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Title: PHASE SYNCHRONIZATION OF MULTIPLE KLYSTRONS IN RF SYSTEMS

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**Title: Phase synchronization of multiple klystrons in RF system**

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**Remark:** The second sections of the paper entitled "Adaptive feedforward ...", the paper entitled "Phase synchronization ...", and the paper entitled "Feedback linearization ..." are describing the klystron model and are almost same. Also, the third sections of the paper entitled "Adaptive feedforward ..." and the paper entitled "Feedback linearization ..." are describing the RF cavity model and almost same. The rest sections of each paper describe the different control techniques and they are derived from the klystron model and the RF cavity model described in the second section and the third section. When at least two papers are accepted for full papers, the second section and the third section will be modified.

# Phase Synchronization of Multiple Klystrons in RF System

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**Abstract**-The Low Energy Demonstration Accelerator(LED A) being constructed at Los Alamos National Laboratory will serve as the prototype for the low energy section of the Acceleration Production of Tritium(AP T) accelerator. The first LED A RF system includes three, 1.2 MW, 350 MHz, continuous wave, klystrons driving a radio frequency quadrupole(RFQ). A phase control loop is necessary for each individual klystron in order to guarantee the phase matching of these klystrons. To meet this objective, we propose adaptive PI controllers which are based on simple adaptive control. These controllers guarantee not only phase matching but also amplitude matching.

## 1 Introduction

The Low Energy Demonstration Accelerator(LED A) being constructed at Los Alamos National Laboratory will serve as the prototype for the low energy section of Acceleration Production of Tritium(AP T) accelerator. The AP T accelerator requires over 244 RF klystrons each with a continuous wave output power of 1 MW. The reliability and availability of these RF systems is a critical to the successful operation of AP T plant and prototypes of these systems are being developed and

demonstrated on LEDA. The first LEDA RF system includes three, 1.2 MW, 350 MHz, continuous wave, klystrons driving a radio frequency quadrupole(RFQ)[11],[12]. The LEDA 350 MHz klystrons are designed to dissipate the full beam power(1.85MW) in the klystron collector in steady state.

In this paper, we address the problem of the control of a multi-klystron RF system. We present the control of amplitudes and phases of multiple klystrons. Previous investigation[16] shows that closed loop amplitude control around each individual klystron is unnecessary and the gain mismatches in klystrons do not affect to the field amplitude control significantly. However, the gain mismatches have influence on the field phase control. A suggestion to counteract this problem is to synchronize multiple klystrons' phases. This paper proposes a method to deal with the synchronization of the phases of multiple klystrons. In order to achieve the objective, we make use of the simple adaptive control(SAC) concept[8].

SAC was first proposed in the work of Sobel, Kaufman, and Mabijs[17] for linear plants. This approach uses a control structure which is a linear combination of feedforward of the model states, and inputs and feedback of the error between plant and model outputs. Asymptotic stability is ensured provided that the plant is almost strictly positive real(ASPR). That is, for a plant represented by the triple  $(A, B, C)$  which is 1)minimum phase, 2)relative degree one, and 3)positive high frequency gain, there exists a feedback gain  $K_e$  such that the closed loop system  $(A - BK_eC, B, C)$  is strictly positive real(STR). The ease of its implementation, non-necessity of identification, and its robustness properties make SAC attractive to the practitioner. SAC was extended to nonlinear plants[2], nonlinear servomechanisms with time varying uncertainties[4], non ASPR plants[3],[6],[7]. Its relation to high gain adaptive control([14] and reference therein) was described in the works of Bar-Kana[1],[2] where high gain adaptive control was described as a special class of SAC.

A klystron is modeled as a two input two output(TITO) nonlinear system where the state equation is linear but the output equation is nonlinear with respect to phase. In order to overcome the nonlinearity of phase in output, a coordinate transformation of the klystron model is addressed. In the new coordinate, the state equation is nonlinear with respect to state and linear with respect to input, and the output equation is linear with respect to phase. Based on the model in the transformed coordinate, we design an adaptive PI controller by applying SAC. In order to set up SAC for multiple klystrons, one of klystrons is taken as the model klystron and the rest of them are taken as plant klystrons. The selected model klystron's input is taken as model input, its output as model output, and its state as model state. For each plant klystron, SAC is designed so that plant klystron's output follows the output of the model klystrons.

## 2 The Klystron Model

We consider a klystron model.

Figure 1 shows a klystron model in Matlab/Simulink. It has 1) **FILTER AND AMPLIFIER**, 2) **AMPLITUDE DEPENDENCE** and **PHASE DEPENDENCE** of ripple, 3) lookup tables, **AMPLITUDE SATURATION** and **PHASE SATURATION**. Also, it has two inputs, LLRF\_I and LLRF\_Q and two outputs HPRF\_I and HPRF\_Q.

Let  $u_1 = \text{LLRF\_I}$  and let  $u_2 = \text{LLRF\_Q}$ . Let  $x_1$  and  $x_2$  be outputs of the system called **FILTER AND AMPLIFIER** whose transfer function are given by

$$\frac{X_1(s)}{U_1(s)} = \frac{1}{3.54e^{-7}s + 1} \quad (1)$$

$$\frac{X_2(s)}{U_2(s)} = \frac{1}{3.54e^{-7}s + 1} \quad (2)$$

In state space, transfer functions (1) and (2) are represented as

$$\dot{x}_1 = -a_1 x_1 + a_1 u_1 \quad (3)$$

$$\dot{x}_2 = -a_1 x_2 + a_1 u_2 \quad (4)$$

where  $a_1 = \frac{1e^{+007}}{3.54}$ .

A klystron is affected by a high voltage power supply ripple. Influence of the ripple on a klystron is represented by **AMPLITUDE DEPENDENCE** and **PHASE DEPENDENCE**. **AMPLITUDE DEPENDENCE** has the form of

$$(c_{rppl}R(t) + 1)^{d_{rppl}}$$

and **PHASE DEPENDENCE** has the form of

$$\phi_{rppl} \frac{\pi}{180} R(t)$$

where  $c_{rppl}$ ,  $d_{rppl}$ , and  $\phi_{rppl}$  are characteristic parameters of a high voltage power supply and a klystron. In this paper,  $c_{rppl} = 0.01$ ,  $d_{rppl} = 1.25$ , and  $\phi_{rppl} = 3.00$ .

A klystron has two look-up tables, called **AMPLITUDE SATURATION** and **PHASE SATURATION**. The input of the two look-up tables is given by

$$A = \frac{K_g}{10\sqrt{KP_m}} (0.01R(t) + 1)^{1.25} \cdot \sqrt{x_1^2 + x_2^2} \quad (5)$$

where  $R(t)$  is the ripple,  $K_g$  is the klystron gain, and  $KP_m$  is the maximum klystron power.  $R(t)$ ,  $K_g$ , and  $KP_m$  are specified for a given klystron. For given A, the output of the look-up table **AMPLITUDE SATURATION** can be represented by

$$A_N = I_1(A) \quad (6)$$

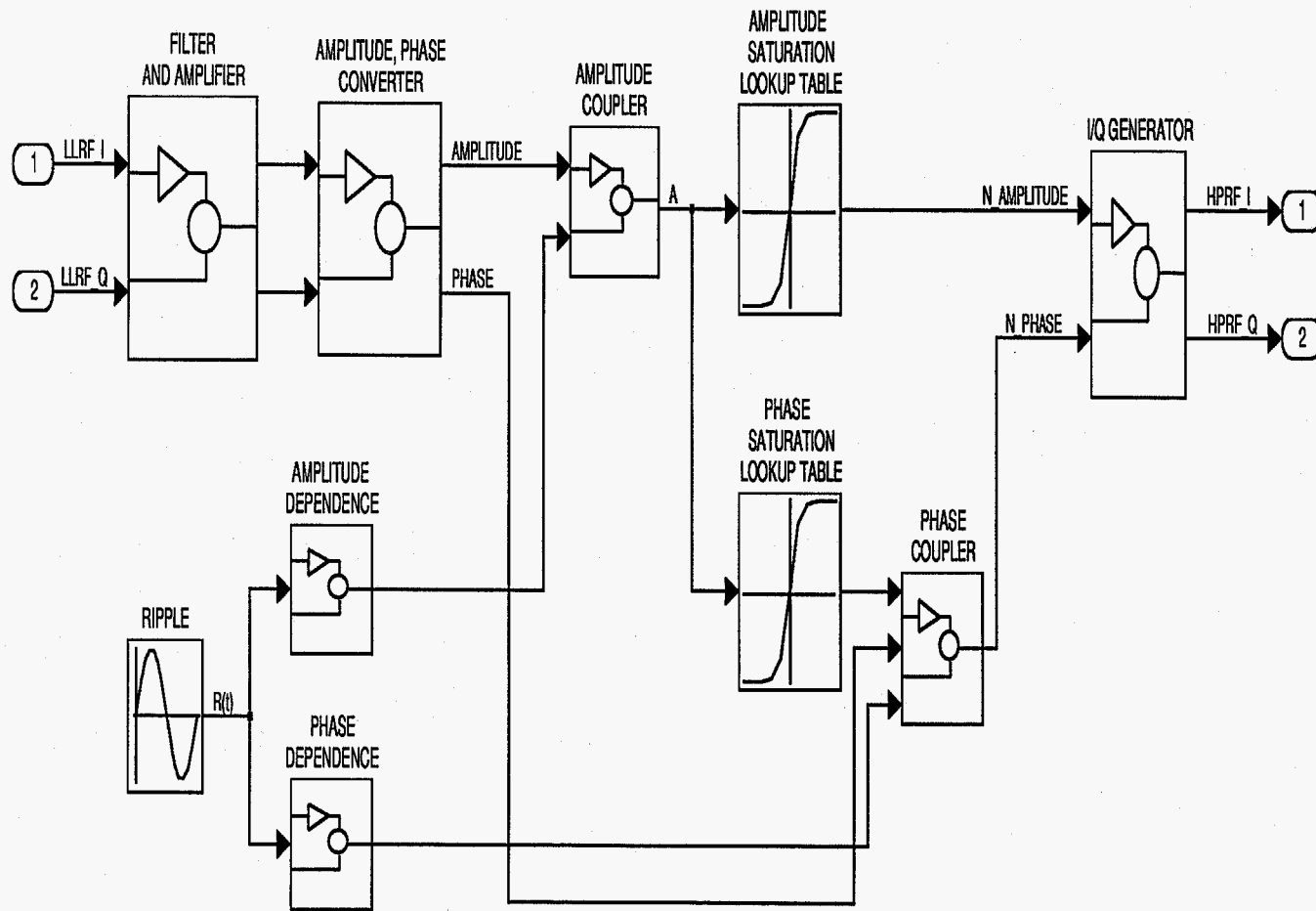


Figure 1: A Klystron Model



and the output of the look-up table **PHASE SATURATION** can be represented by

$$\theta_N = I_2(A) \quad (7)$$

Table 1 and table 2 show data of look-up table **AMPLITUDE SATURATION** and data of look-up table **PHASE SATURATION**, respectively. These data are measured data.

$A$	$A_N$	$A$	$A_N$	$A$	$A_N$	$A$	$A_N$
-0.1000	0.0000	0.0700	0.0000	0.1400	0.1900	0.5700	0.7500
0.7100	0.8700	0.8600	0.9800	0.9000	1.0000	0.9100	1.0000

Table 1. AMPLITUDE SATURATION Measured Data

$A$	$\theta_N$	$A$	$\theta_N$	$A$	$\theta_N$	$A$	$\theta_N$
-0.1000	0.0000	0.0700	0.0000	0.6400	-0.0150	0.7100	-0.0350
0.6400	-0.0150	0.7100	-0.0350	1.0000	-0.4770		

Table 2. PHASE SATURATION Measured Data

In figure 1, the normalized amplitude **N\_AMPLITUDE**, defined by  $y_1$ , and the normalized phase **N\_PHASE**, defined by  $y_2$  are expressed as

$$y_1 = A_N = I_1(A) \quad (8)$$

$$\begin{aligned} y_2 &= \theta_N + \tan^{-1}\left(\frac{x_2}{x_1}\right) + 3 \cdot \frac{\pi}{180} \cdot R(t) \\ &= I_2(A) + \tan^{-1}\left(\frac{x_2}{x_1}\right) + 3 \cdot \frac{\pi}{180} \cdot R(t). \end{aligned} \quad (9)$$

In addition, for given  $y_1$  and  $y_2$ , **HPRF\_I** and **HPRF\_Q** are given by

$$HPRF_I = 10\sqrt{KP_m} \cdot y_1 \cdot \cos(y_2) \quad (10)$$

$$HPRF_Q = 10\sqrt{KP_m} \cdot y_1 \cdot \sin(y_2). \quad (11)$$

Since the look-up tables have the limited number of data, we need to approximate the look-up tables by linear or nonlinear curve fitting equations. Considering the characteristic curve of a klystron, we choose nonlinear equations. We choose curve fitting equations of **AMPLITUDE SATURATION** and **PHASE SATURATION** having the forms

$$A_N = \sum_{i=1}^N c_i e^{-f_i A} \quad (12)$$

$$\theta_N = \sum_{i=1}^N d_i e^{-f_i A} \quad (13)$$

where  $f_i, i = 1, 2, \dots, N$  and parameters  $c_i, i = 1, 2, \dots, N, d_i, i = 1, 2, \dots, N$  are to be determined.

A higher order curve fitting equation may yield more accuracy. For simplicity, we choose  $N = 7$ . Also, in order to reduce the number of coefficients to be determined,  $f_i, i = 1, 2, \dots, N$  are given in Table 3.

$f_1$	$f_2$	$f_3$	$f_4$	$f_5$	$f_6$	$f_7$
0.50	0.75	1.00	1.25	1.50	1.75	2.00

Table 3. Exponents of curve fitting equations

In order to increase the accuracy of curve fitting, we need more data. These data are generated by applying a different set of input pairs (LLRF\_I, LLRF\_Q). Look-up tables generate output data by linear interpolation. Data as given in Table 4 and Table 5 are used for fitting.

By using data given in Table 4 and Table 5, we obtain coefficients  $c_i, i = 1, 2, \dots, N$  and  $d_i, i = 1, 2, \dots, N$ , of the curve fitting equations (12) and (13). Coefficients  $c_i$  and  $d_i$  obtained are given in Table 6. Figure 2 shows plots of data points as given in Table 4, Table 5 and plots of curve fitting equations (12) and (13) whose coefficients,  $f_i, i = 1, 2, \dots, N, c_i, i = 1, 2, \dots, N, d_i, i = 1, 2, \dots, N$ , are given in Table 3 and Table 6 with appropriate domain of  $A$ .

A	$A_N$	A	$A_N$	A	$A_N$	A	$A_N$
-0.1000	0.0000	0.0700	0.0000	0.1400	0.1900	0.5700	0.7500
0.7100	0.8700	0.8600	0.9800	0.9000	1.0000	0.9100	1.0000
0.3122	0.4143	0.3568	0.4724	0.4014	0.5305	0.4461	0.5886
0.4907	0.6467	0.5353	0.7048	0.5799	0.7585	0.5910	0.7680
1.0000	0.9900	0.4461	0.5886	0.6468	0.8158	0.6691	0.8349
0.0446	0.0000	0.0892	0.0521	0.6914	0.8540	0.7360	0.8891
0.1338	0.1732	0.1784	0.2400	0.7806	0.9218	0.8252	0.9545
0.2230	0.2981	0.2676	0.3562	0.8921	0.9961	0.9367	0.9970
0.3122	0.4143	0.3568	0.4724	0.9813	0.9921		

Table 4. AMPLITUDE SATURATION Data

A	$\theta_N$	A	$\theta_N$	A	$\theta_N$	A	$\theta_N$
-0.1000	0.0000	0.0700	0.0000	0.6400	-0.0150	0.7100	-0.0350
0.8600	-0.1370	0.9000	-0.2440	1.0000	-0.4770	0.0446	0.0000
0.4987	-0.0113	0.5445	-0.0125	0.5576	-0.0128	0.6691	-0.0233
0.0892	-5.0552e-4	0.1338	-0.0017	0.5712	-0.0132	0.7140	-0.0377
0.1784	-0.0029	0.2230	-0.0040	0.8921	-0.2229	0.4549	-0.0101
0.2676	-0.0052	0.3122	-0.0064	0.4483	-0.0100	0.4014	-0.0087
0.3568	-0.0075	0.7885	-0.0884	0.9593	-0.3821		

Table 5. PHASE SATURATION Data

$c_1$	0.05680429876058e+006	$d_1$	-0.14120739315590e+005
$c_2$	-0.39264357353961e+006	$d_2$	0.83084262097993e+005
$c_3$	1.12805594234952e+006	$d_3$	-2.01778226478032e+005
$c_4$	-1.72418545240933e+006	$d_4$	2.58441412755651e+005
$c_5$	1.47878241712872e+006	$d_5$	-1.83680595711727e+005
$c_6$	-0.67483667002473e+006	$d_6$	0.68453128529433e+005
$c_7$	0.12802296547207e+006	$d_7$	-0.10399245992504e+005

Table 6. Coefficients of curve fitting equations

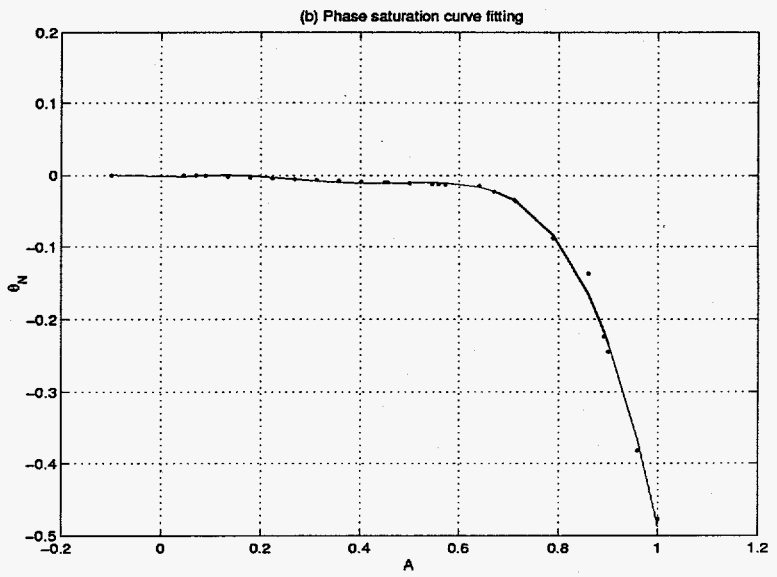
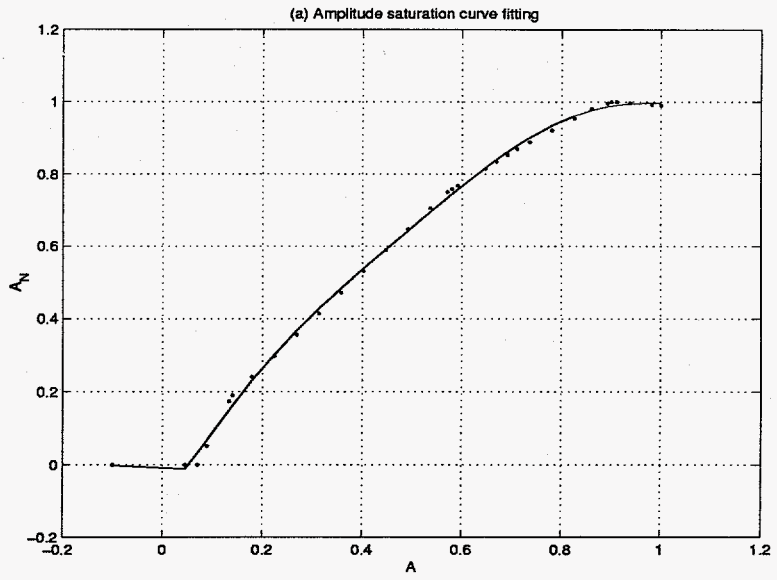


Figure 2: Curve fittings

Plugging (5) into (12) and (13), curve fitting equations (12) and (13) are reduced to

$$A_N = \sum_{i=1}^N c_i e^{-f_i w(t) \sqrt{x_1^2 + x_2^2}} \quad (14)$$

$$\theta_N = \sum_{i=1}^N d_i e^{-f_i w(t) \sqrt{x_1^2 + x_2^2}} \quad (15)$$

where

$$w(t) = \frac{K_g}{10\sqrt{KP_m}} (0.01R(t) + 1)^{1.25}. \quad (16)$$

The normalized amplitude, **N\_AMPLITUDE**  $y_1$  and the normalized phase, **N\_PHASE**  $y_2$  of the klystron are now given by

$$y_1 = A_N = \sum_{i=1}^N c_i e^{-f_i w(t) \sqrt{x_1^2 + x_2^2}} \quad (17)$$

$$\begin{aligned} y_2 &= \theta_N + \tan^{-1}\left(\frac{x_2}{x_1}\right) + 3 \cdot \frac{\pi}{180} \cdot R(t) \\ &= \sum_{i=1}^N d_i e^{-f_i w(t) \sqrt{x_1^2 + x_2^2}} + \tan^{-1}\left(\frac{x_2}{x_1}\right) + 3 \cdot \frac{\pi}{180} \cdot R(t). \end{aligned} \quad (18)$$

The relationship between **HPRF\_I**, **HPRF\_Q** and  $y_1$ ,  $y_2$  remain the same equations as given in (10), (11).

Now, consider the normalized amplitude  $y_1$  and the normalized phase  $y_2$  as given in (17) and (18).

Let

$$z_1 = \sqrt{x_1^2 + x_2^2} \quad (19)$$

$$z_2 = \tan^{-1}\left(\frac{x_2}{x_1}\right) \quad (20)$$

$$z_1 > 0, \quad -\frac{\pi}{2} < z_2 < \frac{\pi}{2}.$$

We consider a transformation from  $x$ -coordinate to  $z$ -coordinate. In  $z$ -coordinate, the state equations (3) and (4) are reduced to

$$\dot{z}_1 = -a_1 z_1 + a_1 \cos(z_2) u_1 + a_1 \sin(z_2) u_2 \quad (21)$$

$$\dot{z}_2 = -a_1 \frac{\sin(z_2)}{z_1} u_1 + a_1 \frac{\cos(z_2)}{z_1} u_2. \quad (22)$$

Also, the curve fitting equations (12) and (13) are reduced to

$$A_N = \sum_{i=1}^N c_i e^{-f_i w(t) z_1} \quad (23)$$

$$\theta_N = \sum_{i=1}^N d_i e^{-f_i w(t) z_1}. \quad (24)$$

The normalized amplitude  $y_1$  and the normalized phase  $y_2$  are represented by

$$y_1 = \sum_{i=1}^N c_i e^{-f_i w(t) z_1} \quad (25)$$

$$y_2 = \sum_{i=1}^N d_i e^{-f_i w(t) z_1} + z_2 + 3 \cdot \frac{\pi}{180} \cdot R(t). \quad (26)$$

Note that the exponents of the first term of (25) are the same as the exponents of the first term of (26). Also, note that the phase  $y_2$  is linear with respect to  $z_2$ .

In the next section, we address the control problem of the normalized amplitudes  $y_1$ 's and the normalized phases  $y_2$ 's of multiple(three) klystrons driving RFQ of LEDA.

### 3 Phase and Amplitude Control of Multiple Klystrons

In this section, we consider the control problem of phases and amplitudes of multiple klystrons which drive the RFQ. As studied in the work of Ziomek [16], phase matching of klystrons is very important. In order to achieve the matching purpose, we develop an adaptive PI controller.

Let  $K_g^m$  and  $KP_m^m$  be the klystron gain and the klystron maximum power of a model klystron and let  $(z^m, y^m, HPRF_I^m, HPRF_Q^m)$  be the representation of the model klystron.

$$z_1^m = -a_1 z_1^m + a_1 \cos(z_2^m) u_1^m + a_1 \sin(z_2^m) u_2^m \quad (27)$$

$$z_2^m = -a_1 \frac{\sin(z_2^m)}{z_1^m} u_1^m + a_1 \frac{\cos(z_2^m)}{z_1^m} u_2^m, \quad (28)$$

$$y_1^m = \sum_{i=1}^N c_i e^{-f_i w^m(t) z_1^m} \quad (29)$$

$$y_2^m = \sum_{i=1}^N d_i e^{-f_i w^m(t) z_1^m} + z_2^m + 3 \cdot \frac{\pi}{180} \cdot R(t) \quad (30)$$

$$z_1^m > 0, \quad -\frac{\pi}{2} < z_2^m < \frac{\pi}{2},$$

and

$$HPRF_I^m = 10 \sqrt{KP_m^m} \cdot y_1^m \cdot \cos(y_2^m) \quad (31)$$

$$HPRF_Q^m = 10 \sqrt{KP_m^m} \cdot y_1^m \cdot \sin(y_2^m), \quad (32)$$

where

$$w^m(t) = \frac{K_g^m}{10 \sqrt{KP_m^m}} (0.01 R(t) + 1)^{1.25}. \quad (33)$$

Similarly, let  $K_g^l$  and  $KP_m^l$  be the klystron gain and the klystron maximum power of the  $l^{th}$  klystron and let  $(z^l, y^l, HPRF_I^l, HPRF_Q^l)$  be its representation. Hence,

$$z_1^l = -a_1 z_1^l + a_1 \cos(z_2^l) u_1^l + a_1 \sin(z_2^l) u_2^l \quad (34)$$

$$z_2^l = -a_1 \frac{\sin(z_2^l)}{z_1^l} u_1^l + a_1 \frac{\cos(z_2^l)}{z_1^l} u_2^l, \quad (35)$$

$$y_1^l = \sum_{i=1}^N c_i e^{-f_i w(t) z_1^l} \quad (36)$$

$$y_2^l = \sum_{i=1}^N d_i e^{-f_i w(t) z_1^l} + z_2^l + 3 \cdot \frac{\pi}{180} \cdot R(t) \quad (37)$$

$$z_1^l > 0, \quad -\frac{\pi}{2} < z_2^l < \frac{\pi}{2},$$

and

$$HPRF_I^l = 10 \sqrt{KP_I^m} \cdot y_1^l \cos(y_2^l) \quad (38)$$

$$HPRF_Q^l = 10 \sqrt{KP_I^m} \cdot y_1^l \sin(y_2^l), \quad (39)$$

where

$$w^l(t) = \frac{K_g^l}{10 \sqrt{KP_m^l}} (0.01 R(t) + 1)^{1.25}. \quad (40)$$

The purpose of the adaptive PI controller is the perfect follow

$$y_1^l = y_1^m, \quad (41)$$

$$y_2^l = y_2^m, \quad (42)$$

or asymptotic follow

$$y_1^l \rightarrow y_1^m, \quad (43)$$

$$y_2^l \rightarrow y_2^m. \quad (44)$$

When the perfect follow is achieved,

$$\sum_{i=1}^N c_i e^{-f_i w^m(t) z_1^m} = \sum_{i=1}^N c_i e^{-f_i w^l(t) z_1^l} \quad (45)$$

and

$$\begin{aligned} \sum_{i=1}^N d_i e^{-f_i w^m(t) z_1^m} + z_2^m + 3 \cdot \frac{\pi}{180} \cdot R(t) \\ = \sum_{i=1}^N d_i e^{-f_i w^l(t) z_1^l} + z_2^l + 3 \cdot \frac{\pi}{180} \cdot R(t). \end{aligned} \quad (46)$$

From (33) and (40), we obtain the relation

$$\begin{aligned} w^m(t)z_1^m &= w^l(t)z_1^l \\ \frac{K_g^m}{\sqrt{KP_m^m}}z_1^m &= \frac{K_g^l}{\sqrt{KP_m^l}}z_1^l. \end{aligned} \quad (47)$$

Hence,

$$z_1^l = \frac{K_g^m}{K_g^l} \sqrt{\frac{KP_m^l}{KP_m^m}} z_1^m. \quad (48)$$

Also, we obtain

$$z_2^l = z_2^m. \quad (49)$$

Similarly, when the asymptotic follow is achieved,

$$z_1^l \rightarrow \frac{K_g^m}{K_g^l} \sqrt{\frac{KP_m^l}{KP_m^m}} z_1^m, \quad (50)$$

and

$$z_2^l \rightarrow z_2^m. \quad (51)$$

Considering (48), (49), (50), and (51), we construct factitious systems which are parts of adaptive PI controller.

For the model klystron, we construct a factitious model given by

$$\dot{z}_1^m = -a_1 z_1^m + a_1 \cos(z_2^m) u_1^m + a_1 \sin(z_2^m) u_2^m \quad (52)$$

$$\dot{z}_2^m = -a_1 \frac{\sin(z_2^m)}{z_1^m} u_1^m + a_1 \frac{\cos(z_2^m)}{z_1^m} u_2^m, \quad (53)$$

$$v_1^m = z_1^m \quad (54)$$

$$v_2^m = z_2^m \quad (55)$$

where  $v_1^m$  and  $v_2^m$  are outputs of the factitious model.

For the  $l^{th}$  klystron, we construct a factitious system given by

$$\dot{z}_1^l = -a_1 z_1^l + a_1 \cos(z_2^l) u_1^l + a_1 \sin(z_2^l) u_2^l \quad (56)$$

$$\dot{z}_2^l = -a_1 \frac{\sin(z_2^l)}{z_1^l} u_1^l + a_1 \frac{\cos(z_2^l)}{z_1^l} u_2^l, \quad (57)$$

$$v_1^l = C_v z_1^l \quad (58)$$

$$v_2^l = z_2^l \quad (59)$$



where  $v_1^l$  and  $v_2^l$  are outputs of the factitious klystron and the coefficient  $C_v$  is given by

$$C_v = \frac{K_g^l}{K_g^m} \sqrt{\frac{K P_m^m}{K P_m^l}}. \quad (60)$$

The adaptive PI controller is designed based on two factitious systems in such a way that

$$v_1^l \rightarrow v_1^m \quad (61)$$

$$v_2^l \rightarrow v_2^m \quad (62)$$

If the adaptive PI controller guarantees convergences (61) and (62), by (50), (51), (54), (55), and (58), (59), we obtain the convergences (43) and (44). The convergence (61) is equivalent to

$$w^l(t)z_1^l \rightarrow w^m(t)z_1^m. \quad (63)$$

Hence, we obviously obtain the convergences (43) and (44),

$$y_1^l \rightarrow y_1^m \quad (64)$$

$$y_2^l \rightarrow y_2^m. \quad (65)$$

Now, consider the adaptive PI controller. In order to design the adaptive PI controller, we make use of the simple adaptive controller concept[8]. To develop the adaptive PI controller, we consider the system as given in (56)-(59). The system as given in (56)-(59) is described by

$$\dot{z}^l = A_l z^l + B(z^l)u^l \quad (66)$$

$$v^l = C_l z^l \quad (67)$$

where

$$A_l = \begin{bmatrix} -a_1 & 0 \\ 0 & 0 \end{bmatrix},$$

$$B(z^l) = \begin{bmatrix} a_1 \cos(z_2^l) & a_1 \sin(z_2^l) \\ -a_1 \frac{\sin(z_2^l)}{z_1^l} & a_1 \frac{\cos(z_2^l)}{z_1^l} \end{bmatrix},$$

$$C_l = \begin{bmatrix} C_v & 0 \\ 0 & 0 \end{bmatrix}$$

with appropriate definitions of state  $z^l$  and input  $u^l$  and output  $v^l$ . Similarly, we can describe the model klystron as given in (52)-(55) as

$$\dot{z}^m = A_m z^m + B(z^m)u^m \quad (68)$$

$$v^m = C_m z^m \quad (69)$$

with similar definitions of system matrices  $A_m (= A_l)$ ,  $B(z^m)$ ,  $C_m$  and state  $z^m$ , input  $u^m$ , output  $v^m$ .

(66), (67) can be considered as a nonlinearly perturbed linear multivariable system with reference signal  $v^m$  generated by the system as given in (68) and (69):

$$\dot{z}^l = A_l z^l + B_l(u + g(z^l, u^l)) \quad (70)$$

$$v^l = C_l z^l \quad (71)$$

$$e_v = v^m - v^l \quad (72)$$

where

$$B_l = \begin{bmatrix} a_1 \cos(z_2^m) & a_1 \sin(z_2^m) \\ -a_1 C_v \frac{\sin(z_2^m)}{z_1^m} & a_1 C_v \frac{\cos(z_2^m)}{z_1^m} \end{bmatrix} \quad (73)$$

$$g(z^l, u^l) = \begin{bmatrix} g_{11}(z^l) & g_{12}(z^l) \\ g_{21}(z^l) & g_{22}(z^l) \end{bmatrix} u^l \quad (74)$$

$$g_{11}(z^l) = a_1 \cos(z_2^m) \cos(z_2^l) + \frac{a_1}{C_v} \frac{z_1^m}{z_1^l} \sin(z_2^m) \sin(z_2^l) - 1$$

$$g_{12}(z^l) = a_1 \cos(z_2^m) \sin(z_2^l) - \frac{a_1}{C_v} \frac{z_1^m}{z_1^l} \sin(z_2^m) \cos(z_2^l)$$

$$g_{21}(z^l) = a_1 \sin(z_2^m) \cos(z_2^l) - \frac{a_1}{C_v} \frac{z_1^m}{z_1^l} \cos(z_2^m) \sin(z_2^l)$$

$$g_{22}(z^l) = a_1 \sin(z_2^m) \sin(z_2^l) + \frac{a_1}{C_v} \frac{z_1^m}{z_1^l} \cos(z_2^m) \cos(z_2^l) - 1.$$

In (70) and (71),

$$C_l B_l = \begin{bmatrix} a_1 C_v \cos(z_2^m) & a_1 C_v \sin(z_2^m) \\ -a_1 C_v \frac{\sin(z_2^m)}{z_1^m} & a_1 C_v \frac{\cos(z_2^m)}{z_1^m} \end{bmatrix}, \quad (75)$$

and  $\det(C_l B_l) \neq 0$  for any bounded  $z_1^m$ . Also,

$$\det \begin{bmatrix} sI - A_l & B_l \\ C_l & 0 \end{bmatrix} = \frac{a_1^2 C_v^2}{z_1^l} \neq 0 \quad (76)$$

for any bounded  $z_1^l$ . Let  $P = P^T$  be a positive definite matrix defined by

$$P = \begin{bmatrix} p_1 & p_2 \\ p_2 & p_3 \end{bmatrix}. \quad (77)$$

If we take

$$p_1 = 1 \quad (78)$$

$$p_2 = 0 \quad (79)$$

$$p_3 = z_1^m, \quad (80)$$

then, since  $-\pi/2 < z_2^m < \pi/2$ ,

$$PC_l B_l + (C_l B_l)^T P = \begin{bmatrix} 2a_1 C_v \cos(z_2^m) & 0 \\ 0 & 2a_1 C_v \cos(z_2^m) \end{bmatrix} > 0. \quad (81)$$

Hence, there exists a static output feedback that achieves stability and solves servomechanism problem guaranteeing boundedness of  $e_v$ [5],[13],[14] and this, in turn, guarantees existence of a simple adaptive control(SAC)[1],[2].

The adaptive PI controller based on SAC is given by the following:

$$r(t) = \begin{bmatrix} v_1^m - v_1^l \\ v_2^m - v_2^l \\ z_1^m \\ z_2^m \\ u_1^m \\ u_2^m \end{bmatrix}, \quad (82)$$

$$K_r(t) = \begin{bmatrix} K_e(t) & K_z(t) & K_u(t) \end{bmatrix}, \quad (83)$$

$$u(t) = \begin{bmatrix} u_1(t) \\ u_2(t) \end{bmatrix} = \begin{bmatrix} LLRF\_L\_I(t) \\ LLRF\_L\_Q(t) \end{bmatrix} = K_r(t)r(t) \quad (84)$$

$$K_r(t) = K_i(t) + K_p(t) \quad (85)$$

$$\dot{K}_i(t) = \begin{bmatrix} v_1^m - v_1^l \\ v_2^m - v_2^l \end{bmatrix} r^T(t) T_i, \quad T_i = T_i^T > 0 \quad (86)$$

$$K_p(t) = \begin{bmatrix} v_1^m - v_1^l \\ v_2^m - v_2^l \end{bmatrix} r^T(t) T_p, \quad T_p = T_p^T \geq 0. \quad (87)$$

The integral gain  $K_e(t)$  is the crucial component of the controller and governs the stability of the controlled system. The integral gain  $K_e(t)$  can be viewed as the gain of high gain adaptive control[10],[14],[18].  $K_z(t)$  and  $K_u(t)$  help improve the performance of the adaptive model-following system and even achieve perfect following in ideal environments[3]. The proportional gain  $K_p(t)$  is added for its immediate action and makes the tracking error small quickly.

Figure 3 depicts an amplitude and phase controller for a klystron. FACTITIOUS MODEL represents the factitious model given by (52)-(55) for the prototype klystron operated as a model, FACTITIOUS SYSTEM\_I is the  $i^{th}$  factitious system represented by (56)-(59). LLRF\_M\_I and LLRF\_M\_Q are used for inputs of FACTITIOUS MODEL. LLRF\_I\_I and LLRF\_I\_Q are generated from the adaptive PI controller and are used for inputs of FACTITIOUS SYSTEM\_I.  $K_{p_i}$  represents the adaptive proportional gain  $K_{p_i}$  and  $K_{i_i}$  represents the adaptive integration gain  $K_{i_i}$ . The adaptive controller makes use of both LLRF\_M\_I and LLRF\_M\_Q, FACTITIOUS MODEL outputs  $v_1^m, v_2^m$ , FACTITIOUS MODEL states  $z_1^m, z_2^m$  and FACTITIOUS SYSTEM\_I outputs  $v_1^i, v_2^i$ .

For a multiple klystron system, we make parallel connections of factitious systems and the corresponding adaptive PI controllers. Of interest is the case when the system is composed of three klystrons. Figure 4 shows the case when the system has three klystrons. One of three klystrons is taken as a model klystron. The closed loop system is composed of a factitious model(FACTITIOUS MODEL) as given in (52)-(55), two adaptive control systems(ADAPTIVE CONTROL SYSTEM 1, ADAPTIVE CONTROL SYSTEM 2), and three klystrons(KLYSTRON\_M, KLYSTRON\_I, KLYSTRON\_2). An adaptive control system is composed of a factitious system and an adaptive PI controller as shown in Figure 3. The first klystron(KLYSTRON\_M) is used for model klystron and its klystron gain and klystron maximum power are used for  $K_g^m$  and  $KP_m^m$ . LLRF\_M\_I and LLRF\_M\_Q are inputs to KLYSTRON MODEL. They are inputs of FACTITIOUS MODEL and the adaptive control systems. States and outputs of FACTITIOUS MODEL are feedback to adaptive PI controllers too. Each adaptive control system generates LLRF\_I and LLRF\_Q which is feedback to the corresponding klystron.

The klystron gains  $K_g^i$ s and klystron maximum powers  $KP_m^i$ s used in the simulation are given in Table 7[16].

$K_g^m$	4410	$KP_m^m$	763980(Watts)
$K_g^1$	5145	$KP_m^1$	590640(Watts)
$K_g^2$	5145	$KP_m^2$	642000(Watts)

Table 7. Klystron gain and klystron maximum power

The gain matrices  $T_i$  and  $T_p$  of two adaptive PI controllers are given by

$$T_i = 5000I_{6 \times 6}$$

$$T_p = 500I_{6 \times 6}.$$

Figure 5 shows unit step inputs LLRF\_M\_I, LLRF\_M\_Q and inputs  $u_1^i$ s, LLRF\_I\_I, LLRF\_I\_Q, and inputs  $u_2^i$ s, LLRF\_I\_Q, LLRF\_I\_Q generated by simple adaptive control systems. Figure

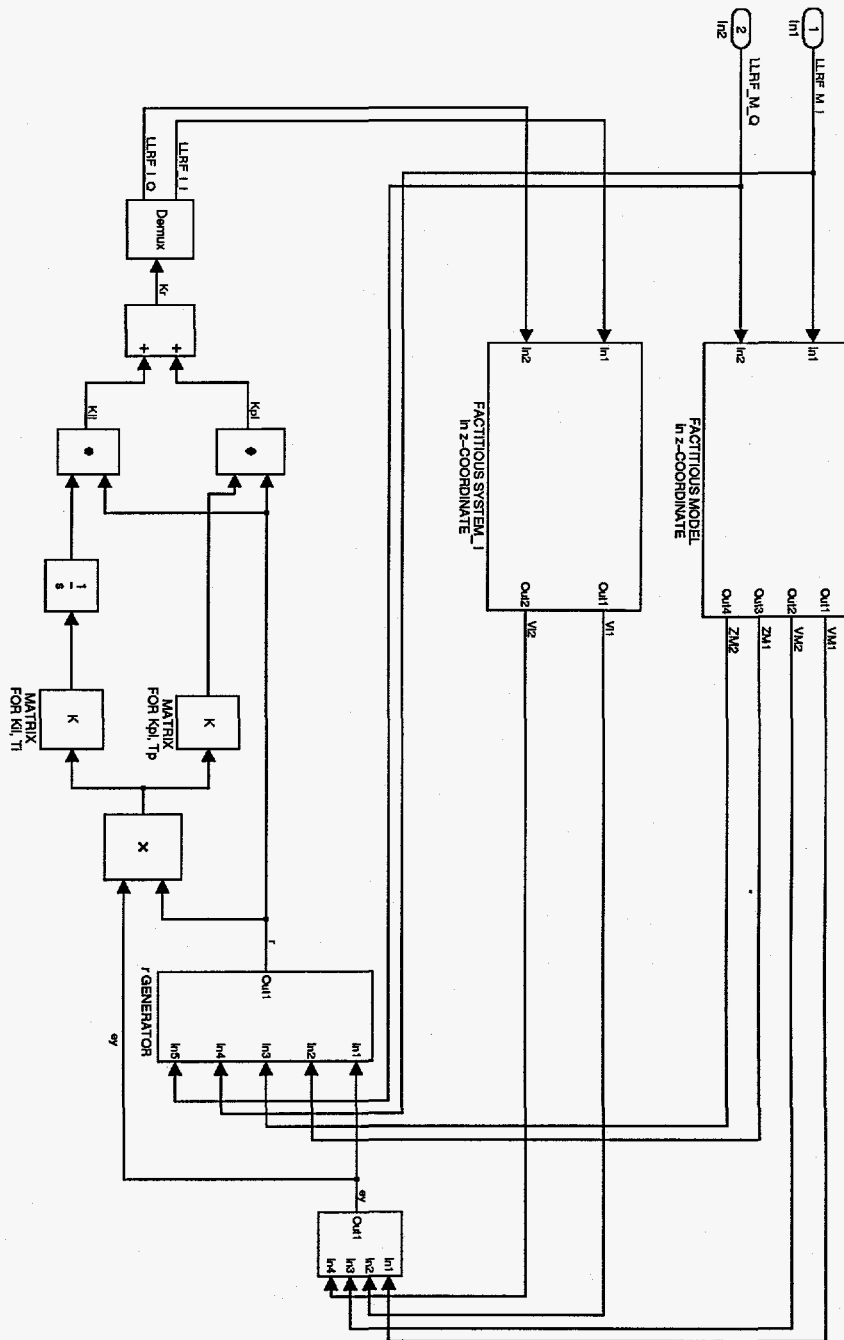


Figure 3: The adaptive PI controller.  $z_1^m=ZM1$ ,  $z_2^m=ZM2$ ,  $v_1^m=VM1$ ,  $v_2^m=VM2$ ,  $u_1^m=LLRF\_M\_I$ ,  $u_2^m=LLRF\_M\_Q$ ,  $u_1^l=LLRF\_I\_I$ ,  $u_2^l=LLRF\_I\_Q$ .

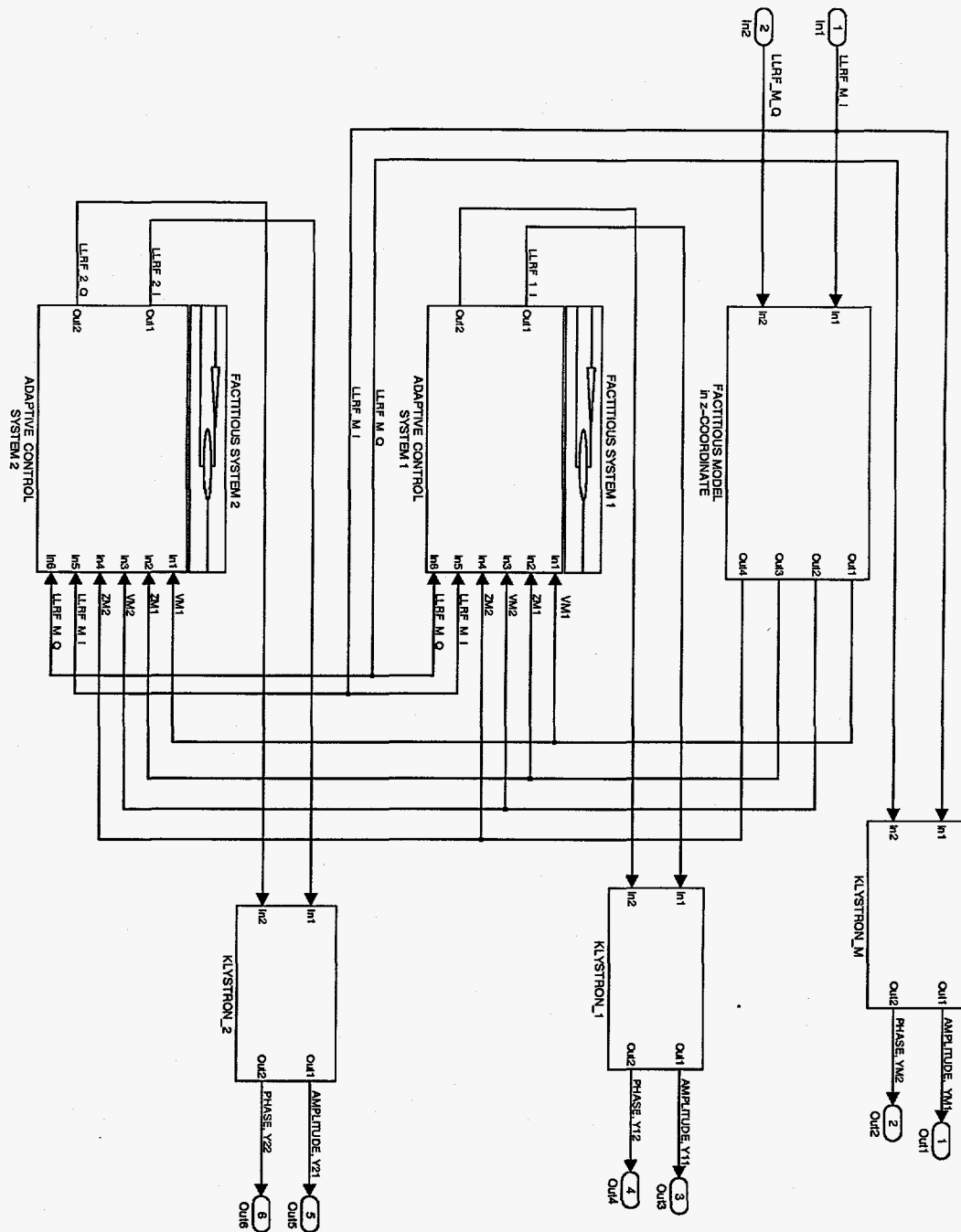


Figure 4: The adaptive PI control closed loop system for three klystrons. Inputs for this closed loop system are LLRF\_M.I and LLRF\_M.Q. LLRF\_1.I, LLRF\_1.Q are generated from ADAPTIVE CONTROL SYSTEM 1; and LLRF\_2.I, LLRF\_2.Q are generated from ADAPTIVE CONTROL SYSTEM 2. These are feedback to KLYSTRON\_1 and KLYSTRON\_2.  $z_1^m=ZM1$ ,  $z_2^m=ZM2$ ,  $v_1^m=VM1$ ,  $v_2^m=VM2$ ,  $u_1^m=LLRF\_M.I$ ,  $u_2^m=LLRF\_M.Q$ ,  $y_1^m=YM1$ ,  $y_2^m=YM2$ ,  $y_1^1=Y11$ ,  $y_2^1=Y12$ ,  $y_1^2=Y21$ ,  $y_2^2=Y22$ .

6 shows klystrons' normalized amplitudes  $y'_1s$  and normalized phases  $y'_2s$  when LLRF\_M\_I and LLRF\_M\_Q are unit step inputs. Figure 6 shows that amplitudes and phases of KLYSTRON\_1 and KLYSTRON\_2 converge to amplitude and phase of klystron model, KLYSTRON\_M.

Figure 7 shows LLRF\_M\_I, LLRF\_M\_Q and inputs  $u'_1s$ , LLRF\_1\_I, LLRF\_2\_I, and inputs  $u'_2s$ , LLRF\_1\_Q, LLRF\_2\_Q generated by simple adaptive control systems when LLRF\_M\_I and LLRF\_M\_Q are

$$\begin{aligned}LLRF\_M\_I &= 1.0e^{-5.0e5t}(\sin(2.0e6t) + 1.0) \\LLRF\_M\_Q &= 1.0e^{-5.0e5t}\sin(1.0e6t).\end{aligned}$$

Figure 8 shows klystrons' amplitudes,  $y'_1s$  and phases,  $y'_2s$ . Figure 8 shows that amplitudes and phases of KLYSTRON\_1 and KLYSTRON\_2 converge to amplitude and phase of klystron model, KLYSTRON\_M.

Figure 9 shows LLRF\_M\_I and LLRF\_M\_Q and inputs  $u'_1s$ , LLRF\_1\_I, LLRF\_2\_I, and inputs  $u'_2s$ , LLRF\_1\_Q, LLRF\_2\_Q generated by simple adaptive control systems when LLRF\_M\_I and LLRF\_M\_Q are

$$\begin{aligned}LLRF\_M\_I &= 1.0 + 0.25\sin(2.0e6t) \\LLRF\_M\_Q &= 0.25\sin(1.0e6t).\end{aligned}$$

Figure 10 shows klystrons' amplitudes,  $y'_1s$  and phases,  $y'_2s$ .

## 4 Conclusion

An adaptive PI controller has been developed for phase synchronization of a multi-klystron driven accelerating cavity. This controller, based on one of the klystrons, modifies the outputs of the others in order to present the *same* drive signals to the cavity.

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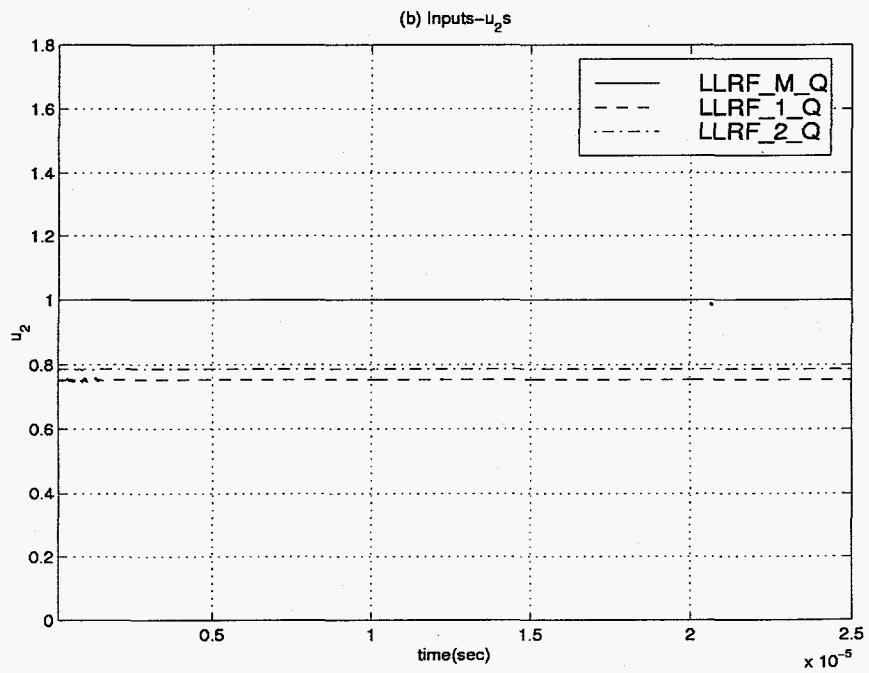
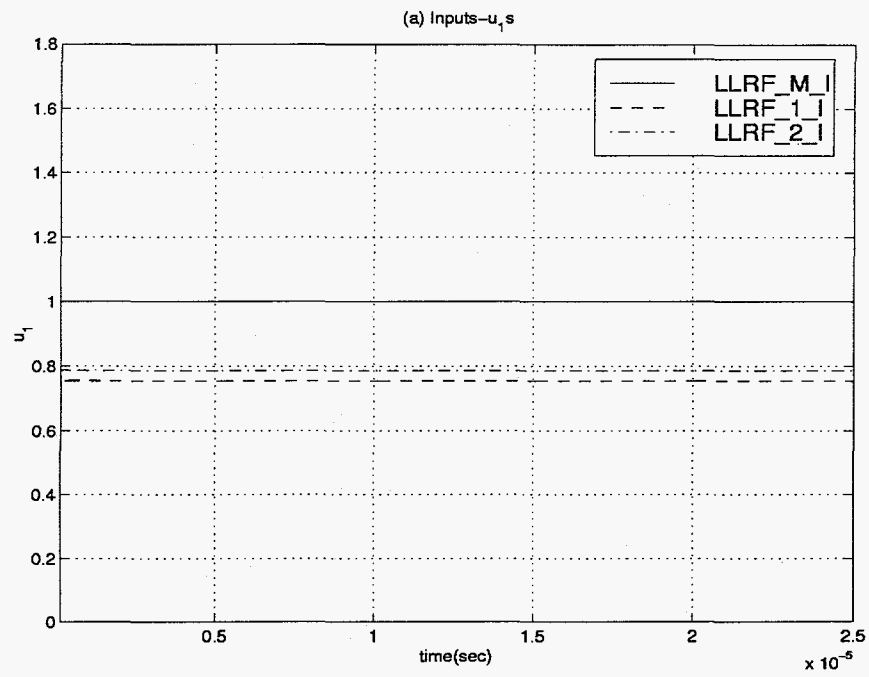


Figure 5: Inputs for three klystrons.

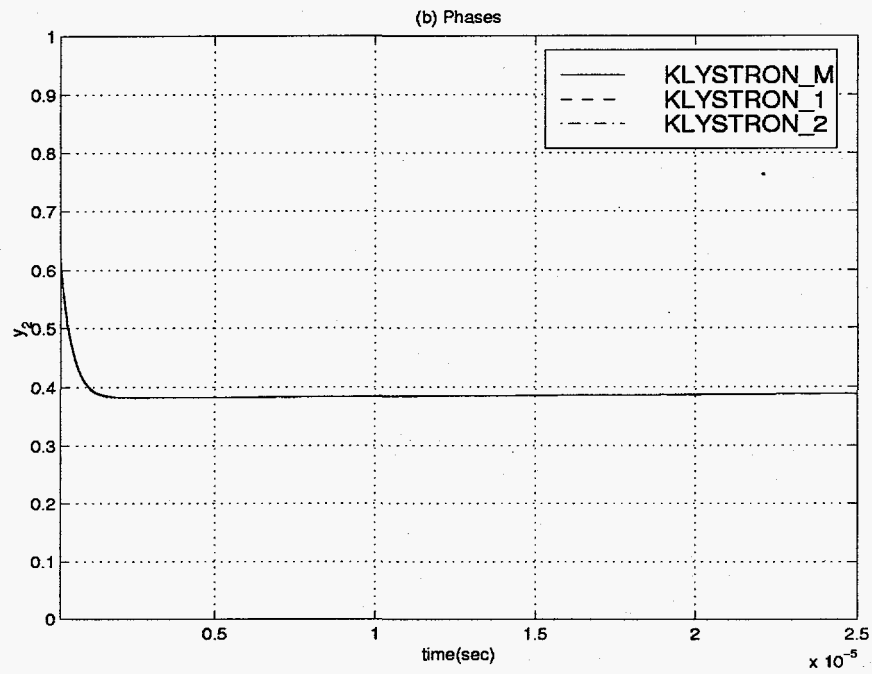
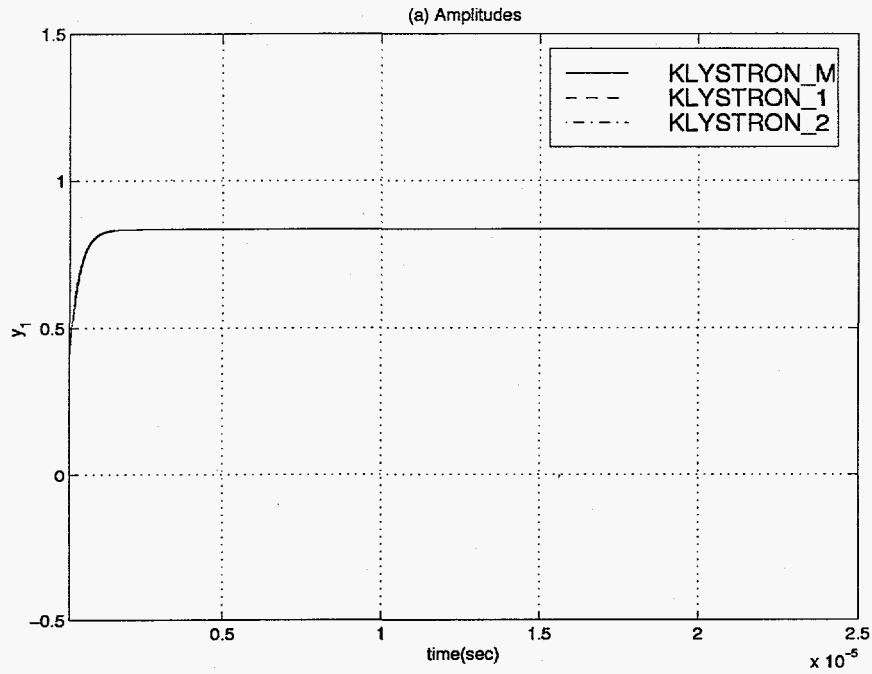


Figure 6: Amplitudes  $y_1$  and Phases  $y_2$  for three klystrons.

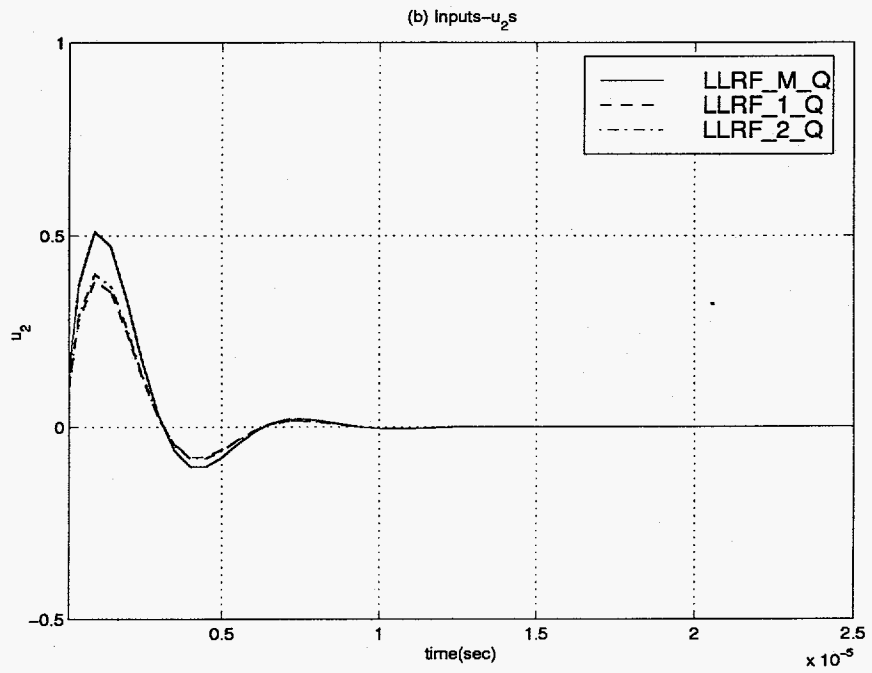
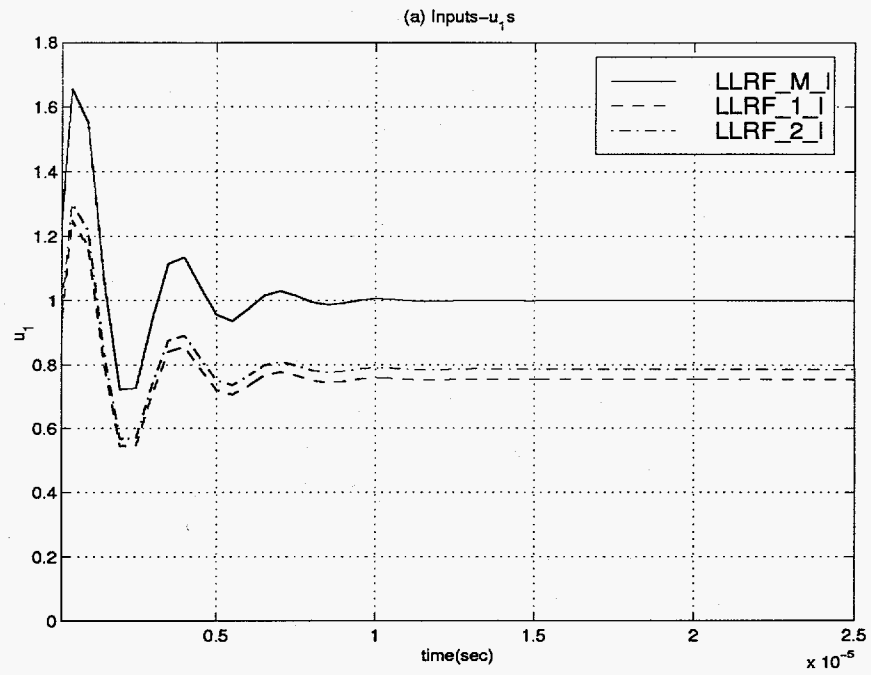


Figure 7: Inputs for three klystrons.

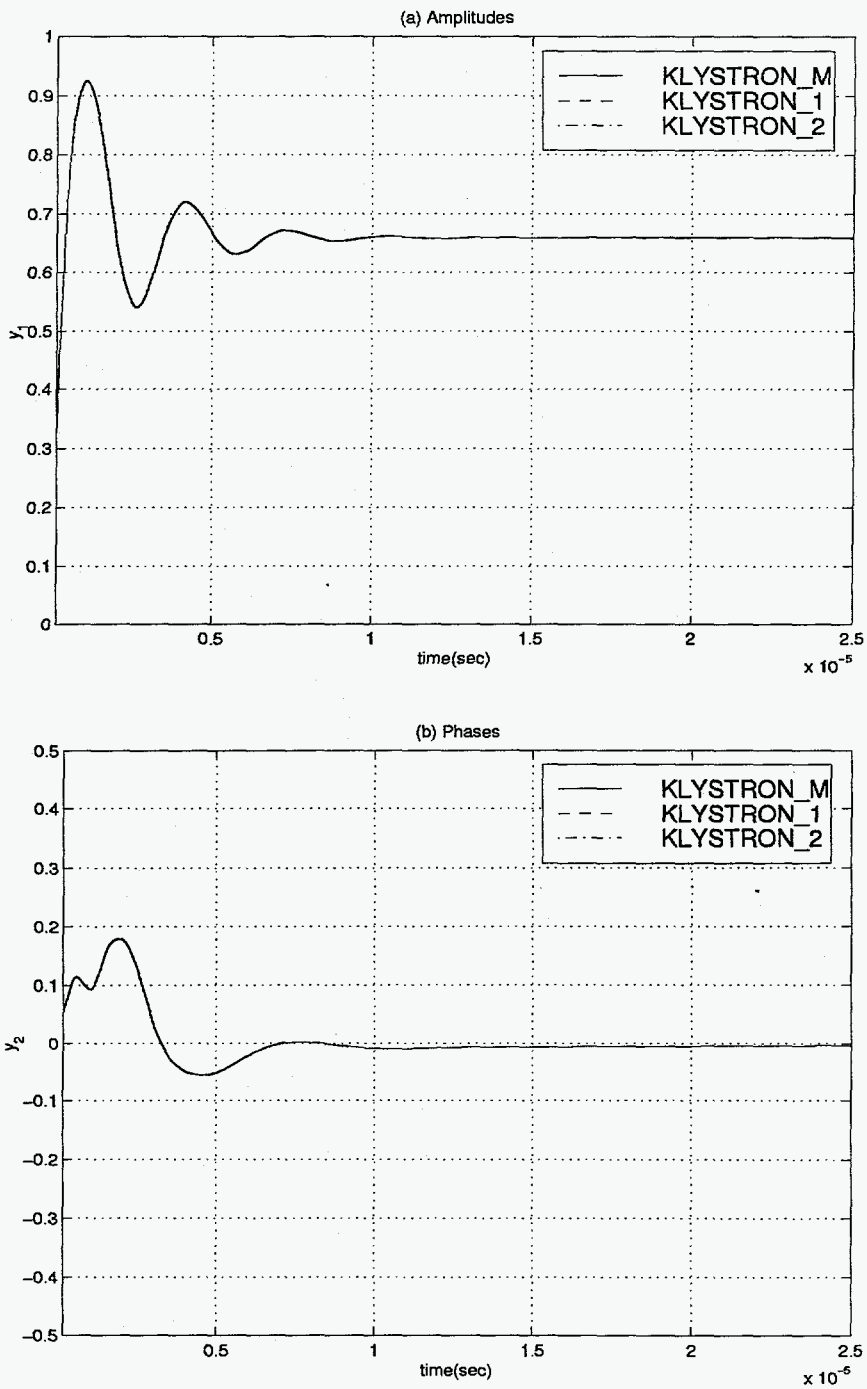


Figure 8: Amplitudes  $y_1$  and Phases  $y_2$  for three klystrons.

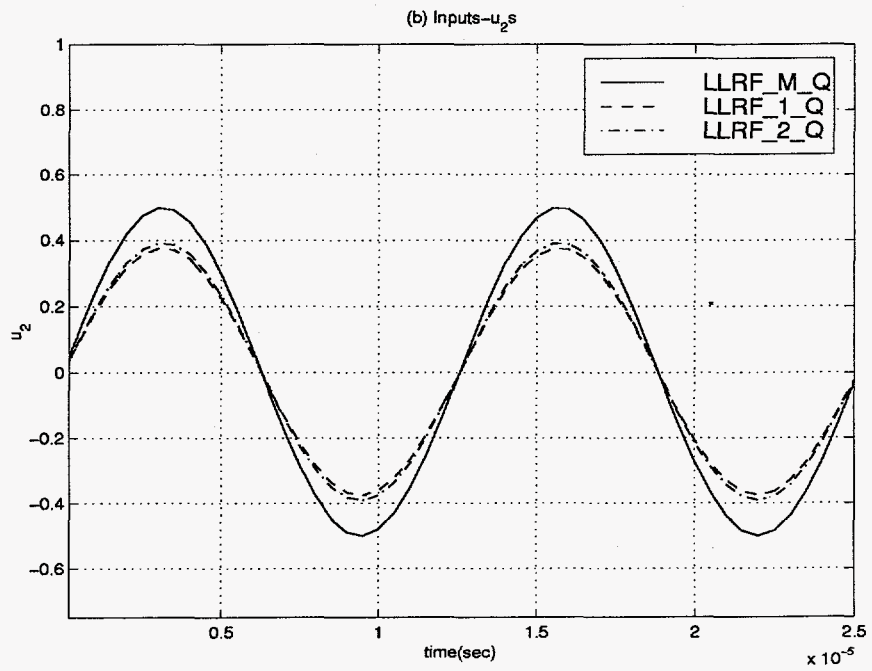
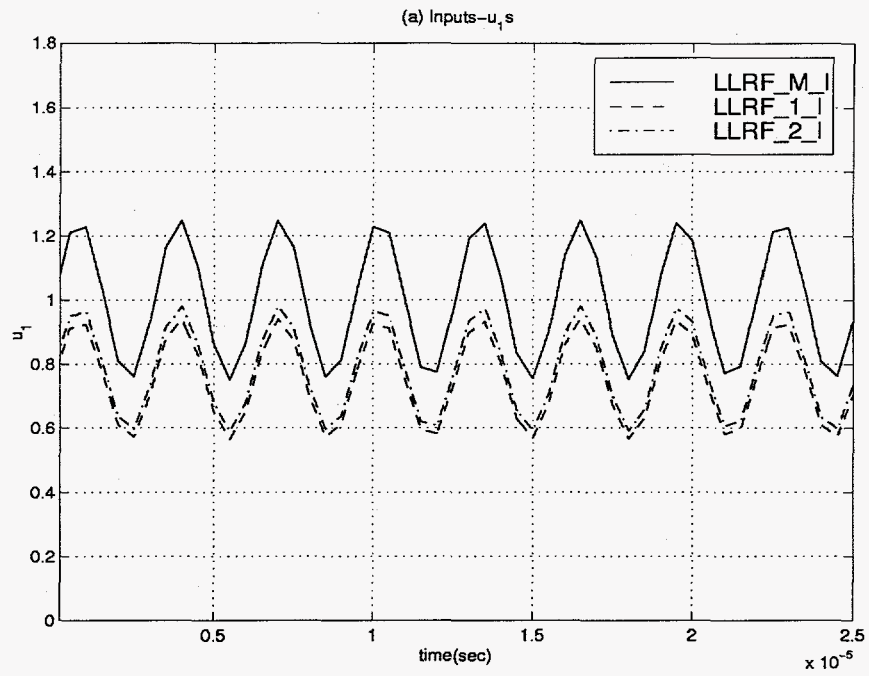


Figure 9: Inputs for three klystrons.

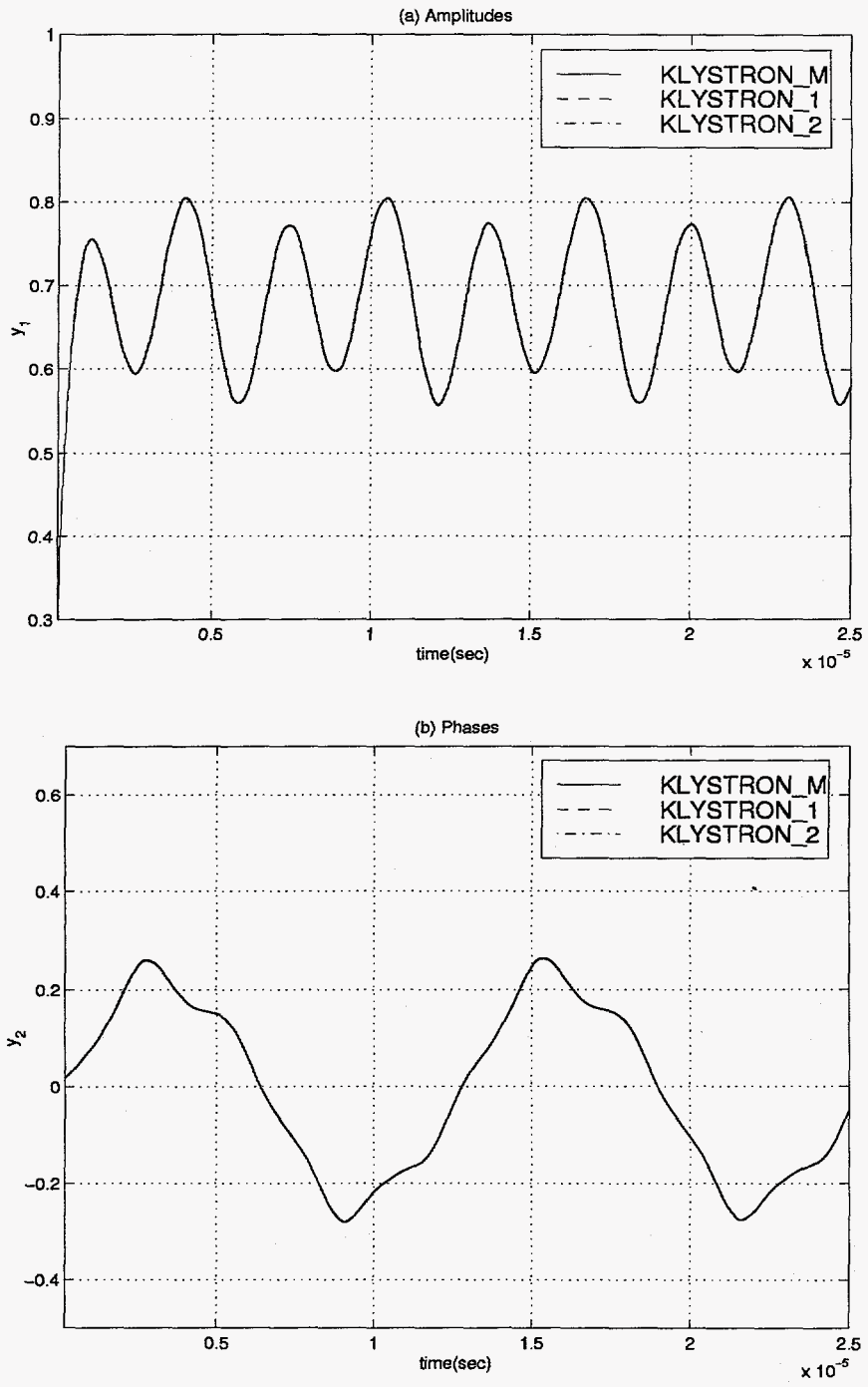


Figure 10: Amplitudes  $y_1$  and Phases  $y_2$  for three klystrons.