

Interface Challenges in Dense Oxide as well as Non-oxide CMCs for Long-Term Applications

Precursor Derived Ceramics and Advanced Hybrid Materials
Sino-German Symposium
Freiberg, Germany, November 6-11, 2016

Martin Frieß, Bernd Mainzer, Sandrine Hönig, Enrico Klatt, Dietmar Koch

German Aerospace Center (DLR), Stuttgart, Germany

Institute of Structures and Design

Department: Ceramic Composites
and Structures (KVS)

Contact:

martin.friess@dlr.de



Knowledge for Tomorrow



The German Aerospace Center (DLR)

Largest aerospace research facility in Europe

- 7800 employees
- 32 institutes
- 16 sites in Germany
- Offices in Brussels, Paris, Tokio, Washington

- Turn over: 1769 M€

- Space program 971 M€

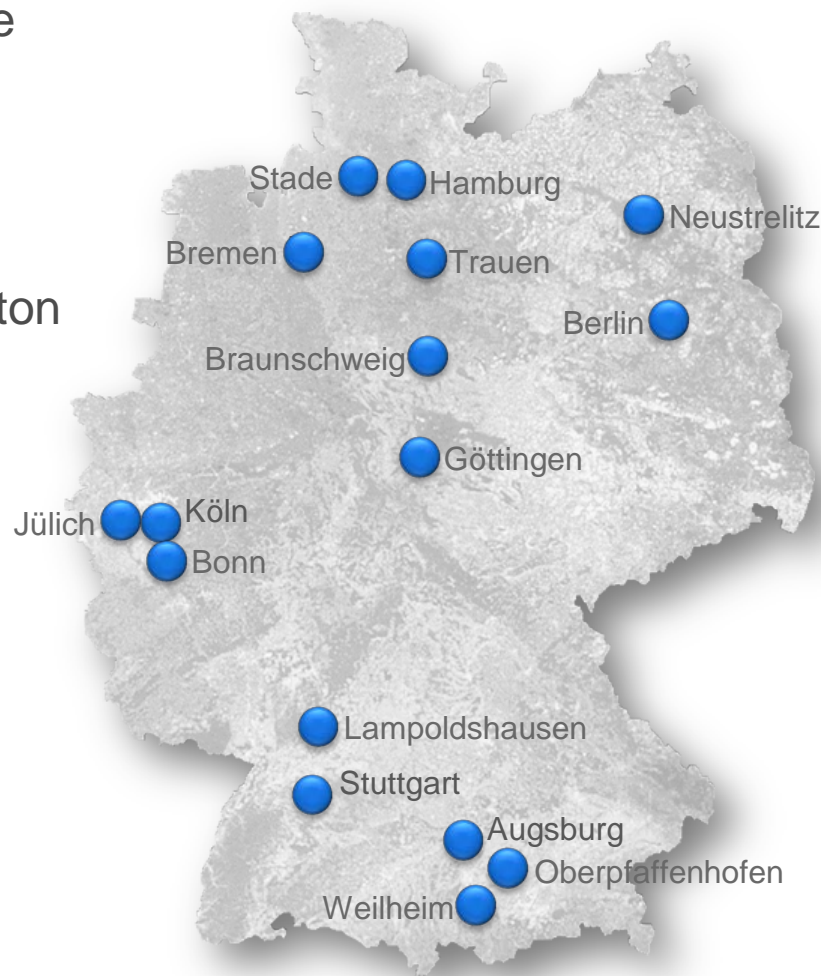
- Research 798 M€

- Space 50%

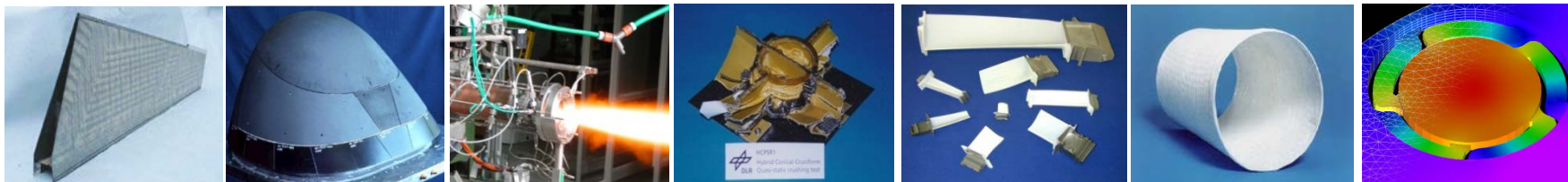
- Aeronautics 34%

- Transport 7%

- Energy 9%



Mission in Research and Education



Research

- High performance materials and structures and relevant processes
- Virtual tools for computational materials engineering and structural design
- Complete engineering chain - from material to automated production

Transfer

- Application focused and close to industry

Education

- Professor chairs and lectures at universities
- Under-graduate and doctoral students

Consulting

- Politics and industry



Outline and Motivation

- Short review on classical CMC manufacturing methods
- Interfaces (interphases) in oxide and non-oxide fibre-reinforced CMCs
- Monazite fibre coating process in principle
- Variation of coating parameters
- Manufacture of CMCs (SiC/SiCN) using PIP and monazite coating
- Physical and mechanical properties of CMCs (SiC/SiCN) and OXIPOL
- HVOF-test and microstructural characterisation of CMCs (SiC/SiCN) before and thereafter
- Summary and outlook



Classic methods to manufacture SiC/SiC(N) composites

CVI

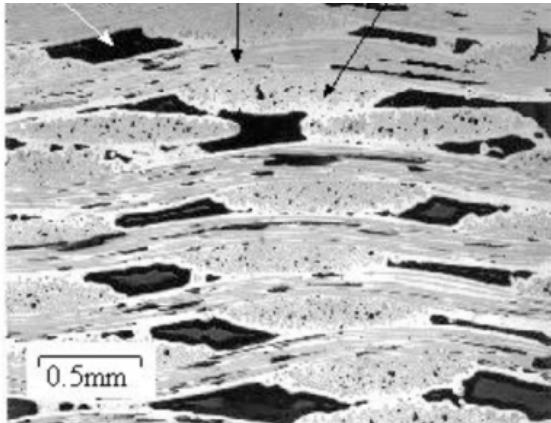
Chemical Vapour Infiltration

Advantages:

- comparably low process temperatures ($\approx 1100\text{ }^{\circ}\text{C}$)
- stoichiometric process is feasible to SiC composites
- fibre coating can be / is part of CMC manufacture

Disadvantages:

- high matrix porosity, large voids in gussets
- release of partly aggressive reaction products
- time-consuming processing to decrease porosity
- Difficulty to process thick-walled parts



Jacques Lamon, 2005

PIP

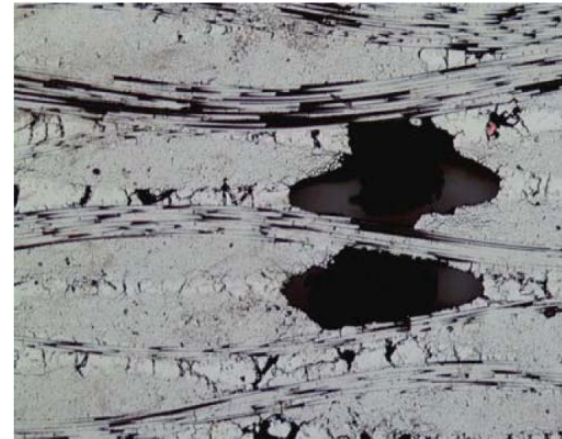
Polymer Infiltration & Pyrolysis

Advantages:

- comparably low process temperatures ($\approx 1100\text{ }^{\circ}\text{C}$)
- matrix composition can be easily influenced by chosen polymer (SiC, SiCN, SiBCN, etc.)

Disadvantages:

- high porosity by loss of mass and increase of density of the matrix during pyrolysis in gussets: in advantage are amorphous matrices (e.g.: SiCN and SiBCN)
- time-consuming processing to decrease porosity



Kazuaki Nishiyabu, 2007



Liquid Silicon Infiltration Process (LSI) at DLR

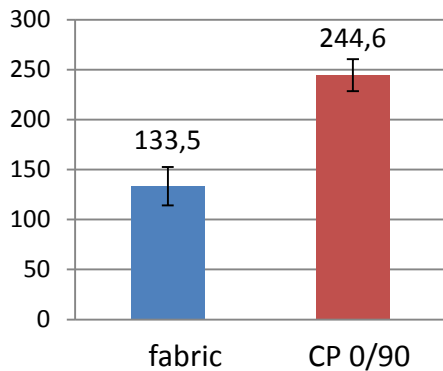
Advantages and requirements for processing

- feasibility of manufacture of shapes and geometries (no limitation to thickness of parts)
- no fibre coating necessary for achievement of damage tolerant behaviour
- intermediate processing (easy machining and substance joining) feasible
- short processing time
- “dense” matrix

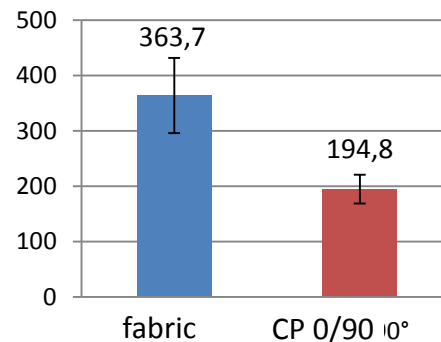
Requirements for future SiC_f/SiC(N) composites

- (near) no porosity
- stoichiometric SiC matrix (no excess of C)
- resistance to corrosion at high temperatures
- necessity of coatings for fibre protection versus chemical attack by liquid Si (or metals)

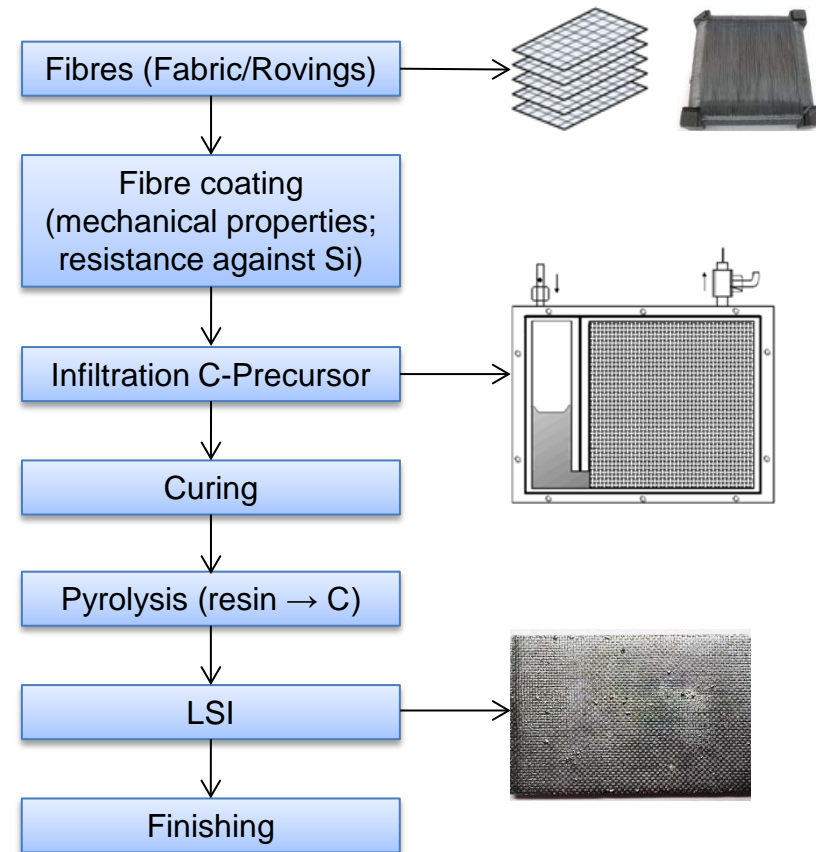
4PB Strength (MPa)



Young's Modulus (GPa)



uncoated SiC fibers



Interfaces in Oxide and Non-oxide Fibre-reinforced CMCs

Interphases based on pyrolytic carbon or boron nitride and dense matrix

- o.k. for short term applications

Interfaces based on porous matrices with no coating

- o.k. for low gas flow and low requirement for gas tightness

Interphases based on fugitive coating and dense matrix

- o.k. for short term applications

All interphases above suffer from:

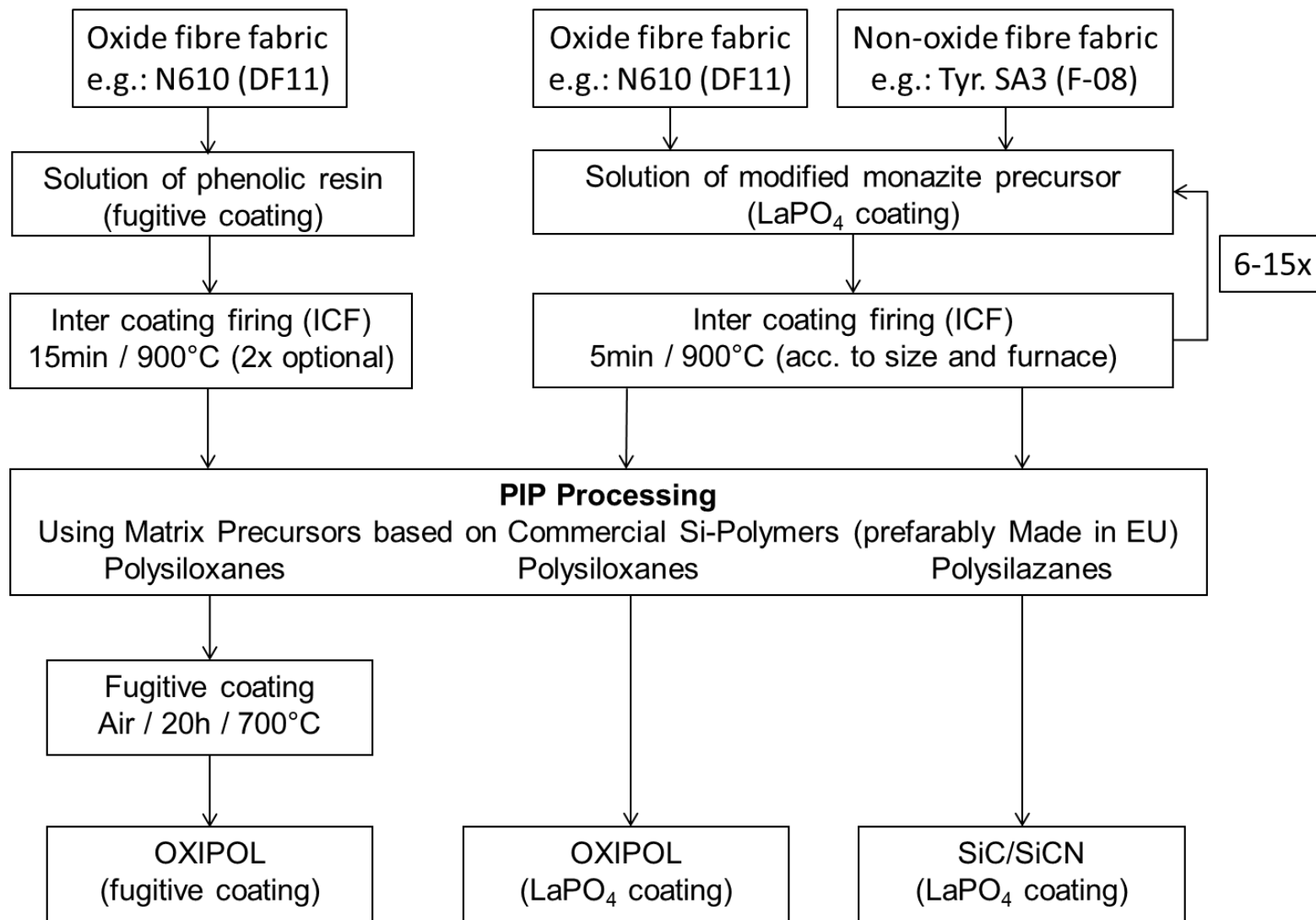
- severe attack on matrix and fibre in aggressive chemical environment
- intensive protective coatings necessary (EBC, TBC)

➤ **CMCs based on stable interphase and dense matrix are mandatory**

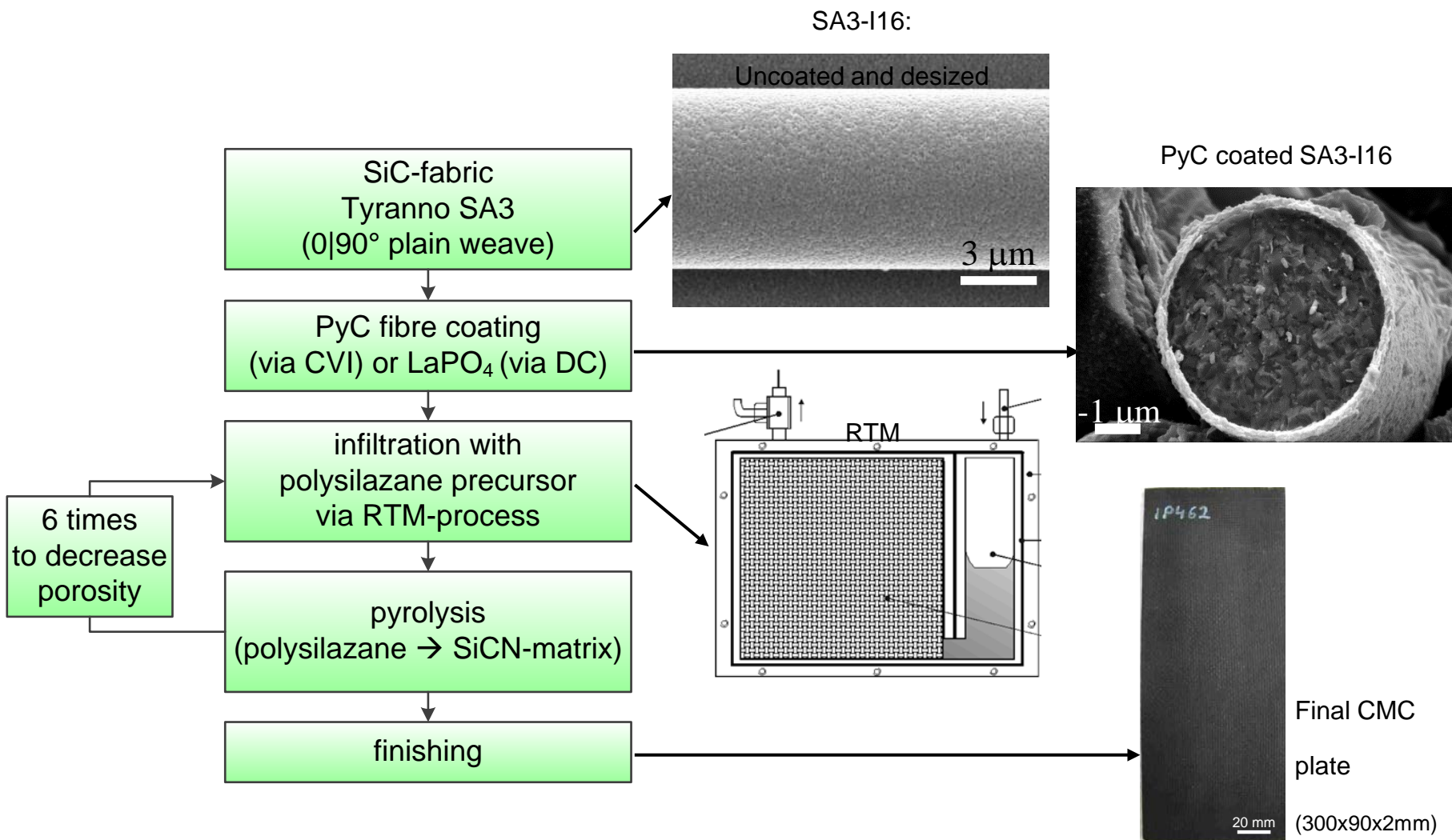
➤ **Monazite (lanthanum phosphate, LaPO_4) as a model candidate**



Manufacture of OXIPOL and SiC/SiCN Using PIP-Processing



Polymer Infiltration and Pyrolysis



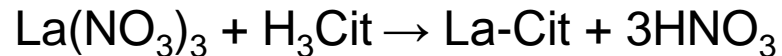
Monazite Fiber Coating Process in Principle

Monazite as oxidation resistant fibre coating for dense CMCs:

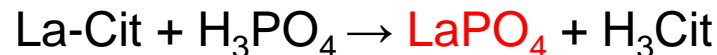
- high melting point: 2073 °C
- low modulus: \approx 130 GPa
- thermodynamically stable and chemically inert to many materials
- low bonding strength to common oxide fibres

Precursor solution made of $\text{La}(\text{NO}_3)_3$, citric (H_3Cit) and phosphoric acid (H_3PO_4)

Stabilisation and inhibition of reaction by citric acid via complex formation:

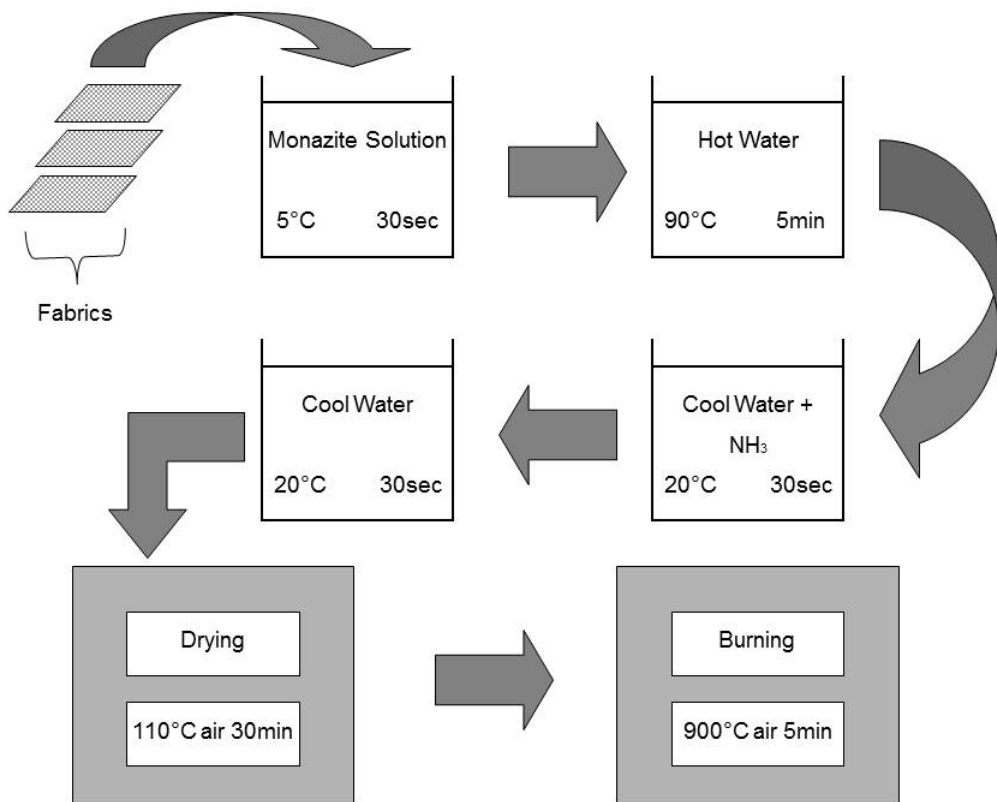


The precipitation reaction in the precursor solution only depends on temperature and concentration, and therefore, can be controlled: **heterogeneous nucleation**

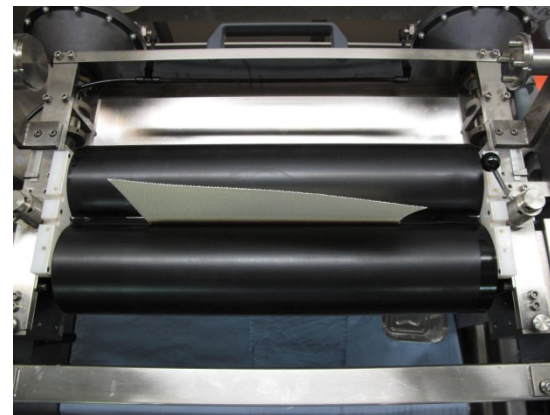
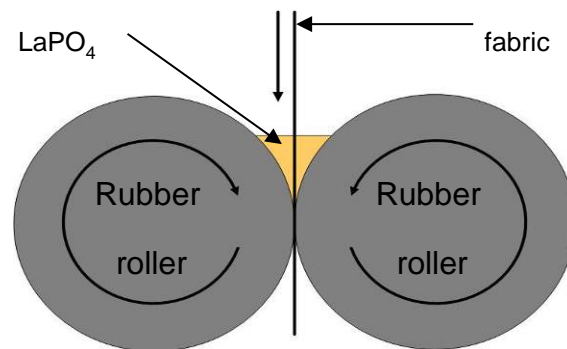


Coating Methods:

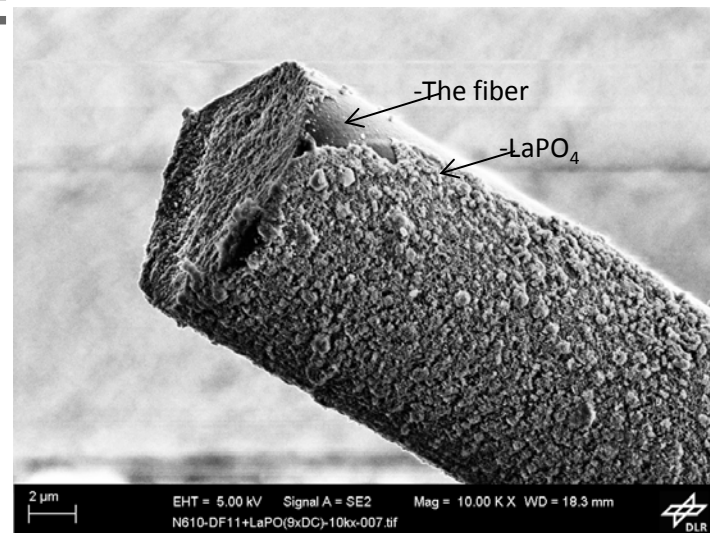
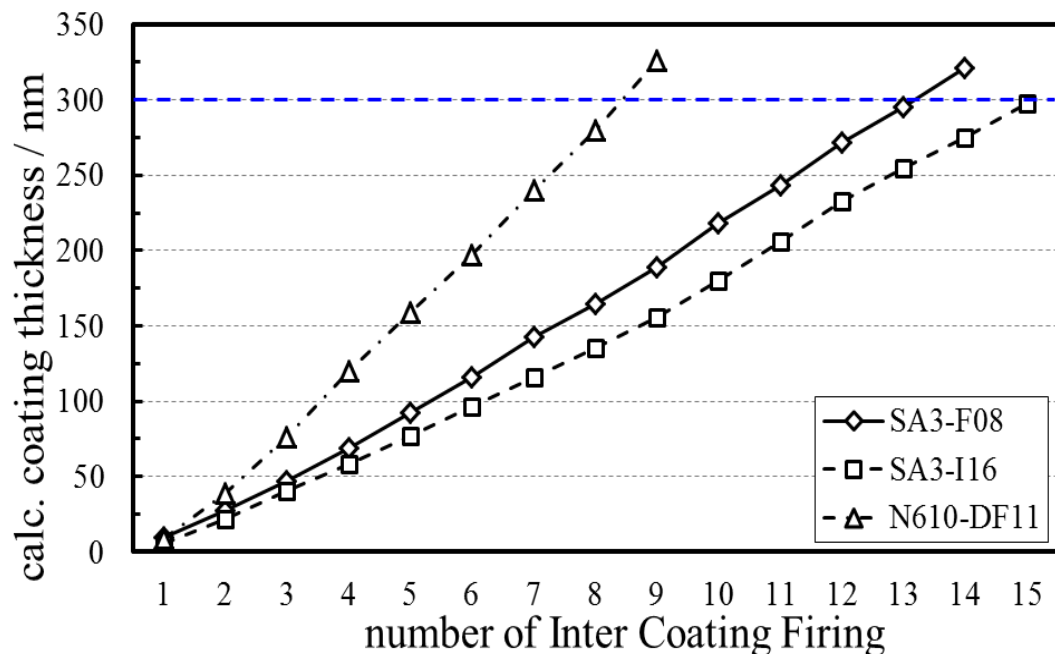
Dip coating (classic)



Foulard Coating (advanced)



Variation of Coating Parameters (1): DC results



SEM picture of Nextel610 fibre after 9 loops

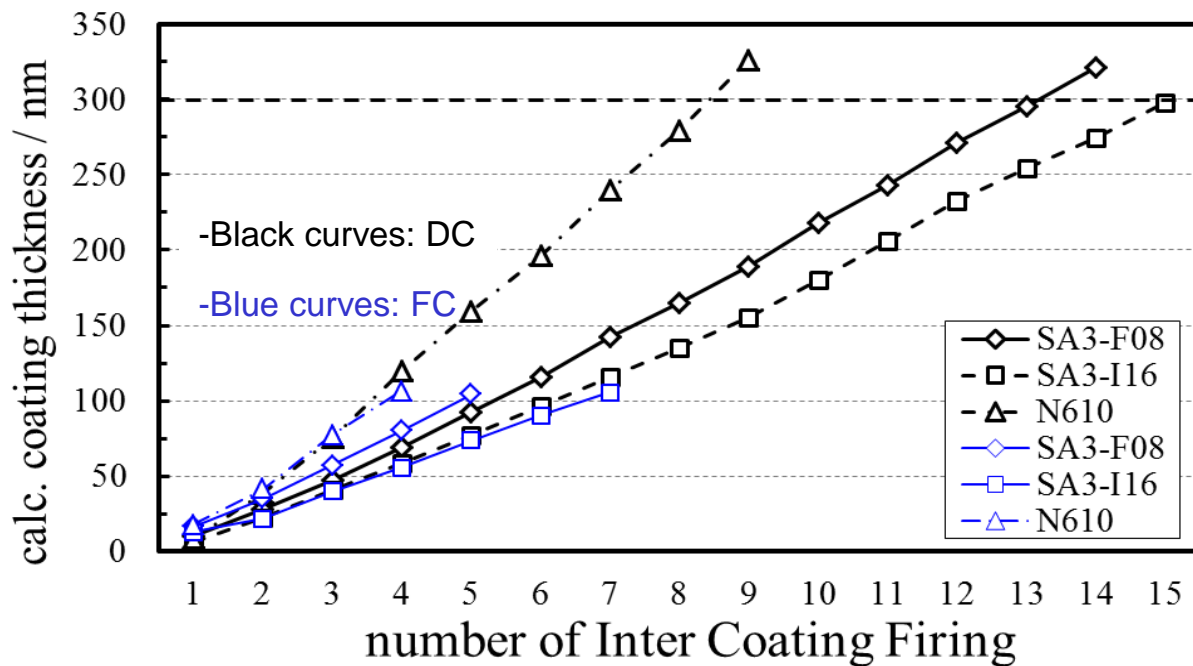
	Fibre diameter	No. of filaments
SA3-I16	7.5	1600
SA3-F08	10	800
N610-DF11	12	400

conclusions:

- the thicker the fibre diameter, the thicker the coating
- the smaller the number of filaments/roving, the thicker the coating

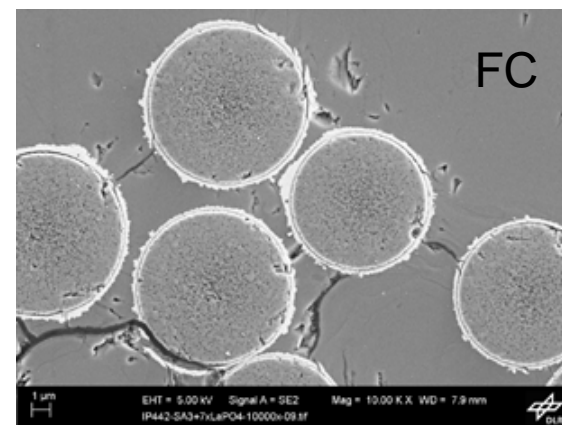
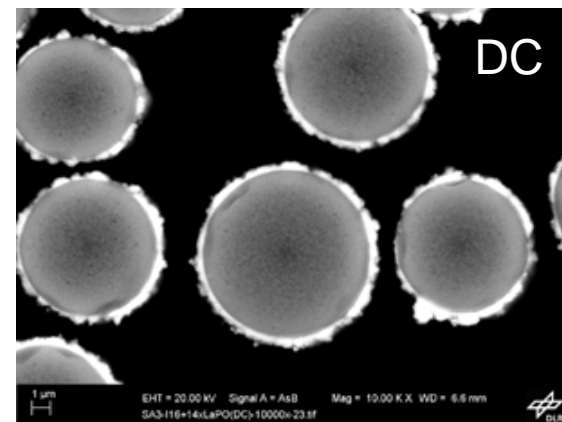


Variation of Coating Parameters (2): Comparison between DC and FC



- DC target: 300nm

- FC target: 100nm



- DC and FC (needs 2 coating cycles) are very similar w.r.t. coating thickness!!!
- FC provides a more homogeneous coating than DC!

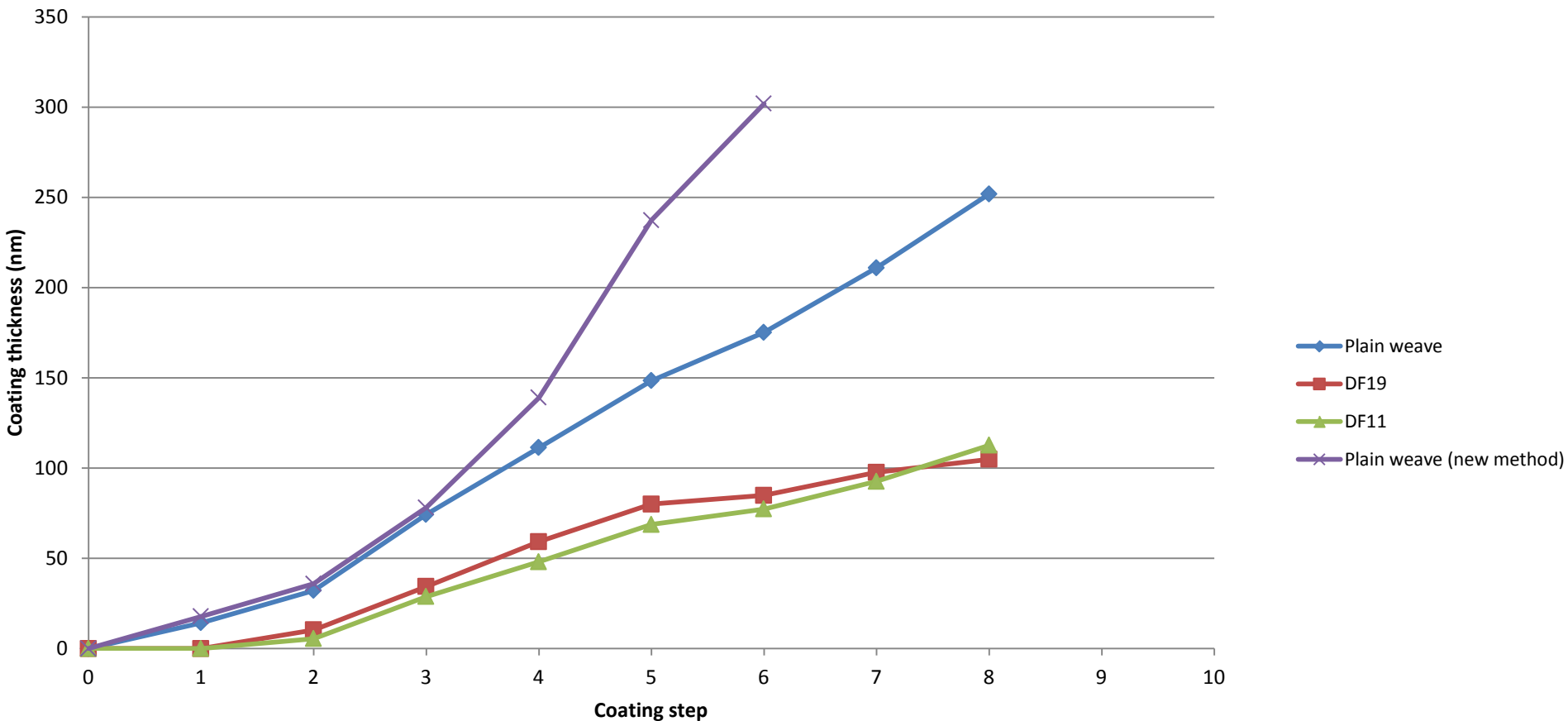


Improved fibre coating processing for fabric types Al_2O_3 (Nextel 610) and SiC (Tyranno SA3)

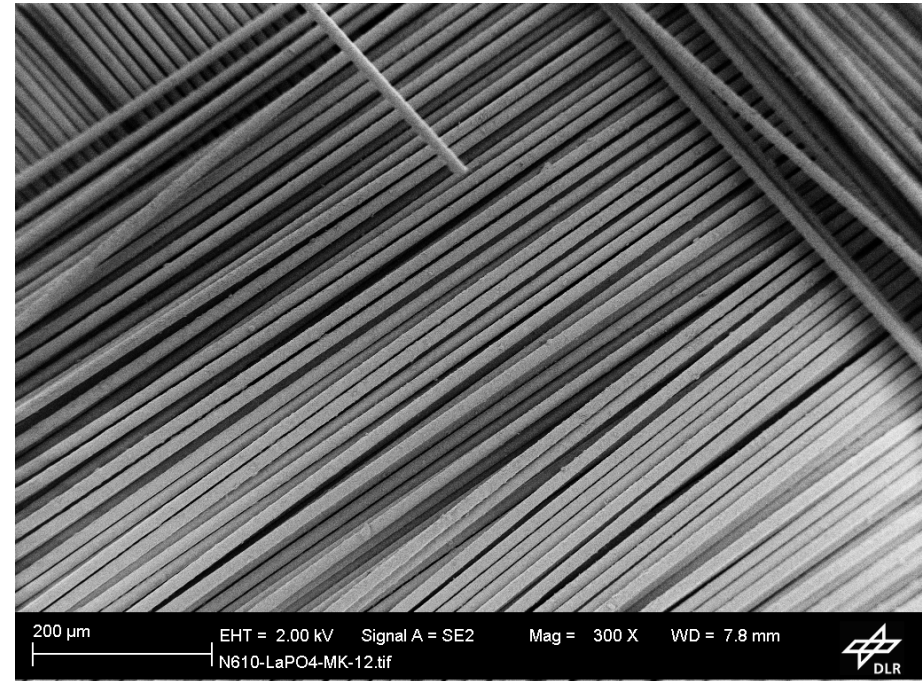
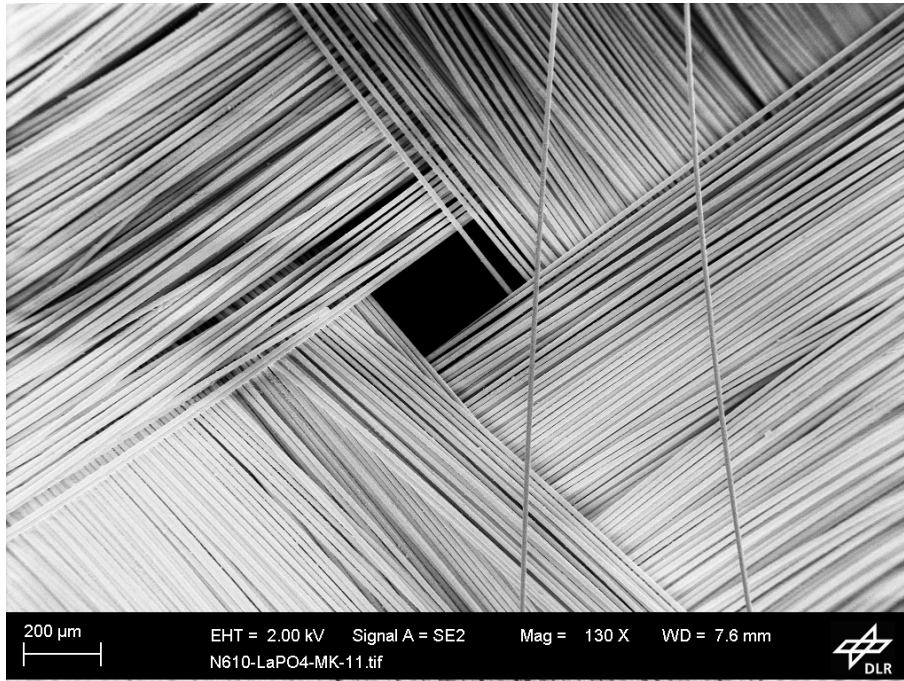
- State-of-the-art: dilute solutions in order to achieve homogeneous coatings:
disadvantage: many time and cost consuming cycles needed for
thick coatings
- Further development at DLR by applying more concentrated solutions via
Foulard technique
- This work shows that Foulard technique using almost saturated solutions
is feasible and provides promising results as well as potential:
 - homogeneous fibre coatings with very low fibre bridging
 - Thick fibre coating steps achievable by low number of cycles
 - Foulard technique has potential for cost-effective fibre coating



Calculated coating evolution from mass gain with respect to fabric type (Al₂O₃)



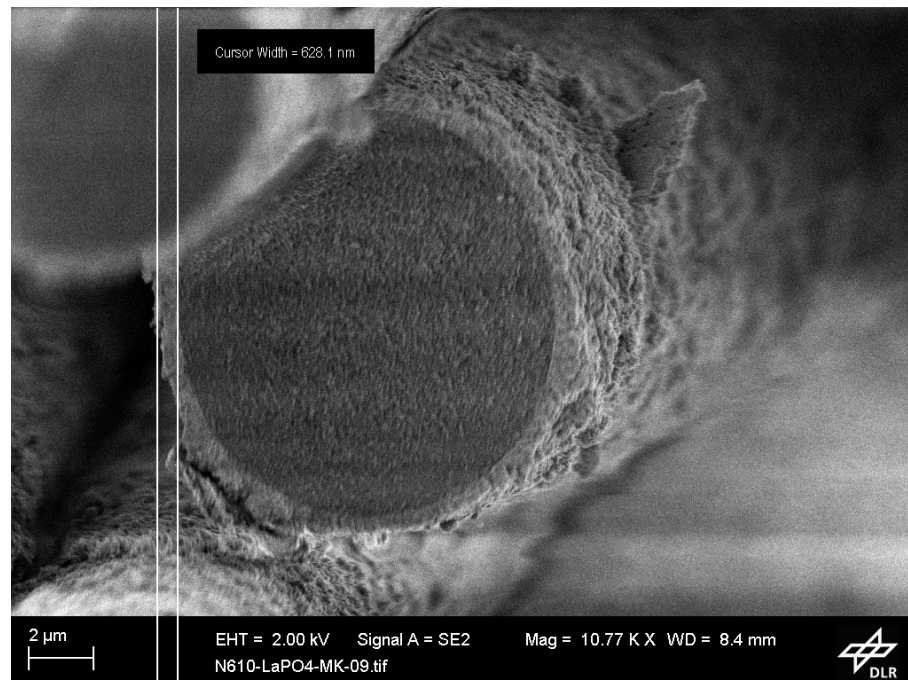
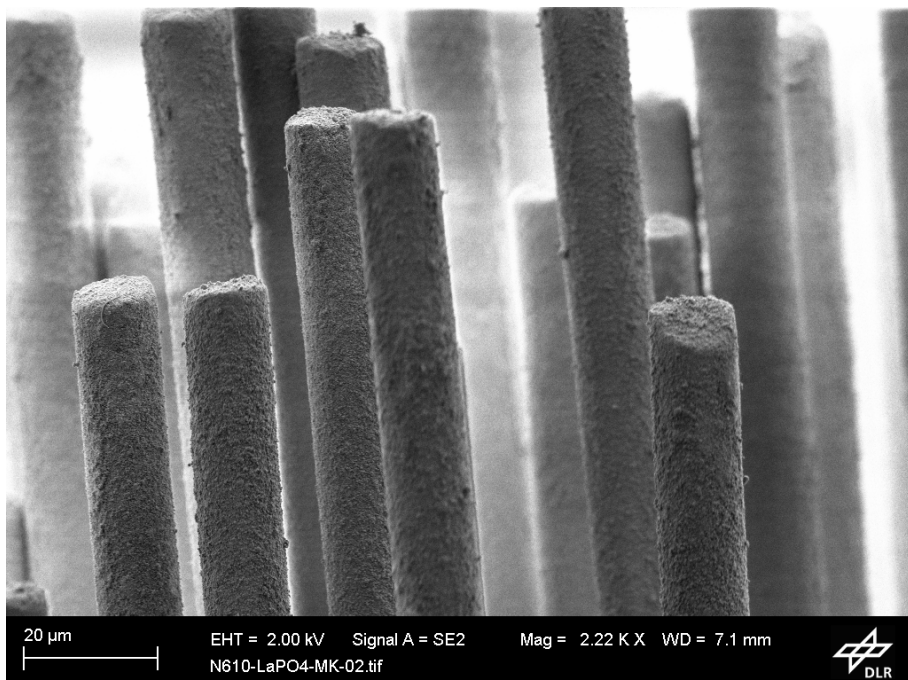
LaPO₄ fibre coating on Al₂O₃ (Nextel 610) I



- **No bridging by fibre coating**
- **Single fibre coating**



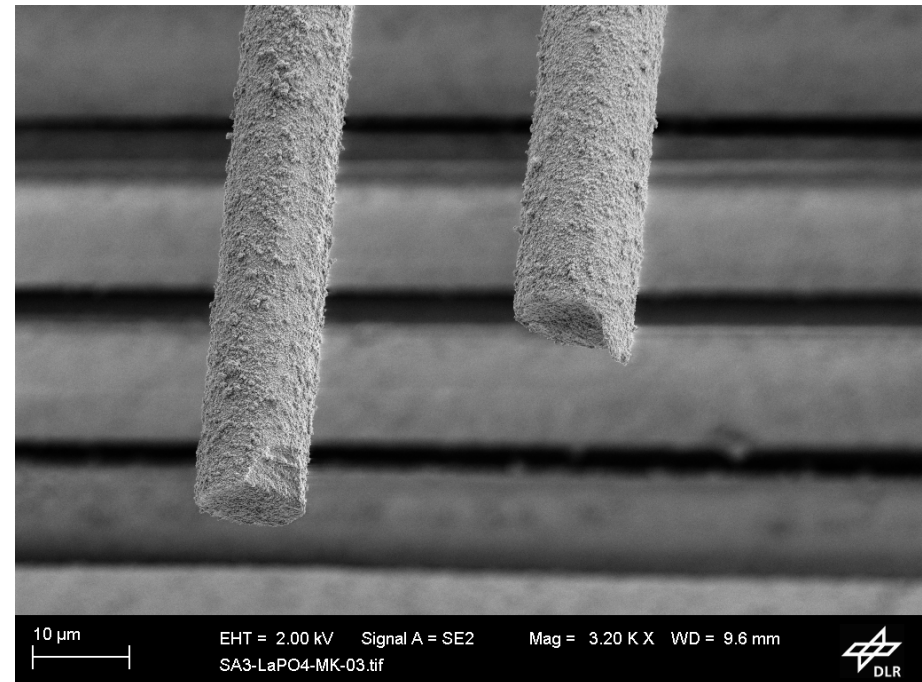
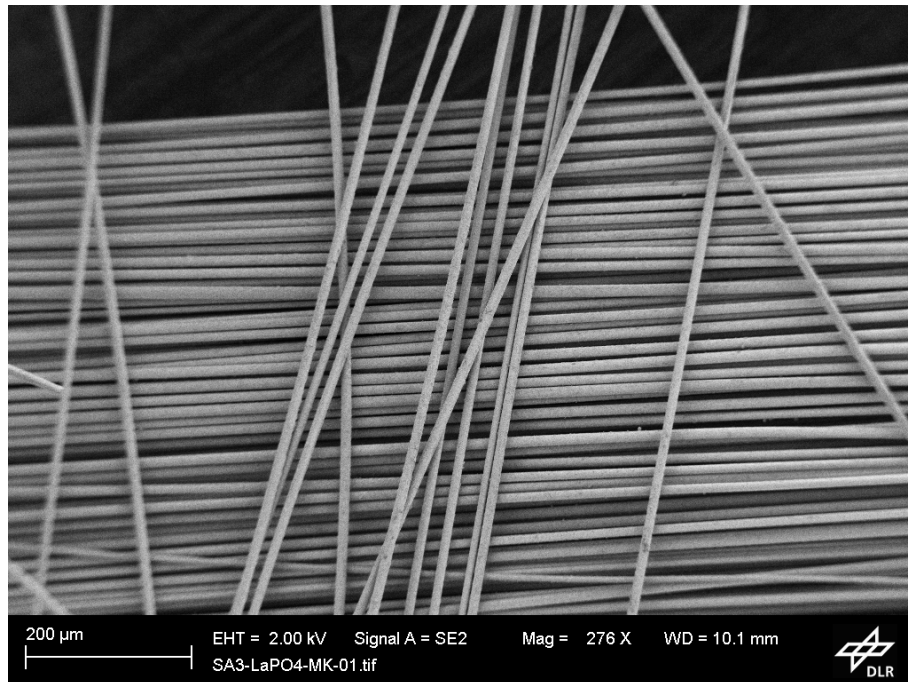
LaPO₄ fibre coating on Al₂O₃ (Nextel 610) II



-Thick homogeneous fibre coatings (628 nm) with six cycles!



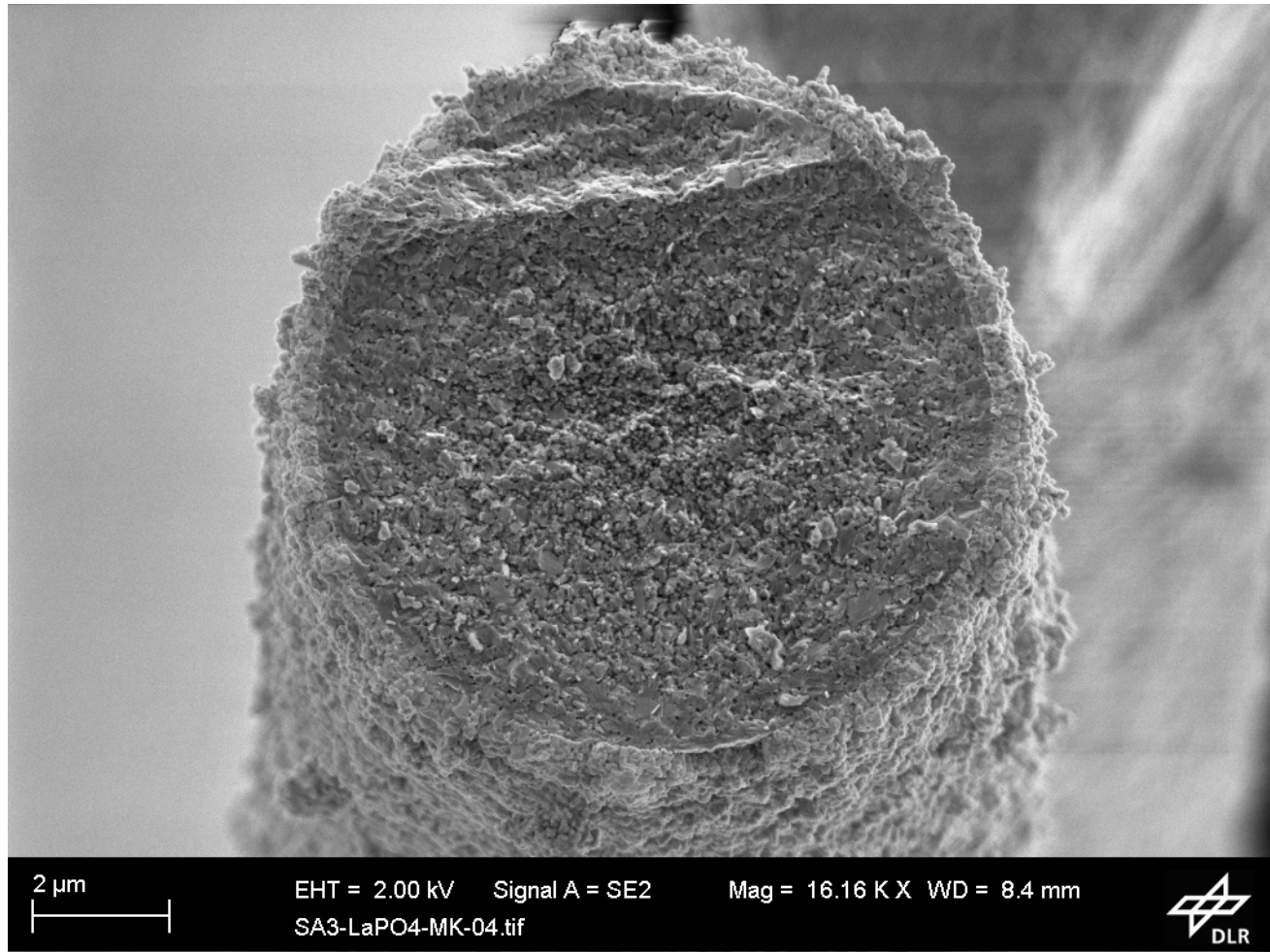
LaPO₄ fibre coating on SiC fabrics (SA3) I



- **No bridging by fibre coating**
- **Single fibre coating**



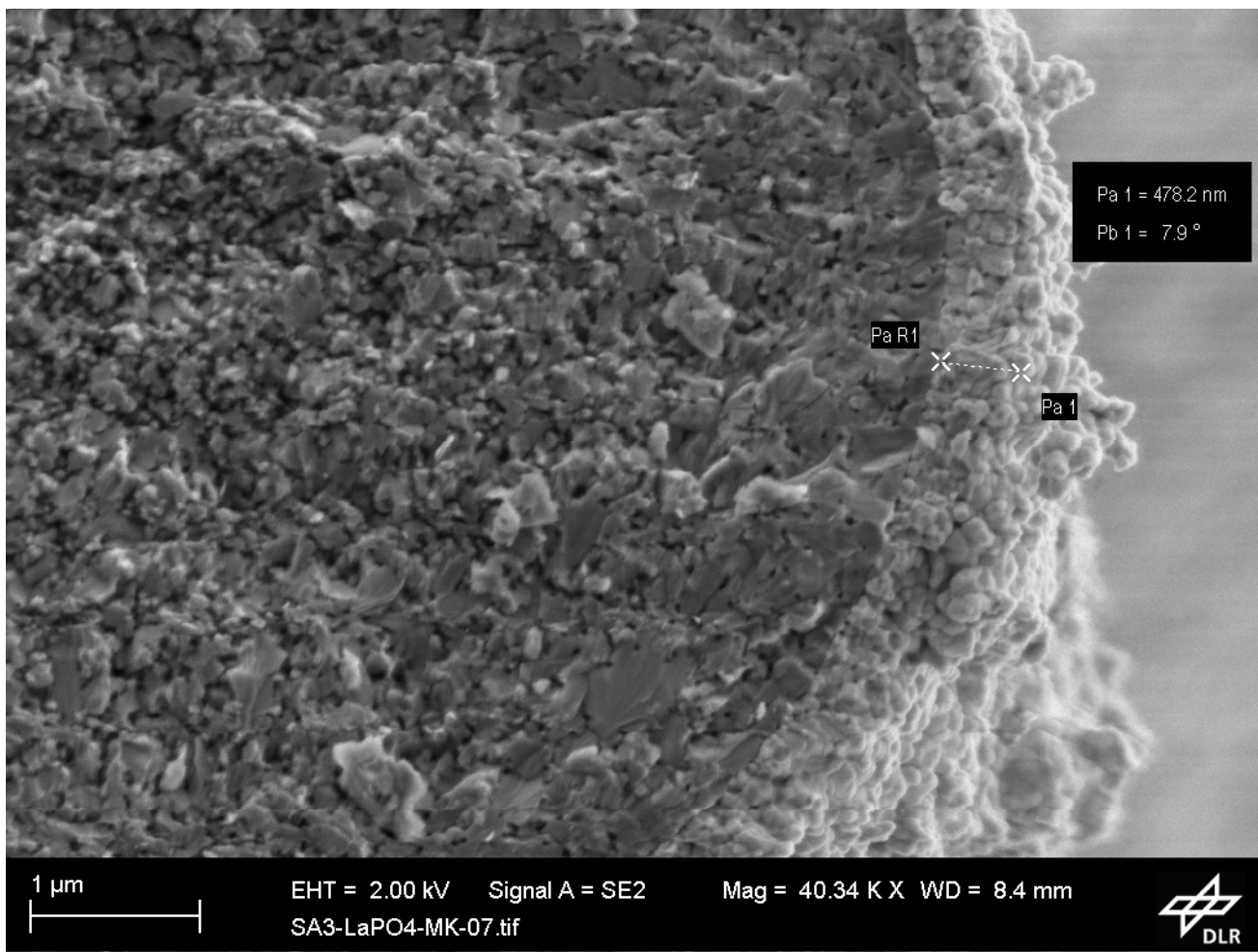
LaPO₄ fibre coating on SiC fabrics (SA3) II



- **Homogeneous fibre coating**



LaPO₄ fibre coating on SiC fabrics (SA3) III



-Thick homogeneous fibre coatings (478 nm) with six cycles!



Properties of OXIPOL (A-D) and SiC/SiCN (E)

Specimen	A	B*	C	D*	E**
Coating type	LaPO ₄	LaPO ₄	-	fugitive	LaPO ₄
Coating loops [-]	9	4	0	2	9
Coating thickness [nm]	300	100	0	<100	300
Fibre volume content [%]	43	45	42	51	45
Fabric type (Nextel 610 fibre) (Tyranno SA3)	DF11 -	DF11 -	DF11 -	DF19 -	- F-08
Aerial weight [g/m ²]	373	373	373	654	240
Density [g/cm ³]	2.77	n.d.	2.88	2.76	2.51
Open porosity [%]	7.0	16.6	6.1	11.7	6.3
Flexural strength (3-point) before exposure [MPa]	132.4	169.5	87.4	197.9	167***
Flexural strength (3-point) after exposure (1100°C, air, 20h) [MPa]	≥ 98.3	≥ 73.3	94.3	117.0	129***
Failure mode after exposure	shear	shear	tensile	tensile	tensile

*S. Hönig et al.: 36th Int. Conf. on Advanced Ceramics and Composites (ICACC) 2012.

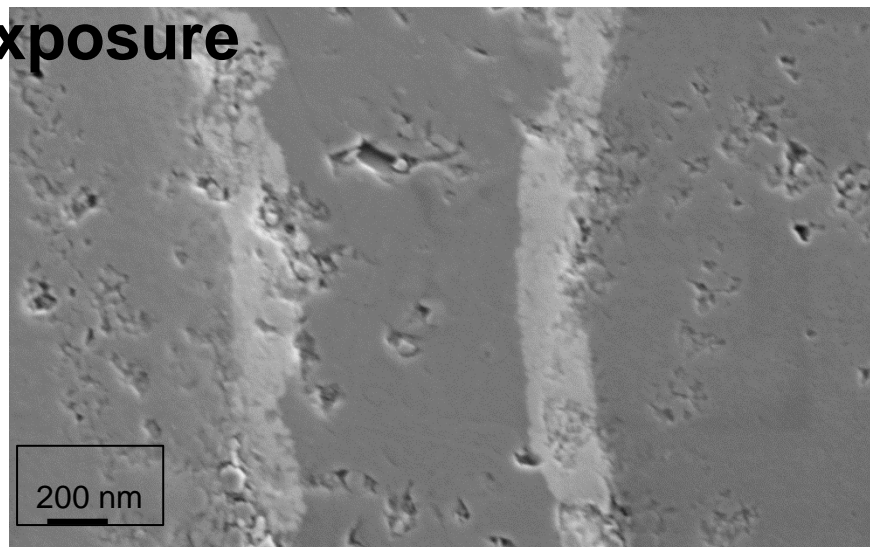
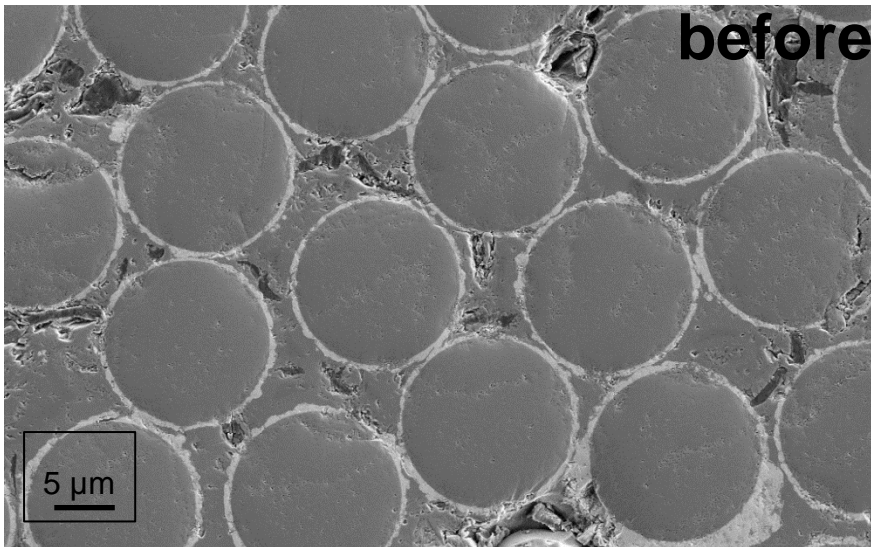
**E. Klatt: PhD Thesis DLR Stuttgart 2013

***measured according DIN ENV 658-3 (l/d=20)

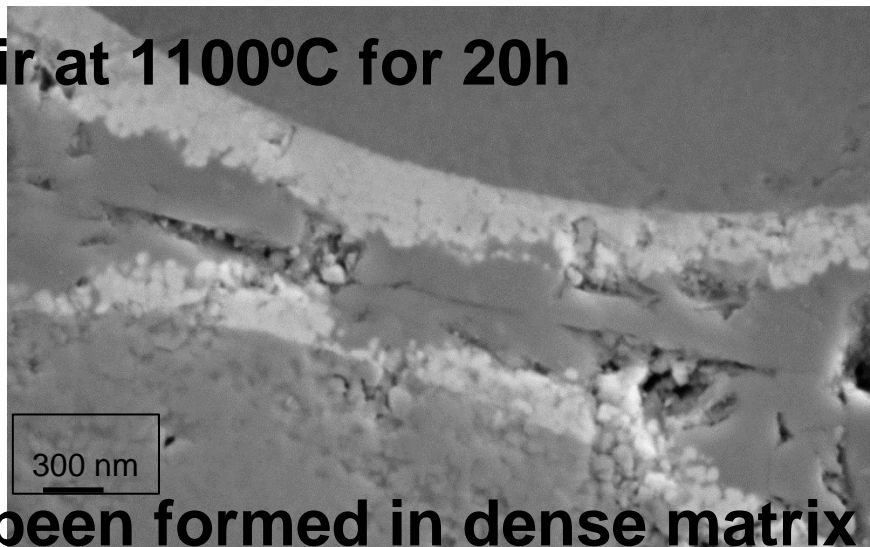
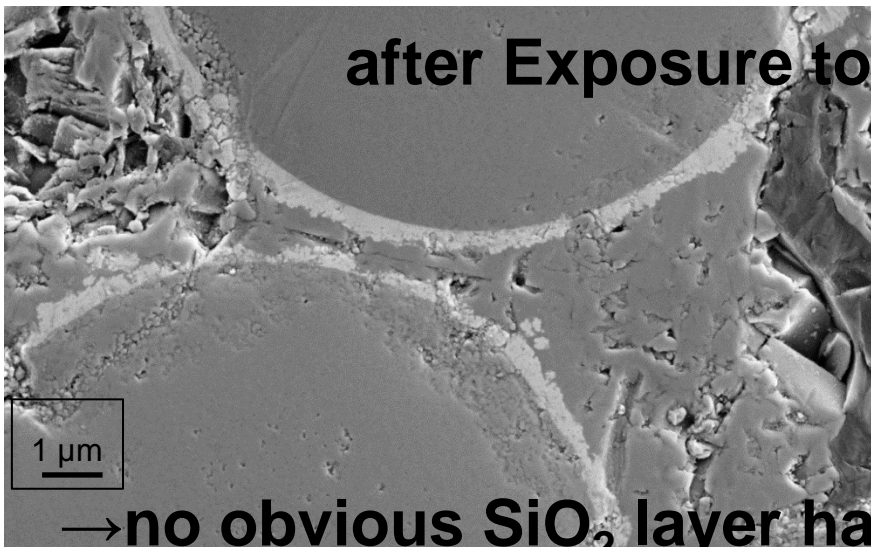


SEM of Polished Surface of OXIPOL with LaPO_4 coating

before Exposure



after Exposure to Air at 1100°C for 20h



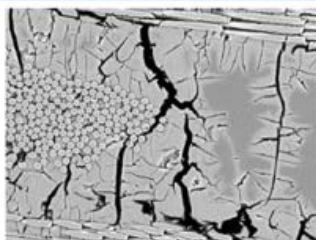
→ no obvious SiO_2 layer has been formed in dense matrix



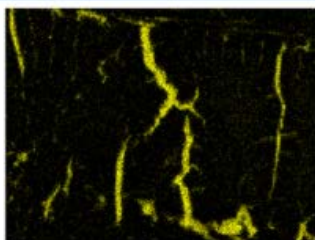
EDX-Analysis of Oxidized OXIPOL @ 1100°C for 20 h in Air

IP477-3

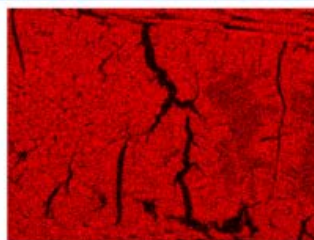
06.06.2012 10:17:22



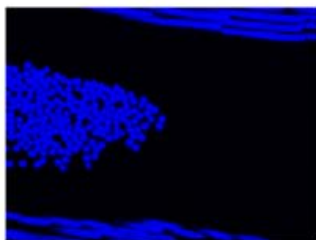
Elektronenbild 1



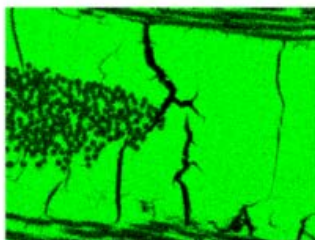
C Ka1_2



O Ka1



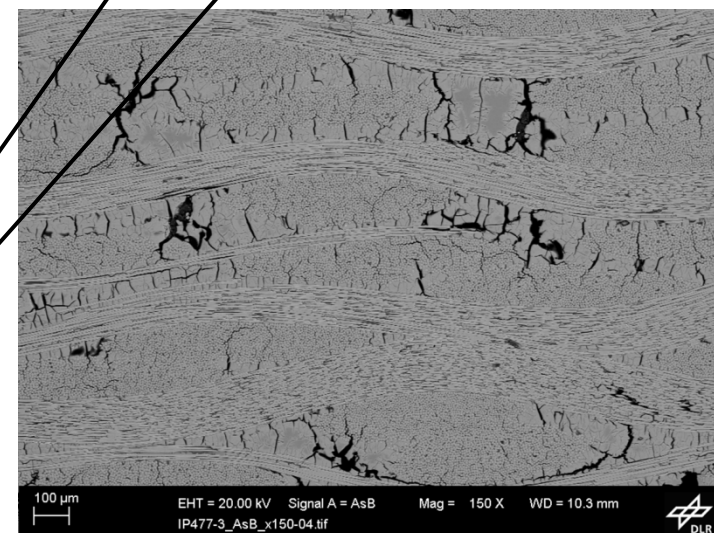
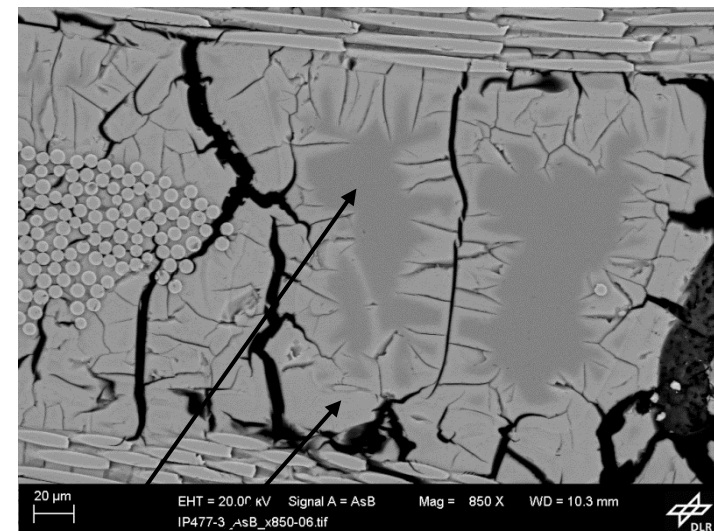
Al Ka1



Si Ka1

confirmation by
elemental
mapping

- matrix composition w.r.t. EDX: → „SiO₂C“
- oxidized matrix composition w.r.t. EDX: → „SiO₂“



HVOF-Test in combustion environment



-HVOF-Test

-(High Velocity Oxygen Fuel)

-by EADS Astrium

(ATLLAS II-project)

- represents combustion typical conditions like combustion chemistry / combustion products, temperature, gas velocity
- test conditions in a temperature range of 1000 °C up to 2050 °C possible
- the HVOF-flame was fueled by Kerosene/Oxygen



Microstructural Characterisation of uncoated C/C-SiC @ T



-C/C-SiC, uncoated, untested



-1500 °C, 1h

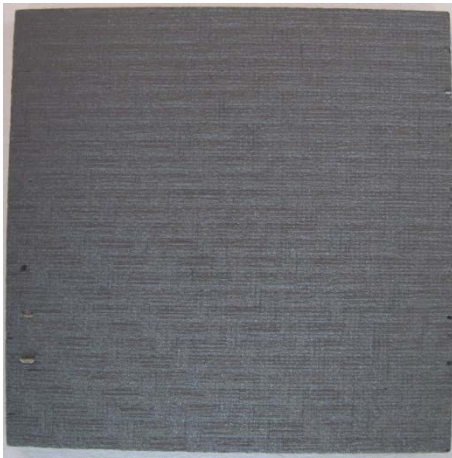


-1700 °C, 1h

- both tested samples are very brittle after exposure
- heavy degradation of the sample tested at 1700 °C



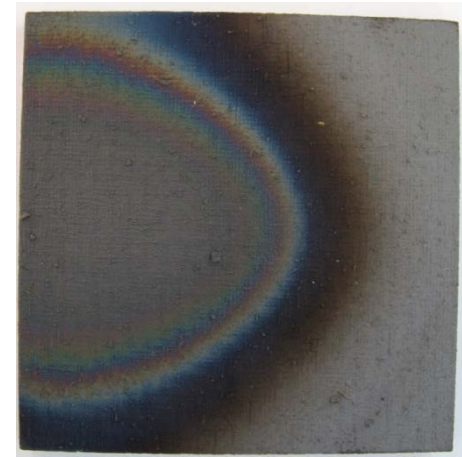
Visual Characterisation of CVD-SiC coated C/C-SiC @ T



-C/C-SiC, SiC-coated, untested



-1500 °C, 4h



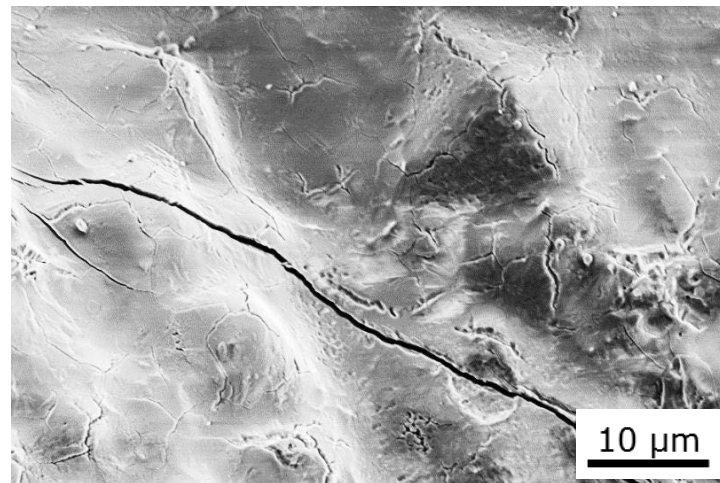
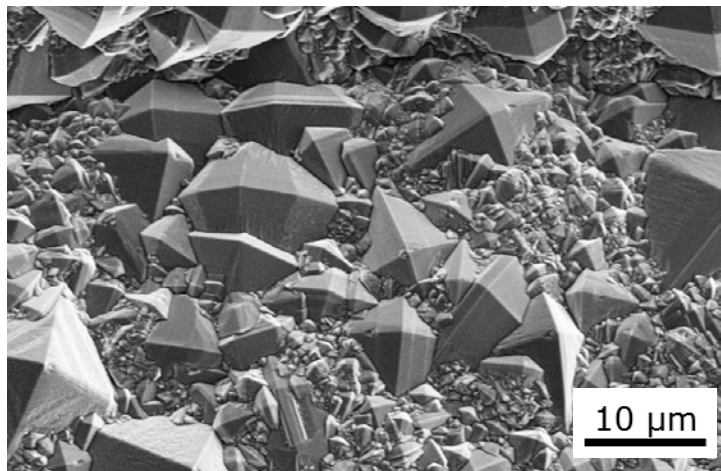
-1700 °C, 1h

- the SiC-coated samples stay intact even at 1700 °C
- some degraded spots can be observed



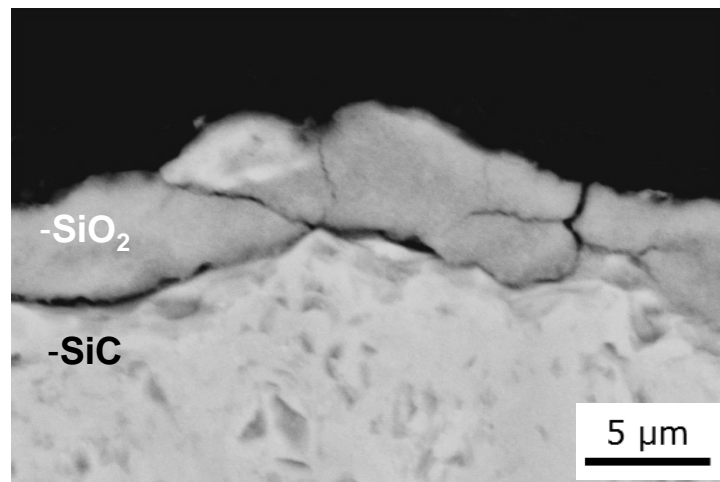
Microstructural Characterisation of C/C-SiC @ 1500 °C

-1500 °C, 4h,
-surface, top view



-C/C-SiC, SiC_{CVD} coating
-(untested sample)

-1500 °C, 4h,
-surface, side view

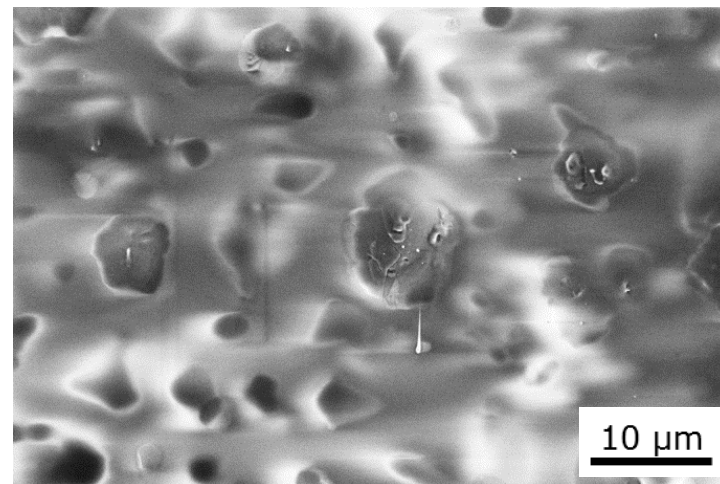
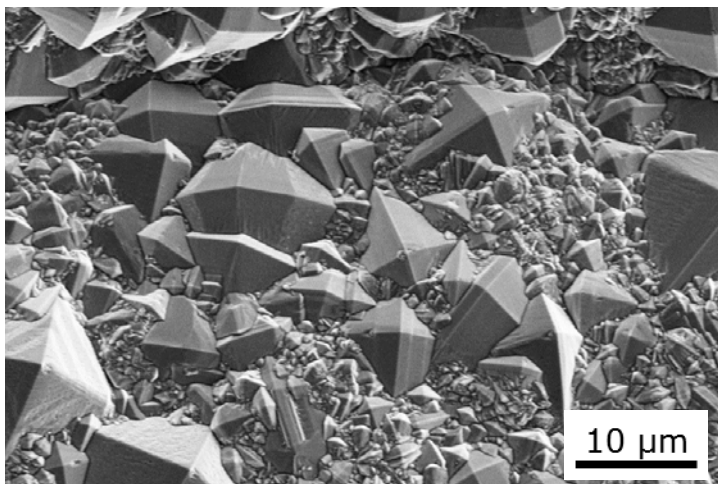


⇒ **Formation of a 5 µm SiO₂-scale**



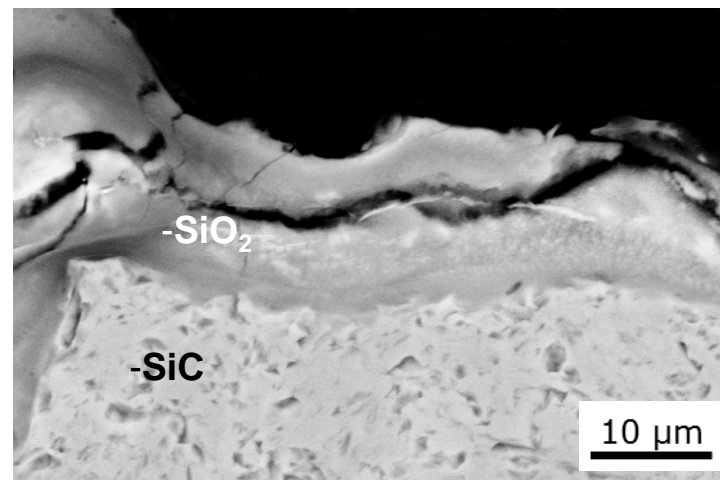
Microstructural Characterisation of C/C-SiC @ 1700 °C

-1700 °C, 1h,
-surface, top view



-C/C-SiC, SiC_{CVD} coating,
-(untested sample)

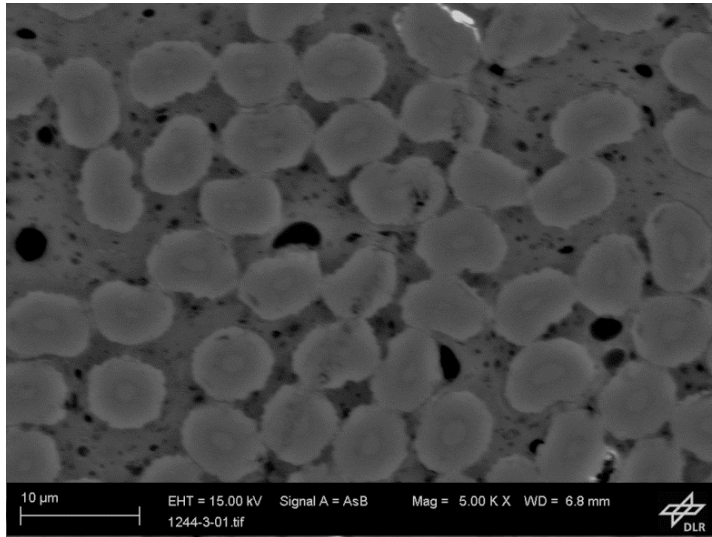
-1700 °C, 1h,
-surface, side view



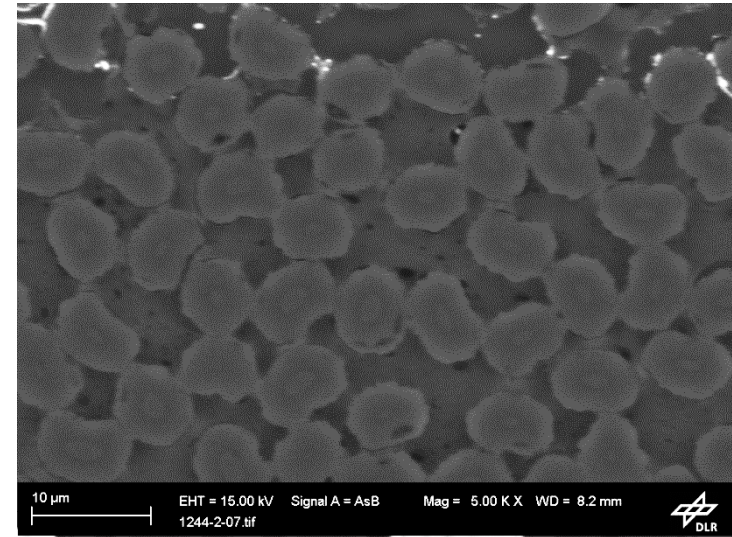
⇒ **Formation of a 10-20 μm SiO₂-scale**



Microstructural Characterisation of C/C-SiC @ T



-C/C-SiC, SiC_{CVD} coating,
-(untested sample)



-C/C-SiC, SiC_{CVD} coating,
-1700 °C, 1h

⇒ **C-matrix is not affected**



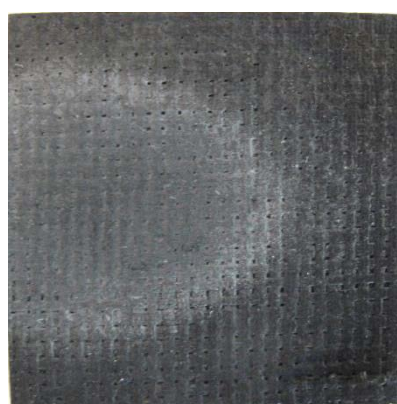
Visual Characterisation of SiC_{pyC}/SiCN @ T



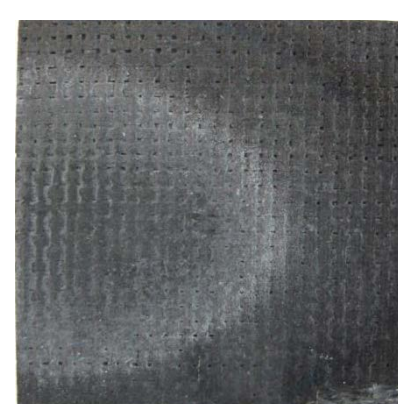
-untested



-1300 °C, 4h



-1500 °C, 4h



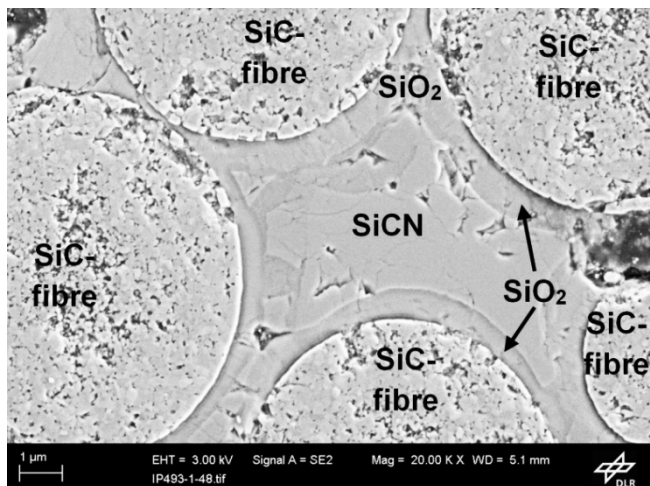
-1700 °C, 4h

e' before exposure [%]	7.04	6.08	6.06
e' after exposure [%]	11.26	11.74	11.31
Δm/m [%]	-2.8	-3.3	-3.3

- **Increased porosity after exposure (independent from test temperature)**
- **no spallation of the surface**

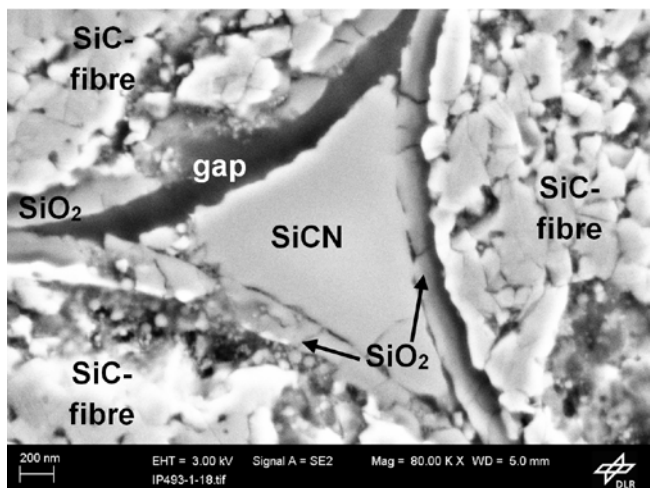


Microstructural Characterisation of SiC_{pyC}/SiCN @ 1300 °C



- Close to surface

- PyC coating was removed due to active oxidation
- Occurring gaps were completely filled by SiO₂



- Center of sample

- PyC coating also was removed
- Occurring gaps were starting to get filled by SiO₂
- @ higher T similar microstructure in center of composite like on surface



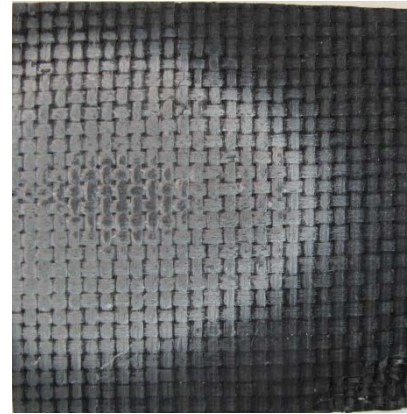
Visual Characterisation of $\text{SiC}_{\text{LaPO}_4}/\text{SiCN}$ @ T



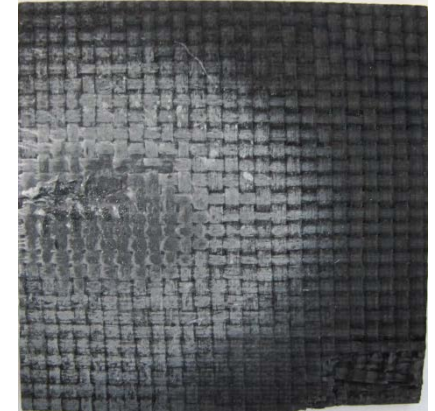
-untested



-1300 °C, 4h



-1500 °C, 4h



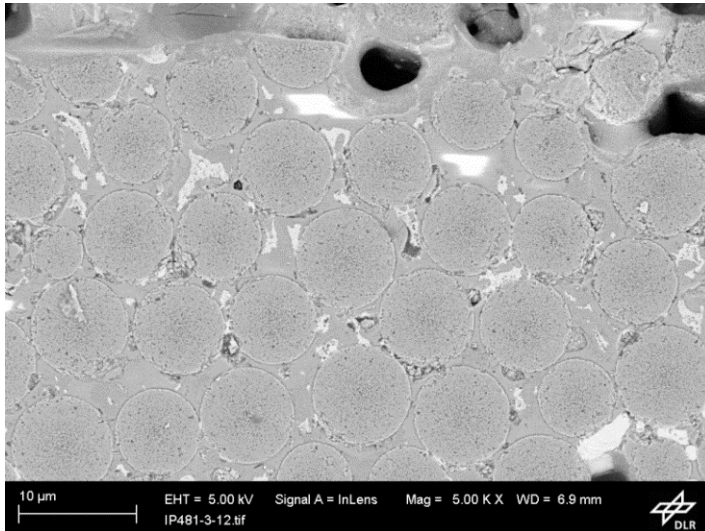
-1700 °C, 1.5h

e' before exposure [%]	11.16	12.12	12.24
e' after exposure [%]	13.21	16.97	16.47
$\Delta m/m$ [%]	-1.8	-2.4	-2.7

- only slight increase of porosity at 1300 °C
- strong increase of porosities at 1500 °C and above
- 1700 °C test aborted after 1.5 hours due to spallation of the surface

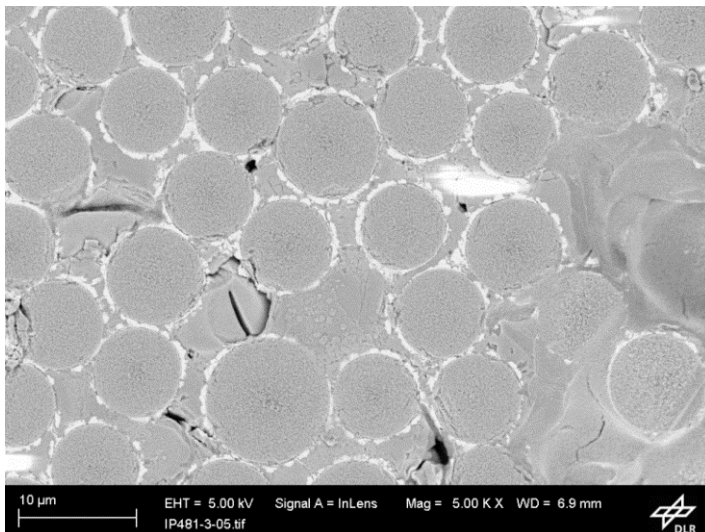


Microstructural Characterisation of $\text{SiC}_{\text{LaPO}_4}/\text{SiCN}$ @ 1700 °C



-Shown is sample tested at 1700 °C, close to surface

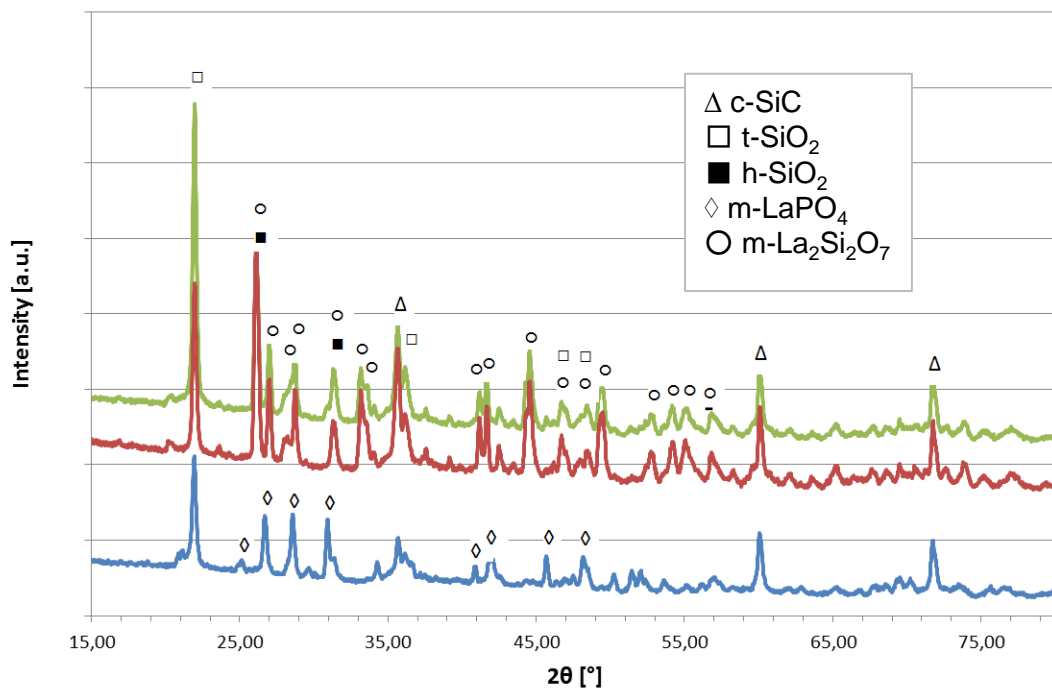
- **Close to the surface: degradation of coating already can be detected at 1300 °C**



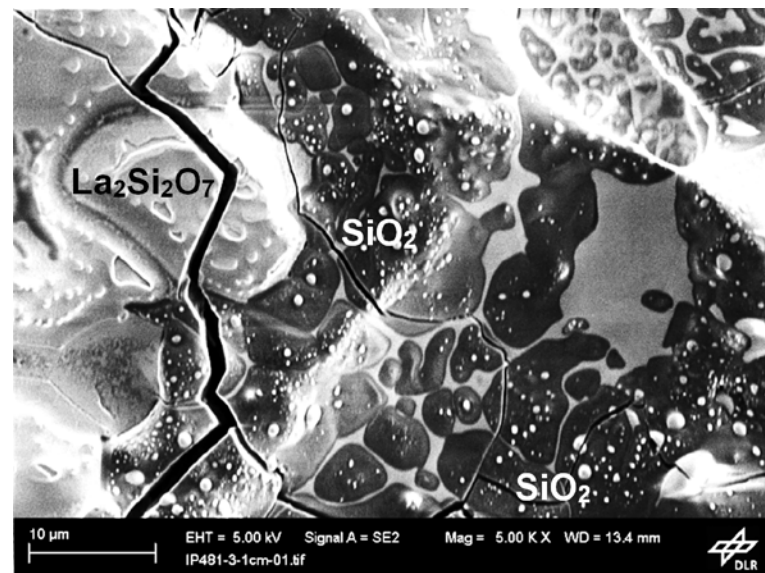
- **At the center of the samples the coating stayed intact at 1700 °C (1.5h)**
- **Samples tested at lower temperatures (4h) showed growth of SiO_2 between fiber and coating**



Microstructural Characterisation of $\text{SiC}_{\text{LaPO}_4}/\text{SiCN}$ @ 1700 °C



- Top view on sample with LaPO_4 -coated fibres
 dark phase (SiO_2)
 bright phase ($\text{La}_2\text{Si}_2\text{O}_7$)

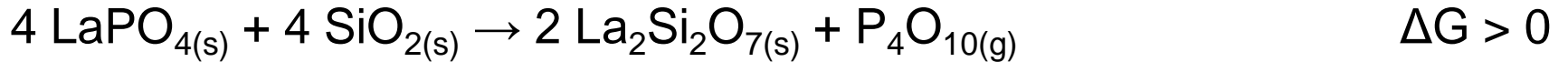


- formation of $\text{La}_2\text{Si}_2\text{O}_7$ at 1500 °C and 1700 °C detected via XRD on the samples surfaces
- Necessity for EBC-protection of structure

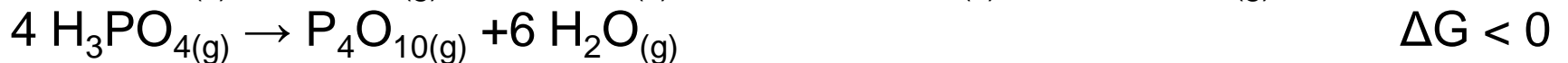


Potential Hydrothermal Decomposition of Monazite

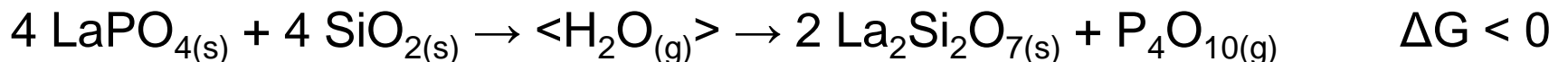
Although, reaction of LaPO_4 with SiO_2 is thermodynamically unfavorable



However, decomposition of LaPO_4 at $T > 1300^\circ\text{C}$ in the **presence of excess water vapor (jet stream) and SiO_2** is feasible due to enforced decomposition reactions. XRD and SEM/EDX of first test samples and thermodynamical calculations reveal:



in total:



Summary

Monazite (LaPO_4) fibre coatings have been successfully applied for

- Oxide (OXIPOL) as well as non-oxide CMCs (SiC/SiCN)

Influences of coating parameters have been demonstrated:

- fibre type and diameter as well as surface roughness
- Foulard (FC) and dip (DC) coating
- Further improvement of coating process w.r.t. coating efficiency

CMC behaviour before and after exposure in hot gas environment (HVOF-Test)

- Potentials and limits of LaPO_4 -coating were demonstrated
- SiO_2 formation observed on outer and crack surface by SEM
- Potential of CVD-SiC as EBC-coating was demonstrated even with oxidation sensitive C/C-SiC
- Hypothesis of potential decomposition of LaPO_4 in water vapor and presence of SiO_2



Outlook

- Further densification of composite (porosity < 5%) may improve composite properties such as strength level before and after exposure to oxidation
- Implementation of fillers serving for „self healing“ properties are in our focus
- Development of EBC/TBC coatings for both type of composites (OXIPOL and SiC/SiCN) are necessary and in development (e.g. $Y_2SiO_5/Y_2Si_2O_7$)
- Mid-term research targets are fibre coatings for CMCs (reaction barrier, low fibre-matrix bonding) suitable for melt-infiltration („dense matrix“)
 - Development of fibre coatings for LSI-processing based on Y_2O_3 , Si_3N_4 and TiB_2 are in progress in collaboration with special partners (Technical University of Chemnitz)
 - Development of fibre coatings for LSI-processing based on polymer-derived ceramics are in progress in collaboration with special partners (Technical University of Darmstadt and University of Stuttgart)



Acknowledgements

- DFG German Science Foundation for Funding SICAFIS-Project in cooperation with Technical University of Chemnitz and last not least in participating this workshop
- German Ministry of Defence (BAAINBw) for general financial support
- EU for Funding ATLLAS II-Project
- JAXA for PostDoc Exchange
- Thanks to the members of Ceramic Composites and Structures

