

# MFH Top-Half Ring Connectors

D-Zero Engineering Note #3740.225-EN-195

Rob Nachtrieb  
10-24-88

Approved by Keith Primdahl

Keith Primdahl

The purpose of the test was to experimentally verify that the connector scheme designed for the top half of the MH ring (end calorimeter) was adequate for the expected structural loads. The test proved that the design was acceptable.

### **Background**

In the end calorimeter of the D-zero experiment, uranium and stainless steel plates are assembled into modules which are to be installed into the cryostat as individual units. A single inner module will be surrounded by 16 middle modules, which are surrounded by 16 outer modules. The 16 modules which comprise the middle ring are to be connected at the outer radius by a link, which is pinned to each front plate. The expected forces at each of the 16 connections vary from 1400 lb to 69,000 lb (Appendix 1). At the inner radius, the forces are known to be compressive; hence, the edges of the front plates will be allowed to bear directly upon one another.

Since it is desirable to minimize the volume of material used, an Inconel connector plate and pins were chosen. Furthermore, it was observed that the forces between the 9 modules of the top half of the ring are all less than 7,700 lb. Accordingly, a connector was designed for the top half of the ring, with the intention that a separate design be performed for the bottom connections.

### **Test Setup**

We tested an Inconel 3" x 8" x .25" connector plate, a stainless steel cover plate, and a portion of a stainless steel front plate, using two Enerpac model RC-251 hydraulic cylinders. The cylinders were powered by the same pump, with an effective total area of 10.30 in<sup>2</sup>. The Inconel plate and pins had been solution annealed, then age-hardened (a heat-treatment) to realize the full capability of Inconel. This process will be described in more detail in a separate report. Dimensions were carefully measured before assembly.

The Inconel connector plate was assembled between the front plate and the stainless steel cover plate, and secured with a 0.75" Inconel pin. The cover plate was attached with two 0.625" Inconel pins and a single 0.5" stainless steel bolt (figure 1). To measure strain of the Inconel connector plate, four 0-60<sup>0</sup>-120<sup>0</sup> rosette strain-gages were employed. The movement of the pin end center points was measured with four separate surveyor's sighting

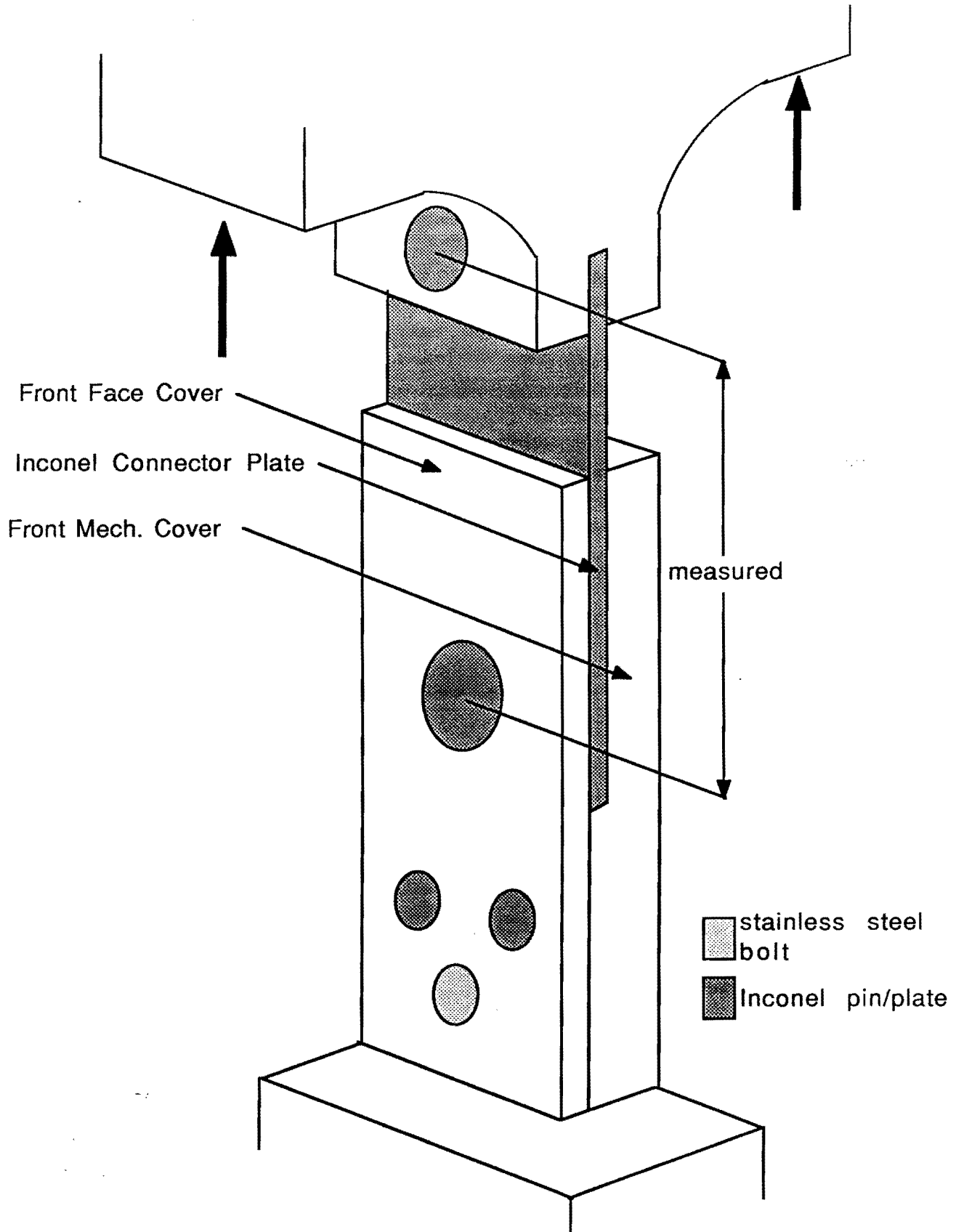


Figure 1: Test Set up

instruments, all calibrated to the same reference point. The difference between the pin heights was used to determine the elongation between the pins.

### Procedure

Starting at 400 psi, the load was increased by increments of 200 psi, recording the strain gage readings and elongation at each interval. The combined surface area of the hydraulic jacks was 10.30 in<sup>2</sup>; accordingly, each successive 200 psi increase generated an additional 2060 lb. Pressure was decreased in similar increments back down to 400 psi. At each increment, we measured the clearance between the bottom of the cover and front plates by using a feeler gage. Two cycles to 10,300 lb resulted in consistent data.

The load was then increased, in 200 psi increments, until permanent deformation was evident. Finally, the connector was disassembled and dimensions compared to those from before testing.

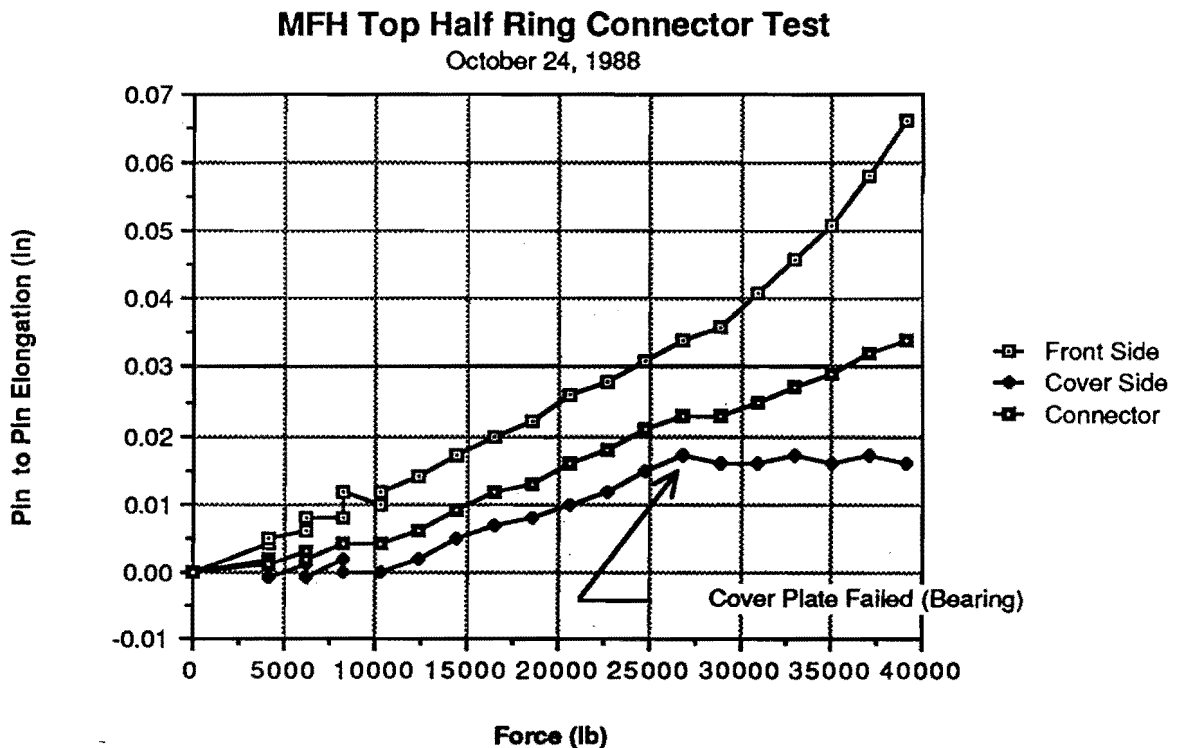


Figure 2: Pin to Pin Elongation

## Results

When tested to 1.25 times the design load, the connector deformed elastically, with nearly duplicate results on the second cycle. The cover plate moved a couple of mills with respect to the front plate, but did not completely return when unloaded. The cover plate slipped until the 0.625" pins came into full bearing, after which the deflection was elastic.

Failure occurred near 27,000 lb. In Figure 2, pin to pin elongation is plotted as a function of load. At the point of failure, the 0.75" pin began to deflect unevenly. The data for the cover plate side indicates that beyond 27,000 lb there is no further pin to pin elongation; in fact, the 0.75" pin was tearing its way through the cover plate. The front plate side's elongation subsequently appears to increase in slope; however, this was merely a result of the changing pitch of the pin.

The strain gages verified the the loading via the hydraulic cylinders was accurately measured. The Inconel performed elasticity as can be seen in figure 3. The permanent deformation of

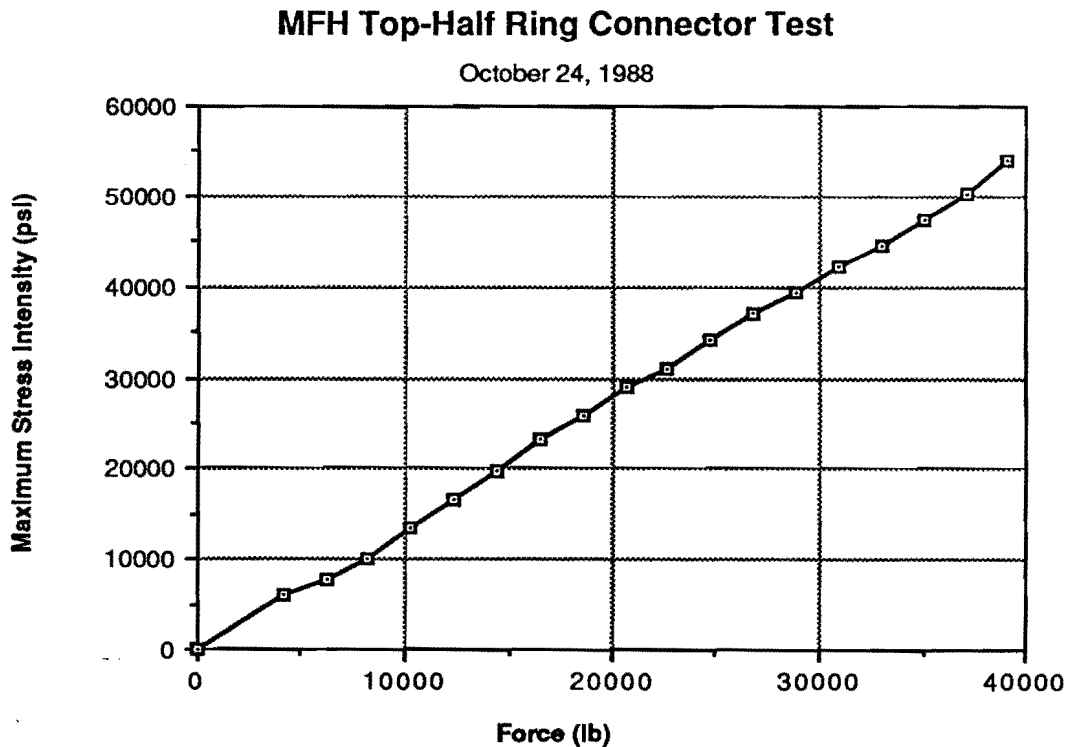


Figure 3: Inconel Link Stress Intensity

the parts (figure 4) shows that the greatest deformation occurred in the stainless steel cover plate about the 0.75" pin. A small local deformation occurred in the Inconel plate due to the twisting effect of the 0.075" pin.

### Conclusion

The connector design is acceptable for the loads expected between modules of the top half of the MH ring.

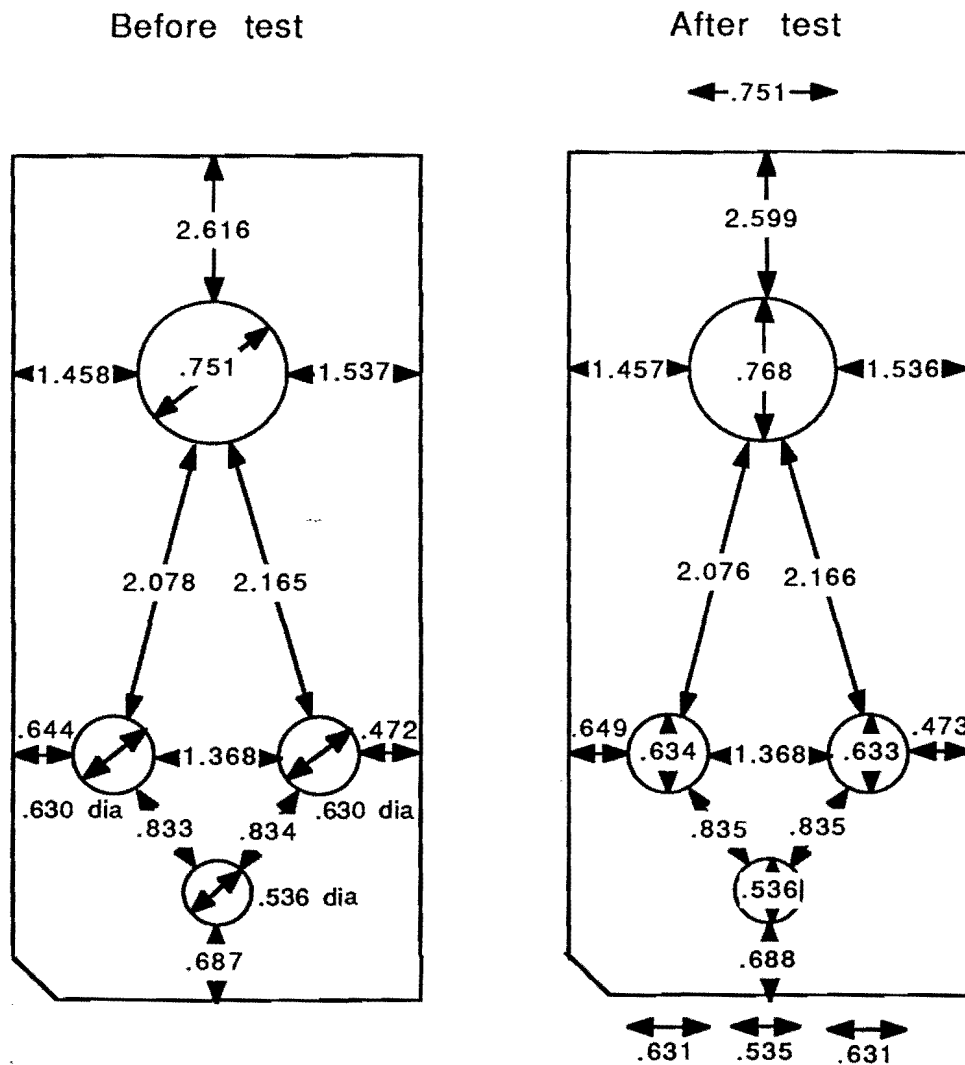


Figure 4: Cover Plate Dimensions

Appendix 1

Connector Force Calculations



CONCENTRATION	...	F <sub>0</sub> (200)	F <sub>0</sub> (200)	F <sub>5</sub> (200)
1-2	...	68.2 + 19.1μ	-48.2	2.6
2-3	...	64.1 + 1.0μ	-65.1	4.9
3-4	...	67.5 - 16.7μ	-58.4	5.9
4-5	...	4.1	-50.5	5.2
5-6	...	7.5	-28.6	4.2
6-7	...	6.8	-20.1	8.9
7-8	...	146.25	-10.0	8.9
8-9	...	168.75	-2.4	5.2



MH FLOOR FLOOR:

20 20 20

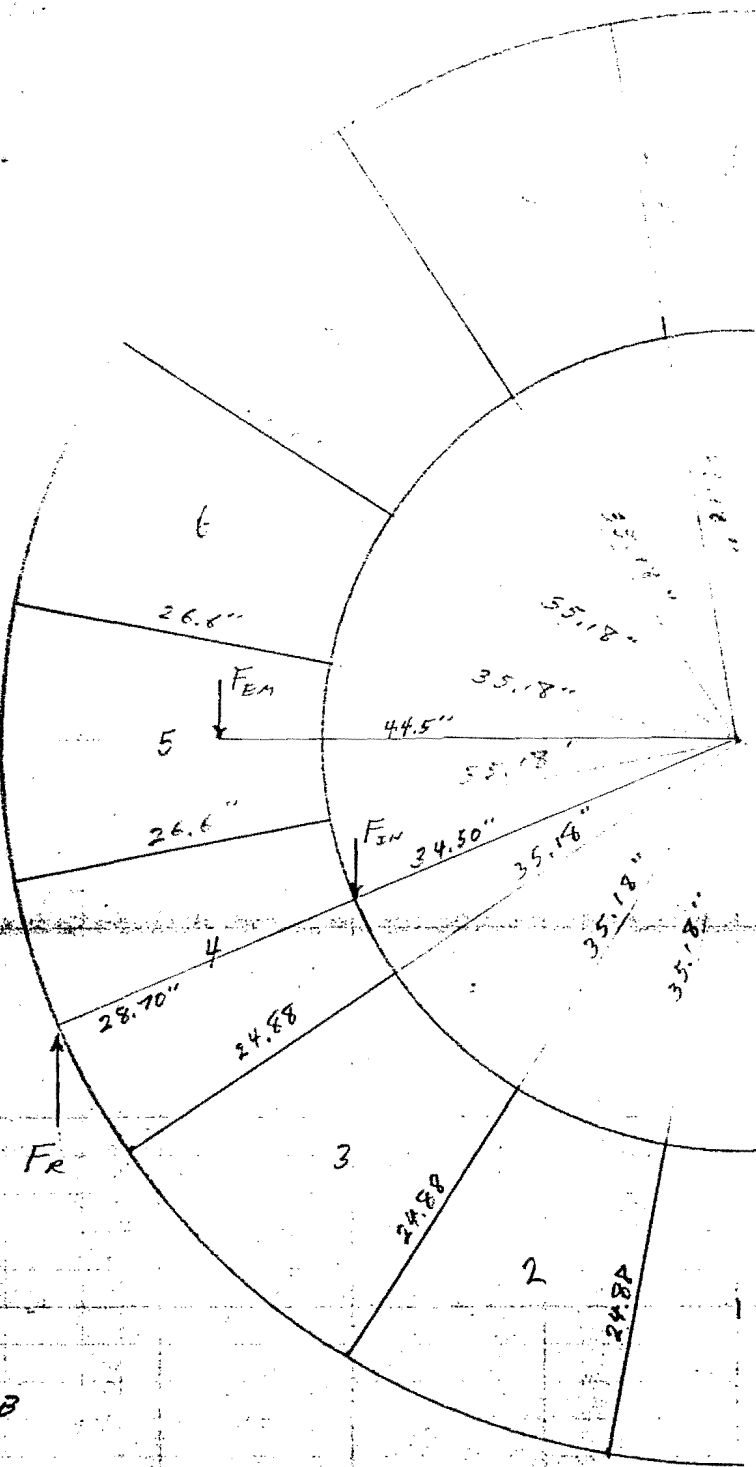
12 20

20 20 20 20 20

20 20 20 20 20

20 20 20 20 20

20 20 20 20 20



$$F_{EM} = 18,000 \text{ LB}$$

$$F_{EM} = 4,000 \text{ LB}$$

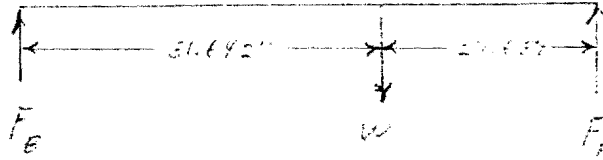
$$F_R = 18,000 \text{ LB} + 4,000 \text{ LB} + (8)(5340 \text{ LB}) = 64,720 \text{ LB}$$

5F



1000 LBS

MH



ASSUME  $W \leq 10,000 \text{ LB}$

$$\sum M_o = -(31.642 \text{ in})(10,000 \text{ LB}) - (59.28 \text{ in})(F_F) = 0$$

$$F_F = 5340 \text{ LB}$$

CENTROID LOCATIONS:

	$\bar{x}$ (in)	$\bar{y}$ (in)	$W$ (LB)
1	-4.80	-49.33	2670
2	-18.97	-45.79	5340
3	-35.06	-35.06	5340
4	-45.79	-18.97	5340
5	-49.56	0	5340
6	-45.79	18.97	5340
7	-35.06	35.06	5340
8	-18.97	45.79	5340
9	0	49.56	5340

COMPRESSION FORCE BETWEEN 2 & 3

$$F_{CA} = 5340 \text{ LB}$$

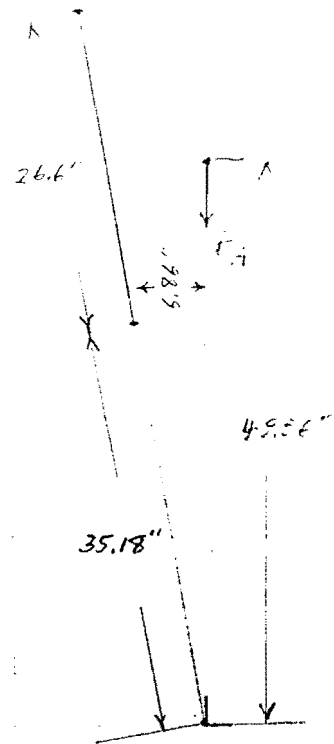
$F_{CB} = (5340 \text{ LB}) \sin 11.25^\circ$   
 $F_{CB} = 1040 \text{ LB}$

$$F_{CC} = 1377 \text{ LB (TENSION)}$$

$$F_{CD} = (1377 \text{ LB} + (5340 \text{ LB}) \sin 11.25^\circ)$$

$$F_{CD} = 2419 \text{ LB (COMPRESSION)}$$

$$F_{CS} = (5340 \text{ LB}) (\cos 11.25^\circ) = 5237 \text{ LB}$$



COMPRESSION MEMBER

$$L = \sqrt{10.0^2 + 9.42^2} = 13.6 \text{ m}$$

$$\bar{F}_c = \left[ \frac{98.56 + 95.75}{2} \right] L = 4717 \text{ m}$$

$$W = 10680 \text{ LB}$$

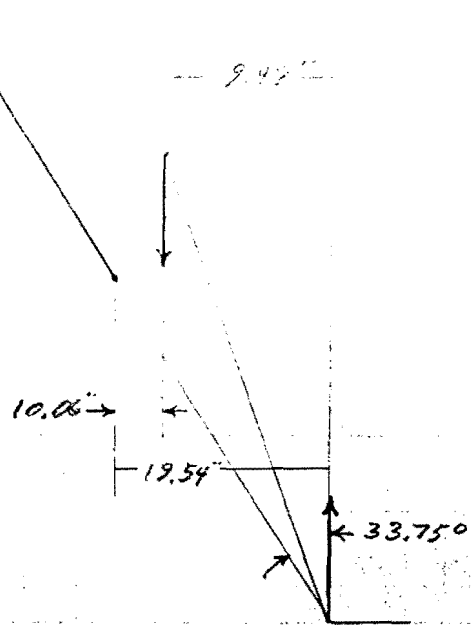
$$\sum M_{SR} = -(10.0 \text{ m})(10680 \text{ LB}) + (26.6 \text{ m})(F_{OR}) = 0$$

$$F_{OR} = 4039 \text{ LB TENSION}$$

$$F_{IR} = (-4039 \text{ LB}) - (10680 \text{ LB}) \sin 33.75^\circ$$

$$F_{IR} = -9972 \text{ LB COMPRESSION}$$

$$F_{IR} = (10680 \text{ LB}) \sin 33.75^\circ = 8880 \text{ LB}$$



CONCENTRIC FORCE

$$F = 16020 \text{ LB}$$

16020 LB

$$16020 \text{ LB}$$

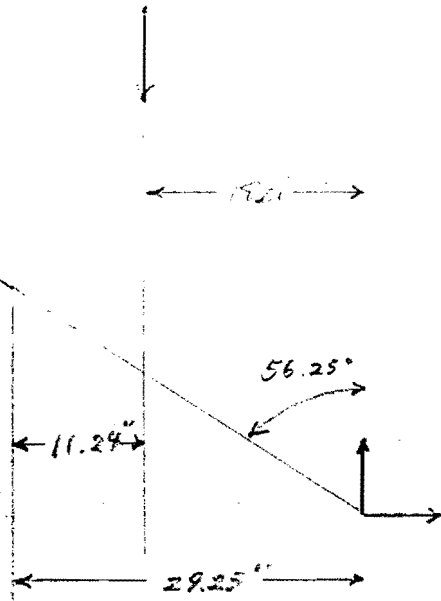
$$\sum M_{20} = -(11.24")(16020 \text{ LB}) + (26.6")(F_{CR}) = 0$$

$$F_{CR} = 6769 \text{ LB TENSION}$$

$$F_{JR} = -6769 \text{ LB} - (16020 \text{ LB}) \sin 56.25$$

$$F_{JR} = -20089 \text{ LB COMPRESSION}$$

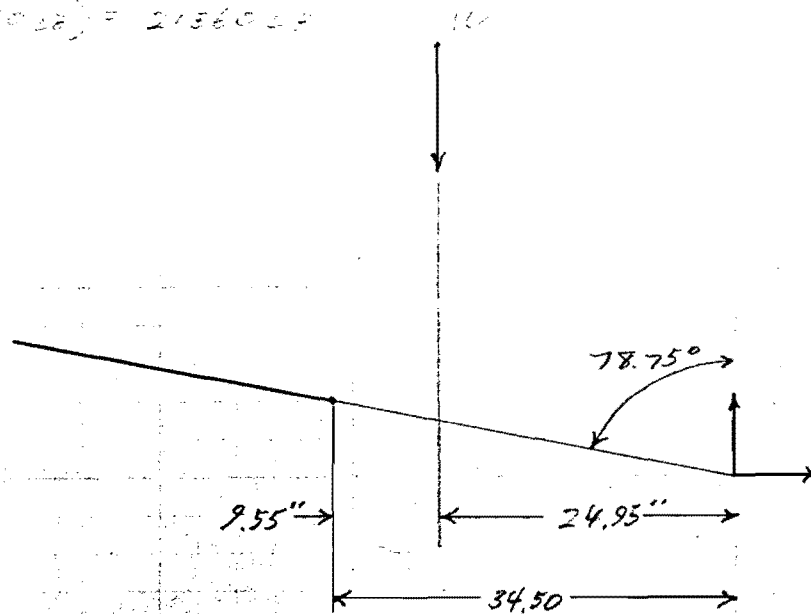
$$F_S = (16020 \text{ LB}) \cos 56.25 = 8900 \text{ LB}$$



MEMBER FORCES

$$\bar{x} = \left[ \frac{4(5.33)(26.6)}{4(5.33) + 26.6} \right] = 16.6 \text{ in}$$

$$W = (4)(5330 \text{ lb}) = 21360 \text{ lb}$$



$$\sum M_{SR} = (9.55 \text{ in})(21360 \text{ lb}) + (26.6 \text{ in})(F_{OR}) = 0$$

$$F_{OR} = 7669 \text{ lb} \quad \text{TENSION}$$

$$F_{FR} = -7669 \text{ lb} - (21360 \text{ lb}) \sin(78.75^\circ)$$

$$F_{FR} = -28619 \text{ lb} \quad \text{COMPRESSION}$$

$$F_S = (21360 \text{ lb}) \cos(78.75^\circ) = 4167 \text{ lb}$$

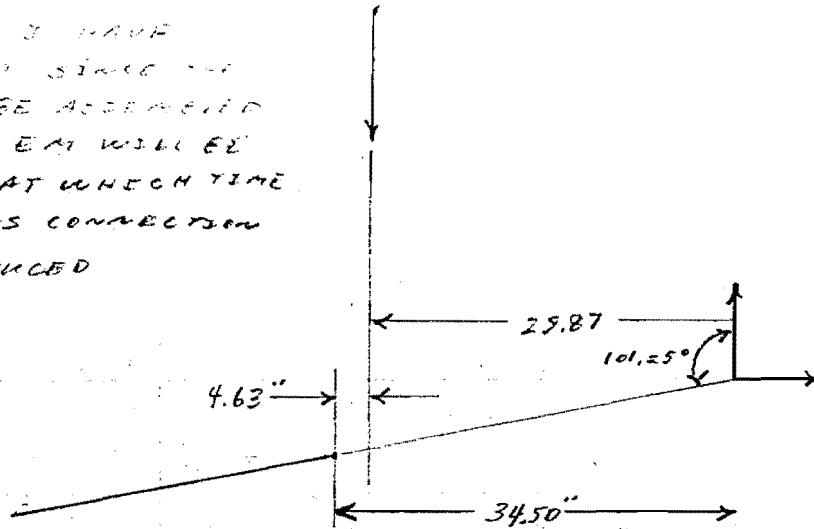
RECALCULATION FOR THE 2021

$$\bar{x} = -29.87 + 4.63 \sin(101.25^\circ) = -29.87 + 4.63(0.98) = -29.87 + 4.54 = -25.33$$

$$\bar{y} = 26.6 + 4.63 \cos(101.25^\circ) = 26.6 + 4.63(-0.17) = 26.6 - 0.79 = 25.81$$

$$W = (5)(19)(90 \text{ LB}) = 8550 \text{ LB}$$

NOTE: MORE I HAVE  
 LOADED EM SINCE THE  
 KING WILL BE ASSEMBLED  
 FIRST, THEN EM WILL BE  
 INSTALLED, AT WHICH TIME  
 FOR FOR THIS CONNECTION  
 WILL BE REDUCED



$$\sum M_{IR} = (-4.63'')(26700 \text{ LB}) + (26.6'')(F_{OR}) = 0$$

$$F_{OR} = 4647 \text{ LB}$$

$$F_{IR} = -4647 \text{ LB} - (26700 \text{ LB}) \sin(101.25^\circ) =$$

$$F_{IR} = -30834 \text{ LB}$$

$$F_S = (26700 \text{ LB}) \cos(101.25^\circ) = 5209 \text{ LB}$$

WITH EM AND IH (IF ATTACHED TO S)

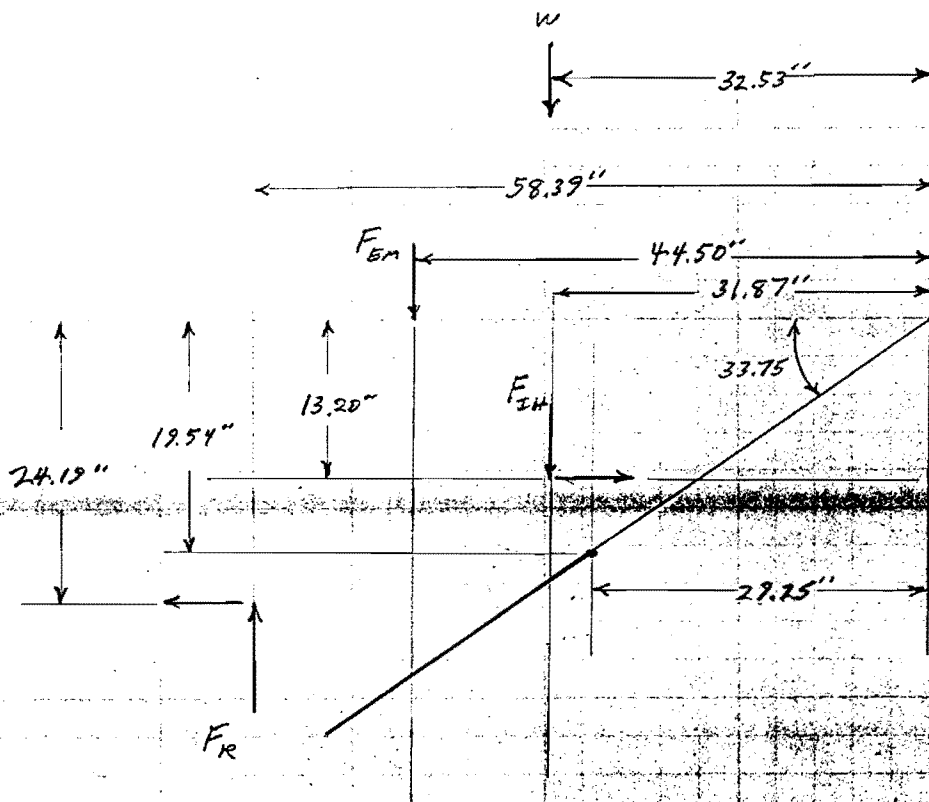
$$F_S = (26700 \text{ LB} + 4000 \text{ LB} + 19,000 \text{ LB}) \cos(101.25^\circ) = 9501 \text{ LB}$$

COMPOSITE SECTION 344

$$T = \left[ -45.79 - 49.24 - 45.79 - 55.06 - 18.91 - 0 \right] \frac{1}{6} = -52.53 \text{ in}$$

$$D = \left[ 19.51 + 0 + 17.97 + 55.06 + 45.79 + 45.56 \right] \frac{1}{6} = 21.75 \text{ in}$$

$$W = (6 \text{ in} \times 21.75) = 32040 \text{ LB}$$



$$\sum M_{IR} = (-3.28") (32040 \text{ LB}) + (15.25") (4000 \text{ LB}) - (24.87") (F_{OR}) + (2.62") (18,000 \text{ LB}) - (29.14") (64,720 \text{ LB}) = 0$$

$$F_{OR} = 67230 \text{ LB} \quad \text{TENSION}$$

$$F_{ER} = -67230 \text{ LB} - (32040 \text{ LB} + 4000 \text{ LB} + 18,000 \text{ LB} - 67230 \text{ LB}) \sin 123.75^\circ$$

$$F_{ER} = -58350 \text{ LB} \quad \text{COMPRESSION}$$

$$F_S = (64720 \text{ LB} - 4000 \text{ LB} - 18,000 \text{ LB} - 32040 \text{ LB}) \sin 123.75^\circ$$

$$F_S = 5933 \text{ LB}$$



CONSIDER THE FOLLOWING

ASSUME  $\mu = 0.1$

ASSUME IH CONNECTION IS STICKING  
OH CONNECTION IS SLIDING

$$\sum M_{EN} = (6.34'')(17,000 \text{ LB}) (\frac{1}{2} \mu) - (4.65'')(47,200 \text{ LB}) (\mu) \\ - (24.88'')(F_{OR,F}) = 0$$

$$F_{OR,F} = (-14,385 \text{ LB}) \mu$$

FRICTION DUE TO COOLDOWN "STRESSNESSES" THIS  
CONNECTOR, ASSUMING WARMUP MEANS ONLY A  
SIGN CHANGE:

$$F_{OR,F} = (67,500 \text{ LB}) + (44,000 \text{ LB}) \mu$$

ALSO, SINCE FRICTION ON IH & OH BOTH  
RESULT IN A MOMENT OF THE SAME SIGN,  
A MORE CONSERVATIVE ASSUMPTION (FOR  
THIS CONNECTOR ONLY) WOULD BE THAT  
BOTH CONNECTIONS ARE STICKING:

$$\sum M_{EN} = (6.34'')(17,000 \text{ LB}) \mu + (4.65'')(64,720 \text{ LB}) \mu \\ + (24.88'')(F_{OR,F}) = 0$$

$$F_{OR,F} = -16,683 \text{ LB}$$

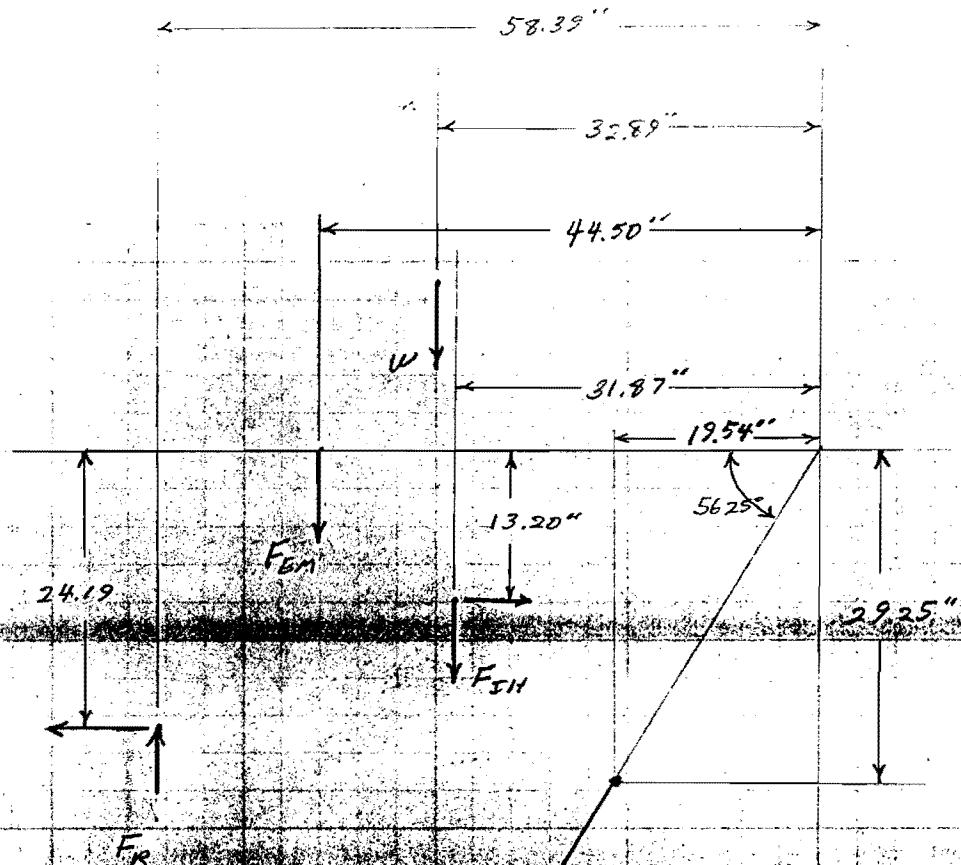
$$F_{OR} = (67,300 \text{ LB}) + (16,700 \text{ LB}) \mu$$

CONNECTION BETWEEN 2 & 3

$$\bar{x} = \left[ -35.06 - 45.79 - 49.56 - 45.79 - 35.06 - 18.97 - 0 \right] \frac{1}{7} = -32.89$$

$$\bar{y} = \left[ -35.06 - 18.97 - 0 + 18.97 + 35.06 + 45.79 + 49.56 \right] \frac{1}{7} = 13.62$$

$$W = (7)(5540 \text{ LB}) = 37380 \text{ LB}$$



$$\sum M_{SR} = (13.35") (37380 \text{ LB}) - (24.19") (4000 \text{ LB}) - (24.88") (F_{vR}) + (12.33") (18,000 \text{ LB}) - (32.89") (64,720 \text{ LB}) = 0$$

$$F_{vR} = 68,069 \text{ LB} \quad \text{TENSION}$$

$$F_{vH} = (-68,069 \text{ LB}) - (37380 \text{ LB} + 4000 \text{ LB} + 18000 \text{ LB} - 64720 \text{ LB}) \text{ at } 146.25^\circ$$

$$F_{vH} = -65103 \text{ LB} \quad \text{COMPRESSION}$$

$$F_s = (64720 \text{ LB} - 4000 \text{ LB} - 17000 \text{ LB} - 137500 \text{ LB}) \mu = 11250 \text{ LB} \mu$$

FOR, F

CONNECTION SECTION

$$\text{ASSUMED } \mu = 0.5$$

ASSUMED 50 CONNECTION TO BRACING  
 CH CONNECTION TO BRACING

$$\sum M_{\text{IK}} = (-16.05") (17,000 \text{ LB}) \left(\frac{1}{2} \mu\right) + (5.06") (64720 \text{ LB}) - (24.88") (F_{OR, F}) = 0$$

$$F_{OR, F} = (7357 \text{ LB}) \mu$$

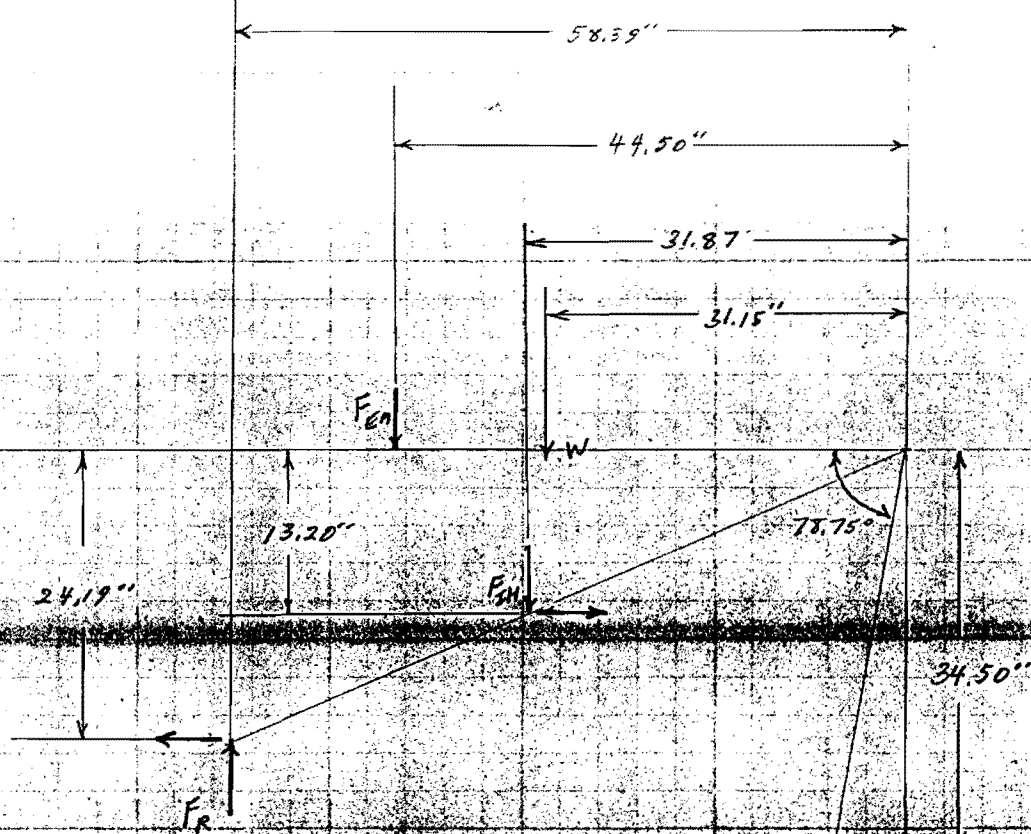
$$F_{OR, F} = (68,100 \text{ LB}) + (7,400 \text{ LB}) \mu$$

CONNECTION BETWEEN 1 & 2

$$\bar{X} = \left[ -18.97 - 35.06 - 45.79 - 49.56 - 45.79 - 35.06 - 18.97 + 0 \right] \frac{1}{8} = -51.15''$$

$$\bar{Y} = \left[ -45.79 - 35.06 - 18.97 - 0 + 18.97 + 35.06 + 45.79 + 49.56 \right] \frac{1}{8} = 6.17''$$

$$W = (8)(5340 \text{ LB}) = 42720 \text{ LB}$$



$$\sum M_{FR} = (-24.29'')(42720 \text{ LB}) + (37.64'')(4000 \text{ LB})$$

$$- (-24.88'')(F_{FR}) + (-25.01'')(18000 \text{ LB})$$

$$- (51.53'')(42720 \text{ LB}) = 0$$

$$F_{FR} = 68,192 \text{ LB}$$

$$F_{FR} = (-68,192 \text{ LB}) - (42720 \text{ LB} + 4000 \text{ LB} + 18000 \text{ LB} - 68720 \text{ LB}) \sin 168.75^\circ$$

$$F_{FR} = -68192 \text{ LB}$$

$F_s \approx 0$  (ONLY WITH ADJUSTED CONNECTION)

$F_s = (5340 LB) \left(\frac{1}{2}\right) \sin 25^\circ$

$F_s = 2619 LB$

OH FIREHEAD  
 BEHIND THE  
 INSTALLATION

CONSIDER FRICTION

ASSUME  $\mu_{SLIDING} = \frac{1}{2} \mu$

ASSUME IH CONNECTION IS SLIDING

ASSUME OH CONNECTION IS STICKING

$$\sum M_{=R} = (-21.30'') (18,000 LB) \left(\frac{1}{2} \mu\right) + (-10.31'') (64,720 LB) \mu$$

$$= (-24.88'') (F_{OH}) = 0$$

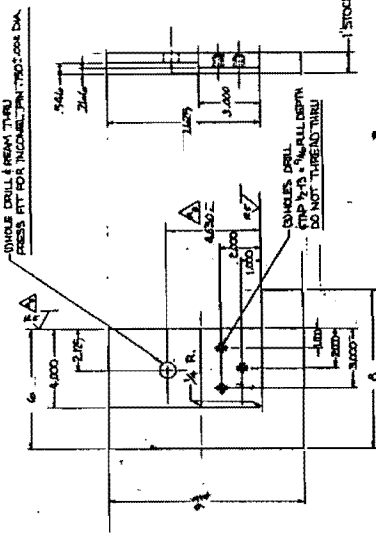
$F_{OH} = (19,114 LB) \mu$

$F_{OH} = (68,200 LB) + (19,100 LB) \mu$

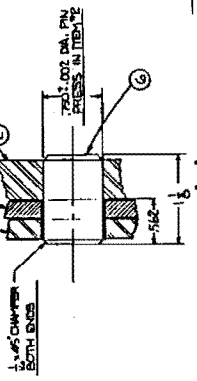
Appendix 2

Drawings

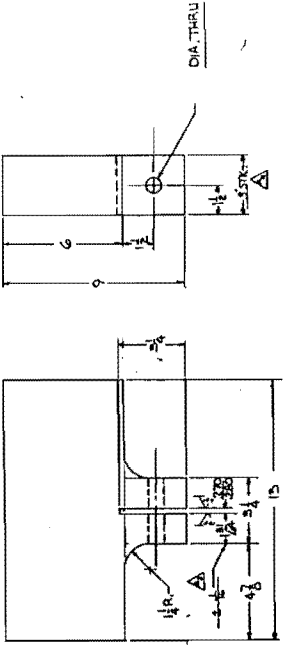
1	ADDED UNITS A, B, C, D, E, F, G, H, I, J, K, L, M, N, O, P, Q, R, S, T, U, V, W, X, Y, Z, AA, AB, AC, AD, AE, AF, AG, AH, AI, AJ, AK, AL, AM, AN, AO, AP, AQ, AR, AS, AT, AU, AV, AW, AX, AY, AZ, BA, BB, BC, BD, BE, BF, BG, BH, BI, BJ, BK, BL, BM, BN, BO, BP, BQ, BR, BS, BT, BU, BV, BW, BX, BY, BZ, CA, CB, CC, CD, CE, CF, CG, CH, CI, CJ, CK, CL, CM, CN, CO, CP, CQ, CR, CS, CT, CU, CV, CW, CX, CY, CZ, DA, DB, DC, DD, DE, DF, DG, DH, DI, DJ, DK, DL, DM, DN, DO, DP, DQ, DR, DS, DT, DU, DV, DW, DX, DY, DZ, EA, EB, EC, ED, EE, EF, EG, EH, EI, EJ, EK, EL, EM, EN, EO, EP, EQ, ER, ES, ET, EU, EV, EW, EX, EY, EZ, FA, FB, FC, FD, FE, FF, FG, FH, FI, FJ, FK, FL, FM, FN, FO, FP, FQ, FR, FS, FT, FU, FV, FW, FX, FY, FZ, GA, GB, GC, GD, GE, GF, GG, GH, GI, GJ, GK, GL, GM, GN, GO, GP, GQ, GR, GS, GT, GU, GV, GW, GX, GY, GZ, HA, HB, HC, HD, HE, HF, HG, HH, HI, HJ, HK, HL, HM, HN, HO, HP, HQ, HR, HS, HT, HU, HV, HW, HX, HY, HZ, IA, IB, IC, ID, IE, IF, IG, IH, II, IJ, IK, IL, IM, IN, IO, IP, IQ, IR, IS, IT, IU, IV, IW, IX, IY, IZ, JA, JB, JC, JD, JE, JF, JG, JH, JI, JJ, JK, JL, JM, JN, JO, JP, JQ, JR, JS, JT, JU, JV, JW, JX, JY, JZ, KA, KB, KC, KD, KE, KF, KG, KH, KI, KJ, KK, KL, KM, KN, KO, KP, KQ, KR, KS, KT, KU, KV, KW, KX, KY, KZ, LA, LB, LC, LD, LE, LF, LG, LH, LI, LJ, LK, LL, LM, LN, LO, LP, LQ, LR, LS, LT, LU, LV, LW, LX, LY, LZ, MA, MB, MC, MD, ME, MF, MG, MH, MI, MJ, MK, ML, MM, MN, MO, MP, MQ, MR, MS, MT, MU, MV, MW, MX, MY, MZ, NA, NB, NC, ND, NE, NF, NG, NH, NI, NJ, NK, NL, NM, NN, NO, NP, NQ, NR, NS, NT, NU, NV, NW, NX, NY, NZ, OA, OB, OC, OD, OE, OF, OG, OH, OI, OJ, OK, OL, OM, ON, OO, OP, OQ, OR, OS, OT, OU, OV, OW, OX, OY, OZ, PA, PB, PC, PD, PE, PF, PG, PH, PI, PJ, PK, PL, PM, PN, PO, PP, PQ, PR, PS, PT, PU, PV, PW, PX, PY, PZ, QA, QB, QC, QD, QE, QF, QG, QH, QI, QJ, QK, QL, QM, QN, QO, QP, QQ, QR, QS, QT, QU, QV, QW, QX, QY, QZ, RA, RB, RC, RD, RE, RF, RG, RH, RI, RJ, RK, RL, RM, RN, RO, RP, RQ, RR, RS, RT, RU, RV, RW, RX, RY, RZ, SA, SB, SC, SD, SE, SF, SG, SH, SI, SJ, SK, SL, SM, SN, SO, SP, SQ, SR, SS, ST, SU, SV, SW, SX, SY, SZ, TA, TB, TC, TD, TE, TF, TG, TH, TI, TJ, TK, TL, TM, TN, TO, TP, TQ, TR, TS, TT, TU, TV, TW, TX, TY, TZ, UA, UB, UC, UD, UE, UF, UG, UH, UI, UJ, UK, UL, UM, UN, UO, UP, UQ, UR, US, UT, UY, UZ, VA, VB, VC, VD, VE, VF, VG, VH, VI, VJ, VK, VL, VM, VN, VO, VP, VQ, VR, VS, VT, VU, VV, VW, VX, VY, VZ, WA, WB, WC, WD, WE, WF, WG, WH, WI, WJ, WK, WL, WM, WN, WO, WP, WQ, WR, WS, WT, WU, WV, WW, WX, WY, WZ, XA, XB, XC, XD, XE, XF, XG, XH, XI, XJ, XK, XL, XM, XN, XO, XP, XQ, XR, XS, XT, XU, XV, XW, XX, XY, XZ, YA, YB, YC, YD, YE, YF, YG, YH, YI, YJ, YK, YL, YM, YN, YO, YP, YQ, YR, YS, YT, YU, YV, YW, YX, YZ, ZA, ZB, ZC, ZD, ZE, ZF, ZG, ZH, ZI, ZJ, ZK, ZL, ZM, ZN, ZO, ZP, ZQ, ZR, ZS, ZT, ZU, ZV, ZW, ZX, ZY, ZZ
---	--



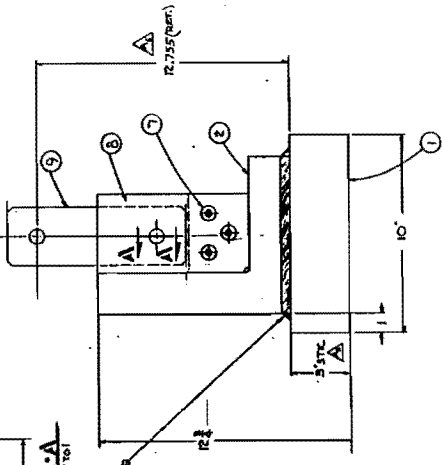
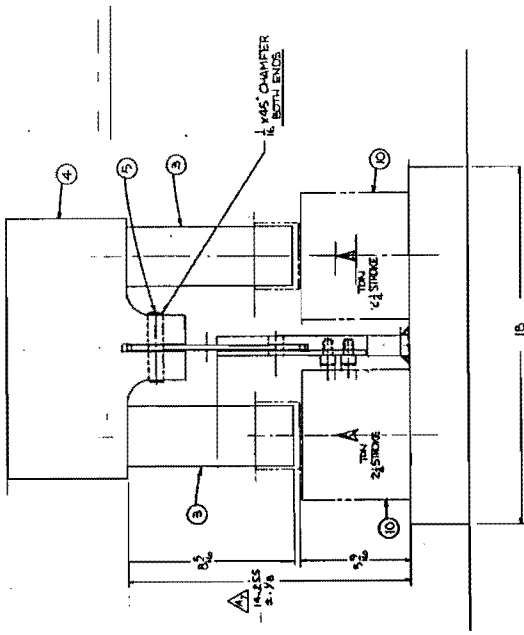
DETAIL ITEM 2  
REF. FRONT R. VIEW 132179



SECT. A-A  
SCALE: 2:1



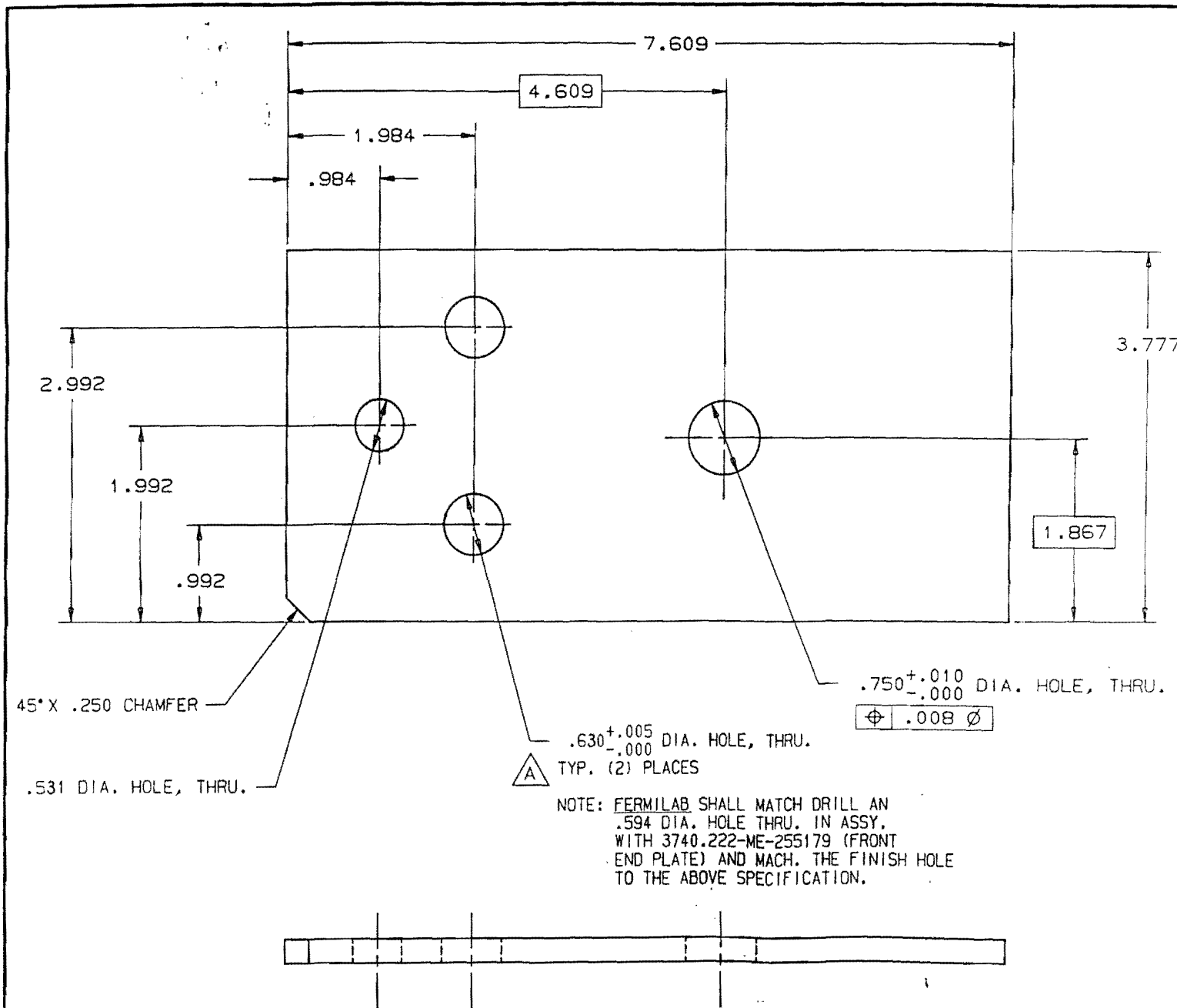
DETAIL ITEM 4



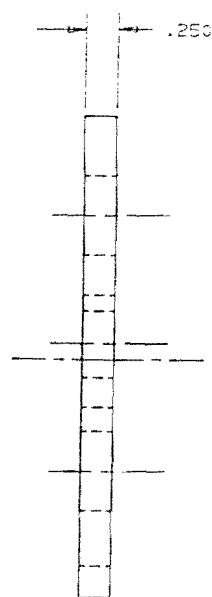
NOTE:  
1) ALL OF PARTS TO BE DONE BY  
2) FINISH PERSONEL  
3) BUBBER (FINAL) TO SUPPLY INCONEL MAT'L FOR REF. S. 1, 6, 9  
4) BUBBER (FINAL) TO SUPPLY FINISHED PARTS REF. 1, 2, 7, 10  
5) ITEM 1, 2 AND 4 MAY BE SAW CUT, EXCEPT WHERE 'G' IS INDICATED.

10	REF. ONLY IN CONE. ON 7903-100	1
9	IN CONE. ON 7903-100	1
8	IN CONE. ON 7903-100	1
7	IN CONE. ON 7903-100	1
6	IN CONE. ON 7903-100	1
5	IN CONE. ON 7903-100	1
4	IN CONE. ON 7903-100	1
3	IN CONE. ON 7903-100	1
2	IN CONE. ON 7903-100	1
1	IN CONE. ON 7903-100	1

FROM NATIONAL ACCELERATION LABORATORY  
MILITARY DIVISION OF AEC  
P.O. BOX 217  
BETHESDA, MARYLAND 20814  
REF. S. 1, 6, 9  
REF. 1, 2, 7, 10  
ITEM 1, 2 AND 4 MAY BE SAW CUT, EXCEPT WHERE 'G' IS INDICATED.



REV.	DESCRIPTION	DRWN.	DATE
A	.630 +.005/-0 WAS .622 +0/-001 AND REMOVED TRUE POSITION DIM. AND ADDED THE NOTE.	S.L.M.	10-12-77



45° X .250 CHAMFER

.531 DIA. HOLE, THRU.

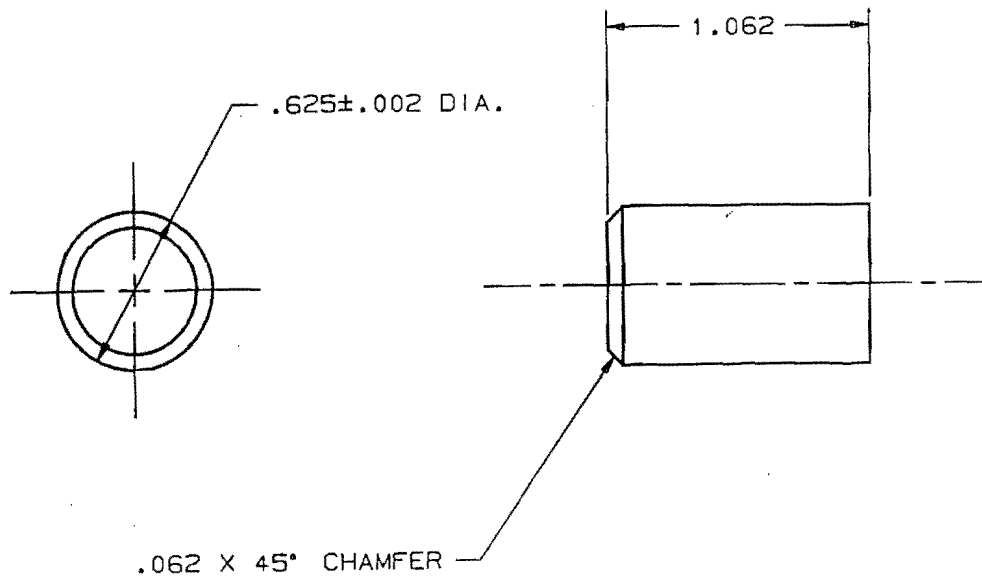
**A** .630<sup>+.005</sup>/<sub>-.000</sub> DIA. HOLE, THRU.  
TYP. (2) PLACES

.750<sup>+.010</sup>/<sub>-.000</sub> DIA. HOLE, THRU.  
 $\varnothing$  .008  $\varnothing$

NOTE: FERMILAB SHALL MATCH DRILL AN .594 DIA. HOLE THRU. IN ASSY. WITH 3740.222-ME-255179 (FRONT END PLATE) AND MACH. THE FINISH HOLE TO THE ABOVE SPECIFICATION.

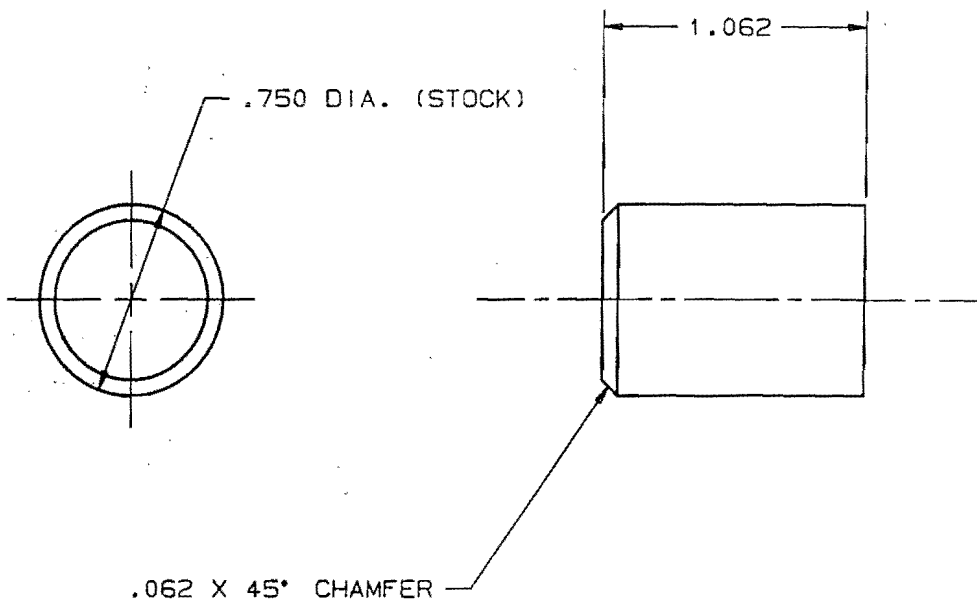
ITEM	PART NO.	DESCRIPTION OR SIZE	QTY.
PARTS LIST			
UNLESS OTHERWISE SPECIFIED		ORIGINATOR	K. KREMPETZ 8-28-80
FRACTIONS DECIMALS		ANILES	DRAWN
± .016 ± 1"		CHECKED	S.L. MEREDITH 8-28-80
1. BREAK ALL SHARP EDGES 1/32" MAX.		APPROVED	
2. DO NOT SCALE DRAWING.		USED ON	
3. DIM. TO ANSI Y14.5			3740.222-ME-255196
4. MAX. ALL MACH. SURFACES		MATERIAL	STAINLESS STEEL TY-304 PLATE
 FERMILAB NATIONAL ACCELERATOR LABORATORY UNITED STATES DEPARTMENT OF ENERGY			
DO DETECTOR END CALORIMETER TOP-HALF MIDDLE HADRONIC MODULE FRONT FACE COVER PLATE			
SCALE	PLD.	PLD. NUMBER	REV.
1:1		3740.222-ME-273214	A
CREATED WITH IGEMODY VERSION 1.64			





REV.	DESCRIPTION	DRAWN	DATE
		APPROV.	

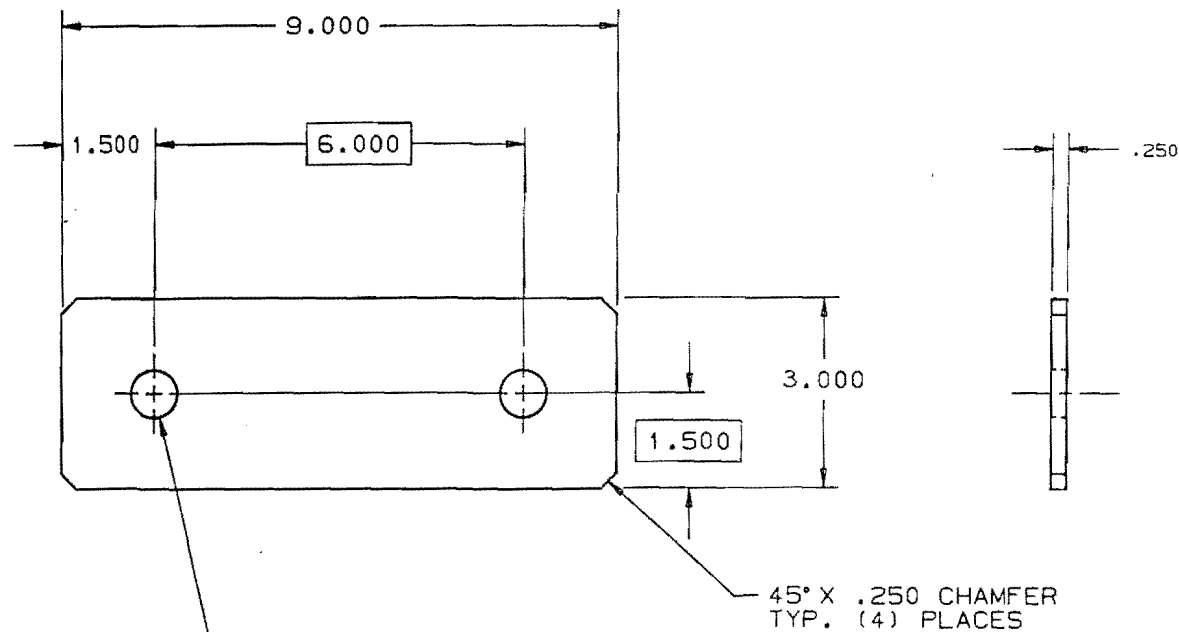
ITEM	PART NO.	DESCRIPTION OR SIZE	QTY.
PARTS LIST			
UNLESS OTHERWISE SPECIFIED: ORIGINATOR: K. KREMPETZ 110-4-86			
DRAWN: S.L. MEREDITH 110-4-86			
FRACTIONAL DECIMALS: ANGLES		CHECKED	
± .005 ± .5°			
1. BREAK ALL SHARP CORNERS 1/8" MIN.		APPROVED	
2. DO NOT SCALE DRAWING.		USED ON	
3. DIM. TO AMBI TTTLS		3740.222-ME-255186	
4. MAX. ALL MACH. SURFACES		MATERIAL: INCONEL 718 (AGED HARDEN)	
 <b>FERMI NATIONAL ACCELERATOR LABORATORY</b> UNITED STATES DEPARTMENT OF ENERGY			
DO DETECTOR END CALORIMETER TOP-HALF MIDDLE HADRONIC MODULE FRONT COVER PLATE PIN DETAIL			
SCALE	FILMED	QUANTITY NUMBER	REV.
2:1		3740.222-MB-273216	
CREATED WITH TCEMVIEW VERSION 1.64			



REV.	DESCRIPTION	DRWN	DATE
		APPD.	DATE

ITEM	PART NO.	DESCRIPTION OR SIZE	QTY.
PARTS LIST			
UNLESS OTHERWISE SPECIFIED		ORIGINATOR	K. KREMPETZ 10-4-88
FRACTIONS DECIMALS	ANGLES	DRWEN	S.L. MEREDITH 10-4-88
±	±.005 ±.5°	CHECKED	
1. BREAK ALL SHARP EDGES TYP. MAT.		APPROVED	
2. DO NOT SCALE DRAWING.		USED ON	3740.222-ME-255186
3. DIM. TO ANG. TYP. S		MATERIAL	INCONEL 718 (AGED HARDEN)
4. MAX. ALL MACH. SURFACES			
<b>FERMI NATIONAL ACCELERATOR LABORATORY</b> UNITED STATES DEPARTMENT OF ENERGY			
DO DETECTOR END CALORIMETER TOP-HALF MIDDLE HADRONIC MODULE FRONT LINKAGE PIN DETAIL			
SCALE	FILED	DRAWING NUMBER	REV.
2:1		3740.222-MB-273215	
CREATED WITH ICEDOWN VERSION 1.64			

REV.	DESCRIPTION	ORIGIN	DATE



.750<sup>+0.005</sup>/<sub>-0.000</sub> DIA. HOLE, THRU.  
 $\varnothing$  .004  $\varnothing$   
 TYP. (2) PLACES

45° X .250 CHAMFER  
 TYP. (4) PLACES

ITEM	PART NO.	DESCRIPTION OR SIZE	QTY.
PARTS LIST			
UNLESS OTHERWISE SPECIFIED		ORIGINATOR	KREMPE 7 9-28-88
FRACTIONS	DECIMALS	ANGLES	DRAWN
			S.L. MEREDITH 9-28-88
±	±.016	± 1°	CHECKED
1. BREAK ALL SHARP CORNERS 1/64 MAX.		APPROVED	
2. DO NOT SCALE DRAWING.		USED ON 3740.220-ME-255181	
3. DIM. TO DIM. LINES			
4. HIDE ALL HIDDEN SURFACES		MATERIAL	
		INCONEL 718	
 FERMI NATIONAL ACCELERATOR LABORATORY UNITED STATES DEPARTMENT OF ENERGY			
DO DETECTOR END CALORIMETER TOP-HALF MIDDLE HADRONIC MODULE FRONT MECHANICAL LINKAGE PLATE			
SCALE	FILMED	DRAWING NUMBER	REV.
1:2		3740.222-MB-273213	
CREATED WITH INTERVIEW VERSION 1.0A			

Appendix 3

Test Data

MH Connector Test, October 24, 1988

Jack Load:	1000 psi		10300 lb		Stress Intensity (psi)	Max Princ Stress (psi)	Min Princ Stress (psi)	Theta (degrees)	Maximum Shear (psi)	X-Dir Stress (psi)	E = 29800000 v = 0.3	
	Measured Strains (x10e6)	Adjust for first zero (x10e6)	Adjusted Strain (x10e6)	Adjusted Strain (x10e6)							Y-Dir Stress (psi)	XY Shear (psi)
# 1	436		436									
Front Plate	49		49		13114	13114	140	3.9	6487	13053	201	887
Side, Edge	-18		-18									
# 2	398		398									
Front Plate	28		28		12052	12052	635	0.5	5709	12051	635	93
Side, Center	21		21									
# 3	385		385									
Cover Plate	35		35		11657	11657	575	1.6	5541	11648	584	318
Side, Edge	11		11									
# 4	350		350									
Cover Plate	18		18		10612	10612	599	-0.7	5006	10610	600	-119
Side, Center	27		27									

MH Connector Test, October 24, 1988

Jack Load:	2600 psi	26780 lb										
	Measured Strains (x10e6)	Adjust for first zero (x10e6)	Adjusted Strain (x10e6)	Stress Intensity (psi)	Max Princ Stress (psi)	Min Princ Stress (psi)	Theta (degrees)	Maximum Shear (psi)	X-Dir Stress (psi)	Y-Dir Stress (psi)	XY Shear (psi)	
												E = 29800000 v = 0.3
# 1	1222		1222									
Front Plate	158		158	36915	36915	718	4.5	18099	36697	937	2806	
Side, Edge	-54		-54									
# 2	1116		1116									
Front Plate	94		94	33821	33821	1826	1.1	15998	33808	1838	635	
Side, Center	46		46									
# 3	1150		1150									
Cover Plate	70		70	34526	34526	808	1.0	16859	34516	819	596	
Side, Edge	25		25									
# 4	1063		1063									
Cover Plate	27		27	32089	32089	1287	-1.5	15401	32069	1307	-781	
Side, Center	86		86									

MH Connector Test, October 24, 1988

Jack Load: 3800 psi 39140 lb

E = 29800000

v = 0.3

	Measured Strains (x10e6)	Adjust for first zero (x10e6)	Adjusted Strain (x10e6)	Stress Intensity (psi)	Max Princ Stress (psi)	Min Princ Stress (psi)	Theta (degrees)	Maximum Shear (psi)	X-Dir Stress (psi)	Y-Dir Stress (psi)	XY Shear (psi)
# 1	1784		1784								
Front Plate	235		235	53980	53980	1391	4.4	26295	53673	1698	4010
Side, Edge	-68		-68								
# 2	1604		1604								
Front Plate	149		149	48889	48889	3587	0.9	22651	48879	3598	701
Side, Center	96		96								
# 3	1747		1747								
Cover Plate	128		128	52475	52475	1194	1.7	25640	52432	1237	1482
Side, Edge	16		16								
# 4	1612		1612								
Cover Plate	38		38	48505	48505	1473	-1.2	23516	48486	1493	-966
Side, Center	111		111								