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### Measuring water level in rivers and lakes from lightweight Unmanned Aerial Vehicles

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1	Journal of Hydrology and in press.
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3	
4	Measuring water level in rivers and lakes from
5	lightweight Unmanned Aerial Vehicles
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10	
11	Highlights
12	•Water level of rivers and lakes can be measured by Unmanned Aerial Vehicles.
13	•Unmanned Aerial Vehicles ensure high accuracy and spatial resolution.
14	•The measuring system consists of a ranging sensor and a GNSS receiver.
15	•Among the ranging sensors, the radar has the highest accuracy and longest range.
16	•The camera-laser sensor is preferred for narrow field of view to water surface.
17	
18	Abstract

19 The assessment of hydrologic dynamics in rivers, lakes, reservoirs and wetlands requires 20 measurements of water level, its temporal and spatial derivatives, and the extent and dynamics of 21 open water surfaces. Motivated by the declining number of ground-based measurement stations, 22 research efforts have been devoted to the retrieval of these hydraulic properties from spaceborne 23 platforms in the past few decades. However, due to coarse spatial and temporal resolutions, 24 spaceborne missions have several limitations when assessing the water level of terrestrial surface 25 water bodies and determining complex water dynamics. Unmanned Aerial Vehicles (UAVs) can 26 fill the gap between spaceborne and ground-based observations, and provide high spatial 27 resolution and dense temporal coverage data, in quick turn-around time, using flexible payload 28 design. This study focused on categorizing and testing sensors, which comply with the weight 29 constraint of small UAVs (around 1.5 kg), capable of measuring the range to water surface. 30 Subtracting the measured range from the vertical position retrieved by the onboard Global 31 Navigation Satellite System (GNSS) receiver, we can determine the water level (orthometric 32 height). Three different ranging payloads, which consisted of a radar, a sonar and an in-house 33 developed camera-based laser distance sensor (CLDS), have been evaluated in terms of 34 accuracy, precision, maximum ranging distance and beam divergence. After numerous flights, 35 the relative accuracy of the overall system was estimated. A ranging accuracy better than 0.5 %36 of the range and a maximum ranging distance of 60 m were achieved with the radar. The CLDS showed the lowest beam divergence, which is required to avoid contamination of the signal from 37 38 interfering surroundings for narrow fields of view. With the GNSS system delivering a relative 39 vertical accuracy better than 3-5 cm, water level can be retrieved with an overall accuracy better 40 than 5-7 cm.

41

42 Keywords: UAV; water level; radar; sonar; laser; GPS;

#### 43 **1. Introduction**

44 Extreme hydro-climatic events such as droughts, floods and heavy precipitation have increased 45 the awareness that knowledge of spatial and temporal variation of open water surfaces is 46 important (Alsdorf et al., 2007). In order to achieve a better quantitative understanding of 47 hydrologic processes and to increase sharpness and reliability of hydrologic predictions, 48 observations of hydrological variables, such as surface water area, water level (h), its slope 49  $(\partial h/\partial x)$  and its temporal change  $(\partial h/\partial t)$  are required. However, ground-based measurements of 50 terrestrial water bodies are limited to networks of measuring stations. In-situ stations provide 51 point observations that are often spaced too far apart to capture spatial patterns. Often, in-situ 52 observation technology fails during extreme events. Furthermore, globally, the availability of in-53 situ hydrologic observation stations has been declining in the recent past (Lawford et al., 2013). 54 Hence, remote sensing datasets have become increasingly popular in hydrology. Remote sensing 55 techniques are presently unable to observe river discharge directly, however spatial and temporal 56 variation of water level has been routinely observed using spaceborne or airborne platforms. 57 Although most satellite altimetry missions were not designed primarily for monitoring 58 continental waters, water levels of continental water surfaces retrieved by Seasat, 59 TOPEX/Poseidon, Jason-1 and 2, GFO, ERS 1 and 2, ENVISAT have a measurement accuracy 60 that is well understood and generally on the order of a few tens of centimeters (Calmant et al., 61 2008). This accuracy can be improved for larger lakes and rivers by averaging over large water 62 surfaces (Birkett, 1998; Birkett et al., 2002; Frappart et al., 2006). The satellite CryoSat-2 carries 63 a Synthetic Aperture Interferometric Radar Altimeter (SIRAL) which is a new generation radar 64 altimeter (Wingham et al., 2006) with a spatial resolution of around 300 m (Villadsen et al.,

65 2015). When operating in SARIn mode, a correction of the cross-track slope can be performed 66 and waveform analysis allows separation between water and surrounding topography 67 (Kleinherenbrink et al., 2014) resulting in an accuracy of the retrieved water level of just a few 68 decimeters (Kleinherenbrink et al., 2015). Spaceborne LIDARs such as the Geoscience Laser 69 Altimeter System (GLAS) have been shown to provide water level measurements with higher 70 accuracy than radar altimeters such as TOPEX/Poseidon (Zhang and Xie, 2010). Still, GLAS 71 has a ground footprint that is around 65 m (Schutz et al., 2005) and retrieves observations at 72 irregular temporal intervals. Therefore, the main limitations of conventional satellite radar and 73 laser altimetry are low spatial resolution, local coverage (for short repeat orbit missions) and low 74 temporal resolution (for long repeat missions such as CryoSat). In order to overcome these 75 limitations, the forthcoming Surface Water and Ocean Topography (SWOT) satellite mission 76 will build on the heritage of the imaging interferometric radars such as the Shuttle Radar 77 Topography Mission (SRTM) (Kiel et al., 2006; LeFavour and Alsdorf, 2005; Rodriguez et al., 78 2006). However, spaceborne sensors will always face problems of: i) large ground footprints, 79 which result in relatively low spatial resolution; ii) fixed orbit configurations, which may be 80 inappropriate for high-resolution coverage of local water bodies; iii) coarse temporal resolution 81 and/or the non-regular revisit intervals. These limitations restrict their ability to measure the 82 temporal and spatial variation of the water level with the accuracy needed for determining the 83 hydraulics of complex rivers and flood waves. 84 Airborne LIDAR techniques have the advantages of better tracking of terrestrial water bodies,

improved spatial resolution, clear segmentation between land and water surfaces and a higher
accuracy (Schumann et al., 2008). However, airborne LIDAR surveys are expensive and their

87 success depends on surveying conditions (e.g. topography and geometry, vegetation cover, size

of the water body). For this reason, digital elevation models and digital surface models retrieved
by airborne LIDAR are not universally available and are normally not retrieved during periods of
hydrological interest such as flood events.

91 UAVs (Unmanned Aerial Vehicles) and in particular micro-UAVs (payload less than 1.5-2 kg), 92 represent the latest frontier in land and water monitoring because of low-altitude flight, low cost 93 and flexible payload design (Anderson and Gaston, 2013). In recent years, miniaturized 94 components (GNSS receivers, inertial measurement units, autopilots) have advanced (Watts et 95 al., 2012), and UAVs have been used also for a wide range of hydrological applications such as 96 fluvial monitoring; river bathymetry and photogrammetric DEM generation using very high 97 resolution (VHR) imagery (Lejot et al., 2007); water velocity measurements using large-scale 98 particle image velocimetry (LSPIV) (Detert and Weitbrecht, 2015; Tauro et al., 2016, 2015). 99 Moreover, UAVs have attracted great interest for monitoring of environmental disasters and 100 floods (Luo et al., 2015). UAVs are low-cost platforms that have unique capabilities to access 101 hostile or inaccessible environments that need to be urgently monitored. Moreover, they ensure 102 tracking of water surfaces better than satellite technology. However, for LIDAR and SAR 103 systems, the tradeoff between performance, cost and size/weight is still a challenge to be solved 104 before their application in UAV remote sensing (Colomina and Molina, 2014). 105 In this paper, we demonstrate the possibility to acquire measurements of water level by a ranging 106 system that includes a ranging sensor (radar, CLDS or sonar) and a GNSS receiver. The ranging 107 technology described in this paper provides water level measurements with higher accuracy than 108 spaceborne or airborne altimetry. Moreover, it ensures a spatial resolution ideal for measuring 109 the two dimensional spatial variability of small rivers and their interaction with floodplains (Lee 110 et al., 2011). Lastly, the newly developed CLDS can acquire ranges to water surfaces when only

111	narrow fields of view are available. The CLDS is specifically developed for applications in
112	vegetated environments or inside sinkholes in karst environments.
113	2. Materials and Methods
114	2.1. General concept
115	To acquire accurate water level (height above mean sea level) of open water surfaces, the UAV
116	must be equipped with: i) accurate lightweight sensors for measuring the range to water surface
117	ii) a high accuracy dual frequency GNSS receiver and antenna. Installation of an in-situ dual
118	frequency GNSS master station is needed for differential corrections. The general concept is
119	illustrated in Fig. 1.
120	
121	
122	<b>Fig. 1</b> .
123	

124 The ellipsoidal height of the water surface is measured by subtracting the range measured by a 125 ranging sensor from the vertical position retrieved by the onboard GNSS receiver. Afterwards 126 the orthometric height can be retrieved from the ellipsoidal height if the geoid height is known 127 (Featherstone, 2001). For the purpose of this work, a hexacopter has been assembled from 128 TAROT-RC components and has been equipped with DJI Naza-M2 flight controller. The 129 hexacopter is able to fly at least 12 minutes carrying a payload of at most 2 kg. The choice of the 130 ranging sensors was constrained by: i) maximum weight of the payload, ii) a reasonable price 131 necessary for flexible operations, iii) sensor interfaces that allow time synchronization with the 132 GNSS receiver through a microprocessor. The selected ranging sensors included two off-the-

133	shelf sensors (a radar, a sonar) and the in-house developed CLDS. The total cost of the platform
134	is ca. 7000 euros. This cost includes the drone, the onboard GNSS system, the inertial
135	measurement unit (IMU), the three tested sensors and the microprocessor unit.
136	Fig. 2 shows the arrangement of the drone payload.
137	
138	Fig 2
130	115.2
140	
141	
142	
143	
144	2.1.1. Radar ranger
145	The radar is the ARS 30X model developed by Continental as anti-collision system for the
146	automotive industry (market price: 3200 EUR). It weighs around 350 g and consists of a 77 GHz
147	radar sensor with a mechanical scanning antenna. It measures the range to targets using FMCW
148	(Frequency Modulated Continuous Wave) with a sampling frequency of 15 measurements per
149	second. It provides up to 32 targets in near range and up to 64 targets in far range with a
150	resolution of 0.10 m. Each individual target angle is provided with a resolution of $0.1^{\circ}$ .
151	212 Sonar ranger
101	
152	The sonar is the MB7386 model from MaxBotix (market price: 150 EUR). It weighs around 50g
153	and consists of a 42 kHz ultrasonic sensor (6 Hz reading rate) with internal temperature

154 compensation, noise tolerance and clutter rejection. Its maximum ranging capability is up to 10155 m.

156

### 2.1.3. Camera-based laser distance sensor (CLDS)

157 This ranging sensor is a laser camera-based solution recently developed at Technical University 158 of Denmark (Reyna Gutierrez, 2013). It weighs around 350 g. It was inspired by the measuring 159 procedure proposed by Danko (2004). The range distance to the target is estimated by measuring 160 the angle at which laser light enters the camera. The original methodology is expanded in this 161 work to include corrections for tilting and rotation angles of the aircraft. An efficient automatic 162 algorithm for identifying the laser dots on the water surface was developed. Our prototype 163 consists of two laser pointers (100 mW laser diodes) and a complementary metal-oxide-164 semiconductor (CMOS) camera. The camera resolution is 20.2 megapixels. The camera is 165 triggered by the on-board single board computer (SBC) with an image rate of 1 frame every 2.5 166 seconds. The total manufacturing cost of this CLDS system is around 800 EUR. The current 167 design of the distance-meter includes a digital camera mounted at the center between the two 168 laser pointers. Fig. 3 shows the geometrical configuration of the camera. Range to water surface 169 is measured by illuminating the water surface with the laser pointers and taking a picture of the 170 illuminated water surface. When light emitted by laser pointers hits the water surface, bright dots 171 are formed at the interface between water and air. Due to scattering processes (in particular 172 Rayleigh and Mie scattering), some portion of the radiation is reflected in the direction of the 173 camera and an estimation of the range to water surface is possible.

174

175

Fig. 3.

176 The angle  $\alpha$  is a design parameter. The CLDS was built with  $\alpha = 90^{\circ}$  to simplify the measuring

177 concept and the derivation of the formulas. The CLDS shown in Fig. *3* is exactly symmetrical.

178 Indeed, only one laser would be sufficient to acquire the range to the surface; nevertheless, two

179 laser pointers improve error assessment and system accuracy.

180 The value of the measured range Hm can be computed by measuring the angle  $\theta'$  at which light

181 enters the camera, i.e. from equation (1).

$$Hm = \frac{A}{\tan \theta'} \tag{1}$$

182

183 Alternatively, the measured range Hm can be obtained through equation (2)

$$Hm = \frac{A \cdot f}{ImD} \tag{2}$$

184 Where ImD (Image distance) is the distance between the center of the image and the recorded

185 light source. A calibration procedure is needed to convert from the number of pixels from the

186 center of the image (PFC) to ImD as shown in equation (3)

$$ImD = PFC \cdot d_{pp1} + d_0 \tag{3}$$

187 Where d<sub>pp1</sub> and d<sub>0</sub> are the coefficients of the first-order polynomial producing the best least-

188 squares fit to the data. Equations (2) and (3) can be applied only when the focal length (f) of the

189 camera is exactly and the focus is constantly set to infinity. Otherwise, the calibration procedure

190 needs to estimate the angle  $\theta'$  directly from the number of pixels (PFC) as shown in equation (4).

$$\theta' = PFC \cdot r_{pp1} + r_0 \tag{4}$$

191 Where  $r_{pp1}$  and  $r_0$  are the coefficients of the first-order polynomial producing the best least-192 squares fit to the data. The calibration procedure, which has to be performed to estimate the  $r_{pp1}$ 193 and  $r_0$  coefficients, is presented in the appendix. The calibration procedure allows estimation of 194 the angle  $\theta'$  by measuring PFC, without having to consider the linear or nonlinear intrinsic 195 camera parameters, such as focal length and lens distortion.

196 Onboard the UAV, tilting and rotation cause a displacement of the light sources from their

197 equilibrium position. The changes in the geometrical relationships generate an error in the

198 estimation of the true range distance (hereafter defined as Ht) between the sensor of the camera

and the water surface. Tilting is the angle between the plane on which the camera and laser are

200 located, i.e. the axis of the CLDS, and the horizontal plane (angle  $\beta$  as shown in Fig. 4). Rotation

201 occurs between the vertical line and the optical axis of the camera (angle  $\delta$  as shown in Fig. 5).

Fig. 4

Fig. 5

If tilting pushes the light source below the axis of the distance meter, formula (5) can be used toobtain the true range (Ht) between the camera and the water surface:

$$Ht = [(Hm + A \cdot \tan \beta) \cos \beta] \cdot \cos \delta$$
<sup>(5)</sup>

204 Conversely, if the tilting pushes the light source above the axis of the CLDS, formula (6) can be205 used:

$$Ht = [(Hm - A \cdot \tan \beta) \cos \beta] \cdot \cos \delta$$
<sup>(0)</sup>

 $( \cap )$ 

If pitch and roll angles are retrieved on board the UAV, the measured range can be corrected according to equation (5) and (6) (Reyna Gutierrez, 2013). If the angles are not retrieved on board, the resulting error on the range can be estimated as shown in Fig. 6. Numerous tests have been conducted in order to determine the best configuration of the CLDS in terms of: i) arm length A, ii) wavelengths of the two laser pointers, iii) optimal camera configuration parameters such as optical zoom and resolution.

- The arm length choice affects the measuring range function, as shown in Fig. 7.
  - Fig. 6

- Fig. 7 shows that the resolution of the measurements depends on the derivative of the range
- 214 function. Hence, a longer arm will result in higher resolution, especially for longer ranges.
- 215 Indeed, in Fig. 7, the smoothest curve is for an arm length of 0.6 m. However, the payload size of
- small UAVs is limited and thus a 30 cm arm was chosen for our tests. The wavelengths of the
- two laser pointers were chosen as 450 nm and 531 nm, because reflectivity of water is relatively

high at these wavelengths as a consequence of the optical proprieties of water as described inHale and Querry (1973).

220 When the laser light hits the water surface, a bright dot is formed at the point of contact. 221 However, additional bright spots might be visible due to reflection from the riverbed and due to 222 additional scattering processes caused by water waves. To identify the two dots formed by laser 223 reflection, an automatic identification algorithm was developed consisting of the following 224 computational steps: i) the RGB image is converted to Hue, Saturation and Value (HSV) image. 225 Quasi-circular shapes in the image are found through circular Hough transform (Yuen et al., 226 1990). In case there are multiple circles in the image, the two circles (one generated by the left 227 laser and one by the right laser) with the highest mean Value (V) are considered to be the contact 228 spots. Thereafter, ii) the brightest pixel (pixel with the highest Value) is identified inside each of 229 the two circles (laser dots). The brightest pixel typically lies in the center of the laser dot in case 230 of normal light incidence. Lastly, iii) the distance (PFC) between the center of the image and the 231 two identified brightest pixels is computed. Post-processing of the images is performed after the 232 flight and takes around 30 seconds per image.

#### 233 2.1.4. GNSS system

The differential GNSS system consists of two NovAtel receivers: one used as master station
(flexpack6) and one as rover (OEM628 board). A NovAtel GPS-703-GGG pinwheel triple
frequency and GLONASS antenna is used as base station and an antcom (3G0XX16A4-XT-1-4Cert) dual frequency GPS and GLONASS flight antenna is used as rover station on the UAV.
Raw pseudoranges and carrier phase measurements are stored at 5 Hz. The position solution is
post-processed using Leica Geomatic Office v 8.1 in kinematic mode. In post-processed mode, a
Kalman filter can be applied both in forward and backward direction for best position

performance. The length of the GPS baseline affects the vertical and horizontal accuracy of the
drone position. Position error is expected to increase by 1-3 ppm (1-3 mm additional error per
km of baseline).

244

245 2.1.5. Payload controller

Data acquired by the different sensors are saved on the SBC (BeagleBone Black) and a time
synchronization of the different sensors can be performed. Synchronization between the position
retrieved by the GNSS system and the range retrieved by the sensors is essential for accurate
water level observations, as described in Appendix B.

250

#### 251 *2.2. Testing of the sensors*

252 To test the accuracy of the system, both static (ground-based) and dynamic (airborne) tests were 253 performed. First, several tests were conducted from bridges of different heights over free-flowing 254 rivers in order to test accuracy, precision and maximum ranging capability. Beam divergence was tested by acquiring measurements inside a water well of small diameter. After the ground-255 256 based tests, numerous flight tests were conducted over a lake. Because the water level in the lake 257 can be assumed to be uniform in space, these flights allowed determination of the accuracy of the 258 full system, which consists of the GNSS receiver and the ranging sensors. Appendix B reports 259 the experimental settings of both static and airborne tests.

#### 261 2.2.1. Ground-based evaluation

Accuracy of the ranging sensors was estimated using as reference a water level dip meter, which has an accuracy better than 0.3% of the range. When tested in static mode, sensors acquired measurements for 30 seconds. Subsequently the average range  $(\bar{x})$  was computed as the weighted arithmetic mean as shown in equation (7) after outlier removal ( $\geq 5\sigma$ ).

$$\bar{x} = \frac{\sum_{i=1}^{N} f_i x_i}{\sum_{i=1}^{N} f_i}$$
(7)

266

In equation (7)  $x_i$  is an observation and  $f_i$  the frequency of that value. N is the total number of measurements which depends on the reading range of the individual sensor.

269 Precision is estimated as standard deviation ( $\sigma$ ) of the measuring stack, and is computed using 270 equation (8):

$$\sigma = \sqrt{\frac{\sum_{i=1}^{N} f_i \cdot (x_i - \bar{x})^2}{\sum_{i=1}^{N} f_i - 1}}$$
(8)

271

272 Maximum ranging capability is the maximum range from which the sensor can retrieve a

273 measurement with a reasonable accuracy (i.e. 5% of the range).

274 Beam divergence is defined as the measure (in angular units) of the increment in beam

275 diameter with distance from the optical aperture or antenna from which the sonic or

276 electromagnetic beam emerges. A larger beam divergence leads to a larger ground footprint of

the signal, which results in contamination of the signal if the surface is inhomogeneous. For the

278 CLDS this parameter is negligible, since its ground footprint directly depends on the arm length

A and the laser beam divergence is very low. Moreover, the CLDS provides images of each

280 individual acquisition and the user can perform a-posteriori supervision to control if the

281 measured target is indeed the water surface. For the radar and the sonar, beam divergence is a 282 critical parameter to ensure that water is measured without interference from the surroundings. 283 This parameter has to be considered in order to monitor water bodies (e.g. large sinkholes, rivers 284 surrounded by dense vegetation), which only expose a narrow stretch of water to aerial view. 285 Indeed, because of loss of GNSS signal, flights under vegetation canopy or inside small cavities 286 (e.g. karst sinkholes) cannot be performed without losing position accuracy. Beam divergence 287 was estimated by acquiring measurements over water wells of small diameter, while water was 288 gradually being pumped out, as described in Appendix B.

289

## 2.2.2. Airborne evaluation

Numerous flights were conducted above a 0.02 km<sup>2</sup> lake located near Holte, Denmark 290 291 (55.821720°N, 12.509067°E). Water level in the lake is practically uniform. Whilst the sonar 292 and the CLDS identify only one target in the field of view, the radar can identify multiple targets 293 and reports the target angle for each of those. This requires an accurate identification of the 294 target, which is representative of the water surface. Indeed, sometimes multiple targets are 295 retrieved at nadir angle, for instance when vegetation is overhanging the water body. In that case, 296 postprocessing requires switching between different targets to obtain a result that is continuous 297 in time. Moreover, a low-pass digital filter was applied on the 15Hz raw radar data. A weighted 298 moving average (WMA) with a temporal window of 0.33 s (five observations) was applied to 299 smoothen the signal as shown in equation (9).

$$WMA_t = w_1A_{t-2} + w_2A_{t-1} + w_3A_t + w_4A_{t+1} + w_5A_{t+2}$$
(9)

301 Weights (w1, w2...w5) are normally set to a high value for the measurement taken at the actual

302 time (At) and to lower values for the previous and subsequent measurements.

303 .The overall accuracy of the system consisting of the GNSS receiver and the ranging sensor

304  $(\sigma_{tot})$  is assumed to be that of two independent normally distributed variables: the ranging sensor

accuracy and the GNSS accuracy (10).

$$\sigma_{tot} = \sqrt{\sigma_s^2 + \sigma_{RTK}^2} \tag{10}$$

306

307 where  $\sigma_s$  is the accuracy of the ranging sensor and  $\sigma_{RTK}$  is the accuracy of the GNSS receiver.

### 308 **3. Results**

The first section of the results describes the technical performance of the ranging sensors when tested from a static position on the ground. Results are based on numerous tests conducted from bridges of different heights to compare the technical performance of the different sensors. The second section describes the results of the flight tests that are intended to evaluate the accuracy of the integrated system, i.e. GNSS receiver and sensors operating on board the UAV.

#### 314 *3.1. Ground-based performance results*

Sensors demonstrated different performance in terms of accuracy and standard deviation of the measuring stack when tested from bridges of different heights. Appendix B lists the experimental settings for the static tests. Fig. 8 shows that the sonar usually tends to overestimate the range to water surface, which is probably caused by a slight penetration of the ultrasonic wave (42 kHz) below the water surface. Conversely, the radar usually tends to underestimate the range. The

320	authors guess that this is due to the post-processing of the raw data by the proprietary radar
321	firmware.
322	
323	
324	Fig. 8
325	
326	
327	Table 1 summarizes the sensors' technical performance in terms of accuracy, standard deviation
328	of the measurement stack, maximum ranging distance and beam divergence.
329	
330	Table 1
331	
551	
332	Table 1 confirms that the sonar is the best sensor in terms of accuracy and standard deviation of
333	the measurement stack. The CLDS has the lowest beam divergence. However, the radar is the
334	sensor that combines the longest ranging capability, with accuracy and standard deviation that
335	are only slightly worse than for the sonar. In Fig.9, two regression lines confirm the systematic
336	error of radar and sonar. Plotted as function of the range, the regression line of the radar absolute
337	error has a slope of -0.0090, while the slope of the sonar is 0.0083. After removal of this
338	systematic error, the radar shows an accuracy of 0.5 % of the range, whilst the accuracy of the
339	sonar is around 0.3%.
340	
341	

342 Fig.9

343

344

345 Finally, the accuracy of the retrieved vertical position has to be assessed. The accuracy of the 346 GNSS height depends mainly on: i) the integer ambiguity solution that has to be fixed to obtain 347 reliable observations, ii) the satellite geometry that affects the dilution of precision (DOP), iii) 348 multipath interference, especially because of signal reflection from the water surface. 349 350 3.2. Airborne performance results 351 In this section, we report the observations of two flights and we show a table summarizing the 352 entire dataset of flights over the lake. The range measured by each of the sensors and the altitude 353 retrieved by the GNSS are shown in Fig.10. The figure contains the entire dataset of observations 354 retrieved by the radar and sonar. Only not-a-number (NaN) values are removed. The sonar 355 outputs NaN when the range exceeds the maximum range capability (10 m). For the CLDS, we 356 only reported the measurements retrieved from images in which the laser dots are clearly 357 identifiable on the water surface. 358

359 Fig.10

360

361 Fig.10 shows an extremely high correlation (Pearson coefficient of 0.9991), between the GNSS 362 and the radar measurements, which indicates the consistency of our ranging technology. The 363 laser dots are generally distinguishable on the water surface only when the range to water surface 364 is less than 12-13 m. Similarly, the sonar provided accurate measurements only when the UAV

was hovering at low altitudes (less than 10 m from the water surface). Indeed, the radar andsonar curves only overlap during these flight maneuvers.

367 In Fig. 11 we display the water level measured by the different sensors. Outliers (> $2\sigma$ ) were 368 removed.

369

370 Fig. 11

371

372 Mode value, mean and standard deviation of water level retrieved by each of the sensors are 373 reported in Table 2 under the column with flight date "04/04". The dispersion in water level 374 measurements retrieved by the system consisting of the radar and the GNSS receiver may be due 375 to multipath errors on the GNSS receiver. The cut-off angle for the elevation of the satellites, 376 which defines the angle below which GPS satellites are excluded, turned out to be a sensitive parameter. The selected values for each flight are reported in Appendix B. 377 378 The water level values retrieved by the sonar had low accuracy, especially during high-speed 379 maneuvers. Since the range to water surface was greater than the maximum range capability of 380 the sonar for a significant portion of flight duration, the sonar retrieved many NaN values and 381 noisy observations. However, the mode value retrieved by the sonar is 24.14 m, which is close to 382 the mean value retrieved by the radar. 383 The CLDS exhibits only few observations due to limited range capability and low frame rate. 384 Moreover, natural light conditions complicate the recognition of the laser dots on the water 385 surface. 386 In order to estimate the absolute accuracy of the sensors, results were compared to in-situ

387 measurements of water level. For the in-situ measurement, an additional accurate RTK (Real

388	Time Kinematic) GNSS rover station was used, which was connected to a Danish GPS network
389	The position was averaged over a period of one minute which resulted in 24.10 m above the
390	DVR90 geoid model (with an estimated accuracy of the GNSS rover station of around 5-6 cm).
391	For this flight, the accuracy of the radar is thus better than 5 cm, the mode value of the sonar is
392	around 4 cm from the ground truth, while the mean value retrieved by the CLDS is within two
393	decimeters.

394 The second flight reported in Fig. 12 evaluated performance for higher drone altitude (up to 60395 m) above the water surface.

396

397 Fig. 12

398 As shown in Fig. 12, the radar and the GNSS show very high correlation for the entire flight.
399 The flight confirmed the limited ranging capability of the sonar (specified as 10 m, but already
400 very noisy beyond 9 m). The CLDS retrieved ranges up to 13 m, however standard deviations
401 increased significantly with range. In Fig. 13 we compare the water level retrieved by the three
402 different sensors for this flight.

403

404 Fig. 13

405

Statistics of the flight are shown in Table 2 under the column "27/05". In-situ water level was
24.01 m. Fig. 13 shows that the sonar measurements were unsuccessful. The CLDS, despite very
high standard deviations, shows a mean value that is very close to the ground truth. The radar
shows higher dispersion for long ranges. Moreover, systematic error is still observable, in fact

410	when the drone is at higher altitude, the retrieved water level increases by a few cm. System
411	performance was confirmed in a number of other flights, as shown in Table 2 . Experimental
412	settings, such as flight speed, illumination conditions, sensor settings for each flight, are
413	explained in appendix B.
414	
415	
416	Table 2
417	
418	Table 2 clearly indicates that the radar is the most reliable sensor, with the lowest standard
419	deviation and good agreement with in-situ measurements. However, during some of the flights,
420	the measured water level exhibits significant standard deviation also for the radar. This
421	dispersion of the water level observations is caused not only by ranging errors but also by the
422	GNSS. Indeed, during some flights, the geometrical configuration of GNSS satellites may have
423	been suboptimal for accurate positioning. In addition to this, multipath of the GNSS signal may
424	occur and degrade the accuracy of water level observations to ca. 7 cm.
425	The sonar provides very noisy measurements and exhibits a skewed distribution with a fat tail
426	around 10 m, which is the maximum range of the sensor. While the mean value of water level
427	does not provide an accurate estimate, the mode values measured by the sonar are very similar to
428	the corresponding values measured by the radar.
429	For the CLDS, the mode value is not relevant because the number of observations is low. The
430	CLDS standard deviation is quite large and in order to obtain accurate results the drone has to
431	hover for several seconds.

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433	The ranging technology showed great potential in terms of accuracy, maximum range and beam
434	divergence. In particular, the radar demonstrated the best performance in terms of accuracy and
435	maximum range. The ranging sensor has to be integrated with carrier phase differential GNSS to
436	retrieve water level. The accuracy of the integrated system consisting of GNSS receiver and
437	radar is estimated to be better than 5-7 cm. This accuracy can be compared with the accuracy
438	achievable with: i) airborne LIDARs, ii) spaceborne laser altimetry, iii) spaceborne radar
439	altimetry and iv) ground-based stations as shown in Table 3.
440	
441	
442	
443	Table 3
444	
445	
446	Few studies report the accuracy of LIDAR system in measuring water surface, but it has been
447	estimated to vary from few cm up to two tens of cm (Hopkinson et al., 2011). For airborne
448	LIDAR systems, the inaccuracy of the onboard positioning systems has to be included. Similarly
449	spaceborne laser altimetry from ICESat, which is the satellite altimeter with the smallest
450	footprint (50-90 m) and the highest along-track resolution (40 Hz, 170 m), provides water
451	surface elevation measurements for rivers with an accuracy at decimeter level. However the
452	accuracy degrades in case of cloud cover (Phan et al., 2012). Additionally, simultaneous return
453	from land and water are inevitable for small rivers and the identification of water surfaces
454	remains problematic. The accuracy of radar altimetry sensors such as the systems on board

455 Jason-2 (Asadzadeh Jarihani et al., 2013), Envisat (Frappart et al., 2006) and Cryosat-2 (Song et 456 al., 2015) is in the order of some tens of dm. Moreover, satellite radar altimetry generally has a 457 spatial resolution lower than satellite laser altimetry and requires that rivers are hundreds of 458 meters wide to avoid signal contamination by interfering land and vegetation (Maillard et al., 459 2015). With UAV-borne monitoring, water surface and interfering surroundings can be clearly 460 separated due to the smaller ground footprint, and the possibility to retrieve individual radar 461 target angles. However, for very narrow fields of view, the CLDS is the only sensor that can 462 provide reliable water level measurements. Image analysis as part of the post-processing 463 workflow ensures that measurement are accepted only if the monitored target is the water 464 surface. This is the case for rivers surrounded by dense riparian vegetation or for small targets 465 such as karst sinkholes, e.g. on the Yucatán Peninsula (Gondwe et al., 2010). Our CLDS solution 466 overcomes the limitations of traditional red wavelength time-of-flight (TOF) laser distance 467 meters, which are not suitable for ranging to water surfaces, because the reflectivity of water is 468 very low for red visible wavelengths.

469 Only ground-based hydrometric stations ensure an accuracy higher than the one achieved with 470 UAV-based monitoring, but coverage and reliability of in-situ monitoring networks have been 471 degrading in many regions of the world. Moreover, despite providing high accuracy and 472 temporal resolution, in-situ stations acquire only local measurements and tend to fail during 473 extreme events. Therefore, UAV-based water level monitoring is beneficial for the monitoring of 474 a wide range of hydrological systems, including small-scale rivers, ephemeral lakes, sinkholes, 475 meltwater lakes, etc... UAV-based water level observations can resolve the spatial 476 multidimensional variability of rivers. Indeed, UAVs can monitor water level along and across 477 the river course, in order to obtain water slope and assess interaction between rivers and adjacent

478 floodplains. Improved sharpness and reliability of estimates of surface water-groundwater 479 interaction using UAV-based monitoring of river water levels have already been reported 480 (Bandini et al., 2016). Furthermore, UAVs can sense water level in unconventional remote 481 sensing targets such as sinkholes or cenotes. This could potentially improve mapping of phreatic 482 surfaces, for instance for the Yucatan peninsula (Bauer-Gottwein et al., 2011). Additionally, 483 UAVs can potentially be used during extreme events when in-situ monitoring stations often fail 484 and satellite observations do not ensure the required spatial and temporal resolution. Thus, UAVs 485 have the potential to improve flood risk assessment. However, the  $\pm 7$ cm accuracy of our 486 technology may still be insufficient for rivers flowing through low-lying terrain. Nonetheless, the 487 accuracy is better than other spaceborne and airborne technologies and UAVs have a great 488 potential in improving flood mapping because they allow optimal timing of the observations and 489 high spatial resolution. UAV-based observations of water level in the flooded areas allow 490 determination of stage-damage curves (Cammerer et al., 2013) which are essential for the design 491 of insurance policies.

## 492 **5.** Conclusions

UAV-based remote sensing of river and lake water level (orthometric height) has the potential to
fill the gap between in-situ measurements and spaceborne remote sensing. It ensures: i) high
accuracy, ii) optimal spatial resolution, iii) flexible timing of the sampling, and iv) precise
tracking of lakes and rivers. Different water surface ranging sensors were tested: a radar, a sonar,
and a CLDS.

498 Static (on ground) and dynamic (airborne) tests demonstrated the following results:

499	• The radar showed the best accuracy and longest maximum range. Despite having a
500	resolution of only 10 cm, averaging the 15 Hz primary data, an accuracy of 0.5% of the
501	range can be achieved after correction of a negative bias of 0.9% of the range.
502	• The sonar provided unreliable results for high ranges or high speeds. Our results show
503	that the sonar generally overestimates the range to water surface. However, when the
504	UAV flies at a stable and low height, the accuracy is down to a few centimeters.
505	• The CLDS is less accurate than the radar. However, it has the lowest beam divergence
506	and is useful when only a narrow field of view to the water surface is available for
507	sensing.
508	Water level can be measured on board UAVs by subtracting the range to water surface from
509	the vertical position retrieved by the GNSS receiver. Dynamic (airborne) tests have been
510	performed on the positioning technology and the GNSS receiver had a vertical accuracy
511	around 4-6 cm (2 $\sigma$ ) and had an expected horizontal accuracy around 2 cm (2 $\sigma$ ). However,
512	multipath of the GNSS signal causes problems above water and the choice of the cut-off
513	satellite elevation angle has a considerable influence on the position accuracy.
514	The integrated system GNSS receiver and radar is able to measure water level with an overall
515	accuracy better than 5-7 cm when the UAV flies at a speed of few km/h.
516	Future research should include different types of sonar sensors, trading off signal penetration
517	below the water surface (more penetration at lower frequencies) and interference of the propeller
518	noise (more interference at higher frequencies). Moreover, research efforts are ongoing to
519	develop new radars with higher measurement resolution, exploiting other region of the
520	microwave spectrum commonly used in radar altimetry such as Ku and Ka bands.
521	

523

- 524 The master thesis from Reyna Gutierrez, J. A. (2013) "Monitoring and modeling of regional
- 525 groundwater flow on the Yucatán Peninsula'' can be obtained from the authors upon request.

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643	
644	Appendix A. Calibration of the CLDS

- 645 The CLDS needs to be calibrated in order to provide a ranging measurement. Calibration has
- been performed acquiring multiple range measurements (from 0 to 12 m) using a black vertical
- 647 wall as calibration target. Since the focal length of the camera is not exactly known, equation (4)
- must be used and the calibration is used to retrieve the coefficients  $r_{pp1}$  and  $r_0$  for converting from
- 649 pixel units to angular units. The relationships between  $\theta'$  and the distance from the laser dots to
- 650 the center of the image (PFC) are shown in Fig. A.1 for each of the laser pointers. Alternatively,
- Fig. A.2 depicts the relationship between the range to the target and PFC.

Fig. A.2

652

653 Fig. A.1 and Fig. A.2 show that the laser pointers' curves are not coincident as a consequence of 654 the slight asymmetry of the layout (imaging sensor of the camera not placed exactly in the 655 middle of the two laser pointers). As confirmed by Fig. A.1, the relationship between PFC and 656 the measured angle is approximately linear for each of the two laser pointers. Calibration has 657 shown an r (Pearson linear correlation coefficient of determination) of 0.99978 and an RMSE 658 (Root Mean Square Error) of 7.16 cm for the blue laser (left laser); an r of 0.99937 and an RMSE 659 of 8.29 cm for the green laser (right laser). Calibration error is displayed in Fig. A.3. 660 Fig. A.3 661 Fig. A.4 662 663

Fig. A.3 demonstrates that the advantage of using two laser pointers is improved error

- assessment. Considering the average of the measurements of the two laser pointers, calibration
  - 30

666 RMSE is reduced to 5.61 cm. When range to water surface has to be retrieved, the precise 667 computation of PFC is more problematic than during the simple calibration procedure. Indeed, 668 while laser dots can be normally identified as in Fig. A.4 (a), laser dots on the water surface 669 might have contours that are less defined as in Fig. A.4 (b). Sometimes even multiple laser dots 670 are visible, as shown in Fig. A.4 (c). This is caused by: i) atmospheric scattering processes, ii) 671 scattering processes due to water waviness iii) vibrations of the UAV. The laser light reflected 672 from the bottom is occasionally visible in the image, especially in case of shallow or very clear 673 water, as shown in Fig. A.4 (d). Experiments showed that the uncertainty in the PFC increases 674 with the range to water surface. This is displayed in Fig. A.5 with the curve PFC- $\sigma_{PFC}$ . Fig. A.5 675 clearly shows that the green laser exhibits larger uncertainty than the blue laser since green 676 wavelengths are scattered to a greater extent than blue wavelengths. The expected uncertainty in 677 the range can be estimated using the derivative of the range function as shown in equation (A.1).

$$\sigma(range) = \frac{\partial range}{\partial (PFC)} \sigma(PFC)$$
(A.1)

Fig. A.6

678	Fig. A.6 shows that the uncertainty of the range estimate increases with the range to water
679	surface. This is a consequence of: i) the derivative of the curve in Fig. A.2 that increases in
680	absolute value for longer ranges (small inaccuracy in PFC determines high imprecision in the
681	range observation). The derivative is lower in absolute terms for the blue laser, because of the
682	prototype layout. ii) Increasing uncertainty of the PFC with increasing range (i.e. decreasing
683	PFC).
684	
685	
686	Appendix B. Experimental settings
687	

In Table B.1 we report the location, the date and time of the day, the environmental conditions
and the water flow speed for each of the static tests. The mean value and the standard deviation
of the measurements are shown in Figure 8.

692 Here Table B.1

693 Illumination conditions are reported in the table because they affect visibility of the laser dots on 694 the water surface. This factor has been critical only in case of sun glint conditions during which 695 laser dots are hardly identifiable. On the other hand, wind stress and current can affect water 696 surface roughness and change the intensity of the backscattered radar signal.

Estimates of beam divergence for the different sensors were obtained from tests above a cylindrical water well of diameter (D) equal to 0.7 m. The sensors were placed exactly in the middle of the water well as shown in Fig. B.1. The initial range between the sensors and the water surface was 0.5 m. Subsequently, the well was pumped to gradually increase the range to the water surface. Beam divergence ( $\varphi$ ) was then computed according to equation B.1.

$$\varphi = 2 * \tan^{-1} \frac{D}{2 \cdot r_c} \tag{B.1}$$

In equation B.1, r<sub>c</sub> is the critical range i.e. the range at which the sensor first produced erroneous
 results because of interference with the well walls. Fig. B.1 provides an illustration of the
 experimental setup.

705

706 Fig. B.1

707

708	While the CLDS was able to retrieve the range to water surface for all water levels (beam width
709	is constant and equal to the arm length), the beam divergences of the radar and the sonar were
710	estimated using this method. For the radar, interferences started to occur at a range of 1.3 m, and
711	for the sonar at 1 m. Equation B.1, then gives beam divergence of the radar as ca. $30^{\circ}$ and beam
712	divergence for the sonar as ca. 40°.
713	Table B.2 shows the flight records for the tests conducted over the lake to estimate the airborne
714	accuracy of the system.
715	
716	Table B.2
717	
718	
719	As Table B.2 shows, the GNSS satellite cut-off angle settings are different between the flights.
720	The cut-off angle showed an influence on the position accuracy, and thus on the water level
721	measurements, up to 1-2 cm. Larger cut-off angles reduce the number of satellites in the field of
722	view of the GNSS antenna, while smaller cut-off angles might increase multi-path effects (e.g.
723	GNSS signal reflected by the water surface).
724	Average and vertical speed was varied between the different flights to test the synchronization
725	between the GNSS system and the different sensors. Indeed, since water level is constant in the
726	lake, when the drone rapidly changes its altitude, equivalent variations should be recorded by the
727	ranging sensors and the GNSS system. Synchronization between the radar, sonar and the GNSS

was obtained at the 30 ms level, while synchronization with the CLDS was obtained at the 0.2 s





Fig. 1. Illustration of measurement principle for retrieving water level. The system includes: i) the UAV, ii) the
sensors to measure the range from the UAV to the water surface, iii) a GNSS receiver on board the UAV providing
accurate vertical and horizontal position. Centimeter-level position accuracy is obtained through the installation of
an in-situ GNSS master station providing corrections for a kinematic post-processed solution.



Fig. 2. Picture of the drone payload. It includes the three tested sensors (CLDS, radar and sonar), the GNSS system
 (antenna and receiver), the IMU, the Single Board Computer (SBC) and the power convertion units (DC/DC
 converters).



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Fig. 3. Geometric configuration of the CLDS solution. A is the distance between the center of the camera and each of the laser pointers.  $\alpha$  is the angle between each of the lasers and the focal plane of the camera. Hm is the distance between the camera and the water surface. ImD is the distance between the center of the image focal plane and each of the recorded laser light dots. f is the focal length of the camera.  $\theta'$  is the reflection angle.  $\theta$  is its angle between the axis of the CLDS and the reflected ray.  $\gamma$  is the angle between incident and reflected ray. If  $\alpha$  is 90° (as in the figure),  $\gamma$  is equal to  $\theta'$ .







Fig. 8. Absolute error as a function of the range measured by each of the ranging sensors. Absolute error is
computed using the water level dip meter as reference. The marker is the average error (bias) of all measurements
taken for a specific range, while the bar shows the standard deviation.



Fig. 9. Sonar and radar errors as a function of the range. Dots represent the measurements acquired by the radar and the sonar. The regression line shows that the absolute error is a function of the range.



Fig. 10. Observations retrieved during the flight on April 4, 2016. The plot shows the range measured by the radar
 (blue), sonar (red), CLDS (green) in meter (m) to the water surface, and the drone altitude retrieved by the GNSS
 (black) in meter above mean sea level (mamsl).



Fig. 11. Water level (mamsl) observations retrieved during the flight on April 4, 2016. Each of plots shows the water
 level observations measured by subtracting the range retrieved by each of the sensors (radar, sonar, CLDS) from the
 GNSS altitude. In each plot, the black line is the mean of the water level observations and the magenta line is the
 mode of those observations.



Fig. 12. Observations retrieved during the flight on May 27, 2016. The plot shows the range measured by the radar
(blue), sonar (red), CLDS (green) in meter (m) to the water surface, and the drone altitude retrieved by the GNSS
(black) in meter above mean sea level (mamsl).



Fig. 13. Water level (mamsl) observations retrieved during the flight on May 27, 2016. Each of plots shows the
water level observations measured by subtracting the range retrieved by each of the sensors (radar, sonar, CLDS)
from the GNSS-derived altitude. In each plot, the black line is the mean of the water level observations and the
magenta line is the mode of those observations.





Fig. A.3. Calibration error for left laser (blue column), right laser (green column) and for the average (red column)
 between the two laser pointers.



Fig. A.4. Airborne image of water surface taken by the CLDS. (a) the two laser dots are clearly identifiable (b)
larger laser dots with contours that are less identifiable (c) multiple green laser dots caused by multiple reflection
and scattering processes (d) laser light is reflected by the bottom (larger dots) and by the surface (smaller dots)





Fig. A.5. Uncertainty ( $\sigma_{PFC}$ ) in computing the number of pixels as a function of PFC, for green and blue laser.







865 Fig. B.1. Schematic representation of the test conducted over the water well to retrieve beam divergence ( $\phi$ ) for each of the sensors. D is the diameter of the water well,  $r_c$  is the critical range.

# **Tables**

- 871 Table 1. Technical performance of the sensors and of the GNSS receiver when tested in static mode.

	mean absolute	standard	Maximum ranging	Beam divergence
	error (percentage	deviation of the	distance	
	of the range)	stack		
Radar	-1.09%	0.064 m	60 m near field	~30°
			200 m far field	
Sonar	0.98%	0.007 m	10 m	>40°
CLDS	1.5%	2.3 % of the range	13 m	negligible
GNSS receiver	negligible	Vertical		

#### coordinates : 4-6

#### cm at 2 sigma

872
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- 875 Table 2. Summary of the test flights over the lake. Each flight is named with the date (corresponding year is 2016). Ground truth
- 876 was measured with a RTK GNSS rover station connected to the network of reference stations. Statistics concern the water level
- 877 observations measured by subtracting the GNSS flight altitude from the range to water surface measured by each of the sensors.
- 878 Statistics are computed after removal of the observations that lie beyond  $2\sigma$ .

#### Flight date (dd/mm/2016)

Flight statistics						
		17/03	04/04	13/04	05/13	27/05
Ground truth (mamsl)		missing	24.10±0.0 6	24.13±0.0 6	24.04±0.0 6	24.01±0.0 6
Mean value (mamsl) of water level retrieved by	radar sonar	24.10 23.50	24.11 23.93	24.20 20.01	24.11 27.05	24.02 38.45
	CLDS	missing	24.29	24.81	24.82	23.93
	radar	24.18	24.13	24.10	24.12	24.00
Mode value (mamsl) of water	sonar	24.40	24.14	24.08	24.65	27.50
level retrieved by	CLDS	missing	21.27	24.56	24.41	20.66
	radar	0.07	0.05	0.08	0.09	0.05
Standard Deviation (m) in water	sonar	0.80	2.31	1.3	0.36	14.42
level retrieved by	CLDS	missing	1.08	0.95	1.68	2.05

889

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# 891 Table 3. Accuracy and ground footprint of different techniques for observing water level

Location	Technique	Footprint	Accuracy	Reference
Airborne	LIDARs	20 cm-1 m	4-22 cm	(Hopkinson et al.,
				2011)
Spaceborne	laser altimetry (e.g.	50–90 m	10 cm	(Phan et al., 2012)
	ICESat)			
Spaceborne	radar altimetry (e.g.	400 m-2 km	30-60 cm	(Frappart et al., 2006)
	ERS2, Envisat,			
	Topex/Poseidon)			
Ground-based	radar/sonar/pressure	negligible	1 mm-10 cm	Widely known
	transducers			metrology
UAV-borne	radar altimetry	negligible	5-7 cm	Methodology described
				in this paper

#### 892

Table B.1. Locations, settings and environmental conditions during static (on ground) tests. Coordinates are in WGS84. Country
is either Denmark (DK) or Italy (IT). Range (m) is the value measured by the water level dip meter. Water speed has qualitatively

is either Denmark (DK) or Italy (IT). Range (m) is the value measured by the water level dip meter. Water speed has qualitatively
been classified into no speed (still water), low (less than 0.4 m/s), medium (between 0.4 and 1 m/s), and high speed (more than 1

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and high wind speed (more than 8 m/s). Illumination has been qualitatively classified into artificial lightening, low (less than 20 000 lux), medium (between 20 000 and 50 000 lux), and high illumination (more than 50 000 lux)

Latitude	Longitude	River	Cou ntry	Range to water	Date (dd/ mm/ 2015)	Time of the day (hh:mm)	Flow speed	Wind	Illumination
55.783431	12.515610	Laboratory	DK	0.63	2/11	11:20	no	no	artificial
55.775211	12.470266	Mølleåen	DK	1.60	20/11	13:30	low	medium	low
55.775211	12.470266	Mølleåen	DK	2.38	20/11	11:41	low	medium	low
55.775211	12.470266	Mølleåen	DK	2.58	20/11	11:52	low	medium	low
55.775211	12.470266	Mølleåen	DK	2.65	10/10	11:20	low	high	low
55.775211	12.470266	Mølleåen	DK	2.98	1/10	14:10	low	medium	low
55.775211	12.470266	Mølleåen	DK	3.10	10/10	11:25	low	high	low
55.775211	12.470266	Mølleåen	DK	3.49	10/10	11:49	low	high	low
44.909645	10.991254	Sabbioncello	IT	3.92	22/12	16:00	low	low	low
55.775211	12.470266	Mølleåen	DK	4.20	10/10	14:10	low	high	medium
55.775211	12.470266	Mølleåen	DK	4.35	1/10	14:33	low	medium	low
45.038994	10.965141	Canale Bonifica	IT	5.32	22/12	13:00	low	low	medium

		Parmigiana,							
45.029723	10.959166	Canale della Bonifica Reggiana Montovana	IT	7.10	22/12	14:05	low	low	low
45.029726	10.960432	Canale della Bonifica Parmigiana	IT	7.33	22/12	9:30	low	low	low
44.650573	10.794755	Secchia	IT	9.79	29/10	12:00	medium	medium	medium
44.821261	10.994579	Secchia	IT	11.16	29/10	12:50	medium	medium	high
44.67578	10.860146	Secchia	IT	12.20	29/10	13:50	medium	medium	medium
45.008365	10.977453	Secchia	IT	12.72	29/10	20:30	medium	medium	low
44.727259	11.045292	Panaro	IT	12.97	29/10	8:30	medium	low	low

# 900 Table B.2. Summary of the test flights over the lake.

	Flight date (dd/mm/2016)							
Flight statistics	17/03	04/04	13/04	05/13	27/05			
Take-off time (hh:mm)	15:00	12:20	13:20	13:00	12:00			
Flight time over water (s)	500	270	200	250	260			
Minimum-Maximum flight height (meter above water surface)	3-28	4-18.5	5-60	8-48	9-58			
Average horizontal speed (m/s)	2	3	4	1	2			
Average vertical speed (m/s)	0.1	1.1	1	0.3	0.3			
Maximum vertical speed (m/s)	1	2	3	1.5	1			
GPS cut-off angle (degree)	10	13	14	15	15			