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Analysis of a Twin Screw Expander for ORC Systems using Computational Fluid Dynamics with a Real Gas Model

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ABSTRACT

With the increase in energy prices and environmental constraints, recovering the energy from low grade waste heat presents an important challenge. The Organic Ranke Cycles (ORCs) are widely used, but there is still need to improve their efficiency (especially for small scale energy production). This paper presents the Computational Fluid Dynamics (CFD) analysis of a twin screw expander which is used for power generation in an ORC system with refrigerant R245fa. The deforming mesh motion is handled by an in-house code which generates a block-structured grid with the help of the solutions of Laplace problems. The properties of refrigerant R245fa are derived from the Augnier Redlich-Kwong cubic equation of state which is incorporated in the computational model.

From the results of a CFD analysis, the pressure-volume diagram, mass flow rates and power output for different pressure ratios and different designs were obtained. In order to evaluate the effects of individual clearance gaps on the performance of the expander, time dependent mass flow diagrams for each of them are provided and studied. It can be concluded that the influence of the leakage flows increases with decrease in rotational speed or with the increase in pressure ratio. To avoid losses during the filling, an optimized design of the inlet port is necessary.

1. INTRODUCTION

Small scale ORC systems have a big potential for waste heat recovery. The ORC is named for its use of an organic, high molecular mass fluid which is characterized by low saturation temperatures. Although these ORC systems are now well developed, research efforts are increasingly directed towards higher efficiencies and powers. Thermodynamic machines for the expansion of the organic fluids are the key element for the power generation in the ORC. The screw expander is a special type of rotary displacement machine, in which two screws rotate inside two overlapping cylinders. The current generation of screw expanders is in fact designed as compressors used the other way around.

Moving the mesh surrounding the rotor surfaces is complex and therefore a powerful grid manipulator is necessary for CFD calculations. In this paper an in-house code presented in Vande Voorde (2004) is used. The validation of CFD calculations using this grid generator for a twin screw compressor has been reported in Vande Voorde (2005). Since the validation showed good agreement between the experimental results and CFD calculations, the same procedure is used in this work, for the simulations of an expander where only the sense of rotation is changed.

Several papers have been published using the zero-dimensional chamber model, based on the ideal gas law and standard thermodynamic relations for compressors and expanders. Seshaiah (2006) presented a model for twin screw oil injected compressor and Bruemmer and Hutker (2009, 2012) a model for the expander. Recently, the influence of the clearance sizes on the performance of a screw expander has been investigated by means of CFD calculations by Rane (2013). Despite these investigations, there is no detailed analysis of the flow through the clearances as a function of the rotation angle. Since the leakage flows have a significant influence on screw expander efficiency, it is of great importance to calculate these flows correctly.

The screw expanders are characterized by a fixed built-in volume ratio v, determined by the form of the inlet and outlet surfaces. Paul (2011) determined the internal pressure ratio as $\pi = v^k$. For the best performance, the built-in volume ratio should match the operating conditions in order to limit the over- and under-expansion losses. If the internal volume ratio is too high for a given set of operating conditions, the fluid remains trapped longer than

necessary, leading to a pressure drop below the discharge pressure (over-expansion). On the contrary, if the internal volume ratio is too low, the fluid in the chamber remains above the discharge pressure when the discharge/outlet port starts to open (under-expansion). Therefore, depending on the internal pressure ratio, under- or over-expansion losses can decrease the efficiency of a screw expander.

Apart from over- and under-expansion, the efficiency of a screw expander is mainly governed by the leakage flows through the clearances and suction pressure losses. In this paper, the performance of a screw expander is studied for different pressure ratios and for two different expander designs. The difference between these two designs is additional injection ports. The leakage flows inside the screw expander and the suction losses are described. Based on this study, a better expander can be designed.

2. CFD ANALYSIS

The geometry used in CFD calculations, is shown in Figure 1. The CFD calculations were performed for two designs, with and without additional injection ports.



Figure 1: Geometry of the twin screw expander

2.1 Grid generation

The domain for CFD calculations is decomposed into three blocks, namely inlet, outlet and injection ports. The grid in the inlet, outlet, and oil injection ports is stationary, since the only moving boundaries to the flow domain are the rotors surfaces.

The grid in the casing is built by stacking two-dimensional structured grids in slices of the casing (Figure 2). Before the CFD simulation, 2D structured meshes are constructed in slices corresponding with different rotation of profiles, with steps of 1.5 degrees of the male rotor. These 2D meshes are generated using the Laplace potential equation $\nabla^2 \phi = 0$. During the entire simulation the cells definitions are the same (the same faces and nodes). So, the nodes of the grid are moved in the housing (ALE, Arbitrary Langrangian-Eulerian method). For each time step in the calculation, a new position of the grid nodes can be found by interpolating between two supplied grids.

The size of the gaps between the rotors and between the rotors and the casing is extremely small (order tens of microns versus rotor diameter of about 70 mm). To reach sufficient numerical accuracy, the number of cells in radial direction is set to 8 (zoomed view in Figure 2).



Figure 2: Grid in the rotor domain

All blocks are joined together by using sliding interfaces between the rotor domain and stationary grid. The complete 3D mesh of the twin screw expander consists of 2060515 cells.

2.2 Boundary conditions and simulation parameters

The fluid flow inside the screw expander is compressible and is described by the set of momentum, energy and mass conservation equations, which are accompanied by the equation of state and the $k - \varepsilon$ turbulence model. The governing differential equations are solved by a commercial package with use of user-defined functions to handle the grid movement and real gas behavior.

The boundary conditions used in calculations for two different designs of the expander are presented in Table 1. During the simulations, the inlet temperature of the working fluid was $\vartheta_{in} = 126.5$ °C. Only the effects of a variation in the pressure ratio are studied. The spatial discretization that is used for the calculations is first-order upwind for the convective terms. The temporal discretization is first-order implicit. The solver calculates the pressure and velocity simultaneously.

Table 1: Boundary conditions and simulation parameters

Design	Evaluated pressure ratios	Rotors speed (male/female)	Boundary conditions	Turbulence model	Solver
Without additional injection ports	6, 5, 4	6000/4000	Pressure inlet	lt o	Coupled
With additional injection ports	6,5,4,3,2	6000/4000	Pressure outlet	К-С	

R245fa is one of the most commonly used fluids in ORC systems with a low temperature heat source. Since the ideal gas equation of state shows big deviations in the ORC working conditions as presented in Luján (2012), an appropriate real gas equation of state should be used. In this paper the Aungier Redlich-Kwong equation of state presented in Aungier (1995) is incorporated in the real gas model through user-defined functions. It represents a modified form of the Redlich-Kwong equation of state. This new correlation includes an acentric factor and the critical point compressibility factor as additional parameters to improve its accuracy. The Aungier Redlich-Kwong equation of state is defined with the following equations:

$$p = \frac{RT}{(v - \tilde{b})} - \frac{a(T)}{V(V + b_0)} \tag{1}$$

Where:

$$V = \frac{1}{\rho}, a(T) = a_0 \left(\frac{T_c}{T}\right)^n, a_0 = 0.42747 \frac{R^2 T_c^2}{p_c}$$
$$b_0 = 0.08664 \frac{RT_c}{p_c}, c_0 = \frac{RT_c}{p_c + \frac{a_0}{V_c(V_c + b_0)}} + b_0 - V_c, \tilde{b} = b_0 - c_0$$

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However, it is important that the flow conditions in every cell are always in the gas phase region. If they fall outside that region, the partial derivative of pressure at constant temperature with respect to specific volume becomes positive. Since the mesh motion is complex and the volume of some cells can be extremely small (gap region), this can cause convergence problems during the start up.

2.3 Geometrical parameters

The configuration of the rotor lobes is 4/6 (male/female). The outer diameter of the male and female rotors is approximately 70 mm.

The volume curve of the screw expander is shown in Figure 3. The formation of a chamber starts when the male and female lobes are in connection with the inlet port (zero degrees in Figure 3). As the rotors rotate, the volume of the chamber is rising together with the increase in inlet surface area. But, it is only for a short time that the surface of the inlet port is completely available for the filling (zoomed view in Figure 3). Since a significant pressure drop occurs during the filling (Section 3.1), the inlet port should be designed in such a way that filling occurs only during a small part of the growing phase of the volume with maximum inlet surface area available.



Figure 3: Volume curve with inlet and outlet surface area as a function of the rotational angle of the male rotor

3. PERFORMANCE ANALYSIS

The design with additional injection ports could also be used for the oil injected twin screw expander with small percentage of oil mixed with the working fluid and injected through the injection ports. Therefore it was of great importance to investigate the influence of the injection of additional working fluid through the injection ports. The performance analysis is done for both configurations in order to indicate how its design influences the performance of the twin screw expander.

The performance of the expander can be evaluated by comparing the mass flow rates and power output for different pressure ratios and different designs. In Figure 4, the mass flow rates and the power output (torque on the rotors walls multiplied by rotational speed) from the CFD calculations are averaged over one period (360 degrees male rotation) which corresponds with 4 expansion cycles. When increasing the pressure ratio, the output power increases together with the mass flow rate.



Figure 4: Power vs. Mass Flow Rate at different filling pressures and two different designs

3.1 Pressure-Volume Diagram

P-V diagrams for different pressure ratios and two different designs are presented in Figure 5 and Figure 6.



Figure 5: P-V diagram of the expansion process (design with additional injection ports)

In both figures, only a female working chamber is represented. Looking from the high pressure side, fluid starts to fill an expanding chamber formed between the lobes of the rotor profiles and the casing of the expander. Since it is only for a short time that the inlet is completely open for working fluid to enter the working chamber (Figure 3), throttling losses and a pressure drop will occur (first pressure drop in a P-V diagram). Closing of the inlet (decrease in inlet surface area) in connection with the increase in volume of the chamber will cause the so called pre-expansion (second pressure drop in a P-V diagram). Once the chamber is disconnected from the inlet the volume increases until the maximum volume is reached. Addition of the working fluid through the injection ports will cause an increase in pressure in a chamber, despite it's increasing volume. This affects the slope of the P-V curve. As the chamber meets the outlet port, the fluid in the chamber is discharged. Depending on the pressure ratio, over- or under-expansion can be seen (e.g. for pressure ratio 2 and pressure ratio 6, respectively).



Figure 6: P-V diagram of the expansion process (design without additional injection ports)

Comparing Figure 5 and Figure 6, it can be seen that with additional injection of the working fluid, the slope of the P-V curve can be changed in a way that it overcomes the over- and under-expansion losses.

3.2 Leakage analysis

In this paper special focus is on a study of mass flow rates through the clearances which are forming leakage paths. Depending on the internal pressure ratio, leakage flows from adjacent chambers can cause re-filling and consequently, affect the volumetric efficiency of the expander as reported in Rane (2011). On the other hand, during the suction, inlet throttling occurs because the inlet surface is blocked by the tips of the rotor and the filling is reduced, which will decrease the volumetric efficiency. Therefore, it is paramount that these leakage flows are assessed correctly.

In the twin screw expander there are four types of leakage paths. It is possible to identify every one of them as it is shown in Figure 7:

1. Clearances between the rotor **tips** and the housing allow the flow between two adjacent chambers with different pressures (m_{tip} in Figure 7).

2. Between the meshing rotors, a sealing line is formed, where leaking occurs (m_{sealing} in Figure 7).

3. **Blowholes** are triangular gaps that are formed at the cusps between two rotors and the housing ($m_{blowhole}$ in Figure 7). Because of manufacturing reasons, it is impossible to have a sealing line that reaches to the cusp of the housing. As a consequence, when looking at a 2D cross section at the end of the sealing line, a triangular region (blowhole) is formed between this end and the nearest positions of the tip clearances between both rotors and the housing.

4. Leaks through the clearances at the **end planes** occur both on the inlet and outlet side (m_{end plane} in Figure 7).



Figure 7: Different leakages during the expansion process

In this paper the focus was only on tip and sealing leakage flows. The total mass flow rates in Figure 8 and Figure 9 are the sum of all mass flows that are going through the leading and trailing clearances of the working chamber on the female side (Figure 7).



Figure 8: Leakage flows through (a) tip and (b) sealing clearance gaps (design with additional injection ports)

With an increase in pressure ratio, the internal leakage flows are rising, as depicted in Figure 8 and Figure 9. In these figures, a positive value means that the flow is leaving the current chamber. During the filling (until 80 degrees of male rotation), the outflows via both leakage paths are rising. During the expansion, the inflow through clearances increases, since the pressure in an adjacent chamber (which in this moment is connected with the inlet) is higher. If in this moment, additional working fluid is injected through injection ports (Figure 8 (a)), the outflow through clearances will increase. After the chamber is connected with the outlet, two cases are possible. In the case of high pressure ratios (between $p_{in} = 6 \text{ bar}$ and $p_{in} = 3 \text{ bar}$), the inflow is dominating (since the pressure in the adjacent chamber is still much higher than at the outlet port). In the case of over-expansion with imposed pressure of $p_{in} = 2 \text{ bar}$ at the inlet, the outflow will increase (since the pressure in the adjacent chamber is lower than the pressure at the outlet port).



Figure 9: Leakage flows through (a) tip and (b) sealing clearance gaps (design without additional injection ports)

3.3 Influence of rotational speed on mass flow through clearances

The variation of the rotational speed has a significant influence on the performance of the twin screw expander. Ideally, the output power should rise linearly with the rotational speed. However, with increase of the rotational speed, the time for filling decrease and throttling losses rise. Therefore there is a small decline (comparing to an ideal linear rise) in output power for higher rotational speed.



Figure 10: Power vs. Mass Flow Rate at different rotational speed with pressure ratio 6, and for the design with additional injection ports

The influence of the leakages is also affected by a change in rotational speed. In Figure 11, the mass flow rates for two types of leakages and for different rotational speeds are presented. It can be concluded, that with a decrease in the rotation speed the influence of the leakage flows is higher.



Figure 11: Leakage flows through (a) tip and (b) sealing clearance gaps (design with additional injection ports and pressure ratio of 6)

4. CONCLUSION

In this paper, transient 3D CFD analysis of a twin screw expander with R245fa as a working fluid is presented. The calculations are performed with the use of a grid manipulation algorithm and a commercial CFD solver accompanied by user-defined functions.

It is shown that during the filling process significant pressure drops occur. In terms of different pressure ratios, overor under- expansion can occur depending on the design of the expander. It is shown that with additional injection of the working fluid, the output power increases. In the analysis of the leakage flows, mass flow rates as the function of rotational angle are presented for two different types of leakage paths. With the increase in the pressure ratio or decrease in rotational speed the influence of the leakage flows is more significant. The study presented in this paper will be used to improve the performance of the expander and optimize the design.

NOMENCLATURE

v	built in volume ratio	V	specific volume
k	specific heat ratio	P	pressure
π	pressure ratio	T	temperature
n	rotational speed	B	gas constant
n	rotational speed	R	gas constant

Subscript

c critical value

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