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#### FULL LENGTH ARTICLE 2

# Algorithm and experiments of six-dimensional force/torque dynamic measurements based on a Stewart platform

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# **KEYWORDS**

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- tion: F/T-driven alignment;
- 17 18 Precision analysis;
- P&O adjusting platform; 19
- 20 Six-dimensional F/T

Abstract Stewart platform (SP) is a promising choice for large component alignment, and interactive force measurements are a novel and significant approach for high-precision assemblies. The designed position and orientation (P&O) adjusting platform, based on an SP for force/torquedriven (F/T-driven) alignment, can dynamically measure interactive forces. This paper presents an analytical algorithm of measuring six-dimensional F/T based on the screw theory for accurate determination of external forces during alignment. Dynamic gravity deviations were taken into consideration and a compensation model was developed. The P&O number was optimized as well. Given the specific appearance of repeated six-dimensional F/T measurements, an approximate cone shape was used for spatial precision analysis. The magnitudes and directions of measured F/Ts can be evaluated by a set of standards, in terms of accuracy and repeatability. Experiments were also performed using a known applied load, and the proposed analytical algorithm was able to accurately predict the F/T. A comparison between precision analysis experiments with or without assembly fixtures was performed. Experimental results show that the measurement accuracy varies under different P&O sets and higher loads lead to poorer accuracy of dynamic gravity compensation. In addition, the preferable operation range has been discussed for high-precision assemblies with smaller deviations.

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The alignment of large-scale and complex components, such as

airframes, satellites, and rockets, typically involves a large

number of assembly fixtures, which control the position and

orientation (P&O) of larger components in order to meet the

accuracy requirement of the final assembly. Traditional fixed

assembly fixtures can only be applied to the alignment of

1. Introduction

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one specific large component. Even a small change in the shape 29 or structure of the large component will lead to redesigning 30 and remanufacturing a new fixed assembly fixture. With the 31 rapid development of the assembly technology toward becom-32 ing digital, more flexible and intelligent, digital flexible align-33 ment systems have gained popularity for large component 34 35 alignment, consisting of both software and hardware. The software includes a control system, a measurement system, a sim-36 ulation system, and a calculation system. The hardware 37 includes a P&O adjusting platform, digital measurement 38 equipment, and an integrated control platform. The large com-39 40 ponents alignment process using a digital flexible alignment 41 system has been transformed from the traditional process, 42 based on manual fixtures and operations, to automatic alignment, which significantly improves aligning precision and 43 44 efficiency.<sup>1</sup>

45 The P&O adjusting platform (such as the electronic mating 46 alignment system, automated positioning systems based on 47 POGOs, and parallel adjusting platforms), as a key section of the large component alignment, can automatically adjusts 48 the P&O of large components.<sup>2</sup> In recent years, parallel robots 49 have been widely used for P&O adjustments of large-scale 50 component assembly, due to their outstanding advantages 51 including high stiffness, high load capacity, fast motion, and 52 high positioning accuracy.<sup>3,4</sup> Being the most frequently used 53 structure of parallel robots, Stewart platforms (SPs) are suit-54 able for machining and manufacturing,<sup>5,6</sup> surgical operations,<sup>7</sup> 55 simulator designing,<sup>8,9</sup> flexible and precise assembly of aircraft sections,<sup>10</sup> spacecraft P&O adjustments,<sup>11</sup> and low-impact alignments.<sup>12</sup> An SP is composed of a moving platform and 56 57 58 a base platform, which are connected with six stretchable limbs 59 through spherical/universal joints. In the operation range, the 60 6-degrees-of-freedom (DOF) motion of the moving platform 61 could be achieved by the motions of the six limbs as a 62 whole.13,14 63

Currently, the main assembly strategy that is followed for a 64 digital flexible alignment system is measurement aided assem-65 bly (MAA), which is based on geometrical control.<sup>15</sup> In order 66 to direct and support the applications of advanced approaches 67 in MAA for wing-fuselage alignment and realize the process 68 69 integration and data fusion, a novel framework of measurement assisted assembly methodology has been proposed, based 70 on key measurement.<sup>16</sup> Aiming to control the geometrical key 71 characteristics and attain the best fit of P&O, which is a key 72 feature in MAA, an optimization algorithm based on key char-73 acteristics for large component assembly has been proposed.<sup>17</sup> 74 75 Using different measurement systems to measure the coordinates of points, the uncertainty of measurement results was 76 analyzed from uncertainty contributions and setup proce-77 dures.<sup>18,19</sup> With the improvement of manufacturing and pro-78 79 cessing accuracy, a phenomenon that the accuracy of the measurement system is lower than that of assembly design 80 takes place.<sup>20</sup> In this case, it will not only lead to an assembly 81 82 failure, but also cause unexpected damages of components. 83 Thus the measurement and control of the interaction force between components have great significance for the quality 84 of the final product. Since six-dimensional force/torque (F/T)85 sensors can measure three-dimensional forces and three-86 dimensional torques with appropriate control techniques, they 87 are commonly utilized to complete the force feedback loop 88 control and high-precision assembly of components. The force 89

measurement and control technology rely on two important parts: sensors and force control.

- Six-dimensional F/T sensors: Based on the elastomeric 92 structure, six-dimensional F/T sensors can be divided into 93 two groups: direct output type without coupling and indi-94 rect output type with coupling (including the SP structure). 95 Structures of both types are fixed and unchangeable. More-96 over, the isotropic configuration of a six-dimensional F/T 97 sensor based on an SP, the task-oriented design method 98 of a six-dimensional F/T sensor, and a six-dimensional F/ 99 T sensor have been introduced to complete peg-in-hole 100 assembly tasks.<sup>21,22</sup> A six-beam sensor based on SP and 101 the idea concept of "joint less" structure and beam sensors 102 have been proposed to improve the precision and sensitivity 103 in measuring a small F/T.<sup>23</sup> A six-dimensional heavy F/T 104 sensor with high stiffness and good linearity based on SP 105 has been presented.<sup>24</sup> Experimental results verified the feasibility and validity of the sensor by an established calibration platform. To summarize, six-dimensional F/T sensors 108 have many types of forms and some advanced features. 109 However, they are limited to the work environment and 110 cannot be open-access designed for specific needs. Finally, 111 they are very expensive. 112
- Force control techniques: A shape recognition algorithm based on a six-dimensional F/T sensor and a hole detection algorithm have been reported.<sup>25</sup> Experimental results showed that the two algorithms could complete the assembly of chamferless square peg-in-hole. The six-dimensional F/T sensor was employed to estimate the contact phases and design the assembling strategy to achieve force-guided robotic assembling.<sup>26-28</sup> The admittance characteristics for a force-guided robotic assembly and analytical derivations for different contact states were presented by Wiemer and Schimmels.<sup>29</sup> A modified control scheme for an SP with compensations for interaction force control and positional error recovery was introduced.<sup>30</sup> A novel strategy of the high-precision chamferless peg-hole insertion with a sixdimensional F/T sensor was introduced.<sup>31</sup> This strategy implemented the relation between a peg and a hole from the force sensor signal, and provided a wide range of initial conditions that affected the insertion. To summarize, a correct use of the interaction force can effectively achieve assembling. Finally, many control strategies have also been studied.

Following a literature review, traditional fixed assembly fix-134 tures have been unable to meet the needs of large component 135 alignment in a digital, flexible, and intelligent assembly pro-136 cess. On the contrary, the SP has gained popularity for its out-137 standing advantages in alignment of large-scale components. 138 However, the measurement and control of the interaction 139 forces between components should be considered. Therefore, 140 a digital flexible alignment system with an SP based on six-141 dimensional F/T feedback and combined with force control 142 techniques has been designed in this study. Due to the high 143 manufacturing costs of six-dimensional F/T sensors and the 144 required large size, they are not suitable for direct use in digital 145 flexible alignment systems. Consequently, a P&O adjusting 146 platform based on an SP and force sensors has been designed, 147 which can adjust the P&O of a component and dynamically 148 measure interactive forces. The platform uses six inexpensive 149

force sensors placed in each limb to measure the forces of limbs
 and calculates the six-dimensional F/T based on measurement
 results. Moreover, combined with force control techniques, a
 precision analysis method of the six-dimensional F/T is
 proposed.

This paper takes into consideration multiple influential fac-155 tors of the measurement accuracy of the interaction forces 156 between components. Among the forces, gravity is of great 157 research interest, and for the first time, this paper provides 158 an analytical algorithm of a six-dimensional F/T with dynamic 159 gravity compensation. The setup of the paper is as follows: 160 Section 1 introduces the digital flexible assembly system and 161 its significance, highlights the applications of the SP, and pro-162 163 vides a new perspective and novel methods of large components alignment. Section 2 provides the analytical algorithm 164 of a six-dimensional F/T, proposes a dynamic gravity compen-165 sation model based on the screw theory, and offers a parame-166 ter which is optimized through experiments. For the spatial 167 168 precision analysis, Section 3 uses an approximate cone shape to evaluate the accuracy and repeatability of the six-169 dimensional F/T. In Section 4, using the designed P&O adjust-170 ing platform to verify the accuracy of the proposed algorithm 171 and perform spatial precision experiments, relevant experimen-172 tal data are analyzed and discussed. Section 5 concludes the 173 paper and assesses the validity and limitations of the present 174 algorithm and model. 175

# 2. Analytical algorithm of the six-dimensional F/T with dynamic gravity compensation

## 178 2.1. Overall research description

The overall study for calculating a six-dimensional F/T with dynamic gravity compensation can be depicted in the flowchart presented in Fig. 1. The P&O adjusting platform offers 6-DOF motion, due to the motions of six limbs as a whole, and the sixdimensional F/T is dynamically calculated by force sensors, which are placed in each limb to measure the forces of limbs. Moreover, due to the barycenter and gravity deviations of the large component, wrong calculation results will be derived. Thus, the dynamic gravity compensation is studied.

As shown in the left part of Fig. 1, a traditional sixdimensional F/T sensor is used to measure the sixdimensional F/T. The structural parameters of the sensor cannot be changed; hence, the measurement process is static. Since the lengths of the limbs remain unchanged after the initial setting, there are no sliding joints on the limbs. The sixdimensional F/T is calculated by measuring the forces of the limbs in  $o_1$ - $x_1y_1z_1$ . As shown in the right part of Fig. 1, the P&O adjusting platform based on an SP is used to calculate the six-dimensional F/T. The length of the structural parameters is changed to adjust the P&O of the component; hence, the calculation process is dynamic. Moreover, the barycenter and gravity of the component must be dynamically compensated. Then, the six-dimensional F/T is calculated by measuring the forces of the limbs based on the dynamic gravity compensation in  $o_1 - x_1 y_1 z_1$ .

# 2.2. Analytical algorithm of a six-dimensional F/T based on an SP

As presented in Fig. 2, the P&O adjusting platform based on an SP consists of a moving platform and a base platform, which are connected to each other with six limbs, adjustable in length through sliding joints. In the operation range, the 6-DOF motion of the moving platform could be achieved by the motions of the six limbs as a whole. Force sensors are placed in each limb to measure the force  $f_i$  applied to the limbs. 208



Fig. 1 Overview of overall study for calculating a six-dimensional F/T with dynamic gravity compensation.

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K. Wen et al.

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Encoders are used to measure the length  $l_i$  of the limbs  $(i = 1, 2, \dots, 6)$ . The Cartesian coordinate system of  $o_0 - x_0 v_0 z_0$ is located in the center of the top surface of the base platform, while the Cartesian coordinate system of  $o_1$ - $x_1y_1z_1$  is located in 216 the center of the bottom surface of the moving platform. The centers of the spherical joints are denoted as  $A_i$  and  $B_i$ . 218

The external load  $[F_s, M_s]^T$  of the moving platform in  $o_1$ -219  $x_1y_1z_1$  could be calculated by the measured  $f_i$  and  $l_i$ . The six-220 dimensional F/T can be defined as follows: 221 222

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$$[\mathbf{F}_{s}, \mathbf{M}_{s}]^{\mathrm{T}} = g(f_{i}, l_{i}) \quad i = 1, 2, 3, 4, 5, 6$$

where  $[\mathbf{F}_{s}, \mathbf{M}_{s}]^{T}$  is the calculation results, and  $f_{i}$  and  $l_{i}$  are the 225 226 measured forces and length data of the limbs, respectively.

Once the distance between  $A_i$  and  $B_i$  (limb length  $l_i$ ) is set, 227 the P&O parameters  $\{x, y, z, \alpha, \beta, \gamma\}$  between  $o_1$ - $x_1y_1z_1$  and 228 229  $o_0 - x_0 y_0 z_0$  could be solved by the newton iteration method.<sup>32</sup>

 $S_i$  and  $S_{0i}$  can be given by (as in Fig. 2):

$$\begin{cases} \boldsymbol{S}_{i} = \frac{A_{i} - B_{i}}{\|A_{i} - B_{i}\|} \\ \boldsymbol{S}_{0i} = A_{i} \times \boldsymbol{S}_{i} \end{cases}$$
(5)

where  $A_i$  and  $B_i$  are the coordinates in  $o_1$ - $x_1y_1z_1$ . However, in the actual calculation,  $A_i$  is the position vector from  $o_1 - x_1 y_1 z_1$ to the *i*th spherical joint and  $B_i$  is the position vector from  $o_0$  $x_0y_0z_0$  to the *i*th universal joint. According to the P&O parameters {x, y, z,  $\alpha$ ,  $\beta$ ,  $\gamma$ }, Eq. (5) can be rewritten as follows:

$$S_{i} = \frac{A_{i} - R^{-1}(B_{i} - M)}{\|A_{i} - R^{-1}(B_{i} - M)\|}$$
(6)

where

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(7)

[	$\int \cos \alpha \cos \beta$	$\cos\alpha\sin\beta\sin\gamma - \sin\alpha\cos\gamma$	$\cos\alpha\sin\beta\cos\gamma+\sin\alpha\sin\gamma$
R =	$\sin \alpha \cos \beta$	$\sin\alpha\sin\beta\sin\gamma+\cos\alpha\cos\beta$	$\sin\alpha\sin\beta\cos\gamma-\cos\alpha\sin\gamma$
	$-\sin\beta$	$\cos\beta\sin\gamma$	$\cos\beta\cos\gamma$

Among the P&O parameters, x, y, and z are the displacements 230 231 of  $o_1$ - $x_1y_1z_1$  with respect to  $o_0$ - $x_0y_0z_0$ , and  $\alpha$ ,  $\beta$ , and  $\gamma$  are the rotation angles of  $o_1 - x_1 y_1 z_1$  with respect to  $o_0 - x_0 y_0 z_0$ . 232

The force equilibrium equation could be defined in  $o_1$ -233  $x_1y_1z_1$  using the screw theory as follows: 234 235

$$\begin{bmatrix} \mathbf{F}_{s} \\ \mathbf{M}_{s} \end{bmatrix} = \sum_{i=1}^{6} f_{i} \mathbf{s}_{ii}$$
<sup>(2)</sup>

where  $\$_{ii}$  is the unit screw along the *i*th leg, and could be 238 239 240 obtained by the following:

$$\boldsymbol{\$}_{ii} = \begin{bmatrix} \boldsymbol{S}_i \\ \boldsymbol{S}_{0i} \end{bmatrix} \tag{3}$$

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where

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$$\begin{cases} \boldsymbol{S}_i \boldsymbol{S}_i = 1\\ \boldsymbol{S}_i \boldsymbol{S}_{0i} = 0 \end{cases}$$
(4)



Fig. 2 Schematic diagram of a flexible fixture based on an SP for F/T-driven assembly.

 $M = [x, y, z]^T$ (8)264

where **R** represents a rotation matrix and **M** represents a translation matrix.

Eq. (2) can be rewritten in the form of matrix equation as follows:

$$\boldsymbol{F} = \boldsymbol{G}\boldsymbol{f} \tag{9} \qquad 271$$

where

$$\boldsymbol{F} = [\boldsymbol{F}_{s}, \boldsymbol{M}_{s}]^{\mathrm{T}} = [F_{x}, F_{y}, F_{z}, \boldsymbol{M}_{x}, \boldsymbol{M}_{y}, \boldsymbol{M}_{z}]^{\mathrm{T}}$$
(10) 275  
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$$\boldsymbol{f} = [f_1, f_2, f_3, f_4, f_5, f_6]^{\mathrm{T}}$$
(11) 278  
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$$G = \begin{bmatrix} S_1 & S_2 & S_3 & S_4 & S_5 & S_6 \\ S_{01} & S_{02} & S_{03} & S_{04} & S_{05} & S_{06} \end{bmatrix}$$

$$= \begin{bmatrix} \frac{A_1 - R^{-1}(B_1 - M)}{\|A_1 - R^{-1}(B_1 - M)\|} & \cdots & \frac{A_6 - R^{-1}(B_6 - M)}{\|A_6 - R^{-1}(B_6 - M)\|} \\ A_1 \times S_1 & A_6 \times S_6 \end{bmatrix}$$
(12)

Hence, the external load  $[F_s, M_s]^T$  can be calculated by Eq. (1). 282

#### 2.3. Dynamic gravity compensation 283

During the assembly process, the moving platform of the SP, 284 assembly fixtures, and components are relatively heavy and 285 bulky, so their barycenter and gravity deviations, which are 286 caused by manufacturing errors and installation errors, will 287 lead to wrong calculation results of the six-dimensional F/T. 288 Additionally, during the measurements, the adjustable motions 289 of the six limbs would also lead to the coordinate changes of 290 barycenter in  $o_0$ - $x_0y_0z_0$  and the direction changes of gravity 291 in  $o_1$ - $x_1y_1z_1$ . To ensure the accuracy of the proposed analytical 292 algorithm in Section 2.2, dynamic gravity compensation is 293 needed. 294

#### 295 2.3.1. Compensation model

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The influential factors of the calculation results, which are the 206 barycenter and gravity of the moving platform of the SP, 207 assembly fixtures, and other components, cannot be ignored. 298 This paper considers them as a rigid system. Eq. (9) can thus 299 be rewritten as follows: 300 301

$$\boldsymbol{F} + \begin{bmatrix} \boldsymbol{S}_{\mathrm{G}} \\ \boldsymbol{S}_{\mathrm{0G}} \end{bmatrix} \boldsymbol{W} = \boldsymbol{G}\boldsymbol{f} \tag{13}$$

where W is the dimensionless value of the gravity, so it is not a 304 305 vector.  $S_{G}$  is the gravity unit vector of the said rigid system, so it is a 3-column vector.  $S_{0G}$  is the torque vector of  $S_G$  with 306 respect to  $o_1$ - $x_1y_1z_1$ , so  $S_{0G}$  is also a 3-column vector. 307

When the external load F = 0, the six-dimensional F/T is caused by the gravity of the rigid system (as in Fig. 3). The coordinate  $C = [x, y, z]^{T}$  indicates the barycenter of the rigid 310 system in  $o_1$ - $x_1y_1z_1$ . The gravity is divided into forces, along the  $x_1$ -,  $y_1$ -, and  $z_1$ -axis ( $F_x$ ,  $F_y$ , and  $F_z$ ), and torques, about 312 the  $x_1$ -,  $y_1$ -, and  $z_1$ -axis ( $M_x$ ,  $M_y$ , and  $M_z$ ), simultaneously. The relation between three-dimensional forces and three-314 dimensional torques is as follows: 315 316

$$\begin{bmatrix} M_x \\ M_y \\ M_z \end{bmatrix} = \begin{bmatrix} 0 & -z & y \\ z & 0 & -x \\ -y & x & 0 \end{bmatrix} \begin{bmatrix} F_x \\ F_y \\ F_z \end{bmatrix}$$
(14)

When the P&O of the moving platform changes,  $F_x$ ,  $F_y$ ,  $F_z$ , 319  $M_x$ ,  $M_y$ , and  $M_z$  also change, while satisfying Eq. (14). 320 According to the least square principle, C and W can be solved 321 by the six-dimensional F/T under three different P&O sets. 322 323 The accuracy of the compensation model can be improved by more measurements under different P&O sets. As an exam-324 ple, four measurements are performed here: 325 326

$$\begin{bmatrix} M_{x1} & M_{x2} & M_{x3} & M_{x4} \\ M_{y1} & M_{y2} & M_{y3} & M_{y4} \\ M_{z1} & M_{z2} & M_{z3} & M_{z4} \end{bmatrix}$$

$$= \begin{bmatrix} 0 & -z & y \\ z & 0 & -x \\ -y & x & 0 \end{bmatrix} \begin{bmatrix} F_{x1} & F_{x2} & F_{x3} & F_{x4} \\ F_{y1} & F_{y2} & F_{y3} & F_{y4} \\ F_{z1} & F_{z2} & F_{z3} & F_{z4} \end{bmatrix}$$
(15)

The resolving process of C and W from Eq. (15) is similar generalized inverse that of the matrix to of  $\begin{bmatrix} F_{x1} & F_{x2} & F_{x3} & F_{x4} \\ F_{y1} & F_{y2} & F_{y3} & F_{y4} \\ F_{z1} & F_{z2} & F_{z3} & F_{z4} \end{bmatrix}$  $F_{x1}$   $F_{x2}$   $F_{x3}$   $F_{x4}$  $\begin{bmatrix} F_{x1} & F_{x2} & F_{x3} & F_{x4} \\ F_{y1} & F_{y2} & F_{y3} & F_{y4} \\ F_{z1} & F_{z2} & F_{z3} & F_{z4} \end{bmatrix}$ Since is a



Schematic diagram of the gravity of the rigid system in  $o_1$ -Fig. 3  $x_1y_1z_1$ .

matrix consisting of real numbers, its generalized inverse matrix is of unique existence. When the P&O changes, C and W of the rigid system do not differ in  $o_1$ - $x_1y_1z_1$ , and the direction  $\mathbf{S} = [0, 0, -1]^{\mathrm{T}}$  of the gravity does not vary in  $o_0 \cdot x_0 y_0 z_0$ , meaning:

$$\begin{cases} S_{\rm G} = \frac{R^{-1} \cdot S}{\|R^{-1} \cdot S\|} \\ S_{\rm BC} = C \times S_{\rm C} \end{cases}$$
(16)

which could serve for the solution of Eq. (13).

For the preparation of a six-dimensional F/T measurement, experiments without external loads were carried out first, and the measured F/Ts could be used for the calculations of C and W using Eq. (15), after which C was substituted into Eq. (16) for the vector  $[S_G, S_{0G}]^T$ . Then, for an arbitrary external load, the six-dimensional F/T in  $o_1$ - $x_1y_1z_1$  could be obtained using Eq. (13) and the gravity of the rigid system could be dynamically compensated.

#### 2.3.2. Parameter optimization

Without any external load, the measurement of the sixdimensional F/T should be equal to  $[0, 0, 0, 0, 0, 0]^{T}$  in  $o_{1}$  $x_1y_1z_1$ . However, the analytical algorithm is affected by the gravity of the rigid system, resulting in errors for actual measurements, which must be compensated. According to Eq. (15), the accuracy of the model could be more efficiently compensated and improved by measurements under additional different P&O sets. The determination of the P&O number is essential for efficient dynamic gravity compensation.

Following the instructions of the Monte Carlo method, nP&O sets were selected for experimental verification. Each set was repeated 200 times measurements, and the average values of the limb lengths and forces were obtained. A sixdimensional F/T can be calculated for each P&O and can be used for the calculations of C and W using Eq. (15). Substituting C into Eq. (16), for the vector  $[S_G, S_{0G}]^T$ , the gravity of the rigid system can be dynamically compensated by using Eq. (13). The designed P&O adjusting platform applied in dynamic gravity compensation is presented in Fig. 4. The parameters of the P&O adjusting platform are listed in Table 1 and the graphical user interface (GUI) of data acquisition for dynamic gravity compensation is presented in Fig. 5.

For the determination of the proper selection of the P&O n number, another 50 P&O sets were selected, and the sixdimensional F/T after dynamic compensation could be



Fig. 4 Designed P&O adjusting platform applied in dynamic gravity compensation.

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Table 1 I	Parameters	of the	P&O	adjusting	platform.
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Value				
$\pm 50 \text{ mm}$				
$\pm$ 50 mm				
$\pm$ 50 mm				
$\pm 5^{\circ}$				
$\pm 5^{\circ}$				
$\pm 5^{\circ}$				
800 kg				
0.05 mm				
$1.5\ m\times 1.5\ m\times 1\ m$				



Fig. 5 GUI of data acquisition for dynamic gravity compensation.

obtained. The fluctuations between the compensated F/T and 375 376  $[0, 0, 0, 0, 0, 0]^{T}$  were calculated. For measurements repeating *n* times (n = 3, 4, ..., 24), the fluctuation range of the error 377 is  $3\sigma$ , where  $\sigma$  is the standard deviation. Experimental data 378 379 are illustrated in Fig. 6.

From Fig. 6, it is noteworthy that the fluctuations after 380 compensation were reduced with a higher n from the torques 381 382 errors, indicating an enhanced compensation effect. The  $F_{r}$ ,  $F_{v}$ , and  $F_{z}$  fluctuations show that the standard deviations vary 383 along with n, yet within a small overall range, implying that the 384 dynamic compensation of forces is of high stability and credi-385 bility. Regarding the  $M_x$ ,  $M_y$ , and  $M_z$  fluctuations, the stan-386 dard deviations decrease when n increases. In particular, the 387

deviations of n = 6 have been significantly reduced compared to those of n = 3, indicating high converging rates of  $M_x$  and  $M_{\nu}$ .

For a stable and efficient compensation, the P&O number was selected to be 18 for the following experiments.

### 3. Spatial precision analysis of the six-dimensional F/T

The force control techniques take the magnitude and direction 394 of the measured F/T into consideration: hence, by measuring 395 the six-dimensional F/T, the magnitude and direction of the 396 measured F/T could be illustrated as an approximate cone 307 shape, as demonstrated in Fig. 7, which is utilized to evaluate 398 the accuracy and repeatability of the six-dimensional F/T in the spatial precision analysis.

In an arbitrary coordinate system, six-dimensional F/T accuracy represents the deviation between an expected sixdimensional F/T and the average value of the measured F/T. The spatial precision standard in Fig. 7 can be described using the following parameters:

- Force direction accuracy  $A_{\rm FD}$ : the angle between the 406 expected direction and the central direction of measurements.
- Force magnitude accuracy  $A_{\rm FM}$ : the difference between the expected magnitude and the average magnitude of 410 measurements. 411



Fig. 7 Approximate cone shape for spatial precision analysis.



Fig. 6 Fluctuation analysis of the six-dimensional F/T after dynamic compensation.

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In an arbitrary coordinate system, six-dimensional F/T repeatability stands for the variation in measurements for one expected six-dimensional F/T, which can be expressed by the following parameters:

- Force direction repeatability  $R_{\rm FD}$ : half of the apex angle of the cone, formed by the measured directions.
- 421 Force magnitude repeatability  $R_{\rm FM}$ : spread of magnitude 422  $\pm 3S_{\rm FM}$  about the mean value  $\overline{m}$ , where  $S_{\rm FM}$  is the standard 423 deviation.
- Torque repeatability  $R_{MX}$ ,  $R_{MY}$ ,  $R_{MZ}$ : spreads of torques  $\pm 3S_{MX}$ ,  $\pm 3S_{MY}$ ,  $\pm 3S_{MZ}$  regarding the mean values  $\overline{M_x}$ ,  $\overline{M_y}$ ,  $\overline{M_z}$ , where  $S_{MX}$ ,  $S_{MY}$ ,  $S_{MZ}$  are the standard deviations, respectively (as  $R_{MY}$  in Fig. 7).

## 428 3.1. Force direction and magnitude accuracy

Let  $\overline{F_x}$ ,  $\overline{F_y}$ ,  $\overline{F_z}$  be the directional vectors of the center of the direction cluster for measurements that are repeated for *n* times,  $F_{xc}$ ,  $F_{yc}$ , and  $F_{zc}$  the directional vectors of the expected force, and  $F_{xj}$ ,  $F_{yj}$ , and  $F_{zj}$  the directional vectors of the *j*th measurement.

434 Then, the force direction accuracy  $A_{\rm FD}$  and the force mag-435 nitude accuracy  $A_{\rm FM}$  can be calculated as follows:

$$A_{\rm FD} = \frac{\sqrt{(\overline{F_x} - F_{xc})^2 + (\overline{F_y} - F_{yc})^2 + (\overline{F_z} - F_{zc})^2}}{\sqrt{F_{xc}^2 + F_{yc}^2 + F_{zc}^2}}$$
(17)

$$_{1} \qquad A_{\rm FM} = \sqrt{\overline{F_{x}}^{2} + \overline{F_{y}}^{2} + \overline{F_{z}}^{2}} - \sqrt{F_{xc}^{2} + F_{yc}^{2} + F_{zc}^{2}}$$

442 where

$$\begin{cases}
\overline{F_x} = \frac{1}{n} \sum_{j=1}^{n} F_{xj} \\
\overline{F_y} = \frac{1}{n} \sum_{j=1}^{n} F_{yj} \\
\overline{F_z} = \frac{1}{n} \sum_{j=1}^{n} F_{zj}
\end{cases}$$
(19)

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# 446 *3.2. Torque accuracy*

Let  $\overline{M_x}$ ,  $\overline{M_y}$ ,  $\overline{M_z}$  be the mean values of the torques measurements that are repeated for *n* times,  $M_{xc}$ ,  $M_{yc}$ ,  $M_{zc}$  the expected torques, and  $M_{xj}$ ,  $M_{yj}$ ,  $M_{zj}$  the torques of the *j*th measurement, respectively. The torque accuracy  $A_{MX}$ ,  $A_{MY}$ ,  $A_{MZ}$ can be expressed as follows:

$$\begin{cases}
A_{MX} = \overline{M_x} - M_{xc} \\
A_{MY} = \overline{M_y} - M_{yc} \\
A_{MZ} = \overline{M_z} - M_{zc}
\end{cases}$$
(20)

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$$\begin{cases} \overline{M_x} = \frac{1}{n} \sum_{j=1}^{n} M_{xj} \\ \overline{M_y} = \frac{1}{n} \sum_{j=1}^{n} M_{yj} \\ \overline{M_z} = \frac{1}{n} \sum_{j=1}^{n} M_{zj} \end{cases}$$
(21)

#### 3.3. Torque accuracy

With the aforementioned  $\overline{F_x}$ ,  $\overline{F_y}$ ,  $\overline{F_z}$  and  $F_{xj}$ ,  $F_{yj}$ ,  $F_{zj}$ , the force direction repeatability  $R_{\rm FD}$  and the force magnitude repeatability  $R_{\rm FM}$  can be defined as follows:

$$R_{\rm FD} = \overline{\theta_{\rm d}} + 3S_{\rm FD} \tag{22}$$

$$R_{\rm FM} = \pm 3S_{\rm FM} \tag{23}$$

where

$$\overline{\theta_{d}} = \frac{1}{n} \sum_{j=1}^{n} \theta_{dj}$$

$$\theta_{dj} = \frac{\sqrt{(F_{xj} - \overline{F_{x}})^{2} + (F_{yj} - \overline{F_{y}})^{2} + (F_{zj} - \overline{F_{z}})^{2}}}{\sqrt{F_{x}^{2} + \overline{F_{y}^{2}} + \overline{F_{z}^{2}}}}$$

$$S_{FD} = \sqrt{\frac{1}{n-1} \sum_{j=1}^{n} (\theta_{dj} - \overline{\theta_{d}})^{2}}$$

$$S_{FM} = \sqrt{\frac{1}{n-1} \sum_{j=1}^{n} \left(\sqrt{F_{xj}^{2} + F_{yj}^{2} + \overline{F_{zj}^{2}}} - \sqrt{\overline{F_{x}}^{2} + \overline{F_{y}}^{2} + \overline{F_{z}^{2}}}\right)^{2}}$$
(24) 472

# 3.4. Torque accuracy

(18)

The torque repeatability  $R_{MX}$ ,  $R_{MY}$ ,  $R_{MZ}$  can be obtained by the following: 474 475 476

$$R_{MX} = \pm 3S_{MX} = \pm 3\sqrt{\frac{1}{n-1}\sum_{j=1}^{n} (M_{xj} - \overline{M}_{x})^{2}}$$

$$R_{MY} = \pm 3S_{MY} = \pm 3\sqrt{\frac{1}{n-1}\sum_{j=1}^{n} (M_{yj} - \overline{M}_{y})^{2}}$$
(25)

$$R_{\rm MZ} = \pm 3S_{\rm MZ} = \pm 3\sqrt{\frac{1}{n-1}\sum_{j=1}^{n}(M_{zj}-\overline{M_z})^2}$$
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4. Experimental results and discussion 479

#### 4.1. Measurement of a known applied load

The designed P&O adjusting platform is presented in Fig. 4. 481 To measure a known applied load, three steps have been fol-482 lowed in this paper. Firstly, a known six-dimensional F/T is 483 applied to the P&O adjusting platform. The through-hole of 484 the moving platform is used to hang the known load. In this 485 paper, the coordinates of the through-hole are [302.874, 486 -175.192, 89.64]<sup>T</sup> in  $o_1$ - $x_1y_1z_1$ , the known load is 15 kg, and 487 the P&O is  $\{0, 0, 0, 0, 0, 0\}$ . Hence, the value of the 488 six-dimensional F/T can be calculated by the proposed algo-489 rithm and be expressed as  $[0, 0, -150, 26278.8, 45431.1, 0]^{T}$ 490 in  $o_1$ - $x_1y_1z_1$ . Secondly, measuring the load 1000 times 491 repeatedly, under dynamic gravity compensation, 1000 mea-492 surement results of the six-dimensional F/T can be obtained. 493

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Thirdly, the analytical predictions of the proposed algorithm are presented in Fig. 8. The accuracy and repeatability analyses are listed in Tables 2 and 3, respectively. Finally, additional 1000 measurement results of the six-dimensional F/T are used to verify the validity of the calculation results and conclusions can be acquired.

From Fig. 8, it is noteworthy that the proposed algorithm 500 could accurately predict the forces and torques in consistency 501 with the theoretical values. Together with the experimental 502 data, the force direction accuracy  $A_{\rm FD}$  is 0.003 rad and the 503 force direction repeatability  $R_{\rm FD}$  is 0.107 + 3 × 0.048 rad; 504 the force magnitude accuracy  $A_{\rm FM}$  is 0.329 N and the force 505 magnitude repeatability  $R_{\rm FM}$  is  $\pm 3 \times 13.668$  N, which are 506 ideal for the six-dimensional F/T measurements. Comparisons 507 between Tables 2 and 3 could also lead to the conclusion that 508 the accuracy and repeatability of force were improved, com-509 pared to those of torque, which is attributed to the difference 510 in their physical properties. For the force measurements, the 511 512 errors could offset due to their directions. However, for the torque measurements, the deviations would be amplified by 513 the arm of force for experiments. 514

515 Measuring the load 1000 times repeatedly, under dynamic 516 gravity compensation, 1000 measurements of the six-517 dimensional F/T are all within the scopes of Tables 2 and 3. Thus, the method of the spatial precision analysis is considered as correct.

#### 4.2. Precision analysis of measuring the six-dimensional F/T 520

Dynamic gravity compensation is critical for high-precision assembly and serves as an efficient tool for preliminary calibration before actual measurements. The designed P&O adjusting platform based on an SP for F/T-driven alignment can adjust the P&O, and dynamically measure interactive forces. However, the measurement accuracy of the six-dimensional F/T is different under different P&O sets and the accuracy of dynamic gravity compensation presented in this paper could be affected by the gravity of the rigid system. In this section, experiments were carried out with the P&O adjusting platform and spatial accuracy analyses were provided. Different P&O parameters { $x, y, z, \alpha, \beta, \gamma$ } were discussed. A comparison between compensations with or without assembly fixtures was also presented.

The P&O number n = 18 has been selected based on Section 2.2. Under limb length variations, 18 groups of sixdimensional F/T were obtained, and the barycenter C and the gravity value W could be determined.



Fig. 8 Measurement results of a known applied load.

Table 3	Repeatability analysis of the six-dimensional F/T.				
Item	$R_{\rm FD}$ (rad)	$R_{\rm FM}$ (N)	$R_{\rm MX}$ (N·mm)	$R_{\rm MY}$ (N·mm)	$R_{\rm MZ}$ (N·mm)
Value	$0.107 + 3 \times 0.048$	$\pm3\times13.668$	$\pm 3 \times 3255.6$	$\pm 3  imes 3305$	$\pm 3  imes 3742.7$

Table 2	Accuracy analysis of the six-dimensional F/T.				
Item	$A_{\rm FD}$ (rad)	$A_{\rm FM}$ (N)	$A_{\rm MX}$ (N·mm)	$A_{\rm MY}$ (N·mm)	$A_{\rm MZ}$ (N·mm)
Value	0.003	0.329	9.689	-93.488	-94.210

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Algorithm and experiments of six-dimensional force/torque dynamic measurements based on a Stewart platform



(a) Forces distribution moving along the x, y, z directions



(b) Torques distribution moving along the x, y, z directions



Fig. 9 Experimental results of measuring the six-dimensional F/T without assembly fixtures.

# 4.2.1. Measuring the six-dimensional F/T without assembly fixtures

541 The P&O adjusting platform without assembly fixtures is pre-542 sented in Fig. 4. The P&O parameters of the moving platform were controlled for a motion of a single degree of freedom. 543 Without assembly fixtures, the gravity value of the rigid system 544 is 773.16 N (n = 18). The moving range in x, y, and z direc-545 tions is  $\pm 40$  with 10 mm variation for each measurement. 546 The angle range is  $\pm 5^{\circ}$  with 1° variation for each measure-547 ment. For every change in P&O parameters, 500 groups of 548 549 the six-dimensional F/T were measured. The average F/T and relevant results are illustrated in Fig. 9. 550

551 Fig. 9(a) and (b) displays the F/T distribution of the P&O 552 adjusting platform without assembly fixtures moving along 553 the x, y, and z directions. Fig. 9(c) and (d) displays the F/T distribution of the P&O adjusting platform without assembly fix-554 555 tures rotating around the x, y, and z directions. The location of 556 the ball in Fig. 9(a) represents the control position, while in Fig. 9(c) the control orientation. The color bar on the right 557 stands for the value of the measured force, with red areas being 558 higher. The arrow of the ball shows the direction of the mea-559 sured force. Fig. 9(b) and (d) presents the torque variations 560 with respect to the pose parameters. Details about the black 561 lines, markers, and line styles are listed in the legend. 562

From Fig. 9(a) and (c), it can be observed that when the P&O adjusting platform is moving along the x, y, and z directions without rotations, the deviations in the z direction are much smaller, compared to those in the other two directions. When the P&O adjusting platform is moving in the range of

 $x \ge 0$  or  $y \ge 0$ , the deviations are smaller, and the force mag-568 nitude accuracy  $A_{\rm FM}$  is 30–35 N. The measured deviations 569 increase when x decreases or y increases, and the biggest devi-570 ation is 60 N for x = -40 mm. When the P&O adjusting plat-571 form is rotating around the x, y, and z directions, the 572 deviations around the z direction are much smaller, compared 573 to those around the other two directions. The deviations are 574 smaller for  $\alpha \ge 0$  or  $\beta \le 0$ , and the force magnitude accuracy 575  $A_{\rm FM}$  is 30–35 N. The biggest deviation appears at  $\alpha = -5^{\circ}$  and 576 its force magnitude accuracy  $A_{\rm FM}$  is 65 N. The force vectors 577 are all heading to the -x and -y directions, which indicates 578 that a directional compensation could be made in the future 579 to improve the accuracy of the algorithm. 580

From Fig. 9(b) and (d), it can be observed that when the P&O adjusting platform is moving along the *x*, *y*, and *z* directions without rotations, the torque accuracy  $A_{MX}$  shows enhanced compensation with smaller deviations. When the P&O adjusting platform without assembly fixtures is rotating around the *x*, *y*, and *z* directions, the torque accuracy  $A_{MX}$  shows enhanced compensation with smaller deviations.

4.2.2. Measuring the six-dimensional F/T with assembly fixtures 588 The P&O adjusting platform with assembly fixtures is pre-589 sented in Fig. 10. The P&O parameters of the moving platform 590 were controlled for a single degree of freedom of motion. 591 Without assembly fixtures, the gravity value of the rigid system 592 is 1912.09 N (n = 18). The moving range in the x, y, and z 593 directions is  $\pm 30$ , with 10 mm variation for each measure-594 ment. The angle range is  $\pm 3^{\circ}$  with  $1^{\circ}$  variation for each mea-595

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Fig. 10 P&O adjusting platform with assembly fixtures.

surement. For every change in P&O parameters, 500 groups of 596 the six-dimensional F/T were measured. The average F/T and 597 598 relevant results are illustrated in Fig. 11.

Fig. 11(a) and (b) displays the F/T distribution of the P&O adjusting platform with assembly fixtures moving along the x, y, and z directions, while Fig. 11(c) and (d) displays the F/Tdistribution of the P&O adjusting platform with assembly fix-602 tures rotating around the x, y, and z directions. The location of the ball in Fig. 11(a) represents the control position, while in Fig. 11(c) the control orientation. The color bar on the right 605 stands for the value of the measured force, with red areas being higher. The arrow of the ball shows the direction of the measured force. Fig. 11(b) and (d) shows the torque variations with respect to the pose parameters. Details about the black lines, markers, and line styles are listed in the legend.

From Fig. 11(a) and (c), it can be observed that when the P&O adjusting platform with assembly fixtures is moving along the x, y, and z directions without rotations, the deviations in the z direction are much smaller, compared to those in the other two directions. When the P&O adjusting platform is moving in the range of  $x \ge 0$  or  $y \ge 0$ , the deviations are smaller, and the force magnitude accuracy  $A_{\rm FM}$  is 30–55 N. The measured deviations increase when x decreases or yincreases, and the highest deviation is 80 N for y = -15 mm. When the P&O adjusting platform is rotating around the x. v, and z directions, the deviations around the z direction are much smaller, compared to those of the other two directions. The deviations are smaller for  $\alpha \ge 0$  or  $\beta \le 0$ , and the force magnitude accuracy  $A_{\rm FM}$  is 30–55 N. The biggest deviation appears at  $\alpha = -3^{\circ}$  and its force magnitude accuracy  $A_{\rm FM}$  is 65 N. The force vectors are all heading to the -x and -y directions, which indicate that a directional compensation could be made in the future to improve the algorithm's accuracy.

From Fig. 11(b) and (d), it can be observed that when the 629 **P&O** adjusting platform is moving along the x, y, and z direc-630 tions without rotations, the torque accuracy  $A_{MX}$  shows 631 enhanced compensation with small deviations. When the 632 P&O adjusting platform with assembly fixtures is rotating 633



Fig. 11 Experimental results of measuring the six-dimensional F/T with assembly fixtures.

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around the x, y, and z directions, the torque accuracy  $A_{MX}$ shows enhanced compensation with small deviations as well.

636 Comparisons between Sections 4.1 and 4.2 lead to the conclusion that the measurement accuracy of the six dimensional 637 F/T is different under different P&O sets. Therefore, a prefer-638 able operation range of the P&O adjusting platform can be 639 selected for high-precision assembly with smaller deviations. 640 The accuracy of measuring the six-dimensional F/T with 641 assembly fixtures is inferior to that observed without assembly 642 fixtures, and the deviations are higher. This is attributed to the 643 644 weight difference between the assembly fixtures and the moving platform. In this case, the assembly fixtures are twice as 645 646 heavy as the moving platform, which lowers the accuracy of 647 dynamic gravity compensation as well as the coupling effect of the six-dimensional F/T in the analytical algorithm. 648

# 649 4.3. F/T-driven alignment of large components

Experiments of the F/T-driven alignment for large compo-650 nents were performed on the designed digital flexible alignment 651 system using aerospace products (as in Fig. 12), and the align-652 ment process can be described by the flowchart shown in 653 Fig. 13. Firstly, according to the precision analysis results of 654 655 Section 4.2, the threshold value of the six-dimensional F/T656 could be obtained. In the experiment, the threshold value of 657 the force magnitude was 60 N, and the threshold value of the torque magnitude was 25 N·m. Secondly, an operator 658 judged the direction of the applied force by visual, and then 659 an external force was applied to the P&O adjusting platform 660 by the operator. Thirdly, the six-dimensional F/T was calcu-661 lated in real time by dynamic gravity compensation. Fourthly, 662 intention recognition methods were designed through the 663 threshold value, the direction and magnitude of the force, 664 and the torque. In the end, the P&O adjusting platform 665 adjusted the P&O of the large component to follow the inten-666 tions of the operator. 667

In the above experiments, the alignment process was completed successfully. The experimental results proved that the precision analysis of the six-dimensional F/T was correct and effective, and the intention recognition was correct. The alignment process met the real-time requirements. The analytical algorithm and precision analysis of the six-dimensional F/T



Fig. 12 Alignment system of large components for F/T-driven assembly.



Fig. 13 Flowchart of the F/T-driven alignment.

based on the P&O adjusting platform laid the foundation for F/T-driven alignment of large components. 674

# 5. Conclusions

The P&O adjusting platform can dynamically measure interac-677 678 tive forces. This paper provides an analytical algorithm of the interaction forces between components and takes into consid-679 eration dynamic gravity deviations as influential factors. The 680 relevant experimental results show that the proposed analytical 681 algorithm can evaluate gravity deviations and make reliable 682 compensations. The contributions of the paper are summa-683 rized as follows: 684

- (1) An analytical algorithm of the six-dimensional F/T based on the screw theory is proposed for accurate determination of external forces during high-precision alignment. Dynamic gravity deviations are taken into consideration and a precise compensation model is provided. Barycenter coordinates and gravity directions are discussed in details. Meanwhile, the choice of the P&O number is optimized for a stable and efficient compensation through experiments.
- (2) An approximation cone shape is used for spatial precision analysis. Given the specific appearance of the repeated six-dimensional F/T measurements, the magnitudes and directions of the measured F/T could be evaluated by a set of standards, regarding accuracy and repeatability.
- (3) Known applied load measurement experiments have 700 been performed on the P&O adjusting platform based 701 on an SP for F/T-driven alignment, and relevant exper-702 imental data adequately prove that the proposed analyt-703 ical algorithm could accurately predict the F/T with 704 small deviations. Precision analysis experiments have 705 been performed on the P&O adjusting platform (without 706 or with assembly fixtures), and relevant experimental 707 data adequately prove that the measurement accuracy 708 of the six-dimensional F/T is different under different 709 P&O sets. Higher loads lead to poorer accuracy of 710

- dynamic gravity compensation. In addition, the preferable operation range is discussed for high-precision alignment with smaller deviations. Based on the above analysis, the experiments of F/T-driven alignment for large components have been completed successfully.
- (4) Interactive force measurements are novel and significant 716 for high-precision assembly, and the present algorithm 717 could fulfill accurate force determination and provide 718 satisfactory dynamic gravity compensation. Measuring 719 the six-dimensional F/T could be further improved with 720 higher motion control of the moving platform or more 721 722 accurate measurements of forces or limb lengths. 723 Besides, the coupling effect for the P&O parameters, varying in synchronization and force control techniques, 724 725 should be studied in future research.
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## 732 Appendix A. Supplementary material

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### 736 References

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765

- 1. Mei ZY, Maropoulos PG. Review of the application of flexible, measurement-assisted assembly technology in aircraft manufacturing. *Proc IME B J Eng Manuf* 2014;228(10):1185–97.
  - Chen ZH, Du FZ, Tang XQ. Research on uncertainty in measurement assisted alignment in aircraft assembly. *Chin J Aeronaut* 2013;26(6):1568–76.
  - Yao R, Zhu WB, Huang P. Accuracy analysis of Stewart platform based on interval analysis method. *Chin J Mech Eng* 2013;26 (1):29–34.
  - 4. Rosenzveig V, Briot S, Martinet P, Ozgur E, Bouton N. A method for simplifying the analysis of leg-based visual servoing of parallel robots*IEEE international conference on robotics & automation* (*ICRA*), May 31-June 7, Hong Kong, China.
  - Pedrammehr S, Mahboubkhah M, Khani N. A study on vibration of Stewart platform-based machine tool table. *Int J Adv Manuf Technol* 2013;65(5):991–1007.
  - Denkenaa B, Holza C, Abdellatifb H. Model-based control of a hexapod with linear direct drives. *Int J Comput Integr Manuf* 2006;19(5):463–72.
  - Dalvand MM, Shirinzadeh B. Motion control analysis of a parallel robot assisted minimally invasive surgery/microsurgery system (PRAMiSS). *Robot Comput-Int Manuf* 2013;29(2):318–27.
  - 8. Tang F. Development of an engineering simulator for armored vehicleInternational conference on automation, mechanical control and computational engineering, Apr 24–26, Jinan, China.
  - Pisla A, Itul T, Pisla D, Szilaghyi A. Considerations upon the influence of manufacturing and assembly errors on the kinematic and dynamic behavior in a flight simulator Stewart-Gough platform. *Mech, Transm Appl: Mech Mach Sci* 2012;3:215–23.
- 10. Lochte C, Dietrich F, Raatz A. A parallel kinematic concept targeting at more accurate assembly of aircraft sections. *Intell Robot Appl* 2011;7101:142–51.

- 11. Xu YF, Yuan JR, Zhao J, Zhao YB. Robust attitude control and simulation of a Stewart spacecraft *The 27th Chinese control and decision conference, May 23–25, Qingdao, China.*
- 12. Zhao H, Zhang SY, Chen XD. Compliant force control in space docking*Proceedings of the 2007 IEEE international conference on mechatronics and automation, Aug 5–8, Harbin, China.*
- 13. Zhang GQ, Du JJ, To S. Calibration of a small size hexapod machine tool using coordinate measuring machine. *Proc IME E J Process Mech Eng* 2014.
- Zhou WY, Chen WY, Liu HD. A new forward kinematic algorithm for a general Stewart platform. *Mech Mach Theory* 2015;87:177–90.
- Jamshidi J, Kayani A, Iravani P, Summers MD. Manufacturing and assembly automation by integrated metrology systems for aircraft wing fabrication. *Proc IME B J Eng Manuf* 2010;**224** (1):25–36.
- Chen ZH, Du FZ, Tang XQ, Zhang X. A framework of measurement assisted assembly for wing-fuselage alignment based on key measurement characteristics. *Int J Manuf Res* 2015;10 (2):107–28.
- 17. Zheng LY, Zhu XS, Liu RW, Wang YW, Maropoulos PG. A novel algorithm of posture best fit based on key characteristics for large components assembly. *Procedia CIRP* 2013;**10**:162–8.
- Galetto M, Mastrogiacomo L. Analysing uncertainty contributions in dimensional measurements of large-size objects by ultrasound sensors. *Int J Comput Integr Manuf* 2014;27(1):36–47.
- Ferria C, Mastrogiacomob L, Farawayc J. Sources of variability in the set-up of an indoor GPS. *Int J Comput Integr Manuf* 2010;23 (6):487–99.
- Muelaner JE, Cai B, Maropoulos PG. Large-volume metrology instrument selection and measurability analysis. *Proc IME B J Eng Manuf* 2010;224(6):853–68.
- Yao JT, Zhang HY, Zhu JL, Xu YD, Zhao YS. Isotropy analysis of redundant parallel six-axis force sensor. *Mech Mach Theory* 2015;91:135–50.
- Yao JT, Li WJ, Zhang HY, Xu YD, Zhao YS. Task-oriented design method and experimental research of six-component force Sensor. *Intell Robot Appl* 2014;8917:1–12.
- Dwarakanath TA, Bhutani G. Beam type hexapod structure based six component force-torque sensor. *Mechatronics* 2011;21 (8):1279–87.
- Liu W, Li Q, Jia ZY, Jiang E. Design and experiment of a parallel six-axis heavy force sensor based on Stewart structure. *Sensors Transd* 2013;151(4):54–62.
- Kim YL, Song HC, Song JB. Hole detection algorithm for chamferless square peg-in-hole based on shape recognition using F/T sensor. Int J Prec End Manuf 2014;15:425–32.
- Jasim IF, Plapper PW. Contact-state monitoring of force-guided robotic assembly tasks using expectation maximization-based Gaussian mixtures models. *Int J Adv Manuf Technol* 2014;73 (5):623–33.
- Shirinzadeh B, Zhong Y, Tilakaratna PDW, Tian YL, Dalvand MM. A hybrid contact state analysis methodology for roboticbased adjustment of cylindrical pair. *Int J Adv Manuf Technol* 2011;**52**(1):329–42.
- 28. Park DI, Park C, Do H, Choi T, Kyung JH. Assembly phase estimation in the square peg assembly process *The 12th international conference on control, automation and systems, Oct 17–21, JeJu Island, Korea.*
- 29. Wiemer SC, Schimmels JM. Optimal admittance characteristics for planar force-assembly of convex polygonal parts2012 IEEE international conference on robotics and automation, May 14–18, Saint Paul, MN, USA.
- Bera TK, Merzouki R, Bouamama BO, Samantaray AK. Force control in a parallel manipulator through virtual foundations. *Proc IME 1 J Syst Control Eng* 2012;**226**(8):1088–106.

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833

Algorithm and experiments of six-dimensional force/torque dynamic measurements based on a Stewart platform

- 31. Qiao H, Dalay BS, Parkin RM. Robotic peg-hole insertion
   operations using a six-dimensional force sensor. *Proc IME C J Mech Eng Sci* 1993;207:289–305.
- 838 32. Geng MC, Zhao TS, Wang C, et al. The study of the direct
  position analysis of parallel mechanism based on quasi-newton
  840 method. *J Mech Eng* 2015;51(9):28–36 [in Chinese].

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