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# Temporal Experiment for Storms and Tropical Systems Technology Demonstration (TEMPEST-D) Mission: Enabling Time- Resolved Cloud and Precipitation Observations from 6U-Class Satellite Constellations

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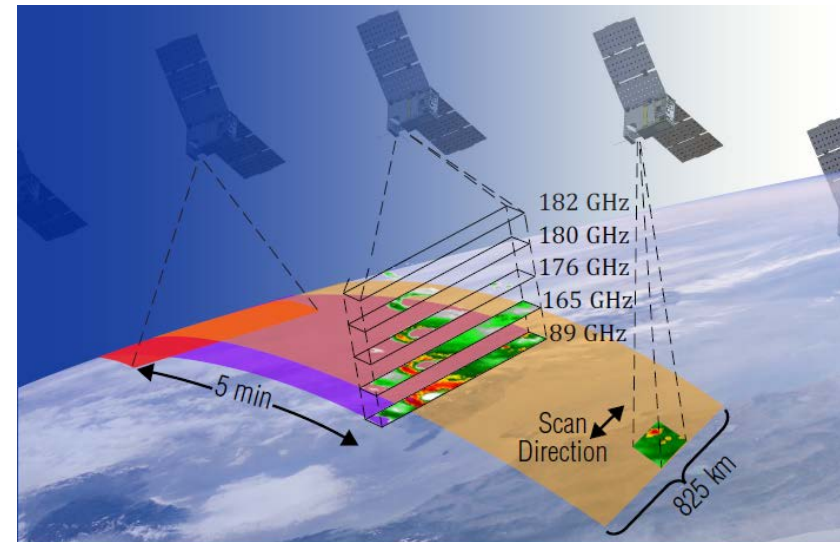
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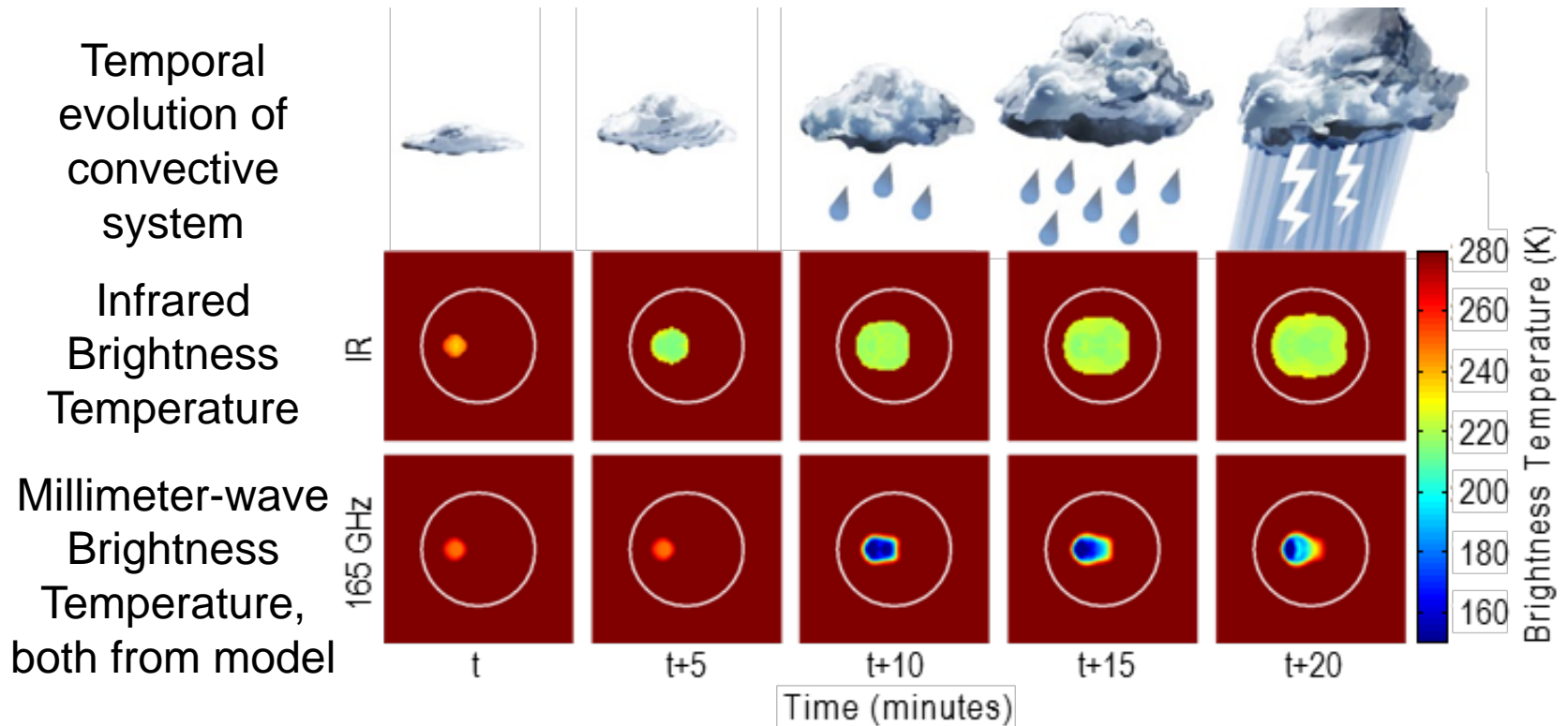
# Temporal Experiment for Storms and Tropical Systems (TEMPEST)

- TEMPEST was proposed to NASA Earth Venture Instrument-2 in Nov. 2013.
  - Low-risk, high-margin approach to use 6U-Class satellites (6U CubeSats) for repeat-pass millimeter-wave radiometry
  - First global temporally-resolved observations of cloud and precipitation processes to improve weather and climate models
  - Selected by Earth Venture for in-space technology demonstration managed by NASA Earth Science Technology Office (ESTO).
- TEMPEST-D started in Aug. 2015 as a partnership among CSU, JPL and BCT, with 2.5-year development cycle.
  - Deliver one complete flight system with integrated payload to NanoRacks for launch integration by Feb. 1, 2018.
- Manifested by NASA CSLI for launch on OA-9 to ISS, planned for Mar. 14, 2018
  - Commercial resupply service to ISS on Cygnus Antares II for deployment via NanoRacks within several months



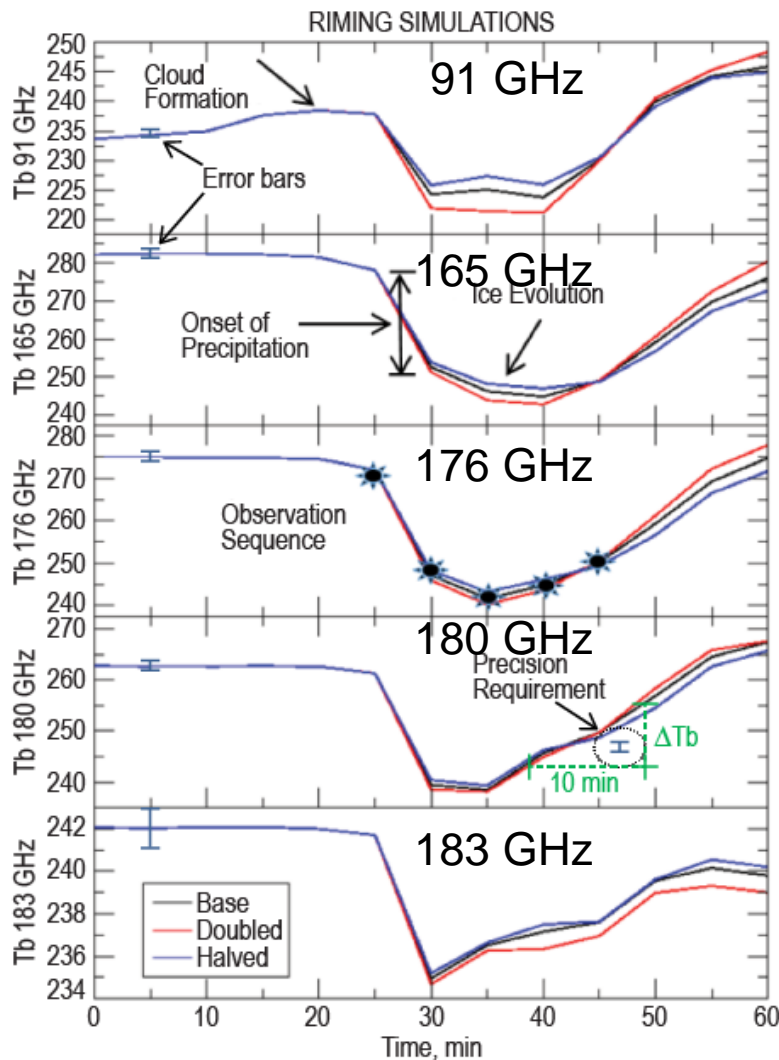
5 identical 6U small sats, each with an identical 5-channel radiometer, flying 5 minutes apart

# Observations of Transition from Clouds to Precipitation



- Infrared brightness temperatures (middle row, available from GEO) show cloud top temperatures, locations and morphology.
- Onset of precipitation clearly detected at millimeter-wave frequencies on TEMPEST constellation, including 165 GHz (bottom row).
- TEMPEST minimum spatial resolution of 25 km is shown (circles).

# Temporal Development of Ice in Cloud-Scale Models

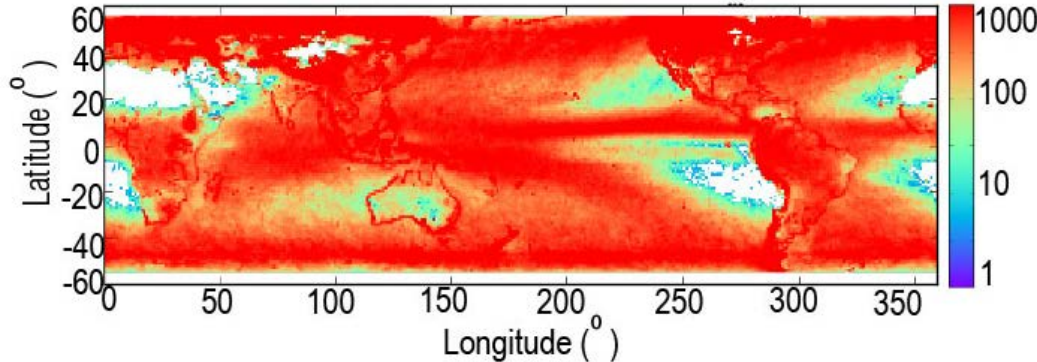


- Modeled brightness temperatures at the five TEMPEST frequencies with 25-km spatial resolution
- Simulations compare different rates of supercooled water droplets collecting on ice crystals (riming efficiency).
- Rate varies from baseline (black) to twice (red) and half (blue).
- Measurable difference between curves is 4 K or greater in 5 minutes at onset of ice formation. Instrument precision requirement is 1 K in 5 minutes.
- Ice remaining in clouds after precipitation has substantial effects on climate. Residual ice can be compared to W-band radar observations from NASA's CloudSat or ESA's EarthCARE.

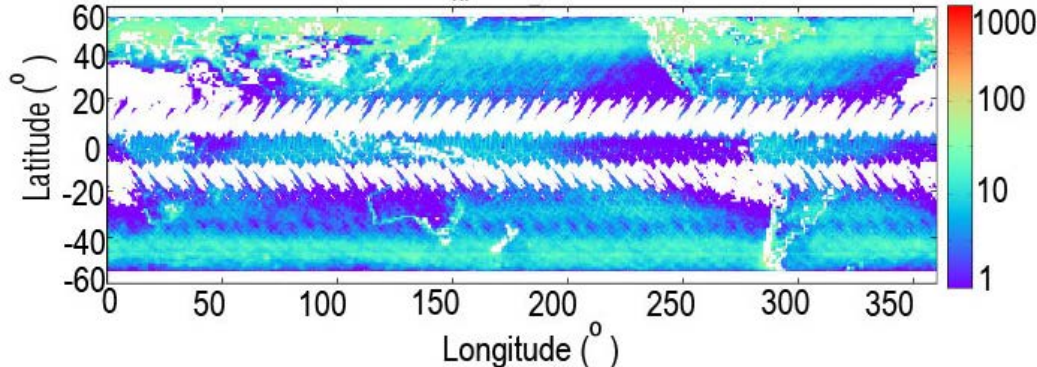


# Global Time-Resolved Observations of Clouds and Precipitation

TEMPEST Number of Rain Events  $> 1 \text{ mm/hr}$  in each  $1^\circ \times 1^\circ$

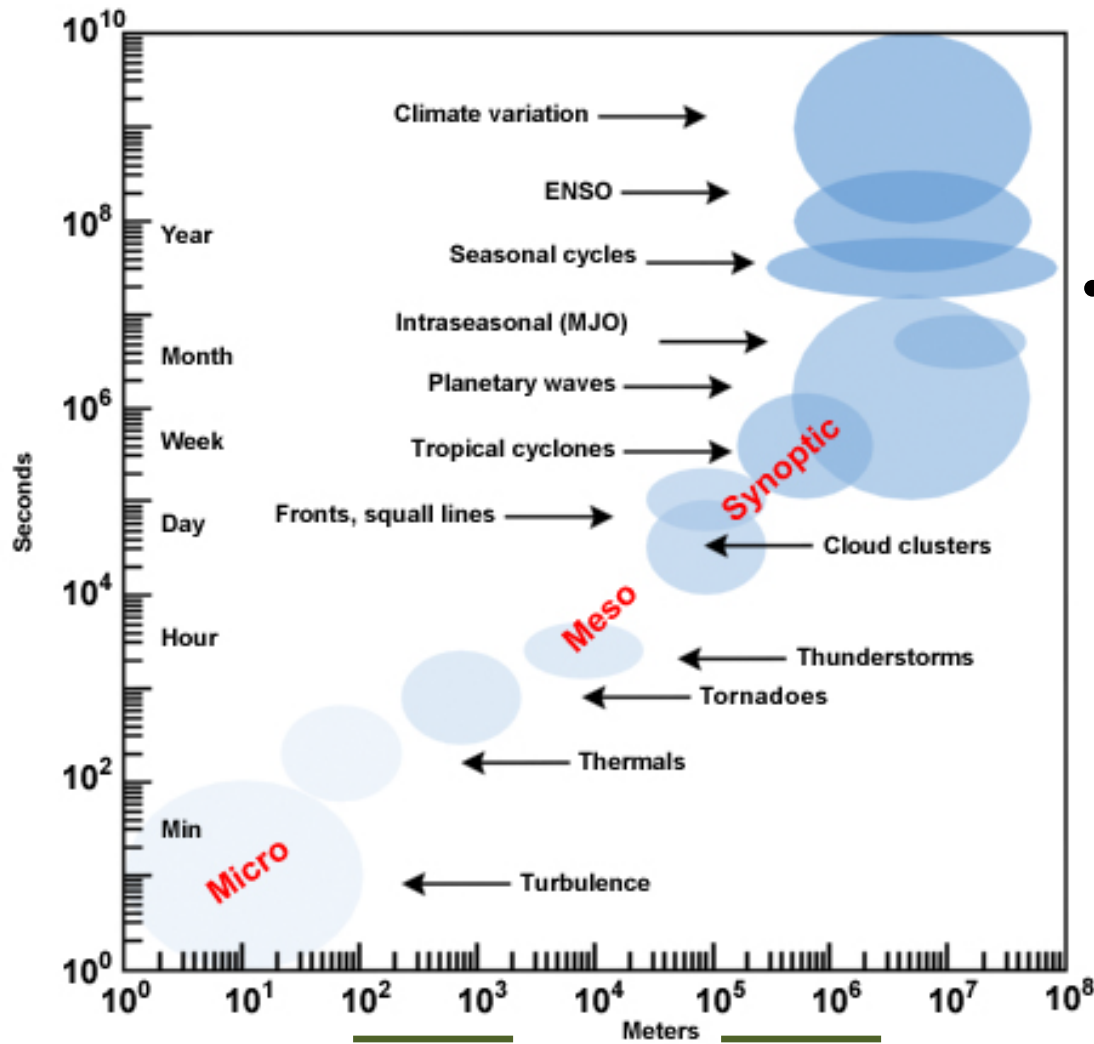


TEMPEST Rain Events  $> 1 \text{ mm/hr}$  in each  $1^\circ \times 1^\circ$ , seen by GPM within 30 min



- During a future one-year mission, TEMPEST constellation could make more than 3,000,000 time-resolved observations of precipitation ( $> 1 \text{ mm/hr}$ ), including 100,000+ deep-convection events
- Could perform more than 50,000 precipitation observations coincident (within 30 minutes) with NASA's Global Precipitation Mission (GPM)
- Assumes nominal TEMPEST orbit for deployment from ISS at 400-km altitude and  $51.6^\circ$  inclination.
- Precipitation estimates from AMSR-E satellite radiometer data with oceanic observations only.

# Spatio-Temporal Scales Observed by Proposed TEMPEST Mission



- TEMPEST enables temporal sampling on the time scale of tens of minutes, corresponding to meteorological structures with spatial scales on the order of hundreds of m to a few km.

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TEMPEST-D

TEMPEST-D + Polar-Orbiting Sats in LEO

# TEMPEST-D Demonstration: Motivation and Objectives

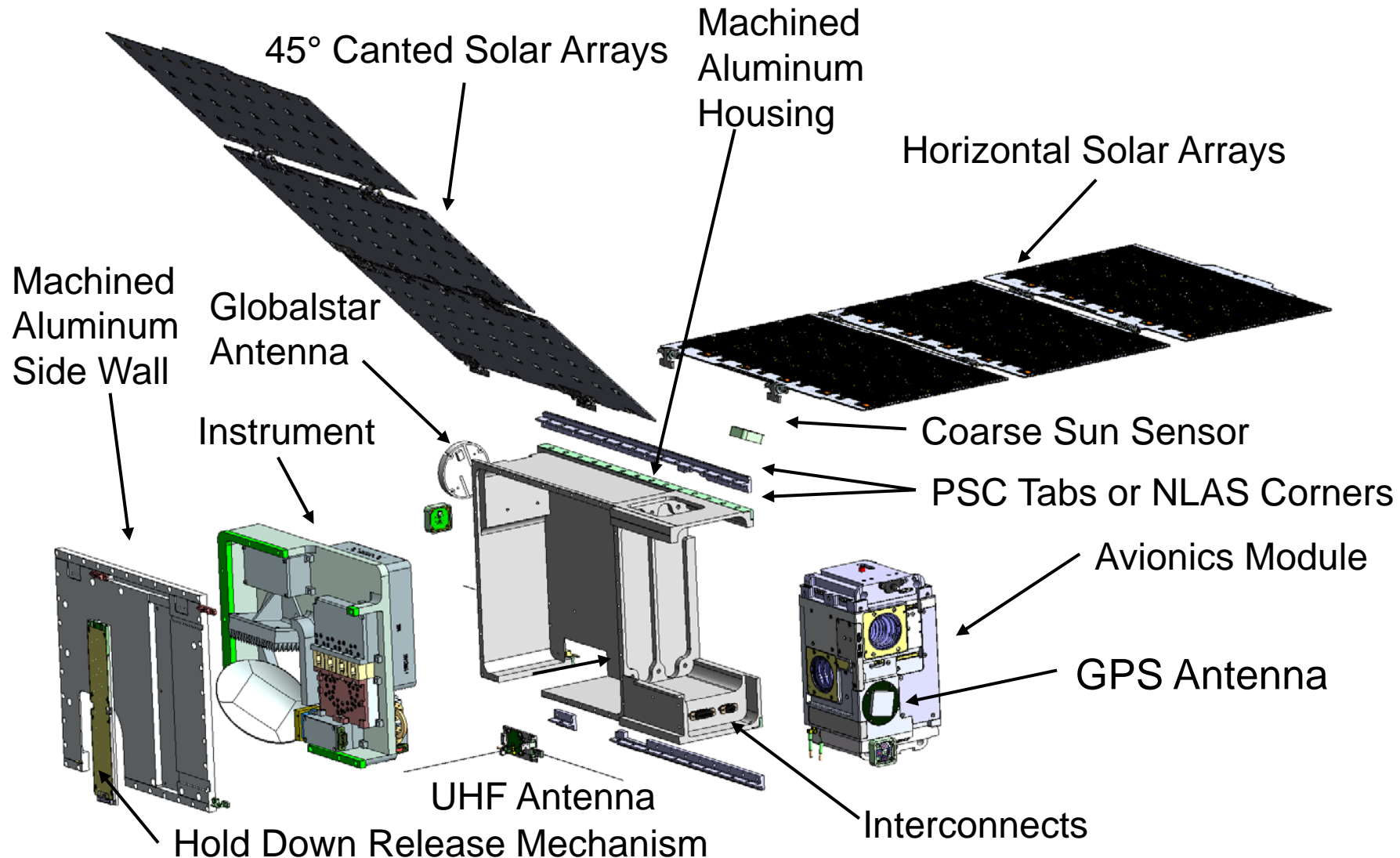


- Demonstrate capability of 6U-Class satellites to contribute to NASA Earth Science measurements in a 90-day technology demonstration mission
- Reduce **risk**, cost and development time for small satellite constellations for Earth Science measurements
- Raise the technology readiness level (TRL) of the TEMPEST mm-wave radiometer instrument from 6 to 9 (scanning reflector to 7)
- Provides the first in-space demonstration of a millimeter-wave radiometer with an InP HEMT low-noise amplifier front-end (LNA) for Earth Science measurements.

## Success Criteria:

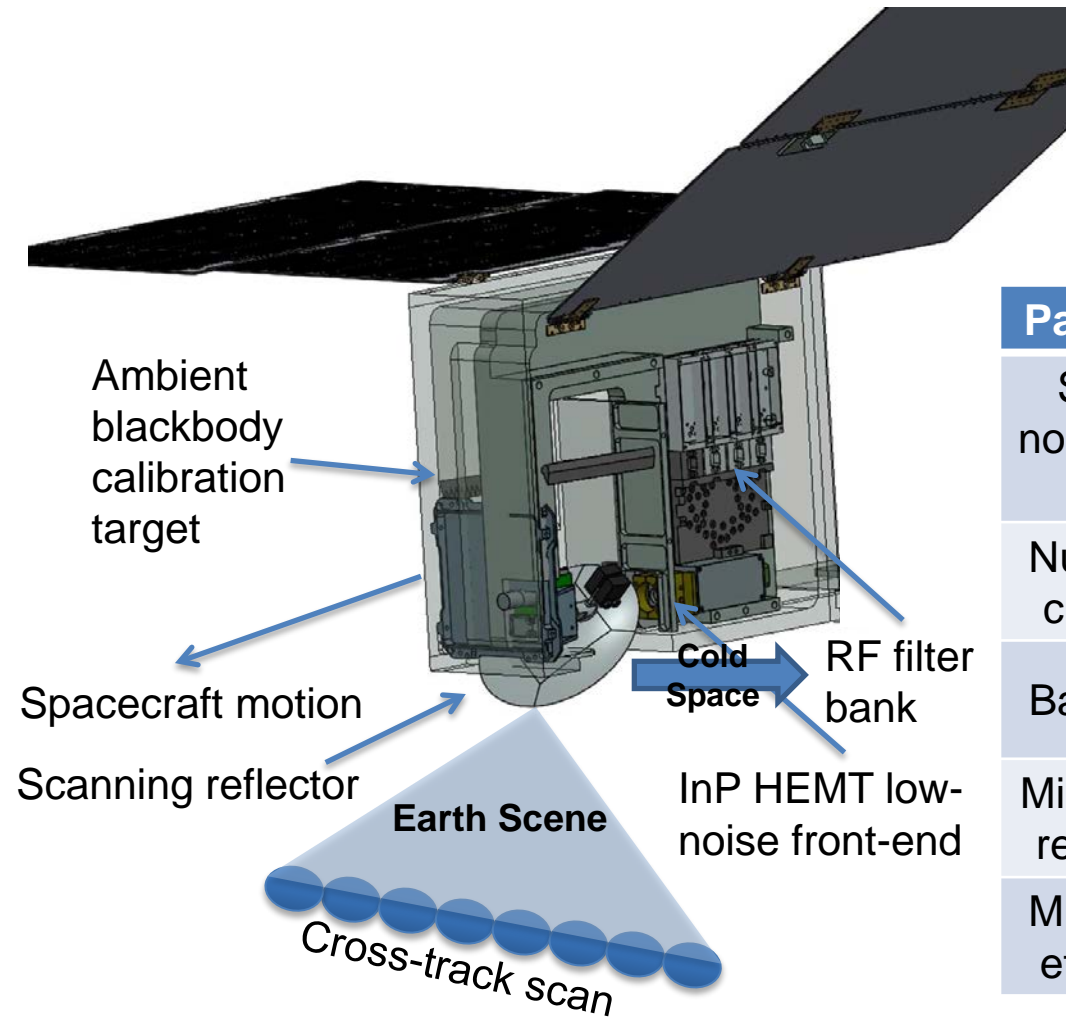
- Demonstrate feasibility of differential drag maneuvers to achieve required time separation of 6U-Class satellites in same orbital plane
- Demonstrate cross-calibration between TEMPEST mm-wave radiometers and NASA/JAXA Global Precipitation Mission Microwave Imager and/or Microwave Humidity Sounder (MHS, on two NOAA satellites and two ESA/EUMETSAT satellites) with 2 K precision and 4 K accuracy.

# TEMPEST-D 6U-Class BCT Spacecraft Bus based on XB1





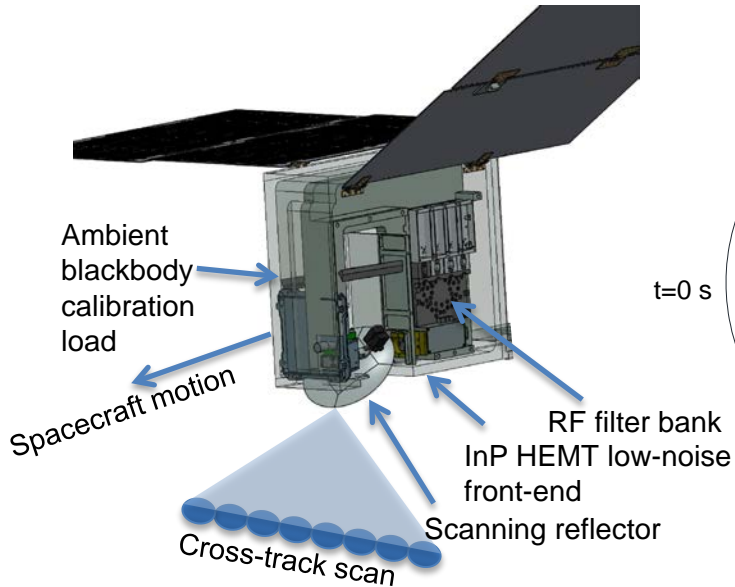
# TEMPEST-D Millimeter-Wave Radiometer for 6U-Class Satellite



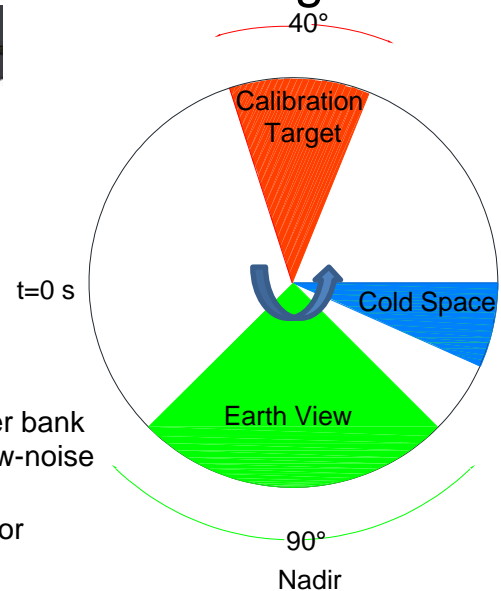
Parameter	Specification	
System noise temp.	< 800 K at 89 GHz < 1300 K at 165, 176, 180 and 182 GHz	
Number of channels	5	
Bandwidth	4 GHz at 89 and 165 GHz 2 GHz at 176, 180 & 182 GHz	
Min. spatial resolution	13 km at 182 GHz	25 km at 89 GHz
Min. beam efficiency	> 90%	> 90%

# TEMPEST-D Instrument: Radiometer Calibration

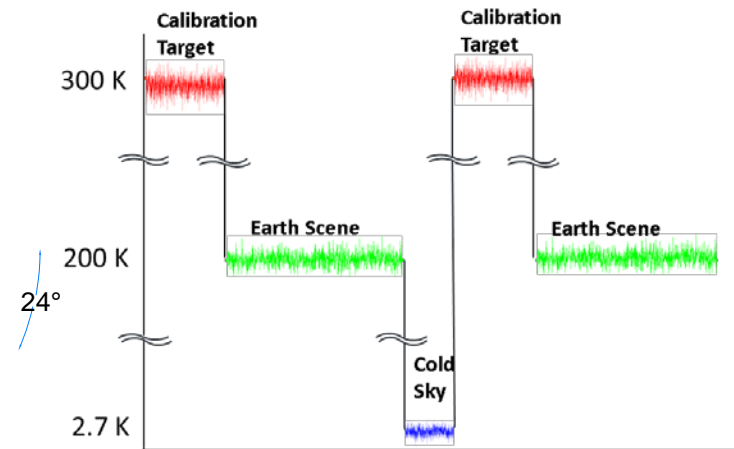
## TEMPEST-D Instrument



## Observing Profile

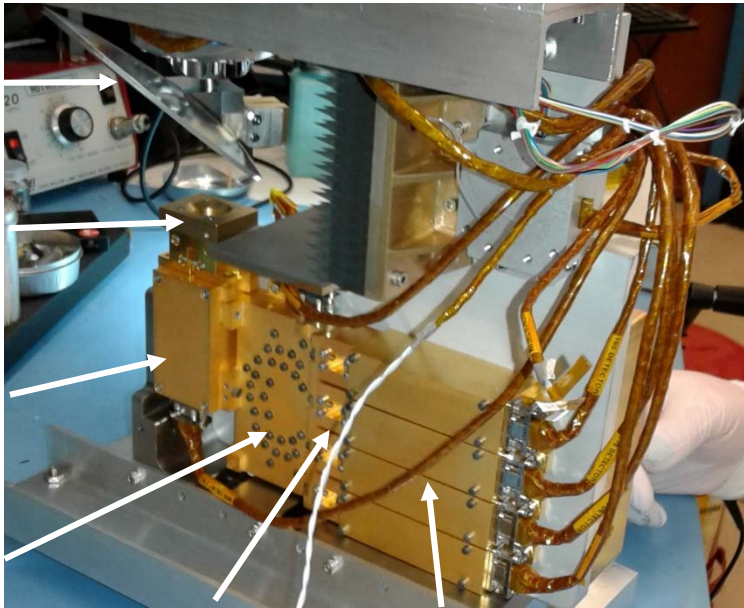


## Time Series of Output Data



- Five-frequency millimeter-wave radiometer measures Earth scene over  $\pm 45^\circ$  nadir angles, providing an 825-km swath width from a nominal orbit altitude of 400 km. Each pixel is sampled for 5 ms.
- Space view observes cosmic microwave background at 2.73 K (“cold sky”). Ambient Blackbody calibration target is measured each revolution to perform two-point external calibration every 2 sec. (scanning at 30 RPM).

# Flight Model Radiometer Instrument Bench-top Integration at JPL



Scanning Reflector

Dual-Frequency Feed horn

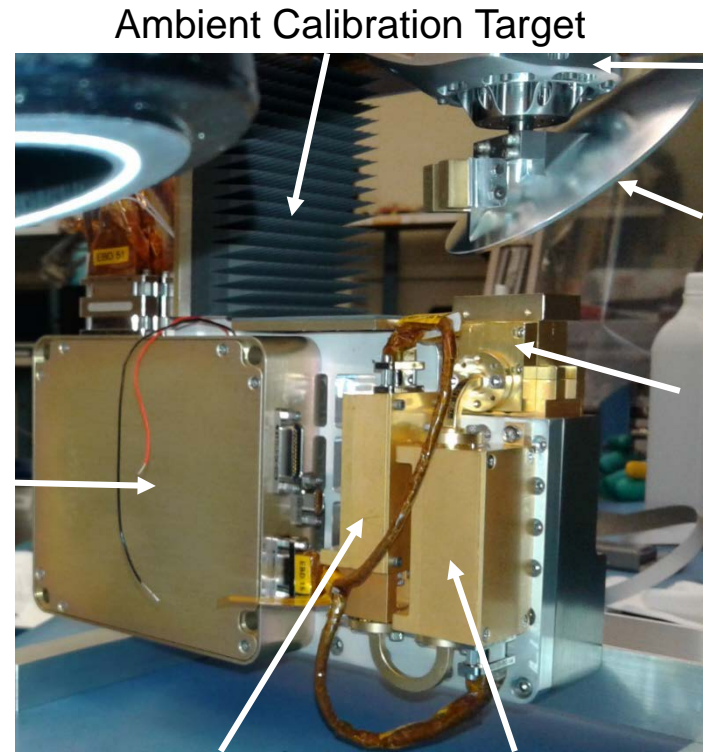
165-182 GHz Radiometer Front-end

165-182 GHz Power Divider

165-182 GHz Filter Bank

165-182 GHz Detectors

Command & Data Handling and Power Distribution Subsystem



Ambient Calibration Target

Scanning Motor  
Scanning Reflector

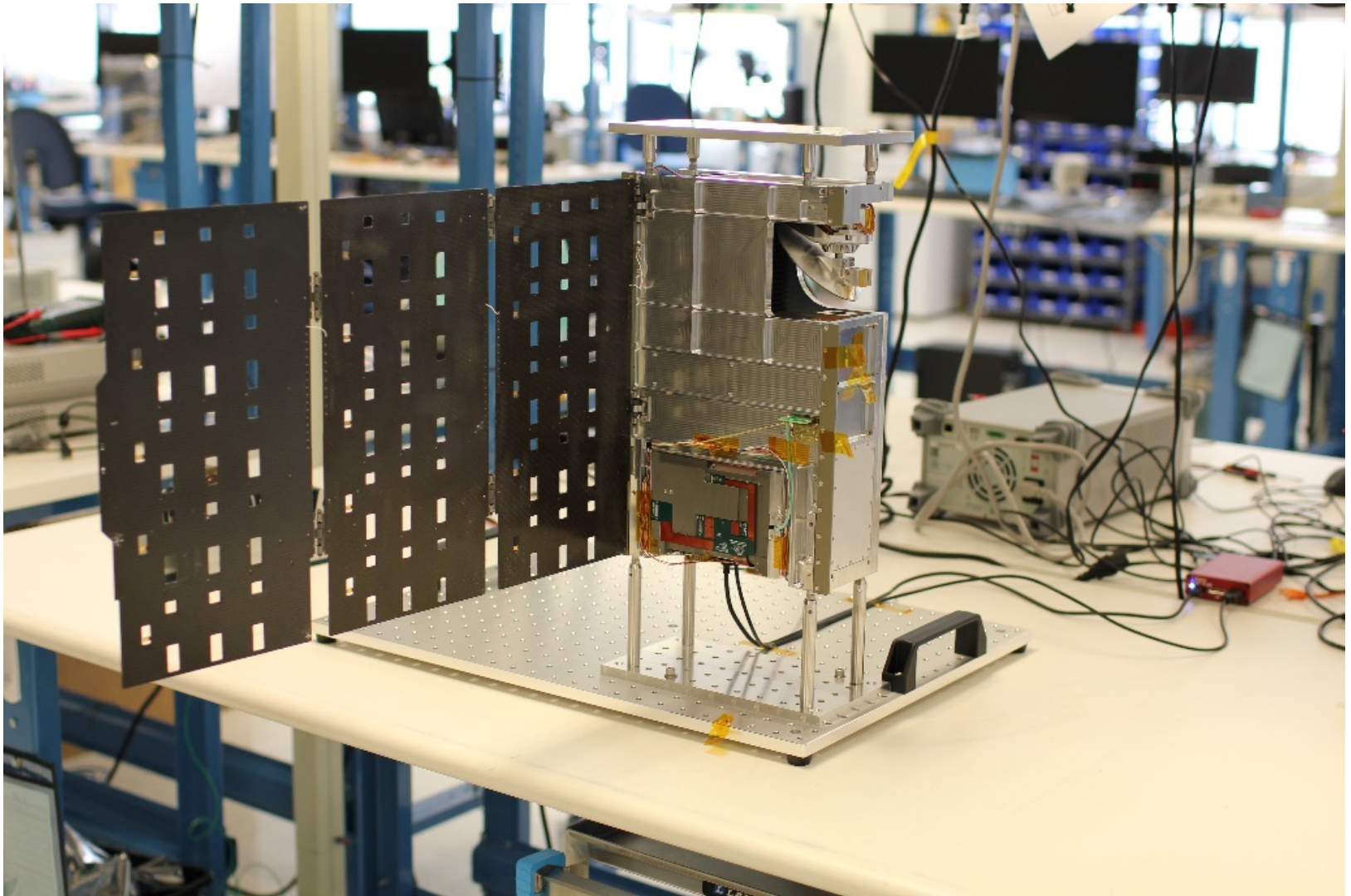
Dual-Frequency Feed horn

89 GHz Detector

89 GHz Radiometer Front-end

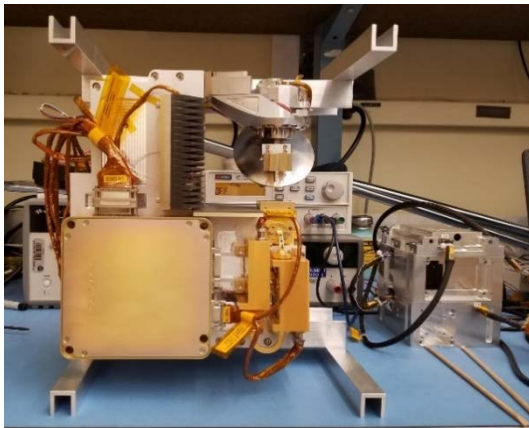


# Spacecraft Bus Integrated with Spare Instrument at BCT

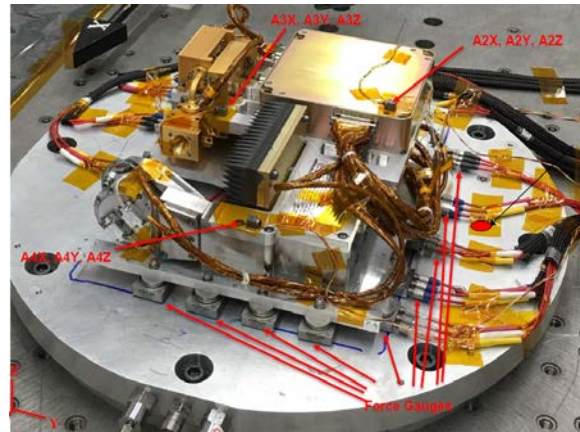




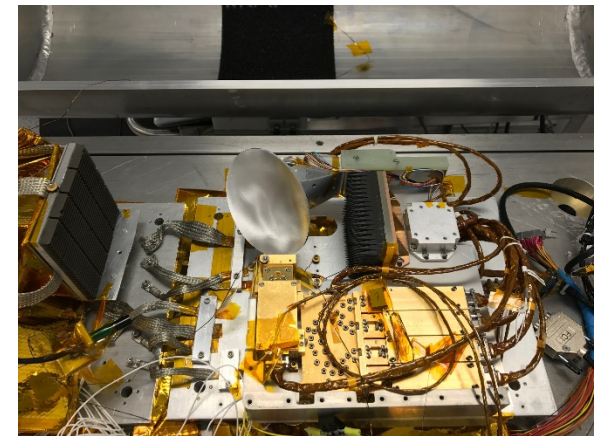
- **Flight spare:** Successfully completed EMI/EMC self-compatibility testing with spacecraft bus. (April 2017)
- **Flight instrument:**
  - Completed end-to-end receiver bandpass and linearity measurements successfully. (June 2017)
  - Integrated and vibration testing successfully completed. (June 2017)
  - Thermal vacuum testing successfully completed. (July 2017)
  - Antenna pattern validation measurements performed. (July 2017)
  - Delivered to BCT for integration with spacecraft bus & testing (July 2017)



Instrument Assembly



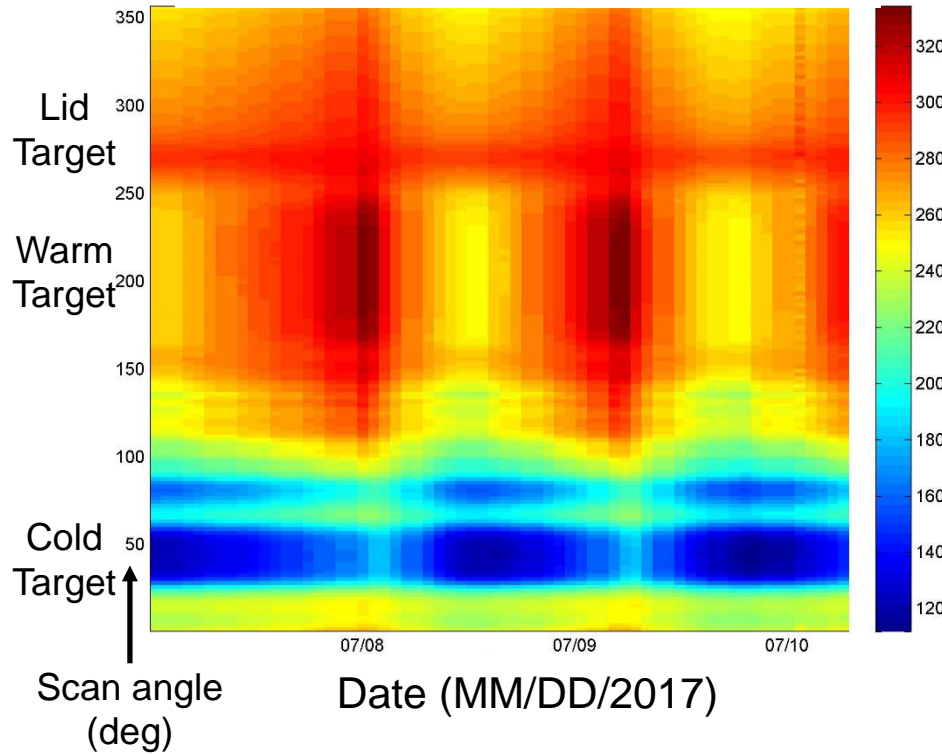
Vibration Testing



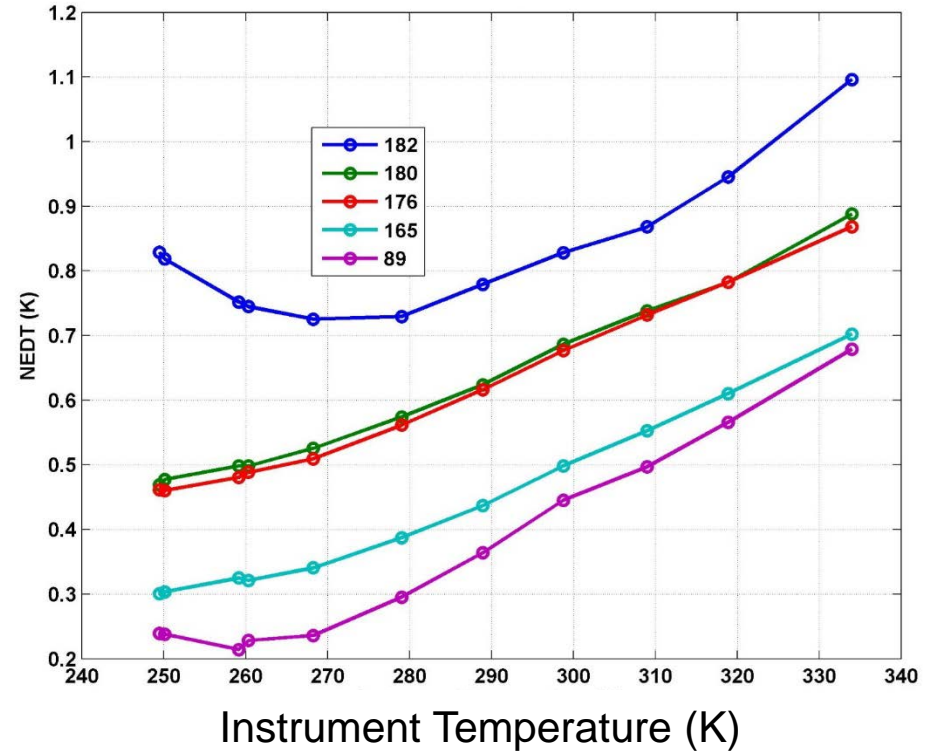
TVAC Testing

# TVAC Testing Results for Flight Instrument

89 GHz Antenna Temperature (K)



NEDT vs. Temperature for 5 channels



NEDT is measured on warm target, which varies with chamber temperature.

# Summary



- TEMPEST-D mission to demonstrate capability of 6U-Class satellites to perform global observations of clouds and precipitation processes
- Reduces risk, cost and development time for repeat-pass radiometry to measure temporal signatures of precipitation using small satellite constellations
- Provides first in-space technology demonstration of a millimeter-wave radiometer based with an InP HEMT low-noise amplifier front-end for Earth Science measurements
- Raises the TRL of the TEMPEST mm-wave radiometer instrument from 6 to 9 (scanning reflector to 7)
- Demonstrates the feasibility of differential drag maneuvers to achieve required time separation of 6U-Class satellites in the same orbital plane
- Demonstrates cross-calibration of TEMPEST radiometers with NASA/JAXA GPM Microwave Imager and/or MHS with 2 K precision and 4 K accuracy
- Features rapid development cycle of 2.5 years from project start to delivery to NanoRacks for integration on Feb. 1, 2018.
- Launch expected on Cygnus Antares II from Wallops to ISS on Mar. 14, 2018





*Thank you for your kind attention. Many thanks to NASA Earth Ventures for their support and to the NASA Earth Science Technology Office for program management.*