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## ABSTRACT

Random loading can be characterised to some extent by the Irregularity Factor - that is the ratio of the number of mean-chossings. to the number of max-a min.. of the signal. It is well known that the distribution of maxima is a function of only this parameter and the standard deviation of the random signal. However other probability density distributions of interest to the fatigue engineer (particularly the distribution of ranges) are not known. This is probably due to the lack of versatile systems for recording and anal, sing random loading histories.

This thesis contains a description of the design and development of such a system. liagnetic tape is used as the recording medium and during replay the system produces punched paper tape for subsequent analysis by computer. The paper tape contains sufficient information about the signal to construct all of the histograms of interest in fatigue.

The system is used to make a series of recordings of typical random signals with various values of Irregularity Factor. These recordings are of filtered 'white' noise and car suspension loads. From the data, various distributions are constructed to indicate the effect of the Irregularity Factor. In particular the distributions of ranges are presented, based both on this recorded data and on other data contained in the literature.

Since the Irregularity Factor describes the distribution of peaks, and is seen to be a major parameter in the experimental femily of range distribution curves, it would seem to indicate that it mey be a controling factor
in the lives obtained under true random loading conditions. Accordingly, an examiration is made of random loading fatigue tests published in the literature to see if there are any consistent trends with the Irregularity Factor.

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Several developments in the field of fatigue have indicated the necessity of a versatile system for recording and analysing continuous random signals. In a series of fatigue tests, Kowaleski (ref. 4) examined the effect of the traditional counting methods (for the contruction of histograms) in comparison with true random loading. In this work comparatively narrow band random noise was used, and Kowaleski introduced a parameter to describe the spectrum width. This parameter is simply the ratio between the number of maxima and rinima and the number of mean crossings of the signal. The parameter is now called the Irregularity Factor.

The shortcomings in programine testing have
stimulated the development of random loading fatigue research. The Irregularity Factor is a convenient and useful parameter to describe randon signals. The work of Rice (ref. I) shows that the probability density distribution of maxima is a function only of the r.m.s. value of the signal and the Irregularity Factor. Thus it is suggested by Swanson (ref. 11) that random fatigue teste using a twin lever machine could be periormed. The lever combinations are simply set to simulate random loading with the same value of irregularity factor as is measured or predicted for the structure. The parameter is readily calculated from the power spectrum of the loading.

With the present investment in programmed fatigue testing equipment, and the use in industry of the linear cumulative demage theory, it is important to be aware of the effect of the spectrum width on the particular counting method used. One point wortin study is the assumption that
the variation of mean stress (around the overall mean) can be neglected. Or, expressing this in terms of the traditional counting methods, will the range-mean count give similar estimated lives, or programme test lives, to the range count? Clearly the greater the spectrum width the greater is the effect of variation in mean stress.

It is also assumed in programmed tests (or life estimations) that very small stress amplitudes have a negligible effect. This omission of small stress amplitudes is performed after the range histogram has bee l constructed. Clearly the effect of leaving out these ranges before constructing the histogram is that a greater number of larger ranges will be detected. In the first case the following reduction takes place:

in the second:


The manner in which the various counting methods break down the data depends on the Irregularity Factor.

There is a further problem concerned with the range count. As is well known the theoretical determination of the distribution of ranges for a Gaussian signal is very difficult, except of course for very narrow bandwidth noise, when there is a Rayleigh distribution. Indeed the empirical distribution of ranges is unknow. It is possible
for example, that the distribution of ranges, like that of peaks, may depend only on the standard deviation of the signal and the Irregularity Factor. In order to find the empirical distribution of ranges a recording system with high resolution is required. A recording system with poor resolution will not find the true distribution due to the effect described above; i.e. the omission of small ranges leads to a completely different distribution of the remaining ranges.

There are, however, practical limits to the effective resolution of any recording systern. For example when recording stress histories the familiar chain of strain gauge, strain-gauge bridge amplifying unit and measuring device leads to overall experimental accuracies of some 2 to $3 \%$, and noise levels of about 0.2 to $0.3 \%$. As far as the range distribution is concerned, a system with very high resolution breaks down the ranges still further because of the noise. A practical compromise is probably a resolution around $\frac{1}{2} \%$ full scale. The range distribution can then be found making the assumption that there are proportionately very few ranges below this level. Indeed for a Rayleigh distribution of ranges (very narrow bandwidth signal) the proportion of ranges below this level is only 3 in 10,000 (assuming a standard deviation of $10 \%$ full scale). For wider bandwidth signals there will clearly be more, but even if there is an order of magnitude more, the range distribution will not be significantly affected.

The development of a recording system which would provide an automatic means of recording and analysing random strain histories is a necessity for studies of this kind. The first section of this thesis describes the design and development of such a system. A major criterion
in the design of the system is cost. Host high resolution systems, particularly digital ones, are very expensive. The design concepts are therefore evolved with cost in mind, consistent, of course, with the required overall accuracy of the system.

The recording system is then used in two series of tests. In the first series two examples of filtered 'white' noise are recorded with different band-pass filter settings for the two cases. In the second series the system is used for the measurement of car suspension loads to provide an example with a wider spectrum. From the recordings the following studies are made.
(i) The signals are examined to see if they are, in fact Gaussian. A comparison is made between actual and theoretical distributions of maxima for the measured values of the irregularity factor. For the filtered random noise recordings, where the spectral density distributions are known, the expected theoretical values of irregularity factor are compared with those obtained experimentally.
(ii) The various counting methods are compared, in particular the effect on the range distribution of omitting small ranges. All these comparisons are made using probebility density distributions. These distributions illustrate the differences between the counting methodsfar more clearly than cumulative plots of the data. Indeed cumulative plots can be misleading in that respect. As an illustration of these comparisons a number of fatigue damage functions are contructed. These functions are used to compare all the above counting
methods assuming linear cumulative damage. This provides a quantitative indication of the effects of both the irregularity factor and the various counting methods on estimated fatigue lives.
(iii) An empirical distribution of ranges is established using, in addition to the data recorded here, data published by Naumann and Leybold (refs. 9 and lO).

Finally a survey is made of random fatigue testing to see if there are any clear trends in the effect of jr gularity factor.

Before describing the digital recording system, however, a brief outline of random noise theory (principally the work of Rice, ref. I) is presented. Particular attention is paid to the role of the irregularity factor in the theory.

## 2 THE IRREGULARITY FACTOR

### 2.1 Introduction

The parameter introduced by Kowaleski (ref. 4) to describe the spectrum width is simply the ratio between the number of maxima and minima and the number of mean crossings. It is a convenient parameter to describe random signals not only because its meaning is immediately obvious and it is easily measured, but because it also has particular relevance to random noise theory. It is now known as the Irregularity Factor.

In the classic random noise theory of Rice (ref. 1) the parameter is shown to be a simple function of three of the moments of the energy spectrum of the signal. from the equation derived by Rice for the distribution of maxima, Cartwright and Longuet-Higgins (ref. 2) have shown that the distribution of maxima depends only on the root-mean-square of the signal and a parameter which represents the frequency spectrum of the signal. This latter parameter is very simply related to the Kowaleski Irregularity Factor. Either parameter also relates the r.m.s. peaks to the r.m.s. of the continuous signal. There follows a brief outline of the theory derived by Rice, and extended by Cartwright and Longuet-Higgins.

The Fourier series representation of random noise is an inininite number of sine-waves;

$$
f(t)=\sum_{n=1}^{\infty} c_{n} \cos \left(\sigma_{n} t+\varepsilon_{n}\right)
$$

where the frequencies $\sigma_{n}$ are distributed throughout 0 to $\infty$ and the phases $\varepsilon_{n}$ are distributed throughout 0 to $2_{r i}$, and
the amplitudes $c_{n}$ are such that

$$
\sum_{\sigma_{n}=0^{-}}^{\sigma+d \sigma}{ }^{\frac{1}{2}} c_{n}^{2}=E(\sigma) \cdot d \sigma
$$

where $\mathbb{E}(\sigma)$ is the energy spectrum of $f(t)$. The summations then become Fourier integrals.

To obtain the expected number of maxima and meancrossings per second, and also the distribution of maxima, use is made of the Central Limit Theorem. This theorem states that the joint probability density distribution of a number of random variables approaches a normal law when the distributions of the variables satisfy certain conditions (see refs. 1 or 5). Thus the multi-variate normal distribution can be used to find the distribution of maxima, etc. Again using Rice's notation the multivariate normal distribution is

$$
p\left(x_{1}, x_{2}, \ldots \cdots x_{k}\right)=2 \pi^{-k / 2}|M|^{-\frac{1}{2}} \exp \left(-\frac{1}{2} x^{\prime} H^{-1} x\right)
$$

where

$$
x=\left\{x_{1}, x_{2}, \ldots . x_{k}\right\}
$$

and
and

$$
M=\left[\begin{array}{lll}
\mu_{11} & \cdots & \mu_{1 k} \\
\mu_{k i} & \cdots & \cdots
\end{array} \mu_{k k} .\right]
$$

$$
\mu_{i j}=\overline{x_{i}} \bar{x}_{j}
$$

Thus any particular distribution is found by calculating the relevant members $\mu_{i j}$ of .

In the cases considered here the members $\mu_{i j}$ of in are simply the various moments of the energy spectrum $E(\sigma)$ of $f(t)$. The $n^{\text {th }}$ moment is

$$
m_{n}=\int_{0}^{\infty} E(\sigma) \cdot \sigma^{n} \cdot d \sigma
$$

where the total energy per unit length of $f(t)$ is

$$
m_{0}=\int_{0}^{\infty} E(\sigma) \cdot d \sigma^{\circ}
$$

2.2 Number of zeros per second

Rice evaluates the expected number of zeros per second by considering the probability that $y$ will pass through zero with positive slope in the interval $t_{1}, t_{1}+d t$ which is
where

$$
P(\eta d t<\xi<0)=d t \int_{0}^{\infty} \eta \cdot p\left(0, \eta ; t_{1}\right) d \eta
$$

$$
\xi=P\left(a_{1}, a_{2} \ldots \ldots a_{N} ; t\right) \quad\left[\text { i.e. } \xi=y \text { at } t_{1}\right]
$$

and

$$
\eta=\left[\frac{\partial F}{\partial t}\right]_{t=t_{1}}
$$

The probability density function for the two random variables $\xi$ and $\eta$ reduces to (using the Central Limit Theorem)

$$
p\left(\xi, \eta ; t_{1}\right)=\frac{\left(-m_{0} m_{2}\right)^{-\frac{1}{2}}}{2 \pi} \exp \left[\frac{-\xi^{2}}{2 m_{0}}+\frac{\eta^{2}}{2 n_{2}}\right]
$$

The probability of having a zero in $t_{1}, t_{1}+d t$ with positive slope is thus

$$
\alpha t \int_{0}^{\infty} \eta \cdot \frac{\left(-m_{0} m_{2}\right)^{-\frac{1}{2}}}{2 \pi} \exp \left(\frac{\eta^{2}}{2 m_{2}}\right) d \eta=\frac{\partial t}{2 \pi}\left[\frac{-m_{2}}{m_{0}}\right]^{\frac{1}{2}}
$$

Thus the expected number or zeros per second with positive slope is

$$
N_{\delta}=\frac{1}{2 \pi}\left[-\frac{m_{2}}{m_{0}}\right]^{\frac{1}{2}}
$$

2.3 Distribution of maxima.

The probability that $y$ has a maximum in the rectangle ( $\left.t_{1}, t_{1}+d t ; y_{1}, y_{1}+d y\right)$ is
where

$$
\begin{aligned}
P\left(\zeta d t<\eta<0 \text { AND } y_{1}\right. & \left.\leqslant \xi \leqslant y_{1}+d y\right)_{0} \\
& =-d t . d y \cdot \int_{-\infty} p\left(y_{1}, 0, \xi\right) d \xi
\end{aligned}
$$

$$
S=\left[\frac{\partial^{2} F}{\partial t^{2}}\right]_{t=t_{1}}^{-\infty}
$$

The probability density function for the three random variables $\xi, \eta$ and 5 required here is

$$
\mathrm{p}(\xi, \dot{0}, \zeta)=-2 \pi^{-3 / 2}|\mathrm{~m}|^{-\frac{1}{2}} \exp \left[-\frac{1}{2|\mathrm{M}|} \xi^{2}+\mathrm{H}_{33} 丂^{2}+2 \mathrm{~m} \cdot \xi \cdot 5\right]
$$

where

$$
M=\left[\begin{array}{ccc}
m_{0} & 0 & m_{2} \\
0 & -m_{2} & 0 \\
m_{2} & 0 & m_{4}
\end{array}\right]
$$

and $M_{r s}$ is the cofactor of $\mu_{r s}$ in I . The probability of a maxima in the rectangle is then obtained by integration with respect to 5 ;

$$
\begin{aligned}
& \times \exp \left(-\frac{\xi^{2}}{2 m_{0}}\right) \quad \ldots \ldots(1)
\end{aligned}
$$

2.4 Number of maxima per second

Rice obtains the expected number of maxima per second by integrating the above with respect to $t$ from $t_{1}$ to $t_{2}$ and with respect to $y$ from $-\infty$ to $+\infty$. The expected number is

$$
\begin{equation*}
N_{1}=\frac{t_{2}-t_{1}}{2 \pi} \cdot\left[\frac{m_{4}}{-m_{2}}\right]^{\frac{1}{2}} \tag{2}
\end{equation*}
$$

and the expected number per second is

$$
N_{1}=\frac{1}{2 \pi_{i}}\left[\frac{m_{4}}{-m_{2}}\right]^{\frac{1}{2}}
$$

2.5 The Irregularity Factor

The Irregularity Factor is then

$$
n=\frac{N_{1}}{\partial} / N_{1}=\frac{m_{2}}{\left(m_{0} m_{4}\right)^{\frac{1}{2}}}
$$

2.6 Distribution of maxima in terms of Irregularity Factor

Cartwright and Longuet-Higgins (ref. 2) obtain the probability density distribution of maxima by dividing the equation for the probability of a maxina in the rectangle $\left(y_{1}, y_{1}+d y ; t_{1}, t_{1}+d t\right)$, given by equation (I), by the total number of maxima in the interval ( $t_{1}, t_{1}+d t$ ), which is (from equation (2) )

$$
\frac{d t}{2 \pi}\left[\frac{m_{4}}{-m_{2}}\right]^{\frac{1}{2}}
$$

Cartwright and Ionguet-Higgins then substitute for $1 /$ and its cofactoxs in terms of the moments $m_{n}$ of the energy spectrum, and then express the moments $m_{n}$ in terms of a spectrum width parameter $\mathcal{E}$, given by

$$
\varepsilon^{2}=\left(m_{0} m_{4}-m_{2}^{2}\right) / m_{0} m_{4}
$$

to obtain the distribution equation in terms of $\varepsilon$. Clearly $\varepsilon$ is related to the irregularity factor ( $n$ ) by

$$
\varepsilon^{2}=1-n^{2}
$$

The probability density distribution of maxima can now be expressed in terms of the irregularity factor by

$$
p(x)=\underbrace{\sqrt{\frac{1-n^{2}}{2 \pi}} \exp \left(-\frac{x^{2}}{1-n^{2}}\right)}_{\text {Gauss term }}+\underbrace{\frac{n}{2} x\left[1+\operatorname{erf}\left(x \sqrt{\frac{n^{2}}{2\left(1-n^{2}\right)}}\right)\right]}_{\text {Rayleigh term }}
$$

where

$$
x=\xi / m_{0}^{\frac{1}{2}}
$$

The distribution equation has two components, a Gauss term and a Rayleigh term. For the case of $n=0$ the distribution is pure Gaussian and for $n=1$ pure Rayleigh. The distributions are plotted in Fig. 2.1 for values of $n$ from 0 to $l$ showing the transition from the Gaussian distribution to the Rayleigh distribution.

Thus the distribution of maxima is a function only of the r.m.s. ( $m_{0}^{\frac{1}{2}}$ ) and the irregularity factor ( $n$ ).
2.7 Relation between r.m.s. maxima and r.m.s. function

After recordings using the Digital Recording System have been made, the data which is extracted to construct histograms consists of only maxima and minima. From this must be calculated the r.m.s. value of the continuous function originally recorded.

Cartwright and Longuet-Higgins (ref. 2) also show that the second moment $\left(\nu_{2}\right)$ of the variate $\xi$ (the maxima) is
ere
i.e.

$$
\begin{aligned}
& \nu_{2}=\left(2-\varepsilon^{2}\right) m_{0} \\
& \varepsilon^{2}=\left(m_{0} m_{4}-m_{2}^{2}\right) / m_{0} m_{4}
\end{aligned}
$$

therefore

$$
m_{0}^{\frac{1}{2}}=\nu_{2}^{\frac{1}{2}} /\left(1+n^{2}\right)^{\frac{1}{2}}
$$

For exanple with a sine-wave ( $n=1$ ) the ratio between the r.m.s. peaks (in this case also the amplitude) and the r.m.s. of the continuous wave is 2 as expected.

### 2.8 Filtered Random Noise

To obtain examples of narrow spectrum noise for the sequence of recordings filtered random noise was used.

For filtered white noise (ie. noise assumed to have constant energy distribution throughout the bandwidth) the irregularity factor is

$$
\begin{equation*}
\left.\left.n=\frac{m_{2}}{\left(m_{0} m_{4}\right.}\right)^{\frac{1}{2}}=\frac{\int_{f_{1}}^{f_{2}} \omega_{0} f^{2} d f}{\left[\int_{f_{1}}^{f_{2}} \omega_{0} d f \int_{f_{1}}^{f_{2}} \omega_{0} f^{4} d f\right.}\right]^{\frac{1}{2}} \tag{3}
\end{equation*}
$$

where $\omega_{0}$ is the constant energy density and $f_{1}$ and $f_{2}$ are the lower and upper limits of the band-pass filter - the latter assumed to be a perfect filter.

$$
\begin{equation*}
\therefore n=\frac{\sqrt{5}}{3} \frac{1-\phi^{3}}{\left[(1-\phi)\left(1-\phi^{5}\right)\right]^{\frac{1}{2}}} \tag{4}
\end{equation*}
$$

where

$$
\phi=f_{1} / f_{2}
$$

This function is shown in Fig. 2.2

In practice because of the relatively poor characteristics of the band-pass filter equation (4) was used only as a guide in selecting pass-bands. The theoretical estimates of irregularity factor (required for comparison with the measured factors) were obtained from equation (3). The necessary integrations were performed using the filter response curves obtained from an independent calibration.



FIG. 2.2 IRREGULARITY FACTOR FOR BAND PASS FILTER.

## 3 DIGITAL RECORDING SYSTEM

3.1 Introduction

From an analogue signal various histograms are required relating to the maximum and minimum values of the signal. Thus a recording system providing automatic dataprocessing capabilities is necessary. Also the design of the recording system should not preclude its use as a general purpose recorder.

Using a digital computer for the generation of histograms the interface between recording system and computer must be established. Because of the continuous and sequential nature of the information it is feasible to use either online replay, or offline with magnetic tape or paper tape. At the time of the development of the recording system the University was about to change to the Atlas computer and in these early days did not recommend an $y$ other than offline usage with paper tape.

In order to reduce the volume of input to the computer only successive maximum and minimum values of the analogue signal are required on the paper tape. To process information up to a nominal 20c/sec. generated through a filter with slope of $20 d B / o c t$. a maximum information rate of about $60 \mathrm{c} / \mathrm{sec}$. must be acceptable for three figure accuracy. Thus if a four character word represents each maximum or minimum this corresponds to a character rate of 480 char/sec. A Creed paper tape punch of 33 char/sec punching rate was available in the department at the time. Clearly the replay of the information for punching must take place at one sixteenth oi the recording speed. The recording is to be done on magnetic tape and to obtain the information from the magnetic tape at
one..sixteenth of the recording speed a flux-sensitive replay system is used.

Two methods of recording are considered; analogue (f.m.) and digital. The former means that digitization occurs on replay, and in the case of the latter, before recording. However f.m. recording is ruled out because of the $\frac{1}{2} \%$ full scale resolution requirement. With digital recording of instantaneous values of the signal the volume of information is very large, necessitating fast (and expensive) digital systems. Conversely if we record only maximum and minimum values the volume of information is substantially reduced. However such a recording system would not be very useful for general purposes.

Incremental digital recording (that is the recording of increments in the signal) substantially reduces the volume of information about the signal. However it has the inherent characteristic of accumulative errors. Because of the limitations of magnetic tape (in respect of quality and wear) low pulse densities, together with careful tape handing and high standards of deck cleanliness, would be necessary.

Successful use in computers of magnetic tape with densities up to $800 \mathrm{p} . \mathrm{p} . \mathrm{i}^{*}$., together with manufacturers' claims (of one or less dropouts over l800ft. for eight tracks at 200 p.p.i.) encouraged the view that performance at 200 p.p.i. or less would be acceptable. Also the bit rate for incremental recording can be reduced by buffering the information. To record 16 full scale cycles per second using increments of $1 / 500$ full scale means accepting a bit rate of $16 \mathrm{kc} / \mathrm{sec}$. Through buffering up to seven pulses and recording a three bit word parallel tracks, the word rate to tape would be $2.3 \mathrm{kc} / \mathrm{sec}$ - quite acceptable at 15 i.p.s. ${ }^{*}{ }^{*}$

* p.p.i. - pulses per inch.
**i.p.s. - inches per second.

In this application where maximum and minimum values are ultimately required, buffering has a negligible effect on accuracy as the signal changes less rapidly in the neighbourhood of maximum and minimum values. As far as general purpose recording is concerned it is equivalent to sampling the signal at a rate acceptable to the magnetic tape medium.

A further advantage of incremental recording is that slower digital systems may be used - thus each digital electronic unit will be cheaper. Wow and flutter characteristics of the tape deck are not critical as the digital information itself contains clock pulses.

Such a digital recording system requires a multichannel tape recorder with acceptable skew characteristics. However at the time of development a recorder with the required tape speed ranges together with flux-sensitive replay facilities had just become comercially available at a reasonable price.

The system developed here is an incremental recording system with a clock frequency of $16 \mathrm{kc} / \mathrm{sec}$ and input signal ronge of $0-1$ volts. The incremental step is $1 / 500$ full scale (i.e. $2 \mathrm{~m} V$ ). The digital unit value used throughout is

1 mV so that the system counts in steps of two digits. The information corresponding to a step of 2 mV is called one bit. The system buffer accepts up to seven bits in every eight clock cycles. The tape speed for recording is 15 i.p.s., corresponding to a pulse density of 133 p.p.i.

There follows a description of the Digital Recording System. Details of the circuits of logic units, amplifiers and other units can be found in Appendix 1.
3.2 Principles of Operation of D.E.S.
3.2.1 Recording

The central unit of the system is an analogue to digital converter. A three figure BCD counter controls a set of current switches such that the current into the virtual earth (e, Fig. 3.1) is proportional to the number in the counter. The feedback in the system maintains the relation
that is

$$
\begin{aligned}
v_{i n} & =i \cdot R \\
v_{e} & =0
\end{aligned}
$$

in the following manner.

An out of balance in the above relation causes the virtual earth potential to change by the same amount. This out of balance is amplified and opens the gate (Fig. 3.1) allowing pulses from the oscillator into the counter. The counter polarity control is directed by the polarity of the out of balance. The gate remains open until the input voltage is balanced. Thus the number in the counter is always proportional to the input voltage.

The pulses passing through the gate are stored and output in batches onto magnetic tape. A.'batch' is the resultant count of pulses received during each cycle of eight oscillator pulses, and can be a count of up to plus or minus seven. The eighth clock pulse is directed from the gate to allow time for the store unit's output and reset operations. The maximun amount of seven means that only three magnetic tape tracks, for binary -coded recording (excluding polarity tracks), are required.
3.2.2 Replay

The batches of pulses are replayed from magnetic tape into a serializer and then input into the counter
with the polarity information. The system therefore integrates the information on tape into digital values which continuousily represents the original data, and which can be sampled when required. Mexima and minima are detected by changes in batch -polarity, and the number then in the counter is stored in a memory unit for subsequent decoding and punching onto paper tape.

To cater for the low speed of paper tape punches the tape recorder is replayed at a fraction of the recording speed. The system was originally designed for a Creed punch ( $33 \mathrm{c} \mathrm{c} / \mathrm{sec}$ ) but was subsequently used with a Westrex ( $120 \mathrm{ch} / \mathrm{sec}$ )

A 'staircase' model of the original analogue signal can be constructed with the system in Fig. 3.2 which uses an operational amplifier.
3.3 Details of D.R.S.
3.3.1 Input Amplifiers, Gate and Polarity Control

The input amplifiers operate the gate and polarity control from the out-of-balance signal on the virtual earth. The amplifiers are d.c. differential; for the gate both output stages drive an OR unit and for the polarity control they drive either side of a bistable unit. There is more gain in the polarity control circuit so that polarity is registered ahead of the operation of the gate (Fig. 3.3).

The detection thresholds of the gate circuit are $\pm 2.5 \mathrm{mV}$. Thus there is a 'dead-space-region' to reduce the incidence of triggering of the system by noise.
3.3.2 Store Unit

The store unit receives the main counter input pulses and polarity information and is controlled by record-command signals from the record control unit. The count of pulses received between record-command signals is output in binary code (with polarity) to the magnetic tape recorder input modules. After each record-command signal the store unit resets itself to zero. When the count is zero a pulse is output on the "+" track so that the "+" and "-" tracks together constitute a clock record.

The unit consists of a four stage binary counter and the coding networks for output. The counter's first three stages store the batch count, the fourth stage is the batch polarity register (Fig. 3.4).

### 3.3.3 Binary Coded Decimal Counter

Four binary units connected in series form a sixteen bit counter, whose counting sequence is scale corrected to form a decimal counter. The scale correction technique used maintains a constant place 'weight' for each binary place of $1,2,4$ and 2 respectively. The $B C D$ counter is shown in Fig. 3.5 together with its counting sequence. The scale corrections are instantaneous transfers from 0001 to 0111 when adding and 1011 to 1110 when subtracting (least significant bit on the left).

Each binary unit controls a current switch whose output into the virtual earth is proportional to the place weight.

Three of these $B C D$ counters connected in series form a three digit counter. By rernoving the binary unit for the
last significant place the resolution of the counter is changod to 2 in 1000. This involves a modification to the scale correction circuitry for this stage (Fig. 3.6).
3.3.4 Control Unit (Record)

The function of the control unit is to provide a clock of $16 \mathrm{kc} / \mathrm{sec}$, to direct seven out of eight clock pulses to the gate, and to provide the record-command signal for the store unit. It consists of a square-wave oscillator and a three stage binary counter (Fig. 3.7).

The square wave oscillator is formed by intercomnecting two logic units with a resistor-capacitor network. The square wave is not symmetric (i.e. does not reverse at half-period) to limit the scatter of the main-counter input pulses (see below).
3.3.5 Gate and Polarity Control Timing Circuits

In Fig. 3.3 the logic unit 1 releases its inhibit on the gate randomly so that the length of the trigger pulse varies. A very short pulse may marginally trigger the main counter but not the store unit, or vice verse. To prevent this the gate is followed by a Schmidt trigger which drives a monostable. From the second (positive going) edge of the monostable output is derived a trigger pulse sufficient to drive both counter and store unit simultaneously and without ambiguity.

Optimun timing of the trigger pulses relative to the record and reset operations of tine store unit is achieved through reducing the scatter on the triggering by reducing the period ( $t_{0}$ ) when the clock is at zero (maintaining a constant clock frequency) Fig. 3.8. To prevent fore than one trigger pulse per clock pulse a second monostable inhibits further triggering.

Whilst the pulses go through the counter, transient spikes appear on the virtual earth. For example in the count 499 to 500 the counter takes the steps $499,490,400,500$. The unbalance is usually of the same polarity as the step taking place, as in the example. However during scale correction transient reversals of polarity can occur whilst a pulse is in transit through the counter. The response of the polarity control is thexefore damped by introducing a delay in polarity switching. Changes in polarity occur only if still required after the delay (during which the gate is kept closed, Fig. 3.9).
3.3.6 Serializer

The principles of operation of the serializer are to set a three stage binary counter to the bit pattern of the batch and to subsequently recycle this counter to zero - the clock pulses effecting this being the serial pulses required.

The bit pattern from tape is received into buifer bistables which set the three stage counter. The setting of a polarity buffer initiates the cycling of the replay control unit. During the second half cycle of the control unit pulses are gated into the three stage counter (and of course into the main counter) until the three stage counter reaches zero. The main counter polarity control is directed from the polarity input buffers. The three binary digit buffers are reset and locked for the second half cycle of the control unit (Fig. 3.10).

### 3.3.7 Control Unit (Replay)

The function of the control unit is to provide a
clock of $8 \mathrm{kc} / \mathrm{sec}$ which controls the replay sequence through a four stage binary counter, cycling over sixteen clock pulses. This cycle takes place whenever a polarity input bistable is set, but after the delay period necessitated by tape skew.

The control unit times the commands to the memory unit and encoder in the data output channel, and directs the second half-cycle of pulses into the serializer. During the second half-cycle the three bit input buffers are reset and locked. At the end of the cycle the polarity buffers are reset. (Fig. 3.11).

### 3.3.8 Data Output

The data output channel is triggered by changes in batch polarity. The number then in the counter is stored in a memory unit and coded to the input requirements of a commercial encoder. This encoder is triggered by a l ms. pulse derived from the second half-cycle of the control unit. Further triggering of the output channel is inhibited until the 'punch-complete' signal is received from the encoder, when the memory unit is also reset. (Fig. 3.12).

The input code of the encoder is ten parallel lines per digit, a signal voltage being. on the line corresponding to the required digit. This is achieved with the decoding network in Fig. 3.13, the code of the memory unit being that of the counter; binary coded decinal.

The encoder punch command signals are synchronized from the punch unit winich cycles at a constant rate corresponding to 120 char./sec (for the Mestrex). The paper tape records consist of three decimal characters and one terminating character. The averase maximum triggering rate of the output channel is therefore about 23 records per second.

The reconstructed analogue signal is used for monitoring the replay operation. The circuit which generates the analogue signal, and its performance, are given in detail in Fig. 3.14.

The data output channel is also triggered by the manual sampling facility on the system console.
3.4 Input and Output
3.4.1 Recording Mode

Five tracks of binary information are to be recorded with the facility for replay at $1 / 16$ of the recording speed, using a flux sensitive replay system. Three modes of recording are considexed, which involve leaving the tape periodically fully saturated (either positively or negatively) or without any magnetization (zero flux). The first mode is non-return-to-zero ( INRZ) where the tape is always fully saturated in one or other direction. A recorded pulse is stored as an instantaneous change of the saturation polarity (Fig. 3.15a). The second is return-to-saturation (RS) where the tape is saturated at one polarity except where a pulse is recorded, when there is a temporary reversal of polarity (Fig. 3.15b). The third is return-to-zero (RZ) where the tape is unmagnetized except where a pulse is stored, when there is a temporary saturation of the tape in one direction (Fig. 3.15c).

Although a flux-sensitive syster can read flux while the tape is stationary it cannot satisfactorily read signals of long wavelength. NRZ saturation recording or RS recording leaves a flux distribution per cycle corresponding to two bar magnets laid end to end in opposite directions.

As the wavelength increases the field strengths at the centres of the effective magnets decrease because their poles move further apart, leading to the replay signal in Figs. 3.15d and 3.15e.

Thus neither NRZ or RS recording can be used here as the gaps between pulses are indeterminate. The RZ mode in Fis. 3. 15 f is used leaving the tape saturated for fixed lengths or otherwise unmagnetized.
3.4.2 Recoder Limitations

The system records sisnal increments and is therefore an accumulative error system. Loss of information must therefore be highly improbable for acceptable performance. To minimise drop-outs from tape the maximum pulse density tolerable is 200 p.p.i. consistent with computer practice - although systems incorporating re-reading, noise detection, parity and such-like allow higher densities. Secondly without sophisticated 'de-skewing' hardware the recording system is skew-limited (see section on I/O circuits).

The recording clock frequency of $16 \mathrm{kc} / \mathrm{s}$. (note; the tape recording frequency of $2 \mathrm{kc} / \mathrm{s}$. leads to a minimum tape speed of 15 i.p.s.) results in a packing density of $133 \mathrm{p} . \mathrm{p} . \mathrm{i}$.

The incidence of drop-outs at this packing density was examined using counting instruments monitoring the data chamels for both output and input when respectively recording and replaying randon noise. In twenty sample recordings lasting one minute each no drop-outs were registered.
3.4.3 Input/Output Circuits

The I/O circuits are designed according to the tape skew and speed characteristics. The manufacturers specification for tape skew is less than . 002 ins. across four tracks which includes the static skew, due to misalignment of the record and replay heads. This misalignment was virtually removed by the manufacturers who specially machined the relevant surfaces. Tests were carried out at the record and replay speeds measuring the effect of skew as the scatter in pulses on replay after being recorded simultaneously on the five tracks used.

A square wave pulse of duration equal to the interval between the first and last pulse was displeyed on an oscilloscope. The persistence and brightness of the oscilloscope were adjusted so that the distribution of the pulse duration could be ascertained. Deviates of similar brightness enclosing most of the distribution were 0.2 and 0.6 msec . Assuming a Rayleigh distribution, a very safe estimate of the standard deviation is 0.4 msec . (corresponding to $10 \%$ and $90 \%$ confidence limits of 0.18 and 0.86 msec . respectively. The maximum scatter the input circuits can accept is $2.5 \mathrm{msec} .$, corresponding to a probability of exceeding of 1 in $3.3 \times 10^{-9}$, or an expectation of one in about 40 hours of recording time. This figure is, of course, very sensitive to the estimate of the stanaro deviation, but this estimate was made extremely conservatively.

To minimise the effect of skew the polarity tracks are positioned on either side of the three digit tracks so that a batch of pulses can occupy a maximum of four adjacent tracks.

Both the mains frequency stability and the wow and flutter characteristics of the tape deck affect the tape speed.

The recorder was powered by a static inverter. Continuous monitoring of the frequency of the inverter (which alters as the batteries drain) enables it to be maintained to well within $2 \mathrm{c} / \mathrm{s}$. The wow and flutter specification for the recording and replay speeds used are $0.25 \%$ r.m.s. and $0.5 \%$ r.m.s. respectively, making an aggregate effect of $\pm 0.75 \% \mathrm{~m} . \mathrm{m} . \mathrm{s}$.

In Fig. 3.16 is shown the timing of the replay sequence. It is evident from the figure that the nominal 8 ms. replay sequence cannot be reduced to below 4.5 ms . For example a permitted variation of $\pm 10 \mathrm{c} / \mathrm{s}$ in mains frequency could lead in the worst case to an interval of 5.3 ms . between batches. Superimposed on this could be 0.8 ms . of wow and flutter, i.e. thirteen standard deviations (or for a Gaussian distribution a probability of exceeding of I in $10^{38}$ ). Clearly there is more than adequate margin for both static and dynamic variation in tape speed.
3.5 Performance of D.R.S.

The basis characteristics of the recording system are an accuracy of $\pm 2.5 \mathrm{mv}$. in a signal range of 0 to I ., and a maximum average following rate of 28 volts $/ \mathrm{sec}$. in 2 mv . steps. The input signal may be either . c. coupled or a. c. coupled to a d. c. bias of half-full scale.

Some characteristics of the system have a reduced effect on its performance when considered only as a recorder of maxima and minima. The effect of recording in batches every eighth clock pulse is negligible in the region of maxima and minima, when the signal is changing least rapidly. In addition the system is more accurate at maxima ( +2.5 , - Onv) and minima ( +0 , -2.5 mv .). By adding or subtracting 1.25 mv . when processing the results the accuracy becomes $\pm 1.25 \mathrm{mv}$.

The performance of the system with respect to its accumulative error characteristic is next considered. Unlike analogue recording data introduced or lost through noise on h.t. lines, drop-outs on tape, etc. will affect the remainder of the record. The tape recorder performed excellently in drop-out tests. (See Appendix l.) Adequate care was always taken in tape handling and deck cleanliness. Noise suppressors are incorporated into the mains supply.

The performance was measured as the drift of d.c. bias during three series each of twelve recordings. The recorded bias should be accurate to within $\pm 2 \mathrm{mv}$.) Two series were recordings of sine-waves and lasted 80 and 200 secs per recording respectively. In the third series filtered random noise was recorded for 80 secs per recording. The whole duration of all recordings was about 70 minutes. The d.c. values for each series lay within the tolerance of the system.

Finally the effect of the rate of following is shown in Fig. 3.17. This graph gives the maximum sine-wave amplitude against frequency for three cases:

1. Complete following of the sine-wave to within the maximum accuracy of the system (i.e. maximum - gradient less than $28 \mathrm{v} / \mathrm{sec}$.
2. Following of the sine-wave such that only the maximum and minimum values are recorded accurately. The solution to this problem was found numerically (using the fact that it lies between the complete following case and the frequency for which the system can generate a saw-tooth waveform of the same amplitude (Fig. 3.18).
3. From actual measurements the following of the
sine-wave to give a peak out-of-balance of
$\pm 20 \mathrm{mv}$. on the virtual earth. (This is a very artificial criterion and is used only because of the difficulties involved in measuring cases $l$ and 2 accurately).

Case 3. demonstrates that the system is at least $12 \frac{1}{2} \%$ slower than Case l. suggests. This is due to the delay in the polarity circuit, particularly during scale correction when transients occuring at the virtual earth cause the polarity circuit to block the gate for $150 \mu \mathrm{~s}$. Whether or not the inhibit occurs during scale correction depends on the overall out-of-balance of the virtual earth. For low unbalances, most transients cause such an inhibit. For high unbalances most of the transients will not cause the virtual earth unbalance to change sign. Thus for the $\pm 20 \mathrm{mv}$. unbalance (Case 3) the following rate is cut to iess then 24.5 volts/sec. (for which complete following coincides with case 3). However for the larger unbalances occuring in Case 2 the effective clock frequency will approach 28 volts/sec. so that Case 2 is more representative of the system's actual performance.

analogue circuit
—— digital circuit

-     -         - control circuit

FIG. 3.1 RECORD SYSTEM


FIG. 3.3 INPUT AMPIIFIERS. GATE AND POLARITY CONTROL


FIG. 3.4 STORE UNIT


FIG. 3.5 BINARY CODED DECIMAL COUNTER



FIG. 3.7 CONTROL UNIT (RECORD)
$\square \square \rightarrow \mid$

FIG. 3.3 CLOCK WAVEFORM







Circuit Performance

$$
\left.\begin{array}{rl}
v_{0} & =-G_{a}(-e) \\
i & =\frac{v_{0} \cdot e}{R} \div \frac{e}{R_{a}} \\
\therefore v_{0} & =\frac{i \cdot R}{1+\frac{1}{G_{a}}\left(1+\frac{R}{R_{a}}\right)} \\
R=500 \Omega \\
\begin{array}{l}
R_{a} \\
G_{a}
\end{array}=200 \mathrm{~K} \Omega
\end{array}\right\} \quad \therefore v_{0}=0.9995 . i R \quad \$
$$

$$
R=500 \Omega
$$

$$
R_{a}=200 \mathrm{k} \Omega
$$



FIG. 3.15 RECORDING MODES


FIG. 3.16 TIMING OF REPLAY SEQUENCE


FIG. 3.17 PERFORMANCE OF D.R.S.

# upper and lover limiting cases <br>  <br> $v=A \sin \omega t: \frac{d V}{d t}=A \omega \cos \omega t: \quad R=A \omega \cos \omega t_{1}$ <br> i.e. $\quad t_{1}=\frac{1}{\omega} \cos ^{-1} \frac{R}{A \omega}$ : Now $\left(t_{1}+\frac{\pi}{2 \omega}\right) R=A\left(\sin \omega t_{1}+1\right)$ <br> But $\sin \omega t_{i}=\sqrt{\omega^{2} A^{2}-R^{2}} / \omega A$ <br> i.e. $\cos ^{-1} \frac{R}{A \omega}=\int\left[\left(\frac{A}{R \omega}\right)-1\right]+\frac{A \omega}{R}-\frac{\pi}{2}$ <br> i.e. $\quad \frac{R}{A \omega}=\sin \left[\sqrt{\left(\frac{A \omega}{R}\right)^{2}-1}+\frac{A \omega}{R}\right]$ 

Solution must lie between $R=A \omega$ (complete following) and $R=4 A f$, ( $u=2 \pi f$ ), ramp of same $A$ and $\omega$

$$
\text { i.e. between } \left.\left.\frac{\pi}{2}\right\rangle \frac{A \omega}{R}\right\rangle_{1}
$$

Solution ( found numerically ) is

$$
\frac{A \omega}{R}=0.4198 \pi
$$

## 4 RECORDINGS

### 4.1 Introduction

Two series of recordings are made. In the first filtered random noise is used as the signal source. In the second series recordings are made of car suspension loads for a wider bandwidth case. The results are presented and discussed in Chapter 5.

For each recording the mean value is half fullscal, (corresponding to a digital value of 500), and the recording levels adjusted to give a signol standard deviation of digital value l00. Thus the crest factor (the ratio between maximum recorded peak and the standard deviation) is five. Assuming a Rayleigh distribution of peaks the probability of the signal having a maxima (or minima) outside the range of the system is therefore 1 in 270,000 . For irrecularity factors less than unity a signal excursion outside the range is even less likely.
4.2 Filtered Random Noise

To match the capabilities of the recording system 'white.' noise over a range of approximately $2-25 \mathrm{c} / \mathrm{s}$ is required. As no such noise generator was available at the time, one was constructed using a miniature neon light as the noise source. A band-pass filter was used to limit the bandwidth of the signal so as to obtain the required values of irregularity factor.

Two sets of recordings were made with filter settings of 2-9 and. $2-25 \mathrm{c} / \mathrm{s}$ respectively. The characteristics of the filter for these two settings are shown in Fig. 4.1 and tabulated in Table 4.1. The values of irregularity factor
obtained were 0.792 and 0.732 which are lower than the corresponding figures for an ideal filter ( 0.835 and 0.728) due to the actual characteristics of the band-pass filter. However these figures obtained do agree remarkably well with the estimates made from the filter characteristics themselves (as discussed in Chapter 5).

### 4.3 Loads in car suspension

For these recordings a 'Mini' belonging to the Advanced School of Automobile Engineering (A.S.A.E.) of Cran:ield College of Aeronautics was used. The car instrumentation consisted of strainmgauged load cells built into various suspension points. Fror the recordings the car was linked by cable to another vehicle containing the recording system. A view of this instrumentation vehicle showing the digital recording system and other ancillary equipment is given in Fig. 4.2.

The tests were performed on the Belgian pavé surface of the inotor Industries Research Association (M.I.R.A.) proving ground at Juneaton. The vehicle containing the recording equipment accompanied the test vehicle on a smooth circuit lying outside the pavé circuit. A view of the pavé surface is show in Fig. 4.3. The 'Mini' was driven on the pavé at a constant speed of $20 \mathrm{~m} . \mathrm{p} . \mathrm{h}$.

The recording system was powered by a static inverter. A transistorized dynamic strain gauge bridge amplifying unit (Peckel T-630) linked the load cell to the recording system. A complete description of the load cells on the 'rini' is given in ref. 7. The data presented here was recorded from the front middle-longitudinal (FIT) load cell. This cell detects fore and aft loads between the car
body and the sub-frame of the front suspension (Fig. 4.4).

The calibration of the load cell when used with the A.S.A.E. standard instrumentation (a 'SEI' unit) was supplied by A.S.A.E. ( $6 \mathrm{lb} / \mathrm{SEL}$ unit deflection a.t SEL attenuation of 24 dB ). Also supplied was the SEL calibration against strain-gauge bridge out-of-balance ( 890 SEL units / $1 \%$ unbalance at attenuation of 21 dB ). Thus the variation of load against bridge \% unbalance is known. As an independent check on this calibration a number of static loads were placed on the vehicle and the simultaneous SEL output and $T-530$ output were recorded (Fig. 4.5). The $T-630$ was then separately calibrated against bridge out-of-balance and this confirmed the calibrations supplied by A.S.A.E. to within $1 \%$ (Fig. 4.6). The T-630 output calibration is finally $1.48 \mathrm{lb} / \mathrm{mV}$. (Pig. 4.6).

For these recordings a two stage $R-C$ filter (corner frequency $160 \mathrm{c} / \mathrm{s}$ ) was incorporated into the input to improve the suppression of the Peekel $\mathbb{T}-630$ carrier frequency ( $1 \mathrm{kc} / \mathrm{s}$ ). The filter resistance ( $500 \Omega$ ) doubles the input resistance of the recording system and so the detection thresholds of the main amplifier were reset to $\pm 5 \mathrm{mV}$, the full scale range being 2 volts. Thus the calibration for these recordings is $2.96 \mathrm{lb} /$ digital unit.

The value of irregularity factor obtained was 0.460. The digital value of r.m.s. load on the first pass of the data was 91.42 , or 270.6 lbs . As explained later the data was then factored by 1.094 to give a r.m.s. digital value of 100 , so that the histogram grids and conditional range counts are based on the r.m.s. level. Thus the calibration for the data presented here is 2.706 lbs/digital unit.



FIG. 4.2 THE INSTRUMENTATION VEHICLE


FIG. 4.3 VIEW OF PAVE SURFACE


FIG.4.4 A.S.A.E MINI


FIG. 4.5 T-630 vs S.EA. CALIERATION


## CALIBRATION CHECK

from figs. 4.5 \& 4.6 :
with attenuation of 12 dB :
S.E.L. deflection $/ \%$ unbalance $=\frac{5100}{16}=318.5$
$\therefore$ with attenuation of 21 dB :
S.EL deflection / \% unbalance

$$
=318.5 \times 2.818=898
$$

i.e. checks calibration supplied (890) to $1 \%$
hence Peckel-T630 O/P calibration is

$$
\begin{aligned}
\text { load }(\mathrm{lb}) / \mathrm{mV} & =6.0 \times 890 \times 1.414 / 5100 \\
& =1.48
\end{aligned}
$$

MABIE 4.1 BATD PASS RILTER CHARACTERISTICS
Setting: 2-25c/s

| $c / s$ | 2 | 4 | 6 | 8 | 10 | 12 | 14 | 16 | 18 | 20 |
| :--- | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| $d B$ | 26 | 9.6 | 5.2 | 2.9 | 1.6 | 0.8 | 0.2 | 0.0 | 0.2 | 1.0 |
| $c / s$ | 22 | 24 | 26 | 28 | 30 | 32 | 34 | 36 | 38 | 40 |
| $d B$ | 2.3 | 3.7 | 5.0 | 6.3 | 7.9 | 9.6 | 11.4 | 12.8 | 14.4 | 16 |
| $\mathrm{c} / \mathrm{s}$ | 41.5 | 43.5 | 45.5 | 47.5 | 49.5 | 51.5 | 53.5 | 55.5 | 57.5 | 59.5 |
| $d B$ | 17.2 | 18.6 | 20.5 | 22 | 23.6 | 25.1 | 26.4 | 28.2 | 29.5 | 31 |
| $c / s$ | 63 | 68 | 73 | 78 | 83 |  |  |  |  |  |
| $d B$ | 33.2 | 36.6 | 39.6 | 42.6 | 45.3 |  |  |  |  |  |

Setting: 2-9c/s

| $c / s$ | 2 | 4 | 6 | 8 | 10 | 11 | 12 | 13 | 14 | .15 |
| :--- | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| $d B$ | 26 | -9.6 | 4.0 | 1.1 | 0.9 | 1.6 | 2.7 | 4.0 | 5.4 | 6.9 |
| $c / s$ | 16 | 17 | 18 | 19 | 20 | 22 | 24 | 26 | 28 | 30 |
| $d B$ | 8.7 | 10.4 | 12 | 13.6 | 15.3 | 18.4 | 21.4 | 24.5 | 27.5 | 30.5 |
| $c / s$ | 31.5 | 33.5 | 35.5 | 37.5 | 39.5 |  |  |  |  |  |
| $d B$ | 32.8 | 35.3 | 38 | 40.6 | 42.6 |  |  |  |  |  |

## 5 RESULTS

5.1 Analysis of Recordings

The data from each recording series was analysed using the University Atlas Computer. Histograms were constructed using the following counting methods - Peak, Hean-Crossing-Peak and Range-Mean counts. The Range-ilean counts were performed with five different minimum range values. The counting methods are described in Appendix 2, and the data analysis programmes given in Appendix 3.

In order to relate the minimum ranges of the conditional range counts to the standard deviation of the signal the data analysis was performed twice. On the first pass the standard deviation of the signal was obtained and on the second the data was factored (relative to the overall mean value) to obtain a standard deviation of $10 \%$ full scale. In the recordings themselves the recording levels were adjusted to give approximately this standard deviation and in the data analysis only very small adjustments were necessary. The standard deviation of $10 \%$ full scale means that the numerical grid is also related to the signal standard deviation.

The data itself consists of successive maxima and minima. The continuous signal standard deviation has therefore to be calculated from the peal (discrete value) standard deviation using the relationship given in Chapter 2.

The tabulations of the data for the counting methods are given in Appendix 5, together with the data corresponding to the probability density distributions referred to in this Chapter. The cumulative distributions corresponding to each counting method are shom in Fig. 5.15.3 for the three recordings (Moise 2-9c/s; Noise 2-25c/s;
car suspension loads). The computed values of irregularity factor for the three cases are $0.792 ; 0.732 ; 0.460$ respectively.

As expected the counting methods lead to essentially similar results for the narrowest bandwidth case (I.F. $=0.792$ ). As the bandwidth widens the counting methods diverge. The peak based methods give essentially different results to the range based methods.

In the random noise analysis it has been assumed that the instantaneous values of the signal have a Gaussian distribution. The distribution of peaks may be used to ascertain whether or not the distribution of the signal is in fact Gaussian. In Figs. 5.4-5.6 are presented the probability density distributions of peaks. The measured distributions agree closely with the expected (theoretical) distributions based on the measured values of irregularity factor. This implies of course that the signal is indeed Gaussian.

Next, comparisons are made between measured values of irregularity factor and the expected (theoretical) values. The expected irregularity factor is given by

$$
\mathrm{n}=\frac{\int_{0}^{\infty} \mathrm{E}(\sigma) \cdot \sigma^{-2} \cdot d \sigma}{\left[\int_{0}^{\infty} \mathrm{E}(\sigma) d \sigma \cdot \int_{0}^{\infty} E(\sigma) \cdot \sigma^{4} \cdot d \sigma\right]^{\frac{1}{2}}}
$$

where $E(\sigma)$ is the energy spectrum of the signal, $\sigma$ being frequency (see chapter 2). The noise generator is assumed to produce 'white' noise ( $\mathrm{E}_{1}\left(\sigma^{\sigma}\right)=$ constant) so that filtering with a filter of receptance $A(i \omega)$ produces a signal with a. new energy spectrum $E_{2}(\sigma)$ given by

$$
E_{2}(\sigma)=|A(i \omega)|^{2} \cdot E_{1}(\sigma)
$$

Since only the modulus of the receptance is used the term $|A(i \omega)|^{2}$ is obtained directly from the filters response curve (Fig. 4.1). The filter curve is obtained over sufficient bandwidth so that the integrations can be completed accurately.

The expected value of I.F. for the 2-9 filter setting is 0.803 (compared to the measured value of 0.792). For the $2-25$ setting it is 0.742 (measured 0.732). Both values are within $1.5 \%$ which shows that if the spectral density is known sufficiently well, very accurate estimates can be made of the irregularity factor. In the case of the car suspension loads the spectrum presented in ref. 8 is not' defined over sufficient bandwidth to perform the integrations

In Figs. 5.7-5.9 are presented the mean-erossing peak distributions. For comparison these distributions are related to the total peak count. In each case the two distributions converge at high peaks (as expected) and also at very small peaks. In the limiting case all amplitudes giving peaks at or near the mean also cross the mean.

In Figs. 5.10-5.12 are presented the range distributions for both the conditional and unconditional counting methods. For comparison the distribution of peaks is also shown in each case. The unconditional distributions are discussed in detail in Chapter 6. The variation of the range counts with the different minimum ranges is clearly evident, even in the nerrowest bandwidth case. The great difference between the unconditional range counts and the corresponding peak count is striking. As expected the distributions converge as the irregularity factor approaches unity.

Figs. 5.13 and 5.14 show the overall distributions
of mean values, where the probability densities are related to the standard deviation of the continuous signal. These distributions are shown to be normal in Chapter 6. In addition the distributions of means for each value of range are also normal as pointed out by schijve (ref. 6). As expected the smaller the bandwidth the closer the individual mean volues are grouped around the overall mean (Fig. 5.15)

It is noteworthy that for the widest bandwidth case (I.F. $=0.46$ ) $50 \%$ of all ranges have mean values lying outside the limits $\pm 0.5 \sigma$, and $20 \%$ outside $\pm \sigma$. In the other two cases the corresponding figures are 38 and IO\% (for I.F. $=0.73$ ) and $28 \div 7 \%$ (for I.F. $=0.79$ ). (Thus the established procedure of neglecting mean value variations in programme fatigue testing based directly or indirectly on raige counts is questionable for wider spectrums).
5.2 Linear Cumulative Damage Estimates

In order to see quantitavely what effects the irregularity factor may have on estimates of fatigue life, a range of typical S-W curves have been selected. Linear cumulative damage estimates have been made using the data from each of the various counting methods. Such estimates are simple to perfom as such form the basis of many design studies, particularly in early project stages. The author considers it relevant to point out further shortcomings in this practice.

Very simple $S-1 y$ curves have been used ( $S=a N^{b}$ ) which have been derived from R.Ae. Soc. data sheets (ref. 7). The derivation of the curves is given in detail in appendix 4, where is also explained the relationship between the numerical grid and the stress scale. Because of the particular type
of $\mathrm{S}-\mathbb{N}$ curve selected, the comparisons between estimates fron differing counting methods depend only on the slopes of the $\mathrm{R}^{5-1 N}$ curves (to within $10 \%$ ) - see Appendix 4.

Three $\operatorname{lot}^{\operatorname{lot}} \log _{\mathrm{N}}$ curve slopes have been used; a high-slope (corresponding to a lug with interference), a medium-slope (lug without interference), and low-slope (al-alloy in bending). In making the comparisons between fatigue life estimates the following points are considered;
(i) the effect of mean stress variations (Range c.f. Range/Nean)
(ii) the effect of omitting small amplitudes in the conditional range counts (Range c.f. Range ( $P>0$ ))
(iii) the difference between range based counts and peak based counts (Range c.f. In-C-P)
(iv) the difference between Peak and H-C-P

Finally since all the data is derived from signals with the same standard deviation it is interesting to see if they lead to similar cumulative damage rates. The estimated lives based on $m-c-p$ counts and referred to the same number of zerocrossings is evaluated. The m-c-p method is used here because it is one of the most comnonly used methods (see for example ref. l2).

Table 5.1 tabulates the above comparisons in the form of ratios between each pair of lives being compared.

The difference between Range and Range-ifean estimates is low, ranging from ratios of 1.0 to 1.7 . For the higher slope or ${ }^{-2}$ curve the difference is in fact completely insignificant so that measurements of mean values in addition to ranges is not justifiable for range based estimates.

The onission of small ranges when making a range count has a surprisingly small effect. With the low-slope curve and the widest bandwidth the life-ratio approaches 1.6 when the smallest detected range ( $P$ ) is equal to the standard deviation. For the more typical case of $P=\frac{1}{4} \sigma$ the effect is completely insignificant even for the widest bandwidth.

Another surprising effect is the vast difference between range based and peak based estimates. The life ratios between $m-c-p$ and range results vary up to 17 depending on slope and bandwidth.

The comparison of peak and m-c-p estimates leads to completely insignificant differences. It appears that the extra complication in obtaining m-c-p results for life estimates is unwarranted - although in programe testing, the testing tine is reduced.

The estimated lives based on m-c-p data (referced to the same number of zero crossings) are given in Table 5.2. The figures can be assumed to be measures of life on an arbitrary scale when exposed to each signal in tum. For each $S-N$ curve the lives are very approximately the same.









FIG. 5.8 PROBABILITY DENSITY DISTRIBUTION OF M.C. PEAKS RELATIVE TO
PEAK DISTRIBUTION


FIG. 5.9 PROBABILITY DENSITY DISTRIBUTION OF M.C. PEAKS RELATIVE






FIG. 5.14 PROBABILITY DENSITY DISTRIBUTION FOR MEAN VALUES ( $\sigma=$ RMS. FUNCTION)


FIG. 5.15 EFFECT OF IRREGULARITY FACTOR ON RM.S VALUE OF MEANS

TABIE 5.1 COMPARATIVE IINEAR CUGUATIVE DAMAGE ESTIMATEG

Ratio Case I.F.
SN2
SN2
SN3 Low-slope Medium-sl Hi-slope

| $I_{r} / I_{r m}$ | $C A R$ | 0.460 | 1.72 | 1.23 | 1.05 |
| :--- | :--- | :--- | :--- | :--- | :--- |
|  | $2-25$ | 0.732 | 1.59 | 1.18 | 1.04 |
|  | $2-9$ | 0.792 | 1.18 | 1.09 | 1.02 |


| $I_{\mathrm{r}} / I_{\text {mcp }}$ CAR | 0.460 | 17.0 | 5.47 | 1.94 |
| :---: | :---: | :---: | :---: | :---: |
| $2-25$ | 0.732 | 5.19 | 2.70 | 1.47 |
| $\therefore$ | $2-9$ | 0.792 | 2.46 | 1.84 |


| $I_{r m} / I_{\text {mopCAR }}$ | 0.460 | 9.90 | 4.44 | 1.95 |
| ---: | ---: | ---: | ---: | ---: |
| $2-25$ | 0.732 | 3.37 | 2.29 | 1.41 |
| $2-9$ | 0.792 | 2.09 | 1.69 | 1.25 |


| $I_{\mathrm{mcp}} / I_{\mathrm{p} \mathrm{CAR}}$ | 0.460 | 1.08 | 1.13 | 1.29 |
| ---: | :--- | :--- | :--- | :--- |
| $2-25$ | 0.732 | 1.0 | 1.0 | 1.0 |
| $2-9$ | 0.792 | 1.0 | 1.0 | 1.0 |

RANGE; $\quad I_{p=0} / L_{p=0}$

| $p=.25 \sigma^{\circ}$ | CAR | 0.460 | 1.16 | 1.14 | 1.12 |
| :--- | :--- | :--- | :--- | :--- | :--- |
| $p=.500^{\circ}$ | CAR | 0.460 | 1.30 | 1.25 | 1.20 |
| $p=.750^{\circ}$ | CAR | 0.460 | 1.35 | 1.32 | 1.25 |
| $p=1.00^{\circ}$ | CAR | 0.460 | 1.56 | 1.44 | 1.30 |

NOTE:

$$
I_{r} / I_{r m}=\frac{\text { Life based on RANGE }}{\text { Life based or RAVGUEAT }} \text {, ETC. }
$$

and $p=$ PEAK
. MCD = REAT-CROSSIITG PEAK

TABLE 5.2 EFFECT OF IRREGULARITY FACTOR ON MOP ESTIMATES

| CASE | CAR | $2-25$ | $2-9$ |  |
| :---: | :---: | :---: | :---: | :---: |
|  | I.F. | 0.460 | 0.732 | 0.792 |
| Effective <br> Lives | High slope | 9662.0 | 12711. | 11279. |
|  | Mow. slope | 1461.0 | 1537.0 | 1404.0 |

To find the empirical distribution of ranges three independent noise sources have been used. These are (i) filtered random noise (two cases), (ii) car suspension loads (one case) and (iii) filtered random numbers (data from Ieybold and Namman - five cases - refs. 9 and 10).

It is the author's submission that the distribution of ranges has not been suggested previously because results have been obscured by the resolution of the counting systems. As is well known range counts are highly sensitive to the smallest range detectable. Figs. 6.1-6.3 show the effect on the range distribution of ignoring $\frac{1}{2}$-ranges of size $0.125 \sigma$ and $0.250^{\circ}$ (for the three recordings of the author's) compared with that for 'all' ranges ( $0.0250^{\circ}$ ). In each case the new distributions have the characteristic shape of the Rayleigh distribution, a fact which may have led to the belief that for fairly high values of irregularity factor (eg. above 0.8 ) the range distribution was Rayleigh (the distribution for I.F. of unity). All recording systems have, of course, a finite resolution - in this case (of $0.025 \sigma$ ), it is assumed that relatively few ranges axe omitted.

To find the range distribution the distribution of all ranges (regardless of mean value) is first considered. Following this an attempt is mede to correlate the distributions for each mean value with this overall distribution. The grounds for attempting this are that the distribution of mean values has already been found (by Schijve ref.6) to be nomal, independent of the value of range (that is constant standard deviation for all values of range).

A difficulty in finding probability density distributions is that the data consists of histograms,
that is areas of columns under the probability density curve. Where the gradations of range are reasonably small, distributions can be constructed accurately even in the region of maxima and minima, by bearing in mind that it is the area under the curve that is known. It has to be assuxaed there are no points of inflexion. It is impossible, of course, to assess the distribution in the immediate neighbourhood. of the vertical axis.

However, these results are based on samples and it would be unrealistic to assume the distributions are the true ones. Hence for the purpose oi illustrating the following discussion the points have merely been joined by smooth curves, truncated at the data point nearest the vertical axis. The histograms have been tabulated separately. Note that Naumann substitutes a new case $D$ for Leybold's original case D. This new case is referred to as case E. Fig. 6.4 gives the distributions for the eight cases of irregularity factor. Clearly case $A$ and $E$ do not fit into the general pattern, having the familiar Rayleigh characteristic shape. The remainder of the curves appear to be part of the same family (Fig. 6.10).

All the Leybold/Naumann data is based on filtered random numbers, where the numbers are considered as equally tined samples of a stationaxy randon process. However, the criteria used by leybold in selecting his random number generator was to choose a generator which gave low correlation between numbers. Clearly this is equivalent to making the sample time-interval longer. In each case the data is based on the same set of approximately 160,000 random numbers. Linear filtering is usea to obtain various 'bandwidth' cases (or differeat spectral aensity shapes). In the five cases the number of maxima and minima obtained are; 52827, 36561, 43850, 44765 and 106426

This amounts to having sampled the random process at the following average number of times per $\frac{1}{2}$-cycle; 3.0, 4.4, 3.7, 3.6, 1.5.

It is submitted that for cases $A$ and $E$ the sampling interval is too long and that the data for these cases is unrepresentative of the assumed spectral shape. This submission can be checked by examining whether the peak distributions lie close to the theoretical distributions. Figs. 6.5-6.9 give the theoretical distribution of peaks for each case together with the corresponding experimental points. For cases $B$ and $C$ the agreement is remarkably good. For the cases $A, D$ and $E$ the agreement is poor, especially case $E$. The distribution for case $E$ (and to a lesser extent cases $A$ and $D$ ) lies completely outside the family of classical distributions for all values of irregularity factor. Thus this certainly supports the hypothesis that the data for cases $A$ and $E$ is unrepresentative, but also indicates that the data for case $D$ must be marginal.

The distributions for the remaining six cases are shown in Fig. 6.10. It is now immediately apparent that these curves lie in the same family. There is a continuous trend away from the Rayleigh distribution (I.F. $=1.0$ ) with decreasing irregularity factor. There is an increasing proportion of smaller ranges and decreasing proportion of larger ranges as the spectrum widens. The proportion in the neighbourhood of $.75 \sigma$ stays roughly constant until the I.F. gets below 0.5. Note that for case $D$ the range distribution departs very slightly from the trend of the remainder of the family, having a relatively higher peak. This clearly indicates that the equivalent sampling interval for case $D$ is marginal.

Another significant feature of the family of distributions is that the data (as stated previously) is derived from completely different sources, indicating that the overall distribution of ranges (like that of peaks) may depend only on the spectrum width parameter (or irregularity factor).

Examined next is the likelihood of the distribution of ranges for each value of mean being independent of the mean value. In Figs. 6.11-6.16 is given the distributions for each value of mean compared with the overall distribution for each case. The departure from the overall distribution is fairly small in all cases. Note here that the data for the relatively higher values of mean (depending on I.T.) is omitted on the grounds that the samples for these are too small. However the data presented here in the figures includes about 95\% of all ranges - that is the ignored data is about one twentieth of the total.

Finally the distribution of the mean values is examined. As found previously by Schijve (ref. 6) the distributions are nomal for the six cases (Figs. 6.17-6.19) The variation of the standard deviation of the means with irregularity factor is shown in Fig. 6.20.

In conclusion it is submitted that an adequate model for the distribution of ranges is that represented by the family given in Fig. 6.10, with the distributions independent of mean value and dependent only on the irregularity factor. The proportion of ranges at each mean is known from the normality of the distribution of mean values, using the empirical relation between mean r.m.s. and the function r.m.s. given in Fig . 6.20.


FIG. 6.1 EFFECT ON RANGE DISTRIBUTION OF IGNORING $1 / 2$-RANGES LESS THAN $P(P=\cdot 025 \sigma, \cdot 125 \sigma, \cdot 25 \sigma)$.


FIG. 6.2 EFFECT ON RANGE DISTRIBUTION OF IGNORING 1/2-RANGES LESS THAN $P(P=.025 \sigma, 125 \sigma,-25 \sigma)$.


FIG. 6.3 EFFECT ON RANGE DISTRIBUTION OF IGNORING $\frac{1}{2}$ RANGES LESS THAN $P(P=0.025 \sigma, 0.125 \sigma, 0.250)$.


FIG. 6.4 RANGE DISTRIBUTIONS - ALL CASES


FIG. 6.5 PEAK DISTRIBUTION CURVE: A $n=0.74$ :


FIG. 6.6 PEAK DISTRIBUTION CURVE : B $n=0.53$


FIG. 6.7 PEAK DISTEIBUTION CURVE $C \quad n=0.88$


FIG. 6.8 PEAK DISTRIBUTION CURVE D $n=0.89$


FIG. 6.9 PEAK DISTRIBUTION CURVE $E \quad n=0.68$.


FIG. S. 10 RANGE DISTRIBUTIONS - ACCEPTED CASES


This includes $95.6 \%$ of all ranges.

FIG. 6.11 DISTRIBUTION OF RANGES FOR EACH MEAN VALUE




FIG. 6.12 DISTRIEUTION OF RANGES FOR EACH -MEAN VALUE.


FIG. 6.13 DISTRIBUTION OF RANGES FOR EACH MEAN VALUE.


FIG. 6.14 DISTRIBUTION OF RANGES FOR EACH MEAN VALUE


FIG. 6.15 DISTRIBUTION OF RANGES FOR EACH MEAN VALUE


FIG. 6.16 DISTRIBUTION OF RANGES FOR EACH MEAN VALUE


FIG. 6.17 DISTRIBUTION OF MEANS


FIG. 6.18 DISTRIBUTION OF MEANS LEYEOLD - CURVE B.


FIG. 6.19 DISTRIBUTION OF MEANS LEYBOLD CURVES C \& D.


FIG. 6.20 VARIATION OF R.M.S. MEANS WITH. I.F.

| SOURCE | $2-9$ | $2-25$ | CAR | $A$ | $B$ | $C$ | $D$ | $E$ |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| I.F. | .792 | .732 | .460 | .741 | .534 | .860 | .885 | .680 |
| $\sigma_{m} / \sigma_{f}$ | .492 | .590 | .804 | .604 | .741 | .350 | .262 | .654. |

The irregularity factor describes the distribution of peaks, and has been demonstrated to be the major parameter in a model of the distribution of ranges. It is an obvious step to investigate its role in random fatigue testing. Random testing has been performed at several research establishments, and a comprehensive bibliography can be found in ref. 3. There follows a brief investigation into the published work to establish any consistent trends in the effect of the irregularity factor.

The fatigue process incorporates several mechanistic
phases. Principally these are the crack initiation phase and the crack propagation phase. The initiation phase may be subdivided into a shear crack initiation phase and shear crack propagation phase, leading to the direct (tensile) stress crack propagation. Thus it may seem unlikely that any one general trend of the effect of irregularity factor will exist.

However the fatigue condition in aircraft is usually high values of stress concentration and relatively high mean loads, so that the fatigue process will be essentially direct stress crack propagation. Secondly random testing may interact the mechanistic phases. For an example, Forsyth's terminology of Stage I (slip plane growth) and Stage II (propagation on plane perpendicular to maximum tensile stress ref. 14) will be used. The transition from Stage I growth to Stage II growth may possibly be delayed owing to periodic low stress cycling during the random sequence which may cause sections of an incomplete crack front to revert to stage $I$ growth, thus reducing the effectiveness of the crack front
as a stress raiser to subsequent higher tensile loads. Such an effect could take place over a range of values of r.m.s. loading.

Another example of the interaction effect could be explained in terms of residual stresses, where earlier higher loads in the random sequence cause residual stresses which reduce the effect of subsequent lower stress cycles. A further example of random loading effects is that brittle striations are very often found in service loading failures, whereas in constant amplitude or programe tests they are not as common (Jacoby, ref. 16). This indicates a certain uniqueness about random loading, as also does the fact that crack propagation rates are highest for random loading tests (Schijve, ref. 17).

In accordance with current practice, fatigue life is considered in the following discussion in terms of counts of zero-crossings as a function of r.m.s. stress. Only worls which includes the testing of similar specimens with at least two different types of random loading is discussed.

Olevenson and Steiner (ref. 13) performed tests on 2024-T4 round bars with a circular notch, with a stress concentration factor oi 2.2 at zero mean stress. Although the tests are classed as axial the stress distribution will clearly not be axial (except for axial symnetry) due to the shape of the specimen. Clevenson and Steiner studied the effect of the spectrum shape on fatigue life and concluded that the spectral shape and irregularity factor heve only a secondary effect. However, two-thirds of their tests were performed with values of irregularity factor between 0.6 and l.O. Only 20\% of the tests were performed with an irregularity factor of less than 0.75 .

Clearly this invites the proposition that spectral shape is irrelevant provided the irregularity factor is approximately constant; that is that the irregularity factor (together with r.m.s. stress) is the controlling parameter in random tests. Their results have therefore been separated into two groups corresponding to irregularity factor bands of 1.0 to 0.8 and 0.8 to 0.6 . In the first group (say irregularity factor $=0.9$ ) are 48 specimens, and in the second group (say irregularity factor $=0.7$ ) are 26 specimens. A linear regression has been performed for both groups to find the $\log \mathrm{S} v \log \mathrm{~N}$ relation.

The two groups regress onto two almost parallel, but distinctly separate, straight lines (fig. 7.1). The lives for the wider spectrun group (irregularity factor $=0.7$ ) are systematically less then those for the narrower spectrum group (irregularity factor $=0.9$ ) by a factor of about 1.8 . The co-efficients of linear correlation for the two groups are 0.98 (larger group, irresularity factor $=0.9$ ) and 0.95 (smaller group, irregularity factor $=0.7$ ) compared with a value of 0.96 in a single regression of 311 the results. Thus the improvement in the correlation of the larger group is only barely significant. The correlation for the smaller group is still good, but note that there is a greater amount of scatter in the results for the smaller group (irregularity factor $=0.7$ ) at the lower stress levels. The statistical significance of the groupings can be further examined from the plots of the results. The line corresponding to the linear regression of the larger group ( 0.9 ) corresponds to the $85 \%$ confidence line for the smaller group ( 0.7 ) and the line for the smaller group ( 0.7 ) corresponds to the $15 \%$ confidence line for the larger group (0.9).

This does not necessexily mean that the effect of
irregularity factor is negligible, as the authors concluded. It surely indicates a need for further empirical data before making an assessment of the effect of irregularity factor, more tests particularly in the wider spectrum group are really necessary. Secondly it may be possible that still lower values of irregularity factor may have an appreciable effect.

Swanson (ref. Il) performed tests on axially loaded 2024-T4 unnotched bars with a mean stress of 16000 psi. These tests were performed using both a single degree of freedom system and a two degree of freedom system to obtain two spectral shapes. The values of irregularity factor obtained were 0.96 and 0.57 . Fig. 7.2 shows the effect of r.m.s. stress on endurance measured in zerocrossings. For lives below $6.10^{5}$ the wider spectrum has an appreciably more severe effect, the spectrums becoming equally severe at that life. These results have therefore a degree of consistency with the more liberal interpretation of Clevenson and Steiner's results in spite of the fact that the testing conditions (mean stress and notching of specimen) are so difierent.
. Naumann (ref. 9) performs individual cycle randomization tests on 2024-T3 notched sheet specimens with a stressmoncentration factor of 4 and mean stress of 17.4 ksi . The specimens are loaded to successive maximum and minimum values of the random signals, whose sources are sequences of random numbers. Thus the time scale when compared to a true continuous random signal has been altered. Secondly when selecting maximum and minimum values, variations (i.e. ranges) less than $20 \%$ of the r.m.s. value of the signal are omitted.

As explained in Chapter 6, these sequences of random numbers are equivalent to periodic samples of a
continuous function. The more maxima and minima that there are in a giren sequence of numbers, the more poorly is the function represented by these numbers (the equivalent sampling intervals being longer). It is shown in Chapter 6 that the samples $A$ and $D$ (as referred to by Naumann) are severly limited in this respect, and give results inconsistent with the continuous signals they are supposed to represent.

The samples $B$ (irregularity factor $=0.534$ ) and C (irregularity factor $=0.856$ ) give average fatigue lives (again based on zero-crossings) almost exactly equal (B; 80800, ( $; 80400$ ) for the same r.m.s. loading. However in considering these results, the time scale limitation must be borne in mind. Although Jacoby (ref. 18) has found that for 2024-T3 clad notched specimens, the triangular-shaped load cycle and sinewave load cycle lead to similar endurances, the rectangular shaped load cycle of the same amplitude leads to shorter endurances, particularly at higher stress levels. Ividently a time dependent mechanism is present. This may be creep a.t the higher stresses.

Another possible time dependent mechanism is cleavage (Forsyth, ref. 14). Thus it may be argued that a. triangular-shaped cycle should lead to longer enarurances than in the case of sine-wave cycles, and that the frequency effect should be greater than that observed by Schijve (ref. 20) However cleavage fracture is not found in metals with e facecentred cubic ( $F C C$ ) siructure, but in those with either bodycentred cubic or close packed hexagonal structures (Schijve, ref. 21). It is significant that tests periormed to find the effects of cycle shave (Jacoby, ref. 18) and frequency (Schijve, ref, 20) have been done vith either 2024 or 7075 alloys, both heving FCC structures. However, such a time dependent mechanism should not affect Nauman's results
(2024 alloy), but triangulization of the loading may affect the endurance obtained for other alloys in this way.

Fuller (ref. 19) performed bending tests on 2024-T3 sheets with a stress concentration of 1 using three random traces with differing spectra, which produced very similar endurances. The values of irregularity factor quoted by Puller are 0.96 (trace A), 0.75 (B) and 0.95 (C). These values are based on electronic counts of zeromorossings and hand counts of peals.

The values of irregularity factor for cases $B$ and C however do not agree with those estimated theoretically from the known power spectral density distributions. These estimates are 0.95 (A), 0.89 (B) and 0.81 (C). It is significant however that the number of zero-crossings (counted electronically) agree remarkably well with the estimates in all three cases. It is the discrepancies in the expected number of peaks causing the disagreement in irregularity factor. However the counts of peaks were not only done by hand but were based on samples which in Fuller's own estimate were small.

In addition the examples of traces that fuller shows add to the confusion. The bandwidth for case $C$ appears from the sample to be a lot wider than the empirical value of irxegularity factor ( 0.96 ) would allow. Indeed an estinate from this sample is 0.59 - so that by either standard (p.s.d. estinate or empirical) it must be an unrepresentative sample.

A further check is the ratio between r.m.s. value of peaks and the r.m.s. of the continuous trace. This ratio is a simple function of irregularity factor. For the three cases the ratios correspond to the following values of
irregularity factor; 0.834 (A), 0.720 (B), 0.925 (C). Thus the measured irregularity factors for cases $B$ and $C$ are substantiated. However case. A - apparently the one least in doubt, is not.

It is clear then that any conclusions regarding the effect of irregularity factor on endurance cannot be made in this case. In fairness to Fuller it must be pointed out that his work deals with far more broader problems in random fatigue than just the effects of spectrum width. His comments on this particular aspect are really minor facets of a very comprehensive treatise.

It may be concluded therefore, from this review of random testing, that the commonly held view that spectrum shape has no effect on endurance is not substantiated. Indeed there are indications that for axial tests on 2024-14 specimens wider spectra lead to lower endurances, with the effect increasing with overall mean stress level. This trend could be explained on the grounds of favourable residual stresses. For narrow bandwidth loading there is a greater proportion of high loads than for mander bandwidth loading at the same r.m.s. level. This fact coupled to higher overall mean stresses may account for a trend of this kind.



## 8. CONCLUSIONS

A digital recording systein has been designed and manufactured enabling the recording and anelysis of continuous random signals. The system incorporates design features which have led to its comparatively low cost - and yet it has been found to be adequate for the purposes for which it was designed. One limiting feature in its design philosophy, namely the accumulative error characteristic, has been found to be acceptable.

This type of design could be used for much faster systems, recording over long periods. The use of higher packing densities on magnetic-tape, or higher tape speeds, or buffering the information to magnetic tape into larger blocks, would mean that much faster following rates could be achieved. The limitation of accumulative errors through dropouts could be overcome by periodic (although relatively infrequent) resetting of the counter to zero and subsequent recycling. This could be on automatic feature of a faster system, and need not require an extra information channel for the recorder. An unused bit pattern (for example a pulse on both polarity channels simultaneously) could be used for the recycling.

The recording system has been used only to find various histograns. However power spectral density distributions are also obtainable. Instead of punching peper tape for maxima and minima, the digital value of the recorded signal can be punched at a constant rate through automatic triggering of the sampling facility which is availeble on the system console. The power spectrum is then obtained by Fourier transform of the auto-correlation function using any
of the normal digital methods. With the present configuration of the system, power spectra of signals with frequency contents up to $100 \mathrm{c} / \mathrm{s}$ or more are obtainable in this way.

Most of the salient features of the results of the recordings have been discussed in Chapter 5, where comparisons of the counting methods are made on the basis of probability density distributions. It is the practice of some authors to compare counting methods using cumalative plots of the data. Probability density distributions show the proportion of events at each stress level, so that comparisons can be made using any chusen criteria in selecting the important ranges of stress for the comparison. Cumulative plots on a. log scale are useful only in high-lighting events at higher stress levels.

The linear curnulative damage estimates have indicated the potentially large differences in estimated lives that can be obtained from different counting methods, particularly with wider bandwidth loading.

The establishing of the range distributions has brought to light several interesting points. The criterion for adequate representation of the ranges should be the agreement of the peak aistribution with the theoretical distribution. For the cases considered here this apparently requires an overall average at at least $7 \frac{1}{2}$ deta points per cycle. This figure will, of course, be directly related to the minimum anount of digital data required to compute power spectral densities; an amount corresponding to a sampling rate of three times the maximum frequency content. Sampling at a rate lover than this leads to the effect known as aliasing, where power at frequencies greater than the

Nyquist frequency (half the sampling rate) is folded back into lower frequency regions of the spectrum.

This figure has an important bearing on random fatigue testing using digital techniques. There is now an increasing tendency towards digital representation of randon loading. The continuous signal is not really fully represented unless the individual ranges are adequately described by the digital information. Thus the criterion for power spectrum calculations (of 3 data points per maximum frequency) is a necessary condition for the representation of the signal (otherwise the spectrum itself, is not properly prescribed, and therefore the distribution of ranges is not properily prescribed.)

The range distributions themselves may be of use in the development of crack propagation theories which up to now have assumed a Rayleigh distribution of ranges. The distributions are clearly not Rayleigh, even for the smallest bandwidth case considered (Irregularity Factor $=0.89$ ) It is interesting to note that the range distributions become more like Rayleigh shape distributions when the size of the smallest detected range increases.

The review of random fatigue-test results has established that the comnonly held view of the irregularity factor having no effect is not really substantiated. Some of the woris which have led to this view have certain short-conings which place limitations on the conclusions which can be drawn from them. The values of irregularity factor quoted by Fuller are not reliaiole, Nauman's data for his cases $A$ and $D$ does not lead to the expected peak distributions. Another work (Clevenson and Steiner) contained a trend which has been either overlooked or considered insignificant. This
trend does however agree with that formed by Swanson. A further conclusion may be drawn from Clevenson and Steiner's work. On the basis of their results the lives under randon loading are similar for any spectrum shape, provided that the irregularity factor is approximately constant.
9.1 Digital Recording System Units
9.1.1 The Logic Unit
(a) Gircuit Design

The requirements of the logic unit are fan-in and fan-out of ten, switching times of $5 \mu s$ or less (for operation at 20 kc ) and low cost. These are achieved with the circuit of Fig. 9.1 which incorporates two diode feedback circuits.

The diode circuits have the following properties

1) They prevent both saturation and cutting-off of the transistor (improving switching times).
2) They limit and staioilize the output voltage by providing feedback circuits for each state of the logic unit, thus providing good fan-in and fan-out characteristics.
(b) Circuit Details

In the circuit of Fig. 9.1 the logic unit is in the state 'l' when its output potential is nominally -2.75 v , and at ' $O$ ' when $-.3 \nabla$. With the unit at ' $O$ ' diode Dl conducts forming the feedback circuit and in the 'I' state D2 conducts.

When switching from either state to the other the conaucting diode in the first state must not close early. Premature closure of the diode would reverse the effect of the input voltage causing the switch.

The bistable unit of Fig. 9.2 can be switched by a positive-going pulse applied through a capacitive network to the base of the transistor which is at 'o'. The capacitive network includes directing circuits controlled by the bistable itself which direct the pulse to that transistor. The binery unit thus formed switches with each input pulse. A similar gating circui.t is directed by the polarity control and selects the output from one or other of the transistors depending on the polarity state. This out lut circuit produces a positive-going pulse for the switch 'l' to 'O' of the selected transistor.

The reversible binary counter is formed by connecting two or more binaries through their respective input and output circuits. In the diagram of Fig. 9.3 the state of the counter is defined by the right-hand collectors. When adding the outputs are derived from the right-hand collectors and when subtracting the left -hand ones are used. The counting sequences (with decimal equivaients) is shown in the diagram.
9.1.3 Input Amplifiers

The main amplifier (operating from the virtual earth potential) is d.c. differential. Germanium $p-n-p$ and silicon $n-p-n$ transistors are used in sequence. The circuit (Fig. 9.4) incorporates both a balance control and balance monitoring meter on the system console.

The main amplifier drives two single stage d.c. differential amplifiers in parallel which control the pulse gate and polarity circuits. The gains of the amplifiers (see Fig. 9.5) are pre-set to operate the pulse gate at $\pm 2.5 \mathrm{nV}$
and the polarity control at $\pm 1.5 \mathrm{mV}$.
9.1.4 Gate Triggexing Circuit

The gate triggering circuit consists of a conventional Schmidt trigger set at -2.2 v followed by a driving stage for the monostable (Fig. 9.6)
9.1.5 Current Switches

The current switches incorporate a silicon $n-p-n$ transistor and wirewound resistors. The transistors have very low leakage current when switched off and low slope resistance and stable collector-to-emitter voltage when saturated. In the circuit of Fig. 9.7 the stability of the output current. is determined by the -4 v supply to the emitter and the transistor $V_{c e}$.

Fig. 9.7 shows the characteristics of the circuit, where the total possible variation in base current is shown in relation to the minimum base current to ensure saturation. In this configuration the variation of $V_{c e}$ due to temperature is $1 \%$ per ${ }^{\circ} \mathrm{C}$ (in the worst possible case). However as $V_{c e}$ is $1 / 25$ of $V_{r}$ the effect of temperature on the current from the switch is $0.04 \%$ per ${ }^{\circ} \mathrm{C}$. The leakage current is small ( $0.3 \mu \mathrm{~A}$ per switch), but in any case has no effect (at constant temperature) as the main amplifiers are balanced at the start of a recording, thus accommodating any static out-of-balance at the virtual earth. The variation of leakage current with temperature (approximately $\pm 0.2 \mu \mathrm{~A}$ for $\pm 10^{\circ} \mathrm{C}$ ) is also negligible.

The -4 v supply is derived from the recording system main d.c. supply and is extremely stable.
9.1.6-4v Supply

The design of this unit is based on similar units made in the department, Fig. 9.8. To obtain a lower variation of $V_{0}$ with $I_{0}$ a higher gain first stage is used, which is also made differential to reduce the variation of $V_{0}$ with temperature. Io virtually eliminate the variation of $V_{0}$ with the main d.c. supply the first and second stages are run off a Zener diode.

The final circuit (Fig. 9.9) has an extremely stajle output voltage (less than $\pm .02 \%$ drift overall).
9.1.7 Operating Facilities

To operate the system several facilities are incorporated on the console.

1. Record/Replay: a switch which selects the mode of operation.
2. D.C./A.C. Input: the input signal may be d.c. or a.c. coupled. In the latter case a d.c. bias of half full-scale is added.
3. Reset:
4. Hold/Record:
5. Bias/Signal:
a. push-button which resets all counters to zero (used prior to a recording or replaying).
a switch which places or removes an inhibit on the gate. This switch is used to lock the recording system prior to a recording whilst the system is reset.
a switch which selects for recording either a built-in constant bias of half full scale, or the input signal. Recording of the bias before and after a
recording provides a means of detecting any drift in the system.
6. Manual Sample: during replay the contents of the counter can be punched onto paper tape by manual triggering of the data output channel. This may be done prior and subsequent to a replay to check the signal bias, thus detecting and drift in the system.
7. Balance potentiometer and meter: this provides the means of balancing the main input amplifier prior to a recording.


FIG. 9.1 CIRCUIT OF LOGIC UNIT


FIG. 9.2 CIRCUIT OF BINARY UNIT


FIG. '9. 3 REVERSIBLE BINARY COUNTER


FIG. 9.4 MAIN I/P AMPLIFIER


FIG. $9.52^{\text {nd }}$ STAGE AMPLIFIER


FIG.9.6 GATE TRIGGERING CIRCUIT


FIG. 9.7 CIRCUIT \& CHARACTERISTICS OF CURRENT SWITCHES


FIG.9.8 ORIGINAL $-4 V$ SUPPLY UNIT


FIG.9.9 FINAL $-4 V$ SUPPLY UNIT

# 9. APPENDICES (continued) 

9.2 Data Processing
9.2.1 Introduction

Three types of counting methods are employed; Range-Mean for ranges greater than $0,2.5,5,7.5$ and $10 \%$ full scale, Peak and Iean-Crossing Peak. An eighth count, the Range, is obtained from the Range-Mean data.
.The following data parameters are computed; Number of data points Mean value of data Number of maxima and minima ( $N_{i}$ ) Number of zero crossings ( $N_{0}$ ) Irregularity Factor $N_{0} / N_{1}$
R.M.S. value of peaks R.In.S. value of continuous function Number of data points rejected

The programmes are run a second time factoring the data so that the r.m.s. of the continuous function is $10 \%$ full scale (to relate the conditional Range-Iean counts to the function) and adjusting the mean value (by typically $.15 \%$ to exactly $50 \%$ full scale.

To illustrate the counting methods the flow diagrams show the logic relevant only to the descriptions of the methods. The programes themselves are given in Appendix 3.
9.2.2 Counting ilethods

1. Conditional Range - Mean Count.

All ranges less than $P$ (see diagram) are
ignored. In the diagram only the range $\mathcal{J}$ - $M$ is counted. Points of inflexion are ignored.


Fig. 9.10 gives the flow diagram of the analyses with $P$ taking the values $0,25,50,75,100$.
2. Peak Count

All maxima above and minima below the mean are counted (In the diagram below the points $J, K, I, M, N, O$.).


The flow diagram for this analysis is in Fig. 9.11.
3. Hean-Crossing Peak Count.

Between each crossing of the mean value the highest maxima above and the lowest minima below the mean are counted (in the diagram below the points $\mathrm{J}, \mathrm{K}, \mathrm{I}, \mathrm{M}$.)

(Flow diagram Fig. 9.12).
9.2.3 The Control Programme.

The data tape is divided naturally by the original discrete magnetic tape files into blocks of up to 3000 records. A read-compute-read cycle is employed with blocks of records to obtain efficient usage of the computer.

The main progranme performs the data-handling; reading an input block, checking for incorrect records and entering the block into an array. It then transfers control to the processing subroutines and repeats the cycle for each input block. After the last file is processed control is transferred to subroutines which calculate the data parameters and print out the results.

The incorrect records encountered either contained spurious characters or were two or four digit numbers. These were ignored (except for the record made of them) and were typically one in five thousand.



FIG. 9.11 FLOW DIAGRAM FOR PEAK COUNT


FIG. 9.12 FLOW DIAGRAM FOR B.C.P. COUNT

## 9. APPENDICES (continued)

9.3 Computer Programmes

The programmes are written for the Atlas computer in Extended Hercury Autocode (ENA).

```
main->9000
auxiliary(0,0)
dumps o
tit!e
fatigue data rrocessing
```

routine 1
70) $n=1(10) \mathrm{II}$
l=j0n-2j
$m=l+450$
$0=17+9$
neruine
reuline
space
$i=t(50) \mathrm{m}$
print(i) 3,0
space
space
repeat
ne:uire
$i=0(20 \%)_{3} 80$
ne:":しま!
spacte.
$j=\mathrm{ar}$ ( I ) 0
$k=i+j+r$
print (bk) 5,0
space
space
rersat
ropeat
roweat
neturn
return
routine

$$
i=n(n) 0
$$

$$
r=i+r
$$

$$
\text { print }(b k)_{3}, o
$$

space
spece
repeat
rereat
returi：
routine 4
＞range mean analysis，ranges＞p

$$
\underset{i n}{0}=j-a t(j n) \quad \gg 00 p \div 0 n j>k
$$

$$
\begin{aligned}
& \left.\mathrm{I}_{7}\right)_{\mathrm{i}}^{2} \mathrm{n}=0(\mathrm{I})_{4} \\
& \begin{array}{l}
p=250 n \\
j=c(2 p: 3)
\end{array} \\
& k=c(2 p: i+z) \\
& t=0 \\
& \sum_{j=0}^{j=4} 45, ~ 53=1 \\
& 32) t=t+1 \\
& \text { in= }=\bmod (i n) \\
& \text { zz) jump 3a, p>1\% } \\
& k=2 \pm \\
& \text { jump } 33, j>1
\end{aligned}
$$

$$
\begin{aligned}
& \text { 75) } n=1(10)_{3} I \\
& \text { L= = } 5 r-25 \\
& \mathrm{~m}=\mathrm{L}+25 \mathrm{~J} \\
& 0=n+9 \\
& \text { ne:utine } \\
& \text { ne:uine } \\
& \text { spece } \\
& i=\because L(2 J) m \\
& \text { print(i) } 5,0 \\
& \text { space } \\
& \text { space } \\
& \text { ropeat } \\
& \text { ne:uine } \\
& \text { newline } \\
& \text { space }
\end{aligned}
$$

3ううrj＝37）
jump $4 j$

$1: 1=k-l$
in＝
jump $36, p>i \pi$
＞＞set lable for j＞\％Loop
$r)=33$ ）
＞scalculate locetion
58）$i n=j=i$
in＝3nod（in）
$k:=j+k$
$L:=I(I)=0$
min＝50l：
jump $5 \ni, m i s i n$
repeat
59） $12 \pi=1(1)_{20}$
min＝ 100 Mi
jump 60, mi：$>16$
repeat
60）$i n=20 n i n+l n+16 p-20$
ふま！＝bin＋1
＞panges analysis，renjes＞ 0
$i n=2040+20 p n+l n$
以立 $n=0 \dot{1} n+1$

```
>>set Lable for j>kloop routine s l33
r)=30)
jump 58
35) k=l
jump j3
45) jump so,t=0
t=七+I
l=at
jump(r)
50)c(2pb)=j
repeat
return
24)p=0(25)=00%:
newline
ne:uline
newline
caption
rance:i mean analysis, ranges >
print(p)3,0
r=15p
jumpdom(ra)
repeat
```


$m=n$
jump $(s)$
22）$x=y$
$y=2$
jump Iz
ne：uline
re：uline
caption
mean crossinj peak anabysis
$r=2000$
jumpdom $\left(r_{3}\right)$

## is）return

$$
\begin{aligned}
& i n=2 I \\
& j: 1=40 \\
& C=y-500 \\
& c I 0=010+c * c \\
& D=!\div 1
\end{aligned}
$$

routine 0
－＞ppear countinj analysis

$$
t=0
$$

$$
\begin{aligned}
& j u m \quad 10, s i \neq 1 \\
& x=30 \\
& y=a 1 \\
& y=a n
\end{aligned}
$$

## $t=1$

| I0）junp $50, y>\%$ | $x=3$ |
| :---: | :---: |
| fump $2 I, y>x$ | $y=2$ |
| $1_{2}$ ）јume $93, t=0$ | jume to |
| t＝さ＋I |  |
| $z=$ at |  |
| jump $22, y>z$ | 62）$x=y$ |
|  | $y=z$ |
| in＝ | jurn je |
| $j \hat{3}=20$ |  |
| jump yo | 93）roturs |
| $\text { 21) jump } 93, t=0$ $t=\dot{t}+1$ |  |
| $x=y$ |  |
| $y=a t$ |  |
| jump ro，${ }^{\text {jox }}$ | 24）newline |
| $\mathrm{c}=\times-j 00$ | no．al幺ne |
| C士O＝CIO＋C＊O | realine |
| $\mathrm{j}=\mathrm{b}+\mathrm{I}$ | caこせion |
| jumplo | Neat crabysio |
|  | $\mathrm{r}=2 \pm \mathrm{r}^{\circ} \mathrm{O}$ |
| 20： $0^{0} 93, t=0$ | juntocua（rッ\％3） |
| $\pm 二 \pm+1$ |  |
| $x=y^{\prime}$ $y=2$ | return |
| $y=a t$ |  |
| junp， 10 |  |


し：＝2 ゴ：
jump gi，lil＞y reveat

勺j
$x=y$
$y=z$
jump．Io

62）$x=y$
$y=z$
jump je

93）rotura

24）newline
no．．．tまne
reaine
くもっだが，
peat arabysia
$\mathrm{r}=2 \mathrm{~L}$ ； 0

return
＞＞compute cumulative results
$\left.{ }_{1}\right) K=c(20) 330$
$i=19(-1) I$
$j=\dot{j}+k+r$
b $j=b j+5(j+r)$
repeat
repeat
returis
＊＊
routire 2
$3>00$ meute cumulative results
$\because I) j=2(I)=0$
$i=r^{+}$
$b \dot{i}=5 \dot{i}+h(\dot{I}-I)$
repeat
$j=30(-I)_{2 I}$
$\dot{i}=r+i$
$\dot{b} i=b \vec{i}+b(i+1)$
repeat
return

## ＊＊

$\gg$ compute cumulative results

1）$\dot{i}=19(-I) I$
$j=i+r$
$b j=b j+b(j+i)$
repeat
retuma
routine 9
＞＞cunulative ranize analysis
I）$p=0(25) 100$
$r=50 p$
ne：iline
revtine
ne：uine
captio
cumulative raize mear results，ranges $>$
print（p）j，o
junpdona $\left(n_{7}\right)$
jumpdonn（ $n=$ ）
nepect

neッしine
realisie
ne：ルine
routine 12
＞＞0mulative peak analysis
1）MO＂Line
ne：ルシャe
caption
cumulative rarse results，ranyes $>$
$\mathrm{p}=5$ ：
print（p） 5,0
$r=2040+47$
jumpdown（re）
repsat
retum
caption
cumulative reat results
$r=2140$
三umaonm（rac）．
fumpdow $\because\left(r_{3}\right)$
return
>print out data paraneters

| 20)ncwine | ne:uine |
| :---: | :---: |
| ception | cartion |
| no. of data tapes = | r. n. s. value of peaks = |
| print(min)3.0 |  |
| nealine | cio=asgnt (0io) |
| caption | print (cso)0,0 |
| no. of deta points = print(0in) m 。 |  |
|  | newine |
| ne:dine | captwion va mention $=$ |
| caption | r.s. s. valie of function $=$ |
| mean value of data $=$ |  |
| do = do:0\% | primet(cio) 0.6 |
| print(do)0,6 |  |


|  | returs |
| :---: | :---: |
| nowline |  |
| caption | 2a) nemline |
| no. of maxima and minima $=$ | rom.s. vue not evaluated, $b=0$ |
|  |  |
| $j=i+6 i^{400}$ | return |
| repeat | ** |
| print(j) |  |
| neuline |  |
| caption | routine 14 |
| no. of zero crossins $=$ | routine 14 |
| $n=0$ $i=2001(I) 2040$ | $\gg$ scale to 100 r.m.s. function |
| $k=k+b$ i |  |
| repeat | 1) $=1-10$ ai ai-s |
| print(k) 5.o | cijai* |
| ne:uine | $a i=2 i+300$ |
| caption | coreat |
| nosmi $=$ | repea |
| do $=1 \times 19$ | recari |
| print (co)0,4 |  |
| ne:uine |  |
| caption |  |
| no. of date points rojocted | $=$ |
| primt(a) ${ }^{\text {a,o }}$ |  |

$0:=I$
3） $0=0$
2） $0=0 \div I$
6）read（a0）
jump 6，co＞1000
jump 6, ？ 30
juma

0）$a=a+1$
$0=0-1$
$j n=0$
fo）readch（in）
jump is，intopode（＊）
12）${ }^{n} n=j n+i$
jume $10, j$ 解；
jump $I_{7}$
15）jump $2, i n=$ code $(n)$
reacich（in）
ju7p $\mathrm{I}_{2}, \dot{i}=\operatorname{qcode}(*)$
jump Ij

```
>>nead no. of tapes and mean value
read(rni)
nead(a;i)
reau(o)
read(f)
read(ao)
Z=100/T
I7)0:3=0:3+0
jumpdowi(nms) (
jumpdo::%(n&)
```

Co $=20$
$\mathrm{sin}=\mathrm{I}$
$S:=I$
$e=-n i$
$j=0$

```
e:に゙
close
```


### 9.4 S-N Curves

The $S-\mathbb{N}$ curves are derived from ref. 7. and are of the form $S=a N^{b}$. In order to refer the numerical data of the histograms to a stress scale, the maximum possible peak value ( $\frac{1}{2}$ full-scale excursion from the mean) plus the mean value is made equal to the ultimate stress divided by a safety factor (1.5). This latter stress will be referred to as the proof stress. The mean stress is the proof stress divided by 3.5. This means that the probability of exceeding the proof load is of the order of $10^{-6}$ for a Rayleigh distribution of peaks and is, of course, lower for irregularity factors less than unity.

The stress scale is arbitrary (ultimate $=700$, therefore Proof $=466.7$, Inean $=133.3$ ) and the numerical scale of the histograms factored by 0.667.

$$
\text { i.e. Fean }+500 \times 0.667=\text { Proồ }
$$

The basic SN curves are then defined by the first co-ordinate (500, $10^{4}$ ) and the second co-ordinate (250, at three successm ive values of $\mathrm{N}, 10^{7}, 10^{6}, 10^{5}$ ). These three curves correspond to the following Al.alloy cases (ref. 7);
(i) High Slope - lugs with interference (E.05.04)
(ii) Medium Slope - lugs without interference (F.05.03)
(iii) Jow slope - alloy in bending (i.07.01)

Although in the data sheets these curves have fatigue limits, the contributions to the estimated lives from the regions of the curves where discrepancies occur are insignificant. The 'damage intensity' is negligibly small in these regions.

The above curves are defined for the zero mean stress cases. The curves corresponding to non-zero mean stress are derived from the R-M diagram of data sheet A.00.01 (ref. 7), using a value of unity for the parameter defining the effect of the mean stress. Thus the relationship between the stress amplitudes (Sa) at mean values (Smi) and the equivalent stress amplitude ( $\mathrm{Sa}_{\mathrm{o}}$ ) giving the same fatigue life at zero mean stress is

$$
\mathrm{Sa}=S a_{0} /\left(I-S m / S_{u l t}\right)
$$

where $S_{u l t}$ is the ultimate stress.
The ratio Mean/Proof has no effect on the comparative lives (eg. peak to range) for these $S-N$ curves, except for comparisons involving mean values (i.e. rangemean method). The comparison between estimates from differing counting methods depends largely only on the slope of the S-N curve.

9 APPETDICES (continued)
9.5 Tables of Data

On the following pages are tabulated the probability density distributions relating to the data from Leybold (ref. 10) and Naumann (ref. 9) followed by the distributions relating to the recordings of filtered random noise and car suspension loads.

Finally the output tabulations are given from the data analysis programes for the recordings of filtered random noise and car suspension loads.

## TABLES OF DATA FROM LEYBOLD AND NAUMANN

Naumann uses the same data as Ieybold except for case $D$, for which Naunann introduces a different case D. Therefore the following letter code for these joint results is used:
SOURCE Nau/Ley A Nau/Ley B Nau/Ley C Ieybold D Naumann D
IETTER CODE
A
B
C
D
E

PEAK DISTRIBUTIONS

| $\frac{X}{\sigma}$ | A | B | C | D | E |
| :---: | :---: | :---: | :---: | :---: | :---: |
| 0.1 | .327 | .359 | .249 | .234 | .370 |
| 0.3 | .394 | .403 | .351 | .339 | .440 |
| 0.5 | .448 | .438 | .446 | .458 | .483 |
| 0.7 | .491 | .447 | .519 | .526 | .510 |
| 0.9 | .491 | .420 | .548 | .575 | .501 |
| 1.1 | .461 | .385 | .541 | .563 | .449 |
| 1.3 | .424 | .350 | .482 | .503 | .392 |
| 1.5 | .363 | .289 | .417 | .429 | .316 |
| 1.7 | .297 | .219 | .351 | .332 | .243 |
| 1.9 | .212 | .184 | .263 | .265 | .179 |
| 2.1 | .158 | .123 | .197 | .193 | .124 |
| 2.3 | .115 | .096 | .132 | .136 | .082 |
| 2.5 | .073 | .061 | .095 | .092 | .052 |
| 2.7 | .048 | .044 | .058 | .054 | .030 |
| 2.9 |  |  |  | .038 |  |


| X | A | B | C | D | E |
| :--- | :---: | :---: | :---: | :---: | :---: |
| $\frac{\sigma}{\sigma}$ | . |  |  |  |  |
| 0.1 | .469 | 1.060 | .449 | .421 | .416 |
| 0.3 | .526 | .811 | .367 | .353 | .634 |
| 0.5 | .625 | .724 | .457 | .426 | .798 |
| 0.7 | .687 | .633 | .555 | .535 | .889 |
| 0.9 | .689 | .525 | .597 | .607 | .742 |
| 1.1 | .605 | .399 | .594 | .589 | .575 |
| 1.3 | .487 | .309 | .535 | .525 | .389 |
| 1.5 | .359 | .212 | .443 | .446 | .234 |
| 1.7 | .244 | .1 .43 | .325 | .345 | .126 |
| 1.9 | .149 | .093 | .240 | .244 | .061 |
| 2.1 | .085 | .049 | .173 | .184 | .028 |
| 2.3 | .045 | .022 | .106 | .125 | .013 |
| 2.5 | .018 | .013 | .071 | .079 |  |
| 2.7 |  |  | .042 | .050 |  |
| 2.9 |  |  | .024 | .030 |  |

tables on data froin ieyboid amd matrann (CONTD)

DISTRIBUTIONS OF MEAN VALUES

|  | B |  | C |  | D |  |
| :--- | :---: | :---: | :---: | :---: | :---: | :---: |
| $\frac{\mathrm{X}}{\sigma_{\mathrm{f}}}$ | $\frac{\mathrm{x}}{\sigma_{m}}$ | $\mathrm{p}\left(\frac{\mathrm{x}}{\sigma_{m}}\right)$ | $\frac{\mathrm{x}}{\sigma_{m}}$ | $\mathrm{p}\left(\frac{\mathrm{x}}{\sigma_{m}}\right)$ | $\frac{\mathrm{x}}{\sigma_{m}}$ | $\mathrm{p}\left(\frac{\mathrm{x}}{\sigma_{\mathrm{m}}}\right)$ |
| 0.0 | 0.0 | .393 | 0.0 | .402 | 0.0 | .409 |
| 0.2 | 0.27 | .390 | 0.572 | .341 | 0.763 | .297 |
| 0.4 | 0.54 | .348 | 1.144 | .204 | 1.527 | .120 |
| 0.6 | 0.81 | .291 | 1.716 | .092 | 2.291 | .028 |
| 0.8 | 1.08 | .216 | 2.288 | .029 | 3.054 | .004 |
| 1.0 | 1.35 | .157 | 2.86 | .007 |  |  |
| 1.2 | 1.62 | .109 |  |  |  |  |
| 1.4 | 1.89 | .063 |  |  |  |  |
| 1.6 | 2.16 | .038 |  |  |  |  |
| 1.8 | 2.43 | .022 |  |  |  |  |
| 2.0 | 2.70 | .012 |  |  |  |  |

RANDOE NOISE $2-9 \mathrm{c} / \mathrm{s}$ : PROBABILITY DENSITY DISTRIBUTIONS

| $\frac{\mathrm{x}}{\sigma}$ | PEAK | RANGE | MCP | $\frac{\mathrm{x}}{\sigma_{m}}$ | $\cdot$ |
| :--- | :--- | :--- | :--- | :--- | :--- |
| 0.125 | .286 | .566 | .247 | 0.507 | .356 |
| 0.375 | .402 | .510 | .273 | 1.523 | .118 |
| 0.625 | .460 | .536 | .348 | 2.538 | .016 |
| 0.875 | .542 | .544 | .463 | 3.553 | .0014 |
| 1.125 | .483 | .548 | .451 |  |  |
| 1.375 | .414 | .447 | .392 |  |  |
| 1.625 | .339 | .324 | .325 |  |  |
| 1.875 | .236 | .231 | .246 |  |  |
| 2.125 | .169 | .136 | .169 |  |  |
| 2.375 | .105 | .079 | .104 |  |  |
| 2.625 | .062 | .038 | .062 |  |  |
| 2.875 | .033 | .023 | .033 |  |  |
| 3.125 | .024 | .010 | .024 |  |  |

RANDOM NOISE $2-25 \mathrm{c} / \mathrm{s}$ : PROBABIIITY DENSITY DISTRIBUTIONS

| $\frac{x}{\sigma}$ | PEAK | RANGE | LCP | $\frac{x}{\sigma_{m}}$ | MEAN |
| :--- | :--- | :--- | :--- | :--- | :--- |
| 0.125 | .329 | .634 | .281 | 0.424 | .373 |
| 0.375 | .423 | .572 | .288 | 1.272 | .168 |
| 0.625 | .430 | .576 | .343 | 2.12 | .042 |
| 0.875 | .489 | .571 | .360 | 2.97 | .006 |
| 1.125 | .455 | .508 | .380 |  |  |
| 1.375 | .405 | .422 | .368 |  |  |
| 1.625 | .314 | .307 | .288 |  |  |
| 1.875 | .218 | .191 | .226 |  |  |
| 2.125 | .165 | .115 | .152 |  |  |
| 2.375 | .093 | .054 | .108 |  |  |
| 2.625 | .062 | .027 | .058 |  |  |
| 2.875 | .037 |  | .037 |  |  |
| 3.125 | .019 |  | .020 |  |  |


| $\frac{x}{\sigma}$ | PEAK | RANGE | I.CP | $\frac{x}{\sigma_{m}}$ | ITAN |
| :--- | :--- | :--- | :--- | :--- | :--- |
| 0.125 | .354 | 1.344 | .312 | 0.311 | .389 |
| 0.375 | .439 | .864 | .242 | 0.933 | .253 |
| 0.625 | .413 | .599 | .214 | 1.555 | .114 |
| 0.875 | .423 | .442 | .197 | 2.177 | .038 |
| 1.125 | .356 | .280 | .202 | 2.799 | .008 |
| 1.375 | .304 | .199 | .196 |  |  |
| 1.625 | .226 | .126 | .150 |  |  |
| 1.875 | .154 | .070 | .113 |  |  |
| 2.125 | .112 | .039 | .082. |  |  |
| 2.375 | .060 | .023 | .052 |  |  |
| 2.625 | .043 | .010 | .035 |  |  |
| 2.875 | .023 |  | .021 |  |  |

The results are tabulated on the following pages as follows:

## Range Mean:

Each column heading is the medium value of ranges in the group
lst column
eg. 25 (Ranges, $0 \rightarrow 50$ )

2nd column
75 (Ranges, $50 \rightarrow 100$ ) etc.

The heading for each row should be taken as the same for the corresponding column, except that each row relates to the grouping of mean values.

| eg. | Ist row | $(25)$ |
| :--- | :--- | :--- |
| 2nd row | $(75)$ | (Heans $0 \rightarrow 50)$ |
|  | (Heans $50 \rightarrow 100)$ |  |

Peak and W.C.P:

The numbers in the odd numbered rows correspond to the peak value for each group. The numbers in the even rows (and below each peak value) are the number of peak counts in the corresponding group of peaks. Each peak value quoted is the lower limit of the peak values in the group.

$$
\text { eg. } 0 \text { (Minima between } 0 \text { and 25) 25 (Minima between } 25 \text { and }
$$ 50)

$$
500 \text { (haxima between } 500 \text { ana 525) } 525 \text { (nexina between }
$$

## Filtered Random Noise $2-25 \mathrm{c} / \mathrm{s}$

Number of maxima and minima ..... 23383
Number of mean crossings ..... 17114
Irregularity Factor ..... 0.732
Ifean Value 500.00
R.M.S. value of signal ..... 100.26
R.M.S. value of maxima ..... 124.24






## MEAN GHOSSING PEAK ANALYSIS

| 0 | 25 | 50 | 75 | 100 | 125 |
| :---: | :---: | :---: | :---: | :---: | :---: |
| 0 | 0 | 2 | 0 | 10 | 16 |
| 250 | 275 | 300 | 325 | 350 | 375 |
| 308 | 425 | 666 | 852 | 1111 | 1044 |
| 501 | 525 | 550 | 575 | 600 | 625 |
| 815 | 857 | 969 | 1027 | 1176 | 1040 |
| 750 | 775 | $80 \%$ | 825 | 850 | 875 |
| 178 | 1.04 | 61 | 24 | 14 | $6$ |

PEAK ANALYSIS

| 0 | 25 | 50 | 75 | 100 | 125 | 150 | 175 | 200 | 225 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 0 | 0 | 2 | 0 | 10 | 16 | 25 | 55 | 110 | 183 |
| 250 | 275 | 300 | 325 | 350 | 375 | 400 | 425 | 450 | 475 |
| 289 | 466 | 648 | 889 | 1231 | 1260 | 1526 | 1271. | 1136 | 984 |
| 500 | 525 | 550 | 575 | 600 | 625 | 650 | 675 | 730 | 725 |
| 941 | 1284 | 1240 | 1330 | 1401 | 1136 | 948 | 624 | 53.1 | 294 |
| 750 | 775 | 800 | 825 | 850 | 875 | 900 | 925 | 950 | 975 |
| 178 | 105 | 61 | 26 | 12 | 6 | 6 | 2 | 0 | 1 |
|  |  |  |  |  |  |  |  | $\begin{aligned} & 4 \\ & 80 \\ & 0 \end{aligned}$ |  |

## Filtered Random Noise $2-9 \mathrm{c} / \mathrm{s}$

Number of maxima and minima ..... 12947
Tumber of mean crossings ..... 10254
Irregularity Factor ..... 0.792
Mean Valve ..... 500.03
R.f.S. value of signal ..... 99.964
R.M.S. value of maxima ..... 127.52



| 25 | 75 | 1.25 | 175 | 225 | 275 | 325 | 375 | 425 | 475 |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| 0 | 0 | 11 | 0 | - 0 | 0 | 0 | 0 | 0 | 0 |
| 0 | 0 | 0 | 0 | 0 | 0 | 0 | 0 | 0 | 0 |
| 0 | 0 | ก | 0 | 0 | 0 | 0 | 1 | 0 | 0 |
| 0 | 0 | 0 | 0 | 0 | 0 | 0 | 0 | 0 | 0 |
| 0 | 0 | 0 | 0 | 0 | 0 | 0 | $1)$ | 0 | 0 |
| 0 | 0 | 0 | 1 | 0 | 0 | 0 | 1 | 0 | ก |
| u | 0 | 0 | 1 | 1 | 0 | 2 | 11 | 0 | 0 |
| -0 | 12 | 4 | 10 | 8 | 2 | 6 | 11 | 5 | 9 |
| . 1 | 65 | 104 | 104 | 127 | 143 | 125 | 94 | 55 | 33 |
| $-0$ | 436 | 547 | 658 | 740 | 643 | 41.6 | 289 | 176 | 99 |
| 0 | 414 | 570 | 717 | 767 | 614 | 417 | 326 | 160 | 8 ? |
| . 1 | 95 | 113 | 123 | 146 | 152 | 137 | 85 | 63 | 30 |
| 1 | 4 | 7 | 3 | 4 | 4 | 7 | 7 | 6 | 1 |
| 6 | 0 | - 2 | 0 | 1 | 1 | 0 | 0 | 0 | 0 |
| U | ! | 0 | 0 | 0 | 0 | 0 | 1 | 0 | 0 |
| i) | n | ก | 0 | 0 | 0 | - 0 | i) | 1 | 0 |
| 10 | 0 | 0 | 0 | 0 | 10 | 0 | 0 | 0 | 0 |
|  | 0 | n | 0 | 0 | 0 | 0 | 0 | 10 | 0 |
| 0 | 0 | 0 | 0 | 0 | 0 | 0 | 0 | 0 | 0 |
| $\cdots$ | 0 | 0 | - 0 | 0 | 0 | 0 | $1)$ | 0 | 0 |
| 525 | 575 | 625 | 675 | 725 | 77.5 | 825 | 875 | 8925 | 975 |
| 0 | 0 | 0 | 0 | 0 | 0 | 0 | 0 | 0 | 0 |
| i) | 0 | 0 | 0 | 0 | O | 0 | 0 | 0 | 0 |
| $1]$ | 0 | 0 | 0 | 0 |  | 0 | 0 | 0 | 0 |
| 4 | 1 | ก | 0 | 0 | 0 | 0 | 0 | 0 | 0 |
| 0 | 0 | 0 | 0 | 0 | 0 | 0 | 19 | 0 | 0 |
|  | 0 | 0 | - 0 | 0 | 0 | 0 | 0 | 0 | 0 |
| -i | 0 | 0 | - 0 | 0 | 0 | 0 | 0 | 0 | 0 |
| 0 | 3 | 2 | 0 | 0 | 0 | 0 | 0 | - 0 | 0 |
| - 16 | 9 |  | $\bigcirc 2$ | 0 | 0 | 0 | 0. | 0 | 0 |
|  | 26 | 11 | 6 | 4 | 2 | 1 | 0 | 0 | 0 |
| $\bigcirc 40$ | 22 | 7 | 8 | 6 | 5 | - 0 | 0 | n | 0 |
| 17 | 13 | 7 | 0 | 1. | 0 | \%n ${ }^{n}$ | " | $\cdots$ | 0 |
| - 6 | 3 | n | 0 | 0 | 0 | RANGE MEAN | ANALYSIS, | RaNGES $>$ | 00 |
| . 1 | 0 | 0 | 0 | 0 | 0 | $\because$ u | ! | $u$ | 0 |
| - 0 | - 0 | - 0 | -10 | 0 | 0 | - 0 | - 2 | 0 | - 0 |
| - 0 | 0 | 0 | 8 | 0 | 0 | 0 | 0 | 0 | 10 |
| $\square 9$ | 0 | 0 | 0 | n | 0 | 0 | 1) | 0 | 0 |




```
MEAN CROSSTNG PEAK ANALYSIS
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PEAK ANALYSTG


## Car Suspension Ioads

Number of maxima and minima ..... 15143
Number of mean crossings ..... 6964
Irregularity Factor ..... 0.460
Mean Value ..... 500.02
R.M.S. value of signal ..... 100.06
R.M.S. value of maxima ..... 110.13







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