

Towards an optimal aiming for molten salt power towers

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Abstract

Finding a suitable aiming strategy for receivers of power towers can be challenging, especially for receivers using molten salt as heat transfer fluid as the allowable flux density decreases dramatically with increasing salt temperature. In this paper a very fast, steady-state model for the molten salt receiver is presented. This model is combined with a ray-tracing software and a metaheuristic optimization procedure. The thermal model is used to calculate the actual temperature and mass flow in the receiver which are then used to calculate the operational limits for the flux density. It is demonstrated that such an optimized aiming strategy can outperform a parameter based aiming strategies by more than 2%.

Keywords: concentrating solar power, molten salt receiver, aim point optimization, ant colony optimization

1. Introduction

One option to utilize solar energy for a renewable electricity production are solar power towers using molten salt as heat transfer fluid in the receiver. Mirrors reflect the sunlight to heat up the salt inside the receiver which is located at the top of the central tower. The hot salt can be easily stored inside

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6 hot storage tanks. This allows a demand-oriented production of electricity.
7 Therefore this technology is a promising option for future energy production.

8 The flux density on the receiver surface is limited due to

- 9 • thermal stresses inside the tube caused by the one-sided radiation
- 10 • an upper temperature limit for the molten salt to avoid rapid corrosion
11 of the tubes and degradation of the salt.

12 The resulting allowable flux density on the absorber tube surface is a key
13 design and operational parameter for the receiver.

14 If all heliostats of the field aimed on the of the receiver or in case of
15 a cylinder on the centerline respectively, almost all radiation would hit the
16 receiver, but the maximum flux density would be beyond the limits. There-
17 fore, the aim points of the heliostats are distributed over the receiver surface,
18 which increases the fraction of radiation, which does not hit the receiver: the
19 so-called spillage losses increase. This is the trade-off for the aiming strat-
20 egy. An optimal aiming should maximize the power on the receiver surface
21 with respect to the allowable flux density. In case of a receiver using molten
22 salt as heat transfer fluid the situation gets even more complicated. The
23 flux limit strongly depends on both temperature and mass flow of the salt.
24 For this reason, the spatially distributed values of flux limits have to be up-
25 dated continuously in accordance with the local salt temperatures and load
26 situation.

27 **2. State-of-the-art**

28 A detailed analysis description of the flux limits and aiming strategy used
29 in the molten salt tower demonstration project Solar Two can be found in
30 Vant-Hull (2002). In this paper, Vant-Hull introduced the concept of the
31 allowable flux density (AFD). In this concept, both limitations, stress and
32 peak salt temperature directly at the wall, are transferred into a flux density
33 limit as a function of local salt temperature and salt velocity. The resulting
34 flux density limits are shown in Fig. 1. With higher salt temperatures the flux
35 limit caused by the film temperature becomes the limiting factor, resulting
36 in the kinks in the curves.

37 Furthermore, Vant-Hull describes the aiming strategy which was used in
38 the Solar Two plant to keep the flux within the limits. The flux coming from
39 each heliostats was approximated by a cone. The opening angle of the cone

Nomenclature

α	Absorptance	f_σ	Extension factor for the assumed beam cone angle
ΔL	Length of the tube element	h	Heat transfer coefficient tube to salt
ϵ	Emissivity	h_{conv}	Convective heat transfer coefficient due to natural and forced convection (wind)
θ	Bulk salt temperature in degree Fahrenheit	k	Thermal conductivity of the tube material
v	Temperature in degree Fahrenheit	LF	Load factor: ratio of local absorbed flux and allowable flux density
\dot{m}	Mass flow	p	Penalty factor
\dot{q}_F''	Flux density	r	Slant range
\dot{q}_{net}''	Net flux density on receiver tube	T_{Fc}	Tube crown temperature
ξ	Second parameter in aiming strategy	T_{F}	Front element temperature
A_p	Tube element projected area	T_{Ic}	Film crown temperature
AFD	Allowable flux density	T_{S}	Fluid temperature
c	Constant for penalty factor	T_{inf}	Ambient temperature
c_p	Specific heat capacity of salt	v	Velocity of the salt in the tubes
d_{I}	Inner diameter	v_{dp}	Velocity of the salt in the tubes for the design point
d_{O}	Outer diameter		
e	Exceedance of limits		
f_0	First parameter in aiming strategy		

40 is chosen in a way that the cone envelopes a predefined amount of power of
 41 the Gaussian distributed flux density of a heliostat. Half of the heliostats
 42 aimed in a way that the assumed outer edge of their beam coincides with
 43 the upper rim of the receiver. The other half aimed on the lower part of the

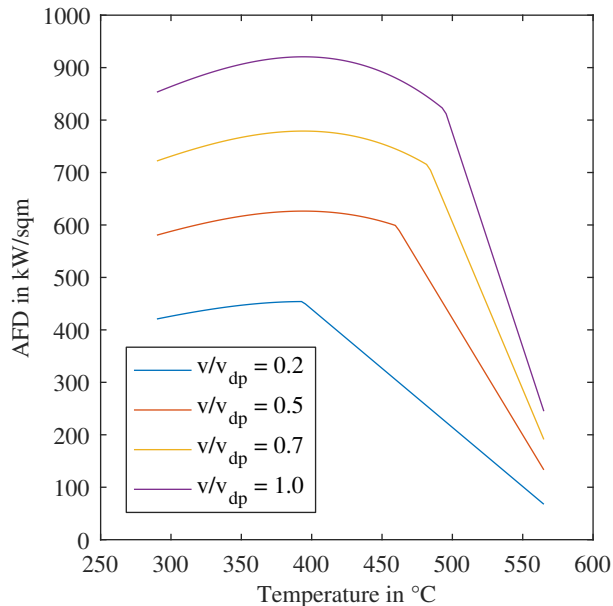


Figure 1: Flux density limits applied in the Solar Two plant as a function of velocity and local salt temperature

44 receiver in an analogous fashion. In order to compute the aiming points for
 45 each heliostat, a respective opening angle of the beam has to be deduced.
 46 This angle can be calculated based on sunshape, mirror and tracking error
 47 as well as astigmatism. The resulting opening angle is then multiplied with
 48 a constant factor; the aiming strategy parameter. During operation the flux
 49 was monitored by numerical simulations and heliostats causing excessive flux
 50 were removed from the receiver. Since removing heliostats reduces the power
 51 on the receiver and therefore also the velocity of the salt flow to maintain
 52 the desired salt outlet temperature level, the AFD is reduced as well. This
 53 might lead to additional areas with overflux and the defocussing of additional
 54 heliostats. Such instability was observed during the operation of Solar Two
 55 (Pacheco et al. (2002)).

56 For the plants having started operation during the past years no detailed
 57 information about the aiming strategy has been published. Nevertheless,
 58 in scientific literature methods and strategies for receivers using molten salt
 59 have been published. Sanchez-Gonzalez et al. (2016) developed a fit algo-
 60 rithm to optimize the aiming strategy for a molten salt receiver. A thermal
 61 model was used to calculate an AFD database including the limits by stresses

62 and temperature. During optimization the mass flow and local temperatures
63 were calculated and the actual flux density was compared to the database val-
64 ues of AFD. The aiming strategy followed the idea of Vant-Hull, but instead
65 of using one factor for the whole field, individual factors for small groups of
66 heliostats were used. These factors were determined during an optimization
67 procedure.

68 Beside the presented methods for molten salt receivers several analyses of
69 aim point optimization for other receiver types were published. Salom et al.
70 (2013) and Besarati et al. (2014) used optimization algorithms to create flux
71 density distribution as homogeneous as possible. Belhomme et al. (2014) used
72 an adapted ant colony algorithm to maximize the output of a concentrated
73 photo-voltaic receiver. García-Martín et al. (1999) developed a heuristic
74 knowledge-based heliostat control strategy for the open volumetric receiver
75 at the Plataforma Solar de Almería.

76 **3. Methods**

77 *3.1. Parameter based aiming*

78 Beside the aim point optimization methodology, which will be described
79 in the next section, a newly developed parameter based aiming strategy is
80 presented, because the Vant-Hull aiming strategy might fail to produce a
81 suitable flux density distribution. The Vant-Hull strategy is based on one
82 single factor: this factor increases the cone angle of each heliostat from its
83 theoretical minimum given by an assumed beam error. With increasing fac-
84 tor the cone angle increases and the cone will envelope a higher amount of
85 the radiation of the heliostat. Therefore a high factor value leads to large
86 angle for the cones. As a result the aim points of the heliostats lie close to
87 the receiver centerline and cause a peak in the flux density at the receiver
88 centerline. Decreasing this factor causes the heliostats to aim more towards
89 the edges of the receiver. In case of heliostats with good beam quality this
90 might cause so-called shoulders at the edges of the receiver, in which the
91 limits are exceeded as well. To overcome this issue, the so-called modified
92 Vant-Hull aiming strategy is developed. In this method the factor for the
93 cone angle f_σ is calculated by

$$f_\sigma = f_0 + \xi \cdot r \quad (1)$$

94 with the two parameters f_0 and ξ and slant range of heliostat and center of
95 the receiver r . When using $\xi = 0$ the aim strategy equals the conventional

96 Vant-Hull aiming. With $\xi > 0$, the first parameter f_0 gives a possibility to
97 limit the amount of spillage due to the inaccuracy of the analytical image
98 size computation: it defines the minimum distance between aiming point and
99 receiver edge as a function of the computed image size. The second parameter
100 ξ introduces the possibility to scale the punishment for the distance between
101 heliostat and receiver. This allows an even distribution of the flux over the
102 receiver surface, also for heliostats with small beam errors. This parameter
103 based aiming strategy will be used as reference for the aim point optimization.

104 *3.2. Aim point optimization*

105 In the current analysis the methodology and tools described in Belhomme
106 et al. (2014) are adapted to be used for a molten salt receiver. In the following
107 of this section the approach for the optical simulation and the optimization
108 approach is described very briefly. Subsequently, the newly developed ther-
109 mal model for the receiver is discussed in detail. The sequence of steps in
110 the optimization procedure is shown in Fig. 2.

111 *3.2.1. Optical model and optimization procedure*

112 Since the approach of the optimization procedure has already been de-
113 scribed in Belhomme et al. (2014) in detail, it is only summarized in this
114 section. The flux density distribution is calculated using the software tool
115 STRAL (Belhomme et al. (2009)), which has been developed at the German
116 Aerospace Center. The tool offers the option to include measured surface
117 data of heliostats and therefore, it is capable to calculate a very accurate
118 prediction of the resulting flux on the receiver surface.

119 By using a discretized grid of aim points the optimization problem is
120 transferred into a combinatorial problem. Therefore, algorithms like the ant
121 colony algorithm can be applied to find an optimal solution.

122 The usage of a ray tracing tool for the calculation of the flux density
123 distribution has the advantage of its high accuracy, but the drawback of
124 the large computational effort. To overcome this issue, the image of each
125 heliostat is calculated and stored for each aim point before starting the opti-
126 mization (precalculation step). During precalculation of the flux distribution
127 of a single heliostat, all other heliostats aim on a predefined aim point. This
128 introduces a slight inaccuracy since the shading and blocking depends on the
129 orientation of the other heliostats. However, the changes in orientation of
130 the heliostats and thus the inaccuracy are very small and the gain in com-
131 putational speed is immense. Therefore this approach is chosen. By using a

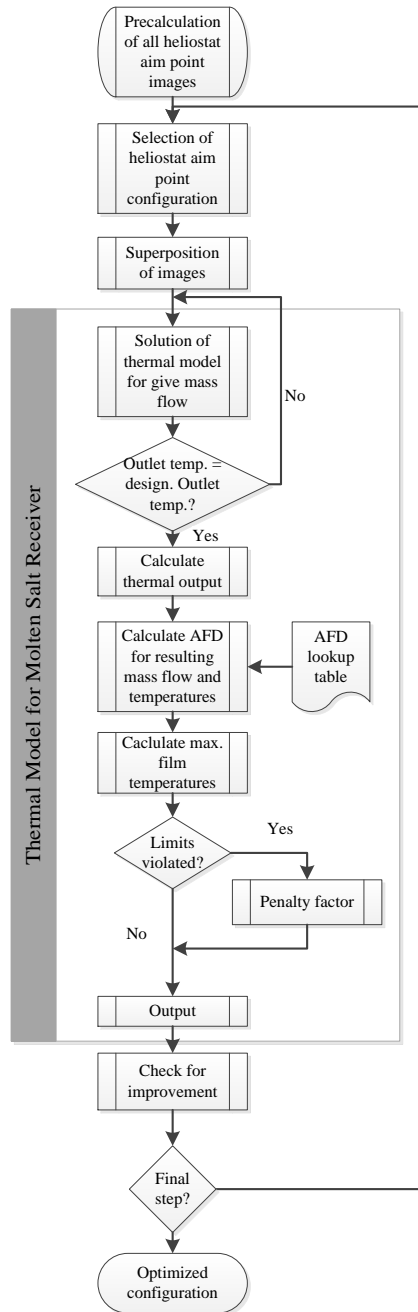


Figure 2: Flow scheme of the optimization procedure

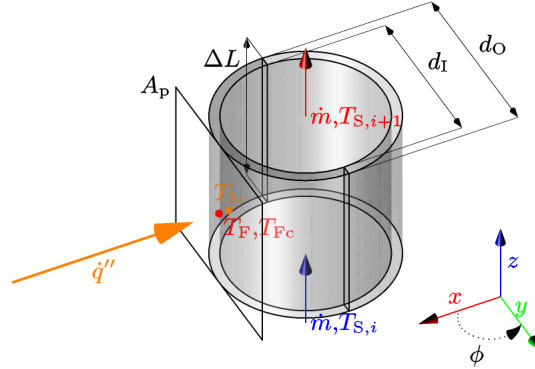


Figure 3: Depiction of one element of a single tube used for the thermal modeling

132 fixed number of aim points with fixed positions the optimization problem is
 133 transferred into a combinatorial problem and algorithms like the ant colony
 134 algorithm can be applied to find an optimal solution with heuristic informa-
 135 tions. Namely the global quality of the current solution, here the receiver
 136 performance for the overall aim point configuration, and the local quality of
 137 a discrete change in the aim point assignment for a single heliostat, here the
 138 intercepted power of one heliostat. Belhomme et al. (2014) transferred the
 139 ant colony optimization metaheuristic to the aim point optimization problem
 140 by defining the aim point configuration as the trail of an ant discovering a for-
 141 age source. Every aim point heliostat assignment is a part of this trail. The
 142 receiver performance as the global quality value corresponds to the length of
 143 the overall trail. The intercepted power of one heliostat as the local quality
 144 value corresponds to the myopic information of a single ant while discovering
 145 the trail. During the optimization process the current aim point configu-
 146 ration is improved according the heuristic informations. Starting point is
 147 a sub-optimal aim point configuration. In each optimization step all corre-
 148 sponding precalculated images are superposed to calculate the resulting flux
 149 density distribution. The resulting flux density distribution is used by the
 150 receiver model, which is described in the following section, to calculate one
 151 single output value.

152 3.2.2. Thermal model

153 In general, the receivers used in solar power towers consist of several
 154 panels. The salt passes the panels sequentially. Each panel is composed of
 155 several parallel tubes. In the following we describe the approach which was

156 chosen to obtain a fast model as it is required for the optimization. The
157 model calculates the relevant local states of the salt flow (mass flow and
158 temperatures) required to calculate the AFD. To obtain a fast model several
159 assumptions are made:

- 160 • The radiative heat exchange between the tubes (reflection and emis-
161 sion) is neglected. Neighboring tubes are exposed to almost the same
162 flux and the temperatures are similar (except the for the neighboring
163 tubes belonging to different panels). Therefore, the thermal radia-
164 tion from one tube to its neighbors equals approximately the radiation
165 coming back from the neighboring tubes. If the radiation exchange is
166 neglected, all tubes of a panel can be treated individually.
- 167 • The heat conduction along the tube axis inside the tube material is
168 neglected. The influence is assumed to be small as the tubes are very
169 thin.
- 170 • The tubes are discretized in two elements in circumferential direction:
171 one element for the front side and one for the backside. We assume
172 that no heat is conducted from the front element to the back element
173 which implies that the total absorbed heat flux is transported radially
174 through the front element.
- 175 • For the flux coming from the field and the radiation exchange with the
176 environment the front element is treated like a flat plate with a surface
177 area corresponding to the projected area of the tube.
- 178 • The total mass flow in one panel is equally split between all absorber
179 tubes.
- 180 • The outlet temperature of a panel is equal to the mixing temperature of
181 the outlet temperatures of its tubes. The assumption implies adiabatic
182 headers. This outlet temperature is used as the inlet temperature for
183 the downstream panel.

184 The axial discretization can be chosen by the user. The different physical
185 values of one tube element are depicted in Fig. 3. Additionally, the user can
186 specify whether each tube of a panel should be simulated or just a user-
187 specified number of representative tubes.

188 Applying the previously described assumptions we can express the heat
 189 which is transferred through the tube \dot{Q}_{cond} with

$$\dot{Q}_{\text{cond}} = \alpha A_p \dot{q}_F'' - \sigma \epsilon A_p \cdot (T_F^4 - T_{\text{inf}}^4) - h_{\text{conv}} A_p (T_F - T_{\text{inf}}) \quad (2)$$

190 using the absorptance α , the emissivity ϵ , the projected area A_p , the flux
 191 density from the field \dot{q}_F'' , the convective coefficient h_{conv} and the front and
 192 ambient temperature T_F and T_{inf} respectively. The heat flux \dot{Q}_{cond} has to be
 193 transported through the tube material into the fluid. The heat transfer to the
 194 fluid is modeled with a one dimensional approach and with a correlation based
 195 calculation of the convective heat transfer coefficient. For the heat transport
 196 through tube and to the fluid the following equation is used

$$\dot{Q}_{\text{cond}} = \frac{1}{\frac{1}{h \frac{\pi}{2} d_I \Delta L} + \frac{1}{\pi k \Delta L} \cdot \ln\left(\frac{d_O}{d_I}\right)} \cdot (T_F - T_{S,i+1}) \quad (3)$$

197 In this equation we have used the symbols h , for the convective heat transfer
 198 inside the tube, k for the thermal conductivity of the tube material, d_I and
 199 d_O for the inner and outer diameter of the tube, ΔL for the length of the
 200 tube element and $T_{S,i+1}$ for the temperature of the fluid. Finally, the heat
 201 transferred to the fluid \dot{Q}_{cond} leads to a temperature rise

$$\dot{Q}_{\text{cond}} = \dot{m} c_p (T_{S,i+1} - T_{S,i}) \quad (4)$$

202 from the inlet temperature $T_{S,i}$ to the outlet temperature $T_{S,i+1}$ with the
 203 mass flow of salt in the tube \dot{m} and the specific heat capacity c_p .

204 By combining the three equations 2, 3 and 4 with the variables \dot{Q}_{cond} ,
 205 T_F and $T_{S,i+1}$ for a given mass flow \dot{m} and inlet temperature $T_{S,i}$ we ob-
 206 tain a nonlinear equation e.g. for the front temperature. If the influence
 207 of the temperature on the properties like k and on those in the calculation
 208 of h is neglected, this equation can be solved analytically. As a first guess
 209 all properties depending on temperatures are evaluated for the known inlet
 210 temperature. The solution which is obtained with this assumption is refined
 211 in a second iteration, in which all properties are evaluated for the tempera-
 212 tures of the initial solution. In a real plant the mass flow is adjusted by the
 213 control system to obtain a predefined outlet temperature. For this reason,
 214 the mass flow is not known prior to the solution. In the model an iterative
 215 approach is used for the calculation of the mass flow. An initial estimate of
 216 the mass flow is calculated by using the gross flux and a guess of the receiver

217 efficiency. For this mass flow all temperatures are calculated sequentially,
 218 leading to an outlet temperature. In the following iterations the mass flow
 219 is adjusted until the outlet temperature is close enough to the given outlet
 220 temperature.

221 Solving the thermal model gives the mass flow of salt and the local salt
 222 temperatures for the given flux density distribution. To check whether the
 223 flux can be tolerated a similar approach to the one described by Vant-Hull
 224 (2002) is used. In the model a look-up table is included in which any flux
 225 limits

$$AFD_i = f(T_{S,i}, \dot{m}) \quad (5)$$

226 can be used. In case of Solar Two both effects limiting the flux, the film
 227 temperature and the stresses in the tube, were transformed into a formula-
 228 tion for the AFD in the form of equation 5. In the presented model we have
 229 implemented an integrated approach calculating the film temperature. The
 230 approach described above is fast, but it calculates only one single temper-
 231 ature for the front side element, which represents a mean temperature. In
 232 reality, the temperature of the inner and outer tube surface changes from the
 233 tube crown to the sides. For energetic calculations the usage of a mean tem-
 234 perature can be appropriate. However, for the limiting film temperature the
 235 highest temperature is relevant which cannot be calculated by the described
 236 approach. To estimate the highest temperature at the crown of the tube we
 237 can use the assumption that the flux density on the outer tube surface is
 238 almost symmetric in circumferential direction and therefore $\partial T / \partial \phi \approx 0$. As
 239 a result the flux hitting the crown has to be transported through the tube
 240 radially and the crown temperature T_{Fc} , which is the highest temperature,
 241 can be calculated by solving the equation

$$\alpha \dot{q}_F'' - \epsilon \sigma (T_{Fc}^4 - T_\infty^4) - h_{conv} (T_{Fc} - T_\infty) = \frac{1}{\frac{d_o - d_i}{2k} + \frac{1}{h}} (T_{Fc} - T_{S,i}) . \quad (6)$$

242 As the heat transfer coefficient h and the temperature of the salt T_S is known,
 243 the equation can be solved directly. Then the crown film temperature T_{Fc}
 244 can be calculated by solving

$$\alpha \dot{q}_F'' - \epsilon \sigma (T_{Fc}^4 - T_\infty^4) - h_{conv} (T_{Fc} - T_\infty) = h (T_{Fc} - T_{S,i}) . \quad (7)$$

245 The thermal model delivers different values representing the output of the
 246 receiver which can be maximized, e.g. thermal output or intercepted power.
 247 In case of a violation of the AFD and/or the maximum salt temperature the
 248 exceedance

$$e = \mathbf{max}(\mathbf{max}(\xi_i - \xi_{\text{lim}}), 0). \quad (8)$$

249 In the equation ξ can be the flux or the maximum film temperature. The
 250 exceedance is used to calculate a penalty factor

$$p = \mathbf{exp}(-c \cdot e) \quad (9)$$

251 with a user defined constant $c > 1$. This penalty factor $0 < p < 1$ is then
 252 applied to make the output values in case of a violation unattractive.

253 4. Validation of the thermal model

254 In order to prove its validity the results of the model are compared to those
 255 of a much more detailed model. In order to validate the accelerated receiver
 256 model, its results are compared to a high detail model. In this study, the
 257 simplified model is compared to the ASTRID© model (Frantz et al. (2016)).
 258 ASTRID© is a thermal FEM model, which considers both the absorber tubes
 259 and the insulation. The local solar irradiation coming from the heliostat
 260 field is simulated by the raytracing software SPRAY and is applied as a
 261 boundary condition. The heat transfer to the fluid is modeled using one-
 262 dimensional fluid flow elements allowing mass and heat transportation. The
 263 local heat transfer coefficients are computed based on Nusselt correlations
 264 as a function of the local fluid temperature and Reynolds number. The
 265 thermal radiative exchange between the absorber tubes and the insulation as
 266 well as the radiation to the ambient is modeled using the radiosity method.
 267 The natural and forced convection losses are modeled by local heat transfer
 268 coefficients issued from CFD simulations. A mass flow control algorithm
 269 adapts the mass flow iteratively for each flow path through the receiver in
 270 order to get the desired outlet temperature for different load scenarios.

271 Both models were used to simulate the thermal behavior of the Solar Two
 272 power plant. From the detailed flux density distribution over the circumfer-
 273 ence of the tube issued from the ASTRID© model a flux density distribution
 274 with just one value per tube in horizontal direction was calculated. The re-
 275 sulting integral values of both models are given in table 1. Additionally,
 276 experimental results of the Solar Two power plant are listed. Both models

Table 1: Comparison of the integral results of both models in the design point (incident power 48 MW). Comparison values of the Solar Two plant are given as well (Pacheco et al. (2002)).

Value	STRAL	ASTRID©	Deviation	Solar Two
Thermal Power	42.45 MW	42.15 MW	0.73 %	42.2 MW
Mass Flow	105.4 kg/s	104.79 kg/s	0.56 %	-
Thermal Efficiency	87.5 %	86.9 %	0.73 %	87 %

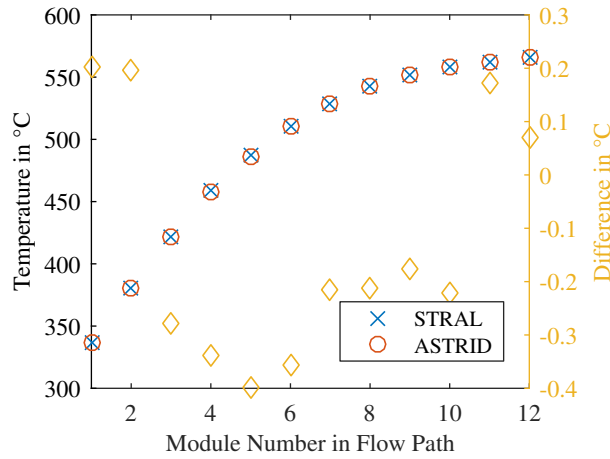


Figure 4: Maximum fluid temperature for each module in the two flow paths

277 give almost the same results and the values are similar to those reported for
 278 the real Solar Two Plant.

279 As described in section 3.2.2 the AFD is calculated based on mass flow
 280 and local temperature. Therefore, in addition to the mass flow the tem-
 281 perature distribution in the receiver is of particular interest. The highest
 282 fluid temperature for each panel is shown in Fig. 4. The fluid temperature
 283 rises from approximately 340 °C behind the first panels to the design outlet
 284 temperature 565 °C behind the last panel. A slight difference between the
 285 two models can be observed, but overall the temperature distributions are
 286 in good agreement. Consequently, the model can be used to calculate the
 287 local allowable flux density based on mass flow and fluid temperature.

288 Beside the limit based on the AFD the model is capable to calculate the
 289 crown film temperature which poses an additional limit for a safe operation
 290 of the plant. To validate the calculation approach for the crown film tem-

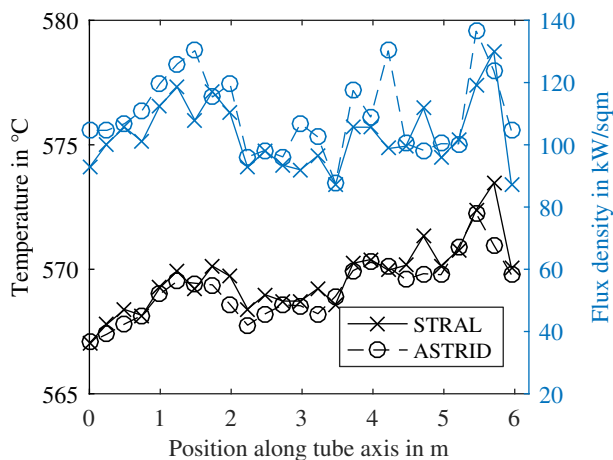


Figure 5: Comparison of the peak film temperature in the tube with the highest film temperatures.

291 perature the values for T_{Ic} calculated by the STRAL model are compared
 292 to the highest film temperature calculated by the ASTRID© model. Fig-
 293 ure 5 shows the comparison of the peak film temperature of a single tube as
 294 function of the position. Additionally, the flux density used in both modes is
 295 presented as well. In case of the ASTRID© model the highest flux density
 296 of all circumferential elements at the given axial position is used, as the film
 297 temperature is expected to correlate with the peak flux. The temperature
 298 distribution calculated by the STRAL model matches the temperatures sim-
 299 ulated by the ASTRID© model quite well. The highest differences occur in
 300 positions where the flux differ as well, which can be explained by the neces-
 301 sary averaging procedure for the flux. But the highest difference is smaller
 302 than 2 °C and therefore acceptable.

303 Altogether, the validation has proven that the model is accurate enough
 304 to be used for calculation of the AFD and the film crown temperature.

305 5. Case Definition

306 For the demonstration of the capabilities of the aim point optimization
 307 a new field layout and receiver is designed. It would have been preferable
 308 to demonstrate the capabilities for an existing plant, but in most cases the
 309 required detailed information on the design are not publicly available. The
 310 Solar Two plant, of which design is documented very well, is too small com-
 311 pared to the plants built nowadays.

Table 2: Summary of properties of the used heliostat design

Total reflective area	121 m ²
Width	12.93 m
Height	9.57 m
Pedestal height	5.2 m
Surface error	1.3 mrad
Tracking error	0.65 mrad
Total reflectivity	90.24 %

Table 3: Summary of the receiver parameter used for demonstration

Optical tower height	190 m
Circumscribing diameter	15.82 m
Radiated height	20 m
Number of flow paths	2
Number of panels	12
Number of tubes per panel	117
Tube inner diameter	30 mm
Tube outer diameter	35 mm
Inlet temperature	290 °C
Outlet temperature	565 °C

312 The location of the Redstone thermal power plant (28° S and 23° E) is
 313 chosen as position for the virtual plant. Heliostats similar to the Sanlu-
 314 car 120 (Osuna et al. (2006)) are used. Their most relevant parameter are
 315 summarized in Tab. 2. A field layout is created using HFLCAL (Schwarzbözl
 316 et al. (2009)). The resulting layout with a total number of 6482 heliostats is
 317 shown in Fig. 6. As indicated by the different colors, 20 groups with different
 318 canting distances are used.

319 The receiver is designed with an approximated thermal output of 450 MW.
 320 Its parameters are summarized in Tab. 3. The salt enters the receiver in the
 321 southern part and flows through the panels as depicted in Fig. 7. The salt
 322 flow is crossed after the third panel. In the simulation not all tubes of
 323 each panel are resolved in order to reduce the computational effort. The
 324 simulations are performed using 5 representative tubes of the 117 tubes per
 325 panel and 25 elements per tube in lengthwise direction. This results in a

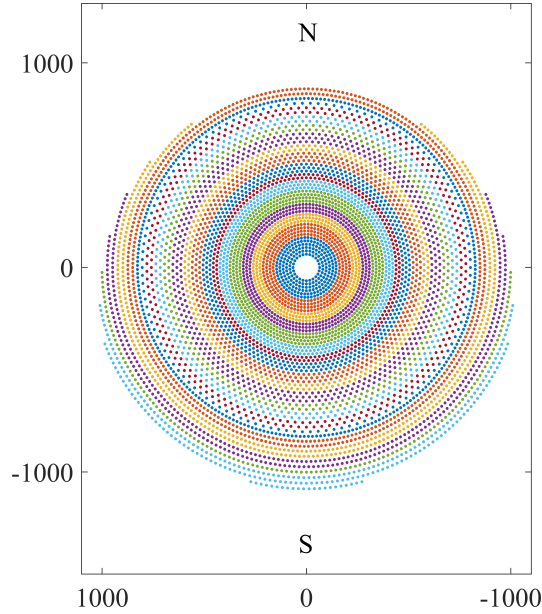


Figure 6: Field layout

326 total amount of 1500 tube elements for the whole receiver.

327 The lookup table for the allowable flux density is calculated using the
 328 formula taken from Vant-Hull (2002) from the bulk salt temperature in degree
 329 Fahrenheit θ

$$AFD = (842.27 - 1.5514 \cdot \theta + 4.613 \cdot 10^{-3} \cdot \theta^2 - 3.2073 \cdot 10^{-6} \cdot \theta^3) \cdot \left(0.3 + 0.7 \cdot \frac{v}{v_{dp}}\right). \quad (10)$$

330 This AFD is applied to limit the net flux density

$$\dot{q}_{net}'' = \frac{\dot{Q}_{cond}}{A_p}. \quad (11)$$

331 For a simpler evaluation we define the load factor

$$LF = \frac{\dot{q}_{net}''}{AFD} \quad (12)$$

332 as the ratio of the local radiation divided by the local AFD . The second
 333 limit resulting from the temperature limit for the fluid is applied directly by
 334 comparing the crown film temperature to the limit of 600 °C.

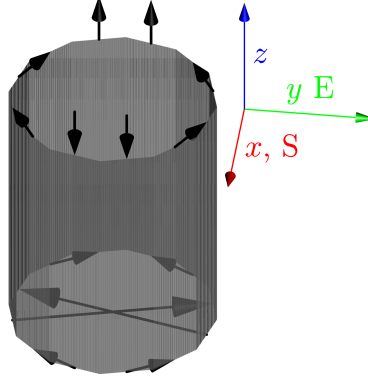


Figure 7: Visualization of the receiver and its flow pattern

335 All simulation are carried out for solar noon on the 21st of march, which
 336 will be referred as the design point.

337 6. Results and Discussion

338 6.1. Parameter based aiming

339 Firstly, we want to find a suitable choice of parameter which combines
 340 a flux which is within the limits and leads to a high intercept factor IC
 341 defined as the ratio of the actual radiation which hits the receiver and the
 342 hypothetical amount of radiation hitting a receiver with infinite dimensions.
 343 A parameter variation is performed for the design point neglecting the track-
 344 ing error. The results are shown in Fig. 8. In the plot the results for the
 345 choice of $\xi = 0$ are shown as well, which corresponds to the original Vant-
 346 Hull aiming. It is clearly observable that independent of the choice of f_0
 347 the original Vant-Hull strategy does not create a flux distribution which is
 348 within the predefined limits. When extending the strategy with the second
 349 parameter one can see that high values for f_0 and ξ are favorable for a high
 350 intercept factor IC . The highest film temperature decreases with increasing
 351 ξ for the lower values of f_0 . In case of the lower values for f_0 the load factor
 352 first decreases with increasing ξ and after becoming minimal around $\xi \approx 2.25$
 353 it increases again. For the highest f_0 shown in the plot the maximum load
 354 factor increases monotonically with increasing ξ . The parameter based aim-
 355 ing strategy barely creates suitable aim point distributions. The parameter

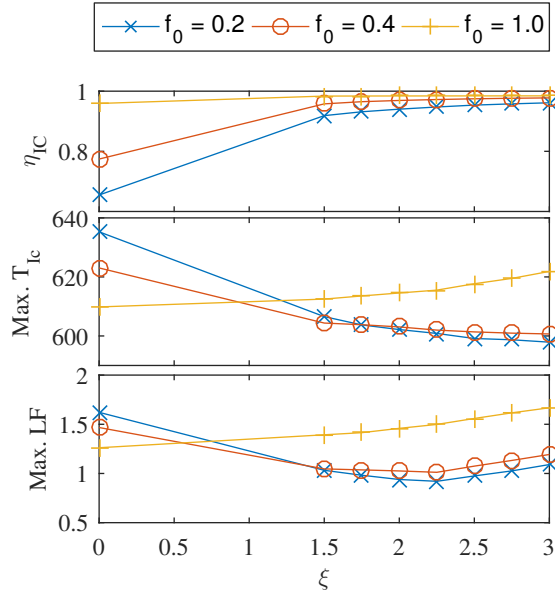


Figure 8: Results of the variation for the parameter based aiming strategy

356 combination $f_0 = 0.2$ and $\xi = 2.5$ creates the best intercept with respect to
 357 all limits. This will be used as reference for the aim point optimization.

358 6.2. Aim point optimization

359 For a first demonstration the aim point optimization is performed for the
 360 design point. An aim point grid with 60 points in circumferential direction
 361 and 30 points in vertical direction is used. Two different strategies are com-
 362 pared: in the first strategy each heliostat is allowed to aim on the vertical
 363 axis of the cylinder. By this restriction the number of possible combination
 364 is limited which reduces the time required for the precalculation step. In
 365 the second strategy each heliostat has the degree of freedom to choose other
 366 aim points in circumferential direction. In both cases the optimization is
 367 initialized with the solution of the modified Vant-Hull aiming. It would have
 368 been possible to start from a random heliostat aim point assignment, but
 369 starting from a good valid solution greatly improves the performance and
 370 makes it easier to compare the results. For the optimization the thermal
 371 output of the receiver is chosen as objective value. The resulting progress of
 372 the optimization is shown in Fig. 9. In the initial phase of the optimization
 373 the output of the receiver increases sharply. In this phase the cylindrical strat-
 374 egy outperforms the on-axis strategy. After the initial phase the progress

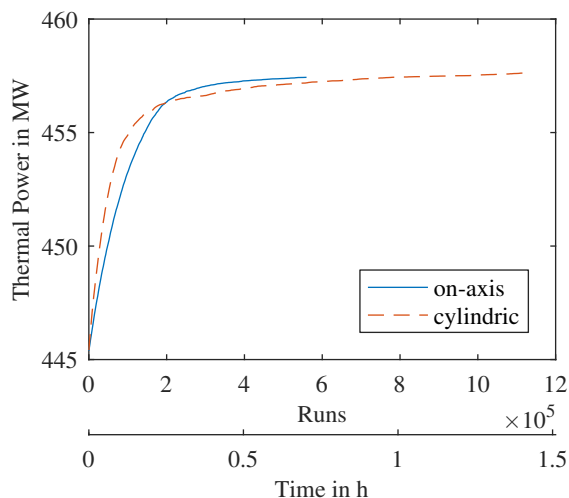


Figure 9: Improvement of the thermal power of the receiver during the optimization using the two different strategies.

375 of the optimization slows down. As this phase begins earlier for the cylin-
 376 dric strategy, the on-axis strategy produces an aim point assignment with a
 377 higher performance after approximately 200 000 runs. In case of the on-axis
 378 strategy the optimization is stopped after 560 000 runs as no significant im-
 379 provement is obtained anymore. The optimization for the cylindric strategy
 380 is carried on for another 560 000 runs. After these runs the optimization
 381 with the cylindric strategy results in an aim point assignment with a slightly
 382 higher performance. In Fig. 9 the duration of the optimization is given on the
 383 second x-axis: the optimization was performed on a workstation computer
 384 with two Intel[®] Xeon[®] E5-2687W processors running at 3.1 GHz using 30
 385 parallel threads. On this machine one run takes less than 5 ms. The initial
 386 phase with the sharp increase of the output takes less than 15 minutes.

387 The intercept factor and thermal output for the modified Vant-Hull aim-
 388 ing and the optimized aim point assignments with the two different strategies
 389 are summed up in Tab. 4 and compared to the results for an aiming strategy,
 390 in which all heliostats aim on the centerline of the receiver. This strategy
 391 maximizes the intercept factor, but the maximum temperature and fluxes
 392 are way beyond the limits. Nevertheless, this strategy gives an upper bound
 393 for the achievable intercept factor. Therefore, in table 4 the so-called aiming
 394 efficiency is given which is defined as the ratio of intercept factors of the
 395 used strategy and the central aiming strategy. All values are given for a cal-

Table 4: Comparison of the key figures for the parameter based aiming and the optimized aim point assignments. As additional reference the results for a central strategy are given in which all heliostats aim on the centerline of the receiver. This aiming strategy does not produce a solution within the limits.

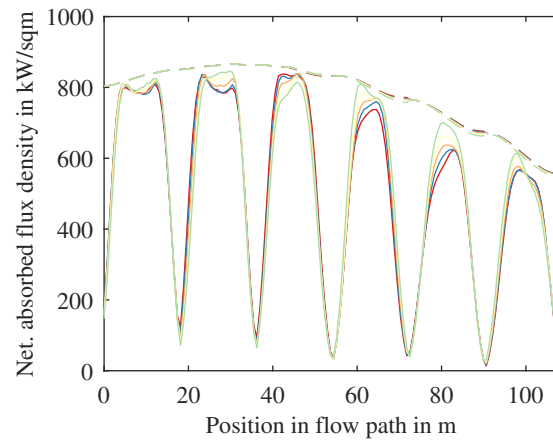
	Central aiming	Parameter based aiming	Optimized on axis	Optimized cylindrical	
w/o err.	Intercept factor	98.5 %	95.4 %	97.8 %	97.8 %
	Thermal power	459 MW	445 MW	457 MW	458 MW
	Aiming efficiency	100 %	96.8 %	99.3 %	99.3 %
w/ err.	Intercept factor	98.2 %	94.7 %	97.3 %	97.3 %
	Thermal power	458 MW	442 MW	455 MW	455 MW
	Aiming efficiency	100 %	96.4 %	99.1 %	99.1 %

396 culation neglecting the tracking error as it is done during the optimization
397 and the mean values of five runs taking the tracking error into account. The
398 aiming efficiency impressively underlines the capabilities of the aim point
399 optimization: less than one percent of the power coming from the field are
400 lost due to the distribution of the aim points in order to obtain a flux density
401 distribution which is within the limits. It is very likely that the value
402 obtained from the optimization is close to the global optimum.

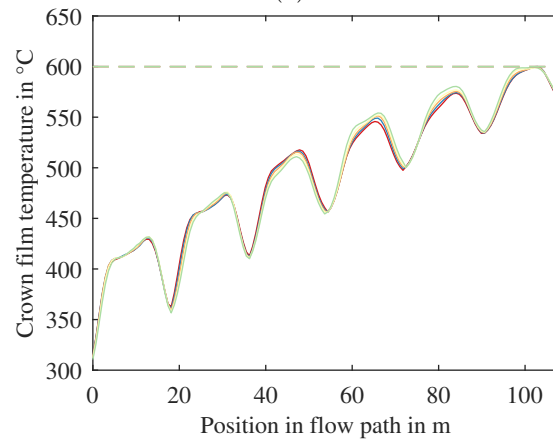
403 When comparing the intercept factors of the different strategies with and
404 without tracking error, we can see that the optimized aim point assignments
405 are more robust to a performance loss caused by the inaccurate tracking,
406 because the intercept factor is only reduced by 0.5 % in the optimized cases
407 instead of 0.7 % in case of the parameter based aiming.

408 Figure 10 gives a deeper insight into the resulting flux density distribution
409 and the crown film temperature for the optimized case with all heliostats
410 aiming on the axis.

411 In Fig. 10a the net absorbed flux density and its limit defined by Eq. 10
412 is shown. The different graphs correspond to the different calculated tubes
413 in the flow path. The AFD is increasing slightly in the first three panels
414 due to the increasing temperature. After that it drops significantly to the
415 outlet. For each panel the actual flux is at least at one position very close
416 to the AFD . In the panels close to the outlet the flux reduces the closer the
417 tubes are to the outlet, because the AFD for the next panel is significantly
418 lower and the flux has to change to the lower values. In the last panel the



(a)



(b)

Figure 10: Net absorbed flux density and film temperature along the flow in one flow path. The different tubes shown are represented by different colors. The dashed line shows the individual limit for the corresponding tube.

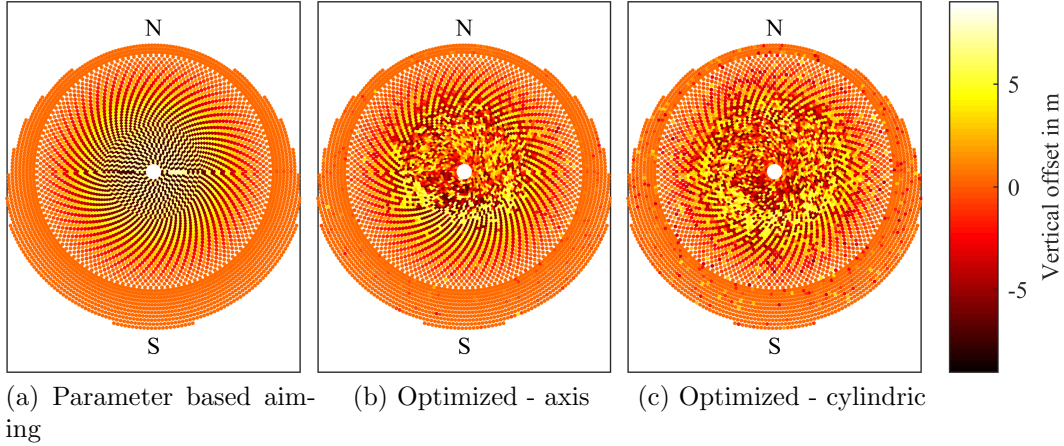


Figure 11: Comparison of the vertical offset for the three analyzed aim point strategies

419 film temperature limit becomes the limiting factor (see Fig. 10b).
 420 Figure 11 gives an overview about the vertical displacement of the aim
 421 point of each heliostat compared to the centerline of the receiver. In case
 422 of the parameter based aiming (Fig. 11a) the vertical offset of the heliostats
 423 follows a regular pattern: neighboring heliostats aim to the lower part and
 424 the upper part of the receiver respectively and with increasing distance the
 425 vertical displacement becomes zero. For both optimized aim point assign-
 426 ments the vertical displacement becomes zero with increasing distance with
 427 some exceptions. The regular pattern of the neighboring heliostats has van-
 428 ished in the central part of the field in both optimized cases. The mean
 429 vertical displacement with a value of 2.3 m is lower for both optimized cases
 430 compared to the parameter based aiming with a mean vertical displacement
 431 of 3.2 m. The lower displacement explains the higher intercept factor and the
 432 higher robustness of the aiming to the tracking error: the closer the heliostat
 433 aims to the centerline the higher the inaccurate tracking has to become to
 434 lead to an increasing spillage. The horizontal shift of the aim points along
 435 the cylindric surface is shown in Fig. 12. This shift follows a rather regular
 436 pattern: the heliostats close to the tower are shifted towards the outlet side
 437 of the receiver. But the performance increase due to that shift is almost
 438 negligible in the presented demonstration case. In case of receiver with a
 439 lower ratio of height to diameter this shift in circumferential direction might
 440 become more important.

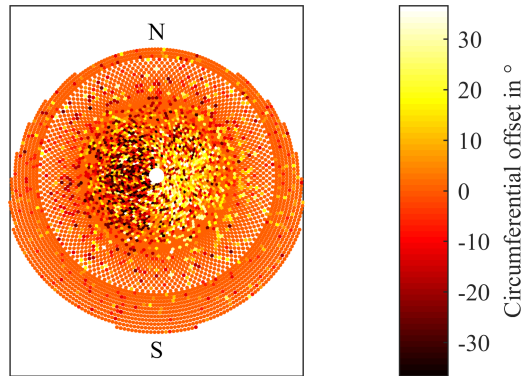


Figure 12: Displacement of the aim points along the circumference of the cylinder.

441 7. Conclusion and Outlook

442 A fast thermal model for external receivers with molten salt is developed
 443 and presented. The model calculates mass flow and local temperatures of the
 444 salt and tube walls. It is validated with an ANSYS based model (ASTRID©)
 445 and the accuracy of the results is proven sufficient. This thermal model is
 446 integrated in a ray tracing based aim point optimization procedure using an
 447 ant colony optimization approach. In the optimization the thermal output
 448 of the receiver is maximized while considering both flux density limits cal-
 449 culated from mass flow and local salt temperature and an additional limit
 450 for the crown film temperature. The capabilities of the aim point optimiza-
 451 tion are demonstrated for a hypothetical power tower with a thermal power
 452 of 450 MW. Two different optimization strategies are analyzed: firstly, a
 453 strategy in which the heliostats had only the degree of freedom to aim on
 454 the vertical axis of the cylinder and one in which they can aim freely on the
 455 cylinder surface. The results are compared to the newly developed modified
 456 Vant-Hull strategy and the hypothetical limit where all heliostats aim on
 457 the receiver centerline. On a state-of-the-art workstation an improvement of
 458 more than 2% points was obtained within an optimization time of 15 min.
 459 In contrast to other proposed strategies the optimization procedure is not
 460 restricted to strategies in which the heliostats aim on the vertical axis of
 461 the cylinder: on the contrary, this method is capable to generate heliostat
 462 aim point assignment where the helisotats aim on any point on the receiver
 463 surface. In the specific analyzed case this unrestricted strategy did not lead
 464 to an improved performance.

465 In the future it would be interesting to use the system to generate aim

466 point assignments for an existing plant to analyze the improvements which
467 can be realized by an optimized aim point distribution under more realistic
468 constraints. The optimization strategy could be integrated as well in a control
469 system of the power plant as it has already been demonstrated in a power
470 tower with an open volumetric receiver (Schwarzbözl et al. (2016)). Such a
471 system could generate new aim points distribution within a given intervall of
472 several minutes to adapt the aim points to the given conditions. As a vision
473 the system could even react to clouds interrupting the normal operation if
474 combined with a nowcasting system.

475 **8. Acknowledgment**

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