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1 **Experimental Study of the Energy and Exergy performance** 2 **for a Pressurized Volumetric Solar Receiver**

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17 **Abstract**

18 This article presents an experimental investigation of the heat transfer
19 characteristics as well as energy and exergy performance for a pressurised volumetric
20 solar receiver under variable mass flow rate conditions. During a two hour period of
21 continuous operation in the morning, the solar irradiance is relatively stable and
22 maintained at approximately 600 W/m^2 , which is beneficial for analyzing the energy
23 and exergy performance of the solar receiver. Experimental results show that the mass
24 flow rate fluctuation has slight effect on the solar receiver outlet temperature.
25 Whereas the mass flow rate plays an important role in the solar receiver power,
26 energy efficiency and exergy efficiency. The efficiency of the solar receiver is
27 normally above 55% with the highest efficiency of 87%, and at steady state, the
28 efficiency is maintained at around 60%. A very low value of the heat loss factor
29 (0.014 kW/K) could be achieved during the current steady state operating conditions.
30 The highest exergy efficiency is approximately 36%. In addition, as the temperature

31 difference increases, the impact of the exergy factor increases. The highest exergy
 32 factor is 0.41 during the entire test.

33 **Keywords:** solar receiver; exergy; energy efficiency; heat transfer; radiation.

34

Nomenclature			
A_{ap}	effective aperture area of dish [m ²]	Ex_s	rate of solar exergy delivery [kw]
A_p	project area [m ²]	G	direct solar radiation [w/m ²]
c_{av}	average specific heat capacity [J/kg·K]	\dot{m}	mass flow rate [kg/s]
D_f	focus point diameter [m]	n_d	parabolic dish combined optical efficiency [-]
E_D	concentrated solar radiation power [kw]	r_c	concentration ratio
E_L	heat loss [kw]	T_{in}	inlet temperature of the air [K]
E_R	receiver power [kw]	T_{out}	outlet temperature of the air [K]
E_S	solar radiation power on the dish [kw]	T_{ave}	average temperature of the air [K]
Ex_D	rate of dish exergy concentrated [kw]	U_L	heat loss coefficient [kw/m ² K]
Ex_f	exergy factor [-]	$\eta_{th.R}$	energy efficiency of the receiver [-]
Ex_R	receiver exergy [kw]	$\eta_{ex.R}$	exergy efficiency of the receiver [-]

35

36 1. Introduction

37 With rapidly increasing energy prices and globalization, process industries seek
 38 opportunities to reduce production costs and improve energy efficiency. Among the
 39 energy-efficient technologies, Concentrated Solar Power (CSP) system is considered
 40 as one of the most attractive ways to solve the energy crisis in the future [1,2]. Many

41 developed countries like the United State and the European Commission have been
42 devoted to the solarised Brayton micro-turbines system over the past decades [3-5].

43 Compared to the traditional gas turbine, solarised Brayton turbines use solar
44 receiver to replace the combustion chamber in the traditional gas turbine [6]. The
45 solar concentration part which is used to provide high temperature air is very crucial
46 for the entire solar power system. The system efficiency and the cost of the power
47 generation are highly depended on the solar concentration conversion efficiency from
48 solar radiation to thermal fluid. Thus, the solar concentration part has to be well
49 designed in order to achieve high efficiency and low pressure loss. Many studies have
50 been devoted to the design and performance of the receiver. Neber and Lee [7]
51 designed a high temperature cavity receiver using silicon carbide. Then a scaled test
52 section was placed at the focal point of a parabolic dish collector and reached a
53 maximum temperature of 1248 K. Fernandez et al. [8] presented a multidisciplinary
54 design optimization of a 5 MW Small Particle Heat Exchange Receiver (SPHER) for
55 a central receiver solar plants. The new developed solar receiver aims to heat air to
56 temperatures in excess of 1300 K and use this high-temperature energy to drive a
57 Brayton cycle or a combined Brayton/Rankine cycle. It was found that the receiver
58 efficiency can be increased by 6% with respect to the previous baseline design from
59 the same author [9]. Buck et al. [10] introduced a receiver module consisting of a
60 secondary concentrator and a volumetric receiver unit which was closed with a domed
61 quartz window to transmit the concentrated solar radiation. Hischer et al. [11, 12]

62 proposed a novel design of a high-temperature pressurized solar air receiver for power
63 generation via combined Brayton-Rankine cycles. It consists of an annular reticulate
64 porous ceramic bounded by two concentric cylinders. The heat transfer mechanism
65 was analyzed by the finite volume technique and by using the Rosseland diffusion, P1,
66 and Monte-Carlo radiation method. It was found that, for a solar concentration ratio of
67 3000 suns, the outlet air temperature can reach 1000 °C at 10 bars, yielding a thermal
68 efficiency of 78%.

69 It is recognized that the flow and heat transfer processes in the solar receiver are
70 very complicated. Over the past years, many studies have been devoted to the
71 optimization of the design using theoretical and numerical method. Cui et al. [13]
72 developed a three-dimensional optical model and simulated the solar transmission
73 process for a pressurized volumetric receiver using the Monte Carlo Ray Tracing
74 (MCRT) method. Wang et al. [14] developed a three-dimensional model of parabolic
75 dish-receiver system with argon gas as the working fluid to simulate the thermal
76 performance of a dish-type concentrated solar energy system. Lu et al. [15] proposed
77 heat and mass transfer models of the solar heat receiver, and associated heat
78 absorption and exergetic performance were further investigated by considering the
79 heat loss outside the receiver and fluid viscous dissipation inside the receiver. Song et
80 al. [16] analyzed the effect of the solar incidence angle in order to accurately simulate
81 the heat flux distribution around the absorber tube outer surface. Le Roux et al. [17]
82 investigated the effect of wind, receiver inclination, rim angle, atmospheric

83 temperature and pressure, recuperator height, solar irradiance and concentration ratio
84 on the optimum geometries and performance. Lu et al. [18] theoretically investigated
85 the nonuniform heat transfer model and performance of parabolic trough solar
86 receiver and the results showed that the heat loss of solar receiver from the
87 nonuniform model is a slight higher than that from the uniform model.

88 On the other hand, many numerical research works are also conducted to simulate
89 the detail heat transfer process. Flesch et al. [19] numerically analyzed the impact of
90 head-on and side-on wind on large cavity receivers with inclination angles ranges
91 from 0° (horizontal cavity) to 90° (vertical cavity) and compared with the data
92 published in the open literature. Wei et al. [20] presented an original CFD-based
93 evolutionary algorithm to determine the optimal fluid distribution in a tubular solar
94 receiver for the minimization of its peak temperature. Tu et al. [21] proposed a
95 modified combined method to simulate the thermal performance of a saturated
96 water/steam solar cavity receiver. Capeillere et al. [22] numerically studied the
97 thermomechanical behavior of a plate solar receiver with asymmetric heating. The
98 numerical results showed that the choice of the shape and levels of the solar irradiance
99 map is crucial. The distribution of the most relevant incident solar flux and the
100 geometry compromise were determined. Wang et al. [23] conducted a numerical
101 study focusing on the thermal performance of porous medium receiver with quartz
102 window. Their results indicated that the pressure distribution and temperature

103 distribution for the condition of fluid inlet located at the side wall was different from
104 that for the condition of fluid inlet located at the front surface.

105 Exergy analysis has been applied in various power studies [24-26].
106 Thermodynamic analyses and optimization of a recompression N₂O Brayton power
107 cycle have been performed [27]. The performance of a regenerative Brayton heat
108 engine has been studied by focusing on the minimization of irreversibility [28]. In the
109 authors' earlier studies [29, 30], a coiled tube solar receiver has been designed and
110 tested in the real solar radiation condition. But due to the limitation of the tube
111 material, the coiled tube solar receiver can not achieve very high temperature. Thus, a
112 pressurized volumetric solar receiver using metal foam as thermal absorbing core is
113 designed in this work. It appears from the previous investigation that the key point for
114 the solarised Brayton micro-turbines is to develop solar receivers which have terrific
115 performance on the pressure loss and heat transfer. To the best of the authors'
116 knowledge, there is a lack of available experimental data under real concentrate solar
117 and variable mass flow conditions especially for the cases of extremely high heat flux
118 and high temperature. To this end, the present research is aimed to experimentally
119 analyze both the efficiency and heat loss of a pressurized volumetric solar receiver
120 under real solar radiation and variable mass flow conditions in more detail.

121

122 **2. Experimental apparatus and method**

123 *2.1. Experimental apparatus*

124 The experimental study was conducted at a location with the geographical
125 position of 30°36' latitude and 120°22' longitude, Hangzhou, China. The whole
126 system, shown schematically in Fig. 1, mainly consists of three components: dish,
127 compressor and receiver. The dish used for the experimental tests of the developed
128 solar heat receiver was listed in Fig. 2. All 40 trapezoidal, pre-bent mirrors are resin
129 molded and laminated. The reflective surface is applied as an adhesive foil. At the
130 bottom of the dish a cut out is made for the tower. The main dish parameters utilized
131 in the current study are illustrated in Table 1, which is provided by the dish reflector
132 manufacturer. To make sure that the light reflected by the mirror focus on the aperture
133 of the receiver, each mirror was adjusted carefully.

134 The dish is controlled by a solar tracker which is embedded in the inner program to
135 make the dish face the Sun automatically. The inner program could accurately
136 calculate the attitude angle in terms of the dish location of the earth and the local time.
137 As illustrated in Fig. 1, the attitude angel is formed between the boom line and
138 horizon line. A stepping motor can be well controlled to change the dish attitude angle
139 slowly. When the dish is in operation during the morning, the attitude angle is lower
140 than 40° for the sun just rising over the horizontal line. Whereas the dish is operated
141 in the noon, the attitude angle is approximately 80°. It should be recognized that the
142 initial setting for the location and accurate time is very crucial during the test. An
143 uninterruptible power supply (UPS) system is also adopted to assist the dish off the
144 solar direction in some emergency to further protect the receiver. During the

145 experiments, a 20 kw piston compressor driven by the electricity is used to compress
146 the air. The pressurized air is compressed at environmental temperature and pressure.
147 After the filter, the air is pressurized into the air tank with the pressure of 0.8 Mpa to
148 guarantee the enough air flow during the experiment process. After that the
149 pressurized air is supplied into the receiver. Two valves are installed at the receiver
150 inlet and outlet to ensure the receiver works under designed pressure about 0.4 Mpa.
151 Because the light incident surface of the receiver is made of quartz glass, too high
152 pressure could damage the receiver. Hence, by adjusting this valve, the pressure of the
153 whole system as well as the portion of the receiver can be well controlled. The output
154 mass flow rate is variable. Thermocouple and pressure sensor are placed at the inlet
155 and outlet of the pipe respectively to obtain the receiver efficiency and heat loss. The
156 receiver itself is mounted onto the cantilever arm. In the current study, the heat flux of
157 the focus power could achieve 1000 kW/m^2 for the dish concentrator has the
158 concentrate ratio of 1750. It can be expected that, except for the receiver and
159 protecting panel, other components of the system would be burned in a short time. To
160 protect other part of the receiver from misaligned radiation an additional protecting
161 panel is mounted circumferentially to the receiver. As shown in Fig. 3, the protecting
162 panel is made of Calcium silicate board with 10 mm in thickness. The diameter of the
163 aperture in the protecting panel is 250 mm. Four K-type thermocouples with an
164 accuracy of $0.5 \text{ }^\circ\text{C}$ are fixed on the back to monitor the temperature of the protecting
165 panel. When the temperature is over $850 \text{ }^\circ\text{C}$, it means that the concentrate solar spot is

166 not located into the aperture. As a result, the inner program has to be reset to adjust
167 the attitude angle in order to prevent fatal damage.

168 *2.2. Solar receiver model*

169 For the current experimental evaluation, as shown in Fig. 4, the solar receiver is
170 designed as a type of pressurized volumetric solar receiver. Fig. 4(a) shows the 3D
171 view of the model and Fig. 4(b) presents the cross sectional view of the receiver. The
172 advantage of the pressurized volumetric solar receiver is high outlet air temperature
173 and high thermal efficiency. It should be stressed here that the key point for the design
174 of the pressurized volumetric solar receiver is the cooling of the light incident glass
175 and the equally distributed mass flow in the heat absorbing core. The light incident
176 glass is made of quartz glass which can endure a temperature up to 1500 C°. But the
177 concentrated solar focuses on the quartz glass directly, the glass cooling using the
178 inlet air could extend the life span of the receiver and make the receiver working
179 process more secure. For this reason, a big inlet tube is used with the diameter of 50
180 mm. The pressured air is injected into the inlet tube, and then, it is divided into three
181 small tubes with the diameter of 20 mm. The three small tubes that circumferentially
182 uniformly distributed are welded at the end of the pressure cavity which forms the
183 main part of the solar receiver. The air flow along the edge of the cavity and inject on
184 the quartz glass forming the cooling of the light incident glass. The diameter and the
185 height of the main part of the receiver are 400 mm and 360 mm, respectively. The
186 concentrate solar radiation (CSR) passes through the quartz glass and heat the

187 absorbing core. As shown in Fig. 5, the material of the absorbing core is Nickel foam
188 which could endure the temperature up to 1453 °C. To increase the absorbing ability,
189 65mm Nickel foam with the PPI (Pores per Inch) value of 75 is selected. PPI which is
190 a common parameter is usually used in industry to indicate the pore diameter of the
191 metal foam. The 75 PPM value means that the pore diameter is about 0.34 mm. One
192 can imagine that the small pore diameter would enhance the heat transfer coefficient
193 and heat transfer area easily. At last, to minimize the heat loss, the receiver is
194 surrounded by Aluminum silicate whose heat conductivity coefficient is
195 0.06 W/m² K⁻¹.

196 2.3. Energy and exergy analysis

197 Experimental energy and exergy parameters to characterize the thermal
198 performance of the receiver are presented in this section.

199 2.3.1. Energy analysis [31, 32]

200 The energy that the whole system receives comes from the solar radiation. The
201 solar radiation power on the parabolic dish reflector can be expressed as:

$$202 E_S = A_{ap}G$$

203 (1)

204 where E_S is the solar radiation power on the dish, A_{ap} is the effective aperture
205 area of the parabolic dish, and G is the direct solar irradiation from the Sun to the dish.
206 G is measured with a normal incidence pyrheliometer (NIP) Hukseflux DR01
207 attached to the solar tracker.

208 The solar radiation is concentrated and delivered to the receiver by the parabolic
209 dish. The concentrated solar radiation power (E_D) can be expressed as:

$$210 \quad E_D = n_d E_S = n_d A_{ap} G \quad (2)$$

211 where E_D is the concentrated solar radiation power from parabolic dish to the
212 receiver, n_d is the parabolic dish combined with optical efficiency described in
213 Table 1.

214 The concentrated solar radiation on the receiver is absorbed by the heat-transfer
215 fluid flowing in the pressurized cavity of the receiver. The energy rate that air absorbs
216 or receives power is given by:

$$217 \quad E_R = \dot{m} c_{av} (T_{out} - T_{in}) \quad (3)$$

218 where \dot{m} is the mass flow rate of the air, c_{av} is the average specific heat
219 capacity of the air which is a function of the average air inlet temperature (T_{in}) and air
220 outlet temperature (T_{out}). The average temperature of the receiver (T_{ave}) can be
221 defined by:

$$222 \quad T_{ave} = (T_{in} + T_{out})/2 \quad (4)$$

223 Thus, the relation between the average specific heat capacity of the air and the
224 average temperature can be obtained as:

$$225 \quad c_{av} = 0.9956 + 0.000093 T_{ave} \quad (5)$$

226 Based on the energy conservation, the receiver power is the difference between
227 the concentrated solar radiation power and the overall heat losses are relative low. The
228 receiver power can also be described as

229 $E_R = E_D - E_L$ (6)

230 where E_L is the rate of the heat loss from the receiver to the surroundings,
 231 which contains the convective heat losses, conductive heat losses and radioactive heat
 232 losses. E_L can be expressed as

233 $E_L = U_L A_R (T_{ave} - T_{amb})$ (7)

234 where U_L is the total heat loss coefficient determined, A_R is the effective
 235 receiver area, and T_{amb} is the ambient temperature. The product $U_L A_R$ is referred as
 236 the heat loss factor given by

237 $U'_L = U_L A_R$

238 (8)

239 Therefore, combination of Eqs (2), (3), (6) and (7) can yield

240 $\dot{m}c_{av}(T_{out} - T_{in}) = n_d A_{ap} G - U'_L (T_{ave} - T_{amb})$

241 (9)

242 The thermal energy efficiency of the receiver is defined as the ratio of the
 243 receiver power to the concentrated solar radiation power from the parabolic dish to
 244 the receiver which is expressed as:

245 $\eta_{th.R} = \frac{E_R}{E_D} = \frac{\dot{m}c_{av}(T_{out}-T_{in})}{n_d A_{ap} G}$

246 (10)

247 By dividing $A_{ap} G$ on both side of Eq. (9) and combine with Eq. (10) leads to

248 $\eta_{th.R} n_d = n_d - \frac{U'_L (T_{ave} - T_{amb})}{A_{ap} G}$ (11)

249 2.3.2 Exergy analysis [31, 32]

250 The exergy rate of the receiver or the quality of the energy delivered to the
 251 circulating fluid with reference to the surroundings can be expressed as

$$252 \quad EX_R = E_R - \dot{m}c_{av}T_{amb} \ln\left(\frac{T_{out}}{T_{in}}\right) \quad (12)$$

253 Substituting Eq. (3) into Eq. (12) yields

$$254 \quad EX_R = \dot{m}c_{av} \left[(T_{out} - T_{in}) - T_{amb} \ln\left(\frac{T_{out}}{T_{in}}\right) \right] \quad (13)$$

255 The rate of the solar exergy delivery by the Sun to the dish and then to the
 256 concentrator is given by the Petela expression [33] and is expressed as

$$257 \quad EX_S = GA_{ap} \left[1 + \frac{1}{3} \left(\frac{T_{amb}}{T_s} \right)^4 - \frac{4T_{amb}}{3T_s} \right] \quad (14)$$

258 where T_s is the surface temperature of the Sun which is approximately 5762 K.

259 So the concentrated solar radiation exergy (EX_D) can be expressed as:

$$260 \quad EX_D = n_d GA_{ap} \left[1 + \frac{1}{3} \left(\frac{T_{amb}}{T_s} \right)^4 - \frac{4T_{amb}}{3T_s} \right]$$

261 (15)

262 The exergy efficiency is defined as the ratio of the receiver exergy rate to the rate
 263 of the concentrated solar radiation exergy and can be determined as follows:

$$264 \quad \eta_{ex.R} = \frac{EX_R}{EX_D} = \frac{\dot{m}c_{av} \left[(T_{out} - T_{in}) - T_{amb} \ln\left(\frac{T_{out}}{T_{in}}\right) \right]}{n_d GA_{ap} \left[1 + \frac{1}{3} \left(\frac{T_{amb}}{T_s} \right)^4 - \frac{4T_{amb}}{3T_s} \right]} \quad (16)$$

265 The exergy factor is defined as the ratio of the receiver exergy rate to the
 266 receiver energy rate and can be represented by equation:

$$267 \quad EX_f = \frac{EX_R}{E_R} = \frac{\dot{m}c_{av} \left[(T_{out} - T_{in}) - T_{amb} \ln\left(\frac{T_{out}}{T_{in}}\right) \right]}{\dot{m}c_{av}(T_{out} - T_{in})} \quad (17)$$

268 3. Uncertainty analysis

269 The uncertainties of the measurement in the experiment are dependent on the
 270 experimental conditions and the measurement instruments. An uncertainty analysis is

271 performed on the receiver power E_R and the receiver exergy Ex_R , which are the
 272 most important derived quantities from the measurements of using the propagation of
 273 error method described by Moffat [34]. The uncertainty of the receiver power could
 274 be calculated by the following equation:

$$275 \quad \delta E_R = \sqrt{\left(\frac{\delta E_R}{\delta \dot{m}}\right)^2 (\delta \dot{m})^2 + \left(\frac{\delta E_R}{\delta T_{out}}\right)^2 (\delta T_{out})^2 + \left(\frac{\delta E_R}{\delta T_{in}}\right)^2 (\delta T_{in})^2} \quad (17)$$

276 While the uncertainty of the receiver exergy rate is given by

$$277 \quad \delta Ex_R = \sqrt{\left(\frac{\delta Ex_R}{\delta \dot{m}}\right)^2 (\delta \dot{m})^2 + \left(\frac{\delta Ex_R}{\delta T_{out}}\right)^2 (\delta T_{out})^2 + \left(\frac{\delta Ex_R}{\delta T_{in}}\right)^2 (\delta T_{in})^2 + \left(\frac{\delta Ex_R}{\delta T_{amb}}\right)^2 (\delta T_{amb})^2} \quad (18)$$

278 In the current study, the main uncertainty parameters are the mass flow rate (\dot{m}),
 279 the inlet temperature (T_{in}), and the outlet temperature (T_{out}). The relative uncertainty
 280 of the mass flow rate is given by the float flowmeter with the value of 2%. Therefore,
 281 $\delta \dot{m} = 2\% \times \dot{m} = \pm 0.001 \text{ kg/s}$. The uncertainty of the temperature is given by the
 282 K-type thermocouple with the value of $\delta T_{out} = \delta T_{in} = \pm 0.5 \text{ K}$.

283 The maximum experimental values for the receiver power and exergy rate are
 284 around 18.5 kW and 7.28 kW, respectively. The uncertainty of the receiver power is
 285 0.372 kW, and the uncertainty of the receiver exergy rate is 0.147 kW. Overall, the
 286 overall uncertainty of the receiver power and exergy rate are 2.01% and 2.02%,
 287 respectively.

288 4. Results and discussion

289 Fig. 6 shows the variation of the solar irradiance (G) during a test period from
 290 10:00 am to 13:30 pm. The experimental data were collected on November 6th, 2015,

291 which is a local autumn season in Hangzhou, China. According to Fig. 6, it is shown
292 that the solar irradiance fluctuates at around 600 W/m^2 all the time. And the solar
293 irradiance data increases slowly with time except two fast drops observed in the
294 afternoon for about 15 mins. The reason could be due to the fact that two short period
295 of passing cloud occurred. From this figure, it can be seen clearly that the solar
296 irradiance is almost stable and maintained at around 600 W/m^2 from 10:00 am to
297 12:00 pm. It is obviously that the stable solar irradiation condition is beneficial for
298 analyzing the energy and exergy performance of the solar receiver. For this purpose, a
299 test period of continuous 2 h from 10:00 am to 12:00 pm is selected. Dynamic
300 acquisition system is used to record the parameters automatically during the test. The
301 ambient temperature is maintained at around $25 \text{ }^\circ\text{C}$ during the experiment process.
302 Fig. 7 shows the variation of the inlet pressure, outlet pressure and mass flow rate. For
303 the design pressure of the solar receiver is 0.4 Mpa , the experiment should be
304 conducted at the same pressure. Since the heating from the concentrate solar
305 irradiation could lead to the rising of the internal pressure, the inlet valve is adjusted
306 during the experimental process to ensure the solar receiver working safety. Therefore,
307 the mass flow rate fluctuates all the time. The average value of the mass flow rate is
308 about 0.036 kg/s . In the current study, the main purpose is to test the energy and
309 exergy performance of the solar receiver under the fluctuant mass flow rate condition.

310 Fig. 8 presents the time series of air temperatures at the inlet and outlet of the
311 receiver. The inlet temperature maintains nearly constant about $42 \text{ }^\circ\text{C}$. The outlet

312 temperature rises very quickly at the starting stage, and achieves the highest
313 temperature of 480 °C at the end of the experiment process. From 10:00 am to 10:30
314 am, it takes about half an hour to rise the solar receiver outlet temperature from 42 °C
315 to 430 °C. After 10:30 am, the outlet temperature increases very slowly with the time.
316 The first half an hour is used for preheating. It is due to the fact that the receiver tubes
317 are surrounded by the insulation materials with high specific heat capability (C_p). It is
318 noteworthy that the rising speed of the receiver efficiency is very high within the first
319 30 mins. This phenomenon is very important and should be stressed here since the
320 sunshine is limited in the day time, quick start up can make the overall solar power
321 generation system to generate more electricity. Therefore, the cost of the power
322 generation will be lower and the investment recovery period could be shorter. There is
323 also another interesting phenomenon that the mass flow rate fluctuation has little
324 effect on the solar receiver outlet temperature. It may be due to the reason that the
325 porous metal is used as the heat absorbing core. The pore size is very small with the
326 value about 0.34 mm. This small size pore could increase the heat transfer coefficient
327 and area obviously. The heat transfer between the porous metal and the air is strong
328 enough that the air outlet temperature could be very close to the temperature of the
329 porous metal. Therefore, the effect of the mass flow rate fluctuation on the solar
330 receiver outlet temperature is very small.

331 Fig. 9 presents a comparison of the power for the concentrated solar radiation
332 and receiver power. For the case of the nearly constant solar irradiance of 600 W/m²,

333 the concentrated solar radiation power (E_D) is maintained at around 22.5 kW with the
334 fluctuation lower than $\pm 10\%$. In addition, the accurate control system can make sure
335 the reflection focus located at the aperture of the receiver. The red line shown in Fig.
336 9 is the receiver power during the testing period. At the first 10 mins, the solar
337 receiver power (E_R) rise quickly from nearly 0 kW to 12 kW. After that, the value of
338 E_R has the same trend with the mass flow rate according to the time. In other words,
339 the mass flow rate has great influence on the solar receiver power. This can be easily
340 explained by the Eq. (3). As previously mentioned, the mass flow fluctuation has little
341 influence on the value of c_{av} , T_{out} and T_{in} , so the solar receiver power is mainly
342 affected by the mass flow rate especially after the starting stage. The maximum solar
343 receiver power is achieved at about 11:50am with the value of 18.5 kW.

344 Fig. 10(a) shows the time series of the solar receiver efficiency. It is found that
345 when the solar receiver turns into steady state, the efficiency of the solar receiver can
346 be above 55%. The peak value of the efficiency is 87%, and finally, the efficiency is
347 maintained at around 60%. And it is also found that the value of $\eta_{th,R}$ has the same
348 trend with the mass flow rate after the receiver entering into steady stage. In other
349 words, the energy efficiency is positively related with the mass flow rate as shown in
350 Fig. 10(b). The main reason for this phenomenon can be explained by Eq. (10). As
351 described above, the concentrated solar radiation power (E_D) is nearly maintained
352 constant at around 22.5 kW, but the mass flow rate has great influence on the solar
353 receiver power (E_R). Therefore, the ratio of E_R and E_D has the positive

354 relationship with the mass flow rate. This is a very beneficial conclusion. Because in
355 out experiment, the mass flow rate is obviously lower than the real solarised gas
356 turbine system. So the efficiency of the real solarised gas turbine system will be very
357 high. Correspondingly, the usage of the gas will be lower and the investment recovery
358 period could be shorter.

359 Fig. 11 demonstrates the evolution of the heat loss factor (U'_L). At the starting
360 point, U'_L is very high (1.17 kW/K) because of the receiver preheating, and then it
361 drops very quickly within the first 5 mins. When the receiver works at steady state,
362 the heat loss becomes lower and U'_L achieves the minimum value of 0.014 kW/K. In
363 the current study, the heat loss mainly consist of the conduction heat loss, convection
364 heat loss and radioactive heat loss. Conduction heat loss could be reduced by using
365 material with low thermal conductivity. In the present work, the receiver is
366 surrounded by Aluminum silicate whose thermal conductivity is $0.06 \text{ W/m}^2 \text{ K}^{-1}$. The
367 Aluminum silicate can be acted as the thermal insulator to minimize the heat loss. The
368 thermal convection between the solar receiver and ambient is very low as well, this is
369 because the absorbing core is sealed in a pressurized cavity. As a result, only small
370 natural convection occurs around the external cavity of the receiver. It is noted that
371 the use of the insulator could obviously reduce the natural convection. The radioactive
372 heat loss is also an important part of the heat loss, but it can be reduced by using small
373 aperture, as shown in Fig 4. In the design of solar receiver, choosing appropriate
374 aperture diameter is very important for the receiver performance.

375 Fig. 12 shows the comparison between the receiver exergy (Ex_R) and the
376 concentrated solar energy as well as the receiver energy. From this figure, the exergy
377 rate and energy rate vary in a similar manner, the mass flow rate also has the same
378 influence on the exergy rate. It is noted that the highest value of the exergy rate during
379 the test period is around 7.28 kW, whereas the maximum energy rate can reach 18.5
380 kW. It can be concluded that the quality of the energy from the receiver is low due to
381 a large amount of irreversible energy changes such as heat losses and the transfer of
382 high quality solar energy to a fluid that circulating at a relatively low temperature.
383 From Eq. (13), it can be concluded that under the same temperature difference ($T_{out} -$
384 T_{in}) and the same energy rate $\dot{m}c_{av}(T_{out} - T_{in})$ condition, increasing the receiver
385 inlet temperature (T_{in}) can achieve higher exergy rate (Ex_R). This will be very
386 helpful for the design of the solar power system. As a result, some recuperator or heat
387 exchanger should be used in the inlet of the solar receiver to recover the waste heat
388 and increase the solar inlet temperature.

389 Fig. 13 presents the comparison between the energy efficiency and exergy
390 efficiency. It is shown from Fig. 13 that similar trends in the exergy efficiency and the
391 energy efficiency are obtained. The highest exergy efficiency is approximately 36%,
392 whereas the highest energy efficiency is around 87%. This suggests that low quality
393 energy obtained from the receiver. It is because the inlet temperature of the receiver is
394 lower than 316 K, whereas the outlet temperature is very high and with the maximum
395 value of 850 K. The temperature ratio (T_{out}/T_{in}) is very high so that too much exergy

396 loss is observed. Therefore, increasing the inlet temperature could be a potential way
397 to increase the exergy efficiency.

398 Fig.14 shows the exergy factor plotted as a function of the temperature
399 difference between the outlet and inlet temperature of the receiver with a linear fitting
400 equation. The exergy factor is also usually used as a measure of the performance of
401 the receiver. Obviously, as the temperature difference increases, the exergy factor also
402 increases. This plot suggests that the higher exergy factor can be obtained when high
403 temperature difference is available. As seen from this figure, the highest exergy factor
404 is 0.41 during the entire test.

405 **5. Conclusions**

406 This paper performed an experimental study to investigate the thermal
407 performance of a pressurized volumetric solar receiver under real solar radiation
408 condition. The mass flow rate is variable during the experimental process. A parabolic
409 dish with solar tracker system is designed and analysed using energy and exergy
410 analyse method. Experimental results reveal that the solar irradiance is almost stable
411 and maintained at around 600 W/m^2 from 10:00 am to 12:00 pm. It takes about half
412 an hour to rise the solar receiver outlet temperature from $42 \text{ }^\circ\text{C}$ to $430 \text{ }^\circ\text{C}$. After 10:30
413 am, the outlet temperature increases very slowly with the time. The mass flow rate
414 fluctuation has little effect on the solar receiver outlet temperature. However, the mass
415 flow rate has great influence on the solar receiver power, energy efficiency and
416 exergy efficiency. The efficiency of the solar receiver can be above 55%. The peak

417 value of the efficiency is 87%, and finally, the efficiency is maintained at around 60%.
418 During the steady state, the heat loss becomes lower and U'_L achieves the minimum
419 value of 0.014 kW/K. The highest exergy efficiency is approximately 36%, whereas
420 the highest energy efficiency is around 87%. As the temperature difference increases,
421 the impact of the exergy factor increases. The highest exergy factor is 0.41 during the
422 entire test.

423

424 **Acknowledgements**

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426 Science Foundation of China (No. 51206164).

427 **References**

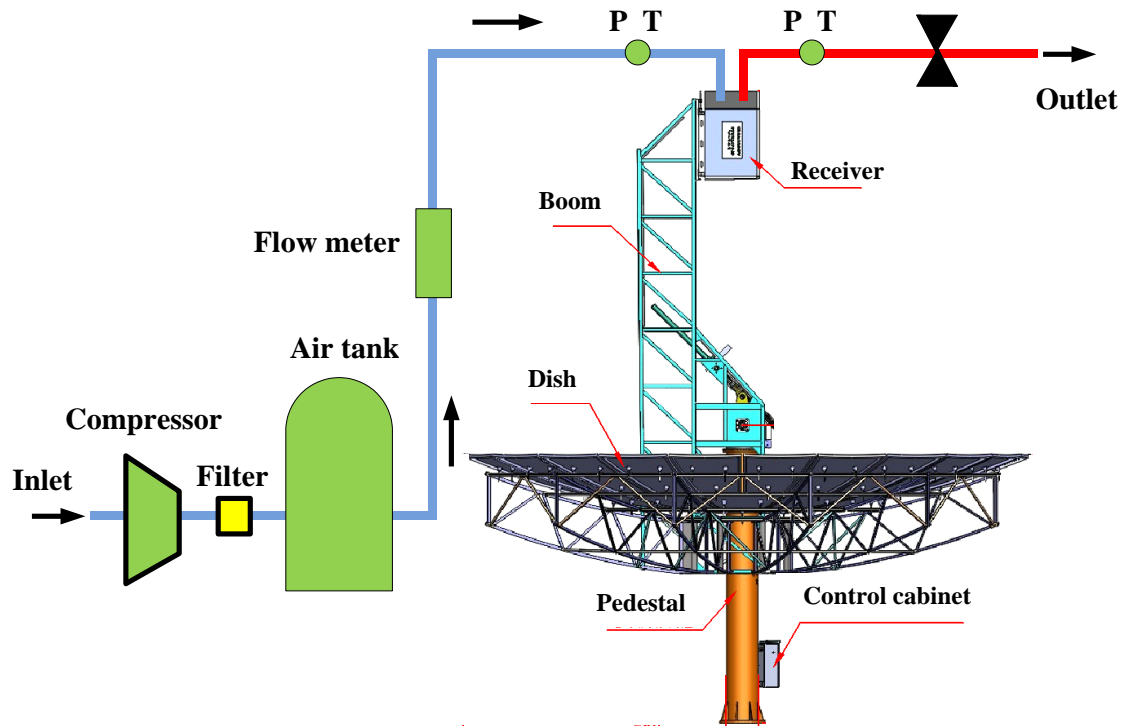
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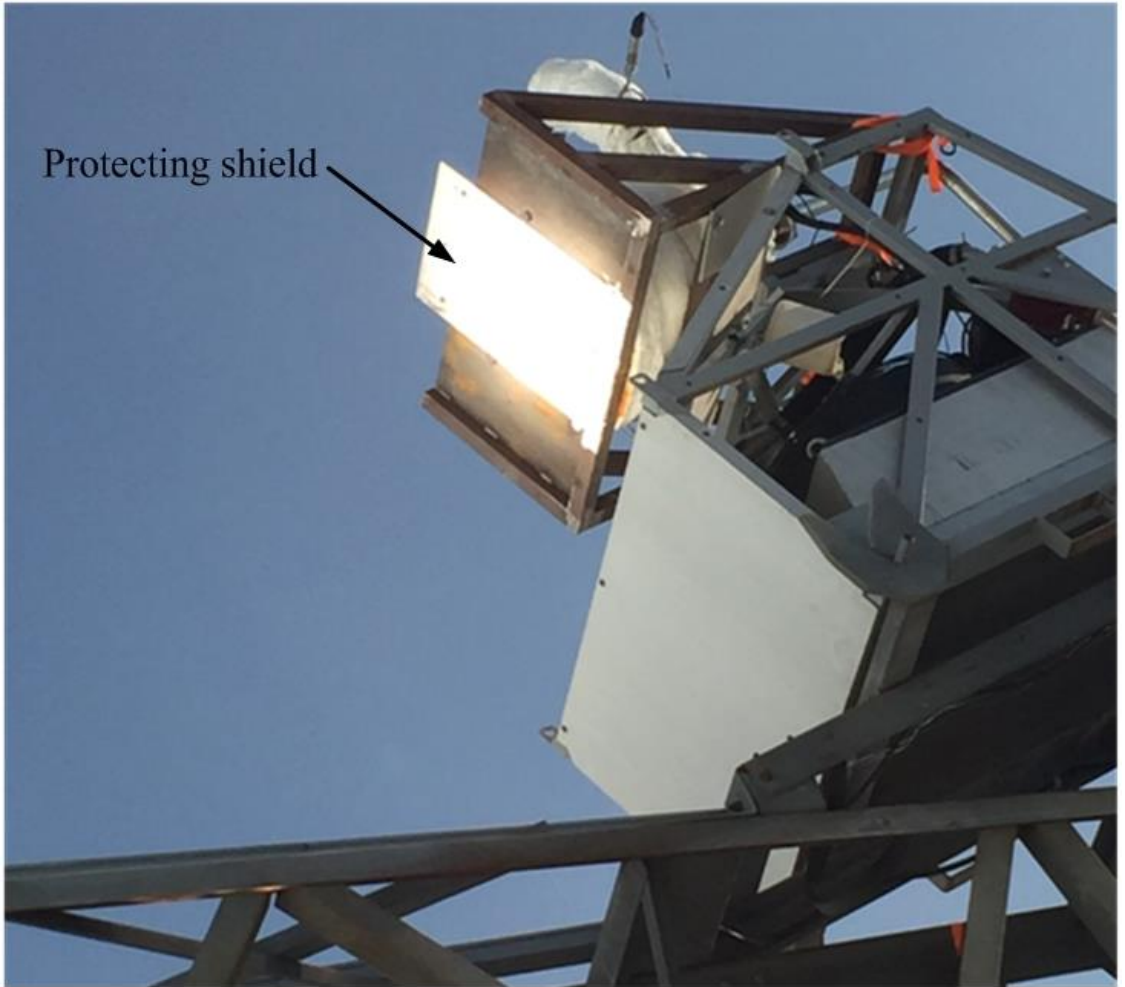
[36] **Fig. 1.** Schematic drawing of experimental test rig.



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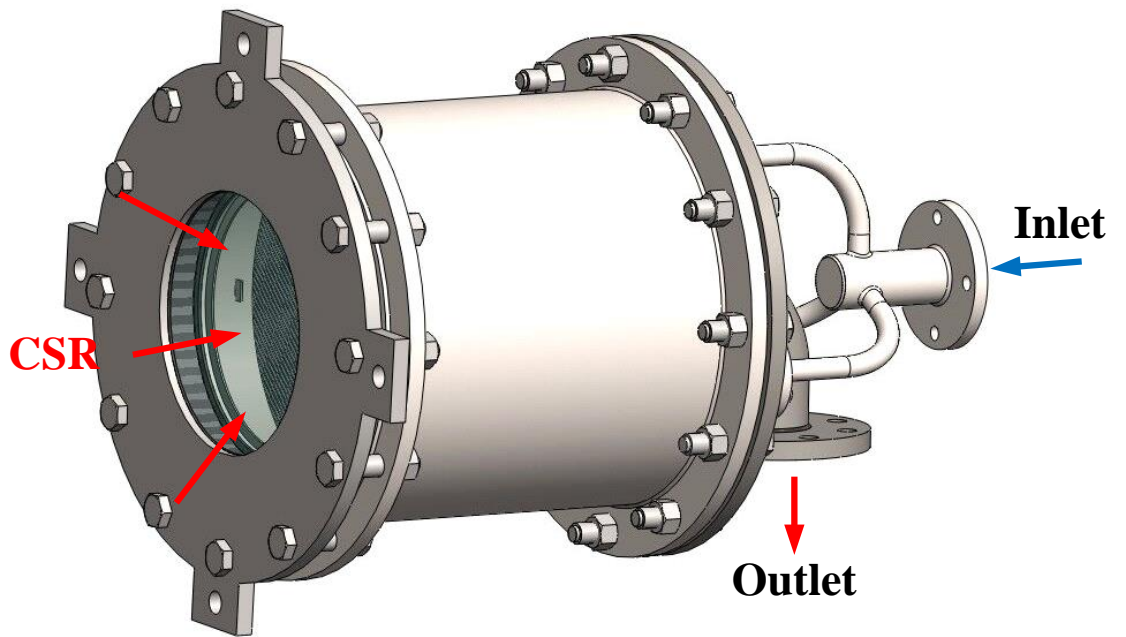
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[38] **Fig. 2.** Parabolic dish.



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[40] **Fig. 3.** Protecting shield under the concentrated solar radiation.

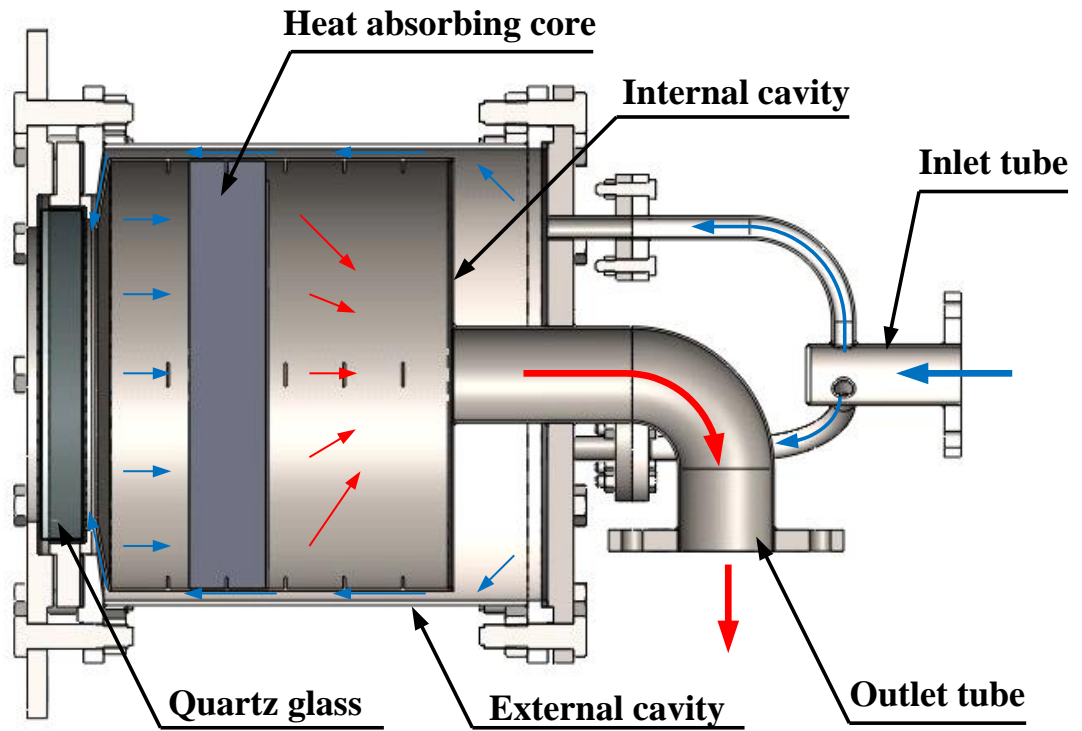


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(a)



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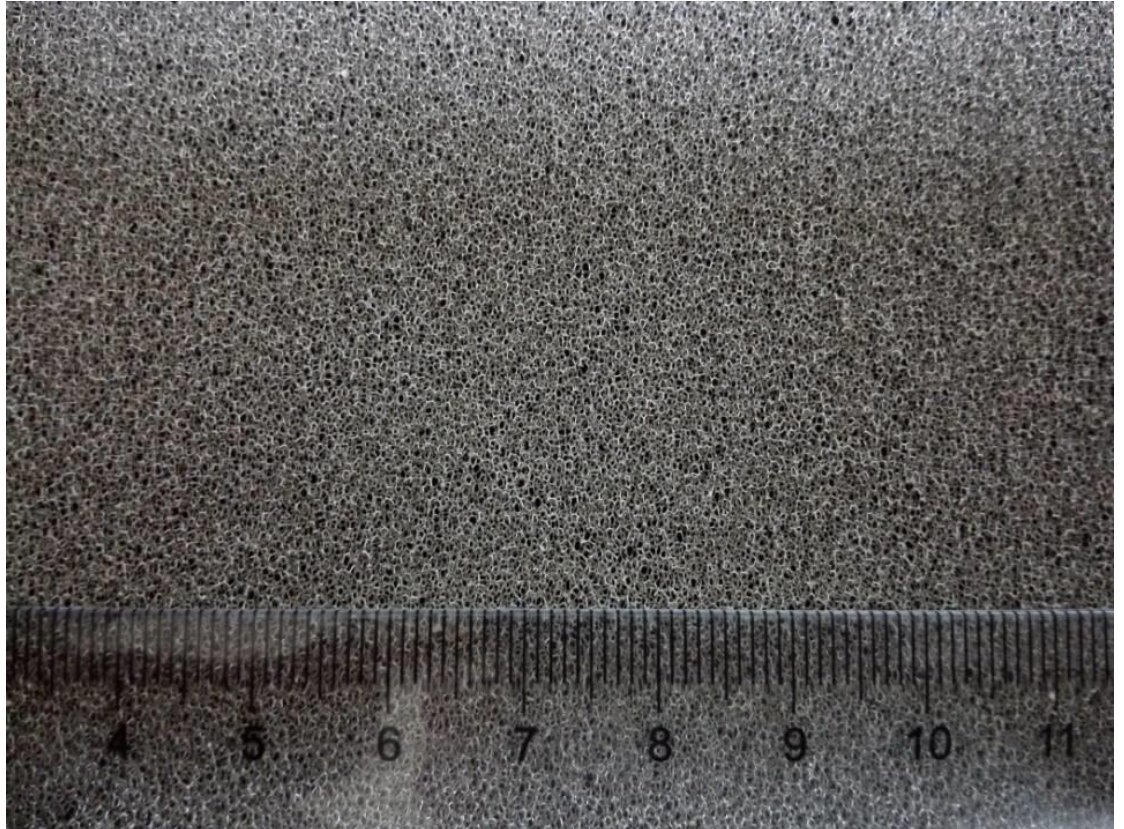
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[45] (b)

530 [46] **Fig. 4.** Pressurized volumetric solar receiver. (a) 3D model, (b) cross sectional

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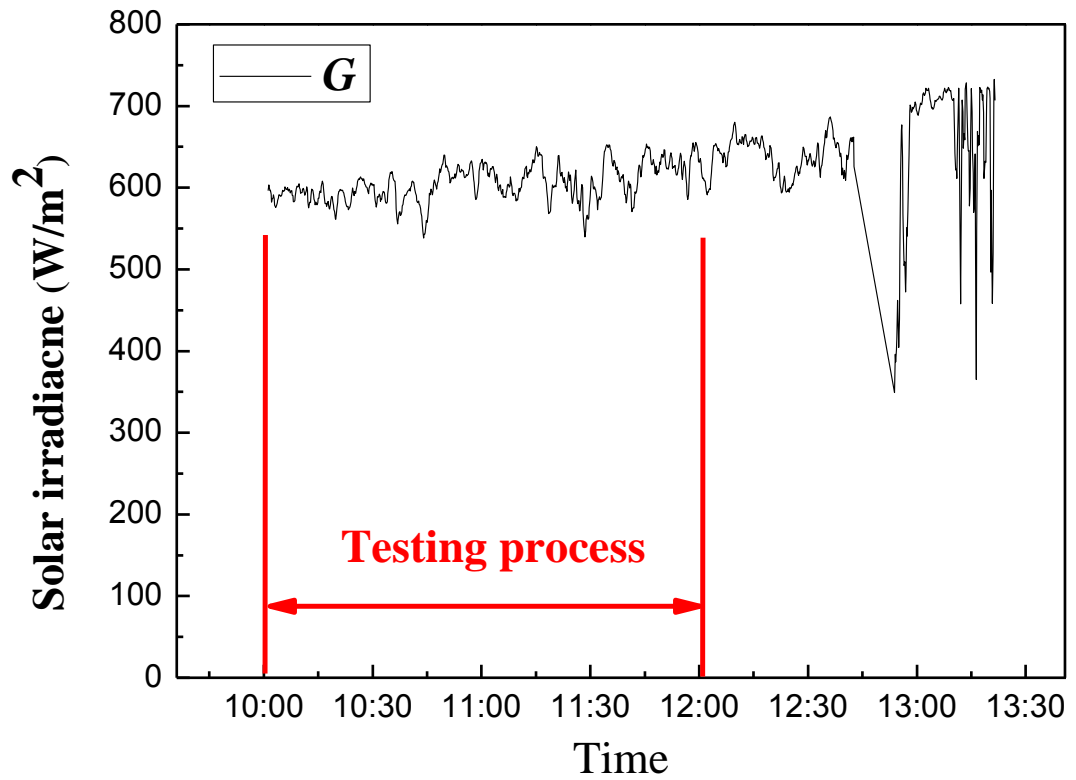
view.



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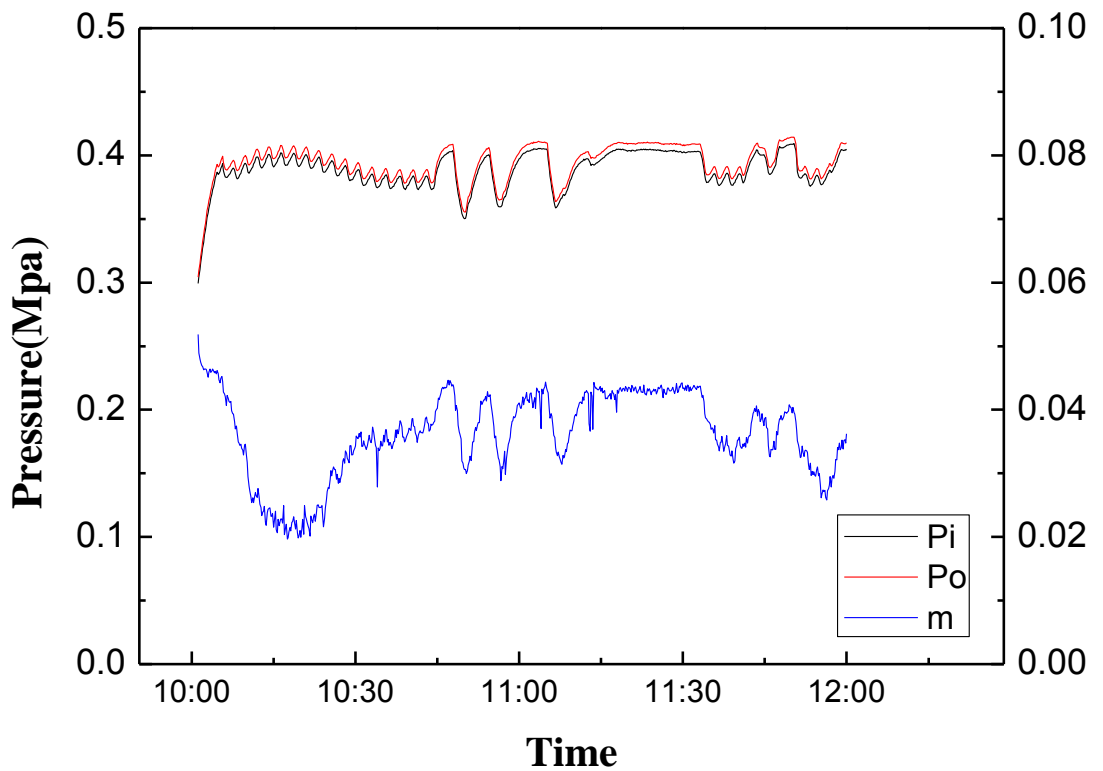
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[48] **Fig.5.** Nickel foam.



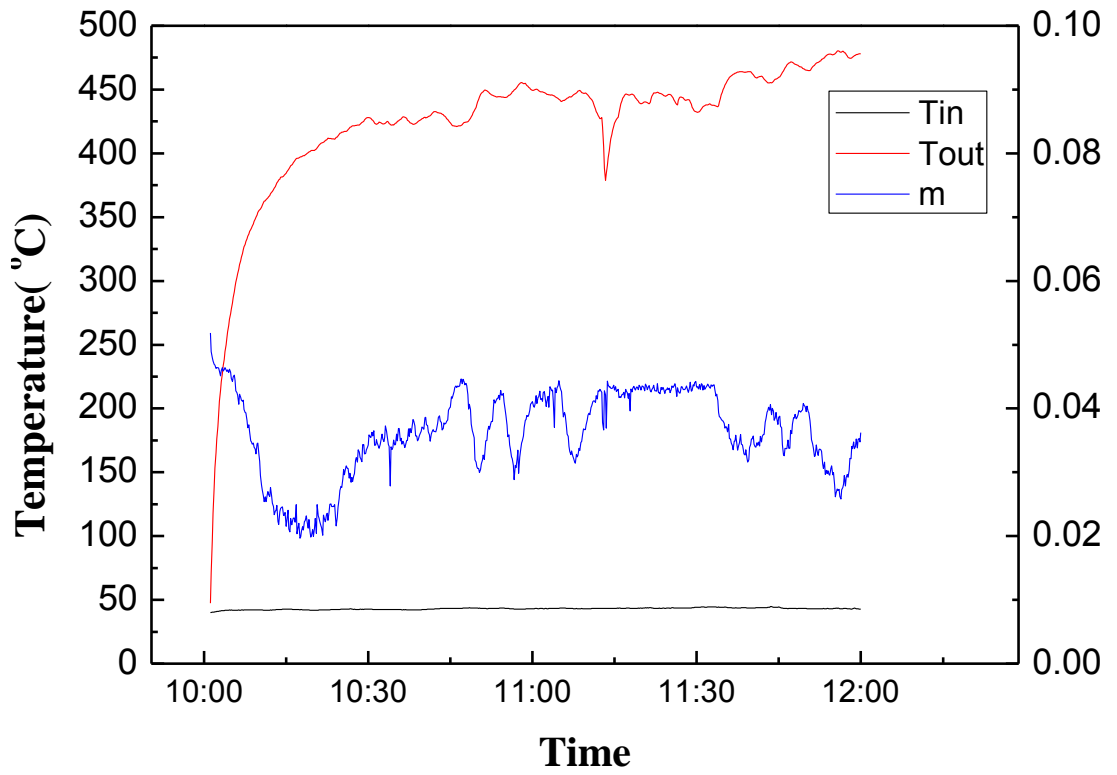
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[50] **Fig. 6.** Solar irradiance data during the day.



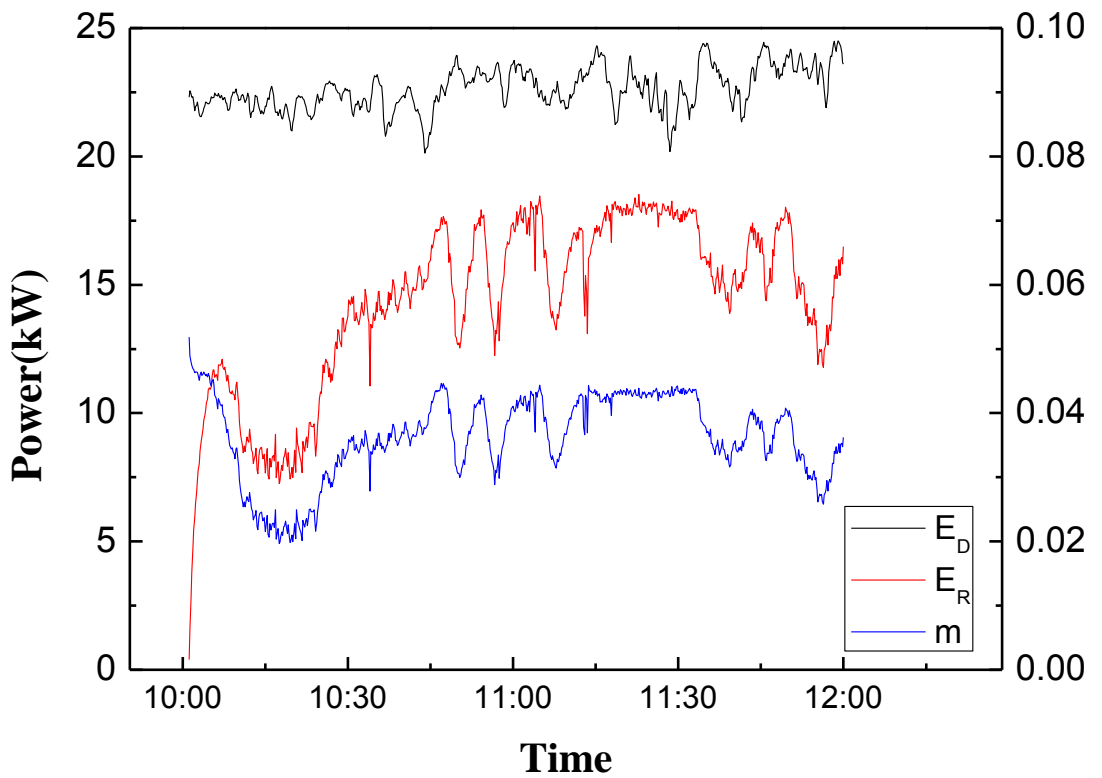
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[52] **Fig. 7.** Variation of the inlet pressure, outlet pressure and mass flow rate.



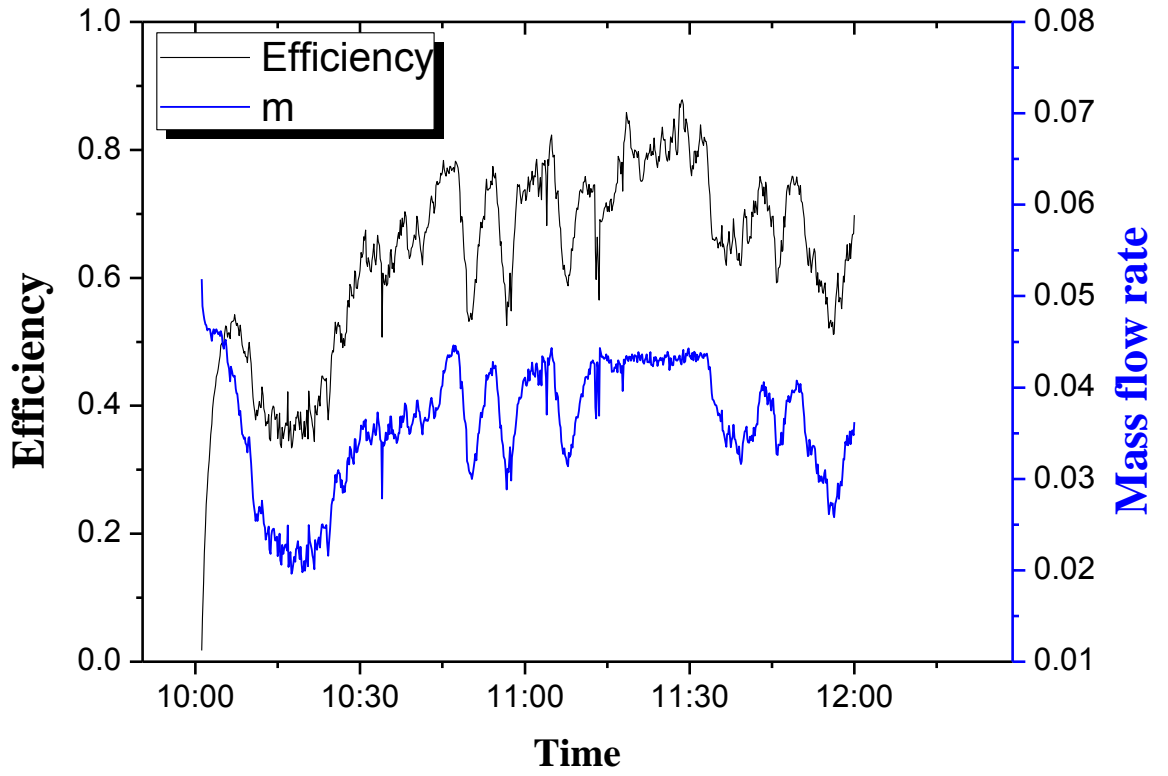
538 [53]

539 [54] **Fig. 8.** Variation of the inlet temperature, outlet temperature and mass flow rate.



540 [55]

541 [56] **Fig. 9.** Variation of the dish power, receiver power and mass flow rate.



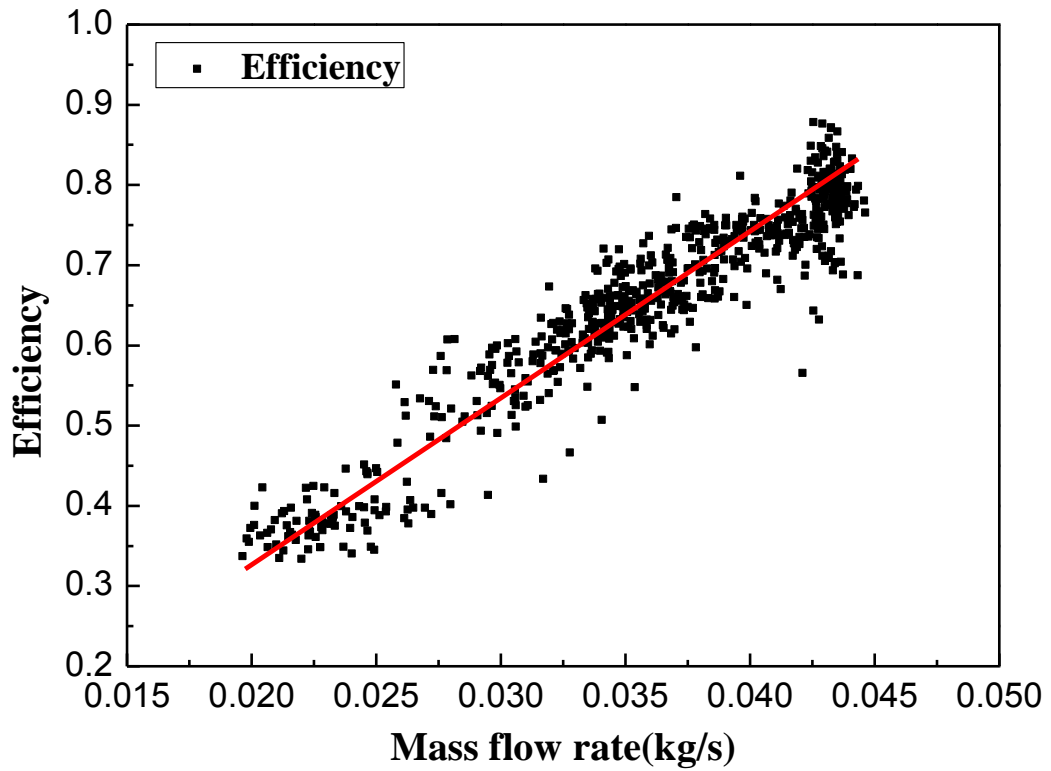
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[58] (a)

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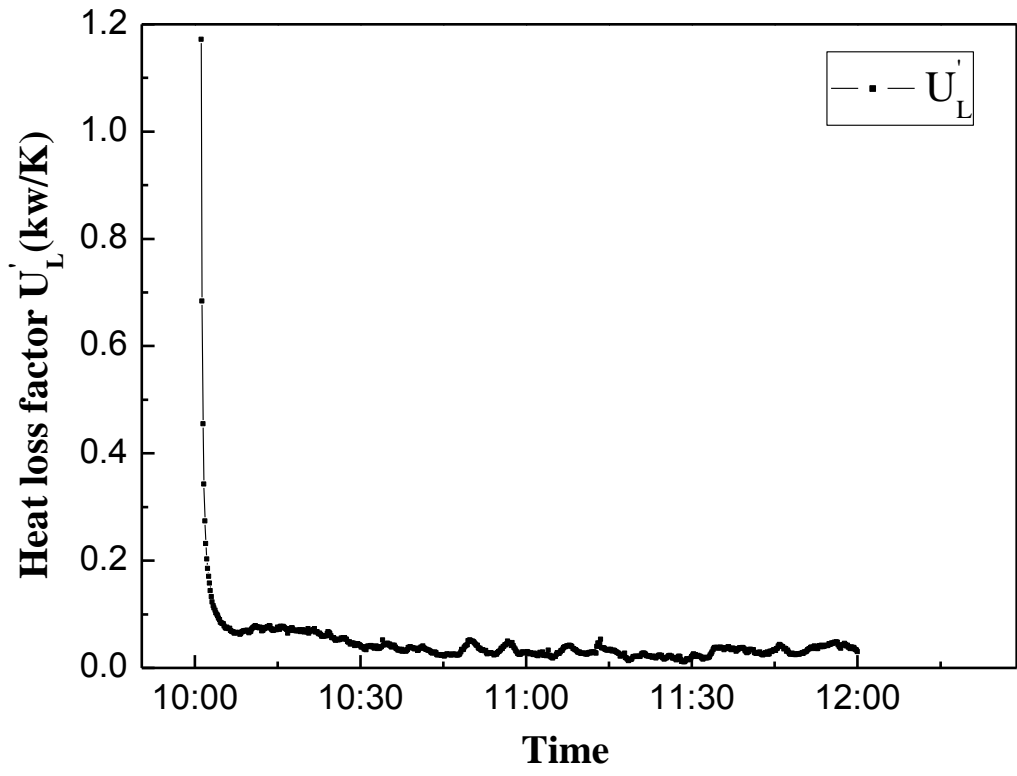


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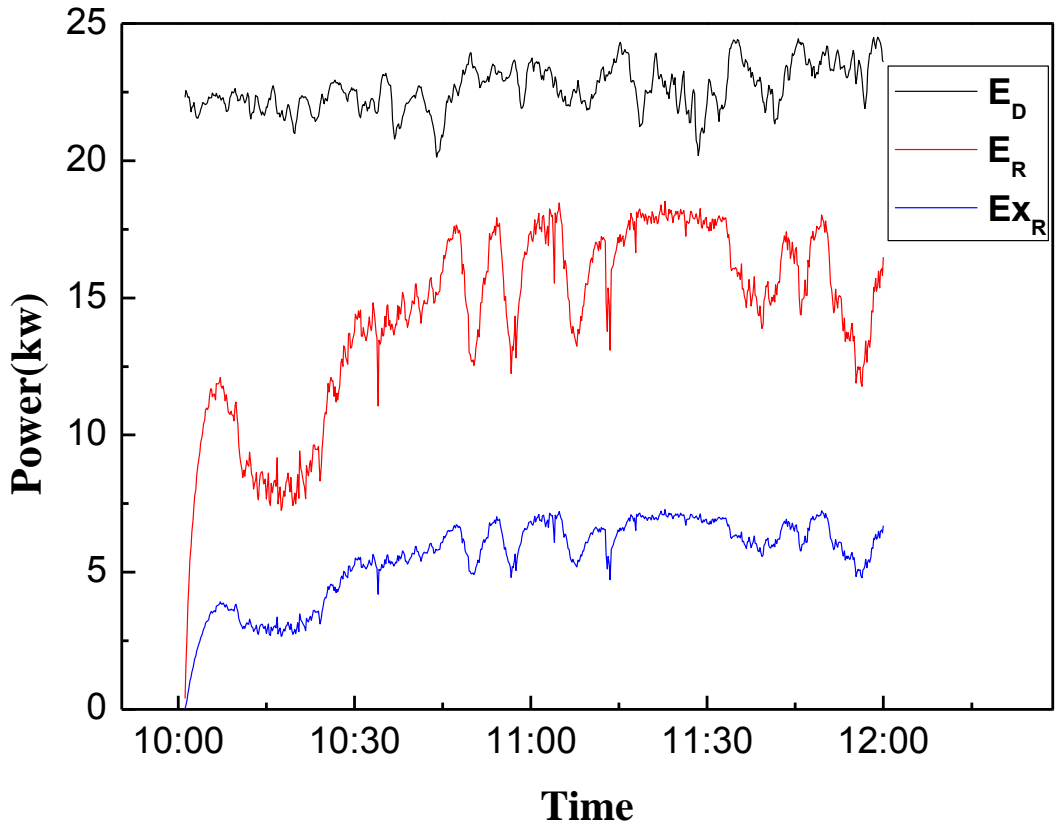
[61] (b)

547 [62] **Fig. 10.** Efficiency results. (a) during the test period, (b) efficiency vs. mass flow
 548 rate.



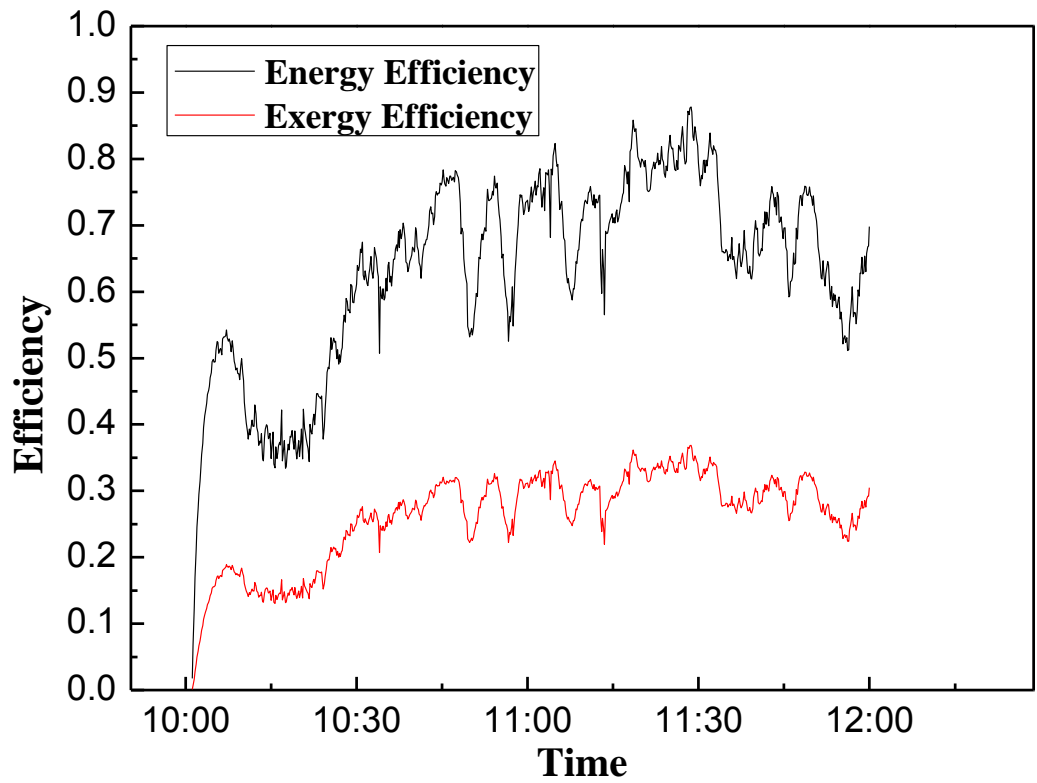
549 [63]
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[64] **Fig. 11.** Heat loss factor profile during the test period.



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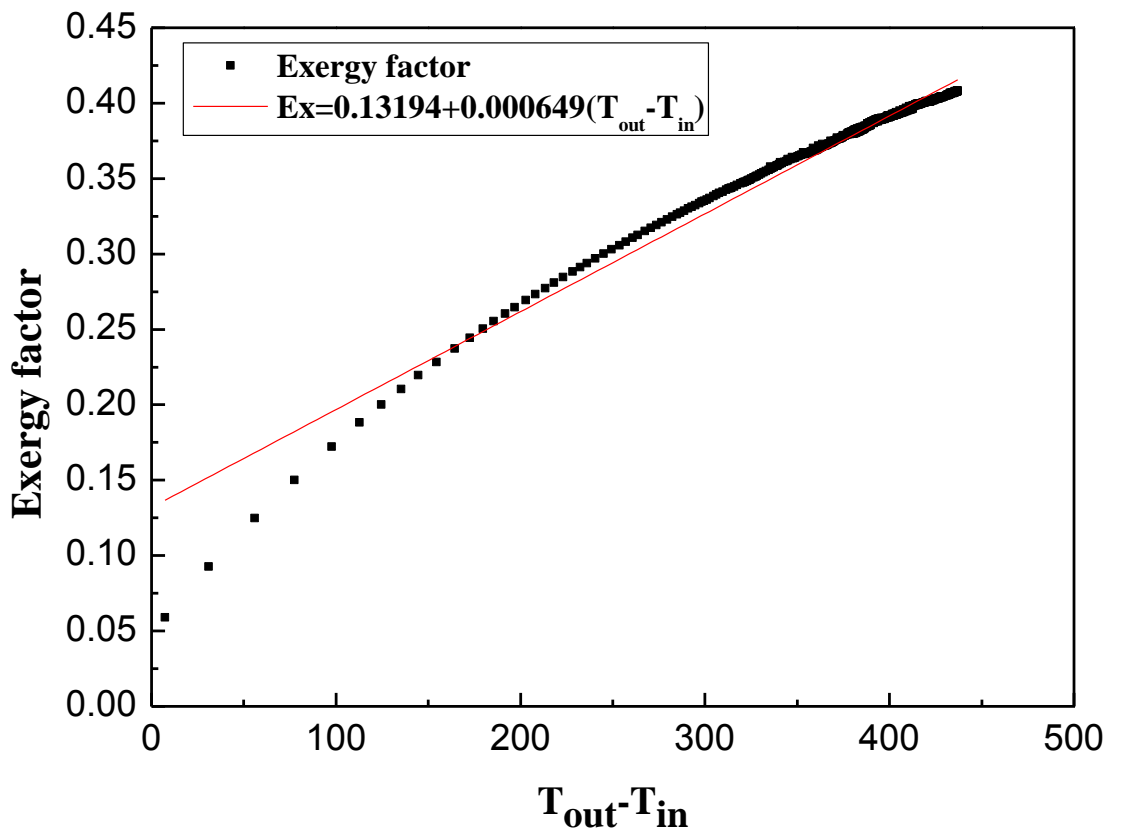
Fig. 12. Variation of the power for the receiver energy and exergy [67] during the test period.



555 [68]

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[69] **Fig. 13.** Energy and Exergy efficiency profile during the test period.



557 [70]

558

[71] **Fig. 14.** The effect of temperature difference on the exergy factor.

559 [72]