

# Vibration response characterisation and fault-size estimation of spalled ball bearings

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*Research efforts have increased to investigate the ability to quantify localised bearing faults, ie spalls. These efforts revolve around extending the useful service life of the bearing after the detection of spalls. A number of studies have investigated a linear correlation between the size of spalls and three geometric points that may be recognised in the vibration response: the entry into the spall; the exit from the spall; and a third impact point between the first two. The time difference between these points, calculated using different signal processing techniques, has been widely exploited for quantifying spall size. Currently, there are two main challenges: the first is to enhance the entry point, which typically has weak excitation; the second is to distinguish the impact and the exit points investigated in the literature based on the spall size. However, for practical applications, there is no prior rough estimation of the fault size (ie small or large) and a method is needed for the interpretation of responses. This paper provides insights into the movement of the rolling element within the spall region and shows that the rolling element strongly strikes the bearing races at a minimum of two points. A new technique is then presented to quantify the spall and determine the inherent scaling factor without comparison to any reference data. The technique is based on evaluating two root-mean-square (RMS) energy envelopes, one for the vibration signal and one for a numerical differentiation of this signal. A geometric scaling factor is then used to give a generalised quantification for the small and large spalls. Serviceable estimations of spall size have been achieved for several seeded faults measured on two dissimilar test-rigs provided by the German Aerospace Centre (DLR) and the University of New South Wales (UNSW).*

## 1. Introduction

Rotating machinery systems are typically equipped with a number of rolling element bearings in order to provide frictionless support for the spindles. The existence of faults in the bearings leads to bearing damage; it can also initiate hazardous vibrational excitations within the whole structure. To prevent such risky scenarios, fault diagnosis and prognosis capabilities may be developed. Fault diagnosis is used to detect emergent faults without evaluating their size or severity in order to launch immediate maintenance. However, more economic benefits can be attained through fault prognosis, in which faulty bearings are tolerated as long as the fault is below a critical limit of severity<sup>[1]</sup>.

Here, the scope is limited to a specific prognosis task: the fault quantification of seeded faults rather than complete remaining useful life estimation. The ball bearings degrade in two manners: localised spalling/pitting and/or distributed wear<sup>[2,3]</sup>. This paper focuses on spall-type degradation because it is the ordinary degradation if the bearing is operated and maintained properly.

Recently, several vibration-based studies have been conducted to investigate spall size<sup>[4-9]</sup> based on the principle that there is a linear correlation between spall width and the geometric features identified in the vibration response. These features consist of three points: the ball entry into the spall zone (the fault), an impact at the spall centre and the ball exit from the spall. The time difference between those impacts, estimated using different signal processing techniques, has been exploited for quantifying the spall.

However, there is a challenge related to which of these features actually exists. The literature has investigated the existence of only two combinations: the entry to the impact or the entry to the exit. It is essential to know which combination is present in order to scale the extracted spacing to the actual fault size using a geometric factor. The existing studies set this factor manually by comparing

reference fault data. However, for new assemblies, this factor cannot be calculated in advance because it requires very complex models and precise loading conditions.

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This paper introduces a new technique for extracting spall size and determining the geometric factor between the extracted size and the actual size without comparison to any reference data. The paper is organised in four sections: Section 2 formulates the spall quantification problem and the current state-of-the-art advances; Section 3 presents a description of the seeded faults and the test-rigs used for the experiments; Section 4 presents the new quantification principle, which is then experimentally verified; followed by a conclusion in Section 5.

## 2. Investigating vibration-based geometric features

A localised spall fault on one of the bearing elements produces a periodical disturbance to the rotating shaft, as shown in Figure 1. This disturbance is due to successive passages of the balls over the spall, which causes a series of impact forces that excite the bearing sub-components and the attached assembly<sup>[10]</sup>.

These impacts are induced at a unique fault characteristic frequency (FCF), which is a function of the spall location (*ie* at the inner race, at the outer race or at the ball), the running speed *N* in Hertz, the bearing geometry, for example the ball and pitch diameters, and the load angle. The spall size, however, is independent of the FCF value, which is utilised only to detect the fault. In addition, the impact's amplitude (*Am* in Figure 1(d)) is not monotonically correlated with the spall size<sup>[11]</sup>.

Specific geometric features of the vibration response have been widely investigated as a promising quantification approach<sup>[5,6,8]</sup>. These features are the entry into the spall zone 'A', an impact at the spall centre 'C' and the exit from the spall 'B', as shown in Figure 2. The entry point A involves a disturbance of the ball's path of motion without significantly striking the spall leading edge; at location C, which may be close to the mid-point between the entry and exit, the ball is found to strongly strike the spall, giving rise to the vibration response, C. The exit B is often difficult to identify due to the complex damping/decaying of the response<sup>[5]</sup>.

The lag time between these points (Figure 2(a)) can be mapped to the spatial domain (Figure 2(b)) in order to estimate the actual spall size in millimetres using the following Formula (1), which is adapted from<sup>[5]</sup>:

$$L = T \frac{G \pi f_r (D_p^2 - d^2)}{2D_p} \dots\dots\dots (1)$$

where *L* (mm) is the spall width, *T* (s) is the lag time between points A and B or A and C depending on the vibration response, *f<sub>r</sub>* (Hz) is the shaft speed, *D<sub>p</sub>* (mm) is the pitch diameter of the bearing and *d* is the ball diameter (mm). This formula can work for faults on both the inner and outer races with a maximum error of 4%, assuming that the load angle of the bearing is neglected as investigated in<sup>[5]</sup>. The term *G* (unitless) represents the geometric factor, which relates the actual size to the extracted size. The *G* value depends on how the

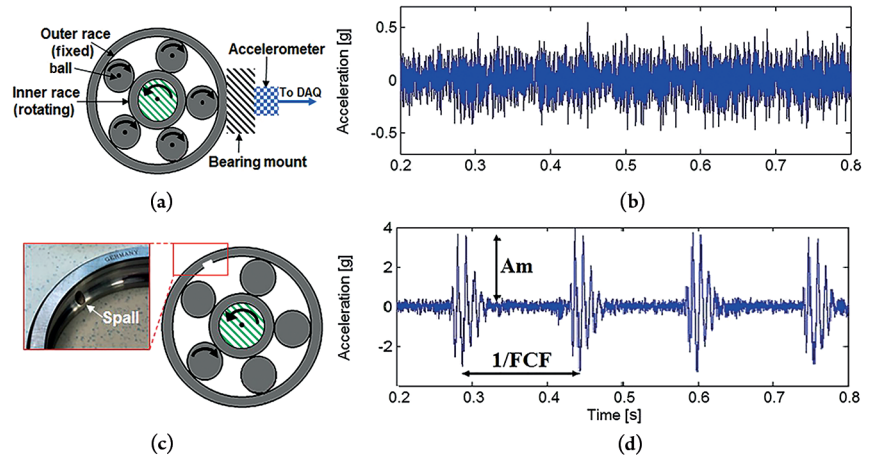


Figure 1. (a) Ball bearing elements and the arrangement used to measure vibration; (b) vibration response of an undamaged bearing; (c) a seeded spall at the outer race; (d) vibration response due to the presence of the fault

Table 1. Summary of the spall quantification literature and geometric factors

	Study	Speed range (r/min)	Load (kN)	Spall sizes (mm)	Geometric ratio, <i>G</i>
1	Epps, 1991 <sup>[4]</sup>	1500-3000	1.5-7	1-1.2	• 2 – manually interpreted
2	Sawalhi, 2011 <sup>[5]</sup>	800-2400	–	0.6-1.2	• 2 – manually interpreted
3	Jena, 2012 <sup>[6]</sup>	1500	–	2.1	• 1 – manually interpreted
4	Moustafa, 2014 <sup>[7]</sup>	10-60	1-10	0.7-6	• 1 and 2 – manually interpreted
5	Ismail, 2015 <sup>[9]</sup>	60-500	5-8.8	1.4-4	• 1 – manually interpreted
6	This work	500-1200	0-5	0.6-4	• 1 and 2 – automatically identified

interval *T* is interpreted.

Two scenarios have been cited in the literature, as presented in Table 1. The first one identifies only the entry point A and the impact point C in the vibration response. The actual size is extracted by multiplying the entry-impact interval, *T* = *T<sub>AC</sub>*, by a factor of two, *ie G* = 2, as shown in Figure 2(a). The other scenario involves the identification of entry point A and exit point B (Figure 2(b)); thus, the actual size is 100% of the extracted size, *ie G* = 1, where *T* = *T<sub>AB</sub>*.

The selection of *G* is performed manually by comparing the extracted size and the actual size. This is because points C and B may have similar responses and thus might not be correctly distinguished. Such confusion can lead to serious problems: (a) assuming incorrect *G* for new non-faulty applications can cause errors of 100%; (b) the extraction of points A, B and C has not been systematically investigated in the literature for more than a test-rig. Both challenges will be investigated.

## 3. Test-rig and seeded fault specifications

In this work, two different test-rigs are used in order to experimentally validate the technique for datasets that have multiple diverse factors, including the size of seeded faults, speeds, load types, resonant frequencies and geometric factors; the latter parameter should be automatically extracted. The seeded faults were obtained by a spark erosion for both cases, as depicted in Figure 3 and Table 2; one of the radial vibration measurements was found to be sufficient for this study.

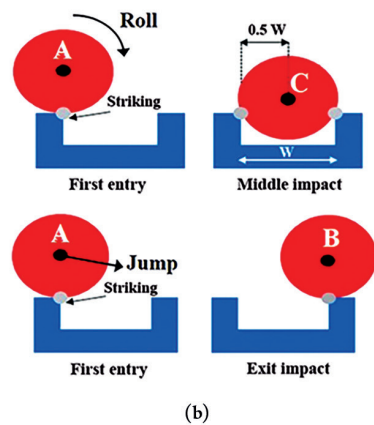
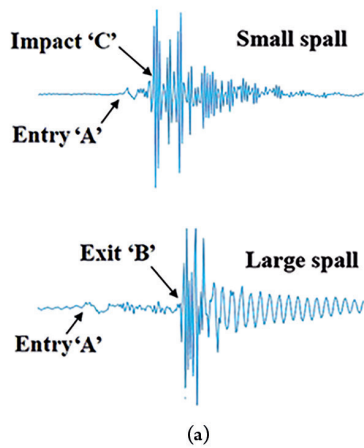


Figure 2. (a) Possible vibration response patterns for spalled bearings (the majority of small spalls match the upper shape, while larger spalls match the bottom case); (b) the corresponding geometric features: A indicates the entry point, B indicates the exit point and C indicates a middle impact point. No signal-based identification between the shown responses has been investigated in the literature

### 4. Spall quantification process

The quantification problem can be formulated in terms of how the geometrical features of the spall (ie A, B and C) are encoded in the vibration response. Basically, there are two encoding forms for any dynamic signal: the amplitude and the frequency modulations. As shown in Figure 2, the exit feature B is mainly encoded as a large amplitude variation in the response, while the entry feature A is more manifested by a short-time variation of the frequency. The middle point C may be identified by both the frequency and the amplitude modulations. The raw vibration response has more distinguishable amplitude-encoding content than the frequency content. Here, a numerical differentiator is proposed to enhance the features that have been encoded in the frequency modulation. An example of this concept is shown in Figure 4.

Based on this investigation, a new investigation to account for both frequency and amplitude encoding types is proposed, as shown in Figure 5. Two processing paths are proposed; the first one involves a root-

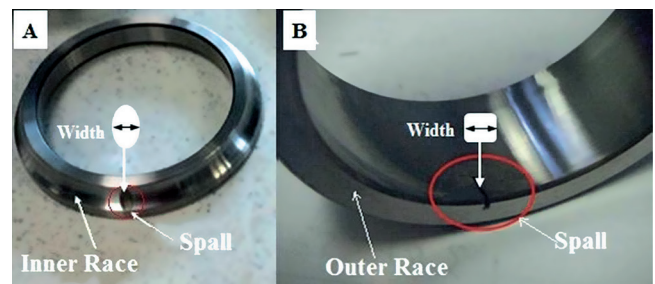


Figure 3. Seeded spall examples for test-rigs: DLR in A and UNSW in B

mean-square (RMS) energy envelope, the first envelope, which is fed by the raw signal. This envelope reflects the vibration energy due primarily to the amplitude modulation. The other path consists of two stages: a numerical differentiator and the second RMS envelope. The differentiator is utilised to extract the inherent frequency modulation in the signal and it is then quantified by the second RMS envelope.

The resultant envelopes are compared in order to locate possible geometric features and determine their relationship to the actual size, ie determine the geometric factor. A study for numerous combinations of spall sizes, loads and speeds was conducted on dissimilar test-rigs (ie two of them are included here), leading to the three main patterns of envelopes depicted in Figure 6. This shape diversity can be understood as a result of different physical contributions, such as loads and their types, shaft speed, spall size and bearing curvatures.

The envelope patterns in Figure 6 can be interpreted by comparing how geometric features are manifested. Here, the interest is in identifying the spall width rather than the depth, which will be considered in future work. Entry feature A has a fixed criterion for all cases, which is the earlier increase of the response, whatever it is, included in the first or second envelopes. In case 2, both envelopes have approximately the same centre. This case indicates that the response is symmetric around the spall centre, thus supporting the entry-impact pattern. The majority of small spalls follow this pattern. In case 3, the centre of the highest peak of the first envelope is significantly shifted away from the virtual centre of the second envelope.

This has been observed for large-spalls. The physical interpretation can be explained based on the main attribute of the first envelope, which is more influenced by higher frequencies than the second envelope. Large separation of the envelopes indicates

Table 2. Test-rig specifications

Test-rig		Fan-gear test-rig <sup>[5]</sup>	Laboratory testing machine <sup>[9]</sup>
Data source		University of New South Wales (UNSW)	German Aerospace Centre (DLR)
Bearing code		NACHI 2206GK	QJ212TVP
Geometric factor		2	1 and 2
Ball diameter	mm	7.93	15.87
Pitch diameter	mm	39	85.15
Speed	r/min	800-1200	500
Sampling rate	kHz	65.5	25.6
Load	N	Radial support load	Axial load of 5000
Inner faults – width	mm	1.2	1, 2.1, 3.8
Outer faults – width	mm	0.6, 1.2	1.4, 2.4, 4



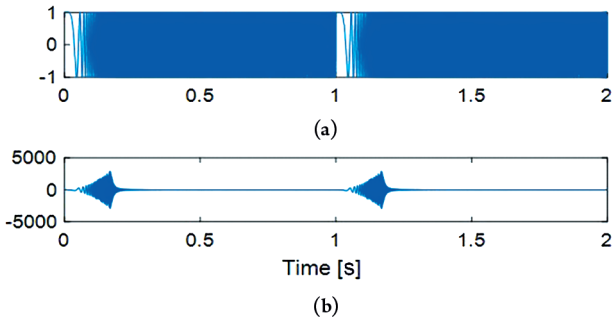


Figure 4. A demonstration of the frequency demodulation by a differentiator: (a) two repeated unity chirp signals, simulating the weak excitation of a spall; (b) a first-order differentiation for the signal of (a), where the frequency content is extracted as an amplitude modulation. The differentiator has a limited bandwidth that stops all frequencies above a certain threshold

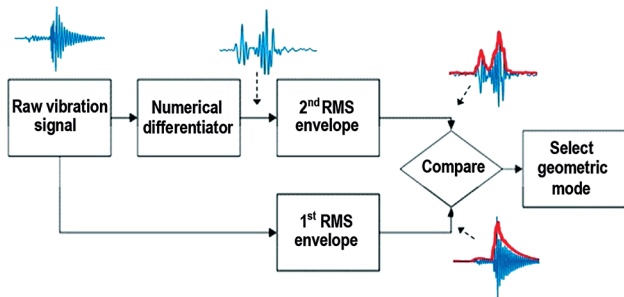


Figure 5. Signal flow of the spall quantification technique

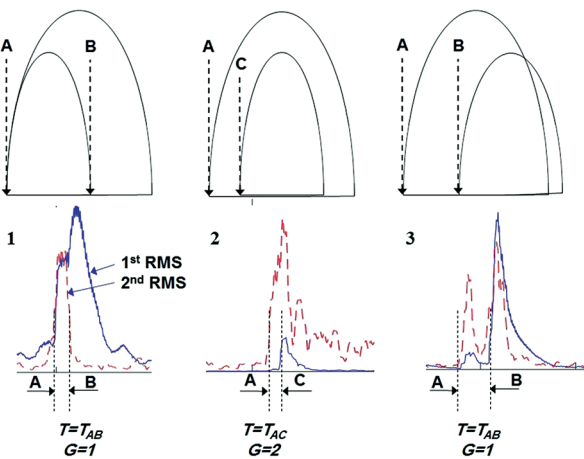


Figure 6. Three patterns are proposed for the envelope modes calculated in Figure 5. The top row represents the conceptual relationship between the outer profiles of the envelopes, while the bottom row shows examples extracted from dissimilar test-rigs: 1 mm inner fault, 1.2 mm inner fault and 3.8 mm outer fault for cases 1 to 3. The envelope magnitudes are not used here because they are mainly influenced by load

a large spall, thus case 3. The remaining case, case 1, represents a special case with high axial loading, where the entry point excites a higher resonant frequency; this excitation is then disturbed as the ball departs the spall.

The Savitzky-Golay differentiator (SGD)<sup>[12]</sup> is used here to perform an adjustable numerical differentiation for the vibration signal. SGDs were introduced in 1964 to smooth and differentiate the data while maintaining the underlying signal's waveform (for

example width and height) obtained from noisy data. The theory of SGDs is to provide an approximation of a set of noisy data points  $x(n)$  of length  $2M + 1$  by fitting them into a polynomial  $p(n)$  of order  $N$ , as in Equation (2)<sup>[12]</sup>:

$$p(n) = \sum_{k=0}^N c_k n^k \quad -M \leq n \leq M \dots\dots\dots (2)$$

where  $c_k$  denotes the polynomial coefficients of the length  $N + 1$ . The input data are defined by a window frame with a length  $f = 2M + 1$  and centred at  $n = 0$ . The process of fitting these input data entails an optimal fitting of  $p(n)$  coefficients. The number of unknown coefficients  $N + 1$  should be selected to be less than the number of the data frame  $2M + 1$ . A demonstration of the SGD principle is depicted in Figure 7. The frequency response of SGD is defined by two parameters, the polynomial order  $k$  and the size of the data window  $f$ , as shown in Figure 8. The term bandwidth (BW) will be used here to indicate the largest amplification interval prior to any spectral leakages.

The determination of the SGD parameters is crucial to separate

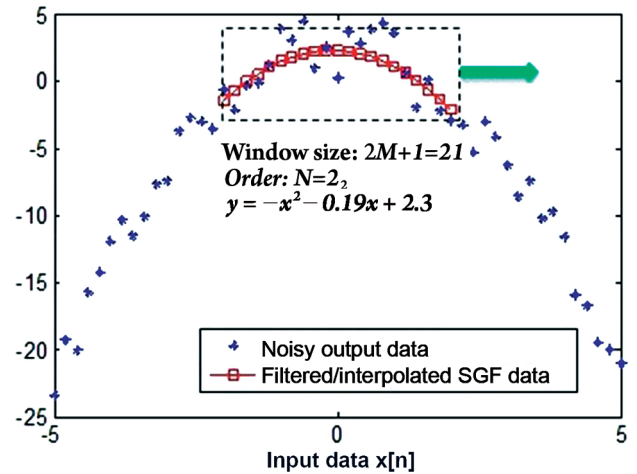


Figure 7. An SGD example for differentiating noisy data. A movable data frame of 21 points is used to fit a polynomial:  $y(x) = -x^2 - 0.19x + 2.3$ . The smoothed value at the frame centre ( $x = 0$ ) equals the absolute term  $c_0 = 2.3$ , while two smoothed derivatives are available at the same point,  $dy/dx = c_1 = -0.19$  and  $d^2y/d^2x = 2c_2 = -2$ . For another  $x$  point, the data window should be centred on this point; in addition, a new polynomial should be fitted

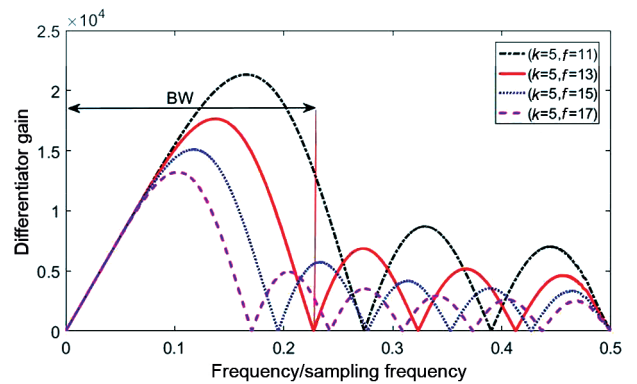


Figure 8. Frequency response of an SGD of  $\{f, k\} = \{11:14, 2\}$ . The bandwidth (BW) is located as an example at  $f = 13$ . The BW is proportional to  $(1/f)$ ; increasing  $f$  permits BW scanning of the response for possible quantification features

a specific bandwidth in which the geometric features are most likely to be distinguishable. We empirically determined that the best window size,  $f$ , is correlated with the kurtosis of the differentiated signal. The kurtosis is a well-known statistical quantity that is quadratically proportional to the variance. The best value of  $f$  can be attained by evaluating increasing values of SGD until the kurtosis rate is approximately zero, as shown in Figure 9.

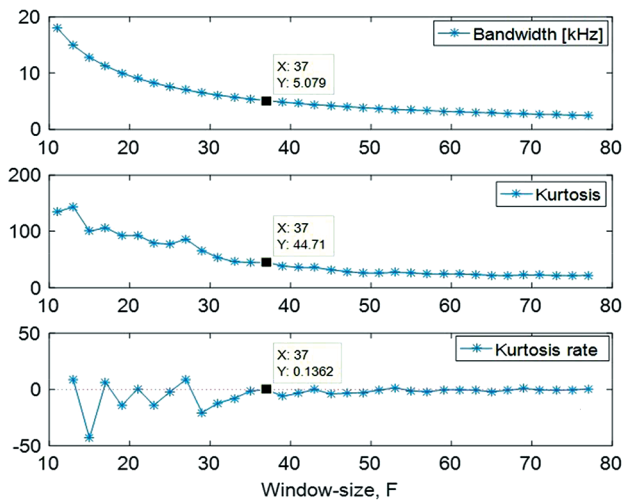


Figure 9. The criterion for selecting the differentiator window,  $f$ . The upper plot shows the inherent F-BW relationship of an SGD. The middle plot indicates the corresponding kurtosis values for each  $f$ . The bottom shows the kurtosis rate, which at  $f = 37$  satisfies a steady-state slow rate below 1

Two examples of resultant envelopes are depicted in Figures 10 and 11. The raw vibrations for both cases are similar, without clear distinguishable features. However, the envelopes in Figure 10 are similar to case 2 in Figure 6, in which the first and second envelopes have approximately the same centre. The entry point A is identified by the second envelope and the middle impact C is identified by the first envelope. Other examples are shown in Figures 11 and 12 for envelopes that follow case 3 and case 1, respectively.

Table 3 and Figure 13 summarise the results for the given datasets. For each case, the data is trimmed to one revolution, or six spall impacts. The average quantification error of DLR data is 12%, while it is 13% for UNSW, 800 and 1200 r/min, respectively.

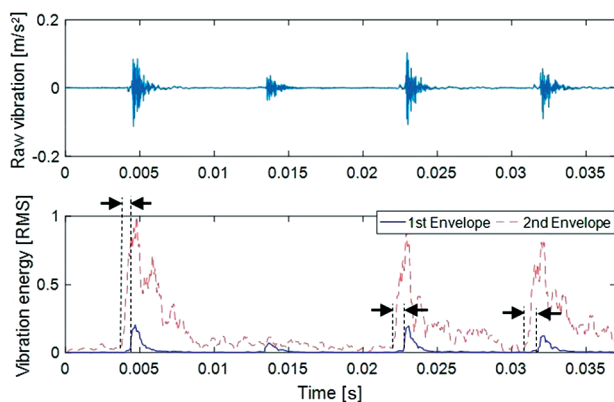


Figure 10. The raw vibration and dual RMS envelopes for an inner spall-width of 1.2 mm at 800 r/min from UNSW. The envelopes match case 2 of Figure 6

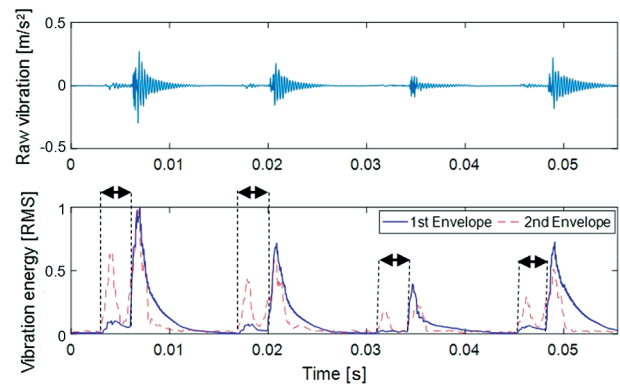


Figure 11. The raw vibration and dual RMS envelopes for an inner spall-width of 3.8 mm at 500 r/min from DLR. The envelopes match case 3 of Figure 6

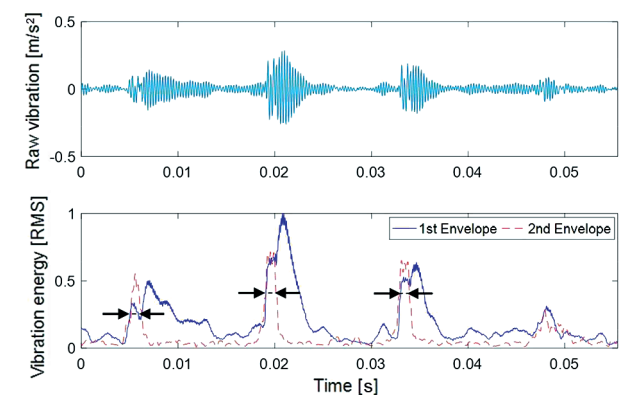


Figure 12. The raw vibration and dual RMS envelopes for an outer spall-width of 2.4 mm at 500 r/min from DLR. The envelopes match case 1 of Figure 6

The maximum error of DLR is 24% because it is biased by a 1 mm inner fault. The maximum error of UNSW is 22%, which is close to the DLR maximum value. The large errors for both cases are observed within the smallest spall sizes, which are the most likely to be masked by the background noise. The quantification error may be predicted based on the standard deviation of the results. As shown in Table 3, higher standard deviation is proportional to the precision of the extracted size.

## 5. Conclusion

This paper has presented a fault quantification technique based on identifying the geometric features of a fault in the vibration response. The technique consists of two energy (RMS) envelopes for the vibration signal and a numerical differentiation of this signal. The differentiator is an adjustable Savitzky-Golay differentiator (SGD) that is automatically tuned based on the kurtosis rate to a wide range of possible parameters. The mutual features of the envelopes are grouped into three generalised patterns, which characterise different excitation forms due to the passage of the ball over the fault zone. In 12 investigated cases from two dissimilar test-rigs, the technique estimated the physical size of the spall with average errors of 12% and 13% and maximum bounds of 24% and 22% for DLR and UNSW data, respectively. The proposed future work is to validate the technique on more diverse loading and speed conditions for naturally induced spalls.

Table 3. Summary of the quantification results for UNSW and DLR test-rigs

Spall width (mm), type, speed (r/min)	Extracted width (one revolution)		Average error		SGD		Envelope pattern
	$\mu$ (mm)	$\sigma$ (mm)	(mm)	(%)	$f(k=5)$	BW [kHz]	
Test-rig: DLR							
1.0, inner, 500	1.24	0.20	0.24	24	49	1.5	2
2.1, inner, 500	2.15	0.14	0.05	2	41	1.8	3
3.8, inner, 500	3.40	0.14	0.4	11	31	2.4	3
1.4, outer, 500	1.28	0.36	0.12	9	45	1.6	1
2.4, outer, 500	2.84	0.51	0.44	18	49	1.5	1
4.0, outer, 500	3.69	0.28	0.31	8	23	3.2	3
Maximum and average error				24, 12			
Test-rig: UNSW							
1.2, inner, 800	0.94	0.11	0.26	22	35	5.4	2
1.2, outer, 800	1.12	0.17	0.08	7	37	5.0	2
0.6, outer, 800	0.47	0.03	0.13	22	35	5.4	2
1.2 outer, 1200	1.06	0.11	0.14	12	33	5.7	2
1.2 inner, 1200	1.07	0.06	0.13	11	41	1.8	2
0.6 outer, 1200	0.57	0.03	0.03	5	35	5.4	2
Maximum and average error				22, 13			

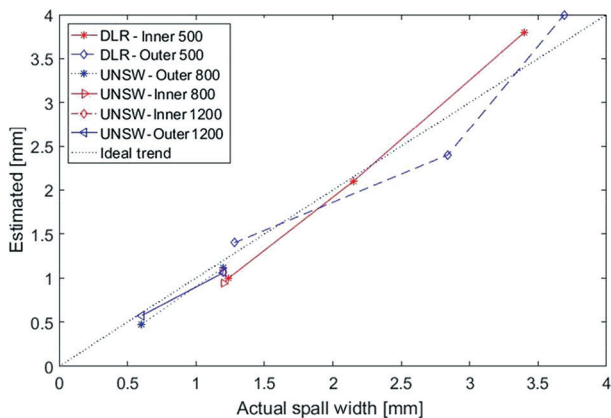


Figure 13. The results of using energy envelopes for estimating spall-width presented in Table 3. The estimations have averages matching the ideal trend

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