

Distribution Category: Water Reactor Safety Research--Fuel Behavior (NRC-3)

ANL-77-30

ARGONNE NATIONAL LABORATORY 9700 South Cass Avenue Argonne, Illinois 60439

UNIAXIAL TENSILE PROPERTIES OF ZIRCALOY CONTAINING OXYGEN: SUMMARY REPORT

by

A. M. Garde, H. M. Chung, and T. F. Kassner

Materials Science Division

June 1977



TABLE OF CONTENTS

			Page
ABS	STR.	ACT	17
•			<u>a</u>
1.	IN .	RODUCTION	17
II.	MA	TERIALS, APPARATUS, AND PROCEDURES	18
	Α.	Materials	18
	В.	Tensile-specimen Geometry	19
	C.	Oxygen Charging and Homogenization	20
		1. Oxygen-charging and Homogenization Furnace	20
		2. Oxidation Procedure	21
		3. Homogenization Procedure	22
	D.	Mechanical Testing Facility	22
	E.	Metallography and Specimen Characterization	23
	F.	Work-hardening Analysis	23
	G.	Strain-rate-sensitivity Determination	24
III.	ME	CHANICAL PROPERTIES OF ZIRCALOY-2	24
	А.	As-received Zircaloy-2	24
		1. Stress-Strain Curves	24
		2. Strain-rate Sensitivity	26
		3. Work-hardening Analysis	28
		4. Superplasticity.	30
		5. Dynamic Strain Aging in Zircaloy near 700°C	30
	В.	Zircaloy-2/Oxygen Alloys	31
IV.	ME	CHANICAL PROPERTIES OF ZIRCALOY-4	33
	А.	As-received Zircaloy-4	33
		1. Stress-Strain Curves	33
		2. Strain-rate Sensitivity	36
		3. Work-hardening Analysis	39
		4. Effect of Texture on the Mechanical Properties	43
		5. Fine-grain-size Zircaloy-4	47
		6. Influence of Specimen Geometry	51
		7. Effect of Grain Size on Tensile Properties	53

TABLE OF CONTENTS

 8. Influence of Microstructural State	54 65
relieved Fubing and Recrystallized Sheet	72
V. PROPERTIES OF ZIRCALOY-4/OXYGEN ALLOYS	74
A. Homogeneous Oxygen Distribution	74
B. Composite Material with Oxygen Gradients in α and β Phases	83
C. Effect of Cooling Rate on Properties of Composite Specimens.	90
VI. ACTIVATION ENERGY FOR HIGH-TEMFERATURE DE-	
FORMATION OF ZIRCALOY	92
VII. SUPERPLASTICITY OF ZIRCALOY.	94
VIII. CONCLUSIONS	105
ACKNOWLEDGMENTS	105
REFERENCES	106

<u>No</u> .	Title	Page
1.	Engineering-stress/Engineering-strain Curves for As-received Zircaloy-2 Specimens Deformed at Temperatures between 750 and 950°C	24
2.	Engineering-stress/Engineering-strain Curves for As-received Zircaloy-2 Specimens Deformed at Temperatures between 1000 and 1400°C	24
3.	Effect of Temperature on 0.2% Yield Stress and Ultimate Tensile Strength of As-received Zircaloy-2	26
4.	Effect of Deformation Temperature on Uniform, Necking, and Total Strains for Zircaloy-2 Specimens	26
5.	Strain Dependence of Strain-rate Sensitivity of Zircaloy-2 at Temperatures between 700 and 1400°C for Strains below ~0.2	27
6.	Strain Dependence of Strain-rate Sensitivity of Zircaloy-2 at 700, 800, 850, and 900°C for Strains to ~1.3	27
7.	Effect of Strain Rate on Strain Dependence of Strain-rate Sensitivity of Zircaloy-2 at 900°C	27
8.	Temperature Dependence of Strain-rate Sensitivity of Zircaloy-2 and -4 Specimens	27
9.	Crussard-Jaoul Plots for Zircaloy-2 Specimens Deformed at 900, 1100, 1200, and 1400°C	28
10.	Temperature Dependence of Work-hardening Exponent Obtained from Zircaloy-2 Tensile Data	28
11.	Comparison between a Zircaloy-2 Specimen Strained under Super- plastic Conditions and an Undeformed Specimen	30
12.	Partial Load-elongation Curves for Zircaloy-2 Specimens De- formed at 700°C after Various Heat Treatments	31
13.	Uniform and Total Strain as a Function of Oxygen Concentration for Zircaloy-2 Specimens at 1300 and 1400°C	32
14.	Oxygen-concentration Dependence of Yield Stress and Ultimate Tensile Strength of Zircaloy-2 at 1300 and 1400°C	32
15.	Total Strain, Uniform Strain, Yield Stress, and Ultimate Tensile Strength of Zircaloy-2 at 1300°C as a Function of Oxygen Concentration	32
16.	Engineering-stress/Engineering-strain Curves for Zircaloy-4 Specimens Deformed at 700, 750, 850, 900, and 950°C	33

No.	Title	Page
17.	Engineering-stress/Engineering-strain Curves for Zircaloy-4 Specimens Deformed at 900, 950, 1000, 1100, 1200, 1300, and 1400°C.	33
18.	Engineering-stress/Engineering-strain Curves for Zircaloy-4 Specimens Deformed at 700, 800, and 900°C	34
19.	Engineering-stress/Engineering-strain Curves for Zircaloy-4 Specimens Deformed at 1000, 1100, 1200, 1300, and 1400°C	34
20.	Ultimate Tensile Strength of Zircaloy-4 at Three Strain Rates over Two Temperature Intervals	34
21.	Uniform Strain and Total Strain for Zircaloy-4 Specimens as a Function of Deformation Temperature at Three Strain Rates	35
22.	True-stress/True-strain Curve for Zircaloy-4 Specimen De- formed at 900°C	36
23.	Strain Dependence of Strain-rate Sensitivity of Zircaloy-4 at Various Temperatures	36
24.	Strain Dependence of Strain-rate Sensitivity of Zircaloy-4 at 700, 800, 850, and 900°C	37
25.,	Effect of Strain and Strain Rate on Strain-rate Sensitivity of Zircaloy-4 at 800, 900, and 1000°C	37
26.	Strain-rate Dependence of Strain-rate Sensitivity of Zircaloy at 800, 900, and 1000°C	38
27.	Strain-rate Sensitivity of As-received Zircaloy-4 at 850°C as a Function of Strain	38
28.	Plots of Log(dg/de) vs Log e for Zircaloy-4 Specimens De- formed at 700, 1000, 1100, 1300, and 1400°C	42
29.	Work-hardening Exponent as a Function of Temperature for Zircaloy-4 at Four Strain Rates	42
30.	Temperature Dependence of Parameter k for Zircaloy-4 at Three Strain Rates	42
31.	Variation of σ_0 with Temperature for Zircaloy-4 at Three Strain Rates	43
32.	Effect of Texture on Uniform Strain of Zircaloy-4 at Tempera- tures between Ambient and 1400°C	43
33.	Influence of Texture on Total Strain of Zircaloy-4 at Tempera- tures between Ambient and 1400°C	44

i. T

<u>No</u> .	Title	Page
34.	Temperature Dependence of Ultimate Tensile Strength for Longitudinal, Transverse, and Diagonal Zircaloy-4 Specimens	44
35.	Strain-rate Sensitivity of Longitudinal and Transverse Zircaloy-4 Specimens as a Function of Temperature	44
36.	Strain Dependence of Strain-rate Sensitivity of Transverse and Longitudinal Zircaloy-4 Specimens at 850°C	45
37.	Work-hardening Exponent as a Function of Temperature for Zircaloy-4 Specimens with Three Textures	45
38.	Total Strain as a Function of Temperature for 5-µmGrain-size Zircaloy-4 Specimens at Three Strain Rates	47
39.	Strain-rate Sensitivity of Fine-grain Zircaloy-4 as a Function of Temperature	48
40.	Effect of Strain Rate on Total and Uniform Strain of Zircaloy-4 at 900°C	49
41.	Strain-rate Dependence of Strain-rate Sensitivity of Zircaloy-4 at 900°C	49
42.	Work-hardening Exponent of Fine-grain-size Zircaloy-4 as a Function of Temperature at Three Strain Rates	51
43.	Effect of Width-to-Thickness Ratio of Tensile Specimen on Strain-rate Dependence of Strain-rate Sensitivity of Zircaloy-4 at 850°C	51
44.	Dependence of Total Strain on Strain Rate for 11- and 50-µm Grain-size Zircaloy-4 Specimens at 850°C	52
45.	Strain-rate Dependence of Total Strain of 5- and 11-µm Grain- size Zircaloy-4 Specimens at 850°C	52
46.	Variation of Uniform Strain with Strain Rate for 5- and 11-µm Grain-size Zircaloy-4	53
47.	Yield Stress as a Function of Grain Size for Zircaloy-4 at 850°C .	53
48.	Effect of Hold Time before Deformation on Total Strain of Zircaloy-4	54
49.	Microstructure of a Zircaloy-4 Specimen Oil-quenched from 850°C after 0.6-ks Hold Time	55
50.	Microstructure of a Zircaloy-4 Specimen Oil-quenched from 850°C after 7.2-ks Hold Time	55

<u>No</u> .	Title	Page
51.	Microstructure of As-received Zircaloy-4 after Cooling through $\beta \rightarrow \alpha'$ Transformation at 0.1°C/s from 1000°C	56
52.	Parallel-plate Structure of As-received Zircaloy-4 after Cooling through $\beta \rightarrow \alpha'$ Transformation at ~2°C/s from 1250°C	56
53.	Basketweave Structure of As-received Zircaloy-4 after Cooling through $\beta \rightarrow \alpha'$ Transformation at ~500°C/s from 1000°C	57
54,	Martensitic Structure of As-received Zircaloy-4 after Cooling through $\beta \rightarrow \alpha'$ Transformation at ~10,000°C/s from 1000°C	57
55.	Effect of α -phase Morphology on Engineering-stress/Engineering- strain Curve of Zircaloy-4 at 400°C	57
56.	SEM Fractograph of As-received Zircaloy-4 Specimen Fractured at 400°C; $\dot{\epsilon} = 3.3 \times 10^{-3} \text{ s}^{-1} \dots \dots$	58
57.	SEM Fractograph of Transformed β -phase Zircaloy-4 Specimen Fractured at 400°C; $\dot{\epsilon} = 3.3 \times 10^{-3} \text{ s}^{-1}$	58
58.	Influence of α -phase Morphology on Total Strain of Zircaloy-4 at Temperatures between Ambient and 950°C	59
59.	Temperature Dependence of Ultimate Tensile Strength of Zircaloy-4 with Equiaxed and Basketweave Microstructures	59
60.	Strain-rate Sensitivity as a Function of Temperature for Zircaloy-4 Specimens with Equiaxed Grains and Basketweave Structure; $\epsilon \approx 0.02$	59
61.	Strain-rate Dependence of Strain-rate Sensitivity of Zircaloy-4 with Equiaxed Grains and Basketweave Structure at 850°C; $\epsilon \simeq 0.02$	60
62.	Temperature Dependence of Ultimate Tensile Strength of Transformed β-phase Basketweave Zircaloy-4 at Two Strain Rates	60
63.	Effect of Strain Rate on Total Strain of Zircaloy-4 with Basket- weave Structure at Temperatures between Ambient and 1000°C	61
64.	Ultimate Tensile Strength as a Function of Temperature for Transformed β -phase Zircaloy-4 with Basketweave and Parallel- plate Structures	61
65.	Temperature Dependence of Total Strain of Transformed β -phase Zircaloy-4 with Basketweave and Parallel-plate Structures	61
66.	Effect of Cooling Rate on Yield Stress of Transformed β -phase Zircaloy-4 at Several Temperatures	62

No.	Title	Page
67.	Influence of Cooling Rate on Ultimate Tensile Strength of Transformed β -phase Zircaloy-4 at Several Temperatures	62
68.	Cooling-rate Dependence of Uniform Strain of Transformed β -phase Zircaloy-4 at Several Temperatures	62
69.	Total Strain as a Function of Cooling Rate for Transformed β-phase Zircaloy-4 at Several Temperatures	63
70.	Uniform Strain vs Temperature for Transformed 8-phase Zircaloy-4 at Various Cooling Rates	64
71.	Total Strain as a Function of Temperature for Transformed β -phase Zircaloy-4 at Various Cooling Rates	64
72.	Temperature Dependence of Work-hardening Exponent of Various Transformed β -phase Zircaloy-4 Structures	67
73.	Work-hardening Exponent of Basketweave-structure Zircaloy-4 as a Function of Temperature at 1 wo Strain Rates	68
74.	Partial Load-elongation Curves for Zircaloy-4 Specimens De- formed at 750°C after Various Heat Treatments	68
75.	Strain-rate Dependence of Work-hardening Exponent of Zircaloy at 700°C	70
76.	Engineering-stress/Engineering-strain Curves for Zircaloy-4 at 750°C after Various Heat Treatments	71
77.	Strain Dependence of Strain-rate Sensitivity of Zircaloy-4 at 750°C after Two Heat Treatments	72
78.	Ultimate Tensile Strength and Yield Stress of Zircaloy-4 Stress- relieved Tube and Recrystallized Sheet as a Function of	
	Temperature	72
79.	Total Strain of Zircaloy-4 Tube and Recrystallized Sheet as a Function of Temperature	73
80.	Uniform Strain of Zircaloy-4 Tube and Recrystallized Sheet as a Function of Temperature	73
81.	Effect of Oxygen on Ultimate Tensile Strength of Zircaloy-4 at Several Temperatures between 200 and 1100°C	74
82.	Variation of Ultimate Tensile Strength of Zircaloy-4 with Oxygen Concentration at 1200, 1300, and 1400°C	74
83.	Oxygen-concentration Dependence of Uniform Strain of Zircaloy-4 at Several Temperatures	75

<u>No</u> .	Title	Page
84.	Oxygen-concentration Dependence of Uniform Strain of Zircaloy-4 at 1200, 1300, and 1400°C	75
85.	Total Strain of Zircaloy-4 as a Function of Oxygen Concentration at Several Temperatures	76
86.	Effect of Oxygen Concentration on Total Strain of Zircaloy-4 at 1200, 1300, and 1400°C	76
87.	Effect of Oxygen on Temperature Dependence of Total Strain of Zircaloy-4	77
88.	Pseudobinary Zircaloy-Oxygen Phase Diagram Determined from Metallographic and Resistivity Measurements	77
89.	SEM Fractograph of Zircaloy-4 Specimen with 2.3 wt % Oxygen Fractured at 850°C; $\dot{e} = 3.3 \times 10^{-3} \text{ s}^{-1}$	78
90.	Work-hardening Exponent of Zircaloy-4 at 1000°C as a Function of Oxygen Concentration	80
91.	Work-hardening Exponent of Zircaloy-4 at 1100°C as a Function of Oxygen Concentration	80
92.	Work-hardening Exponent of Zircaloy-4 at 1200°C as a Function of Oxygen Concentration.	81
93.	Work-hardening Exponent of Zircaloy-4 at 1300°C as a Function of Oxygen Concentration.	81
94.	Work-hardening Exponent of Zircaloy-4 at 1400°C as a Function of Oxygen Concentration.	81
95.	Temperature Dependence of Work-hardening Exponent of Zircaloy-4 with Several Oxygen Concentrations	81
96.	Strain Dependence of Strain-rate Sensitivity of Zircaloy-4 with 0.5 wt %Oxygen at 700 and 850°C	82
97.	Effect of Strain on Strain-rate Sensitivity of Zircaloy-4 at 1000°C with 0.11, 0.21, and 0.34 wt % Oxygen	82
98.	Effect of Strain on Strain-rate Sensitivity of Zircaloy-4 with 0.11, 0.22, and 0.35 wt % Oxygen at 1200°C	83
99.	Effect of Strain on Strain-rate Sensitivity of Zircaloy-4 with 0.25 wt % Oxygen at 1000, 1200, and 1400°C	83
100.	Strain Dependence of Strain-rate Sensitivity of Zircaloy-4 with 0.11, 0.2, and 1.0 wt % Oxygen at 1400°C	83

<u>No</u> .	Title	Page
101.	Oxygen-concentration Dependence of Strain-rate Sensitivity of Zircaloy-4 at 1200, 1300, and 1400°C	84
102.	Temperature Dependence of Total Strain of Homogenized and Composite Zircaloy-4 with 0.8 wt % Oxygen	84
103.	Ultimate Tensile Strength as a Function of Temperature for Homogenized and Composite	84
104.	Strain Dependence of Strain-rate Searitivity of Zircaloy-4/ Oxide Composites with 0.78 and 3 00 wt % Total Oxygen at 800°C	85
105.	Strain-rate Sensitivity as a Function of Strain for Zircaloy-4/ Oxide Composites with 0.43 and 0.82 wt % Total Oxygen at 850°C	85
106.	Strain-rate Sensitivity as a Function of Strain for Zircaloy-4/ Oxide Composites with 0.77, 1.12, and 1.88 wt % Total Oxygen at 900°C	86
107.	Strain Dependence of Strain-rate Sensitivity of Zircaloy-4/ Oxide Composites with 0.78 and 1.17 wt % Total Oxygen at 1100°C	86
108.	Strain-rate Sensitivity of Zircaloy-4/Oxide Composites as a Function of Total Oxygen Concentration at 850, 900, and 1100°C.	86
109.	Microstructure of Zircaloy-4/Oxide Composite Specimen De- formed at 800°C Showing Cracks in Oxide and α Layers	87
110.	Microstructure of Zircaloy-4/Oxide Composite Specimen De- formed at 1100°C Showing No Cracks in a Layer	87
111.	Oxygen-concentration Dependence of Yield Stress of Zircaloy-4/ Oxide Composites at Several Temperatures	88
112.	Ultimate Tensile Strength of Zircaloy-4/Oxide Composites as a Function of Oxygen Concentration at Several Temperatures	88
113.	Effect of Oxygen Concentration on Uniform Strain of Zircaloy-4/ Oxide Composites at Several Temperatures	88
114.	Variation of Total Strain of Zircaloy-4/Oxide Composites with Oxygen Concentration at Several Temperatures	88
115.	Temperature Dependence of Work-hardening Exponent of Zircaloy-4/Oxide Composites with Several Oxygen Concentrations	90

<u>No.</u>	Title	Page
116.	Uniform and Total Strain of Zircaloy-4/Oxide Composites with 4.0 wt % Oxygen as a Function of Cooling Rate	90
117.	Ultimate Tensile Strength and Yield Stress as a Function of Cooling Rate for Zircaloy-4/Oxide Composites with 4.0 wt % Oxygen	90
118.	Effect of Cooling Rate on Yield Stress and Ultimate Tensile Strength of Composite Zircaloy-4 Specimens with 4.4 wt % Oxygen	91
119.	Influence of Cooling Rate on Uniform and Total Strain of Composite Zircaloy-4 Specimens with 4.4 wt % Oxygen	91
120.	Strain-rate Dependence of 0.2% Yield Stress of 5-µm Grain- size Zircaloy-4 at Several Temperatures between 700 and 950°C.	92
121.	Strain-rate Dependence of 0.2% Yield Stress of 11-µm Grain- size Zircaloy-4 at Several Temperatures between 700 and 1000°C	92
122.	Temperature Dependence of Strain Rate for Stage-III De- formation of 5-µm Grain-size Zircalcy-4	93
123.	Temperature Dependence of Strain Rate for Stage-III De- formation of 11-µm Grain-size Zircaloy-4	93
124.	Microstructure near Shoulder Region and Fracture Region of a Zircaloy-4 Specimen Deformed at 850°C at a Strain Rate of $3.3 \times 10^{-3} \text{ s}^{-1}$	94
125.	Microstructure of an 11- μ m Grain-size Specimen Deformed to 0.7 Strain at a Strain Rate of 3.0 x 10 ⁻¹ s ⁻¹ at 850°C	95
126.	Microstructure of a 5- μ m Grain-size Specimen Deformed to -1.5 Strain at a Strain Rate of 6.0 x 10 ⁻⁶ s ⁻¹ at 850°C	95
127.	SEM Fractograph of a Zircaloy-4 Specimen Fractured at 800°C;	96
128.	SEM Fractograph of a Zircaloy-4 Specimen Fractured at 950°C; 4 = 3.3 x 10 ⁻³ s ⁻¹	96
129.	Micrograph of Scratched Surface of a 5-µm Grain-size Zircaloy-4 Specimen Deformed at a Strain Rate of 3.3 x 10 ⁻⁴ s ⁻¹ at 900°C Showing Offsets of Inscribed Scratches at Grain Boundar!es	98
1 30.	Strain-rate Dependence of Work-hardening Exponent of Zircaloy-4 at 850°C	100

No.	Title	Page
131.	Logarithmic Plot of Strain Rate as a Function of Grain Size for Stage-II Deformation of Zircaloy-4 at 850°C	101
1 32.	Comparison of Calculated and Experimental Strain-rate De- pendence of 0.2% Yield Stress for 5- and 11-µm Grain-size Zircaloy-4 at 850 and 900°C	102
133.	Slip Lines on Surface of a Zircaloy-4 Specimen Deformed at 850° C; $t = 3.3 \times 10^{-1} s^{-1}$	103
134.	Facets due to Thermal Etching of Surface of a Zircaloy-4 Specimen Deformed at 850°C; $\dot{\epsilon} = 3.3 \times 10^{-6} \text{ s}^{-1}$	104

LIST OF TABLES

No.	Title	Page
I.	Source and Composition of Zircaloy-2 and -4 Materials Used in Fabrication of Uniaxial Test Specimens	19
Ш.	Summary of Oxidation and Homogenization Conditions for Zircaloy Test Specimens	21
Ш.	Least-squares Work-hardening Constants for Deformation Stages of As-received Longitudinal Zircaloy-2 Specimens from Crussard-Jaoul Analysis	29
IV.	Effect of Oxygen on Least-squares Work-hardening Constants for Deformation Stages of Longitudinal β -phase Zircaloy-2 Specimens from Crussard-Jaoul Analysis; $\dot{\epsilon} = 3.3 \times 10^{-3} \text{ s}^{-1}$.	32
v.	Least-squares Work-hardening Constants for Deformation Stages of As-received Longitudinal Zircaloy-4 Specimens from Crussard-Jaoul Analysis; $\dot{\epsilon} = 3.3 \times 10^{-2} s^{-1} \dots \dots$	39
VI.	Let st-squares Work-hardening Constants for Deformation Stages of As-received Longitudinal Zircaloy-4 Specimens from Crussard-Jaoul Analysis; $\dot{\epsilon} = 3.3 \times 10^{-3} \text{ s}^{-1} \dots \dots$	4 0
VII.	Least-squares Work-hardening Constants for Deformation Stages of As-received Longitudinal Zircaloy-4 Specimens from Crussard-Jaoul Analysis; $\dot{\epsilon} = 3.3 \times 10^{-4} s^{-1}$	41
VIII.	Least-squares Work-hardening Constants for Deformation Stages of As-received Transverse Zircaloy-4 Specimens; $\xi = 3.3 \times 10^{-3} s^{-1}$	46
DX.	Least-squares Work-hardening Constants for Deformation Stages of As-received Diagonal Zircaloy-4 Specimens; $\epsilon = 3.3 \times 10^{-3} s^{-1}$	46
Х.	Least-squares Work-hardening Constants for Deformation Stages of 5- μ m Grain-size Zircaloy-4; $\epsilon = 3.3 \times 10^{-2} \text{ s}^{-1} \dots$	49
XI.	Least-squares Work-hardening Constants for Deformation Stages of 5- μ m Grain-size Zircaloy-4; $\dot{\epsilon} = 3.3 \times 10^{-3} \text{ s}^{-1} \dots$	50
ХШ.	Least-squares Work-hardening Constants for Deformation Stages of 5- μ m Grain-size Zircaloy-4; $\dot{\epsilon} = 3.3 \times 10^{-4} \text{ s}^{-1} \dots$	50
ХШ.	Least-squares Work-hardening Constants for Deformation Stages of Transformed β -phase Zircaloy-4; Cooling Rate -0.1°C/s; $\dot{\epsilon} = 3.3 \times 10^{-3} s^{-1}$	65
XIV.	Least-squares Work-hardening Constants for Deformation Stages of Longitudinal Zircaloy-4 Specimens with Parallel- plate Structure; Cooling Rate $\simeq 1^{\circ}C/s$; $\epsilon = 3.3 \times 10^{-3} s^{-1}$	66

LIST OF TABLES

No.	Title	Page
XV.	Least-squares Work-hardening Constants for Deformation Stages of Longitudinal Zircaloy-4 Specimens with Basket- weave Structure; Cooling Rate $\approx 3^{\circ}C/s$; $\dot{\varepsilon} = 3.3 \times 10^{-3} s^{-1}$	66
X V I.	Least-squares Work-hardening Constants for Deformation Stages of Longitudinal Zircaloy-4 Specimens with Martens- itic Structure; Cooling Rate $\approx 10,000^{\circ}$ C/s; $\dot{e} = 3.3 \times 10^{-3} \text{ s}^{-1}$.	67
XVII.	Least-squares Work-hardening Constants for Deformation Stages of Longitudinal Zircaloy-4 Specimens with Basket- weave Structure; Cooling Rate $\approx 3^{\circ}C/s$; $\epsilon = 3.3 \times 10^{-5} s^{-1} \dots$	67
XVIII.	Least-squares Work-hardening Constants for Longitudinal Zircaloy-2 Specimens Deformed at 700°C and Various Strain Rates	69
XIX.	Least-squares Work-hardening Constants for 5-µm Grain- size Longitudinal Zircaloy-4 Specimens Deformed at 700°C and Various Strain Rates	69
XX.	Least-squares Work-hardening Constants for 11-µm Grain- size Longitudinal Zircaloy-4 Specimens Deformed at 700°C and Various Strain Rates	70
XXI.	Least-squares Work-hardening Constants for $11-\mu m$ Grain- size Longitudinal Zircaloy-4 Specimens Deformed at 750°C after Various Heat Treatments; $\dot{e} = 3.3 \times 10^{-3} \text{ s}^{-1}$	71
XXII.	Least-squares Work-hardening Constants for Longitudinal Specimens of Zircaloy-4/Oxygen Alloys; $\dot{\epsilon} = 3.3 \times 10^{-3} \text{ s}^{-1}$.	78
XXIII.	Least-squares Work-hardening Constants for Longitudinal Specimens of Zircaloy-4/Oxygen Alloys; $\dot{\epsilon} = 3.3 \times 10^{-4} \text{ s}^{-1}$.	79
XXIV.	Least-squares Work-hardening Constants for Longitudinal Specimens of Zircaloy-4/Oxygen Alloys; $\dot{\epsilon} = 3.3 \times 10^{-5} \text{ s}^{-1}$.	79
XXV.	Effect of Oxygen Concentration on Least-squares Work- hardening Constants of Longitudinal Equiaxed α -phase Zircaloy-4 Specimens; $\dot{\epsilon} = 3.3 \times 10^{-3} \text{ s}^{-1}$	80
XXV I.	Least-squares Work-hardening Constants for Zircaloy-4/ Oxide Composite Specimens; $\dot{\epsilon} = 3.3 \times 10^{-3} \text{ s}^{-1}$	89
XXVII.	Least-squares Work-hardening Constants for 5-µm Grain- size Zircaloy-4 Specimens Deformed at 850°C and Various Strain Rates	99

LIST OF TABLES

<u>No.</u>	No. <u>Title</u>			
XXVIII.	Least-squares Work-hardening Constants for 11-µm Grain- size Zircaloy-4 Specimens Deformed at 850°C and Various Strain Rates	100		
XXIX.	Least-squares Work-hardening Constants for 50-µm Grain- size Zircaloy-4 Deformed at 850°C and Various Strain Rates.	100		

UNIAXIAL TENSILE PROPERTIES OF ZIRCALOY CONTAINING OXYGEN: SUMMARY REPORT

A. M. Garde, H. M. Chung, and T. F. Kassner

ABSTRACT

The uniaxial stress-strain behavior of Zircaloy-2 and -4, Zircaloy-oxygen alloys with a uniform oxygen distribution, and composite specimens with a $ZrO_2/\alpha/\beta$ layer structure was investigated over the range of experimental conditions: temperature 25-1400°C; strain rate 10^{-6} - 10^{-1} s⁻¹; oxygen content 0.11-4.4 wt %; grain size 5-50 µm; texture longitudinal, transverse, and diagonal orientations; and microstructural state, which consists of the equiaxed α phase and various transformed β acicular structures. The work-hardening and strain-rate sensitivity parameters were determined from the experimental results, and the tensile properties were correlated with oxygen concentration, oxygen distribution in the material, and microstructure. Dynamic strain-aging phenomena were observed in Zircaloy at 200, 400, and 700°C, and superplastic deformation occurred at 850 and 1000°C. An increase in the oxygen concentration in homogeneous Zircaloy-oxygen alloys increased the ultimate tensile strength and decreased the total strain. particularly below ~900°C. In composite specimens with the $ZrO_2/\alpha/\beta$ structure, the total oxygen content had little effect on the ultimate tensile strength below ~1000°C, but the strength increased with oxygen content at higher temperatures. Information on the effects of grain size, oxygen content, texture, and strain rate on the stress-strain behavior suggests that the dominant mechanism of superplastic deformation in Zircaloy near ~850°C is grain-boundary sliding at the α - β interface with accommodation by diffusional creep, dislocation slip, and grainboundary migration. Good correlation was obtained between ductility and values of the strain-rate sensitivity parameter.

I. INTRODUCTION

This report consists of a compilation of Zircaloy tensile data reported over the past two years in Light-Water-Reactor Safety Research Program quarterly progress reports,¹⁻⁸ along with some additional results. The information represents a portion of the mechanical-property data on Zircaloy and Zircaloy-oxygen alloys generated in a U.S. Nuclear Regulatory Commissionsponsored program in which the effect of oxygen on the uniaxial, biaxial, and impact properties of Zircaloy are being determined. The objective of the program is to establish a quantitative cladding embrittlement criterion applicable to postulated loss-of-coolant accident (LOCA) situations in light-water-reactors (LWR's). The mechanical-property information will be incorporated into fuel-element modeling codes that will provide a quantitative basis for evaluating cladding deformation over a wide range of LOCA and power-coolant-mismatch (PCM) conditions. Most of the reported data are for Zircaloy-4 material. However, scoping tests conducted on Zircaloy-2 material show no significant difference between the deformation behavior of the two grades of Zircaloy.

The uniaxial tensile stress-strain behavior of longitudinal and transverse texture Zircaloy-2 and -4 specimens was investigated over the range of experimental conditions: temperature, 25-1400°C; strain rate, $10^{-6}-10^{-1}$ s⁻¹; oxygen concentration, 0.11-4.4 wt % (both homogeneous and composite materials); grain size, 5-50 μ m; microstructural state, equiaxed α and various transformed β acicular microstructures. The tensile data provide information on the ductility, work-hardening rate, strain-rate sensitivity, and the relationship between the microstructure of the material (prior history) and these properties.

II. MATERIALS, APPARATUS, AND PROCEDURES

A. Materials

Zircaloy-2 and -4 sheet material was used to fabricate the tensile test specimens. Table I lists the compositions of the materials obtained from AMAX Specialty Metals and Teledyne Wah Chang along with the grain size, thickness, and heat treatment. An X-ray diffractometer scan of the Zircaloy-2 sheet indicated the expected high (00ℓ) intensity in the plane of the rolled sheet.

A 6.35-mm-thick hot-rolled and annealed Zircaloy-4 plate was also obtained from Teledyne Wah Chang. The composition of the material was practically identical to that of the 0.635-mm-thick 11-µm grain-size foil. This plate was processed by the Materials Processing and Development Group of the Materials Science Division, to obtain fine grain-size (~5-µm) Zircaloy-4 by the following sequence. The 6.35-mm plate was cleaned and sheared into pieces. Each piece was cold-rolled from a 6.35- to a 4.88-mm thickness, cleaned, and vacuum annealed (3×10^{-3} Pa) at 700°C for 3.6 ks. Then the material was cold-rolled from a 4.88- to a 2.92-mm thickness, cleaned, and vacuum-annealed as before. A metallographic examination of the Zircaloy-4 sheet revealed an average recrystallized grain size of 20 µm. Variations in the amount of cold work and different combinations of annealing times and temperatures in the range of 650-750°C failed to produce a grain size ≤ 12 µm.

After a suggestion by Picklesimer⁹ to warm-roll at 400°C and anneal at 550°C, the fabrication sequence was modified to obtain a 5-µm grain size.

The 20- μ m grain-size sheet was warm-rolled in air at 400°C from a 2.92- to a 0.635-mm thickness in 14 passes. Approximately 10% reduction in area was achieved per pass, and the material wis reheated after each pass. After hot rolling, a slight yellow-brown discoloration of the surface due to oxidation occurred; however, a heavy scale was not formed. The surface was cleaned by vapor-blasting with 400-grit alumina suspended in water.

	Zircaloy-2	Zircaloy-4 Teledyne Wah Chang Albany, Oregon						
	AMAX Specialty Metals,							
Source	Akron, New York							
Composition of	Sn, % 1.49	1.55						
as-received material	Fe,% 0.13	0.225						
	Cr, % 0.10	0.105						
	Ni, % 0.04	<35 ppm						
	Al, ppm <35	46						
	B, ppm <0.25	0.2						
	C, ppm 110 (111) ^a	145 (120) ^a						
	Cd, ppm <0.25	<0.2						
	Co, ppm <10	<10						
	Cu, ppm <25	16						
	Hf, ppm 82	41						
	H, ppm 12 (31) ^a	10 (9) ^a						
	Mn, ppm <25	<25						
	N, ppm 38 (27)a	33 (25) ^a						
	O, ppm 1420 (1365) ^a	1100 (1045) ^a						
	Si, ppm 64	50						
	Ti, ppm <20	<25						
	W,ppm <50	<25						
	U, ppm 0.025	1.0						
	Zr Balance	Balance						
Grain size, µm	~15	~ []						
Thickness, mm	0.686	0.635						
Specification	ASTM Z-59	ASTM B352-67						
Heat treatment	Cold-rolled and annealed	Cold-rolled and vacuum-annealed						

TABLE I. Source and Composition of Zircaloy-2 and -4
Materials Used in Fabrication of Uniaxial
Test Specimens

ANL analysis results:

After cleaning with soap and water, the surface was polished with Scotch Bright (alumina-impregnated nylon) and rinsed with water, acetone, and alcohol. The sheets were given a final vacuum anneal $(3 \times 10^{-3} \text{ Pa})$ at 550°C for 1.8 ks, which resulted in an average recrystallized grain size of ~5 μ m.

B. Tensile-specimen Geometry

Uniaxial tensile specimens with gauge lengths 25.4 and 12.7 mm (gauge widths of 6.35 and 3.18 mm, respectively) were fabricated from Zircaloy sheet material according to ASTM Specification E8-69. These specimens had width-to-thickness ratios of 10 and 5, respectively. All specimens had a gauge length-to-width ratio of four.

Specimens with three different orientations were used. Most of the testing was performed on specimens with a longitudinal texture in which the tensile direction coincided with the rolling direction. Transverse and diagonal texture specimens were also tested in which the tensile direction was at 90 and 45° to the rolling direction, respectively. The shoulder region of each specimen was reinforced on both sides by spot-welding supporting coupons of the same material to the grip area.

C. Oxygen Charging and Homogenization

The high-temperature oxidation of Zircaloy results in the formation of a ZrO_2 oxide layer on α -phase Zircaloy at temperatures below the α - β transition temperature of ~810°C and a $ZrO_2/\alpha/\beta$ composite at temperatures above this transition. Thus, the test-specimen production scheme essentially consists of two parts: (1) oxygen charging to a desired concentration and (2) homogenization by high-temperature annealing to eliminate both the phase boundaries and concentration gradients. A Sieverts apparatus for the purpose of oxygen charging and two high-vacuum annealing furnaces for homogenization were used.

1. Oxygen-charging and Homogenization. Furnace

A modified Sieverts apparatus with a Marshall furnace that has a 62-mm-dia, 450-mm-long Pt-30% Rh heating element and a 250-mm-long uniform-temperature zone was used for charging oxygen into the Zircaloy test specimens. A vacuum in the range of 1.3×10^{-4} Pa was achieved at temperatures up to ~1300°C with the diffusion pump and the liquid-nitrogen trap. Thermocouples of Pt-Pt 10% Rh calibrated against the melting points of aluminum and copper, were used for temperature measurements and control sensing. The temperature-sensing thermocouple was placed at the outer surface of the alumina muffle tube, and a direct temperature reading was displayed on a strip-chart recorder. With a combination of a silicon-controlled rectifier power supply and a Digiset null detector, temperature could be stabilized within $\pm 5^{\circ}$ C up to 1200°C. Specimens were placed inside a recrystallized, impervious alumina muffle tube, which was connected to the pumping and gashandling system by a standard tapered joint with a silicone sleeve. The Zircaloy specimens were suspended from a 1-mm-dia Pt-40% Rh wire.

Two high-temperature. high-vacuum Centorr furnaces for homogenization of the Zircaloy specimens were calibrated and operated at temperatures up to 1600°C. Both systems have tungsten heating elements and tantalum heat shields that can operate in vacuum, inert-gas, or reducing atmospheres. Uniform temperatures were maintained over half the length of the 61-mm-dia, 125-mm-long and 75-mm-dia, 200-mm-long heat zones. The

.

temperatures were measured with Pt-10% Rh thermocouples and compared with optical-pyrometer and thermal-converter readings. At 1400°C, the temperature could be maintained constant within $\pm 10^{\circ}$ C for up to 108 ks. Rapid heating and cooling of the specimens could be achieved; however, it took ~0.9 ks before the power level and temperature stabilized at 1400°C in the automatic control mode. Temperature readings were recorded on a stripchart recorder.

2. Oxidation Procedure

The mechanical-property test specimens were degreased in toluene and acetone, dried, and then weighed on a microbalance. Six to ten specimens were placed inside the alumina muffle tube and the system was evacuated to 1.3×10^{-4} Pa. The specimens were heated to the charging temperatures given in Table II, and a measured quantity of a 1:3 oxygen-helium gas mixture was admitted to the reaction tube through a needle valve. The pressure was increased from 1.3 Pa at the onset of the reaction to 260 Pa in ~30 s to prevent an appreciable temperature increase in the specimens. After this initial period, the gas pressure was increased to 6.5×10^4 Pa, i.e., an oxygen partial pressure of 1.62×10^4 Pa. In a few instances when the gasadmission rate was too high, an appreciable temperature increase was recorded for ~30 s; however, autoignition did not occur.

Matrix Phase	C, wt %	Т _с , ^е С	t _c , ks	т _h . °С	th. ks	
β	0.2	900	0.90	1100	108	
B	0.3	900	1.20	1200	72	
ß	0.4	900	1.80	1200	72	
ß	0.5	900	2.40	1200	72	
B	0.6	900	3.24	1300	54	
B	0.8	900	4.80	1300	54	
ß	1.0	900	6.60	1400	21.6	
0 + B	1.0	900	6.60	1300	72	
a + B	1.0	900	6.60	1200	72	
0 + B	1.0	900	6.60	1100	108	
a + 8	1.0	900	6.60	1000	108	
α+β	1.0	900	6.60	950	144	
0 + B	0.7	900	4.20	1100	72	
a + 8	0.7	900	4.20	1000	108	
α+β	0.7	900	4.20	950	144	
or + B	0.3	900	1.20	950	144	
0 + B	0.3	900	1.20	900	216	
a	1.0	900	6.00	800	432	
đ	1.5	1000	2.40	850	360	
a	2.0	1000	3.60	900	216	
a	2.5	1000	7.20	1000	86.4	
a	3.0	1000	9.00	1000	86.4	
a	3.5	1100	18.00	1100	86.4	
a	4.0	1100	7.80	1200	72	
٥	5.0	1100	13.20	1300	54	
a	6.0	1100	19.80	1300	54	
a	6.7	1200	10.80	1300	54	

TABLE II. Summary of Oxidation and Homogenization Conditions for Zircaloy Test Specimens^a

^aC = oxygen concentration; $T_c = oxygen-charging temperature; t_c = oxygen$ $charging time; <math>T_h = homogenization temperature; and t_h = homogenization time.$ After the desired period of oxygen charging, the specimens were cooled and the weight gain was determined. The variation of weight gains between the specimens in the same group was within $\pm 4\%$. No reaction between the Pt-40% Rh specimen suspension wires and the Zircaloy was observed at 900 or 1000°C.

3. Homogenization Procedure

The oxygen-charged specimens were sealed in a 56-mm-dia, 125-mm-long tantalum capsule by electron-beam welding in a 1.3×10^{-3} -Pa vacuum and then annealed in either of the Centorr furnaces. The tantalum capsule acts as an effective getter in removing any reactive impurities in the residual vacuum inside the capsule. Specimens that were oxidized at 1000°C for 1.8 ks to an average oxygen concentration of $3.6 \cdot \text{wt}$ % and then homogenized for 90 ks at 1000°C showed a weight gain corresponding to only 0.012 wt % oxygen. The Pt-40% Rh wire used to suspend the Zircaloy specimens was found to be satisfactory at temperatures up to 1200°C.

At 1400°C and above, the wire forms a low-melting eutectic with Zircaloy and melts during the homogenization anneal. The Pt-10% Rh wire was satisfactory for temperatures up to 1400°C and 108 ks of annealing. Above 1400°C, a significant reaction occurs with either Zircaloy or tantalum and platinum, Pt-10% Rh, Pt-40% Rh, Mo, and W-Re alloys. However, rhenium was found to be satisfactory up to 25.2 ks at 1500°C. In view of this, homogenization at 1400°C was limited to the production of β -phase Zircaloy with 1 wt % oxygen, which required the high temperature because of phaseequilibrium considerations.

According to the phase diagram of Savitskii et al.,¹⁰ Zr-Re forms a eutectic at ~26 wt % Zr composition, which melts at ~1600°C. From our information obtained by homogenization treatments and the subsequent hardness profile measurements, a rather conservative estimate of required annealing temperature and time is given in Table II for each set of conditions.

D. Mechanical Testing Facility

Two Instron machines (Models TTDML and 1125) were used for the mechanical testing. These machines were equipped with high-temperature/ high-vacuum furnaces with split tungsten heating elements that can operate at temperatures >1400°C at a vacuum <1 x 10^{-3} Pa. The heating and cooling rates of the furnace were controlled by a Data-Trak programmer. The load-elongation data were converted to true-stress/true-strain data by a computer program that made use of the constant volume approximation. Tensile tests at variable speeds were conducted to determine the strain-rate sensitivity parameter.

E. Metallography and Specimen Characterization

After oxygen charging or the homogenization treatment, it was necessary to characterize the specimen in terms of both the phase-boundary locations and the oxygen-concentration gradient. Oxidized specimens with ZrO_2/α and α/β boundaries were sectioned by a carborundum wheel or an electron-beam technique, cold-mounted, polished in a vibratory polisher, and etched in a 50% distilled H₂O: 45% HNO₃: 5% HF solution to observe the phase boundaries and grain structure. Anodization of the specimens in a solution containing 0.06 m³ absolute ethanol:0.035 m³ H₂O:0.02 m³ glycerine:0.01 m³ lactic acid (85%):0.005 m³ phosphoric acid (85%):0.02 kg citric acid was also used to develop the structure for metallographic examination. The specimen color and the current drop provided convenient criteria for adequate anodization.¹¹⁻¹³ In most cases, polarized light was required to obtain satisfactory contrast. The optimum procedures for specimen preparation, which involve chemical etching and/or anodizing, were highly dependent upon the oxygen concentration in the alloy, the phase distributions, and whether the specimen was Zircaloy-2 or -4. Similar metallographic techniques were used to examine microstructures of deformed samples.

To determine the presence of oxygen-concentration gradients in the specimens, microhardness measurements were made with a Leitz hardness tester. The microhardness was found to be independent of load, in the range of 0.025-0.5 kg, and the Vickers hardness data were judged to be more reliable than the Knoop measurements. Both were calibrated against standard hardness materials. Hardness measurements in the α phase in the immediate vicinity of the ZrO_2/α phase boundary frequently produced cracks that resulted in nonreproducible hardness values. For the homogenized specimens, the absence of phase boundaries was verified by metallographic observation and the hardness profile across the thickness.

F. Work-hardening Analysis

The work-hardening analysis of the Zircaloy tensile data was conducted according to the Crussard-Jaoul method,¹⁴⁻¹⁶ which is based on the Ludwik equation,¹⁷

$$\sigma = \sigma_0 + k e^n, \tag{1}$$

where σ is the true stress, ϵ is the true strain, n is the work-hardening exponent, and k and σ_0 are constants introduced to obtain a best fit with the experimental data. Differentiating Eq. 1 with respect to strain and taking the logarithm of both sides, we obtain

$$\log \frac{d\sigma}{d\epsilon} = \log(kn) + (n - 1)\log \epsilon.$$
 (2)

Therefore, when $\log(d\sigma/d\varepsilon)$ is plotted as a function of $\log \varepsilon$, the slope of the curve is n - 1, and the intercept on the log (slope) axis is $\log(kn)$. To obtain a better fit between Eq. 1 and the experimental stress-strain curve, the stress-strain curve can be divided into different deformation stages¹⁸ so that each stage has a different set of parameters σ_0 , k, and n. The stages are numbered so that stage 2 generally represents a major portion of the stressstrain curve. A least-squares straight-line fit was used to calculate the slope (and therefore n) and the intercept for each stage. Since Eq. 1 does not take into account the change in strain rate during a tensile test, the Crussard-Jaoul work-hardening analysis can be applied only to the uniform deformation region (i.e., before the onset of necking).

G. Strain-rate-sensitivity Determination

The strain-rate-sensitivity parameter, m, is defined by

$$(\sigma_2/\sigma_1) = (\dot{\epsilon}_2/\dot{\epsilon}_1)^{\rm m}, \tag{3}$$

where σ_1 and σ_2 are the flow stresses at strain rates $\dot{\epsilon}_1$ and $\dot{\epsilon}_2$, respectively. The strain-rate sensitivity m was determined from constant-temperature differential-crosshead velocity tests where a load change associated with a crosshead velocity change was measured at constant strain.

III. MECHANICAL PROPERTIES OF ZIRCALOY-2

A. As-received Zircaloy-2

1. Stress-Strain Curves

Figures 1 and 2 show the engineering stress-strain curves for Zircaloy-2 specimens fractured at the strain rate of $3.3 \times 10^{-3} \text{ s}^{-1}$. In this



Fig. 1. Engineering-stress/Engineering-strain Curves for As-received Zircaloy-2 Specimens Deformed at Temperatures between 750 and 950°C. Neg. No. MSD-60897.



Fig. 2. Engineering-stress/Engineering-strain Curves for As-received Zircaloy-2 Specimens Deformed at Temperatures between 1000 and 1400°C. Neg. No. MSD-60895.

report, the engineering stress is based on the original cross-sectional area, and the engineering strain is calculated from the original gauge length, i.e., the total strain is defined as $(\ell - \ell_0)/\ell_0$, where ℓ_0 is the initial gauge length of the specimen and ℓ is the specimen gauge length at fracture. The maximum load point (corresponding to the uniform strain) on all these curves in Figs. 1 and 2 occurred at a strain less than 0.1. Therefore, true-stress and true-strain calculations beyond the uniform strain require a necking correction to account for the triaxial state of stress. To apply the necking correction, neck-profile measurements during the test are necessary. Such measurements were not possible with the present equipment because of the high-temperature/highvacuum environmental chamber. The measurement of the fracture area, to calculate the true fracture stress, also had a number of practical difficulties for test temperatures above 1000°C that resulted from excessive grain growth. These include the unevenness of the specimen surface, which made it difficult to measure the thickness, double-neck formation, and the small specimen thickness at the point of fracture.

A linear-intercept grain-size estimation in the gauge length of the fractured specimens revealed that, up to 1000°C, the grain size was close to that of as-received material, i.e., 15 µm. At higher temperatures, grain size during deformation was estimated from the prior β grain size, which was deduced from the room-temperature transformed β structure. On a number of occasions, it was difficult to estimate prior β -grain size from the metallographically polished samples because of the acicular structure obtained during the $\beta \rightarrow \alpha'$ transformation. Thermal etching of the specimens during deformation in the high-temperature/high-vacuum environment provided a means for estimating the prior β -grain size. The estimates from the shoulder region of the samples deformed at $\epsilon = 3.3 \times 10^{-3} \text{ s}^{-1}$ are as follows: 1100°C (440 µm), 1200°C (600 µm), 1300°C (750 µm), and 1400°C (850 µm). Thus, the stress-strain curves in Fig. 2 correspond to different grain sizes. Grain coarsening during simulated LOCA conditions has been observed in Zircaloy-2.¹⁹

Figure 3 is a plot of the temperature dependence of the 0.2% offset yield stress and ultimate tensile strength. Both the yield stress and the ultimate tensile strength decrease continuously with an increase in temperature. Figure 4 shows the variations of the uniform strain, total strain, and necking strain (from the maximum load point to the fracture point) with temperature. Note that below 1100°C, the uniform elongation is almost zero and that necking elongation almost completely accounts for the total elongation. Another significant feature of Fig. 4 is the minimum in the necking strain and total strain at 950°C.

The elongation peaks near 850 and 1100°C are manifestations of superplasticity. This point will be discussed in detail in subsequent sections.



Fig. 3. Effect of Temperature on 0.2% Yield Stress and Ultimate Tensile Strength of As-received Zircaloy-2. Neg. No. MSD-60902.



Fig. 4. Effect of Deformation Temperature on Uniform. Necking, and Total Strains for Zircaloy-2 Specimens. Neg. No. MSD-60901.

2. Strain-rate Sensitivity

Figure 5 shows the effect of strain on the strain-rate sensitivity of Zircaloy-2. Note that the strain-rate sensitivity decreases with strain at large strains in all cases, except at 800 and 900°C, where superplastic deformations are observed. Once localized deformation due to necking starts, the strain-rate cycling technique gives m values at a higher starting strain rate. For this reason, the measurements reported in Fig. 5 were limited to $\epsilon = 0.20$. Between 700 and 900°C, however, necking was not easily detected in the profile of the specimens deformed to a considerable strain beyond the maximum load point on the load-elongation curve. In these cases, therefore, the strain-rate cycling was continued to higher strains. The results are shown in Fig. 6. Note that at 850 and 900°C, where superplastic deformations are observed, the strain-rate sensitivity continues to increase with strain at large values of strain. The rate of increase of m with strain $(\partial m/\partial e)$, is less significant at 700 and 800°C. Figure 7 shows the effect of strain rate and strain on the 900°C. strain-rate sensitivity of Zircaloy-2. At slower strain rates, the value of m is higher and more sensitive to strain; this probably explains the larger superplastic deformations at slower strain rates.

The temperature dependence of the strain-rate sensitivity of Zircaloy-2 is plotted in Fig. 8. Normally. m increases as the temperature increases; however, at high temperature, grain growth tends to decrease m. Excessive grain growth is probably responsible for the poorly defined minimum near 1100°C and the decrease in m above 1200°C. Figure 8 also shows Zircaloy-4 data that exhibit a larger variation with temperature and a maximum near 900°C. Results presented in later sections show that superplastic elongations are observed in Zircaloy-4 deformation near 900°C. Also, above 1200°C, the Zircaloy-4 strain-rate sensitivity is higher than that of Zircaloy-2. This is probably due to the higher oxygen concentration in Zircaloy-2 (0.14 wt %) than in Zircaloy-4 (0.11 wt %).



Fig. 5. Strain Dependence of Strain-rate Sensitivity of Zircaloy-2 at Temperatures between 700 and 1400°C for Strains below ~0.2. Neg. No. MSD-61490.





Fig. 7. Effect of Strain Rate on Strain Dependence of Strain-rate Sensitivity of Zircaloy-2 at 900°C. Neg. No. MSD-61496.



Fig. 8. Temperature Dependence of Strain-rate Sensitivity of Zircaloy-2 and -4 Specimens. Neg. No. MSD-61502.

3. Work-hardening Analysis

Figure 9 shows plots of log(slope) versus log(strain) for Zircaloy-2 specimens deformed at 900, 1100, 1200, and 1400°C. A portion of each plot can be approximated by a straight line that designates a stage.



Crussard-jaoul Plots for Zircaloy-2 Specimens Deformed at 900, 1100, 1200, and 1400°C. Neg. No. MSD-61488.

Fig. 9

Table III lists the results of the analysis, and Fig. 10 is a plot of a portion of these data. Note that in Fig. 10, the work-hardening exponent assumes negative values and a minimum in the temperature interval ~800-1000°C. This tendency is greater at the lower strain rate. The variation of k with temperature is similar to that of n.

A negative value of n (or k) implies a rapid decrease in the slope of the stress-strain curve with an increase in strain. The processes during deformation that can lead to a negative value of n are: dynamic recovery, dynamic recrystallization, phase transformation, grain-boundary migration, and grain-boundary sliding (intergranular deformation). The $\alpha \rightarrow \beta$ phase



Temperature Dependence of Work-hardening Exponent Obtained from Zircaloy-2 Tensile Data. ANL Neg. No. 306-77-65 Rev. 1.



Strain Rate, é, s ⁻¹	Temp, °C	Stage Number	Strain Interval	n	k, Pa	Average ₀₀ , Pa
3.3 x 10 ⁻⁴	700	2	0.0028-0.035	0.35	3.1×10^7	3.5 x 10 ⁷
	900	1 2	0.0004-0.004 0.005-0.012	0.70 -1.50	7.2×10^{7} -1.9 x 10 ²	1.2 x 10 ⁷ 1.4 x 10 ⁷
	950	1 2	0.0004-0.0024 0.0028-0.006	-0.18 -2.10	-7.0 x 10 ⁵ -7.0 x 10 ⁻¹	7.9 x 10 ⁶ 6.0 x 10 ⁶
	1100	l Z	0.0004-0.012 0.012-0.014	0.73 -1.10	1.6×10^{7} -2.1 x 10 ³	5.6 x 10 ⁵ 1.5 x 10 ⁶
	1 300	1 2 3	0.0004-0.0028 0.0032-0.014 0.014-0.070	1.03 0.17 0.77	8.5 x 10 ⁶ 3.7 x 10 ⁵ 8.0 x 10 ⁵	7.8 x 10 ⁵ 6.7 x 10 ⁵ 8.2 x 10 ⁵
3.3 x 10 ⁻³	200	1 2 3	0.0004-0.0032 0.0036-0.03 0.035-0.09	-1,10 1,10 -0,14	-4.4×10^{3} 1.3 x 10 ⁹ -1.4 x 10 ⁸	2.3 x 10 ⁸ 2.3 x 10 ⁸ 4.9 x 10 ⁸
	400	1 2 3	0.0004-0.0024 0.0028-0.014 0.014-0.13	-0.77 1.10 0.03	-4.4 x 10 ⁴ 2.2 x 10 ⁹ 8.5 x 10 ⁹	1.1 x 10 ⁸ 1.1 x 10 ⁸ -8.3 x 10 ⁹
	700	2	0.007-0.05	0.41	3.1×10^{7}	4.8 x 107
	750	2	0.0004-0.03	-0.06	-1.7×10^{7}	6.9×10^{7}
	825	2	0.0004-0.02	0.01	9.4×10^{7}	-6.8×10^{7}
	850	1 2	0.0004-0.0029 0.0032-0.025	0.50 -0.55	1.1 x 10 ⁸ -1.7 x 10 ⁵	1.3 x 10 ⁷ 2.4 x 10 ⁷
	900	1 2	0.0004-0.0024 0.0024-0.011	0.77 -0.54	6.3 x 10 ⁷ -3.3 x 10 ⁴	1.8 x 10 ⁷ 1.9 x 10 ⁷
	950	2	0.0004-0.002	-0.46	-1.7×10^{4}	1.1 x 10 ⁷
	1100	2	0.025-0.07	0.56	1.9×10^{6}	2.4×10^{6}
	1200	1 2	0.0004-0.004 0.004-0.11	0.92 0.36	1.3 x 10 ⁷ 1.8 x 10 ⁶	1.6 x 10 ⁶ 1.4 x 10 ⁶
	1 300	1 2	0.0004-0.014 0.014-0.11	0.88 0.23	8.2 x 10 ⁶ 2.1 x 10 ⁶	1.5 x 10 ⁶ 9.3 x 10 ⁵
	1400	· 1 2	0.0004-0.013 0.013-0.09	1.00 0.48	4.3 x 10 ⁶ 8.8 x 10 ⁵	1.1 x 10 ⁶ 1.0 x 10 ⁶

 TABLE III.
 Least-squares Work-hardening Constants for Deformation

 Stages of As-received Longitudinal Zircaloy-2 Specimens from

 Crussard-Jaoul Analysis

transformation occurs extremely rapidly; therefore it is expected that the transformation is complete within the 0.6 ks hold period at the test temperature prior to deformation. Phase transformation is not likely to affect the work-hardening rate.

The remaining four factors influence the work-hardening rate to a different extent depending on the strain-rate and deformation temperature. Grain-boundary sliding is significant at higher temperatures ($\gtrsim 800^{\circ}$ C) and lower strain rates ($\leq 10^{-3} \text{ s}^{-1}$). The substantial grain growth during deformation at temperatures >1000°C increases the work-hardening rate, since grain growth decreases the extent of grain-boundary sliding and grain-boundary migration. Dynamic recovery probably occurs at all temperatures, and its extent increases as the temperature increases. Recrystallization of the α phase occurs above 725°C.

4. Superplasticity

Several scoping tests have been made at initial strain rates between 10^{-4} and 10^{-5} s⁻¹. At 900°C, the Zircaloy-2 specimens exhibited superplastic behavior characterized by a small (practically zero) uniform elongation, quite large necking elongation (and therefore a large total elongation), zero or negative work-hardening exponents, and a large value (>0.2) for the strain-rate



Fig. 11

Comparison between a Zircaloy-2 Specimen Strained under Superplastic Conditions (i.e., 900°C, $\dot{\epsilon} = 6.6 \times 10^{-5} \text{ s}^{-1}$) and an Undeformed Specimen. Neg. No. MSD-60891.

sensitivity parameter. The macrophotograph in Fig. 11 compares a Zircaloy-2 specimen deformed at 900°C $(\dot{\mathbf{c}} = 6.6 \times 10^{-5} \text{ s}^{-1})$ and an undeformed specimen. Even after deformation to a strain of 220%, no visible concentrated neck occurs in the gauge section of the specimen. A considerable strain in the width and thickness directions is also present. As seen in Fig. 11 (deformed specimen), the strain in the width direction is ~5^% in the center of the gauge length. At the same point the measured strain in the thickness direction was ~40%. Similar behavior was observed at a strain rate of 3.3 x 10^{-4} s^{-1} .

The factors responsible for this superplastic behavior are probably the fine grain size and two-phase structure. As stated earlier, grain coarsening occurs above 1000°C. At 900°C, on the other hand, some grain-size refinement occurs during deformation. For example, in the gauge section of the specimen deformed at 900°C, the average grain size was $~7 \mu m$. The undeformed shoulder region of the same specimen had a grain size of 15 μm , the same as the as-received material. A similar effect was observed by Backofen et al.²⁰ in the deformation of a two-phase Ti-6Al-4V alloy. A metallographic examination of our 900°C specimens also revealed evidence of a two-phase structure. Superplasticity was reported²⁰ for fine-grain Zircaloy-4 deformed in the transformation region. A detailed discussion of possible mechanisms of superplastic deformation of Zircaloy is included in Sec. VII.

5. Dynamic Strain Aging in Zircaloy near 700°C

Strain aging has been observed in commercial zirconium,²¹ Zircaloy-2,^{22,23}, zirconium alloys,²⁴ and zirconium-oxygen alloys²⁵ in the temperature range of 150-500°C. Strain aging at 250-300°C is generally attributed²⁵ to the presence of oxygen, whereas some controversy exists with regard to the element responsible for strain aging at ~450°C. Although Kelly and Smith²⁵ suggest that strain aging at 450°C is due to carbon, Veevers²⁴ attributes it to iron and, to some extent, oxygen. Our survey of the literature found no reference to strain aging in Zircaloy above 500°C.

Since a yield point was consistently observed in the tensile tests of both Zircaloy-2 and -4 near 700°C, additional tests were conducted to evaluate phenomena responsible for this strain-aging effer Figure 12 shows results for Zircaloy-2 specimens deformed at 70(\therefore Curve A is the partial load-elongation curve of an as-received sample (5.4-ks final anneal at 760°C) strained to fracture at 700°C that shows a yield point. Curve B indicates the behavior of another as-received sample at 700°C strained to ~ $\epsilon = 0.02$ and unloaded. Curve C represents the deformation behavior of the sample from curve B after a 14.4-ks anneal at 800°C. Pronounced servations at regular intervals, which is a manifestation of dynamic strain aging (DSA), are observed in curve C. A complete discussion of DSA effect in Zircaloy is included after presentation of the Zircaloy-4 DSA data.



B. Zircaloy-2/Oxygen Alloys

Fig. 12

Partial Load-elongation Curves for Zircaloy-2 Specimens Deformed at 700°C after Various Heat Treatments. (A) As-received, strained to fracture: (B) as-received, partially strained; and (C) restrained to fracture following a 14.4-ks annealed at 800°C after B. Neg. No. MSD-63143.

Scoping tests were conducted to investigate the effect of oxygen on the properties of β Zircaloy-2 at 1300 and 1400°C. Figure 13 shows the oxygen-concentration dependence of uniform and total strain of β Zircaloy at 1300 and 1400°C at the strain rate of 3.3 x 10⁻³ s⁻¹. Near 0.34 wt % oxygen concentration, the uniform strain shows a minimum and the total strain shows a maximum. The oxygen-concentration dependence of yield stress and ultimate tensile strength is presented in Fig. 14. The significant grain growth observed at 1300 and 1400°C may be responsible for the complex nature of the data presented in Figs. 13 and 14. Similar data for a lower strain-rate of 3.3 x 10⁻⁴ s⁻¹ are shown in Fig. 15.

In general, oxygen increases the yield and ultimate tensile stresses and decreases the total and uniform strains. The influence of oxygen on the work-hardening parameters is presented in Table IV. Over the oxygenconcentration range investigated, oxygen appears to decrease the workhardening rate.



Fig. 13. Uniform and Total Strain as a Function of Oxygen Concentration for Zircaloy-2 Specimens at 1300 and 1400°C. ANL Neg. No. 306-77-64 Rev. 1.



Fig. 14. Oxygen-concentration Dependence of Yield Stress and Ultimate Tensile Strength of Zircaloy-2 at 1300 and 1400°C. ANL Neg. No. 306-77-63 Rev. 1.





Temp, °C	Oxygen, wt%	Stage Number	Strain Interval	n	k, Pa	Average ₀₀ , Pa
1300	0.14	1	0.0004-0.014	0.88	8.2 x 10 ⁶	1.5 x 10 ⁶
		2	0.014-0.11	0.23	Z.1 x 10 ⁶	9.3 x 10
1300	0,33	1	0.0004-0.004	0.17	6.6 x 10 ⁵	18×106
		2	0.004-0.007	-1.30	-2.7×10^{1}	2.0 x 10 ⁶
1400	0.14	2	0.0004-0.013	1.00	4.3 x 10 ⁶	1.1×10^{6}
		Z	0.013-0.09	0.48	88×10	1.0 x 10 ⁶
1400	0.34	L	0.0004-0.009	0.60	1.7×10^{7}	6.8 x 10 ⁴
		2	0.01-0.025	-1.70	-1.2 × 10 ²	2.1 x 10 ⁶
1400	0.59	1	0.0004-0.013	0.81	1.2 × 107	7.4×10^{5}
		Z	0.014-0.090	-0.07	-3.1×10^{6}	5.3 x 10 ⁶

IV. MECHANICAL PROPERTIES OF ZIRCALOY-4

A. As-received Zircaloy-4

1. Stress-Strain Curves

Figures 16 and 17 represent the engineering-stress/engineeringstrain curves of Zircaloy-4 specimens deformed at a strain rate of 3.3 x 10^{-3} s⁻¹ for different temperatures. Figures 18 and 19 show Zircaloy-4 stressstrain curves at a faster strain rate. The general shape of these curves is similar to those for Zircaloy-2 in Sec. III.A.1. At both strain rates, the Zircaloy-4 specimens show superplastic elongations between 800 and 1000°C.

The temperature variation of the ultimate tensile strength of Zircaloy-4 at three strain rates is plotted in Fig. 20. A tendency toward a plateau formation in the temperature interval 400-500°C (Fig. 20a) is a manifestation of dynamic strain aging. An increasing extent of dynamic recovery may be responsible for the rapid drop in the ultimate tensile strength as the temperature increases from 700 to 1000°C (Fig. 20b).

Figure 21 shows the effect of temperature and strain rate on the uniform and total strain of Zircaloy-4. In Fig. 21a, the tendency for a maximum uniform strain near 200°C ($\dot{\varepsilon} = 3.3 \times 10^{-3} \text{ s}^{-1}$) and a minimum in total strain near 550°C ($\dot{\varepsilon} = 3.3 \times 10^{-2} \text{ s}^{-1}$) provide indications of dynamic strain aging. Note that, at the three strain rates in Fig. 21b, a maximum in total



Fig. 16. Engineering-stress/Engineeringstrain Curves for Zircaloy-4 Specimens Deformed at 700, 750, 850, 900, and 950°C. Neg. No. MSD-61494.





Fig. 18. Engineering-stress/Engineering-strain Curves for Zircaloy-4 Specimens Deformed at 700, 800, and 900°C. Neg. No. MSD-6149'(.



Fig. 19. Engineering-stress/Engineering-strain Curves for Zircaloy-4 Specimens Deformed at 1000, 1100, 1200, 1300, and 1400°C. Neg. No. MSD-61503.



Fig. 20. Ultimate Tensile Strength of Zircaloy-4 at Three Strain Rates over Two Temperature Intervals. (a) 23-700°C and (b) 700-1400°C. ANL Neg. No. 306-77-67 Rev. 1 and MSD-61498.



Fig. 21. Uniform Strain and Total Strain for Zircaloy-4 Specimens as a Function of Deformation Temperature at Three Strain Rates. (a) 23-700°C and (b) 700-1400°C. ANL Neg. Nos. 306-77-54 Rev. 1 and 306-75-199 Rev. 1.

strain occurs in the temperature interval 800-850°C, i.e., in the two-phase region near the α -phase boundary, which increases as the strain rate decreases. For a strain rate of 3.3 x 10^{-4} s⁻¹, failure did not occur after 285% strain and no concentrated neck was observed in the deformed gauge region. The test was terminated because the specimen length exceeded the uniform temperature zone in the furnace. The larger total strains at lower strain rates are consistent with the well-documented fact that superplasticity is enchanced at slower strain rates.

Note that the maxima at 800-850 and at ~1000°C and the minimum at 900-950°C in Fig. 21 resemble the hoop-strain-versus-temperature data obtained on internally pressurized tubes.²⁶⁻²⁹ Although the correlation between uniaxial and biaxial mechanical properties is not simple, the same microstructural factors may be responsible for the strain-temperature behavior in the two cases. Although the deformation behavior above 1100°C is complicated by a substantial amount of grain growth, the elongation peak at ~1300°C is probably representative of the material behavior, since a similar peak was also observed during tube-burst tests on Zircaloy under transient-heating conditions,³⁰ in which grain growth occurred to a lesser extent.
Deriving true-stress/true-strain curves for Zircaloy-4 specimens (i.e., from the data shown in Figs. 16-19) was difficult because the maximum load point on each curve appeared at small strains. The neck profile could not be measured because the specimens were enclosed in the vacuum chamber during tensile testing. The true-stress/true-strain data could be obtained only by straining several samples to different levels at a constant temperature, unloading, and measuring the minimum cross section. Data obtained in this fashion at 900°C are shown in Fig. 22. The true stress is almost constant from 0.2 to 0.97 (fracture point) true strain. A visible neck was observed in specimens strained beyond $\varepsilon = 0.30$.



2. Strain-rate Sensitivity

The effect of strain on the strain-rate sensitivity of Zircaloy-4 is shown in Figs. 23 and 24. Below ε = 0.05, the strain-rate sensitivity decreases



0.16

STRAIN, C

0.24

0.32



Fig. 24

Strain Dependence of Strain-rate Sensitivity of Zircaloy-4 at 700, 800, 850, and 900°C. Neg. No. MSD-61492.

at all temperatures as the strain increases. Beyond $\epsilon = 0.05$, at temperatures where superplastic deformation is observed (i.e., at 800, 900, and 1000°C), the strain-rate sensitivity either remains constant or increases as the strain increases. For other temperatures at higher strains, strain-rate sensitivity continues to decrease as the strain increases. As stated earlier, necking is not evident even after several percent strain beyond the maximum load point under superplastic deformation conditions. The strain-rate increase due to neck formation is not significant in such cases; therefore, at 700, 800, 850, and 900°C strain-rate cycling was continued to much larger strain values (Fig. 24). Note that at 850 and 900°C, where the extent of superplastic elongation is large, m actually increases with strain up to $\epsilon \simeq 0.9$.

Figure 25 shows the effect of strain and strain rate on the strainrate sensitivity of Zircaloy-4 at 800, 900, and 1000°C. In the single-phase region, at 800 and 1000°C, the strain-rate sensitivity is almost insensitive to strain rate, whereas in the two-phase region, at 900°C, the strain-rate



Fig. 25

Effect of Strain and Strain Rate on Strainrate Sensitivity of Zircaloy-4 at 800, 900, and 1000°C. Neg. No. MSD-61499. sensitivity is rather sensitive to strain rate. These data are presented in a different format in Fig. 26. Note that at 900°C, both Zircaloy-2 and -4 data show similar behavior. The temperature dependence of the strain-rate sensitivity of Zircaloy-4 was shown in Fig. 8.



Figure 27 shows the effect of strain on the strain-rate sensitivity of as-received Zircaloy-4 at 850°C. The four curves are for different ranges of strain-rate change. As the starting strain rate decreases from 3.3×10^{-3} to $3.3 \times 10^{-6} \text{ s}^{-1}$, the average value of the strain-rate sensitivity increases from 0.2 to 0.75. This observation is consistent with the data in Fig. 21, which show that the superplastic elongation peak at 850°C is strongly dependent on strain rate. The total elongation at 850°C increases with a decrease in strain rate. These results show that the amount of the tensile elongation under superplastic deformation conditions is related to the magnitude of the strain-rate sensitivity.



Fig. 27

Strain-rate Sensitivity of As-received Zircaloy-4 at 850°C as a Function of Strain. The four curves are for different strain-rate ranges: (A) 3.3×10^{-6} to $3.3 \times 10^{-5} \text{ s}^{-1}$, (B) 3.3×10^{-5} to $3.3 \times 10^{-4} \text{ s}^{-1}$, (C) 3.3×10^{-4} to $3.3 \times 10^{-3} \text{ s}^{-1}$, (D) 3.3×10^{-3} to $3.3 \times 10^{-2} \text{ s}^{-1}$. ANL Neg. No. 306-76-10.

3. Work-hardening Analysis

The Crussard-Jaoul work-hardening analysis was also applied to Zircaloy-4 tensile data. The results are shown in Tables V-VII and Figs. 28-31. The temperature dependence of the work-hardening exponent for the stage covering the major portion of the uniform strain region at three strain rates is plotted in Fig. 29. In the α -phase region ($\leq 805^{\circ}$ C), peak values of the work-hardening exponent are observed at 200° and near 700°C. At these two temperature regions, dynamic-strain-aging effects are observed. Positive work-hardening exponents are observed below 600°C, where transgranular deformation is important. In the two-phase region (810-980°C), negative workhardening exponents are observed due to intergranular deformation (grainboundary sliding). At lower strain rates, the extent of grain-boundary sliding is greater, and the work-hardening exponent therefore assumes larger negative values over a wider temperature interval.

Temp, °C	Stage Number	Strain Interval	n	k, Pa	Average σ ₀ , Pa
23	2	0.0004-0.07	0.31	3.7 x 10 ⁸	3.3×10^8
200	2	0.011-0.03	0.96	8.9×10^{8}	2.5 x 10 ⁸
	3	0.035-0.07	-0.36	-3.0 x 10 ⁷	3.8 x 10 ⁸
400	2	0.0004-0.015	0.94	1.1×10^9	1.2 x 10 ⁸
	3	0.020-0.09	-0.16	-8.4 x 10 ⁷	3.0 x 10 ⁸
550	2	0.0004-0.013	0.92	9.3×10^8	8.9×10^7
	3	0.014-0.11	0.03	5.7 x 10 ⁸	-3.9 x 10 ⁸
700	2	0.0004-0.05	0.86	8.1×10^{7}	6.8×10^{7}
750	2	0.011-0.05	0.89	7.7×10^{7}	6.1×10^{7}
1000	1	0.0004-0.0020	0.83	1.7 x 10 ⁸	4.3 x 10 ⁶
	2	0.0024-0.07	-0.26	-6.9 x 10 ⁵	8.7 x 10 ⁶
1100	1	0.0004-0.014	0.98	1.2×10^7	4.3×10^{6}
	2	0.015-0.11	0.64	4.5 x 10 ⁶	4.2×10^{6}
1200	1	0.004-0.025	0.97	8.3 x 10 ⁶	3.3×10^{6}
	2	0.03-0.11	0.39	2.8 x 10 ⁶	2.9 x 10 ⁶
1300	1	0.0004-0.014	0.98	7.7 x 10 ⁶	2.6 x 10 ⁶
	2	0.015-0.11	0.59	2.7 x 10 ⁶	2.5 x 10 ⁶
1400	1	0.0004-0.015	0.98	5.1 x 10 ⁶	2.3 x 10 ⁶
	2	0.02-0.09	0.61	1.9 x 10 ⁶	2.2 x 10 ⁶

TABLE V. Least-squares Work-hardening Constants for Deformation Stages of As-received Longitudinal Zircaloy-4 Specimens from Crussard-Jaoul Analysis; & = 3.3 x 10⁻² s⁻¹

Temp, °C	Stage Number	Strain Interval	n	k, Pa	Average σ ₀ , Pa
23	1	0.0004-0.0028	0.11	2.5×10^{8}	2.4×10^{8}
	2	0.0032-0.09	0.54	4.4×10^8	3.5 x 10 ⁸
200	2	0.0036-0.23	0.86	4.4×10^{8}	1.8×10^{8}
400	2	0.0004-0.15	0.26	2.0×10^8	6.5×10^{7}
550	1	0.0004-0.0036	0.05	1.2×10^{8}	4 l x 10 ⁶
	2	0.0040-0.014	0.82	5.6×10^8	8.7×10^{7}
	3	0.015-0.17	0.21	1.3×10^{8}	5.0×10^{1}
700	2	0.006-0.05	0.46	3.2×10^7	4.3×10^{7}
750	2	0.008-0.035	0.68	2.8×10^{7}	3.7 x 10 ⁷
800	1	0.0004-0.0012	0.36	7.1×10^{7}	1.9×10^{-1}
	2	0.0016-0.025	-0.04	-2.2×10^7	5.5×10^{7}
850	1	0.0004-0.0016	0.06	4.4×10^{7}	-8.8×10^6
	2	0.0016-0.015	-0.30	-5.5×10^{5}	2.5×10^{7}
900	1	0.0004-0.002	-0.08	-6.6×10^{6}	2.5 x 10 ²
	2	0.0024-0.0032	-7.00	-5.8×10^{-14}	1.5×10^{7}
950	2	0.0004-0.0016	-0.65	-8.5×10^3	9.5×10^6
1000	1	0.0004-0.004	0.37	6.5×10^{6}	2.8×10^{6}
	2	0.005-0.025	-0.85	-4.6×10^3	4.1×10^{6}
1050	1	0.0004-0.0012	-0.46	-3.1×10^3	2.9×10^{6}
	2	0.0016-0.0036	0.49	1.9×10^{6}	2.8×10^{6}
	3	0.0036-0.006	-1.80	-1.5×10^{0}	2.9 x 10 ⁶
1100	2	0.0004-0.09	0.04	3.2×10^{6}	-3.0×10^{4}
1150	1	0.0004-0.005	0.52	1.4×10^{6}	1.9×10^{6}
	2	0.005-0.01	0.77	3.5×10^{6}	1.9×10^{6}
	3	0.01-0.11	0.55	1.9×10^{6}	1.9×10^{6}
1200	1	0.0004-0.0032	0.91	4.1×10^{7}	1.2×10^{6}
	2	0.0026-0.13	0.17	2.3×10^{6}	6.0×10^{9}
1250	1	0.0004-0.005	0.67	5.9×10^{6}	1.2×10^{6}
	2	0.005-0.13	0.28	1.7 x 10 ⁶	9.5 x 10
1300	2	0.0024-0.09	0.68	3.1×10^{6}	1.3×10^{4}
1350	2	0.0028-0.11	0.64	1.4×10^{6}	1.2 x 10
1400	2	0.0032-0.09	0.89	1.8×10^{6}	1.3×10^{6}

TABLE VI.Least-squares Work-hardening Constants for DeformationStages of As-received Longitudinal Zircaloy-4 Specimens from
Crussard-Jaoul Analysis; $\dot{\epsilon} = 3.3 \times 10^{-3} \text{ s}^{-1}$

Temp, °C	Stage Number	Strain Interval	n	k, Pa	Average σ_0 , Pa
23	2	0.0004-0.09	0.31	3.5×10^8	2.8 x 10 ⁸
200	2	0.005-0.02	0.98	2.1×10^9	1.8 x 10 ⁸
	3	0.025-0.11	-0.30	-6.5 x 10 ⁷	4.3 x 10 ⁸
400	1	0.0004-0.0028	0.34	1.7×10^8	7.0×10^7
	2	0.0032-0.014	0.81	8.7 × 10 ⁸	8.7 x 10 ⁷
	3	0.015-0.11	-0.08	-2.3 × 10 ⁸	4.5 x 10 ⁸
550	1	0.0004-0.0024	-0.33	-1.9×10^{6}	9.2×10^7
	2	0.0028-0.02	0.78	4.3 x 10 ⁸	7.5×10^7
	3	0.025-0.09	-0.18	-3.8 x 10 ⁷	1.7×10^8
700	1	0.0004-0.002	-0.53	-3.2×10^4	3.2×10^7
	2	0.002-0.013	1.03	2.7 x 10 ⁸	3.1×10^7
	3	0.013-0.045	-0.24	-3.6 x 10 ⁶	4.4×10^7
750	1	0.0004-0.0024	-0.94	-1.9×10^{3}	2.8 x 10 ⁷
	2	0.0024-0.05	0.40	1.5 x 10 ⁷	2.6 x 10 ⁷
800	1	0.0004-0.0012	0.40	5.6×10^7	1.1×10^{7}
	2	0.0012-0.015	-0.52	-6.5 x 10 ⁴	1.7×10^{7}
850	1	0.0004-0.0016	0.55	1.0 x 10 ⁸	8.3×10^6
	2	0.002-0.025	-0.52	-1.1 x 10 ⁵	1.5×10^7
900	1	0.0004-0.0028	0.40	1.3×10^7	4.4 x 10 ⁶
	2	0.0028-0.015	-0.49	-5.7 x 10 ⁴	6.6 x 10 ⁶
950	1	0.0004-0.0012	0.34	1.1×10^7	1.3×10^{6}
	2	0.0016-0.009	-1.20	-1.3 x 10 ²	2.8 x 10 ⁶
1000	1	0.0004-0.0028	0.83	1.9×10^7	2.0 x 10 ⁶
	2	0.0032-0.009	-0.99	-4.6 x 10^2	2.3 x 10 ⁶
1100	1	0.0004-0.01	0.57	1.3×10^7	2.0 x 10 ⁵
	2	0.012-0.05	-0.86	-6.6 x 10 ³	1.5 x 10 ⁶
1200	1	0.0004-0.02	0.21	4.4 x 10 ⁵	8.1 x 10 ⁵
	2	0.025 、 17	0.73	8.9 x 10 ⁵	9.5 x 10 ⁵
1300	1 2	0.0004-0.03	0.24	9.9 x 10⁵ 7.2 x 10⁵	4.8×10^5 6.1×10^5
1400	2	0.002-0.07	0.34	5.6 x 10 ⁵	6.2×10^{5}

TABLE VII. Least-squares Work-hardening Constants for Deformation Stages of As-received Longitudinal Zircaloy-4 Specimens from Crussard-Jaoul Analysis; έ = 3.3 x 10⁻⁴ s⁻¹

-





Fig. 30

Temperature Dependence of Parameter k for Zircaloy-4 at Three Strain Rates. ANL Neg. No. 306-77-73 Rev. 1.



Fig. 31

Variation of σ_0 with Temperature for Zircaloy-4 at Three Strain Rates. ANL Neg. No. 306-77-112.

In the β -phase field (above 1000°C), extensive grain growth during deformation results in positive work-hardening exponents. The variation of k and n with temperature is similar (Fig. 30). Since the product kn is proportional to the slope of the stress-strain curve ($d\sigma/d\varepsilon$), the data in Figs. 29 and 30 imply a similar variation of the work-hardening rate with temperature. The parameter σ_0 decreases with an increase in temperature (Fig. 31).

4. Effect of Texture on the Mechanical Properties

Figure 32 shows the influence of texture on the temperature dependence of uniform strain of Zircaloy-4. The Zircaloy sheet had a texture in which most of the grains had basal poles tilted at ~30° away from the normal (thickness) direction toward the transverse direction. The longitudinal specimens therefore had more grains favorably oriented for prism slip than the transverse specimens. The diagonal orientation is midway between the longitudinal and transverse orientations. The results in Fig. 32 indicate that, below 800°C, the uniform strain is somewhat texture-dependent, whereas above this temperature the texture influence appears to be negligible.



The texture dependence of the total strain of Zircaloy-4 is shown in Fig. 33. The texture effect on the total strain is significant at the lowtemperature superplastic elongation peak at 850°C; this implies that prism slip plays a role in superplastic deformation at $\dot{\epsilon} = 3.3 \times 10^{-3} \text{ s}^{-1}$. Above 1000°C, grain growth during tensile testing is considerable; this makes it difficult to deduce the effect of texture on total elongation from the data in Fig. 33.



Fig. 33

Influence of Texture on Total Strain of Zircaloy-4 at Temperatures between Ambient and 1400°C. ANL Neg. No. 306-76-20 Rev. 1.

Figure 34 shows the variation of the ultimate tensile strength with temperature, which appears to be independent of texture at all temperatures. The strain-rate sensitivity is plotted against temperature at $\epsilon = 0.02$ for longitudinal and transverse specimens in Fig. 35. For both specimen orientations, the strain-rate sensitivity has a maximum value at 900°C. Except at 1400°C, texture does not have a significant effect on strain-rate sensitivity. The difference in the values of strain-rate sensitivity at this temperature may be related to variations in the grain growth.



Fig. 34. Temperature Dependence of Ultimate Tensile Strength for Longitudinal, Transverse, and Diagonal Zircaloy-4 Specimens. ANL Neg. No. 306-76-13 Rev. 1.

Fig. 35. Strain-rate Sensitivity of Longitudinal and Transverse Zircaloy-4 Specimens as a Function of Temperature. ε = 0.02. Neg. No. MSD-62581.

Figure 36 shows the variation of strain-rate sensitivity with strain for longitudinal and transverse Zircaloy-4 specimens. Although the strain-rate sensitivity of a transverse specimen is always greater than that of a longitudinal specimen, the strain dependence of strain-rate sensitivity is almost identical in both cases.



The effect of texture on the work-hardening exponent is shown in Fig. 37 (deduced from the data in Tables VI, VIII, and IX for longitudinal, transverse and diagonal specimens, respectively). Above 800°C, the work-hardening rate does not seem to be a strong function of the texture.



Fig. 37

Work-hardening Exponent as a Function of Temperature for Zircaloy-4 Specimens with Three Textures. ANL Neg. No. 306-77-78 Rev. 1.

Temp, °C	Stage Nunber	Strain Interval	n	ĸ, Pa	Average σ ₀ , Pa
22	2	0.0004-0.007	-0.15	-4.5×10^{7}	4.8 x 10 ⁸
	3	0.008-0.09	0.51	$4.6 \times 10^{\circ}$	3.4 x 10°
200	2	0.01-0.17	0.50	2.6×10^8	1.9 x 10 ⁸
400	2	0.009-0.17	0.21	2.2×10^8	3.4×10^7
700	2	0.01-0.07	0.42	3.2×10^{7}	4.7×10^{7}
750	2	0.008-0.05	0.67	3.0×10^{7}	4.1×10^{7}
850	2	0.0004-0.02	-0.06	-1.3×10^{7}	4.1×10^{7}
900	1	0.0004-0.0024	0.2	2.9×10^{7}	7.2 x 10 ⁶
	2	0.0024-0.01	-1.2	-1.3×10^3	1.7 x 10 ⁷
950	2	0.0004-0.0028	-0.65	-3.4×10^3	6.3 x 10 ⁶
1050	2	0.0004-0.025	-0.43	-2.5×10^4	3.5 x 10 ⁶
1200	ĩ	0.0004-0.007	-0.06	-1.1×10^{6}	3.3×10^{6}
	2	0.007-0.11	0.47	1.8×10^{6}	1.6 ж 10 ⁶
1300	2	0.0004-0.014	0.91	6.4×10^{6}	1.3×10^{6}
	3	0.015-0.11	0.29	1.6×10^{6}	9.4 × 10 ⁵
1400	2	0.0004-0.11	0.75	1.6×10^{6}	1.1 x 10 ⁶

TABLE VIII. Least-squares Work-hardening Constants for Deformation Stages of As-received Transverse Zircaloy-4 Specimens; $\dot{\varepsilon} = 3.3 \times 10^{-3} \text{ s}^{-1}$

TABLE IX. Least-squares Work-hardening Constants for Deformation Stages of As-received Diagonal Zircaloy-4 Specimens; $\dot{\epsilon} = 3.3 \times 10^{-3} \text{ s}^{-1}$

Temp, °C	Stage Number	Strain Interval	n	k, Pa	Average ₀₀ , Pa
23	2	0.0004-0.014	0.09	2.9 x 10 ⁸	2.0 x 10 ⁶
	3	0.015-0.09	0.47	4.1×10^{6}	3.4×10^{6}
400	2	0.0012-0.012	0.88	8.1 x 10^{8}	1.0 x 10 ⁸
	3	0.013-0.19	0.21	2.3×10^8	2.3×10^7
700	2	0.008-0.045	0.36	2.7×10^{7}	4.9×10^{7}
800	2	0.009-0.025	0.35	9.6 x 10 ⁶	2.9×10^{7}
850	1	0.0004-0.002	0.64	2.3×10^8	1.6×10^{7}
	2	0.002-0.02	-0.56	-1.2×10^{5}	2.4×10^{7}
950	1	0.0004-0.0024	0.18	1.5×10^{7}	6.6×10^5
	2	0.0024-0.009	-1.31	-2.2×10^2	6.3×10^{6}
1000	1	0.0004-0.0028	0.37	1.5×10^{7}	1.9×10^{6}
	2	0.0028-0.008	-2.0	-1.7×10^{0}	3.8×10^{6}
	3	0.008-0.015	0.85	3.1×10^{6}	3.8×10^{6}
1200	2	0.0004-0.004	0.83	4.4×10^{7}	1.1×10^{6}
	3	0.005-0.11	0.11	3.4×10^{6}	-2.7 x 10 ⁵
1300	2	0.0004-0.008	0.95	8.0×10^{6}	1.3×10^{6}
	3	0.008-0.09	0.21	1.1×10^{6}	9.8 x 10 ⁵
1400	2	0.0004-0.09	0.98	1.8×10^{6}	1.3 x 10 ⁶

The results presented in Figs. 32-36 indicate that texture effects in the deformation of Zircaloy are significant below 850°C and negligible above 1000°C. Factors that contribute to this type of behavior are (a) the greater symmetry of the body-centered-cubic (bcc) structure of β -phase Zircaloy (>1000°C) when compared with the low-temperature (<810°C) α -phase hexagonalclose-packed (hcp) material, (b) the change in the dominant deformation mode of α -phase Zircaloy as the temperature is increased³¹ (i.e., below 500°C, slip occurs on first-order prism planes; at higher temperatures, both basal and pyramidal slip take place), and (c) superplastic deformation at 850 and 1000°C, where possible grain rotation due to grain-boundary sliding diminishes texture effects.

5. Fine-grain-size Zircaloy-4

Figure 38 shows the total strain of $5-\mu m$ grain-size specimens as a function of deformation temperature at three initial strain rates. Note that, in Fig. 38a, an elongation minimum exists near 400°C that is associated with dynamic-strain-aging effects. In Fig. 38b, at all strain rates, elongation maxima occur at 800-900 and 1000-1050°C separated by a strain minimum near 950°C. The ~850°C total strain peak is highly dependent on strain rate. As the strain rate decreases, the magnitude and temperature of this peak increase. The elongation peak near 1000°C does not appear to be a strong function of strain rate. This observation is consistent with the negligible effect of heating rate on the circumferential elongation peak near 1000°C in transient -heating burst tests on internally pressurized stress-relieved fine-grain Zircaloy tubes.⁶ The tests on the 5- μ m grain-size Zircaloy were restricted to temperatures below 1100°C, since this material showed extraordinary grain growth above 1000°C.



Fig. 38. Total Strain as a Function of Temperature for 5-µm Grain-size Zircaloy-4 Specimens at Three Strain Rates (a) 23-700°C and (b) 700-1100°C. ANL Neg. No. 306-77-58 Rev. 1 and Neg. No. MSD-64145.

The total elongation-versus-temperature data for the 5- μ m grainsize material with a width-to-thickness ratio (w/t) of 5 and those for the 11- μ m grain-size material (w/t = 10) in Sec. IV.A.1 are consistent and imply that grain-boundary sliding at the α - β interface is an important factor in superplastic deformation of Zircaloy between 850 and 900°C. The shift in the temperature of the superplastic elongation maximum from 850 to 900°C as the grain size decreases from 11 to 5 μ m can be rationalized in terms of the increase in the α - α grain-boundary area. To achieve a grain-boundary film of β phase, the fine-grain material must have a larger volume fraction of β phase, which is present at the higher temperature. The increase in the magnitude of the superplastic elongation maximum at lower strain rates is consistent with the greater importance of grain-boundary sliding under these conditions.

Figure 39 shows the effect of temperature on the strain-rate sensitivity of fine-grain Zircaloy-4. The strain-rate sensitivity shows a peak at 900°C. These data are similar to the results presented earlier (Fig. 35) for the 11- μ m grain-size material. However, a comparison of the two sets of data at 900°C reveals that the strain-rate sensitivity increases from 0.35 to 0.60 as the grain size decreases from 11 to 5 μ m.



Figure 40 shows the effect of strain rate on the total and uniform strain of the 5- μ m grain-size material at 900°C. Both the uniform and total strains increase as the strain rate decreases. The large uniform strain, i.e., ~0.5 at a strain rate of 2.6 x 10⁻⁵ s⁻¹, may be due to the grain growth observed at the lower strain rates. Figure 41 shows that the strain-rate sensitivity also increases as the strain rate decreases.

The work-hardening characteristics of fine-grain Zircaloy at three strain rates are presented in Tables X-XII and Fig. 42. Similar to the 11- μ m grain-size material, the work-hardening exponent (Fig. 42) is positive below 600°C and negative for the temperature interval ~700-1000°C.



Fig. 40. Effect of Strain Rate on Total and Uniform Strain of Zircaloy-4 at 900°C. Neg. No. MSD-63139.

Fig. 41. Strain-rate Dependence of Strainrate Sensitivity of Zircaloy-4 at 900°C. Neg. No. MSD-63140.

TABLE X.	Least-squares Work-hardening Constants for Deformation
Stage	es of 5-µm Grain-size Zircaloy-4; $\dot{\epsilon} = 3.3 \times 10^{-2} \text{ s}^{-1}$

Temp, °C	Stage Number	Strain Interval	n	k, Pa	Average σ_0 , Pa
25	2	0.0004-0.09	0.57	5.2 x 10 ⁸	4.2×10^8
200	2	0.0002-0.13	0.34	2.8×10^8	1.8×10^{8}
400	2	0.02-0.17	0.36	2.5×10^8	1.2×10^8
550	2	0.013-0.15	0.28	1.7×10^8	8.3×10^{7}
800	2	0.0004-0.0016	-0.46	-2.1×10^5	6.5×10^{7}
850	2	0.0004-0.005	-0.57	-9.4×10^4	4.5×10^{7}
900	1 2	0.0004-0.002 0.0024-0.004	0.04 -3.3	5.9×10^7 -9.8 × 10 ⁻⁴	-1.2×10^7 3.4 x 10 ⁷
950	2	0.0004-0.0012	-0.76	-5.1×10^3	2.4×10^{7}
1000	1	0.0004-0.0016	-0.19	-4.4×10^4	8.6 x 10 ⁶
1050	1 2 3	0.0004-0.0032 0.004-0.011 0.012-0.09	1.0 -1.2 0.73	2.5×10^8 -5.3 × 10 ² 5.7 × 10 ⁶	5.1×10^{6} 6.2 x 10 ⁶ 5.8 x 10 ⁶

Temp, °C	Stage Number	Strain Interval	п	k, Pa	Average _{co} , Pa
25	1	0.0004-0.004	0.41	3.6×10^8	3.8×10^{8}
	2	0.005-0.15	0.64	5.6 x 10^8	4.0×10^{8}
200	2	0.025-0.25	0.62	5.0×10^{8}	2.0×10^{6}
400	2	0.015-0.19	0.50	3.0×10^8	1.3 x 10 ⁸
550	1	0.0004-0.0032	-0.32	$-1.5 \times 10^{\circ}$	1.3×10^{8}
	2	0.0036-0.13	0.72	1.6×10^{8}	1.2×10^{6}
700	2	0.012-0.05	1.2	9.8×10^7	5.3×10^{7}
800	2	0.0004-0.002	-0.77	-7.4×10^3	3.9×10^{7}
850	2	0.0004-0.0024	-0.96	-2.0×10^{3}	2.6×10^{7}
900	2	0.0004-0.004	0.34	3.2×10^{7}	1.3×10^{12}
950	2	0.005-0.012	-1.9	-3.4×10^{1}	1.8×10^{7}
1000	2	0.0004-0.0024	-0.13	-1.6×10^{5}	4.9×10^{6}
1050	1	0.0004-0.0032	-0.42	-6.9×10^3	3.4×10^{6}
	2	0.0036-0.05	0.92	3.3×10^{6}	3.3×10^{6}
1100	1	0.0004-0.012	0.01	5.8×10^{6}	-2.7 x 10 ⁶
	2	0.013-0.07	0.94	4.1×10^{6}	2.8×10^{6}

TABLE XI. Least-squares Work-hardening Constants for Deformation Stages of 5- μ m Grain-size Zircaloy-4; $\dot{e} = 3.3 \times 10^{-3} \text{ s}^{-1}$

TABLE XII. Least-squares Work-hardening Constants for Deformation Stages of 5-µm Grain-size Zircaloy-4; $\dot{\varepsilon} = 3.3 \times 10^{-4} \text{ s}^{-1}$

Temp, °C	Stage Number	Strain Interval	n	k, Pa	Average ₀₀ , Pa
25	2	0.011-0.13	0.53	5.2 x 10 ⁸	3.4×10^{8}
200	2	0.025-0.15	~0	1.6 x 10 ¹⁰	-1.5 x 10 ¹⁰
400	2	0.013-0.15	0.41	2.7 x 10 ⁸	1.3 x 10 ⁸
700	2	0.0004-0.0028	-0.54	-4.3×10^4	4.8 x 10 ⁷
800	1	0.0004-0.006	-0.23	-3.6 x 10 ⁵	2.2 x 10 ⁷
850	1	0.0004-0.002	0.55	1.1×10^{8}	7.5×10^6
	Z	0.0024-0.013	-1.1	-1.5×10^{3}	1.3×10^{7}
906	1	0.0004-0.0024	0.81	1.4×10^{9}	2.6×10^6
	~ 2	0.0028-0.09	-0.16	-2.3 x 10 ⁶	9.9×10^{6}
950	1	0.0004-0.045	-0,11	-1.8×10^{6}	7.6×10^{6}
1000	1	0.0004-0.002	-0.35	-9.8×10^{3}	2.4×10^{6}
	2	0.0024-0.007	0.47	9.6 x 10 ⁵	2.3×10^{6}
1050	1	0.0004-0.004	0.63	1.6 x 10 ⁶	1.8 x 10 ⁶
1100	1	0.0004-0.004	0.56	8.2×10^{6}	1.1×10^{6}
	2	0.005-0.07	-0.03	-4.7×10^{6}	7.0×10^{6}



6. Influence of Specimen Geometry

Since the total elongation observed in a tension test is a function of the gauge length-to-width (l/w) ratio of the specimen, the ASTM standard tensile specimen has a constant l/w ratio of four. Although the ASTM standard does not restrict the thickness of the sheet tensile sample, the w/t ratio has a significant influence on the ductility of Zircaloy. Figure 43 shows the effect of the w/t ratio on the strain-rate dependence of the strain-rate sensitivity. At low strain rates (<10⁻⁵ s⁻¹), a lower w/t ratio decreases the value of m significantly. Since the total strain measured during superplastic tensile deformation is a strong function of the strain-rate sensitivity, the value of w/t influences the superplastic ductility.



The strain-rate dependence of the total strain is shown in Figs. 44 and 45 for Zircaloy-4 with several grain sizes at two values of w/t. A comparison of the $11-\mu m$ grain-size curve in both figures indicates that w/t has a

51

strong influence on the total strain at intermediate strain rates $(10^{-5} \text{ to } 10^{-3} \text{ s}^{-1})$. These results directly show that the value of w/t influences the extent of superplastic ductility.





Dependence of Total Strain on Strain Rate for 11- and 50- μ m Grain-size Zircaloy-4 Specimens at 850°C. Neg. No. MSD-63153.

Fig. 45

Strain-rate Dependence of Total Strain of 5- and 11-µm Grain-size Zircaloy-4 Specimens at 850°C. Neg. No. MSD-63154.



7. Effect of Grain Size on Tensile Properties

As shown in Figs. 44 and 45, a larger grain size results in a smaller total elongation for Zircaloy-4, particularly at the intermediate strain



Fig. 46. Variation of Uniform Strain with Strain Rate for 5- and 11-µm Grain-size Zircaloy-4. Neg. No. MSD-63138.

rates $(10^{-5} - 10^{-3} \text{ s}^{-1})$, Fig. 44) at 850°C where superplastic ductilities are observed. Figure 46 shows the effect of grain size on the strain-rate dependence of the uniform strain of Zircaloy-4. At 850°C, as the strain rate decreases below 10^{-4} s⁻¹, the uniform strain increases for both grain sizes. The increase in uniform strain at 900°C for the 5-µm grainsize material is even more dramatic and occurs at strain rates below 10^{-3} s⁻¹. The high uniform strain at low strain rates is related to the substantial grain growth that occurs during deformation. When grain-boundary sliding contributes significantly to the overall defor-

0.0

mation, as is the case at these low strain rates, grain growth results in a hardening effect that increases the uniform strain. The larger uniform strain at the strain rate of $6.6 \times 10^{-6} \text{ s}^{-1}$ for 5-µm grain-size material at 850°C can be rationalized in terms of a greater contribution of grain-boundary sliding for the smaller grain size.

Figure 47 shows the variation of the yield stress with grain size for Zircaloy-4 at 850°C. As the strain rate decreases from 10^{-2} to 10^{-6} s⁻¹, the slope of the curves changes from small negative to large positive values.



Yield Stress as a Function of Grain Size for Zircaloy-4 at 850°C. Neg. No. MSD-63155.

Fig. 47

At high strain rates (>10⁻⁴ s⁻¹), transgranular deformation due to dislocation glide and creep is important. This case is similar to deformation at lower temperatures ($T < 0.4T_m$, where T_m is the absolute melting temperature), where the finer-grain-size material exhibits a higher flow stress than the coarse-grain material. At low temperatures the well-known Petch effect is observed, in which the slope of the logarithmic stress-grain size curve has a value of -0.5. In Fig. 47, as the strain rate decreases below 10^{-5} s⁻¹, grainboundary sliding becomes the predominant deformation mechanism and the slopes of the curves assume positive values that are consistent with viscous deformation at the grain boundaries.

8. Influence of Microstructural State

a. <u>Phase Distribution</u>. The effect of hold time at high temperature prior to deformation on the tensile elongation of Zircaloy illustrates the influence of the microstructural state on the tensile properties.

The experimental procedure for determining the tensile properties of Zircaloy in the present program involved a short (0.6-ks) hold time at the various test temperatures before the deformation was started. Figure 48 shows the effect of increasing the hold time from 0.6 to 7.2 ks before deformation on the total elongation of Zircaloy-4 specimens for temperatures between 850 and 1000°C. The additional hold time does not have a significant effect on the total strain at 900 and 950°C; however, the strain decreases considerably at 850 and 1000°C. The decrease in the total elongation at 1000°C is probably due to the grain coarsening that becomes significant at this temperature. At lower temperatures, grain growth is less important.



Metallographic examination of the specimens tested at 850° C revealed that the α grains in the deformed region of the specimen with a 0.6-ks hold time were completely equiaxed, whereas the grains in the specimen with the 7.2~ks hold time were elongated.

The dependence of the ductility on hold time is related to the microstructural changes that occur in the material. To examine the microstructures that exist at high temperature, specimens were oil-quenched from 850° C after 0.6- and 7.2-ks hold times. For a 0.6-ks hold time (Fig. 49), the microstructure consists of a large volume fraction of α -phase grains with a small volume fraction of β phase distributed both at the α -grain boundaries and within the α grains. During the 7.2-ks hold time, oxygen redistribution occurs and the thickness of the β -phase film at the α -grain boundaries increases (Fig. 50).



Fig. 49. Microstructure of a Zircaloy-4 Specimen Oil-quenched from 850°C after 0.6-ks Hold Time. A small volume fraction of β phase is distributed at the α -grain boundaries and within the α grains. Neg. No. MSD-63406.



Fig. 50. Microstructure of a Zircaloy-4 Specimen Oil-quenched from 850°C after 7.2-ks Hold Time. The volume fraction of β phase concentrated at the α -grain boundaries is greater than that in Fig. 49. Neg. No. MSD-63404.

Apparently the fine β -phase film at the α -grain boundaries in Fig. 49 was more conducive to grain-boundary sliding and hence to superplastic ductility. To determine the role of β phase at the α -grain boundaries on grain-boundary sliding, we conducted the following experiment. The asreceived Zircaloy was prestrained to $\varepsilon = 0.003$ at 700°C, heated to 850°C, held for 7.2 ks, and then strained to fracture. As a result of the small prestrain at 700°C, the large total-strain behavior was reestablished, even after the 7.2-ks hold time; i.e., the specimen fractured at $\varepsilon = 2.00$. Prestraining at 700°C introduced additional defects and dislocations in the structure. When this specimen was heated to the two-phase region, β phase precipitated on a finer scale as a result of the additional nucleation sites. During the 7.2-ks hold time at 850°C, β -phase coalescence (and also that of the χ phase present)* was difficult because of the interaction between β phase and the additional

^{*}The X phase exists below $870^{\circ}C^{32}$ and has the approximate composition of 5-9 wt % Fe, 0-2 wt % Cr, 1.5 wt % Sn, and the balance is zirconium.³³ As-received Zircaloy in the cold-rolled and the annealed condition contains a small volume fraction of randomly distributed ~1- μ m X-phase particles.

defects introduced during prestraining at 700°C. Since the β -phase film thickness at the α -grain boundary remained small, this specimen exhibited superplastic deformation similar to the specimen with the 0.6-ks hold time at 850°C without low-temperature prestrain.

b. α -phase Morphology. The morphology of transformed β phase (α' phase) strongly depends on the cooling rate through the two-phase region (~980-805°C). The types of transformed β -phase microstructure associated with the various approximate cooling rates are as follows: elongated wide grains (0.1°C/s, Fig. 51), parallel-plate structure (2°C/s, Fig. 52), basketweave structure (~500°C/s, Fig. 53), and martensitic structure (~10,000°C/s, Fig. 54). At intermediate cooling rates, i.e., ~100°C/s, the microstructure of transformed β phase consists of a mixture of the parallel-plate and basketweave structures. For a hypothetical LOCA situation in an LWR, the cladding microstructure during the heating transient is equiaxed, and if the maximum temperature exceeds 1000°C, the β phase transforms during cooling to either the parallel-plate or basketweave structure.

Figure 55 shows the effect of α -phase morphology on the engineering stress-strain behavior of Zircaloy-4 at 400°C. Without the hightemperature anneals at 850 and 1000°C, the microstructure of the as-received material is completely equiaxed α . This material has an ultimate tensile strength of ~150 MPa and a total strain of 0.42. After a 0.6-ks anneal at 850°C and cooling to 400°C, the structure will contain ~10% α in the transformed



Fig. 51

Microstructure of As-received Zircaloy-4 after Cooling through $\beta \rightarrow \alpha'$ Transformation at 0.1°C/s from 1000°C. ANL Neg. No. 306-77-38.



Fig. 52

Parallel-plate Structure of As-received Zircalcy-4 after Cooling through $\beta \rightarrow \alpha'$ Transformation at ~2°C/s from 1250°C. ANL Neg. No. 306-77-35.





Basketweave Structure of As-received Zircaloy-4 after Cooling through $\beta \rightarrow \alpha'$ Transformation at ~500°C/s from 1000°C. ANL Neg. No. 306-77-36.





Martensitic Structure of As-received Zircaloy-4 after Cooling through $\beta \rightarrow \alpha^{i}$ Transformation at ~10,000°C/s from 1000°C. ANL Neg. No. 306-77-34.



Fig. 55

Effect of α -phase Morphology on Engineeringstress/Engineering-strain Curve of Zircaloy-4 at 400°C. Neg. No. MSD-61851. β form, and the remainder will be equiaxed. This heat treatment will result in redistribution of oxygen and possibly other alloying elements during cooling at a rate of ~10°C/s through the two-phase region (980-805°C). Introduction of a small fraction of α' raises the ultimate tensile strength to ~160 MPa and lowers the total strain to ~0.34. After a 0.6-ks anneal at 1000°C, the entire microstructure at 400°C will be α' ; the ultimate tensile strength increases to ~190 MPa, and the ductility drops to ~0.2 total strain. Thus, apart from oxidation effects, the $\beta \rightarrow \alpha$ microstructural transformation also tends to decrease the ductility in Zircaloy that has undergone a high-temperature excursion into the β -phase field.

Figures 56 and 57 show scanning-electron-microscopy (SEM) fractographs of the as-received and annealed (0.6 ks at 1000°C) specimens, respectively, that were fractured at 400°C. The as-received specimen (Fig. 56) showed the typical equiaxed dimples that failed as a result of normal stress; the fracture surface was perpendicular to the tensile axis. The fracture region of the α' material (Fig. 57) showed elongated dimples that failed as a result of shear stress; the fracture surface was ~45° to the tensile axis.



Fig. 56

SEM Fractograph of As-received Zircaloy-4 Specimen Fractured at 400°C; $\dot{\epsilon} = 3.3 \times 10^{-3} \text{ s}^{-1}$. ANL Neg. No. 306-75-119 Rev. 1.





SEM Fractograph of Transformed β -phase Zircaloy-4 Specimen Fractured at 400°C; $\dot{\epsilon} = 3.3 \times 10^{-3} \text{ s}^{-1}$. ANL Neg. No. 306-75-125 Rev. 1.

Figure 58 shows the effect of α -phase morphology on the temperature dependence of the total strain. The lower curve in Fig. 58 shows the total strain of samples that were heated to 1000°C and then cooled at ~3°C/s through the two-phase region to the deformation temperatures, which resulted in a basketweave-type structure at the start of deformation. A comparison of these data shows that morphology has a pronounced effect on the total elongation. This is particularly evident between 800 and 850°C, where the large superplastic elongations were not observed for the transformed $\beta(\alpha')$ material.





Influence of α -phase Morphology on Total Strain of Zircaloy-4 at Temperatures between Ambient and 950°C. ANL Neg. No. 306-75-195 Rev. 2.

Figure 59 shows the temperature dependence of the ultimate tensile strength of Zircaloy-4 with equiaxed α and transformed β basketweave microstructures. The ultimate tensile strength for both types of specimens decreases continously as the temperature increases. Both curves show a hump near 400°C, which is a manifestation of dynamic strain aging. Figure 60 shows the temperature dependence of the strain-rate sensitivity of specimens with the two types of microstructure. Both curves show a minimum at 400°C, another indication of dynamic strain aging. The strain-rate sensitivity of



specimens with basketweave microstructure does not show a peak at 900°C (Fig. 60), which is consistent with the observed (Fig. 58) decrease in the extent of superplasticity in the transformed β phase. Figure 61 shows the strain-rate dependence of the strain-rate sensitivity of Zircaloy-4 at 850°C. Although the strain-rate sensitivity of the equiaxed material increases rapidly as the strain rate decreases, the basketweave structure does not show a significant strain-rate dependence. This is another indication that superplasticity is associated with an equiaxed structure at lower strain rates.



Figure 62 shows the variation of the ultimate tensile strength with temperature for the basketweave structure. Above ~500°C, the strength is greater for the higher strain rate; i.e., the normal strain-rate dependence is exhibited. However, at 200 and 500°C, dynamic strain aging results in a higher strength for the slower strain rate.



Variations of the total strain with temperature for basketweave structure are shown in Fig. 63 at two strain rates. The large elongation peak observed in equiaxed material at 850°C is not present in the data from the α'

TEMPERATURE (*C)



Fig. 63

Effect of Strain Rate on Total Strain of Zircaloy-4 with Basketweave Structure at Temperatures between Ambient and 1000°C. ANL Neg. No. 306-76-24 Rev. 1. specimens at $\dot{\epsilon} = 3.3 \times 10^{-3} \text{ s}^{-1}$. A microstructural examination of the α' specimen deformed at 850°C, $\dot{\epsilon} = 3.3 \times 10^{-3} \text{ s}^{-1}$, did not show a significant number of grains with equiaxed shape. When the strain rate was decreased to $3.3 \times 10^{-5} \text{ s}^{-1}$, however, the elongation increased to ~105%. The microstructure in the deformed region, in this case, revealed a large fraction of equiaxed grains. These results imply that, at sufficiently low strain rates (<10⁻⁵ s⁻¹), superplastic deformation reappears in transformed β -phase material as the equiaxed grain shape is achieved by grain-boundary migration.

Figure 64 shows the temperature dependence of the ultimate tensile strength of Zircaloy transformed from the β phase at cooling rates of 3 and 1°C/s (coarse basketweave and coarse parallel-plate structures, respectively). The strength behavior is almost identical for the two types of microstructure.

Figure 65 shows the corresponding total strain variation with temperature. At all temperatures, the basketweave structure is more ductile than the parallel-plate structure.



Fig. 64

Ultimate Tensile Strength as a Function of Temperature for Transformed β -phase Zircaloy-4 with Basketweave and Parallel-plate Structures. ANL Neg. No. 306-77-113.



Fig. 65

Temperature Dependence of Total Strain of Transformed β -phase Zitcaloy-4 with Basket-weave and Parallel-plate Structures. ANL Neg. No. 306-77-114.

Figures 66-69 show the cooling-rate dependence of yield stress, ultimate tensile strength, uniform strain, and total strain, respectively, at various temperatures. These data refer to tensile specimens with a widthto-thickness ratio (w/t) of 5. The transformed β structure was obtained by heating the as-received equiaxed homogeneous α structure into the β field (1000°C) and then cooling the specimen through the two-phase region (805-980°C) at the designated cooling rate. Cooling rates of less than 3°C/s could





Fig. 69

Total Strain as a Function of Cooling Rate for Transformed 8-phase Zircaloy-4 at Several Temperatures. Neg. No. MSD-63844.

be achieved in the Instron furnace prior to mechanical testing. Specimens with the faster cooling rate (>3°C/s) were annealed and quenched in another furnace and were heated to the mechanicaltest temperature in the Instron furnace.

The results show that, as the cooling rate increases, both the yield stress (Fig. 66) and the ultimate tensile strength (Fig. 67) increase somewhat. Two factors responsible for the higher strength at faster cooling rates are finer structure and less oxygen (and other alloying elements) redistribution. The acicular structure obtained at the cooling rate of 2.5 x 10² °C/s is very fine

compared to the lenticular structure produced at the slower cooling rate of 9×10^{-2} °C/s. Consistent with the well-known grain-size effect on the flow stress (Petch effect), the finer structure is stronger.

Solute redistribution during the phase transformation is another factor that contributes to the cooling rate dependence of the flow stress. It was shown² that considerable oxygen redistribution takes place at the slower cooling rates (<1°C/s) such that the central portion of the α' plates become enriched in oxygen and the boundary region between two adjacent α' plates becomes depleted in oxygen. At faster cooling rates (>3°C/s), the extent of oxygen (and other alloying elements) redistribution is less. The inhomogeneous structures produced at low cooling rates are weaker than the fairly homogeneous structures produced at the faster cooling rates.

The data in Figs. 68 and 69 show that the basketweave structure produced at the cooling rate of ~3°C/s exhibits uniform strain and total strain that are higher than those of the microstructures formed at the other cooling rates. This effect is believed to be due to the interlocking nature of the basketweave structure. Since the matrix orientation within two adjacent nonparallel subgrains ("fibers") is significantly different, the interlocking nature of basketweave structure gives rise to a work-hardening effect. The uniform strain of the basketweave structure at 200°C (Fig. 68) is higher than that at 25 and 400°C. This is probably due to the higher work-hardening rate at 200°C associated with dynamic strain aging. From a comparison of the mechanical properties of different microstructures in Figs. 66-69, we can conclude that the basketweave structure exhibits both good strength and good ductility.

Figures 70 and 71, respectively, show the temperature dependence of the uniform strain and total strain for four cooling rates (microstructures) and two specimen geometries. The peak uniform strain at 200°C (Fig. 70) is due to the high work-hardening rate associated with dynamic strain aging. At temperatures above 400°C, the uniform strain decreases rapidly due to the decrease in work-hardening rate because of the larger extent of dynamic recovery at higher temperatures. The plot of total strain versus temperature (Fig. 71) shows a minimum in the temperature interval





1000

Total Strain as a Function of Temperature for Transformed 8-phase Zircaloy-4 at Various Cooling Rates. ANL Neg. No. 306-77-110.

TOTAL

400-550°C, which can be attributed to dynamic strain aging. Above 600°C, the total strain increases due to dynamic recovery.

The work-hardening rate analysis of Zircaloy-4 with three types of transformed β microstructures is presented in Tables XIII-XVII and Figs. 72 and 73. The work-hardening exponent shows a maximum value between 200 and 400°C due to dynamic strain aging and negative values with a minimum between 700 and 900°C due to intergranular deformation.

9. Dynamic Strain Aging near 750°C

Figure 74 represents partial Instron load-elongation curves obtained from Zircaloy-4 specimens deformed at 750°C after various heat treatments. Curve A shows the partial load-elongation curve for as-received material (cold-rolled and vacuum-annealed at 732°C for 7.2 ks) deformed at 750°C after a 0.6-ks hold time at 750°C. Note the pronounced yield point. Curve B represents another specimen, which was deformed to a strain of ~0.02 under identical conditions and unloaded. This specimen was then held at 750°C for 14.4 ks and strained again at the same rate at 750°C. No yield point is evident in the load-elongation response in curve C.

TABLE XIII.	Least-squares Work-hardening Constants for Deformation
Stages	of Transformed β-phase Zircaloy-4; Cooling Rate
	$\simeq 0.1^{\circ} C/s; \dot{\epsilon} = 3.3 \times 10^{-3} s^{-1}$

Temp, °C	Stage Number	Strain Interval	n	k, Pa	Average σ ₀ , Pa
25	1	0.0004-0.0032	0.71	4.6×10^{8}	3.8×10^7
	2	0.0036-0.015	-0.86	-4.9×10^4	5.2×10^{7}
200	1	0.0004-0.0028	0.71	2.8×10^{9}	1.1×10^{8}
	2	0.0032-0.014	-0.25	-2.5×10^{7}	2.6×10^8
	3	0.015-0.13	0.38	2.6×10^8	1.4×10^{8}
400	1	0.0004-0.0028	0.68	1.2 x 10 ⁹	9.1×10^{7}
	2	0.0032-0.11	0.31	2.2×10^8	8.0×10^{7}
550	1	0.0004-0.0024	0.74	1.7×10^{9}	8.5×10^{7}
	2	0.0028-0.05	0.04	4.6×10^8	-2.6×10^8
700	1	0.0004-0.0036	0.83	1.1×10^{9}	6.1×10^{7}
	2	0.004-0.025	-0.57	-6.0×10^{5}	8.6×10^7
800	2	0.0004-0.01	-0.3	-1.5×10^{6}	5.1×10^{7}

Temp, °C	Stage Number	Strain Interval	n	k, Pa	Average ₀ , Pa
23	2	0.0004-0.13	0.16	3.9×10^8	2.0 x 10 ⁸
200	1	0.0004-0.02	-0.04	-3.8×10^8	7.0×10^{8}
	2	0.025-0.15	0.70	3.1×10^{8}	2.3×10^{8}
400	1	0.0004-0.0032	~0	6.3×10^{11}	-6.3 x 10 ¹¹
	2	0.0036-0.09	0.36	2.2×10^{8}	1.0×10^{8}
550	1	0.0004-0.0032	0.66	1.3×10^{9}	9.4×10^{7}
	2	0.0036-0.05	-0.13	-6.1×10^{7}	2.5×10^{8}
700	1	0.0004-0.008	0.07	1.6×10^8	-3.2×10^{7}
	2	0.009-0.015	-2.1	-2.0×10^{2}	9.0×10^{7}
800	1	0.0004-0.0016	0.69	6.8×10^{9}	3.3×10^{7}
	2	0.0020-0.012	-0.88	-2.5×10^4	4.8×10^{7}
850	1	0.0004-0.0016	0.61	3.0×10^{8}	2.0×10^{7}
	2	0.0020-0.015	-0.58	-9.4×10^4	3.1×10^{7}
900	1	0.0004-0.0016	0.15	1.8×10^{7}	7.3×10^{6}
	2	0.0020-0.015	-0.63	-3.1×10^4	1.6×10^{7}
950	1	0.0004-0.0028	0.58	5.9×10^{7}	4.9×10^{6}
-3	2	0.0032-0.014	-1.5	-1.5×10^{2}	7,8 x 10 ⁶

TABLE XIV. Least-squares Work-hardening Constants for Deformation Stages of Longitudinal Zircaloy-4 Specimens with Parallel-plate Structure; Cooling Rate $\simeq 1^{\circ}C/s$; $\epsilon = 3.3 \times 10^{-3} s^{-1}$

TABLE XV. Least-squares Work-hardening Constants for Deformation Stages of Longitudinal Zircaloy-4 Specimens with Basketweave Structure; Cooling Rate ~ 3°C/s; έ = 3.3 x 10⁻³ s⁻¹

Temp, °C	Stage Number	Strain Interval	n	k, Pa	Average o ₀ , Pa
23	2	0.0004-0.008	0.05	5.8 x 10 ⁸	-1.4×10^{7}
	3	0.008-0.09	0.37	3.3×10^8	3.8×10^8
200	2	0.0004-0.045	0.07	3.8×10^8	-3.9×10^{7}
	3	0.05-0.17	0.71	3.1×10^{8}	2.3 x 10 ⁸
400	1	0.0004-0.0012	0.84	4.0×10^{9}	1.1×10^{8}
	2	0.0016-0.11	0.17	2.3 x 10 ⁸	5.5×10^{7}
550	2	0.0004-0.04	0.03	5.6 x 10 ⁸	-3.5 x 10 ⁸
700	2	0.0004-0.004	0.17	1.8×10^{8}	1.9 x 10 ⁷
	3	0,005-0.01	-2,60	-7.6 x 10 ⁰	1.0 x 10 ⁸
800	1	0.0004-0.0012	0.64	4.5×10^{8}	2.6×10^7
	2	0.0016-0.01	-0,76	-3.5×10^4	3.8 x 10 ⁷
850	2	0.0004-0.015	-0.51	-1.7×10^{5}	3.1 x 10 ⁷
900	1	0.0004-0.002	0.16	3.2×10^{7}	-7.7×10^{5}
	2	0.0024-0.025	-0.58	-7.6×10^4	1.4×10^{7}
950	1	0.0004-0.002	-0.003	-2.0×10^{8}	2.1×10^8
	2	0.0024-0.015	-0.71	-1.0×10^4	6.5 x 10 ⁶

Temp, °C	Stage Number	Strain Interval	n	k, Pa	Average o ₀ , Pa
25		0.0004-0.0036	0.75	4.7 x 10 ⁹	4.1 x 10 ⁶
	2	0.004-0.07	0.07	1.1×10^{9}	-2.5 x 10 ⁸
200	2	0.0004-0.09	0.22	4.5 x 10 ⁸	1.8 x 10 ⁸
400	1	0.0004-0.0032	0.68	2.7 x 10 ⁹	1.9 x 10 ⁸
	2	0.0036-0.05	-0.09	-2.0×10^{8}	5,9 x 10 ⁸
550	1	0.0004-0.004	0,65	1.9×10^{9}	1.6×10^{6}
	2	0.005-0.025	-0.80	-7.1 x 10 ⁵	2.6×10^8
700	1	0.0004-0.006	0.16	1.5×10^{8}	5.3×10^{7}
	2	0.007-0.015	- 1.90	-4.7×10^{2}	1.3×10^8

TABLE XVI. Least-squares Work-hardening Constants for Deformation Stages of Longitudinal Zircaloy-4 Specimens with Martensitic Structure; Cooling Rate ~ 10,000°C/s; έ = 3.3 x 10⁻³ s⁻¹

TABLE XVII. Least-squares Work-hardening Constants for Deformation Stages of Longitudinal Zircaloy-4 Specimens with Basketweave Structure; Cooling Rate ~ 3°C/s; é = 3.3 x 10⁻⁵ s⁻¹

Temp, °C	Stage Number	Strain Interval	n	k, Pa	Average 0 ₀ , Pa
23	2	0,0004-0.13	0.23	5.3 x 10 ⁸	1.9 × 10 ⁶
200	2	0.0004-0.15	0.26	4.5 x 10 ⁸	1.0 x 10 ⁸
400	2	0.0004-0.07	0.12	2.8×10^8	1.8 x 10 ⁷
550	1	0.0004-0.0016	0.71	2.1×10^{9}	9.1 x 10 ⁷
	2	0.0020-0.02	-0.55	-1.2×10^{6}	1.5×10^8
700	1	0.0004-0.0016	0.90	2.8×10^{9}	3.7 x 10 ⁷
	2	0.0020-0.01	-1.30	-2.4×10^{3}	5.4×10^{7}
800	1	0.0004-0.0020	0.15	3.7×10^{7}	4.3 x 10 ⁶
	2	0.0024-0.01	-1.15	-1.8×10^{3}	2.1×10^{7}
850	1	0.0004-0.0024	0.64	1.4×10^{8}	5.8 x 10 ⁶
	2	0.0028-0.015	-1.21	-1.3×10^{3}	1.1×10^{7}
900	1	0.0004-0.0028	0.46	3.5×10^{7}	2.3 x 10 ⁶
	2	0.0032-0.02	-1.05	-2.3×10^{3}	5.6 x 10 ⁶
950	1	0.0004-0.0032	0.75	9.0×10^{7}	1.2×10^{6}
	Z	0.0036-0.015	-1.10	-1.1×10^{3}	3.2 × 10 ⁶



Fig. 72

Temperature Dependence of Work-hardening Exponent of Various Transformed β -phase Zircaloy-4 Structures. ANL Neg. No. 306-77-50 Rev. 1.



Fig. 73. Work-hardening Exponent of Basketweave-structure Zircaloy-4 as a Function of Temperature at Two Strain Rates. ANL Neg. No. 306-77-51 Rev. 1.



Fig. 74. Partial Load-elongation Curves for Zircaloy-4 Specimens Deformed at 750°C after Various Heat Treaments. (A) As-received strained to fracture; (B) as-received, partially strained; (C) strained to fracture after a 14.4-ks anneal at 750°C following B; (D) as-received, partially strained; (E) strained to fracture after a 14.4-ks anneal at 800°C following D. Neg. No. MSD-63142.

These results show that the yield point is probably not associated with a breakaway mechanism at 750°C. Curve D represents another specimen prestrained in an identical manner as curve B. However, this specimen was annealed at 800°C for 14.4 ks before additional straining at 750°C. Curve E shows the resultant load-elongation response. Notice the presence of serrations at fairly regular intervals on the load-elongation curve. This observation implies that some impurity element (or an intermediate phase containing multiple elements) went into solution during the anneal at 800°C and then, during subsequent straining at 750°C, the element (or elements) interacted with dislocations.

Since the yield-point phenomenon occurred at a relatively high temperature (700-750°C) in both Zircaloy-2 and -4, substitutional elements are likely to be involved. The observation that the yield point was equally pronounced in the two materials suggests that nickel is not likely to be responsible. The difference in the load-elongation curves at er anneals at 750 and 800°C suggests that the element responsible for strain aging is in solution at 800°C but not at 750°C. This implies a precipitation or disorder-order phase boundary between 750 and 800°C. An examination of binary-phase diagrams³⁴ of zirconium with the three remaining substitutional elements, viz., tin, iron, and chromium, reveals that such a phase boundary exists only in the zirconium-tin system.

For an alloy of zirconium with $\sim 1.5\%$ tin, the α phase exists above ~700°C, and below this temperature, two phases (α + Zr₄Sn) exist in equilibrium. Whether the presence of tin can cause a vield-point effect depends on the distribution of the tin atoms in the material during deformation. When the as-received material is given a final anneal above 700°C, tin goes into solution and remains in solution during furnace cooling. When this material is strained for the first time at temperatures between 700 and 750°C, a yield point is observed. Deformation in this temperature range induces the precipitation or ordering process of tin in the α -Zircalov matrix. Additional annealing below ~800°C does not result in the re-solution of the tin, and, thus, the yield point (or serrations on the load-elongation curve) is not observed. However, if the material is annealed at a temperature >800°C, tin goes back into solution and dynamic strain-aging effects are observed. Thus, the final heat treatment of the Zircaloy sample and the tin content of the material will determine whether the yield point is observed. These factors may be responsible for the absence of yield points in the data of Luton and Jonas.³⁵ We are presently conducting metallographic examinations to determine whether precipitation or ordering of tin actually occurs.

•

The work-hardening analysis of Zircaloy specimens deformed at 700°C and at various strain rates is presented in Tables XVIII-XX and Fig. 75. The yield-point effect was significant at strain rates greater than 10^{-3} s⁻¹ where the work-hardening exponent had positive values. At strain rates lower than 10^{-3} s⁻¹, generally the work-hardening exponent had lower

Strain Rate, é, s ⁻¹	Stage Number	Strain Interval	n	k, Pa	Average o _o , Pa
3.3×10^{-2}	Z	0.013-0.05	0.38	4.5×10^{7}	6.6 x 10 ⁷
3.3×10^{-3}	2	0.007-0.05	0.41	3.1×10^{7}	4.8×10^{7}
3.3×10^{-4}	Z	0.0028-0.035	0.35	3.1×10^{7}	3.5 x 10 ⁷
3.3 x 10 ⁻⁵	Z	0.0004-0.014	-0.25	-1.2×10^{6}	3.2×10^{7}
3.3×10^{-6}	2	0.0004-0.02	-0.30	-3.8×10^{5}	$2.1 \ge 10^{7}$

TABLE XVIII.Least-squares Work-hardening Constants for LongitudinalZircaloy-2Specimens Deformed at 700°C and Various Strain Rates

TABLE XIX. Least-squares Work-hardening Constants for 5-µm Grain-size Longitudinal Zircaloy-4 Specimens Deformed at 700°C and Various Strain Rates

Strain Rate, é, s ⁻¹	Stage Number	Strain Interval	n	k, Pa	Average ₀₀ , Pa
3.3×10^{-2}	2	0.03-0.05	0.79	5.2 x 10 ⁷	7.2 x 10 ⁷
3.3×10^{-3}	2	0.012-0.05	1.20	9.8×10^7	5.3×10^{7}
3.3 x 10 ⁻⁴	1 2 3	0.0004-0.006 0.007-0.025 0.03-0.05	-0.67 1.25 0.58	-8.6×10^{3} 8.9 x 10 ⁷ 2.3 x 10 ⁷	3.9 x 10 ⁷ 3.8 x 10 ⁷ 3.6 x 10 ⁷

Strain Rate, ć, s ⁻¹	Stage Number	Strain Interval	n	k, Pa	Average, o ₀ , Pa
3.3 x 10 ⁻²	2	0.0004-0.05	0.86	8.1×10^{7}	6.8×10^7
3.3×10^{-3}	2	0.006-0.05	0.46	3.2×10^7	4.3×10^{7}
3.3×10^{-4}	1	0.0004-0.002	-0.53	-3.2×10^4 2.7 × 10 ⁸	3.2×10^7 3.1×10^7
	3	0.013-0.045	-0.24	-3.6×10^{6}	4.4×10^{7}
2.6 x 10 ⁻⁵	2	0.0004-0.04	-0.04	-2.2×10^7	5.6×10^{7}
3.3 x 10 ⁻⁶	1	0.0004-0.004	-0.63	-1.7×10^4 1.0 × 10 ⁷	1.6×10^{7} 1.4×10^{7}

TABLE XX. Least-squares Work-hardening Constants for 11-µm Grain-size Longitudinal Zircaloy-4 Specimens Deformed at 700°C and Various Strain Rates





Strain-rate Dependence of Workhardening Exponent of Zircaloy at 700°C. ANL Neg. No. 306-77-66.

values (often negative) and the yield point behavior was absent. Thus, dynamicstrain-aging behavior is associated with peak values of work-hardening exponent.

Figure 76 represents engineering-stress/engineering-strain curves of Zircaloy 4 specimens deformed at 750°C after various heat treatments. Figure 74 presents some of these data. The stress-strain curve corresponding to a 0.6-ks hold time at 750°C prior to deformation (curve A) shows a load drop only at the start of plastic deformation (yield point); subsequently, there are no load drops or serrations. Curve B represents the stress-strain behavior during final loading of a specimen that was prestrained to 0.02 at 750°C, unloaded, and held for 14.4 ks at 750°C prior to restraining at 750°C. Note that the stress-strain curve is smooth with no load drops and the yield point is absent. Curve C corresponds to a specimen that was prestrained to 0.02 at 750°C, unloaded, annealed for 14.4 ks at 800°C, and then deformed at 750°C. Note that curve C has a higher flow-stress level than the two other specimens and shows load drops at regular intervals throughout the curve (serrations). This difference in the stress-strain behavior is related to the difference in the work-hardening characteristics presented in Table XXI. All specimens were held for 0.6 ks prior to deformation at the prestraining or restraining temperature. Specimens prestrained and heated to 800°C show positive and higher n values indicative of dynamic strain aging. The strainrate sensitivity does not seem to depend on the heat treatment (Fig. 77). The heat-treatment designation is the same as in Fig. 76.



Fig. 76

Engineering-stress/Engineering-Strain Curves for Zircaloy-4 at 750°C after Various Heat Treatments, ANL Neg. No. 306-77-75.

TABLE XXI. Least-squares Work-hardening Constants for 11-µm Grainsize Longitudinal Zircaloy-4 Specimens Deformed at 750°C after Various Heat Treatments; é = 3.3 x 10⁻³ s⁻¹

	2 C C C C C C C C C C C C C C C C C C C	1 (iz)			
Stage Number	Strain Interval	n	k, Pa	Average o ₀ , Pa	Comment ^a
2	0.008-0.035	0.68	2.8 x 10 ⁷	3.7 x 10 ⁷	A
Z	0.0004-0.03	0.05	2.8×10^{7}	2.4×10^{7}	В
2	0.013-0.04	0.08	2.5×10^{7}	2.4×10^{7}	С
2	0.0004-0.04	-0.07	-1.7×10^{7}	6.2 x 10 ⁷	D
2	0.0004-0.045	0.08	3.1×10^{7}	1.9 x 10 ⁷	E
2	0.0004-0.04	0.11	3.0×10^7	2.3 x 10 ⁷	F
2	0.0004-0.04	0.19	2.1×10^{7}	3.3 x 10 ⁷	G
2	0.009-0.04	0.46	3.1×10^{7}	4.7 x 10 ⁷	Н
2	0.009-0.035	0.49	3.2×10^{7}	6.0 x 10 ⁷	I

 $^{\mathbf{a}}\mathbf{A}$ = As-received.

B = 0.02 prestrain at 750°C, unload and immediately reload.

C = 14.4 ks hold time at 750°C prior to deformation.

D = 0.02 prestrain at 750°C, unload, 14.4-ks hold at 750°C.

E = 0.02 prestrain at 750°C, unload, 1.2-ks hold at 800°C.

F = 0.02 prestrain at 750°C, unload, 14.4-ks hold at 800°C.

G = 0.02 prestrain at 750°C, unload, 0.6-ks hold at 800°C.

H = Equiaxed homogeneous α -phase containing 0.46 wt % oxygen.

I = Equiaxed homogeneous α -phase containing 0.90 wt % oxygen.


10. Comparison of Zircaloy-4 Tensile Properties of Stressrelieved Tubing and Recrystallized Sheet

For comparison with the properties of Zircaloy sheet material, uniaxial-tensile tests have been performed on the Zircaloy-4 tubing³⁶ that is used in the ANL biaxial tube-burst tests and in other Nuclear Regulatory Commission research programs. The microstructure of the material in the as-received stress-relieved condition consists of fine (~2- μ m-wide) elongated grains. The wall thickness of the central 10-mm gauge length of the 70-mmlong specimens was reduced from 0.635 to 0.432 mm to ensure that fracture occurred away from the Swageloks that gripped the samples.

Figure 78 shows the temperature dependence of the ultimate tensile strength and the 0.2% yield stress of the tube material. The properties

Fig. 78

Ultimate Tensile Strength and Yield Stress of Zircaloy-4 Stress-relieved Tube and Recrystallized Sheet as a Function of Temperature. Neg. No. MSD-63395.



of the recrystallized sheet are shown for comparison. The fine grain size and the elongated grain shape of the cold-worked material are primarily responsible for the higher strength of the tubing below ~800°C. At higher temperatures, complete recrystallization of the tubing occurs during the tensile test, and the flow stress of the sheet and tube materials is essentially the same.

The total strain of the Zircaloy tube and sheet is plotted in Fig. 79 as function of the deformation temperature. The materials exhibit similar behavior, although the tubing is somewhat less ductile below ~400°C and more ductile at higher temperatures. The temperature dependence of the uniform strain of the two materials is shown in Fig. 80. The uniform strains are comparable below ~850°C; however, at 900°C, the tube material exhibits a large strain (~0.7) that was confirmed by duplicate tests. This behavior can be related to the microstructural changes that occur during deformation. Although the sheet material has an equiaxed grain structure before and during deformation, and grain growth is not significant, the tube recrystallizes and the small equiaxed grains grow considerably during the tensile test. The higher work-hardening rate, associated with grain growth, and the grainboundary-sliding deformation mechanism result in a larger uniform strain in the tube material at ~900°C.



Fig. 79

Total Strain of Zircaloy-4 Tube and Recrystallized Sheet as a Function of Temperature. Neg. No. MSD-63396.



Fig. 80

Uniform Strain of Zircaloy-4 Tube and Recrystallized Sheet as a Function of Temperature. Neg. No. MSD-63401.

A. Homogeneous Oxygen Distribution

Uniform oxygen distribution was achieved before deformation by homogenizing the oxygen-charged samples at an appropriate temperature for sufficent time. The effect of oxygen in Zircaloy-4 on the ultimate tensile strength is shown in Fig. 81. In α -phase Zircaloy-4 (Fig. 81a), strengthening due to oxygen is more significant at lower temperatures. Oxygen increases the strength of Zircaloy monotonically at these temperatures. Similar data, presented in Fig. 82 for β -Zircaloy at higher temperatures (>1200°C), showed a less systematic trend because with increasing oxygen concentration, the



Fig. 81. Effect of Oxygen on Ultimate Tensile Strength of Zircaloy-4 at Several Temperatures between 200 and 1100°C. ANL Neg. Nos. 306-77-62 Rev. 1 and 306-76-15.



Fig. 82

Variation of Ultimate Tensile Strength of Zircaloy-4 with Oxygen Concentration at 1200, 1300, and 1400°C. Neg. No. MSD-61848. two-phase $(\alpha + \beta)$ region is approached. Moreover, complications associated with the grain growth that occurred during homogenization and deformation of the Zircaloy-oxygen alloys also exist. In some instances, a single grain encompassed the width and thickness of the fracture region of the specimen. Therefore, differences in grain size and orientation may contribute to the complex behavior.

Figures 83 and 84 demonstrate the influence of oxygen on the uniform strain of Zircaloy-4. For temperatures below 750°C (Fig. 83a), an increase in the oxygen concentration decreases the uniform strain. Except for the



Fig. 83. Oxygen-concentration Dependence of Uniform Strain of Zircaloy-4 at Several Temperatures: (a) 200, 400, and 750°C and (b) 850, 900, 950, 1000, and 1050°C. ANL Neg. Nos. 306-77-61 Rev. 1 and 306-76-17.



Fig. 84

Oxygen-concentration Dependence of Uniform Strain of Zircaloy-4 at 1200, 1300, and 1400°C. Neg. No. MSD-61849. initial portion of the 850°C curve at Fig. 83b (≤ 0.5 wt % oxygen), the data indicate that the uniform strain increases as the oxygen concentration increases for temperatures between 850 and 1050°C. This is probably the result of the work-hardening rate, which increases with oxygen concentration. The effect of oxygen on the uniform strain at temperatures >1200°C is complicated (Fig. 84) by the same factors mentioned in connection with Fig. 82.

The effect of oxygen concentration in Zircaloy-4 on the total strain is shown in Figs. 85 and 86. In Fig. 85, the total strain decreases as the oxygen concentration increases, except for the 900, 950, and 1000°C curves. The temperature range of the two-phase ($\alpha + \beta$) field for Zircaloy is altered by oxygen additions; thus, the phase distributions as well as the compositions of the phases must be considered in an attempt to rationalize the results presented in Fig. 85. The influence of oxygen on the total strain at higher temperatures >1200°C (Fig. 86) is complicated by the same factors mentioned earlier.



Fig. 85. Total Strain of Zircaloy-4 as a Function of Oxygen Concentration at Several Temperatures: (a) 200, 400, and 750°C and (b) 700, 850, 900, 1000, and 1050°C. ANL Neg. Nos. 306-77-80 Rev. 1 and 306-76-16.



Fig. 86

Effect of Oxygen Concentration on Total Strain of Zircaloy-4 at 1200, 1300, and 1400°C. Neg. No. MSD-61850.

÷



Fig. 87. Effect of Oxygen on Temperature Dependence of Total Strain of Zircaloy-4. ANL Neg. No. 306-76-18 Rev. 1.

Figure 87 shows the influence of oxygen on the total strain of Zircaloy-4 as a function of temperature. As the oxygen concentration increases, the low-temperature superplastic elongation peak (i.e., at 850°C for 0.11 wt % oxygen) shifts to higher temperatures. This observation is consistent with the Zircaloy-oxygen phase diagram (Fig. 88) since the $\alpha/(\alpha + \beta)$ boundary shifts to higher temperatures with increasing oxygen concentration. Figure 87 also shows that the amount of elongation at the peak decreases as the oxygen concentration increases (i.e., 2.1 total strain at 0.11 wt % oxygen, compared with 1.4 total strain at 1.1 wt % oxygen). This decrease in elongation occurs because oxygen reduces the ductility of both the α and β phases. We could not determine the influence of oxygen on the

high-temperature superplastic elongation peak (i.e., the peak at ~1000°C for 0.11 wt % oxygen) because of the long homogenization times required at temperatures below ~850°C to obtain a small equiaxed grain structure with large oxygen concentrations.



Fig. 88. Pseudobinary Zircaloy-Oxygen Phase Diagram Determined from Metallographic and Resistivity Measurements. Dashed line corresponds to the zirconium-oxygen system. ANL Neg. No. 306-76-82 Rev. 1,

Figure 89 shows an SEM fractograph of a Zircaloy-4 specimen, with 2.3 wt % oxygen, fractured at 850°C. At 850°C, Zircaloy with this oxygen content has a single-phase α structure. The α phase exhibits transgranular cleavage fracture as a result of the high oxygen concentration. The river patterns are clearly visible in the grains.



Fig. 89

SEM Fractograph of Zircaloy-4 Specimen with 2.3 wt % Oxygen Fractured at 850°C; $\dot{\epsilon} = 3.3 \times 10^{-3} \text{ s}^{-1}$. ANL Neg. No. 306-75-132 Rev. 1.

The effect of oxygen on the work-hardening behavior of Zircaloy-4 is shown in Tables XXII-XXV and Figs. 90-95. The data in Tables XXII-XXIV and Figs. 90-94 refer to specimens that were cooled through the two-phase region after charging and/or a homogenization anneal before the mechanical testing. Therefore, these specimens contained nonequiaxed tranformed β -phase grains with various degrees of oxygen (and probably other alloying elements) redistribution.

Temp, °C	Oxygen, wt %	Stage Number	Strain Interval	л	k, Pa	Average ₀₀ , Pa
1000	0.25	1	0.0004-0.0024	0.71	2.3 x 10 ⁸	4.3 x 10 ⁶
		2	0.0028-0.0090	-2.20	-2.6×10^{0}	8.7×10^{6}
	0.40	1	0.0004-0.0028	0.78	1.2 x 10 ⁸	1.1×10^{7}
		2	0.0032-0.0150	-0.59	-4.5 x 10 ⁴	1.3×10^{7}
1100	0.25	1	0.0004-0.0016	0.21	2.4×10^{7}	-1.6×10^{6}
		2	0.0020-0.0050	-0.82	-2.8×10^{3}	5.4 x 10 ⁶
	0.39	2	0.0004-0.0700	0.21	1.5 x 10 ⁶	2.5×10^{6}
1200	0.24	1	0.0004-0.0050	0.26	2.3×10^{6}	1.6 x 10 ⁶
		2	0.0060-0.0250	-0.46	-3.1×10^4	2.5×10^{6}
	0.40	1	0.0004-0.0020	0.68	9.6×10^7	2.2 ж 10 ⁶
		2	0.0024-0.0060	- 3. 50	-1.7×10^{-4}	4.1×10^{6}
	0.59	1	0.0004-0.0020	0.75	1.1×10^{7}	4.4 ж 10 ⁶
		2	0.0020-0.0050	-0.56	-2.9×10^3	4.6 x 10 ⁶
1 300	0.39	2	0.0004-0.0140	0.92	6.1 x 10 ⁶	1.5×10^{6}
		3	0.0150-0.0500	0.11	1.5×10^{6}	7.0 x 10 ⁵
	0.60	2	0.0004-0.0500	0.14	2.1×10^{6}	7.1×10^{5}
	0.92	2	0.0004-0.0040	0.82	1.8×10^{8}	3.4×10^{6}
		3	0.0050-0.0200	-1.50	-4.2×10^2	6.8 x 10 ⁶
1400	0.27	2	0.0012-0.0150	1.10	1.2×10^{7}	1.1 x 10 ⁶
		3	0.0200-0.0700	-0.31	-1.4×10^{5}	1.8 x 10 ⁶
	0.39	2	0.0004-0.0040	0.87	1.4×10^{7}	1.0 x 10 ⁶
		3	0.0050-0.0900	0.19	1.5×10^{6}	5.9 x 10 ⁵
	0.90	2	0.0004-0.0060	0.93	3.4×10^7	2.4×10^{6}
		3	0.0070-0.1300	0.20	3.9×10^{6}	1.3×10^{6}
	1.00	2	0.0004-0.0032	0.50	1.4×10^{7}	3.2 x 10 ⁶
		3	0.0036-0.0080	- 1.90	-5.6 x 10°	4.3×10^{6}

TABLE XXII.Least-squares Work-hardening Constants for LongitudinalSpecimens of Zircaloy-4/Oxygen Alloys; & = 3.3 x 10⁻³ s⁻¹

Temp, °C	Oxygen, wt %	Stage Number	Strain Interval	n	k, Pa	Average σ ₀ , Pa
1000	0.25	1	0.0004-0.0028	0.53	6.0×10^{7}	1.8 x 10 ⁶
1000		2	0.0032-0.0150	-1.30	-6.1×10^{2}	5.7 x 10 ⁶
	0.32	1	0.0004-0.0016	0.56	7.4×10^{7}	3.5 x 10 ⁶
		2	0.0020-0.0130	-1.06	-1.2×10^{3}	6.6×10^{6}
	0.41	1	0.0004-0.0028	0.06	2.2×10^{7}	-8.0×10^{6}
		2	0.0032-0.0150	-1.20	-7.4 x 10 ²	7.6×10^6
1100	0.25	1	0.0004-0.0020	0.50	2.1×10^{7}	1.2×10^{6}
		2	0.0024-0.0050	-2.16	-5.1×10^{-1}	2.4×10^{6}
	0.31	1	0.0004-0.0020	0.15	9.8×10^6	-9.6×10^{5}
		2	0.0024-0.0040	- 3.90	-6.9×10^{-6}	3.0×10^{6}
	0.42	1	0.0004-0.0032	0.73	1.6×10^7	2.6×10^6
		2	0.0036-0.0100	-1.10	-2.7×10^{2}	3.0×10^{6}
1200	0.23	1	0.0008-0.0080	1.00	2.9×10^{7}	7.9 x 10 ⁵
		2	0.0090-0.0450	-0.30	-1.2×10^{5}	1.5×10^{6}
	0.59	1	0.0004-0.0028	0.44	7.0×10^{6}	2.2×10^{6}
		Z	0.0032-0.0050	-2.30	-1.4×10^{-1}	2.8×10^{6}
1 300	0.24	1	0.0004-0.0040	1.06	2.5 x 10 ⁷	7.1 x 10 ⁵
		2	0.0040-0.0700	~0.00	9.9 x 10 ⁷	-9.8×10^{7}
	0.55	1.	0.0004-0.0016	0.86	1.5×10^{7}	9.7 x 10 ⁵
		2	0.0020-0.0300	0.05	6.3 x 10 ⁵	5.9 x 10 ⁵
	0.87	1	0.0004-0.0050	0.79	5.3 x 10^7	1.3×10^{6}
		2	0.0060-0.0300	-0.85	-4.8×10^{3}	2.6×10^6
1400	0.31	1	0.0004-0.0130	1.05	3.7×10^{6}	7.6 x 10 ⁵
		2	0.0140-0.0500	0.28	4.7×10^{5}	6.6 x 10 ⁹
	0.80	1	0.0004-0.0020	0.56	1.0×10^{7}	1.1×10^{6}
		Z	0.0024-0.0070	-1.40	-3.6×10^{1}	1.7×10^{6}
	0.99	1	.0.0004-0.0050	0.57	1.7×10^{7}	1.5×10^{6}
	20 22	2	0.0060-0.0120	-2.05	$-5.3 \times 10^{\circ}$	2.6 x 10 ⁶

i

TABLE XXIII. Least-squares Work-hardening Constants for Longitudinal Specimens of Zircaloy-4/Oxygen Alloys; $\dot{\varepsilon} = 3.3 \times 10^{-4} \text{ s}^{-1}$

TABLE XXIV. Least-squares Work-hardening Constants for LongitudinalSpecimens of Zircaloy-4/Oxygen Alloys; $\dot{e} = 3.3 \times 10^{-5} \, s^{-1}$

•

Temp. °C	Oxygen, wt %	Stage Number	Strain Interval	n	k, Pa	Average ₀₀ , Pa
1000	0.23	1	0.0004-0.0028	0.85	5.1×10^{7}	1.2 x 10 ⁶
		2	0.0032-0.0060	-0.97	-1.3×10^{3}	1.9 x 10 ⁶
	0.43	1	0.0004-0.0024	0.63	6.7×10^{7}	1.6 x 10 ⁶
		2	0.0028-0.0500	-0.16	-1.4×10^{6}	7.0×10^{6}
1100	0.43	1	0.0004-0.0032	0.76	3.4×10^{7}	2.1×10^{6}
		Z	0.0036-0.0250	-0.72	-8.7×10^{3}	3.1 x 10 ⁶
1200	0.43	1	0.0004-0.0036	0.84	2.1×10^{7}	1.2 x 10 ⁶
		2	0.0040-0.0070	-1.30	-1.1 × 10 ²	1.5×10^{6}
1 300	0.79	1	0.0004-0.0032	0.83	2.6×10^{7}	4.4×10^{5}
		2	0.0036-0.0300	-0.37	-5.9×10^4	1.2 x 10 ⁶
	0.94	1	0.0004-0.0040	0.86	3.5×10^7	1.7×10^{6}
		2	0.0050-0.0450	-0.50	-2.7×10^4	2.5×10^{6}
1400	0.31	1	0.0004-0.0150	1.05	1.9 × 10 ⁶	3.8×10^{5}
		2	0.0200-0.1100	0.46	6.2 x 10 ⁵	3.2×10^{5}
	1.00	1	0.0004-0.0040	0.96	1.0×10^8	5.9 x 10 ⁵
		2	0.0050-0.0250	-0.21	-6.8×10^5	3.3 x 10 ⁶

Temp. °C	Oxygen Concentration, wt %	Stage Number	Strain Interval	n	k, Pa	Average ₀₀ , Pa
200	0.50	2	0.0004-0.0024	-0.13	-4.7 x 10 ⁷	6.2 x 10 ⁸
400	0.50	2	0.0080-0.09	0.88	5.4 x 10 ⁸	1.9 x 10 ⁶
750	0.46	2	0.009-0.04	0.46	3.1×10^{7}	4.7 x 10 ⁷
8 50	0.46	2	0.0004-0.014	-0.49	-1.1 × 10 ⁵	2.9×10^{7}
900	0.45	1 2	0.0004-0.002 0.0024-0.015	0.20 -0.57	2.5×10^7 - 5.0 × 10 ⁴	1.1 x 10 ⁷ 2.0 x 10 ⁷
950	0.47	Z	0.0004-0.005	-0.44	-6.1 x 10 ⁴	1.7×10^{7}
1000	0.49	2	0.0004-0.0036	-1.00	-7.6 x 10 ²	1.4 x 10 ⁷
1050	0.50	Z	0.0004-0.0036	-1.50	-1.4×10^{1}	1.1×10^{7}
1100	0.50	2	0.0004-0.002	-1.10	-3.5×10^2	8.8 x 10 ⁶
750	0.90	2	0.009-0.035	0.49	3.2 x 10 ⁷	6.0×10^{7}
900	0.80	2	0.0004-0.02	-0.18	-1.3×10 ⁶	2.9×10^{7}
950	0.82	1 2	0.0004-0.0012 0.0016-0.02	0.67 -0.45	2.7 x 10 ⁸ -2.4 x 10 ⁵	1.3 x 10 ⁷ 2.1 x 10 ⁷
1000	0.81	1 2	0.0004-0.0012 0.0016-0.02	0.75 -0.41	4.0 x 10 ⁸ -2.6 x 10 ⁵	1.3 x 10 ⁷ 1.9 x 10 ⁷
1050	0.82	Z	0.0004-0.008	-0.37	-2.8 x 10 ⁵	1.6×10^{7}
1100	0.82	2 3	0.0004-0.005 0.006-0.011	0.83 - 3.70	2.1×10^{8} 4.0×10^{-3}	8.6 x 10 ⁶ 1.2 x 10 ⁷
850	1.15	2	0.011-0.03	0.30	1.8×10^{7}	4.0×10^{7}
9 50	1.14	1 2	0.0004-0.0036 0.004-0.03	0.82 -0.89	5.9×10^8 - 3.5 × 10 ⁴	1.9 x 10 ⁷ 3.0 x 10 ⁷
1000	1.14	2	0.0004-0.025	-0.08	-1.1×10^{7}	4.0 x 10 ⁷
1050	1.10	1 2	0.0004-0.0024 0.0028-0.02	0.23 -0.47	3.1 ж 10 ⁷ -1.9 ж 10 ⁵	9.4 x 10 ⁶ 2.0 x 10 ⁷

TABLE XXV. Effect of Oxygen Concentration on Least-squares Workhardening Constants of Longitudinal Equiaxed α -phase Zircaloy-4 Specimens; $\dot{e} = 3.3 \times 10^{-3} s^{-1}$



Fig. 90

Work-hardening Exponent of Zircaloy-4 at 1000°C as a Function of Oxygen Concentration, ANL Neg. No. 306-77-60 Rev. 1.



Fig. 91

Work-hardening Exponent of Zircaloy-4 at 1100°C as a Function of Oxygen Concentration. ANL Neg. No. 306-77-55 Rev. 1.





Work-hardening Exponent of Zircaloy-4 at 1200°C as a Function of Oxygen Concentration. ANL Neg. No. 306-77-56 Rev. 1.



Fig. 93

Work-hardening Exponent of Zircaloy-4 at 1300°C as a Function of Oxygen Concentration. ANL Neg. No. 306-77-68 Rev. 1.





Fig. 95

Work-hardening Exponent of Zircaloy-4 at 1400°C as a Function of Oxygen Concentration. ANL Neg. No. 306-77-53 Rev. 1.

Fig. 94

Temperature Dependence of Work-hardening Exponent of Zircaloy-4 with Several Oxygen Concentrations. ANL Neg. No. 306-77-77 Rev. 1.

The work-hardening rate of Zircaloy-oxygen alloys depends on the influence of oxygen on the microstructure of the material. From the phase diagram (Fig. 88), an increase in the oxygen concentration changes the microstructure of the alloy from β to $\alpha + \beta$ at temperatures above 1000°C. The data in Tables XXII-XXIV indicate that small additions of oxygen increase the workhardening rate of β -phase Zircaloy as long as the composition of the alloy is not close to the two-phase boundary. Near the two-phase boundary the workhardening rate shows a minimum (negative value of n). The data in Figs. 90-92 all show a minimum value of n at intermediate oxygen concentrations (0.25-0.4 wt % oxygen), and the oxygen concentration at which the minimum occurs increases as the temperature increases from 1000 to 1200°C. In Figs. 93 and 94, the minimum is not readily apparent, since the oxygen concentration range is away from $\beta/(\alpha + \beta)$ phase boundary. With a higher work-hardening rate, the Considere criterion for maximum load point on the load-elongation curve $d\sigma/d\epsilon = \sigma$, is satisfied at the highest strain. Therefore, the uniform strain behavior in Fig. 84 is consistent with Figs. 92-94.

The influence of strain rate on the work-hardening rate is complicated by the concomitant grain growth that occurs during deformation. At slower strain rates, the grain-boundary sliding component is expected to be more significant, which may lower the work-hardening rate. However, the longer test times result in significant grain growth, which tends to increase the workhardening rate. These opposing factors operate to various degrees and result in a complicated behavior.

The data in Table XXV and Fig. 95 were obtained from specimens that were oxygen-charged and homogenized in the α -phase region and therefore had an equiaxed grain structure at the start of deformation. Consistent with the phase diagram, the region of negative work-hardening exponent shifts to higher temperature with increasing oxygen concentration.

Figures 96-100 show the effect of strain on the strain-rate sensitivity of Zircaloy-4 containing different oxygen concentrations in the range of



Fig. 96. Strain Dependence of Strainrate Sensitivity of Zircaloy-4 with 0.5 wt % Oxygen at 700 and 850°C. Strain-rate change between 3.3 x 10⁻⁴ and 3.3 x 10⁻³ s⁻¹. ANL Neg. No. 306-75-196 Rev. 1.



Fig. 97. Effect of Strain on Strain-rate Sensitivity of Zircaloy-4 at 1000°C with 0.11, 0.21, and 0.34 wt % Oxygen. Strainrate change between 3.3 x 10⁻⁴ and 3.3 x 10⁻³ s⁻¹, ANL Neg. No. 306-75-197 Rev. 1.









Effect of Strain on Strain-rate Sensitivity of Zircaloy-4 with 0.11, 0.22, and 0.35 wt % Oxygen at 1200°C. Strain-rate Change between 3.3×10^{-4} and 3.3×10^{-3} s⁻¹. ANL Neg. No. 306-75-193 Rev. 1 Effect of Strain on Strain-rate Sensitivity of Zircaloy-4 with 0.25 wt % Oxygen at 1000, 1200, and 1400°C. Strain-rate change between 3.3 x 10⁻⁴ and 3.3 x 10⁻³ s⁻¹. ANL Neg. No. 306-75-191 Rev. 1.

Fig. 99

Fig. 100

Strain Dependence of Strain-rate Sensitivity of Zircaloy-4 with 0.11, 0.2, and 1.0 wt % Oxygen at 1400°C. Strain-rate change between 3.3 x 10^{-4} and 3.3 x 10^{-3} s⁻¹. ANL Neg. No. 306-75-189 Rev. 1.

0.11-1.0 wt % for temperatures between 700 and 1400°C. Near the lowtemperature peak in the curve of total elongation versus temperature, i.e., 850°C in Fig. 21, small additions of oxygen do not decrease the strain-rate sensitivity of Zircaloy. The m values in Fig. 96 for an alloy with 0.5 wt % oxygen are essentially the same as those for the as-received material with 0.11 wt % oxygen (see Fig. 24). Similar to the behavior of as-received material, the strain-rate sensitivity of Zircaloy with 0.5 wt % oxygen is not straindependent at 850°C (near the peak on the strain-temperature diagram), but decreases as the strain increases at 700°C (away from the superplastic elongation peak).

At higher temperatures (≥ 1000 °C), an increase in the oxygen concentration decreases the strain-rate sensitivity to the extent that the superplastic elongation behavior is diminished or eliminated. Grain growth during homogenization of the oxygen-charged specimens may also contribute to the decrease in elongation. Figure 101 shows the oxygen dependence of the strain-rate sensitivity parameter m for temperatures of 1200, 1300, and 1400°C. These results are consistent with Zircaloy tensile data obtained in an air environment³⁷ in which oxidation during the test eliminated the superplastic elongation behavior at temperatures >1000°C.

B. Composite Material with Oxygen Gradients in α and β Phases

Since the duration of a hypothetical LOCA in an LWR is quite short, steam oxidation of the Zircaloy cladding will produce $ZrO_2/\alpha/\beta$, $ZrO_2/\alpha/(\alpha + \beta)$,



Fig. 101

Oxygen-concentration Dependence of Strain-rate Sensitivity of Zircaloy-4 at 1200, 1300, and 1400°C. ANL Neg. No. 306-76-9 Rev. 1.

or ZrO_2/α composites with oxy gen-concentration gradients in each phase. Although an understanding of the influence of oxygen on the mechanical properties of the individual α and β phases will be useful in interpreting the mechanical properties of composite materials, it is essential to conduct tensile tests on composites.

Tensile specimens of the as-received Zircaloy-4 (with 0.11 wt % oxygen) were heated in oxygen for 0.72 ks at 800°C to increase the oxygen level to 0.8 wt %. This charging process produced a surface oxide layer and α phase with a steep oxygen-concentration gradient. The oxide surface layer was quite thin, so that ~95% of the total thick ess was α phase. However, the number of phase layers and/or the oxygen-concentration gradient in each case can change during hightemperature deformation in the Instrom furnace. Metallographic examinations of the fractured specimens were used to determine the final layer thicknesses. Similarly, composite specimens

with different total oxygen concentrations were produced with the following charging conditions: [0.45 wt % (700°C, 0.72 ks); 1.12 wt % (800°C, 2.7 ks); and 1.9 wt % (900°C, 0.9 ks].

Figure 102 compares the temperature dependence of the total strain for the homogenized and composite specimens. Figure 103 shows the





Temperature Dependence of Total Strain of Homogenized and Composite Zircaloy-4 with 0.8 wt % Oxygen. ANL Neg. No. 306-76-23 Rev. 1.





Ultimate Tensile Strength as a Function of Temperature for Homogenized and Composite Zircaloy-4 with 0.8 wt % Oxygen. ANL Neg. No. 306-76-22.

corresponding comparison of the ultimate tensile strength as a function of temperature. The composition and volume fraction of each phase in the homogenized specimens at the test temperature varied according to the phase diagram, although no macroscopic oxygen gradient existed across the specimen. The structures of the composite specimens at the deformation temperature were as follows: oxide $|\alpha|$ oxide (below 800°C), oxide $|\alpha/(\alpha + \beta)/\alpha|$ oxide (between 800 and 1000°C, with smaller thicknesses of the oxide and α layers as the temperature increased), and $\alpha/\beta/\alpha$ (above 1000°C). An oxygenconcentration gradient was present in each phase. Metallographic examination of the deformed samples revealed that the oxide surface was cracked severely below 1000°C and the oxide/ α interface was separated at several places. Thus, only the α and β layers of the composite were load-bearing. The results in Fig. 102 demonstrate that the presence of a brittle layer on the surface decreases the total strain (ductility) of Zircaloy considerably, although the ultimate tensile strength does not change significantly (Fig. 103).

Figures 104-107 show the strain dependence of the strain-rate sensitivity of Zircaloy-oxide composites for various total oxygen contents at 800, 850, 900, and 1100°C, respectively. At 800°C (Fig. 104), the strain-rate sensitivity is relatively independent of strain; however, the magnitude of the strain-rate sensitivity decreases as the oxygen concentration increases. At 850 and 900°C (Figs. 105 and 106), the strain-rate sensitivity increases with strain due to the superplastic characteristics of the load-bearing α phase or







Fig. 105

Strain Dependence of Strain-rate Sensitivity of Zircaloy-4/Oxide Composites with 0.78 and 1.00 wt % Total Oxygen at 800°C. Strainrate change from 3.3 x 10⁻⁴ to 3.3 x 10⁻³ s⁻¹. Neg. No. MSD-62579.

Strain-rate Sensitivity as a Function of Strain for Zircaloy-4/Oxide Composites with 0.43 and 0.82 wt % Total Oxygen at 850°C. Strainrate change from 3.3 x 10^{-4} to 3.3 x 10^{-3} s⁻¹. Neg. No. MSD-62585.



86





Strain-rate Sensitivity as a Function of Strain for Zircaloy-4/Oxide Composites with 0.77, 1.12, and 1.88 wt % Total Oxygen at 900°C. Strain-rate change from 3.3 x 10⁻⁴ to 3.3 x 10⁻³ s⁻¹. Neg. No. MSD-62587.





Strain Dependence of Strain-rate Sensitivity of Zircaloy-4/Oxide Composites with 0.78 and 1.17 wt % Total Oxygen at 1100°C. Strain-rate change from 3.3 x 10⁻⁴ to 3.3 x 10⁻³ s⁻¹. Neg. No. MSD-62589.

 $\alpha + \beta$ layer. At 1100°C, the superplastic effect diminishes and the strain dependence of strain-rate sensitivity shows a normal behavior, i.e., an initial



Fig. 108

Strain-rate Sensitivity of Zircaloy-4/ Oxide Composites as a Function of Total Oxygen Concentration at 850, 900, and 1100°C. Strain-rate change from 3.3 x 10^{-4} to 3.3 x 10^{-3} s⁻¹; $\epsilon = 0.02$. Neg. No. MSD-62584. decrease in the m value, which then becomes independent of strain (Fig. 107).

Figure 108 indicates the effect of total oxygen concentration on the strain-rate sensitivity of Zircaloy-4/oxide composites. The data points for an oxygen concentration of 0.11 wt % correspond to the as-received homogeneous material. At all three temperatures, the strainrate sensitivity decreases as the oxygen concentration increases. These results do not agree with measurements by Boček,³⁷ which showed that the magnitude of the strain-rate sensitivity of Zircaloy-oxide composites increased during simultaneous oxidation and deformation in an air environment.

Specimens with oxygen concentrations <1.75 wt % were prepared by charging oxygen into the as-received 11- μ m grain-size Zircaloy-4 material at temperatures below the α -phase boundary (i.e., below 805°C). Thus, before heating in the mechanical testing furnace, the specimens had an oxide $|\alpha|$ oxide composite structure. For mechanical-test temperatures <805°C, this composite structure was maintained during ϵ formation with only minor changes in the oxygen gradient in the α phase because of the low diffusivity of oxygen. As the deformation temperature was increased into the $\alpha + \beta$ (>805°C) and β -phase (>980°C) regions, the composite structure changed significantly during deformation due to the large diffusivity of oxygen in the β phase. When the deformation temperature increases above ~1000°C, the oxide layer dissolves and the relative thicknesses of the α and β layers vary as functions of holding time, deformation time, and deformation temperature.

The microstructures of deformed composite specimens showed significant differences for deformation temperatures below and above ~1000°C. In specimens deformed below ~1000°C, both the oxide and α layers exhibited significant cracking (Fig. 109). In specimens deformed above ~1000°C, the oxide layer in most cases was absent and the α layer did not show cracking (Fig. 110).



|--|



Microstructure of Zircaloy-4/Oxide Composite Specimen Deformed at 800°C Showing Cracks in Oxide and α Layers. ANL Neg. No. 306-77-89 Rev. 1. Microstructure of Zircaloy-4/Oxide Composite Specimen Deformed at 1100°C Showing No Cracks in α Layer. ANL Neg. No. 306-77-88 Rev. 1.

The effect of total oxygen concentration in the composite on the yield stress, ultimate tensile strength, uniform strain, and total strain is presented in Figs. 111-114. respectively. Below 900°C, the yield stress (Fig. 111) and ultimate tensile strength (Fig. 112) are not sensitive to oxygen concentration. This is the net result of two compensating factors. As the oxygen concentration increases, the oxide-layer thickness also increases. Since the layer cracks during deformation, the load-bearing area decreases. On the other hand, the diffusion of oxygen into the base α (or $\alpha + \beta$) material increases its strength. Above 950°C, both the yield stress and ultimate strength increase as the oxygen concentration increases since the α layer does not crack and carries a portion of the load.





Oxygen-concentration Dependence of Yield Stress of Zircaloy-4/Oxide Composites at Several Temperatures. ANL Neg. No. 306-77-57 Rev. 1.



Fig. 112

Ultimate Tensile Strength of Zircaloy-4/Oxide Composites as a Function of Oxygen Concentration at Several Temperatures. ANL Neg. No. 306-77-70 Rev. 1.



Fig. 113

Effect of Oxygen Concentration on Uniform Strain of Zircaloy-4/Oxide Composites at Several Temperatures. ANL Neg. No. 306-77-52 Rev. 1.







The effect of oxygen on the uniform strain is complicated (Fig. 113), but generally, the uniform strain decreases as the oxygen increases. As shown in Fig. 114, oxygen decreases the total strain of composite material.

The influence of oxygen on the work-hardening rate of composite specimens is shown in Table XXVI and Fig. 115. Generally, an increase in the

Temp. °C	Oxygen Concentration, wt %	Stage Number	Strain Interval	n	k, Pa	Average o ₀ , Pa
200	0.43	 Z	0.008-0.02	0.87	1 3 x 10 ⁹	2.0×10^{8}
		3	0.025-0.045	-1.40	-1.3×10^{5}	2.8×10^{8}
	0.77	2	0.0004-0.025	0.51	7.3×10^8	1.8×10^{8}
	1.10	2	0.0004-0.025	0.39	5.7 × 10 ⁶	1.6×10^8
400	0.45	1	0.0004-0.0016	0.12	7.3 x 10 ⁷	7.5×10^{7}
		2	0.002-0.015	1.10	3.2×10^9	1.1×10^{8}
		3	0.02-0.05	- 1.40	-1.2×10^{5}	1.9 x 10 ⁸
	0.75	2	0.0004-0.013	0.83	2.1×10^{9}	1.0 x 10 ⁸
		3	0.014-0.04	-0.88	$-1.4 \times 10^{\ell}$	2.3 x 10 ⁸
	1.11	2	0.0004-0.011	0.41	7.4×10^8	6.5×10^{7}
700	0.45	2	0.008-0.04	-0.09	-6.2×10^{7}	1.6 x 10 ⁸
	0.75	1	0.0004-0.008	0.55	2.8×10^{8}	4.9 x 10 ⁷
		2	0.009-0.045	-0.36	-4.8×10^{6}	9.6 x 10 ⁷
	1.12	1	0.0004-0.011	0.28	1.6×10^{8}	2.8×10^7
		2	0.012-0.045	-0.66	-1.2×10^{6}	9.7×10^{7}
800	0.45	2	0.0004-0.04	-0.05	-4.7 x 10 ⁷	1.0 x 10 ⁸
	0.82	2	0.0004-0.04	-0.09	-3.4×10^{7}	9.7 x 10 ⁷
	1.12	1	0.0004-0.012	0.36	9.6×10^7	2.9×10^{7}
		2	0.013-0.035	-0.96	-9.6×10^4	5.5 x 10 ⁷
850	0.45	2	0.0004-0.025	-0.23	-3.2×10^{6}	4.3×10^{7}
	0.81	2	0.0004-0.02	-0.12	-1.3×10^{7}	5.8×10^7
900	0.45	2	0.0004-0.025	-0.41	-6.6 x 10 ⁵	3.5 x 10 ⁷
	1.11	1	0.0004-0.007	-0.17	-4.6 x 10 ⁶	4.4×10^{7}
		2	0.008-0.012	- 2. 50	-3.9×10^{0}	3.4×10^7
	1.89	1	0.0004-0.0012	0.36	1.5 x 10 ⁸	1.2×10^{7}
		2	0.0012-0.004	-0.67	-7.7 x 10 ⁴	3.2×10^7
		3	0.004-0.008	-2.50	-8.4×10^{-1}	3.0×10^{7}
950	0.45	1	0.0004-0.006	-0.12	-3.7×10^{5}	2.1×10^{7}
		2	0.007-0.013	-0.10	-2.4 x 10°	2.2×10^{7}
	0.82	1	0.0004-0.0024	0.69	5.2 x 10°	1.0×10^{7}
		2	0.0024-0.009	-0.71	-8.4×10^{-1}	2.4×10^{7}
		3	0.009-0.015	- 1.90	-1.4 x 10	2.3 x 10 ⁻
1000	0.44	2	0.0004-0.03	-0.38	-1.5×10^{3}	$1.1 \times 10'$
	0.82	1	0.0004-0.0024	0.34	7.1 x 10'	$1.1 \times 10'$
		2	0.0028-0.025	-0.59	-1.4×10^{-1}	2.6 x 10'
	1.14	2	0.0004-0.008	-0.25	-2.2×10^{-3}	3.1 x 10 ⁷
	1.84	2	0.0004-0.0036	-0.87	-4.7 x 10°	2.9 x 10'
1100	0.44	2	0.0004-0.07	0.05	7.4 x 10 ⁶	-2.1 x 10°
	0.85	1	0.0004-0.004	0.90	2.7 x 10"	8.2 x 10°
		2	0.005-0.05	-0.45	$-3.1 \times 10^{\circ}$	1.4 x 10'
	1.16	Z	0.0004-0.012	-0,54	-1.3 x 10 ⁻	2.1 x 10'
1200	0.76	2	0.0004-0.05	-0.20	-5.3 x 10 ⁵	7.1 x 10 ⁶
	1.88	1	0.0004-0.0028	0.73	8.6 x 10 ⁸	6.2 x 10 ⁶
		2	0.0032-0.025	-1.10	-4.8 x 10'	2.2×10^7
1400	0.43	2	0.0004-0.11	0.89	1.8 × 10 ⁶	1.3×10^{6}
	0.77	1	0.0004-0 0028	-0.71	-1.1×10^{3}	1.6×10^{6}
		2	0.0032+0.11	0.50	1.6 x 10"	1.5 x 10°

TABLE XXVI. Least-squares Work-hardening Constants for Zircaloy-4/Oxide Composite Specimens; $\dot{e} = 3.3 \times 10^{-3} \text{ s}^{-1}$.



Fig. 115

Temperature Dependence of Work-hardening Exponent of Zircaloy-4/Oxide Composites with Several Oxygen Concentrations. ANL Neg. No. 306-77-71 Rev. 1. oxygen content decreases the work-hardening rate of the major deformation stage. This observation is consistent with the data in Fig. 113, which tend to show lower uniform strains for higher oxygen concentrations.

C. Effect of Cooling Rate on Properties of Composite Specimens

The furnace used to charge oxygen into Zircaloy specimens does not have a quenching facility. After the power to the furnace is switched off, the cooling rate is slow, i.e., ~0.15°C/s. Specimens with oxygen concentrations >1.75 wt % were oxidized in the β -phase field above 1000°C, and the slow cooling rate resulted in a coarse lenticulor structure. These composite specimens were placed in another furnace with a quenching facility, heated into the β -phase field, quenched to room temperature at various cooling rates,

and then deformed in the Instron furnace. Because of this heat-treatment cycle, the microstructure and oxygen distribution at the start of deformation are a function of several parameters apart from the cooling rate in the annealing furnace, viz., the cooling rate in the charging furnace, the heating rate in the annealing furnace, and the temperature and time of annealing. There-fore, the data points for the 250°C/s cooling rate in Figs. 116-119 refer to a complex microstructure not representative of the fast cooling rate alone.



Fig. 116. Uniform and Total Strain of Zircaloy-4/ Oxide Composites with 4.0 wt % Oxygen as a Function of Cooling Rate. Neg. No. MSD-63836.



Fig. 117. Ultimate Tensile Strength and Yield Stress as a Function of Cooling Rate for Zircaloy-4/ Oxide Composites with 4.0 wt % Oxygen. Neg. No. MSD-63837.



Fig. 118. Effect of Cooling Rate on Yield Stress and Ultimate Tensile Strength of Composite Zircaloy-4 Specimens with 4.4 wt % Oxygen. Neg. No. MSD-63839.





The strength and ductility of the material with 4.0 wt % oxygen, in Figs. 116 and 117, respectively, are quite low, particularly at the 0.09°C/s cooling rate. Oxygen penetration into the material at the 1100°C charging temperature and the significant amount of oxygen redistribution that occurred during slow cooling through the phase transformation cause a degradation in the properties. As the cooling rate increases, both the strength and ductility increase; however, the total strain is quite small (<0.02) at all cooling rates.

The yield stress and ultimate tensile strength of the specimens with 4.4 wt % oxygen, in Fig. 118, are not strongly dependent on the cooling rate. The ductility data in Fig. 119 show a peak at a cooling rate of $\sim 17^{\circ}$ C/s. Despite the somewhat higher total oxygen concentration (4.4 wt %) in these specimens, the ductility is significantly higher than for the composite material with 4.0 wt % oxygen. This observation implies that the time-temperature history is an important variable with regard to the mechanical properties. For specimens with 4.4 wt % oxygen, the oxidation temperature was 1000°C. Consequently, oxygen diffusion into the central β (or two-phase) region occurred to a lesser extent than in the material with 4.0 wt % oxygen, for which the oxidation temperature was 1100°C.

VI. ACTIVATION ENERGY FOR HIGH-TEMPERATURE DEFORMATION OF ZIRCALOY

If the deformation of Zircaloy is assumed to be thermally activated, the strain rate \dot{e} during steady-state deformation can be expressed in terms of a constant A, shear modulus G, Boltzmann constant k, Burgers Vector b, stress σ , grain size D, absolute temperature T, grain-size exponent p, stress exponent n (i.e., n = 1/m in Eq. 3), activation energy for deformation Q, and the gas constant R; i.e.,

$$\dot{\epsilon} = \frac{AG}{kT} \left(\frac{b}{D}\right)^{p} \left(\frac{\sigma}{G}\right)^{n} \exp\left(-\frac{Q}{RT}\right).$$
(4)

A plot of log ekT/G versus reciprocal temperature at a constant value of σ/G and grain size can be used to evaluate Q. As pointed out by Gifkins,³⁸ two precautions should be taken to obtain a meaningful value for Q; i.e., the temperature range should be selected so that the stress exponent (and hence, the strain-rate sensitivity) is approximately constant and no change in the deformation mechanism occurs for the particular value of σ/G .

More complete results for the strain-rate dependence of the 0.2% yield stress of 5- and 11- μ m grain-size Zircaloy-4 at several temperatures between 700 and 1000°C are shown in Figs. 120 and 121, respectively. These data suggest a tendency toward an S-shape sigmoidal curve, as observed in other superplastic materials, although a constant threshold stress³⁹ at low strain rates is not evident. The comparatively large slope of the curves at temperatures between 850 and 950°C and stress levels ≤ 6 MPa is indicative of a high value of m (stage-II deformation), whereas at higher stress levels the m-value is lower (stage-III deformation).



Fig. 120. Strain-rate Dependence of 0.2% Yield Stress of 5-µm Grain-size Zircaloy-4 at Several Temperatures between 700 and 950°C. ANL Neg. No. 306-76-256 Rev. 1.



Fig. 121. Strain-rate Dependence of 0.2% Yield Stress of 11-µm Grain-size Zircaloy-4 at Several Temperatures between 700 and 1000°C. ANL Neg. No. 306-76-242 Rev. 1.

The deformation stages II and III refer to the intermediate and high strain-rate ranges, respectively, on a plot of the flow stress σ versus the strain rate $\dot{\epsilon}$, e.g., Figs. 120 and 121, in contrast to the work-hardening stages 1, 2, and 3, obtained from plots of $d\sigma/d\epsilon$ versus strain ϵ . As the temperature increases, the transition from stage-II to -III behavior occurs at higher strain rates.

The temperature dependence for the stage-III deformation process for the 5- and 11- μ m grain-size material is shown in Figs. 122 and 123, respectively, in which the strain rate and compensated strain rate, at constant stress σ or σ/G , are plotted as a function of reciprocal temperature. The activation energies for stage-III deformation, defined by either

$$Q = -R \frac{d \ln(\dot{\epsilon})}{d(1/T)} \Big|_{\sigma}$$
(5)

or

$$Q = -R \frac{d \ln(\dot{e}kT/G)}{d(1/T)} \Big|_{\sigma/G}$$
(6)

were computed from the slopes of the respective curves.



Fig. 122. Temperature Dependence of Strain Rate for Stage-III Deformation of 5-µm Grain-size Zircaloy-4. ANL Neg. No. 306-76-250 Rev. 2.



Fig. 123. Temperature Dependence of Strain Rate for Stage-III Deformation of 11-µm Grain-size Zircaloy-4. Neg. No. MSD-63431.

The shear modulus of zirconium (calculated from the relationship given by Hearmon⁴⁰ and the elastic-constant data of Fisher and Renken⁴¹) was used to obtain the compensated strain-rate curves. The activation energies in the range of 270-352 kJ/mole obtained from Eqs. 5 and 6 are stressdependent; i.e., the Q values in Figs. 122 and 123 decrease with an increase in stress. This behavior is consistent with the increase in the m value with temperature over the 700 to 900°C range (Figs. 39 and 60 for the 5- and 11-µm grain-size material, respectively). The values of the activation energy are in good agreement with those obtained by Kearns et al.⁴² for the deformation of Zircaloy-hydrogen alloys (304-365 kJ/mole) and in reasonable agreement with the activation energies for self-diffusion in α -Zircaloy (218,⁴³ 276,⁴⁴ and 282⁴⁵ kJ/mole).

VII. SUPERPLASTICITY OF ZIRCALOY

The Zircaloy-2 and -4 tensile data show that both exhibit superplastic deformation between 800 and 1000°C. Maximum superplasticity is observed between 850 and 900°C. The exact temperature of maximum elongation is a function of strain rate (Figs. 21 and 38). At this temperature, Zircaloy is in the two-phase region with a high volume fraction of α phase. The small amount of β , located at the α -grain boundaries, is similar to a grain-boundary film (Fig. 124) and restricts the growth of the α grains.



(a)

(b)

Fig. 124. Microstructure near (a) Shoulder Region ($\varepsilon \simeq 0$) and (b) Fracture Region ($\varepsilon \simeq 2.1$) of a Zircaloy-4 Specimen Deformed at 850°C at a Strain Rate of $3.3 \times 10^{-3} \text{ s}^{-1}$. Etched and anodized. Polarized light. Neg. Nos. MSD-61479 and -61455.

The microstructures of the specimens deformed at 800-850°C were dependent on strain rate. At 850°C and $\dot{\varepsilon} = 3.3 \times 10^{-3} \text{ s}^{-1}$, the deformed region exhibited completely equiaxed α grains surrounded by a β film at the grain boundaries (Fig. 124). Moreover, the deformed region has a smaller grain size (~7 µm) than the starting material (~11 µm). Because of the fine equiaxed structure, substantial superplasticity occurred at this strain rate.

At 850°C and a strain rate of $3.3 \times 10^{-1} \text{ s}^{-1}$, the deformed region of the specimen had elongated grains (see Fig. 125), grain-boundary sliding was less pronounced, and the ductility was low. At low strain rates and in fine-grain-size material, where superplastic phenomena are significant, equiaxed grain structure is maintained after extensive deformation (Fig. 126).



Fig. 125. Microstructure of an 11-µm Grainsize Specimen Deformed to 0.7 Strain at a Strain Rate of 3.0 x 10⁻¹ s⁻¹ at 850°C. Neg. No. MSD-63405.



Fig. 126. Microstructure of a 5-µm Grain-size Specimen Deformed to ~1.5 Strain at a Strain Rate of 6.0 x 10⁻⁶ s⁻¹ at 850°C, Neg. No. MSD-63403.

For the as-received Zircaloy containing 0.11 wt % oxygen at 1000°C, the structure is single-phase β . Because of the high temperature, the lowoxygen-concentration β phase is superplastic. This elongation peak was not detected by German workers³⁴ because the high-temperature testing in air resulted in an increase in the oxygen concentration of the material.

Scanning-electron-microscopy (SEM) fractographs of specimens fractured at temperatures corresponding to the maxima and minimum in the curves of total elongation versus temperature in Fig. 21 provide additional insight into the deformation behavior. The SEM fractograph (Fig. 127) of a specimen fractured at 800°C and a strain rate of $3.3 \times 10^{-2} \text{ s}^{-1}$ exhibits tensile dimples characteristic of a ductile failure. A large ductility is evident from the depth of the dimples and the sharpness of the dimple ridges.



Fig. 127. SEM Fractograph of a Zircaloy-4 Specimen Fractured at 800°C; $\dot{z} = 3.3 \times 10^{-2} \text{ s}^{-1}$. ANL Neg. No. 306-75-123 Rev. 1.

Figure 128 is a fractograph of a specimen fractured at 950°C and a strain rate of $3.3 \times 10^{-3} \text{ s}^{-1}$. This temperature corresponds to the minimum in the curve of total elongation versus temperature in Fig. 21. The photograph was taken viewing down on the fracture edge, which is the line joining the two arrows. The fractograph shows an α particle on the fracture edge that failed in a brittle manner (evidenced by the characteristic river pattern inside the black region), whereas the surrounding β phase failed in a ductile manner



Fig. 128. SEM Fractograph of a Zircaloy-4 Specimen Fractured at 950°C; ė = 3.3 x 10⁻³ s⁻¹. Fracture edge is the line joining the arrows. ANL Neg. No. 306-75-127 Rev. 1.

(manifested by the white tear ridge). Such particles were observed all along the fracture edge. From phase-diagram considerations, the structure of the material at this temperature would consist of a higher fraction of β phase with a low oxygen content and a smaller fraction of α phase with relatively high oxygen concentration. This structure would be equivalent to a distribution of hard particles (α) in a soft matrix (β). When this type of structure is strained, the hard particles would be expected to fail in a brittle manner, which initiates total fracture.

In a high-temperature tensile test, unusually large elongations can occur as a result of the strain-rate dependence of either the flow stress or the work-hardening rate. The former is attributed to superplasticity; the latter is attributed to dynamic strain aging.⁴⁶ Superplasticity is generally associated with a large value of the strain-rate-sensitivity parameter m ($m \ge 0.3$).

There are two types of superplasticity: structural and environmental. Structural superplasticity is observed in single-phase or two-phase fine-grain material (grain size $\leq 10 \ \mu$ m) deformed at temperatures above ~0.5 Tm. The presence of the second phase is not essential, but it can stabilize the grain size of the material. Environmental superplasticity results from anisotropic dimensional changes that occur when a material is thermally cycled at a transformation temperature under a small load. Evidence for structural superplasticity in Zircaloy,⁴⁷ Zircaloy-4/hydrogen alloys,⁴² and zirconiumniobium alloy,⁴⁸ and environmental superplasticity in zirconium⁴⁹ has been reported. Results in this report include a detailed investigation of micrograin superplasticity in Zircaloy.

We conducted a scoping test to check the extent of environmental superplasticity in Zircaloy. Since Zircaloy undergoes an $\alpha \rightarrow \beta$ phase transformation above ~805°C, transformation-induced superplastic deformation may be obtained by thermal-cycling Zircaloy under a small load between 800 and 1000°C. Such an experiment was conducted with a 0.11-MPa (16-psi) stress. After 50 cycles, only 14% strain was observed when compared with noncycled material. Since the high-temperature β phase always transforms to nonequiaxed α' grains when Zircaloy is cooled to lower temperatures, the nonequiaxed grains may be responsible for the low transformation plasticity in Zircaloy. The redistribution of alloying elements during phase transformation will also influence the environmental superplasticity of Zircaloy.

Data in this report clearly demonstrate that micrograin superplasticity is important in Zircaloy (a) at low strain rates ($<10^{-4} s^{-1}$) and (b) for temperatures and oxygen concentrations that result in a two-phase structure consisting of ~10% volume fraction of β phase distributed in the form of a fine-grainboundary film between α grains. The elongation peaks are typical of a superplastic material in which most of the total strain results from the necking strain that follows uniform elongation. The equiaxed grain structure observed after extensive superplastic deformation can be a result of either (1) dynamic recrystallization, in which new equiaxed grains are generated continuously during deformation, or (2) deformation, mainly by grain-boundary sliding (with some accommodation mechanism), so that grains do not elongate during deformation. As we will discuss shortly, the second possibility seems more likely.

A decrease in the strain rate increases the extent of superplasticity at ~850°C (Figs. 21 and 38), which indicates the importance of grain-boundary



Fig. 129

Micrograph of Scratched Surface of a 5- μ m Grain-size Zircaloy-4 Specimen Deformed at a Strain Rate of 3.3 x 10⁻⁴ s⁻¹ at 900°C Showing Offsets of Inscribed Scratches at Grain Boundaries. Neg. No. MSD-63402: sliding. Direct evidence for grainboundary sliding is presented in Fig. 129, a micrograph of the surface of a 5-µm specimen deformed at a strain rate of $3 \times 10^{-4} \text{ s}^{-1}$ at 900°C. Before the test, several light scratches were inscribed on the gauge-length region, perpendicular to the tensile axis. Scratch-direction offsets at the grain boundaries after deformation imply grain-boundary sliding. With a decrease in strain rate (Figs. 21 and 38), the elongation peak shifts to higher temperature, i.e., further inside the two-phase region. This observation suggests that the grain-boundary-sliding contribution to superplasticity probably occurs at the α -B interface.

A change in the deformation mechanism is apparent from the dependence of the vield stress on grain size

at the different strain rates in Fig. 47. As the strain rate decreases, a transition from normal to superplastic deformation occurs. The results in Fig. 47 imply that at large strain rates $(>10^{-4} \text{ s}^{-1})$ dislocation creep is predominant, whereas at low strain rates $(<10^{-4} \text{ s}^{-1})$ grain-boundary sliding becomes important during the deformation of Zircaloy at 850°C.

To maintain coherency of the material during grain-boundary sliding, some accommodation is necessary at the triple points of adjacent grains. This accommodation can be achieved⁵⁰ by (1) diffusional flow, (2) dislocation slip, (3) grain-boundary migration, or (4) recrystallization. Dynamic recrystallization is expected to be more difficult in a two-phase structure than in a singlephase alloy, since recrystallization involves a partitioning of alloying elements. Moreover, if dynamic recrystallization occurs simultaneously during deformation, servations are normally observed on the load-elongation curve. Since no serrations were observed on the Zircaloy load-elongation curves at 850°C and Zircaloy has a two-phase structure at this temperature, dynamic recrystallization is not expected to be the accommodation mechanism during superplastic deformation. The present evidence suggests that the three remaining accommodation processes operate to various degrees at different strain-rate and flow-stress levels during the superplastic deformation of Zircaloy. The high value of m at low strain rates is indicative of diffusional flow. The grain-boundary traces at the triple points in the mic rostructure of Fig. 126 provide evidence for grain-boundary migration. The lew value of :m and elongated grains at high-strain rates are indicative of dislocation creep. The small extent of grain-size refinement after deformation at intermediate strain rates also suggests grain-boundary migration.

Significant differences observed in the microstructures of specimens deformed at high and low strain rates at 850°C are related to the deformation mechanisms. At fast strain rates (10^{-1} s^{-1}) , in which transgranular deformation due to dislocation motion was important and the ductility was low, an elongated grain structure with minimal grain growth was typically observed, e.g., Fig. 125. At slow strain rates (10^{-6} s^{-1}) , where intergranular deformation due to grain-boundary sliding was dominant and the ductility was higher, the deformed region of the specimen exhibited equiaxed grains, in which considerable grain growth is evident (Fig. 126).

The influence of strain rate and grain size on the work-hardening characteristics of Zircaloy at 850°C is presented in Tables XXVII-XXIX and Fig. 130. The figure shows that superplasticity is associated with a minimum value of the work-hardening exponent that occurs at intermediate strain rates $(10^{-4}-10^{-3} \text{ s}^{-1})$. At higher strain rates $(>10^{-2} \text{ s}^{-1})$, the work-hardening exponent is higher due to intergranular deformation, and superplasticity is less significant. At lower strain rates $(<10^{-4} \text{ s}^{-1})$, although grain-boundary sliding is significant, extensive grain growth due to substantial test times gives rise to hardening, and superplasticity is less significant. An optimum combination of grain-boundary sliding and a stable equiaxed grain structure is obtained at intermediate strain rates, where superplasticity is predominant.

Strain Rate, ¢, s ⁻¹	Stage Number	Strain Interval	n	k, Pa	Average, co, Pa
3.3 × 10-2	2	0.0004-0.005	-0.57	-9.4×10^{4}	4.5 x 10 ⁷
3.3 x 10 ⁻³	2	0.0004-0.0024	-0.96	-2.0 × 103	2.6 x 107
3.3×10^{-4}	1	0.0004-0.002	0.55	1 1 x 10 ⁶	7.5 x 10 ⁴
	2	0.0024-0.013	-1.10	-1.5 x 10 ¹	1.3 x 107
3.3 x 10 ⁻⁵	ĩ	0.0004-0.006	C.90	1.1×10^{6}	3.3×10^{6}
	2	0.007-0.09	-0.26	-6.1 x 10 ⁵	6.7 x 10 ⁶
3.3 x 10 ⁻⁶	L	0.0004-0.0032	0.62	1.2 × 10"	1.0 x 10 ⁴
	2	0.0036-0.13	0.32	2.0 x 10 ²	1.4 x 10 ⁷
3,3 x 10-6	1	0.0004-0.014	0.05	6.9 x 10 ⁴	-2.4×10^{6}
	2	0.015-0.37	0.99	7.8 x 10 ⁶	3.1 x 10 ⁶

TABLE XXVII. Least-squares Work-hardening Constants for 5-um Grain-size Zircaloy-4 Specimens Deformed at 850°C and Various Strain Rates

Strain Rate, ¢, s ⁻¹	Stage Number	Strain Interval	n	k, Pa	Average ₀₀ , Pa
3.3 x 10 ⁻²	2	0.000 1-0.009	0.85	4.7 x 10 ⁸	2.9 x 10 ²
	3	0.01-0.035	-0.83	1.3×10^{5}	4.4×10^{7}
3.3 x 10 ⁻³	1	0.0004-0.0012	0.70	3.5 x 10 ⁸	1.9×10^{7}
	2	0.0016-0.03	-0,27	-1.2×10^{6}	2.9 x 107
3.3×10^{-4}	1	0.0004-0.0016	0.55	1.0×10^{8}	8.3×10^{6}
	2	0.002-0.025	-0,52	-1.1 x 10 ⁵	1.5×10^{7}
2.6 x 10 ⁻⁵	Z	0.0004-0.015	-0.23	-5.3 x 10 ⁵	8.4 x 10 ⁶
3.3 x 10 ⁻⁶	2	0.0004-0.39	0.53	5.8 x 10 ⁶	1.1 x 10 ⁶

 TABLE XXVIII. Least-squares Work-hardening Constants for 11-µm Grain-size

 Zircaloy-4 Specimens Deformed at 850°C and Various Strain Rates

TABLE XXIX. Least-squares Work-hardening Constants for 50-µm Grain-size Zircaloy-4 Deformed at 850°C and Various Strain Rates

Strain Rate, è, g ⁻¹	Stage Number	Strain Interval	n	k, Pa	Average, σ_0 , Pa
3,3 x 10 ⁻²	I	0.0004-0.0028	-0.88	-5.3×10^3	3.8 x 10 ⁷
	2	0.0032-0.035	1.10	1.2×10^{9}	3.6 x 10 ⁷
3.3×10^{-3}	1	0.0004-0.304	0.89	3.0×10^{8}	2.0×10^7
	Z	0.004-0.025	-0.48	-3.0 x 10 ⁵	2.7×10^7







The large effect of hold time on the superplastic ductility demonstrates the importance of phase distribution in Zircaloy. A fine distribution of \$ phase both at the α -grain boundaries and within the α grains is conducive to superplastic elongations. A continuous thin film of soft β phase at the boundaries of the relatively hard α grains appears to be necessary for the maximum superplasticity in Zircaloy. This observation supports the core-mantle model of superplastic deformation proposed by Gifkins.⁵¹ Longer hold times before deformation result in coalescence of the B phase at the α -grain boundaries, and the thickness of the 8 film increases with time. Large 8-film thicknesses presumably decrease the efficiency of the "mantle" to assist grain-

boundary sliding at the α - β interface. At higher volume fractions of the β phase, β grains begin to appear in the microstructure, the continuity of the β film at the grain boundary is broken, and the α - β interface area decreases. These microstructural observations indicate that grain-boundary sliding at the α - β interface is important in the superplastic deformation of Zircaloy near 850°C. The α -phase morphology provides additional evidence for the importance of grain-boundary sliding on superlasticity of Zircaloy (Fig. 58). The total elongation of the acicular α -grain structure with a large aspect ratio (i.e., transformed β -phase) is considerably smaller than for the equiaxed α grains. The overall accommodation of grain-boundary sliding is easier in the equiaxed structure than in a structure that contains elongated grains.

The ductility of Zircaloy increases as the grain size of the material decreases, primarily because more grain-boundary area is available for grain-boundary sliding in the small-grain-size material. As the grain size decreases from 1 i to 5 μ m, the superplastic elongation peak shifts from 850 to 900°C. In other words, maximum superplasticity is observed in a finer-grain-size material at a higher volume fraction of β phase. The higher volume fraction of β phase is required to achieve a continuous β film at the α - α grain boundaries of the fine-grain material, which has a larger initial α - α grain-boundary area.

Regarding possible mechanisms of superplasticity in Zircaloy, Nabarro⁵²-Herring⁵³ bulk-diffusion vacancy creep and Coble⁵⁴ grain-boundary vacancy creep are not likely to be directly rate-controlling in Zircaloy, since the strain-rate sensitivity is never close to unity and the grain shape after deformation remains equiaxed.

An experimental determination of the stress and grain-size exponents, n and p in Eq. 4, is frequently used to evaluate the relative contributions of different mechanisms of superplastic deformation. Figure 131 is a logarithmic plot of the strain rate versus grain size for stage-II deformation of Zircaloy at 850°C. The value of p, in Eq. 4, determined from the slope of the curve is -1.



Fig. 131

Logarithmic Plot of Strain Rate as a Function of Grain Size for Stage-II Deformation of Zircaloy-4 at 850°C. ANL Neg. No. 306-76-248 Rev. 1. To evaluate the deformation process in the two-phase material at 850 and 900°C, values of the total strain rate ϵ_T were calculated from the contributions from dislocation creep, in Eq. 4, and from the grain-boundary sliding mechanism with diffusional accommodation,³⁹ in Eq. 7,

$$\boldsymbol{\varepsilon}_{\mathbf{D}-\mathbf{A}} = \frac{100\Omega D_{\mathbf{V}}}{D^2 kT} \left(\boldsymbol{\sigma} - \frac{0.72\Gamma}{D} \right) \left(1 + \frac{3.3 W D_{\mathbf{B}}}{D D_{\mathbf{V}}} \right); \tag{7}$$

that is,

$$\dot{\boldsymbol{\varepsilon}}_{\mathrm{T}} = \dot{\boldsymbol{\varepsilon}} + \dot{\boldsymbol{\varepsilon}}_{\mathrm{D-A}}. \tag{8}$$

The definition of the symbols in Eq. 7 and their values are as follows: atomic volume of zirconium $\Omega = 2.37 \times 10^{-29} \text{ m}^3$;⁵⁴ volume diffusion coefficient for zirconium $D_V = 5.9 \times 10^{-2} \exp(-Q_V/RT) \text{ m}^2/\text{s}$, where $Q_V = 218 \text{ kJ/mole}$;^{43*} grain-boundary diffusion coefficient of α -zirconium $D_B = 5.9 \times 10^{-2} \exp(-Q_B/RT) \text{ m}^2/\text{s}$, where $Q_B = 0.60Q_V$; grain size $D = (5 \text{ or } 11) \times 10^{-6} \text{ m}$; grain-boundary free energy Γ is assumed to be zero; and the grain-boundary layer thickness $W = 1 \times 10^{-9} \text{ m}$.

With regard to the additional parameters in Eq. 4, the following values were used for both grain sizes: n = 4.35, p = -1, Q = 316 kJ/mole, and $G = (1.94 \text{ and } 1.82) \times 10^{10}$ Pa at 350 and 900°C, respectively. The constant A was adjusted so that the computed and measured yield stresses were the same at strain rates $\gtrsim 10^{-3} \text{ s}^{-1}$. (The strain-rate contribution \dot{e}_{D-A} from Eq. 7 was negligible in this range.) The experimental data and calculated curves for the dependence of yield stress on the strain rate at 850 and 900°C are shown in Fig. 132 for the 5- and 11-µm grain-size materials. The curves, based upon Eqs. 4, 7, and 8 and the values for the parameters listed above, are in good agreement with the experimental data at strain rates $\leq 10^{-3} \text{ s}^{-1}$ at which the transition to stage-II behavior occurs.





Comparison of Calculated and Experimental Strain-rate Dependence of 0.2% Yield Stress for 5- and 11-µm Grain-size Zircaloy-4 at 850 and 900°C. ANL Neg. No. 306-76-253 Rev. 1.

[•]The diffusion coefficient of zirconium in α -zirconium and zirconium-tin alloys is sensitive to the dislocation density in the material. The lower activation energy for zirconium diffusion. in Ref. 43, is more appropriate for a material undergoing plastic deformation.

Although the Ashby-Verrall model³⁹ was originally proposed for single-phase material, the information in Fig. 132 and microstructural observations indicate that it is also applicable for two-phase materials. Figure 12: shows that the grains do not elongate during deformation and new grains (free of scratches) appear on the surface following deformation at $\dot{\epsilon} = 3.3 \times 10^{-4} \text{ s}^{-1}$ at 900°C (stage II). In addition, a tendency for grain rotation and grain-boundary sliding is observed. Figure 125 shows a microstructure that is characteristic of stage-III behavior, i.e., elongated grains that result from dislocation activity, for Zircaloy-4 deformed at a strain rate of $3 \times 10^{-1} \text{ s}^{-1}$ at 850°C. These observations, in addition to the large value of the strain-rate sensitivity for the two-phase material (m ~ 0.8 at low-strain rates) and the dependence of the strain rate on grain size (-3 consistent with the Ashby-Verrall model. The model also predicts the shift in the stress-versus-strain-rate curves, in Fig. 132, to higher strain rates as the temperature increases.



Fig. 133

Slip Lines on Surface of a Zircaloy-4 Specimen Deformed at 850°C; $\dot{\varepsilon} = 3.3 \times 10^{-1} \text{ s}^{-1}$. Neg. No. MSD-63846.

To supplement the work concerning the mechanism of superplastic deformation in Zircaloy at 850°C, surfaces of deformed tensile specimens were examined by scanning-electron microscopy. Figure 133 represents the surface of the thinnest side of a Zircaloy-4 specimen deformed at 850°C at the strain rate of $3.3 \times 10^{-1} \text{ s}^{-1}$. The observed broad slip lines within the grain imply that transgranular deformation due to dislocation slip was important at this fast strain rate, which is within the region of stage-III deformation. Note that the offset at the grain boundary separating two grains is not pronounced; this indicates that grain-boundary sliding is not important in stage-III deformation.

Similar surface microstructure for a specimen deformed at a lower strain rate of $3.3 \times 10^{-6} \text{ s}^{-1}$, in the

stage-II region, is shown in Fig. 134. In this case, the step formation at the grain boundary is significant and shows the importance of grain-boundary sliding. The fine crystalline faceting within the grains is due to thermal etching of the surface during the long-test times at high temperature in a vacuum environment. Faceting was also observed on samples held in a similar vacuum at 850°C without deformation. Note the multiplicity of the crystal-line planes along which thermal etching occurs. Compared to the slip lines in Fig. 133, the steps that formed during thermal etching, in Fig. 134, are

sharper and finer. The absence of slip lines in Fig. 134 implies that, at this low strain rate, transgranular deformation due to dislocation slip is unimportant. The observed grain-boundary sliding is accommodated by diffusion creep. These observations are consistent with conclusions that, at 850°C in stage II, grain-boundary sliding accommodated by diffusion creep is important, and that, in stage III, transgranular deformation due to dislocation motion predominates.





Facets due to Thermal Etching of Surface of a Zircaloy-4 Specimen Deformed at 850°C; $\dot{\epsilon}$ 3.3 x 10^{-6} s⁻¹. Neg. No. MSD-63845. The uniaxial tensile behavior of Zircaloy-2 and -4 and Zircaloy-oxygen alloys was investigated over a wide range of temperatures, strain rates, oxygen concentrations, and microstructural states. The following conclusions can be drawn from the experimental results:

1. Superplastic strain maxima were observed in Zircaloy at 850, 1000, and ~1300°C. Maximum superplastic elongations occurred at 850°C in the two-phase region, in which the microstructure consists of a thin film of soft β phase at the boundaries of the α grains. The ductility values in excess of several hundred percent are consistent with large values of the strain-ratesensitivity parameter. The dominant mechanism of superplasticity in Zircaloy at 850°C is grain-boundary sliding with accommodation by diffusional flow, dislocation slip, and grain-boundary migration.

2. Dynamic strain-aging manifestations such as yield points and high work-hardening rates were observed in Zircaloy near 200 and 700°C.

3. The mechanical properties of Zircaloy are a function of texture below 600°C. However, at higher temperatures, the properties are independent of texture.

4. The microstructure (and hence the mechanical properties) of transformed β -phase Zircaloy depends on the cooling rate through the $\alpha + \beta$ region.

5. In a homogeneous Zircaloy-oxygen alloy, an increase in the oxygen concentration increases the ultimate tensile strength and decreases the total strain.

6. Oxygen increases the ultimate tensile strength of Zircaloy-oxide composite specimens. The effect is more pronounced at temperatures above 1000°C because the α -phase surface layer does not crack.

7. Below 600°C, where transgranular deformation of α phase is important, the work-hardening exponent shows positive values. In the temperature region of ~700-1100°C, where intergranular deformation is significant, the work-hardening exponent assumes negative values. Above 1100°C, the work-hardening exponent is positive.

ACKNOWLEDGMENTS

This work was sponsored by the U. S. Nuclear Regulatory Commission, Office of Nuclear Regulatory Research. We are grateful to M. L. Picklesimer for helpful discussions. Experimental assistance of W. K. Soppet, L. J. Marek, and L. P. Burkel is acknowledged. F. J. Karasek fabricated the fine-grainsize Zircaloy-4 sheet material.

REFERENCES

- Light-water-reactor Safety Research Program: Quarterly Progress Report, January-March 1975, Sec. III, "Mechanical Properties of Zircaloy Containing Oxygen," ANL-75-28 (June 13, 1975).
- Light-water-reactor Safety Research Program: Quarterly Progress Report, April-June 1975, Sec. III, "Mechanical Properties of Zircaloy Containing Oxygen," ANL-75-58 (Sept 26, 1975).
- 3. Light-water-reactor Safety Research Program: Quarterly Progress Report, July-September 1975, Sec. III, "Mechanical Properties of Zircaloy Containing Oxygen," ANL-75-72.
- 4. Light-water-reactor Safety Research Program: Quarterly Progress Report, October-December 1975, Sec. III, "Mechanical Properties of Zircaloy Containing Oxygen," ANL-76-15.
- 5. Light-water-reactor Safety Research Program: Quarterly Progress Report, January-March 1976, Sec. III, "Mechanical Properties of Zircaloy Containing Oxygen," ANL-76-49.
- 6. Light-water-reactor Safety Research Program: Quarterly Progress Report, April-June 1976, Sec. III, "Mechanical Properties of Zircaloy Containing Oxygen," ANL-76-87.
- Light-water-reactor Safety Research Program: Quarterly Progress Report, July-September 1976, Sec. III, "Mechanical Properties of Zircaloy Containing Oxygen," ANL-76-121.
- 8. Light-water-reactor Safety Research Program: Quarterly Progress Report, October-December 1976, Sec. III, "Mechanical Properties of Zircaloy Containing Oxygen," ANL-77-10.
- 9. M. L. Picklesimer, Fuel Behavior Research Branch, Nuclear Regulatory Commission, Washington, D.C., private communication.
- 10. E. M. Savitskii, M. A. Tylkina, and I. A. Cyganova, Zustandsdiagramm des Systems Zirkonium-Rhenium, Kernenergie 3, 641 (1960).
- 11. M. L. Picklesimer, Anodizing for Controlled Microstructural Contrast by Color, The Microscope 15, 472 (1967).
- 12. M. L. Picklesimer, Anodizing as a Metallographic Technique for Zirconium Base Alloys, ORNL-2296 (May 1957).
- 13. P. D. Kaufmann, P. Danielson, and E. F. Baroch, *Zirconium in Nuclear* Applications, ASTM-STP 551, pp. 52-62, American Society for Testing and Materials, Philadelphia, Pa. (1974).
- C. Crussard and B. Jaoul, Contribution à l'étude de la forme des courbes de traction des métaux et à son interpretation physique, Rev. Metall. 47, 589 (1950).
- C. Crussard, Rapport entre la forme exacte des courbes de traction des métaux et les modifications concomitantes de leur structure, Rev. Metall. 50, 697 (1953).
- 16. B. Jaoul, Etude de la Forme des Courbes de Deformation Plastique, J. Mech. Phys. Solids 5, 95 (1957).

- 17. P. Ludwik, Elemente der Technologischen Mechanik, Julius Springer, Berlin, p. 32 (1909).
- A. M. Garde and R. E. Reed-Hill, Dual Analysis of Longitudinal and Transverse Zirconium Tensile Stress-Strain Data, ASTM-STP 551, 75-91 (Aug 1974).
- 19. R. H. Meservey and R. Herzel, Brittle Behavior of Zircaloy in an Emergency Core Cooling Environment, IN-1389 (Sept 1970).
- W. A. Backofen, D. L. Holt, D. Lee, P. Martin, and F. J. Azzarto, Deformation Processes in Anisotropic Meto¹s, MIT Final Report to Naval Air Systems Command under Contract NOW-66 V068-d (July 1966).
- A. M. Garde, E. Aigeltinger, B. N. Woodruf, and R. E. Reed-Hill, Concerning the Strength of Dynamic Strain 19 and Virconium, Metall. Trans. 6A, 1183-1188 (June 1975).
- 22. K. Veevers and W. B. Rotsey, Effect of Irradiation on Strain Aging in Annealed Zircaloy-2, J. Nucl. Mater. 27, 108 (1968).
- K. Veevers, W. B. Rotsey, and K. U. Snowden, The Effect of Neutron Irradiation and Cold Work on the Strain Aging Behavior of Zircaloy-2, ASTM-STP 458, American Society for Testing and Materials, p. 194 (1970).
- 24. K. Veevers, Strain Aging in Zirconium Alloys, J. Nucl. Mater. 55, 109 (1975).
- P. M. Kelly and P. D. Smith, Strain Aging in Zirconium-Oxygen Alloys, J. Nucl. Mater. 46, 23 (1973).
- D. O. Hobson, M. F. Osborne, and G. W. Parker, Comparison of Rupture Data from Irradiated Fuel Rods and Unirradiated Cladding, Nucl. Tech. 11, 479 (1971).
- 27. D. O. Hobson and P. L. Rittenhouse, Deformation and Rupture Behavior of Light-water Reactor Fuel Cladding, ORNL-4727 (Oct 1971).
- D. G. Hardy, High Temperature Expansion and Rupture Behavior of Zirculoy Tubing, Topical Meeting on Water-reactor Safety, Salt Lake City, Utah, March 26-28, 1973, G. A. Freund, compiler, pp. 254-273.
- 29. C. C. Busby and K. B. Marsh, High Temperature Deformation and Burst Characteristics of Recrystallized Zircaloy-4 Tubing, WAPD-TM-900 (Jan 1970).
- H. M. Chung, A. M. Garde, and T. F. Kassner, Deformation and Rupture Behavior of Zircaloy Cladding under Simulated Loss-of-coolant Accident Conditions, ASTM-STP 633, American Society for Testing and Materials, pp. 82-97 (1977).
- 31. E. Tenokhoff, Operable Deformation Systems and Mechanical Behavior of Textured Zircaloy Tubing, Zirconium in Nuclear Applications, ASTM-STP 551, American Society for Testing and Materials, p. 179 (1974).
- 32. G. Östberg, Metallographic Study of Isothermal Transformation of Beta Phase in Zircaloy-2, Jernkontorets Ann. 145, 119 (1963).
- 33. G. Östberg, Determination of the Composition of the Second Phase in Zircaloy, J. Nucl. Mater. 7, 103 (1962).
- 34. M. Hansen and K. Anderko, Constitution of Binary Alloys, McGraw-Hill, New York, pp. 1219, 741, and 573 (1958).
- 35. M. J. Luton and J. J. Jonas, Solute Strengthening at High Temperature in Zirconium-tin Alloys, Can. Metall. C. 11, 79 (1972).
- 36. R. H. Chapman, Compiler, Characterization of Zircaloy-4 Tubing Procured for Fuel Cladding Research Programs, ORNL/NUREG/TM-29 (July 1976).
- 37. M. Bocek, Kernforschungszentrum Karlsruhe, Institute für Material und Festkörperforschung, West Germany, private communication (May 1975).
- 38. R. C. Gifkins, Scr. Metall. 10, 433 (1976).
- 39. M. F. Ashby and R. A. Verrall, Diffusional-accommodated Flow and Superplasticity, Acta Metall. 21, 149 (1973).
- 40. R. F. S. Hearmon, The Elastic Constants of Anisotropic Materials, Adv. Phys. 5, 323 (1956).
- 41. E. S. Fisher and C. J. Renken, Single Crystal Elastic Moduli and the hcp+bcc Transformation in Ti, Zr, and Hf, Phys. Rev. 135, A482 (1964).
- 42. J. J. Kearns, J. E. McCauley, and F. A. Nichols, Effect of Alpha/Beta Phase Constitution on Superplasticity and Strength of Zircaloy-4, J. Nucl. Mater. 61, 169 (1976).
- V. S. Lyashenko, V. N. Bykov, and L. V. Pavlinov, A Study of Self-Diffusion in Zivecnium and Its Alloys with Tin, Fiz. Metall. Metalloved. 8, 362 (1959), English Translation 8, 40 (1960).
- G. M. Hood and R. J. Schultz, Tracer Diffusion in α-2r, Acta Metall.
 22, 459 (1974).
- 45. M. C. Naik and R. P. Agarwala, Self- and Impurity Diffusion in Alpha-Zirconium, Acta Metall. 15, 1521 (1971).
- 46. A. T. Santhanam and R. E. Reed-Hill, Metall. Trans. 2, 2619 (1971).
- 47. D. Lee and W. A. Backofen, Trans. AIME 239, 1034 (1967).
- 48. K. Nuttall, Scr. Metall. 10, 835 (1976).
- 49. G. W. Greenwood and R. H. Johnson, Proc. Roy. Soc. 283A, 403 (1965).
- 50. D. L. Holt and W. A. Backofen, Trans. ASM 59, 755 (1966).
- 51. R. C. Gifkins, Grain-boundary Sliding and Its Accommodation during Creep and Superplasticity, Metall. Trans. 7A, 1225 (1976).
- F. R. N. Nabarro, "Deformation of Crystals by the Motion of Single Ions," Proc. Conf. on Strength of Solids, Phys. Soc. of London, Cambridge, p. 75 (1948).
- 53. C. Herring, Diffusional Viscosity of a Polyorystalline Solid, J. Appl. Phys. 21, 437 (1950).
- 54. R. L. Coble, A Model for Boundary Diffusion Controlled Creep in Polycrystalline Materials, J. Appl. Phys. 34, 1679 (1963).
- 55. L. T. Lloyd, Thermal Expansion of Alpha-Zirconium Single Crystals, ANL-6591 (Jan 1963).