



University of Dundee

Vortex shedding suppression and wake control

Rashidi, Saman; Hayatdavoodi, Masoud; Esfahani, Javad Abolfazli

Published in:
Ocean Engineering

DOI:
[10.1016/j.oceaneng.2016.08.031](https://doi.org/10.1016/j.oceaneng.2016.08.031)

Publication date:
2016

Document Version
Peer reviewed version

[Link to publication in Discovery Research Portal](#)

Citation for published version (APA):

Rashidi, S., Hayatdavoodi, M., & Esfahani, J. A. (2016). Vortex shedding suppression and wake control: A review. *Ocean Engineering*, 126, 57-80. DOI: [10.1016/j.oceaneng.2016.08.031](https://doi.org/10.1016/j.oceaneng.2016.08.031)

General rights

Copyright and moral rights for the publications made accessible in Discovery Research Portal are retained by the authors and/or other copyright owners and it is a condition of accessing publications that users recognise and abide by the legal requirements associated with these rights.

- Users may download and print one copy of any publication from Discovery Research Portal for the purpose of private study or research.
- You may not further distribute the material or use it for any profit-making activity or commercial gain.
- You may freely distribute the URL identifying the publication in the public portal.

Take down policy

If you believe that this document breaches copyright please contact us providing details, and we will remove access to the work immediately and investigate your claim.

Vortex Shedding Suppression and Wake Control: A Review

Saman Rashidi^a, Masoud Hayatdavoodi^b and Javad Abolfazli Esfahani^{a*}

^aDepartment of Mechanical Engineering, Ferdowsi University of Mashhad, Mashhad 91775-1111, Iran

Division of Civil Engineering, School of Science and Engineering, University of Dundee, Dundee DD1 4HN,

UK

*Corresponding author e-mail: abolfazl@um.ac.ir

Abstract:

Wake destructive behavior and vortex shedding behind bluff bodies may be controlled by use of active and passive methods. Computational fluid dynamics, experimental and analytical techniques have been utilized to study this problem. In this survey, existing studies on different methods of controlling the wake destructive behavior and suppression of vortex shedding behind bluff bodies are discussed, including the very recent developments. These methods are classified into two groups. In the first group, these methods are discussed according to the type of external source or modification of the geometry of bluff body for controlling the flow. In the second group, the methods are classified according to the part of the flow, boundary layer or wake, that is modified by the method. Advantages, limitations, energy efficiency, and particular applications of each method are discussed and summarized, followed by some conclusions and recommendations. Moreover, the effectiveness of each technique on the drag reduction is discussed.

Keywords: Suppression of vortex shedding; Bluff bodies; Wake control; Computational fluid dynamics; Experimental and analytical techniques

1. Introduction

1
2
3
4
5
6
7
8
9
10
11
12
13
14
15
16
17
18
19
20
21
22
23
24
25
26
27
28
29
30
31
32
33
34
35
36
37
38
39
40
41
42
43
44
45
46
47
48
49
50
51
52
53
54
55
56
57
58
59
60
61
62
63
64
65

Studies on wake structure and vortex shedding behind bluff bodies and analysis of flow separation have been a topic of interest due to their fundamental significance and practical importance in aerodynamics and hydrodynamics applications. Examples of such applications are vibration of pipelines lying on the seabottom under the influence of currents and waves, bridges, interaction of currents and waves with offshore structures, flow tubular or tube banks heat exchangers, skyscrapers, chimneystacks, suspension bridges and chimneys near tall buildings, structures in the atmospheric boundary layers and submarine pipe, see e.g., Zang et al. (2013), Zang and Gao (2014), Bovand et al. (2015a), Rashidi et al. (2014a) and Rashidi et al. (2015a). In such applications, bluff bodies experience unsteady loads due to the flow separation and formation of the wake region and vortices, which in some cases, play an important role on their design. Control of wake structures and flow separations in such cases can significantly decrease the unsteady forces on the structure; resulting in less structural vibrations. Also, the offshore industry is particularly concerned with the vortex induced vibration suppression due to the failure of components of offshore structures, like risers and spars, which leads to catastrophic environmental and economic consequences.

There are various methods to control the wake structure and vortex shedding, such as employing an external small element, modifying geometry of the obstacle (by placing tabs or streaks on the obstacle), injection and suction in porous surface, and surface roughness elements. These methods have been introduced and discussed by researchers in the past decades, however, to the author's knowledge, there has not been any attempt to classify these techniques. For example, Zdravkovich (1981) performed a review on various aerodynamic and hydrodynamic means for suppressing vortex shedding, and classified these approaches in three categorie: surface protrusions, shrouds and near wake stabilizers. The various aerodynamic and hydrodynamic means have been highlighted in this paper rather than various techniques in this field. Also, Zdravkovich (1981) reviewed different means for

1 suppressing the vortex shedding behind circular obstacle, while other shapes, such as square,
2 diamond, triangular, trapezium, etc. did not receive any attention. Gad-el-Hak and Bushnell
3 (1991) performed a review on separation control. There are several control techniques that
4 the wake characteristics are modified directly, but in paper of Gad-el-Hak and Bushnell
5 (1991), attentions are focused on boundary layer control. In addition, many new devices and
6 approaches are introduced after 1991. Williamson (1996) reviewed new developments and
7 the discoveries of several new phenomena of vortex dynamics. Overview of vortex shedding
8 regimes, 3D vortex patterns in different regimes including the laminar, transition, and higher
9 Reynolds number regimes, wave interactions in the far wake, and 3D effects in other wake
10 flows were also reviewed by Williamson (1996). Modi (1997) and King (1997) performed a
11 review on moving surface boundary-layer control and vortex shedding, respectively.
12 Williamson and Govardhan (2004) reviewed the fundamental results and discoveries on
13 vortex-induced vibration. Free and forced vibrations of a cylinder are presented separately in
14 their review. In another research, Kumar et al. (2008) classified recent patents on passive
15 control techniques for vortex-induced vibrations. Choi et al. (2008) performed a review study
16 about control of flow over the bluff bodies. However, this paper is published in the year 2008
17 and this means that the recent developments in this field are not covered by this paper.
18 Moreover, some important factors about this topic such as energy efficiency of control
19 methods and the effectiveness of each technique on the drag reduction should be highlighted
20 with more details. These factors are necessary for assessing different control techniques.
21 Shmilovich and Yadlin (2011) investigated flow control techniques for transport aircraft.

22
23
24
25
26
27
28
29
30
31
32
33
34
35
36
37
38
39
40
41
42
43
44
45
46
47
48
49
50
51 It appears that the literature lacks from a comprehensive study on classification of
52 different methods for control of wake destructive behavior and suppression of vortex
53 shedding behind bluff bodies. The current study focuses on this goal. In Section 2, a
54 classification of various methods for controlling the wake destructive behavior and
55
56
57
58
59
60
61
62
63
64
65

1 suppression of vortex shedding is provided. Section 3 presents some conclusions and
2 suggestions based on the literature review.
3

4 **2. Classification of various methods**

5
6
7 In this study, we classify the existing aerodynamic and hydrodynamic methods used to
8 control the wake destructive behavior and suppress the vortex shedding into nine categories.
9
10 These methods are summarized below.
11

- 12 a) Control of vortex shedding by electrical methods
- 13
- 14 b) Control of vortex shedding by utilizing an external small element
- 15
- 16 c) Control of vortex shedding by feedback control methods
- 17
- 18 d) Control of vortex shedding by use of a magnetic field (MHD)
- 19
- 20 e) Control of vortex shedding by rotary oscillations
- 21
- 22 f) Control of vortex shedding by generating a secondary flow (suction, blowing, bleed
23 and synthetic jets)
- 24
- 25 g) Control of vortex shedding by surface roughness
- 26
- 27 h) Control of vortex shedding by use of thermal effects
- 28
- 29 i) Control of vortex shedding by other methods
- 30

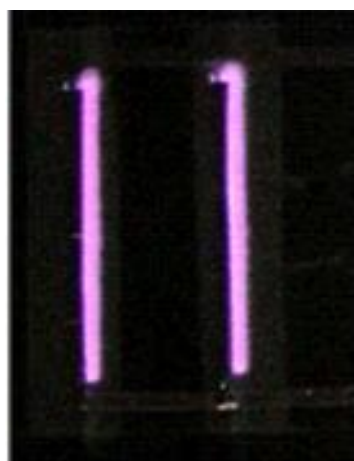
31
32 These approaches can be classified as passive and active control methods. The passive
33 control techniques are dependent on modifications of the bluff body geometry, which affect
34 the formation of the vortex shedding. These passive methods, class b and g, do not need any
35 external energy during application and have a simpler implementation, which favour them
36 over the active techniques for engineering applications. Active control methods, class a, c-f,
37 and h, need external energy to affect the fluid flow. Another classification of these methods is
38 based on the parts of flow that are modified in control process: the boundary layer control
39 methods and the wake control (Choi et al. 2008). For the former method, the boundary-layer
40 flow characteristics are changed by using the control method but the wake characteristics are
41
42
43
44
45
46
47
48
49
50
51
52
53
54
55
56
57
58
59
60
61
62
63
64
65

1 modified directly by using the later control method. Next, we will introduce and discuss each
2 method in detail.
3

4 **2.1. Active control methods (energy consuming)**

5 **2.1.1. Control of vortex shedding by electrical methods (EHD)**

6
7 This method is classified as active and boundary layer control method. In this method, an
8 electric discharge creates an electric force acting on fluid particles, resulting in a change in
9 the fluid velocity field. This force leads to a delay in separation of the flow in the trailing side
10 of the obstacle. The ions in the fluid flow, used in this method, are created by an electrical
11 discharge in an otherwise electrically neutral fluid. The ions are under the action of columbic
12 forces due to the electrode of opposite sign and this leads to exchange momentum with the
13 neutral fluid particles. This method is suitable for aerodynamic applications such as airfoils
14 typically used in wind turbine blades, civil air traffic projects and aircraft. Figure 1 shows the
15 surface plasma generated to modify the flow over a circular cylinder. Figure 2 shows a smoke
16 visualization of plasma actuation on flow surrounding a 2-D circular cylinder at $Re= 1.8 \times 10^4$,
17 with and without plasma actuation. It is observed that the EHD method reduces the re-
18 circulated smoke, resulting in Karman vortex shedding suppresses.
19
20
21
22
23
24
25
26
27
28
29
30
31
32
33
34
35
36
37
38



39
40
41
42
43
44
45
46
47
48
49
50
51
52
53
54
55 **Fig. 1.** Surface plasma generated to modify the flow over a circular cylinder (Figure reprinted
56 from Sung et al. (2006) with permission from the publisher)
57

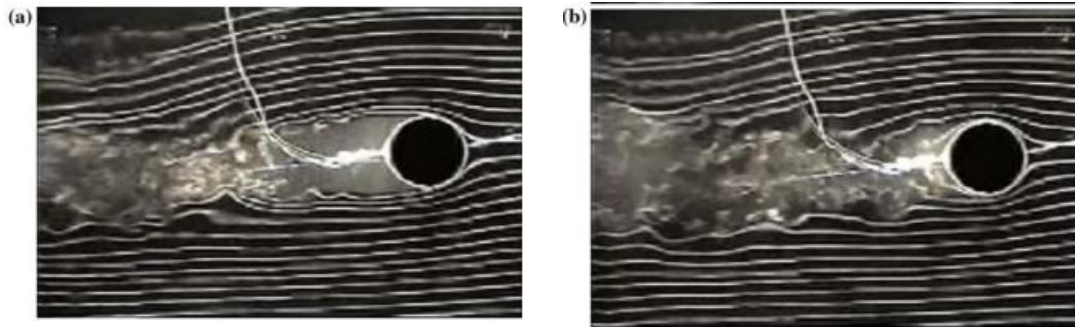


Fig. 2. Smoke visualization of plasma actuation on flow surrounding a circular cylinder at $Re = 1.8 \times 10^4$ for two cases of (a) with and (b) without plasma actuation (Flow is from right to left; Figure reprinted from Sung et al. (2006) with permission from the publisher)

This method is used by Artana et al. (2003) to experimentally control the near-wake flow around a circular obstacle with electrohydrodynamic actuators for Reynolds number range of $2.3 \times 10^3 < Re < 5.8 \times 10^4$. In these experiments, two electrodes were flush-mounted on the obstacle wall and were charged with DC power supplies to produce a plasma sheet contouring the body. The discharge electrodes modify the shear stresses of the fluid layers increase and the base pressure increases by creating the electric force, modifying the size of the mean recirculation region. Most of the studies focusing on this method are conducted experimentally and are for a specific range of Reynolds number flow.

Hyun and Chun (2003) performed an experimental study on the wake flow control behind a circular obstacle using ion wind for Reynolds number range of $4.0 \times 10^3 < Re < 8.0 \times 10^3$. It was found that presence of ion wind could affect the wake structure behind an obstacle significantly. Also, it was shown that the pressure drag decreases by superimposing ion wind. Control of flow separation from an airfoil (NACA 66₃-018) on high angle of attack was studied by Post and Corke (2004) using plasma actuators for Reynolds number range of $7.7 \times 10^4 < Re < 3.33 \times 10^5$. In this study, two types of plasma actuators were utilized. The first type creates a spanwise array of streamwise vortices, and the second type creates a 2-D jet in the flow direction along the airfoil wall. These actuators were placed at the leading edge of

1 the airfoil. It was found that the reattached flow created by the actuator leads to a decrease in
2 drag and significant suction-pressure recovery.
3

4
5 In the experiments conducted by Sung et al. (2006), flow over circular obstacles was
6 modified by plasma actuation for the Reynolds number range of $1.0 \times 10^4 < Re < 4.0 \times 10^4$. A PIV
7 technique was applied to quantify the changes in the wake structure. It was shown that
8 modest discharge power into the boundary layer, created by a dielectric barrier discharge, can
9 decrease the wake momentum deficit significantly by influencing flow separation. Less
10 recirculation was observed in the recirculation region when the plasma is actuated. Thomas et
11 al. (2008) used plasma actuators for the cylinder flow control and noise reduction. Both
12 steady and unsteady actuation were used in the experiments for $Re = 3.3 \times 10^4$. It was shown
13 that steady actuation is more suitable for noise-control applications; the unsteady actuation is
14 affected by an unsteady body force and creates a tone at the actuation frequency. Note that
15 there is a less power dissipation for each actuator for unsteady actuation due to a lower duty
16 cycle. Chaos controlling is referred to the stabilization of unstable periodic orbits by means of
17 small system perturbations. It is widely used in general non-linear dynamics problems.
18 Muddada and Patnaik (2011) reviewed some chaos control strategies and discussed their
19 relevance to some fluid turbulence applications. Through a computational study, vortex
20 shedding behind a blunt trailing edge was controlled experimentally using plasma actuators
21 by Nati et al. (2013). They focused on laminar boundary layer regime on an elongated D-
22 shaped flat plate of 12 mm thickness with a free-stream velocity of 10 m/s. Their results show
23 that the vortex shedding frequency peak is decreased during the steady plasma actuation by
24 10 dB.
25
26
27
28
29
30
31
32
33
34
35
36
37
38
39
40
41
42
43
44
45
46
47
48
49
50
51

52 Finally, this method is classified as active and boundary layer control method. In this
53 method, the electrodes create electric force acting on fluid particles, resulting in a change in
54 the fluid velocity field. This force leads to a delay in separation of the flow in the trailing side
55
56
57
58
59
60
61
62
63
64
65

of the obstacle. As mentioned by e.g. Artana et al. (2003), it is more efficient to mount the electrodes on the surface of the bluff bodies because most of the momentum should be transferred in fluid layers that have a very low kinetic energy. This leads to important changes in the fluid dynamics with minimum power added to the flow. EHD mitigates some of the drawbacks of active flow control methods. This method has some favorable features as an active flow control such as low power consumption, simple construction, robustness, and high frequency response. Moreover, the electric discharge varies in space and time and so creates an easy changeable force. The electric discharge can be employed only when needed, and when not in use, they have no destructive effects.

Table 1 summarizes researches on application of EHD to control the vortex shedding behind a bluff body. As shown in this table, these researches have covered only flows with moderate Reynolds number. More studies are necessary for higher range of Reynolds numbers (i.e. $Re > 10^5$). All studies for this method are performed in air. Application of this method in ocean engineering, with water being as fluid, is questionable. The results show a lack of information about application of numerical approaches in this method. However, vortex induced vibration phenomenon is observed in slender flexible structures as a destructive aerodynamic excitation. Studies on the control of the flows with this phenomenon by EHD method have remained interesting.

Table 1: Researches on application of EHD to control the vortex shedding behind a bluff body

Authors	Type of research	Type of fluid	Reynolds number
Artana et al. (2003)	Experimental	air	$2.3 \times 10^3 - 5.8 \times 10^4$
Hyun and Chun (2003)	Experimental	air	$4.0 \times 10^3 - 8.0 \times 10^3$
Post and Corke (2004)	Experimental	air	$7.7 \times 10^4 - 3.3 \times 10^5$
Sung et al. (2006)	Experimental	air	$1.0 \times 10^4 - 4.0 \times 10^4$
Thomas et al. (2008)	Experimental	air	3.3×10^4
Nati et al. (2013)	Experimental	air	$3.8 \times 10^3 - 7.3 \times 10^3$

2.1.2. Control of vortex shedding by feedback control methods

This method is classified as active and wake control method. It is possible to apply feedback from a suitable sensor to an actuator for stabilizing the wake and suppressing the vortex shedding at Reynolds numbers close to the onset of vortex shedding, see e.g. Park et al. (1994). In this method, the control input is modified successively according to the response of the flow system. This method is widely used in aircraft and projectile aerodynamics including dynamic stall control, marine structures, chemical mixing improvement, submarine periscopes, increase mixing and heat transfer in combustion. A sample set up of this control method is shown in figure 3. The localized cross-stream velocity measurement in the cylinder wake region is feedback to the compensator, $R(s)$, which drives the cylinder rotation. The actuation is understood by unsteady angular rotations of the circular obstacle surface $\varphi(t)$.

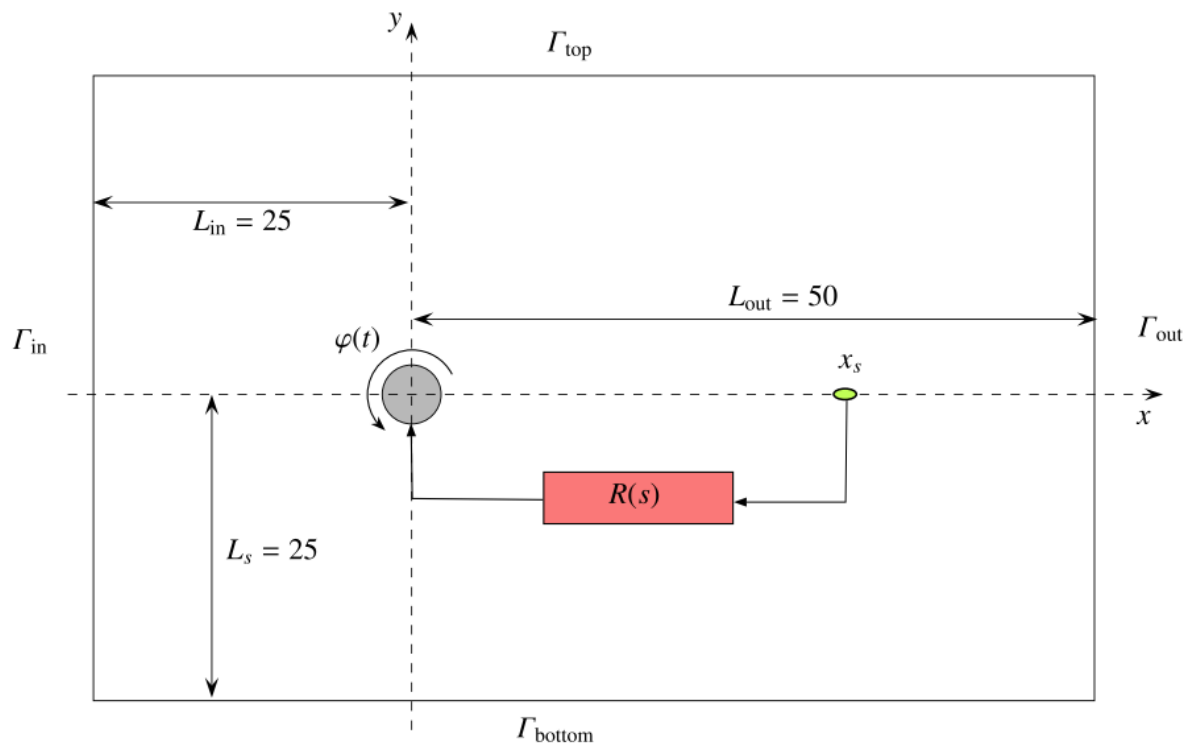


Fig. 3. A sample set up of a feedback control method (Flow is from left to right; Figure reprinted from Carini et al. (2015) with permission from the publisher)

Feedback control method can be used to control the wake and vortex shedding behind a bluff body. For example, Williams and Zhao (1989) presented an active experimental method

1 for controlling the vortex shedding behind a circular obstacle at $Re=4.0 \times 10^2$, sending
2 acoustic feedback signals. These signals were taken from hot-wires in the wake of the
3 obstacle. The shedding was suppressed by switching on the controller after two or three
4 vortex shedding periods. Tao et al. (1996) experimentally presented a feedback control
5 method to suppress and excite the vortex shedding behind a circular obstacle. The feedback
6 perturbations were created inside the wake by oscillating the obstacle transversely to the
7 ongoing flow. They observed that the peaks at the vortex shedding frequency decrease under
8 the feedback suppression. Roussopoulos and Monkewitz (1996) proposed a nonlinear model
9 for vortex shedding control in cylinder wakes. They used the feedback control of Karman
10 vortex shedding for critical value of $Re \sim 47$. It was observed that the vortex shedding is
11 suppressed only at the spanwise location of the sensor for long cylinders. Fujisawa et al.
12 (2001) controlled experimentally the vortex shedding behind a circular obstacle at moderate
13 Reynolds numbers. They used a rotary cylinder oscillations controlled by the feedback signal
14 of a reference velocity in the main obstacle. They showed that the fluid forces and the
15 velocity fluctuations decreases by the feedback control with optimum values of the phase lag
16 and feedback gain. The feedback gain is the ratio of the maximum of the circumferential
17 velocity of the cylinder to the maximum filtered reference velocity, and the phase lag is
18 defined as the time lag in degrees between the filtered velocity signal and the monitored
19 angular velocity signal.

20
21
22
23
24
25
26
27
28
29
30
31
32
33
34
35
36
37
38
39
40
41
42
43
44
45
46
47
48
49
50
51
52
53
54
55
56
57
58
59
60
61
62
63
64
65

Fujisawa and Nakabayashi (2002) experimentally controlled the vortex shedding behind a circular obstacle by the rotational feedback oscillations. They used the neural network approach for calculating the optimum condition. Their results showed that the drag force is decreased by about 16% and the lift force is suppressed about 70% in comparison with the stationary obstacle. Wolfe and Ziada (2003) performed an experimental study on the effect of feedback control on the vortex shedding behind two tandem obstacles in cross-flow. The aim

1 of this study was to decrease the downstream obstacle response to vortex shedding and
2 turbulence excitations. Feedback control is utilized for resonant and non-resonant cases.
3
4 Frequency of vortex shedding coincides with the resonance frequency of the downstream
5 obstacle for resonant case, but the shedding frequency is about 30% higher than the
6 downstream obstacle resonance frequency for non-resonant case. They observed that the
7 feedback control can significantly decrease the downstream obstacle response to turbulence
8 excitations and vortex shedding. Leclerc et al. (2006) presented a numerical feedback control
9 method for controlling the laminar vortex shedding behind a circular obstacle due to a
10 compressible flow. This method was based on the incomplete sensitivities and gradient
11 evaluation by complex variable method. Muddada and Patnaik (2010) proposed an algorithm
12 to model and control vortex shedding behind a circular obstacle. Multiple-feedback sensors,
13 actuators and a control strategy are used in this approach. The state of the flow is reported by
14 some sensors that are located in the downstream of the body. Two external actuators are used
15 to control the fluid flow. The actuators respond with a rotation parameter that this parameter
16 temporally varies by the control algorithm. They considered the range of $Re= 1.0 \times 10^2 -$
17 3.0×10^2 . They observed that the wake gradually becomes weaker and the vortex shedding
18 completely suppresses for higher rotation rate of the control cylinders. Lu et al. (2011) was
19 able to reduce the lift force on a circular cylinder by about 50% at low Reynolds number,
20 using the feedback rotary oscillation method. Carini et al. (2015) presented a linear feedback
21 control for the unsteady cylinder wake at low Reynolds numbers. The classical small-gain has
22 been used to design a full-dimensional stabilising compensator of the linearized Navier–
23 Stokes equations. Also, they investigated the effect of sensor placement on the compensator
24 performance. It was found that the amount of control energy for stabilizing the flow
25 characteristics by a small, lower plateau for $11 < x_s < 14$ when the sensor is moved along the
26
27
28
29
30
31
32
33
34
35
36
37
38
39
40
41
42
43
44
45
46
47
48
49
50
51
52
53
54
55
56
57
58
59
60
61
62
63
64
65

flow centerline, where x_s is the streamwise locations of the sensor in the far-wake region, see the green elliptic in figure 3.

There is a feedback control method called linear Proportional Integral Differential (PID) control. In this control method, only the output is available and used for feedback. Figure 4 shows a block diagram of the PID control. It should be noted that the controller is a combination of a simple gain (P control), an integrator (I control), a differentiator (D control) or some weighted combination of these possibilities, see e.g. Son et al. (2011).

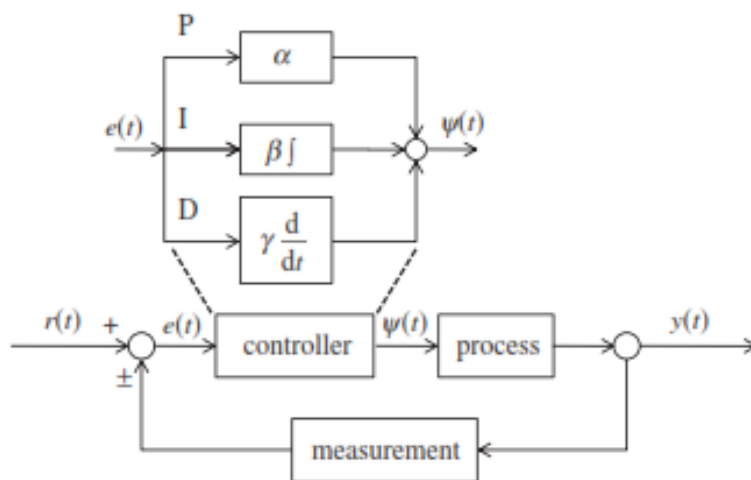


Fig. 4. Block diagram of the PID control (Figure reprinted from Son et al. (2011) with permission from the publisher)

In this figure, $\psi(t)$, $e(t)$, α , β and γ are the control input, error, proportional, integral and differential gains, respectively. Moreover, r and y indicate the reference input (or desired response) and process output, respectively. Son et al. (2011) summarized the researches on the application of the PID control to annihilate the vortex shedding behind a bluff body. The results of these researches are presented in table 2. The ranges of Reynolds number, actuation type and feedback law (P control, P control with phase shift, PI and PID) for each research are presented in this table.

Table 2. Previous studies of the P, PI and PID controls for flow over a cylinder (Table reprinted from Son et al. (2011) with permission from the publisher)

Author	Reynolds number	Actuation	Feedback law
--------	-----------------	-----------	--------------

Berger (1967)	laminar	Cylinder displacement	P
Williams and Zhao (1989)	4.0×10^2	Loudspeakers	P with phase shift
Baz and Ro (1991)	1.716×10^4 – 2.6555×10^4	Cylinder displacement	P
Roussopoulos (1993)	50–65 and 1.2×10^2	Loudspeakers	P with phase shift
Park et al. (1994)	60 and 80	Blowing/suction	P
Warui and Fujisawa (1996)	6.7×10^3	Cylinder displacement	P with phase shift
Huang (1996)	4.95×10^3 6.5×10^3 and 1.3×10^4	Blowing/suction	P with phase shift
Zhang et al. (2004)	3.5×10^3	Cylinder displacement	P, PI, PID
Hiejima et al. (2005)	2.0×10^2	Blowing/suction	P with phase shift

From these studies, Son et al. (2011) concluded that the P controls are very sensitive to the amount of phase shift and sensing position. Therefore, P controls may be used as an effective method to suppress the vortex shedding or decrease its strength by considering suitable values for these variables. Son et al. (2011) also concluded that the reductions in velocity fluctuations in the wake and the strength of vortex shedding are intensified with increase in the proportional gain. However, a large gain leads to more system instability. As shown in this table, the research of Zhang et al. (2004) is the only study where the PI and PID controls are utilized to control the flow over a bluff body. They decreased both the cylinder vibration amplitude and streamwise velocity fluctuations by PID controls.

Son et al. (2011) utilized the P, PI and PD to control the flow over a circular cylinder for the Reynolds numbers of 60 and 1.0×10^2 . They measured the transverse velocity at a centreline position in the wake for sensing, and determined the actuation velocity from the P, PI or PD control. They observed that the velocity, lift fluctuations and the mean drag decrease by addition of a D or an I control to the P control for the cases that the P control does not completely suppress the vortex shedding.

Extremum is another feedback control method used to suppress the vortex shedding by seeking control. This method is used as an online optimization technique to control the dynamic (nonlinear) systems characterised by an output extremum in a steady state. These

1 controllers are used to find an optimal set point to modify the flow around the bluff bodies.
2 Beaudoin et al. (2006) controlled the drag exerted on a bluff-body at large Reynolds number
3
4 ($Re=2.0 \times 10^4$) by extremum-seeking control. The set point in this study was defined as the
5
6 minimum cost of global energy consumption of a turbulent separated flow (drag reduction
7
8 versus actuator's energy). They used the extremum-seeking control to find the lowest value
9
10 for cost of global energy consumption. They observed that the system works around its
11
12 optimal state of lower energy consumption by using this algorithm. Becker and King (2006)
13
14 controlled the separation on a high-lift configuration by using extremum seeking control.
15
16 Their results showed the capability of this method to minimize the separation even in the
17
18 presence of disturbances.
19
20
21
22
23

24 Feedback method was classified as an active wake control method. With the development
25
26 of Microelectromechanical systems and smart materials, intelligent sensing and actuator
27
28 techniques, this method has been popular rapidly among the researchers in last years. Indeed,
29
30 this method is a combinatorial method that directly modifies the wake characteristics. Note
31
32 that this method is not easy to simulate even with the rapid developments in computer power
33
34 and CPU. There are some numerical techniques to decrease the amount of computations. For
35
36 example, it is possible to replace the Navier Stokes equation by a low order dynamical
37
38 system, or to find new methods to calculate the gradient of the cost function, see e.g. Leclerc
39
40 et al. (2006). We note that this is a costly method. Therefore, an energetic and optimization
41
42 analyses are necessary for this method. Flow visualization is a good technique to provide an
43
44 overall picture of the change in wake by using feedback control.
45
46
47
48
49
50

51 Table 3 provides a summary on the research conducted on application of feedback
52
53 method to control the vortex shedding and flow separation of a bluff body. As presented in
54
55 this table, researchers used different actuations for this method. However, it can be seen that
56
57 the numerical methods have been used for only very low Reynolds number. Moreover, the
58
59
60
61
62
63
64
65

potential of this method at moderate and high Reynolds numbers is very significant from an engineering viewpoint and requires future study. A relatively low cost method in this classification is application of sound field as actuation. More researches are needed about this technique as the ability, advantages and disadvantages of this technique are still unclear.

Table 3: Researches on application of feedback method to control the vortex shedding and flow separation of a bluff body

Author	Reynolds number	Actuation	Spatial dimensions	Type of research	Type of fluid
Williams and Zhao (1989)	4.0×10^2	Loudspeakers	----	Experimental	Air
Roussopoulos and Monkewitz (1996)	50	Loudspeakers	2D	Numerical	----
Tao et al. (1996)	48.5 and 51	Oscillating the cylinder transverse to the oncoming flow	----	Experimental	Water
Fujisawa et al. (2001)	6.7×10^3 and 2.0×10^4	Rotary cylinder oscillations	----	Experimental	Air
Wolfe and Ziada (2003)	4.11×10^4 - 5.79×10^4	Synthetic jet	----	Experimental	Air
Leclerc et al. (2006)	1.0×10^2	Blowing/suction	2D	Numerical	----
Lu et al. (2011)	60-200	Rotary cylinder oscillations	2D	Numerical	----
Carini et al. (2015)	50-90	Rotations of the cylinder surface	2D	Numerical	----

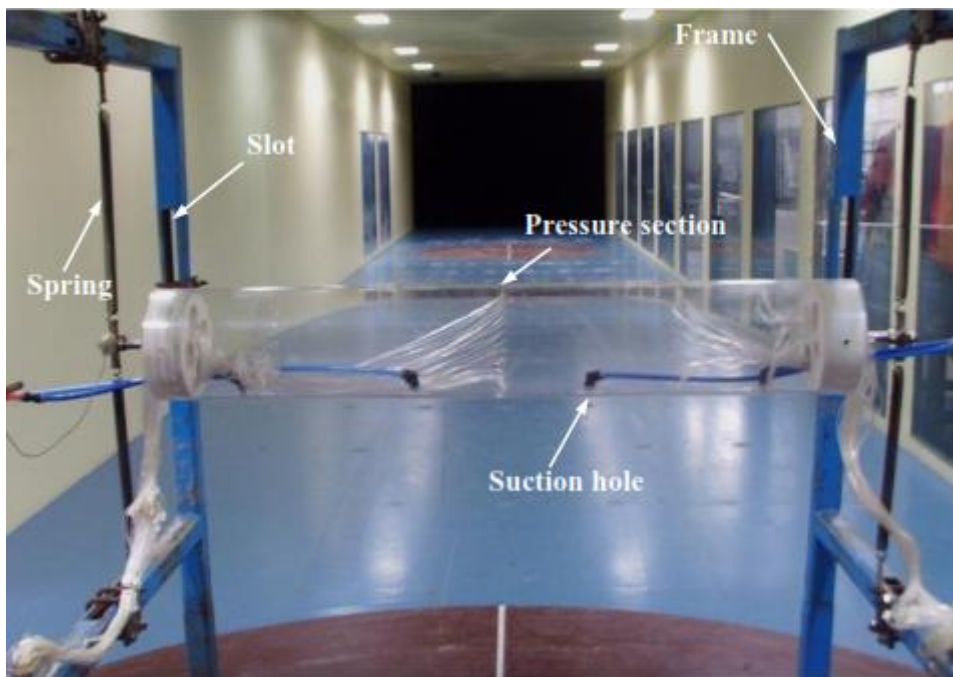
2.1.3. Control of vortex shedding by generating a secondary flow (Suction, blowing, base bleed and synthetic jets)

These methods are classified as active wake control method. It is possible to control the flow around bluff bodies by generating a secondary flow (suction, blowing, base bleed). For suction and blowing, the wall of obstacle is penetrable and the suction or blowing velocity is defined. For base bleed method, the wall of obstacle is not penetrable and the secondary flow is injected at the edge of obstacle. The suction, blowing and bleed are used as energy input for controlling the flow. These techniques are widely used in long-span suspension bridges and cable-stayed bridges, see e.g. Chen et al. (2013). Figure 5 shows an experimental setup used to suppress the vortex-induced vibration of a circular cylinder by means of suction flow.

1 The time-averaged streamline contours with injection through the surfaces of a square
 2 cylinder for different Reynolds numbers are shown in figure 6. Here, Γ represents the
 3 injection parameter, defined as the ratio of the fluid velocity through the porous wall to inlet
 4 streamwise velocity. It is observed that the wake structure changes significantly when using
 5 injection; the vortices disappear for higher injection parameter. In this figure, the pressure
 6 coefficient is defined by:
 7
 8
 9
 10
 11
 12
 13

$$14 \quad C_p = \frac{(p_{in} - p_w)}{0.5\rho U_{in}^2} \quad (1)$$

15
 16
 17
 18 where “in” and “w” subscripts demonstrate inlet and cylinder wall, respectively. Moreover,
 19
 20
 21 p and U are the pressure and velocity, respectively.
 22



46
 47 **Fig. 5.** Experimental setup utilized to suppress the vortex-induced vibration of a circular
 48 cylinder by using suction flow. (Figure reprinted from Chen et al. (2013) with permission
 49
 50
 51
 52
 53
 54
 55
 56
 57
 58
 59
 60
 61
 62
 63
 64
 65
 66
 67
 68
 69
 70
 71
 72
 73
 74
 75
 76
 77
 78
 79
 80
 81
 82
 83
 84
 85
 86
 87
 88
 89
 90
 91
 92
 93
 94
 95
 96
 97
 98
 99
 100
 101
 102
 103
 104
 105
 106
 107
 108
 109
 110
 111
 112
 113
 114
 115
 116
 117
 118
 119
 120
 121
 122
 123
 124
 125
 126
 127
 128
 129
 130
 131
 132
 133
 134
 135
 136
 137
 138
 139
 140
 141
 142
 143
 144
 145
 146
 147
 148
 149
 150
 151
 152
 153
 154
 155
 156
 157
 158
 159
 160
 161
 162
 163
 164
 165
 166
 167
 168
 169
 170
 171
 172
 173
 174
 175
 176
 177
 178
 179
 180
 181
 182
 183
 184
 185
 186
 187
 188
 189
 190
 191
 192
 193
 194
 195
 196
 197
 198
 199
 200
 201
 202
 203
 204
 205
 206
 207
 208
 209
 210
 211
 212
 213
 214
 215
 216
 217
 218
 219
 220
 221
 222
 223
 224
 225
 226
 227
 228
 229
 230
 231
 232
 233
 234
 235
 236
 237
 238
 239
 240
 241
 242
 243
 244
 245
 246
 247
 248
 249
 250
 251
 252
 253
 254
 255
 256
 257
 258
 259
 260
 261
 262
 263
 264
 265
 266
 267
 268
 269
 270
 271
 272
 273
 274
 275
 276
 277
 278
 279
 280
 281
 282
 283
 284
 285
 286
 287
 288
 289
 290
 291
 292
 293
 294
 295
 296
 297
 298
 299
 300
 301
 302
 303
 304
 305
 306
 307
 308
 309
 310
 311
 312
 313
 314
 315
 316
 317
 318
 319
 320
 321
 322
 323
 324
 325
 326
 327
 328
 329
 330
 331
 332
 333
 334
 335
 336
 337
 338
 339
 340
 341
 342
 343
 344
 345
 346
 347
 348
 349
 350
 351
 352
 353
 354
 355
 356
 357
 358
 359
 360
 361
 362
 363
 364
 365
 366
 367
 368
 369
 370
 371
 372
 373
 374
 375
 376
 377
 378
 379
 380
 381
 382
 383
 384
 385
 386
 387
 388
 389
 390
 391
 392
 393
 394
 395
 396
 397
 398
 399
 400
 401
 402
 403
 404
 405
 406
 407
 408
 409
 410
 411
 412
 413
 414
 415
 416
 417
 418
 419
 420
 421
 422
 423
 424
 425
 426
 427
 428
 429
 430
 431
 432
 433
 434
 435
 436
 437
 438
 439
 440
 441
 442
 443
 444
 445
 446
 447
 448
 449
 450
 451
 452
 453
 454
 455
 456
 457
 458
 459
 460
 461
 462
 463
 464
 465
 466
 467
 468
 469
 470
 471
 472
 473
 474
 475
 476
 477
 478
 479
 480
 481
 482
 483
 484
 485
 486
 487
 488
 489
 490
 491
 492
 493
 494
 495
 496
 497
 498
 499
 500
 501
 502
 503
 504
 505
 506
 507
 508
 509
 510
 511
 512
 513
 514
 515
 516
 517
 518
 519
 520
 521
 522
 523
 524
 525
 526
 527
 528
 529
 530
 531
 532
 533
 534
 535
 536
 537
 538
 539
 540
 541
 542
 543
 544
 545
 546
 547
 548
 549
 550
 551
 552
 553
 554
 555
 556
 557
 558
 559
 560
 561
 562
 563
 564
 565
 566
 567
 568
 569
 570
 571
 572
 573
 574
 575
 576
 577
 578
 579
 580
 581
 582
 583
 584
 585
 586
 587
 588
 589
 590
 591
 592
 593
 594
 595
 596
 597
 598
 599
 600
 601
 602
 603
 604
 605
 606
 607
 608
 609
 610
 611
 612
 613
 614
 615
 616
 617
 618
 619
 620
 621
 622
 623
 624
 625
 626
 627
 628
 629
 630
 631
 632
 633
 634
 635
 636
 637
 638
 639
 640
 641
 642
 643
 644
 645
 646
 647
 648
 649
 650
 651
 652
 653
 654
 655
 656
 657
 658
 659
 660
 661
 662
 663
 664
 665
 666
 667
 668
 669
 670
 671
 672
 673
 674
 675
 676
 677
 678
 679
 680
 681
 682
 683
 684
 685
 686
 687
 688
 689
 690
 691
 692
 693
 694
 695
 696
 697
 698
 699
 700
 701
 702
 703
 704
 705
 706
 707
 708
 709
 710
 711
 712
 713
 714
 715
 716
 717
 718
 719
 720
 721
 722
 723
 724
 725
 726
 727
 728
 729
 730
 731
 732
 733
 734
 735
 736
 737
 738
 739
 740
 741
 742
 743
 744
 745
 746
 747
 748
 749
 750
 751
 752
 753
 754
 755
 756
 757
 758
 759
 760
 761
 762
 763
 764
 765
 766
 767
 768
 769
 770
 771
 772
 773
 774
 775
 776
 777
 778
 779
 780
 781
 782
 783
 784
 785
 786
 787
 788
 789
 790
 791
 792
 793
 794
 795
 796
 797
 798
 799
 800
 801
 802
 803
 804
 805
 806
 807
 808
 809
 810
 811
 812
 813
 814
 815
 816
 817
 818
 819
 820
 821
 822
 823
 824
 825
 826
 827
 828
 829
 830
 831
 832
 833
 834
 835
 836
 837
 838
 839
 840
 841
 842
 843
 844
 845
 846
 847
 848
 849
 850
 851
 852
 853
 854
 855
 856
 857
 858
 859
 860
 861
 862
 863
 864
 865
 866
 867
 868
 869
 870
 871
 872
 873
 874
 875
 876
 877
 878
 879
 880
 881
 882
 883
 884
 885
 886
 887
 888
 889
 890
 891
 892
 893
 894
 895
 896
 897
 898
 899
 900
 901
 902
 903
 904
 905
 906
 907
 908
 909
 910
 911
 912
 913
 914
 915
 916
 917
 918
 919
 920
 921
 922
 923
 924
 925
 926
 927
 928
 929
 930
 931
 932
 933
 934
 935
 936
 937
 938
 939
 940
 941
 942
 943
 944
 945
 946
 947
 948
 949
 950
 951
 952
 953
 954
 955
 956
 957
 958
 959
 960
 961
 962
 963
 964
 965
 966
 967
 968
 969
 970
 971
 972
 973
 974
 975
 976
 977
 978
 979
 980
 981
 982
 983
 984
 985
 986
 987
 988
 989
 990
 991
 992
 993
 994
 995
 996
 997
 998
 999
 1000

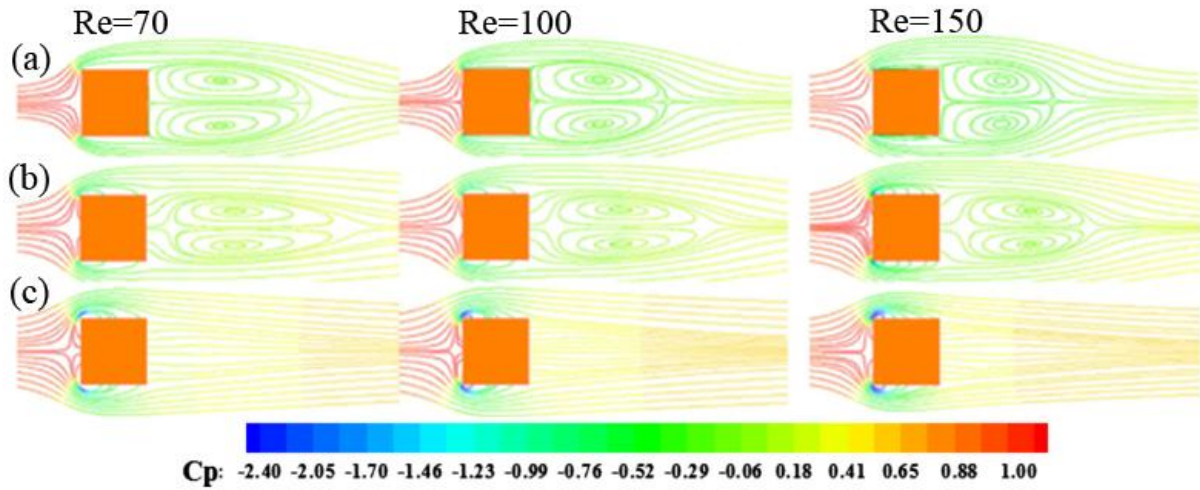


Fig. 6. Time-averaged streamlines colored by the pressure coefficient for different suction/blowing parameter and Reynolds number for case (a) without control ($|\Gamma| = 0$), (b) with control ($|\Gamma| = 0.05$) (c) with control ($|\Gamma| = 0.15$) (Flow is from left to right; Figure reprinted from Sohankar et al. (2015) with permission from the publisher)

Formation of turbulent boundary layers around a cylinder by using surface suction is investigated experimentally by Seal and Smith (1999). It is shown that the surface suction leads to i) weakening the instantaneous turbulent vortex and its associated surface interactions and ii) weakening the average downstream extensions of the vortex. Delaunay and Kaiktsis (2001) studied effects of steady base suction and blowing on the stability and dynamics of the wake behind circular obstacle at low Reynolds numbers, using a numerical method. In this paper, all terms in continuity and momentum equations were discretized based on a Legendre spectral element method except the time term that was discretized based on second-order accurate mixed stiffly stable scheme. They found that slight blowing or high enough suction stabilizes the wake for supercritical Reynolds number regime ($Re > 47$). However, they observed that the wake could be destabilized for the subcritical regime by suction, whereas blowing had no detectable effect on the flow stability.

Optimal control of 2-D cylinder wakes via suction and blowing is investigated numerically by Li et al. (2003) for Reynolds numbers up to 1.1×10^2 . They observed that an

1 open-loop control can be used for suppressing the vortex shedding in the wake of a circular
2 obstacle in a robust way. Kim and Choi (2005) investigated the distributed forcing of flow
3 over a circular obstacle for the range of Reynolds number from 40 to 3900. The distributed
4 forcing is found by a blowing and suction from the slots placed at upper and lower surfaces
5 of the obstacle. The optimal conditions for maximum drag reduction are occurred for the
6 Reynolds number of 1.0×10^2 . Layek et al. (2008) numerically investigated the influence of
7 suction and blowing on vortex shedding behind a square obstacle placed inside a channel. In
8 this study, the Navier–Stokes equations were solved by a finite-difference method using
9 staggered grid arrangement. It is concluded that the amplitude of the lift coefficient decreases
10 by increasing the blowing velocity. Also, the flow becomes steady and symmetric for a
11 specific value of the blowing parameter. Warjito et al. (2012) used this approach in a
12 numerical and experimental study and modified turbulent flow structure around Ahmed's
13 body suction flow, resulting in a reduction of the drag force by about 15%. It was shown that,
14 using the suction in the rear part of the model results in a reduction of the wake and vortex
15 formation. Chen et al. (2013) conducted experiments on suppression of the vortex-induced
16 vibration of a circular obstacle by using a suction flow method. Four different steady suction
17 flow rates were used in this study. Following this approach, they were able to decrease the
18 amplitude of the oscillation, fluctuating wall pressure, and the unsteady aerodynamic forces
19 acting on the obstacle. A given suction flow rate was found to optimize the control. Sohankar
20 et al. (2015) numerically controlled fluid flow and heat transfer around a square obstacle by
21 uniform suction and blowing at low Reynolds numbers ($Re=70-1.5 \times 10^2$). They considered
22 three cases for the location of the blowing and suction, including the front surface, rear
23 surface and top/bottom surface. They found that when the suction is employed on the top and
24 bottom surfaces, and blowing is applied on the front and rear faces, optimum control can be
25 achieved. Also, their study showed that the drag and lift fluctuations for the optimum
26
27
28
29
30
31
32
33
34
35
36
37
38
39
40
41
42
43
44
45
46
47
48
49
50
51
52
53
54
55
56
57
58
59
60
61
62
63
64
65

1 configuration decrease and the maximum deduction in drag force are 61%, 67% and 72% for
2 $Re = 70, 1.0 \times 10^2, 1.5 \times 10^2$, respectively.
3

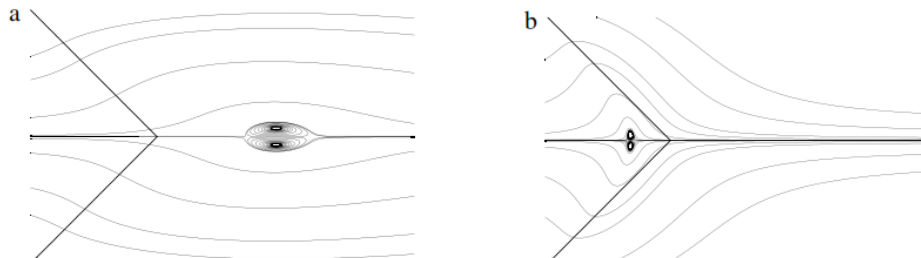
4 In addition, synthetic jet (unsteady blowing) is used in some cases to control the flow and
5 separation phenomena. For example, Tensi et al. (2002) performed experimental tests to
6 control the flow around an obstacle by synthetic jet (unsteady blowing) through a single slot
7 disposed on the surface of the obstacle for $Re=1.05 \times 10^2$. The effects of this jet on delaying
8 separation and modifying the drag coefficient were investigated through a series of
9 experiments. It was suggested that zero-mass-flux actuator types, which require little energy,
10 could be used for further flow control around various bodies including airfoils. Effect of
11 novel synthetic jet on wake vortex shedding modes of a circular cylinder is investigated
12 experimentally by Feng et al. (2010). They found that the control effect of the synthetic jet
13 upon the flow around a circular obstacle can increase with an increase in the suction duty
14 cycle factor, i.e., the ratio between the time duration of the suction cycle and the blowing
15 cycle of the actuator signal and increase in the momentum coefficient. Feng and Wang (2010,
16 2011) experimentally controlled vortex-synchronization behind a circular obstacle with a
17 synthetic jet positioned at the rear stagnation point at $Re=9.5 \times 10^2$. They observed that the
18 symmetric shedding mode weakens the interaction between the upper and lower wake
19 vortices and decreases turbulent kinetic energy produced by the vortices. Also, this mode has
20 an influence on the velocity fluctuations. Joseph et al. (2013) performed an experimental flow
21 control on the Ahmed body using micro-electro mechanical system (MEMS) pulsed micro
22 jets. They found that the drag reduction (up to 10%) is occurred with using the micro-jets
23 located upstream the recirculation wake created over the rear slant of the body. Chaligné et
24 al. (2014) controlled the wake-flow behind a 2-D square obstacle by a fluidic control system.
25 This system is made of pulsed jets placed at the upper edge of the model. This method can
26 modify the wake flow development and the static pressure distribution around the obstacle.
27
28
29
30
31
32
33
34
35
36
37
38
39
40
41
42
43
44
45
46
47
48
49
50
51
52
53
54
55
56
57
58
59
60
61
62
63
64
65

1 Liu and Feng (2015) numerically suppressed the lift fluctuations on a circular obstacle by two
 2 synthetic jets for $Re=5.0 \times 10^2$. In this study, the unsteady Navier–Stokes equations were
 3 solved by utilizing a finite volume method. These jets were placed at the mean separation
 4 points. They focused on reduction of the lift fluctuations by changing the vortex shedding
 5 mode. These jets were placed at the mean separation
 6 points. They focused on reduction of the lift fluctuations by changing the vortex shedding
 7 mode. Their results showed that the complete suppression of lift fluctuations can be achieved
 8 when the typical Karman type vortex shedding are converted into the symmetric shedding
 9 modes. In some cases, base bleed or base suction effects are used for controlling the physics
 10 of flow around an obstacle.
 11
 12
 13
 14
 15
 16
 17
 18

19 The base bleed is secondary flow that is injected at the base of the truncated plug, see e.g.
 20 Rashidi et al. (2013). The streamline contours past a porous diamond obstacle are shown in
 21 figure 7. As shown in figure 7(a), the wake is completely detached from the cylinder wall for
 22 this range of parameters ($Da=1.0 \times 10^{-3}$ and $Re=29$), due to the base bleed effects. In this
 23 figure, the Darcy number is defined as:
 24
 25
 26
 27
 28
 29
 30

$$31 \quad Da == K / D^2 \quad (2)$$

32 where, K and D are the permeability of porous medium and cylinder diameter, respectively.
 33
 34 Also, figure 7(b) shows that the wake is permeated into the porous obstacle for $Da=10^{-6}$ and
 35
 36
 37
 38
 39
 40
 41
 42
 43
 44
 45
 46
 47
 48
 49
 50
 51
 52



53 **Fig. 7.** Configuration of streamlines around a porous diamond obstacle for (a) $Da=1.0 \times 10^{-3}$
 54 and $Re=29$; (b) $Da=1.0 \times 10^{-6}$ and $Re=6$ (Flow is from left to right; Figure reprinted from
 55
 56
 57
 58 Valipour et al. (2014b) with permission from the publisher)
 59
 60
 61
 62
 63
 64
 65

1 Fu and Rockwell (2005) experimentally controlled the vortex formation in the near wake
 2 for the shallow flow past a vertical cylinder by base bleed through a very narrow slot. They
 3 observed that the patterns of streamline topology and Reynolds stress at the bed change even
 4 by small bleed and these changes have important consequences for the bed loading.
 5
 6
 7
 8

9 Finally, this method is classified as active and wake control method. This technique is an
 10 important method to control the flow separation. Moreover, this method significantly affects
 11 the stability of the boundary layer and the transition to turbulence. As an active method,
 12 typically a high source of energy is required for this method especially for underwater flows
 13 in ocean engineering applications. Beside this, base bleed, among other techniques, requires a
 14 relatively high power input because it is applied inside the recirculation region. As mentioned
 15 by researchers, the forcing close to the separation point generally has the foremost control
 16 efficiency. Hence, it is recommended to consider this point for improving the efficiency of
 17 this method. Researchers reported a drag reduction up to 30% by synthetic jets (Feng et al.
 18 (2010)). Due to the alternative effects of unsteady blowing and suction, synthetic jets have
 19 good efficiency in delaying separation and reducing the drag coefficient. The mass-flux
 20 produced by these techniques is an important point that affects the energy consumption by
 21 these methods. The effect of the mass-flux requires further studies.
 22
 23
 24
 25
 26
 27
 28
 29
 30
 31
 32
 33
 34
 35
 36
 37
 38
 39
 40

41 Table 4 summarizes the researches on application of secondary flows to control the
 42 vortex shedding and flow separation of a bluff body. As presented in this table, most of these
 43 researches cover low Reynolds number regimes. The future researches should be focused on
 44 the critical or supercritical Reynolds numbers.
 45
 46
 47
 48
 49
 50

51 **Table 4:** Researches on application of generating a secondary flow (suction, blowing, bleed
 52 and synthetic jets) to control the vortex shedding and flow separation of a bluff body
 53
 54
 55

Authors	Type of research	Type of fluid	Reynolds number	Spatial dimensions	Type of bluff body	Type of injection
Seal and Smith (1999)	Experimental	Water	1.09×10^4	----	Circular	Suction insert at a streamwise location 90 cm from the

						leading edge of cylinder
Delaunay and Kaiktsis (2001)	Numerical	----	1-90	2D	Circular	Suction and blowing applied at the cylinder base
Tensi et al. (2002)	Experimental	Air	1.0×10^5	----	Circular	Synthetic jet at the wall of the cylinder
Li et al. (2003)	Numerical	----	Less than 1.1×10^2	2D	Circular	Injection and suction at the surface of the cylinder
Layek et al. (2008)	Numerical	----	6.0×10^2	2D	Square	Uniform suction and blowing at channel walls
Feng and Wang (2010)	Experimental	Water	9.5×10^2	----	Circular	Synthetic jet at the rear stagnation point
Feng et al. (2010)	Experimental	Water	9.5×10^2 and 1.6×10^3	----	Circular	Synthetic jet at the rear stagnation point
Feng et al. (2011)	Experimental	Water	9.5×10^2 and 1.8×10^3	----	Circular	Synthetic jet at the back stagnation point
Chen et al. (2013)	Experimental	Air	Unknown	----	Circular	Suction from lowest positions of the cylinder
Joseph et al. (2013)	Experimental	Air	1.1×10^6 - 2.8×10^6	----	Ahmed body	Micro-jets
Muralidharan et al. (2013)	Numerical	----	1.0×10^2 and 3.9×10^3	2D	Circular	Suction and blowing from the cylinder surface
Chaligné et al. (2014)	Experimental	Air	1.76×10^5	----	Square	Pulsed jets at upper edge of the cylinder
Liu and Feng (2015)	Numerical	----	5.0×10^2	2D	Circular	Two synthetic jet at the mean separation points
Sohankar et al. (2015)	Numerical	----	70 - 1.5×10^2	2D	Square	Uniform suction and blowing from the cylinder surface

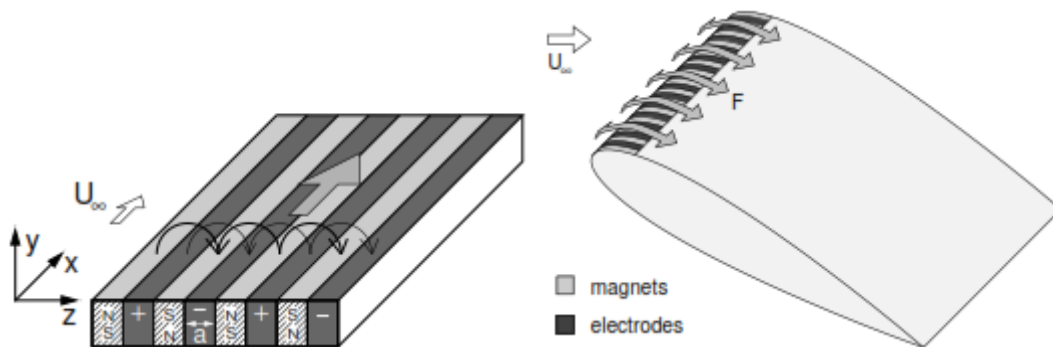
2.1.4. Control of vortex shedding by a magnetic field (MHD)

This method is classified as active and boundary layer control method.

Magnetohydrodynamics (MHD) is the physical-mathematical framework that is concerned with the dynamics of magnetic fields in electrically conducting fluids. Any movement of a conducting fluid in a magnetic field, B , generates electric currents, j . Each unit volume of liquid having j and B experiences a resistive type force, known as the Lorentz force. This force can be calculated by, see e.g. Rashidi et al. (2015b) and Valipour et al. (2014a):

$$F_{\text{Lorentz force}} = \sigma(V \times B \times B) \quad (3)$$

1 where V , B and σ are the fluid velocity vector, uniform magnetic field strength vector and
 2 electrical conductivity of the fluid, respectively. This force acts on the downstream flow in
 3
 4 opposite of the flow direction and reduces the flow velocity, leading to flow stabilization.
 5
 6 This method is shown to be effective in controlling the flows over tubular heat exchangers,
 7
 8 pipelines, suspension wires, and suspension bridges. Figure 8 shows the electrodes and
 9
 10 permanent magnets to generate Lorentz force to control the separated flows around a
 11
 12 hydrofoil. Figure 9 shows the effects of an external magnetic field in horizontal direction
 13
 14 (streamwise magnetic field) on vorticity contours. As shown in this figure, the flow stabilizes
 15
 16 and changes its distribution from the time-dependent behavior with vortex shedding to the
 17
 18 steady state with a symmetric shape along the centerline for strong magnetic field. It should
 19
 20 be noted that if the magnetic field is used in the horizontal direction, the vertical force acts in
 21
 22 the negative y -direction and this leads to retard the motion of the fluid (See Eq. 3).
 23
 24
 25
 26
 27
 28
 29
 30
 31
 32
 33
 34
 35
 36
 37
 38
 39
 40
 41



42 **Fig. 8.** Electrodes and permanent magnets to generate Lorentz force to control the separated
 43
 44 flows around a hydrofoil (Figure reprinted from Weier and Gerbeth (2004) with permission
 45
 46
 47 from the publisher)
 48
 49
 50
 51
 52
 53
 54
 55
 56
 57
 58
 59
 60
 61
 62
 63
 64
 65

1
2
3
4
5
6
7
8
9
10
11
12
13
14
15
16
17
18
19
20
21
22
23
24
25
26
27
28
29
30
31
32
33
34
35
36
37
38
39
40
41
42
43
44
45
46
47
48
49
50
51
52
53
54
55
56
57
58
59
60
61
62
63
64
65

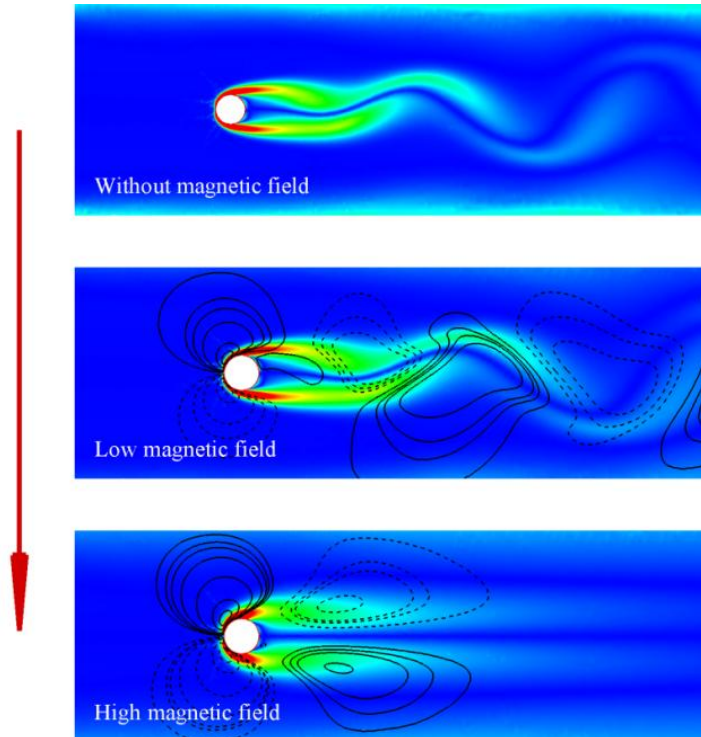


Fig. 9. Contours of vorticity at different magnetic field strength at $Re=100$ (Flow is from left to right)

Mutschke et al. (1998) studied the problem of controlling wake behind a circular cylinder by utilizing magnetic fields in liquid metal flows. They performed a numerical simulation on the unsteady two-dimensional flow. Weier and Gerbeth (2004) experimentally investigated the application of time periodic Lorentz forces in controlling the suction side flow on a NACA 0015 hydrofoil for the low Reynolds number range of $10^4 < Re < 10^5$. It was found that the Lorentz force allows for a great flexibility to provide the time dependency of the forcing. Chen and Aubry (2005) numerically developed an active control algorithm for manipulating wake flows past a two-dimensional circular obstacle. They used control actuators in their algorithm, where the actuators exert Lorentz forces over the entire surface of the obstacle. Their results showed that vortex shedding may completely disappear and the total drag coefficient would decrease as the Lorentz force increases for $N > 3$, where N is the interaction parameter, defined as the ratio of the electromagnetic force to the inertial force. This was found to be due to the predominant role played by the pressure on the total drag coefficient.

1 Control of vortex shedding from a bluff body using imposed magnetic field was studied
2 numerically by Singha et al. (2007). A 2-D incompressible laminar viscous flow past a square
3 obstacle placed inside the channel was considered. This study showed that the periodic vortex
4 shedding can be completely eliminated by applying a strong magnetic field. Singha and
5 Sinhamahapatra (2011) numerically investigated effects of an imposed transverse magnetic
6 field on control of vortex shedding from a circular cylinder for Reynolds numbers varying
7 from 50 to 250. They observed that the separated zone behind the obstacle in a steady flow
8 decreases with increase in the magnetic field strength. Chatterjee et al. (2012) controlled flow
9 separation around 2-D bluff bodies by exerting a transverse magnetic field. The computations
10 were performed for low Reynolds numbers in the range of 10 to 40, and two-dimensional
11 circular and square cross sections. It was found that length of the wake region and separation
12 angle decrease with the increase in the magnetic field strength.

13 Zhang et al. (2014) presented a suppression mechanism for vortex-induced vibration by
14 symmetric Lorentz forces. They used a 2-D moving circular cylinder and numerically studied
15 the effects of Lorentz force for controlling the vortex-induced vibration (VIV) of a circular
16 cylinder at $Re=150$. They concluded that the applied Lorentz force affects the phase angles
17 among the obstacle displacement, the total lift force and the vortex induced lift force. Rashidi
18 et al. (2015c) controlled the wake structure behind a square obstacle in channel flow by
19 exerting a horizontal magnetic field. They defined two parameters for controlling the flow:
20 critical Stuart number, defined as the Stuart number where unsteady flow becomes steady;
21 and disappearance Stuart number, defined to measure the strength of the inserted magnetic
22 field required for disappearing the recirculating wake. They observed that critical and
23 disappearance Stuart numbers increase for higher Reynolds numbers. Bovand et al. (2015c)
24 repeated this problem for a porous circular obstacle and they used a cross-radial magnetic
25 field to control the flow. They found that the value of critical Stuart number for suppressing

1 the vortex shedding decreases when Darcy number increases. Also, the disappearance Stuart
2 number decreases with increase in Darcy numbers.
3

4 Rashidi and Esfahani (2015) controlled instabilities of heat transfer from an obstacle in a
5 channel by a horizontal magnetic field, and showed that existence of channel wall decreases
6 the effects of Stuart number on heat transfer compared with corresponding infinite-domain
7 case. Also, they found that unsteadiness in temperature field increases with increase in
8 Reynolds number. Therefore, a stronger magnetic field is needed to control the instabilities of
9 heat transfer at higher Reynolds number.
10
11
12
13
14
15
16
17
18

19 Finally, this method is classified as an active boundary layer control method. This method
20 is required an electrically conducting fluid (i.e. seawater) as working fluid. Therefore, this
21 method is very practical in ocean engineering applications. A resistive force (Lorentz force)
22 is created by applying a magnetic field in a moving fluid. This force tends to retard the
23 motion of the fluid and can be used as a damping force to suppress the vortex shedding from a
24 bluff body. The magnetic field can vary in space and time and therefore, it can create a
25 flexible force, which can be adjusted to achieve a specific purpose. No mechanically moving
26 parts are necessary to persuade this changeable force, so there is no reason for additional
27 attrition. As discussed in literature, the drag force can be decreased by a suitable selection of
28 Lorentz force distribution that can be used in marine vehicles, see e.g. Bovand et al. (2015c).
29
30
31
32
33
34
35
36
37
38
39
40
41
42
43

44 Table 5 summarizes the researches on application of MHD to control the vortex shedding
45 and flow separation of a bluff body. As shown in this table, the ranges of Reynolds numbers
46 are very low. Such low values of Reynolds number are only practical for some liquid metal
47 flows in nuclear or semi-conductor applications and microfluidic applications. Therefore,
48 more studies are required to cover the critical and transcritical Reynolds numbers (i.e. larger
49 than 1.0×10^5). Moreover, most of researches performed using this method, are numerical.
50
51
52
53
54
55
56
57
58
59
60
61
62
63
64
65

as a volume force on entire the domain. It would be interesting to have a tunable force to act only in specific regions of the flow. This supposition can be used in future researches on applications of this method in flow control.

Table 5: Researches on application of MHD to control the vortex shedding and flow separation of a bluff body

Authors	Type of research	Type of fluid	Reynolds number	Spatial dimensions
Mutschke et al. (1998)	Numerical	---	$50-3.0 \times 10^2$	2D/3D
Weier et al. (1998)	Experimental/ Numerical	Solution of 10% copper sulphate (CuSO_4) and 5% sulphuric acid (H_2SO_4)	2.0×10^2	2D/3D
Posdziech and Grundmann (2001)	Numerical	---	$10-3.0 \times 10^2$	2D
Weier and Gerbeth (2004)	Experimental	Electrolyte (NaOH solution)	$5.2 \times 10^4 - 1.5 \times 10^5$	---
Chen and Aubry (2005)	Numerical	---	2.0×10^2	2D
Mutschke et al. (2006)	Numerical	---	5.0×10^2 and 6.0×10^2	2D
Singha et al. (2007)	Numerical	---	$50-2.5 \times 10^2$	2D (Square cylinder)
Singha and Sinhamahapatra (2011)	Numerical	---	$50-2.5 \times 10^2$	2D (Circular cylinder)
Chatterjee et al. (2012)	Numerical	---	10-40	2D
Zhang et al. (2014)	Numerical	---	1.5×10^2	2D
Bovand et al. (2015c)	Numerical	---	$1-2.0 \times 10^2$	2D
Rashidi et al. (2015c)	Numerical	---	$1-2.5 \times 10^2$	2D

2.1.5. Control of vortex shedding by rotary oscillations or rotary non-oscillations

This method is classified as active and boundary layer control method. A positive or negative lift can be created by rotating a cylinder. The rotation in this method can be both oscillating or non-oscillating. Note that constant rotation results in no oscillations. The direction of lift depends on the direction of the spin. This force can lead to a complete vortex shedding behind the obstacle. In this method, the near-wake structure is significantly modified by the interaction between the rotationally oscillating surface and the surrounding fluid. It is worth mentioning that the oscillation of an obstacle leads to weaken the wake and transfers the formation region closer behind the obstacle. This interaction leads to acceleration or deceleration of the fluid flow around the obstacle according to the direction of rotation. This method is widely used to control the flow of tube heat exchangers, shafts,

drilling of oil wells, nuclear reactor fuel rods and steel suspension bridge cables. Figure 10 shows an experimental setup to control the wake behind a circular cylinder by rotary oscillations. Figure 11 shows the streamlines and vorticity at different rotational speeds at $Re = 1.0 \times 10^2$. As shown in this figure, a complete suppression of flow unsteadiness can be achieved with an increase in rotation rate. In this figure, α is the dimensionless rotation rate given by:

$$\alpha = \frac{D\omega}{2U} \quad (4)$$

where ω , U , and D are the rotational speed of cylinder, free flow velocity and cylinder diameter, respectively.

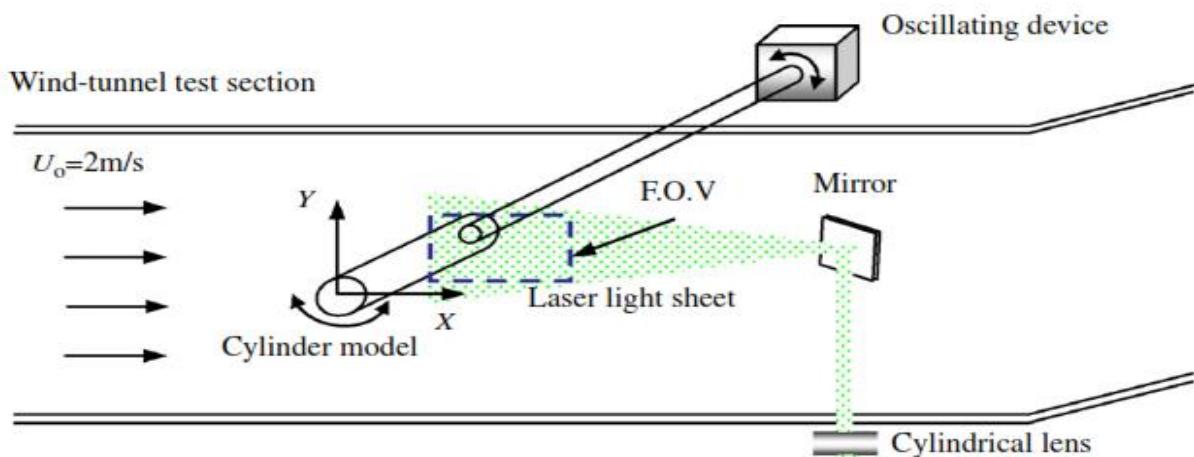


Fig. 10. An experimental setup to control the wake behind a circular cylinder by rotary oscillations (Figure reprinted from Joon Lee and Yeop Lee (2008) with permission from the publisher).

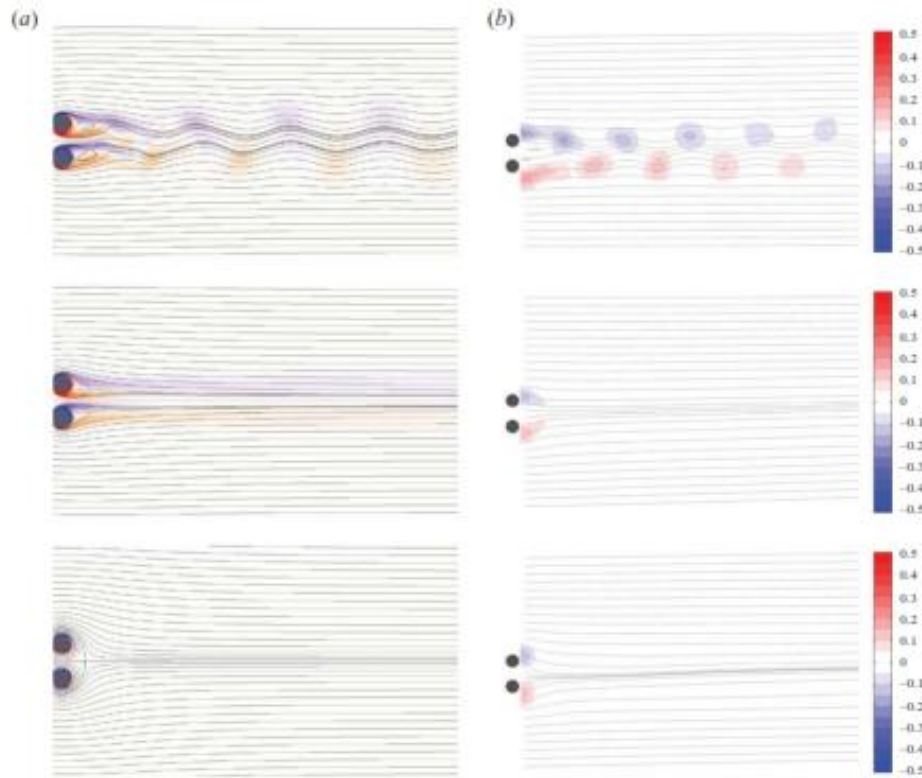


Fig. 11. Configuration of streamlines and vorticity at different rotational speeds $Re = 100$ a)

Computational results; b) Experimental results. First row, $\alpha=1.2$; second row, $\alpha=1.5$; third row $\alpha=3$ (Flow is from left to right; Figure reprinted from Chan et al. (2011) with permission from the publisher).

Baek and Sung (1998) performed a numerical study on the flow behind a rotary oscillating circular obstacle at $Re=1.1 \times 10^2$. They found that the counterclockwise rotation leads to large-scale vortex move downward. Sengupta et al. (2007) controlled the flow for a 2D circular obstacle executing rotary oscillation using genetic algorithm at $Re=1.5 \times 10^4$. The 2D Navier–Stokes equations are solved numerically by using a fast viscous-vortex method. They calculated the maximum rotation rate and the forcing frequency of the rotary oscillation for the purpose of drag reduction. Joon Lee and Yeop Lee (2008) presented experimentally the PIV measurements of the wake behind a rotationally oscillating circular obstacle at $Re=4.14 \times 10^3$. They observed that the length of the vortex formation region decreases with increase in the forcing frequency when the frequency ratio is smaller than 1.0. Also, they

1 showed that the rotational oscillatory motion of a circular obstacle is an effective and
2 promising technique for controlling the near-wake flow structure. The method is used by
3 others, e.g., Korkischko and Meneghini (2012). Chan et al. (2011) investigated numerically
4 and experimentally the vortex suppression and drag reduction around a pair of counter-
5 rotating circular cylinders for $Re=1.0\times 10^2-2.0\times 10^2$. They observed that the unsteady wake
6 can be suppressed by creating a rotation for each cylinder in the opposite direction. They
7 claimed that the drag force can be reduced to zero for a higher rotational speeds. Flinois and
8 Colonius (2015) suppressed the vortex shedding of a circular cylinder using body rotation.
9 They were able to reduce the drag by about 19% and effectively suppressed the vortex
10 shedding.
11
12
13
14
15
16
17
18
19
20
21
22
23

24 Finally, this method is classified as an active boundary layer control method. The
25 oscillation of an obstacle leads to weakening the wake and transfers the formation region
26 closer behind the obstacle. Moreover, the interaction between the rotationally oscillating
27 cylinder and the surrounding fluid leads to modification of the near-wake structure
28 substantially by accelerating or decelerating the flow near the cylinder, depending on the
29 rotation direction. The drag force may be reduced up to 80% at a high forcing frequency.
30 Since the rotation rates can be adjusted by a simple electronic device and mechanical means,
31 this control method is suitable for practical applications. This method can be used to control
32 the near-wake flow structure.
33
34
35
36
37
38
39
40
41
42
43
44
45

46 Table 6 summarizes researches on the application of rotary oscillations to control the
47 vortex shedding and flow separation of a bluff body. A comparison between different
48 directions of oscillating, i.e. in-line or transverse linear oscillations, would be interesting for
49 future researches. In addition, there are some other types of oscillation such as flow
50 perturbations and imposed sound field that can be used as a flow control method. Although,
51
52
53
54
55
56
57
58
59
60
61
62
63
64
65

there is a few research about these, e.g. Armstrong et al. (1986) and Blevins (1985), but there is need for further studies as these types are more applicable in ocean engineering.

Table 6: Researches on application of rotary oscillations to control the vortex shedding and flow separation of a bluff body

Authors	Type of research	Type of fluid	Reynolds number	Spatial dimensions	Type of rotation
Tokumaru and Dimotakis (1989)	Experimental	water	1.5×10^4	---	Oscillating
Baek and Sung (1998)	Numerical	----	1.1×10^2	2D	Oscillating
Sengupta et al. (2007)	Numerical	---	1.5×10^4	2D	Oscillating
S.J. Lee and J.Y. Lee (2006)	Experimental	Air	4.14×10^3	---	Oscillating
Chan et al. (2011)	Experimental / Numerical	water	1.0×10^2 - 2.0×10^2	2D for numerical part	Non-oscillating
Flinois and Colonius (2015)	Numerical	---	75-200	2D	Oscillating/ Non-oscillating

2.1.6. Control of vortex shedding by thermal effects

This method is classified as active and boundary layer control method. Buoyancy effects around a bluff body appear in the near wake when the level of heating is high. Buoyancy effect creates inertia and viscous forces around the obstacle and this results in to delay in flow separation and suppression of the vortex shedding. Increase of the heat input in the obstacle leads to a suppression of the vortex shedding for air flow while for the liquid flowing around a heated obstacle this effect is inverse. This opposite behaviour for gases and liquids suggest that this control is due to both changes of dynamic viscosity and density with temperature. The changes of the viscosity and the density lead to slight changes in velocity profiles and consequently amplitude of the shed vortices and instability conditions. The thermal buoyancy is a natural way to control the boundary layer separation over a cylinder. Therefore, it can be used in many engineering applications. Figure 12 shows the application of thermocouple to measure the temperature gradient on the surface of the cylinder for controlling the flow.

Figure 13 shows the contours of streamlines around a square obstacle for different Richardson numbers is defined as $Ri=Gr/Re^2$, where $Gr = g\beta(T_w - T_\infty)D^3 / \nu^2$ and g, β, T_∞, T_w and ν are the gravitational acceleration, volumetric expansion coefficient, free stream temperature, cylinder temperature and kinematic viscosity of fluid, respectively. As the buoyancy effect increases, Ri becomes larger. In this method, location of separation point changes due to the increase of fluid dynamic viscosity. As shown in figure 13, as Ri increases, and for a constant Re , the flow stabilizes and changes distribution from a time-dependent flow with vortex shedding to steady state flow with a symmetric shape along the centerline. In this figure, spatial coordinates in the horizontal and vertical directions are dimensionless.

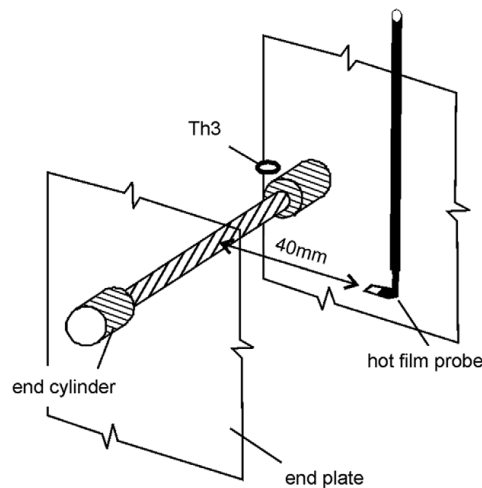


Fig. 12. Application of thermocouple to measure the temperature gradient on the surface of the cylinder for controlling the flow (Figure reprinted from Vit et al. (2007) with permission from the publisher)

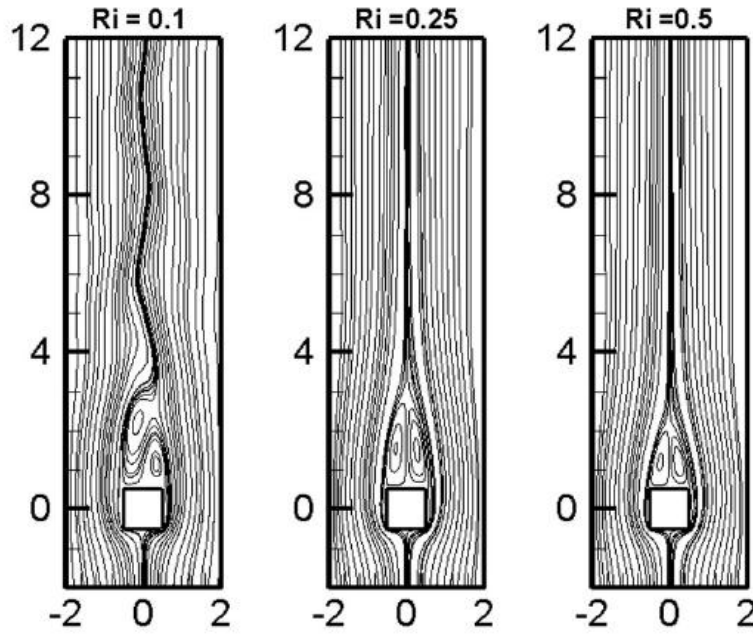


Fig. 13. Contours of streamlines around square obstacle for different Richardson numbers

(Ri) at a constant Reynolds number, $Re = 1.0 \times 10^2$ (Flow is from down to up; Figure reprinted from Chatterjee (2014) with permission from the publisher)

This method was used by Lecordier et al. (1991) in an experimental study on control of vortex shedding behind circular cylinders at low Reynolds numbers by heating the cylinder. They found that the thermal effect method can lead to a complete suppression of periodic oscillations in the wake region for $90 < Re < 1.2 \times 10^2$. Later, Lecordier et al. (2000) experimentally controlled the formation of vortex shedding behind two 2-D bluff bodies (circular cylinder and a flat ribbon) by thermal effect at low Reynolds numbers. It was found that the method mainly depends on the type of the fluid. They showed that a suppression of vortex shedding occurs with increase in the heat input to the cylinder or ribbon for air flow, however, an opposite effect was observed in a water flowing around a heated cylinder or ribbon. Therefore, both cooling and heating can be used in this method.

In a numerical study, Chatterjee (2014) investigated the dual role of thermal effects for controlling boundary layer separation around cylinders. Circular and square shaped cylinders were considered in this research. They observed that the thermal buoyancy can change the

1 flow from time-dependent pattern with vortex shedding in the wake to steady re-circulatory
2 flow with complete suppression of vortex shedding. Also, the steady flow may turn into an
3 unsteady periodic flow with initiation of vortex shedding by the action of thermal buoyancy
4 when the buoyancy acts in a cross wise direction to the incoming flow. Therefore, the
5 buoyancy has positive or negative effects on the flow control. It acts as a destabilizer when
6 acting perpendicular to the flow direction. They showed that the minimum heating that is
7 needed to suppress the vortex shedding increases with increase in Reynolds number.
8 Moreover, they observed that a square obstacle requires more heating to suppress the vortex
9 shedding in comparison to the circular one.

10
11
12
13
14
15
16
17
18
19
20
21
22 Chatterjee and Mondal (2014) numerically controlled the flow separation around bluff
23 obstacles by superimposing thermal buoyancy. Results show that, for lower Prandtl number,
24 higher heating is required for a complete suppression of flow separation around the bluff
25 bodies. Circular and square cross-sections were used, and it was found that the square
26 obstacle requires more heating when compared with the corresponding circular obstacle for
27 complete suppression of flow separation. Chatterjee and Ray (2014) repeated the study for a
28 triangular obstacle, and found that the wake length decreases with increasing strength of
29 buoyancy and it completely vanishes with further increase in strength of buoyancy.

30
31
32
33
34
35
36
37
38
39
40
41
42 Finally, this method is classified as active boundary layer control method. According to
43 this method, the density and viscosity of the fluid are changed due heating the fluid. This
44 leads to creation of buoyancy forces and slight changes in velocity profiles in near wake and
45 consequently instability of the shedding vortices. We considered this method in active
46 classifications, however thermal buoyancy can be inherently a natural way to control the
47 flow. Note that the thermal buoyancy can have a dual role. It stabilizes the flow when acting
48 along the flow direction and conversely it destabilizes the flow when acting perpendicular to
49 the flow direction. As shown by researchers, e.g. Chatterjee and Ray (2014), the change in the
50
51
52
53
54
55
56
57
58
59
60
61
62
63
64
65

fluid type has considerable effects on the suppression behaviour in this method. More heating is required to completely suppress the flow separation around the bluff bodies for the fluids with low values of Prandtl number. Therefore, this method is more suitable for water rather than air from the viewpoint of energy consumption.

Table 7 summarizes researches about application of thermal effects to control the vortex shedding and flow separation of a bluff body. As shown in this table, these researches covered very low Reynolds number and two-dimensional objects. As a result, the ability of this method to suppress the vortex shedding for moderate and high Reynolds numbers is questionable and more research is necessary to answer this question. Note that the flow separation is controlled at lower values of Reynolds number (i.e. $Re < 40$) while, the vortex shedding is suppressed for larger Reynolds number ($50 < Re < 1.5 \times 10^2$). None of these researches (except the paper of Chatterjee and Sinha (2014)) focused on the effects of heating on aerodynamic or hydrodynamic forces such as drag and lift forces. Therefore, further studies are necessary for these forces.

Table 7: Researches on application of thermal effects to control the vortex shedding and flow separation of a bluff body

Authors	Type of research	Type of fluid	Reynolds number	Spatial dimensions
Lecordier et al. (1991)	Experimental	Air	$45-1.2 \times 10^2$	---
Lecordier et al. (2000)	Experimental	Air and water	34-75	---
Chatterjee (2014)	Numerical	Air	$10-1.5 \times 10^2$	2D
Chatterjee and Mondal (2014)	Numerical	Air and water	10-40	2D
Chatterjee and Ray (2014)	Numerical	Pr=50(Pr denotes the Prandtl number)	5-30	2D
Chatterjee and Sinha (2014)	Numerical	Air	5-45	2D

2.1.7. Control of vortex shedding by other active methods

Huang (1995) experimentally suppressed the vortex shedding from a circular obstacle by internal acoustic excitation for Reynolds numbers in a range of $Re=4 \times 10^3$ to $Re=8.0 \times 10^3$. Their results showed that the vortex shedding peak decreases noticeably only in a very

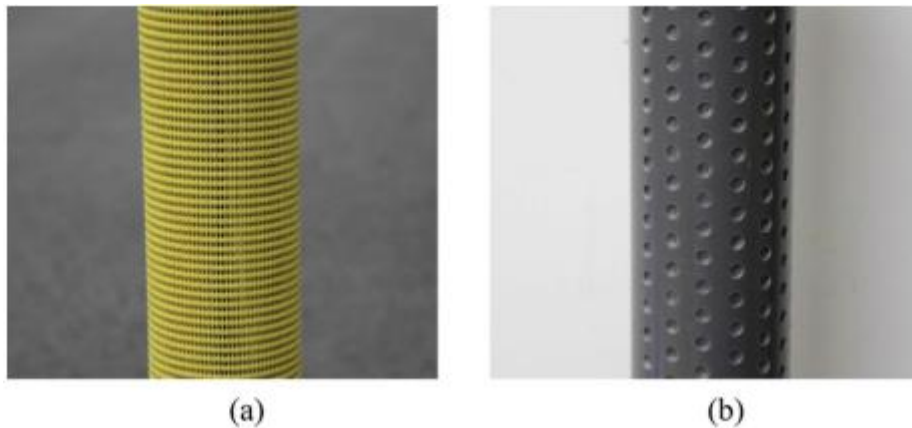
1 narrow excitation range. The optimal sound level is located at this range. Bimbato et al.
2 (2013) numerically controlled vortex shedding on a 2-D circular obstacle by using a moving
3 ground plane for $Re=1.0\times 10^5$. Obstacle is placed near this ground plane. Their results showed
4 that when a circular obstacle is placed closer to the ground, the vortex shedding suppresses by
5 Venturi effect, and drag force decreases due to this suppression.
6
7
8
9
10

11 **2.2. Passive method**

12 **2.2.1. Control of vortex shedding by surface roughness**

13
14 Surface roughness is classified as passive and boundary layer control method. Surface
15 roughness affects the location of boundary layer separation point and consequently the
16 unsteady and steady force around the obstacle. These effects are created in boundary layer by
17 two approaches. The transition moves forward on the surface of obstacle due to the roughness
18 on the surface; and the roughness makes the velocity profile less full in beneath the turbulent
19 boundary layer, see e.g. Shih et al. (1993). This method has been typically used as a passive
20 control approach, and is utilize in offshore and marine application, such as platform pillars,
21 pipelines and risers, see e.g. Zhou et al. (2015). Figure 14 shows two rough cylinders that
22 were used in the experiments conducted by Zhou et al. (2015) to control the flow. Chang et
23 al. (2011) performed an experimental passive turbulence control (PTC) by surface roughness
24 for a circular obstacle in a steady flow at $3.0\times 10^4 < Re < 1.2\times 10^5$. They showed that for a
25 smooth obstacle about seven vortices shed per cycle, but it decreased to about five vortices
26 per cycle for a rough obstacle. Kiu et al. (2011) investigated the effects of uniform surface
27 roughness on vortex-induced vibration of towed vertical cylinders. They found that the rough
28 obstacles have a higher Strouhal number than a smooth obstacle. Gao et al. (2015)
29 experimentally studied the effects of surface roughness on the vortex-induced vibration
30 response of a flexible obstacle. They observed that the displacement response decreases and
31 the vortex-shedding frequency increases with increase in surface roughness. Also, rough
32
33
34
35
36
37
38
39
40
41
42
43
44
45
46
47
48
49
50
51
52
53
54
55
56
57
58
59
60
61
62
63
64
65

1 obstacles have a narrower lock-in region than a smooth obstacle. In another study, Zhou et al.
2 (2015) experimentally investigated the force and flow characteristics of a circular obstacle
3 with uniform surface roughness at $6.0 \times 10^3 < Re < 8.0 \times 10^4$. They found that the mean drag
4 coefficient on the cylinder and the root-mean-square of lift coefficient decrease with increase
5 in surface roughness.
6
7
8
9
10



27 **Fig. 14.** The rough circular cylinders with (a) netting and (b) dimples. (Figure reprinted from
28 Zhou et al. (2015) with permission from the publisher)
29

30 Finally, this method is classified as passive boundary layer control method. As mentioned
31 in the literatures, surface roughness leads to delay in boundary layer separation. Moreover,
32 there is a faster transition from laminar to turbulent regime for a bluff body with rough
33 surfaces. This leads to an abrupt reduction in the drag coefficient, see e.g. Achenbach (1971).
34 Therefore, this method is suitable for subcritical Reynolds numbers. This method has a more
35 practical application due to its easier manufacturing, simpler installation and lower costs. The
36 marine organisms grow on the surface of the marine structures over a period of time. This
37 leads to increase in the surface roughness of the structure. As a result, it is important to study
38 the effects of surface roughness.
39
40
41
42
43
44
45
46
47
48
49
50
51
52
53

54 Gao et al. (2015) summarized the recent researches on the application of the surface
55 roughness to control the vortex shedding behind a bluff body. The results of these researches
56 are presented in table 8. The ranges of Reynolds number, aspect ratio, surface roughness, and
57
58
59
60
61
62
63
64
65

cylinder conditions for each research are presented in this table. There are very few researches on the effect of surface roughness for oscillating cylinder in wind. The VIV (vortex-induced vibration) for a bluff body is important in wind flows. These papers focused on determining the pressure distribution, flow separation, and Strouhal number characteristics. However, the aerodynamic excitation of the body is not discussed in these papers. Structural–dynamic response is important from the practical point of view. Accordingly, many questions remain unanswered in wind flows. Moreover, most of the studies on this control method are conducted experimentally, while the numerical studies in this field are very limited. Most of previous studies focused on the uniform distribution of roughness over the entire of the surface. More attention is required to study the effects of roughness distributions and local roughness. It can also be seen that this method works well at subcritical Reynolds numbers but does not work very good at a lower values of Reynolds number, see e.g. Achenbach (1971), Achenbach and Heinecke (1981).

Table 8: Previous studies on the surface roughness controls for flow over a cylinder (Table reprinted from Geo et al. (2015) with permission from the publisher)

Authors	Type of research	Aspect ratio (Cylinder length/Cylinder diameter)	Cylinder conditions	Reynolds number	Surface roughness (equivalent roughness height/Cylinder diameter)	Type of roughness
Air						
Achenbach (1971)	Experimental	3.33	Stationary	$4.0 \times 10^4 - 3.0 \times 10^6$	$1.1 \times 10^{-3} - 9.0 \times 10^{-3}$	Sand roughness
Achenbach and Heinecke (1981)	Experimental	3.38	Stationary	$6.0 \times 10^3 - 5.0 \times 10^6$	$7.5 \times 10^{-4} - 3.0 \times 10^{-2}$	Pyramidal roughness
Nakamura and Tomonari (1982)	Experimental	3.33	Stationary	$4.0 \times 10^4 - 1.7 \times 10^6$	$9.0 \times 10^{-4} - 1.0 \times 10^{-2}$	Roughness strips
Ribeiro (1991a, b)	Experimental	6.1	Stationary	$5.0 \times 10^4 - 4.0 \times 10^5$	$1.8 \times 10^{-3} - 1.2 \times 10^{-2}$	Sand paper, wire screen and nylon ribs
Bearman and Harvey (1993)	Experimental	12.26	Stationary	$2.0 \times 10^4 - 3.0 \times 10^5$	$4.5 \times 10^{-3} - 9.0 \times 10^{-3}$	Dimpled surface
Shih et al. (1993)	Experimental	8	Stationary	$1.0 \times 10^5 - 1.0 \times 10^7$	$3 \times 10^{-4} - 1.01 \times 10^{-2}$	Screens
Okajima et al. (1999)	Experimental	1.83	Oscillating	$2.5 \times 10^4 - 3.2 \times 10^5$	$5.0 \times 10^{-3} - 3.8 \times 10^{-8}$	Spherical glass bead particles or 64 stranded cables with 6 mm diameter that are attached to the cylinder surface
Water						
Allen and Henning (2001)	Experimental	84.6	Oscillating	$1.8 \times 10^5 - 6.5 \times 10^5$	$5.1 \times 10^{-5} - 5.8 \times 10^{-3}$	Rough pipes that consisted simply of

1						macro-spheres glued to their surface	
2	Bernitsas et al. (2008); Bernitsas and Raghavan (2008)	Experimental	7.2-14.4	Oscillating	$8.0 \times 10^3 - 2.0 \times 10^5$	$1.4 \times 10^{-3} - 4.2 \times 10^{-3}$	Sandpaper strips
6	Kiu et al. (2011)	Experimental	8.0	Oscillating	$1.7 \times 10^4 - 8.3 \times 10^4$	$2.8 \times 10^{-4} - 1.4 \times 10^{-2}$	Sandpapers with known mean particle diameters
7	Chang et al. (2011)	Experimental	10.3	Oscillating	$3.0 \times 10^4 - 1.2 \times 10^5$	$5.4 \times 10^{-4} - 4.68 \times 10^{-3}$	Roughness strips
10	Zhou et al. (2015)	Experimental	10	Stationary	$6.0 \times 10^3 - 8 \times 10^4$	$2.8 \times 10^{-3} - 2.5 \times 10^{-2}$	Netting and dimples
11	Gao et al. (2015)	Experimental	48.32	Oscillating	$2.5 \times 10^4 - 1.8 \times 10^5$	$1.14 \times 10^{-4} - 1.2 \times 10^{-2}$	Sand with different particle diameters

2.2.2. Control of vortex shedding by porous and permeable walls

This method is classified as passive wake control method. Gozmen et al. (2013) experimentally controlled the downstream flow behind a circular obstacle in deep water by using an outer permeable obstacle. They reported that the outer permeable obstacle significantly suppresses the vortex shedding behind two obstacles with increase in the porosity defined as ratio of the gap area on the region to the whole body surface area. Also, they found that porosity of 0.7 is the most suitable case to control the vortex shedding behind the obstacle. A fixed value of Reynolds number, $Re=5.0 \times 10^3$, was considered in all experiments in this study. In another research, Mimeau et al. (2014) numerically controlled the flow around a semi-circular cylinder by using a porous layer for laminar and transitional flows. The incompressible Navier-Stokes equations were used to describe the dynamics of flow. They achieved relevant control performances by introducing porous layers only at the top and bottom of the solid body permits. Also, they observed a reduction in drag by using a thin layer with intermediate permeability in both edges of the back wall.

Ozkan et al. (2012) experimentally controlled the flow characteristics around a circular cylinder by a porous outer cylinder in shallow water. They used Particle Image Velocimetry technique and suppressed the formation of an organized vortex street by porous outer cylinder. They observed that the effect of porous outer cylinder on the characteristics of inner cylinder decreases with the increase in the porosity. Pinar et al. (2015) performed an

1 experimental study on the flow structure around permeable circular obstacles. They used
2 Particle Image Velocimetry technique and found that the porosity had a significant effect on
3 the control of large-scale vortical structures downstream of the obstacle. They attenuated
4 significantly the fluctuations and prevented the formation of Karman Vortex Street by use of
5 permeable cylinders.
6
7
8
9

10 Valipour et al. (2014b) numerically controlled the flow around and through a porous
11 obstacle with diamond cross section for $Re < 50$. They observed that the critical Reynolds
12 numbers to onset of the re-circulating wake for a porous obstacle is larger than that of a solid
13 obstacle for intermediate and large Darcy numbers. Also, they found that this critical
14 Reynolds number for a square obstacle is smaller than that of the diamond obstacle. Rashidi
15 et al. (2014b) repeated this study for a porous diamond obstacle with different apex angles at
16 $Re < 50$. The computed results showed that the critical Reynolds number for the onset of the
17 wake in a porous obstacle decreases with an increase in the apex angle. Note that finite
18 volume method was used to solve the continuity and Navier–Stokes equations in above two
19 papers.
20
21
22
23
24
25
26
27
28
29
30
31
32
33
34
35

36 Finally, this method is classified as a passive wake control method. The main control
37 parameter in this method is porosity or void fraction of porous medium. Highly permeable
38 media (higher porosity) are more efficient, as presented by researchers, see e.g. Rashidi et al.
39 (2014b).
40
41
42
43
44
45

46 Table 9 summarizes researches on application of permeable wall to control the vortex
47 shedding and flow separation of a bluff body. The existing studies need to be complemented
48 by 3D researches to match the control trends that were observed in the 2D observations.
49 Generally, there are limited studies on the 3D applications.
50
51
52
53
54
55

56 **Table 9:** Researches on application of permeable wall to control the vortex shedding and
57 flow separation of a bluff body
58
59
60
61
62
63
64
65

Authors	Type of research	Type of fluid	Reynolds number	Spatial dimensions
Ozkan et al. (2012)	Experimental	Water	8.5×10^3	----
Gozmen et al. (2013)	Experimental	Water	5.0×10^3	----
Mimeau et al. (2013)	Numerical	----	5.5×10^2 and 3.0×10^3	2D
Rashidi et al. (2014b)	Numerical	----	1-45	2D
Valipour et al. (2014b)	Numerical	----	1-45	2D

2.2.3. Control of vortex shedding by other passive methods

Bearman and Owen (1998) experimentally controlled the vortex shedding from a plate by a spanwise sinusoidal form at Reynolds numbers of about 4.0×10^4 . They observed at least 30% reduction in drag for wavy plate in comparison with the equivalent straight bodies. Also, the vortex shedding is completely suppressed for specific ratios of peak-to-peak wave height to wavelength ratio. Lei et al. (2000) suppressed numerically vortex shedding behind a circular cylinder near a plane boundary by changing the gap ratios between cylinder and plate for different Reynolds numbers ranging from 80 to 1.0×10^3 . They observed that the critical gap ratio for suppressing vortex shedding is identified at different Reynolds numbers. Also, the effect of the gap ratio on the vortex shedding frequency increases by an increase in Reynolds numbers. In another research, Straatman and Martinuzzi (2002) performed a numerical study on vortex shedding phenomena from a square obstacle near a wall. This research indicated that the suppression of vortex shedding could not be expressed simply by a cancellation of vorticity mechanism, because the vorticity generated at the wall. Graham and Huang (2010) presented a passive flow control device by flow visualization techniques for a 2-D bluff body. They tested two endplates on the bluff body (with and without the passive control system). It was found that the flow visualization model did not succeed in fully demonstrating the characteristics of the entire wake due to a flaw in the dye system design. Huang (2011) experimentally suppressed vortex induced vibration (VIV) of a circular cylinder by using the helical grooves. Author introduced the triple-start helical grooves as an effective method to suppress the cylinder VIV response. Author observed 64% reduction in the

1 maximum cross-flow VIV amplitude by this method for the cases considered in the study.
2 Kunze and Brücker (2012) experimentally controlled vortex shedding behind a circular
3 cylinder using self-adaptive hairy-flaps method for a Reynolds number range of
4 $5.0 \times 10^3 < \text{Re} < 3.1 \times 10^4$. It was reported that the flow fluctuations are considerably decreased
5 when compared to the reference cases without hairy-flaps. The deductions were about 42%
6 and 35% for streamwise and transversal directions, respectively. Also, using self-adaptive
7 hairy-flaps, length of the separation wake behind the obstacle and rms-values (Root mean
8 square) of velocity components in horizontal and vertical directions within the wake region
9 were decreased. Xu et al. (2014) numerically suppressed the vortex-induced vibration of an
10 elastically mounted 2-D cylinder by a traveling wave wall at $\text{Re} = 2.0 \times 10^2$. They observed that
11 a series of small scale vortices and formed inside the troughs of the wave downstream. These
12 vortices control the flow separation from the obstacle wall, remove the oscillating wake and
13 suppress the vortex-induced vibration of the obstacle.
14
15
16
17
18
19
20
21
22
23
24
25
26
27
28
29
30

31 **2.3. Active/passive method**

32 **2.3.1. Control of vortex shedding by employing an external element**

33
34
35
36 Generally, external element method is classified as wake control method. For energy
37 consuming, this method can be placed in both classifications (Active or passive).
38

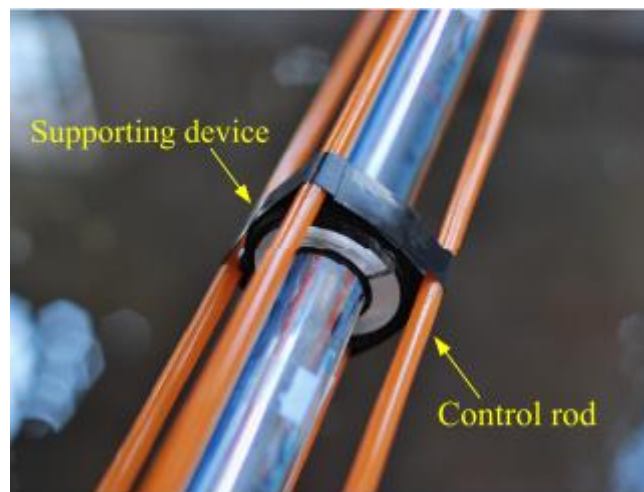
39
40
41 First, active methods are reviewed. Nishiyama et al. (1991) conducted experiments to control
42 the vortex shedding generated from three in-line elliptic cylinders in a uniform flow, with one
43 of the cylinders undergoing forced transverse vibration. Efficiency of this method depends on
44 the amplitude and frequency of the forced vibration, arrangement of the elliptic obstacles,
45 location of the vibrating obstacle and Reynolds number ($\text{Re} = \frac{U_\infty C}{\nu}$, where C is major axis
46 length of elliptic cylinder). Patnaik and Wei (2002) presented a control strategy for taming
47 the wake turbulence behind a square obstacle. In their strategy, two circular cylinders are cast
48 at the centers of the first and fourth quadrants of the main obstacle. These control cylinders
49
50
51
52
53
54
55
56
57
58
59
60
61
62
63
64
65

1 rotate in clockwise and counter clockwise fashions, respectively. It is found that the wake
2 turbulence suppresses significantly when the system is synchronized. Mittal (2001) utilized
3 finite element method and controlled the flow past a 2-D circular obstacle by using small
4 rotating cylinders for $Re=1.0\times 10^2$ and $Re=1.0\times 10^4$ and two values for the gap between the
5 main cylinder and the control cylinders. The computed results showed that the overall drag
6 coefficient and the unsteady aerodynamic forces acting on the main cylinder decrease
7 considerably by using this method. Such reduction in the unsteady forces, results in smaller
8 flow-induced vibrations from the body. Korkischko and Meneghini (2012) suppressed
9 experimentally the vortex-induced vibration using moving surface boundary-layer control for
10 the Reynolds number ranges from 1.6×10^3 to 7.5×10^3 . They used two small rotating
11 cylinders, placed in the boundary layer of the main cylinder for delaying the separation of the
12 boundary layer. They observed that the maximum amplitude of oscillation for the cylinder
13 decreases significantly by using this method. Reddy et al. (2013) analytically controlled the
14 flow past a circular cylinder by use of two counter-rotating control cylinders. They identified
15 the control cylinder position by using a Foppl vortex model and found that the vortex
16 formation behind the cylinder can be suppressed by this method.

17 After review of the active type of this method, the passive ones are reviewed in the
18 following:

19 This method is classified as passive and wake control method. External element with
20 different shapes can be used to control the vortex shedding behind an obstacle, e.g., Kwon
21 and Choi (1996); Huera-Huarte (2014) used Splitter plates, Oruc (2012) used Control
22 elements with a streamlined shape, Lee and Kim (1997)) used helical wires, and Strykowski
23 and Sreenivasan (1990); Wu et al. (2012); Zhu and Yao (2015) applied Control
24 cylinders/rods. When an external element is placed in the vicinity of an obstacle, the vortex
25 shedding characteristics, the mean structure and dynamic behaviour of the wake and transport

1 phenomena in the wake region change. Gap width between the element and the main
2 obstacle, the shapes of main obstacle and control element and the placement of element are
3 different parameters that influence the flow. The element can be placed at upstream or
4 downstream of the bluff body in this method. This method is widely used in deepwater risers
5 and free-spanning submarine pipelines, flourishing extraction of ocean oil and gas resources,
6 see e.g. Wu et al. (2012). Figure 15 shows control rods that are used to suppress the vortex-
7 induced vibration of long flexible riser.
8
9
10
11
12
13
14
15



16
17
18
19
20
21
22
23
24
25
26
27
28
29
30
31
32
33
34 **Fig. 15.** Control rods that are used to suppress the vortex-induced vibration of long flexible
35 riser (Figure reprinted from Wu et al. (2012) with permission from the publisher)
36

37 Ozkan and Akilli (2014) experimentally controlled the flow around circular obstacles by
38 use of attached permeable plates on the obstacle wall. The plate was tilted at five different
39 angles ($\theta=0^\circ, 15^\circ, 30^\circ, 45^\circ$ and 60°), defined as the angle between the centerline of the
40 cylinder and the plate. It was found that the maximum Reynolds shear stress (depends on the
41 mean velocity fluctuation on two axes in turbulent flow) and turbulent kinetic energy
42 decreased by about 72% and 66%, respectively, for the plate angles of $\theta=45^\circ$ and 60° . Control
43 of vortex shedding behind a circular obstacle for flows at low Reynolds numbers is
44 performed numerically by Mittal and Raghuvanshi (2001). They used a stabilized finite
45 element method to solve the governing equations (Navier-Stokes equations). They performed
46
47
48
49
50
51
52
53
54
55
56
57
58
59
60
61
62
63
64
65

1 this work by using a second, much smaller, control cylinder in the near wake of the main
2 obstacle. The computed results shown that the control obstacle provides a local favourable
3 pressure gradient in the wake region and stabilizes the shear layer. Also, it was found that a
4 proper placement of the control obstacle leads to a complete suppression of the vortex
5 shedding behind the main obstacle. Maiti and Bhatt (2014) used a numerical method and
6 suppressed the vortex shedding behind a square obstacle by utilizing a series of upstream
7 rectangular cylinders. Their results showed that the aerodynamic forces acting downstream,
8 decreases with an increase in the height of the upstream obstacle for a specific gap height
9 (distance between the cylinder and wall). Bao and Tao (2013) utilized finite element
10 formulations and controlled the wake flow behind a circular obstacle due to a laminar flow by
11 use of parallel dual plates, attached at the trailing side of the obstacle. The stabilization of the
12 free shear layer fluctuation and the basal cavity effect were two control mechanisms in this
13 study. The maximum drag reduction was about 13% for $Re=1.6 \times 10^2$. Figure 16 shows the
14 configuration of streamlines (upper) and pressure contours (lower) that are presented by Bao
15 and Tao (2013) for (a) attached dual plates control, (b) detached dual plates control, and (c)
16 detached dual cylinders control. Shown in this figure, such mechanism is not applicable for the
17 detached control devices because the reverse flow can cross the gap between the cylinder and device
18 without any disturbance. This flow is driven by the adverse pressure gradient, see figures 16(b) and
19 (c). Moreover, the modifications of the shear layer by these devices are presented in figure 17.
20 It should be noted that the solid and dash-dot lines in this figure refer to two time instants at
21 which the lift force achieves its maximum and minimum, respectively. In this figure, A/D is
22 the dimensionless fluctuation amplitude. It can be seen that the separated shear layer for the
23 plain cylinder has the maximum fluctuation with the largest value of A/D , immediately
24 followed by the value for the single splitter plate. This shows that the traditional splitter plate
25 has a negligible effect on the stabilization of the fluctuation. As a result, the attached dual
26

plates control is the most effective approach to suppress the shear layer fluctuations with the minimum amplitude. Moreover, the detached flat plate and circular cylinder devices have considerable stabilizing influences with moderate fluctuation amplitude.

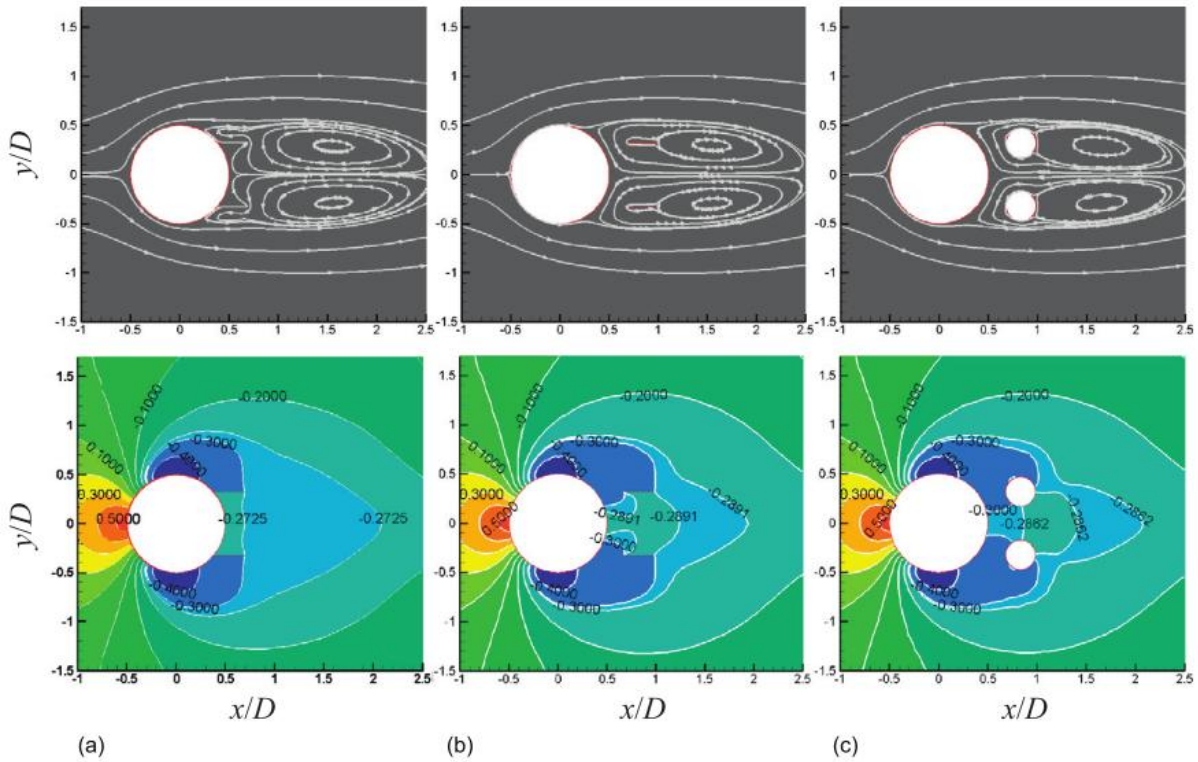


Fig. 16. Configuration of streamlines (upper) and pressure contours (lower) for (a) Attached dual plates control; (b) Detached dual plates control; (c) Detached dual cylinders control at $Re=1.0 \times 10^2$ (Flow is from left to right; Figure reprinted from Bao and Tao (2013) with permission from the publisher)

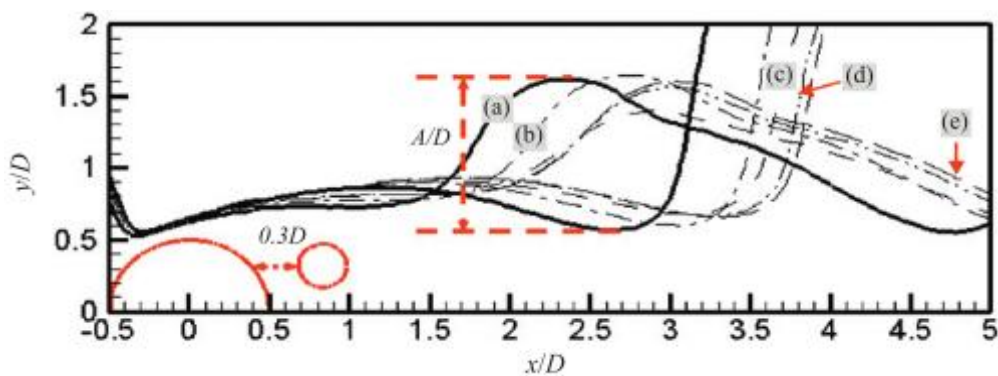


Fig. 17. The isocontour of streamwise velocity $U=U_{\infty}$ for cylinders with and without control devices at $Re=100$: (a) plain cylinder; (b) splitter plate control; (c) attached dual plates

control; (d) detached dual plates control; and (e) dual cylinders control (Flow is from left to right; Figure reprinted from Bao and Tao (2013) with permission from the publisher)

Akilli et al. (2005) experimentally suppressed the vortex shedding of circular obstacle in shallow water by a splitter plate at $Re=5.0\times 10^3$. Their results showed that the splitter plate has a considerable effect on the suppression of the vortex shedding for the gap ratio between 0 and 1.75D (where D is the diameter of the cylinder). The gap is defined between the base of the obstacle and the leading edge of the splitter plate. It was found that among different shapes of elements, the splitter plate was one of the most successful external devices to control the vortex shedding behind an obstacle. Chen and Shao (2013) suppressed the vortex shedding from a rectangular cylinder at low Reynolds numbers by using four kinds of elements: circular, regular triangular, square and rectangular. They performed a comparison on the application of different objects on the suppression of vortex shedding from a main cylinder at $Re=1.1\times 10^2$. The mechanism of the suppression is considered from the viewpoints of stress distribution and velocity profile stability. Their study indicated that the triangular element has the smallest effective zone, whereas the square element presents the largest. The effective zone is defined as a certain location of element which vortex shedding can be suppressed and the drag and lift fluctuations can be greatly decreased.

Miau et al. (1993) experimentally suppressed low-frequency variations in vortex shedding behind a trapezoidal obstacle by a splitter plate for the Reynolds numbers in a range of 5.0×10^3 to 4.5×10^4 . They found that the degree of 2-D vortex shedding is improved with suppression of low-frequency variations. Kuo and Chen (2009) numerically controlled the wake flow by two small control cylinders at $Re=80$. The computed results show that the symmetric standing eddies at downstream of the main obstacle and the delay of the vortex shedding lead to a 70–80% deduction of the fluctuating lift on the main obstacle. Vilaplana et al. (2013) experimentally controlled the turbulent wake behind a sphere by a small sphere at

1
2
3
4
5
6
7
8
9
10
11
12
13
14
15
16
17
18
19
20
21
22
23
24
25
26
27
28
29
30
31
32
33
34
35
36
37
38
39
40
41
42
43
44
45
46
47
48
49
50
51
52
53
54
55
56
57
58
59
60
61
62
63
64
65

$Re=3.3 \times 10^4$. They reported that the vortex loops shed from only one side of the sphere for the reference case (without the control sphere) but the vortex loops shed closer to the symmetry axis for the control sphere placed at the center of the wake. Wu et al. (2012) experimentally suppressed the vortex-induced vibration of long flexible riser by multiple control rods for the Reynolds numbers ranging from $Re=2.4 \times 10^3$ to $Re=7.6 \times 10^3$. They found that this control method performs well in mitigating the vortex-induced vibration (VIV). Also, the smaller spacing ratio and the larger coverage rates lead to better VIV suppression.

17
18
19
20
21
22
23
24
25
26
27
28
29
30
31
32
33
34
35
36
37
38
39
40
41
42
43
44
45
46
47
48
49
50
51
52
53
54
55
56
57
58
59
60
61
62
63
64
65

Gozmen et al. (2013) experimentally controlled the vortex shedding behind a circular obstacle in shallow water using a splitter plate located in the downstream region at a Reynolds number of 6.25×10^3 . They found that the turbulent quantities and mean flow, change significantly with length and height ratios of the splitter plates in shallow flow. Also, it is shown that L/D ration of 2, where D is cylinder diameter and L is splitter plate length, provides optimum control. Weickgenannt and Monkewitz (2000) controlled experimentally the vortex shedding in an axisymmetric bluff body wake by using a control disc mounted at the rear of an axisymmetric blunt-based body of revolution for the Reynolds number range $3.0 \times 10^3 < Re < 5.0 \times 10^4$. El-Gammal et al. (2007) controlled the vortex shedding in a sectional bridge model by a spanwise sinusoidal perturbation method (SPPM). The same amplitude with two different SPPM configurations and two different wavelengths were used in this study. Their results showed that the SPPM with the higher wave steepness distinctively suppresses the vortex-induced vibrations (VIV). Ozono (2003) experimentally controlled the flow around a circular obstacle by using a few interference elements shifted along the wake. These elements were short and long splitter-plates and a circular cylinder. They found that the spectral peak associated with the vortex shedding weakens by using splitter-plates. Also, a base suction was created by shifting the downstream cylinder to upstream direction in two-cylinder case. Shi et al. (2010) experimentally investigated the effects of wall proximity on

1 the characteristics of the wake downstream a 2-D square obstacle for a turbulent regime.
2 They found that the time-averaged streamwise velocity in the wake region modifies
3 significantly with the reduction in the gap width between the cylinder and the wall.
4
5

6
7 Finally, this method could be classified as both active and passive methods. Moreover,
8 this method is classified in the group of wake control methods. This method is important
9 from two viewpoints. First, the existence of applying an external element modifies the
10 velocity profiles in the wake region of the bluff body and changes their stability nature.
11 Second, applying external element alters the pressure and shear stress distributions in the
12 wake region. This method may create undesirable forces due to the use of external body.
13 Moreover, the performance of this method is very sensitive to the element arrangements such
14 as the distance between the element and main bluff body and the placement of element with
15 respect to the incoming flow.
16
17
18
19
20
21
22
23
24
25
26
27
28

29 Table 10 summarizes the researches on application of external element to control the
30 vortex shedding and flow separation of a bluff body. Vortex induced vibration is a basic
31 reason for the fatigue damage of deepwater risers and free spanning submarine pipelines.
32 This phenomenon has received more attention during recent years due to the increase in
33 exploration and extraction of offshore oil and gas resources. Utilizing a control rod is shown
34 to be very good method to suppress the vortex induced vibration of deepwater drilling riser
35 with a large aspect ratio, see e.g. Wu et al. (2012). In practical ocean engineering cases,
36 multiple auxiliary pipelines are often available. Most of the previous studies in this method
37 (Control rod) have used a single control rod, and studies on the use of multiple control rods
38 cases are very limited. Finally, it should be noted that it is not practical to apply external
39 elements as an integral part of the main configuration.
40
41
42
43
44
45
46
47
48
49
50
51
52
53
54
55

56 **Table 10:** Researches on application of external element to control the vortex shedding and
57 flow separation of a bluff body
58
59
60
61
62
63
64
65

Authors	Type of research	Type of fluid	Reynolds number	Spatial dimensions	Classification	Controlled device	Controller
Strykowski and Sreenivasan (1990)	Experimental/ Numerical	Air/water	$30-1.2 \times 10^2$	2D	Passive	Circular cylinder	Small non-rotating circular cylinder
Nishiyama et al. (1991)	Experimental	Air	$4.0 \times 10^3-8.0 \times 10^3$	----	Active	Elliptic cylinder	Elliptic vibrating cylinder
Miau et al. (1993)	Experimental	Air	$5.3 \times 10^3-4.5 \times 10^4$	----	Passive	Trapezoidal cylinder	Splitter plate
Kwon and Choi (1996)	Numerical	Unknown	$80-1.6 \times 10^2$	2D	Passive	Circular cylinder	Splitter plate
Lee and Kim (1997)	Experimental	Air	$5.0 \times 10^3-5.0 \times 10^4$	----	Passive	Two circular cylinder	Three helically wrapped small rubber wires
Weickgenant and Monkewitz (2000)	Experimental	Air	$3.0 \times 10^3-5.0 \times 10^4$	----	Passive	Axisymmetric blunt-based body	Disc
Mittal (2001)	Numerical	Unknown	1.0×10^2 and 1.0×10^4	2D	Active	Circular cylinder	Rotating cylinder
Mittal and Raghuvansh (2001)	Numerical	Unknown	$60-1.0 \times 10^2$	2D	Passive	Circular cylinder	Small non-rotating circular cylinder
Patnaik and Wei (2002)	Numerical	Unknown	$2.0 \times 10^2-4.0 \times 10^2$	2D	Active	D shape cylinder	Two small rotating circular cylinders
Ozono (2003)	Experimental	Air	$6.7 \times 10^3-2.5 \times 10^4$	----	Passive	Circular cylinder	Short or long splitter-plate or circular cylinder
Akilli et al. (2005)	Experimental	Water	5.0×10^3	----	Passive	Circular cylinder	Splitter plate
Kuo and Chen (2009)	Numerical	Unknown	80	2D	Passive	Circular cylinder	Two fixed circular cylinders
Muddada and Patnaik (2010)	Numerical	Unknown	$1.0 \times 10^2-3.0 \times 10^2$	2D	Active	Circular cylinder	Two rotating cylinder+feedback
Shi et al. (2010)	Experimental	Air	1.32×10^4	----	Passive	Square cylinder	Plane wall
Korkischko and Meneghini (2012)	Experimental	Water	3×10^3	----	Active	Circular cylinder	Two small rotating circular cylinders
Wu et al. (2012)	Experimental	Water	$2.4 \times 10^2-7.6 \times 10^2$	----	Passive	Slender riser	Multiple rods
Oruc (2012)	Experimental	Water	5.2×10^3	----	Passive	Circular cylinder	Screen
Bao and Tao (2013)	Numerical	Air	$20-1.6 \times 10^2$	2D	Passive	Circular cylinder	Parallel dual plates
Chen and Shao (2013)	Numerical+ Experimental	Water	$75-1.3 \times 10^2$	2D	Passive	Rectangular cylinder	Small elements of circular, square, triangular and thin-strip cross-sections
Gozmen et	Experimental	Water	6.25×10^3	----	Passive	Circular	Splitter plate

al. (2013)						cylinder	
1 Vilaplana et al. (2013)	Experimental	Air	3.3×10^4	3D	Passive	Sphere	Small sphere
3 Reddy et al. (2013)	Analytically	Unknown	100	2D	Active	Circular cylinder	Two counter-rotating cylinders
6 Huera-Huarte (2014)	Experimental	Water	1.0×10^3 - 8.0×10^3	----	Passive	Circular cylinder	Splitter plate
9 Maiti and Bhatt (2014)	Numerical	Unknown	1.25×10^2 - 2×10^2	2D	Passive	Square cylinder	Rectangular cylinders and plane wall
12 Ozkan and Akilli (2014)	Experimental	Water	5.0×10^3	----	Passive	Circular cylinder	Attached permeable plates
15 Zhu and Yao (2015)	Numerical	Water	1.16×10^3 - 6.4×10^3	2D	Passive	Circular cylinder	Rods

3. Energetic efficiency of control methods

The cost, energetic efficiency and optimization analyses of each active control method are significant parameters that are considered to evaluate and compare the various techniques. A key factor of an optimal flow control method is the minimization of input energy required for this method. Such analyses help ocean engineers to select the most applicable and efficient method. In this section, recent studies in this field are reviewed. Choi et al. (2008) defined the control efficiency as the ratio of the save power to the control input power for the case of drag reduction. Recently a new parameter, namely Power Loss Coefficient, has been proposed to quantify the energetic efficiency. Power loss coefficient is employed for optimal PID control and it obtains the amount of the energy lost in the wake of the translating body. It is applicable to different flow configurations including active drag reduction, self-propulsion and thrust generation, see Arakeri and Shukla, (2013). Shukla and Arakeri (2013) determined the proper tangential velocity profiles which would result in minimum values of the drag force acting on the cylindrical body, and subsequently, minimum value of the net power consumption. They concluded that the flow tangential velocity enables energetically efficient propulsion. Das et al. (2016) proposed an energetically efficient active flow control strategy (PID) to control the wake vortices behind a circular cylinder. They used a linear quadratic

optimal control formulation to minimise the cost functional. Their results showed that this system was energetically efficient, even when the twin eddies are still persisting behind the obstacle. Jukes and Choi (2009) optimally modified the vortex shedding cycle behind a circular cylinder by using a short plasma excitation. They expressed the effectiveness of plasma in reducing drag in terms of the energy efficiency. They concluded that a power saving ratio over 1000 can be achieved, while the energy efficiency was 51%. Note that the power saving ratio was defined as a ratio of the power saved by drag reduction to the fluidic power introduced by the plasma.

Table 11 summarizes researches performed on different algorithms to achieve an optimal control approach. Reynolds number, type of optimal target, and quantity of optimized parameters for each research are included in this table. As shown in this table, the optimization analyses are very limited for higher Reynolds numbers ($Re > 1.0 \times 10^5$) and three-dimensional flows that are more applicable in ocean engineering. However, most attentions are focused on rotary oscillation control method and other methods, especially MHD and thermal methods, are needed to perform such analysis. Finally, an economic analysis is strongly advised, especially for experimental works to obtain the risks and gains of each technique. Such analysis has not been performed in previous works.

Table 11: Recent researches on different algorithms to achieve an optimal control approach

Authors	Reynolds number	Optimized parameter	Quantity of optimized parameters	Spatial dimensions
Rotary				
He et al. (2000)	2.0×10^2 - 6.0×10^2	Drag	60% reduction in drag	2D
Homescu et al. (2002)	$60-1.0 \times 10^3$	Drag	Unknown	2D
Protas and Styczek (2002)	75 and 1.5×10^2	Power drag+ Power control	93% reduction in drag	2D
Bergmann et al. (2005)	2.0×10^2	Drag	25% reduction in drag	2D
Sengupta et al. (2007)	1.5×10^4	Drag	66% reduction in drag	2D
Shukla and Arakeri (2012)	1-300	Drag	77% reduction in drag	2D
Arakeri and Shukla (2013)	100	Drag	79% reduction in drag	2D
Flinois and Colonius (2015)	$75-2.0 \times 10^2$	Power drag+ Power control	19% reduction in drag	2D
Feedback control				
Min and Choi (1999)	1.0×10^2 and	Drag/Lift	28% reduction in drag	2D

	1.6×10^2				
1	Leclerc et al. (2006)	1.0×10^2	Drag/Lift	12% reduction in drag	2D
2	Joe et al. (2009)	3.0×10^2	Lift	85% augmentation in lift	2D
3	Das et al. (2016)	100	Power loss coefficient	Unknown	2D
4	Blowing/suction				
5	Li et al. (2003)	1.1×10^2	Objective functional (a function of the initial condition, the model parameters, and the boundary parameter)	47% reduction in objective functional	2D
6	EHD				
7	Jukes and Choi (2009)	15000	Drag/Lift	8% reduction in drag/50% augmentation in lift	3D
8	D'Adamo et al. (2011)	2.0×10^2	Energy consumption of EHD	Not mentioned	2D

4. Drag analysis

An important factor, which should be considered for assessing different control techniques, is the effectiveness of each technique on the drag reduction. Table 12 presents a compilation of the available data on the impact of these techniques on the drag reduction. Note that the maximum drag reductions are presented in this table for each relevant method. As shown in this table, some methods, such as suction and blowing, external element (especially splitter plate), rotationally oscillating, and surface roughness are more active for drag reduction. Some other techniques need more attentions. For example, none of researches about thermal method, except the study of Chatterjee and Sinha (2014) have focused on the effects of heating on drag and lift forces. Note that some of these methods have both positive and negative impacts on the drag reduction. For example, Chatterjee and Sinha (2014) stated that heating can leads to drag reduction for lower heating flux, while it shows some opposite behavior for higher heating flux. There is a similar trend for MHD method. Singha and Sinhamahapatra (2008) showed that the mean drag coefficient decreases slowly for lower magnetic field strength and increases rapidly for higher magnetic field strength. It is worth mentioning that some researchers used two very simple passive techniques to control the vortex shedding and especially reduce the drag coefficient. First technique applies small tabs

on the lower and upper trailing edges of the bluff bodies (Park et al. 2006). This leads to perturb the wake, subsequently attenuate the vortex shedding, and decrease in drag coefficient. Park et al. (2006) examined both experimentally and numerically this technique successfully for the Reynolds number in the range of 2.0×10^4 to 8.0×10^4 . As second technique, some researchers used wavy walls for bluff bodies to suppress the vortex shedding and decrease the drag coefficient. For example, Darekar and Sherwin (2001) used in a numerical study the wavy stagnation face for flow around a square cylinder at low Reynolds numbers in the range of 10 to 1.5×10^2 . They found that this technique leads to suppression of the vortex shedding by the spanwise waviness and subsequently a significant reduction in drag coefficient. In another research, Xu et al. (2014) applied traveling wave wall for a circular cylinder to reduce the drag coefficient and suppress the vortex-induced vibration. They performed a numerical study for a fixed low Reynolds number (i.e. $Re = 2.0 \times 10^2$). They presented that this control technique can lead to reduction in the fluctuation of the lift and drag coefficients. As a result, these low cost and simple techniques can be used to reduce the drag coefficient and control of flow around the bluff bodies. As a result, the drag coefficient significantly decreases (up to 90%) by using some methods such as rotationally oscillating or steady blowing with two actuators on a cylinder.

Table 12: A compilation on available data about the impact of the control methods on the drag reduction

Author	Relative drag reduction	Range of Reynolds number	Method
Thomas et al. (2005)	90%	$1.28 \times 10^4 - 2.54 \times 10^4$	Steady blowing with two actuators on a cylinder
Amitay et al. (1997)	30%	7.55×10^4	Synthetic jets
Sohankar et al. (2015)	72%	$70 - 1.5 \times 10^2$	Uniform suction and blowing
Delaunay and Kaiktsis (2001)	14%	< 90	Steady base suction and blowing
Joseph et al. (2013)	10%	$1.1 \times 10^6 - 2.8 \times 10^6$	Micro-jets
Mimeau et al. (2014)	30%	$5.5 \times 10^2 - 3.0 \times 10^3$	Porous coatings
Muddada and Patnaik (2010)	53%	$1.0 \times 10^2 - 3.0 \times 10^2$	Two rotating cylinder+feedback

Lee et al. (2004)	29%	2.0×10^4	Small control rod installed upstream
Kuo and Chen (2009)	5%	80	Two fixed circular cylinders
Hwang et al. (2003)	23%	$30-1.6 \times 10^2$	Detached splitter plate
Gu et al. (2012)	30%	$3.0 \times 10^4-6.0 \times 10^4$	Rotatable splitter plates
Bao and Tao (2013)	23%	$20-1.6 \times 10^2$	Parallel dual plates
Korkischko and Meneghini (2012)	60%	3.0×10^3	Two small rotating circular cylinders
Bearman and Owen (1998)	30%	4.0×10^4	Spanwise sinusoidal forms and with sinusoidal front faces
He et al. (2000)	60%	$2.0 \times 10^2-6.0 \times 10^2$	Rotationally oscillating
Tokumar and Dimotakis (1989)	80%	1.5×10^4	Rotationally oscillating
S.J. Lee and J.Y. Lee (2006)	32.82%	4.14×10^3	Rotationally oscillating
Bergmann et al. (2005)	25%	2.0×10^2	Rotationally oscillating
Sengupta et al. (2007)	66%	1.5×10^4	Rotationally oscillating
Kim and Choi (2005)	23%	$40-3.9 \times 10^3$	Blowing and suction from the slots placed at upper and lower surfaces of the obstacle
Huang et al. (2007)	25%	$3 \times 10^4-4 \times 10^5$	Surface roughness
Zhou et al. (2015)	30%	$6.0 \times 10^3-8 \times 10^4$	Surface roughness
Singha and Sinhamahapatra (2011)	4%	$50-2.5 \times 10^2$	MHD
Chen and Aubry (2005)	46%	2.0×10^2	MHD
Min and Choi (1999)	28%	1.0×10^2 and 1.6×10^2	Feedback
Leclerc et al. (2006)	12%	1.0×10^2	Feedback
He et al. (2000)	60%	$2.0 \times 10^2-6.0 \times 10^2$	Rotationally oscillating
Protas and Styczek (2002)	93%	75 and 1.5×10^2	Rotationally oscillating
Bergmann et al. (2005)	25%	2.0×10^2	Rotationally oscillating
Sengupta et al. (2007)	66%	1.5×10^4	Rotationally oscillating
Flinois and Colonius (2015)	19%	$75-2.0 \times 10^2$	Rotationally oscillating

5. Summary and concluding remarks

This paper focused on reviews of the vortex shedding suppression and wake control. Some of the applications of these methods are summarized in table 13.

Table 13: Applications of existing approaches for vortex shedding suppression and wake control.

Name of method	Applications
EHD	This method is suitable for aerodynamic applications such as airfoils typically used in wind turbine blades, civil air traffic projects and aircraft.
External element	This method is used in ocean structures, bridges, high voltage lines, deepwater risers and free spanning submarine pipelines, flourishing

	extraction of ocean oil and gas resources.
Feedback control	This method is used in aircraft and projectile aerodynamics include dynamic stall control, marine structures, chemical mixing improvement, submarine periscopes, increase mixing and heat transfer in combustion, and civil and engineering applications.
MHD	This method can be used to control the flow over ship hull, tubular heat exchangers, pipelines, suspension wires, oblique shocks and suspension bridges.
Rotary oscillations	This method can be used to control the flow for tube heat exchangers, shafts, drilling of oil wells, nuclear reactor fuel rods and steel suspension bridge cables.
Secondary flow	These techniques are used in long-span suspension bridges and cable-stayed bridges.
Surface roughness	This method is utilized in offshore and marine application, such as platform pillars, pipelines and risers.
Thermal effects	As a natural mean to control the boundary layer separation over a cylinder, it can be used in many engineering applications.

Several existing approaches and techniques used to control the wake destructive behavior and suppression of vortex shedding behind bluff bodies are discussed and reviewed. These techniques are classified into nine categories based on the fundamental methods and applications utilized. The following conclusions and suggestions can be drawn:

- We presented the ranges of Reynolds number that were used by researchers for each category of control technique. However, some techniques work better for special flow regimes and some are suitable for wider flow regimes. As an example, surface roughness approach works well at subcritical Reynolds numbers ($Re=1\times 10^3$ to 1×10^4) but does not work at lower values of Reynolds number, see e.g. Achenbach 1971. For low values of Reynolds number that have applications in microsystems, the following methods have been tested successfully: Splitter plates, second (control) cylinder, thermal method, secondary flow (base bleed and suction), permeable wall, feedback control law based on a single-sensor measurement, cylinder rotation or transverse oscillations, and apply of steady magnetic fields.

- Most of the researches in the literature and reviewed here, are performed for a single bluff body, while in practice, vertical cylinder and structures exist in groups. Oil and gas transmission lines in the oceans and other marine pipelines, vertical columns of offshore platforms, and heat exchanger tube bundles are a few examples of this application.
- Some type of oscillation, such as the flow perturbations and imposed sound field, have not received a full attention and hence have good potential for future researches. These types of control methods are more applicable in ocean engineering.
- The future attentions should be more focused on modifying the near wake flow and vortex shedding to use them as the advantage instead of effort for removing them completely.
- There are some technologies such as micro-electro-mechanical systems (MEMS) to miniaturize the synthetic jet. Such technologies lead to save the power needed for controlling the flow.
- A very good method to suppress the vortex induced vibration of deepwater drilling riser with a large aspect ratio is the use of a control rod. In practical ocean engineering cases, multiple auxiliary pipelines are often available. Most of the previous studies in this field (Control rod) have used a single control rod, and studies using multiple control rods cases are very limited.
- A good method for controlling the wake turbulence behind an obstacle could be angular momentum injection as tested successfully by researchers. They showed that the wake turbulence could be completely suppressed. This method has a good potential for future researches as it works well for turbulent flows.
- Distributed force in spanwise direction to flow over a bluff body modifies the vertical evolution in the wake and leads to drag reduction significantly for laminar and

1
2
3
4
5
6
7
8
9
10
11
12
13
14
15
16
17
18
19
20
21
22
23
24
25
26
27
28
29
30
31
32
33
34
35
36
37
38
39
40
41
42
43
44
45
46
47
48
49
50
51
52
53
54
55
56
57
58
59
60
61
62
63
64
65

turbulent flows. This force can produce by blowing, suction, perturbations, and phase excitation by surface heaters.

- In summary, the target of control methods in boundary layer classification are to prevent or provoke boundary layer separation, control the spanwise and streamwise vortices in a turbulent boundary layer, transition delay, and decrease the skin-friction drag. Subsequently, the purposes of wake control methods are to suppress the pressure drag and control the wake structures, and the pressure distribution on the rear portion of cylinders.
- With regards to VIV control, the following parameters are to be considered before selecting the control method:
 - ✓ To suppress the oscillations, higher quantum of control is required for objects in water than in air, due to the damping effects.
- The energetic efficiency and optimization analyses for each active control method must be considered to evaluate and compare the various techniques. Such analyses help ocean engineers to select the best method. Although, there are some researches about this topic, but most of the studies have focused on rotary oscillation control methods; other methods, especially MHD and thermal methods, are required such optimization analysis.
- An economic analysis is strongly advised especially for experimental works to obtain the risks and gains of each technique. Such analysis has not been performed in previous works.
- The transition of flow pass a bluff body from 2D to 3D is an important phenomenon to understand in an engineering context because the wake-induced forces can have potentially detrimental structural or local effects on the bodies or their surroundings.

1 The effects of aforesaid method on this transition have a good potential for future
2 studies.
3

- 4 • Applying zero-mass-flux actuator types, which need little energy, is a remarkable
5 method for further investigations in the field of flow control around bluff bodies.
6
- 7 • New optical technologies in flow visualization especially in the wake region include
8 the laser-induced fluorescence and particle-image-velocimetry, are advised to be used
9 as they are tested successfully by some researches, Lee and Lee (2008).
10
- 11 • There are two very simple passive techniques to control the vortex shedding and
12 especially reduce the drag coefficient. Small tabs are applied on the lower and upper
13 trailing edges of the bluff bodies as first technique. For second technique, some
14 researchers used wavy walls for bluff bodies to suppress the vortex shedding and
15 decrease the drag coefficient.
16
- 17 • The researches in this field should be extended to a pragmatic approach. Such
18 approach covers other disciplines including the noise, structures, and systems
19 integration.
20
- 21 • Developments in smart sensors and actuators with their integration into active flow
22 control methods are interesting solutions for minimizing the objective functional.
23
- 24 • Microelectro mechanical systems can be used to reduce the drag coefficient and
25 control the flow separation.
26
- 27 • Piezo-electric can be used as a new actuation for controlling the flow.
28
- 29 • Development of low-order models, such as Galerkin model, can be useful for the
30 suppression of vortex shedding.
31
32
33
34
35
36
37
38
39
40
41
42
43
44
45
46
47
48
49
50
51
52

53 **References**

54
55
56
57
58
59
60
61
62
63
64
65

- 1
2
3
4
5
6
7
8
9
10
11
12
13
14
15
16
17
18
19
20
21
22
23
24
25
26
27
28
29
30
31
32
33
34
35
36
37
38
39
40
41
42
43
44
45
46
47
48
49
50
51
52
53
54
55
56
57
58
59
60
61
62
63
64
65
- [1] H. Akilli, B. Sahin, N. Filiz Tumen, “Suppression of vortex shedding of circular cylinder in shallow water by a splitter plate.” *Flow Measurement and Instrumentation*, 16 (2005) 211–219.
- [2] E. Achenbach, “Influence of surface roughness on the cross-flow around a circular cylinder.” *Journal of Fluid Mechanics*, 46 (2) (1971) 321–335.
- [3] E. Achenbach, A. Heinecke, “On vortex shedding from smooth and rough cylinders in the range of Reynolds numbers 6×10^3 to 5×10^6 .” *Journal of Fluid Mechanics*, 109 (1981) 239–251.
- [4] D.W. Allen, D.L. Henning, “Surface roughness effects on vortex-induced vibration of cylindrical structures at critical and supercritical Reynolds numbers.” 2001, In: *Proceedings of the Offshore Technology Conference*, Houston, Texas, USA OTC, p. 13302.
- [5] M. Amitay, A. Honohan, M. Trautman, A. Glezer, “Modification of the aerodynamic characteristics of bluff bodies using fluidic actuators.” 1997, 28th Fluid Dynamics Conference, Fluid Dynamics and Co-located Conferences.
- [6] J.H. Arakeri, R.K. Shukla, “A unified view of energetic efficiency in active drag reduction, thrust generation and self-propulsion through a loss coefficient with some applications.” *Journal of Fluids and Structures*, 41 (2013) 22–32.
- [7] B. J. Armstrong, F. H. Barnes, I. Grant, “The effect of a perturbation on the flow over a cylinder.” *Physics of Fluids*, 29 (1986) 2095-2102.
- [8] G. Artana, R. Sosa, E. Moreau, G. Touchard, “Control of the near-wake flow around a circular cylinder with electrohydrodynamic actuators.” *Experiments in Fluids*, 35 (2003) 580–588.
- [9] Y. Bao, J. Tao, “The passive control of wake flow behind a circular cylinder by parallel dual plates.” *Journal of Fluids and Structures*, 37 (2013) 201–219.

- 1
2
3
4
5
6
7
8
9
10
11
12
13
14
15
16
17
18
19
20
21
22
23
24
25
26
27
28
29
30
31
32
33
34
35
36
37
38
39
40
41
42
43
44
45
46
47
48
49
50
51
52
53
54
55
56
57
58
59
60
61
62
63
64
65
- [10] S.J. Baek and H. J. Sung, “Numerical simulation of the flow behind a rotary oscillating circular cylinder.” *Physics of Fluids*, 10 (4) (1998) 869–876.
- [11] A. Baz, J. Ro, “Active control of flow-induced vibrations of a flexible cylinder using direct velocity feedback.” *Journal of Sound and Vibration*, 146 (1) (1991) 33-45.
- [12] P.W. Bearman, J.K. Harvey, “Control of circular cylinder flow by the use of dimples.” *AIAA Journal*, 31 (10) (1993) 1753–1756.
- [13] P.W. Bearman, J.C. Owen, “Reduction of Bluff-Body Drag and Suppression of Vortex Shedding by the Introduction of Wavy Separation Lines.” *Journal of Fluids and Structures*, 12 (1998) 123–130.
- [14] J.F. Beaudoin, O. Cadot, J.L. Aider, J.E. Wesfreid, “Bluff-body drag reduction by extremum-seeking control.” *Journal of Fluids and Structures*, 22 (2006) 973–978.
- [15] R. Becker, R. King, “Adaptive closed-loop separation control on a high-lift configuration using extremum seeking.” 3rd AIAA Flow Control Conference 5-8 June 2006, San Francisco, California.
- [16] E. Berger, “Suppression of vortex shedding and turbulence behind oscillating cylinders.” *Physics of Fluids*, 10 (1967) S191–S193.
- [17] M. Bergmann, L. Cordier, J.P. Brancher, “Optimal rotary control of the cylinder wake using proper orthogonal decomposition reduced-order model.” *Physics of Fluids*, 17 (9) (2005) 097101.
- [18] M.M. Bernitsas, K. Raghavan, “Reduction/suppression of VIV of circular cylinders through roughness distribution at $8 \times 10^3 < \text{Re} < 2.0 \times 10^5$.” In: *Proceedings of the Offshore Mechanics and Arctic Engineering Conference, Estoril, Portugal, OMAE, 2008*, p. 58024.
- [19] M.M. Bernitsas, K. Raghavan, G. Duchene, “Induced separation and vorticity using roughness in VIV of circular cylinders at $8 \times 10^3 < \text{Re} < 2.0 \times 10^5$.” In: *Proceedings of the*

1 Offshore Mechanics and Arctic Engineering Conference, Estoril, Portugal, OMAE, 2008, p.
2 58023.
3

4 [20] A. M. Bimbato, L. A. Alcântara Pereira, M. Hiroo Hirata “Suppression of vortex
5 shedding on a bluff body.” *Journal of Wind Engineering and Industrial Aerodynamics*, 121
6 (2013) 16–28.
7

8 [21] R. D. Blevins, “The effect of sound on vortex shedding from cylinder.” *Physics of*
9 *Fluids*, 161 (1985) 217-237.
10

11 [22] M. Bovand, S. Rashidi, J.A. Esfahani, “Enhancement of heat transfer by nanofluids and
12 orientations of the equilateral triangular obstacle.” *Energy Conversion and Management*, 97
13 (2015a) 212-223.
14

15 [23] M. Bovand, S. Rashidi, J.A. Esfahani, “Control of wake destructive behavior for
16 different bluff bodies in channel flow by Magnetohydrodynamics.” *Journal of Magnetism and*
17 *Magnetic Materials*, under review (2015b).
18

19 [24] M. Bovand, S. Rashidi, M. Dehghan, J.A. Esfahani, M.S. Valipour, “Control of wake
20 and vortex shedding behind a porous circular obstacle by exerting an external magnetic
21 field.” *Journal of Magnetism and Magnetic Materials*, 385 (2015c) 198–206.
22

23 [25] M. Carini, J.O. Pralits, P. Luchini, “Feedback control of vortex shedding using a full-
24 order optimal compensator.” *Journal of Fluids and Structures*, 53 (2015) 15–25.
25

26 [26] S. Chaligné, T. Castelain, M. Michard, D. Chacaton, D. Juvé, “Fluidic control of wake-
27 flow behind a two-dimensional square back bluff body.” *Comptes Rendus Mécanique*, 342
28 (2014) 349–355.
29

30 [27] A. S. Chan, P. A. Dewey, A. Jameson, C. Liang, A. J. Smits, “Vortex suppression and
31 drag reduction in the wake of counter-rotating cylinders.” *Journal of Fluid Mechanics*, 679
32 (2011) 343- 382.
33

- 1
2
3
4
5
6
7
8
9
10
11
12
13
14
15
16
17
18
19
20
21
22
23
24
25
26
27
28
29
30
31
32
33
34
35
36
37
38
39
40
41
42
43
44
45
46
47
48
49
50
51
52
53
54
55
56
57
58
59
60
61
62
63
64
65
- [28] C.C. Chang, R. A. Kumar, M. M. Bernitsas, “VIV and galloping of single circular cylinder with surface roughness at $3.0 \times 10^4 < \text{Re} < 1.2 \times 10^5$.” *Ocean Engineering*, 38 (2011) 1713–1732.
- [29] D. Chatterjee, K. Chatterjee, B. Mondal “Control of flow separation around bluff obstacles by transverse magnetic field.” *ASME Journal of Fluids Engineering*, 134 (2012) 091102-1.
- [30] D. Chatterjee, “Dual role of thermal buoyancy in controlling boundary layer separation around bluff obstacles.” *International Communications in Heat and Mass Transfer*, 56 (2014) 152–158.
- [31] D. Chatterjee, B. Mondal, “Control of flow separation around bluff obstacles by superimposed thermal buoyancy.” *International Journal of Heat and Mass Transfer*, 72 (2014) 128–138.
- [32] D. Chatterjee, S. Ray, “Influence of thermal buoyancy on boundary layer separation over a triangular surface.” *International Journal of Heat and Mass Transfer*, 79 (2014) 769–782.
- [33] D. Chatterjee, C. Sinha, “Influence of thermal buoyancy on vortex shedding behind a rotating circular cylinder in cross flow at subcritical Reynolds numbers.” *Journal of Heat Transfer*, 136 (2014) 051704-1.
- [34] Z. Chen, N. Aubry, “Active control of cylinder wake.” *Communications in Nonlinear Science and Numerical Simulation*, 10 (2005) 205–216.
- [35] Y. J. Chen, C.P. Shao, “Suppression of vortex shedding from a rectangular cylinder at low Reynolds numbers.” *Journal of Fluids and Structures*, 43 (2013) 15–27.
- [36] W.L. Chen, D.B. Xin, F. Xu, H. Li, J. P. Ou, H. Hu, “Suppression of vortex-induced vibration of a circular cylinder using suction-based flow control.” *Journal of Fluids and Structures*, 42 (2013) 25–39.

- 1
2
3
4
5
6
7
8
9
10
11
12
13
14
15
16
17
18
19
20
21
22
23
24
25
26
27
28
29
30
31
32
33
34
35
36
37
38
39
40
41
42
43
44
45
46
47
48
49
50
51
52
53
54
55
56
57
58
59
60
61
62
63
64
65
- [37] H. Choi, W.P. Jeon, J. Kim, "Control of flow over a bluff body." *Annual review of fluid mechanics*, 40 (2008) 113-139.
- [38] J. D'Adamo, R. Sosa, M. Barcelo, G. Artana, "Wake flow stabilization with DBD plasma actuators for low Re numbers." *Journal of Physics: Conference Series*, 296 (1) (2011) 1-11.
- [39] R.M. Darekar, S.J. Sherwin, "Flow past a square-section cylinder with a wavy stagnation face." *Journal of Fluid Mechanics*, 426 (2001) 263- 295.
- [40] P.K. Das, S. Mathew, A.J. Shaiju, B.S.V. Patnaik, "Energetically efficient proportional-integral differential (PID) control of wake vortices behind a circular cylinder." *Fluid Dynamics Research*, 48 (2016) 015510.
- [41] Y. Delaunay, L. Kaiktsis, "Control of circular cylinder wakes using base mass transpiration." *Physics of fluids*, 13 (2001) 3285–3301.
- [42] M. El-Gammal, H. Hangan, P. King, "Control of vortex shedding-induced effects in a sectional bridge model by spanwise perturbation method." *Journal of Wind Engineering and Industrial Aerodynamics*, 95 (2007) 663–678.
- [43] L. H. Feng, J. J. Wang, "Circular cylinder vortex-synchronization control with a synthetic jet positioned at the rear stagnation point." *Journal of Fluids Mechanics*, 662 (2010) 232–259.
- [44] L. H. Feng, J. J. Wang, C. Pan, "Effect of novel synthetic jet on wake vortex shedding modes of a circular cylinder." *Journal of Fluids and Structures*, 26 (2010) 900–917.
- [45] L. H. Feng, J. J. Wang, C. Pan, "Proper orthogonal decomposition analysis of vortex dynamics of a circular cylinder under synthetic jet control." *Physics of Fluids*, 23 (2011) 014106-1-13.
- [46] T.L.B. Flinois, T. Colonius, "Optimal control of circular cylinder wakes using long control horizons." *Physics of Fluids*, 27 (2015) 087105.

- 1
2
3
4
5
6
7
8
9
10
11
12
13
14
15
16
17
18
19
20
21
22
23
24
25
26
27
28
29
30
31
32
33
34
35
36
37
38
39
40
41
42
43
44
45
46
47
48
49
50
51
52
53
54
55
56
57
58
59
60
61
62
63
64
65
- [47] H. Fu, D. Rockwell, "Shallow flow past a cylinder: control of the near wake." *Journal of Fluid Mechanics*, 539 (2005) 1–24.
- [48] N. Fujisawa, Y. Kawaji, K. Ikemoto, "Feedback control of vortex shedding from a circular cylinder by rotational oscillations." *Journal of Fluids and Structures*, 15 (2001) 23-37.
- [49] N. Fujisawa, T. Nakabayashi, "Neural network control of vortex shedding from a circular cylinder using rotational feedback oscillations." *Journal of Fluids and Structures*, 16 (2002) 113–119.
- [50] M. Gad-el-Hak, D. M. Bushnell "Separation control: review." *Journal of Fluids Engineering*, 113 (1991) 5–30.
- [51] Y. Gao, S. Fu, J. Wang, L. Song, Y. Chen, "Experimental study of the effects of surface roughness on the vortex-induced vibration response of a flexible cylinder." *Ocean Engineering*, 103 (2015) 40–54.
- [52] B. Gozmen, H. Akilli, B. Sahin, "Vortex control of cylinder wake by permeable cylinder." *Çukurova University Journal of the Faculty of Engineering and Architecture*, 28 (2013), 77-85.
- [53] B. Gozmen, H. Akilli, B. Sahin, "Passive control of circular cylinder wake in shallow flow." *Measurement*, 46 (2013), 1125–1136.
- [54] C. Graham, P. Huang, "Analysis of a passive flow control Device via flow visualization techniques." A thesis for BS in Aerospace Engineering, California Polytechnic State University, Spring 2010.
- [55] F. Gu, J.S. Wang, X.Q. Qiao, Z. Huang, "Pressure distribution, fluctuating forces and vortex shedding behavior of circular cylinder with rotatable splitter plates." *Journal of Fluids and Structures*, 28 (2012) 263–278.

- 1
2
3
4
5
6
7
8
9
10
11
12
13
14
15
16
17
18
19
20
21
22
23
24
25
26
27
28
29
30
31
32
33
34
35
36
37
38
39
40
41
42
43
44
45
46
47
48
49
50
51
52
53
54
55
56
57
58
59
60
61
62
63
64
65
- [56] J.W. He, R. Glowinski, R. Metcalfe, A. Nordlander, J. Périaux, “Active control and drag optimization for flow past a circular cylinder. Part 1. Oscillatory cylinder rotation.” *Journal of Computational Physics*, 163 (2000) 83-117.
- [57] S. Hiejima, T. Kumao, T. Taniguchi, “Feedback control of vortex shedding around a bluff body by velocity excitation.” *International journal of computational fluid dynamics*, 19 (2005) 87–92.
- [58] C. Homescu, I. M. Navon, Z. Li, “Suppression of vortex shedding for flow around a circular cylinder using optimal control.” *International Journal for Numerical Methods in Fluids*, 38 (1) (2002) 43–69.
- [59] S. Huang, “VIV suppression of a two-degree-of-freedom circular cylinder and drag reduction of a fixed circular cylinder by the use of helical grooves.” *Journal of Fluids and Structures*, 27 (2011) 1124–1133.
- [60] S. Huang, D. Clelland, S. Day, R. James, “Drag reduction of deepwater risers by the use of helical grooves.” 2007, International conference on offshore mechanics & arctic engineering.
- [61] X.Y. Huang, “Suppression of vortex shedding from a circular cylinder by internal acoustic excitation.” *Journal of Fluids and Structures*, 9 (1995) 563–570.
- [62] X.Y. Huang, “Feedback control of vortex shedding from a circular cylinder.” *Experiments in Fluids*, 20 (1996) 218–224.
- [63] F.J. Huera-Huarte, “On splitter plate coverage for suppression of vortex-induced vibrations of flexible cylinders.” *Applied Ocean Research*, 48 (2014) 244–249.
- [64] J.Y. Hwang, K.S. Yang, S.H. Sun, “Reduction of flow-induced forces on circular cylinder using a detached splitter plate.” *Physics of Fluids*, 15 (8) (2003) 2433–2436.
- [65] K.T. Hyun, C.H. Chun, “The wake flow control behind a circular cylinder using ion wind.” *Experiments in Fluids*, 35 (2003) 541–552.

- 1
2
3
4
5
6
7
8
9
10
11
12
13
14
15
16
17
18
19
20
21
22
23
24
25
26
27
28
29
30
31
32
33
34
35
36
37
38
39
40
41
42
43
44
45
46
47
48
49
50
51
52
53
54
55
56
57
58
59
60
61
62
63
64
65
- [66] W. T. Joe, T. Colonius, D. G. MacMynowski, “Optimized waveforms for feedback control of vortex shedding.” 39th AIAA Fluid Dynamics Conference, June 2009 San Antonio, Texas.
- [67] P. Joseph, C. Edouard, X. Amandolese, J. L. Aider, “Flow control using MEMS pulsed micro-jets on the Ahmed body.” *Experiments in Fluids*, 54 (2013) 1–12.
- [68] T.N. Jukes, K.S. Choi, “Long lasting modifications to vortex shedding using a short plasma excitation.” *Physical Review Letters*, 102 (2009) 254501.
- [69] J. Kim, H. Choi, “Distributed forcing of flow over a circular cylinder.” *Physics of Fluids*, 17 (2005), 0331031-16.
- [70] R. King, “A review of vortex shedding research and its application.” *Ocean Engineering*, 4 (1997) 141-171.
- [71] K.Y. Kiu, B. Stappenbelt, K.P. Thiagarajan, “Effects of uniform surface roughness on vortex-induced vibration of towed vertical cylinders.” *Journal of Sound and Vibration*, 330 (2011) 4753–4763.
- [72] I. Korkischko, J.R. Meneghini, “Suppression of vortex-induced vibration using moving surface boundary-layer control.” *Journal of Fluids and Structures*, 34 (2012) 259–270.
- [73] R. A. Kumar, C.H. Sohn, B.H.L. Gowda, “Passive control of vortex-induced vibrations: an overview.” *Recent Patents on Mechanical Engineering*, 1 (2008) 1-11.
- [74] S. Kunze, C. Brücker, “Control of vortex shedding on a circular cylinder using self-adaptive hairy-flaps.” *C. R. Mecanique*, 340 (2012) 41–56.
- [75] C. H. Kuo, C. C. Chen, “Passive control of wake flow by two small control cylinders at Reynolds number 80.” *Journal of Fluids and Structures*, 25 (2009) 1021–1028.
- [76] K. Kwon, H. Choi, “Control of laminar vortex shedding behind a circular cylinder using splitter plates.” *Physics of Fluids*, 8 (1996) 479–486.

- 1
2
3
4
5
6
7
8
9
10
11
12
13
14
15
16
17
18
19
20
21
22
23
24
25
26
27
28
29
30
31
32
33
34
35
36
37
38
39
40
41
42
43
44
45
46
47
48
49
50
51
52
53
54
55
56
57
58
59
60
61
62
63
64
65
- [77] G.C. Layek, C. Midya, A.S. Gupta, “Influences of suction and blowing on vortex shedding behind a square cylinder in a channel.” *International Journal of Non-Linear Mechanics*, 43 (2008) 979-984.
- [78] E. Leclerc, P. Sagaut, B. Mohammadi, “On the use of incomplete sensitivities for feedback control of laminar vortex shedding.” *Computers & Fluids*, 35 (2006) 1432–1443.
- [79] J.C. Lecordier, L. Hamma, P. Paranthoen, “The control of vortex shedding behind heated circular cylinders at low Reynolds numbers.” *Experiments in Fluids*, 10 (1991) 224-229.
- [80] J.C. Lecordier, L.W.B. Browne, S. Le Masson, F. Dumouchel, P. Paranthoen, “Control of vortex shedding by thermal effect at low Reynolds numbers.” *Experimental Thermal and Fluid Science*, 21 (2000) 227-237.
- [81] S. Lee, H. Kim, “The effect of surface protrusions on the near wake of a circular cylinder.” *Journal of Wind Engineering and Industrial Aerodynamics*, 69–71 (1997) 351-361.
- [82] S.J. Lee, S.I. Lee, C.W. Park, “Reducing the drag on a circular cylinder by upstream installation of a small control rod.” *Fluid Dynamics Research*, 34 (2004) 233–250.
- [83] S.J. Lee, J.Y. Lee, “Flow structure of wake behind a rotationally oscillating circular cylinder.” *Journal of Fluids and Structures*, 22 (2006) 1097–1112.
- [84] S.J. Lee, J.Y. Lee, “PIV measurements of the wake behind a rotationally oscillating circular cylinder.” *Journal of Fluids and Structures*, 24 (2008) 2–17.
- [85] C. Lei, L. Cheng, S.W. Armfield, K. Kavanagh, “Vortex shedding suppression for flow over a circular cylinder near a plane boundary.” *Ocean Engineering*, 27 (2000) 1109–1127.
- [86] Z. Li, I.M. Navon, M.Y. Hussaini, F. X. Le Dimet, “Optimal control of cylinder wakes via suction and blowing.” *Computers & Fluids*, 32 (2003) 149–171.
- [87] Y.G. Liu, L. H. Feng, “Suppression of lift fluctuations on a circular cylinder by inducing the symmetric vortex shedding mode.” *Journal of Fluids and Structures*, 54 (2015) 743–759.

- 1
2
3
4
5
6
7
8
9
10
11
12
13
14
15
16
17
18
19
20
21
22
23
24
25
26
27
28
29
30
31
32
33
34
35
36
37
38
39
40
41
42
43
44
45
46
47
48
49
50
51
52
53
54
55
56
57
58
59
60
61
62
63
64
65
- [88] L. Lu, J.M. Qin, B. Teng, Y.C. Li, “Numerical investigations of lift suppression by feedback rotary oscillation of circular cylinder at low Reynolds number.” *Physics of fluids*, 23 (2011) 033601.
- [89] D.K. Maiti, R. Bhatt, “Vortex shedding suppression and aerodynamic characteristics of square cylinder due to offsetting of rectangular cylinders towards a plane.” *Ocean Engineering*, 82 (2014) 91-104.
- [90] J.J. Miao, C.C. Yang, J.H. Chou, K.R. Lee, “Suppression of Low-frequency Variations in Vortex Shedding by a Splitter Plate Behind a Bluff Body.” *Journal of Fluids and Structures*, 7 (1993) 897–912.
- [91] C. Mimeau, I. Mortazavi, G. H. Cottet, “Passive flow control around a semi-circular cylinder using porous coatings.” *International Journal of Flow Control*, 6 (2014) 15-21.
- [92] C. Min, H. Choi, “Suboptimal feedback control of vortex shedding at low Reynolds numbers.” *Journal of Fluid Mechanics*, 401 (1999) 123-156.
- [93] S. Mittal, “Control of flow past bluff bodies using rotating control cylinders.” *Journal of Fluids and Structures*, 15 (2001) 291-326.
- [94] S. Mittal, A. Raghuvanshi, “Control of vortex shedding behind circular cylinder for flows at low Reynolds numbers.” *International Journal of Numerical Method in Fluids*, 35 (2001) 421-447.
- [95] V.J. Modi, “Passive control of vortex-induced vibrations: an overview.” *Journal of Fluids and Structures*, 11 (1997) 627–663.
- [96] S. Muddada, B.S.V. Patnaik, “An active flow control strategy for the suppression of vortex structures behind a circular cylinder.” *European Journal of Mechanics B/Fluids*, 29 (2010) 93–104.

- 1
2
3
4
5 [97] S. Muddada, B.S.V. Patnaik, “Application of chaos control techniques to fluid
6 turbulence.” Applications of chaos and nonlinear dynamics in engineering, Part of the series
7 Understanding Complex Systems, 1 (2011) 87-136.
- 8
9 [98] K. Muralidharan, Sridhar Muddada, B.S.V. Patnaik, “Numerical simulation of vortex
10 induced vibrations and its control by suction and blowing.” Applied Mathematical Modelling,
11 37 (2013) 284–307.
- 12
13 [99] G. Mutschke, V. Shatrov, G. Gerbeth, “Cylinder wake control by magnetic fields in
14 liquid metal flows.” Experimental Thermal and Fluid Science, 16 (1998) 92–99.
- 15
16 [100] G. Mutschke, G. Gerbeth, T. Albrecht, R. Grundmann, “Separation control at hydrofoils
17 using Lorentz forces.” European Journal of Mechanics B/Fluids, 25 (2006) 137–152.
- 18
19 [101] Y. Nakamura, Y. Tomonari, “The effect of surface roughness on the flow past circular
20 cylinders at high Reynolds numbers.” Journal of Fluid Mechanics, 123 (1982) 363–378.
- 21
22 [102] G. Nati, M. Kotsonis, S. Ghaemi, F. Scarano, “Control of vortex shedding from a blunt
23 trailing edge using plasma actuators.” Experimental Thermal and Fluid Science, 46 (2013)
24 199–210.
- 25
26 [103] H. Nishiyama, T. Ota, N. Kon, “Vortex shedding controlled by the transverse vibration
27 of three in-line elliptic cylinders in a uniform flow.” Journal of Wind Engineering and
28 Industrial Aerodynamics, 37 (1991) 141–153.
- 29
30 [104] A. Okajima, T. Nagamori, F. Matsunaga, T. Kiwata, “Some experiments on flow-
31 induced vibration of a circular cylinder with surface roughness.” Journal of Fluids and
32 Structures, 13 (1999) 853–864.
- 33
34 [105] V. Oruc, “Passive control of flow structures around a circular cylinder by using
35 screen.” Journal of Fluids and Structures, 33 (2012) 229–242.
- 36
37 [106] G. M. Ozkan, V. Oruc, H. Akilli, B. Sahin, “Flow around a cylinder surrounded by a
38 permeable cylinder in shallow water.” Experiments in Fluids, 53 (6) (2012) 1751-1763.
- 39
40
41
42
43
44
45
46
47
48
49
50
51
52
53
54
55
56
57
58
59
60
61
62
63
64
65

- 1
2
3
4
5
6
7
8
9
10
11
12
13
14
15
16
17
18
19
20
21
22
23
24
25
26
27
28
29
30
31
32
33
34
35
36
37
38
39
40
41
42
43
44
45
46
47
48
49
50
51
52
53
54
55
56
57
58
59
60
61
62
63
64
65
- [107] G.M. Ozkan, H. Akilli, “Flow control around bluff bodies by attached permeable plates.” *International Journal of Mechanical, Aerospace, Industrial and Mechatronics Engineering*, 8 (2014) 1027–1031.
- [108] S. Ozono, “Vortex suppression of the cylinder wake by deflectors.” *Journal of Wind Engineering and Industrial Aerodynamics*, 91 (2003) 91–99.
- [109] D.S. Park, D.M. Ladd, E.W. Hendricks, “Feedback control of von Kármán vortex shedding behind a circular cylinder at low Reynolds numbers.” *Physics of Fluids*, 6 (1994) 2390–2405.
- [110] H. Park, D. Lee, W.P. Jeon, S. Hahn, J. Kim, J. Kim, J. Choi, H. Choi, “Drag reduction in flow over a two-dimensional bluff body with a blunt trailing edge using a new passive device.” *Journal of fluid mechanics*, 563 (2006) 389–414.
- [111] B.S.V. Patnaik, G.W. Wei, “Controlling Wake Turbulence.” *Physical Review Letters*, 88 (5) (2002) 054502-4.
- [112] E. Pinar, G.M. Ozkan, T. Durhasan, M.M. Aksoy, H. Akilli, B. Sahin, “Flow structure around perforated cylinders in shallow water.” *Journal of Fluids and Structures*, 55 (2015) 52–63.
- [113] O. Posdziech, R. Grundmann, “Electromagnetic control of seawater flow around circular cylinders.” *European Journal of Mechanics B/Fluids*, 20 (2001) 255-274.
- [114] M.L. Post, T.C. Corke, “Separation control on high angle of attack airfoil using plasma actuators.” *AIAA J.*, 42 (11) (2004) 2177–2184.
- [115] B. Protas, A. Styczek, “Optimal rotary control of the cylinder wake in the laminar regime.” *Physics of Fluids*, 14 (7) (2002) 2073.
- [116] S. Rashidi, A. Tamayol, M.S. Valipour, N. Shokri, “Fluid flow and forced convection heat transfer around a solid cylinder wrapped with a porous ring.” *International Journal of Heat and Mass Transfer*, 63 (2013) 91–100.

- 1
2
3
4
5
6
7
8
9
10
11
12
13
14
15
16
17
18
19
20
21
22
23
24
25
26
27
28
29
30
31
32
33
34
35
36
37
38
39
40
41
42
43
44
45
46
47
48
49
50
51
52
53
54
55
56
57
58
59
60
61
62
63
64
65
- [117] S. Rashidi, M. Bovand, I. Pop, M.S. Valipour, “Numerical simulation of forced convective heat transfer past a square diamond-shaped porous cylinder.” *Transport in Porous Media*, 102 (2) (2014a) 207-225.
- [118] S. Rashidi, R. Masoodi, M. Bovand, M.S. Valipour, “Numerical study of flow around and through a porous diamond cylinder with different apex angels.” *International Journal of Numerical Methods for Heat and Fluid Flow*, 24 (7) (2014b) 1504-1518.
- [119] S. Rashidi, J.A. Esfahani, “The effect of magnetic field on instabilities of heat transfer from an obstacle in a channel.” *Journal of Magnetism and Magnetic Materials*, 391(2015e) 5–11.
- [120] S. Rashidi, A. Nouri-Borujerdi, M.S. Valipour, R. Ellahi, I. Pop, “Stress-jump and continuity interface conditions for a cylinder embedded in a porous medium.” *Transport in Porous Media*, 107 (1) (2015a) 171-186.
- [121] S. Rashidi, M. Dehghan, R. Ellahi, M. Riaz, M.T. Jamal-Abad, “Study of stream wise transverse magnetic fluid flow with heat transfer around an obstacle embedded in a porous medium.” *Journal of Magnetism and Magnetic Materials*, 378 (2015b) 128–137.
- [122] S. Rashidi, M. Bovand, J.A. Esfahani, H.F. Öztop, R. Masoodi, “Control of wake structure behind a square cylinder by Magnetohydrodynamics.” *ASME Journal of Fluids Engineering*, 137 (6) (2015c) 061102-8.
- [123] L.J.D. Ribeiro, “Effects of surface roughness on the two-dimensional flow past circular cylinders I: mean forces and pressures.” *Journal of Wind Engineering and Industrial Aerodynamics*, 37 (3) (1991a) 299–309.
- [124] L.J.D. Ribeiro, “Effects of surface roughness on the two-dimensional flow past circular cylinders II: fluctuating forces and pressures.” *Journal of Wind Engineering and Industrial Aerodynamics*, 37 (3) (1991b) 311–326.

- 1 [125] K. Roussopoulos, "Feedback control of vortex shedding at low Reynolds numbers."
2 Journal of Fluid Mechanics, 24 (1993) 267–296.
3
4 [126] K. Roussopoulos, P.A. Monkewitz, "Nonlinear modelling of vortex shedding control in
5 cylinder wakes." Physica D: Nonlinear Phenomena, 97 (1996) 264–273.
6
7 [127] M. S. Reddy, S. Muddada, B. S. V. Patnaik, "Flow past a circular cylinder with
8 momentum injection: optimal control cylinder design." Fluid dynamics research, 45 (2013)
9 015501.
10
11 [128] C.V. Seal, C.R. Smith, "The control of turbulent end-wall boundary layers using
12 surface suction." Experiments in Fluids, 27 (1999) 484-496.
13
14 [129] T.K. Sengupta, K. Deb, S.B. Talla, "Control of flow using genetic algorithm for a
15 circular cylinder executing rotary oscillation." Computers & Fluids, 36 (2007) 578–600.
16
17 [130] A. Shmilovich, Y. Yadlin, "Flow control techniques for transport aircraft." AIAA
18 Journal, 49 (3) (2011) 489-502.
19
20 [131] S. Singha, K.P. Sinhamahapatra, S.K. Mukherjea, "Control of vortex shedding from a
21 bluff body using imposed magnetic field." ASME Journal of Fluids Engineering, 129 (2007)
22 517-523.
23
24 [132] S. Singha, K.P. Sinhamahapatra, "Control of vortex shedding from a circular cylinder
25 using imposed transverse magnetic field." International Journal of Numerical Methods for
26 Heat & Fluid Flow, 21 (1) (2011) 32-45.
27
28 [133] L.L. Shi, Y.Z. Liu, H.J. Sung, "On the wake with and without vortex shedding
29 suppression behind a two-dimensional square cylinder in proximity to a plane wall." Journal
30 of Wind Engineering and Industrial Aerodynamics, 98 (2010) 492–503.
31
32 [134] W.C.L. Shih, C. Wangl, D. Coles, A. Roshko, "Experiments on flow past rough
33 circular cylinders at large Reynolds numbers." Journal of Wind Engineering and Industrial
34 Aerodynamics, 49 (1993) 351-368.
35
36
37
38
39
40
41
42
43
44
45
46
47
48
49
50
51
52
53
54
55
56
57
58
59
60
61
62
63
64
65

- 1
2
3
4
5
6
7
8
9
10
11
12
13
14
15
16
17
18
19
20
21
22
23
24
25
26
27
28
29
30
31
32
33
34
35
36
37
38
39
40
41
42
43
44
45
46
47
48
49
50
51
52
53
54
55
56
57
58
59
60
61
62
63
64
65
- [135] R.K. Shukla, J. H. Arakeri, “Minimum power consumption for drag reduction on a circular cylinder by tangential surface motion.” *Journal of Fluid Mechanics*, 715 (2013) 597-641.
- [136] A. Sohankar, M. Khodadadi, E. Rangraz, “Control of fluid flow and heat transfer around a square cylinder by uniform suction and blowing at low Reynolds numbers.” *Computers & Fluids*, 109 (2015) 155–167.
- [137] D. Son, S. Jeon, H. Choi, “A proportional–integral–differential control of flow over a circular cylinder.” *Philosophical Transactions of the Royal Society A*, 369 (2011) 1540–1555.
- [138] A.G. Straatman, R.J. Martinuzzi, “A study on the suppression of vortex shedding from a square cylinder near a wall.” *Engineering Turbulence Modelling and Experiments*, 5 (2002) 403–411.
- [139] P.J. Strykowski, K.R. Sreenivasan, “On the formation and suppression of vortex ‘shedding’ at low Reynolds numbers.” *Journal of Fluid Mechanics*, 218 (1990) 71–107.
- [140] Y. Sung, W. Kim, M.G. Mungal, M.A. Cappelli, “Aerodynamic modification of flow over bluff objects by plasma actuation.” *Experiments in Fluids*, 41 (2006) 479–486.
- [141] J.S. Tao, X.Y. Huang, W.K. Chan, “a flow visualization study on feedback control of vortex shedding from a circular cylinder.” *Journal of Fluids and Structures*, 10 (1996) 965–970.
- [142] J. Tensi, I. Boue, F. Paille, and G. Dury, “Modification of the wake behind a circular cylinder by using synthetic jets.” *Journal of Visualization*, 5 (2002) 37-44.
- [143] F.O. Thomas, A. Kozlov, T.C. Corke, “Plasma actuators for landing gear noise control.” *Journal of Fluids and Structures*, 2005, 11th AIAA/CEAS Aeroacoustics Conference Monterey, California.

- 1
2
3
4
5
6
7
8
9
10
11
12
13
14
15
16
17
18
19
20
21
22
23
24
25
26
27
28
29
30
31
32
33
34
35
36
37
38
39
40
41
42
43
44
45
46
47
48
49
50
51
52
53
54
55
56
57
58
59
60
61
62
63
64
65
- [144] F. Thomas, A. Kozlov, T. Corke, “Plasma actuators for cylinder flow control and noise reduction.” *AIAA J.*, 46 (8) (2008) 1921-1931.
- [145] P.T. Tokumaru, P. E. Diimotakis, “Rotary oscillation control of a cylinder wake.” *AIM 2nd Shear Flow Conference* March 13-16, 1989 1 Tempe, AZ.
- [146] M.S. Valipour, S. Rashidi, R. Masoodi, “Magnetohydrodynamics flow and heat transfer around a solid cylinder wrapped with a porous ring.” *ASME Journal of Heat Transfer*, 136 (2014a) 062601-9.
- [147] M.S. Valipour, S. Rashidi, M. Bovand, R. Masoodi, “Numerical modeling of flow around and through a porous cylinder with diamond cross section.” *European Journal of Mechanics B/Fluids*, 46 (2014b), 74-81.
- [148] G. Vilaplana, M. Grandemange, M. Gohlke, O. Cadot, “Global mode of a sphere turbulent wake controlled by a small sphere.” *Journal of Fluids and Structures*, 41 (2013) 119–126.
- [149] T. Vit, M. Ren, Z. Travnicek, F. Marsik, C.C.M. Rindt, “The influence of temperature gradient on the Strouhal–Reynolds number relationship for water and air.” *Experimental Thermal and Fluid Science*, 31 (2007) 751–760.
- [150] H.B. Warjito, E. A. Kosasih, R. Tarakka, S. P. Simanungkalit, I. G. Made Fredy Lay Teryanto, “Modification of flow structure over a van model by suction flow control to reduce aerodynamics drag.” *Makara, Teknologi*, 16 (2012) 15-21.
- [151] H.M. Warui, N. Fujisawa, “Feedback control of vortex shedding from a circular cylinder by cross-flow cylinder oscillations.” *Experiments in Fluids*, 21 (1996) 49–56.
- [152] A. Weickgenannt, P. A. Monkewitz, “Control of vortex shedding in an axisymmetric bluff body wake.” *European Journal of Mechanics B/Fluids*, 19 (2000) 93–104.

- 1
2
3
4
5
6
7
8
9
10
11
12
13
14
15
16
17
18
19
20
21
22
23
24
25
26
27
28
29
30
31
32
33
34
35
36
37
38
39
40
41
42
43
44
45
46
47
48
49
50
51
52
53
54
55
56
57
58
59
60
61
62
63
64
65
- [153] T. Weier, G. Gerbeth, G. Mutschke, E. Platacis, O. Lielausis “Experiments on cylinder wake stabilization in an electrolyte solution by means of electromagnetic forces localized on the cylinder surface.” *Experimental Thermal and Fluid Science*, 16 (1998) 84-91.
- [154] T. Weier, G. Gerbeth, “Control of separated flows by time periodic Lorentz forces.” *European Journal of Mechanics B/Fluids*, 23 (2004) 835-849.
- [155] J.E.F. Williams, B.C. Zhao, “The active control of vortex shedding.” *Journal of Fluids and Structures*, 3 (2) (1989) 115-122.
- [156] C.H.K. Williamson, “Vortex Dynamics in the Cylinder Wake.” *Annual Review of Fluid Mechanics*, 28 (1996) 477-539.
- [157] C.H.K. Williamson, R. Govardhan, “Vortex-induced vibrations.” *Annual Review of Fluid Mechanics*, 36 (2004) 413-455.
- [158] D. Wolfe, S. Ziada, “Feedback control of vortex shedding from two tandem cylinders.” *Journal of Fluids and Structures*, 17 (2003) 579–592.
- [159] H. Wu, D.P. Sun, L. Lua, B. Teng, G.Q. Tang, J.N. Song, “Experimental investigation on the suppression of vortex-induced vibration of long flexible riser by multiple control rods.” *Journal of Fluids and Structures*, 30 (2012) 115–132.
- [160] F. Xu, W.L. Chen, Y.Q. Xiao, H. Li, J.P. Ou, “Numerical study on the suppression of the vortex-induced vibration of an elastically mounted cylinder by a traveling wave wall.” *Journal of Fluids and Structures*, 44 (2014) 145–165.
- [161] Z.P. Zang, F.P. Gao, J.S. Cui, “Physical modeling and swirling strength analysis of vortex shedding from near-bed piggyback pipelines.” *Applied Ocean Research*, 40 (2013) 50–59.
- [162] Z.P. Zang, F.P. Gao, “Steady current induced vibration of near-bed piggyback pipelines: Configuration effects on VIV suppression.” *Applied Ocean Research* 46 (2014) 62–69.

- 1
2
3
4
5
6
7
8
9
10
11
12
13
14
15
16
17
18
19
20
21
22
23
24
25
26
27
28
29
30
31
32
33
34
35
36
37
38
39
40
41
42
43
44
45
46
47
48
49
50
51
52
53
54
55
56
57
58
59
60
61
62
63
64
65
- [163] M.M. Zdravkovich, “Review and classification of various aerodynamic and hydrodynamic means for suppressing vortex shedding.” *Journal of Wind Engineering and Industrial Aerodynamics*, 7 (1981) 145-189.
- [164] H. Zhang, B. Fan, Z. Chen, Y. Li, “Numerical study of the suppression mechanism of vortex-induced vibration by symmetric Lorentz forces.” *Journal of Fluids and Structures*, 48 (2014) 62–80.
- [165] M.M. Zhang, L. Cheng, Y. Zhou, “Closed-loop-controlled vortex shedding and vibration of a flexibly supported square cylinder under different schemes.” *Physics of Fluids*, 16 (2004) 1439–1448.
- [166] B. Zhou, X. Wang, W.M. Gho, S.K. Tan, “Force and flow characteristics of a circular cylinder with uniform surface roughness at subcritical Reynolds numbers.” *Applied Ocean Research*, 49 (2015) 20–26.
- [167] H. Zhu, J. Yao, “Numerical evaluation of passive control of VIV by small control rods.” *Applied Ocean Research*, 51 (2015) 93–116.