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# Development of a design engineering in-situ sensor payload optimization tool

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DEVELOPMENT OF A DESIGN ENGINEERING IN-SITU SENSOR PAYLOAD  
OPTIMIZATION TOOL

A Thesis

Presented to

The Faculty of the Department of Mechanical and Aerospace Engineering  
San Jose State University

In Partial Fulfillment

of the Requirements for the Degree

Master of Science

by

Keith Ronald Schreck

May 2008

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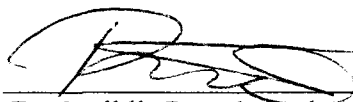
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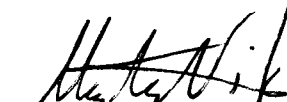
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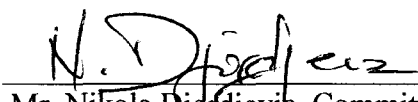
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## ABSTRACT

### DEVELOPMENT OF A DESIGN ENGINEERING IN-SITU SENSOR PAYLOAD OPTIMIZATION TOOL

By Keith R. Schreck

Determination of instrumentation for space science missions is an involved, complex procedure. Components are chosen to meet mission science requirements, creating an initial sensor package design. Design reviews iterate on the initial concept and options are evaluated until a final design solution is determined. Trade studies are traditionally performed at the component level. A final design solution is determined at the end of this often lengthy process. The analysis performed in this work looks at mission requirements and generates a mission sensor package using design engineering relations. Given a set of data to be returned from a science mission, a sensor package that meets mission requirements can be generated for a design solution. A tool for in-situ measurements is developed using systems engineering design relations to deliver a sensor payload configuration.

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## **I. Introduction**

The driving interest behind the development of this program comes from material presented during a short course on In-Situ Instruments for Planetary Probes and Aerial Platforms hosted as part of the 4<sup>th</sup> International Planetary Probe Workshop [1]. Attendees were given a mass and power budget for a planetary probe mission that used an aerial platform, and were tasked to develop a sensor package that would meet the mission requirement and fit within the constraints. Long before a mission is launched, during the initial planning stages, a series of studies are conducted to create a sensor package custom tailored to meet the mission requirements. Components are chosen to survive the operating environment and meet mission requirements.

Design of the sensor payload package for any mission addresses several issues. The final optimal payload configuration is a result of individual case studies and design engineering studies. The scope of the tool, is limited to optimization techniques within the sensor payload, however, a higher system level criteria may impact the component level design, resulting in a different component selection. At some point, a human decision is still included in the design process, as final selection between competing elements is made. The decision to use one component over another can arise from many factors – functionality, heritage, Technology Readiness Level (TRL) [2], etc. The objective of this work is to combine all of the selection techniques for mission hardware into a single tool that can be used to generate a preliminary sensor package configuration.



## **II. Methodology**

The component selection algorithms implemented in this tool trades sensor component characteristics (operational parameters such as range, performance, weight, accuracy, etc.) to arrive at an optimum component choice, based on a set of mission sensor requirements (e.g., planetary atmospheric data collection). Initial configurations are developed using top level mission requirements. As the solution for each sensor type progresses, the properties of the sensor are evaluated at finer levels of analysis. If a component no longer satisfies a requirement it is eliminated from analysis. If no suitable solution can be determined, a work-around strategy must be made to find a way to modify existing hardware to satisfy the mission requirements, either via a custom built specialized sensor, modifying a commercially available component to allow it to meet the requirement, or by making a modification to the mission requirement. The end result of this design tool for each type of mission science data is a unique commercially available sensor component.

A database of commercially available components is developed for each type of sensor. The down-selection process will employ several methods to eliminate incompatible sensors. Primary selection methods are based on the operational range of the sensor type (e.g., temperature range for Thermocouples, atmospheric gasses for mass spectrometers). Special consideration is given to heritage system components, to further select from multiple sensors that operate over similar operational ranges. Use of heritage materials implies a high level of technological development behind the sensor. The Technology Readiness Level (TRL) employed in a sensor's design, relates the

development level and risk associated with the hardware. While technology with a higher TRL is desirable, there are other advantages to a lower TRL device. A given device could be at a high TRL, but be a heavy component or involve a complicated mechanism. A similar device could have a lower TRL, but be significantly lighter. However, the lower TRL device has an increased level of inherent risk in its use, compared to a more mature design. Multi-role components can also be evaluated for their useful properties. These types of units complete the tasks assigned to multiple sensors with the benefit of a single unit capable of recording several data types. The use of these selection factors will allow for the determination of a component that will meet the mission sensor requirements.

As each component is selected for the sensor package, additional interactions between sensors will come into play. Design constraints may limit the use of certain types of sensors. Once a preliminary sensor design package has been completed, interactions between sensors at a system level may determine if any components are incompatible with other sensors. If this condition occurs, individual sensor requirements will have to be modified and evaluated via another iteration with all the sensors until the sensors are compatible with each other and a final design solution exists, or the program will determine that there is no commercially existing solution that meets the mission requirements. In this event a suitable solution sensor design will be a close result to the final sensor that will have to be custom modified.

### **III. ISSPO Tool Subroutines**

The ISSPO Program is comprised of multiple subroutines being called from the main program. Each subroutine is called and returns specific pieces of data back to the main program. Sensor type subroutines are developed as a self contained model only requiring inputs from the main program to select the correct component or, when necessary, obtain data from another module to select the component. The subroutine for each sensor type is only called if a corresponding type of data is requested in the main program (e.g., ACCELERATION for acceleration data, OPTICS for imaging data, GAS ANALYZER for gas properties). This modular development allows for the program to include all sensor types, yet reduce running time to only relevant sensor types. Many planetary science missions feature a basic atmospheric properties sensor pack that monitor temperature, pressure, density, etc. Within the ISSPO tool these atmospheric sensors have been coupled into a single input option selection that calls all the individual sensors automatically. The function and properties of each of the ISSPO Tool subroutines is discussed here. Descriptions here are not meant as an exhaustive description of the design flow through each module, but to detail the key elements of the modules, limitations, design logic and the data used.

#### **A. ISSPO Program**

The ISSPO routine is the primary program call entered at the MatLAB command prompt. From here, the input data is loaded to the main program and the sensor payload is configured from the sensor modules called for in this program. Each module contains

its own set of variables needed by the program. At the end of each module, the relevant data is written to a '.mat' binary data file. In the main ISSPO program, the data file is loaded into memory and the data is made available for all subroutines to use. A summary program flowchart in Figure 1 outlines the operation of the ISSPO Tool.

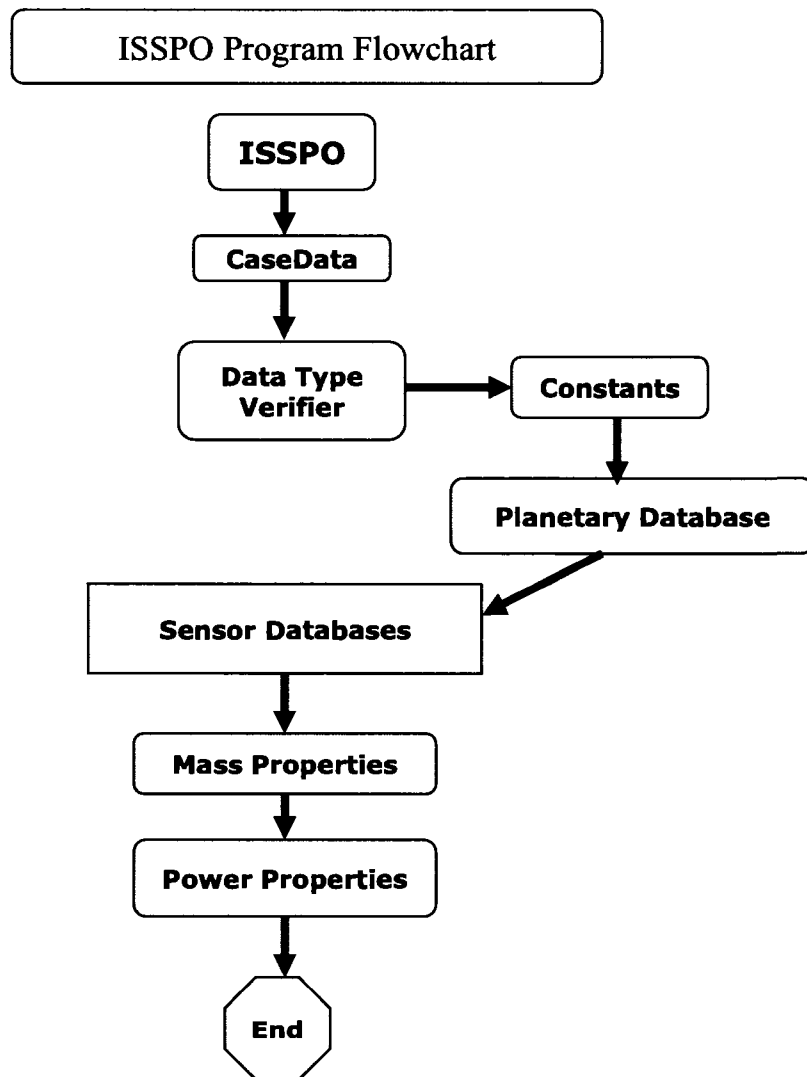


Figure 1 ISSPO Tool Flowchart Diagram

The development and design of the ISSPO program is keyed toward a simple, minimal design. To save time in execution of multiple runs, all the program input variables are preloaded into a case data file with the MatLAB '.m' file extension. This allows the file to be evaluated as a program and it loads all the relevant case data. When ISSPO is executed from MatLAB, a brief introduction to the program is displayed and the user is asked to enter the name of the data file that contains the design information for that case. A logic check for the file is made and continues the program. If the file doesn't exist, an error message is printed to the user and the program ends. Assuming the file exists and is set up correctly, the ISSPO program creates a case directory in a working directory folder. From there a program directory file folder is made and copies of the main program files are copied into the working directory folder. The main program then runs a planetary database program to obtain reference data values for the intended mission location; such as bulk parameters, orbital properties, and atmospheric data. Specific modules are executed for each sensor type depending on the required mission data.

## **B. Input File**

The input file is not, in reality, a program executed as part of the ISSPO program. Instead it is intended to serve as a file to load the basic requirements of the sensor system design into the main program. The main benefit here is to quickly load all of the program inputs from a single file, saving the user from having to enter the data from scratch each time the program is executed. Each time a design change is made, only the variable has

to be updated in the input file. A default set of case inputs and required input data is included in the CaseData MatLAB file.

### **C. Constants**

The constants subroutine is as straightforward as it sounds. Constant data values or conversion factors for different units are stored within two different data sets. These sets correspond to the chosen unit system. Data values and labels are stored for use by the various programs invoked from the main ISSPO program. All of the constants' data properties are written to a MatLAB '.mat' binary data file and loaded into memory after the program is executed.

### **D. Data Type Verifier**

The Data Type Verifier program is a simple subroutine used to verify inputs. The type of sensor data to be collected by the mission is stored within a cell data array in the input file program. The Input File sensor data array is compared to a list of allowed data types in the ISSPO Program. This verifies the data entered for any misspellings or unknown data types. If the program finds a data type that is not within the program's database, it prints out the unknown data type to screen, prints an error message, and terminates the program. Additional logic checks are performed on the input file for the different input sensor for any required additional information needed to select the final sensor. Required information varies with each sensor type and is discussed in each section.

## **E. Planetary Database**

The primary purpose of this module is to provide data to the main ISSPO program on the planetary environment that the mission would encounter. Planetary data is crucial in the decision process of sensors for the mission. A database of all the planetary values is generated based on the object relevant parameters, and is stored in a data file loaded by the main ISSPO program to be used in selection of the sensor components. Intended as a planetary object database to determine the environment for a mission, the database provides enough information on planetary bodies that it can be of use to any interested researcher. To aide the common user, the program is sufficiently developed to be used as a component in the ISSPO program or serve as a stand-alone model with its own interactive command structure. From within ISSPO, the planetary database program is called to load the parameters into the main program.

The program consists of a database of planetary bulk parameters (e.g., mass, volume, radius, gravity), orbital parameters (e.g., period, velocity, orbital inclination) and atmospheric properties. Composition of the atmosphere includes major components by percent and trace element composition by particle concentration. These parameters are used to aid further along in the design process in selecting other components.

The planetary database currently contains the full set of properties of 27 celestial objects including the sun, the eight major planets, the moon, the recently reclassified minor planet Pluto, and the eight largest moons of Jupiter and Saturn. A sparse set of data is currently available from NASA's Planetary Database for Jupiter's and Saturn's moons. With increased interest in these moons, the currently available data for these

objects is included in the database with temporary placeholder values for the unknown properties. The data for these bodies is taken only from NASA's Planetary Database [3] to represent a consistent planetary dataset. Planetary values are available from many sources online and in print; however, there are differences in the data that vary from source to source based on the measurements. Additional objects can be added to the database by simply including the additional parameters, no significant program modification is needed to handle the larger data array. Two versions of the database values exist within the planetary database program, one containing SI metric units, the other British Imperial units. With the planned upcoming missions to the asteroid belt and the outer planets, object data for the Saturn and Jovian moons can be added to the database. All of the data for the body is not needed to update the database. Placeholder values can be entered into the array based on analytical relationships or estimates until verified data values can be obtained.

While the main planetary data is stored in one database, the atmospheric properties are split into two different data arrays each using a different data structure. The first consists of the major atmospheric components and is stored in pairs of column elements. For each added element, two columns of data are needed to be accurately read into the program. The first location is a string value containing the element of the atmosphere (e.g., N, O, CO<sub>2</sub>) and the second is the numerical percent composition of the planet's atmosphere (e.g., Earth 'N<sub>2</sub>' 78.084). Data values for the atmospheric compositions are determined from NASA's Planetary Database and represent average values in the atmosphere. Certain planets exhibit large variations in the concentration of



compounds in the atmosphere over the course of the year, thus average values are used in the database. Additionally small portions of the atmosphere are comprised of trace elements or due to the uncertainty in the primary composition the total composition of the atmosphere may not equal 100 percent.

For each planet, there are additional trace elements present in the atmosphere, but not in sufficient amounts to be measured as a primary component in the atmosphere. Trace components are stored in a secondary data array using a trinary column format. Each trace component requires three columnar elements to describe its composition. The first column is allotted to a data string containing the name of the gas, the same as the major atmospheric component array. The second and third columns hold the number of particles and the concentration, respectively. For example, on Earth the primary trace gas is displayed as 'Ar' for argon in the first column, 9340 in the second column, and 'ppm' in the third indicating the concentration of argon gas in Earth's atmosphere to be 9340 parts per million. Concentration amounts for the trace elements for this example are expressed as a function of the number of particles in the atmosphere, but can also as easily be expressed as a number per unit volume of the atmosphere (e.g., ppccm - parts per cubic centimeter). The Planetary Database program does not actually place any restriction on the concentration unit type in the array. This allows for future corrections to the database to update the concentrations, or if the concentration is updated, to reflect a different unit basis.

For any of the atmospheric elements, either major or trace, the component is shown via its chemical formula, but could be rearranged to display the component's

chemical or common name. This was merely done to save space in the database for storage, and to simplify the program output when displaying the components. Although no fixed database size is allocated for the atmospheric properties, an additional column space has been added to the major and trace gas arrays to allow for the addition of newly determined atmospheric components and also serves as a limit in the program when displaying data to the screen or writing the results to a summary file to be loaded later in the main ISSPO program.

#### **F. Sensor Databases**

The following sections detail the sensor types contained in the ISSPO Tool. Each database was developed in a similar manner. Inputs from the main ISSPO program are loaded into each sensor database file along with any requirements the sensor must be able to meet, based on the planet the mission is going to. Data arrays contain the sensor properties in a two-dimensional array format. Sensors are listed in row format, with each column designated to a property. The number of data arrays in each sensor database varies based on the number of available properties for each sensor type. The data arrays contain both numerical values and textual properties, such as power requirements of alternating or direct current, or comments on the use of the sensor in certain environments or other restrictions. The final selected sensor properties are recorded to a MatLAB '.mat' data file and saved for use by the main ISSPO program.

### *1. AC (Magnetic) Field*

The AC\_FIELD\_SENSORS program loads information on magnetic field sensors used to map an object's magnetic field. The magnetic fields on several planets are understood, but others are too small to maintain a magnetic field. The strength of the field helps to answer questions about the nature of the planet's history and if it were ever able to support life. A planet's magnetic field helps to protect the surface from the sun's harmful radiation and can absorb and block high energy cosmic radiation. By mapping the object's magnetic field, the levels of surface radiation can be determined. Missions to the Asteroid Belt, smaller moons, and comets reveal important information about the nature of the object's core. Additionally, the shape of the field around small objects needs to be understood to safely land small probes and how the probe will react to the object's field.

Magnetic field sensor data [4, 5, 6] is recorded into several data arrays. Properties that have common units in both unit systems are recorded into a single data array using rows for each sensor, and columns for each separate property. Sensor properties that depend on the chosen unit system are loaded into separate data arrays and concatenated onto the common property arrays. A final array containing comments on each sensor adds any relevant data about the sensor into the database, which can be searched for specific keywords to aid in the selection process for a compatible sensor. A planet's magnetic field strength is tracked in the PLANETARY\_DATABASE program and the existence or strength of the field is input to the magnetic field sensor program, along with planetary atmospheric properties that aid in the selection of a sensor based on the

environment the sensor is subject to. The properties shown in Table 1 are recorded for each sensor and stored within the database.

**Table 1 AC (Magnetic) Field Sensor Data recorded in ISSPO Tool.**

<b>AC Field Sensor Database Properties</b>	
<b>Common Sensor Properties</b>	
Sensor Type	Number of Axis
Low Magnetic Field Limit	High Magnetic Field Limit
Operational Frequency Range	Accuracy
Sampling Time Interval	Sensitivity
Voltage Requirements	
<b>Unit System Specific Properties</b>	
Sensor Operational Temperature Range	
Electronics Operational Temperature Range	
Sensor Dimensions	Electronics Dimensions
Sensor Weight	Electronics Weight
Sensor Comments	

## 2. *Accelerometers*

The ACCELEROMETER\_SENSORS program loads database properties for acceleration sensors into the ISSPO Program. Accelerometers are used to map atmospheric profiles during planetary entry. Acceleration profile data can be used to determine the mass of the science mission object. There is a wide variety of available accelerometers for a number of uses. Accelerometer data is also a key parameter used to determine sequence timings during planetary entry. Events during planetary entry are triggered by accelerometer data, such as parachute deployment, and retro rocket firings to slow landers to safely arrive on the surface. Often multiple sensors are used at different

locations to obtain differential accelerations on the spacecraft to keep it stable, and act as redundant systems to offset any erroneous data recorded by a single accelerometer, when used as event triggers. Sensors [ 7, 8 ] are available with a wide range of performance, from high acceleration limits at low fidelity to small acceleration limits at high fidelity. Common properties for each sensor are recorded into a single data and appended to by data from a secondary data array holding values specific to the chosen unit system. Many of the available sensors can be ordered to meet several different acceleration ratings with different sensitivity. For these sensors the highest available rating was input into the database with a comment in the data that it is available at other calibrations. The different calibrations for a single sensor don't affect the dimensions or mass of the sensor.

The selection of an accelerometer is guided in part by the nature of the mission's objectives. Mission with low acceleration tolerances will usually require a high sensitivity. In contrast missions requiring high accelerations, a lower sensitivity is available. A required input to the ISSPO program when acceleration data is required is the type of accelerometer data required, whether for a mission with soft accelerations, high accelerations or extreme acceleration profiles like NASA's Deep Impact. The mission profile type is entered into the second column of the sensor data variable as 'SOFT', 'MEDIUM', 'HIGH', 'IMPACT', or 'BALLISTIC' in the input file. Data on the power supply type to the unit is entered into the third column. A summary of the recorded values in the database is shown in Table 2. Proper input file format for acceleration sensors is shown in Table 3.

**Table 2 Accelerometer Sensor Data recorded in ISSPO Tool.**

<b>Accelerometer Database Properties</b>	
<b>Common Sensor Properties</b>	
Sensor Type	Dynamic Range
Sensitivity	Operational Frequency Range
Linearity	Resonance Frequency
Shock Limits	Circuit Configuration
Current Range	Voltage Requirements
Electrical Signal Noise	
<b>Unit System Specific Properties</b>	
Sensor Temperature Range	Sensor Weight
Sensor Dimensions	
Temperature Sensitivity Correction	

**Table 3 Acceleration Sensor Input File Format**

1 <sup>st</sup> Column	2 <sup>nd</sup> Column	3 <sup>rd</sup> Column
'ACCELERATION'	'LOW'	'Voltage'
	'MEDIUM'	'Const Current'
	'HIGH'	'Self Generating'
	'IMPACT'	
	'BALLISTIC'	

3. *Acoustic Sensors*

The ACOUSTIC\_SENSORS program loads acoustic sensor data into the ISSPO program. Acoustic sensors [9, 10] have several different functions and can monitor several different aspects of the mission. A microphone-based acoustic sensor can be used to listen to the sounds of the mission location. They are also useful for monitoring vibration on components onboard the vehicle. Moving parts on the vehicle can send information in their vibration and be used to send back information on the health of the

part. Flow detection-type acoustic sensors can be used to monitor blockages in pipes in soil or air sampling. An acoustic sensor can also be used to measure particle velocity and sound pressure in three orthogonal directions. Recorded database properties for the acoustic sensors are shown in Table 4.

**Table 4 Acoustic Sensor Data recorded in ISSPO Tool.**

<b>Acoustic Sensor Database Properties</b>	
<b>Common Sensor Properties</b>	
Sensor Type	Dynamic Range
Frequency Range	Sensitivity
Resonance	Inherent Noise
Clipping Limit	Voltage Range
Current Range	Stability
<b>Unit System Specific Properties</b>	
Operational Temperature Range	
Storage Temperature Range	
Temperature Coefficient	Pressure Coefficient
Sensor Dimensions	Weight

Due to the wide array of different uses and configurations for acoustic sensors, there is a smaller amount of consistent property data amongst different models. Since acoustic sensors are capable of operating in different modes additional input is required to determine the specific use of the sensor in that case. The type of operation is entered into the input file in the second column position as either, 'SENSOR', 'ARRAY', or 'VELOCITY'. A sample of the input file format is shown in Table 5.

**Table 5 Acoustic Sensor Input File Format**

1 <sup>st</sup> Column	2 <sup>nd</sup> Column
'ACOUSTICS'	'SENSOR'
	'ARRAY'
	'VELOCITY'

#### 4. *Density Sensors*

The DENSITY\_SENSORS program loads and selects the correct density sensor needed to survive the environment. This module is called when the atmospheric data suite option is chosen from the sensor data array. Inputs to this program are the chosen unit system and the planetary bodies' surface atmospheric density. Using a general definition from Wikipedia [11], the densitometer is basically a light source aimed at a photoelectric cell, which determines the density of the sample from differences in the readings. Understanding of the atmospheric profile provides valuable data into seasonal fluctuations in the atmospheric properties. By understanding the seasonal fluctuations, future missions can predict how the atmospheric profile will affect the trajectory of an entry probe or lander system.

There are a variety of units with different applications [12], material, features, construction, and other specifications. Recorded data values for the sensors are specific to the chosen unit system and loaded upon program execution. Comments included with the data array provide additional information on the use and applications of the densitometer. Due to larger sizes, density sensors can tend to be heavy and consume a higher portion of a mission's science equipment budget. As an alternate solution, given the object's atmospheric composition, atmospheric density can be determined from other known planetary properties via the Ideal Gas Equation in Equation 1.

$$\rho = \frac{P}{RT} \quad (1)$$



Planetary Atmospheric properties are assigned in the PLANETARY\_DATABASE subroutine. A logic check within the program determines if an atmosphere exists, and will calculate an approximate density based on the Ideal Gas Equation and override a zero input value for the atmospheric density, if it is unknown. The database contains gas constants for different gasses, and the value used is based on the gas constant for the primary constituent in the atmosphere. This incorporates an element of error into the planetary atmosphere density calculation that varies based on the number of known constituents in the atmosphere. The planetary atmospheric density value is fed into the density sensor program to determine a sensor capable of operating in the atmosphere. Database properties for the density sensors are shown in Table 6.

**Table 6 Density Sensor Data recorded in ISSPO Tool.**

<b>Density Sensor Database Properties</b>	
<b>Unit System Specific Properties</b>	
Sensor Type	Density Range
Accuracy	Repeatability
Temperature Range	Maximum Pressure
Weight	Dimensions
Voltage Requirements	Current Requirements
Vibration Limits	Tolerance
Viscosity	Flowrates

In development of mission plane for the outer planets, there is a fair amount of uncertainty of the exact nature of the planetary surface material, whether the vehicle will land on solid ground, a frozen slush, or a vast ocean of liquid. To aid the ISSPO Tool in selection of a sensor suited to the particular application, knowledge of the operating

environment is required. Sensors are chosen based on sampling gas density, i.e., during atmospheric entry, or liquid, if the sensor is to take samples on the ground. By default the density sensor module is called as part of the “ATMOSPHERIC” sensor package suite. To modify the type of sensor selected for a particular application, the ISSPO tool needs to know which sensor package to associate the modification with. To modify density sensors, enter the term “DENSITY” into any column following the ATMOSPHERIC sensor package call. In the following column, enter the type of density sensor required, either “GAS,” or “LIQUID.” The program will err off if it detects the “DENSITY” modifier flag in the input variable array without a sensing medium type. If the “DENSITY” flag is not entered into the ATMOSPHERIC sensor package call, the program defaults to “GAS” sensing type and continues the program. A sample input format for a density sensor is shown in Table 7.

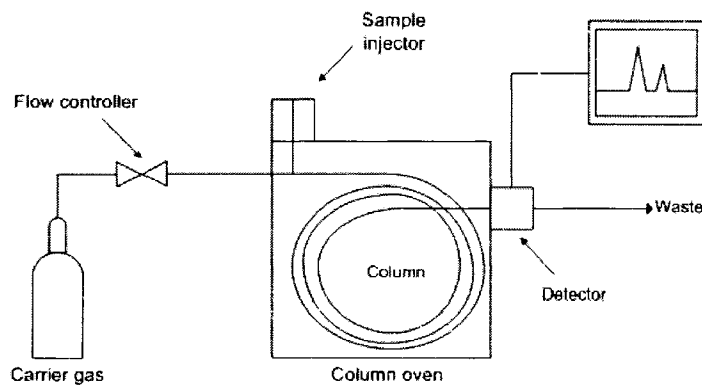
**Table 7 Density Sensor Input File Format**

1 <sup>st</sup> Column	ANY Subsequent Column	Next Column
“ATMOSPHERIC”	“DENSITY”	“GAS”
		“LIQUID”
“ATMOSPHERIC”		“GAS”

#### 5. *GCMS Sensors*

The GCMS\_SENSORS program loads Gas Chromatograph Mass Spectrometer (GCMS) sensor data into the ISSPO program. These units are typically a combination of two different sensor components: Gas Chromatograph and Mass Spectrometers. These

two components work together to perform similar tasks but operate via different methods. Both components determine the chemical makeup of a gas or solid sample by breaking up the samples into its smallest components and determining the amounts of the different chemicals to known test samples. These units are calibrated periodically to verify that consistent results are obtained. By analyzing the output of these devices the composition of the original sample can be precisely determined. In Gas Chromatography [13], an unknown gas sample is carried into a sampling chamber along with an inert carrier gas and heated. As the gas is heated the different components begin to ionize into its simplest elements. The ionized particles are tracked as they leave the sampling chamber and recorded for the amount of time within the oven and their exit temperature as these can be used to determine the composition of the sample.



**Figure 2 Gas Chromatograph Reference Diagram [13]**

Reprinted from [http://en.wikipedia.org/wiki/Gas\\_chromatography](http://en.wikipedia.org/wiki/Gas_chromatography)

In Mass Spectrometry [14], the object under analysis is often a solid sample instead of a gas. Here an ion source is used to bombard the surface of the sample and ionizes the sample. The beam of the resultant out-gassed particles is then sent through a magnetic field to deflect the particle stream. A detector at the far end measures the

amount of the beam deflection and correlates the amount of deflection with a specific element. Lighter elements experience a greater amount of deflection in the magnetic field. A charge to mass ratio is then determined by the detector and plotted to show the intensity on the detector over different charge to mass ratios. An operational diagram of a mass spectrometer is shown in Figure 3.

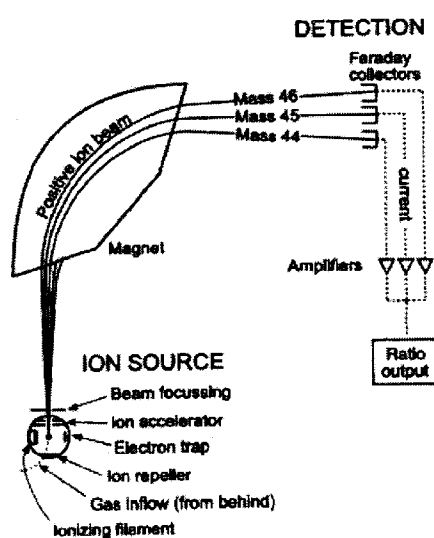


Figure 3 Mass Spectrometer Reference Diagram [14]

Reprinted from [http://en.wikipedia.org/wiki/Mass\\_spectrometry](http://en.wikipedia.org/wiki/Mass_spectrometry)

Intensity signal spikes indicate the presence of different chemical elements. For each known compound a unique series of peaks is generated. The intensity data is recorded and transmitted back to scientists in a laboratory that compare the data to known compounds to determine the sample's original components and concentration. Use of these two sensors in combination allow for the determination of both atmospheric and ground samples.

The use of a GCMS sensor system in space missions is still fairly limited, so the reliability of space rated units is fairly limited. A unit's volume, mass, and power are all at premiums on any space mission, so efforts are made to reduce these parameters when selecting components. These types of units are often mission specifically built to meet mission's requirements. A large number of different models exist, but are primarily used in laboratory setups, and thus not as restricted by mass, volume and power requirements. The data for these sensors [15, 16] stored in the ISSPO database records some of the key selections parameters for a viable unit and records additional information on the performance capabilities of the sensors in the comments section.

Two different operational types exist for these devices. To select the correct operational type within the ISSPO program additional information is required in the case input file. Their operation and specifications are tied the manner under which they report the results of the analysis. One type displays the wavelengths and intensity of the spectral emission band by filtering the samples gasses. By observing the wavelength and intensity of the samples spectral peaks, the exact chemical composition of the unknown sample can be determined by comparing the results against known gas samples. The other type uses a charge to mass ratio to determine the samples composition. To select the different types of sensor to use enter "WAVELENGTH," or "MASS-CHARGE" in the 2<sup>nd</sup> column position in the case input file. A sample of the input format is shown in Table 8. If the mass charge option is used, an additional range value of "LOW", "MEDIUM", or "HIGH" is required in the third column position in the input file.

**Table 8 GCMS Sensor Input File Format**

1 <sup>st</sup> Column	2 <sup>nd</sup> Column	3 <sup>rd</sup> Column
"GAS ANALYSIS"	"MASS-CHARGE"	"LOW"
		"MEDIUM"
		"HIGH"
"GAS ANALYSIS"	"WAVELENGTH"	

The chemical composition of the planetary atmosphere is loaded via the PLANETARY\_DATABASE program and input to the GCMS\_SENSOR program. Each component of the atmosphere emits absorption spectrum lines at distinct wavelengths. The strong signal peaks for each component of the atmosphere are loaded into a data array and used to determine the minimum and maximum wavelengths for the components in the atmosphere. These values are then used to select a sensor whose operational spectral range covers all the components in the atmosphere. Units based on detection of wavelength spectral peaks allow for the detection of complex molecules beyond simple elements, and offer the ability to determine the presence of tholins or organic components within a planetary atmosphere. Recorded database parameters are shown in Table 9.

**Table 9 GCMS - Wavelength Sensor Data recorded in ISSPO Tool.**

<b>GCMS Sensor Database Properties</b>	
<b>Unit System Specific Properties</b>	
Sensor Type	Wavelength Range
Spectral Width	Resolution
Scan Time	Detector Size
Weight	Dimensions
Voltage	Sample Size

Selection of the “MASS-CHARGE” option loads sensor properties from a different database and contain different properties. A separate set of selection criteria is used to determine the correct sensor. Use of mass charge based analyzers are best suited towards missions monitoring basic atmospheric elements, and are less well suited to analyze the high molecular weight of organic compounds in a planetary atmosphere. Database properties for these configurations are shown in Table 10.

**Table 10 GCMS – Mass-Charge Sensor Data recorded in ISSPO Tool.**

<b>GCMS Sensor Database Properties</b>	
<b>Unit System Specific Properties</b>	
Sensor Type	Mass-Charge Range
Mass Filter Material	Detector Type
Dynamic Scan Range	Resolution
Sensitivity	Minimum Partial Pressure
Number of Ion Sources	Construction Material
Filament Material	Field Range
Electron Energy	Ion Energy
Focus Voltage	Electron Current
Dimensions	Weight
Mass Stability – Mass	Mass Stability – Time
Input Voltage	Voltage Type
Current	Data Rate
Power Requirements – Min/Max/Mean	

## 6. *Humidity Sensors*

The HUMIDITY\_SENSORS program loads humidity sensor data into the ISSPO program. This module is called when the atmospheric data suite option is chosen from the sensor data array. Humidity sensors [17] track the amount of water vapor in the

planets atmosphere. Several of the other sensor types have constraints on the relative humidity that they will operate in. For most of the known objects in our solar system, the relative humidity of the atmosphere is a minor concern as only Earth has high water vapor content in the atmosphere. Since many planetary objects do not contain water vapor in any great quantity, inclusion of this sensor may not be required to obtain a better model of the planet’s atmospheric properties. However, humidity sensors are small and lightweight, so it would not use much of the mission’s mass, volume, and power budget to include a sensor in the design package. For planetary objects that do not contain any known amounts of water vapor in the atmosphere, the inclusion of humidity sensors is not required and the sensor design is overridden to zero values. Humidity sensor data arrays are created based on the chosen unit system, and the recorded properties are shown in Table 11.

**Table 11 Humidity Sensor Data recorded in ISSPO Tool.**

<b>Humidity Sensor Database Properties</b>	
<b>Unit System Specific Properties</b>	
Sensor Type	Relative Humidity Range
Sensitivity	Hysteresis
Response Time	Stability
Temperature Range	Frequency Range
Dimensions	Weight

7. *Inclinometer Sensors*

The INCLINOMETER\_SENSORS program loads inclinometer sensor data into the ISSPO program. This sensor allows for the determination of tilt angles from the



vertical or horizontal plane. Tilt sensors can provide valuable information about planetary surface terrain and orientation to lander or rover systems. Rovers need information about the local terrain conditions to determine if it is able to maneuver around steep grade terrains. Use of tilt sensors can provide local terrain information and determine if the rover has sufficient power to climb out of craters or on steep slopes traversing over hilly terrain. Sensors selection is driven in part by the application and the expected operational range the sensor is expected to encounter. Selection of a sensor for an entry probe would differ from that for a lander vehicle. During planetary entry the probe can tumble through high angles or orientations, whereas a lander vehicle will travel over fairly level terrain and is not required to be able to tilt to large angles.

The sensor data recorded [18, 19, 20] comes from many industrial use tilt sensors. There are a wide variety of configurations available with multiple options. Additional features and unit specifications are included in the database. Units are available with multi axis detection abilities, various voltage requirements, and computer data interfaces. Desired sensor ranges for the chosen sensor mission are entered as an additional input into the second column of the sensor data array in the input file as either “LOW,” “MEDIUM,” or “HIGH.” Inclinator sensor data arrays are created based on the chosen unit system, and the recorded properties are shown in Table 12. A sample of the input file format for inclinometers is shown in Table 13.

**Table 12 Inclinometer Sensor Data recorded in ISSPO Tool.**

<b>Inclinometer Sensor Database Properties</b>	
<b>Unit System Specific Properties</b>	
Sensor Type	Inclination Range
Output Voltage Range	Resolution
Sensitivity	Non-linearity
Input Voltage	Response Time
Bandwidth	Operating Temperatures
Storage Temperatures	Dimensions
Weight	

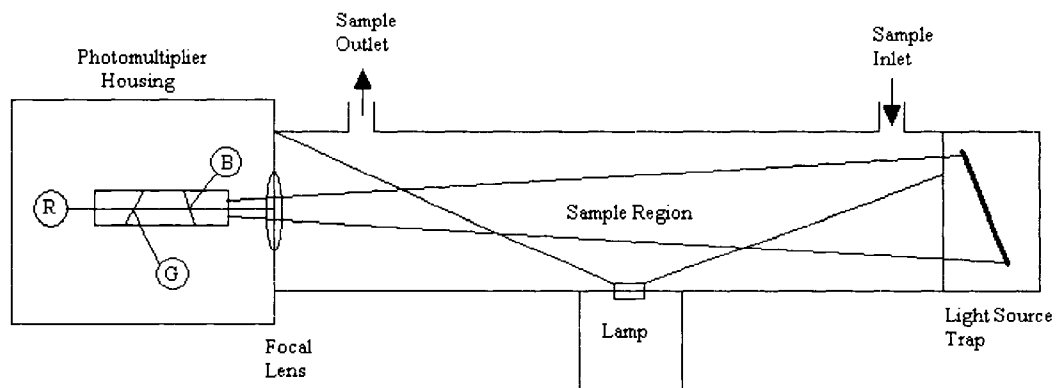
**Table 13 Inclinometer Sensor Input File Format**

1 <sup>st</sup> Column	2 <sup>nd</sup> Column
“INCLINATION”	“LOW”
	“MEDIUM”
	“HIGH”

8. *Nephelometer Sensor*

The NEPHELOMETER\_SENSORS program loads nephelometer sensor data into the ISSPO program. This sensor allows for the measurement of suspended particles in a gas. These make it possible to obtain atmospheric data such as visibility and particle size in the atmosphere. There are few commercially available nephelometers [ 21, 22]. Most space missions that would require one onboard are likely to be custom designed for that particular mission. Size and power requirements of commercially available nephelometers make them impractical for use in space exploration missions. Smaller, more power efficient sensors are producible, but are usually custom designed on an ‘as needed’ basis.

The sensor models included in the database are based on an integrating nephelometer design as shown in the simplified operating schematic in Figure 4. A sample chamber is filled with an atmospheric sample and analyzed over a time period while the particles in the chamber are counted. At the end of the integration cycle, the sample is expelled out of the chamber and is ready to begin again. While the sample is being recorded, light is used to illuminate the sample while a series of filters determines the amount of light blocked by the sample. Certain models can offer the advantage of additional atmospheric sensor data recorded during the sample. The units include pressure, temperature, and humidity sensors within the test chamber. However, these units incorporate a complex design and contain many moving parts that increase risk involved with its operation due to failure or clog. The design incorporates inlet and exit valves and an internal fan to maintain the particles suspended in the air. Additionally, the built-in pressure, temperature, humidity sensors have reduced operational ranges to those of individual sensors. The possibility exists to upgrade these internal sensors and re-qualify its capability to satisfy the mission requirements.



**Figure 4. Simplified Nephelometer Reference Diagram**

Due to the large mass, volume and power requirements of nephelometer sensors, several other design considerations must be taken into account. During the design phase, it must be determined whether there is enough budget in the design to accommodate this sensor package. For smaller spacecraft, the mission planners may have to consider the workload time for this sensor, as the electrical power system may not be able to power all the components all the time. Several other sensors may have to be shutdown for a period of time to divert enough power to the nephelometer for it to function.

Commercially available nephelometers are designed as stand-alone units, with an integrated power supply unit. These units can be removed to reduce the sensors weight volume, and power requirements. The unit can then be directly connected to the spacecraft's power distribution system. Nephelometer sensor properties tracked in the database are shown in Table 14.

**Table 14 Nephelometer Sensor Data recorded in ISSPO Tool.**

<b>Nephelometer Sensor Database Properties</b>	
<b>Nephelometer Properties</b>	
Sensor Type	Operating Wavelengths
Sensor Bandwidth	Weight
Power Requirements	Voltage
Dimensions	Integration Time
Sensor drift	Response time
Sample Flow Rates	Operating Temperature
Relative Humidity Limits	
<b>Power Supply Properties</b>	
Weight	Power Supply
Voltage	Power Supply Dimensions

## 9. *Optic Sensors*

The IMAGING\_SENSORS program loads and selects the correct optical imaging sensor data into the main ISSPO program. Optical imaging cameras are a staple component of nearly all planetary missions. The ability to record images from another planet and relay back data has truly given us insight to what occurs on other worlds. Imagers have become so crucial, that often multiple sensors are used to take a variety of different image formats. Optical imagers are not only used to relay images back to earth but also to “see” obstacles in its path and be able to determine how to maneuver around objects by analyzing the image data. Optical imagers can provide a wealth of information beyond the visible image. Given a reference object length in an image, the size of geographic formations can be determined. Observing geologic formations can provide insight into the past history of the planet and indicate how the planet environment is changing over time. Recording variations in terrain over seasonal periods provide additional valuable information on environmental cycles on a planet.

Imaging sensor units have a wide variety of capabilities and operating ranges. Use of different lens types allow for wide field images, panoramic, fish eye, magnification, etc. Commonly used imagers feature a black and white format, but the operating range can be extended to other regions of the spectrum via the use of color filters to record images in each color spectrum later be combined to create a single false color image. Use of additional filters can extend the imagers range to include the ultraviolet, infrared and x-ray regions of the electromagnetic spectrum. These extended

capabilities make use to imaging systems a highly valued component for many planetary missions.

Selection of optical system components can also have profound effects on the design of the overall system. Digital imaging systems typically use either a CCD or a CMOS sensor to detect an image. Recently imaging sensor array sizes have grown significantly and deliver several mega pixel image resolution. A tradeoff here is made between the camera resolution size and the data relay system used on the spacecraft. For a fixed rate data communications system, the designer can choose a high resolution camera that will fill up the available data bandwidth quickly, with a limited number of images. Alternatively, a low resolution array will only be able to image a small area at a time however; the system would be able to handle the bandwidth of all the small images. A similar situation arises in sizing on board memory for the mission, to store a number of high resolution images requires a large amount of memory, which must be factored into the system design and will use up portions of the spacecraft's mass, volume and power budget. Each of these factors must be taken into account in the overall system design.

A common practice in including imagers in mission payloads is the use of multiple imagers, with several different types. On board the Mars Exploration Rovers, is a suite of optical imaging camera types [23] that aide in the collection of science data and help maneuver the rovers around obstacles. Five different imaging camera systems are used for a variety of purposes. The PANCAM array consists of two panoramic cameras situated atop the mast and is capable of recording 360 degree azimuth and +/- 90 degree elevation range. A pair of NAVCAMS sits atop the mast next to the PAMCAM and

provides a wide field of view to aid in driving the rovers. A micro-imager camera is mounted at the tip of the extendible arm and allows for close up images to be taken of the geology and terrain surrounding the rover. Hazard cameras are mounted on the front and rear of the rover under the solar panels with a view looking at the front and rear wheels. These lenses aid in steering the rovers around small obstacles near the wheels. A SUNCAM CCD array is used to provide an accurate orientation measurement of the rover's position on Mars relative to Earth to indicate the direction to aim the high gain antennae to obtain the best signal to Earth. Most of these cameras are custom built units that were required to operate in a harsh environment. Several sensors on the exterior surface are comprised solely of a CCD imaging array and a optical lens to resolve the image. Each of these sensors provides additional functions to the rover and was selected carefully to meet the requirements of its function.

Selection of the appropriate CCD array to capture the image is only a part of sensor. The array must be coupled to an appropriate camera lens to obtain the proper image type. Data on sensors [ 24, 25, 26] entered into the ISSPO database is based on a CCD array, or complete system with array and camera lens. Selection of the optical sensor option in the input file requires three additional columns of input in the same sensor row. The second column represents the type of system desired, and is entered either as "CAMERA," or "ARRAY," or "LINEAR." The third column corresponds to the desired resolution of the sensor as either "LOW," "MEDIUM," or "HIGH." Additionally required by the program is the desired type of image to be taken. There are many different types of image range available that extend beyond the visual spectrum.

Additional input in the fourth column is the desired type of image to be returned. Valid inputs to the optical sensor program are “X-RAY,” “VISUAL,” “UV,” “NIR,” and “MICRO.” These terms correspond to the x-ray, visual, ultraviolet, near-infrared, and microscopic images. The desired image type for the optical sensor can allow for multiple image type sensor configurations. Enter the required image type into additional columns on the optical sensor row within the sensor data variable in the input file to configure the correct sensor. Since most of the sensor elements are common between the use of a CCD array and a complete camera system, a single database exists within the program. Optical imaging sensor properties tracked in the database are shown in Table 15. A sample of the input format required for the optical sensor program is shown in Table 16.

**Table 15 Optical Imaging Sensor Data recorded in ISSPO Tool.**

<b>Optical Imaging Sensor Database Properties</b>	
Sensor Type	Optical Range Type
CCD Array Dimensions	Pixel Size
Image Area	Signal Read Noise
Full Well Capacity	Gain
Linearity	ADC Dynamic Range
Readout Rates	Readout Time
Frame Rate	Dimensions
Operating Temperature	Cooling Temperature
Cooling Method	ADC Dynamic Range
Weight	

**Table 16 Optical Sensor Input File Format**

1 <sup>st</sup> Column	2 <sup>nd</sup> Column	3 <sup>rd</sup> Column	>4 <sup>th</sup> Column
“OPTICS”	“CAMERA”	“LOW”	“X-RAY”
	“ARRAY”	“MEDIUM”	“VISUAL”
		“HIGH”	“UV”
			“NIR”
			“MICRO”



## *10. Pressure Sensors*

The PRESSURE\_SENSORS program loads and selects the correct pressure sensor needed to survive the environment. This module is called when the atmospheric data suite option is chosen from the sensor data array. Inputs to this program are the chosen unit system and the planetary body's surface pressure. There exists a myriad of pressure sensors with different calibrations, functions and measurement types [27, 28, 29, 30, 31]. The selection of hardware is highly driven by the application. Pressure sensors can operate in several different types – vacuum, gage, or differential. They can be used to monitor atmospheric pressure or used as a trigger switch to activate a system once a pressure limit is reached. Several other driving factors in the selection of components include the operational and storage temperatures, maintenance of any moving parts or routine oiling, and analog or digital signal outputs. Operational properties for the pressure sensor database are compiled from several different suppliers and incorporate a spectrum of different sensor technologies.

Sensors are chosen to represent a wide variety of configurations, instead of all calibrations of a single sensor range to generate a broad database. Data arrays are generated for each unit system and include a comment array containing additional information on use, limitations, maintenance, etc. for each sensor. Several pressure sensors have multi-function capability with built in temperature sensors used in special applications. Models with this capability include the thermocouple temperature limits in the database. Operational performance range for the optimal pressure sensor is based on the known planetary surface pressure defined within the planetary database program.

Additional information on the planetary environment is used to reduce the number of sensors available in the database to those meeting the environmental specifications. Recorded data properties for each type of pressure sensor are shown in Table 17.

**Table 17 Pressure Sensor Data recorded in ISSPO Tool.**

<b>Pressure Sensor Database Properties</b>	
Sensor Type	Accuracy
Stability	Over Pressure Rating
Burst Pressure Rating	Power Requirements
Number of Life Cycles	Pressure Range
Shock Limits	Impulse Time
Operating Frequency Range	Humidity Range
Operating Temperature	Storage Temperature
Voltage Requirements	Current Requirements
Sensor Material Construction	Dimensions
Weight	

### *11. Radiation Sensors*

The RADIATION\_SENSORS program loads and selects the correct radiation sensor into the main ISSPO program. This module is called when information on the planets radiation exposure is desired. Radiation effects pose several significant hazards to science missions; first and foremost, it's deleterious effect on biological organisms, and second, its effects on science hardware. Large portions of the suns incident radiations is absorbed and blocked by the earth's magnetic field and ozone layer. This effect protects us from the sun's output of radiation. On other planets without a planetary magnetic field the surface environment is subject to the full effect of the suns radiation and any incident cosmic radiation. Certain types of radiation can be effectively blocked

by various mediums. Mapping of a planet's radiation exposure will help determine methods to safely and effectively block harmful radiation, and allow for safe human exploration. Radiation has also negatively affected flight hardware. Stray high energy particles can interfere with electronic systems on spacecraft, and without control to handle these cases, can render the equipment unusable. During the Gravity Probe B science experiment to prove Einstein's theory of curved space-time, [32] radiation events caused by high energy photons from solar coronal ejections interfered with the delicate electronics onboard the satellite. This resulted in computer "safemode" holds to the mission timeline, to clear out any errors in the flight computer. Safely handling radiation events, and understanding the occurrence of radiation events can allow for expanded science opportunities.

Radiation sensors [33, 34, 35, 36, 37, 38, 39, 40] operate over several different ranges and feature the ability to track different event types. Detectors can track and record single high energy protons and electrons, as well as larger alpha and beta particle radiation forms. Sensors also detect high energy photons existing as gamma and x-ray radiation which is lethal to humans in high doses, or over prolonged periods of exposure. A preliminary knowledge of the expected forms of radiation that will be encountered is required to determine the correct type of detector required for the mission. Radiation sensors can be selected by specific types or general detectors can be used to measure the energy levels of the different particles and the type of particle can be determined later. Selection of the radiation sensor option within the sensor data variable array requires additional program inputs. The different radiation types are entered into the second

column location of the radiation sensor row. Allowable types include “Charged Particle,” “Alpha,” “Beta,” “Gamma,” and “X-Ray.” Multiple types of radiation can be entered into the sensor array with each additional radiation type being entered into the next column location. A sample input file format for radiation sensors is shown in Table 18. Recorded data properties in the radiation sensor program for each type of sensor are shown in Table 19.

**Table 18 Radiation Sensor Input File Format**

1 <sup>st</sup> Column	>2 <sup>nd</sup> Column
“RADIATION”	“Charged Particle”
	“Alpha”
	“Beta”
	“Gamma”
	“X-Ray”

**Table 19 Radiation Sensor Data recorded in ISSPO Tool.**

<b>Radiation Sensor Database Properties</b>	
Sensor Type	
Proton Energy Range	Number of Proton Detectors
Electron Energy Range	Number of Electron Detectors
Alpha Particle Energy Range	Number of Alpha Particle Detectors
Beta Particle Energy Range	Number of Beta Particle Detectors
Gamma Particle Energy Range	Number of Gamma Particle Detectors
X-Ray Energy Range	Number of X-Ray Detectors
Particle Count Rate	G-Factor
Weight	Dimensions
Power Requirements	

## *12. Refraction Sensors*

The REFRACTION\_SENSORS program loads and selects the correct refraction sensor into the main ISSPO program. This module is called when information on the nature of a planetary body's sample refractive index is being determined. Refractometers measure the optical refraction properties of a sample and can determine an unknown samples composition, the purity of a sample, or the concentration of different substances. Samples can be either liquids or gasses. Selection of the sensor will depend on the nature of the data to be returned and known information about the location where the sensor will be operating.

Refractometers operate through the use of a light source and a prism. Prisms can be made of a number of optically pure materials. Photodetectors within the sensor record a "sensed" light level. Given the known refractive properties of the prism the composition of the original sample material can be determined, from the observed optical properties. The refractive index measured, is a measure of angle of light deflection as it passes through the prism, and the detectors register the transition between the dark / light line.

Selection of refractometer sensors is based on the environmental conditions the sensor must be able to survive during the operation mission phase. Database [41, 42] elements consist of industrial sensors and flight models representing commercially available optical designs used to determine concentration or purity, of gasses, liquids, or trans-lucent solids. Database properties tracked in the sensor module are shown in Table 20.

**Table 20 Refraction Sensor Data recorded in ISSPO Tool.**

<b>Refraction Sensor Database Properties</b>	
Sensor Type	Refraction Index Range
Resolution	Sensor Drift
Accuracy	Weight
Voltage	Current
Power	Dimensions
Operating Temperatures	

*13. Temperature Sensors*

The TEMP\_SENSORS program loads and selects the correct temperature sensor needed to survive the environment. This module is called when the atmospheric data suite option is chosen from the sensor data array. Inputs to this program are the chosen unit system, the planetary body’s surface temperature, and the detection type. Temperature sensors operate primarily over two different methods, either a voltage change or a resistance measurement over the operational temperature range. The operational sensor type is fed to the TEMP\_SENSORS program. Inputs for the sensor type are “VOLTAGE” or “RESISTANCE”. The ISSPO program defaults to use “VOLTAGE” if the temperature term and a sensor type are not found in the SENSOR\_DATA variable in the same row as the atmosphere option. A sample input for the temperature sensor option is shown in Table 21.

**Table 21 Temperature Sensor Input File Format**

1 <sup>st</sup> Column	ANY Subsequent Column	Next Column
“ATMOSPHERIC”	“TEMPERATURE”	“VOLTAGE”
		“RESIATANCE”
“ATMOSPHERIC”		“VOLTAGE”

Standardized sets of temperature sensors exist for most applications. Thermal performance values for different sensor types are consistent amongst different vendors, with minor variations in values at the extreme thermal limits. Performance specifications for thermocouples are regulated by several national standard organizations and, while small differences are observed in data between suppliers, the performance of each type of thermal sensor is assumed to be consistent. The temperature sensor database for voltage based thermocouples, consists of currently available calibrations from Omega sensors [43], and contains a wide variety of data on each thermocouple type that can be used to determine the appropriate temperature sensor. Combinations of Thermocouple material and Insulation material must be chosen to survive the planet’s surface temperature and the effects of the atmosphere. Selection of thermal sensors is based on the operational range of each type and the chosen planetary body’s known surface temperature. Additional information on the sensor entered in the comments section provides additional selection criteria for sensor based on popular calibration ranges or calibration for cryogenic temperature ranges. Insulation material for the thermocouple wires is given a rating for its resistance to different environmental factors. An aggregate score is determined for each material and used to determine the appropriate insulation material.

Resistance type thermocouple sensors [44, 45, 46] track the same thermocouple properties, however, they are bare wire elements and do not use insulating material on the sensor leads. Sensor properties recorded in the database are shown in Table 22.

**Table 22 Temperature Sensor Data recorded in ISSPO Tool.**

<b>Temperature Sensor Database Properties</b>	
<b>Thermocouple Properties</b>	
Sensor Type	Cold Temperature Limit
Hot Temperature Limit	Error Tolerance Percent (%)
Error Tolerance Temperature (deg)	Positive Lead Material
Negative Lead Material	Minimum EMF Voltage
Maximum EMF Voltage	Usage/Restriction Comments
<b>Insulator Properties</b>	
Insulator Name	Cold Limit
Hot Limit	Abrasion Resistance
Flexibility	Water Submersion
Resistance to Solvents	Resistance to Acids
Resistance to Bases	Resistance to Flame
Resistance to Humidity	Material Construction: Overall
Material Construction: Conductors	

#### *14. Wind Velocity*

The WIND\_VELOCITY\_SENSORS program loads and selects the correct wind velocity sensor needed to record air speed in the object's environment. This module is called when the atmospheric data suite option is chosen from the sensor data array. Understanding of the atmospheric winds adds a key element to the understanding of the nature of the environment on planetary bodies. Atmospheric wind speeds are a key parameter in landing and launching probes to other planets. To ensure safe launch



conditions, limits are placed on wind speeds to ensure that a launch vehicle can safely ascend through the atmosphere without being pushed off course by high winds. Wind speed measurements also add key data to understanding seasonal variations on other planets. Selection of wind velocity measurement is based on the mission operation.

Sensors within the ISSPO database will function for two different operating regimes. For planetary surface measurements for a lander or rover based mission, an anemometer type measurement may present a viable option. For atmospheric wind profile mapping during an atmospheric descent, or any fast moving observational platform, a Doppler based system would present a better solution. However, this also places a mission requirement that some sort of orbital platform is available. Doppler based systems communicate between a sensor located on the entry vehicle and one attached to an orbiting satellite. A series of frequency pulses between the two components can be used to determine the speed of winds between the two elements as the signals will be Doppler shifted slightly by the wind movement. The operating type of the wind velocity sensor is entered into the SENSOR\_DATA variable in the input file. Valid sensor configurations within the ISSPO tool are “ANEMOMETER” or “DOPPLER.” If the wind velocity term is not defined within any columns in the atmospheric sensor option in the input file, ISSPO will default to “ANEMOMETER.” A sample input file format for wind velocity sensors is shown in Table 23.

**Table 23 Wind Velocity Sensor Input File Format**

1 <sup>st</sup> Column	ANY Subsequent Column	Next Column
"ATMOSPHERIC"	"WIND VELOCITY"	"ANEMOMETER"
		"DOPPLER"
"ATMOSPHERIC"		"ANEMOMETER"

Recent mission data from the Mars Exploration Rovers (MER), Spirit and Opportunity, have shown the influence that the atmosphere can have on the success of the mission. Originally slated for 90 days, the two rovers have far surpassed their expected operating time. This was due to mission planners accounting for the deposition of dust on the rovers' solar panels obscuring the sunlight and reducing the amount of power generated. However, seasonal spring winds [47, 48] cleaned some of the dust off the panels and allowed additional power to be generated. During Earth's summer of 2007, a global dust storm [49] encompassed a large portion of the Martian sky, effectively blocking the rovers' ability to collect sunlight and generate electrical power to keep the electronics warm and communicate with Earth. Improved understanding of how these dust storms evolved will allow future missions to better predict the formation of these storms and successfully maintain power in the vehicles.

Separate database arrays [50, 51] for anemometer type sensors are created for each unit system and loaded into the main program via the initial function call. Different models of sensors include different recording capabilities for minimum, maximum, and average values over a sampling time interval. The sensors also offer different output formats via digital or analog signals and offer the ability to transfer data to desktop

computers for analysis. Selection of the wind velocity sensor is based on data from the planetary database for the desired body. The selection procedure is bypassed and zeroed out if the planetary object under analysis is devoid of an atmosphere. Sensor properties recorded in the database are shown in Table 24.

**Table 24 Wind Velocity Anemometer Sensor Data recorded in ISSPO Tool.**

<b>Wind Velocity Sensor Database Properties</b>	
Sensor Type	Wind Velocity
Accuracy	Volumetric Flow Rate Ability
Probe Temperature Range	Instrument Temperature Range
Power Requirements	Weight
Electronics Dimensions	Probe Dimensions

Doppler based velocity sensors operate under a set of drastically different specifications. Properties [52, 53] for these sensors are stored in separate arrays based on the chosen unit system. Sensor properties recorded in the database are shown in Table 25.

**Table 25 Wind Velocity Doppler Sensor Data recorded in ISSPO Tool.**

<b>Wind Velocity Sensor Database Properties</b>	
Sensor Type	Output Frequency
Output Level	Aging Data
Spectral Purity – Harmonics/Non-Harmonics	
Accuracy	Stability
Warm up Time	System Voltage
Operating Temperature Range	Storage Temperature Range
Temperature Sensitivity	Orientation Sensitivity
Pressure Sensitivity	Power Requirements
Weight	Dimensions

## 15. *Digital Signal Processing*

The DIGITAL\_SIGNAL\_PROCESSING module is executed when the “DATA PROCESSING” option is entered in the SENSOR\_DATA array variable in the case input file. This device does not constitute a sensor system that records any sort of data. Instead its function is to interpret and manipulate telemetry data received from other onboard sensors and process the data to determine the next set of actions to be taken. This component is crucial when a mission requires for example, altimetry data to determine when to execute mission events during a mission timeline. The altimetry data must be processed within a very short period of time to determine when events must be performed, such as collecting air samples at specific altitudes, or determining when to activate other components in the vehicle.

Digital signal processing devices (DSP) [54, 55] are essentially microprocessor computers that are highly specialized to their application. These devices are built with special handling capability for data processing and to handle large amounts of processing requests from in-flight hardware. Selection of a DSP component requires some additional about the nature of the mission and the amount and speed of data it needs to be able to process. Specialized versions of DSP’s are tuned to perform specific calculation types with very fast processing times. The number of components connected to the device is also a crucial selection factor as a single input source may not overload the processor for speed, but a high number of inputs may overload the sensors processing capacity / bandwidth and either bog down the processor to a point where it cannot

process the data quickly enough to meet mission time events, or signals may get crossed between different computational cycles and possible data corruption could result.

With information on the nature and amount of data the processor will be receiving the selection of correct DSP unit can be performed. Selection of the “DATA ANALYSIS” option within the SENSOR\_DATA variable in the input file requires two additional program inputs to aid in sizing the correct sensor. The second column positions relates to the desired operational computational speed of the processor. Acceptable input values for this column are “LOW,” “MEDIUM,” and “HIGH.” The third column position requires a numeric value for the number of devices connected to the processor. A sample input for the digital signal processing option is shown in Table 26. Component properties tracked within the ISSPO database are shown in Table 27.

**Table 26 Digital Signal Processing Input File Format**

1 <sup>st</sup> Column	2 <sup>nd</sup> Column	3 <sup>rd</sup> Column
“DATA ANALYSIS”	“LOW”	Number of devices
	“MEDIUM”	
	“HIGH”	

**Table 27 Digital Signal Processing Data recorded in ISSPO Tool.**

<b>Digital Signal Processor Database Properties</b>	
Sensor Type	CPU Clock Speed
CPU Cores	Data Memory
Program Memory	Computational Units
Cycle Time	External Interrupts
System Voltage	Operating Temperature Range
Storage Temperature Range	CPU Power Requirement
Weight	Voltage
Current	Dimensions

## **G. Error Program**

The ERROR\_PRG module can be executed from any point within the ISSPO Program. It is designed to handle any program errors, which may occur, in a clean manner. When an error is encountered within the program, the error program is called with an error number. Each error number is unique and corresponds to a single program fault source. This format is used to simplify the cause of the error when executing the program, instead of simply terminating the program. Error codes are stored as integer numbers and correspond to a single action in the error program.

When an error is recorded anywhere in the program, the error program is executed and sent an error code. The error program then prints a message to the MatLAB command screen with the error number that occurred, a description of the error or possible cause, and a possible solution to resolve the error. Depending on the type of error that occurred, the program may simply terminate, or an alternate solution displayed. Program error codes are organized into several different categories, based on the nature of the error that occurred: a program structure error, invalid variable inputs entered into the program, sensor program errors, files not created, etc. Occasionally an unexpected error may be encountered that will trigger a MatLAB internal compiler error. This is usually the result of a syntax error within the code or illegal variable definition, for example, defining the SENSOR\_DATA variable as a single value in the first row and defining a sensor in the second row that requires additional information. A detailed list of program errors and possible solutions is included in Appendix B.

## **H. Mass Properties**

The MASS\_PROPERTIES subroutine is called as part of the ISSPO program upon completion of all of all the individual sensor cases. Its function is to collect and maintain information on the mass properties and physical dimensions of the final sensor chosen for each sensor type entered in the input file. At the end of each sensor file, a MatLAB “.mat” binary file is written containing the final sensor data array and unit labels. In the mass properties program the sensor data file is loaded for each sensor in the input file sensor data array, and returns a program errors if the file does not load properly. Within the program, sensor mass and dimensions are extracted from each sensor data array and compiled into a single mass property array. Mass properties and dimensions for the different sensor types are defined within each sensor database file. If the values are unknown a default zero (0) value is stored in the database location. Sensor data is also converted into the chosen unit system base units. If the UNITS variable is set to “SI” this calculates the sensor masses in kilograms (kg) and dimensions in meters (m). If UNITS is equal to “British” sensor masses are set to pounds mass (lbm) and dimensions are set to feet (ft). These units are used to calculate the total sensor package limits and displays the results in units equivalent to the units set for the MASS\_LIMIT variable in the user input file. Additionally the sensors dimensions are stored and converted to sensor volumes in the main ISSPO program.

## **I. Power Properties**

The POWER\_PROPERTIES subroutine is called as part of the ISSPO program upon completion of the MASS\_PROPERTIES program. It functions to collect and maintain information on the power requirements of the final sensor chosen for each sensor type entered in the input file. A MatLAB “.mat” binary file is written containing the final sensor data array and unit labels. In the power properties program, sensor data files are loaded for each sensor in the input file sensor data array. Within the program, sensor power requirements are extracted from the sensor data array. Power consumption data for all different sensor types is not always consistent or available. When available, for each product, power consumption requirements are recorded. When both voltage and current data are available the system power loads are determined from basic relations. In determining power consumption, the largest power requirement for each sensor is used. This represents a worst case design in the amount of power needed to operate the equipment. If neither voltage, nor current data, is available for a sensor the resulting power consumption is set to zero (0) within the POWER\_PROPERTIES program. A check within the main program determines if the total power requirements meets the sensor payload power limit entered into the input file. If the total power limit of the system is exceeded, it does not mean that the configuration is invalid, merely that an insufficient power amount is available to operate all the sensors simultaneously. In this case a power load budget must be established to determine when during the mission is data needed from a specific sensor.



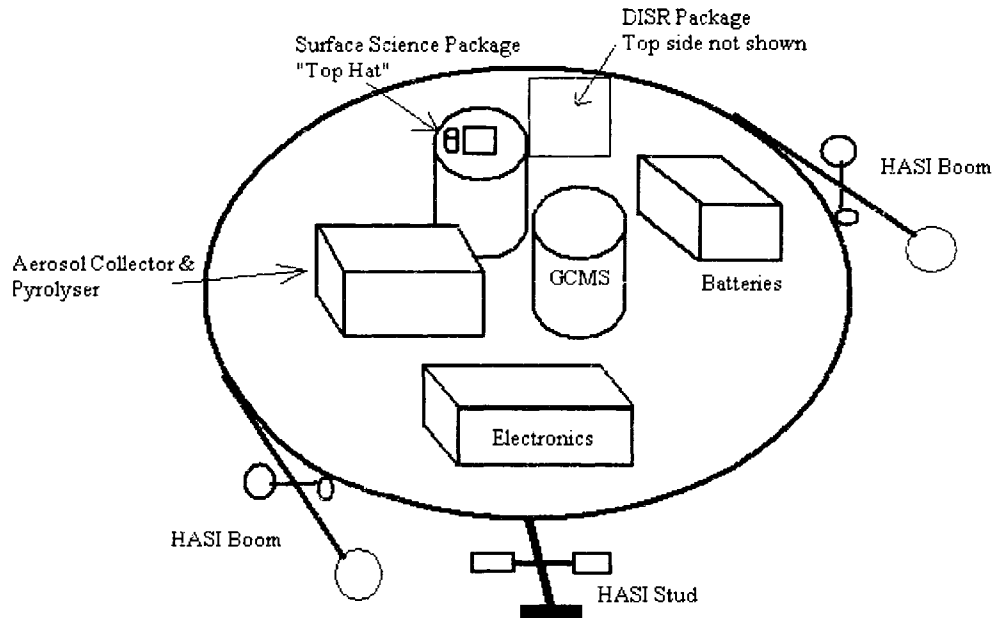
The mission design will have to allow for a contingency to operate a single high power sensor at the expense of several low power sensors, not collecting active data continuously. Sensor power data is converted into the correct unit system. In “SI” UNITS the variable is set to calculate power usage in watts (W). If UNITS is equal to “British” sensor power requirements are set to British Thermal Units per hour (BTU’s/hr). These units are used to calculate the total sensor package power limits and displays the results in units’ equivalent to the units set for the POWER\_LIMIT variable in the user input file. Total system power consumption levels are returned to the main ISSPO program and compared to the input system to determine if system constraints are met.

#### **IV. Huygens Probe Benchmark Case**

As a benchmark case to validate the methodology developed within the ISSPO program, the interplanetary probe mission to Saturn’s Titan moon carried out by the Huygens probe was used. This spacecraft consisted of several complex sensor packages with independent mission objectives. [56, 57, 58] Data on the spacecraft systems and top level packages is obtained from various references to determine which of the hardware components were based on available components, and which were custom designs built to suit a specific science goal for this mission.

Custom designed sensor configurations will always be required when the mission science question is truly unique or the level of fidelity of a commercially developed sensor is not readily available. In this case an available sensor may only need slight modification by the developer to meet the mission objective. The base model design can act as a reference point that can be modified to handle mission requirements that it was not originally designed to meet. In evaluating the component designs for the different sensor packages on the Huygens Lander, there is a combination of commercially available sensors, modified versions, and custom functioning units used.

Each subsystem of the Huygens Planetary Probe was analyzed separately as stand alone design system. Six primary science packages comprised the science payload for the entire mission. The requirements for each package will be described in turn and evaluated using the mass and power limits for each subsystem. A comparison of the results to the published data available will be made. A schematic layout of the six science payloads on the inverted Huygens probe is shown in Figure 5 looking at the bottom of the probe. The Descent Imager Spectral Radiometer (DISR) and Doppler Wind Experiment (DWE) are mounted on the top surface of the spacecraft and not shown in the figure. Despite the large number of instruments contained in the science payload, it represented only a small portion of the total spacecraft mass and volume. The mission profile was for a short lifetime to collect and relay the data thus a number of components has redundant systems or multiple ways to obtain the different data in case a system failed.



**Figure 5 Schematic Diagram of Huygens Probe Instrument Layout**

### **A. Aerosol Collector and Pyrolyser (ACP)**

The Aerosol Collector and Pyrolyser (ACP) Package is one of the science measurements chartered with mapping the properties of Titan's Atmosphere. It samples the atmosphere twice during the entry and descent phase of the mission, in the tropopause (160 – 40 km), and in the cloud layer (23 – 17 km) and prepares the samples for analysis by other components on the probe. Samples are passed to the Gas Chromatograph Mass Spectrometer for analysis. The chemical composition of the samples is analyzed looking for specific combinations of aerosols in the atmosphere. A summary of the ACP sensor system is shown in Table 28.

**Table 28 Mission Properties of the Aerosol Collector and Pyrolyser Sensor**

<b>Aerosol Collector and Pyrolyser Properties</b>	
<b>Scientific Objectives:</b>	
Chemical composition of photochemical Aerosols – Hydrogen (H), carbon (C), nitrogen (N) and oxygen(O)	
Relative concentrations of the organic condensates inside the lower stratosphere (C2, H2, C2, H, HC3N, HCN)	
Relative concentrations of the organic condensates within the troposphere (mainly CH4, C2H6)	
Non condensable constituents trapped in the collected particles (CO2)	
<b>Instrument Characteristics</b>	
Sampling of the particles (direct impact plus capture by filtration)	
Transfer of the evaporates and pyrolysis products to the GCMS (via the special ACP inlet)	
Analysis (direct MS and GCMS)	
Designed to operate with precise timing	
<b>Payload Properties:</b>	
Mass = 6.18 kg	Power (typical/peak) = 3 / 85 W
Energy = 78 Wh	Data Rate = 128 bits/sec

The ACP on its own does not make any direct samples of the atmosphere. Instead, it prepares samples of the atmosphere for the GCMS. The design of the ACP was a collaboration of multiple research institutes and industry. Use of this type of device is not common in planetary probe missions and no broad database of similar devices exists that are commercially available. Calibration of this package type was excluded from the rest of the sensors built into the model.

**B. Descent Imager / Spectral Radiometer (DISR)**

The Descent Imager / Spectral Radiometer (DISR) sensor suite onboard the Huygens probe records images as the probe enters the Titan atmosphere. It is responsible

for recording both descent imagery to monitor the surrounding terrain during the descent phase to provide background context, and upward looking data to observe the optical properties of the atmosphere. The Huygens probe spun as it entered Titan's atmosphere. With the DISR mounted on the side of the probe, the rotation allowed the probe to record smaller strip and area images that were later compiled into a 360 degree image of the terrain under the craft during descent. As the probe descended towards the surface, the imaging pattern of the spacecraft provided continuously updated terrain images. Upon arrival on the surface, the DISR is also responsible for recording images of the planetary surface and visually determine the landing location of the probe. An overview of the DISR instrument properties and objectives is included in Table 29.

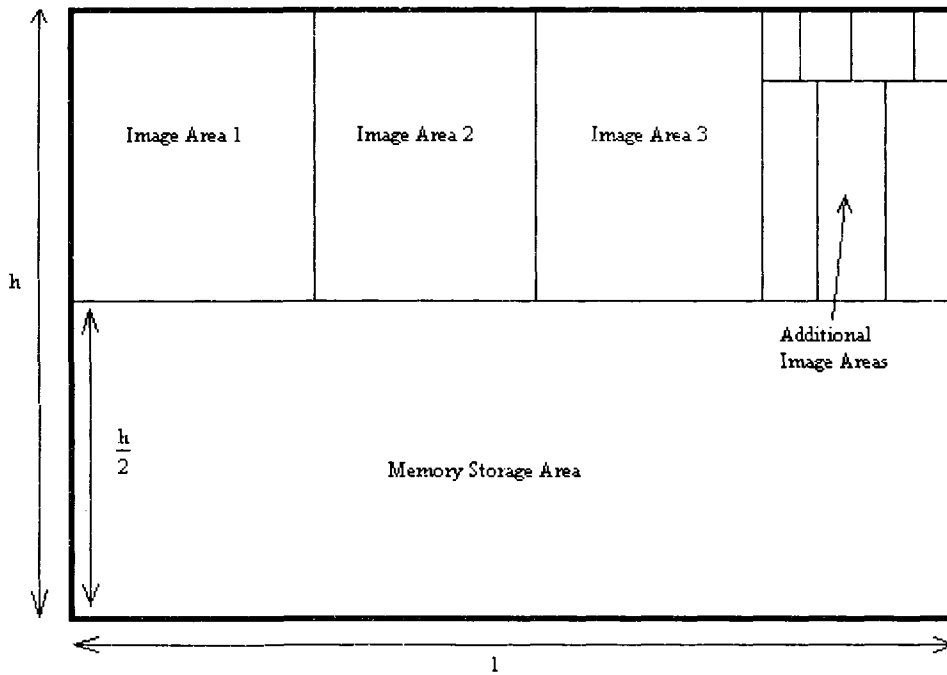
**Table 29 Mission Properties of the Descent Imager / Spectral Radiometer**

<b>Descent Imager / Spectral Radiometer Properties</b>	
<b>Instrument Overview:</b>	
CCD Detector Array	Two Linear IR detector Arrays
Surface Science Lamp	Sun Sensor
In-flight Calibration System	Hardware and software data compression systems
Flexible data collection software	Three frame imagers
Upward and downward-looking IR spectrometer	Upward and downward-looking visible spectrometer
Upward and downward-looking violet photometer	4-channel solar aureole camera
<b>Payload Properties:</b>	
Mass = 8.07 kg	Power (typical/peak) = 13 / 70 W
Energy = 42 Wh	Data Rate = 4800 bits/sec

Assembling each of these devices as a complete camera package the payload weight budget would have been exceeded. Instead to meet all the optical observation

requirements commercial grade CCD detectors were coupled to multiple camera lenses, each built to meet a specific observation requirement. This configuration allowed for a higher instrument density focusing multiple devices onto a single CCD array. The Side Looking Imager (SLI), Medium Resolution Imager (MRI), and High Resolution Imager (HRI) are connected by fiber optic cable to focus on different sections of the CCD array. This allows it to simultaneously record data from multiple imagers into a single unit. The custom configuration of the lenses does not represent a commercial viable solution for optical data recording. In this case the ISSPO tool was tasked to select a CCD based on the required types and amount of data to be recorded by the probe during the mission timeframe.

The CCD array used on the Huygens Probe is a modified version of a commercially available CCD array rated for aerospace applications developed by Fairchild Imaging. [60] The flight unit carried aboard the Huygens probe utilized a 512 x 256 pixel frame transfer CCD. The flight unit chosen supports optical binning modes in which areas of the CCD array can be designated to separate images. The use of this option allowed for greatly reduced sensor weights instead of carrying a separate camera system for each image type required. An example of the use of software binning and multiple imaging areas on a CCD array is shown in Figure 6.



**Figure 6 Schematic Diagram of CCD Array Image Layout and Binning**

Evaluating the ISSPO tool for the CCD array used in the DISR sensor package the following input data in Table 30 was used to determine the correct sensor.

**Table 30 ISSPO - Huygens DISR Input Data**

1 <sup>st</sup> Column	2 <sup>nd</sup> Column	3 <sup>rd</sup> Column	>4 <sup>th</sup> Column
“OPTICS”	“ARRAY”	“LOW”	“VISUAL”
			“UV”
			“NIR”

Based on these criteria for a CCD array design, the ISSPO tool determined a similar CCD array configuration to the design flown on the Huygens Probe. A condensed summary of the sensor data is shown in Table 31. Data on the specific CCD array flown on the Huygens probe was not available for a direct comparison of the CCD

design. The selected design has a larger CCD area, and offers the same frame transfer capability of the model flown on Huygens. The model 424 CCD array is a flight proven model developed for NASA's Deep Impact Mission. [61]

**Table 31 ISSPO System Architecture DISR CCD Array**

Sensor Type	CCD 424
Imaging Spectrum:	X-RAY UV VISUAL NIR MICRO
Array Dimensions - Length:	1024 # Pixels
Width:	1024 # Pixels
Sensor Pixel Size:	21.0000 micro-m
Imaging Array Size - Length:	21.500 mm
Width:	21.500 mm
Dimensions - Length:	73.15 mm
Width:	52.83 mm
Height:	6.10 mm

**C. Doppler Wind Experiment (DWE)**

The Doppler Wind Experiment (DWE) carried aboard the Huygens probe is used to determine wind speeds in the atmosphere as the probe descended towards Titan's Surface. The DWE consists of two Ultra-Stable Oscillators, with the properties shown in Table 32, one carried aboard the Huygens Probe, while the other remains aboard the Cassini Orbiter. Measured wind velocity is back calculated from Doppler shifts in the transmitted radio signal frequencies between the two receiver transmitters. The experiment had several objectives.



**Table 32 Huygens DWE Sensor Properties**

<b>Doppler Wind Experiment Properties</b>		
<b>Scientific Objectives:</b>		
Determine the height profile of Titan zonal wind velocity over the altitude range from 0 – 160 km with an accuracy of ~1.m/s		
Measure Doppler fluctuations to determine the level and spectral index of turbulence and possible wave activity in Titan's atmosphere		
Measure Doppler and signal level modulation to monitor Probe Descent Dynamics, including its rotation rate and phase, parachute swing and post-impact status		
<b>Physical Properties:</b>		
Mass (g)	1898	radiation shielding: 150 g
Dimensions (mm)	170 × 117 × 119	L × W × H
<b>DC power:</b>		
Warm-up power (W)	< 18.4	< 30 min
DC consumption (mA)	< 675	System limit: 0.7 A
Energy (Wh)		worst case (minimum temp)
<b>Frequency Parameters:</b>		
Output frequency (MHz)	10	± 0.1 Hz
Frequency long term drift	$1.4 \times 10^{-9}$	$\delta f_0/f_0$ over 3 hours
Allen Variation	$3 \times 10^{-11}$	$\tau = 1$ s
	$6 \times 10^{-12}$	$\tau = 10$ s

The input file setup used by the ISSPO program to determine the properties of the sensor to use for the DWE sensor package is shown in Table 33.

**Table 33 ISSPO Huygens DWE Sensor Input File**

1 <sup>st</sup> Column	2 <sup>nd</sup> Column	3 <sup>rd</sup> Column
“ATMOSPHERIC”	“WIND VELOCITY”	“DOPPLER”

The use of Doppler based velocity detections [62] involves tracking shifts in radio frequencies between two receivers. To determine the velocity to first order accuracy for a Doppler shift the primary governing equations are:

$$\Delta f = -\frac{f}{c} \Delta V \quad (2)$$

$$\Delta V = (\vec{V}_p - \vec{V}_o) \cdot \vec{R}_{op} \quad (3)$$

where p and o are the probe and orbiter elements, respectively. And  $\vec{R}_{op}$  is a unit vector path from the orbiter to the probe defined as:

$$\vec{R}_{op} = \frac{\vec{R}_p - \vec{R}_o}{|\vec{R}_p - \vec{R}_o|} \quad (4)$$

The  $\Delta V$  is selected to be negative at the beginning of the descent phase resulting in a blue shifted (increased) frequency in Equation 2. The resulting velocity equation is then written as a sum of four velocity components as summed and defined below.

$$\vec{R}_{op} \cdot \vec{V}_p = V_1 + V_2 + V_3 + V_4 \quad (5)$$

where,

$V_1$  defined as zonal wind u (positive towards East)

$V_2$  defined as Titan's rotation  $\Omega a \cdot \cos \lambda$  (co-aligned with  $V_1$ )

$V_3$  defined as meridional wind v (positive toward North)

$V_4$  defined as descent velocity  $v_T$  + vertical wind w (positive upwards)

Finally the orbiter's velocity projection onto the radio path direction is:

$$\vec{R}_{op} \cdot \vec{V}_o = V_5 \quad (6)$$

Based on the different sensor types in the database and the planetary environmental properties that the sensor must cope with the ISSPO determined a sensor package with similar properties to the actual flight unit. The DWE package [63] built by Daimler-Benz Aerospace (DASA), now EADS Germany used a commercially developed

space-qualified rubidium oscillator built by Ball Efratom Elektronik GmbH. Detailed information on the specific sensor configuration flown is not available and a similar performing model built by Symmetricom [52] uses a militarized Rubidium Oscillator.

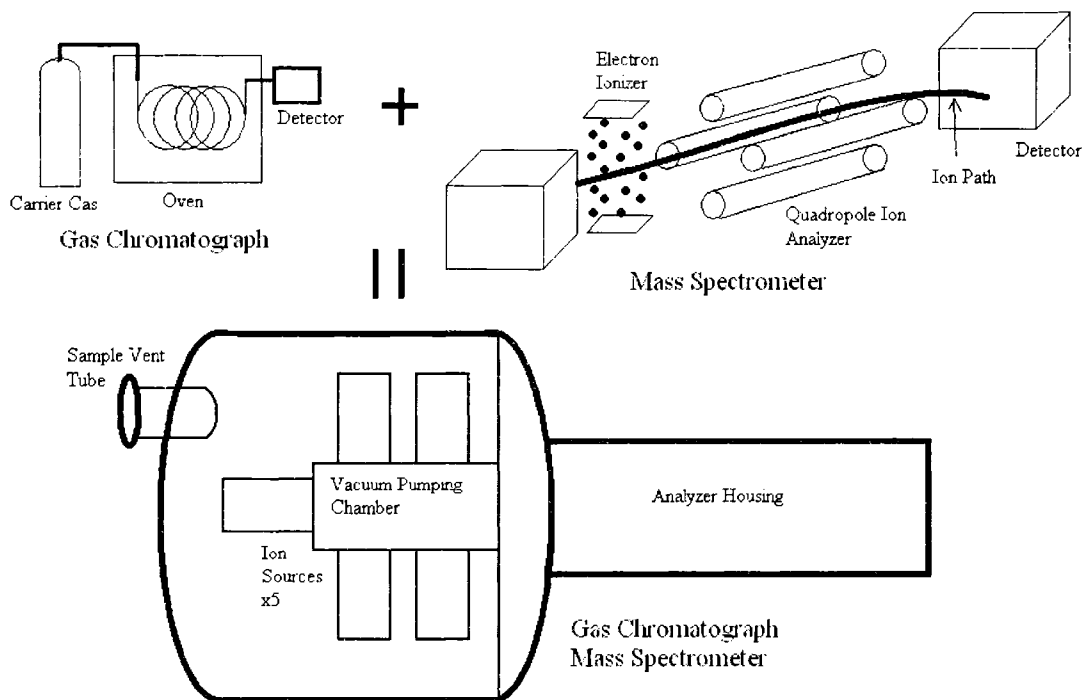
There are slight performance differences between the actual flight unit and the ISSPO design solution. The largest difference is the unit is about half the weight of the Huygens model but consumes more power at max input, but better matches the sensor performance during steady state operation. The model has a significantly shorter warm up period resulting in lower total power consumption. Brief summary results on the DWE sensor package are provided in Table 34 with full details of the sensor design and operation provided in Appendix B.

**Table 34 ISSPO DWE Sensor Configuration**

<b>ISSPO DWE Sensor System Architecture Result</b>		
<b>Model : Symmetricom 8130A</b>		
<b>Physical Properties:</b>		
Mass (g)	900	
Dimensions (mm)	102.6 × 74.1 × 72.8	L × W × H
<b>DC power:</b>		
Warm-up power (W)	<35.0	< 10 min
DC consumption (mA)	1094	At 32 V DC input
Energy (Wh)		worst case (minimum temp)
<b>Frequency Parameters:</b>		
Output frequency (MHz)	10	
Frequency long term drift	$<5.0 \times 10^{-11}$	After 1 month
Allen Variation	$3 \times 10^{-11}$	$\tau = 1 \text{ s}$
	$3 \times 10^{-11}$	$\tau = 10 \text{ s}$

#### **D. Gas Chromatograph Mass Spectrometer (GCMS)**

Use of Gas Chromatograph Mass Spectrometer (GCMS) in planetary missions is a relatively new option. Typically these units have been custom built and designed specifically for a single mission. As such, a database of flight rated configurations is limited. Numerous laboratory models exist but are not built with the same considerations that are given to flight systems. In aerospace missions mass, power, and volume are all at a premium whereas laboratory models do not have these constraints. Additionally flight rated models must be completely automated as there is no possibility to intervene should a problem arise. The design of such a device is also a combination of multiple separate instruments. The configuration flown on the Huygens probe is a custom configuration due to the payload constraints but based on commercially available individual products. The completed flight unit now represents a flight proven configuration with enough available performance data that it can represent a stand alone product to be used on other future missions with minimal changes for other planetary atmospheres. The GCMS onboard Huygens was built by NASA's Goddard Space Flight Center (GSFC), [64, 65, 66] and can be made available to reproduce for any future missions needing a similar device. The device is built around the two basic components: a gas chromatograph, and mass spectrometer. The layout of these components and associated hardware resulted in one of the most complex instruments GSFC has ever built as shown in Figure 7.



**Figure 7 Schematic Diagram of Huygens GCMS Sensor**

The science objectives for the GCMS are defined relatively broad [67], and include the following goals: to determine the noble gas abundance, isotopic ratios, and identify the high molecular weight trace organic compounds. Sufficient data from these sources provided enough design information to add the sensor configuration to the GCMS database. The flight unit is comprised of commercially available products including five ion sources attached to the mass spectrometer. The different components were machined to fit and assembled together to create the final flight unit.

The final assembled version houses all the sensor components and associated electrical hardware to sample the Titan atmosphere and analyze the data. Information required for the ISSPO program to determine the appropriate sensor is shown in Table 35.

**Table 35 ISSPO GCMS Sensor Input File Format**

1 <sup>st</sup> Column	2 <sup>nd</sup> Column	3 <sup>rd</sup> Column
“GAS ANALYSIS”	“MASS-CHARGE”	“MEDIUM”

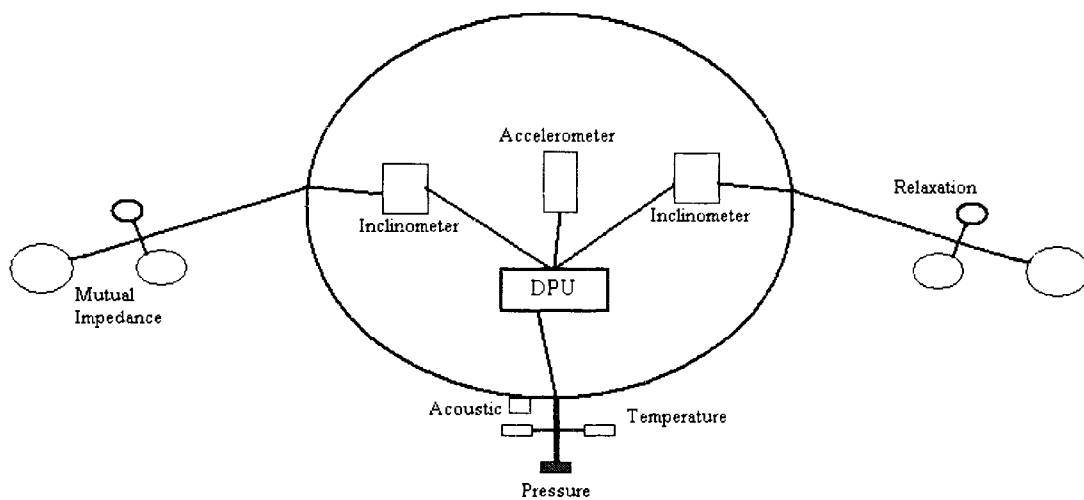
The data in Table 36 corresponds to the final published data for this component. The sensing range is defined as “MEDIUM” in comparison to other models with larger or smaller scanning ranges. It is not clear, from the reports whether this range is the maximum sensing range of unit or merely the range of chemical compounds expected to be found within the Titan atmosphere. In total 5634 mass spectra recordings were made during the descent phase with an additional 2692 spectra samples collected on the surface with a range of 2 -141 amu’s. [68] Designs within the ISSPO tool can be set to a high range, if the chemical contents of the planetary atmosphere are unknown; however with a better understanding of the compounds under investigation the required sensing range can be reduced.

**Table 36 ISSPO GCMS Sensor Configuration**

<b>ISSPO GCMS System Architecture Design Result</b>		
<b>Model : GCMS - Huygens</b>		
<b>Physical Properties:</b>		
Mass (kg)	17.30	
Dimensions (mm)	470 × 198 × 198	L × W × H
<b>DC power:</b>		
Typical Power (W)	28	After 36 min warm-up period
Average Power (W)	41	At 28 V DC input
Peak Power (W)	71	
<b>Sensing Parameters:</b>		
Mass – Charge Range	2-141	amu
Number of Ion Sources	5	units
Ion Source Charge	1.00	W degas
Field Range	16.00	W degas

### E. Huygens Atmosphere Structure Instrument (HASI)

The Huygens Atmosphere Structure Instrument (HASI) is a key multi-sensor component to mapping the Titan atmosphere. It incorporated multiple atmospheric sensor elements into an overall package that would monitor most atmospheric properties during the planetary entry. Sensors within this package have been categorized into the type of environmental data that is returned. These categories include: acceleration, pressure, temperature, and Permittivity, Wave, and Altimetry. Mission science objectives for the HASI sensor suite are to determine the density, temperature, and pressure through the descent profile to the surface, determine the nature of the surface contact, whether solid or liquid, determine atmospheric electrical conductivity, electric fields and atmospheric lightning, and surface topography by monitoring surface dielectric fields. Within the design of this sensor package a certain amount of redundancy is included; in case a single element failed the secondary unit would continue to provide information.



**Figure 8 Schematic Diagram of Huygens HASI Sensor Package**

A diagram of the different sensor components contained in the HASI package is shown in Figure 8. A summary of the different science data to be returned, and sensor system components, is detailed in Table 37. The sensors in the PWA subsystem or mounted external to the spacecraft, and shielded from the electric field generated within the probe body by all the electronic components. Two deployable boom arms are released prior to atmospheric entry, which contain the mutual impedance and relaxation probe sensors. These sensors consist of a pair of metal wire coils and flat disks mounted on the deployed boom arms shown in Figure 14.

**Table 37 Huygens HASI Science Package Objectives**

<b>HASI Sensor Packages</b>		
<b>Sensor Package</b>	<b>Sensor Type</b>	<b>Measured Parameter</b>
Accelerometers (ACC)	3-Axis Acc	Atmospheric deceleration, Descent Monitoring, Impact Response
Pressure Profile Instrument (PPI)	Kiel probe, capacitive gauges	Atmospheric Pressure
Temperature Sensor	2-Dual Element Platinum Thermometers	Atmospheric Temperature
Permittivity, Wave & Altimetry (PWA)	Mutual impedance	Atmospheric electric conductivity
	AC field measurement	Wave electric fields & Lightning
	Relaxation probe	Ion conductivity and DC electric field
	Acoustic sensor	Acoustic noise due to turbulence of storms
	Radar signal processing (FFT)	Radar echoes below 60 km altitude

Performance data on the sensitivity of the PWA sensor components is available in several documents [69,70] however, no data on the physical specification (dimensions,



weight, materials, etc.) of the sensors is available. The scope of the ISSPO tool is determine the necessary sensor components based on mission requirements from databases of existing commercially available components. The mutual impedance sensors and relaxation probe design used do not represent a broad enough spectrum of uses to develop a database of configuration for interplanetary missions. These components are excluded from the rest of the sensor design analysis.

Information on sensors in this package based on commercially available sensors was incorporated into the sensor databases and analyzed to determine the selection reasoning for each component in this sensor package. For each of the separate science requirements the appropriate sensor configuration is selected by the ISSPO tool out of the databases based on the information required by the program for each different sensor type. A comparison of the ISSPO tool results to the known flight units for each component of the HASI sensor package is shown in Table 38. Variations in the data between the flight version and the ISSPO determined sensor result based on the required input data, is discussed in each separate section.

**Table 38 HASI Sensor Package Summary Results**

<b>HASI Sensor Package Components</b>		
<b>Sensor Package</b>	<b>Flight Unit</b>	<b>ISSPO Results</b>
Accelerometer – 3 axis servo	Sundstrand QA-2000-030	QA2000-030
Accelerometer - piezo-resistive	Endevco 7264A-2000T	Endevco 7264A-2000T
Pressure Profile Instrument	Vaisala - Barocap	Series 48-0025
Temperature Sensor	dual element platinum resistance thermometers Rosemount Aerospace Inc.	Goodrich Model 0146MD
PWA – Acoustic Sensor	Kulite CT-190M	Kulite CT-190M
PWA – Digital Signal Processor(FFT)	Analog Devices ADSP-2100A	ADSP-2100A

## 1. Accelerometers

Two different configurations of accelerometers are carried aboard the HASI sensor package. The selected sensors feature single axis calibration, one featuring a high sensitivity used along the probe primary x-axis to precisely measure the deceleration profile of the probe through the entry phase, the other configuration three single-axis sensors to measure acceleration loads along all three primary axis at reduced resolution. Sundstrand, makers of the three-axis servo accelerometer [71], was acquired by Allied Signal in 1994, and operated for 5 years before being acquired by Honeywell sensors in 1999 [72]. Since, the launch of the Huygens probe additional versions of the sensors have been developed based on the same sensor platform and sensor performance data varies in sources. Due to the nature of the designs of these elements the maximum operational range of these components is shown on datasheets, however, through signal conditioning the maximum range of the models can be reduced with a corresponding increase in sensor resolution. For the piezoresistive accelerometer the design comes from the Endevco 7264 accelerometer line [73], the optional “T” configuration of the design reduces the transverse sensitivity to one percent from the default three percent calibration. The resulting HASI accelerometer sensor design data is shown in Table 39. Accelerometer data was also used to determine the atmospheric density during the planetary descent from Equation 7:

$$\rho = -\frac{2ma}{C_D AV^2} \quad (7)$$

where  $a$  is the magnitude of the acceleration,  $m$  is the entry probe mass,  $CD$  is the vehicle's drag coefficient,  $A$  is the frontal cross sectional area, and  $V$  is the velocity of the probe relative to the atmosphere direction.

**Table 39 ISSPO HASI Accelerometer Sensor Configuration**

<b>ISSPO HASI Accelerometer Sensor Package Results</b>		
<b>Model :</b>	QA-2000-030	Endevco 7264A-2000T
<b>Physical Properties:</b>		
Mass (g)	71	1.0
Dimensions (mm)	27.18 × 25.53 × 25.53	5.08 × 10.16 × 10.16
<b>Sensing Parameters:</b>		
Acceleration Range (g's)	+/- 60	+/- 2000
Transverse Sensitivity	3.0 %	1 %
Repeatability	≤ 160 μg	0.2 g

## 2. *Pressure Profile Instrument*

The Pressure Profile Instrument's objective is to map Titan's atmosphere pressure profile during the descent phase of the mission. Available data [74] on the configuration of the PPI sensor indicates that the silicon capacitive absolute pressure sensor flown is a variant of the Barocap design manufactured by Vaisala Co., in Helsinki Finland. Two types of silicon diaphragm were used, constructed based on different thicknesses to yield different sensitivities to the pressures in different regions of the atmosphere. Searches for specific information on the configuration flown aboard Huygens yielded no successful results. Available configurations on the Vaisala website, do not represent the flight rated model, or cover the same operational range. Without the specific flight configuration

data to incorporate into the database, the planetary environmental data is used to determine an alternate sensor with similar performance capabilities.

Contrary data [72] on the performance of the pressure sensing range proposes an upper limit on the sensing range of 1600 mbar. Onboard the spacecraft the pressure profile instrument was not exposed directly to the atmosphere, but atmospheric samples were funneled through a Kiel Probe into the instrument. The performance data for the flight model is compared to the ISSPO sensor configuration in Table 40.

**Table 40 ISSPO HASI Pressure Profile Instrument Sensor Configuration**

<b>ISSPO HASI PPI Sensor Package Comparison</b>		
<b>Model :</b>	Barocap – Flight Unit	ASCO 48-0025
<b>Physical Properties:</b>		
Dimensions (mm)	70.00 × 50.0 × 17.00	48.56 × 22.23 × 22.23
<b>Sensing Parameters:</b>		
Pressure Range (g's)	0 – 2000 mbar	0 – 1724 mbar
Accuracy	1.0 %	0.5 %

### 3. Temperature Sensors

The flight unit temperature sensors carried as part of the HASI sensor package features a coarse and fine sensor design. The designs features a platinum resistance thermometer mounted onto a platinum-rhodium truss frame that is mounted on a small STUB boom that extends past the spacecraft boundary layer, shown in Figure 15, during entry to measure free stream temperatures. The original flight units were produced by Rosemount Aerospace, Inc. based in Minnesota, U.S.A. Since the qualification effort and

launch of the Cassini- Huygens mission, they were acquired by Goodrich Sensor Systems [75], and data on the specific model is no longer available.

Based on the design of the TEM structure the measured temperature is not the actual ambient temperature and must be corrected for dynamic effects while moving in the atmosphere. The relationship between measured and corrected temperature [76] is given by Equation 8.

$$T_{corr} = T_{meas} - r \left[ \frac{V_r^2}{2C_p} \right] \quad (8)$$

Where “r” is a recovery factor,  $V_r$  is the relative velocity of the sensor to the atmosphere, and  $C_p$  is the specific heat ratio at constant pressure. A number of similar capable platinum resistance thermocouples are available within the known performance specifications of the original model. Information provide to the ISSPO tool to aid in selecting a similar sensor type is shown in Table 41. The performance properties of the selected sensor are shown in Table 42.

**Table 41 ISSPO HASI Temperature Sensor Input File Format**

1 <sup>st</sup> Column	2 <sup>nd</sup> Column	3 <sup>rd</sup> Column
“ATMOSPHERIC”	“TEMPERATURE”	“RESISTANCE”

Table 42 ISSPO HASI Temperature Sensor Configuration

<b>ISSPO HASI Temperature Sensor Package Results</b>		
<b>Model : M-0146MD</b>		
<b>Physical Properties:</b>		
Mass (g)	0.35	
Dimensions (mm)	45.72 × 1.524 × 1.524	L × W × H
<b>Sensing Parameters:</b>		
Temperature Range	-269 – 400	deg C
Stability	0.03	%
Repeatability	≤ 0.1	deg C

4. *PWA Acoustic Sensor*

The Acoustic sensor flight unit carried aboard the Huygens probe HASI experiment package [77] is tasked to measure the acoustic noise due to the turbulence of storms. However the environmental conditions on Titan placed additional constraints on the selection of the sensor. Due to the cold thermal environment, the reasoning [78] was made to use a pressure sensor capable of surviving the harsh environment, and sacrificing the ability to detect low level noise environments. This reduced ability to detect sounds prevents the pressure sensor from monitoring quiet sounds but allowed it sufficient resolution to detect strong winds and thunder as specified in the mission requirement. Information provided to the ISSPO tool to select this sensor is shown in Table 43. A summary of the relevant performance parameters is listed in Table 44.

**Table 43 ISSPO HASI Acoustic Sensor Input File Format**

1 <sup>st</sup> Column	2 <sup>nd</sup> Column
“ACOUSTICS”	“SENSOR”

**Table 44 ISSPO HASI Acoustic Sensor Configuration**

<b>ISSPO HASI Acoustic Sensor Package Results</b>		
<b>Model :</b> Kulite CT-190M		
<b>Physical Properties:</b>		
Mass (g)	4.0	
Dimensions (mm)	34.3 × 9.5 × 9.5	L × W × H
<b>Sensing Parameters:</b>		
Acoustic Range	56.8 – 135	dB <sub>SPL</sub>
Resonance	500	kHz
Temperature Range	-195.5 – 37.0	deg C

5. *PWA Digital Signal Processing*

Data collected from the various sensors in the PWA package that required real time processing to trigger events, or interpret the data was processed via a Digital Signal Processor. In essence these components are micro-computers, with fixed amounts of program and data memory. In selecting the right component for any use, several key operational parameters must be taken into account – including processor speed, memory, number of signal inputs, and special functionality. In selecting a processor speed, faster is usually better, however several limitations are placed on this. The device chosen must be fast enough to read in the sampled data at a fast enough rate, and be able to generate the results quickly enough to be useful. For time critical events, the data must be sampled, analyzed, and results output before too much time has elapsed or else a critical

time event may be missed. The counterpoint to this is to opt for the fastest processor speed, as this would provide more than sufficient computational power. This trade-off comes with a price, as the faster processor is usually heavier, larger, and requires more power to operate. Additionally the processor might be so over-powered that the processor may never use its full capacity and end up wasting precious spacecraft resources.

An additional critical selection criteria is the amount of available memory for the both the program and for data. In development aspects the amount of program code should be kept to a minimum to ensure all the data will be able fit onto a single processor's memory, as difficulties are often encountered when program data has to be stored in separate memory locations. Attempts to reduce the amount of code used and optimize the organizational pattern can improve the performance of the overall component. Additional care, in the processing of data is also crucial to ensure the program efficiently uses the available memory to prevent possible data corruption if the amount of available data memory is overflowed.

An additional crucial selection consideration in determining the correct DSP is the type of tasks the processor is required to complete. Components are selected based on the specific tasks the processor is required to perform. Special versions of processors are available, tailored with specific computational processing abilities, or algorithms to be able to perform specific types of calculations.

Selection of the DSP unit flown on the Huygens Probe is based on the required computation speed of the processor and the attached components. Sensor data from the



mutual impedance, relaxation probes, acoustic sensor, and radar altimeter unit is passed to the DSP unit and undergoes Fast Fourier Transform analysis before transmitting the data to the Huygens orbiter for the return trip to Earth. Data provided to the ISSPO tool to select the DSP component is shown in Table 45. A summary of properties of the ISSPO resulting selection is in Table 46.

**Table 45 ISSPO DSP Component Input File Format**

1 <sup>st</sup> Column	2 <sup>nd</sup> Column	3 <sup>rd</sup> Column
“DATA ANALYSIS”	“LOW”	4

**Table 46 ISSPO HASI DSP Component Configuration**

<b>ISSPO HASI DSP Sensor Package Results</b>		
<b>Model :</b> Analog Devices ADSP-2100		
<b>Physical Properties:</b>		
Output Power (mW)	790.0	
Dimensions (mm)	33.83 × 33.83 × 9.12	L × W × H
<b>Operating Parameters:</b>		
CPU Operating Speed	8.192	MHz
Data Memory	16 K-Words	16 Bins
Program Memory	32 K-Words	24 Bins
Temperature Range	-55.0 – 125.0	deg C

**F. Surface Science Package (SSP)**

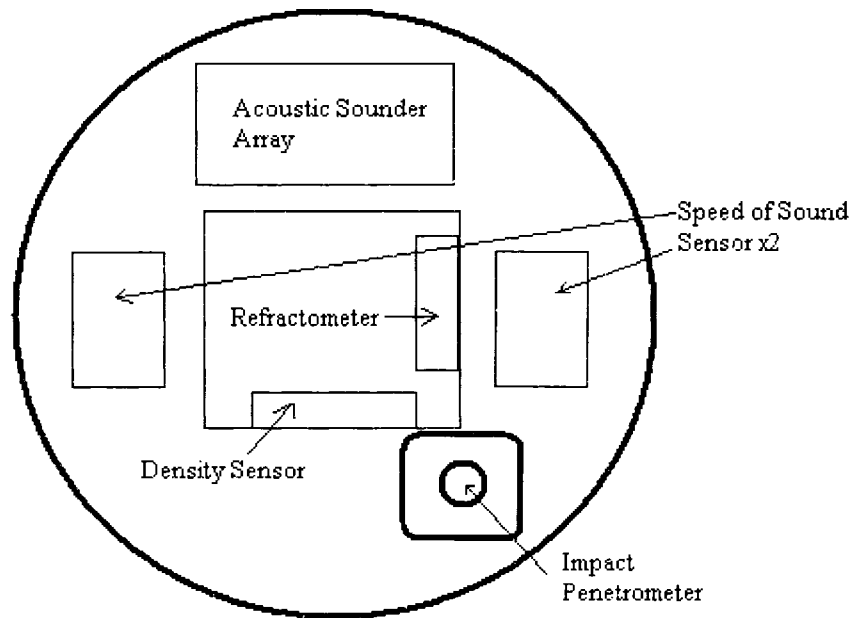
The Surface Science Package (SSP) carried aboard the Huygens Titan probe is comprised of nine different sensors monitoring seven different types of planetary science data. The charge of the SSP is to measure and determine the nature of Titan’s surface

properties. Most of the sensors were mounted on the external surface of the probe to allow for direct measure of the surface environment. These sensors were mounted on a cylindrical unit dubbed “Top Hat” [79] that extended slightly forward of the surface to provide direct contact with the surface. A summary of the SSP’s scientific objectives is shown in Table 47. The main components of the SSP are located along the probes x-axis at the center of the front of the vehicle. The remaining sensors are mounted internal to the spacecraft. A vent tube that extends through the probe’s structure is attached to the Top Hat to allow Titan’s fluid to the internal sensors if Huygens lands in an ocean environment.

**Table 47 Huygens Surface Science Package Scientific Objectives**

<b>Surface Science Package (SSP) Scientific Objectives</b>
Determine the physical nature and condition of Titan’s surface at the landing site
Determine the abundances of the major constituents, placing bounds on atmospheric and ocean evolution
Measure the thermal, optical, acoustic and electrical properties and density of any ocean, providing data to validate physical and chemical models
Determine wave properties and ocean/atmosphere interaction
Provide ground truth for interpreting the large-scale Orbiter Radar Mapper and other experimental data

A schematic layout of the sensor contained within the “Top Hat” is shown in Figure 9. The two sensor elements not shown in the diagram below are the two tilt sensor components which are mounted internal to the spacecraft housing atop the SSP electronics box.



**Figure 9 Schematic Diagram of Huygens SSP Top Hat Sensor Cavity**

The location of the SSP within the overall Huygens probe structure is shown in Figure 10 with the impact penetrometer extending out beyond the surface of the probe vehicle. Based on the summary of mission requirements input into the ISSPO program, a summary sensor design package is detailed in Table 48. The inputs and summary comparison for each sensor design is discussed in the following sections. Information on commercial components used in the SSP is scarce and several of the sensors are approximated in the final ISSPO design solution by similar components.

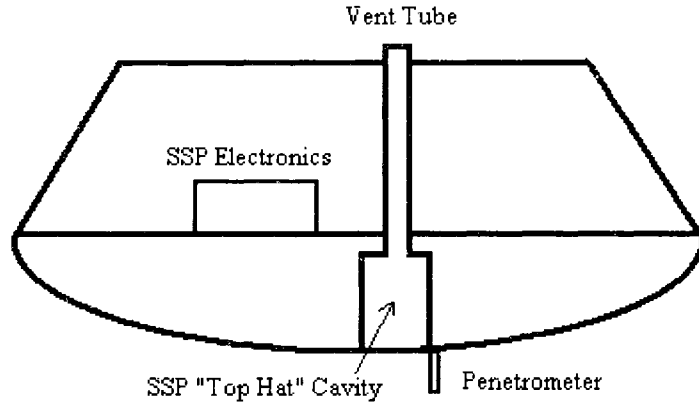


Figure 10 Schematic Diagram of Huygens SSP Component Layout

Table 48 ISSPO - Huygens SSP Sensor Component Comparison

SSP Sensor Components		
Sensor	Flight Unit	ISSPO Results
Accelerometer – Impact penetrometer ACC-E	Piezoelectric Ceramic PZT-5A	N/A
Accelerometer - Impact accelerometer ACC-I	Endevco 2271AM20 0 – 100 g's	Endevco 2271AM20
Tilt Sensor TIL	Spectron L-211U +/- 60°	Spectron L-211U
Temperature Sensor THP	Hot Wire 65 – 100 K	M-0146MD
Velocity of Sound API-V	Piezoelectric Transducers – 2 150 – 2000 m/s	Physical Acoustics R80 Alpha
Acoustic Sounder API-S	Piezoelectric Transducers Array	N/A
Fluid Permittivity PER	Capacitance Sensor	N/A
Density of Fluid DEN	Archimedes Sensor 400 – 700 kg/m <sup>3</sup>	N/A
Refractive Index REF	Critical Angle Refractometer 1.25 – 1.45	Huygens Design

Several of the components utilized in the SSP represented custom design solutions for the specific application they were tasked for on the Huygens probe.

1. *Accelerometers ACC*

To determine the impact acceleration loads on the vehicle during the final phase of the atmospheric descent, two different types of units are used. One unit (ACC-I) will measure the landing acceleration loads dependent on the surface material composition. The other probe is based on a penetrometer design and extends forward of the outer skin of the vehicle and is designed to measure the surface resistance force on the penetrometer tip as the probe lands on the surface. The ACC-I unit flown is an Endevco 2271AM20 piezoelectric accelerometer [80]. The unit is mounted internal to the spacecraft at the base of the SSP electronics box and records the acceleration profile along the spacecraft's primary x-axis. Properties of the resulting ISSPO sensor configuration are shown in Table 50.

**Table 49 ISSPO SSP Accelerometer Sensor Input File Format**

1 <sup>st</sup> Column	2 <sup>nd</sup> Column	3 <sup>rd</sup> Column
“ACCELERATION”	“IMPACT”	“Self Generating”

**Table 50 ISSPO SSP Accelerometer Configuration**

<b>ISSPO SSP Accelerometer Sensor Package Results</b>		
<b>Model :</b> Endevco 2271AM20		
<b>Physical Properties:</b>		
Dimensions (mm)	29.2 × 15.88 × 15.88	L × W × H
<b>Operating Parameters:</b>		
Sensing Range	0 – 10,000	g's
Transverse Sensitivity	3	%
Temperature Range	-269.0 – 260.0	deg C

2. *Tilt Sensor TIL*

The tilt sensor [81] carried aboard the Huygens probe primary purpose was to return orientation data about the spacecraft if it were to land in an ocean. Combined with data from the onboard accelerometer, the position and movement of the probe could be obtained. If the probe lands on solid ground, it will return the final resting orientation. The unit was also designed to operate during the entry phase, and provide orientation data that would improve the fidelity of data collected in the DWE by adding probe orientation data to determine atmospheric wind speeds.

The sensor component flown aboard Huygens consists of two single axis tilt sensors mounted in a dual axis configuration on a small mounting block [82]. Sensors used on the SSP were chosen for their quality, performance and high reliability. The selection criteria required as input information to the ISSPO tool is shown in Table 51. A summary of the resultant sensor design is shown in Table 52.

**Table 51 ISSPO SSP Tilt Sensor Input File Format**

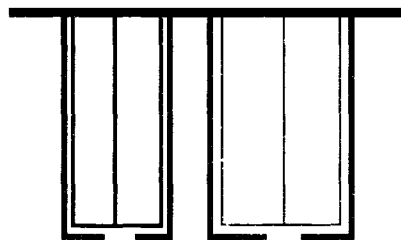
1 <sup>st</sup> Column	2 <sup>nd</sup> Column
“INCLINATION”	“HIGH”

**Table 52 ISSPO SSP Tilt Sensor Configuration**

<b>ISSPO SSP Tilt Sensor Package Results</b>		
<b>Model :</b> Spectron L-211U		
<b>Physical Properties:</b>		
Dimensions (mm)	41.0 × 18.4 × 15.9	L × W × H
<b>Operating Parameters:</b>		
Tilt Range	+/- 60	Degrees
Resolution	0.0083	Degrees
Temperature Range	-54.0 – 124.0	deg C

### 3. *Temperature Sensor THP*

The Temperature sensor carried aboard the Huygens probe as part of the SSP is a dual operation hot wire thermometer to measure surface thermal conductivity. The sensor configuration consists of a redundant pair of cylinders housing a platinum wire resistor. Since the final landing environment was unknown, and the possibility existed that the probe could land on either solid ground or in an ocean environment, each cylinder is designed for a liquid measurement or gaseous atmospheric sample. Each cylinder contains a platinum wire resistor configured to operate as a four wire resistance thermometer, and operates via a transient hot wire method. [83] A schematic drawing of a pair of the four sensing cylinders with a cut-a-way view of the internal structure is shown in Figure 11. The left-hand thinner cylinder is configured for sampling in a liquied environment while the right-hand wider cylinder is configured with a thinner sensing wire to sample the atmosphere.



**Figure 11 Schematic Diagram of Huygens SSP THP Sensor**

The thermal conductivity is back calculated from the temperature change in the wire when a known power is applied to the wire. Calculation of the thermal conductivity of the air / fluid is based on equations 9 – 11.

$$\Delta T = \frac{q}{4\pi\lambda} \ln\left(\frac{4\kappa t}{a^2 C}\right) \quad (9)$$

$$\lambda = \frac{q}{4\pi A} \quad (10)$$

$$\kappa = \left(\frac{a^2 C}{4}\right) \exp\left(\frac{B}{A}\right) \quad (11)$$

In Equations. 9 – 11,  $\Delta T$  is the temperature rise in the wire,  $\lambda$  is the thermal conductivity of the medium,  $\kappa$  is the thermal diffusivity of the medium,  $q$  is the power per unit length of wire,  $a$  is the radius of the wire,  $t$  is the time interval of the sampling period, and  $C$  is a constant with a value of  $e^g$ , and  $g$  is Euler’s constant. The  $A$  and  $B$  terms in Equation 4, represent the gradient and intercept values, respectively, of the plot of  $\Delta T$  vs.  $\ln(t)$ .

The material construction of this sensor is the same platinum as the thermal sensor carried in the HASI sensor package. The configuration of the components here is custom but the same basic configuration can be converted to suite the purpose of the experiment. The selection of this same type of sensor is achieved by loading the same input criteria into the ISSPO tool as shown in Table 53.

**Table 53 ISSPO SSP THP Sensor Input File Format**

1 <sup>st</sup> Column	2 <sup>nd</sup> Column	3 <sup>rd</sup> Column
“ATMOSPHERIC”	“TEMPERATURE”	“RESISTANCE”



Table 54 ISSPO SSP THP Sensor Configuration

<b>ISSPO HASI Temperature Sensor Package Results</b>		
<b>Model : M-0146MD</b>		
<b>Physical Properties:</b>		
Mass (g)	0.35	
Dimensions (mm)	45.72 × 1.524 × 1.524	L × W × H
<b>Sensing Parameters:</b>		
Temperature Range	-269 – 400	deg C
Stability	0.03	%
Repeatability	≤ 0.1	deg C

#### 4. *Acoustic Sensor API*

The Acoustic Properties Instrument (API) is comprised of two separate instruments. A pair of piezoelectric ceramic transducers [80] comprising the API-V element are mounted on opposite sides of the Top Hot facing each other. The two units transmit and receive ultrasonic pulses [84] over the unobstructed path. A timing circuit is used to measure the time it takes the pulses to travel between the sensors. With the known distance between the sensor elements and the signal delay time the speed of sound could be resolved to 8 cm/s. The calculation of the speed of sound is made via several different forms depending on the nature of the medium that the sensors are interacting with, for solids Equation 12a applies, and Equation 12b applies for liquids and gasses.

$$c = \sqrt{\frac{B}{\rho}} \quad (12a)$$

$$c = \sqrt{\frac{\gamma RT}{M}} \quad (12b)$$

In Equation 12a, B represents the Young’s modulus of the medium, and Equation 12b is a variation of the Ideal Gas equation, where  $\gamma$  is the ratio of specific heats ( $\sim 1$  for liquids, 1.1-1.4 for gasses), R is the universal gas constant and M is the molecular weight of the gas sample. The other component to the API sensor is an array of 10 resonant piezoelectric plates and operates similar to a SODAR array (Sound Detection and Ranging). This component uses an acoustic beam transmitted to the surface to detect from the signal scattering the number of particles in the atmosphere. The data results will be compared to the radar altimeter and verify results. The use of an acoustic array in this manner is not widely flown on planetary missions, thus a database of commercially available components featuring this ability is limited and was excluded from direct comparison in this analysis.

The API components were designed and built under research collaborations between academia and industry. Data on the configuration of the flight unit version of the components for the sound velocity sensor was unavailable. However, a wide enough database of acoustic transducers, allow the ISSPO tool to determine a similarly capable option that could measure the speed of sound. Input criterion for the sensor selection by the ISSPO tool is shown in Table 53. The performance properties of the resultant ISSPO design configuration are shown in Table 54.

**Table 55 ISSPO SSP Acoustic Properties Instrument Speed of Sound Input File Format**

1 <sup>st</sup> Column	2 <sup>nd</sup> Column
“ACOUSTICS”	“VELOCITY”

**Table 56 ISSPO SSP Acoustic Sensor Configuration**

<b>ISSPO SSP Acoustic Sensor Package Results</b>		
<b>Model :</b> Physical Acoustics R80 Alpha		
<b>Physical Properties:</b>		
Mass (g)	32	
Dimensions (mm)	19.0 × 19.0 × 21.4	L × W × H
<b>Operating Parameters:</b>		
Frequency Range	200 – 1000	kHz
Temperature Range	-54.0 – 124.0	deg C

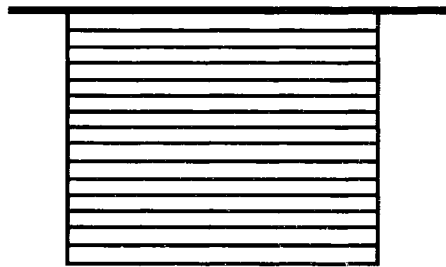
5. *Permittivity PER*

The task of the permittivity sensor flown aboard Huygens is to measure the static permittivity of any fluid samples found on the surface. In part this was to aide in the determination of the mixing ratio of ethane to methane in any surface liquids. From the permittivity of the sample, and data from the density sensor, an estimate of the mean molecular weight of the sample can be obtained from the Clausius-Mossotti function, Equation 13. Where  $\rho$  is the density,  $\alpha$  is the molecular polarisability,  $N_A$  is Avogadro's number,  $\epsilon_0$  is the electrical permittivity of free space and  $\epsilon_r$  is the refractive index permittivity from the refractometer.

$$\alpha = \frac{3\epsilon_0(\epsilon_r - 1)}{\rho N_A(\epsilon_r + 2)} \quad (13)$$

The Huygens design consists of 22 stacked open plate capacitors. [80, 81] A schematic layout of the sensor design is shown in Figure 12. The design of this type of

sensor is specifically tied to the mission and would be custom built for each application. The design can be based on any number of mission constraints or tuned specifically to determine a number of fluid or gaseous properties can be determined. These units serve a variety of functions that serve other industries and uses, such that no standard design configurations exist. Capacitance units for this application can be configured via many different parameters and are best suited being designed for each application.



**Figure 12 Schematic Diagram of Huygens SSP Permittivity Sensor Design**

6. *Density DEN*

The density sensor flown aboard the Huygens probe employs Archimedes' principle in its operation. A buoyant float is suspended off the Top Hat inner wall by a thin flexible beam [81] and the resulting forces on the beam are measured by strain gauges. Additional data on the specific configuration of the sensor are not available. The unit was configured as a Wheatstone bridge circuit, and the unit was designed to operate in the region of  $400 - 700 \text{ kg/m}^3$  with a science mission requirement of a resolution of  $\pm 2 \text{ kg/m}^3$  at a density of  $550 \text{ kg/m}^3$ . The design of the electronic circuit can be based on

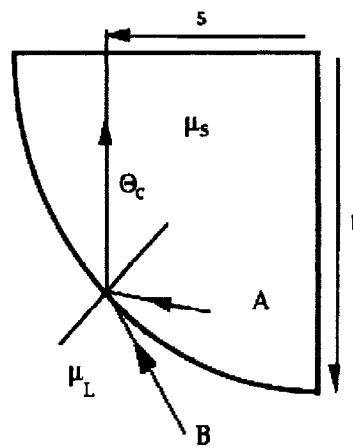
any circuit configuration and no data on the cylindrical float unit incorporated into the design is available. Only a constant voltage source is required and the resulting deflection of the float alters the Wheatstone bridge circuit resistance, and changes the output voltage. Knowledge of the beam material and its properties allows the determination of the density of the fluid that caused the resulting deflection and thus the circuit voltage change. Density sensor configurations like this are simple common components, but usually custom configured to the exact system use. Sensor configurations currently within the ISSPO database are based on larger scale Archimedes buoyancy float designs for measuring density in pipe flows or other large liquid systems.

#### 7. *Refractive Index REF*

The Refractive Index (REF) sensor carried aboard the Huygens probe measured the refractive index of the Titan atmosphere determine the abundance of methane and ethane in any liquid samples on the surface. The flight model is a linear critical angle refractometer [81], which uses a sapphire prism to aid in determining the refractive index range. The flight unit also incorporates a detector array which incorporates a self-scanning linear photodiode array. This detector then measures the light incident upon it and indicates the transition between light and dark on the array. A schematic representation of the operation of a refractometer is shown in Figure 13. Two LED's transmit light through and external to the prism to measure the refraction index. Light from the internal source "A" refracts of the prism's external surface and terminates in the

detector array. Knowing the “r” and “s” distances allows the calculation of the refractive index angle  $\Theta_c$  of the prism. Rearranged the equation [85] for “s” is given as a function of the prism ( $\mu_s$ ) and fluid ( $\mu_L$ ) refractive indexes, in Equation 14. The external light source “B” passes through the sample, either gaseous or liquid, before entering the prism and landing on the detector array.

$$s = r \times (\mu_L / \mu_S) \quad (14)$$



**Figure 13 Schematic Diagram of a Refractometer Prism**

The final flight configuration is a combination of commercial components which was designed for laboratory use and custom elements to aid in detection and operation. The component was built as a collaboration between the University of Manchester Institute of Science and Technology (UMIST) and Rutherford Appleton Laboratory. The overall design of the refractometer flight unit includes the refractometer sensor, detector, light sources, and associated electronics to record the data.

The location of the refractometer component is within the center of the Top Hat on the Huygens probe. It is oriented with the sapphire prism face facing downward; to interact with the gaseous atmosphere or liquid surface once it lands. [80]

Data on the original laboratory model was not available to include in the ISSPO database, however enough data is available on the performance and configuration of the flight unit to add it to the database for refractometer sensors and is the resulting design from the ISSPO tool. The selection of the “REFRACTION” sensor option in the ISSPO input file requires no additional modifiers, and the resulting design is chosen by the refraction index range and data based on the planetary bodies environment. The summary of the resulting sensor data is then shown in Table 57. The photodiode array [85, 86] is a commercially available unit mounted to the refractometer to record the data sampled during the mission.

**Table 57 ISSPO SSP Refractometer Summary Configuration**

<b>ISSPO SSP Refractometer Sensor Package Results</b>		
<b>Model :</b> Huygens - Ref		
<b>Physical Properties:</b>		
Dimensions (cm)	10.0 × 10.0 × 11.61	L × W × H
Power (mW)	10	Photodiode array requirement
<b>Operating Parameters:</b>		
Refractive Index Range	1.25 – 1.45	Refractive Index Units
Accuracy	0.001	Refractive Index Units

## **G. ISSPO Huygens Summary**

The results of the ISSPO tool comparison with the Huygens mission requirements shows a high correlation of the ISSPO tool results with the flown sensors. Given the mission requirements for science data and the sensor package power and mass limits, the ISSPO tool was able to assemble a sensor configuration capable of meeting the requirements. Data on sensor configurations is based on product specified data sheets available from commercial suppliers. Variations in reported data amongst different vendors make for difficult product comparisons. Data not available for all sensor configurations amongst suppliers is zeroed out within the sensor databases.

For each sensor package the payload totals were used for ISSPO program inputs. Often power or mass properties of the individual components were not available for the individual components to compare against the total sensor package weight. Total sensor package weight used in the inputs file also includes mounting hardware for the components, electrical wiring components, and miscellaneous elements. Component level mass properties for the different sensor elements are not all available for individual level comparisons.

Power requirements for each specific component are also not available and made direct component level comparison difficult. If data is available for both sensor voltage and current then the sensor power level is derived. Some datasheets provide voltage or current, not both making comparison difficult. Power consumption levels for some components are available for different levels of operation, i.e. minimum, typical, and maximum. When maximum power data was available the data is passed to the power



properties function as the total maximum sensor power must be less than the input power limit to pass the program power limit check.

Custom configured sensors for some of the sensor carried aboard the Huygens probe were not included in the benchmark case, as they represent a single point design or use of the sensor has not been used enough to develop a database of sensor configurations. Additional space missions with similar science objectives are required to develop additional sensor databases.

## **V. Venus Atmospheric Properties Mission**

With the development of the ISSPO tool, a study for an interplanetary mission can be made. Much interest as of late has been focused on Mars and the outer gas giant planets. Often forgotten are the inner planets from which there is still a wealth of information to be determined. If the inner planets, one still poses many interesting questions, that have yet to be answered. Venus offers many additional atmospheric questions, and represents one of the most extreme environments to survive.

Venus's atmosphere [1] represents an environment with a surface pressure about 90 times that of earth, planetary surface temperature of 730 K, and a carbon dioxide (CO<sub>2</sub>) atmosphere with cloud layers compromised of sulfuric acid. The upper region in the atmosphere also rotates about the planet in a period of about 4 days, about 60 times faster than at the surface. Currently it is unknown what the driving mechanism is behind this observation. Flow around the planet also tends toward the poles, but no observed

flows head away from the polar regions. The nature of the harsh environment on Venus makes long term observation of atmospheric properties difficult, and the underlying operation of some of the atmospheric properties are likely to remain a mystery until new methods can be found to survive the environment.

The objective goals of the mission will be to provide additional information on the structure of Venus's atmosphere, and to try to provide insight into additional mechanisms that could be causing the observed atmospheric properties. Along with the atmospheric suite, additional sensors carried aboard the craft can provide additional contextual information to exactly what is occurring during the mission timeframe. Unfortunately due to the extremely hot and corrosive nature of Venus's atmosphere, a long term mission that would operate over several weeks or months is not possible, so as much information on what is occurring must be recorded in the mission time available. The mission window would be on the order of several hours Earth time, and last during the atmospheric entry and for a period of time on the surface.

The sensor elements required during this planetary mission are shown in Table 58, along with information on the nature of the observation and its benefit to the planned mission concept. With the relative closeness of Venus, the mission timeframe is reduced, and the transmission time delay is significantly shorter than missions to the outer planets.

Table 58 Venus Atmospheric Mission Objectives

Venus Atmospheric Mission Objectives	
Mission Data	Objective Reasoning
ATMOSPHERE - Temperature	Provide Atmospheric Temperature consistent with mission trajectory, provides temperature calibration data for other sensors onboard
ATMOSPHERE - Pressure	Provide Atmospheric Pressure consistent with mission trajectory, provides pressure calibration data for other sensors onboard
ATMOSPHERE – Wind Velocities	Measure Wind Velocity profiles throughout the atmospheric descent, providing data to validate physical models
INCLINATION	Provides spacecraft orientation information to determine relative wind flows around spacecraft
ACCELERATION	Provide descent deceleration profile information and use for event sequence timings
EM FIELD	Map planetary magnetic field to determine if it accounts for observed wind flows
RADIATION	Detect charged particle occurrences in the atmosphere looking for energetic reactions in the atmosphere
OPTICS	Provide in-situ visual observations of wind flows in visual, ultraviolet, and infrared wavelengths to track cloud motion patterns

These sensor data types are then loaded into an ISSPO input file to investigate the destination planet Venus, and use the “SI” unit system. In this case the mission is being planned and there are no fixed limits to apply to the sensor design for a preliminary configuration. Values are still required by the program, but can be set to any values. If the resulting package is less than the input values, the mission design will “appear” to “succeed”, and if the input limits the tool will “appear” to “fail”. However, since there is no driving goal to meet for this mission concept, a failure case merely means that the input limits are too low. This can be resolved by finding ways to reduce the weight, volume, or power consumption, through a custom designed configuration, or find ways to

increase the payload budget limits. Either of these two methods usually means additional development time and additional budgetary costs.

**Table 59 Venus Atmosphere Concept Mission ISSPO Input Format**

1 <sup>st</sup> Column	2 <sup>nd</sup> Column	3 <sup>rd</sup> Column	4 <sup>th</sup> Column	5 <sup>th</sup> Column	6 <sup>th</sup> Column
"ATMOSPHERE"	"WIND VELOCITY"	"ANEMOMETER"	"TEMPERATURE"	"VOLTAGE"	
"INCLINATION"	"MEDIUM"				
"ACCELERATION"	"MEDIUM"	"Voltage"			
"EM FIELD"					
"RADIATION"	"Charged Particle"				
"OPTICS"	"ARRAY"	"MEDIUM"	"VISUAL"	"UV"	"NIR"

A summary of the resulting design for the different sensor is shown in Table 59. It includes a minimum design for each sensor, and the quality of the data returned would be improved by incorporating all spacecraft axis and some redundancy into the sensor design. Sensors that record one-dimensional data, i.e. wind velocity, inclination, and acceleration would require at least one sensor along each primary spacecraft axis to provide a complete description of the data being recorded. With the primary data interest in wind velocity, use of a redundant design would be beneficial to the returned science data.

Table 60 ISSPO Venus Atmosphere Mission Sensor Package Summary Results

<b>ISSPO Venus Atmosphere Sensor Package Results</b>						
<b>SENSOR</b>	<b>NAME</b>	<b>RANGE</b>	<b>MASS</b>	<b>VOLUME</b>	<b>POWER</b>	
ATMOSPHERE - Temperature	K	- 200 -- 1250 deg C	0	0	0	
ATMOSPHERE - Pressure	13(C,U)2000P	0 -- 13789.5 kPa	0	29.41 cm <sup>3</sup>	0.030 W	
ATMOSPHERE -- Wind Velocities	FT 702	0 -- 70 m/s	0.500 kg	0.0004 m <sup>3</sup>	0	
INCLINATION	L-212T	-45 -- 45 deg	0	12.00 cm <sup>3</sup>	0	
ACCELERATION	MA15	-50 -- 50 g's	0.1417 kg	47.72 cm <sup>3</sup>	0.64 W	
EM FIELD	TAM-1	0 -- 1000 mG	1.080 kg	0.0016 m <sup>3</sup>	0	
RADIATION	LPD	0 -- 250 MeV	7.00	0.0067 m <sup>3</sup>	15 W	
OPTICS	CCD 3041	UV VISUAL NIR	0	3.384 cm <sup>3</sup>	0	
SUMMARY			8.7217 kg	0.00871 m <sup>3</sup>	15.77 W	

The full details of the assembled sensor package are noted in Appendix C. Data for some of the components is not available, and the results might vary with additional sensor data. With the sensor configuration known it can be incorporated into the spacecraft development process and provide information for the size and power of the vehicle needed to carry the science payload.

## **VI. Summary**

The ISSPO Tool is developed as a preliminary tool to determine the sensors required to answer planetary science mission questions. Given a planetary body and a description of planetary science objectives to be accomplished, the tool will determine the sensors needed to return the results. ISSPO selects the correct sensor components based on the input criteria and the planetary environmental properties. Selection of the final sensor components are then based on additional operational parameters that vary for each different sensor component. Databases for each type of sensor are based on commercially available sensors. The goal of this is to reduce the amount of time spent performing trades studies to determine the correct sensor to use. The use of commercially developed products and flight rated hardware reduces the time and cost of new space missions by having to develop new custom hardware components for each mission.

A benchmark case proving the sensor design algorithms was performed on the Huygens probe mission to Titan. Science objectives for each sensor package were

broken down into the different types of sensor data that was to be provided. The requirements for the different sensors were input into the ISSPO input file along with the sensor package weight and power limits. Evaluating each sensor package resulted in selection of the Huygens hardware for known flight components. For sensors where the commercial component was unknown, similar performing sensors were chosen based on operational range, environmental properties and the other sensor requirements input into the ISSPO program file. The resulting design configurations agree within a reasonable margin to the sensor package mass and power requirements.

A sensor payload configuration was developed for a conceptual mission design to Venus to answer additional question on the mechanisms that drive the intense atmospheric winds. The sensor package was designed to monitor atmospheric winds, probe orientation, acceleration, planetary magnetic fields, charged particle radiation, and optical imaging to provide a visual record of atmospheric cloud movements. At this conceptual level no restrictions on payload mass, volume, or power constraints are applied to the mission design. In this manner the ISSPO tool can provide preliminary budgets, for mission requirements, to determine the minimum sensor configuration needed to obtain the mission science data.

The summary package comprises commercially available design components, and could be modified further by the use of custom designed sensors to provide higher fidelity data or modified to reduce mass or power requirements. The use of custom mission components, adds complexity, time, and cost into the development time frame for

any mission, as the unit must be fully qualified and tested before being certified for flight use.

The ISSPO tool also aids in mission design. Based on the knowledge of the planetary science mission a preliminary payload mass budget can be determined. With this knowledge the associated spacecraft hardware can be determined required to support the mission. The size of the spacecraft will determine the required size of the launch vehicle and the operating capabilities it will need to have to send the probe to its target. This information then can be used to determine the operating timeline for the mission.

## **VII. Future Work**

The framework has been established here for a tool to determine the optimal sensor configuration based on the mission science requirements. Refinements can be made to each available sensor database, update existing database values, add additional performance values, incorporate sensors based on different governing equations, new sensor types, etc. As each new planetary mission is executed, sensor components within the database can be updated, and the selection algorithms refined. Additional refinements can be made to the tool at system level parameters and account for interactions between the sensors and the spacecraft systems.

With each new planetary mission the databases of flight proven equipment will grow, and new methods will be evaluated to obtain science data. Development of custom configurations today will lead to common components in the future. There



always exists a need for new custom sensor components to answer planetary science questions or deliver new levels of data fidelity that were unobtainable before. The science and technology for these new sensor designs can be built into the ISSPO Program and expanded as new methods are found to determine answers to new scientific questions.

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## APPENDIX A:

### N2 Systems Diagram

The N2 systems diagram demonstrates the hierarchical nature of the program's operation and the flow of requirements and parameters between the program modules. It visually illustrates the dependent nature of the design selection based on all the other sensor component elements within the system.

The diagonal flow nature of the chart details the order of operation of the program and illustrates the direction movement of program inputs and outputs between the modules. The goal is begin with the broadest goals for the final system and use the top level programs to determine all the inputs needed by the sub-functions and process them in the most efficient manner. Arrows in the upper right diagonal portion of the diagram indicate the inputs and out puts of the different modules and how the chosen components provide inputs to other modules. Arrows below the main diagonal shape indicate a direct relational dependency on a higher level component by another component lower along the diagonal. The organization of the individual components is to reduce the number of "up-flowing" program elements in the illustration.

By reducing the number of up-flowing elements the design cycle is reduced and fewer iterations are required before a final design is achieved. The input file format required by the ISSPO program, loads all the required program data and determines which sensor to run and provides them in a simple method, reducing the number of design cycle iterations. The full system diagram detailing the relationship between all the program modules, and flow down of data is detailed in Figure A-1. The atmospheric

suite is shown as the single sensor package system as it is called in main program. The individual sensors contained within this module are tightly coupled together, and more dependent on the output of other sensor within this module. The N<sup>2</sup> diagram showing the interconnectivity of atmospheric sensors is shown in Figure A-2.

# N<sup>2</sup> Systems Diagram

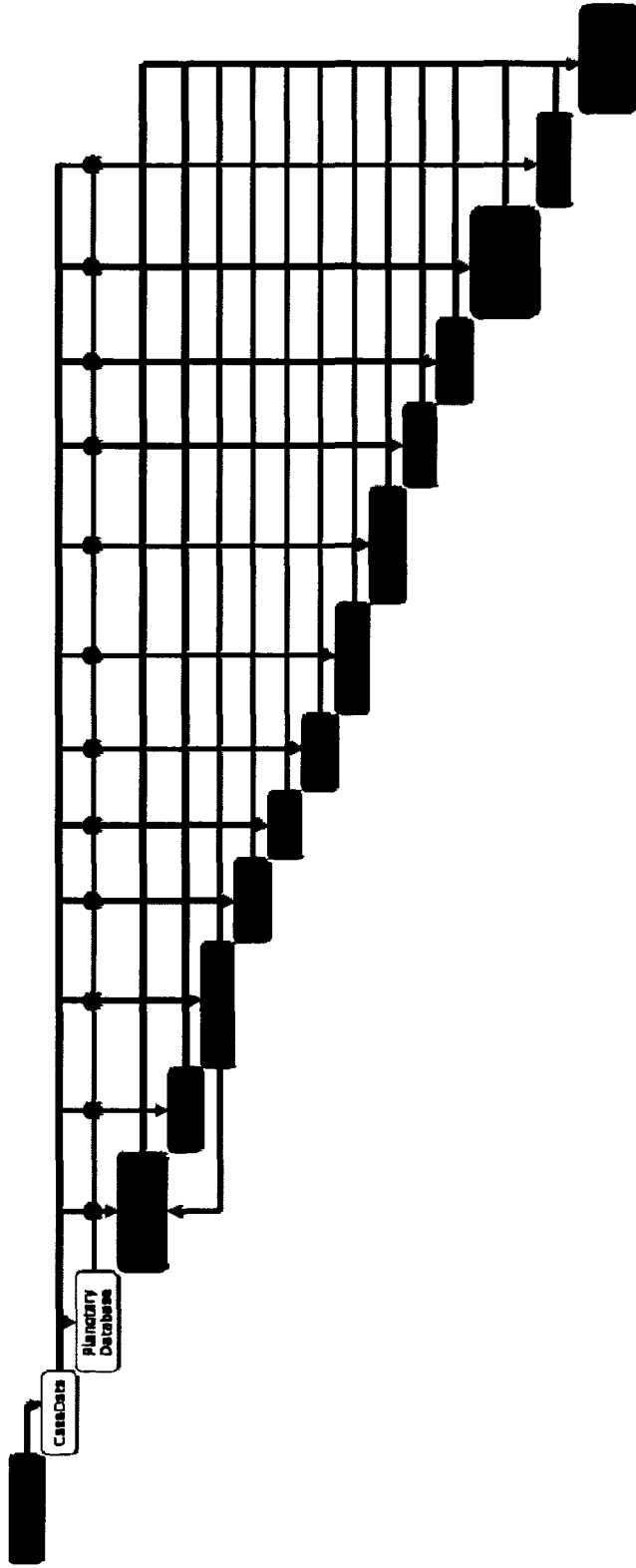


Figure A-1 ISSPO Systems N2 Flowchart

## N2 Diagram Atmospheric Suite

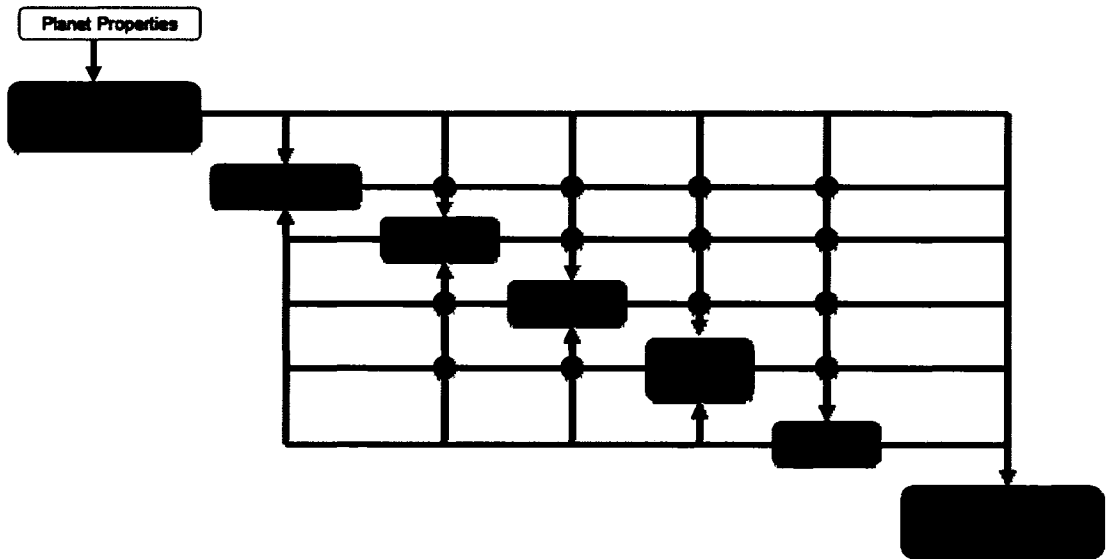


Figure A-2 Atmospheric Suite N<sup>2</sup> Diagram

Circular nodes in the two diagrams indicate a connection between data flow between the different sensor modules. These sensors are often design to survive harsh environments and include the ability to monitor multiple atmospheric parameters. Depending on the environment and desired sensor properties use of a single sensor may prove to a superior option than separate sensors (e.g. pressure sensor monitoring temperature).

## APPENDIX B

### Huygens Probe Benchmark Results

#### 1. Descent Imager / Spectral Radiometer

The resultant input design files for the Descent Imager / Spectral Radiometer package in the ISSPO comparison of the Huygens Mission is detailed in the subsequent data files.

#### Input File

```
function Huygens_DISR
*** Declare and Load all Input Variables ***
BODY = 'Titan';
UNITS = 'SI';
PRINTFLG = 'N';
% 'SI' or 'British' Units
% 'Y' or 'N'
MASS_LIMIT = 8.1; % Max Allowable Sensor Package Wt 'lbm' or 'kg'
POWER_LIMIT = 70; % Max Total Power Available 'BTU's/hr' or 'W'
VOLUME_LIMIT = 0.3; % Max Allowable Volume 'ft^3' or 'm^3'
% Type of Science Data to be returned
SENSOR_DATA = { 'OPTICS' 'ARRAY' 'LOW' 'VISUAL' 'UV' 'NIR' };
```

#### Output File:

```
*****
*** IN - S I T U S E N S O R P A Y L O A D O P T I M I Z A T I O N T O O L ***
*****
*** P R O G R A M I N P U T S ***
User Input Filename: Huygens_DISR
Planetary Body Name: Titan
Unit System: SI
Print Summary Option: N
Payload Mass Limit: 8.100 kg
Payload Volume Limit: 0.3000 m^3
Payload Power Limit: 70.000 W
```

\*\*\*\*\*

SENSOR PACKAGES DESIGNED:

OPTICS

- ARRAY
- LOW
- VISUAL
- UV
- NIR

\*\*\* PAYLOAD SENSOR BREAKDOWN \*\*\*

SENSOR	MASS	VOLUME	POWER
	kg	m <sup>3</sup>	W
OPTICS	0.0000	0.0000	0.0000
SENSOR PAYLOAD TOTAL	0.0000	0.0000	0.0000

\*\*\* CASE LIMITS SUMMARY \*\*\*  
 \*\*\*\*\*  
 CASE MASS LIMIT: SUCCEEDED!

CASE VOLUME LIMIT: SUCCEEDED!

CASE POWER LIMIT: SUCCEEDED!

Designed Mission Sensor Package Meets Mass Limit!  
 Minimum Sensor Requirements met allows for System Redundancy in Components.  
 Designed Mission Sensor Package Meets Volume Limit!  
 Minimum Sensor Requirements met allows for System Redundancy in Components.  
 Designed Mission Sensor Package Meets Power Limit!  
 Minimum Sensor Requirements met allows for System Redundancy in Components.

\*\*\* PLANETARY DATABASE SUMMARY RESULTS \*\*\*

\*\*\* INPUTS \*\*\*

Planetary Body Name: TITAN  
 Unit System: SI

\*\*\* PLANETARY BULK PROPERTIES \*\*\*

Planetary Mass : 1.34550e+023 kg  
 Planetary Volume (10<sup>10</sup>) : 7.15188e+009 m<sup>3</sup>  
 Planetary Equatorial Radius : 2.57500e+006 m  
 Planetary Polar Radius : 2.57500e+006 m

Planetary Gravity : m/s^2  
 Planetary Density : kg/m^2  
 Planetary Ellipticity :  
 Planetary Escape Velocity : m/s  
 Planetary GM : km^3/s^2  
 Planetary Bond Albedo :  
 Planetary Visual Geometric Albedo :  
 Planetary Visual Magnitude :  
 Planetary Solar Irradiance : W/m^2  
 Planetary Black Body Temperature : deg C  
 Planetary Magnetic Field Strength: mGauss  
 Planetary Moment of Inertia :  
 Planetary J2 Parameter :  
 Planetary Satellites : 0

\*\*\* PLANETARY ORBITAL PARAMETERS \*\*\*

Semi-Major Axis : m  
 Sidereal Orbit Period : Earth Days  
 Tropical Orbit Period : Earth Days  
 Perihelion Distance : m  
 Aphelion Distance : m  
 Synodic Period :  
 Mean Orbital Velocity : Earth Days  
 Maximum Orbital Velocity : m/s  
 Minimum Orbital Velocity : m/s  
 Orbital Inclination : deg  
 Orbital Eccentricity :  
 Sidereal Rotation Period : Earth Hours  
 Length of Day : Earth Hours  
 Obliquity to Orbit : deg

\*\*\* ATMOSPHERIC PROPERTIES \*\*\*

Planetary Surface Temperature : -179.45 deg C  
 Planetary Surface Pressure : 1.46700e+005 Pa  
 Planetary Surface Density : 5.50000e+000 kg/m^2

ATMOSPHERIC COMPOSITION

Major Elements

-----  
 Element % Composition  
 N2 98.400 %  
 CH4 1.600 %

Trace Elements

-----  
 Element Amount Concentration

\*\*\* IMAGING SENSOR SUMMARY RESULTS \*\*\*

\*\*\* INPUTS \*\*\*



```

System Type:          ARRAY
Resolution:          LOW
Image Type:          VISUAL
Unit System:        SI
                    UV      NIR
*****
** Sensor Properties **
Imaging Sensor Type: CCD 424
Imaging Spectrum:   X-RAY
Array Dimensions - Length: 1024
                   Width:  1024 # Pixels
Sensor Pixel Size:   21.0000 micro-m
Imaging Array Size - Length: 21.500 mm
                   Width:  21.500 mm
Read Noise at 1 MHz: 0 e-
                   250 kHz: 0 e-
Full Well Capacity - Pixel: 0 ke-
Register:             500 ke-
Sensor Gain:         0.00 e-/ADU
Sensor Linearity:   N/A
ADC Dynamic Range:  N/A
Readout Rates:     0.0000 MHz
Readout Time:      0.000000 sec
Frame Rates:       0.00 fps

Environmental Properties
Operating Temperature Range - Low: -30.00 deg C
                               High:  100.00 deg C
Cooled Temperature:          0.00 deg C
Cooling Method:              N/A

Physical Properties
Dimensions - Length: 73.15 mm
                Width: 52.83 mm
                Height: 6.10 mm
Sensor Mass:      0.000 kg

Power Requirements
Input Voltage:    24.000 V
Input Voltage Type: DC
Input Amperage:  0.000 mA
Power Requirement: 0.000 mW

Sensor Comments:
Split Frame Transfer CCD Array
2x - 1024 x 530 pixel Storage Areas
Rapid image capture capability
Multiple output image formats
Four-port readout
Low Read Noise
Space Qualified - Deep Impact Mission
Software Gain & Binning
Scientific precision and accuracy

```

## 2. Doppler Wind Experiment (DWE)

The resultant input design files for the Doppler Wind Experiment (DWE) package in the ISSPO comparison of the Huygens Mission is detailed in the subsequent data files. In the ISSPO configuration the Wind Sensor option is part of the Atmospheric suite and runs all components. The other options can be disregarded and look at the Wind Velocity component only.

### Input File:

```
function Huygens_DWE
*** Declare and Load all Input Variables ***
BODY = 'Titan';
UNITS = 'SI';
PRINTFLG = 'N';
% 'SI' or 'British' Units
% 'Y' or 'N'
MASS_LIMIT = 1.9; % Max Allowable Sensor Package Wt 'lbm' or 'kg'
POWER_LIMIT = 18; % Max Total Power Available 'BTU's/hr' or 'W'
VOLUME_LIMIT = .0025; % Max Allowable Volume 'ft^3' or 'm^3'

% Type of Science Data to be returned
SENSOR_DATA = { 'ATMOSPHERE', 'WIND VELOCITY', 'DOPPLER' };
```

### Output File:

```
*****
*** I N - S I T U S E N S O R P A Y L O A D O P T I M I Z A T I O N T O O L ***
*****

*** P R O G R A M I N P U T S ***
User Input Filename: Huygens_DWE
Planetary Body Name: Titan
Unit System: SI
Print Summary Option: N

Payload Mass Limit: 1.900 kg
Payload Volume Limit: 0.0025 m3
Payload Power Limit: 18.000 W

*****

SENSOR PACKAGES DESIGNED:
-----
ATMOSPHERE
```

- Temperature
- Pressure
- Wind Velocity
- Humidity
- Density

\*\*\* P A Y L O A D S E N S O R B R E A K D O W N \*\*\*

SENSOR	MASS	VOLUME	POWER
	kg	m <sup>3</sup>	W
ATMOSPHERE			
- Temperature	0.0000	0.0000	0.0000
- Pressure	0.0000	0.0000	0.0250
- Wind Velocity Probe	1.8000	0.0006	35.0000
- Humidity	0.0000	0.0000	0.0000
- Density	5.0000	0.0070	0.8250
SENSOR PAYLOAD TOTAL	5.9000	0.0076	35.8500

\*\*\* CASE LIMITS SUMMARY \*\*\*  
 \*\*\*\*\*  
 CASE MASS LIMIT: FAILED!

Designed Mission Sensor Package EXCEEDS Mass Limits  
 Database Configuration for Required Sensors Exceeds Input Sensor Package Mass Limit  
 Reconfigure Sensor Package for Less Components or Select Multiple Data Type Sensors in

Design.  
 CASE VOLUME LIMIT: FAILED!

Designed Mission Sensor Package EXCEEDS Mass Limits  
 Database Configuration for Required Sensors Exceeds Input Sensor Package Volume Limit  
 Reconfigure Sensor Package for Less Components or Select Multiple Data Type Sensors in

Design.  
 CASE POWER LIMIT: FAILED!

Designed Mission Sensor Package EXCEEDS Power Limits  
 Database Configuration for Required Sensors Exceeds Input Sensor Package Power Limit  
 Total Power Limit Achievable by Selecting Sensor Component Duty Cycles with Mission

Planning.  
 Design.

Reconfigure Sensor Package for Less Components or Select Multiple Data Type Sensors in

\*\*\* P L A N E T A R Y D A T A B A S E S U M M A R Y R E S U L T S \*\*\*

\*\*\* I N P U T S \*\*\*  
 Planetary Body Name: TITAN  
 Unit System: SI

\*\*\*\*\*

\*\*\* PLANETARY BULK PROPERTIES \*\*\*

Planetary Mass : 1.34550e+023 kg  
Planetary Volume (10^10) : 7.15188e+009 m^3  
Planetary Equatorial Radius : 2.57500e+006 m  
Planetary Polar Radius : 2.57500e+006 m  
Planetary Gravity : 1.35200e+000 m/s^2  
Planetary Density : 1.88000e+003 kg/m^2  
Planetary Ellipticity : 0.00000e+000  
Planetary Escape Velocity : 2.63900e+003 m/s  
Planetary GM : 0.00000e+000 km^3/s^2  
Planetary Bond Albedo : 2.20000e-001  
Planetary Visual Geometric Albedo : 2.20000e-001  
Planetary Visual Magnitude : 7.90000e+000  
Planetary Solar Irradiance : 1.50440e+001 W/m^2  
Planetary Black Body Temperature : 0.00000e+000 deg C  
Planetary Magnetic Field Strength : 0.00000e+000 mGauss  
Planetary Moment of Inertia : 0.00000e+000  
Planetary J2 Parameter : 0.00000e+000  
Planetary Satellites : 0

\*\*\* PLANETARY ORBITAL PARAMETERS \*\*\*

Semi-Major Axis : 1.22183e+006 m  
Sidereal Orbit Period : 1.59450e+001 Earth Days  
Tropical Orbit Period : 1.59450e+001 Earth Days  
Perihelion Distance : 0.00000e+000 m  
Aphelion Distance : 0.00000e+000 m  
Synodic Period : 1.59450e+001 Earth Days  
Mean Orbital Velocity : 0.00000e+000 m/s  
Maximum Orbital Velocity : 0.00000e+000 m/s  
Minimum Orbital Velocity : 0.00000e+000 m/s  
Orbital Inclination : 3.30000e-001 deg  
Orbital Eccentricity : 2.92000e-002  
Sidereal Rotation Period : 3.82690e+002 Earth Hours  
Length of Day : 3.82690e+002 Earth Hours  
Obliquity to Orbit : 0.00000e+000 deg

\*\*\* ATMOSPHERIC PROPERTIES \*\*\*

Planetary Surface Temperature : -179.45 deg C  
Planetary Surface Pressure : 1.46700e+005 Pa  
Planetary Surface Density : 5.50000e+000 kg/m^2

ATMOSPHERIC COMPOSITION

Major Elements

-----  
Element % Composition  
N2 98.400 %  
CH4 1.600 %

Trace Elements

-----  
Element Amount Concentration

\*\*\* WIND VELOCITY SENSOR SUMMARY RESULTS \*\*\*

\*\*\* INPUTS \*\*\*

Wind Velocity Sensor Type: DOPPLER  
 Unit System: SI

\*\*\*\*\*

\*\* Sensor Properties \*\*

Wind Velocity Sensor Type: 8130A  
 Sensor Output Frequency: 10.000 MHz  
 Sensor Output Level: 7.000 dBm  
 Output Tolerance: 1.500 dBm  
 Spectral Purity - Harmonics: -30.000 dBc  
 Spectral Purity - Non-Harmonics: -80.000 dBc  
 Aging Data - Month: 5.00E-011 Hz/month  
 Aging Data - 10 yrs: 1.00E-009 10 yrs  
 Sensor Accuracy: 1.00E-011 Hz @ 25 deg C  
 Sensor Stability: 3.00E-012 Hz @ 100 sec  
 Sensor Warm Up Time: 14.000 min  
 Sensor Lifetime: 20.0 years

Environmental Properties

Operating Temperature Range - Low: -40.000 deg C  
 High: 68.000 deg C  
 Storage Temperature Range - Low: -62.000 deg C  
 High: 85.000 deg C  
 Temperature Sensitivity - Op Range: 3.00E-010 Hz  
 Orientation Sensitivity: 5.00E-011 Hz  
 Pressure Sensitivity: 1.00E-013 Hz/mbar

Power Requirements

Voltage Type: DC Type  
 System Voltage Levels: 22.000 32.000 V  
 Warm Up Time Max Power: 35.000 W  
 Steady State Power: 22.000 W  
 Input Power - Quiescent: 12.000 W

Physical Properties

Sensor Mass: 0.900 Kg  
 Sensor Dimensions - Length: 10.260 cm  
 Width: 7.410 cm  
 Height: 7.280 cm

Sensor Comments:

Modern Militarized Design  
 5 and 10 MHz Sinewave Outputs  
 RS-232 Digital Control and Monitoring  
 Ruggedized High Performance Rb Physics Package

Meets many Mil-Spec Standards  
 Data for Single Unit  
 Two Units required for Doppler Tracking  
 Space Qualified Hardware Production capability

### 3. Gas Chromatograph Mass Spectrometer (GCMS)

The resultant input design files for the Gas Chromatograph Mass Spectrometer package in the ISSPO comparison of the Huygens Mission is detailed in the subsequent data files.

#### Input File:

```
function Huygens_GCMS
  *** Declare and Load all Input Variables ***
  BODY = 'Titan';
  UNITS = 'SI';
  PRINTFLG = 'N';
  % 'SI' or 'British' Units
  % 'Y' or 'N'
  MASS_LIMIT = 17.3;
  POWER_LIMIT = 79.0;
  VOLUME_LIMIT = .0185;
  % Max Allowable Sensor Package Wt 'lbm' or 'kg'
  % Max Total Power Available 'BTU's/hr' or 'W'
  % Max Allowable Volume 'ft^3' or 'm^3'

  % Type of Science Data to be returned
  SENSOR_DATA = { 'GAS ANALYSIS' 'MASS-CHARGE' 'MEDIUM' };
};
```

#### Output File:

```
*****
*** I N - S I T U S E N S O R P A Y L O A D O P T I M I Z A T I O N T O O L ***
*****
*** P R O G R A M I N P U T S ***
User Input Filename: Huygens_GCMS
Planetary Body Name: Titan
Unit System: SI
Print Summary Option: N
Payload Mass Limit: 17.300 kg
Payload Volume Limit: 0.0185 m^3
Payload Power Limit: 79.000 W
*****
SENSOR PACKAGES DESIGNED:
-----
```

GAS ANALYSIS

\*\*\* PAYLOAD SENSOR BREAKDOWN \*\*\*

SENSOR	MASS kg	VOLUME m <sup>3</sup>	POWER W
GAS ANALYSIS	17.3000	0.0184	71.0000
SENSOR PAYLOAD TOTAL	17.3000	0.0184	71.0000

\*\*\* CASE LIMITS SUMMARY \*\*\*  
 \*\*\*\*\* SUCCEEDED!  
 CASE MASS LIMIT: SUCCEEDED!  
 CASE VOLUME LIMIT: SUCCEEDED!  
 CASE POWER LIMIT: SUCCEEDED!  
 Designated Mission Sensor Package Meets Mass Limit!  
 Minimum Sensor Requirements met allows for System Redundancy in Components.  
 Designated Mission Sensor Package Meets Volume Limit!  
 Minimum Sensor Requirements met allows for System Redundancy in Components.  
 Designated Mission Sensor Package Meets Power Limit!  
 Minimum Sensor Requirements met allows for System Redundancy in Components.

\*\*\* PLANETARY DATABASE SUMMARY RESULTS \*\*\*

\*\*\* INPUTS \*\*\*  
 Planetary Body Name: TITAN  
 Unit System: SI

\*\*\*\*\*

\*\*\* PLANETARY BULK PROPERTIES \*\*\*  
 Planetary Mass : 1.34550e+023 kg  
 Planetary Volume (10<sup>10</sup>) : 7.15188e+009 m<sup>3</sup>  
 Planetary Equatorial Radius : 2.57500e+006 m  
 Planetary Polar Radius : 2.57500e+006 m  
 Planetary Gravity : 1.35200e+000 m/s<sup>2</sup>  
 Planetary Density : 1.88000e+003 kg/m<sup>3</sup>  
 Planetary Ellipticity : 0.00000e+000  
 Planetary Escape Velocity : 2.63900e+003 m/s  
 Planetary GM : 0.00000e+000 km<sup>3</sup>/s<sup>2</sup>  
 Planetary Bond Albedo : 2.20000e-001  
 Planetary Visual Geometric Albedo : 2.20000e-001  
 Planetary Visual Magnitude : 7.90000e+000  
 Planetary Solar Irradiance : 1.50440e+001 W/m<sup>2</sup>  
 Planetary Black Body Temperature : 0.00000e+000 deg C

Planetary Magnetic Field Strength: 0.00000e+000 mGauss  
 Planetary Moment of Inertia : 0.00000e+000  
 Planetary J2 Parameter : 0.00000e+000  
 Planetary Satellites : 0

\*\*\* PLANETARY ORBITAL PARAMETERS \*\*\*

Semi-Major Axis : 1.22183e+006 m  
 Sidereal Orbit Period : 1.59450e+001 Earth Days  
 Tropical Orbit Period : 1.59450e+001 Earth Days  
 Perihelion Distance : 0.00000e+000 m  
 Aphelion Distance : 0.00000e+000 m  
 Synodic Period : 1.59450e+001 Earth Days  
 Mean Orbital Velocity : 0.00000e+000 m/s  
 Maximum Orbital Velocity : 0.00000e+000 m/s  
 Minimum Orbital Velocity : 0.00000e+000 m/s  
 Orbital Inclination : 3.30000e-001 deg  
 Orbital Eccentricity : 2.92000e-002  
 Sidereal Rotation Period : 3.82690e+002 Earth Hours  
 Length of Day : 3.82690e+002 Earth Hours  
 Obliquity to Orbit : 0.00000e+000 deg

\*\*\* ATMOSPHERIC PROPERTIES \*\*\*

Planetary Surface Temperature : -179.45 deg C  
 Planetary Surface Pressure : 1.46700e+005 Pa  
 Planetary Surface Density : 5.50000e+000 kg/m^2

ATMOSPHERIC COMPOSITION

Major Elements  
 -----  
 Element % Composition  
 N2 98.400 %  
 CH4 1.600 %

Trace Elements

-----  
 Element Amount Concentration

\*\*\* G C M S S E N S O R S U M M A R Y R E S U L T S \*\*\*

\*\*\* I N P U T S \*\*\*  
 GCMS Operational Type: MASS-CHARGE  
 GCMS Operational Range: MEDIUM  
 Unit System: SI

\*\*\*\*\*

\*\* Sensor Properties \*\*  
 GCMS Sensor Type: GCMS-Huygens  
 Sensing Mass/Charge Range - Low: 2.00 amu



High: 141.00 amu

Mass Filter Type: Quadrupole

Detector Operational Type: Electron Multiplier

Dynamic Mass-Charge Scanning Range: g N/A

Scanning Resolution: 0.00 amu

Scanning Sensitivity: 0.00 A/Torr

Minimum Detected Partial Pressure: 0.00 Torr

Operating Pressure: 100.00 Torr

Maximum Operating Temperature: 100.00 deg C

Number of Ion Sources: 5 N/A

Charge Field Range - Low: 1.00 W Degas

High: 16.00 W Degas

Electron Source Voltage - Low: 25.00 V

High: 70.00 V

Ion Energy: 8.00 12.00 V

Electron Source Current - Low: 0.00 mA

High: 0.08 mA

Warm Up Period Mass Stability: 0.00 amu

Time: 36.00 min

Sensor Return Data Rate: 960.00 bits/sec

Physical Properties

Sensor Mass: 17.30 kg

Sensor Dimensions - Length: 47.00 cm

Width: 19.80 cm

Height: 19.80 cm

Power Source Voltage: 28.00 V DC

Power Source Voltage Type: 1.68 A

Power Source Current: 28.00 W

Power Requirements - Typical: 41.00 W

Average: 71.00 W

Peak:

Sensor Comments:

Combined Gas Chromatograph; Mass Spectrometer

Probe design consists of an 5 ion sources, quadrupole mass filter, and 2 Electron Multipliers

Leak Management Holes

Hydrogen gas used as Transport medium in sampling chambers

Huygens Probe GCMS System - Custom Flight Proven configuration

#### 4. Huygens Atmosphere Structure Instrument (HASI)

The resultant input design files for the Huygens Atmosphere Structure Instrument package in the ISSPO comparison of the Huygens Mission is detailed in the subsequent data files. In the ISSPO configuration the Atmospheric suite runs Temperature, Pressure, Wind Velocity, Humidity, and Density. The results for components in the atmospheric suite not part of the HASI suite can be subtracted from the results.

##### Input File:

```
function Huygens_HASI
  %*** Declare and Load all Input Variables ***
  BODY = 'Titan';
  UNITS = 'SI';
  PRINTFLG = 'N';
  % 'SI' or 'British' Units
  % 'Y' or 'N'
  MASS_LIMIT = 6.3;
  POWER_LIMIT = 85;
  VOLUME_LIMIT = 0.15;
  % Max Allowable Sensor Package Wt 'lbm' or 'kg'
  % Max Total Power Available 'BTU's/hr' or 'W'
  % Max Allowable Volume 'ft^3' or 'm^3'

  % Type of Science Data to be returned
  SENSOR_DATA = {
    'ATMOSPHERE', 'TEMPERATURE', 'RESISTANCE' ;
    'EM_FIELD', 'N/A', 'N/A', 'N/A', 'N/A' ;
    'ACCELERATION', 'MEDIUM', 'Const Current' ;
    'ACOUSTICS', 'SENSOR', 'N/A', 'N/A' ;
    'DATA_PROCESSING', 'LOW', '4' ;
  };
endfunction
```

##### Output File:

```
*****
*** IN - S I T U S E N S O R   P A Y L O A D   O P T I M I Z A T I O N   T O O L *****
*****

*** P R O G R A M   I N P U T S ***
User Input Filename:    Huygens_HASI
Planetary Body Name:   Titan
Unit System:           SI
Print Summary Option:  N

Payload Mass Limit:    6.300   kg
Payload Volume Limit:  0.5000  m^3
Payload Power Limit:   85.000  W

*****
```

SENSOR PACKAGES DESIGNED:

- 
- ATMOSPHERE
- Temperature
  - Pressure
  - Wind Velocity
  - Humidity
  - Density
- ACCELERATION
- ACOUSTICS
- DATA PROCESSING

\*\*\* P A Y L O A D S E N S O R B R E A K D O W N \*\*\*

SENSOR	MASS kg	VOLUME m <sup>3</sup>	POWER W
ATMOSPHERE			
- Temperature	0.0004	0.0000	0.0600
- Pressure	0.0000	0.0000	0.0250
- Wind Velocity Probe	0.2270	0.0004	0.0000
Handheld Unit	0.0000	0.0003	0.0000
- Humidity	0.0000	0.0000	0.0000
- Density	5.0000	0.0070	0.8250
ACCELERATION	0.0710	0.0000	0.4800
ACOUSTICS	0.0040	0.0000	0.0000
DATA PROCESSING	0.0000	0.0000	0.7900
SENSOR PAYLOAD TOTAL	5.3023	0.0077	2.1800

\*\*\* CASE LIMITS SUMMARY \*\*\*  
 \*\*\*\*\* SUCCEEDED!  
 CASE MASS LIMIT: SUCCEEDED!

Designed Mission Sensor Package Meets Mass Limit!  
 Minimum Sensor Requirements met allows for System Redundancy in Components.

CASE VOLUME LIMIT: SUCCEEDED!

Designed Mission Sensor Package Meets Volume Limit!

Minimum Sensor Requirements met allows for System Redundancy in Components.  
 Designed Mission Sensor Package Meets Power Limit!  
 Minimum Sensor Requirements met allows for System Redundancy in Components.

CASE POWER LIMIT: SUCCEEDED!

\*\*\* P L A N E T A R Y   D A T A B A S E   S U M M A R Y   R E S U L T S   \*\*\*

\*\*\* I N P U T S   \*\*\*  
 Planetary Body Name: TITAN  
 Unit System: SI

\*\*\*\*\*

\*\*\* PLANETARY BULK PROPERTIES \*\*\*

Planetary Mass :	1.34550e+023	kg
Planetary Volume (10 <sup>10</sup> ) :	7.15188e+009	m <sup>3</sup>
Planetary Equatorial Radius :	2.57500e+006	m
Planetary Polar Radius :	2.57500e+006	m
Planetary Gravity :	1.35200e+000	m/s <sup>2</sup>
Planetary Density :	1.88000e+003	kg/m <sup>2</sup>
Planetary Ellipticity :	0.00000e+000	
Planetary Escape Velocity :	2.63900e+003	m/s
Planetary GM :	0.00000e+000	km <sup>3</sup> /s <sup>2</sup>
Planetary Bond Albedo :	2.20000e-001	
Planetary Visual Geometric Albedo :	2.20000e-001	
Planetary Visual Magnitude :	7.90000e+000	
Planetary Solar Irradiance :	1.50440e+001	W/m <sup>2</sup>
Planetary Black Body Temperature :	0.00000e+000	deg C
Planetary Magnetic Field Strength:	0.00000e+000	mGauss
Planetary Moment of Inertia :	0.00000e+000	
Planetary J2 Parameter :	0.00000e+000	
Planetary Satellites :	0	

\*\*\* PLANETARY ORBITAL PARAMETERS \*\*\*

Semi-Major Axis :	1.22183e+006	m
Sidereal Orbit Period :	1.59450e+001	Earth Days
Tropical Orbit Period :	1.59450e+001	Earth Days
Perihelion Distance :	0.00000e+000	m
Aphelion Distance :	0.00000e+000	m
Synodic Period :	1.59450e+001	Earth Days
Mean Orbital Velocity :	0.00000e+000	m/s
Maximum Orbital Velocity :	0.00000e+000	m/s
Minimum Orbital Velocity :	0.00000e+000	m/s
Orbital Inclination :	3.30000e-001	deg
Orbital Eccentricity :	2.92000e-002	
Sidereal Rotation Period :	3.82690e+002	Earth Hours
Length of Day :	3.82690e+002	Earth Hours
Obliquity to Orbit :	0.00000e+000	deg

\*\*\* ATMOSPHERIC PPROPERTIES \*\*\*

Planetary Surface Temperature :	-179.45	deg C
Planetary Surface Pressure :	1.46700e+005	Pa
Planetary Surface Density :	5.50000e+000	kg/m <sup>2</sup>

ATMOSPHERIC COMPOSITION

Major Elements

```

-----
Element % Composition
N2      98.400 %
CH4     1.600 %

```

Trace Elements

```

-----
Element      Amount      Concentration

```

\*\*\* TEMPERATURE SENSOR SUMMARY RESULTS \*\*\*

\*\*\* INPUTS \*\*\*

Planetary Surface Temperature: -179.45 deg C  
 Unit System: SI

\*\*\*\*\*

\*\* Sensor Properties \*\*

Thermal Sensor Type: M-0146MD  
 Thermal Sensor Low Limit: -269.000 deg C  
 Thermal Sensor High Limit: 400.000 deg C  
 Power Requirements: 0.060 W  
 Mass Requirements: 0.350 g  
 Dimensions - Length: 45.720 mm  
                   Width: 1.524 mm  
                   Height: 1.524 mm  
 Operating Pressure: 0.000 Pascals  
 Error Tolerance: 0.100 %  
 TRL Level: 0.260 deg C  
                   0 TRL

Thermocouple Construction Materials

Positive Terminal: Platinum +Lead  
 Negative Terminal: Platinum -Lead

EMF Voltage spread over full Temperature Range: 0.000 mV

Sensor Comments:

Resistance type Thermal Sensor. High Reliability, Linear Output, Miniature Size, Wide Temperature Range, Recommended for Space Vehicles.

\*\*\* PRESSURE SENSOR SUMMARY RESULTS \*\*\*

\*\*\* I N P U T S \*\*\*  
 Planetary Surface Pressure: 146700.00 Pa  
 Unit System: SI

\*\*\*\*\*

\*\* Sensor Properties \*\*  
 Pressure Sensor Type: Ser 48-0025  
 Pressure Sensor Accuracy: 0.500 %  
 Pressure Sensor Stability: 0.250 % @ 1 Yr  
 Over Pressure Rated Value: 2.000 N/A  
 Burst Pressure Rating: 5.000 N/A  
 Life Cycles: 1.000E+008 Cycles  
 Sensing Pressure Range - Low: 0.000 kPa  
 High: 172.369 kPa  
 Burst Pressure: 344.738 kPa  
 Proof Pressure: 861.845 kPa

Environmental Properties  
 Maximum Shock Load: 100.000 g's  
 Shock Impulse Time: 0.011 sec  
 Vibration Shock Limit: 20.000 g's  
 Operating Frequency Range - Low: 20.000 Hz  
 High: 2000.000 Hz  
 Operating Humidity Range - Low: 0.000 % RH  
 High: 95.000 % RH  
 Operating Temperature Range - Low: -40.000 deg C  
 High: 104.440 deg C  
 Storage Temperature Range - Low: -51.100 deg C  
 High: 121.110 deg C

Power Requirements  
 Input Voltage: 5.000 V  
 Input Amperage: 5.000 mA  
 Sensing Bandwidth: -3.000 dB

Physical Properties  
 Sensor Material: Stainless Steel  
 Case Material: Stainless Steel  
 Dimensions - Length: 48.560 mm  
 Width: 22.230 mm  
 Height: 22.230 mm  
 Sensor Mass: 0.000 gr

Temperature Sensing Range  
 Lower Sensing Limit: -53.900 deg C  
 Upper Sensing Limit: 121.100 deg C  
 Accuracy: 1.100 deg C

Sensor Comments:  
 Combination Pressure/Temperature Sensor. No silicone oil, no internal O-rings, no welds. Low static / thermal errors.  
 Rugged design for harsh environments. Vacuum Calibration available.

\*\*\* WIND VELOCITY SENSOR SUMMARY RESULTS \*\*\*

\*\*\* INPUTS \*\*\*  
Wind Velocity Sensor Type: ANEMOMETER  
Unit System: SI

\*\*\*\*\*

\*\* Sensor Properties \*\*  
Wind Velocity Sensor Type: HH-30A  
Velocity Sensing Range - Low: 0.2032 m/s  
High: 39.6240 m/s  
Velocity Accuracy: 0.5000 % FS  
Volumetric Flow Rate: NO  
Environmental Properties  
Probe Temperature Range - Low: -20.000 deg C  
High: 100.000 deg C  
Instrument Temperature Range - Low: 0.000 deg C  
High: 50.000 deg C

Power Requirements  
Number of Batteries: 2 N/A  
Battery Type: AA Type

Physical Properties  
Sensor Mass: 0.227 kg  
Handheld Dimensions - Length: 180.340 mm  
Width: 76.200 mm  
Height: 20.320 mm  
Probe Dimensions - Length: 124.000 mm  
Width: 70.000 mm  
Height: 41.000 mm

Sensor Comments:  
Compact Design. 3 Extension Rods. 2 or 16 sec Averaging Period. 1"/2.75" Probe Diameters.

\*\*\* HUMIDITY SENSOR SUMMARY RESULTS \*\*\*

\*\*\* INPUTS \*\*\*  
Unit System: SI

\*\*\* WARNING IN HUMIDITY SENSOR PROGRAM \*\*\*  
Humidity Sensor Not Required In Sensor Package Design!!  
Water Component not found in Selected Planetary Atmosphere.

\*\*\*\*\*

```

** Sensor Properties **
Humidity Sensor Type:
Humidity Sensing Range - Low:
High:
Capacitance at 5% RH - Low:
Typical:
Max:
Sensor Sensitivity:
Sensor Hysteresis:
Response Time:
Sensor Stability:

Environmental Properties
Temperature Range - Low:
High:
Sensing Frequency Range - Min:
Max:

Physical Properties
Sensor Dimensions - Length:
Width:
Height:

Sensor Mass:

```

```

N/A
0.000 % RH
0.000 % RH
0.000 pF
0.000 pF
0.000 pF
0.000 pF/%RH
0.000 +/-%RH
0.000 sec
0.000 %RH/yr
0.000 deg C
0.000 deg C
0.000 kHz
0.000 kHz
0.000 mm
0.000 mm
0.000 mm
0.000 gr

```

\*\*\* D E N S I T Y S E N S O R S U M M A R Y R E S U L T S \*\*\*

```

*** I N P U T S ***
Density Sensor Operating Type: GAS
Planetary Surface Density: 5.50 kg/m^3
Unit System: SI
*****

```

```

** Sensor Properties **
Density Sensor Type:
Density Sensing Range - Low:
High:
Sensor Accuracy:
Sensor Repeatability:
Sensor Tolerance:
Sensor Viscosity:
Sensing Flowrates:

Environmental Properties
Temperature Range - Low:
High:
Test Pressure:
Max Operating Pressure:

```

```

GDI 7812
0.001 g/cc
4.000 g/cc
0.001000 g/cc
0.0020000 g/cc
0.000600 g/cc
0.000 cP
216.000 ltr/hr
-20.000 deg C
85.000 deg C
0.000 kPa
15000.000 kPa

```



Vibration Loads: 0.000 g's

Physical Properties

Sensor Mass: 5.000 kg  
Sensor Dimensions - Length: 364.000 mm  
Width: 139.000 mm  
Height: 139.000 mm

Power Requirements

Input Voltage: 33.000 V-DC  
Input Amperage: 25.000 mA

Sensor Comments:

Gas blending & Direct measurement of ethylene density - Process Gas Must be dry

\*\*\* A C C E L E R O M E T E R S E N S O R S U M M A R Y R E S U L T S \*\*\*

\*\*\* I N P U T S \*\*\*

Mission Acceleration Profile: MEDIUM  
Accelerometer Power Type: Const Current  
Unit System: SI

\*\*\*\*\*

\*\* Sensor Properties \*\*

Accelerometer Sensor Type: QA2000-030  
Accelerometer Sensing Range - Low: -60.00 g's  
High: 60.00 g's  
Sensor Sensitivity: 0.00 mV/g  
Transverse Sensitivity: 0.00 %  
Operating Frequency Range - Low: 20.00 Hz  
High: 2000.00 Hz  
Sensor Linearity: 0.00 %  
Resonance Frequency: 0.00 Hz  
Shock Limits: 250.00 g's  
Vibration Limits: 15.00 g's

Power Requirements

Input Type:	Const	Current	Type
Current Range - Low:	16.00	mA	
High:	16.00	mA	
Bias Voltage:	28.00	V	
Bias Voltage Type:	DC	Type	
Electrical Noise:	0.30	mG's	
Electrical Power:	0.48	W	

Environmental Properties

Sensor Temperature Range - Low: -55.00 deg C  
High: 95.00 deg C  
Temperature Sensitivity: 0.000 g's/deg C

Physical Properties  
 Sensor Mass: 71.000 gr  
 Sensor Dimensions - Length: 3.350 cm  
 Width: 3.350 cm  
 Height: 2.718 cm

Sensor Comments:  
 Single Axis, Piezoresistive accelerometer. Long-term repeatability and superior reliability. Dual built-in self-test.  
 Flight Proven - Huygens Lander Descent phase HASI.

\*\*\* ACOUSTIC SENSOR SUMMARY RESULTS \*\*\*

\*\*\* INPUTS \*\*\*  
 Unit System: SI

\*\*\*\*\*

\*\* Sensor Properties \*\*

Acoustic Sensor Type: CT-190M  
 Dynamic Sensing Range - Low: 56.80 dB  
 High: 135.00 dB  
 Operating Frequency Range - Low: 10.00 Hz  
 High: 2000.00 Hz  
 Sensitivity at 250 Hz - Range: 0.00 dB  
 Tolerance - Low: 0.00 dB  
 High: 0.00 dB  
 Resonance Frequency: 500000.00 Hz  
 Inherent Noise: 56.80 dB  
 Clipping Limit: 135.00 dB  
 Stability Conditions - Time: 0.00 dB/yr  
 Temperature: 0.00 deg C  
 Relative Humidity: 0.00 %RH

Power Requirements  
 Voltage Range - Min: 10.00 V  
 Max: 15.00 V  
 Nominal Current: 0.00 mA

Environmental Properties

Operational Temperature Range - Low: -195.50 deg C  
 High: 37.00 deg C  
 Storage Temperature Range - Low: -195.50 deg C  
 High: 120.00 deg C  
 Temperature Coefficient at 250 Hz: 0.00 dB/deg C  
 Pressure Coefficient at 250 Hz: 0.00 dB/kpa

Physical Properties

Sensor Dimensions - Length: 34.300 mm  
 Width: 9.500 mm

Sensor Mass: 9.500 mm  
4.000 Gram

Sensor Comments:  
Pressure Sensor. Excellent for extreme conditions - temperature, pressure. Not suited for quiet sounds. Flight Proven - Huygens Lander.

\*\*\* DIGITAL SIGNAL PROCESSING SUMMARY RESULTS \*\*\*

\*\*\* INPUTS \*\*\*  
CPU Operating Speed: LOW  
Unit System: SI  
\*\*\*\*\*

\*\* Sensor Properties \*\*  
Digital Signal Processing Unit Type: ADSP-2100  
CPU Operating Speed: LOW  
CPU Clock Speed: 8.192 MHz  
Number of CPU Processing Cores: 1 #  
Amount of On-Board Data Memory: 16.0 K-Words  
Number of Data Bins: 16 #  
Amount of On-Board Program Memory: 32.0 K-Words  
Number of Memory Bins: 24 #  
Number of Computational Units: 3 #  
CPU Cycle Time: 80.000 ns  
Number of External Interrupts: 4 #

Power Requirements  
CPU Output Power: 790.000 mW  
Input Voltage - Min: -0.300 V  
Max: 7.000 V  
Voltage Type: DC N/A  
Input Current: 100.000 mA

Physical Properties  
Processor Core Mass: 0.00 kg  
Sensor Dimensions - Length: 33.830 mm  
Width: 33.830 mm  
Height: 9.120 mm

Environmental Properties  
Operating Temperature Range - Low: -55.00 deg C  
High: 125.00 deg C  
Storage Temperature Range - Low: -65.00 deg C  
High: 150.00 deg C

Sensor Comments:  
Dual Purpose Program Memory for Both Instruction and Data Storage  
Three Independent Computational Units: ALU, Multiplier/Accumulator and Barrel Shifter

Two Independent Data Address Shifters  
Powerful Program Sequencer  
Internal Instruction Cache  
Provisions for Multiprecision Computation and Saturation Logic  
Single-Cycle Instruction Execution  
Multifunction Instructions

APPLICATIONS

- Optimized for DSP Algorithms including: Digital Filtering, Fast Fourier Transforms
- Image Processing
- Radar, Sonar
- Speech Processing
- Telecommunications

## 5. Surface Science Package (SSP)

The resultant input design files for the Surface Science Package in the ISSPO comparison of the Huygens Mission is detailed in the subsequent data files. In the ISSPO configuration the Atmospheric suite runs Temperature, Pressure, Wind Velocity, Humidity, and Density. The results for components in the atmospheric suite not part of the SSP can be subtracted from the results.

### Input File:

```
function Huygens_SSP
*** Declare and Load all Input Variables ***
BODY = 'Titan';
UNITS = 'SI';
PRINTFLG = 'N';
% 'SI' or 'British' Units
% 'Y' or 'N'
MASS_LIMIT = 3.9; % Max Allowable Sensor Package Wt 'lbm' or 'kg'
POWER_LIMIT = 11; % Max Total Power Available 'BTU's/hr' or 'W'
VOLUME_LIMIT = 0.1; % Max Allowable Volume 'ft^3' or 'm^3'

% Type of Science Data to be returned
SENSOR_DATA = {
    'ATMOSPHERE', 'TEMPERATURE', 'RESISTANCE', ;
    'INCLINATION', 'HIGH', 'N/A', ;
    'ACCELERATION', 'IMPACT', 'Self Generating', ;
    'ACOUSTICS', 'VELOCITY', 'N/A', ;
    'REFRACTION', 'N/A', 'N/A', ;
};
```

### Output File:

```
*****
*** J N - S I T U S E N S O R P A Y L O A D O P T I M I Z A T I O N T O O L ***
*****
*** P R O G R A M I N P U T S ***
User Input Filename: Huygens_SSP
Planetary Body Name: Titan
Unit System: SI
Print Summary Option: N
Payload Mass Limit: 3.900 kg
Payload Volume Limit: 0.1000 m^3
Payload Power Limit: 11.000 W
*****
```

SENSOR PACKAGES DESIGNED:

- ATMOSPHERE
- Temperature
- Pressure
- Wind Velocity
- Humidity
- Density
- INCLINATION
- ACCELERATION
- ACOUSTICS
- REFRACTION

\*\*\* P A Y L O A D S E N S O R B R E A K D O W N \*\*\*

SENSOR	MASS kg	VOLUME m <sup>3</sup>	POWER W
ATMOSPHERE			
- Temperature	0.0004	0.0000	0.0600
- Pressure	0.0000	0.0000	0.0250
- Wind Velocity Probe	0.2270	0.0004	0.0000
- Handheld Unit	0.0000	0.0003	0.0000
- Humidity	0.0000	0.0000	0.0000
- Density	5.0000	0.0070	0.8250
INCLINATION	0.0000	0.0000	0.0000
ACCELERATION	0.0270	0.0000	0.0000
ACOUSTICS	0.0320	0.0000	0.0000
REFRACTION	0.0000	0.0012	10.0000
SENSOR PAYLOAD TOTAL	5.2864	0.0089	10.9100

\*\*\* CASE LIMITS SUMMARY \*\*\*  
 \*\*\*\*\*

CASE MASS LIMIT: FAILED!

Design. Mission Sensor Package EXCEEDS Mass Limits  
 Database Configuration for Required Sensors Exceeds Input Sensor Package Mass Limit  
 Reconfigure Sensor Package for Less Components or Select Multiple Data Type Sensors in

CASE VOLUME LIMIT: SUCCEEDED!

Design. Mission Sensor Package Meets Volume Limit!  
 Minimum Sensor Requirements met allows for System Redundancy in Components.

CASE POWER LIMIT: SUCCEEDED!

Design. Mission Sensor Package Meets Power Limit!  
 Minimum Sensor Requirements met allows for System Redundancy in Components.

\*\*\* PLANETARY DATABASE SUMMARY RESULTS \*\*\*

\*\*\* INPUTS \*\*\*  
 Planetary Body Name: TITAN  
 Unit System: SI

\*\*\*\*\*

\*\*\* PLANETARY BULK PROPERTIES \*\*\*

Planetary Mass :	1.34550e+023	kg
Planetary Volume (10 <sup>10</sup> ) :	7.15188e+009	m <sup>3</sup>
Planetary Equatorial Radius :	2.57500e+006	m
Planetary Polar Radius :	2.57500e+006	m
Planetary Gravity :	1.35200e+000	m/s <sup>2</sup>
Planetary Density :	1.88000e+003	kg/m <sup>3</sup>
Planetary Ellipticity :	0.00000e+000	
Planetary Escape Velocity :	2.63900e+003	m/s
Planetary GM :	0.00000e+000	km <sup>3</sup> /s <sup>2</sup>
Planetary Bond Albedo :	2.20000e-001	
Planetary Visual Geometric Albedo :	2.20000e-001	
Planetary Visual Magnitude :	7.90000e+000	
Planetary Solar Irradiance :	1.50440e+001	W/m <sup>2</sup>
Planetary Black Body Temperature :	0.00000e+000	deg C
Planetary Magnetic Field Strength:	0.00000e+000	mGauss
Planetary Moment of Inertia :	0.00000e+000	
Planetary J2 Parameter :	0.00000e+000	
Planetary Satellites :	0	

\*\*\* PLANETARY ORBITAL PARAMETERS \*\*\*

Semi-Major Axis :	1.22183e+006	m
Sidereal Orbit Period :	1.59450e+001	Earth Days
Tropical Orbit Period :	1.59450e+001	Earth Days
Perihelion Distance :	0.00000e+000	m
Aphelion Distance :	0.00000e+000	m
Synodic Period :	1.59450e+001	Earth Days
Mean Orbital Velocity :	0.00000e+000	m/s
Maximum Orbital Velocity :	0.00000e+000	m/s
Minimum Orbital Velocity :	0.00000e+000	m/s
Orbital Inclination :	3.30000e-001	deg
Orbital Eccentricity :	2.92000e-002	
Sidereal Rotation Period :	3.82690e+002	Earth Hours

Length of Day : 3.82690e+002 Earth Hours  
Obliquity to Orbit : 0.000000e+000 deg

\*\*\* ATMOSPHERIC PROPERTIES \*\*\*  
Planetary Surface Temperature : -179.45 deg C  
Planetary Surface Pressure : 1.46700e+005 Pa  
Planetary Surface Density : 5.50000e+000 kg/m^2

ATMOSPHERIC COMPOSITION

Major Elements

-----  
Element % Composition  
N2 98.400 %  
CH4 1.600 %

Trace Elements

-----  
Element Amount Concentration

\*\*\* TEMPERATURE SENSOR SUMMARY RESULTS \*\*\*

\*\*\* INPUTS \*\*\*  
Planetary Surface Temperature: -179.45 deg C  
Unit System: SI

\*\*\*\*\*

\*\* Sensor Properties \*\*

Thermal Sensor Type: M-0146MD  
Thermal Sensor Low Limit: -269.000 deg C  
Thermal Sensor High Limit: 400.000 deg C  
Power Requirements: 0.060 W  
Mass Requirements: 0.350 g  
Dimensions - Length: 45.720 mm  
Width: 1.524 mm  
Height: 1.524 mm  
Operating Pressure: 0.000 Pascals  
Error Tolerance: 0.100 %  
TRL Level: 0.260 deg C  
TRL 0 TRL

Thermocouple Construction Materials

Positive Terminal: Platinum +Lead  
Negative Terminal: Platinum -Lead

EMF Voltage spread over full Temperature Range: 0.000 mV

Sensor Comments:



Resistance type Thermal Sensor. High Reliability, Linear Output, Miniature Size, Wide Temperature Range, Recommended for Space Vehicles.

\*\*\* P R E S S U R E S E N S O R S U M M A R Y R E S U L T S \*\*\*

\*\*\* I N P U T S \*\*\*  
 Planetary Surface Pressure: 146700.00 Pa  
 Unit System: SI

\*\*\*\*\*

\*\* Sensor Properties \*\*  
 Pressure Sensor Type: Ser 48-0025  
 Pressure Sensor Accuracy: 0.500 %  
 Pressure Sensor Stability: 0.250 % @ 1 yr  
 Over Pressure Rated Value: 2.000 N/A  
 Burst Pressure Rating: 5.000 N/A  
 Life Cycles: 1.000E+008 Cycles  
 Sensing Pressure Range - Low: 0.000 kPa  
 High : 172.369 kPa  
 Proof Pressure: 344.738 kPa  
 Burst Pressure: 861.845 kPa

Environmental Properties

Maximum Shock Load: 100.000 G's  
 Shock Impulse Time: 0.011 sec  
 Vibration Shock Limit: 20.000 G's  
 Operating Frequency Range - Low: 20.000 Hz  
 High: 2000.000 Hz  
 Operating Humidity Range - Low: 0.000 % RH  
 High: 95.000 % RH  
 Operating Temperature Range - Low: -40.000 deg C  
 High: 104.440 deg C  
 Storage Temperature Range - Low: -51.100 deg C  
 High: 121.110 deg C

Power Requirements

Input Voltage: 5.000 V  
 Input Amperage: 5.000 mA  
 Sensing Bandwidth: -3.000 dB

Physical Properties

Sensor Material: Stainless Steel  
 Case Material: Stainless Steel  
 Dimensions - Length: 48.560 mm  
 Width: 22.230 mm  
 Height: 22.230 mm  
 Sensor Mass: 0.000 gr

Temperature Sensing Range

Lower Sensing Limit: -53.900 deg C  
 Upper Sensing Limit: 121.100 deg C

Accuracy: 1.100 deg C

Sensor Comments:  
 Combination Pressure/Temperature Sensor. No silicone oil, no internal O-rings, no welds. Low static / thermal errors.  
 Rugged design for harsh environments. Vacuum Calibration available.

\*\*\* WIND VELOCITY SENSOR SUMMARY RESULTS \*\*\*

\*\*\* INPUTS \*\*\*  
 Wind Velocity Sensor Type: ANEMOMETER  
 Unit System: SI

\*\*\*\*\*

\*\* Sensor Properties \*\*  
 Wind Velocity Sensor Type: HH-30A  
 Velocity Sensing Range - Low: 0.2032 m/s  
 High: 39.6240 m/s  
 Velocity Accuracy: 0.5000 % FS  
 Volumetric Flow Rate: NO

Environmental Properties  
 Probe Temperature Range - Low: -20.000 deg C  
 High: 100.000 deg C  
 Instrument Temperature Range - Low: 0.000 deg C  
 High: 50.000 deg C

Power Requirements  
 Number of Batteries: 2 N/A  
 Battery Type: AA Type

Physical Properties  
 Sensor Mass: 0.227 kg  
 Handheld Dimensions - Length: 180.340 mm  
 Width: 76.200 mm  
 Height: 20.320 mm  
 Probe Dimensions - Length: 124.000 mm  
 Width: 70.000 mm  
 Height: 41.000 mm

Sensor Comments:  
 Compact Design. 3 Extension Rods. 2 or 16 sec Averaging Period. 1"/2.75" Probe Diameters.

\*\*\* HUMIDITY SENSOR SUMMARY RESULTS \*\*\*

\*\*\* INPUTS \*\*\*  
 Unit System: SI

\*\*\* WARNING IN HUMIDITY SENSOR PROGRAM \*\*\*  
 Humidity Sensor Not Required In Sensor Package Design!!  
 Water Component not found in Selected Planetary Atmosphere.

\*\*\*\*\*

\*\* Sensor Properties \*\*  
 Humidity Sensor Type: N/A  
 Humidity Sensing Range - Low: 0.000 % RH  
 High: 0.000 % RH  
 Capacitance at 55% RH - Low: 0.000 pF  
 Typical: 0.000 pF  
 Max: 0.000 pF  
 Sensor Sensitivity: 0.000 PF/%RH  
 Sensor Hysteresis: 0.000 +- %RH  
 Response Time: 0.000 sec  
 Sensor Stability: 0.000 %RH/yr

Environmental Properties  
 Temperature Range - Low: 0.000 deg C  
 High: 0.000 deg C  
 Sensing Frequency Range - Min: 0.000 kHz  
 Max: 0.000 kHz

Physical Properties  
 Sensor Dimensions - Length: 0.000 mm  
 Width: 0.000 mm  
 Height: 0.000 mm  
 Sensor Mass: 0.000 gr

\*\*\* D E N S I T Y S E N S O R S U M M A R Y R E S U L T S \*\*\*

\*\*\* I N P U T S \*\*\*  
 Density Sensor Operating Type: GAS  
 Planetary Surface Density: 5.50 kg/m^3  
 Unit System: SI

\*\*\*\*\*

\*\* Sensor Properties \*\*  
 Density Sensor Type: GDT 7812  
 Density Sensing Range - Low: 0.001 g/cc  
 High: 4.000 g/cc  
 Sensor Accuracy: 0.001000 g/cc  
 Sensor Repeatability: 0.0020000 g/cc  
 Sensor Tolerance: 0.000600 g/cc  
 Sensor Viscosity: 0.000 cP  
 Sensing Flowrates: 216.000 ltr/hr

Environmental Properties  
 Temperature Range - Low: -20.000 deg C  
                           High: 85.000 deg C  
 Test Pressure: 0.000 kPa  
 Max Operating Pressure: 15000.000 kPa  
 Vibration Loads: 0.000 g's

Physical Properties  
 Sensor Mass: 5.000 kg  
 Sensor Dimensions - Length: 364.000 mm  
                           Width: 139.000 mm  
                           Height: 139.000 mm

Power Requirements  
 Input Voltage: 33.000 V-DC  
 Input Amperage: 25.000 mA

Sensor Comments:  
 Gas blending & Direct measurement of ethylene density - Process Gas Must be dry

\*\*\* I N C L I N O M E T E R   S E N S O R   S U M M A R Y   R E S U L T S   \*\*\*

\*\*\* I N P U T S \*\*\*  
 Inclination Angle Range: HIGH  
 Unit System: SI

\*\*\*\*\*

\*\* Sensor Properties \*\*  
 Inclinator Sensor Type: L-211U  
 Sensing Inclination Range - Low: -60.000 Deg  
                                   High: 60.000 Deg  
 Output Data Format: Analog mV  
 Sensor Resolution: 0.0083000 Deg  
 Sensor Sensitivity: 100.000 mV/deg  
 Sensor Non-Linearity: 3.3300 %  
 Response Time: 0.000 sec  
 Sensor Bandwidth: 0.00 Hz

Environmental Properties  
 Operating Temperature Range - Low: -54.000 deg C  
                                   High: 124.000 deg C  
 Storage Temperature Range - Low: -54.000 deg C  
                                   High: 124.000 deg C

Power Requirements  
 Input Voltage Type: AC  
 Input Voltage: 0 V  
                                   N/A

Physical Properties  
Dimensions - Length: 41.000 mm  
Width: 18.400 mm  
Height: 15.900 mm  
Sensor Mass: 0.000 gram

Sensor Comments:  
Two Single Axis Sensors Mounted on Single Unit  
Wide angular range, high accuracy, dynamic output attenuation  
L Series incorporates a specially designed dampening orifice  
Flight Proven Product - Cassini-Huygens Mission Huygens Probe  
Includes hermetic sealing, compact size, and are available in a variety of housing configurations  
Applications - Industrial, Aerospace, Military, Photonics, Geotechnical, Oceanographic, Construction.  
Flight Rated Production Company

\*\*\* A C C E L E R O M E T E R S E N S O R S U M M A R Y R E S U L T S \*\*\*

\*\*\* I N P U T S \*\*\*  
Mission Acceleration Profile: IMPACT  
Accelerometer Power Type: Self Generating  
Unit System: SI

\*\*\*\*\*  
\*\* Sensor Properties \*\*  
Accelerometer Sensor Type: 2271A/AM20  
Accelerometer Sensing Range - Low: 0.00 g's  
High: 10000.00 g's  
Sensor Sensitivity: 0.00 mV/g  
Transverse Sensitivity: 3.00 %  
Operating Frequency Range - Low: 0.50 Hz  
High: 7000.00 Hz  
Sensor Linearity: 1.00 %  
Resonance Frequency: 27000.00 Hz  
Shock Limits: 10000.00 g's  
Vibration Limits: 1000.00 g's

Power Requirements  
Input Type: Self Generating Type  
Current Range - Low: 0.00 mA  
High: 0.00 mA  
Bias Voltage: 0.00 V  
Bias Voltage Type: Charge Type  
Electrical Noise: 0.30 mG's  
Electrical Power: 0.00 W

Environmental Properties  
Sensor Temperature Range - Low: -269.00 deg C  
High: 260.00 deg C  
Temperature Sensitivity: 0.000 g's/deg C

Physical Properties  
 Sensor Mass: 27.000 gr  
 Sensor Dimensions - Length: 2.920 cm  
 Width: 1.588 cm  
 Height: 1.588 cm

Sensor Comments:  
 Single Axis, wide temperature range piezoelectric accelerometer for cryogenic applications. Self generating power device.  
 Flight Proven - Huygens Lander SSP.

\*\*\* A C O U S T I C S E N S O R S U M M A R Y R E S U L T S \*\*\*

\*\*\* I N P U T S \*\*\*  
 Unit System: SI  
 \*\*\*\*\*

\*\* Sensor Properties \*\*  
 Acoustic Sensor Type: R80-Alpha  
 Dynamic Sensing Range - Low: 30.00 dB  
 High: 62.00 dB  
 Operating Frequency Range - Low: 200000.00 Hz  
 High: 1000000.00 Hz  
 Sensitivity at 250 Hz - Range: 0.00 dB  
 Tolerance - Low: 0.00 dB  
 High: 0.00 dB  
 Resonance Frequency: 800000.00 Hz  
 Inherent Noise: 58.00 dB  
 Clipping Limit: 120.00 dB  
 Stability Conditions - Time: 0.00 dB/yr  
 Temperature: 0.00 deg C  
 Relative Humidity: 0.00 %RH

Power Requirements  
 Voltage Range - Min: 0.00 V  
 Max: 0.00 V  
 Nominal Current: 0.00 mA

Environmental Properties  
 Operational Temperature Range - Low: -65.00 deg C  
 High: 175.00 deg C  
 Storage Temperature Range - Low: -65.00 deg C  
 High: 175.00 deg C  
 Temperature Coefficient at 250 Hz: 0.00 dB/deg C  
 Pressure Coefficient at 250 Hz: 0.00 dB/kPa

Physical Properties  
 Sensor Dimensions - Length: 19.000 mm  
 Width: 19.000 mm

Sensor Mass:                      Height:                      21.400 mm  
 32.000 gram

Sensor Comments:  
 General Purpose Acoustic Sensor. High Frequency sensor. Robust, Reliable Sensor.

\*\*\* REFRACTION SENSOR SUMMARY RESULTS \*\*\*

\*\*\* INPUTS \*\*\*  
 Unit System: SI

\*\*\*\*\*

\*\* Sensor Properties \*\*  
 Refraction Sensor Type:                      REF-Huy  
 Sensing Refraction Range - Min:                      1.25000 N/A  
    Middle:                      1.45000 N/A  
 Sensing Resolution:                      1.0000E-003 RIU  
 Sensing Drift Rate:                      0.00000 RIU  
 Sensor Accuracy:                      1.00000 %

Power Requirements  
 Input Voltage:                      15.000 V DC Type  
 Voltage Type:                      6.660 mA  
 Input Current:                      10.000 mW  
 Input Power:

Physical Properties  
 Sensor Mass:                      0.00 g  
 Sensor Dimensions - Length:                      10.00 cm  
    Width:                      10.00 cm  
    Height:                      11.61 cm

Environmental Properties  
 Temperature Range - Low:                      -40.00 deg C  
    High:                      65.00 deg C

Sensor Comments:  
 Unique Design Solution - Combination CMOS Linear Image Sensor and Sapphire Refraction Prism  
 Self-Scanning Photodiode Array  
 Linear critical-angle refractometer.  
 Flight Proven Hardware configuration - Huygens Lander SSP.

## APPENDIX C

### Venus Atmospheric Properties Mission Design

The attached design file and output file are for the Venus Atmospheric Properties Mission Design. The goal is to better understand the atmospheric properties, and will investigate: Atmospherics, Inclination, Acceleration, EM Field, Radiation, and Optical properties.

#### Input File:

```
function Venus_ATM
%*** Declare and Load all Input Variables ***
BODY = 'Venus';
UNITS = 'SI';
PRINTFLG = 'N';
% 'SI' or 'British' Units
% 'Y' or 'N'
% Max Allowable Sensor Package Wt 'lbm' or 'kg'
% Max Total Power Available 'BTU's/hr' or 'W'
% Max Allowable Volume 'ft^3' or 'm^3'
MASS_LIMIT = 30;
POWER_LIMIT = 200;
VOLUME_LIMIT = 0.50;

% Type of Science Data to be returned
SENSOR_DATA = {
    'ATMOSPHERE', 'WIND VELOCITY', 'ANEMOMETER', 'TEMPERATURE',
    'INCLINATION', 'MEDIUM', 'N/A', 'N/A',
    'ACCELERATION', 'MEDIUM', 'N/A', 'Voltage',
    'EM FIELD', 'N/A', 'N/A', 'N/A',
    'RADIATION', 'Charged Particle', 'N/A', 'N/A',
    'OPTICS', 'ARRAY', 'MEDIUM', 'VISUAL',
    'UV'
};
```

#### Output File:

```
*****
*** I N - S I T U S E N S O R P A Y L O A D O P T I M I Z A T I O N T O O L ***
*****
*** P R O G R A M I N P U T S ***
User Input Filename: Venus_ATM
Planetary Body Name: Venus
Unit System: SI
Print Summary Option: N
Payload Mass Limit: 30.000 kg
Payload Volume Limit: 0.5000 m^3
```



Payload Power Limit: 200.000 W

\*\*\*\*\*

SENSOR PACKAGES DESIGNED:

ATMOSPHERE

- Temperature
- Pressure
- Wind Velocity
- Humidity
- Density

INCLINATION  
ACCELERATION

EM FIELD

RADIATION

- Charged Particle

OPTICS

- ARRAY
- MEDIUM
- VISUAL
- UV
- NIR

\*\*\* P A Y L O A D S E N S O R B R E A K D O W N \*\*\*

SENSOR	MASS kg	VOLUME m <sup>3</sup>	POWER W
ATMOSPHERE			
- Temperature	0.0000	0.0000	0.0000
- Pressure	0.0000	0.0000	0.0300
- Wind Velocity Probe	0.5000	0.0004	0.1000
- Handheld Unit	0.0000	0.0000	0.0000
- Humidity	0.0000	0.0000	0.0000
- Density	5.0000	0.0070	0.8250
INCLINATION	0.0000	0.0000	0.0000
ACCELERATION	0.1417	0.0000	0.6400
EM FIELD Handheld Unit	0.3100 0.7700	0.0004 0.0012	0.0000 0.0000

RADIATION	7.0000	0.0067	15.0000
OPTICS	0.0000	0.0000	0.0000
SENSOR PAYLOAD TOTAL	13.7217	0.0159	16.5950

```

*** CASE LIMITS SUMMARY ***
*****
CASE MASS LIMIT: SUCCEEDED!

CASE VOLUME LIMIT: SUCCEEDED!

CASE POWER LIMIT: SUCCEEDED!

Designed Mission Sensor Package Meets Mass Limit!
Minimum Sensor Requirements met allows for System Redundancy in Components.

Designed Mission Sensor Package Meets Volume Limit!
Minimum Sensor Requirements met allows for System Redundancy in Components.

Designed Mission Sensor Package Meets Power Limit!
Minimum Sensor Requirements met allows for System Redundancy in Components.

```

\*\*\* PLANETARY DATA SUMMARY RESULTS \*\*\*

```

*** INPUTS ***
Planetary Body Name: VENUS
Unit System: SI

```

```

*** PLANETARY BULK PROPERTIES ***
Planetary Mass : 4.86900e+024 kg
Planetary Volume (10^10) : 9.28430e+010 m^3
Planetary Equatorial Radius : 6.05180e+006 m
Planetary Polar Radius : 6.05180e+006 m
Planetary Gravity : 8.87000e+000 m/s^2
Planetary Density : 5.24300e+003 kg/m^2
Planetary Ellipticity : 0.00000e+000
Planetary Escape Velocity : 1.03600e+004 m/s
Planetary GM : 3.24900e+005 km^3/s^2
Planetary Bond Albedo : 7.50000e-001
Planetary Visual Geometric Albedo : 6.50000e-001
Planetary Visual Magnitude : -4.40000e+000
Planetary Solar Irradiance : 2.61390e+003 W/m^2
Planetary Black Body Temperature : 2.31700e+002 deg C
Planetary Magnetic Field Strength: 0.00000e+000 mGauss
Planetary Moment of Inertia : 3.30000e-001
Planetary J2 Parameter : 4.45800e+000

```

Planetary Satellites : 0

\*\*\* PLANETARY ORBITAL PARAMETERS \*\*\*

Semi-Major Axis : 1.08210e+011 m  
Sidereal Orbit Period : 2.24701e+002 Earth Days  
Tropical Orbit Period : 2.24695e+002 Earth Days  
Perihelion Distance : 1.07480e+011 m  
Aphelion Distance : 1.08940e+011 m  
Synodic Period : 5.83920e+002 Earth Days  
Mean Orbital Velocity : 3.50200e+004 m/s  
Maximum Orbital Velocity : 3.52600e+004 m/s  
Minimum Orbital Velocity : 3.47900e+004 m/s  
Orbital Inclination : 3.39000e+000 deg  
Orbital Eccentricity : 6.70000e-002  
Sidereal Rotation Period : -5.83250e+003 Earth Hours  
Length of Day : 2.80200e+003 Earth Hours  
Obliquity to Orbit : 1.77360e+002 deg

\*\*\* ATMOSPHERIC PROPERTIES \*\*\*

Planetary Surface Temperature : 464.00 deg C  
Planetary Surface Pressure : 9.20000e+006 Pa  
Planetary Surface Density : 6.44000e+001 kg/m^2

ATMOSPHERIC COMPOSITION

Major Elements

Element % Composition  
CO2 96.500 %  
N2 3.500 %

Trace Elements

Element	Amount	Concentration
SO2	150.000	ppm
Ar	70.000	ppm
H2O	20.000	ppm
CO	17.000	ppm
He	12.000	ppm
Ne	7.000	ppm

\*\*\* TEMPERATURE SENSOR SUMMARY RESULTS \*\*\*

\*\*\* INPUTS \*\*\*

Planetary Surface Temperature: 464.00 deg C  
Unit System: SI

\*\*\*\*\*

\*\* Sensor Properties \*\*

Thermal Sensor Type: K  
Thermal Sensor Low Limit: -200.000 deg C  
Thermal Sensor High Limit: 1250.000 deg C  
Power Requirements: 0.000 W  
Mass Requirements: 0.000 g  
Dimensions - Length: 0.000 mm  
Width: 0.000 mm  
Height: 0.000 mm  
Operating Pressure: 0.000 Pascals  
Error Tolerance: 0.750 %  
2.200 deg C  
TRL Level: 0 TRL

Thermocouple Construction Materials

Positive Terminal: Ni-Cr +Lead  
Negative Terminal: Ni-Al -Lead

EMF Voltage spread over full Temperature Range: 48.428 mV

Sensor Comments:

Clean Oxidizing and Inert. Limited Use in Vacuum or Reducing. Wide Temperature Range, Most Popular Calibration.

INSULATION MATERIAL PROPERTIES

Insulation Type: GG  
Insulation Lower Limit: -73.000 deg C  
Insulation Upper Limit: 482.000 deg C

Environmental Properties

Abrasion Resistance: Poor  
Flexibility: Good  
Water Submersion: Poor  
Resistance to Solvents: Excellent  
Resistance to Acids: Excellent  
Resistance to Bases: Excellent  
Resistance to Flame: Excellent  
Resistance to Humidity: Fair  
Overall Material Construction: Glass Braid  
Conductor Material Construction: Glass Braid

\*\*\* P R E S S U R E S E N S O R S U M M A R Y R E S U L T S \*\*\*

\*\*\* I N P U T S \*\*\*

Planetary Surface Pressure: 9200000.00 Pa  
Unit System: SI

\*\*\*\*\*

\*\* Sensor Properties \*\*

Pressure Sensor Type: 13 (C,U)2000P

Pressure Sensor Accuracy: 2.000 %  
 Pressure Sensor Stability: 0.600 % @ 1 Yr  
 Over Pressure Rated Value: 3.000 N/A  
 Burst Pressure Rating: 5.000 N/A  
 Life Cycles: 1.000E+006 Cycles  
 Sensing Pressure Range - Low: 0.000 kPa  
 High: 13789.500 kPa  
 Proof Pressure: 41368.500 kPa  
 Burst Pressure: 68947.600 kPa

**Environmental Properties**  
 Maximum Shock Load: 100.000 g's  
 Shock Impulse Time: 0.011 sec  
 Vibration Shock Limit: 10.000 g's  
 Operating Frequency Range - Low: 20.000 Hz  
 High: 2000.000 Hz  
 Operating Humidity Range - Low: 0.000 % RH  
 High: 0.000 % RH  
 Operating Temperature Range - Low: -40.000 deg C  
 High: 125.000 deg C  
 Storage Temperature Range - Low: -40.000 deg C  
 High: 125.000 deg C

**Power Requirements**  
 Input Voltage: 15.000 V  
 Input Amperage: 2.000 mA  
 Sensing Bandwidth: 0.000 dB

**Physical Properties**  
 Sensor Material: Stainless Steel  
 Case Material: Stainless Steel  
 Dimensions - Length: 81.030 mm  
 Width: 19.050 mm  
 Height: 19.050 mm  
 Sensor Mass: 0.000 gr

**Temperature Sensing Range**  
 Lower Sensing Limit: 0.000 deg C  
 Upper Sensing Limit: 0.000 deg C  
 Accuracy: 0.000 deg C

**Sensor Comments:**  
 Low cost, Rugged Isolated Stainless Steel Design. Calibrated and temperature compensated. Oil-isolated housing. Measurement in hostile environments.

\*\*\* W I N D V E L O C I T Y S E N S O R S U M M A R Y R E S U L T S \*\*\*

\*\*\* I N P U T S \*\*\*  
 Wind Velocity Sensor Type: ANEMOMETER  
 Unit System: SI

\*\*\*\*\*

```

** Sensor Properties **
Wind Velocity Sensor Type: FT 702
Velocity Sensing Range - Low: 0.0000 m/s
High: 70.0000 m/s
Velocity Accuracy: 4.0000 % FS
Volumetric Flow Rate: NO

Environmental Properties
Probe Temperature Range - Low: -40.000 deg C
High: 85.000 deg C
Instrument Temperature Range - Low: -40.000 deg C
High: 85.000 deg C

Power Requirements
Number of Batteries: 0 N/A
Battery Type: N/A Type
Voltage Input: 7.000 V
Voltage Type: DC N/A
Current Input: 14.000 mA
Input Power: 100.000 mW

Physical Properties
Sensor Mass: 0.500 kg
Handheld Dimensions - Length: 0.000 mm
Width: 0.000 mm
Height: 0.000 mm
Probe Dimensions - Length: 162.000 mm
Width: 50.000 mm
Height: 50.000 mm

Sensor Comments:
High accuracy wind speed and direction sensing. WIND DIRECTION MEASUREMENT. Compact, unobtrusive solid-state design with no
moving parts

*** H U M I D I T Y S E N S O R S U M M A R Y R E S U L T S ***

*** I N P U T S ***
Unit System: SI

*** WARNING IN HUMIDITY SENSOR PROGRAM ***
Humidity Sensor Not Required In Sensor Package Design!!
Water Component not found in Selected Planetary Atmosphere.

*****

** Sensor Properties **
Humidity Sensor Type: N/A
Humidity Sensing Range - Low: 0.000 % RH
High: 0.000 % RH
Capacitance at 55% RH - Low: 0.000 pF
Typical: 0.000 pF

```

Max: 0.000 PF  
 0.000 PF/%RH  
 Sensor Sensitivity: 0.000 PF/%RH  
 Sensor Hysteresis: +- %RH  
 Response Time: 0.000 sec  
 Sensor Stability: 0.000 %RH/yr  
  
 Environmental Properties  
 Temperature Range - Low: 0.000 deg C  
 High: 0.000 deg C  
 Sensing Frequency Range - Min: 0.000 kHz  
 Max: 0.000 kHz  
  
 Physical Properties  
 Sensor Dimensions - Length: 0.000 mm  
 Width: 0.000 mm  
 Height: 0.000 mm  
 Sensor Mass: 0.000 gr

\*\*\* D E N S I T Y S E N S O R S U M M A R Y R E S U L T S \*\*\*

\*\*\* I N P U T S \*\*\*  
 Density Sensor Operating Type: GAS  
 Planetary Surface Density: 64.40 kg/m<sup>3</sup>  
 Unit System: SI  
 \*\*\*\*\*

\*\* Sensor Properties \*\*  
 Density Sensor Type: GDT 7812  
 Density Sensing Range - Low: 0.001 g/cc  
 High: 4.000 g/cc  
 Sensor Accuracy: 0.001000 g/cc  
 Sensor Repeatability: 0.0020000 g/cc  
 Sensor Tolerance: 0.000600 g/cc  
 Sensor Viscosity: 0.000 CP  
 Sensing Flowrates: 216.000 ltr/hr  
  
 Environmental Properties  
 Temperature Range - Low: -20.000 deg C  
 High: 85.000 deg C  
 Test Pressure: 0.000 kPa  
 Max Operating Pressure: 15000.000 kPa  
 Vibration Loads: 0.000 g's  
  
 Physical Properties  
 Sensor Mass: 5.000 kg  
 Sensor Dimensions - Length: 364.000 mm  
 Width: 139.000 mm  
 Height: 139.000 mm

Power Requirements

Input Voltage: 33.000 V-DC  
Input Amperage: 25.000 mA

Sensor Comments:

Gas blending & Direct measurement of ethylene density - Process Gas Must be dry

\*\*\* I N C L I N O M E T E R S E N S O R S U M M A R Y R E S U L T S \*\*\*

\*\*\* I N P U T S \*\*\*

Inclination Angle Range: MEDIUM  
Unit System: SI

\*\*\*\*\*

\*\* Sensor Properties \*\*

Inclinometer Sensor Type: L-212T  
Sensing Inclination Range - Low: -45.000 Deg  
High: 45.000 Deg  
Output Data Format: Analog mV  
Sensor Resolution: 0.0056000 Deg  
Sensor Sensitivity: 150.000 mV/deg  
Sensor Non-Linearity: 2.2200 %  
Response Time: 0.000 sec  
Sensor Bandwidth: 0.00 Hz

Environmental Properties

Operating Temperature Range - Low: -54.000 deg C  
High: 124.000 deg C  
Storage Temperature Range - Low: -54.000 deg C  
High: 124.000 deg C

Power Requirements

Input Voltage Type:  
Input Voltage: 0 AC N/A V

Physical Properties

Dimensions - Length: 41.000 mm  
Width: 18.400 mm  
Height: 15.900 mm  
Sensor Mass: 0.000 gram

Sensor Comments:

Two Single Axis Sensors Mounted on Single Unit  
Wide angular range, high accuracy, dynamic output attenuation  
L Series incorporates a specially designed dampening orifice  
Includes hermetic sealing, compact size, and are available in a variety of housing configurations  
Applications - Industrial, Aerospace, Military, Photonics, Geotechnical, Oceanographic, Construction.  
Flight Rated Production Company



\*\*\* ACCELEROMETER SENSOR SUMMARY RESULTS \*\*\*

\*\*\* INPUTS \*\*\*

Mission Acceleration Profile: MEDIUM  
 Accelerometer Power Type: Voltage  
 Unit System: SI

\*\*\*\*\*

\*\* Sensor Properties \*\*

Accelerometer Sensor Type: MA15  
 Accelerometer Sensing Range - Low: -50.00 g's  
 High: 50.00 g's  
 Sensor Sensitivity: 0.00 mV/g  
 Transverse Sensitivity: 5.00 %  
 Operating Frequency Range - Low: 2.00 Hz  
 High: 1000.00 Hz  
 Sensor Linearity: 1.00 %  
 Resonance Frequency: 5000.00 Hz  
 Shock Limits: 0.00 g's  
 Vibration Limits: 0.00 g's

Power Requirements

Input Type:	Voltage	Type
Current Range - Low:	4.00	mA
High:	20.00	mA
Bias Voltage:	32.00	V
Bias Voltage Type:	DC	Type
Electrical Noise:	0.30	mG's
Electrical Power:	0.64	W

Environmental Properties

Sensor Temperature Range - Low:	-23.30	deg C
High:	98.90	deg C
Temperature Sensitivity:	0.261	g's/deg C

Physical Properties

Sensor Mass:	141.700	gr
Sensor Dimensions - Length:	5.690	cm
Width:	2.896	cm
Height:	2.896	cm

Sensor Comments:

General Purpose. 4 - 20 mA Proportional to G.

\*\*\* AC FIELD SENSOR SUMMARY RESULTS \*\*\*

\*\*\* INPUTS \*\*\*

Magnetic Field Strength: 0.000 mGauss

Unit System: SI  
\*\*\*\*\*

\*\* Sensor Properties \*\*  
AC Field Sensor Type: TAM - 1 3 N/A  
Sensor Axis Number: 0.00 mg  
AC Field Sensing Range - Low: 1000.00 mg  
High: 0.00 Hz  
Operating Frequency Range - Low: 60.00 Hz  
High: 1.00 %  
Sensor Accuracy: 0 N/A  
Number of Digit Accuracy: 0.0000 sec  
Sampling Time Interval: 10.00 mV/mg  
Sensor Tolerance:

Environmental Properties  
Sensor Temperature Range - Low: -90.000 deg C  
High: 80.000 deg C  
Electronics Temperature Range - Low: -34.000 deg C  
High: 71.000 deg C

Physical Properties  
Sensor Dimensions - Length: 4.760 cm  
Width: 6.680 cm  
Height: 11.900 cm  
Sensor Mass: 310.000 gram  
Electronics Dimensions - Length: 5.230 cm  
Width: 14.020 cm  
Height: 16.380 cm  
Sensor Mass: 770.000 gram

Power Requirements  
Input Voltage: 9.000 V  
Voltage Type: DC N/A

Sensor Comments:  
TAM-1 series parts meet NASA and military programs (MIL-STD-975). Fully Space Qualified

\*\*\* R A D I A T I O N S E N S O R S U M M A R Y R E S U L T S \*\*\*

\*\*\* I N P U T S \*\*\*  
Unit System: SI  
\*\*\*\*\*

\*\* Sensor Properties \*\*  
Radiation Sensor Type: LPD  
Radiation Detected: Charged Particle

Proton Energy Range - Low: 1.000 MeV  
                           High: 250.000 MeV  
 Number of Detection Bins: 15 # of Bins  
 Electron Energy Range - Low: 0.500 MeV  
                           High: 20.000 MeV  
 Number of Detection Bins: 7 # of Bins  
 Alpha Particle Range - Low: 6.000 MeV  
                           High: 250.000 MeV  
 Number of Detection Bins: 6 # of Bins  
 Nucleon/Heavy Ion Range - Low: 1.500 MeV  
                           High: 0.000 MeV  
 Number of Detection Bins: 1 # of Bins  
 Beta Particle Range - Low: 0.000 keV  
                           High: 0.000 keV  
 Number of Detection Bins: 0 # of Bins  
 Gamma Particle Range - Low: 0.000 keV  
                           High: 0.000 keV  
 Number of Detection Bins: 0 # of Bins  
 X-Ray Energy Range - Low: 0.000 keV  
                           High: 0.000 keV  
 Number of Detection Bins: 0 # of Bins  
 Particle Detection Rate: 100.000 kcps  
 G-Factor: 0.050 0.100 0.300 cm<sup>2</sup> sr

Physical Properties  
 Sensor Mass: 7.000 kg  
 Dimensions - Length: 150.000 mm  
                   Width: 150.000 mm  
                   Height: 300.000 mm  
 Power Requirement: 15.000 W  
 Sensor Comments:  
 16 Different Operational Modes  
 Excellent Particle Specificity  
 High Level Of Redundancy  
 Radiation Hardened Components  
 Fully Flight Qualified: DRTS-E, MDS-1, and ISS

\*\*\* I M A G I N G S E N S O R S U M M A R Y R E S U L T S \*\*\*

\*\*\* I N P U T S \*\*\*  
 System Type: ARRAY  
 Resolution: MEDIUM  
 Image Type: VISUAL UV NIR  
 Unit System: SI  
 \*\*\*\*\*

\*\* Sensor Properties \*\*  
 Imaging Sensor Type: CCD 3041

	X-RAY	UV	VISUAL	NIR	MICRO
Imaging Spectrum:	2048	# Pixels			
Array Dimensions - Length:	2048	# Pixels			
Width:	15.0000	micro-m			
Sensor Pixel Size:	30.720	mm			
Imaging Array Size - Length:	30.720	mm			
Width:	7	e-			
Read Noise at 1 MHz:	0	e-			
250 kHz:	0	ke-			
Full Well Capacity - Pixel:	750	ke-			
Register:	0.00	e-/ADU			
Sensor Gain:	0.00	%			
Sensor Linearity:	N/A	N/A			
ADC Dynamic Range:	2.0000	1.0000	MHz		
Readout Rates:	1.100000	0.364000	0.180000	sec	
Readout Time:	0.90	2.70	5.60	fps	
Frame Rates:					
Environmental Properties					
Operating Temperature Range - Low:		-100.00	deg C		
High:		40.00	deg C		
Cooled Temperature:		0.00	deg C		
Cooling Method:	N/A	N/A			
Physical Properties					
Dimensions - Length:	45.72	mm			
Width:	36.83	mm			
Height:	2.01	mm			
Sensor Mass:	0.000	kg			
Power Requirements					
Input Voltage:	24.000	V			
Input Voltage Type:	DC	Type			
Input Amperage:	0.000	mA			
Power Requirement:	0.000	mW			
Sensor Comments:					
Back-illuminated CCD Array					
Highest quantum efficiency					
Multiple output image formats					
Four-port readout					
Optimal design for speed and sensitivity					
Dual Digitizer					
Separate, optimized readout channels for lowest noise and highest speed					
Space Qualified - Production Company					
Software Gain & Binning					
Scientific precision and accuracy					



```

if exist(InputFile) == 2
    fprintf('\n\n**Loading Input Design Parameters from File.***\n\n');
    InputFileM = { InputFile '.m' };
else
    cd PRGM_FILES;
    ERROR_PRG(1);
    %cd .;
    return;
end

% Delete Summary File from Prior Case if It Exists
if exist('ISSPO_SUMMARY.txt') == 0
    delete('ISSPO_SUMMARY.txt');
end

% MAKE Case Working Directory
cd WORK_DIR;
if exist(InputFile) == 7
    cd(InputFile);
    delete *;
    cd DIR_FILES;
    delete_*;
    cd .;
    % Changes Directory to WORK DIR
    cd .;
    % Returns to Main Directory
else
    mkdir(InputFile);
    cd(InputFile);
    mkdir DIR_FILES;
    cd .;
end
cd .;
% Return to Main Directory

% Get Program Files from Program Folder
cd PRGM_FILES;
dirpath = [ './WORK_DIR/' InputFile 'DIR_FILES' ];
copyfile('*.*', dirpath);
cd .;
dp2 = [ 'WORK_DIR/' InputFile 'DIR_FILES/InputDeck.m' ];
copyfile(InputFileM, dp2);

% Change Directory to DIR_FILES Working Directory to Run Programs
cd WORK_DIR;
cd(InputFile);
cd DIR_FILES;

% Loads InputFile Parameters from InputDeck
InputDeck;

```

```

Fname = [ InputFile '.mat' ];
if exist(Fname) == 2
    load(Fname);
else
    ERROR_PRG(2);
    return;
end

% Verify InputDeck Program Values
if (strcmpi(UNITS,'SI') == 0) && (strcmpi(UNITS,'British') == 0)
    ERROR_PRG(50);
    return;
end

if (strcmpi(PRINTFLG,'Y') == 0) && (strcmpi(PRINTFLG,'N') == 0)
    ERROR_PRG(51);
    return;
end

% If MASS LIMIT is undefined Call ERROR Program
if exist('MASS_LIMIT') == 0
    ERROR_PRG(53);
    return;
end

% If POWER LIMIT is undefined Call ERROR Program
if exist('POWER_LIMIT') == 0
    ERROR_PRG(54);
    return;
end

% If VOLUME LIMIT is undefined Call ERROR Program
if exist('VOLUME_LIMIT') == 0
    ERROR_PRG(55);
    return;
end

% Sensor Data Verifier Program
% Checks through input Sensors for invalid types
DataTypeERR_FLG = DataTypeVerifier(SENSOR_DATA);

% Error Flag Conditions Calls ERROR Program
if DataTypeERR_FLG == 1
    ERROR_PRG(52);
    return;
end
if DataTypeERR_FLG == 2
    ERROR_PRG(56);
    return;
end
if DataTypeERR_FLG == 3
    ERROR_PRG(57);
    return;
end

```

```
if DataTypeERR_FLG == 4
    ERROR_PRG(58);
return;
end
if DataTypeERR_FLG == 5
    ERROR_PRG(59);
return;
end
if DataTypeERR_FLG == 6
    ERROR_PRG(60);
return;
end
if DataTypeERR_FLG == 7
    ERROR_PRG(61);
return;
end
if DataTypeERR_FLG == 8
    ERROR_PRG(62);
return;
end
if DataTypeERR_FLG == 9
    ERROR_PRG(63);
return;
end
if DataTypeERR_FLG == 10
    ERROR_PRG(64);
return;
end
if DataTypeERR_FLG == 11
    ERROR_PRG(65);
return;
end
if DataTypeERR_FLG == 12
    ERROR_PRG(66);
return;
end
if DataTypeERR_FLG == 13
    ERROR_PRG(67);
return;
end
if DataTypeERR_FLG == 14
    ERROR_PRG(68);
return;
end
if DataTypeERR_FLG == 15
    ERROR_PRG(69);
return;
end
if DataTypeERR_FLG == 16
    ERROR_PRG(70);
return;
end
if DataTypeERR_FLG == 17
    ERROR_PRG(71);
```



```

return;
end
if DataTypeERR_FLG == 18
  ERROR_PRG(72);
return;
end
if DataTypeERR_FLG == 19
  ERROR_PRG(73);
return;
end
if DataTypeERR_FLG == 20
  ERROR_PRG(74);
return;
end
if DataTypeERR_FLG == 21
  ERROR_PRG(75);
return;
end
if DataTypeERR_FLG == 22
  ERROR_PRG(76);
return;
end
if DataTypeERR_FLG == 23
  ERROR_PRG(77);
return;
end
if DataTypeERR_FLG == 24
  ERROR_PRG(78);
return;
end
if DataTypeERR_FLG == 25
  ERROR_PRG(79);
return;
end
if DataTypeERR_FLG == 26
  ERROR_PRG(80);
return;
end
if DataTypeERR_FLG == 27
  ERROR_PRG(81);
return;
end
if DataTypeERR_FLG == 28
  ERROR_PRG(82);
return;
end
if DataTypeERR_FLG == 29
  ERROR_PRG(83);
return;
end
if DataTypeERR_FLG == 30
  ERROR_PRG(84);
return;
end

```

```

% Load Constants Subroutine
Constants(UNITS);
if exist('Constants_DB.mat') == 2
    load('Constants_DB.mat');
else
    ERROR_PRG(3);
    return;
end

% Load Planetary Data File
PlanetaryDatabase(BODY,UNITS,PRINTFLG);
if exist('Planet_Data.mat') == 2
    load('Planet_Data.mat');
else
    ERROR_PRG(4);
    return;
end

% Copies Summary file to Working Directory and Main directory
movefile('PlanetDB_Summary.txt','../');

% Design Loop for Each Sensor Element
for i = 1:size(SENSOR_DATA,1)
    % Runs the Atmospheric Data Sensor Suite Package
    if strcmpi(SENSOR_DATA{i,1}, 'ATMOSPHERE') == 1
        % Prints Program Sensor Header to Screen
        disp(' *** Executing ATMOSPHERIC Suite! ***');

        % Display Program Module Header
        disp(' *** Executing TEMPERATURE Sensor! ***');
        % Load Temperature Sensor Data into program
        % Logic Check for Temperature Sensor Type
        for temp = 2:2:size(SENSOR_DATA,2)
            if strcmpi(SENSOR_DATA{i,temp}, 'TEMPERATURE') == 1
                TEMP_SENS_TYPE = SENSOR_DATA{i,temp+1};
            end
        end
    end
    if exist('TEMP_SENS_TYPE') == 0
        TEMP_SENS_TYPE = 'VOLTAGE';
    end
    % Load Temperature Sensor Data into program
    TEMP_SENSORS(TEMP_SENS_TYPE,Planet_Properties{7},UNITS,BODY,PRINTFLG);
    if exist('TEMP_SENSOR_Data.mat') == 2
        load('TEMP_SENSOR_Data.mat');
        % Verify Sensor Summary File is created Properly
        if exist('TEMPERATURE_SENSOR_Summary.txt') == 2
            % Copies Summary file to Working Directory and Main directory
            movefile('TEMPERATURE_SENSOR_Summary.txt','../');
        else

```

```

        ERROR_PRG(200);
        return;
    end
else
    ERROR_PRG(100);
    return;
end

% Display Program Module Header
disp(' *** Executing PRESSURE SENSOR! ***');
% Load Pressure Sensor Data into Program
PRESSURE_SENSORS(Planet_Properties{8},UNITS,PRINTFLG);
if exist('PRESSURE_SENSOR_Data.mat') == 2
    load('PRESSURE_SENSOR_Data.mat');
% Verify Sensor Summary File is created Properly
if exist('PRESSURE_SENSOR_Summary.txt') == 2
% Copies Summary file to Working Directory and Main directory
    movefile('PRESSURE_SENSOR_Summary.txt','../');
else
    ERROR_PRG(201);
    return;
end
else
    ERROR_PRG(101);
    return;
end

% Display Program Module Header
disp(' *** Executing WIND VELOCITY Sensor! ***');
% Load Wind Velocity Sensor Data into program
% Logic Check for Wind Velocity Sensor Type
for pos = 2:size(SENSOR_DATA,2)
    if strcmpi(SENSOR_DATA{i,pos},'WIND VELOCITY') == 1
        WV_SENS_TYPE = SENSOR_DATA{i,pos+1};
    end
end
if exist('WV_SENS_TYPE') == 0
    WV_SENS_TYPE = 'ANEMOMETER';
end

WIND_VELOCITY_SENSORS(WV_SENS_TYPE,UNITS,PRINTFLG);
if exist('WIND_VELOCITY_SENSOR_Data.mat') == 2
    load('WIND_VELOCITY_SENSOR_Data.mat');
% Verify Sensor Summary File is created Properly
if exist('WIND_VELOCITY_SENSOR_Summary.txt') == 2
% Copies Summary file to Working Directory and Main directory
    movefile('WIND_VELOCITY_SENSOR_Summary.txt','../');
else
    ERROR_PRG(207);
    return;
end
else
    ERROR_PRG(107);
end

```

```

return;
end

% Display Program Module Header
disp(' *** Executing HUMIDITY Sensor! ***');
% Load Humidity Sensor Data into program
HUMIDITY_SENSORS(UNITS,PRINTFLG);
if exist('HUMIDITY_SENSOR_Data.mat') == 2
load('HUMIDITY_SENSOR_Data.mat');
% Verify Sensor Summary File is created Properly
if exist('HUMIDITY_SENSOR_Summary.txt') == 2
% Copies Summary file to Working Directory and Main directory
movefile('HUMIDITY_SENSOR_Summary.txt','../');
else
ERROR_PRG(203);
return;
end
else
ERROR_PRG(103);
return;
end

% Display Program Module Header
disp(' *** Executing DENSITY Sensor! ***');
% Load Density Sensor Data into program
% Logic Check for Density Sensor Type
for temp = 2:2:size(SENSOR_DATA,2)
if strcmpi(SENSOR_DATA{i,temp},'DENSITY') == 1
DENS_SENS_TYPE = SENSOR_DATA{i,temp+1};
end
end
if exist('DENS_SENS_TYPE') == 0
DENS_SENS_TYPE = 'GAS';
end
DENSITY_SENSORS(DENS_SENS_TYPE,Planet_Properties{18},UNITS,PRINTFLG);
if exist('DENSITY_SENSOR_Data.mat') == 2
load('DENSITY_SENSOR_Data.mat');
% Verify Sensor Summary File is created Properly
if exist('DENSITY_SENSOR_Summary.txt') == 2
% Copies Summary file to Working Directory and Main directory
movefile('DENSITY_SENSOR_Summary.txt','../');
else
ERROR_PRG(204);
return;
end
else
ERROR_PRG(104);
return;
end
end

end

```

```

% Ends Atmospheric sensor Suite Package

% Runs the Acoustic Sensor Program
if strcmpi(SENSOR_DATA{i,1}, 'ACOUSTICS') == 1

% Prints Program Sensor Header to Screen
disp(' *** Executing ACOUSTIC Sensor! ***');

% Load Acoustic Sensor Data into program
ACS_SENS_TYPE = SENSOR_DATA{i,2};
ACOUSTIC_SENSORS(ACS_SENS_TYPE, UNITS, PRINTFLG);
if exist('ACOUSTIC_SENSOR_Data.mat') == 2
load('ACOUSTIC_SENSOR_Data.mat');
% Verify Sensor Summary File is created Properly
if exist('ACOUSTIC_SENSOR_Summary.txt') == 2
% Copies Summary file to Working Directory and Main directory
movefile('ACOUSTIC_SENSOR_Summary.txt', './');
else
ERROR_PRG(205);
return;
end
else
ERROR_PRG(105);
return;
end
end

% Runs the EM Field Sensor Program
if strcmpi(SENSOR_DATA{i,1}, 'EM FIELD') == 1

% Prints Program Sensor Header to Screen
disp(' *** Executing EM FIELD Sensor! ***');

% Load AC Field Sensor Data into program
AC_FIELD_SENSORS(Planet.Properties{19}, UNITS, PRINTFLG);
if exist('AC_FIELD_SENSOR_Data.mat') == 2
load('AC_FIELD_SENSOR_Data.mat');
% Verify Sensor Summary File is created Properly
if exist('AC_FIELD_SENSOR_Summary.txt') == 2
% Copies Summary file to Working Directory and Main directory
movefile('AC_FIELD_SENSOR_Summary.txt', './');
else
ERROR_PRG(206);
return;
end
else
ERROR_PRG(106);
return;
end
end
end

```

```

% Runs the Acceleration Sensor Program
if strcmpi(SENSOR_DATA{i,1},'ACCELERATION') == 1

% Prints Program Sensor Header to Screen
disp(' *** Executing ACCELEROMETER Sensor! ***');

%Assigns SENSOR_DATA Inputs to Program Variables
A_TYPE = { SENSOR_DATA{i,2} };
P_TYPE = { SENSOR_DATA{i,3} };
%NUM_ACCS = SENSOR_DATA{i,4}

% Load Accelerometer Sensor Data into program
ACCELEROMETER_SENSORS(A_TYPE,P_TYPE,UNITS,PRINTFLG);
if exist('ACCELEROMETER_SENSOR_Data.mat') == 2
load('ACCELEROMETER_SENSOR_Data.mat');
% Verify Sensor Summary File is created Properly
if exist('ACCELEROMETER_SENSOR_Summary.txt') == 2
% Copies Summary file to Working Directory and Main directory
movefile('ACCELEROMETER_SENSOR_Summary.txt','./');
else
ERROR_PRG(208);
return;
end
else
ERROR_PRG(108);
return;
end
end

% Runs the GCMS Sensor Program
if strcmpi(SENSOR_DATA{i,1},'GAS ANALYSIS') == 1

% Prints Program Sensor Header to Screen
disp(' *** Executing GAS ANALYSIS Sensor! ***');

%Assigns SENSOR_DATA Inputs to Program Variables
GCMS_SENS_TYPE = SENSOR_DATA{i,2};
GCMS_SENS_RANGE = SENSOR_DATA{i,3};
% Load GCMS Sensor Data into program
GCMS_SENSORS(GCMS_SENS_TYPE,GCMS_SENS_RANGE,UNITS,PRINTFLG);
if exist('GCMS_SENSOR_Data.mat') == 2
load('GCMS_SENSOR_Data.mat');
% Verify Sensor Summary File is created Properly
if exist('GCMS_SENSOR_Summary.txt') == 2
% Copies Summary file to Working Directory and Main directory
movefile('GCMS_SENSOR_Summary.txt','./');
else
ERROR_PRG(209);
return;
end
end

```

```

else
ERROR_PRG(109);
return;
end
end

% Runs the Nephelometer Sensor Program
if strcmpi(SENSOR_DATA{i,1},'NEPHELOMETRY')==1
% Load Nephelometer Sensor Data into program
% Prints Program Sensor Header to Screen
disp(' *** Executing NEPHELOMETER Sensor! ***');
NEPHELOMETER_SENSORS(UNITS,PRINTFLG);
if exist('NEPHELOMETER_SENSOR_Data.mat')==2
load('NEPHELOMETER_SENSOR_Data.mat');
% Verify Sensor Summary File is created Properly
if exist('NEPHELOMETER_SENSOR_Summary.txt')==2
% Copies Summary File to Working Directory and Main directory
movefile('NEPHELOMETER_SENSOR_Summary.txt','../');
else
ERROR_PRG(202);
return;
end
else
ERROR_PRG(102);
return;
end
end

% Runs the Radiation Sensor Program
if strcmpi(SENSOR_DATA{i,1},'RADIATION')==1
% Load Radiation Sensor Data into program
% Prints Program Sensor Header to Screen
disp(' *** Executing RADIATION Sensor! ***');
%Assigns SENSOR_DATA Inputs to Program Variables
% Counts the length of the Radiation Type Array till End or sees N/A
cnt = 0;
for pt = 2:size(SENSOR_DATA,2)
if strcmpi(SENSOR_DATA{i,pt},'N/A')==0
cnt = cnt + 1;
else
break;
end
end
RAD = cellstr(SENSOR_DATA{i,2:1+cnt});
RADIATION_SENSORS(RAD,UNITS,PRINTFLG);
if exist('RADIATION_SENSOR_Data.mat')==2
load('RADIATION_SENSOR_Data.mat');
% Verify Sensor Summary File is created Properly

```

```

if exist('RADIATION_SENSOR_Summary.txt') == 2
    % Copies Summary file to Working Directory and Main directory
    movefile('RADIATION_SENSOR_Summary.txt', './');
else
    ERROR_PRG(210);
    return;
end
else
    ERROR_PRG(110);
    return;
end
end

% Runs the Inclinator Sensor Program
if strcmpi(SENSOR_DATA{i,1}, 'INCLINATION') == 1
    % Prints Program Sensor Header to Screen
    disp(' *** Executing INCLINATION Sensor! ***');
    %Assigns SENSOR_DATA Inputs to Program Variables
    INCL_RANGE = { SENSOR_DATA{i,2} };
    % Load Inclinator Sensor Data into program
    INCLINOMETER_SENSORS(INCL_RANGE, UNITS, PRINTFLG);
    if exist('INCLINOMETER_SENSOR_Data.mat') == 2
        load('INCLINOMETER_SENSOR_Data.mat');
        % Verify Sensor Summary File is created Properly
        if exist('INCLINOMETER_SENSOR_Summary.txt') == 2
            % Copies Summary file to Working Directory and Main directory
            movefile('INCLINOMETER_SENSOR_Summary.txt', './');
        else
            ERROR_PRG(211);
            return;
        end
    else
        ERROR_PRG(111);
        return;
    end
end

% Runs the Optical Imaging Sensor Program
if strcmpi(SENSOR_DATA{i,1}, 'OPTICS') == 1
    % Prints Program Sensor Header to Screen
    disp(' *** Executing OPTICS Sensor! ***');
    %Assigns SENSOR_DATA Inputs to Program Variables
    SYS_FUNC_TYPE = SENSOR_DATA{i,2};
    RES = SENSOR_DATA{i,3};
    % Counts the length of the Image Type Array till End or sees N/A
    cntnr = 0;
    cntnr = 1;
    for pt = 4:size(SENSOR_DATA,2)

```



```

if strcmpi(SENSOR_DATA{i,pt},'N/A') == 0
    IMAGE{cntr} = SENSOR_DATA{i,pt};
    cntr = cntr + 1;
else
    break;
end
end
%
IMAGE = cellstr(SENSOR_DATA{i,4:4+cntr});
IMAGING_SENSORS(SYS_FUNC_TYPE,RES,IMAGE_UNITS,PRINTFLG);
if exist('IMAGING_SENSOR_Data.mat') == 2
    load('IMAGING_SENSOR_Data.mat');
% Verify Sensor Summary File is created Properly
if exist('IMAGING_SENSOR_Summary.txt') == 2
% Copies Summary file to Working Directory and Main directory
    movefile('IMAGING_SENSOR_Summary.txt','../');
else
    ERROR_PRG(212);
    return;
end
else
    ERROR_PRG(112);
    return;
end
end
end
%
% Runs the Refraction Sensor Program
if strcmpi(SENSOR_DATA{i,1},'REFRACTION') == 1
% Prints Program Sensor Header to Screen
disp(' *** Executing REFRACTION Sensor! ***');
% Load Refraction Sensor Data into Program
REFRACTION_SENSORS(UNITS,PRINTFLG);
if exist('REFRACTION_SENSOR_Data.mat') == 2
    load('REFRACTION_SENSOR_Data.mat');
% Verify Sensor Summary File is created Properly
if exist('REFRACTION_SENSOR_Summary.txt') == 2
% Copies Summary file to Working Directory and Main directory
    movefile('REFRACTION_SENSOR_Summary.txt','../');
else
    ERROR_PRG(213);
    return;
end
else
    ERROR_PRG(113);
    return;
end
end
end
%
% Runs the Digital Signal Processing Program
if strcmpi(SENSOR_DATA{i,1},'DATA PROCESSING') == 1

```

```

% Prints Program Sensor Header to Screen
disp(' *** Executing DIGITAL SIGNAL PROCESSING! ***');

%Assigns SENSOR_DATA Inputs to Program Variables
DIG_SIG_PROG_TYPE = SENSOR_DATA{i,2};
% Load Digital Signal Processing Data into program
DIGITAL_SIGNAL_PROCESSING(DIG_SIG_PROG_TYPE,UNITS,PRINTFLG);
    if exist('DIGITAL_SIGNAL_PROCESSING_Data.mat') == 2
        load('DIGITAL_SIGNAL_PROCESSING_Data.mat');
    % Verify Sensor Summary File is created Properly
    if exist('DIGITAL_SIGNAL_PROCESSING_Summary.txt') == 2
        % Copies Summary file to Working Directory and Main directory
        movefile('DIGITAL_SIGNAL_PROCESSING_Summary.txt','../');
    else
        ERROR_PRG(214);
        return;
    end
else
    ERROR_PRG(114);
    return;
end
end

end
% Ends FOR Program Design Loop

% Main Program Components After Individual Optimization Programs is Run

% ISSPO Summary Optimization Section

% Determine the number of Sensor Programs Runs
NUM_SENSORS = size(SENSOR_DATA,1);
% Offset for Atmospheric Suite comprised of 5 Sensors
for i = 1:size(SENSOR_DATA,1)
    if strcmpi(SENSOR_DATA{i},'ATMOSPHERE') == 1
        NUM_SENSORS = NUM_SENSORS + 4;
    end
end

% Runs MASS PROPERTIES Program Obtains Sensor Mass and Dimensions
MASS_PROPERTIES(SENSOR_DATA,UNITS);
if exist('MASS_PROPERTIES_Data.mat') == 2
    load('MASS_PROPERTIES_Data.mat');
else
    ERROR_PRG(215);
    return;
end

% Determine Each Sensors Power Requirements
POWER_PROPERTIES(SENSOR_DATA,UNITS);
if exist('POWER_PROPERTIES_Data.mat') == 2
    load('POWER_PROPERTIES_Data.mat');
else

```

```

ERROR_PRG(216);
return;
end

% Determine Total Mass of all Selected Sensors
TOTAL_MASS = sum(MASS_PROPS(:,1));

% Determine Total Volume of all Selected Sensors
for i = 1:size(MASS_PROPS,1)
    VOLUME_SUM(i) = MASS_PROPS(i,2) * MASS_PROPS(i,3) * MASS_PROPS(i,4);
end
TOTAL_VOLUME = sum(VOLUME_SUM);
POWER_TOTAL = sum(POWER_PROPS);

% Determine Volume for Wind Velocity Sensor Handheld Unit if Chosen in Analysis
if exist('MASS_PROPS2') == 1
    TOTAL_MASS = TOTAL_MASS + MASS_PROPS2(1,1);
    VOLUME_SUM2 = MASS_PROPS2(1,2) * MASS_PROPS2(1,3) * MASS_PROPS2(1,4);
    TOTAL_VOLUME = TOTAL_VOLUME + VOLUME_SUM2;
    POWER_SUM2 = 0;
    POWER_TOTAL = POWER_TOTAL + POWER_PROPS2;
end

% Determine Volume for EM Field Sensor Handheld Unit if Chosen in Analysis
if exist('MASS_PROPS3') == 1
    TOTAL_MASS = TOTAL_MASS + MASS_PROPS3(1,1);
    VOLUME_SUM3 = MASS_PROPS3(1,2) * MASS_PROPS3(1,3) * MASS_PROPS3(1,4);
    TOTAL_VOLUME = TOTAL_VOLUME + VOLUME_SUM3;
    POWER_SUM3 = 0;
    POWER_TOTAL = POWER_TOTAL + POWER_PROPS3;
end

% Payload Requirements Logic Check
% Logic Checks for MASS, VOLUME, POWER meet MISSION INPUT Constraints.
if TOTAL_MASS <= MASS_LIMIT
    MASS_FLAG = 'SUCCEEDED!';
    MASS_FLAG_STR = {'Designated Mission Sensor Package Meets Mass Limit!';
                    'Minimum Sensor Requirements met allows for System Redundancy in Components.'};
else
    MASS_FLAG = 'FAILED!';
    MASS_FLAG_STR = {'Designated Mission Sensor Package EXCEEDS Mass Limits';
                    'Database Configuration for Required Sensors Exceeds Input Sensor Package Mass Limit';
                    'Reconfigure Sensor Package for less Components or Select Multiple Data Type Sensors in Design.'};
end

if TOTAL_VOLUME <= VOLUME_LIMIT
    VOLUME_FLAG = 'SUCCEEDED!';
    VOLUME_FLAG_STR = {'Designated Mission Sensor Package Meets Volume Limit!';
                      'Minimum Sensor Requirements met allows for System Redundancy in Components.'};
else
    VOLUME_FLAG = 'FAILED!';

```







```

|');
|-----|-----|-----|-----|
|fprintf('\n|
|fprintf('\n|
|};
|k = k + 1;
|-----|-----|-----|-----|
end
|if strcmpi(SENSOR_DATA{i,1}, 'NEPHELOMETRY') == 1
|fprintf('\n|
|');
|fprintf('\n| %17s | %10.4f | %10.4f
|', SENSOR_DATA{i,1}, MASS_PROPS(k,1), VOLUME_SUM(k), POWER_PROPS(k));
|fprintf('\n|
|');
|fprintf('\n|
|-----|-----|-----|-----|
|k = k + 1;
|-----|-----|-----|-----|
end
|if strcmpi(SENSOR_DATA{i,1}, 'OPTICS') == 1
|fprintf('\n|
|');
|fprintf('\n| %17s | %10.4f | %10.4f
|', SENSOR_DATA{i,1}, MASS_PROPS(k,1), VOLUME_SUM(k), POWER_PROPS(k));
|fprintf('\n|
|');
|fprintf('\n|
|-----|-----|-----|-----|
|k = k + 1;
|-----|-----|-----|-----|
end
|if strcmpi(SENSOR_DATA{i,1}, 'RADIATION') == 1
|fprintf('\n|
|');
|fprintf('\n| %17s | %10.4f | %10.4f
|', SENSOR_DATA{i,1}, MASS_PROPS(k,1), VOLUME_SUM(k), POWER_PROPS(k));
|fprintf('\n|
|');
|fprintf('\n|
|-----|-----|-----|-----|
|k = k + 1;
|-----|-----|-----|-----|
end
|if strcmpi(SENSOR_DATA{i,1}, 'REFRACTION') == 1
|fprintf('\n|
|');
|fprintf('\n| %17s | %10.4f | %10.4f
|', SENSOR_DATA{i,1}, MASS_PROPS(k,1), VOLUME_SUM(k), POWER_PROPS(k));
|fprintf('\n|
|');
|fprintf('\n|
|-----|-----|-----|-----|
|k = k + 1;
|-----|-----|-----|-----|
end
|if strcmpi(SENSOR_DATA{i,1}, 'DATA PROCESSING') == 1
|fprintf('\n|
|');

```

```

fprintf('\n| %17s | %10.4f | %10.4f | %10.4f |
|', SENSOR_DATA{i,l}, MASS_PROPS(k,l), VOLUME_SUM(k), POWER_PROPS(k));
fprintf('\n|');
fprintf('\n|-----|-----|-----|-----|
| k = k + 1;
end
end

fprintf('\n|-----|-----|-----|-----|-----|
|');
fprintf('\n| SENSOR PAYLOAD TOTAL | %10.4f | %10.4f | %10.4f |
|', TOTAL_MASS, TOTAL_VOLUME, POWER_TOTAL);
fprintf('\n|-----|-----|-----|-----|-----|
|');

% PRINT OUT CASE RESULTS FOR MASS VOLUME AND POWER LIMITS
fprintf('\n\n *** CASE LIMITS SUMMARY ***');
fprintf('\n*****');
fprintf('\n CASE MASS LIMIT: %s', MASS_FLAG);
for ln = 1:length(MASS_FLAG_STR)
    fprintf('\n\t\t\t\t\t %s', MASS_FLAG_STR{ln});
end
fprintf('\n CASE VOLUME LIMIT: %s', VOLUME_FLAG);
for ln = 1:length(VOLUME_FLAG_STR)
    fprintf('\n\t\t\t\t\t %s', VOLUME_FLAG_STR{ln});
end
fprintf('\n CASE POWER LIMIT: %s', POWER_FLAG);
for ln = 1:length(POWER_FLAG_STR)
    fprintf('\n\t\t\t\t\t %s', POWER_FLAG_STR{ln});
end

cd ...; % Return to Case Working Directory

% Create ISSPO SUMMARY RESULTS FILE
% ISSPO SUMMARY FILE ID
FISSPO = fopen('ISSPO_SUMMARY.txt', 'w+');

fprintf(FISSPO, '\n *****');
fprintf(FISSPO, '\n *** I N - S I T U S E N S O R P A Y L O A D O P T I M I Z A T I O N T O O L ***');
fprintf(FISSPO, '\n *****');
fprintf(FISSPO, '\n\n *** P R O G R A M I N P U T S ***');
fprintf(FISSPO, '\n User Input Filename: %s', InputFile);
fprintf(FISSPO, '\n Planetary Body Name: %s', BODY);
fprintf(FISSPO, '\n Unit System: %s', UNITS);
fprintf(FISSPO, '\n Print Summary Option: %s', PRINTFLAG);

fprintf(FISSPO, '\n\n Payload Mass Limit: %10.3f %s', MASS_LIMIT, MassUnit);
fprintf(FISSPO, '\n Payload Volume Limit: %10.4f %s', VOLUME_LIMIT, VolUnit);
fprintf(FISSPO, '\n Payload Power Limit: %10.3f %s', POWER_LIMIT, PowerUnit);

```







```

%10.4f fprintf(FISSPO, '\n|         Handheld Unit |         |         |
|         |, MASS_PROPS3(1), VOLUME_SUM3, POWER_PROPS3);
|');
-----|-----
fprintf(FISSPO, '\n|
|         |-----|');
k = k + 1;
end
if strcmpi(SENSOR_DATA{i,1}, 'GAS ANALYSIS') == 1
fprintf(FISSPO, '\n|         |         |         |
|');
fprintf(FISSPO, '\n|         |%17s |         |
|, SENSOR_DATA{i,1}, MASS_PROPS(k,1), VOLUME_SUM(k), POWER_PROPS(k));
|');
-----|-----
fprintf(FISSPO, '\n|
|         |-----|');
k = k + 1;
end
if strcmpi(SENSOR_DATA{i,1}, 'INCLINATION') == 1
fprintf(FISSPO, '\n|         |         |         |
|');
fprintf(FISSPO, '\n|         |%17s |         |%10.4f
|, SENSOR_DATA{i,1}, MASS_PROPS(k,1), VOLUME_SUM(k), POWER_PROPS(k));
|');
-----|-----
fprintf(FISSPO, '\n|
|         |-----|');
k = k + 1;
end
if strcmpi(SENSOR_DATA{i,1}, 'NEPHELOMETRY') == 1
fprintf(FISSPO, '\n|         |         |         |
|');
fprintf(FISSPO, '\n|         |%17s |         |         |%10.4f
|, SENSOR_DATA{i,1}, MASS_PROPS(k,1), VOLUME_SUM(k), POWER_PROPS(k));
|');
-----|-----
fprintf(FISSPO, '\n|
|         |-----|');
k = k + 1;
end
if strcmpi(SENSOR_DATA{i,1}, 'OPTICS') == 1
fprintf(FISSPO, '\n|         |         |         |
|');
fprintf(FISSPO, '\n|         |%17s |         |         |%10.4f
|, SENSOR_DATA{i,1}, MASS_PROPS(k,1), VOLUME_SUM(k), POWER_PROPS(k));
|');
-----|-----
fprintf(FISSPO, '\n|
|         |-----|');
k = k + 1;
end
if strcmpi(SENSOR_DATA{i,1}, 'RADIATION') == 1

```

```

|');
    fprintf(FISSPO, '\n|
    fprintf(FISSPO, '\n|          %17s          |          %10.4f          |
|', SENSOR_DATA{i,1}, MASS_PROPS(k,1), VOLUME_SUM(k), POWER_PROPS(k));
    fprintf(FISSPO, '\n|
|');
    fprintf(FISSPO, '\n|-----|-----|-----|-----|
    k = k + 1;
end
if strcmpi(SENSOR_DATA{i,1}, 'REFRACTION') == 1
    fprintf(FISSPO, '\n|
|');
    fprintf(FISSPO, '\n|          %17s          |          %10.4f          |
|', SENSOR_DATA{i,1}, MASS_PROPS(k,1), VOLUME_SUM(k), POWER_PROPS(k));
    fprintf(FISSPO, '\n|
|');
    fprintf(FISSPO, '\n|-----|-----|-----|-----|
    k = k + 1;
end
if strcmpi(SENSOR_DATA{i,1}, 'DATA PROCESSING') == 1
    fprintf(FISSPO, '\n|
|');
    fprintf(FISSPO, '\n|          %17s          |          %10.4f          |
|', SENSOR_DATA{i,1}, MASS_PROPS(k,1), VOLUME_SUM(k), POWER_PROPS(k));
    fprintf(FISSPO, '\n|
|');
    fprintf(FISSPO, '\n|-----|-----|-----|-----|
    k = k + 1;
end
end

fprintf(FISSPO,
'\n|-----|-----|-----|-----|-----|-----|-----|-----|-----|-----|
=|');
fprintf(FISSPO, '\n|          SENSOR PAYLOAD TOTAL          |          %10.4f          |          %10.4f
|', TOTAL_MASS, TOTAL_VOLUME, POWER_TOTAL);
fprintf(FISSPO,
'\n|-----|-----|-----|-----|-----|-----|-----|-----|-----|-----|
=|');

% PRINT OUT CASE RESULTS FOR MASS VOLUME AND POWER LIMITS
fprintf(FISSPO, '\n\n\n *** CASE LIMITS SUMMARY ***');
fprintf(FISSPO, '\n*****');
fprintf(FISSPO, '\n CASE MASS LIMIT:          %s', MASS_FLAG);
for ln = 1:length(MASS_FLAG_STR)
    fprintf(FISSPO, '\n\t\t\t\t\t %s', MASS_FLAG_STR{ln});
end
fprintf(FISSPO, '\n CASE VOLUME LIMIT:          %s', VOLUME_FLAG);
for ln = 1:length(VOLUME_FLAG_STR)
    fprintf(FISSPO, '\n\t\t\t\t\t %s', VOLUME_FLAG_STR{ln});
end
end

```

```

fprintf(FISSPO, '\n CASE POWER LIMIT: %s', POWER_FLAG);
for ln = 1:length(POWER_FLAG_STR)
    fprintf(FISSPO, '\n\t\t\t\t\t %s', POWER_FLAG_STR{ln});
end

% Planetary Database File ID
FPDB = fopen('PlanetDB_Summary.txt', 'r');

% Prints Planetary Database Information to Summary File
PDB_Data = fread(FPDB);
fwrite(FISSPO, PDB_Data, 'char');
fprintf(FISSPO, '\n\n');

for i = 1:size(SENSOR_DATA, 1)
    switch SENSOR_DATA{i,1}
        case 'ATMOSPHERE'
            %Reads and copies TEMPERATURE SENSOR SUMMARY to ISSPO Summary File
            FIDT = fopen('TEMPERATURE_SENSOR_Summary.txt', 'r');
            DataT = fread(FIDT);
            write(FISSPO, DataT, 'char');
            fprintf(FISSPO, '\n\n');
            fclose(FIDT);

            %Reads and copies PRESSURE SENSOR SUMMARY to ISSPO Summary File
            FIDP = fopen('PRESSURE_SENSOR_Summary.txt', 'r');
            DataP = fread(FIDP);
            write(FISSPO, DataP, 'char');
            fprintf(FISSPO, '\n\n');
            fclose(FIDP);

            %Reads and copies WIND VELOCITY SUMMARY to ISSPO Summary File
            FIDW = fopen('WIND_VELOCITY_SENSOR_Summary.txt', 'r');
            DataW = fread(FIDW);
            write(FISSPO, DataW, 'char');
            fprintf(FISSPO, '\n\n');
            fclose(FIDW);

            %Reads and copies HUMIDITY SUMMARY to ISSPO Summary File
            FIDH = fopen('HUMIDITY_SENSOR_Summary.txt', 'r');
            DataH = fread(FIDH);
            write(FISSPO, DataH, 'char');
            fprintf(FISSPO, '\n\n');
            fclose(FIDH);

            %Reads and copies DENSITY SENSOR SUMMARY to ISSPO Summary File
            FIDD = fopen('DENSITY_SENSOR_Summary.txt', 'r');
            DataD = fread(FIDD);
            write(FISSPO, DataD, 'char');
            fprintf(FISSPO, '\n\n');
            fclose(FIDD);

        case 'ACCELERATION'

```

```

%Reads and copies ACCELERATION SUMMARY to ISSPO Summary File
FIDACC = fopen('ACCELEROMETER_SENSOR_Summary.txt','r');
DataACC = fread(FIDACC);
write(FISSPO,DataACC,'char');
fprintf(FISSPO,'\n\n\n');
fclose(FIDACC);

case 'ACOUSTICS'
%Reads and copies ACOUSTICS SUMMARY to ISSPO Summary File
FIDACS = fopen('ACOUSTICS_SENSOR_Summary.txt','r');
DataACS = fread(FIDACS);
write(FISSPO,DataACS,'char');
fprintf(FISSPO,'\n\n\n');
fclose(FIDACS);

case 'EM FIELD'
%Reads and copies AC FIELD SUMMARY to ISSPO Summary File
FIDACF = fopen('AC_FIELD_SENSOR_Summary.txt','r');
DataACF = fread(FIDACF);
write(FISSPO,DataACF,'char');
fprintf(FISSPO,'\n\n\n');
fclose(FIDACF);

case 'GAS ANALYSIS'
%Reads and copies GCMS SUMMARY to ISSPO Summary File
FIDGC = fopen('GCMS_SENSOR_Summary.txt','r');
DataGC = fread(FIDGC);
write(FISSPO,DataGC,'char');
fprintf(FISSPO,'\n\n\n');
fclose(FIDGC);

case 'INCLINATION'
%Reads and copies INCLINATION SUMMARY to ISSPO Summary File
FIDIN = fopen('INCLINOMETER_SENSOR_Summary.txt','r');
DataIN = fread(FIDIN);
write(FISSPO,DataIN,'char');
fprintf(FISSPO,'\n\n\n');
fclose(FIDIN);

case 'NEPHELOMETRY'
%Reads and copies NEPHELOMETER SUMMARY to ISSPO Summary File
FIDNPH = fopen('NEPHELOMETER_SENSOR_Summary.txt','r');
DataNPH = fread(FIDNPH);
write(FISSPO,DataNPH,'char');
fprintf(FISSPO,'\n\n\n');
fclose(FIDNPH);

case 'OPTICS'
%Reads and copies IMAGING SUMMARY to ISSPO Summary File
FIDIMG = fopen('IMAGING_SENSOR_Summary.txt','r');
DataIMG = fread(FIDIMG);
write(FISSPO,DataIMG,'char');
fprintf(FISSPO,'\n\n\n');
fclose(FIDIMG);

case 'RADIATION'
%Reads and copies RADIATION SUMMARY to ISSPO Summary File
FIDRAD = fopen('RADIATION_SENSOR_Summary.txt','r');
DataRAD = fread(FIDRAD);
write(FISSPO,DataRAD,'char');
fprintf(FISSPO,'\n\n\n');

```

```

fclose(FIDRAD);
case 'REFRACTION'
    %Reads and copies REFRACTION SUMMARY to ISSPO Summary File
    FIDREF = fopen('REFRACTION_SENSOR_Summary.txt','r');
    DataREF = fread(FIDREF);
    write(FISSPO,DataREF,'char');
    fprintf(FISSPO,'\n\n');
    fclose(FIDREF);
case 'DATA PROCESSING'
    %Reads and copies DIGITAL SIGNAL PROCESSING SUMMARY to ISSPO Summary File
    FIDDSP = fopen('DIGITAL_SIGNAL_PROCESSING_Summary.txt','r');
    DataDSP = fread(FIDDSP);
    write(FISSPO,DataDSP,'char');
    fprintf(FISSPO,'\n\n');
    fclose(FIDDSP);
otherwise
    fprintf('\n\nSENSOR_DATA(%d): %s\n', i, SENSOR_DATA{i,1} );
    ERROR_PRG(52);
    return;
end % Ends SWITCH Case Block
end % Ends For loop Block
fclose(FISSPO);

% Moves Summary File to Main Directory
copyfile('ISSPO_SUMMARY.txt', './.');
cd ..; % Return to WORK_DIR
cd ..; % Return to Main Directory

fprintf('\n\n*****\n\n*****\n\n*****');
fprintf('\n ISSPO DESIGN TOOL FINISHED!!');
fprintf('\n*****\n\n*****\n\n*****');

```

## D2. Constants.

```

% function Constants(Units)
% ISSPO Tool Constants
%
% Developed by: Keith Schreck
%              Mechanical and Aerospace Engineering
%              San Jose State University
% Date:       Fall 2007
%
% This file contains Constant data values for the ISSPO program.
% Data here is loaded into the main program to allow for use of common
% properties amongst all the subfunctions. The Values here are conversion
% factors or other needed values to be used by any section of the code.
% Values are assigned to a variable name and saved to a Matlab '.mat' file
% and loaded when calling the ISSPO program.

```

```

% Some of the values are based on the Unit system chosen and are loaded
% in separate sections.

function Constants(UNITS)

if strcmpi(UNITS, 'SI') == 1

    Guniv = 6.67E-11;
    Guniv_lbl = 'Nm^2/kg^2';
    m2m = 1;
    m2m_lbl = 'm/m';
    cm2m = 100;
    cm2m_lbl = 'cm/m';
    mm2m = 1000;
    mm2m_lbl = 'mm/m';
    gr2kg = 1000;
    gr2kg_lbl = 'gram/kg';
    kg2kg = 1;
    kg2kg_lbl = 'kg/kg';
    cm3_to_m3 = 1000000;
    cm3_to_m3_lbl = 'cm^3/m^3';
    kgm3_to_gcm3 = 0.001;
    kgm3_to_gcm3_lbl = 'kg/m^3 / g/cm^3';
    mA2A = 1000;
    mA2A_lbl = 'mA/A';
    mW2W = 1000;
    mW2W_lbl = 'mW/W';
    Press_Cnvt = 1000;
    Press_Cnvt_lbl = 'pa to kPa';
    Temp_C2K = 273.15;
    Temp_C2K_lbl = 'deg C to deg K';
    MassUnit = 'kg';
    VolUnit = 'm^3';
    PowerUnit = 'W';
    DistUnit = 'm';
    GravUnit = 'm/s^2';
    TempUnit = 'deg C';
    PressureUnit = 'Pa';
    DensityUnit = 'kg/m^3';
    VelUnit = 'm/s';
    GMUnit = 'km^3/s^2';
    SolIradUnit = 'W/m^2';
    MagfieldUnit = 'mGauss';

elseif strcmpi(UNITS, 'British') == 1
    Guniv = 3.32E-11;
    Guniv_lbl = 'lbf*ft^2/lb^2';
    ft2mi = 5280;
    ft2mi_lbl = 'ft/mi';
    in2ft = 12;
    in2ft_lbl = 'in/ft';
    oz2lbm = 16;
    oz2lbm_lbl = 'oz/lbm';

```



```

slug2lbm = 32.2;
slug2lbm_lbl = 'slug/lbm';
in3_to_ft3 = 1728;
in3_to_ft3_lbl = 'in^3/ft^3';
lbft3_to_lbft3 = 1;
lbft3_to_lbft3_lbl = 'lb/ft^3 / lb/ft^3';
mAZA = 1000;
mAZA_lbl = 'mA/A';
BTU_hr2W = 3.412141;
BTU_hr2W_lbl = 'BTU/hr/W';
Press_Cnvt = 1;
Press_Cnvt_lbl = 'psi to psi';
Temp_F2R = 459.67;
Temp_F2R_lbl = 'deg F to deg R';
MassUnit = 'lbm';
VolUnit = 'ft^3';
PowerUnit = 'BTU''s/hr';
DistUnit = 'mi';
GravUnit = 'ft/s^2';
TempUnit = 'deg F';
PressureUnit = 'psi';
DensityUnit = 'lbm/ft^3';
VelUnit = 'ft/s';
GMUnit = 'ft^3/s^2';
SolIradUnit = 'BTUs/h*ft^2';
MagFieldUnit = 'mGauss';

else
    fprintf('\n\nERROR: Unknown Unit Type!');
    fprintf('\n\nTerminating Program. . .');
end

save('Constants_DB.mat');

```

### D3. Data Type Verifier

```

% function DataVerifier(DataArray)
% ISSPC Tool Sensor Data Type Verifier
%
% Developed by: Keith Schreck
%               Mechanical and Aerospace Engineering
%               San Jose State University
% Date:        Fall 2007
%
% This file verifies the entered Sensor Data Types for the ISSPO program.
% The requested Sensor Data Types are entered into a cell string array.
% This program verifies that the entered types correspond to a valid
% sensor type to eliminate typographical errors or that an invalid sensor
% type was asked for. The program only generates an error message if it

```

```

% finds an unknown data type and sends an error message to the user.
%
function DataTypeERR_FLG = DataTypeVerifier(DataArray);

DataTypeERR_FLG = 0;

for i = 1:size(DataArray,1)
    switch DataArray{i,1}
        case 'ATMOSPHERE'
        case 'ACCELERATION'
        case 'ACOUSTICS'
        case 'EM FIELD'
        case 'GAS ANALYSIS'
        case 'INCLINATION'
        case 'NEPHELOMETRY'
        case 'OPTICS'
        case 'RADIATION'
        case 'REFRACTION'
        case 'DATA PROCESSING'
        otherwise
            fprintf('\nSENSOR_DATA{%d,1}: %s\n', i, DataArray{i,1} );
            DataTypeERR_FLG = 1;
        end
    end

%-----
% For Sensors where additional specifications are required
if strcmpi(DataArray{i,1},'ATMOSPHERE') == 1
    if size(DataArray,2) > 1
        for pos = 2:size(DataArray,2)
            if strcmpi(DataArray{i,pos},'WIND VELOCITY') == 1
                if size(DataArray,2) >= pos+1
                    if strcmpi(DataArray{i,pos+1},'ANEMOMETER') ~= 1
                        fprintf('\nSENSOR_DATA{%d,%d}: %s\n', i, pos+1, DataArray{i,pos+1} );
                        DataTypeERR_FLG = 15;
                    end
                    return;
                end
            else
                DataTypeERR_FLG = 14;
                return;
            end
        end
    end
    if strcmpi(DataArray{i,pos},'TEMPERATURE') == 1
        if size(DataArray,2) >= pos+1
            if strcmpi(DataArray{i,pos+1},'VOLTAGE') ~= 1
                if strcmpi(DataArray{i,pos+1},'RESISTANCE') ~= 1
                    fprintf('\nSENSOR_DATA{%d,%d}: %s\n', i, pos+1, DataArray{i,pos+1} );
                    DataTypeERR_FLG = 22;
                end
            end
        end
    else
        DataTypeERR_FLG = 21;
        return;
    end
end
end
end
end

```

```

if strcmpi(DataArray{i,pos}, 'DENSITY') == 1
    if size(DataArray,2) >= pos+1
        if strcmpi(DataArray{i,pos+1}, 'GAS') ~= 1 && strcmpi(DataArray{i,pos+1}, 'LIQUID') ~= 1)
            fprintf('\n\nSENSOR_DATA{%d,%d}: %s\n', i, pos+1, DataArray{i,pos+1});
            DataTypeERR_FLG = 26;
            return;
        end
    else
        DataTypeERR_FLG = 25;
        return;
    end
end
end
end

if strcmpi(DataArray{i,1}, 'GAS ANALYSIS') == 1
    if size(DataArray,2) == 1
        fprintf('\n\nSENSOR_DATA{%d,1}: %s\n', i, DataArray{i,1});
        DataTypeERR_FLG = 2;
        return;
    end
end
if (strcmpi(DataArray{i,2}, 'WAVELENGTH')==1) || (strcmpi(DataArray{i,2}, 'MASS-CHARGE')==1)
    if strcmpi(DataArray{i,2}, 'MASS-CHARGE')==1
        if size(DataArray,2) < 3
            DataTypeERR_FLG = 17;
            return;
        end
    end
end
if (strcmpi(DataArray{i,3}, 'LOW')==1) || (strcmpi(DataArray{i,3}, 'MEDIUM')==1) ||
(strcmpi(DataArray{i,3}, 'HIGH')==1)
    else
        fprintf('\n\nSENSOR_DATA{%d,3}: %s\n\n', i, DataArray{i,3});
        DataTypeERR_FLG = 18;
        return;
    end
end
else
    fprintf('\n\nSENSOR_DATA{%d,2}: %s\n\n', i, DataArray{i,2});
    DataTypeERR_FLG = 16;
    return;
end
end

if strcmpi(DataArray{i,1}, 'OPTICS') == 1
    if size(DataArray,2) == 1
        fprintf('\n\nSENSOR_DATA{%d,1}: %s\n', i, DataArray{i,1});
        DataTypeERR_FLG = 2;
        return;
    end
end
if size(DataArray,2) == 2
    fprintf('\n\nSENSOR_DATA{%d,1}: %s\n', i, DataArray{i,2});

```

```

        DataTypeERR_FLG = 3;
        return;
    end
    if size(DataArray,2) == 3
        fprintf('\n\nSENSOR_DATA{%d,1}: %s\n', i, DataArray{i,3});
        DataTypeERR_FLG = 4;
        return;
    end
    if (strcmpi(DataArray{i,2}, 'CAMERA')==1) || (strcmpi(DataArray{i,2}, 'ARRAY')==1) ||
    (strcmpi(DataArray{i,2}, 'LINEAR')==1)
    else
        fprintf('\n\nSENSOR_DATA{%d,2}: %s\n\n', i, DataArray{i,2});
        DataTypeERR_FLG = 6;
        return;
    end
    if (strcmpi(DataArray{i,3}, 'LOW')==1) || (strcmpi(DataArray{i,3}, 'MEDIUM')==1) || (strcmpi(DataArray{i,3}, 'HIGH')==1)
    else
        fprintf('\n\nSENSOR_DATA{%d,3}: %s\n\n', i, DataArray{i,3});
        DataTypeERR_FLG = 5;
        return;
    end
    for r = 4:size(DataArray,2)
        % Reads Data Array Variables Till end of Array or Encounters N/A
        if strcmpi(DataArray{i,r}, 'N/A') == 0
            if (strcmpi(DataArray{i,r}, 'X-RAY')==1) || (strcmpi(DataArray{i,r}, 'VISUAL')==1) ||
            (strcmpi(DataArray{i,r}, 'UV')==1) || (strcmpi(DataArray{i,r}, 'NIR')==1) || (strcmpi(DataArray{i,r}, 'MICRO')==1)
            else
                fprintf('\n\nSENSOR_DATA{%d,%d}: %s\n\n', i, r, DataArray{i,r});
                DataTypeERR_FLG = 7;
                return;
            end
        else
            break;
        end
    end % Closes For loop
end% Closes OPTICS Sensor Definition Area
% Additional Information Required for RADIATION SENSORS Program
if strcmpi(DataArray{i,1}, 'RADIATION') == 1
    if size(DataArray,2) == 1
        fprintf('\n\nSENSOR_DATA{%d,1}: %s\n', i, DataArray{i,1});
        DataTypeERR_FLG = 8;
        return;
    end
    for r = 2:size(DataArray,2)
        % Reads Data Array Variables Till end of Array or Encounters N/A
        if strcmpi(DataArray{i,r}, 'N/A') == 0

```

```

        if (strcmpi(DataArray{i,r}, 'Charged Particle')==1) || (strcmpi(DataArray{i,r}, 'Alpha')==1) ||
        (strcmpi(DataArray{i,r}, 'Beta')==1) ...
            || (strcmpi(DataArray{i,r}, 'Gamma')==1) || (strcmpi(DataArray{i,r}, 'X-Ray')==1)
        else
            fprintf('\n\nSENSOR_DATA{%d,%d}: %s\n\n', i, r, DataArray{i,r});
            DataTypeERR_FLG = 9;
            return;
        end
    else
        % if N/A is Encountered in DataArray, Stops Reading in Values
        break;
    end
end % Closes For loop

end % Closes RADIATION Sensor Definition Area

% Additional Information Required for INCLINOMETER SENSORS Program
if strcmpi(DataArray{i,1}, 'INCLINATION') == 1
    if size(DataArray,2) == 1
        fprintf('\n\nSENSOR_DATA{%d,1}: %s\n\n', i, DataArray{i,1});
        DataTypeERR_FLG = 10;
        return;
    end
end

if (strcmpi(DataArray{i,2}, 'LOW')==1) || (strcmpi(DataArray{i,2}, 'MEDIUM')==1) || (strcmpi(DataArray{i,2}, 'HIGH')==1)
else
    fprintf('\n\nSENSOR_DATA{%d,2}: %s\n\n', i, DataArray{i,2});
    DataTypeERR_FLG = 11;
    return;
end

end % Closes INCLINOMETER Sensor Definition Area

% Additional Information Required for ACCELEROMETER SENSORS Program
if strcmpi(DataArray{i,1}, 'ACCELERATION') == 1
    if size(DataArray,2) == 1
        fprintf('\n\nSENSOR_DATA{%d,1}: %s\n\n', i, DataArray{i,1});
        DataTypeERR_FLG = 12;
        return;
    end
end

if size(DataArray,2) < 3
    fprintf('\n\nSENSOR_DATA{%d,1}: %s\n\n', i, DataArray{i,1});
    DataTypeERR_FLG = 19;
    return;
end

if (strcmpi(DataArray{i,2}, 'SOFT')==1) || (strcmpi(DataArray{i,2}, 'MEDIUM')==1) || ...
    (strcmpi(DataArray{i,2}, 'HIGH')==1) || (strcmpi(DataArray{i,2}, 'IMPACT')==1) || ...
    (strcmpi(DataArray{i,2}, 'BALLISTIC')==1)
else
    fprintf('\n\nSENSOR_DATA{%d,2}: %s\n\n', i, DataArray{i,2});
end

```

```

        DataTypeERR_FLG = 13;
        return;
    end
    if (strcmpi(DataArray{i,3}, 'Voltage')==1) || (strcmpi(DataArray{i,3}, 'Const Current')==1) || ...
        (strcmpi(DataArray{i,3}, 'Self Generating')==1)
    else
        fprintf('\n\nSENSOR_DATA{%d,3}: %s\n\n', i, DataArray{i,3});
        DataTypeERR_FLG = 20;
        return;
    end

end % Closes ACCELEROMETER Sensor Definition Area

% Additional Information Required for ACOUSTIC SENSORS Program
if strcmpi(DataArray{i,1}, 'ACOUSTICS') == 1
    if size(DataArray,2) == 1
        fprintf('\n\nSENSOR_DATA{%d,1}: %s\n\n', i, DataArray{i,1});
        DataTypeERR_FLG = 23;
        return;
    end

    if (strcmpi(DataArray{i,2}, 'SENSOR')==1) || (strcmpi(DataArray{i,2}, 'ARRAY')==1) ||
        (strcmpi(DataArray{i,2}, 'VELOCITY')==1)
    else
        fprintf('\n\nSENSOR_DATA{%d,2}: %s\n\n', i, DataArray{i,2});
        DataTypeERR_FLG = 24;
        return;
    end

end % Closes ACOUSTIC Sensor Definition Area

% Additional Information Required for Digital Signal Processing Program
if strcmpi(DataArray{i,1}, 'DATA PROCESSING') == 1
    if size(DataArray,2) == 1
        fprintf('\n\nSENSOR_DATA{%d,1}: %s\n\n', i, DataArray{i,1});
        DataTypeERR_FLG = 27;
        return;
    end

    if size(DataArray,2) < 3
        fprintf('\n\nSENSOR_DATA{%d,1}: %s\n\n', i, DataArray{i,1});
        DataTypeERR_FLG = 28;
        return;
    end

    if (strcmpi(DataArray{i,2}, 'LOW')==1) || (strcmpi(DataArray{i,2}, 'MEDIUM')==1) || (strcmpi(DataArray{i,2}, 'HIGH')==1)
    else
        fprintf('\n\nSENSOR_DATA{%d,2}: %s\n\n', i, DataArray{i,2});
        DataTypeERR_FLG = 29;
        return;
    end

end

```

```

if isnumeric(DataArray{i,3}) == 0
    fprintf('\n\nSENSOR_DATA{%d,3}:  %s\n\n', i, DataArray{i,3} );
    DataTypeERR_FLG = 30;
    return;
end

end % Closes DIGITAL SIGNAL PROCESSING Sensor Definition Area

end

% return;

```

## D4. Planetary Database

```

% function PlanetaryDatabase(Body, Units, Print)
%
% Developed by: Keith Schreck
% Mechanical and Aerospace Engineering
% San Jose State University
% Date: Fall 2007
%
% This Program is a Planetary Database Model for the ISSPO Tool. Given a
% Planetary Name and a Unit System (S.I. or British) it will return the
% associated planetary properties.
%
% Program can be evaluated in two modes, either part of the ISSPO Tool or
% as a standalone tool to return Planetary Body data. The Program is
% either called from the MATLAB command line via:
% PlanetaryDatabase('Planet Name', 'Unit System', 'Print')
% where, Planet Name is it name of the object under scrutiny, and
% Unit System, is the desired Unit System either 'SI', or 'British'.
% Print is 'Y', or 'N' whether or not to print results to the screen.
% when running ISSPO, Print is 'N' to suppress intermediate printouts,
% but when running PlanetaryDatabase as a standalone program, the
% user will probably want the results printed to the screen.
%
% If these values are not input at the command line the program will then
% prompt the user for the proper inputs before continuing.
%
% References
% http://nssdc.gsfc.nasa.gov/planetary/planetfact.html

function PlanetaryDatabase(varargin)

% Setup for Program
%clc;
%clear all;
%close all;
format compact;

% Define Program Input variables

```





```

% Does not Print Anything does Not exit
else
  fprintf('\n\n ***ERROR: Invalid Number of Input Parameters');
  fprintf('\n\n Terminating Program Operation . . .');
  return;
end

if (nargin == 3) && (strcmpi(Print,'Y') == 1)
  fprintf('\n\n *** Loading Input Program Values ***\n\n');
end

if nargin == 0
  fprintf('\n*** Enter Program Values ***');
  fprintf('\nPlanetary Bodies Available to Analyze:');
  fprintf('\n 0 - Sun \n 1 - Mercury \n 2 - Venus \n 3 - Earth \n 4 - Mars');
  fprintf('\n 5 - Jupiter \n Jovian Satellites \n 5.1 - Io \n 5.2 - Europa');
  fprintf('\n 5.3 - Ganymede \n 5.4 - Callisto \n 5.5 - Metis \n 5.6 -Adrastea');
  fprintf('\n 5.7 - Amalthea \n 5.8 - Thebe');
  fprintf('\n 6 - Saturn \n Saturnian Satellites \n 6.1 - Mimas \n 6.2 - Enceladus');
  fprintf('\n 6.3 - Tethys \n 6.4 - Dione \n 6.5 - Rhea \n 6.6 - Titan');
  fprintf('\n 6.7 - Hyperion \n 6.8 - Iapetus');
  fprintf('\n 7 - Uranus \n 8 - Neptune \n 9 - Pluto');
end

while strcmpi(BODY,'default') == 1
  SwitchCase = input('\n\nEnter a Planetary Body Name to Analyze: ','s');

  switch SwitchCase
    case {'Sun','sun'}; BODY = 'Sun'; LOC = 1;
    case {'Mercury','mercury'}; BODY = 'Mercury'; LOC = 2;
    case {'Venus','venus'}; BODY = 'Venus'; LOC = 3;
    case {'Earth','earth'}; BODY = 'Earth'; LOC = 4;
    case {'Moon','moon'}; BODY = 'Moon'; LOC = 5;
    case {'Mars','mars'}; BODY = 'Mars'; LOC = 6;
    case {'Jupiter','jupiter'}; BODY = 'Jupiter'; LOC = 7;
    case {'Io','io'}; BODY = 'Io'; LOC = 8;
    case {'Europa','europa'}; BODY = 'Europa'; LOC = 9;
    case {'Ganymede','ganymede'}; BODY = 'Ganymede'; LOC = 10;
    case {'Callisto','callisto'}; BODY = 'Callisto'; LOC = 11;
    case {'Metis','metis'}; BODY = 'Metis'; LOC = 12;
    case {'Adrastea','adrastea'}; BODY = 'Adrastea'; LOC = 13;
    case {'Amalthea','amalthea'}; BODY = 'Amalthea'; LOC = 14;
    case {'Thebe','thebe'}; BODY = 'Thebe'; LOC = 15;
    case {'Saturn','saturn'}; BODY = 'Saturn'; LOC = 16;
    case {'Mimas','mimas'}; BODY = 'Mimas'; LOC = 17;
    case {'Enceladus','enceladus'}; BODY = 'Enceladus'; LOC = 18;
    case {'Tethys','tethys'}; BODY = 'Tethys'; LOC = 19;
    case {'Dione','dione'}; BODY = 'Dione'; LOC = 20;
    case {'Rhea','rhea'}; BODY = 'Rhea'; LOC = 21;
    case {'Titan','titan'}; BODY = 'Titan'; LOC = 22;
    case {'Hyperion','hyperion'}; BODY = 'Hyperion'; LOC = 23;
    case {'Iapetus','iapetus'}; BODY = 'Iapetus'; LOC = 24;
    case {'Uranus','uranus'}; BODY = 'Uranus'; LOC = 25;
    case {'Neptune','neptune'}; BODY = 'Neptune'; LOC = 26;
  end
end

```

```

case ('Pluto', 'pluto');          BODY = 'Pluto';          LOC = 27;
otherwise
    fprintf('\nERROR: Unknown Planetary Object in Planetary Database Program!\n');
end % Ends Body Declaration Loop

while strcmpi(UNITS,'Default') == 1
    UnitCase = input('\nEnter a Unit System to Use ( 'SI' or 'British' ): ', 's');

    switch UnitCase
        case {'SI', 'si'}
            UNITS = 'SI';
        case {'British', 'british'}
            UNITS = 'British';
        otherwise
            fprintf('\nERROR: Unknown Unit System!\n');
    end % Ends UNITS Declaration Loop
end % Ends UNITS Declaration Loop

PrintCase = input('\nPrint Planetary Summary Results to Screen ( 'Y' or 'N' ): ', 's');

switch PrintCase
    case {'Y', 'y'}
        Print = 'Y';
    case {'N', 'n'}
        N = 'N';
    otherwise
        % Ends Print Declaration Loop
end % Ends Input Section

% Commented out Verification of Inputs for Debugging
% BODY;
% LOC;
% UNITS;
% Print;

% Set Unit Labels
if strcmpi(UNITS,'SI') == 1
    MassUnit = 'kg';
    VolUnit = 'm^3';
    DistUnit = 'm';
    GravUnit = 'm/s^2';
    TempUnit = 'deg C';
    PressureUnit = 'Pa';
    DensityUnit = 'kg/m^2';
    VelUnit = 'm/s';
    GMUnit = 'km^3/s^2';
    SolIradUnit = 'W/m^2';
elseif strcmpi(UNITS,'British') == 1
    MassUnit = 'lbm';
    VolUnit = 'ft^3';
    DistUnit = 'mi';
    GravUnit = 'ft/s^2';

```

```

TempUnit = 'deg F';
PressureUnit = 'psi';
DensityUnit = 'lbm/ft^3';
VelUnit = 'ft/s';
GMUnit = 'ft^3/s^2';
SolIrradUnit = 'BTUs/h*ft^2';

else
end

%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%
%% Planetary Data Base Created Based on Unit System
%% 2 - D Matrix Planet_DB(BODY_LOC, Property)

if strcmpi(UNITS,'SI')==1
% Properties | OBJECT
DENSITY ELLIPTICITY VELOCITY-Esc GM VOLUME RADIUS-EQ GEO-ALBEDO VIS-MAG RADIUS-Pol TEMPERATURE PRESSURE
MAG FIELD Mom-Inertia J2 NATURAL ] BOND-ALBEDO GEO-ALBEDO VIS-MAG SOL-Irradiance BB-TEMP ATM DENSITY
% Units NAME m/s km^3/s^2 N/A N/A m V(1,0) W/m^2 m/s^2 deg C deg K Pascals
kg/m^3 N/A I/MR^2 x10^-6 SATELLITES ] 10^10 m^3 N/A N/A m m m/m^2 m/m^2
mGauss I/MP^2 'SUN' 1.9891E+30 1.41200E+17 6.9600E+08 6.9600E+08 6.9600E+08 5778.00 8.680E+01
1.4080E+03 5.0000E-05 6.1760E+05 1.3271E+11 0.000E+00 0.000E+00 -26.74 0.0000E+00 5.778E+03 0.000E+00
0.00 5.900E-02 0.0000E+00 0 'MERCURY' 3.3022E+23 6.08300E+09 2.4397E+06 2.4397E+06 2.4397E+06 167.00 1.000E-10
5.4270E+03 0.0000E+00 4.3000E+03 2.2030E+04 1.190E-01 1.060E-01 -0.42 9.1266E+03 4.425E+02 0.000E+00
0.00 3.300E-01 6.0000E+01 0 'VENUS' 4.8690E+24 9.28430E+10 6.0518E+06 6.0518E+06 6.0518E+06 464.00 9.200E+06
5.2430E+03 0.0000E+00 1.0360E+04 3.2490E+05 7.500E-01 6.500E-01 -4.40 2.6139E+03 2.317E+02 6.440E+01
0.00 3.300E-01 4.45800E+00 0 'EARTH' 5.9742E+24 1.08321E+11 6.3781E+06 6.3781E+06 6.3781E+06 6.3781E+06 9.800E+00 15.00
1.013E+05 5.5150E+03 3.3500E-03 1.1200E+04 3.9860E+05 3.060E-01 3.670E-01 -3.86 1.3676E+03 2.543E+02
1.225E+00 310.00 3.308E-01 1.08263E+03 1 'MOON' 7.3490E+22 2.19580E+09 1.7381E+06 1.7381E+06 1.7381E+06 1.7360E+06 1.620E+00 -23.15
3.000E-10 3.3500E+03 1.2000E-03 2.3800E+03 4.9000E+03 1.100E-01 1.200E-01 +0.21 1.3676E+03 2.745E+02
0.000E+00 0.00 3.940E-01 2.02700E+02 0 'MARS' 6.4185E+23 1.63180E+10 3.3970E+06 3.3970E+06 3.3970E+06 3.3750E+06 3.710E+00 -63.00
6.360E+02 3.9330E+03 6.4800E-03 5.0300E+03 4.2830E+04 2.500E-01 1.500E-01 -1.52 5.8920E+02 2.101E+02
0.000E+00 0.00 3.660E-01 1.96045E+03 2 'JUPITER' 1.8986E+27 1.43128E+14 7.1492E+07 7.1492E+07 7.1492E+07 6.6854E+07 2.479E+01 -108.00
1.000E+08 1.3260E+03 6.4870E-02 5.9500E+04 1.2669E+08 3.430E-01 5.200E-01 -9.40 5.0500E+01 1.100E+02
0.000E+00 4280.00 2.540E-01 1.47360E+04 63 'IO' 8.9320E+22 2.53191E+09 1.8216E+06 1.8216E+06 1.8216E+06 1.8216E+06 0.000E+00 0.00
0.000E+00 3.5300E+03 0.0000E+00 0.0000E+00 0.0000E+00 0.000E+00 6.200E-01 6.200E-01 0.00 0.0000E+00 0.000E+00
0.000E+00 0.00 'EUROPA' 4.8000E+22 1.59269E+09 1.5608E+06 1.5608E+06 1.5608E+06 0.000E+00 0.00 0.000E+00
3.0100E+03 0.0000E+00 0.0000E+00 0.0000E+00 0.0000E+00 0.000E+00 6.800E-01 6.800E-01 0.00 0.000E+00 0.000E+00
0.00 0.000E+00 0.0000E+00 0 'GANYMEDE' 1.4819E+23 7.63045E+09 2.6312E+06 2.6312E+06 2.6312E+06 0.000E+00 0.00 0.000E+00
1.9400E+03 0.0000E+00 0.0000E+00 0.0000E+00 0.0000E+00 0.000E+00 4.400E-01 4.400E-01 0.00 0.000E+00 0.000E+00
0.00 0.000E+00 0.0000E+00 0 'CALLISTO' 1.0759E+23 5.86546E+09 2.4103E+06 2.4103E+06 2.4103E+06 0.000E+00 0.00 0.000E+00
1.8300E+03 0.0000E+00 0.0000E+00 0.0000E+00 0.0000E+00 0.000E+00 1.900E-01 1.900E-01 0.00 0.0000E+00 0.000E+00
0.00 0.000E+00 0.00600E+00 0 'CALLISTO' 0.0000E+00 0.0000E+00 0.0000E+00 0.0000E+00 0.0000E+00 0.000E+00 0.00 0.000E+00

```

MAX-VEL	MIN-VEL	ORB-INCLIN	SEMIMAJOR-AXIS	SIDEREAL-PERIOD	TROPICAL-PERIOD	PERIHELION	APHELION	SYNODIC-Per	MEAN-VEL
m/s	m/s	deg	m	E. Days	E. Days	m	m	E. Days	m/s
%	%	Units	ECCENTRICITY	ROTATION-Per	LENGTH-DAY	OBLIQUITY to Orbit			
			E. Days	E. Hours	E. Hours	deg			
			N/A						
0.0000E+00	0.0000E+00	0.0000E+00	1.0000E+17	3.35103E+04	2.0000E+04	2.0000E+04	0.000E+00	0.00	0.000E+00
0.00	0.0000E+00	0.0000E+00	0.0000E+00	0.0000E+00	0.000E+00	6.000E-02	0.00	0.0000E+00	0.000E+00
0.0000E+00	0.0000E+00	0.0000E+00	0	% Metis					
0.0000E+00	0.0000E+00	0.0000E+00	2.0000E+16	4.35634E+02	1.3000E+04	8.0000E+03	0.000E+00	0.00	0.000E+00
0.00	0.0000E+00	0.0000E+00	0.0000E+00	0.0000E+00	0.000E+00	1.000E-01	0.00	0.0000E+00	0.000E+00
0.0000E+00	0.0000E+00	0.0000E+00	0	% Adrastea					
3.1000E+03	0.0000E+00	0.0000E+00	7.5000E+18	2.68385E+05	1.3100E+05	6.7000E+04	0.000E+00	0.00	0.000E+00
0.00	0.0000E+00	0.0000E+00	0.0000E+00	0.0000E+00	0.000E+00	9.000E-02	0.00	0.0000E+00	0.000E+00
0.0000E+00	0.0000E+00	0.0000E+00	0	% Amalthea					
0.0000E+00	0.0000E+00	0.0000E+00	8.0000E+17	5.70199E+04	5.5000E+04	4.5000E+04	0.000E+00	0.00	0.000E+00
0.00	0.0000E+00	0.0000E+00	0.0000E+00	0.0000E+00	0.000E+00	5.000E-02	0.00	0.0000E+00	0.000E+00
0.0000E+00	0.0000E+00	0.0000E+00	0	% Thebe					
6.8700E+02	9.7960E-02	3.5500E+04	5.6846E+26	8.27130E+13	6.0268E+07	5.4364E+07	1.044E+01	-139.15	1.000E+08
220.00	2.100E-01	1.62980E+04	3.7931E+07	3.420E-01	4.700E-01	-8.88	1.4900E+01	8.110E+01	0.000E+00
1.1500E+03	0.0000E+00	0.0000E+00	3.7900E+19	3.27736E+06	2.0900E+05	1.9100E+05	0.000E+00	0.00	0.000E+00
0.00	0.0000E+00	0.0000E+00	0.0000E+00	0.0000E+00	0.000E+00	6.000E-01	0.00	0.0000E+00	0.000E+00
0.0000E+00	0.0000E+00	0.0000E+00	0	% Mimas					
1.6100E+03	0.0000E+00	0.0000E+00	1.0800E+20	6.48921E+06	2.5600E+05	2.4500E+05	0.000E+00	0.00	0.000E+00
0.00	0.0000E+00	0.0000E+00	0.0000E+00	0.0000E+00	0.000E+00	1.000E+00	0.00	0.0000E+00	0.000E+00
0.0000E+00	0.0000E+00	0.0000E+00	0	% Enceladus					
9.6000E+02	0.0000E+00	0.0000E+00	6.1800E+20	6.23553E+07	5.3600E+05	5.2600E+05	0.000E+00	0.00	0.000E+00
0.00	0.0000E+00	0.0000E+00	0.0000E+00	0.0000E+00	0.000E+00	8.000E-01	0.00	0.0000E+00	0.000E+00
0.0000E+00	0.0000E+00	0.0000E+00	0	% Tethys					
1.4700E+03	0.0000E+00	0.0000E+00	1.1000E+21	7.35619E+07	5.6000E+05	5.6000E+05	0.000E+00	0.00	0.000E+00
0.00	0.0000E+00	0.0000E+00	0.0000E+00	0.0000E+00	0.000E+00	7.000E-01	0.00	0.0000E+00	0.000E+00
0.0000E+00	0.0000E+00	0.0000E+00	0	% Dione					
1.2300E+03	0.0000E+00	0.0000E+00	2.3100E+21	1.86796E+08	7.6400E+05	7.6400E+05	0.000E+00	0.00	0.000E+00
0.00	0.0000E+00	0.0000E+00	0.0000E+00	0.0000E+00	0.000E+00	7.000E-01	0.00	0.0000E+00	0.000E+00
0.0000E+00	0.0000E+00	0.0000E+00	0	% Rhea					
1.8800E+03	0.0000E+00	0.0000E+00	1.3455E+23	7.15188E+09	2.5750E+06	2.5750E+06	1.352E+00	-179.45	1.467E+05
0.00	0.0000E+00	0.0000E+00	2.6390E+03	0.0000E+00	2.200E-01	2.200E-01	1.5044E+01	0.000E+00	5.500E+00
0.0000E+00	0.0000E+00	0.0000E+00	0	% Titan					
5.7000E+02	0.0000E+00	0.0000E+00	5.5000E+18	1.22593E+06	1.8500E+05	1.1300E+05	0.000E+00	0.00	0.000E+00
0.00	0.0000E+00	0.0000E+00	0.0000E+00	0.0000E+00	0.000E+00	3.000E-01	0.0000E+00	0.0000E+00	0.000E+00
0.0000E+00	0.0000E+00	0.0000E+00	0	% Hyperion					
1.0900E+03	0.0000E+00	0.0000E+00	1.8100E+21	1.55046E+08	7.1800E+05	7.1800E+05	0.000E+00	0.00	0.000E+00
0.00	0.0000E+00	0.0000E+00	0.0000E+00	0.0000E+00	0.000E+00	5.000E-01	0.0000E+00	0.0000E+00	0.000E+00
0.0000E+00	0.0000E+00	0.0000E+00	0	% Iapetus					
1.000E+08	1.2700E+03	2.2930E-02	'URANUS'	8.6832E+25	6.83300E+12	2.5559E+07	2.4973E+07	8.870E+00	-197.00
0.000E+00	230.00	2.250E-01	2.1300E+04	5.7940E+06	3.000E-01	5.100E-01	-7.19	3.7100E+00	5.820E+01
1.000E+08	1.6380E+03	1.7080E-02	3.34343E+03	27	% Uranus				
0.000E+00	140.00	0.000E+00	'NEPTUNE'	1.0243E+26	6.25400E+12	2.4764E+07	2.4341E+07	1.115E+01	-201.00
3.000E-01	0.00	0.000E+00	2.3500E+04	6.8351E+06	2.900E-01	4.100E-01	-6.87	1.5100E+00	4.660E+01
0.000E+00	0.00	0.000E+00	3.41100E+03	13	% Neptune				
0.000E+00	0.00	0.000E+00	'PLUTO'	1.2500E+22	7.15000E+09	1.1950E+06	1.1950E+06	5.800E-01	-223.00
0.000E+00	0.00	0.000E+00	1.2000E+03	8.3000E+02	5.000E-01	6.000E-01	-1.00	8.9000E-01	3.750E+01
0.000E+00	0.00	0.000E+00	0.0000E+00	3	% Pluto				



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UnitLabelStr1 = { 'N/A', 'kg', '10^10 m^3', 'm', 'm/s^2', 'deg C', 'Pascals', 'kg/m^3', 'N/A', 'm/s',
'kg^3/s^2', 'N/A', 'N/A', 'V(1,0)', 'W/m^2', 'deg K', 'mGauss', '1/MR^2', 'x10^-6', 'N/A', '};
UnitLabelStr2 = { 'm', 'E. Days', 'm', 'E. Days', 'm/s', 'm/s', 'deg', 'N/A', 'E.
Hours', 'E. Hours', 'deg', '};

end

if strcmpi(UNITS,'British')== 1
% Properties [ OBJECT
DENSITY ELLIPTICITY VELOCITY-Esc MASS VOLUME RADIUS-EQ RADIUS-Pol GRAVITY TEMP PRESSURE
MAG FIELD Mom-Inertia NAME ft/s x10^-6 SATELLITES ] BOND-ALBEDO GEO-ALBEDO VIS-MAG SOL-Irradiance BB-TEMP ATM DENSITY
% Units
lbm ft^3/s^2 10^10 ft^3 N/A ft N/A ft BTUs/h*ft^2 deg F deg F psi
mGauss I/MR^2 I/MR^2
Planet_DB = { 'SUN', 4.3852E+30 4.98643E+18 2.2834E+09 2.2834E+09 9.012E+02 9940.73 1.015E+01
8.7899E+01 5.000E-05 2.0263E+06 4.6866E+21 0.000E+00 0.000E+00 -26.74 0.0000E+00 9.941E+03 0.000E+00
0.00 5.900E-02 0.0000E+00 0 % Sun
3.3880E+02 0.0000E+00 1.4107E+04 7.7798E+14 1.190E-01 8.0045E+06 8.0045E+06 1.214E+01 332.33 1.450E-13
0.00 3.300E-01 6.0000E+01 0 % Mercury
3.2731E+02 0.0000E+00 3.3990E+04 1.1474E+16 7.500E-01 1.9855E+07 1.9855E+07 2.910E+01 854.33 1.334E+03
0.00 3.300E-01 4.4580E+00 0 % Venus
'EARTH', 1.3171E+25 3.82532E+12 2.0926E+07 2.0926E+07 3.219E+01 59.00
3.4429E+02 3.3500E-03 1.470E+04 1.4076E+16 3.060E-01 3.670E-01 -3.86 4.3352E+02 -1.926E+00
2.377E-03 310.00 3.308E-01 1.08263E+03 1 % Earth
'MOON', 1.6202E+23 7.75440E+10 5.7003E+06 5.7003E+06 5.315E+00 -9.67
4.351E-14 2.0913E+02 1.2000E-03 7.8084E+03 1.7304E+14 1.100E-01 1.200E-01 +0.21 4.3352E+02 3.443E+01
0.000E+00 0.00 3.940E-01 2.02700E+02 0 % Moon
'MARS', 1.4150E+24 5.76265E+11 1.1145E+07 1.1145E+07 1.217E+01 -85.00
9.224E+01 2.4553E+02 6.4800E-03 1.6502E+03 1.5125E+15 2.500E-01 1.500E-01 -1.52 1.8678E+02 -8.149E-01
0.000E+00 0.00 3.660E-01 1.96045E+03 2 % Mars
'JUPITER', 4.1857E+27 5.05452E+15 2.3455E+08 2.3455E+08 7.585E+01 58.73
1.450E+04 8.2779E+02 6.4870E-02 1.9521E+05 4.4740E+18 3.430E-01 5.200E-01 -9.40 1.6009E+01 -2.617E-02
0.000E+00 4280.00 2.540E-01 1.47360E+04 63 % Jupiter
'IO', 1.9692E+23 8.94136E+10 5.9764E+06 5.9764E+06 0.000E+00 0.00 0.000E+00 0.000E+00 0.000E+00
2.2037E+02 0.0000E+00 0.0000E+00 0.0000E+00 0.000E+00 6.200E-01 0.00 0.000E+00 0.000E+00 0.000E+00
0.00 0.000E+00 0.0000E+00 0 % Io
'EUROPA', 1.0582E+23 5.62453E+10 5.1207E+06 5.1207E+06 0.000E+00 0.00 0.000E+00 0.000E+00
1.8791E+02 0.0000E+00 0.0000E+00 0.0000E+00 0.000E+00 6.800E-01 0.00 0.000E+00 0.000E+00 0.000E+00
0.00 0.000E+00 0.0000E+00 0 % Europa
'GANYMEDE', 3.2670E+23 2.69467E+11 8.6325E+06 8.6325E+06 0.000E+00 0.00 0.000E+00 0.000E+00
1.2111E+02 0.0000E+00 0.0000E+00 0.0000E+00 0.000E+00 4.400E-01 0.00 0.000E+00 0.000E+00 0.000E+00
0.00 0.000E+00 0.0000E+00 0 % Ganymede
'CALLISTO', 2.3720E+23 2.07137E+11 7.9078E+05 7.9078E+06 0.000E+00 0.00 0.000E+00 0.000E+00
1.1424E+02 0.0000E+00 0.0000E+00 0.0000E+00 0.000E+00 1.900E-01 0.00 0.000E+00 0.000E+00 0.000E+00
0.00 0.000E+00 0.0000E+00 0 % Callisto
'METIS', 2.2046E+17 1.18341E+06 6.5617E+04 6.5617E+04 0.000E+00 0.00 0.000E+00 0.000E+00
0.0000E+00 0.0000E+00 0.0000E+00 0.000E+00 6.300E-02 0.00 0.000E+00 0.000E+00 0.000E+00
0.00 0.000E+00 0.0000E+00 0 % Metis

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0.0000E+00	4.4092E+16	1.53843E+04	4.2651E+04	2.6247E+04	0.000E+00	0.00	0.000E+00	0.000E+00
0.00	0.0000E+00	0.0000E+00	0.000E+00	1.000E-01	0.00	0.000E+00	0.000E+00	0.000E+00
0.00	0.000E+00	0.0000E+00	0.000E+00	0.000E+00	0.00	0.000E+00	0.000E+00	0.000E+00
1.9353E+02	1.6535E+19	9.47793E+06	4.2979E+05	2.1982E+05	0.000E+00	0.00	0.000E+00	0.000E+00
0.00	0.000E+00	0.0000E+00	0.000E+00	9.000E-02	0.00	0.000E+00	0.000E+00	0.000E+00
0.0000E+00	1.7637E+18	2.01364E+06	1.8045E+05	1.4764E+05	0.000E+00	0.00	0.000E+00	0.000E+00
0.00	0.000E+00	0.0000E+00	0.000E+00	5.000E-02	0.00	0.000E+00	0.000E+00	0.000E+00
0.0000E+00	0.0000E+00	0.0000E+00	0.000E+00	0.000E+00	0.00	0.000E+00	0.000E+00	0.000E+00
1.450E+04	4.2888E+01	9.7960E-02	1.2532E+27	2.92098E+15	1.9773E+08	1.7836E+08	2.940E+01	-218.47
0.000E+00	220.00	2.100E-01	1.62980E+04	56	% Saturn	4.700E-01	-8.88	4.7233E+00
0.000E+00	0.000E+00	0.000E+00	0.000E+00	0.000E+00	0.000E+00	0.000E+00	0.00	0.000E+00
7.1792E+01	0.000E+00	0.000E+00	0.000E+00	6.000E-01	0.00	0.000E+00	0.000E+00	0.000E+00
0.00	0.000E+00	0.0000E+00	0.000E+00	% Mimas				
1.0051E+02	0.000E+00	0.000E+00	2.3810E+20	2.29164E+08	8.3989E+05	8.0380E+05	0.00	0.000E+00
0.00	0.000E+00	0.0000E+00	0.000E+00	0.000E+00	1.000E+00	0.00	0.000E+00	0.000E+00
0.00	0.000E+00	0.0000E+00	0.000E+00	% Enceladus				
5.9931E+01	0.000E+00	0.000E+00	1.3625E+21	2.20206E+09	1.7585E+06	1.7257E+06	0.00	0.000E+00
0.00	0.000E+00	0.0000E+00	0.000E+00	% Tethys				
9.1769E+02	0.000E+00	0.000E+00	2.4251E+21	2.59781E+09	1.8373E+06	1.8373E+06	0.00	0.000E+00
0.00	0.000E+00	0.0000E+00	0.000E+00	% Dione				
7.6786E+02	0.000E+00	0.000E+00	5.0927E+21	6.59664E+09	2.5066E+06	2.5066E+06	0.00	0.000E+00
0.00	0.000E+00	0.0000E+00	0.000E+00	% Rhea				
0.00	0.000E+00	0.000E+00	2.9663E+23	2.52566E+11	8.4481E+06	8.4481E+06	-291.01	2.128E+01
1.1736E+02	0.000E+00	0.000E+00	0.000E+00	2.200E-01	2.200E-01	7.90	0.000E+00	0.343E+00
0.00	0.000E+00	0.0000E+00	0.000E+00	% Titan				
3.5584E+01	0.000E+00	0.000E+00	1.2125E+19	4.32933E+07	6.0695E+05	3.7073E+05	0.00	0.000E+00
0.00	0.000E+00	0.0000E+00	0.000E+00	0.000E+00	3.000E-01	0.00	0.000E+00	0.000E+00
0.00	0.000E+00	0.0000E+00	0.000E+00	% Hyperion				
6.8046E+01	0.000E+00	0.000E+00	3.9904E+21	5.47540E+09	2.3556E+06	2.3556E+06	0.00	0.000E+00
0.00	0.000E+00	0.0000E+00	0.000E+00	0.000E+00	5.000E-01	0.00	0.000E+00	0.000E+00
0.00	0.000E+00	0.0000E+00	0.000E+00	% Iapetus				
1.450E+04	7.9284E+01	2.2930E-02	1.9143E+26	2.41305E+14	8.3857E+07	8.1932E+07	2.851E+01	-322.87
0.000E+00	230.00	2.250E-01	2.0461E+17	3.000E-01	5.100E-01	-7.19	1.1761E+00	-3.549E-02
0.000E+00	0.000E+00	0.000E+00	0.000E+00	% Uranus				
1.450E+04	1.0226E+01	1.7080E-02	2.2562E+26	2.20858E+14	8.1249E+07	7.9858E+07	3.609E+01	-328.27
0.000E+00	140.00	0.000E+00	2.4138E+17	2.900E-01	4.100E-01	-6.87	4.7867E-01	-3.758E-02
4.351E-05	1.0925E+01	0.000E+00	2.7558E+22	2.52499E+11	3.7763E+06	3.9206E+06	2.657E+00	-355.63
0.000E+00	0.00	0.000E+00	2.9311E+13	6.000E-01	6.000E-01	-1.00	2.8213E-01	-3.922E-02
0.00	0.000E+00	0.000E+00	% Pluto					
% Properties	[ SEMIMAJOR-AXIS	SIDEREAL-PERIOD	TROPICAL-PERIOD	PERIHELION	APHELION	SYNODIC-Per	MEAN-VEL	
MAX-VEL	MIN-VEL	ORB-INCLIN	ROTATION-Per	LENGTH-DAY	OBLIQUITY to Orbit	Orbit	ft/s	
% Units	ft/s	deg	E. Days	E. Hours	ft	E. Days	ft/s	
Planet_DB2 = {	0.0000E+00	0.0000E+00	0.00000E+00	0.0000E+00	deg	0.0000E+00	0.0000E+00	6.365E+04
0.000E+00	0.000E+00	0.000E+00	0.00000E+00	0.00000E+00	% Sun	0.0000E+00	0.000E+00	
1.89993E+11	1.89993E+11	8.796900E+01	8.796900E+01	1.50918E+11	2.26115E+11	1.1588E+02	1.571E+05	
1.935E+05	1.275E+05	7.000E+00	1.407600E+03	4.22260E+03	1.000E-02	% Mercury		

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1.15E+05 1.141E+05 3.55019E+11 2.247010E+02 2.247010E+02 6.700E-02 2.246950E+02 3.52623E+11 3.57414E+11 5.8392E+02 1.149E+05
9.770E+04 9.938E+04 9.610E+04 0.0000E+00 4.90812E+11 3.652560E+02 3.652420E+03 2.80200E+02 1.7736E+02 4.82577E+11 4.99014E+11 0.0000E+00
3.356E+03 3.530E+03 3.163E+03 1.26115E+09 0.000000E+00 1.670E-02 2.393450E+01 2.40000E+01 2.40000E+01 2.3450E+01 2.3450E+01 0.0000E+00
7.917E+04 8.694E+04 4.501E+04 1.3040E+00 2.55436E+12 4.890E-02 2.462290E+01 6.869730E+02 6.77885E+11 8.17681E+11 7.7994E+02
4.288E+04 0.000E+00 0.000E+00 4.000E-02 1.38320E+06 0.000000E+00 4.890E-02 9.92500E+00 9.92500E+00 3.1300E+00 3.1300E+00 0.000E+00
0.000E+00 0.000E+00 0.000E+00 4.000E-03 6.70900E+05 0.000000E+00 4.245931E+01 0.00000E+00 0.00000E+00 0.0000E+00 0.0000E+00 0.000E+00
0.000E+00 0.000E+00 4.7000E-01 1.010E-02 0.000000E+00 8.522834E+01 0.00000E+00 0.00000E+00 0.00000E+00 0.0000E+00 0.0000E+00 0.000E+00
0.000E+00 0.000E+00 2.2011E+06 0.000000E+00 2.2011E+06 0.000000E+00 1.71709E+02 1.71709E+02 0.0000E+00 0.0000E+00 0.0000E+00 0.000E+00
0.000E+00 0.000E+00 6.17682E+06 0.000000E+00 6.17682E+06 0.000000E+00 0.00000E+00 0.00000E+00 0.00000E+00 0.0000E+00 0.0000E+00 0.000E+00
0.000E+00 0.000E+00 5.1000E-01 7.000E-03 0.000000E+00 4.00536E+02 0.00000E+00 0.00000E+00 0.00000E+00 0.0000E+00 0.0000E+00 0.000E+00
0.000E+00 0.000E+00 4.19946E+05 0.000000E+00 4.19946E+05 0.000000E+00 0.00000E+00 0.00000E+00 0.00000E+00 0.0000E+00 0.0000E+00 0.000E+00
0.000E+00 0.000E+00 4.23227E+05 0.000000E+00 4.23227E+05 0.000000E+00 7.07470E+00 7.07470E+00 0.0000E+00 0.0000E+00 0.0000E+00 0.000E+00
0.000E+00 0.000E+00 3.0000E-02 1.500E-03 0.000000E+00 7.158240E+00 7.158240E+00 0.0000E+00 0.0000E+00 0.0000E+00 0.0000E+00 0.000E+00
0.000E+00 0.000E+00 5.95143E+05 0.000000E+00 5.95143E+05 0.000000E+00 0.00000E+00 0.00000E+00 0.00000E+00 0.0000E+00 0.0000E+00 0.000E+00
0.000E+00 0.000E+00 4.0000E-01 3.000E-03 0.000000E+00 1.19563E+01 1.19563E+01 0.0000E+00 0.0000E+00 0.0000E+00 0.0000E+00 0.000E+00
0.000E+00 0.000E+00 8.0000E-01 1.800E-02 0.000000E+00 1.61880E+01 1.61880E+01 0.0000E+00 0.0000E+00 0.0000E+00 0.0000E+00 0.000E+00
3.34E+04 2.982E+04 2.4850E+00 5.650E-02 1.074694E+04 1.06560E+01 1.06560E+01 2.6730E+01 2.6730E+01 3.7809E+02 3.7809E+02 3.179E+04
0.000E+00 0.000E+00 6.08660E+05 0.000000E+00 6.08660E+05 0.000000E+00 0.00000E+00 0.00000E+00 0.00000E+00 0.0000E+00 0.0000E+00 0.000E+00
0.000E+00 0.000E+00 1.5300E+00 2.020E-02 2.020E-02 2.21618E+01 2.21618E+01 0.0000E+00 0.0000E+00 9.4242E-01 9.4242E-01 0.000E+00
0.000E+00 0.000E+00 0.0000E+00 0.000000E+00 0.000000E+00 0.000000E+00 0.000000E+00 0.00000E+00 0.00000E+00 1.3702E+00 1.3702E+00 0.000E+00
0.000E+00 0.000E+00 0.0000E+00 4.500E-03 4.500E-03 3.28852E+01 3.28852E+01 0.0000E+00 0.0000E+00 0.0000E+00 0.0000E+00 0.000E+00
0.000E+00 0.000E+00 9.66729E+05 0.000000E+00 9.66729E+05 0.000000E+00 0.00000E+00 0.00000E+00 0.00000E+00 1.8878E+00 1.8878E+00 0.000E+00
0.000E+00 0.000E+00 1.8600E+00 0.0000E+00 1.8600E+00 4.530725E+01 4.530725E+01 0.0000E+00 0.0000E+00 2.7369E+00 2.7369E+00 0.000E+00
0.000E+00 0.000E+00 1.23819E+06 0.000000E+00 1.23819E+06 0.000000E+00 0.00000E+00 0.00000E+00 0.00000E+00 0.0000E+00 0.0000E+00 0.000E+00
0.000E+00 0.000E+00 2.000E-02 2.200E-03 2.200E-03 6.56860E+01 6.56860E+01 0.0000E+00 0.0000E+00 4.5175E+00 4.5175E+00 0.000E+00
0.000E+00 0.000E+00 3.5000E-01 1.000E-03 1.000E-03 1.08420E+02 1.08420E+02 0.0000E+00 0.0000E+00 0.0000E+00 0.0000E+00 0.000E+00
0.000E+00 0.000E+00 1.22183E+06 1.594500E+01 1.594500E+01 1.594500E+01 1.594500E+01 0.0000E+00 0.0000E+00 1.5945E+01 1.5945E+01 0.000E+00
0.000E+00 0.000E+00 3.3000E-01 2.920E-02 2.920E-02 3.826901E+02 3.826901E+02 0.00000E+00 0.00000E+00 0.0000E+00 0.0000E+00 0.000E+00
0.000E+00 0.000E+00 4.00862E+06 0.000000E+00 4.00862E+06 0.000000E+00 0.00000E+00 0.00000E+00 0.00000E+00 2.1277E+00 2.1277E+00 0.000E+00
0.000E+00 0.000E+00 1.16840E+07 0.000000E+00 1.16840E+07 0.000000E+00 0.00000E+00 0.00000E+00 0.00000E+00 7.9330E+00 7.9330E+00 0.000E+00
0.000E+00 0.000E+00 1.4720E+01 9.42405E+12 9.42405E+12 3.058874E+04 3.058874E+04 8.99374E+12 8.99374E+12 9.85437E+12 9.85437E+12 3.6966E+02
2.23E+04 2.333E+04 2.129E+04 7.7200E-01 7.7200E-01 4.570E-02 -1.72400E+01 1.72400E+01 1.72400E+01 1.72400E+01 1.72400E+01 9.7770E+01 3.6749E+02
1.78E+04 1.804E+04 1.762E+04 1.47475E+13 1.47475E+13 6.018900E+04 5.979990E+04 1.45815E+13 1.45815E+13 1.49136E+13 1.49136E+13 3.6749E+02
1.549E+04 2.000E+04 1.217E+04 1.93778E+13 1.93778E+13 9.046500E+04 9.046500E+04 1.61100E+01 1.61100E+01 2.8320E+01 2.8320E+01 3.6673E+02
0.000E+00 0.000E+00 0.000E+00 1.7160E+01 1.7160E+01 2.488E-01 -1.532928E+02 1.532928E+02 1.532928E+02 1.532928E+02 1.532928E+02 1.2253E+02 };
UnitLabelStr1 = { 'N/A', 'lbm', '10^10 ft^3', 'ft', 'ft/s^2', 'deg F', 'psi', 'lbm/ft^3', 'N/A', 'ft/s',
'lbm^3/s^2', 'N/A', 'N/A', 'V(1,0)', 'BTUs/h*ft^2', 'deg F', 'lbm/ft^3', 'mGauss', 'I/MR^2', 'x10^-6', 'N/A',
};

```



```
UnitLabelStr2 = { 'ft' 'E. Days' 'E. Days' 'ft' 'ft/s' 'ft/s' 'ft/s' 'deg' 'N/A' 'E.  
Hours' 'E. Hours' 'deg' };
```

```
end
```

```
%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%  
%% PLANETARY ATMOSPHERE PROPERTIES DEFINITION
```

```
%% Form Planet_Atmosphere(BODY_LOC, :) = { Element, %composition, Element, %composition, ... }  
% Order of Decreasing Composition  
{ Element %Concentration Element %Concentration Element %Concentration ... }  
Planet_AtmosphereDB = {  
'0.0' 'N/A' '0.0' / % H 'He' '8.889' 'N/A' '0.0' 'N/A'  
'0.5' 'N/A' '0.0' / % Sun 'Na' '29' 'He' '6' 'K'  
/ % Mercury 'N2' '3.5' 'N/A' '0.0' 'N/A'  
'0.0' 'N/A' '0.0' / % Venus 'O2' '20.946' 'H2O' '1.0' 'N/A' '0.0' 'N/A'  
'0.0' 'N/A' '0.0' / % Earth 'N/A' '0.0' 'N/A' '0.0' 'N/A'  
'0.0' 'N/A' '0.0' / % Moon 'N2' '2.7' 'O2' '0.13' 'CO'  
'0.08' 'N/A' '0.0' / % Mars 'He' '10.2' 'N/A' '0.0' 'N/A'  
'0.0' 'N/A' '0.0' / % Jupiter 'N/A' '0.0' 'N/A' '0.0' 'N/A'  
'0.0' 'N/A' '0.0' / % Io 'N/A' '0.0' 'N/A' '0.0' 'N/A'  
'0.0' 'N/A' '0.0' / % Europa 'N/A' '0.0' 'N/A' '0.0' 'N/A'  
'0.0' 'N/A' '0.0' / % Ganymede 'N/A' '0.0' 'N/A' '0.0' 'N/A'  
'0.0' 'N/A' '0.0' / % Callisto 'N/A' '0.0' 'N/A' '0.0' 'N/A'  
'0.0' 'N/A' '0.0' / % Metis 'N/A' '0.0' 'N/A' '0.0' 'N/A'  
'0.0' 'N/A' '0.0' / %Adrastea 'N/A' '0.0' 'N/A' '0.0' 'N/A'  
'0.0' 'N/A' '0.0' / %Amalthea 'N/A' '0.0' 'N/A' '0.0' 'N/A'  
'0.0' 'N/A' '0.0' / %Thebe 'He' '3.25' 'N/A' '0.0' 'N/A'  
'0.0' 'N/A' '0.0' / %Saturn 'N/A' '0.0' 'N/A' '0.0' 'N/A'  
'0.0' 'N/A' '0.0' / %Mimas 'N/A' '0.0' 'N/A' '0.0' 'N/A'  
'0.0' 'N/A' '0.0' / %Enceladus 'N/A' '0.0' 'N/A' '0.0' 'N/A'  
'0.0' 'N/A' '0.0' / %Tethys 'N/A' '0.0' 'N/A' '0.0' 'N/A'  
'0.0' 'N/A' '0.0' / %Dione 'N/A' '0.0' 'N/A' '0.0' 'N/A'  
'0.0' 'N/A' '0.0' / %Rhea 'N/A' '0.0' 'N/A' '0.0' 'N/A'
```

```

'0.0' 'N/A' '0.0' 'CH4' '1.6' 'N/A' '0.0' 'N/A' '0.0' 'N/A'
'0.0' 'N/A' '0.0' 'N/A' '0.0' 'N/A' '0.0' 'N/A' '0.0' 'N/A'
'0.0' 'N/A' '0.0' 'N/A' '0.0' 'N/A' '0.0' 'N/A' '0.0' 'N/A'
'0.0' 'N/A' '0.0' 'He' '15.2' 'CH4' '2.3' 'N/A' '0.0' 'N/A'
'0.0' 'N/A' '0.0' 'He' '19.0' 'CH4' '1.5' 'N/A' '0.0' 'N/A'
'0.0' 'N/A' '0.0' 'N2' '0.0' 'N/A' '0.0' 'N/A' '0.0' 'N/A'
} ; % Pluto

```

```

%
Trace elements
Element Conc Conc Unit Element Conc Conc Unit
Planet_AtmosphereDB2 = {
'Fe' '43' 'ppm' 'Mg' '35' 'ppm' 'S' '15' 'ppm' 'Ne' '112' 'ppm' 'N' '102' 'ppm'
'Ar' '43' 'ppm' 'Ar' '774' 'ppm' 'Ne' '15' 'ppm' 'CO2' '330' 'ppm' 'N/A' '0' 'ppm' 'N2' '102' 'ppm'
'He' '12' 'ppm' 'SO2' '150' 'ppm' 'Ar' '774' 'ppm' 'Ne' '15' 'ppm' 'H2O' '112' 'ppm' 'CO' '17' 'ppm'
'CH4' '1.7' 'ppm' 'Ar' '774' 'ppm' 'Ne' '15' 'ppm' 'H2O' '112' 'ppm' 'N/A' '0' 'ppm' '% Mercury'
'Ne22' '5000' 'ppccm' 'He4' '40000' 'ppccm' 'H2' '55' 'ppm' 'CO2' '330' 'ppm' 'N/A' '0' 'ppm' '% Venus'
'Kr' '0.3' 'ppm' 'H2O' '210' 'ppm' 'N/A' '0' 'ppm' 'Ar' '774' 'ppm' 'Ne' '15' 'ppm' 'H2O' '112' 'ppm' 'He' '5.24' 'ppm'
'H2O' '4' 'ppm' 'Xe' '0.08' 'ppm' 'N/A' '0' 'ppm' 'N/A' '0' 'ppm' 'N/A' '0' 'ppm' 'N/A' '0' 'ppm' 'Ar40' '30000' 'ppccm'
'N/A' '0' 'ppm' 'CH4' '3000' 'ppm' 'NH3' '260' 'ppm' 'N/A' '0' 'ppm' 'N/A' '0' 'ppm' 'N/A' '0' 'ppm' 'HDO' '0.85' 'ppm'
'N/A' '0' 'ppm' 'N/A' '0' 'ppm' 'N/A' '0' 'ppm' 'N/A' '0' 'ppm' 'N/A' '0' 'ppm' 'N/A' '0' 'ppm' 'C2H6' '5.8' 'ppm'
'N/A' '0' 'ppm' 'N/A' '0' 'ppm' 'N/A' '0' 'ppm' 'N/A' '0' 'ppm' 'N/A' '0' 'ppm' 'N/A' '0' 'ppm' 'N/A' '0' 'ppm' '% Jupiter'
'N/A' '0' 'ppm' 'N/A' '0' 'ppm' 'N/A' '0' 'ppm' 'N/A' '0' 'ppm' 'N/A' '0' 'ppm' 'N/A' '0' 'ppm' 'N/A' '0' 'ppm' 'Io'
'N/A' '0' 'ppm' 'N/A' '0' 'ppm' 'N/A' '0' 'ppm' 'N/A' '0' 'ppm' 'N/A' '0' 'ppm' 'N/A' '0' 'ppm' 'N/A' '0' 'ppm' 'N/A' '0' 'ppm' '% Europa'
'N/A' '0' 'ppm' 'N/A' '0' 'ppm' 'N/A' '0' 'ppm' 'N/A' '0' 'ppm' 'N/A' '0' 'ppm' 'N/A' '0' 'ppm' 'N/A' '0' 'ppm' 'N/A' '0' 'ppm' '%
Ganymede
'N/A' '0' 'ppm' 'N/A' '0' 'ppm' 'N/A' '0' 'ppm' 'N/A' '0' 'ppm' 'N/A' '0' 'ppm' 'N/A' '0' 'ppm' 'N/A' '0' 'ppm' 'N/A' '0' 'ppm'
Callisto
'N/A' '0' 'ppm' 'N/A' '0' 'ppm' 'N/A' '0' 'ppm' 'N/A' '0' 'ppm' 'N/A' '0' 'ppm' 'N/A' '0' 'ppm' 'N/A' '0' 'ppm' 'N/A' '0' 'ppm' 'N/A' '0' 'ppm'
'N/A' '0' 'ppm' 'N/A' '0' 'ppm' 'N/A' '0' 'ppm' 'N/A' '0' 'ppm' 'N/A' '0' 'ppm' 'N/A' '0' 'ppm' 'N/A' '0' 'ppm' 'N/A' '0' 'ppm' 'N/A' '0' 'ppm'
Adrastea
'N/A' '0' 'ppm' 'N/A' '0' 'ppm' 'N/A' '0' 'ppm' 'N/A' '0' 'ppm' 'N/A' '0' 'ppm' 'N/A' '0' 'ppm' 'N/A' '0' 'ppm' 'N/A' '0' 'ppm' 'N/A' '0' 'ppm'
Amalthea
'N/A' '0' 'ppm' 'N/A' '0' 'ppm' 'N/A' '0' 'ppm' 'N/A' '0' 'ppm' 'N/A' '0' 'ppm' 'N/A' '0' 'ppm' 'N/A' '0' 'ppm' 'N/A' '0' 'ppm' 'N/A' '0' 'ppm'
'N/A' '0' 'ppm' 'CH4' '4500' 'ppm' 'NH3' '125' 'ppm' 'N/A' '0' 'ppm' 'N/A' '0' 'ppm' 'N/A' '0' 'ppm' 'N/A' '0' 'ppm' 'N/A' '0' 'ppm' 'N/A' '0' 'ppm'
'N/A' '0' 'ppm' 'N/A' '0' 'ppm' 'N/A' '0' 'ppm' 'N/A' '0' 'ppm' 'N/A' '0' 'ppm' 'N/A' '0' 'ppm' 'N/A' '0' 'ppm' 'N/A' '0' 'ppm' 'N/A' '0' 'ppm'
'N/A' '0' 'ppm' 'N/A' '0' 'ppm' 'N/A' '0' 'ppm' 'N/A' '0' 'ppm' 'N/A' '0' 'ppm' 'N/A' '0' 'ppm' 'N/A' '0' 'ppm' 'N/A' '0' 'ppm' 'N/A' '0' 'ppm'
} ; % Mimas

```



```

end
% Ends R Definition Switch Case

% Determine Atmospheric Density From Database Values P = Rho*R*T
Planet_DB{LOC,18} = Planet_DB{LOC,8}/(R * (Planet_DB{LOC,7}+Temp_C2K)) ;

end
% Ends 'SI' R Gas Constant Definition Block

if strcmpi(UNITS,'British')== 1
    switch PlanAtm_Prop{1}
        case 'Are', R = 0.0; % ft*lb/slug*R - not Defined
        case 'CO' ; R = 0.0; % ft*lb/slug*R - not Defined
        case 'CO2' ; R = 1130.0; % ft*lb/slug*R
        case 'He' ; R = 12420.0; % ft*lb/slug*R
        case 'H2' ; R = 24660.0; % ft*lb/slug*R
        case 'CH4' ; R = 3095.0; % ft*lb/slug*R
        case 'N2' ; R = 1775.0; % ft*lb/slug*R
        case 'O2' ; R = 1554.0; % ft*lb/slug*R
        case 'SO2' ; R = 0.0; % ft*lb/slug*R - not Defined
        case 'H2O' ; R = 2760.0; % ft*lb/slug*R
        otherwise
            fprintf('\n*** WARNING: Undefined Gas Constant In Database for Atmospheric Component: %s ***',
                PlanAtm_Prop{1} );
            R = input('Enter the corresponding Component Gas Constant to determine Planetary Surface Density: ');
        end
    % Ends R Definition Switch Case

% Determine Atmospheric Density From Database Values P = Rho*R*T
Planet_DB{LOC,18} = Planet_DB{LOC,8}/(R * (Planet_DB{LOC,7} + Temp_F2R)) ;

end
% Ends 'SI' R Gas Constant Definition Block

end
% Ends Definition of Planetary Atmospheric Density Section

%*****
%
% *** PRINT SUMMARY Output Section for 'Print' = 'Y'
%*****

if strcmpi(Print,'Y')== 1
    fprintf('\n\n *** P L A N E T A R Y D A T A B A S E S U M M A R Y R E S U L T S ***');
    fprintf('\n\n*** I N P U T S ***');
    fprintf('\n Planetary Body Name: %s', Planet_DB{LOC,1});
    fprintf('\n Unit System: %s', UNITS);
    fprintf('\n\n*****\n\n*****\n\n*****\n\n*****\n\n*****');
    fprintf('**** PLANETARY BULK PROPERTIES ****');
    fprintf('\n Planetary Mass: %14.5e %s', Planet_DB{LOC,2}, MassUnit );
    fprintf('\n Planetary Volume (10^10): %14.5e %s', Planet_DB{LOC,3}, VolUnit );
end

```

```

fprintf('\n\n Planetary Equatorial Radius : %14.5e DistUnit );
fprintf('\n\n Planetary Polar Radius : %14.5e DistUnit );
fprintf('\n\n Planetary Gravity : %14.5e GravUnit );
fprintf('\n\n Planetary Density : %14.5e DensityUnit );
fprintf('\n\n Planetary Ellipticity : %14.5e );
fprintf('\n\n Planetary Escape Velocity : %14.5e VelUnit );
fprintf('\n\n Planetary GM : %14.5e GMUnit );
fprintf('\n\n Planetary Bond Albedo : %14.5e );
fprintf('\n\n Planetary Visual Geometric Albedo : %14.5e );
fprintf('\n\n Planetary Visual Magnitude : %14.5e );
fprintf('\n\n Planetary Solar Irradiance : %14.5e );
fprintf('\n\n Planetary Black Body Temperature : %14.5e TempUnit );
fprintf('\n\n Planetary Magnetic Field Strength : %14.5e );
fprintf('\n\n Planetary Moment of Inertia : %14.5e );
fprintf('\n\n Planetary J2 Parameter : %14.5e );
fprintf('\n\n Planetary Satellites : %14d );

fprintf('\n\n\n*** PLANETARY ORBITAL PARAMETERS ***');
fprintf('\n\n Semi-Major Axis : %14.5e );
fprintf('\n\n Sidereal Orbit Period : %14.5e );
fprintf('\n\n Tropical Orbit Period : %14.5e );
fprintf('\n\n Perihelion Distance : %14.5e );
fprintf('\n\n Aphelion Distance : %14.5e );
fprintf('\n\n Synodic Period : %14.5e );
fprintf('\n\n Mean Orbital Velocity : %14.5e );
fprintf('\n\n Maximum Orbital Velocity : %14.5e );
fprintf('\n\n Minimum Orbital Velocity : %14.5e );
fprintf('\n\n Orbital Inclination : %14.5e );
fprintf('\n\n Orbital Eccentricity : %14.5e );
fprintf('\n\n Sidereal Rotation Period : %14.5e );
fprintf('\n\n Length of Day : %14.5e );
fprintf('\n\n Obliquity to Orbit : %14.5e );

fprintf('\n\n\n*** ATMOSPHERIC PROPERTIES ***');
fprintf('\n\n Planetary Surface Temperature : %14.2f );
fprintf('\n\n Planetary Surface Pressure : %14.5e );
fprintf('\n\n Planetary Surface Density : %14.5e );

fprintf('\n\n\n ATMOSPHERIC COMPOSITION ');
fprintf('\n\n Major Elements ');
fprintf('\n\n-----');
for i = 1:2:length(PlanAtm_Prop)-1
    if strcmpi(PlanAtm_Prop{i}, 'N/A') ~= 1
        fprintf('\n %6s %7.3f %%', PlanAtm_Prop{i}, str2double(PlanAtm_Prop{i+1}));
    end
end

fprintf('\n\n Trace Elements ');
fprintf('\n\n-----');
for i = 1:3:length(PlanAtm_MEProp)-2
    fprintf('\n Element Amount Concentration ');

```

```

    if strcmpi(PlanAtm_MEProp{i}, 'N/A') ~= 1
        fprintf('\n %s %9.3f %s ', PlanAtm_MEProp{i}, str2double(PlanAtm_MEProp{i+1}), PlanAtm_MEProp{i+2} );
    end
end

Planet_Properties = { Planet_DB{LOC,:} Planet_DB2{LOC,:} };
UnitLabelStr = [ UnitLabelStr1 UnitLabelStr2 ];

save('Planet_Data.mat', 'Planet_Properties', 'UnitLabelStr', 'PlanAtm_Prop', 'PlanAtm_MEProp' );

%***** Writes a text file summary of the Planetary Properties
%***
Fout = fopen('PlanetDB_Summary.txt','w+');

fprintf(Fout, '\n\n *** P L A N E T A R Y   D A T A   B A S E   S U M M A R Y   R E S U L T S   * * * *');
fprintf(Fout, '\n\n\n *** I N P U T S * * *');
fprintf(Fout, '\n Planetary Body Name: %s', Planet_DB{LOC,1});
fprintf(Fout, '\n Unit System: %s', UNITS);

fprintf(Fout, '\n\n\n *****\n\n\n *****\n\n\n *****\n\n\n *****\n\n\n *****\n\n\n *****');

fprintf(Fout, ' *** PLANETARY BULK PROPERTIES * * *');
fprintf(Fout, '\n Planetary Mass : %14.5e %s', Planet_DB{LOC,2}, MassUnit );
fprintf(Fout, '\n Planetary Volume (10^10) : %14.5e %s', Planet_DB{LOC,3}, VolUnit );
fprintf(Fout, '\n Planetary Equatorial Radius : %14.5e %s', Planet_DB{LOC,4}, DistUnit );
fprintf(Fout, '\n Planetary Polar Radius : %14.5e %s', Planet_DB{LOC,5}, DistUnit );
fprintf(Fout, '\n Planetary Gravity : %14.5e %s', Planet_DB{LOC,6}, GravUnit );
fprintf(Fout, '\n Planetary Density : %14.5e %s', Planet_DB{LOC,9}, DensityUnit );
fprintf(Fout, '\n Planetary Ellipticity : %14.5e ', Planet_DB{LOC,10} );
fprintf(Fout, '\n Planetary Escape Velocity : %14.5e %s', Planet_DB{LOC,11}, VelUnit );
fprintf(Fout, '\n Planetary GM : %14.5e %s', Planet_DB{LOC,12}, GMUnit );
fprintf(Fout, '\n Planetary Bond Albedo : %14.5e ', Planet_DB{LOC,13} );
fprintf(Fout, '\n Planetary Visual Geometric Albedo : %14.5e ', Planet_DB{LOC,14} );
fprintf(Fout, '\n Planetary Visual Magnitude : %14.5e %s', Planet_DB{LOC,15} );
fprintf(Fout, '\n Planetary Solar Irradiance : %14.5e %s', Planet_DB{LOC,16}, SolIradUnit );
fprintf(Fout, '\n Planetary Black Body Temperature : %14.5e %s', Planet_DB{LOC,17}, TempUnit );
fprintf(Fout, '\n Planetary Magnetic Field Strength : %14.5e %s', Planet_DB{LOC,19}, UnitLabelStr1{19} );
fprintf(Fout, '\n Planetary Moment of Inertia : %14.5e ', Planet_DB{LOC,20} );
fprintf(Fout, '\n Planetary J2 Parameter : %14.5e ', Planet_DB{LOC,21} );
fprintf(Fout, '\n Planetary Satellites : %14d ', Planet_DB{LOC,22} );

fprintf(Fout, '\n\n\n *** PLANETARY ORBITAL PARAMETERS * * *');
fprintf(Fout, '\n Semi-Major Axis : %14.5e %s', Planet_DB2{LOC,1}, DistUnit );
fprintf(Fout, '\n Sidereal Orbit Period : %14.5e Earth Days', Planet_DB2{LOC,2} );
fprintf(Fout, '\n Tropical Orbit Period : %14.5e Earth Days', Planet_DB2{LOC,3} );
fprintf(Fout, '\n Perihelion Distance : %14.5e %s', Planet_DB2{LOC,4}, DistUnit );
fprintf(Fout, '\n Aphelion Distance : %14.5e %s', Planet_DB2{LOC,5}, DistUnit );
fprintf(Fout, '\n Synodic Period : %14.5e Earth Days', Planet_DB2{LOC,6} );

```

```

fprintf(Fout, '\n Mean Orbital Velocity : %14.5e %s', Planet_DB2{LOC,7}, VelUnit );
fprintf(Fout, '\n Maximum Orbital Velocity : %14.5e %s', Planet_DB2{LOC,8}, VelUnit );
fprintf(Fout, '\n Minimum Orbital Velocity : %14.5e %s', Planet_DB2{LOC,9}, VelUnit );
fprintf(Fout, '\n Orbital Inclination : %14.5e deg', Planet_DB2{LOC,10} );
fprintf(Fout, '\n Orbital Eccentricity : %14.5e ', Planet_DB2{LOC,11} );
fprintf(Fout, '\n Sidereal Rotation Period : %14.5e Earth Hours', Planet_DB2{LOC,12} );
fprintf(Fout, '\n Length of Day : %14.5e Earth Hours', Planet_DB2{LOC,13} );
fprintf(Fout, '\n Oblliquity to Orbit : %14.5e deg', Planet_DB2{LOC,14} );

fprintf(Fout, '\n\n*** ATMOSPHERIC PPROPERTIES ***');
fprintf(Fout, '\n Planetary Surface Temperature : %14.2f %s', Planet_DB{LOC,7}, TempUnit );
fprintf(Fout, '\n Planetary Surface Pressure : %14.5e %s', Planet_DB{LOC,8}, PressureUnit );
fprintf(Fout, '\n Planetary Surface Density %14.5e %s', Planet_DB{LOC,18}, DensityUnit );

fprintf(Fout, '\n\n ATMOSPHERIC COMPOSITION ');
fprintf(Fout, '\n Major Elements ');
fprintf(Fout, '\n-----');
fprintf(Fout, '\n Element % Composition ');
for i = 1:2:length(PlanAtm_Prop)-1
    if strcmpi(PlanAtm_Prop{i}, 'N/A') ~= 1
        fprintf(Fout, '\n %6s %7.3f %s', PlanAtm_Prop{i}, str2double(PlanAtm_Prop{i+1}) );
    end
end

fprintf(Fout, '\n\n Trace Elements ');
fprintf(Fout, '\n-----');
fprintf(Fout, '\n Element Amount Concentration ');
for i = 1:3:length(PlanAtm_MEProp)-2
    if strcmpi(PlanAtm_MEProp{i}, 'N/A') ~= 1
        fprintf(Fout, '\n %6s %9.3f %6s ', PlanAtm_MEProp{i}, str2double(PlanAtm_MEProp{i+1}), PlanAtm_MEProp{i+2}
);
    end
end
fclose(Fout);

```

## D5. AC Field Sensors

```

% function AC_FIELD_SENSORS(PLACFIELD,UNITS,Print)
% AC Field Sensor Database
%
% Developed by: Keith Schreck
% Mechanical and Aerospace Engineering
% San Jose State University
% Date: Fall 2007
%
% This file contains AC Field Sensor values for the ISSPO program.
% Data here is loaded into the main program to determine the proper sensor
% to use for Missions based on the sensor requirements.

```

```

% Values are assigned to a variable name and saved to a Matlab '.mat' file
% and loaded when calling the ISSPO program.

% Some of the values are based on the Unit system chosen and are loaded
% in separate sections.

% References
% http://www.magneticciences.com/MSI95Gaussmeter.html
% http://meda-sam.meda.com/catalog/cat4_1.htm
% http://www.ssec.honeywell.com/magnetic/magnetometers.html
function AC_FIELD_SENSORS(PLACFIELD,UNITS,Print)

% Load Constants values into Local Program
load('Constants_DB.mat');
load('Planet_Data.mat');

PLTEMP = Planet_Properties{7};

%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%
%% AC Field Sensor Data Base Created Based on un Unit System
%% 2 - D Matrix AC_FIELD_DB(AC_Type, Property)
%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%
% AC Field Sensor Common Properties
% Properties [ SENSOR
TOLERANCE VOLTAGE TYPE ]
% Units [ #
mV/mG V N/A ]
AC_FIELD_SENSOR_DB1 = { 'Bell-4080'
0 'DC' ;
1.00 9 'DC' ;
10.0 2 'DC' ;
10.0 2 'DC' ;
0 5 'DC' ;
0 5 'DC' ;
0 5 'DC' ;
0 5 'DC' ;
10.0 36 'DC' ;
10.0 38 'DC' ;
0 15 'DC' };

%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%
%% AC Field Sensor Data Base Created Based on un Unit System
%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%
%% 2 - D Matrix AC_FIELD_DB(AC_Type, Property)
%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%
% AC Field Sensor Common Properties
% Properties [ SENSOR
TOLERANCE VOLTAGE TYPE ]
% Units [ #
mV/mG V N/A ]
AC_FIELD_SENSOR_DB1 = { 'Bell-4080'
0 'DC' ;
1.00 9 'DC' ;
10.0 2 'DC' ;
10.0 2 'DC' ;
0 5 'DC' ;
0 5 'DC' ;
0 5 'DC' ;
0 5 'DC' ;
10.0 36 'DC' ;
10.0 38 'DC' ;
0 15 'DC' };

```



```

if strcmpi(UNITS,'SI')==1
% Properties {
WEIGHT ELECT WEIGHT }
% Units { Low deg C High
Grams }
AC_FIELD_SENSOR_DB2 = {
141.748 ;
0 ;
177.00 ;
177.00 ;
0 ;
0 ;
0 ;
0 ;
0 ;
770.0 ;
0 ;
98.0 };
ACfieldUnitLabel1Str2 = { 'deg C' 'deg C' 'deg C' 'deg C' 'cm' 'cm' 'cm' 'cm' 'cm' 'gram' 'gram' };
end

```

SENS	ELECT DIMENSIONS (cm)			SENSOR DIMENSIONS (cm)			ELECTRONICS TEMP		SENS WEIGHT
	L	W	H	L	W	H	Low deg C	High	
141.748	6.096	11.938	2.540	6.096	11.938	2.540	0	0	141.748
0	5.000	3.700	2.700	5.000	3.700	2.700	0	0	0
177.00	12.00	7.600	3.700	12.00	7.600	3.700	-10.0	50.0	177.00
177.00	12.00	7.600	3.700	12.00	7.600	3.700	-10.0	50.0	177.00
0	14.047	1.811	1.811	14.047	1.811	1.811	-25.0	70.0	0
0	15.494	0.889	0.889	15.494	0.889	0.889	-25.0	70.0	0
0	15.494	0.889	0.889	15.494	0.889	0.889	-25.0	70.0	0
0	15.494	0.889	0.889	15.494	0.889	0.889	-25.0	70.0	0
770.0	4.76	6.68	11.9	4.76	6.68	11.9	-34.0	71.0	310.0
0	4.45	7.62	14.3	4.45	7.62	14.3	-39.0	76.0	500.0
98.0	10.67	3.81	2.23	10.67	3.81	2.23	-40.0	85.0	98.0

```

if strcmpi(UNITS,'BRITISH')==1
% Properties {
ELECT WEIGHT }
% Units { Low deg F High
OZ }
AC_FIELD_SENSOR_DB2 = {
5.00 ;
0 ;
6.24 ;
6.24 ;
0 ;
0 ;
0 ;
0 ;
0 ;
0 ;
0 };
ACfieldUnitLabel1Str2 = { 'deg C' 'deg C' 'deg C' 'deg C' 'cm' 'cm' 'cm' 'cm' 'cm' 'gram' 'gram' };
end

```

SENS	ELECT DIMENSIONS (in)			SENSOR DIMENSIONS (in)			ELECTRONICS TEMP		SENS WEIGHT
	L	W	H	L	W	H	Low deg F	High	
5.00	2.40	4.70	1.00	2.40	4.70	1.00	0	0	5.00
0	2.00	1.50	1.00	2.00	1.50	1.00	0	0	0
6.24	4.70	3.00	1.75	4.70	3.00	1.75	14.0	122.0	6.24
6.24	4.70	3.00	1.75	4.70	3.00	1.75	14.0	122.0	6.24
0	5.53	0.71	0.71	5.53	0.71	0.71	32.0	158.0	0
0	6.10	0.35	0.35	6.10	0.35	0.35	32.0	158.0	0
0	6.10	0.35	0.35	6.10	0.35	0.35	32.0	158.0	0
0	6.10	0.35	0.35	6.10	0.35	0.35	32.0	158.0	0

```

27.2      ;
0         ;
0         };

```

	-130.0	176.0	-29.2	159.8	1.88	2.63	4.69	2.06	5.52	6.45	12.64
	-38.2	168.8	-38.2	168.8	1.75	3.00	5.63	0	0	0	17.60
	-40.0	185.0	-40.0	185.0	4.20	1.50	0.876	0	0	0	3.46

```

ACFieldUnitLabelStr2 = { 'deg F' 'deg F' 'deg F' 'in' 'in' 'in' 'in' 'in' 'oz' 'oz' };
end

%Comments on sensors - uses
AC_FIELD_SENSOR_DB3{1,1} = ' ';
AC_FIELD_SENSOR_DB3{2,1} = ' ';
AC_FIELD_SENSOR_DB3{3,1} = ' ';
AC_FIELD_SENSOR_DB3{4,1} = ' ';
AC_FIELD_SENSOR_DB3{5,1} = ' ';
AC_FIELD_SENSOR_DB3{6,1} = ' ';
AC_FIELD_SENSOR_DB3{7,1} = ' ';
AC_FIELD_SENSOR_DB3{8,1} = ' ';
AC_FIELD_SENSOR_DB3{9,1} = 'TAM-1 series parts meet NASA and and military programs (MIL-STD-975). Fully Space Qualified';
AC_FIELD_SENSOR_DB3{10,1} = 'TAM-2 series parts meet NASA and and military programs (MIL-STD-975). Fully Space Qualified.
Single Analog Output per axis';
AC_FIELD_SENSOR_DB3{11,1} = 'Single Package design Sensor & Electronics self contained on PCB';

```

```

% Combine Data Arrays to a Single Matrix
AC_FIELD_SENSOR_DB = cat(2,AC_FIELD_SENSOR_DB1,AC_FIELD_SENSOR_DB2,AC_FIELD_SENSOR_DB3);
ACFieldUnitLabelStr = cat(2,ACFieldUnitLabelStr1,ACFieldUnitLabelStr2);

```

```

% Print Sensor Alert Warning if Planetary Magnetic Field is Zero
if PLACFIELD == 0
    if strcmpi(Print,'Y') == 1
        fprintf('\n\n *** WARNING: PLANETARY MAGNETIC FIELD NOT DEFINED! ***');
        fprintf('\n Sensor Design Based on other properties!');
    end
end

```

```

c = 1;
% Build Up Database of Possible Magnetic Field Sensors Based on Planetary Sensor Range
for d = 1:size(AC_FIELD_SENSOR_DB,1)
    if (PLACFIELD >= AC_FIELD_SENSOR_DB(d,3)) && (PLACFIELD <= AC_FIELD_SENSOR_DB(d,4))
        for a = 1:size(AC_FIELD_SENSOR_DB,2)
            AC_MATRIX{c,a} = AC_FIELD_SENSOR_DB{d,a};
        end
        c = c+1;
    end
end

```

```

% Copies MATRIX to MATRIX 2 if all Sensors Eliminated based on Temperature Range
if exist('AC_MATRIX') == 0

```

```

AC_MATRIX = AC_FIELD_SENSOR_DB;
end

% Down Select a Single Sensor if Multiple Options Exist
if size(AC_MATRIX,1) > 1

% Select Via Temperature Operational Range
c = 1;
% Build Up Database of Possible AC field Sensors Based on Temperature Range
for t = 1:size(AC_MATRIX,1)
if (PLTEMP >= AC_MATRIX{t,13}) && (PLTEMP <= AC_MATRIX{t,14})
for a = 1:size(AC_MATRIX,2)
AC_MATRIX2{c,a} = AC_MATRIX{t,a};
end
c = c+1;
end
end

% Copies AC_MATRIX to AC_MATRIX2 if all Sensors Eliminated based on Temperature Range
if exist('AC_MATRIX2') == 0
AC_MATRIX2 = AC_MATRIX;
end

if size(AC_MATRIX2,1) > 1
% Account for Fluctuations in Planetary Magnetic Field - Adds 20% to Sensor Max Range
PLACFIELD2 = PLACFIELD+0.2*PLACFIELD;
c = 1;
for d = 1:size(AC_MATRIX2,1)
if (PLACFIELD2 >= AC_MATRIX2{d,3}) && (PLACFIELD2 <= AC_MATRIX2{d,4})
for a = 1:size(AC_MATRIX2,2)
AC_MATRIX3{c,a} = AC_MATRIX2{d,a};
end
c = c+1;
end
end

end

% Copies AC_MATRIX2 to AC_MATRIX3 if all Sensors Eliminated based on Fluctuation Range
if exist('AC_MATRIX3') == 0
AC_MATRIX3 = AC_MATRIX2;
end

% Run Case if More than One Sensor is Viable
if size(AC_MATRIX3,1) > 1
% Sort Sensors by Max Operational Temperature Range
for t = 1:size(AC_MATRIX3,1)
AC_TEMP(t) = AC_MATRIX3{t,14} - AC_MATRIX3{t,13};
end
[TempR, AC_Loc ] = max(AC_TEMP);

% Build Sensor Matrix Based on the Temp Range
k = 1;
for c = 1:length(AC_TEMP)
if AC_TEMP(c) == TempR

```

```

for j = 1:size(AC_MATRIX3,2)
    AC_MATRIX4{k,j} = AC_MATRIX3{c,j};
end
k = k + 1;
end
end

end
% Copies AC_MATRIX3 to AC_MATRIX4 if all Sensors Eliminated based on Fluctuation Range
if exist('AC_MATRIX4') == 0
    AC_MATRIX4 = AC_MATRIX3;
end

% Run Case if More than One Sensor is Viable
if size(AC_MATRIX4,1) > 1
    % Create Mass / Volume Scaling Factor Optimization
    for mv = 1:size(AC_MATRIX4,1)
        AC_SCALE_FAC(mv) = AC_MATRIX4{mv,17} * AC_MATRIX4{mv,18} * AC_MATRIX4{mv,19} * AC_MATRIX4{mv,23} + ...
            AC_MATRIX4{mv,20} * AC_MATRIX4{mv,21} * AC_MATRIX4{mv,22} * AC_MATRIX4{mv,24};

        if (AC_MATRIX4{mv,17} == 0) || (AC_MATRIX4{mv,18} == 0) || (AC_MATRIX4{mv,19} == 0) || (AC_MATRIX4{mv,23} == 0)
            AC_SCALE_FAC(mv) = 9.99E99;
        elseif (AC_MATRIX4{mv,20} == 0) || (AC_MATRIX4{mv,21} == 0) || (AC_MATRIX4{mv,22} == 0) && (AC_MATRIX4{mv,23} ~= 0)
            AC_SCALE_FAC(mv) = 9.99E99;
        elseif (AC_MATRIX4{mv,20} ~= 0) || (AC_MATRIX4{mv,21} ~= 0) || (AC_MATRIX4{mv,22} ~= 0) && (AC_MATRIX4{mv,23} == 0)
            AC_SCALE_FAC(mv) = 9.99E99;
        end
    end
end

[ AC_MIN_SCL, INDEX ] = min(AC_SCALE_FAC);

% Build Sensor Matrix Based on Scale Factors
k = 1;
for c = 1:length(AC_SCALE_FAC)
    if AC_SCALE_FAC(c) == AC_MIN_SCL
        for j = 1:size(AC_MATRIX4,2)
            AC_MATRIX5{k,j} = AC_MATRIX4{c,j};
        end
        k = k + 1;
    end
end

end

% Copies AC_MATRIX4 to AC_MATRIX5 if all Sensors Eliminated based on Dimensions Range
if exist('AC_MATRIX5') == 0
    AC_MATRIX5 = AC_MATRIX4;
end

else
% Assign Null values to Array if No sensor is viable
for ac = 1:size(AC_FIELD_SENSOR_DB,2)
    AC_MATRIX5{1,ac} = 0.0;
end

```





```

fprintf(Fout, '\n Sensor Mass: %9.3f %s', AC_MATRIX5{1,24}, ACFieldUnitLabelStr{24} );
fprintf(Fout, '\n\n Power Requirements');
fprintf(Fout, '\n Input Voltage: %9.3f %s', AC_FIELD_SENSOR_DB{1,11}, ACFieldUnitLabelStr{11} );
fprintf(Fout, '\n Voltage Type: %s %s', AC_FIELD_SENSOR_DB{1,12}, ACFieldUnitLabelStr{12} );

fprintf(Fout, '\n\n Sensor Comments:');
fprintf(Fout, '\n %s', AC_MATRIX5{1,25});
fclose(Fout);

```

## D6. Accelerometer Sensors

```

% function ACCELEROMETER_SENSORS(ACC_TYPE,POW_TYPE,UNITS, Print)
% Accelerometers Sensor Database
%
% Developed by: Keith Schreck
% Mechanical and Aerospace Engineering
% San Jose State University
% Date: Fall 2007
%
% This file contains Accelerometers Sensor values for the ISSPO program.
% Data here is loaded into the main program to determine the proper sensor
% to use for Missions based on the Planetary Atmosphere composition.
% Values are assigned to a variable name and saved to a Matlab '.mat' file
% and loaded when calling the ISSPO program.
%
% Some of the values are based on the Unit system chosen and are loaded
% in separate sections.
%
% References
% http://www.sensotec.com/catpages.asp
% http://www.omega.com/toc_esp/subsectionsC.asp?subsection=K02&book=Pressure
% http://www.endevco.com/feature/Featured.aspx?page=71
% http://www.bksv.com/pdf/2271A_AM20.pdf
% http://www.honeywell.com/sites/servlet/com.merx.npoint.servlets.DocumentServlet?docid=D732BF1BD-AFCB-3204-9D96-
% 2292C529E86F
%
function ACCELEROMETER_SENSORS(ACC_TYPE,POW_TYPE,UNITS, Print)
%
% Load Constants values into Local Program
load('Constants_DB.mat');
load('Planet_Data.mat');
%
% Define Local Temperature Variable
PLTEMP = Planet_Properties{7};
%
% Define Program Variables

```

```

if strcmpi(ACC_TYPE,'BALLISTIC')== 1
    ACC_RANGE = 10000;
    TOL_LOWER = 0;
end
if strcmpi(ACC_TYPE,'IMPACT')== 1
    ACC_RANGE = 10000;
    TOL_LOWER = 1001;
end
if strcmpi(ACC_TYPE,'HIGH')== 1
    ACC_RANGE = 1000;
    TOL_LOWER = 101;
end
if strcmpi(ACC_TYPE,'MEDIUM')== 1
    ACC_RANGE = 100;
    TOL_LOWER = 11;
end
if strcmpi(ACC_TYPE,'SOFT')== 1
    ACC_RANGE = 10;
    TOL_LOWER = 0;
end

% Common Accelerator Properties
% Properties [
% RESONANCE_FREQ. SHOCK_LIMITS VIB_LIMITS SENSOR
% Units g's DB1 = { 'JTF-GPA' 10000.0 10000.0 10000.0 }
Hz ACCELEROMETER_SENSOR_DB1 = { 'JTF-GPA' 10000.0 10000.0 10000.0 }
5000.00 10000.0 2000.0 2000.0 2000.0 2000.0 2000.0 2000.0 2000.0 2000.0
18000.00 2000.0 2000.0 2000.0 2000.0 2000.0 2000.0 2000.0 2000.0 2000.0
18000.00 0 0 0 0 0 0 0 0 0
5000.00 0 0 0 0 0 0 0 0 0
18000.00 0 0 0 0 0 0 0 0 0
25000.00 0 0 0 0 0 0 0 0 0
18000.00 1000.0 1000.0 1000.0 1000.0 1000.0 1000.0 1000.0 1000.0 1000.0
18000.00 1000.0 1000.0 1000.0 1000.0 1000.0 1000.0 1000.0 1000.0 1000.0
18000.00 1000.0 1000.0 1000.0 1000.0 1000.0 1000.0 1000.0 1000.0 1000.0
22000.00 0 0 0 0 0 0 0 0 0
30000.00 0 0 0 0 0 0 0 0 0
18000.00 0 0 0 0 0 0 0 0 0
45000.00 0 0 0 0 0 0 0 0 0
10000.00 0 0 0 0 0 0 0 0 0

```

Hz	ACCELEROMETER_SENSOR_DB1	g's	SHOCK_LIMITS	VIB_LIMITS	SENSOR	DYNAMIC_RANGE	SENSITIVITY	TRANSVERSE_SENS	FREQ_RANGE	LINEARITY
			Low	High	Type	Low	High	V	Elect. Noise	%
5000.00	10000.0	10000.0	-500.0	500.0	'N/A'	0	0.08	5.00	0.00	1.00
18000.00	2000.0	2000.0	-80.0	80.0	'Const Current'	0	100.0	10.0	2.00	1.00
18000.00	0	0	-80.0	80.0	'Const Current'	0.500	8.000	12.0	2.00	1.00
5000.00	0	0	-50.0	50.0	'Voltage'	0	0	32.0	2.00	1.00
18000.00	0	0	-80.0	80.0	'Const Current'	0.500	8.000	12.0	5.00	1.00
25000.00	0	0	-10.0	10.0	'Const Current'	5.000	8.000	12.0	5.00	1.00
18000.00	1000.0	1000.0	-50.0	50.0	'Voltage'	7.000	7.000	24.0	1.00	1.00
18000.00	1000.0	1000.0	-50.0	50.0	'Voltage'	4.000	20.000	30.0	1.00	1.00
18000.00	1000.0	1000.0	-50.0	50.0	'Voltage'	7.000	7.000	5.0	1.00	1.00
18000.00	1000.0	1000.0	-80.0	80.0	'Const Current'	0.500	8.000	12.0	2.00	1.00
22000.00	0	0	-800.0	800.0	'Self Generating'	0	20.0	5.00	0	1.00
30000.00	0	0	-1000.0	1000.0	'Self Generating'	0	50.0	5.00	1.00	1.00
18000.00	0	0	-2000.0	2000.0	'Self Generating'	0	5.0	5.00	1.00	1.00
45000.00	0	0	-100.0	100.0	'Self Generating'	0	1100.0	5.00	0	1.00
10000.00	0	0	-100.0	100.0	'Self Generating'	0	0	5.00	0	1.00







```

fprintf(Fout, '\n Sensor Mass: %9.3f %s', AC_MATRIX5{1,24}, ACfieldUnitLabelStr{24} );
fprintf(Fout, '\n\n Power Requirements');
fprintf(Fout, '\n Input Voltage: %9.3f %s', AC_FIELD_SENSOR_DB{1,11}, ACfieldUnitLabelStr{11} );
fprintf(Fout, '\n Voltage Type: %s %s', AC_FIELD_SENSOR_DB{1,12}, ACfieldUnitLabelStr{12} );

fprintf(Fout, '\n\n Sensor Comments:');
fprintf(Fout, '\n %s', AC_MATRIX5{1,25});
fclose(Fout);

```

## D6. Accelerometer Sensors

```

% function ACCELEROMETER_SENSORS(ACC_TYPE,POW_TYPE,UNITS, Print)
% Accelerometers Sensor Database
%
% Developed by: Keith Schreck
% Mechanical and Aerospace Engineering
% San Jose State University
% Date: Fall 2007
%
% This file contains Accelerometers Sensor values for the ISSPO program.
% Data here is loaded into the main program to determine the proper sensor
% to use for Missions based on the Planetary Atmosphere composition.
% Values are assigned to a variable name and saved to a Matlab '.mat' file
% and loaded when calling the ISSPO program.
%
% Some of the values are based on the Unit system chosen and are loaded
% in separate sections.
%
% References
% http://www.sensotec.com/catpages.asp
% http://www.omega.com/toc_esp/subsectionsC.asp?subsection=K02&book=Pressure
% http://www.endevco.com/feature/Featured.aspx?page=71
% http://www.bksv.com/pdf/2271A_AM20.pdf
% http://www.honeywell.com/sites/servlet/com.merx.npoint.servlets.DocumentServlet?docid=D732BF1BD-AFCB-3204-9D96-
% 2292C529E86F
%
function ACCELEROMETER_SENSORS(ACC_TYPE,POW_TYPE,UNITS, Print)
%
% Load Constants values into Local Program
load('Constants_DB.mat');
load('Planet_Data.mat');
%
% Define Local Temperature Variable
PLTEMP = Planet_Properties{7};
%
% Define Program Variables

```

```

if strcmpi(ACC_TYPE,'BALLISTIC')== 1
    ACC_RANGE = 10000;
    TOL_LOWER = 0;
end
if strcmpi(ACC_TYPE,'IMPACT')== 1
    ACC_RANGE = 10000;
    TOL_LOWER = 1001;
end
if strcmpi(ACC_TYPE,'HIGH')== 1
    ACC_RANGE = 1000;
    TOL_LOWER = 101;
end
if strcmpi(ACC_TYPE,'MEDIUM')== 1
    ACC_RANGE = 100;
    TOL_LOWER = 11;
end
if strcmpi(ACC_TYPE,'SOFT')== 1
    ACC_RANGE = 10;
    TOL_LOWER = 0;
end

% Common Accelerator Properties
% Properties [
% RESONANCE_FREQ. SHOCK_LIMITS VIB_LIMITS SENSOR
% Units [ [ [ [
Hz ACCELEROMETER_SENSOR_DB1 = {'JTF-GPA' 10000.0 10000.0 0 'MA11'
5000.00 2000.0 2000.0 0 'MA12'
18000.00 2000.0 2000.0 0 'MA15'
5000.00 0 0 'MA21'
18000.00 0 0 'MA23'
25000.00 0 0 'MA311'
18000.00 1000.0 1000.0 0 'MA321'
18000.00 1000.0 1000.0 0 'MA331'
18000.00 1000.0 1000.0 0 'MA342'
22000.00 0 0 'MAQ13'
30000.00 0 0 'MAQ14'
18000.00 0 0 'MAQ36'
45000.00 0 0 'MAQ41'
10000.00 0 0 'MAQ41'

% Dynamic Range
% INPUT Low g's High
Type 'N/A'
'-80.0 80.0
'Const Current' 0.500 8.000
'-80.0 80.0
'Const Current' 0.500 8.000
'-50.0 50.0
'Voltage' 4.000 20.000
'-80.0 80.0
'Const Current' 0.500 8.000
'-10.0 10.0
'Const Current' 5.000 8.000
'-50.0 50.0
'Voltage' 7.000 7.000
'-50.0 50.0
'Voltage' 4.000 20.000
'-50.0 50.0
'Voltage' 7.000 7.000
'-80.0 80.0
'Const Current' 0.500 8.000

% Sensitivity
% CURRENT RANGE Low mA High
mv/G 0 0.08
100.0 100.0
0.500 8.000
0.500 8.000
4.000 20.000
100.0 100.0
7.000 7.000
4.000 20.000
7.000 7.000
0.500 8.000
0 20.0
0 50.0
0 5.0
0 1100.0
0 0

% Transverse Sens
% BIAS VOLTAGE V Type
5.00 5.00 'DC'
10.0 10.0 'DC'
12.0 12.0 'DC'
12.0 12.0 'DC'
5.00 32.0 'DC'
5.00 5.00 'DC'
12.0 12.0 'DC'
12.0 12.0 'DC'
5.00 24.0 'DC'
5.00 30.0 'DC'
5.00 5.00 'DC'
5.00 12.0 'DC'
5.00 5.00 'Charge'
5.00 5.00 'Charge'
5.00 5.00 'Charge'
5.00 5.00 'Charge'

% Frequency Range
% ELECT_NOISE Low Hz High mg's
0.00 0 2400.00 1.00
2.00 0 10000.00 1.00
2.00 0.10 8000.00 1.00
2.00 0.10 1000.00 1.00
2.00 0.30 9000.00 1.00
5.00 0.10 14000.00 1.00
1.00 0.10 1000.00 1.00
1.00 10.00 1000.00 1.00
1.00 10.00 1000.00 1.00
1.00 10.00 15000.00 1.00
0 0 0 1.00
1.00 10000.00 1.00
1.00 0 30000.00 1.00
0 0 0 1.00
0 0 0 1.00

% Linearity
% LINEARITY %
%1
%2
%3
%4
%5
%6
%7
%8
%9
%10
%11
%12
%13
%14

```

```

18000.00      0      'MAT53'      -80.0      80.0      100.0      5.00      'DC'      2.00      10000.00      1.00
0      'Const Current'      0.500      8.000
'PA'      -1000.0      1000.0      5.0      12.0      'DC'      3.00      5000.00      0
2000.0      3000.0      2000.0      0      16.5      'DC'      3.00      10000.00      2.00
'ACC301'      -500.0      500.0      10.0      5.00      'DC'      3.00      10000.00      0
0      'Const Current'      2.000      2.000      1.50      'DC'      0.00      400.00      0
'OM-CP-EB'      -100.0      100.0      0      9.0      'DC'      0.00      100000.00      2.00
0      'N/A'      0      0      5.00      10.0      'DC'      0.50      7000.00      1.00
'71-60K'      -60000.0      60000.0      3000.0      3.00      'Charge'      20.00      6000.00      1.00
0      'Voltage'      0      0      1.00      10.0      'DC'      20.00      6000.00      1.00
'2271A/AM20'      1000.0      'Self Generating'      0      0      10.0      'DC'      20.00      2000.00      0
10000.0      10000.0      -2000.0      2000.0      0.20      0      0.30      0      0
'7264D'      1000.0      'Voltage'      0      0      10.0      'DC'      0.30      0
40000.00      10000.0      'QA2000-030'      -60.0      60.0      28.0      'DC'      0.30      0
0      250.0      'Const Current'      16.000      16.000

```

```

AccUnitLabelStr1 = { 'N/A' 'g's' 'g's' 'mV/g' '%' 'Hz' 'Hz' '%' 'Hz' 'g's' 'g's' 'Type' 'mA' 'mA' 'V'
'Type' 'mG's' };

```

```

***
*** Accelerometer Sensor Data Base Created Based on un Unit System
*** 2 - D Matrix ACCELEROMETER_SENSOR_DB(A_Type, Property)
if strcmpi(UNITS, 'SI') == 1
% Properties
% Units
% ACCELEROMETER_SENSOR_DB2 = {
PROBE TEMP RANGE      WEIGHT      DIMENSIONS PROBE      TEMP SENS      POWER
Low deg C      High      gr      Length      Width      Height      g's/deg C      W
21.1      93.3      129.3      3.556      3.556      3.556      0      0      %1
-56.7      137.8      150.3      6.706      2.489      2.489      0.261      0      %2
-56.7      137.8      170.1      5.283      2.896      2.540      0.261      0      %3
-23.3      98.9      141.7      5.690      2.896      2.896      0.261      0.640      %4
-12.2      137.8      110.6      4.140      1.702      1.702      0.261      0      %5
-56.7      121.1      8.5      1.397      1.397      1.397      0.261      0      %6
-17.8      82.2      39.7      2.540      2.540      2.540      0.261      0.168      %7
-17.8      82.2      39.7      2.540      2.540      2.540      0.261      0      %8
-17.8      82.2      39.7      2.540      2.540      2.540      0.261      0      %9
-23.3      137.8      28.3      4.089      1.880      1.880      0.261      0      %10
-56.7      248.9      28.3      2.388      1.500      1.500      0.261      0      %11
-56.7      248.9      34.0      2.007      1.905      1.905      0.261      0      %12
-56.7      248.9      7.1      1.194      0.838      0.838      0.261      0      %13
-56.7      248.9      99.2      4.191      3.200      3.200      0.261      0      %14
-23.3      137.8      107.7      5.004      2.489      2.489      0.261      0      %15
0      0      85.0      4.064      2.540      2.540      0      0      %16
-55.0      120.0      10.0      2.030      2.030      2.030      0      0      %17
-20.0      54.0      2300.0      8.900      11.100      2.600      0      0      %18
-54.0      121.1      0.06      0.635      0.635      0.635      72.000      0      %19
-269.0      260.0      27.0      1.588      1.588      1.588      0      0      %20
-18.0      66.0      1.0      1.143      1.016      0.513      2.000      0      %21
-55.0      95.0      71.0      3.350      3.350      2.718      0.0003      0.480      %22
}
AccUnitLabelStr2 = { 'deg C' 'deg C' 'gr' 'cm' 'cm' 'g's/deg C' 'W' };

```

```

end

```

```

if strcmp(UNITS,'British')== 1
% Properties
% Units
% ACCELEROMETER_SENSOR_DB2 = {
PROBE TEMP RANGE          DIMENSIONS PROBE - in          TEMP SENS          POWER
Low deg F High           Length Width Height g's/deg F BTUs/hr
70.0 200.0                1.400 1.400 1.400 0 0
-70.0 280.0                2.640 0.980 0.980 0 0
-70.0 280.0                2.080 1.140 1.000 0 0
-10.0 210.0                1.630 1.140 1.140 0 2.186
10.0 280.0                1.630 0.670 0.670 0 0
-70.0 250.0                0.550 0.550 0.550 0 0
0.0 180.0                 1.000 1.000 1.000 0 0.574
0.0 180.0                 1.400 1.000 1.000 0 0
0.0 180.0                 1.400 1.000 1.000 0 0
-10.0 280.0                1.610 0.740 0.740 0 0
-70.0 480.0                0.940 0.590 0.590 0 0
-70.0 480.0                1.200 0.750 0.750 0 0
-70.0 480.0                0.250 0.330 0.330 0 0
-70.0 480.0                3.500 1.260 1.260 0 0
-10.0 280.0                1.970 0.980 0.980 0 0
32.0 160.0                 1.600 1.000 1.000 0 0
-4.0 129.0                 80.000 0.800 0.800 0 0
-65.0 250.0                0.002 3.500 1.000 40.000 0
-452.0 500.0               1.150 0.625 0.625 0 0
0.0 150.0                  0.400 0.400 0.202 1.200 0
-67.0 203.0               1.320 1.320 1.070 0.00017 1.640
};
%22
AccUnitLabelStr2 = { 'deg F' 'oz' 'in' 'in' 'in' 'g's/deg F' 'BTUs/hr' };
end
% Combine Data Arrays to a Single Matrix
ACCELEROMETER_SENSOR_DB = cat(2,ACCELEROMETER_SENSOR_DB1,ACCELEROMETER_SENSOR_DB2);
AccUnitLabelStr = cat(2,AccUnitLabelStr1,AccUnitLabelStr2);
% Additional Comments on Sensor
ACCELEROMETER_SENSOR_DB{ 1,26} = '3 JTF Units mounted on Triaxial Mounting block Records 3 axis data simultaneously.';
ACCELEROMETER_SENSOR_DB{ 2,26} = 'General Purpose.';
ACCELEROMETER_SENSOR_DB{ 3,26} = 'General Purpose.';
ACCELEROMETER_SENSOR_DB{ 4,26} = 'General Purpose. 4 - 20 mA Proportional to G.';
ACCELEROMETER_SENSOR_DB{ 5,26} = 'General Purpose. Miniature Size, Low Profile. Sensitivity up to 100mV/G special order.';
ACCELEROMETER_SENSOR_DB{ 6,26} = 'Miniature Size, Low Profile. Sensitivity down to 100mV/G special order.';
ACCELEROMETER_SENSOR_DB{ 7,26} = 'DC response. Amplified Voltage Output. Range +/-1, 2, 5, 10, 20, 50 g's available';
ACCELEROMETER_SENSOR_DB{ 8,26} = 'DC response. 4-20 mA output proportional to G. Built in signal conditioning. 12 mA @ 0 G sensitivity scaled at full range. Range +/-1, 2, 5, 10, 20, 50 g's available';
ACCELEROMETER_SENSOR_DB{ 9,26} = 'DC response. 15-75 mV Full Scale Output Range. Use with Strain Gage Amps. Range +/-1, 2, 5, 10, 20, 50 g's available';
ACCELEROMETER_SENSOR_DB{ 10,26} = 'High Frequency. Sensitivity up to 1000mV/G special order.';
ACCELEROMETER_SENSOR_DB{ 11,26} = 'General Purpose. Small Size. Charge Output format requires Charge Amplifier sets Frequency Range. Sensitivity in pc/G at 10, 20, 30 special order.';
ACCELEROMETER_SENSOR_DB{ 12,26} = 'General Purpose. Small Size. Charge Output format requires Charge Amplifier. Sensitivity in pc/G at 10, 30, 50 special order.';

```

```

ACCELEROMETER_SENSOR_DB{13,26} = 'Sub-Miniature Size. Charge Output format requires Charge Amplifier. Sensitivity in pC/G at
0.5, 5, 50 special order.';
ACCELEROMETER_SENSOR_DB{14,26} = 'Enhanced Output. Charge Output format requires Charge Amplifier sets Frequency Range.
Sensitivity in pC/G.';
ACCELEROMETER_SENSOR_DB{15,26} = 'Temperature and Vibration Output. Sensitivity in mV/G at 10, 30, 100, 1000 special order.
Temp range 0 - 200 deg F.';
ACCELEROMETER_SENSOR_DB{16,26} = 'Rugged, High Impact. Enhanced Sensitivity mV/G in 1000, 500, 100, 50, 10, 5 at 5, 10, 50,
100, 500, 1000 G's. 13.5-16.5 VDC Source 5VDC Output.';
ACCELEROMETER_SENSOR_DB{17,26} = 'Triaxial Lightweight Accelerometer. Low Noise, Wide Frequency, Rugged Design.';
ACCELEROMETER_SENSOR_DB{18,26} = 'Temp, Humidity, Pressure, & Tri-Axial Shock Data Logger. Specific Operating Properties each.
60 day Battery Operation. 180 day start delay. PC Software Interface.';
ACCELEROMETER_SENSOR_DB{19,26} = 'Single Axis Accelerometer. Similar Model to type flown on Mars Deep Space 2 Mission. Three
Axis model available with different ratings each direction.';
ACCELEROMETER_SENSOR_DB{20,26} = 'Single Axis, wide temperature range piezoelectric accelerometer for cryogenic applications.
Self generating power device. Flight Proven - Huygens Lander SSP.';
ACCELEROMETER_SENSOR_DB{21,26} = 'Single Axis, piezoresistive accelerometer. Small size, rugged design. Ideal for long
duration transient shocks. Flight Proven - Huygens Lander Penetrator SSP.';
ACCELEROMETER_SENSOR_DB{22,26} = 'Single Axis, piezoresistive accelerometer. Long-term repeatability and superior reliability.
Dual built-in self-test. Flight Proven - Huygens Lander Descent phase HASI.';

```

```

% Select Acceleration Sensor by Input Operation Range Value

```

```

c = 1;
for a = 1:size(ACCELEROMETER_SENSOR_DB,1)
    if strcmpi(ACC_TYPE,'BALLISTIC')==1
        if (ACCELEROMETER_SENSOR_DB{a,3} > ACC_RANGE)
            for s = 1:size(ACCELEROMETER_SENSOR_DB,2)
                ACC_MATRIX{c,s} = ACCELEROMETER_SENSOR_DB{a,s};
            end
            c = c + 1;
        end
    end
    if strcmpi(ACC_TYPE,'IMPACT')==1
        if (ACCELEROMETER_SENSOR_DB{a,3} <= ACC_RANGE) && (ACCELEROMETER_SENSOR_DB{a,3} >= TOL_LOWER)
            for s = 1:size(ACCELEROMETER_SENSOR_DB,2)
                ACC_MATRIX{c,s} = ACCELEROMETER_SENSOR_DB{a,s};
            end
            c = c + 1;
        end
    end
    if strcmpi(ACC_TYPE,'HIGH')==1
        if (ACCELEROMETER_SENSOR_DB{a,3} <= ACC_RANGE) && (ACCELEROMETER_SENSOR_DB{a,3} >= TOL_LOWER)
            for s = 1:size(ACCELEROMETER_SENSOR_DB,2)
                ACC_MATRIX{c,s} = ACCELEROMETER_SENSOR_DB{a,s};
            end
            c = c + 1;
        end
    end
    if strcmpi(ACC_TYPE,'MEDIUM')==1
        if (ACCELEROMETER_SENSOR_DB{a,3} <= ACC_RANGE) && (ACCELEROMETER_SENSOR_DB{a,3} >= TOL_LOWER)
            for s = 1:size(ACCELEROMETER_SENSOR_DB,2)
                ACC_MATRIX{c,s} = ACCELEROMETER_SENSOR_DB{a,s};
            end
        end
    end
end

```

```

end
c = c + 1;
end
end
if strcmpi(ACC_TYPE,'SOFT')== 1
if (ACCELEROMETER_SENSOR_DB{a,3} <= ACC_RANGE)
for s = 1:size(ACCELEROMETER_SENSOR_DB,2)
ACC_MATRIX{c,s} = ACCELEROMETER_SENSOR_DB{a,s};
end
c = c + 1;
end
end
tmp = 1;
for a = 1:size(ACC_MATRIX,1)
if strcmpi(ACC_MATRIX{a,12},POW_TYPE) == 1
for s = 1:size(ACC_MATRIX,2)
ACC_MATRIX2{tmp,s} = ACC_MATRIX{a,s};
end
tmp = tmp + 1;
end
end
end

% Continue Selection Process From Viable Options - If Multiple Sensor Options Exist
if size(ACC_MATRIX2,1) > 1

% Select Via Temperature Operational Range
c = 1;
% Build Up Database of Possible Acceleration Sensors Based on Temperature Range
for t = 1:size(ACC_MATRIX2,1)
if (PLTEMP >= ACC_MATRIX2{t,18}) && (PLTEMP <= ACC_MATRIX2{t,19})
for a = 1:size(ACC_MATRIX2,2)
ACC_MATRIX3{c,a} = ACC_MATRIX2{t,a};
end
c = c+1;
end
end

% Copies ACC_MATRIX2 to ACC_MATRIX3 if all Sensors Eliminated based on Temperature Range
if exist('ACC_MATRIX3')== 0
ACC_MATRIX3 = ACC_MATRIX2;
end

if size(ACC_MATRIX3,1) >1
% Down Select by Min Electrical Noise
for d = 1:size(ACC_MATRIX3,1)
NOISE_ACC_DELTA(d) = ACC_MATRIX3{d,24};
end
% Find the Minimum Accelerometer Electrical Noise Sensor
MIN_NOISE_ACC_DELTA = min(NOISE_ACC_DELTA);

```



```

% For Multiple Sensors with the Same Range Reduce Sensor Matrix to Those
% with the Least Range
k = 1;
for c = 1:length(ACC_DELTA)
    if NOISE_ACC_DELTA(c) == MIN_NOISE_ACC_DELTA
        for j = 1:size(ACC_MATRIX3,2)
            ACC_MATRIX4{k,j} = ACC_MATRIX3{c,j};
        end
        k = k + 1;
    end
end

end

% Copies ACC_MATRIX3 to ACC_MATRIX4 if all Sensors Eliminated based on Temperature Range
if exist('ACC_MATRIX4') == 0
    ACC_MATRIX4 = ACC_MATRIX3;
end

if size(ACC_MATRIX4,1) > 1

    % Down Select by Max Temperature Range
    for d = 1:size(ACC_MATRIX4,1)
        TEMP_ACC_DELTA(d) = ACC_MATRIX4{d,19} - ACC_MATRIX4{d,18};
    end

    % Find the Maximum Accelerometer Temperature Range Sensor
    MAX_TEMP_ACC_DELTA = max(TEMP_ACC_DELTA);

    % For Multiple Sensors with the Same Range Reduce Sensor Matrix to Those
    % with the Greatest Range
    k = 1;
    for c = 1:length(TEMP_ACC_DELTA)
        if TEMP_ACC_DELTA(c) == MAX_TEMP_ACC_DELTA
            for j = 1:size(ACC_MATRIX4,2)
                ACC_MATRIX5{k,j} = ACC_MATRIX4{c,j};
            end
            k = k + 1;
        end
    end

end

end

% Copies ACC_MATRIX4 to ACC_MATRIX5 if all Sensors Eliminated based on Temperature Range
if exist('ACC_MATRIX5') == 0
    ACC_MATRIX5 = ACC_MATRIX4;
end

end

if size(ACC_MATRIX5,1) > 1
    % Create Mass / Volume Scaling Factor Optimization
    for mv = 1:size(ACC_MATRIX5,1)

```

```

ACC_SCALE_FAC(mv) = ACC_MATRIX5{mv,20} * ACC_MATRIX5{mv,21} * ACC_MATRIX5{mv,22} * ACC_MATRIX5{mv,23};
if (ACC_MATRIX5{mv,20} == 0) || (ACC_MATRIX5{mv,21} == 0) || (ACC_MATRIX5{mv,22} == 0) || (ACC_MATRIX5{mv,23} == 0)
    ACC_SCALE_FAC(mv) = 9.99E99;
end
end
[ ACC_MIN_SCL, INDEX ] = min(ACC_SCALE_FAC);
% Build Sensor Matrix Based on Scale Factors
k = 1;
for c = 1:length(ACC_SCALE_FAC)
    if ACC_SCALE_FAC(c) == ACC_MIN_SCL
        for j = 1:size(ACC_MATRIX4,2)
            ACC_MATRIX6{k,j} = ACC_MATRIX5{c,j};
        end
        k = k + 1;
    end
end
end
% Copies ACC_MATRIX5 to ACC_MATRIX6 if all Sensors Eliminated based on Mass Volume Scale Factor
if exist('ACC_MATRIX6') == 0
    ACC_MATRIX6 = ACC_MATRIX5;
end
end
% Copies ACC_MATRIX2 to ACC_MATRIX6 if all Sensors Eliminated based on Mass Volume Scale Factor
if exist('ACC_MATRIX6') == 0
    ACC_MATRIX6 = ACC_MATRIX2;
end
end

% Save Accelerometer Sensor Properties to Data File
save('ACCELEROMETER_SENSOR_Data.mat');

save('ACCELEROMETER_SENSOR_Final.mat','ACC_MATRIX6','AccUnitLabel1str');
%*****
%
% *** PRINT SUMMARY Output Section for 'Print' = 'Y'
if strcmpi(Print,'Y') == 1
    fprintf('\n\n *** ACCELEROMETER SENSOR SUMMARY RESULTS ***\n');
    fprintf('\n\n *** I N P U T S ***\n');
    fprintf('\n\n Mission Acceleration Profile: %s', char(ACC_TYPE));
    fprintf('\n\n Accelerometer Power Type: %s', char(POW_TYPE));
    fprintf('\n\n Unit System: %s', UNITS);
    fprintf('\n\n\n*****\n\n\n*****\n\n\n');
% Accelerometer Sensor Properties

```

```

fprintf('\n\n** Sensor Properties **');
fprintf('\n Accelerometer Sensor Type:');
fprintf('\n Accelerometer Sensing Range - Low:');
fprintf('\n Accelerometer Sensing Range - High:');
fprintf('\n Sensor Sensitivity:');
fprintf('\n Transverse Sensitivity:');
fprintf('\n Operating Frequency Range - Low:');
fprintf('\n Operating Frequency Range - High:');
fprintf('\n Sensor Linearity:');
fprintf('\n Resonance Frequency:');
fprintf('\n Shock Limits:');
fprintf('\n Vibration Limits:');

    %s', ACC_MATRIX6{1,1} };
    %s', ACC_MATRIX6{1,2} };
    %s', ACC_MATRIX6{1,3} };
    %s', ACC_MATRIX6{1,4} };
    %s', ACC_MATRIX6{1,5} };
    %s', ACC_MATRIX6{1,6} };
    %s', ACC_MATRIX6{1,7} };
    %s', ACC_MATRIX6{1,8} };
    %s', ACC_MATRIX6{1,9} };
    %s', ACC_MATRIX6{1,10} };
    %s', ACC_MATRIX6{1,11} };
    %s', ACC_MATRIX6{1,12} };
    %s', ACC_MATRIX6{1,13} };
    %s', ACC_MATRIX6{1,14} };
    %s', ACC_MATRIX6{1,15} };
    %s', ACC_MATRIX6{1,16} };
    %s', ACC_MATRIX6{1,17} };
    %s', ACC_MATRIX6{1,25} };

    %s', ACC_MATRIX6{1,12} };
    %s', ACC_MATRIX6{1,13} };
    %s', ACC_MATRIX6{1,14} };
    %s', ACC_MATRIX6{1,15} };
    %s', ACC_MATRIX6{1,16} };
    %s', ACC_MATRIX6{1,17} };
    %s', ACC_MATRIX6{1,25} };

    %s', ACC_MATRIX6{1,18} };
    %s', ACC_MATRIX6{1,19} };
    %s', ACC_MATRIX6{1,24} };

    %s', ACC_MATRIX6{1,20} };
    %s', ACC_MATRIX6{1,21} };
    %s', ACC_MATRIX6{1,22} };
    %s', ACC_MATRIX6{1,23} };

    %s', ACC_MATRIX6{1,26} };

    % Environmental Properties
    fprintf('\n\n Environmental Properties');
    fprintf('\n Sensor Temperature Range - Low:');
    fprintf('\n Sensor Temperature Range - High:');
    fprintf('\n Temperature Sensitivity:');

    %s', ACC_MATRIX6{1,18} };
    %s', ACC_MATRIX6{1,19} };
    %s', ACC_MATRIX6{1,24} };

    %s', ACC_MATRIX6{1,20} };
    %s', ACC_MATRIX6{1,21} };
    %s', ACC_MATRIX6{1,22} };
    %s', ACC_MATRIX6{1,23} };

    %s', ACC_MATRIX6{1,26} };

    %*** Writes a text file summary of the Planetary Properties
    Fout = fopen('ACCELEROMETER_SENSOR_Summary.txt','w+');

    fprintf(Fout, '\n\n *** A C C E L E R O M E T E R S E N S O R S U M M A R Y R E S U L T S ***');
    fprintf(Fout, '\n\n *** I N P U T S ***');
    fprintf(Fout, '\n Mission Acceleration Profile: %s', char(ACC_TYPE));
    fprintf(Fout, '\n Accelerometer Power Type: %s', char(POW_TYPE));
    fprintf(Fout, '\n Unit System: %s', UNITS);

```

end



```

D7. Acoustic Sensors
% function ACOUSTIC_SENSORS(ACS_TYPE,UNITS,Print)
% Acoustic Sensor Database
%
% Developed by: Keith Schreck
% Mechanical and Aerospace Engineering
% San Jose State University
% Date: Fall 2007
%
% This file contains Acoustic Sensor values for the ISSPO program.
% Data here is loaded into the main program to determine the proper sensor
% to use for Missions based on the sensor requirements.
% Values are assigned to a variable name and saved to a Matlab '.mat' file
% and loaded when calling the ISSPO program.
%
% Some of the values are based on the Unit system chosen and are loaded
% in separate sections.
%
% References
% http://www.bksv.com/pdf/bp2016.pdf
% http://www.bksv.com/pdf/bp1803.pdf
% http://www.bksv.com/pdf/bp1844.pdf
% http://www.pcb.com/Linked_Documents/Vibration/Mics_1006.pdf
% http://www.microflown.com/data/manuals_datasheets/probes/Datasheet_PU-mini_v1.0.pdf
% http://www.pacndt.com/index.aspx?go=products&focus=/sensors/general_purpose.htm
%
function ACOUSTIC_SENSORS(ACS_TYPE,UNITS,Print)
%
% Load Constants values into Local Program
load('Constants_DB.mat');
load('Planet_Data.mat');
PLTEMP = Planet_Properties{7};
%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%
%% ACOUSTIC Sensor Data Base Created Based on un Unit System
%% 2 - D Matrix ACOUSTIC_DB(A_Type, Property)
%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%
% Acoustic Sensor Common Properties
% Properties [ SENSOR
INHERENT NOISE CLIPPING LIMIT VOLTAGE SUPPLY NOM CURRENT FREQUENCY RANGE STABILITY RESONANCE
% Units [ # Min V Max dB Low dB High Hz High Hz Low Tol High Hz
ACOUSTICSENSOR_DB1 = { 'B&K Type 4948', 55 160 4 0.001 20 20000 50
58 165 22 30 30 100 5000 50 'SENSOR' ; 2 24000
35 140 0 0 0 0 0 0 'SENSOR' ; 3 0
37 172 200 30 172 4 0.001 20 70000 0 'SENSOR' ; 3 60000
}

```

```

28      165      'PCB 377B01'      28      165      4      80000      0      -2      2      0
15      146      'PCB 377B02'      15      146      3.15      20000      0      -2      2      0
30      170      'PCB 377B10'      30      170      4      70000      0      -2      2      0
15      146      'PCB 377B11'      15      146      3.15      10000      0      -2      2      0
30      187      'PCB 377A12'      30      187      4      20000      0      -2      2      0
15      146      'PCB 377B20'      15      146      3.15      12500      0      -2      2      0
20      160      'PCB 377B40'      20      160      4      40000      0      -2      2      0
15      146      'PCB 377B41'      15      146      3.15      20000      0      -2      2      0
10      146      'PCB 377A42'      10      146      2.60      20000      0      -2      2      0
40      178      'PCB 377A50'      40      178      6.50      140000      0      -2      2      0
31      192      'PCB 377A51'      31      192      10.0      20000      0      -2      2      0
10      146      'PCB 377A53'      10      146      2.60      8000      0      -2      2      0
15      146      'PCB 377A54'      15      146      3.15      10000      0      -2      2      0
15      146      'PCB 377A60'      15      146      3.15      10000      0      -2      2      0
56.8    'CT-190M'      56.8    135      10      2000      0      0      0      500000
10      135      'PU-Mini'      10      125      1.00      20000      0      0      0      0
35      'Array Type 4935'      35      140      100      5000      0      -2      2      0
0      135      'USP-Mini'      0      135      0.3      40      90      0      0      0
58      120      'R80-Alpha'      0      62      200000      1000000      0      0      0      800000

```

```

AcousticUnitLabelStr1 = { 'N/A' 'dB' 'Hz' 'dB' 'dB' 'Hz' 'dB' 'dB' 'Hz' 'dB' 'dB' 'V' 'V' 'mA' 'dB/yr' 'deg C' '%RH'
'N/A' };

```

```

if strcmpi(UNITS,'SI')==1
% Properties
WEIGHT      [
% Units      [ Low deg C High Low deg C High Low deg C High Low deg C High Low deg C High Low deg C High
Grams      ACOUTICSSENSOR_DB2 = { -50.0 100.0 100.0 70.0 70.0 70.0 70.0 70.0 70.0 70.0 70.0 70.0 70.0
2.3      ;
10.0      ;
0      ;

```

```

0 / -40.0 150.0 -40.0 150.0 0 0 6.35 6.35 0
0 / -40.0 120.0 -40.0 120.0 0 0 12.7 12.7 0
0 / -40.0 150.0 -40.0 150.0 0 0 6.35 6.35 0
0 / -40.0 120.0 -40.0 120.0 0 0 12.7 12.7 0
0 / -40.0 120.0 -40.0 120.0 0 0 6.35 6.35 0
0 / -40.0 120.0 -40.0 120.0 0 0 12.7 12.7 0
0 / -40.0 150.0 -40.0 150.0 0 0 12.7 12.7 0
0 / -40.0 150.0 -40.0 150.0 0 0 12.7 12.7 0
0 / -40.0 150.0 -40.0 150.0 0 0 25.4 25.4 0
0 / -40.0 150.0 -40.0 150.0 0 0 3.175 3.175 0
0 / -40.0 150.0 -40.0 150.0 0 0 6.35 6.35 0
0 / -40.0 150.0 -40.0 150.0 0 0 25.4 25.4 0
0 / -40.0 150.0 -40.0 150.0 0 0 12.7 12.7 0
0 / -40.0 150.0 -40.0 150.0 0 0 12.7 12.7 0
0 / -195.5 37.0 -195.5 120.0 0 0 34.3 9.5 9.5
4.0 / 0.0 60.0 0.0 60.0 0 0 40.0 12.7 12.7
25.0 / -10.0 55.0 -25.0 70.0 0 0 65.0 7.0 7.0
10.0 / 0.0 60.0 0.0 60.0 0 0 62.0 12.7 12.7
30.0 / -65.0 175.0 -65.0 175.0 0 0 19.0 19.0 21.4
32.0 / };

```

```

AcousticUnitLabelStr2 = { 'deg C' 'deg C' 'deg C' 'dB/deg C' 'dB/kPa' 'mm' 'mm' 'gram' };
end
if strcmp(UNITS,'BRITISH')==1
% Properties [ OPERATING TEMP STORAGE TEMP TEMP COEFFICIENT PRESSURE COEFFICIENT DIMENSIONS (in)
WEIGHT [ Units [ Low deg F High Low deg F High @ 250 Hz dB/psi @ 250 Hz dB/psi L W H
oz ]
ACOUSTICSENSOR_DB2 = { -67.0 212.0 -22.0 158.0 0.0072 -0.4826 0.79 0.79 0.10
0.08 ;
0.35 ;
0 ;

```

```

0      ?      -40.0      302.0      -40.0      302.0      0      0      0.25      0.25      0
0      ?      -40.0      248.0      -40.0      248.0      0      0      0.50      0.50      0
0      ?      -40.0      302.0      -40.0      302.0      0      0      0.25      0.25      0
0      ?      -40.0      248.0      -40.0      248.0      0      0      0.50      0.50      0
0      ?      -40.0      248.0      -40.0      248.0      0      0      0.25      0.25      0
0      ?      -40.0      248.0      -40.0      248.0      0      0      0.50      0.50      0
0      ?      -40.0      302.0      -40.0      302.0      0      0      0.50      0.50      0
0      ?      -40.0      302.0      -40.0      302.0      0      0      1.00      1.00      0
0      ?      -40.0      302.0      -40.0      302.0      0      0      0.125      0.125      0
0      ?      -40.0      302.0      -40.0      302.0      0      0      0.25      0.25      0
0      ?      -40.0      302.0      -40.0      302.0      0      0      1.00      1.00      0
0      ?      -40.0      302.0      -40.0      302.0      0      0      0.50      0.50      0
0      ?      -40.0      302.0      -40.0      302.0      0      0      0.50      0.50      0
0      ?      -40.0      302.0      -40.0      302.0      0      0      1.35      0.38      0.38
0.14      ?      32.0      140.0      32.0      140.0      0      0      1.58      0.50      0.50
0.88      ?      14.0      131.0      -13.0      158.0      0      0      2.6      0.25      0.25
0.35      ?      32.0      140.0      32.0      140.0      0      0      2.44      0.50      0.50
1.06      ?      -85.0      347.0      -85.0      347.0      0      0      0.75      0.75      0.84
1.13      ?      ;

```

```

end
AcousticUnitLabelStr2 = { 'deg F' 'deg F' 'deg F' 'deg F' 'dB/psi' 'in' 'in' 'in' 'oz' };

```

```

%Comments on sensors - uses
ACOUSTICSENSOR_DB3{ 1,1} = 'Measurement of sound pressure on surfaces, Acoustic-fatigue testing of airplanes, Wind-tunnel
measurements, Medium- to high-level measurements, Measurement in confined spaces';
ACOUSTICSENSOR_DB3{ 2,1} = 'Simultaneous recording of time signals in medium to large microphone arrays, for example, simulated
pass-by measurements, Spatial Transformation of Sound Fields (STSF) measurements';
ACOUSTICSENSOR_DB3{ 3,1} = 'For measurements in confined spaces and small cavities. High level, high frequency measurements.
Optimised for flush mounting';
ACOUSTICSENSOR_DB3{ 4,1} = 'Free-field Precision Condenser Microphone Cartridges. Exceptional performance in high humidity.';
ACOUSTICSENSOR_DB3{ 5,1} = 'Free-field Precision Condenser Microphone Cartridges. Exceptional performance in high humidity.';
ACOUSTICSENSOR_DB3{ 6,1} = 'Pressure Precision Condenser Microphone Cartridges. Exceptional performance in high humidity.';

```



```

ACOUSTICSSENSOR_DB3{ 7,1 } = 'Pressure Precision Condenser Microphone Cartridges. Exceptional performance in high humidity.';
ACOUSTICSSENSOR_DB3{ 8,1 } = 'Pressure Precision Condenser Microphone Cartridges. Exceptional performance in high humidity.';
ACOUSTICSSENSOR_DB3{ 9,1 } = 'Random Incidence Precision Condenser Microphone Cartridges. Exceptional performance in high humidity.';
ACOUSTICSSENSOR_DB3{10,1} = 'Free-Field Externally Polarized Precision Condenser Microphone Cartridges. Exceptional performance in high humidity.';
ACOUSTICSSENSOR_DB3{11,1} = 'Free-Field Externally Polarized Precision Condenser Microphone Cartridges. Exceptional performance in high humidity.';
ACOUSTICSSENSOR_DB3{12,1} = 'Free-Field Externally Polarized Precision Condenser Microphone Cartridges. Exceptional performance in high humidity.';
ACOUSTICSSENSOR_DB3{13,1} = 'Pressure Externally Polarized Precision Condenser Microphone Cartridges. Exceptional performance in high humidity.';
ACOUSTICSSENSOR_DB3{14,1} = 'Pressure Externally Polarized Precision Condenser Microphone Cartridges. Exceptional performance in high humidity.';
ACOUSTICSSENSOR_DB3{15,1} = 'Pressure Externally Polarized Precision Condenser Microphone Cartridges. Exceptional performance in high humidity.';
ACOUSTICSSENSOR_DB3{16,1} = 'Pressure Externally Polarized Precision Condenser Microphone Cartridges. Exceptional performance in high humidity.';
ACOUSTICSSENSOR_DB3{17,1} = 'Random Incidence Externally Polarized Precision Condenser Microphone Cartridges. Exceptional performance in high humidity.';
ACOUSTICSSENSOR_DB3{18,1} = 'Pressure Sensor. Excellent for extreme conditions - temperature, pressure. Not suited for quiet sounds. Flight Proven - Huygens Lander.';
ACOUSTICSSENSOR_DB3{19,1} = 'Microflown Acoustic Sensor. Measures both particle velocity and sound pressure. Designed for the use in arrays.';
ACOUSTICSSENSOR_DB3{20,1} = 'Microphone Array Sensor. Simultaneous recording of time signals in medium to large microphone arrays. Mass, volume, power reqs per each sensor.';
ACOUSTICSSENSOR_DB3{21,1} = 'Microflown Acoustic Sensor. Measures both particle velocity and sound pressure. Designed for the use in arrays. Mass, volume, power reqs per each sensor.';
ACOUSTICSSENSOR_DB3{22,1} = 'General Purpose Acoustic Sensor. High Frequency sensor. Robust, Reliable Sensor.';

% Combine Data Arrays to a Single Matrix
ACOUSTICSSENSOR_DB = cat(2,ACOUSTICSSENSOR_DB1,ACOUSTICSSENSOR_DB2,ACOUSTICSSENSOR_DB3);
AcousticUnitLabelStr = cat(2,AcousticUnitLabelStr1,AcousticUnitLabelStr2);

tmp = 1;
for a = 1:size(ACOUSTICSSENSOR_DB,1)
    if strcmpi(ACOUSTICSSENSOR_DB{a,18},ACS_TYPE) == 1
        for s = 1:size(ACOUSTICSSENSOR_DB,2)
            ACS_MATRIX{tmp,s} = ACOUSTICSSENSOR_DB{a,s};
        end
        tmp = tmp + 1;
    end
end

% Sort Sensors by Operational Temperature Range Based on Target Planetary Temperature
c = 1;
for s = 1:size(ACS_MATRIX,1)
    if (PLTEMP >= ACS_MATRIX{s,19}) && (PLTEMP <= ACS_MATRIX{s,20})
        for a = 1:size(ACS_MATRIX,2)
            ACS_MATRIX2{c,a} = ACS_MATRIX{s,a};
        end
        c = c+1;
    end
end

```

```

end
if exist('ACS_MATRIX2') == 0
    fprintf('\n\n *** WARNING ***\n');
    fprintf('\n NO VIABLE ACOUSTIC SENSORS EXISTS IN DATABASE BASED ON OPERATIONAL TEMPERATURE RANGE!');
    fprintf('\n Insulation methods must be used to insulate Acoustic Sensors. ');
    ACS_MATRIX2 = ACS_MATRIX;
end

% Find the Max Operational Range
for d = 1:size(ACS_MATRIX2,1)
    TEMPACS_DELTA(d) = ACS_MATRIX2(d,20) - ACS_MATRIX2(d,19);
end

% Find the Maximum Acoustic Temperature Range of the Sensor
MAX_TEMPACS_DELTA = max(TEMPACS_DELTA);

% For Multiple Sensors with the Same Range Reduce Sensor Matrix to Those
% with the Greatest Range
k = 1;
for c = 1:length(TEMPACS_DELTA)
    if TEMPACS_DELTA(c) == MAX_TEMPACS_DELTA
        for j = 1:size(ACS_MATRIX2,2)
            ACS_MATRIX3{k,j} = ACS_MATRIX2{c,j};
        end
        k = k + 1;
    end
end

if size(ACS_MATRIX3,1) > 1
    % Create Volume Scaling Factor Optimization
    for v = 1:size(ACS_MATRIX2,1)
        ACS_SCALE_FAC(v) = ACS_MATRIX3{v,25} * ACS_MATRIX3{v,26} * ACS_MATRIX3{v,27};
        if (ACS_MATRIX3{v,25} == 0) || (ACS_MATRIX3{v,26} == 0) || (ACS_MATRIX3{v,27} == 0)
            ACS_SCALE_FAC(v) = 9.99E99;
        end
    end

    [ ACS_MIN_SCL, INDEX ] = min(ACS_SCALE_FAC);

    % Build Sensor Matrix Based on Scale Factors
    k = 1;
    for c = 1:length(ACS_SCALE_FAC)
        if ACS_SCALE_FAC(c) == ACS_MIN_SCL
            for j = 1:size(ACS_MATRIX3,2)
                ACS_MATRIX4{k,j} = ACS_MATRIX3{c,j};
            end
            k = k + 1;
        end
    end
end

```





```

fprintf(Fout, '\n Operational Temperature Range - Low: %10.2f %s', ACS_MATRIX4{1,19}, AcousticUnitLabelStr{19} );
fprintf(Fout, '\n High: %10.2f %s', ACS_MATRIX4{1,20}, AcousticUnitLabelStr{20} );
fprintf(Fout, '\n Storage Temperature Range - Low: %10.2f %s', ACS_MATRIX4{1,21}, AcousticUnitLabelStr{21} );
fprintf(Fout, '\n High: %10.2f %s', ACS_MATRIX4{1,22}, AcousticUnitLabelStr{22} );
fprintf(Fout, '\n Temperature Coefficient at 250 Hz: %10.2f %s', ACS_MATRIX4{1,23}, AcousticUnitLabelStr{23} );
fprintf(Fout, '\n Pressure Coefficient at 250 Hz: %10.2f %s', ACS_MATRIX4{1,24}, AcousticUnitLabelStr{24} );

fprintf(Fout, '\n\n Physical Properties');
fprintf(Fout, '\n Sensor Dimensions - Length: %9.3f %s', ACS_MATRIX4{1,25}, AcousticUnitLabelStr{25} );
fprintf(Fout, '\n Width: %9.3f %s', ACS_MATRIX4{1,26}, AcousticUnitLabelStr{26} );
fprintf(Fout, '\n Height: %9.3f %s', ACS_MATRIX4{1,27}, AcousticUnitLabelStr{27} );
fprintf(Fout, '\n Sensor Mass: %9.3f %s', ACS_MATRIX4{1,28}, AcousticUnitLabelStr{28} );

fprintf(Fout, '\n\n Sensor Comments:');
fprintf(Fout, '\n %s', ACS_MATRIX4{1,29});

fclose(Fout);

```

## D8. Density Sensors

```

% function DENSITY_SENSORS(DENS_TYPE, PLDENSITY, UNITS, Print)
% Density Sensor Database
%
% Developed by: Keith Schreck
% Mechanical and Aerospace Engineering
% San Jose State University
% Date: Fall 2007
%
% This file contains Density Sensor values for the ISSPO program.
% Data here is loaded into the main program to determine the proper sensor
% to use for Missions based on the Planetary Atmosphere composition.
% Values are assigned to a variable name and saved to a Matlab '.mat' file
% and loaded when calling the ISSPO program.
%
% Some of the values are based on the Unit system chosen and are loaded
% in separate sections.
%
% References
% http://www.mobrey.com/downloads/view_all.php?type=datasheets&classid=density
% http://www.foxboro-eckardt.com/products/167ip_en.html
%
function DENSITY_SENSORS(DENS_TYPE, PLDENSITY, UNITS, Print)

% Load Constants values into Local Program
Load('Constants_DB.mat');

```

```

load('Planet_Data.mat');

% Assign Planetary Surface Properties to Local Variable
PTEMP = Planet_Properties{7};
PLP = Planet_Properties{8};

%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%
%% Density Sensor Data Base Created Based on un Unit System
%% 2 - D Matrix DENSITYSENSOR_DB(DS_Type, Property)

if strcmpi(UNITS,'SI') == 1
    DENSITY_CNVT = kgm3_to_gcm3;
    % Properties [ { SENSOR RANGE ACCURACY REPEATABILITY TEST PRESSURE
    MAX OP. PRESSURE WEIGHT DIMENSIONS - mm POWER CURRENT VIBRATION VISCOSITY
    FLOWRATES OPERATIONAL }
    % Units kg Len TYPE Wid Hgt Low g/cc High g/cc g/cc g/cc kPa
    Type DENSITYSENSOR_DB = { 'MMS 7826' 0 0 0 25 3 0.001 0.0001 0 0 0
    'LIQUID' ;
    'MMS 7826L' 0 0 0 28 3 0.001 0.0001 0 0 0
    'LIQUID' ;
    'MMS 7828' 0 0 0 28 3 0.001 0.0001 0 0 0
    'LIQUID' ;
    'MMS 7828L' 0 0 0 28 3 0.001 0.0001 0 0 0
    'LIQUID' ;
    'MMS 7835' 0 114 3 0.00015 0.00002 0 0 0 22500
    'LIQUID' ;
    'MMA 7835' 0 114 3 0.00015 0.00002 0 0 0 22500
    'LIQUID' ;
    'EGL 7845' 0 114 3 0.005 0 0 0 15000
    'LIQUID' ;
    'EGL 7847' 0 114 3 0.005 0 0 0 3000
    'LIQUID' ;
    'MMT 7845' 0 114 3 0.00035 0.00005 0 0 0 15000
    'GAS' ;
    'MMT 7846' 0 114 3 0.00050 0.00005 0 0 0 7500
    'GAS' ;
    'MMT 7847' 0 114 3 0.00035 0.00005 0 0 0 3000
    'GAS' ;
}

```

```

15000      'GDT 7812' 0.001 4 0.001 0.002 -20.0 85.0 0 216
'GAS'      364 139 139 33 25 0 0.0006 0 0
16200      '167 LP' 0.1 1.6 0.015 0 400.0 0 0
'LIQUID'   362 270 210 0 0 -196.0 0 0 1200
DensityUnitLabelStr = { 'N/A' 'g/cc' 'g/cc' 'deg C' 'deg C' 'kPa' 'kPa' 'mm' 'mm' 'mm'
'V-DC' 'mA' 'g/s' 'g/cc' 'ltr/hr' 'Type' };
end

if strcmp(UNITS,'British') == 1
    DENSITY_CNVT = lbft3_to_lbft3;
    % Properties {
    PRESSURE MAX Op. PRESSURE WEIGHT RANGE DIMENSIONS - in ACCURACY POWER REPEATABILITY TEMPERATURE RANGE TEST
    VISCOSITY FLOWRATES OPERATIONAL ] lb/ft^3 High lb/ft^3 mA lb/ft^3 g's lb/ft^3 Low deg F High TOLERANCE
    % Units {
    Type lb Len Wid Low Hgt Hgt V DC
    DENSITYSENSOR_DB = { 'MMS 7826' 0 0 187.284 0.06243 0.06243 42 28 0.06243 0 0.6243 0 392.0 0
    'LIQUID' ;
    1450.38 0 0 'MMS 7826L' 0 0 187.284 0.06243 0.06243 45 28 0.06243 0 0.6243 0 392.0 0
    'LIQUID' ;
    3002.28 0 0 'MMS 7826' 0 0 187.284 0.06243 0.06243 45 28 0.06243 0 0.6243 0 20000 0
    'LIQUID' ;
    1450.38 0 0 'MMS 7828L' 0 0 187.284 0.06243 0.06243 45 28 0.06243 0 0.6243 0 392.0 0
    'LIQUID' ;
    3263.35 2175.57 2175.57 48.50 0 187.284 0.009364 28 0.0012485 230.0 0.00181
    0 792.516 'LIQUID' ;
    3263.35 2175.57 2175.57 48.50 0 187.284 0.009364 28 0.0012485 230.0 0.00181
    0 792.516 'LIQUID' ;
    2175.57 1450.38 1450.38 48.50 0 187.284 0.31214 28 0 320.0 0.03745
    0 792.516 'LIQUID' ;
    435.113 290.075 290.075 48.50 0 187.284 0.31214 28 0 320.0 0.03745
    0 792.516 'LIQUID' ;
    2175.57 1450.38 1450.38 61.73 0 187.284 0.02185 28 0.0031214 230.0 0.03745
    0 792.516 'MMT 7845' ;
    1087.78 725.189 725.189 61.73 0 187.284 0.031214 28 0.0031214 230.0 0.03745
    0 792.516 'MMT 7846' ;
    
```

```

435.113      290.075      'MMT 7847'      0      187.284      0.02185      0.0031214      230.0
0            792.516      'GAS'      ;      40.43      6.73      28      17      0.03745
2175.57      11.02      'GDT 7812'      0.06243      249.712      0.06243      0.124856      185.0
57.0612      14.33      5.47      5.47      33      25      0      0.03745      0
2349.61      40.79      '167 LP'      6.2428      99.885      0.93641      0      752.0
317.006      14.25      10.63      8.27      0      0      2.0      0
DensityUnitLabelStr = { 'N/A' 'lb/ft^3' 'lb/ft^3' 'lb/ft^3' 'lb/ft^3' 'deg F' 'psi' 'psi' 'lb' 'in'
'in' 'in' 'V-DC' 'mA' 'g's' 'lb/ft^3' 'lb/ft^3' 'CP' 'gal/hr' 'Type' };
end

```

```

end
DENSITYSENSOR_DB{1,21} = 'continuous real time measurement of fluid density in pipelines, open or closed tanks';
DENSITYSENSOR_DB{2,21} = 'possible of top entry only';
DENSITYSENSOR_DB{3,21} = 'continuous real time measurement of fluid density in pipelines, open or closed tanks';
DENSITYSENSOR_DB{4,21} = 'possible of top entry only';
DENSITYSENSOR_DB{5,21} = 'metering of crude and refined hydrocarbons or non-aggressive process liquids - in pipeline operating
conditions';
DENSITYSENSOR_DB{6,21} = '';
DENSITYSENSOR_DB{7,21} = 'General process, Measurement of liquids with up to 100% entrained gas';
DENSITYSENSOR_DB{8,21} = 'Hygienic, Measurement of liquids with up to 100% entrained gas';
DENSITYSENSOR_DB{9,21} = 'general process';
DENSITYSENSOR_DB{10,21} = 'corrosive applications';
DENSITYSENSOR_DB{11,21} = 'hygienic applications';
DENSITYSENSOR_DB{12,21} = 'Gas blending & Direct measurement of ethylene density - Process Gas Must be dry';
DENSITYSENSOR_DB{13,21} = 'Measuring of liquid level, interface or density with displacer (Archimedes principle) and torque
tube as transmitting element';

```

```

% Sort Database Sensors by GAS or Liquid Function Capability
tmp = 1;
for d = 1:size(DENSITYSENSOR_DB,1)
    if strcmpi(DENSITYSENSOR_DB{d,20},DENS_TYPE) == 1
        for s = 1:size(DENSITYSENSOR_DB,2)
            D_MATRIX{tmp,s} = DENSITYSENSOR_DB{d,s};
        end
        tmp = tmp + 1;
    end
end
end

```

```

c = 1;
% Build Up Database of Possible Density Sensors Based on Density Range
for d = 1:size(D_MATRIX,1)
    if (PLDENSITY*DENSITY_CNVT >= D_MATRIX{d,2}) && (PLDENSITY*DENSITY_CNVT <= D_MATRIX{d,3})
        for a = 1:size(D_MATRIX,2)
            D_MATRIX2{c,a} = D_MATRIX{d,a};
        end
        c = c+1;
    end
end
end

```



```

end

if exist('D_MATRIX2') == 0
    fprintf('\n\n *** WARNING ***');
    fprintf('\n NO VIABLE DENSITY SENSOR EXISTS IN DATABASE!');
    fprintf('\n Other methods to determine Planetary Atmospheric Density must be used.\n\n');
end

% Continue Selection Process From Viable Options - Bypass if None Exists
if exist('D_MATRIX2') == 1

    % Select Via Temperature Operational Range
    c = 1;
    % Build Up Database of Possible Density Sensors Based on Temperature Range
    for t = 1:size(D_MATRIX2,1)
        if (PLTEMP >= D_MATRIX2{t,6}) && (PLTEMP <= D_MATRIX2{t,7})
            for a = 1:size(D_MATRIX2,2)
                D_MATRIX3{c,a} = D_MATRIX2{t,a};
            end
            c = c+1;
        end
    end

    % Copies D_MATRIX2 to D_MATRIX3 if all Sensors Eliminated based on Temperature Range
    if exist('D_MATRIX3') == 0
        D_MATRIX3 = D_MATRIX2;
    end

    % Select Via Pressure Operational Range
    c = 1;
    % Build Up Database of Possible Density Sensors Based on Pressure Range
    for t = 1:size(D_MATRIX3,1)
        if (PLPRESSURE >= D_MATRIX3{t,6}) && (PLPRESSURE <= D_MATRIX3{t,7})
            for a = 1:size(D_MATRIX3,2)
                D_MATRIX4{c,a} = D_MATRIX3{t,a};
            end
            c = c+1;
        end
    end

    % Copies D_MATRIX3 to D_MATRIX4 if all Sensors Eliminated based on Pressure Range
    if exist('D_MATRIX4') == 0
        D_MATRIX4 = D_MATRIX3;
    end

    % Create Mass / Volume Scaling Factor Optimization
    for mv = 1:size(D_MATRIX4,1)
        SCALE_FAC(mv) = D_MATRIX4{mv,10} * D_MATRIX4{mv,11} * D_MATRIX4{mv,12} * D_MATRIX4{mv,13};
        if (D_MATRIX4{mv,10} == 0) || (D_MATRIX4{mv,11} == 0) || (D_MATRIX4{mv,12} == 0) || (D_MATRIX4{mv,13} == 0)
            SCALE_FAC(mv) = 9.99E99;
        end
    end
end

```

```

[ MIN_SCL, INDEX ] = min(SCALE_FAC);
% Build Sensor Matrix Based on Scale Factors
k = 1;
for c = 1:length(SCALE_FAC)
    if SCALE_FAC(c) == MIN_SCL
        for j = 1:size(D_MATRIX4,2)
            D_MATRIX5{k,j} = D_MATRIX4{c,j};
        end
        k = k + 1;
    end
end

% Copies D_MATRIX4 to D_MATRIX5 if all Sensors Eliminated based on Pressure Range
if exist('D_MATRIX5') == 0
    D_MATRIX5 = D_MATRIX4;
end

if size(D_MATRIX5,1) > 1
    % Find the Min Accuracy Value
    for n = 1:size(D_MATRIX5,1)
        D_DELTA_RNG(n) = D_MATRIX5{n,4};
    end

    % Find the Minimum Refraction Operating Range of the Sensor
    MIN_D_DELTA_RNG = min(D_DELTA_RNG);

    % For Multiple Sensors with the Same Range Reduce Sensor Matrix to Those
    % with the Greatest Range
    k = 1;
    for c = 1:length(D_DELTA_RNG)
        if D_DELTA_RNG(c) == MIN_D_DELTA_RNG
            for j = 1:size(D_MATRIX5,2)
                D_MATRIX6{k,j} = D_MATRIX5{c,j};
            end
            k = k + 1;
        end
    end

% Copies D_MATRIX5 to D_MATRIX6 if all Sensors Eliminated
if exist('D_MATRIX6') == 0
    D_MATRIX6 = D_MATRIX5;
end

% If NO Viable Sensor Exists Load Zero Values into Data Matrix
if exist('D_MATRIX6') == 0
    D_MATRIX6{1,1} = 'N/A';
    for d = 2:size(DENSITYSENSOR_DB,2)-1

```





```

D9. GCMS Sensors
% function GCMS_SENSORS(GCMS_TYPE,GCMS_RANGE,UNITS,Print)
% Density Sensor Database
%
% Developed by: Keith Schreck
% Mechanical and Aerospace Engineering
% San Jose State University
% Date: Fall 2007
%
% This file contains Gas Chromatograph Mass Spectrometer Sensor (GCMS)
% values for the ISSPO program. Data here is loaded into the main program
% to determine the proper sensor to use for Missions based on the Planetary
% Atmosphere composition. Values are assigned to a variable name and saved
% to a Matlab '.mat' file and loaded when calling the ISSPO program.
%
% Some of the values are based on the Unit system chosen and are loaded
% in separate sections.
%
% References
% http://www.hitachi-hita.com/pageloader.aspx?type=product&id=390&orgid=45&gclid=CMCQxI2pzcACFRpdagodVmMrXg
% http://www.spectrometer.com/
% http://www.thinkers.com/products/RGA.htm
% http://www.netzsch-thermal-analysis.com/en/products/detail/pid,33.html
% http://www.restek.com/

function GCMS_SENSORS(GCMS_TYPE,GCMS_RANGE,UNITS,Print)

% Load Constants values into Local Program
load('Constants_DB.mat');
load('Planet_Data.mat');

if strcmpi(GCMS_TYPE,'WAVELENGTH')==1
%*****
% GCMS Sensor Data Base Created Based on un Unit System
% 2 - D Matrix GCMSSENSOR_DB(DS_Type, Property)

if strcmpi(UNITS,'SI')==1
% Properties [ SENSOR
WEIGHT % Units [ TYPE Wavelength Range Spectral Width Resolution Scan Time Detector Array
kg Len Wid Hgt POWER Low nm High Volts N/A N/A sec N/A # of Elements
GCMSSENSOR_DB = { 'U-0080D', 190 1100 'AC' [ 1 2 5 10 20 50 ] 0 'Photodiode Array' ;
10 420 294 120 'U-1900', 190 1100 'AC' [ 1 2 5 10 20 50 ] 0 ;
27 643 380 311 [ 100 120 220 230 240 ] 'AC' [ 0 ] 'Silicon photodiode' ;
31 605 500 283 [ 100 120 220 230 240 ] 'AC' [ 5 25 50 ] 0 ;
50 680 690 330 [ 100 120 220 230 240 ] 'AC' [ 0 ] 'Single Photodiode' ;
120 800 730 880 [ 100 120 220 230 240 ] 'AC' [ 0 ] 'Photomultiplier(UV-Vis region) Cooled
PBs(NIR region)';

```

```

35 630 500 270 100 120 220 230 240 ] 730 'AC' 20 [ 0 ] 0 10.20 1
'F-2500' 220 230 240 ] 'Fluorescence Spectrophotometer' ;
'F-7000' 200 750 'AC' 20 [ 6000 ] 0 0.55 1
'Z-2000' 190 900 'AC' 2.6 [ 100 ] 0 0 'Fluorescence Spectrophotometer' ;
'Z-2300' 190 900 'AC' 2.6 [ 100 ] 0 0 'Photomultiplier Flame/Graphite Furnace' ;
'Z-2700' 190 900 'AC' 2.6 [ 0 ] 0 'Photomultiplier Flame' ;
'Z-2700' 190 900 'AC' 2.6 [ 100 ] 0 0 'Photomultiplier Graphite Furnace' ;
'H10' 200 1600 'AC' 4 [ 100 ] 1 0 'Monochromator' ;
'CP140' 190 2500 'N/A' 0 [ 0 ] 0 2048 'Imaging Spectrograph' ;
'VS140' 190 1100 'N/A' 0 [ 0 ] 1.4 0.01 3864 'Imaging Spectrograph' ;
58 [ 7.5 ] 'DC' 0 [ 0 ] 1.4 0.01 3864 'Linear CCD 3864 pixels, Photo Diode Array
1024 pixels' ;
'MicroHR' 150 1500 'N/A' 0 [ 0 ] 0.3 3 3648 '2D CCD 3684 pixels, Photo Diode Array 1024
pixels' ;
'TRIAX 180' 185 4000 'N/A' 0 [ 0 ] 0.3 0 1 'Imaging Monochromator/Spectrograph' ;
'H-20' 100 3200 'N/A' 0.5 [ 0 ] 0.5 0 1 'High Luminescence Monochromator' ;
'HR320' 150 4000 'N/A' 0 [ 0 ] 0.06 24.06 1 'Imaging Spectrometer' ;
'HR550' 150 4000 'N/A' 0 [ 0 ] 0.025 24.06 1 'Imaging Spectrometer' ;
'FHR 640' 185 1500 'N/A' 0 [ 0 ] 0.016 5.0 1 'Imaging Spectrometer' ;
'750M' 185 4000 'N/A' 0 [ 0 ] 0.01 0 1 'Monochromator/Spectrograph' ;
838 311 330 [ 0 ] 'N/A' 'kg' 'mm' 'mm' 'V' 'N/A' 'microL'
GCMSUnitLabelStr = { 'N/A' 'nm' 'nm' 'nm' 'sec' 'N/A' 'kg' 'mm' 'mm' 'V' 'N/A' 'microL'
};
end

```

```

if strcmp(UNITS,'British')== 1
% Properties
% DIMENSIONS - in
% Units Len Wid Hgt
% GCMSSENSOR_DB =
22.05 16.54 11.57 4.72 [ 100 120 230 240 ] 'AC' 1.1811 10^-7 in 10^-9 in 0 0.05 1024
59.5 22.5 15 12.3 [ 100 120 220 230 240 ] 'AC' 1.5748 1.5748 [ 0 ] 0 'Photodiode Array' ;
68 24 20 11.5 [ 100 120 220 230 240 ] 'AC' 0.5906 0.5906 [ 5 25 50 ] 0 'Silicon photodiode' ;
110 26.8 27.2 13 [ 100 120 220 230 240 ] 'AC' 1.9685 1.9685 [ 0 ] 0 'Silicon photodiode' ;
'U-0080D' 74.80 433.07 'AC' [ 1 2 5 10 20 50 ] 0 'Photodiode Array' ;
'U-1900' 74.80 433.07 'AC' 1.5748 [ 0 ] 0 'Silicon photodiode' ;
'U-2900' 74.80 433.07 'AC' 0.5906 [ 5 25 50 ] 0 'Silicon photodiode' ;
'U-3010' 74.80 354.33 'AC' 1.9685 [ 0 ] 0 'Single Photomultiplier' ;
end

```

```

264      31.50  28.74      68.89      1023.62      3.1496      0      60.63      1
PBs (NIR region) , ;      'U-4100' [ 100 120 220 230 240 ] 'AC' 'Photomultiplier(UV-Vis region) Cooled
77      24.8   19.7      86.61      287.40      7.8740      0      10.20      1
      10.6 [ 100 120 220 230 240 ] 'AC' 'Fluorescence Spectrophotometer' ;
90.39   20.47  24.41      78.74      295.27      7.8740      0      0.55      1
      11.81 [ 100 120 220 230 240 ] 'AC' 'Fluorescence Spectrophotometer' ;
416.67  24.80  43.31      74.80      354.33      1.0236      0      0      1
      'Z-2000' [ 220 230 240 ] 'AC' 'Photomultiplier Flame/Graphite Furnace' ;
233.70  24.80  29.53      74.80      354.33      1.0236      0      0      1
      'Z-2300' [ 115 220 230 240 ] 'AC' 'Photomultiplier Flame' ;
306.44  24.80  29.53      74.80      354.33      1.0236      0      0      1
      'Z-2700' [ 220 230 240 ] 'AC' 'Photomultiplier Graphite Furnace' ;
2.20    5.04  4.49      78.74      629.92      1.5748      39.37      0      1
      'H10' [ 0 ] 'N/A' 'Monochromator' ;
0        0      0      74.80      984.25      0      0      0      2048
      'CP140' [ 0 ] 'N/A' 'Imaging Spectrograph' ;
1.5     7.0   4.73      74.80      433.07      0      55.12      0.01      3864
1024 pixels' ;      2.25 [ 7.5 ] 'DC' 'Linear CCD 3864 pixels, Photo Diode Array
11.0    7.0   10.5      59.05      590.55      0      11.81      3      3648
pixels' ;      5.5 [ 0 ] 'N/A' '2D CCD 3684 pixels, Photo Diode Array 1024
0        10.24  9.06      72.83      1574.80      0      11.81      0      1
      'TRIAx 180' [ 0 ] 'N/A' 'Imaging Monochromator/Spectrograph' ;
7.05    13.35  3.31      39.37      1258.94      0.1969      19.69      0      1
      'H-20' [ 0 ] 'N/A' 'High Luminosity Monochromator' ;
45      16.4   16.6      59.05      1574.80      0      2.362      24.06      1
      'iHR320' [ 0 ] 'N/A' 'Imaging Spectrometer' ;
62      25.51  18.09      59.05      1574.80      0      0.9843      24.06      1
      'iHR550' [ 0 ] 'N/A' 'Imaging Spectrometer' ;
143.3   29.13  13.78      72.83      590.55      0      0.6299      5.0      1
      'FHR 640' [ 0 ] 'N/A' 'Monochromator/Spectrograph' ;
0        32.99  12.24      72.83      1574.80      0      0.3937      0      1
      '750M' [ 0 ] 'N/A' 'High resolution spectrometer' };
'V' 'N/A' 'microL' 'N/A' };
end

% Comments on Features / Use
GCMSSENSOR_DB{1,16} = { 'DNA/RNA measurement'
concentration (Nucleotide), Background correction (no correction, fixed wavelength, Tangent), Purity, Concentration, Molar
dilution magnification' ;
Other quantitative analysis with reagent' ;
'Calculation of protein concentration'
'Labeled DNA'
before DNA sequencing' ;
detection, Recalculation, Printing' ;
'Wavelength scan'
'Time scan'
scale, Recalculation, Printing' ;
'Concentration calculation by Warburg-Christian method,
'Verification that a fluorescent dye coupled with a primer
'Baseline correction, Overlay of 20 channel data, Peak
'Baseline correction, Overlay of 20 channel data, Auto

```

```

One, two and three wavelength calculations, Saving and printing of calibration curve and measurement data' ;
'Other analysis'
sample name, comment and file name, constant display of measurement parameters, Data saving and retrieval' };
GCMSSENSOR_DB{2,16} = { 'Modes of operation' 'Scanning, Time (Kinetics), Photometry and Ratio (DNA/RNA Analysis)' };
GCMSSENSOR_DB{3,16} = { 'Resolution' 'Satisfies European Pharmacopoeia Bandpass' };
GCMSSENSOR_DB{4,16} = { 'Band pass'
'Scan speed'
'Automated analysis, specular and diffuse reflectance, turbid
samples, thin films, long path cells, and continuous flow-through monitoring' ;
'Hitachi's U-3010 Model'
'Hitachi's U-3310 Model'
'NIR Region: Switching in 0.2 nm steps from 0.24 to 8.0 nm '
'prism/grating or grating/grating' };
'Variable: 2.5, 5, 10 and 20 nm Excitation and Emission
monochromators' ;
'15, 60, 300, 1,500 and 3,000 nm/min';
'photometry, wavelength scanning and time scanning' };
'Excitation side: 1, 2.5, 5, 10, 20 nm Emission side: 1,
2.5, 5, 10, 20 nm ' ;
'Scan speed '
'Response'
'Applications'
'Optics '
'Slit width'
'Sample Volume'
'Sample Number'
GCMSSENSOR_DB{8,16} = { 'Measurement Mode'
'Optics '
'Slit width'
'Sample Volume'
'Sample Number'
'UV-Range 200 - 750 nm, Vis-Range 350 - 890 nm, IR-Range
400 - 1600' ;
'Optical Configuration'
'Slit width'
'Application'
'Models'
'Detector Array'
pixels' ;
'Optical Configuration'
'Aberration-corrected concave holographic diffraction grating'
'Fixed width, interchangeable. 3 pairs supplied: 0.5, 1 and 2 mm'
'Low resolution absorption transmission and fluorescence' };
'Multiple Models ranges from 190 to 2500 nm' ;
'CCD - Linear 2D array <2048 pixels, FDA - NMOS & CMOS 128-1024
'Aberration-corrected concave holographic diffraction grating'

```





```

% Define Sensing Operational Wavelengths based on Atmospheric Composition
% Spectral Data From: http://laserstars.org/data/elements/
% NIST ASD: http://physics.nist.gov/PhysRefData/ASD/lines\_form.html
k = 1;
for L = 1:2:length(PlanAtm_Prop)
    switch PlanAtm_Prop{L}
    case 'H'
        if strcmpi(UNITS,'SI') == 1;
            Lambdas(k) = 91.13;
            Lambdas(k+1) = 364.51;
            Lambdas(k+2) = 820.14;
            Lambdas(k+3) = 1458.17;
            Lambdas(k+4) = 2278.17;
            Lambdas(k+5) = 3280.56;
            k = k+6;
        end
        if strcmpi(UNITS,'British') == 1;
            Lambdas(k) = 35.877; % 10^-7 inches
            Lambdas(k+1) = 143.507; % 10^-7 inches
            Lambdas(k+2) = 322.889; % 10^-7 inches
            Lambdas(k+3) = 574.083; % 10^-7 inches
            Lambdas(k+4) = 896.917; % 10^-7 inches
            Lambdas(k+5) = 1291.56; % 10^-7 inches
            k = k+6;
        end
    case 'H2'
        if strcmpi(UNITS,'SI') == 1;
            Lambdas(k) = 397.007;
            Lambdas(k+1) = 410.174;
            Lambdas(k+2) = 434.047;
            Lambdas(k+3) = 486.133;
            Lambdas(k+4) = 656.285;
            k = k+5;
        end
        if strcmpi(UNITS,'British') == 1;
            Lambdas(k) = 156.302; % 10^-7 inches
            Lambdas(k+1) = 161.486; % 10^-7 inches
            Lambdas(k+2) = 170.885; % 10^-7 inches
            Lambdas(k+3) = 191.391; % 10^-7 inches
            Lambdas(k+4) = 258.380; % 10^-7 inches
            k = k+5;
        end
    case 'CO2'
        if strcmpi(UNITS,'SI') == 1;
            Lambdas(k) = 2600.00;
            Lambdas(k+1) = 4000.00;
            Lambdas(k+2) = 13000.00;
            k = k+3;
        end
        if strcmpi(UNITS,'British') == 1;
            Lambdas(k) = 1023.62; % 10^-7 inches
            Lambdas(k+1) = 1574.80; % 10^-7 inches
            Lambdas(k+2) = 5118.09; % 10^-7 inches
            k = k+3;
        end
    end
end

```

```

end
case 'N2'
if strcmpi(UNITRS,'SI')==1;
Lambdas(k) = 399.500;
Lambdas(k+1) = 404.131;
Lambdas(k+2) = 444.703;
Lambdas(k+3) = 463.054;
Lambdas(k+4) = 500.148;
Lambdas(k+5) = 500.515;
Lambdas(k+6) = 566.663;
Lambdas(k+7) = 567.956;
Lambdas(k+8) = 575.250;
Lambdas(k+9) = 594.165;
Lambdas(k+10) = 648.205;
k = k+1;
end
if strcmpi(UNITRS,'British')==1;
Lambdas(k) = 157.283;
Lambdas(k+1) = 159.107;
Lambdas(k+2) = 175.078;
Lambdas(k+3) = 182.305;
Lambdas(k+4) = 196.909;
Lambdas(k+5) = 197.053;
Lambdas(k+6) = 223.096;
Lambdas(k+7) = 223.605;
Lambdas(k+8) = 226.476;
Lambdas(k+9) = 233.923;
Lambdas(k+10) = 255.199;
k = k+1;
end
case '02'
if strcmpi(UNITRS,'SI')==1;
Lambdas(k) = 391.196;
Lambdas(k+1) = 397.326;
Lambdas(k+2) = 407.587;
Lambdas(k+3) = 418.979;
Lambdas(k+4) = 441.491;
Lambdas(k+5) = 464.914;
Lambdas(k+6) = 615.818;
Lambdas(k+7) = 645.598;
Lambdas(k+8) = 700.223;
k = k+9;
end
if strcmpi(UNITRS,'British')==1;
Lambdas(k) = 154.014;
Lambdas(k+1) = 156.428;
Lambdas(k+2) = 160.467;
Lambdas(k+3) = 164.952;
Lambdas(k+4) = 173.815;
Lambdas(k+5) = 183.037;
Lambdas(k+6) = 242.448;
Lambdas(k+7) = 254.172;
Lambdas(k+8) = 275.678;
k = k+9;
end

```

```

end
case 'CH4'
if strcmpi(UNITS,'SI') == 1;
Lambdas(k) = 350.000;
Lambdas(k+1) = 800.000;
k = k+2;
end
if strcmpi(UNITS,'British') == 1;
Lambdas(k) = 137.795; % 10^-7 inches
Lambdas(k+1) = 314.960; % 10^-7 inches
k = k+2;
end
case 'He'
if strcmpi(UNITS,'SI') == 1;
Lambdas(k) = 396.474;
Lambdas(k+1) = 447.149;
Lambdas(k+2) = 501.569;
Lambdas(k+3) = 587.562;
Lambdas(k+4) = 667.814;
Lambdas(k+5) = 706.518;
k = k+6;
end
if strcmpi(UNITS,'British') == 1;
Lambdas(k) = 156.092; % 10^-7 inches
Lambdas(k+1) = 176.043; % 10^-7 inches
Lambdas(k+2) = 197.468; % 10^-7 inches
Lambdas(k+3) = 231.324; % 10^-7 inches
Lambdas(k+4) = 262.919; % 10^-7 inches
Lambdas(k+5) = 278.157; % 10^-7 inches
k = k+6;
end
case 'Na'
if strcmpi(UNITS,'SI') == 1;
Lambdas(k) = 371.107;
Lambdas(k+1) = 411.370;
Lambdas(k+2) = 430.881;
Lambdas(k+3) = 439.281;
Lambdas(k+4) = 444.670;
Lambdas(k+5) = 453.332;
Lambdas(k+6) = 455.153;
Lambdas(k+7) = 519.165;
Lambdas(k+8) = 654.575;
Lambdas(k+9) = 780.978;
k = k+10;
end
if strcmpi(UNITS,'British') == 1;
Lambdas(k) = 156.092; % 10^-7 inches
Lambdas(k+1) = 161.957; % 10^-7 inches
Lambdas(k+2) = 169.638; % 10^-7 inches
Lambdas(k+3) = 172.945; % 10^-7 inches
Lambdas(k+4) = 175.067; % 10^-7 inches
Lambdas(k+5) = 178.477; % 10^-7 inches
Lambdas(k+6) = 179.194; % 10^-7 inches
Lambdas(k+7) = 204.396; % 10^-7 inches

```

```

    Lambdas(k+8) = 257.707; % 10^-7 inches
    Lambdas(k+9) = 307.472; % 10^-7 inches
    k = k+10;
end
case 'H2O'
    if strcmpi(UNITS,'SI') == 1;
        Lambdas(k) = 730.00;
        Lambdas(k+1) = 820.00;
        Lambdas(k+2) = 930.00;
        Lambdas(k+3) = 970.00;
        Lambdas(k+4) = 1200.00;
        Lambdas(k+5) = 1450.00;
        Lambdas(k+6) = 1950.00;
        Lambdas(k+7) = 2500.00;
        k = k+8;
    end
    if strcmpi(UNITS,'British') == 1;
        Lambdas(k) = 287.402; % 10^-7 inches
        Lambdas(k+1) = 322.835; % 10^-7 inches
        Lambdas(k+2) = 366.142; % 10^-7 inches
        Lambdas(k+3) = 381.890; % 10^-7 inches
        Lambdas(k+4) = 472.441; % 10^-7 inches
        Lambdas(k+5) = 570.866; % 10^-7 inches
        Lambdas(k+6) = 767.717; % 10^-7 inches
        Lambdas(k+7) = 984.252; % 10^-7 inches
        k = k+8;
    end
end
case 'Air'
    if strcmpi(UNITS,'SI') == 1;
        Lambdas(k) = 392.572;
        Lambdas(k+1) = 407.201;
        Lambdas(k+2) = 413.172;
        Lambdas(k+3) = 415.859;
        Lambdas(k+4) = 420.067;
        Lambdas(k+5) = 427.753;
        Lambdas(k+6) = 434.806;
        Lambdas(k+7) = 442.600;
        Lambdas(k+8) = 454.505;
        Lambdas(k+9) = 460.957;
        Lambdas(k+10) = 472.687;
        Lambdas(k+11) = 487.986;
        Lambdas(k+12) = 696.543;
        Lambdas(k+13) = 706.722;
        Lambdas(k+14) = 714.704;
        k = k+15;
    end
    if strcmpi(UNITS,'British') == 1;
        Lambdas(k) = 154.556; % 10^-7 inches
        Lambdas(k+1) = 160.315; % 10^-7 inches
        Lambdas(k+2) = 162.666; % 10^-7 inches
        Lambdas(k+3) = 163.724; % 10^-7 inches
        Lambdas(k+4) = 165.381; % 10^-7 inches
        Lambdas(k+5) = 168.407; % 10^-7 inches
        Lambdas(k+6) = 171.183; % 10^-7 inches
    end
end

```

```

Lambda(k+7) = 174.252; % 10^-7 inches
Lambda(k+8) = 178.939; % 10^-7 inches
Lambda(k+9) = 181.479; % 10^-7 inches
Lambda(k+10) = 186.097; % 10^-7 inches
Lambda(k+11) = 192.120; % 10^-7 inches
Lambda(k+12) = 274.230; % 10^-7 inches
Lambda(k+13) = 278.237; % 10^-7 inches
Lambda(k+14) = 281.380; % 10^-7 inches
k = k+15;
end
case 'K'
if strcmpi(UNITS,'SI')== 1;
Lambda(k) = 360.888;
Lambda(k+1) = 404.414;
Lambda(k+2) = 464.237;
Lambda(k+3) = 496.503;
Lambda(k+4) = 533.969;
Lambda(k+5) = 535.957;
Lambda(k+6) = 580.175;
Lambda(k+7) = 583.189;
Lambda(k+8) = 691.108;
Lambda(k+9) = 693.877;
Lambda(k+10) = 696.467;
k = k+11;
end
if strcmpi(UNITS,'British')== 1;
Lambda(k) = 142.082; % 10^-7 inches
Lambda(k+1) = 159.218; % 10^-7 inches
Lambda(k+2) = 182.770; % 10^-7 inches
Lambda(k+3) = 195.474; % 10^-7 inches
Lambda(k+4) = 210.224; % 10^-7 inches
Lambda(k+5) = 211.007; % 10^-7 inches
Lambda(k+6) = 228.415; % 10^-7 inches
Lambda(k+7) = 229.602; % 10^-7 inches
Lambda(k+8) = 272.090; % 10^-7 inches
Lambda(k+9) = 273.180; % 10^-7 inches
Lambda(k+10) = 274.200; % 10^-7 inches
k = k+11;
end
case 'CO'
if strcmpi(UNITS,'SI')== 1;
Lambda(k) = 200.079;
Lambda(k+1) = 892.624;
k = k+2;
end
if strcmpi(UNITS,'British')== 1;
Lambda(k) = 78.771; % 10^-7 inches
Lambda(k+1) = 351.427; % 10^-7 inches
k = k+2;
end
otherwise
end
end
end

```

```

% Based on Atmospheric Contents Determine Range Of Wavelengths of Atmosphere
Min_Lambda = min(Lambdas);
Max_Lambda = max(Lambdas);

% Build Up Sensor Database of Sensors whose Range Exceeds Atmospheric
% Components Range
c = 1;
for g = 1:size(GCMSSENSOR_DB,1)
    if (Min_Lambda >= GCMSSENSOR_DB{g,2}) && (Max_Lambda <= GCMSSENSOR_DB{g,3})
        for a = 1:size(GCMSSENSOR_DB,2)
            GCMS_MATRIX{c,a} = GCMSSENSOR_DB{g,a};
        end
        c = c+1;
    end
end

% Down Select a Single Sensor If Multiple Options Exist in the Database
if size(GCMS_MATRIX,1) >1
    % Create Mass / Volume Scaling Factor Optimization
    for mv = 1:size(GCMS_MATRIX,1)
        GCMS_SCALE_FAC(mv) = GCMS_MATRIX{mv,8} * GCMS_MATRIX{mv,9} * GCMS_MATRIX{mv,10} * GCMS_MATRIX{mv,11};
        if (GCMS_MATRIX{mv,8} == 0) || (GCMS_MATRIX{mv,9} == 0) || (GCMS_MATRIX{mv,10} == 0) || (GCMS_MATRIX{mv,11} == 0)
            GCMS_SCALE_FAC(mv) = 9.99E99;
        end
    end

[ GCMS_MIN_SCL, INDEX ] = min(GCMS_SCALE_FAC);

% Build Sensor Matrix Based on Scale Factors
k = 1;
for c = 1:length(GCMS_SCALE_FAC)
    if GCMS_SCALE_FAC(c) == GCMS_MIN_SCL
        for j = 1:size(GCMS_MATRIX,2)
            GCMS_MATRIX2{k,j} = GCMS_MATRIX{c,j};
        end
        k = k + 1;
    end
end

% Copies GCMS_MATRIX to GCMS_MATRIX2 if all sensors Eliminated based on Mass Volume Scale Factor
if exist('GCMS_MATRIX2') == 0
    GCMS_MATRIX2 = GCMS_MATRIX;
end

% Down Select Sensor Based On Smallest Resolution Range
if size(GCMS_MATRIX2,1) >1
    for r = 1:size(GCMS_MATRIX2,1)
        GCMS_RES(r) = GCMS_MATRIX2{r,5};
        if GCMS_RES(r) == 0
            GCMS_RES(r) = 9.99E99;
        end
    end
end

```





```

fprintf('\n Resolution: %s', GCMS_MATRIX3{1,5}, GCMSUnitLabelStr{5} );
fprintf('\n Scan Time: %10.2f %s', GCMS_MATRIX3{1,6}, GCMSUnitLabelStr{6} );
fprintf('\n Sample Volume: ');
for s = 1:length(GCMS_MATRIX3{1,14})
    fprintf(' %3d', GCMS_MATRIX3{1,14}(1,s) );
end
fprintf(' %s', GCMSUnitLabelStr{14} );
fprintf('\n Detector Array # of Elements: %10d %s', GCMS_MATRIX3{1,7}, GCMSUnitLabelStr{7} );
fprintf('\n Detector Type: %s', GCMS_MATRIX3{1,15} );

fprintf('\n\n Power Requirements');
fprintf('\n Voltage Source: ');
for v = 1:length(GCMS_MATRIX3{1,12})
    fprintf(' %4d', GCMS_MATRIX3{1,12}(1,v) );
end
fprintf(' %s', GCMSUnitLabelStr{12} );
fprintf('\n Voltage Type: %s %s', GCMS_MATRIX3{1,13}, GCMSUnitLabelStr{13} );

fprintf('\n\n Physical Properties');
fprintf('\n Sensor Mass: %10.2f %s', GCMS_MATRIX3{1,8}, GCMSUnitLabelStr{8} );
fprintf('\n Sensor Dimensions - Length: %10.2f %s', GCMS_MATRIX3{1,9}, GCMSUnitLabelStr{9} );
fprintf('\n Width: %10.2f %s', GCMS_MATRIX3{1,10}, GCMSUnitLabelStr{10} );
fprintf('\n Height: %10.2f %s', GCMS_MATRIX3{1,11}, GCMSUnitLabelStr{11} );

fprintf('\n\n Sensor Comments:');
for n = 1:length(GCMS_MATRIX3{1,16})
    fprintf('\n %40s %s', GCMS_MATRIX3{1,16}{n,1}, GCMS_MATRIX3{1,16}{n,2} );
end
end

%*****
%*** Writes a text file summary of the Planetary Properties

Fout = fopen('GCMS_SENSOR_Summary.txt','w+');
fprintf(Fout, '\n\n *** G C M S S E N S O R S U M M A R Y R E S U L T S ***');
fprintf(Fout, '\n\n*** I N P U T S ***');
fprintf(Fout, '\n Unit System: %s', UNITS);

fprintf(Fout, '\n\n*****\n\n*****\n\n*****\n\n');

% GCMS Sensor Properties
fprintf(Fout, '\n\n** Sensor Properties **');
fprintf(Fout, '\n GCMS Sensor Type: %s', GCMS_MATRIX3{1,1});
fprintf(Fout, '\n Sensing Wavelength Range - Low: %10.2f %s', GCMS_MATRIX3{1,2}, GCMSUnitLabelStr{2} );
fprintf(Fout, '\n High: %10.2f %s', GCMS_MATRIX3{1,3}, GCMSUnitLabelStr{3} );
fprintf(Fout, '\n Spectral Width: %10.4f %s', GCMS_MATRIX3{1,4}, GCMSUnitLabelStr{4} );
fprintf(Fout, '\n Resolution: %10.4f %s', GCMS_MATRIX3{1,5}, GCMSUnitLabelStr{5} );
fprintf(Fout, '\n Scan Time: %10.2f %s', GCMS_MATRIX3{1,6}, GCMSUnitLabelStr{6} );

```

```

fprintf(Fout, '\n Sample Volume:
');
for s = 1:length(GCMS_MATRIX3{1,14})
    fprintf(Fout, '%3d', GCMS_MATRIX3{1,14}{1,s} );
end
fprintf(Fout, ' %s', GCMSUnitLabelStr{14} );
fprintf(Fout, '\n Detector Array # of Elements: %10d
');
fprintf(Fout, '\n Detector Type: %s', GCMS_MATRIX3{1,15} );

fprintf(Fout, '\n\n Power Requirements');
fprintf(Fout, '\n Voltage Source:
');
for v = 1:length(GCMS_MATRIX3{1,12})
    fprintf(Fout, '%4d', GCMS_MATRIX3{1,12}{1,v} );
end
fprintf(Fout, ' %s', GCMSUnitLabelStr{12} );
fprintf(Fout, '\n Voltage Type: %s %s', GCMS_MATRIX3{1,13}, GCMSUnitLabelStr{13} );

fprintf(Fout, '\n\n Physical Properties');
fprintf(Fout, '\n Sensor Mass: %10.2f %s', GCMS_MATRIX3{1,8}, GCMSUnitLabelStr{8} );
fprintf(Fout, '\n Sensor Dimensions - Length: %10.2f %s', GCMS_MATRIX3{1,9}, GCMSUnitLabelStr{9} );
fprintf(Fout, '\n Width: %10.2f %s', GCMS_MATRIX3{1,10}, GCMSUnitLabelStr{10} );
fprintf(Fout, '\n Height: %10.2f %s', GCMS_MATRIX3{1,11}, GCMSUnitLabelStr{11} );

fprintf(Fout, '\n\n Sensor Comments:');
for n = 1:length(GCMS_MATRIX3{1,16})
    fprintf(Fout, '\n %40s %s', GCMS_MATRIX3{1,16}{n,1}, GCMS_MATRIX3{1,16}{n,2} );
end
fclose(Fout);

end

if strcmpi(GCMS_TYPE, 'MASS-CHARGE') == 1

% DEFINE MASS-CHARGE VARIABLE LIMITS INTERNAL
if strcmpi(GCMS_RANGE, 'HIGH') == 1
    GCMS_RANGE_NUM = 300;
    G_TOL_LOWER = 99;
end
if strcmpi(GCMS_RANGE, 'MEDIUM') == 1
    GCMS_RANGE_NUM = 200;
    G_TOL_LOWER = 99;
end
if strcmpi(GCMS_RANGE, 'LOW') == 1
    GCMS_RANGE_NUM = 100;
    G_TOL_LOWER = 99;
end

% GCMS Sensor Data Base Created Based on Unit System
%*****
%***** GCMS Sensor Data Base Created Based on Unit System
%*****

```

```

%% 2 - D Matrix GCMSSENSOR_DB(DS_Type, Property)
if strcmp(UNITS,'SI')==1
% Properties
% Properties { SENSOR MASS RANGE MASS FILTER DETECTOR DYNAMIC
SCAN RANGE RESOLUTION SENSITIVITY MIN PARTIAL PRESSURE OPERATING PRESSURE MAX OPERATING TEMP
BAKEOUT TEMP }
Units [ TYPE Low amu High Torr TYPE TYPE Orders
of Magnitude ] amu
deg C GCMSSENSOR_DB1 = { 'RGA100' 1 100 'Quadrupole' { 'Faraday Cup' 'Electron Multiplier' }
[ 2.E-4 200 ] [ 5.0E-11 5.0E-14 ] [ 1.0E-4 1.0E-6 ] 70.00
;
'RGA200' 1 200 'Quadrupole' { 'Faraday Cup' 'Electron Multiplier' }
[ 2.E-4 200 ] [ 5.0E-11 5.0E-14 ] [ 1.0E-4 1.0E-6 ] 70.00
;
'RGA300' 1 300 'Quadrupole' { 'Faraday Cup' 'Electron Multiplier' }
[ 2.E-4 200 ] [ 5.0E-11 5.0E-14 ] [ 1.0E-4 1.0E-6 ] 70.00
;
'Aeolos' 1 300 'Quadrupole' { 'Electron Multiplier' }
[ 0 0 ] [ 1.50E-14 ] [ 0 0 ] 300.00
;
'GCMS-Huygens' 2 141 'Quadrupole' { 'Electron Multiplier' }
[ 1E-6 0 ] [ 1.275E-16 ] [ 0 0 ] 100.00
};

GCMSSensorLabelStr1 = { 'N/A' 'amu' 'N/A' 'N/A' 'amu' 'amu' 'amu' 'amu' 'amu' 'amu' 'amu' 'amu' 'amu' 'amu' 'amu' 'amu'
};

% Properties [ ION SOURCE MATERIAL FILAMENT FIELD RANGE ELECTRON
% ION ENERGY FOCUS VOLTAGE ELECTRON CURRENT Type Type Type Type Type Type Type Type Type Type Type Type Type Type Type
Units # Min mA Max ]
GCMSSENSOR_DB2 = {
[ 8 12 ] 0 150 0.0 1 'SS304' ; '2-Thoriated-Iridium' 1 10 25 105
[ 8 12 ] 0 150 0.0 1 'SS304' ; '2-Thoriated-Iridium' 1 10 25 105
[ 8 12 ] 0 150 0.0 1 'SS304' ; '2-Thoriated-Iridium' 1 10 25 105
[ 0 ] 0 0 0.0 1 'N/A' ; '2-Iridium Cathodes' 0 0 70 70
[ 8 12 ] 0 0 0.0 5 'N/A' ; 'N/A' 1 16 25 70
};

GCMSSensorLabelStr2 = { 'N/A' 'N/A' 'N/A' 'N/A' 'W Degas' 'W Degas' 'W Degas' 'V' 'V' 'V' 'V' 'V' 'V' 'V' 'V' 'V' 'V' 'V' 'V' 'V'
};

% Properties [ DIMENSIONS - cm
% POWER-TYP POWER-AVE POWER-MAX ]
DATA RATE Units Length Width Height
bits/sec W W W
GCMSSENSOR_DB3 = {
28800 0 0 0 46.61 10.42 7.88
};

```

```

28800      0      0      46.61 10.42 7.88 0.1 30 24 'DC' 2.5 2.72
28800      0      0      46.61 10.42 7.88 0.1 30 24 'DC' 2.5 2.72
0          0      0      0 0 0 0.0 00 0 'N/A' 0 0
960      28.0 41.0      47.00 19.80 19.80 0 36 28 'DC' 1.68 17.30
      71.0

```

```

end
GCMSUnitLabelStr3 = { 'cm' 'cm' 'amu' 'min' 'V' 'N/A' 'A' 'kg' 'bits/sec' 'W' 'W' 'W' };

```

```

if strcmp(UNITS,'British')==1
% Properties
SCAN_RANGE RESOLUTION SENSITIVITY MASS RANGE MASS FILTER DETECTOR DYNAMIC
BAKEOUT TEMP Units [ TYPE Low amu High psi TYPE psi TYPE deg F Orders
of Magnitude ] amu
deg F
GCMSSENSOR_DB1 = { 'RGA100' 1 [ 9.67E-13 9.67E-16 ] { 'Paraday Cup' 'Electron Multiplier' }
0.5 [ 3.87E-6 3.87 ] [ 100 'Quadrupole' [ 1.93E-6 1.93E-8 ] 158.00 }
'RG200' 1 [ 9.67E-13 9.67E-16 ] { 'Paraday Cup' 'Electron Multiplier' }
0.5 [ 3.87E-6 3.87 ] [ 200 'Quadrupole' [ 1.93E-6 1.93E-8 ] 158.00 }
'RG300' 1 [ 9.67E-13 9.67E-16 ] { 'Paraday Cup' 'Electron Multiplier' }
0.5 [ 3.87E-6 3.87 ] [ 300 'Quadrupole' [ 1.93E-6 1.93E-8 ] 158.00 }
'Aeolos' 1 [ 300 'Quadrupole' [ 1.93E-6 1.93E-8 ] 158.00 }
0.5 [ 0 ] [ 2.90E-16 ] [ 'Electron Multiplier' ] 572.00
'GCMS-Huygens' 2 [ 141 'Quadrupole' [ 1.450E-1 ] 212.00 }
1E-6 [ 0 ] [ 2.465E-18 ] [ 'Electron Multiplier' ] 572.00
}
}

```

```

GCMSUnitLabelStr1 = { 'N/A' 'amu' 'amu' 'N/A' 'N/A' 'N/A' 'amu' 'psi' 'psi' 'deg F'
};
% Properties
ION ENERGY [ ION SOURCE MATERIAL FILAMENT FIELD RANGE ELECTRON Min Volts Max
Units [ FOCUS VOLTAGE ELECTRON CURRENT Type Type Low W Degas High Min Volts Max
GCMSSENSOR_DB2 = {
0 150 1 0.0 0.0 'SS304' '2-Thoriated-Iridium' 1 10 25 105
[ 8 12 ] 0 150 1 0.0 3.5 'SS304' '2-Thoriated-Iridium' 1 10 25 105
[ 8 12 ] 0 150 1 0.0 3.5 'SS304' '2-Thoriated-Iridium' 1 10 25 105
[ 0 ] 0 0 0.0 0 'N/A' '2-Iridium Cathodes' 0 0 70 70
[ 8 12 ] 0 0 0.0 0.08 'N/A' 'N/A' 1 16 25 70
}
}

```

```

GCMUnitLabelStr2 = { 'N/A' 'N/A' 'N/A' 'W Degas' 'W Degas' 'V' 'V' 'V' 'V' 'V' 'mA' 'mA' 'mA' };

% Properties
% POWER-TYP POWER-AVE DIMENSIONS - in
% Units [ POWER-MAX ]
% BTUs/hr [ Length Width Height ]
GCMSENSOR_DB3 = {
    0 0 0 18.35 4.10 3.10 0.1 30 24 'DC' 2.5 6.00
    0 0 0 18.35 4.10 3.10 0.1 30 24 'DC' 2.5 6.00
    0 0 0 18.35 4.10 3.10 0.1 30 24 'DC' 2.5 6.00
    0 0 0 0 0 0 0.0 00 0 'N/A' 0 0
    95.63 140.02 18.50 7.80 7.80 0 36 28 'DC' 1.68 38.14
};

GCMUnitLabelStr3 = { 'in' 'in' 'in' 'amu' 'min' 'V' 'N/A' 'A' 'lbm' 'bits/sec' 'BTUs/hr' 'BTUs/hr' };
end

% Comments on Features /
GCMSENSOR_DB4{1,1} = {
    'Gas Chromatograph Residual Gas Analyzer' ;
    'Probe design consists of an ionizer, quadrupole mass filter, and detector' ;
    'Built-in degassing feature' ;
    'Faraday Cup or Electron Multiplier option' ;
    'RS-232c Computer Interface' ;
    'Gas Chromatograph Residual Gas Analyzer' ;
    'Probe design consists of an ionizer, quadrupole mass filter, and detector' ;
    'Built-in degassing feature' ;
    'Faraday Cup or Electron Multiplier option' ;
    'RS-232c Computer Interface' ;
    'Gas Chromatograph Residual Gas Analyzer' ;
    'Probe design consists of an ionizer, quadrupole mass filter, and detector' ;
    'Built-in degassing feature' ;
    'Faraday Cup or Electron Multiplier option' ;
    'RS-232c Computer Interface' ;
    'Thermogravimetry and Mass Spectrometry' ;
    'Multiple Components Coupled Together' ;
    'Loss-free gas transfer to the OMS' ;
    'Faraday Cup or Electron Multiplier option' ;
    'RS-232c Computer Interface' ;
    'Combined Gas Chromatograph Mass Spectrometer' ;
    'Probe design consists of an 5 ion sources, quadrupole mass filter, and 2 Electron Multipliers' ;
};

% Leak Management Holes' ;
% Hydrogen gas used as Transport medium in sampling chambers' ;
% Huygens Probe GCMS System - Custom Flight Proven configuration' };

% Combine Data Arrays into a Single Data Matrix

```

```

GCMSSENSOR_DB = cat(2,GCMSSENSOR_DB1,GCMSSENSOR_DB2,GCMSSENSOR_DB3,GCMSSENSOR_DB4);
GCMSUnitLabelStr = cat(2,GCMSUnitLabelStr1,GCMSUnitLabelStr2,GCMSUnitLabelStr3);

% Build Up Sensor Database of Sensors whose Range Meets Input Operational Range
c = 1;
for g = 1:size(GCMSSENSOR_DB,1)
    if (GCMSSENSOR_DB{g,3} <= GCMS_RANGE_NUM) && (GCMSSENSOR_DB{g,3} > (GCMS_RANGE_NUM-G_TOL_LOWER))
        for a = 1:size(GCMSSENSOR_DB,2)
            GCMS_MATRIX{c,a} = GCMSSENSOR_DB{g,a};
        end
        c = c+1;
    end
end

% Down Select from database if Multiple Sensors Exist
if size(GCMS_MATRIX,1) > 1

    % Down Select Sensor by Max Dynamic Range (Orders Of Magnitude)
    for g = 1:size(GCMS_MATRIX,1)
        GCMS_DynRange(g) = GCMS_MATRIX{g,6};
    end

    % Find the Max Dynamic Range Sensors
    [ MAX_GCMS_RANGE, INDEX ] = max(GCMS_DynRange);

    C = 1;
    for g = 1:size(GCMS_MATRIX,1)
        if GCMS_MATRIX{g,6} == MAX_GCMS_RANGE
            for a = 1:size(GCMS_MATRIX,2)
                GCMS_MATRIX2{c,a} = GCMS_MATRIX{g,a};
            end
            C = C+1;
        end
    end

    if exist('GCMS_MATRIX2') == 0
        GCMS_MATRIX2 = GCMS_MATRIX;
    end

    if size(GCMS_MATRIX2,1) > 1
        % Down Select a Single Sensor If Multiple Options Exist in the Database
        % Create Mass / Volume Scaling Factor Optimization
        for mv = 1:size(GCMS_MATRIX2,1)
            GCMS_SCALE_FAC(mv) = GCMS_MATRIX2{mv,25} * GCMS_MATRIX2{mv,26} * GCMS_MATRIX2{mv,27} * GCMS_MATRIX2{mv,33};
            if (GCMS_MATRIX{mv,25} == 0) || (GCMS_MATRIX{mv,26} == 0) || (GCMS_MATRIX{mv,27} == 0) || (GCMS_MATRIX{mv,33}
                == 0)
                GCMS_SCALE_FAC(mv) = 9.99E99;
            end
        end
    end
    [ GCMS_MIN_SCL, INDEX ] = min(GCMS_SCALE_FAC);

```



```

end
fprintf('\n\n Dynamic Mass-Charge Scanning Range:
fprintf('\n\n Scanning Resolution:
fprintf('\n\n Scanning Sensitivity:
for r = 1:length(GCMS_MATRIX3{1,8})
    fprintf('%5.2f', GCMS_MATRIX3{1,8}(r) );
end
fprintf(' %s', GCMSUnitLabelStr{8} );
fprintf('\n\n Minimum Detected Partial Pressure:
for r = 1:length(GCMS_MATRIX3{1,9})
    fprintf('%5.2f', GCMS_MATRIX3{1,9}(r) );
end
fprintf(' %s', GCMSUnitLabelStr{9} );
fprintf('\n\n Operating Pressure:
for r = 1:length(GCMS_MATRIX3{1,10})
    fprintf('%5.2f', GCMS_MATRIX3{1,10}(r) );
end
fprintf(' %s', GCMSUnitLabelStr{10} );
fprintf('\n\n Maximum Operating Temperature:
fprintf('\n\n Bakeout Temperature:
fprintf('\n\n Number of Ion Sources:
fprintf('\n\n Construction Material:
fprintf('\n\n Filament Material:
fprintf('\n\n Charge Field Range - Low:
fprintf('\n\n High:
fprintf('\n\n Electron Source Voltage - Low:
fprintf('\n\n High:
fprintf('\n\n Ion Energy:
for r = 1:length(GCMS_MATRIX3{1,20})
    fprintf('%5.2f', GCMS_MATRIX3{1,20}(r) );
end
fprintf(' %s', GCMSUnitLabelStr{20} );
fprintf('\n\n Focus Voltage - Low:
fprintf('\n\n High:
fprintf('\n\n Electron Source Current - Low:
fprintf('\n\n High:
fprintf('\n\n Warm Up Period Mass Stability:
fprintf('\n\n Time:
fprintf('\n\n Sensor Return Data Rate:
fprintf(' %s', GCMS_MATRIX3{1,21}, GCMSUnitLabelStr{21} );
fprintf(' %s', GCMS_MATRIX3{1,22}, GCMSUnitLabelStr{22} );
fprintf(' %s', GCMS_MATRIX3{1,23}, GCMSUnitLabelStr{23} );
fprintf(' %s', GCMS_MATRIX3{1,24}, GCMSUnitLabelStr{24} );
fprintf(' %s', GCMS_MATRIX3{1,28}, GCMSUnitLabelStr{28} );
fprintf(' %s', GCMS_MATRIX3{1,29}, GCMSUnitLabelStr{29} );
fprintf(' %s', GCMS_MATRIX3{1,34}, GCMSUnitLabelStr{34} );

fprintf('\n\n\n Physical Properties');
fprintf('\n\n Sensor Mass:
fprintf('\n\n Sensor Dimensions - Length:
fprintf('\n\n Width:
fprintf('\n\n Height:
fprintf('\n\n Power Source Voltage:
fprintf('\n\n Power Source Voltage Type:
fprintf('\n\n Power Source Current:
fprintf('\n\n Power Requirements - Typical:
fprintf('\n\n Average:
fprintf('\n\n Peak:

```





```

fprintf(Fout, '\n High:');
fprintf(Fout, '\n Electron Source Voltage - Low:');
fprintf(Fout, '\n High:');
fprintf(Fout, '\n Ion Energy:');
for r = 1:length(GCMS_MATRIX3{1,20})
    fprintf(Fout, '%5.2f', GCMS_MATRIX3{1,20}(r) );
end
fprintf(Fout, ' %s', GCMSUnitLabelStr{20} );
fprintf(Fout, '\n Focus Voltage - Low:');
fprintf(Fout, '\n High:');
fprintf(Fout, '\n Electron Source Current - Low:');
fprintf(Fout, '\n High:');
fprintf(Fout, '\n Warm Up Period Mass Stability:');
fprintf(Fout, '\n Time:');
fprintf(Fout, '\n Sensor Return Data Rate:');

fprintf(Fout, '\n\n Physical Properties');
fprintf(Fout, '\n Sensor Mass:');
fprintf(Fout, '\n Sensor Dimensions - Length:');
fprintf(Fout, '\n Width:');
fprintf(Fout, '\n Height:');
fprintf(Fout, '\n Power Source Voltage:');
fprintf(Fout, '\n Power Source Current:');
fprintf(Fout, '\n Power Requirements - Typical:');
fprintf(Fout, '\n Average:');
fprintf(Fout, '\n Peak:');

fprintf(Fout, '\n\n Sensor Comments:');
for n = 1:length(GCMS_MATRIX3{1,38})
    fprintf(Fout, '\n %s', GCMS_MATRIX3{1,38}(n) );
end

fclose(Fout);

end

%10.2f %s', GCMS_MATRIX3{1,17}, GCMSUnitLabelStr{17} );
%10.2f %s', GCMS_MATRIX3{1,18}, GCMSUnitLabelStr{18} );
%10.2f %s', GCMS_MATRIX3{1,19}, GCMSUnitLabelStr{19} );
);

%10.2f %s', GCMS_MATRIX3{1,21}, GCMSUnitLabelStr{21} );
%10.2f %s', GCMS_MATRIX3{1,22}, GCMSUnitLabelStr{22} );
%10.2f %s', GCMS_MATRIX3{1,23}, GCMSUnitLabelStr{23} );
%10.2f %s', GCMS_MATRIX3{1,24}, GCMSUnitLabelStr{24} );
%10.2f %s', GCMS_MATRIX3{1,28}, GCMSUnitLabelStr{28} );
%10.2f %s', GCMS_MATRIX3{1,29}, GCMSUnitLabelStr{29} );
%10.2f %s', GCMS_MATRIX3{1,34}, GCMSUnitLabelStr{34} );

%10.2f %s', GCMS_MATRIX3{1,33}, GCMSUnitLabelStr{33} );
%10.2f %s', GCMS_MATRIX3{1,25}, GCMSUnitLabelStr{25} );
%10.2f %s', GCMS_MATRIX3{1,26}, GCMSUnitLabelStr{26} );
%10.2f %s', GCMS_MATRIX3{1,27}, GCMSUnitLabelStr{27} );
%10.2f %s', GCMS_MATRIX3{1,30}, GCMSUnitLabelStr{30} );
%10.2f %s', GCMS_MATRIX3{1,31} );
%10.2f %s', GCMS_MATRIX3{1,32}, GCMSUnitLabelStr{32} );
%10.2f %s', GCMS_MATRIX3{1,35}, GCMSUnitLabelStr{35} );
%10.2f %s', GCMS_MATRIX3{1,36}, GCMSUnitLabelStr{36} );
%10.2f %s', GCMS_MATRIX3{1,37}, GCMSUnitLabelStr{37} );

%
function HUMIDITY_SENSORS(UNITS,Print)
Humidity Sensor Database
Developed by: Keith Schreck
Mechanical and Aerospace Engineering
San Jose State University
Fall 2007
Date:
%

```

```

% This file contains Humidity Sensor values for the ISSPO program.
% Data here is loaded into the main program to determine the proper sensor
% to use for Missions based on the sensor requirements.
% Values are assigned to a variable name and saved to a MatLAB '.mat' file
% and loaded when calling the ISSPO program.

% Some of the values are based on the Unit system chosen and are loaded
% in separate sections.
%
% References
% http://content.honeywell.com/sensing/prodinfo/humiditymoisture/
function HUMIDITY_SENSORS(UNITS, Print)

% Load Constants values into Local Program
load('Constants_DB.mat');
load('Planet_Data.mat');

%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%
%% Humidity Sensor Data Base Created Based on un Unit System
%% 2 - D Matrix HUMIDITYSENSOR_DB(H_Type, Property)

if strcmpi(UNITS, 'SI') == 1
% Properties [ Sensor Humidity Range Capacitance at 55% RH
Stability Temp Range [ Frequency Range Length Width Min pf Typ pf Max pf
% Units # Min kHz Max %RH Max mm mm in pf Typ pf Max pf
%RH/yr Min deg C Max Min kHz Max %RH Max mm mm in pf Typ pf Max pf
HUMIDITYSENSOR_DB = {
0.2 -40 85 1 100 0 18.68 100 6.52 310 330 0 ;
'HIH-3610' 0 21.66 100 5.08 2.03 0 ;
0.2 -40 85 0 0 21.66 100 5.08 2.03 0 ;
'HIH-4000' 0 21.66 100 5.08 2.03 0 ;
1.2 -40 85 0 0 21.66 100 5.08 2.03 0 ;
};

HumUnitLabelStr = { 'sensor' % RH' % RH' 'pf' 'pf' 'pf/%RH' '+-%RH' 'sec' '%RH/yr' 'deg C' 'deg C' 'kHz' 'kHz'
'mm' 'mm' 'mm' 'gr' };
end

if strcmpi(UNITS, 'BRITISH') == 1
% Properties [ Sensor Humidity Range Capacitance at 55% RH
Stability Temp Range [ Frequency Range Length Width Min pf Typ pf Max pf
% Units # Min kHz Max %RH Max mm mm in pf Typ pf Max pf
%RH/yr Min deg F Max Min kHz Max %RH Max mm mm in pf Typ pf Max pf
HUMIDITYSENSOR_DB = {
0.2 -40 248 1 100 0 0.735 100 0.257 310 330 0 ;
'HIH-3610' 0 0.853 100 0.200 0.080 0 ;
0.2 -40 185 0 0 0.853 100 0.200 0.080 0 ;
'HIH-4000' 0 0.853 100 0.200 0.080 0 ;
1.2 -40 185 0 0 0.853 100 0.200 0.080 0 ;
};

HumUnitLabelStr = { 'N/A' % RH' % RH' 'pf' 'pf' 'pf' 'pf' '%RH/yr' '+-%RH' 'sec' '%RH/yr' 'deg F' 'deg F' 'kHz' 'in'
'in' 'in' 'oz' };
end

```

```

end

% Find the Max Operational Range
for h = 1:size(HUMIDITYSENSOR_DB,1)
    DELTA(h) = HUMIDITYSENSOR_DB{h,3} - HUMIDITYSENSOR_DB{h,2};
end

% Find the Maximum Humidity Range of the Sensor
MAX_DELTA = max(DELTA);

% For Multiple Sensors with the Same Range Reduce Sensor Matrix to Those
% with the Greatest Range
k = 1;
for c = 1:length(DELTA)
    if DELTA(c) == MAX_DELTA
        for j = 1:size(HUMIDITYSENSOR_DB,2)
            H_MATRIX{k,j} = HUMIDITYSENSOR_DB{c,j};
        end
        k = k + 1;
    end
end

% Run Case if More than One Sensor is Viable
if size(H_MATRIX,1) > 1
    % Sort Sensors by Max Operational Range
    for h = 1:size(H_MATRIX,1)
        H_TEMP(h) = H_MATRIX{h,12} - H_MATRIX{h,11};
    end
    [Temp, H_Loc ] = max(H_TEMP);

% Build Sensor Matrix Based on the Temp Range
k = 1;
for c = 1:length(H_TEMP)
    if H_TEMP(c) == Temp
        for j = 1:size(HUMIDITYSENSOR_DB,2)
            H_MATRIX2{k,j} = H_MATRIX{c,j};
        end
    end
    k = k + 1;
end
clear H_MATRIX;
H_MATRIX = H_MATRIX2;

end

% Sort Sensors by Volume if there is more than one sensor
if size(H_MATRIX,1) > 1
    % Sort Sensors by Min Volume
    for h = 1:size(H_MATRIX,1)
        H_Vol(h) = H_MATRIX{h,15} * H_MATRIX{h,16} * H_MATRIX{h,17};
    end
end

```

```

[Vol, H_Loc ] = min(H_Vol);
% Build Sensor Matrix Based on Volumes
k = 1;
for c = 1:length(H_Vol)
    if H_Vol(c) ~= Vol
        for j = 1:size(HUMIDITYSENSOR_DB,2)
            H_MATRIX3{k,j} = H_MATRIX{c,j};
        end
    end
    k = k + 1;
end
clear H_MATRIX;
H_MATRIX = H_MATRIX3;
end

H2O_FLG = 0;
% Search Planetary Atmosphere for Water Component
for w = 1:2:length(PlanAtm_Prop)
    if strcmpi(PlanAtm_Prop{1,w}, 'H2O') == 1
        PlanAtm_Prop{1,w};
        H2O_FLG = 1;
        break;
    end
end

if H2O_FLG == 0;
    if strcmpi(Print, 'Y') == 1
        fprintf('\n\n *** WARNING IN HUMIDITY SENSOR PROGRAM ***');
        fprintf('\n\n Humidity Sensor Not Required In Sensor Package Design!');
        fprintf('\n\n Water Component not found in Selected Planetary Atmosphere. ');
    end
end

H_MATRIX{1,1} = 'N/A';
for w = 2:size(HUMIDITYSENSOR_DB,2)
    H_MATRIX{1,w} = 0.0;
end

end

% Save Humidity Sensor Properties to Data File
save('HUMIDITY_SENSOR_Data.mat');
save('HUMIDITY_SENSOR_Final.mat', 'H_MATRIX', 'HumUnitLabelStr');

%*****
%
% *** PRINT SUMMARY Output Section for 'Print' = 'Y'
if strcmpi(Print, 'Y') == 1

```

```
fprintf('\n\n *** H U M I D I T Y   S E N S O R   S U M M A R Y   R E S U L T S   ***');
fprintf('\n\n *** I N P U T S   ***');
fprintf('\n\n Unit System:   %s', UNITS);

if H2O_FLG == 0;
    fprintf('\n\n *** WARNING IN HUMIDITY SENSOR PROGRAM   ***');
    fprintf('\n\n Humidity Sensor Not Required In Sensor Package Design!!');
    fprintf('\n\n Water Component not found in Selected Planetary Atmosphere. ');
end

fprintf('\n\n *****\n\n *****\n\n *****\n\n *****\n\n *****\n\n *****\n\n *****\n\n *****\n\n *****\n\n *****\n\n *****\n\n *****\n\n *****\n\n *****\n\n *****');

% Humidity Sensor Properties
fprintf('\n\n ** Sensor Properties **');
fprintf('\n\n Humidity Sensor Type:          %s', H_MATRIX{1,1});
fprintf('\n\n Humidity Sensing Range - Low:      %9.3f %s', H_MATRIX{1,2}, HumUnitLabelStr{2});
fprintf('\n\n Humidity Sensing Range - High:     %9.3f %s', H_MATRIX{1,3}, HumUnitLabelStr{3});
fprintf('\n\n Humidity Sensing Range - Min:     %9.3f %s', H_MATRIX{1,4}, HumUnitLabelStr{4});
fprintf('\n\n Humidity Sensing Range - Max:     %9.3f %s', H_MATRIX{1,5}, HumUnitLabelStr{5});
fprintf('\n\n Humidity Sensing Range - Typical: %9.3f %s', H_MATRIX{1,6}, HumUnitLabelStr{6});
fprintf('\n\n Humidity Sensing Range - Typical: %9.3f %s', H_MATRIX{1,7}, HumUnitLabelStr{7});
fprintf('\n\n Sensor Sensitivity:              %9.3f %s', H_MATRIX{1,8}, HumUnitLabelStr{8});
fprintf('\n\n Sensor Hysteresis:               %9.3f %s', H_MATRIX{1,9}, HumUnitLabelStr{9});
fprintf('\n\n Response Time:                   %9.3f %s', H_MATRIX{1,10}, HumUnitLabelStr{10});
fprintf('\n\n Sensor Stability:                %9.3f %s', H_MATRIX{1,11}, HumUnitLabelStr{11});

% Environmental Properties
fprintf('\n\n Environmental Properties');
fprintf('\n\n Temperature Range - Low:         %9.3f %s', H_MATRIX{1,11}, HumUnitLabelStr{11});
fprintf('\n\n Temperature Range - High:        %9.3f %s', H_MATRIX{1,12}, HumUnitLabelStr{12});
fprintf('\n\n Temperature Range - Min:         %9.3f %s', H_MATRIX{1,13}, HumUnitLabelStr{13});
fprintf('\n\n Temperature Range - Max:         %9.3f %s', H_MATRIX{1,14}, HumUnitLabelStr{14});

fprintf('\n\n Physical Properties');
fprintf('\n\n Sensor Dimensions - Length:      %9.3f %s', H_MATRIX{1,15}, HumUnitLabelStr{15});
fprintf('\n\n Sensor Dimensions - Width:       %9.3f %s', H_MATRIX{1,16}, HumUnitLabelStr{16});
fprintf('\n\n Sensor Dimensions - Height:      %9.3f %s', H_MATRIX{1,17}, HumUnitLabelStr{17});
fprintf('\n\n Sensor Mass:                     %9.3f %s', H_MATRIX{1,18}, HumUnitLabelStr{18});

end

%***
%*** Writes a text file summary of the Planetary Properties
Fout = fopen('HUMIDITY_SENSOR_Summary.txt','w+');

fprintf(Fout, '\n\n *** H U M I D I T Y   S E N S O R   S U M M A R Y   R E S U L T S   ***');
fprintf(Fout, '\n\n *** I N P U T S   ***');
fprintf(Fout, '\n\n Unit System:   %s', UNITS);

if H2O_FLG == 0;
    fprintf(Fout, '\n\n *** WARNING IN HUMIDITY SENSOR PROGRAM   ***');
end
```



```

% Values are assigned to a variable name and saved to a MatLAB '.mat' file
% and loaded when calling the ISSFO program.

% Some of the values are based on the Unit system chosen and are loaded
% in separate sections.

% References
% http://www.fairchildimaging.com/applications/aerospace.htm
% http://www.opttronics.com/products/commerce.exe?search=action&category=2050&keywords=all&template=Templates/Category.html
% http://bssc.sel.sony.com/BroadcastandBusiness/docs/brochures/dxc990.pdf
% http://www.fairchildimaging.com/library/

```

```
function IMAGING_SENSORS(SYS_TYPE,RESOLUTION,IMG_TYPE,UNITS,Print)
```

```

% Load Constants values into Local Program
load('Constants.DB.mat');
load('Planet_Data.mat');

```

```

% Define Local Variables
%RESOLUTION = OPT_DATA{2};
%IMG_TYPE = OPT_DATA{3:};
%IMG_TYPE = cellstr(IMG_TYPE);

```

```

%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%
%% IMAGING Sensor Data Base Created Based on un Unit System
%% 2 - D Matrix IMAGINGSENSOR_DB(IS_Type, Property)

```

```

if strcmpi(UNITS,'SI')==1
% Properties [
% Units
% IMAGINGSENSOR_DB1 = {
WIDTH microm LENGTH mm WIDTH mm @ 250kHz FULL WELL CAPACITY IMAGE TYPE IMAGING TYPE LINEARITY LENGTH # pixels ARRAY DIMENSIONS
IMAGINGSENSOR_DB1 = { 'Condor 486:90' 'CAMERA' e- @ 250kHz Pixel ke- Register e-/ADU ; ;
13 61.44 61.44 10 'Condor 486:135' 'CAMERA' 800 { 'X-Ray' } 1.5 1.0 4096 ; %1 4096
15 92.16 92.16 10 'Condor 486:200' 'CAMERA' 800 { 'X-Ray' } 1.5 1.0 4096 ; %2 4096
15 135.10 135.10 10 'Harrier 447' 'CAMERA' 800 { 'VISUAL' } 1.5 1.0 2048 ; %3 2048
15 31.00 31.00 12 'Perigrine 3041' 'CAMERA' 0 { 'VISUAL' } 2.5 1.0 2048 ; %4 2048
15 30.72 30.72 10 'Perigrine 486' 'CAMERA' 800 { 'UV' 'VISUAL' 'NIR' } 1.4 1.0 4096 ; %5 4096
15 61.40 61.40 10 'Perigrine 486SX' 'CAMERA' 800 { 'X-RAY' 'UV' 'VISUAL' 'NIR' } 1.5 1.0 4096 ; %6 4096
15 61.40 61.40 10 'Macrofire - MC' 'CAMERA' { 800 'MICRO' 'VISUAL' } 1.5 1.0 2048 ; %7 2048
7.4 15.20 15.20 0 'Macrofire-MC 1600x1200P' 'CAMERA' 0 { 'MICRO' 'VISUAL' } 0 0 1600 ; %8 1200
7.4 11.84 8.88 0 'QuantifIRE XI' 'CAMERA' 0 { 'MICRO' 'VISUAL' 'NIR' } 0 0 2048 ; %9 2048
7.4 15.20 15.20 0 'QuantifIRE XI' 'CAMERA' 0 { 'MICRO' 'VISUAL' 'NIR' } 0 0 2048 ; %10 2048

```



```

6.45      8.98      'Pixelfly QE'      'CAMERA'      { 'UV' 'VISUAL' 'NIR' }      1392      1024
6.60      0      0      'CAMERA'      18      ;
'Sony DXC-990P'      0      { 'VISUAL' }      752      %11
4.80      0      0      0      0      ;
2048
15      30.72      'CCD 3041'      'ARRAY'      { 'X-RAY' 'UV' 'VISUAL' 'NIR' 'MICRO' }      2048      2048
7      0      0      ;
15      15.36      'CCD 412'      'ARRAY'      { 'X-RAY' 'UV' 'VISUAL' 'NIR' 'MICRO' }      512      1024
45      85      ;
21      21.50      'CCD 424'      'ARRAY'      { 'X-RAY' 'UV' 'VISUAL' 'NIR' 'MICRO' }      1024      1024
0      0      0      ;
15      30.72      'CCD 442A'      'ARRAY'      { 'X-RAY' 'UV' 'VISUAL' 'NIR' 'MICRO' }      2048      2048
0      0      0      ;
15      30.72      'CCD 447'      'ARRAY'      { 'X-RAY' 'UV' 'VISUAL' 'NIR' 'MICRO' }      2048      2048
12      80      1.0      ;
15      61.20      'CCD 485'      'ARRAY'      { 'X-RAY' 'UV' 'VISUAL' 'NIR' 'MICRO' }      4096      4096
12      0      0      ;
13      0.013      'CCD 111'      'LINEAR'      { 'X-RAY' 'UV' 'VISUAL' 'NIR' 'MICRO' }      1      256
0      0      0      ;
13      0.013      'CCD 153A'      'LINEAR'      { 'X-RAY' 'UV' 'VISUAL' 'NIR' 'MICRO' }      1      512
0      0      0      ;

ImgUnitLabelStr1 = { 'N/A' 'N/A' 'N/A' '# Pixels' '# Pixels' 'mm' 'micro-m' 'mm' 'mm' 'e-' 'e-' 'ke-' 'ke-' 'e-/ADU'
};

```

```

% Properties [ OPERATING TEMP COOLING TEMP COOLING METHOD ADC DYNAMIC RANGE READOUT
RATES [ FRAME RATE Length Width Height V N/A mA mW ]
POWER [ Low deg C High deg C ] DIMENSIONS (mm) WEIGHT VOLTAGE V TYPE CURRENT
% Units { fps Length Width Height V N/A } MHz
sec IMAGINGSENSOR_DB2 = { 15.0 30.0 -60.0 'Thermoelectric w/Chilled Water' '16-bit' [ 4 2.5 2.1
1.6 ] [ 6.5 1.90 1.18 0.55 ] [ 0.15 0.52 0.85 1.8 ] 152 145 266 7.7 0 'N/A'
0 ; %1 15.0 30.0 -60.0 'Thermoelectric w/Cryo Cooler' '16-bit' [ 4 2.5 2.1
1.6 ] [ 6.5 1.90 1.18 0.55 ] [ 0.15 0.52 0.85 1.8 ] 165 131 395 16.4 0 'N/A'
0 ; %2 15.0 30.0 -60.0 'Thermoelectric w/Cryo Cooler' '16-bit' [ 4 2.5 2.1
1.6 ] [ 6.5 1.90 1.18 0.55 ] [ 0.15 0.52 0.85 1.8 ] 0 0 0 29.5 0 'N/A'
[ 0 ] [ 0 ] 0 0 0 'N/A' '14-bit' [ 4 ]
[ 0 ] 15.0 30.0 -25.0 'Thermoelectric w/Forced Air' '16-bit' 0 ; %4
[ 1.1 0.364 0.180 ] [ 0.9 2.7 5.6 ] 131 112 180 4.5 0 'N/A' 0 0 ;
%5 15.0 30.0 -60.0 'Thermoelectric w/Chilled Water' '16-bit' [ 4 2.5 2.1
1.6 ] [ 6.5 1.90 1.18 0.55 ] [ 0.15 0.52 0.85 1.8 ] 152 131 239 9.3 0 'N/A'
0 ; %6 15.0 30.0 -60.0 'Thermoelectric w/Chilled Water' '16-bit' [ 4 2.5 2.1
1.6 ] [ 6.5 1.90 1.18 0.55 ] [ 0.15 0.52 0.85 1.8 ] 152 131 239 9.3 0 'N/A'
0 ; %7 [ 0 60 ] [ 5 9 12 15 18 38 ] 0 134.6 0.0 58.4 0 'N/A' '12-bit' [ 0 ]
[ 0 60 ] [ 5 9 12 15 18 38 ] 0 134.6 134.6 58.4 0 'N/A' 0 0 ; %8

```

```

[ 0.0005 60 ] [ 10 18 24 30 48 ] 0 134.6 0.0 134.6 63.5 0 0 'N/A' '12-bit' [ 0 ]
%9
[ 0 1200 ] [ 5 9 12 15 18 38 ] 0 -20.0 'Thermoelectric to -20 Ambient' '12-bit' [ 0 ]
%10
[ 0.000005 65 ] [ 12 23 ] 40.0 39.0 0 'Digital Temperature Compensation' '12-bit' [ 20 ]
%11
] [ 0 ] [ 0 ] 45.0 0 70.0 72.0 123.5 0.630 'N/A' '10-bit' [ 0.0157
0 ; %12
[ 1.1 0.364 0.180 ] [ 0.9 2.7 5.6 ] 40.0 45.72 36.83 2.01 24 'N/A' 'DC' [ 2 1 ]
%13
[ 0 ] [ 0 ] 125.0 0 0 0 'N/A' 'DC' [ 0 ]
%14
[ 0 ] [ 0 ] 100.0 73.15 52.83 6.096 'N/A' 'DC' [ 0 ]
%15
[ 0 ] [ 0 ] 0 40.64 40.64 11.05 'N/A' 'DC' [ 0 ]
%16
[ 0 ] [ 0 ] 0 50.80 50.80 0 'N/A' 'DC' [ 0 ]
%17
[ 0 ] [ 0 ] -90.0 40.0 71.00 71.00 5.08 'N/A' 'DC' [ 0 ]
%18
[ 0 ] [ 0 ] -25.0 55.0 22.86 6.35 7.44 'N/A' 'DC' [ 10 ]
%19
[ 0 ] [ 0 ] -25.0 70.0 30.48 15.24 8.64 'N/A' 'DC' [ 20 ]
%20

```

```

ImgUnitLabelStr2 = { 'deg C' 'deg C' 'deg C' 'N/A' 'N/A' 'MHz' 'sec' 'fps' 'mm' 'mm' 'kg' 'V' 'Type'
'mA' 'mW' };
end

```

```

if strcmp(UNITS,'British')==1 SENSOR
% Properties
PIXEL SIZE IMAGE AREA [ {
% Units
milli-in LENGTH in WIDTH @1MHz e- @ 250kHz FULL WELL CAPACITY IMAGE GAIN ARRAY DIMENSIONS
IMAGINGSENSOR_DB1 = { 'Condor 486:90' 'CAMERA' Type Pixel ke- Register e-/ADU LINEARITY LENGTH # pixels WIDTH
0.5118 2.419 10 5 'CAMERA' 100 100 1.5 1.0 4096 ; %1
0.5905 3.628 3.628 10 5 'CAMERA' 100 100 1.5 1.0 4096 ; %2
0.5905 5.319 5.319 10 5 'CAMERA' 100 100 1.5 1.0 4096 ; %3
}
}

```

```

0.5905 1.220 1.220 'Harrrier 447' 'CAMERA' 0 { 'VISUAL' } 2048 %4 2048
0.5905 1.209 1.209 'Peregrine 3041' 'CAMERA' 0 { 'VISUAL' } 2048 %5 2048
0.5905 2.417 2.417 'Peregrine 486' 'CAMERA' 5 { 'UV' 'VISUAL' 'NIR' } 4096 %6 4096
0.5905 2.417 2.417 'Peregrine 486SX' 'CAMERA' 5 { 'X-RAY' 'UV' 'VISUAL' 'NIR' } 4096 %7 4096
0.2913 0.598 0.598 'Macrofire - MC' 'CAMERA' 0 { 'MICRO' 'VISUAL' } 2048 %8 2048
0.2913 0.466 0.466 'Macrofire-MC 1600x1200P' 'CAMERA' 0 { 'MICRO' 'VISUAL' } 1600 %9 1200
0.2913 0.598 0.598 'QuantifIRE XI' 'CAMERA' 0 { 'MICRO' 'VISUAL' 'NIR' } 2048 %10 2048
0.2539 0.353 0.260 'Pixelfly QE' 'CAMERA' 0 { 'UV' 'VISUAL' 'NIR' } 1392 %11 1024
0.3150 0.252 0.189 'Sony DXC-990P' 'CAMERA' 0 { 'VISUAL' } 752 %12 582
0.5905 1.210 1.210 'CCD 3041' 'ARRAY' 0 { 'X-RAY' 'UV' 'VISUAL' 'NIR' 'MICRO' } 2048 %13 2048
0.5905 0.605 0.605 'CCD 412' 'ARRAY' 0 { 'X-RAY' 'UV' 'VISUAL' 'NIR' 'MICRO' } 512 %14 1024
0.8268 0.846 0.846 'CCD 424' 'ARRAY' 0 { 'X-RAY' 'UV' 'VISUAL' 'NIR' 'MICRO' } 1024 %15 1024
0.5905 1.209 1.209 'CCD 442A' 'ARRAY' 0 { 'X-RAY' 'UV' 'VISUAL' 'NIR' 'MICRO' } 2048 %16 2048
0.5905 1.220 1.220 'CCD 447' 'ARRAY' 0 { 'X-RAY' 'UV' 'VISUAL' 'NIR' 'MICRO' } 2048 %17 2048
0.5905 2.410 2.410 'CCD 485' 'ARRAY' 0 { 'X-RAY' 'UV' 'VISUAL' 'NIR' 'MICRO' } 4096 %18 4096
0.5118 5.118E-03 0.131 'CCD 111' 'LINEAR' 0 { 'X-RAY' 'UV' 'VISUAL' 'NIR' 'MICRO' } 1 %19 256
0.5118 5.118E-03 0.262 'CCD 153A' 'LINEAR' 0 { 'X-RAY' 'UV' 'VISUAL' 'NIR' 'MICRO' } 1 %20 512
/ADU' '%';
ImgUnitLabelStr1 = { 'N/A' 'N/A' '# Pixels' '# Pixels' 'milli-in' 'in' 'in' 'e-' 'e-' 'ke-' 'ke-' 'e-

```

```

% Properties [ OPERATING TEMP COOLING TEMP COOLING METHOD ADC DYNAMIC RANGE READOUT
RATES [ FRAME RATE deg F width Height lbm N/A V N/A ma BTUs'hr ] CURRENT
POWER ] [ 6.5 1.90 1.18 0.55 ] [ 0.15 0.52 0.85 1.8 ] 6.00 5.70 10.50 17.0 0 'N/A' [ 4 2.5 2.1
% Units { Low deg F High Length 86.0 .76.0 6.00 5.70 10.50 17.0 0 'N/A' [ 4 2.5 2.1
sec { fps 59.0 86.0 .76.0 6.00 5.70 10.50 17.0 0 'N/A' [ 4 2.5 2.1
IMAGINGSENSOR_DB2 = { 59.0 86.0 .76.0 6.00 5.70 10.50 17.0 0 'N/A' [ 4 2.5 2.1
%1 [ 6.5 1.90 1.18 0.55 ] [ 0.15 0.52 0.85 1.8 ] 59.0 86.0 -76.0 'Thermoelectric w/Cryo Cooler' '16-bit' [ 4 2.5 2.1
%2 [ 6.5 1.90 1.18 0.55 ] [ 0.15 0.52 0.85 1.8 ] 59.0 86.0 -76.0 'Thermoelectric w/Cryo Cooler' '16-bit' [ 4 2.5 2.1
%3 [ 6.5 1.90 1.18 0.55 ] [ 0.15 0.52 0.85 1.8 ] 59.0 86.0 -76.0 'Thermoelectric w/Cryo Cooler' '16-bit' [ 4 2.5 2.1
%4 [ 6.5 1.90 1.18 0.55 ] [ 0.15 0.52 0.85 1.8 ] 59.0 86.0 -76.0 'Thermoelectric w/Cryo Cooler' '16-bit' [ 4 2.5 2.1
%5 [ 6.5 1.90 1.18 0.55 ] [ 0.15 0.52 0.85 1.8 ] 59.0 86.0 -76.0 'Thermoelectric w/Cryo Cooler' '16-bit' [ 4 2.5 2.1
%6 [ 6.5 1.90 1.18 0.55 ] [ 0.15 0.52 0.85 1.8 ] 59.0 86.0 -76.0 'Thermoelectric w/Cryo Cooler' '16-bit' [ 4 2.5 2.1
%7 [ 6.5 1.90 1.18 0.55 ] [ 0.15 0.52 0.85 1.8 ] 59.0 86.0 -76.0 'Thermoelectric w/Cryo Cooler' '16-bit' [ 4 2.5 2.1
%8 [ 6.5 1.90 1.18 0.55 ] [ 0.15 0.52 0.85 1.8 ] 59.0 86.0 -76.0 'Thermoelectric w/Cryo Cooler' '16-bit' [ 4 2.5 2.1
%9 [ 6.5 1.90 1.18 0.55 ] [ 0.15 0.52 0.85 1.8 ] 59.0 86.0 -76.0 'Thermoelectric w/Cryo Cooler' '16-bit' [ 4 2.5 2.1
%10 [ 6.5 1.90 1.18 0.55 ] [ 0.15 0.52 0.85 1.8 ] 59.0 86.0 -76.0 'Thermoelectric w/Cryo Cooler' '16-bit' [ 4 2.5 2.1
%11 [ 6.5 1.90 1.18 0.55 ] [ 0.15 0.52 0.85 1.8 ] 59.0 86.0 -76.0 'Thermoelectric w/Cryo Cooler' '16-bit' [ 4 2.5 2.1
%12 [ 6.5 1.90 1.18 0.55 ] [ 0.15 0.52 0.85 1.8 ] 59.0 86.0 -76.0 'Thermoelectric w/Cryo Cooler' '16-bit' [ 4 2.5 2.1
%13 [ 6.5 1.90 1.18 0.55 ] [ 0.15 0.52 0.85 1.8 ] 59.0 86.0 -76.0 'Thermoelectric w/Cryo Cooler' '16-bit' [ 4 2.5 2.1
%14 [ 6.5 1.90 1.18 0.55 ] [ 0.15 0.52 0.85 1.8 ] 59.0 86.0 -76.0 'Thermoelectric w/Cryo Cooler' '16-bit' [ 4 2.5 2.1
%15 [ 6.5 1.90 1.18 0.55 ] [ 0.15 0.52 0.85 1.8 ] 59.0 86.0 -76.0 'Thermoelectric w/Cryo Cooler' '16-bit' [ 4 2.5 2.1
%16 [ 6.5 1.90 1.18 0.55 ] [ 0.15 0.52 0.85 1.8 ] 59.0 86.0 -76.0 'Thermoelectric w/Cryo Cooler' '16-bit' [ 4 2.5 2.1
%17 [ 6.5 1.90 1.18 0.55 ] [ 0.15 0.52 0.85 1.8 ] 59.0 86.0 -76.0 'Thermoelectric w/Cryo Cooler' '16-bit' [ 4 2.5 2.1
%18 [ 6.5 1.90 1.18 0.55 ] [ 0.15 0.52 0.85 1.8 ] 59.0 86.0 -76.0 'Thermoelectric w/Cryo Cooler' '16-bit' [ 4 2.5 2.1
%19 [ 6.5 1.90 1.18 0.55 ] [ 0.15 0.52 0.85 1.8 ] 59.0 86.0 -76.0 'Thermoelectric w/Cryo Cooler' '16-bit' [ 4 2.5 2.1
%20 [ 6.5 1.90 1.18 0.55 ] [ 0.15 0.52 0.85 1.8 ] 59.0 86.0 -76.0 'Thermoelectric w/Cryo Cooler' '16-bit' [ 4 2.5 2.1

```

```

[ 0 ] [ 0 ] 0 0 0 0 0 0 'N/A' '14-bit' 0 [ 4 ]
[ 1.1 0.364 0.180 ] 59.0 86.0 0 -13.0 'Thermoelectric w/Forced Air' 'N/A' '16-bit' 0 ; %4 [ 2 1 ]
%5 [ 0.9 2.7 5.6 ] 4.40 7.10 10.0 0 'N/A' 0 ;
1.6 ] [ 6.5 1.90 1.18 0.55 ] [ 0.15 0.52 0.85 1.8 ] 6.00 5.20 9.40 20.5 0 'N/A' [ 4 2.5 2.1
0 ; %6 'Thermoelectric w/Chilled Water'
1.6 ] [ 6.5 1.90 1.18 0.55 ] [ 0.15 0.52 0.85 1.8 ] 6.00 5.20 9.40 20.5 0 'N/A' [ 4 2.5 2.1
0 ; %7 'Thermoelectric w/Chilled Water'
[ 0 60 ] [ 5 9 12 15 18 38 ] 0 5.30 2.30 0 0 'N/A' '12-bit' 0 [ 0 ]
[ 0.0005 60 ] [ 10 18 24 30 48 ] 0 5.30 2.50 0 0 'N/A' '12-bit' 0 ; %8 [ 0 ]
%9 'N/A' '12-bit' 0 [ 0 ]
[ 0 1200 ] [ 5 9 12 15 18 38 ] 0 5.40 -20.0 'Thermoelectric to -20 Ambient' 'N/A' '12-bit' 0 0 ;
%10 50.0 104.0 1.54 0 'Digital Temperature Compensation' 'N/A' '12-bit' 0 [ 20 ]
[ 0.000005 65 ] [ 12 23 ] 1.54 1.54 2.68 9.53 0 'N/A' '10-bit' 0 [ 0.0157
%11 23.0 [ 0 ] 2.88 2.88 4.88 1.38 0 'N/A' 0
0 ; %12 [ 0 ] 1.45 0.079 'N/A' 0 24 'DC' 'N/A' 0 [ 2 1 ]
%13 [ 1.1 0.364 0.180 ] -148.0 104.0 1.80 0 24 'DC' 'N/A' 0
[ 0 ] [ 0 ] -67.0 257.0 0 0 24 'DC' 'N/A' 5 0.17 [ 0 ]
%14 [ 0 ] [ 0 ] -22.0 212.0 0 0 24 'DC' 'N/A' 0 [ 0 ]
[ 0 ] [ 0 ] 0 2.88 0.24 24 'DC' 'N/A' 0 0 ; %15
[ 0 ] [ 0 ] 0 1.60 0.435 22 'DC' 'N/A' 0 0 ; %16
[ 0 ] [ 0 ] 0 2.00 0 24 'DC' 'N/A' 0 0 ; %17
[ 0 ] [ 0 ] -130.0 104.0 2.80 0 20 'DC' 'N/A' 0 [ 0 ]
%18 [ 0 ] [ 0 ] -13.0 131.0 0.20 20 'DC' 'N/A' 0 [ 10 ]
[ 0 ] [ 0 ] -13.0 158.0 0.25 15 'DC' 'N/A' 0 20.83 ; %19
%20 [ 0 ] [ 0 ] -13.0 158.0 0.34 15 'DC' 'N/A' 0 0.581 [ 20 ]
];

ImgUnitLabelStr2 = { 'deg F' 'deg F' 'deg F' 'N/A' 'N/A' 'MHz' 'sec' 'fps' 'in' 'in' 'in' 'lbm' 'V' 'Type'
'mA' 'BTUs/hr' };
end

% Comments on Features / Use
IMAGINGSENSOR_DB3{1,1} = { 'Ultra-Sensitive' ;
'Fiber-Optic coupled Camera' ;
'Scientific Grade' ;

```

```

'Low-noise, dual-speed, four-port readout architecture' ;
'USB 2.0 Computer Interface' };
IMAGINGSENSOR_DB3{2,1} = {
'Largest Commercially Available CCD' ;
'Scientific X-ray imaging and electron imaging' ;
'Ultra-Sensitive' ;
'Fiber-Optic coupled Camera' ;
'Scientific Grade' ;
'Low-noise, dual-speed, four-port readout architecture' ;
'USB 2.0 Computer Interface' };
IMAGINGSENSOR_DB3{3,1} = {
'Flight Rated, used on I-SPOT Mission' ;
'Largest Commercially Available CCD' ;
'Scientific X-ray imaging and electron imaging' ;
'Ultra-Sensitive' ;
'Fiber-optic coupled Camera' ;
'Scientific Grade' ;
'Low-noise, dual-speed, four-port readout architecture' ;
'USB 2.0 Computer Interface' };
IMAGINGSENSOR_DB3{4,1} = {
'Anti-blooming and integration control' };
'Multiport Readout' ;
'Software Selectable Gain' ;
'TTL Trigger I/O' ;
'USB 2.0 interface' ;
'High Optical Throughput' ;
'Supports standard Pmount lenses and mounting hardware' ;
IMAGINGSENSOR_DB3{5,1} = {
'Low noise, multi-speed, multiport readout architecture' ;
'Back-illuminated CCD option' ;
'Highest quantum efficiency' ;
'Four-port readout' ;
'Optimal design for speed and sensitivity' ;
'Dual Digitizer' ;
'Separate, optimized readout channels for lowest noise and highest speed' ;
'Deep Cooling & Low Noise Readout' ;
'Software Gain & Binning' ;
'Scientific precision and accuracy' };
IMAGINGSENSOR_DB3{6,1} = {
'Back-illuminated CCD option' ;
'Highest quantum efficiency' ;
'Four-port readout' ;
'Optimal design for speed and sensitivity' ;
'Deep thermoelectric cooling' ;
'Minimize dark noise' ;
'High-performance low-noise electronics' ;
'Minimize readout noise' ;
'Scientific precision and accuracy' ;
'Software-controlled binning & windowing' ;
'Optimize speed versus resolution' };
IMAGINGSENSOR_DB3{7,1} = {
'Back-illuminated CCD option' ;
'Highest quantum efficiency' ;
'Four-port readout' ;
'Optimal design for speed and sensitivity' ;
'Deep thermoelectric cooling' ;
'Minimize dark noise' ;
'High-performance low-noise electronics' ;
'Minimize readout noise' ;

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```

'Scientific precision and accuracy' ;
'Software-controlled binning & windowing' ;
'Optimize speed versus resolution' ;
'Microscope Monochrome Camera' ;
'500% increase in Field-of-View' ;
'Progressive Scan Interline CCD Array' ;
'Optional Automated RGB 3-Pass Color Filter' ;
'Software image editing tools' ;
'IEEE 1394 FireWire 6-pin Computer Interface' ;
'F-Mount Optical Interface' ;
'Electronic Shutter: Integration time to 60 seconds' ;
'ISO 9001:2000 CERTIFIED' ;
'Microscope Monochrome Camera' ;
'2-Megapixel Monochrome Imager' ;
'Progressive Scan Interline CCD Array' ;
'Optional Automated RGB 3-Pass Color Filter' ;
'Software image editing tools' ;
'IEEE 1394 FireWire 6-pin Computer Interface' ;
'C-Mount Optical Interface' ;
'Electronic Shutter: Integration time 500 micro-sec to 60 seconds' ;
'ISO 9001:2000 CERTIFIED' ;
'Microscope Monochrome Camera' ;
'500% increase in Field-of-View' ;
'Visual to Near Infra-Red Spectrum Range' ;
'Optional Automated RGB 3-Pass Color Filter' ;
'Software image editing tools' ;
'IEEE 1394 FireWire 6-pin Computer Interface' ;
'F-Mount Optical Interface' ;
'Electronic Shutter: Integration time to 20 minutes' ;
'ISO 9001:2000 CERTIFIED' ;
'Optional Color CCD Sensor' ;
'Readout noise: 7e- rms typical' ;
'Spectral Range: 290 - 1100 nm' ;
'12 W Power Consumption' ;
'Software camware and software development kit included' ;
'High speed serial LVDS shielded ethernet patch cable RJ45 connector computer interface' ;
'Hardened against high magnetic fields' ;
'3-CCD color video camera' ;
'Bayonet mount adapts various high quality, professional lenses' ;
'850 TV line Horizontal resolution' ;
'Approx. 8 W Power Consumption' ;
'6 Automatic Exposure Modes' ;
'+30 dB sensitivity gain for low light conditions' ;
'RGB, component, Y/C and composite video outputs, RS-232C controllable' ;
'Applications for microscopy, industrial, inspection and remote camera systems' ;
'Back-illuminated CCD Array' ;
'Highest quantum efficiency' ;
'Multiple output image formats' ;
'Four-port readout' ;
'Optimal design for speed and sensitivity' ;
'Dual Digitizer' ;
'Separate, optimized readout channels for lowest noise and highest speed' ;
'Space Qualified - Production Company' ;
'Software Gain & Binning' ;

```

```

IMAGINGSENSOR_DB3{8,1} = {
IMAGINGSENSOR_DB3{9,1} = {
IMAGINGSENSOR_DB3{10,1} = {
IMAGINGSENSOR_DB3{11,1} = {
IMAGINGSENSOR_DB3{12,1} = {
IMAGINGSENSOR_DB3{13,1} = {

```



```

ImgUnitLabelStr = cat(2,ImgUnitLabelStr1,ImgUnitLabelStr2);
IMG_MATRIX = IMAGINGSENSOR_DB;
% Select Sensors By System Type - 'CAMERA' or 'ARRAY' or 'LINEAR'
tmp = 1;
for typ = 1:size(IMG_MATRIX,1)
    if strcmpi(IMG_MATRIX{typ,2},SYS_TYPE) == 1
        for s = 1:size(IMG_MATRIX,2)
            IMG_MATRIX2{tmp,s} = IMG_MATRIX{typ,s};
        end
        tmp = tmp + 1;
    end
end
% Down Select Sensors that Match Imaging Types
for loc = 1:length(IMG_TYPE)
    STR1 = IMG_TYPE{loc};
    if strcmpi(STR1, 'X-RAY') == 1
        C = 1;
        FLAG = 0;
        for img = 1:size(IMG_MATRIX2,1)
            for pos = 1:length(IMG_MATRIX2{img,3})
                STR2 = IMG_MATRIX2{img,3}{pos};
                if strcmpi(STR1,STR2) == 1
                    for s = 1:size(IMG_MATRIX2,2)
                        IMG_MATRIX_TMP{C,s} = IMG_MATRIX2{img,s};
                    end
                    FLAG = 1;
                end
            end
        end
        if FLAG == 1
            C = C + 1;
        end
        FLAG = 0;
    end
end
if exist('IMG_MATRIX_TMP') == 0
    if strcmpi(print,'Y') == 1
        fprintf('\n\n *** ERROR: NULL DATABASE ***');
        fprintf('\n\n No Sensor Exist in Database which Meet Imaging Requirements!');
        fprintf('\n\n Image Type: %s', STR1);
        IMG_MATRIX_TMP = IMG_MATRIX2;
    end
end
clear IMG_MATRIX2;
IMG_MATRIX2 = IMG_MATRIX_TMP;
clear IMG_MATRIX_TMP;
end
if strcmpi(STR1, 'VISUAL') == 1
    C = 1;
    FLAG = 0;

```



```

for img = 1:size(IMG_MATRIX2,1)
    for pos = 1:length(IMG_MATRIX2{img,3})
        STR2 = IMG_MATRIX2{img,3}{pos};
        if strcmpi(STR1,STR2) == 1
            for s = 1:size(IMG_MATRIX2,2)
                IMG_MATRIX_TMP{c,s} = IMG_MATRIX2{img,s};
            end
            FLAG = 1;
        end
    end
    if FLAG == 1
        C = C + 1;
    end
    FLAG = 0;
end

if exist('IMG_MATRIX_TMP') == 0
    if strcmpi(Print,'Y') == 1
        fprintf('\n\n *** ERROR: NULL DATABASE ***');
        fprintf('\n\n No Sensor Exist in Database which Meet Imaging Requirements!');
        IMG_MATRIX_TMP = IMG_MATRIX2;
    end
end

clear IMG_MATRIX2;
IMG_MATRIX2 = IMG_MATRIX_TMP;
clear IMG_MATRIX_TMP;

if strcmpi(STR1, 'UV') == 1
    C = 1;
    FLAG = 0;
    for img = 1:size(IMG_MATRIX2,1)
        for pos = 1:length(IMG_MATRIX2{img,3})
            STR2 = IMG_MATRIX2{img,3}{pos};
            if strcmpi(STR1,STR2) == 1
                for s = 1:size(IMG_MATRIX2,2)
                    IMG_MATRIX_TMP{c,s} = IMG_MATRIX2{img,s};
                end
            end
            FLAG = 1;
        end
    end
    if FLAG == 1
        C = C + 1;
    end
    FLAG = 0;
end

if exist('IMG_MATRIX_TMP') == 0
    if strcmpi(Print,'Y') == 1
        fprintf('\n\n *** ERROR: NULL DATABASE ***');
        fprintf('\n\n No Sensor Exist in Database which Meet Imaging Requirements!');
        IMG_MATRIX_TMP = IMG_MATRIX2;
    end
end

```

```

        IMG_MATRIX_TMP = IMG_MATRIX2;
    end
end

clear IMG_MATRIX2;
IMG_MATRIX2 = IMG_MATRIX_TMP;
clear IMG_MATRIX_TMP;
end

if strcmpi(STR1, 'MICRO') == 1
    C = 1;
    FLAG = 0;
    for img = 1:size(IMG_MATRIX2,1)
        for pos = 1:length(IMG_MATRIX2{img,3})
            STR2 = IMG_MATRIX2{img,3}{pos};
            if strcmpi(STR1,STR2) == 1
                for s = 1:size(IMG_MATRIX2,2)
                    IMG_MATRIX_TMP{c,s} = IMG_MATRIX2{img,s};
                end
            end
            FLAG = 1;
        end
    end
end
if FLAG == 1
    C = C + 1;
end
FLAG = 0;
end

if exist('IMG_MATRIX_TMP') == 0
    if strcmpi('print','Y') == 1
        fprintf('\n\n *** ERROR: NULL DATABASE ***');
        fprintf('\n\n No Sensor Exist in Database which Meet Imaging Requirements!');
        fprintf('\n\n Image Type: %s', STR1);
        IMG_MATRIX_TMP = IMG_MATRIX2;
    end
end

clear IMG_MATRIX2;
IMG_MATRIX2 = IMG_MATRIX_TMP;
clear IMG_MATRIX_TMP;
end

end

% Down Select from Multiple Sensor Options
if size(IMG_MATRIX2,1) > 1
    if strcmpi(RESOLUTION, 'HIGH') == 1
        MPixels = 8;
        TOL_Low = 1;
        TOL_High = 100;
    end
    if strcmpi(RESOLUTION, 'MEDIUM') == 1
        MPixels = 4;

```

```

TOL_Low = 2;
TOL_High = 4;
end
if strcmpi(RESOLUTION, 'LOW') == 1
    MPixels = 1;
    TOL_Low = 1;
    TOL_High = 0;
end

for img = 1:size(IMG_MATRIX2,1)
    % Convert Number of Pixels to MegaPixel resolutions
    IMG_SIZE(img) = ( IMG_MATRIX2{img,4} * IMG_MATRIX2{img,5})/1024 / 1024;
end

% Sort Sensors by Resolution Scale
k = 1;
for img = 1:size(IMG_MATRIX2,1)
    if (IMG_SIZE(img) >= (MPixels-TOL_Low)) && (IMG_SIZE(img) <= (MPixels+TOL_High))
        for j = 1:size(IMG_MATRIX2,2)
            IMG_MATRIX3{k,j} = IMG_MATRIX2{img,j};
        end
        k = k + 1;
    end
end

% Copies IMG_MATRIX2 to IMG_MATRIX3 if all Sensors Eliminated
if exist('IMG_MATRIX3') == 0
    if strcmpi(Print, 'Y') == 1
        fprintf('\n *** WARNING: Optical Sensors Eliminated based on Chosen Resolution!');
    end
    IMG_MATRIX3 = IMG_MATRIX2;
end

if size(IMG_MATRIX3,1) > 1
    % Create Mass / Volume Scaling Factor Optimization
    for mv = 1:size(IMG_MATRIX3,1)
        IMG_SCALE_FAC(mv) = IMG_MATRIX3{mv,23} * IMG_MATRIX3{mv,24} * IMG_MATRIX3{mv,25} + IMG_MATRIX3{mv,26};
        if (IMG_MATRIX3{mv,23} == 0) || (IMG_MATRIX3{mv,24} == 0) || (IMG_MATRIX3{mv,25} == 0) || (IMG_MATRIX3{mv,26} == 0)
            IMG_SCALE_FAC(mv) = 9.99E99;
        end
    end

    [ IMG_MIN_SCL, INDEX ] = min(IMG_SCALE_FAC);

    % Build Sensor Matrix Based on Scale Factors
    k = 1;
    for c = 1:length(IMG_SCALE_FAC)
        if IMG_SCALE_FAC(c) == IMG_MIN_SCL
            for j = 1:size(IMG_MATRIX3,2)
                IMG_MATRIX4{k,j} = IMG_MATRIX3{c,j};
            end
            k = k + 1;
        end
    end
end

```

```

end
end
end
% Copies IMG_MATRIX3 to IMG_MATRIX4 if all Sensors Eliminated
if exist('IMG_MATRIX4') == 0
    IMG_MATRIX4 = IMG_MATRIX3;
end
if size(IMG_MATRIX4,1) > 1
    for we = 1:size(IMG_MATRIX4,1)
        Max_WE_Des(we) = IMG_MATRIX4{we,12};
    end
    Max_WE_Sens = max(Max_WE_Des);
    k = 1;
    for we = 1:size(IMG_MATRIX4,1)
        if IMG_MATRIX4{we,12} == Max_WE_Sens
            for j = 1:size(IMG_MATRIX4,2)
                IMG_MATRIX5{k,j} = IMG_MATRIX4{we,j};
            end
            k = k + 1;
        end
    end
end
% Copies IMG_MATRIX4 to IMG_MATRIX5 if all Sensors Eliminated
if exist('IMG_MATRIX5') == 0
    IMG_MATRIX5 = IMG_MATRIX4;
end
end
% Copies IMG_MATRIX2 to IMG_MATRIX5 if all Sensors Eliminated
if exist('IMG_MATRIX5') == 0
    IMG_MATRIX5 = IMG_MATRIX2;
end
end
% Save Density Sensor Properties to Data File
save('IMAGING_SENSOR_Data.mat');
save('IMAGING_SENSOR_Final.mat', 'IMG_MATRIX5', 'ImgUnitLabelStr');
%*****
% *** PRINT SUMMARY Output Section for 'Print' = 'Y'
if strcmpi(Print, 'Y') == 1
    fprintf('\n\n *** I M A G I N G S E N S O R S U M M A R Y R E S U L T S ***');
    fprintf('\n\n *** I N P U T S ***');

```



```

fprintf('\n\n Width: %10.2f %s', IMG_MATRIX5{1,24}, ImgUnitLabelStr{24} );
fprintf('\n\n Height: %10.2f %s', IMG_MATRIX5{1,25}, ImgUnitLabelStr{25} );
fprintf('\n\n Sensor Mass: %10.3f %s', IMG_MATRIX5{1,26}, ImgUnitLabelStr{26} );

fprintf('\n\n Power Requirements');
fprintf('\n\n Input Voltage: %9.3f %s', IMG_MATRIX5{1,27}, ImgUnitLabelStr{27} );
fprintf('\n\n Input Voltage Type: %s %s', IMG_MATRIX5{1,28}, ImgUnitLabelStr{28} );
fprintf('\n\n Input Amperage: %9.3f %s', IMG_MATRIX5{1,29}, ImgUnitLabelStr{29} );
fprintf('\n\n Power Requirement: %9.3f %s', IMG_MATRIX5{1,30}, ImgUnitLabelStr{30} );

fprintf('\n\n Sensor Comments:');
for r = 1:length(IMG_MATRIX5{1,31})
    fprintf('\n %s', IMG_MATRIX5{1,31}{r} );
end

end

%***** Writes a text file summary of the Planetary Properties
%*****
Fout = fopen('IMAGING_SENSOR_Summary.txt','w+');

fprintf(Fout, '\n\n *** I M A G I N G S E N S O R S U M M A R Y R E S U L T S ***\n\n');
fprintf(Fout, '\n\n*** I N P U T S ***\n\n');
fprintf(Fout, '\n System Type: %s', SYS_TYPE);
fprintf(Fout, '\n Resolution: %s', RESOLUTION);
fprintf(Fout, '\n Image Type: ');
for r = 1:length(IMG_TYPE)
    fprintf(Fout, '%s', IMG_TYPE{r} );
end
fprintf(Fout, '\n Unit System: %s', UNITS);

fprintf(Fout, '\n\n*****\n\n*****\n\n*****\n\n*****\n\n');

% Imaging Sensor Properties
fprintf(Fout, '\n\n** Sensor Properties **\n\n');
fprintf(Fout, '\n Imaging Sensor Type: %s', IMG_MATRIX5{1,1});
fprintf(Fout, '\n Imaging Spectrum: ');
for r = 1:length(IMG_MATRIX5{1,3})
    fprintf(Fout, '%s', IMG_MATRIX5{1,3}{r} );
end
fprintf(Fout, '\n\n Array Dimensions - Length: %10d %s', IMG_MATRIX5{1,4}, ImgUnitLabelStr{4} );
fprintf(Fout, '\n\n Width: %10d %s', IMG_MATRIX5{1,5}, ImgUnitLabelStr{5} );
fprintf(Fout, '\n\n Sensor Pixel Size: %10.4f %s', IMG_MATRIX5{1,6}, ImgUnitLabelStr{6} );
fprintf(Fout, '\n\n Imaging Array Size - Length: %10.3f %s', IMG_MATRIX5{1,7}, ImgUnitLabelStr{7} );
fprintf(Fout, '\n\n Width: %10.3f %s', IMG_MATRIX5{1,8}, ImgUnitLabelStr{8} );
fprintf(Fout, '\n\n Read Noise at 1 MHz: %10d %s', IMG_MATRIX5{1,9}, ImgUnitLabelStr{9} );
fprintf(Fout, '\n\n 250 kHz: %10d %s', IMG_MATRIX5{1,10}, ImgUnitLabelStr{10} );
fprintf(Fout, '\n\n Full Well Capacity - Pixel: %10d %s', IMG_MATRIX5{1,11}, ImgUnitLabelStr{11} );
fprintf(Fout, '\n\n Register: %10d %s', IMG_MATRIX5{1,12}, ImgUnitLabelStr{12} );
fprintf(Fout, '\n\n Sensor Gain: %10.2f %s', IMG_MATRIX5{1,13}, ImgUnitLabelStr{13} );
fprintf(Fout, '\n\n Sensor Linearity: %10.2f %s', IMG_MATRIX5{1,14}, ImgUnitLabelStr{14} );
fprintf(Fout, '\n\n ADC Dynamic Range: %s', IMG_MATRIX5{1,19}, ImgUnitLabelStr{19} );

```

```

fprintf(Fout, '\n Readout Rates:
for i = 1:length(IMG_MATRIX5{1,20})
    fprintf(Fout, '%7.4f', IMG_MATRIX5{1,20}(i) );
end
fprintf(Fout, ' %s', ImgUnitLabelStr{20} );
fprintf(Fout, '\n Readout Time:
for i = 1:length(IMG_MATRIX5{1,21})
    fprintf(Fout, '%9.6f', IMG_MATRIX5{1,21}(i) );
end
fprintf(Fout, ' %s', ImgUnitLabelStr{21} );
fprintf(Fout, '\n Frame Rates:
for i = 1:length(IMG_MATRIX5{1,22})
    fprintf(Fout, '%6.2f', IMG_MATRIX5{1,22}(i) );
end
fprintf(Fout, ' %s', ImgUnitLabelStr{22} );

% Environmental Properties
fprintf(Fout, '\n\n Environmental Properties');
fprintf(Fout, '\n Operating Temperature Range - Low: %10.2f %s', IMG_MATRIX5{1,15}, ImgUnitLabelStr{15} );
fprintf(Fout, '\n High: %10.2f %s', IMG_MATRIX5{1,16}, ImgUnitLabelStr{16} );
fprintf(Fout, '\n Cooled Temperature: %10.2f %s', IMG_MATRIX5{1,17}, ImgUnitLabelStr{17} );
fprintf(Fout, '\n Cooling Method: %s %s', IMG_MATRIX5{1,18}, ImgUnitLabelStr{18} );

fprintf(Fout, '\n\n Physical Properties');
fprintf(Fout, '\n Dimensions - Length: %10.2f %s', IMG_MATRIX5{1,23}, ImgUnitLabelStr{23} );
fprintf(Fout, '\n Width: %10.2f %s', IMG_MATRIX5{1,24}, ImgUnitLabelStr{24} );
fprintf(Fout, '\n Height: %10.2f %s', IMG_MATRIX5{1,25}, ImgUnitLabelStr{25} );
fprintf(Fout, '\n Sensor Mass: %10.3f %s', IMG_MATRIX5{1,26}, ImgUnitLabelStr{26} );

fprintf(Fout, '\n\n Power Requirements');
fprintf(Fout, '\n Input Voltage: %9.3f %s', IMG_MATRIX5{1,27}, ImgUnitLabelStr{27} );
fprintf(Fout, '\n Input Voltage Type: %s %s', IMG_MATRIX5{1,28}, ImgUnitLabelStr{28} );
fprintf(Fout, '\n Input Amperage: %9.3f %s', IMG_MATRIX5{1,29}, ImgUnitLabelStr{29} );
fprintf(Fout, '\n Power Requirement: %9.3f %s', IMG_MATRIX5{1,30}, ImgUnitLabelStr{30} );

fprintf(Fout, '\n\n Sensor Comments:');
for r = 1:length(IMG_MATRIX5{1,31})
    fprintf(Fout, '\n %s', IMG_MATRIX5{1,31}{r} );
end

fclose(Fout);

```

```

D12. Inclinometer Sensors
% function INCLINOMETER_SENSORS(INCR,UNITS,Print)
% Inclinometer Sensor Database
%
% Developed by: Keith Schreck
% Mechanical and Aerospace Engineering
% San Jose State University
% Date: Fall 2007
%
% This file contains Inclinometer Sensor values for the ISSPO program.
% Data here is loaded into the main program to determine the proper sensor
% to use for Missions based on the sensor requirements.
% Values are assigned to a variable name and saved to a MatLAB '.mat' file
% and loaded when calling the ISSPO program.
%
% Some of the values are based on the Unit system chosen and are loaded
% in separate sections.
%
% References
% http://www.riekerinc.com/ElectronicInclinometers.htm
% http://www.jewellinstruments.com/inclinometer.htm
% http://www.spectrionsensors.com/tilt.html
function INCLINOMETER_SENSORS(INCR,UNITS,Print)
%
% Load Constants values into Local Program
load('Constants_DB.mat');
load('Planet_Data.mat');
PLTEMP = Planet_Properties{7};
%
% Define Program Variables For Max Inclination Range
if strcmpi(INCR,'HIGH') == 1
    INC_ANGLE = 90;
    TOL_LOWER = 46;
end
if strcmpi(INCR,'MEDIUM') == 1
    INC_ANGLE = 45;
    TOL_LOWER = 30;
end
if strcmpi(INCR,'LOW') == 1
    INC_ANGLE = 30;
    TOL_LOWER = 0;
end

%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%
%% INCLINOMETER Sensor Data Base Common Properties
%% 2 - D Matrix INCLINOMETERSENSOR_DB(IS_Type, Property)
%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%
% Properties [ SENSOR INCLINATION RANGE OUTPUT DATA RESOLUTION SENSITIVITY NON-LINEARITY
VOLTAGE TYPE VOLTAGE RESPONSE TIME BANDWIDTH ]

```



```

% N/A Units Volts
INCLINOMETERSENSOR_DB1 = {
'DC' [ 8 30 ]
'DC' [ 8 30 ]
'DC' [ 8 30 ]
'DC' [ 12 30 ]
'DC' [ 12 30 ]
'DC' [ 12 30 ]
'DC' [ 12 30 ]
'DC' [ 12 30 ]
'DC' [ 3 6 ]
'DC' [ 3 6 ]
'DC' [ 3 6 ]
'DC' [ 8 30 ]
'DC' [ 8 30 ]
'DC' [ 8 30 ]
'DC' [ 12 18 ]
'DC' [ 12 18 ]
'AC' [ 0 ]
'AC' [ 0 ]
'AC' [ 0 ]
}
TYPE
'H4A1-30'
'H4A1-45'
'H4A1-70'
'H5A1-30'
'H5A1-45'
'H5A1-60'
'H5A1-90'
'N2'
'N3'
'N4'
'NG2U'
'NG3U'
'NG4U'
'L-196'
'LSRP-90'
'L-210'
'L-211U'
'L-212T'
}
sec
{
0.30
0.30
0.30
0.30
0.30
0.30
0.30
0.30
0.30
0.30
0.30
0.30
0.30
0
0
0
0
0
}
Low
HZ
-30.00
-45.00
-70.00
-30.00
-45.00
-60.00
-90.00
-10.00
-30.00
-70.00
-10.00
-30.00
-80.00
-90.00
-80.00
-60.00
-45.00
0
0
0
}
Deg
Deg
30.00
45.00
70.00
30.00
45.00
60.00
90.00
10.00
30.00
70.00
10.00
30.00
80.00
90.00
80.00
60.00
45.00
0
0
0
}
High
High
30.00
45.00
70.00
30.00
45.00
60.00
90.00
10.00
30.00
70.00
10.00
30.00
80.00
90.00
80.00
60.00
45.00
0
0
0
}
N/A
'Analog 0-5V'
'Analog 0-5V'
'Analog 0-5V'
'Analog 0-5V, 0-10V'
'Analog 0-5V, 0-10V'
'Analog 0-5V, 0-10V'
'Analog 0-5V, 0-10V'
'Analog mV'
'Analog mV'
'Analog mV'
'Analog 0-5V'
'Analog 0-5V'
'Analog 0-5V'
'Analog 0-5V'
'Analog mV'
'Analog mV'
'Analog mV'
0
0
0
}
deg
deg
0.02
0.03
0.04
0.02
0.03
0.03
0.05
0.002
0.005
0.010
0.001
0.003
0.010
0.00017
0.0000573
0.0083
0.0083
0.0056
}
mV/deg
mV/deg
67.70
44.40
28.60
67.70
44.40
33.30
22.20
12.00
5.75
3.65
200.00
67.00
25.00
0
0
100.00
100.00
150.00
}
%
0.50
0.50
0.20
0.50
0.50
0.20
0.20
0.50
0.20
0.20
0.10
0.10
0.10
0.10
0.05
1.25
3.33
2.22
}

```

```

IncUnitLabelStr1 = { 'N/A' 'Deg' 'Deg' 'N/A' 'Deg' 'mV/deg' '%' 'N/A' 'V' 'sec' 'Hz' };

```

```

%% INCLINOMETER Sensor Data Base Created Based on un Unit System
%% 2 - D Matrix INCLINOMETERSENSOR_DB(IS_Type, Property)

```

```

if strcmpi(UNITS, 'SI') == 1
% Properties
% Units
INCLINOMETERSENSOR_DB2 = {
    OPERATING TEMP
    Low deg C High
    -40.0 85.0
    -40.0 85.0
    -40.0 85.0
    STORAGE TEMP
    Low deg C High
    -45.0 90.0
    -45.0 90.0
    -45.0 90.0
    DIMENSIONS
    Length Width Height
    66.0 66.0 28.0
    66.0 66.0 28.0
    66.0 66.0 28.0
    WEIGHT
    gram
    227
    227
    227
    212.6
}

```



```

'EMC protected' ;
'Mechanical Zero adjustment' };
'Single Axis Angle Measurement' ;
'2 Levels of Temperature Compensation' ;
'Analog Voltage 0-5V Output' ;
'Serial Digital RS232 Output' ;
'Vibration and shock resistant' ;
'Environmentally sealed to IP66' ;
'Rugged die-cast Zinc housing' ;
'EMC protected' ;
'Mechanical Zero adjustment' };
'Single Axis Angle Measurement' ;
'2 Levels of Temperature Compensation' ;
'Analog Voltage 0-5V Output' ;
'Serial Digital RS232 Output' ;
'Vibration and shock resistant' ;
'Environmentally sealed to IP66' ;
'Rugged die-cast Zinc housing' ;
'EMC protected' ;
'Mechanical Zero adjustment' };
'Single Axis Angle Measurement' ;
'Temperature Compensation' ;
'2 Analog Outputs: 0-5V, 0-10V' ;
'Digital Serial RS232 Output' ;
'Vibration and Shock Resistant' ;
'Environmentally sealed to IP66' ;
'Rugged Machined Aluminum Housing' ;
'EMC protected' ;
'Mechanical Zero adjustment' };
'Single Axis Angle Measurement' ;
'Temperature Compensation' ;
'2 Analog Outputs: 0-5V, 0-10V' ;
'Digital Serial RS232 Output' ;
'Vibration and Shock Resistant' ;
'Environmentally sealed to IP66' ;
'Rugged Machined Aluminum Housing' ;
'EMC protected' ;
'Mechanical Zero adjustment' };
'Single Axis Angle Measurement' ;
'Temperature Compensation' ;
'2 Analog Outputs: 0-5V, 0-10V' ;
'Digital Serial RS232 Output' ;
'Vibration and Shock Resistant' ;
'Environmentally sealed to IP66' ;
'Rugged Machined Aluminum Housing' ;
'EMC protected' ;
'Mechanical Zero adjustment' };
'Single Axis Angle Measurement' ;
'Temperature Compensation' ;
'2 Analog Outputs: 0-5V, 0-10V' ;
'Digital Serial RS232 Output' ;
'Vibration and Shock Resistant' ;
'Environmentally sealed to IP66' ;
'Rugged Machined Aluminum Housing' ;

```

```
INCLINOMETERSENSOR_DB3{2,1} = {
```

```
INCLINOMETERSENSOR_DB3{3,1} = {
```

```
INCLINOMETERSENSOR_DB3{4,1} = {
```

```
INCLINOMETERSENSOR_DB3{5,1} = {
```

```
INCLINOMETERSENSOR_DB3{6,1} = {
```

```
INCLINOMETERSENSOR_DB3{7,1} = {
```

```

'EMC protected' ;
'Mechanical zero adjustment' ;
'Linear output characteristics' ;
'Minimal zero offset drift' ;
'Hysteresis free output signal' ;
'High measurement accuracy' ;
'Very low relative linearity errors' ;
'Integrated sensor electronics' ;
'Long-term stability' ;
'Low power consumption' ;
'Analog mv output signal' ;
'Hermetically sealed to IP65' ;
'Zero offset mechanically adjustable through 360 within mounting ring' ;
'No interference by ambient electromagnetic fields' ;
'Shockproof to 10,000g - no moving mechanical parts' ;
'Sensor electrically isolated from point of measurement using high quality PBT plastic housing

- no ground connections' ;
INCLINOMETERSENSOR_DB3{9,1} = {
'Linear output characteristics' ;
'Minimal zero offset drift' ;
'Hysteresis free output signal' ;
'High measurement accuracy' ;
'Very low relative linearity errors' ;
'Integrated sensor electronics' ;
'Long-term stability' ;
'Low power consumption' ;
'Analog mv output signal' ;
'Hermetically sealed to IP65' ;
'Zero offset mechanically adjustable through 360 within mounting ring' ;
'No interference by ambient electromagnetic fields' ;
'Shockproof to 10,000g - no moving mechanical parts' ;
'Sensor electrically isolated from point of measurement using high quality PBT plastic housing

- no ground connections' ;
INCLINOMETERSENSOR_DB3{10,1} = {
'Linear output characteristics' ;
'Minimal zero offset drift' ;
'Hysteresis free output signal' ;
'High measurement accuracy' ;
'Very low relative linearity errors' ;
'Integrated sensor electronics' ;
'Long-term stability' ;
'Low power consumption' ;
'Analog mv output signal' ;
'Hermetically sealed to IP65' ;
'Zero offset mechanically adjustable through 360 within mounting ring' ;
'No interference by ambient electromagnetic fields' ;
'Shockproof to 10,000g - no moving mechanical parts' ;
'Sensor electrically isolated from point of measurement using high quality PBT plastic housing

- no ground connections' ;
INCLINOMETERSENSOR_DB3{11,1} = {
'Liquid capacitive based inclinometers' ;
'Analog DC Output' ;
'Single Supply Input' ;
'Low Power Consumption' ;
'IP65 Packaging' ;
'Shock Resistant' ;
'Hysteresis Free' ;

```

```

INCLINOMETERSENSOR_DB3{12,1} = {
'Liquid capacitive based inclinometers' ;
'Analog DC Output' ;
'Single Supply Input' ;
'Low Power Consumption' ;
'IP65 Packaging' ;
'Shock Resistant' ;
'Hysteresis Free' ;
'Applications - Heavy Construction, Grading, Ship & Barge Leveling, Deviation Surveys,
Continuous Casting, Weapons Platform Leveling' ;
};

INCLINOMETERSENSOR_DB3{13,1} = {
'Liquid capacitive based inclinometers' ;
'Analog DC Output' ;
'Single Supply Input' ;
'Low Power Consumption' ;
'IP65 Packaging' ;
'Shock Resistant' ;
'Hysteresis Free' ;
'Applications - Heavy Construction, Grading, Ship & Barge Leveling, Deviation Surveys,
Continuous Casting, Weapons Platform Leveling' ;
};

INCLINOMETERSENSOR_DB3{14,1} = {
'Biaxial inclinometer - two-axis tilt sensor' ;
'Excellent turn-on-repeatability' ;
'Very low hysteresis' ;
'Seals meet MIL-STD 202, Method 112 Requirements' ;
'Compact, cylindrical shape' ;
'Stacking Feature allows for several unit use for multi-axis measurement' ;
'Precise readings with high outputs at lower range frequencies' ;
'Applications - Heavy Construction, Grading, Ship & Barge Leveling, Deviation Surveys,
Continuous Casting, Weapons Platform Leveling' ;
};

INCLINOMETERSENSOR_DB3{15,1} = {
'Seals meet MIL-STD 202, Method 112 Requirements' ;
'Two Single Axis Sensors Mounted on Single Unit' ;
'Wide angular range, high accuracy, dynamic output attenuation' ;
'L Series incorporates a specially designed dampening orifice' ;
'Includes hermetic sealing, compact size, and are available in a variety of housing
configurations' ;
'Applications - Industrial, Aerospace, Military, Photonics, Geotechnical, Oceanographic,
Construction.' ;
};

INCLINOMETERSENSOR_DB3{17,1} = {
'Flight Rated Production Company' ;
'Two Single Axis Sensors Mounted on Single Unit' ;
'Wide angular range, high accuracy, dynamic output attenuation' ;
'L Series incorporates a specially designed dampening orifice' ;
'Flight Proven Product - Cassini-Huygens Mission Huygens Probe' ;
'Includes hermetic sealing, compact size, and are available in a variety of housing
configurations' ;
'Applications - Industrial, Aerospace, Military, Photonics, Geotechnical, Oceanographic,
Construction.' ;
};

INCLINOMETERSENSOR_DB3{18,1} = {
'Flight Rated Production Company' ;
'Two Single Axis Sensors Mounted on Single Unit' ;
'Wide angular range, high accuracy, dynamic output attenuation' ;
'L Series incorporates a specially designed dampening orifice' ;
'Includes hermetic sealing, compact size, and are available in a variety of housing
configurations' ;
'Applications - Industrial, Aerospace, Military, Photonics, Geotechnical, Oceanographic,
Construction.' ;
};

% Combine Data Arrays into a Single Data Matrix
INCLINOMETERSENSOR_DB = cat(2,INCLINOMETERSENSOR_DB1,INCLINOMETERSENSOR_DB2,INCLINOMETERSENSOR_DB3);
IncUnitLabelStr = cat(2,IncUnitLabelStr1,IncUnitLabelStr2);

```

```

% Select Inclination Sensor by Input Operation Range Value
c = 1;
for I = 1:size(INCLINOMETERSENSOR_DB,1)
    if (INCLINOMETERSENSOR_DB{I,3} <= INC_ANGLE) && (INCLINOMETERSENSOR_DB{I,3} >= TOI_LOWER)
        for s = 1:size(INCLINOMETERSENSOR_DB,2)
            INCL_MATRIX{c,s} = INCLINOMETERSENSOR_DB{I,s};
        end
        c = c + 1;
    end
end

% Continue Selection Process From Viable Options - If Multiple Sensor Options Exist
if size(INCL_MATRIX,1) > 1

    % Select Via Temperature Operational Range
    c = 1;
    % Build Up Database of Possible Inclinometer Sensors Based on Temperature Range
    for t = 1:size(INCL_MATRIX,1)
        if (PLTEMP >= INCL_MATRIX{t,12}) && (PLTEMP <= INCL_MATRIX{t,13})
            for a = 1:size(INCL_MATRIX,2)
                INCL_MATRIX2{c,a} = INCL_MATRIX{t,a};
            end
            c = c + 1;
        end
    end

% Copies INCL_MATRIX to INCL_MATRIX2 if all Sensors Eliminated based on Temperature Range
if exist('INCL_MATRIX2') == 0
    INCL_MATRIX2 = INCL_MATRIX;
end

if size(INCL_MATRIX2,1) > 1

    % Down Select by Max Temperature Range
    for d = 1:size(INCL_MATRIX2,1)
        TEMP_INCL_DELTA(d) = INCL_MATRIX2{d,13} - INCL_MATRIX2{d,12};
    end

    % Find the Maximum Inclinometer Temperature Range Sensor
    MAX_TEMP_INCL_DELTA = max(TEMP_INCL_DELTA);

    % For Multiple Sensors with the Same Range Reduce Sensor Matrix to Those
    % with the Greatest Range
    k = 1;
    for c = 1:length(TEMP_INCL_DELTA)
        if TEMP_INCL_DELTA(c) == MAX_TEMP_INCL_DELTA
            for j = 1:size(INCL_MATRIX2,2)
                INCL_MATRIX3{k,j} = INCL_MATRIX2{c,j};
            end
            k = k + 1;
        end
    end
end
end

```

```

end
% Copies INCL_MATRIX2 to INCL_MATRIX3 if all Sensors Eliminated based on Temperature Range
if exist('INCL_MATRIX3') == 0
    INCL_MATRIX3 = INCL_MATRIX2;
end
if size(INCL_MATRIX3,1) >1
    % Create Mass / Volume Scaling Factor Optimization
    for mv = 1:size(INCL_MATRIX3,1)
        INC_SCALE_FAC(mv) = INCL_MATRIX3{mv,10} * INCL_MATRIX3{mv,12} * INCL_MATRIX3{mv,13} * INCL_MATRIX3{mv,13};
        if (INCL_MATRIX3{mv,10} == 0) || (INCL_MATRIX3{mv,11} == 0) || (INCL_MATRIX3{mv,12} == 0) || (INCL_MATRIX3{mv,13}
            == 0)
            INC_SCALE_FAC(mv) = 9.99E99;
        end
    end
end
[ INC_MIN_SCL, INDEX ] = min(INC_SCALE_FAC);
% Build Sensor Matrix Based on Scale Factors
k = 1;
for c = 1:length(INC_SCALE_FAC)
    if INC_SCALE_FAC(c) == INC_MIN_SCL
        for j = 1:size(INCL_MATRIX3,2)
            INCL_MATRIX4{k,j} = INCL_MATRIX3{c,j};
        end
        k = k + 1;
    end
end
end
% Copies INCL_MATRIX3 to INCL_MATRIX4 if all Sensors Eliminated based on Mass Volume Scale Factor
if exist('INCL_MATRIX4') == 0
    INCL_MATRIX4 = INCL_MATRIX3;
end
if size(INCL_MATRIX4,1) >1
    k = 1;
    % Search for Flight Rated Components
    for c = 1:size(INCL_MATRIX4,1)
        for pos = 1:length(INCL_MATRIX4{c,20})
            if findstr(INCL_MATRIX4{c,20}{pos}, 'Flight Proven Product') == 1
                for loc = 1:size(INCL_MATRIX4,2)
                    INCL_MATRIX5{k,loc} = INCL_MATRIX4{c,loc};
                end
                k = k + 1;
            end
        end
    end
end
end
end

```

```

% Copies INCL_MATRIX4 to INCL_MATRIX5 if all Sensors Eliminated based on Temperature Range
if exist('INCL_MATRIX5') == 0
    INCL_MATRIX5 = INCL_MATRIX4;
end
end

% Save Density Sensor Properties to Data File
save('INCLINOMETER_SENSOR_Data.mat');

save('INCLINOMETER_SENSOR_Final.mat', 'INCL_MATRIX5', 'IncUnitLabelStr');

%*****
%
% *** PRINT SUMMARY Output Section for 'Print' = 'Y'

if strcmpi(Print,'Y') == 1
    fprintf('\n\n\n *** I N C L I N O M E T E R   S E N S O R   S U M M A R Y   R E S U L T S   ****');
    fprintf('\n\n\n *** I N P U T S ***');
    fprintf('\n Inclination Angle Range: %s', INCR{1});
    fprintf('\n Unit System: %s', UNITS);

    fprintf('\n\n\n*****\n\n*****\n\n*****\n\n*****\n\n');

    % Inclinometer Sensor Properties
    fprintf('\n\n\n** Sensor Properties **');
    fprintf('\n Inclinometer Sensor Type: %s', INCL_MATRIX5{1,1});
    fprintf('\n Sensing Inclination Range - Low: %10.3f %s', INCL_MATRIX5{1,2}, IncUnitLabelStr{2});
    fprintf('\n High: %10.3f %s', INCL_MATRIX5{1,3}, IncUnitLabelStr{3});
    fprintf('\n Output Data Format: %s', INCL_MATRIX5{1,4});
    fprintf('\n Sensor Resolution: %10.7f %s', INCL_MATRIX5{1,5}, IncUnitLabelStr{5});
    fprintf('\n Sensor Sensitivity: %10.3f %s', INCL_MATRIX5{1,6}, IncUnitLabelStr{6});
    fprintf('\n Sensor Non-Linearity: %10.4f %s', INCL_MATRIX5{1,7}, IncUnitLabelStr{7});
    fprintf('\n Response Time: %10.3f %s', INCL_MATRIX5{1,10}, IncUnitLabelStr{10});
    fprintf('\n Sensor Bandwidth: %10.2f %s', INCL_MATRIX5{1,11}, IncUnitLabelStr{11});

    % Environmental Properties
    fprintf('\n\n\n Environmental Properties');
    fprintf('\n Operating Temperature Range - Low: %9.3f %s', INCL_MATRIX5{1,12}, IncUnitLabelStr{12});
    fprintf('\n High: %9.3f %s', INCL_MATRIX5{1,13}, IncUnitLabelStr{13});
    fprintf('\n Storage Temperature Range - Low: %9.3f %s', INCL_MATRIX5{1,14}, IncUnitLabelStr{14});
    fprintf('\n High: %9.3f %s', INCL_MATRIX5{1,15}, IncUnitLabelStr{15});

    fprintf('\n\n\n Power Requirements');
    fprintf('\n Input Voltage Type: %s %s', INCL_MATRIX5{1,8}, IncUnitLabelStr{8});
    fprintf('\n Input Voltage: ');
    for i = 1:length(INCL_MATRIX5{1,9})
        fprintf('%2d', INCL_MATRIX5{1,9}(i));
    end
    fprintf('%s', IncUnitLabelStr{9});
end

```





```

fprintf(Fout, '\n\n Physical Properties');
fprintf(Fout, '\n Dimensions - Length: %10.3f %s', INCL_MATRIX5{1,16}, IncUnitLabelStr{16} );
fprintf(Fout, '\n Width: %10.3f %s', INCL_MATRIX5{1,17}, IncUnitLabelStr{17} );
fprintf(Fout, '\n Height: %10.3f %s', INCL_MATRIX5{1,18}, IncUnitLabelStr{18} );
fprintf(Fout, '\n Sensor Mass: %10.3f %s', INCL_MATRIX5{1,19}, IncUnitLabelStr{19} );

fprintf(Fout, '\n\n Sensor Comments:');
for r = 1:length(INCL_MATRIX5{1,20})
    fprintf(Fout, '\n %s', INCL_MATRIX5{1,20}{r} );
end

fclose(Fout);

```

### D13. Nephelometer Sensors

```

% function NEPHELOMETER_SENSORS(UNITS,Print)
% Nephelometer Sensor Database
%
% Developed by: Keith Schreck
% Mechanical and Aerospace Engineering
% San Jose State University
% Date: Fall 2007
%
% This file contains Nephelometer Sensor values for the ISSPO program.
% Data here is loaded into the main program to determine the proper sensor
% to use for Missions based on the sensor requirements.
% Values are assigned to a variable name and saved to a Matlab '.mat' file
% and loaded when calling the ISSPO program.
%
% Some of the values are based on the Unit system chosen and are loaded
% in separate sections.
%
% References
% http://www.tsi.com/Product.aspx?pid=64
% http://www.optecinc.com/optec_041.htm
function NEPHELOMETER_SENSORS(UNITS,Print)

% Load Constants values into Local Program
load('Constants_DB.mat');
load('Planet_Data.mat');

%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%
% Nephelometer Sensor Data Base Created Based on un Unit System

```

```

%% 2 - D Matrix NEPHELOMETER_DB(N_Type, Property)

if strcmpi(UNITS,'SI')==1
% Properties [ Sensor
Length Avg Time Drift Wavelength1 Wavelength2 Wavelength3 Bandwidth Weight Power Voltage Width Height
% Units [ # Angle Integration nm nm nm nm kg W V mm mm
mm sec per m @ 60 sec low deg high deg j 550 700 40 40 18 175 24 300 1100
NEPHELOMETER_DB = { 'TSI-3563' 450 550 700 40 40 18 175 24 300 1100
250 4096 2.000E-07 7 170 ; 730 100 12.247 62.5 13.8 419
271 600 0 'NGN-2a' 5 320 550 ; 0 200 12.247 62.5 13.8 419
271 810 0 'NGN-3a' 5 175 ; ; ; ; ; ;
% Properties [ Response Time Rec Flow Rate Operating Temperature Ambient Rel Humidity ]
% Units [ sec L/min L/min Low deg C High deg C Low % High % ]
NEPHELOMETER_DB2 = { 10 200 40 95 ;
4 2 2 -20 0 0 ;
4 7 9 -10 0 100 ;
};

NephUnitLabelStr1 = { 'sensor' 'nm' 'nm' 'nm' 'kg' 'W' 'V' 'mm' 'mm' 'mm' 'sec' 'per m @ 60 sec' 'deg' 'deg'
};
NephUnitLabelStr2 = { 'sec' 'L/min' 'L/min' 'deg C' 'deg C' '% RH' '% RH' };

% Required Power Supply Specifications
% Properties [ Weight Power Type Voltage Width Height Length ]
% Units [ kg W N/A V mm mm mm mm ]
NephPS_DB = { 18 175 'DC' 24 305 178 102 ;
0 62.5 'DC' 13.8 203 152 51 ;
0 62.5 'DC' 13.8 203 152 51 ;
};

NephPSLabelStr = { 'kg' 'W' 'N/A' 'V' 'mm' 'mm' 'mm' };

end

if strcmpi(UNITS,'BRITISH')==1
% Properties [ Sensor
Height Length Avg Time Drift Wavelength1 Wavelength2 Wavelength3 Bandwidth Weight Power Voltage Width
% Units [ # Angle Integration inch inch inch inch lb FTU's/hr V inch
inch inch sec per ft @ 60 sec low deg high deg inch deg 16.5 8.2 600 0 5 175 ; 0 2.1653E-05 2.8740E-05 3.937E-06 27 213.449 13.8 10.7
9.84 43.3 4096 6.096E-08 'TSI-3563' 1.7716E-05 2.1653E-05 2.7559E-05 1.574E-06 39.68 597.656 24 11.81
'NGN-2a' 1.2598E-05 2.1653E-05 2.8740E-05 3.937E-06 27 213.449 13.8 10.7
16.5 8.2 600 0 5 175 ; 0 2.1653E-05 2.8740E-06 7.874E-06 27 213.449 13.8 10.7
16.5 8.2 810 0 5 175 ; ; ; ; ; ;
% Properties [ Response Time Rec Flow Rate Operating Temperature Ambient Rel Humidity ]
% Units [ sec gal/min gal/min Low deg F High deg F Low % High % ]
NEPHELOMETER_DB2 = { 10 5.28344 52.8344 50 104 0 95 ;
4 0.528344 0.528344 -4 113 0 0 ;
4 1.8492 2.37755 14 113 0 100 ;
};

```

```

NephUnitLabelStr1 = { 'N/A' 'inch' 'inch' 'inch' 'inch' 'lb' 'BTUs/hr' 'V' 'inch' 'inch' 'inch' 'sec' 'per ft @ 60 sec'
'deg' 'deg' };
NephUnitLabelStr2 = { 'sec' 'L/min' 'L/min' 'deg F' 'deg F' '% RH' '% RH' };

% Required Power Supply Specifications
% Properties {
% Units {
NephPS_DB = {
0 597.656 'DC' 24 7 4 ;
0 213.5 'DC' 13.8 8 6 2 ;
0 213.5 'DC' 13.3 8 6 2 ;
}

NephPSLabelStr = { 'lb' 'BTU/hr' 'N/A' 'V' 'in' 'in' 'in' };

end

NEPHELOMETER_DB = cat(2,NEPHELOMETER_DB,NEPHELOMETER_DB2);
NephUnitLabelStr = cat(2,NephUnitLabelStr1,NephUnitLabelStr2);

%comments on sensors
NEPHELOMETER_DB{ 1,23} = 'High sensitivity';
NEPHELOMETER_DB{ 2,23} = 'Unit may be outdated, floppy drive data recovery';
NEPHELOMETER_DB{ 3,23} = 'Unit may be outdated, floppy drive data recovery';

% Find the Max Operational Range
for n = 1:size(NEPHELOMETER_DB,1)
N_DELTA(n) = NEPHELOMETER_DB{n,4} - NEPHELOMETER_DB{n,2};
end

% Find the Maximum Nephelometer Operating Range of the Sensor
MAX_N_DELTA = max(N_DELTA);

% For Multiple Sensors with the Same Range Reduce Sensor Matrix to Those
% with the Greatest Range
k = 1;
for c = 1:length(N_DELTA)
if N_DELTA(c) == MAX_N_DELTA
for j = 1:size(NEPHELOMETER_DB,2)
N_MATRIX{k,j} = NEPHELOMETER_DB{c,j};
end
k = k + 1;
end
end

if exist('N_MATRIX') == 0
N_MATRIX = NEPHELOMETER_DB;
end

% Select Sensor Based on Integration Range if More than One Sensor Exists In Database
if size(N_MATRIX,1) > 1
% Find the Max Operational Range
for n = 1:size(N_MATRIX,1)
N_DELTA_RNG(n) = N_MATRIX{n,15} - N_MATRIX{n,14};
end
end

```

```

% Find the Maximum Nephelometer Operating Range of the Sensor
MAX_N_DELTA_RNG = max(N_DELTA_RNG);

% For Multiple Sensors with the Same Range Reduce Sensor Matrix to Those
% with the Greatest Range
k = 1;
for c = 1:length(N_DELTA_RNG)
    if N_DELTA_RNG(c) == MAX_N_DELTA_RNG
        for j = 1:size(NEPHELOMETER_DB,2)
            N_MATRIX2{k,j} = N_MATRIX{c,j};
        end
        k = k + 1;
    end
end

if exist('N_MATRIX2') == 0
    N_MATRIX2 = N_MATRIX;
end

% Select Sensor Based on Minimum Volume if More than One Sensor Exists In Database
if size(N_MATRIX,1) > 1
    % Create Volume Scaling Factor Optimization
    for v = 1:size(N_MATRIX2,1)
        N_SCALE_FAC(v) = N_MATRIX2{v,9} * N_MATRIX2{v,10} * N_MATRIX2{v,11} * N_MATRIX2{v,6};
    end
    if (N_MATRIX2{v,9} == 0) || (N_MATRIX2{v,10} == 0) || (N_MATRIX2{v,11} == 0) || (N_MATRIX2{v,6} == 0)
        N_SCALE_FAC(v) = 9.99E99;
    end
end

{ N_MIN_SCL, INDEX } = min(N_SCALE_FAC);

% Build Sensor Matrix Based on Scale Factors
k = 1;
for c = 1:length(N_SCALE_FAC)
    if N_SCALE_FAC(c) == N_MIN_SCL
        for j = 1:size(N_MATRIX2,2)
            N_MATRIX3{k,j} = N_MATRIX2{c,j};
        end
        k = k + 1;
    end
end

% Copies N_MATRIX2 to N_MATRIX3 if all Sensors Eliminated
if exist('N_MATRIX3') == 0
    N_MATRIX3 = N_MATRIX2;
end

% Determine the Corresponding Power Supply Array

```

```

for n = 1:size(NEPHELOMETER_DB,1)
    if strcmpi(NEPHELOMETER_DB{n,1},N_MATRIX3{1,1}) == 1
        break;
    end
end

% Save Nephelometer Sensor Properties to Data File
save('NEPHELOMETER_SENSOR_Data.mat');

save('NEPHELOMETER_SENSOR_Final.mat', 'N_MATRIX3', 'NephUnitLabelStr');

%*****
%
% *** PRINT SUMMARY Output Section for 'Print' = 'Y'

if strcmpi('Print', 'Y') == 1

    fprintf('\n\n\n *** NEPHELOMETER SENSOR SUMMARY RESULTS ****');
    fprintf('\n\n\n***** I N P U T S *****');
    fprintf('\n Unit System:          %s', UNITS);

    fprintf('\n\n\n*****\n\n\n*****\n\n\n*****\n\n\n*****\n\n\n*****');

    % Nephelometer Sensor Properties
    fprintf('\n\n\n***** Sensor Properties ****');
    fprintf('\n Nephelometer Sensor Type:          %s', N_MATRIX3{1,1});
    fprintf('\n Sensing Wavelengths - Min:          %12.4E %s', N_MATRIX3{1,2}, NephUnitLabelStr{2});
    fprintf('\n                               Middle:          %12.4E %s', N_MATRIX3{1,3}, NephUnitLabelStr{3});
    fprintf('\n                               Max:              %12.4E %s', N_MATRIX3{1,4}, NephUnitLabelStr{4});
    fprintf('\n Sensing Bandwidth:                %12.4E %s', N_MATRIX3{1,5}, NephUnitLabelStr{5});
    fprintf('\n Sampling Time:                    %12.2f %s', N_MATRIX3{1,12}, NephUnitLabelStr{12});
    fprintf('\n Sensing Drift Rate:               %12.3E %s', N_MATRIX3{1,13}, NephUnitLabelStr{13});
    fprintf('\n Integration Angles - Min:         %12.2f %s', N_MATRIX3{1,14}, NephUnitLabelStr{14});
    fprintf('\n                               Max:              %12.2f %s', N_MATRIX3{1,15}, NephUnitLabelStr{15});
    fprintf('\n Response Time:                    %12.2f %s', N_MATRIX3{1,16}, NephUnitLabelStr{16});
    fprintf('\n Sampling Flow Rate - Min:         %12.2f %s', N_MATRIX3{1,17}, NephUnitLabelStr{17});
    fprintf('\n                               Max:              %12.2f %s', N_MATRIX3{1,18}, NephUnitLabelStr{18});

    fprintf('\n\n\n Power Requirements');
    fprintf('\n Power Source:          %12.3f %s', N_MATRIX3{1,7}, NephUnitLabelStr{7});
    fprintf('\n Input Voltage:        %12.3f %s', N_MATRIX3{1,8}, NephUnitLabelStr{8});

    fprintf('\n\n\n Physical Properties');
    fprintf('\n Sensor Mass:          %12.2f %s', N_MATRIX3{1,6}, NephUnitLabelStr{6});
    fprintf('\n Sensor Dimensions - Length: %12.2f %s', N_MATRIX3{1,11}, NephUnitLabelStr{11});
    fprintf('\n                               Width: %12.2f %s', N_MATRIX3{1,9}, NephUnitLabelStr{9});
    fprintf('\n                               Height: %12.2f %s', N_MATRIX3{1,10}, NephUnitLabelStr{10});

    % Environmental Properties

```



```

fprintf(Fout, '\n Input Voltage:      %12.3f      %s', N_MATRIX3{1,8}, NephUnitLabelStr{8} );

fprintf(Fout, '\n\n Physical Properties');
fprintf(Fout, '\n Sensor Mass:      %12.2f      %s', N_MATRIX3{1,6}, NephUnitLabelStr{6} );
fprintf(Fout, '\n Sensor Dimensions - Length: %12.2f      %s', N_MATRIX3{1,11}, NephUnitLabelStr{11} );
fprintf(Fout, '\n                               Width:  %12.2f      %s', N_MATRIX3{1,9}, NephUnitLabelStr{9} );
fprintf(Fout, '\n                               Height:  %12.2f      %s', N_MATRIX3{1,10}, NephUnitLabelStr{10} );

% Environmental Properties
fprintf(Fout, '\n\n Environmental Properties');
fprintf(Fout, '\n Temperature Range - Low:  %12.2f      %s', N_MATRIX3{1,19}, NephUnitLabelStr{19} );
fprintf(Fout, '\n                               High:    %12.2f      %s', N_MATRIX3{1,20}, NephUnitLabelStr{20} );
fprintf(Fout, '\n Relative Humidity - Min:    %12.2f      %s', N_MATRIX3{1,21}, NephUnitLabelStr{21} );
fprintf(Fout, '\n                               Max:      %12.2f      %s', N_MATRIX3{1,22}, NephUnitLabelStr{22} );

fprintf(Fout, '\n\n Sensor Comments:');
fprintf(Fout, '\n      %s', N_MATRIX3{1,23} );

% Power Supply Properties
fprintf(Fout, '\n\n Power Supply Properties');
fprintf(Fout, '\n Power Generated:      %12.3f      %s', NephPS_DB{n,2}, NephPSLabelStr{2} );
fprintf(Fout, '\n Supplied Voltage:      %12.2f      %s', NephPS_DB{n,4}, NephPSLabelStr{4} );
fprintf(Fout, '\n Voltage Type:          %s      %s', NephPS_DB{n,3}, NephPSLabelStr{3} );
fprintf(Fout, '\n Sensor Mass:          %12.2f      %s', NephPS_DB{n,1}, NephPSLabelStr{1} );
fprintf(Fout, '\n Sensor Dimensions - Length: %12.2f      %s', NephPS_DB{n,7}, NephPSLabelStr{7} );
fprintf(Fout, '\n                               Width:  %12.2f      %s', NephPS_DB{n,5}, NephPSLabelStr{5} );
fprintf(Fout, '\n                               Height:  %12.2f      %s', NephPS_DB{n,6}, NephPSLabelStr{6} );

fclose(Fout);

```

## D14. Pressure Sensors

```

% function PRESSURE_SENSORS(PLPRESS,UNITS,Print)
% Pressure Sensor Database
%
% Developed by: Keith Schreck
%               Mechanical and Aerospace Engineering
%               San Jose State University
% Date:        Fall 2007

```

```

% This file contains Pressure Sensor values for the ISSPO program.
% Data here is loaded into the main program to determine the proper sensor
% to use for Missions based on the planetary Atmosphere composition.
% Values are assigned to a variable name and saved to a Matlab '.mat' file
% and loaded when calling the ISSPO program.

```



```

% Some of the values are based on the Unit system chosen and are loaded
% in separate sections.
%
% References
% http://www.ascovalve.com/Applications/Products/SensorsPressureSensorsData.aspx
% http://sensing.honeywell.com/index.cfm?ci_id=140264&defID=121030
% http://www.omron.com/ecb/products/sensor/8.html
% http://www.geensing.com/products/pdcr3500.htm?bc=bc_drucks
% http://www.climatronics.com/Products/Sensors/atmospheric_pressure.php
% TO BE ADDED
% http://www.taberindustries.com/transducer/Aerospace.asp
% http://www.mksinst.com/product/catalog.aspx?catalogID=4
%
function PRESSURE_SENSORS(PLPRESS,UNITS,Print)

% Load Constants values into Local Program
load('Constants_DB.mat');
load('Planet_Data.mat');

% Assign Planetary Surface Properties to Local Variable
PLTEMP = Planet_Properties{7};

%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%
%% Pressure Sensor Data Base Created Based on un Unit System
%% 2 - D Matrix PRESSENSOR_DB(P_Type, Property)
%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%

if strcmpi(UNITS,'SI') == 1
%
ENVIRONMENTAL_DATA
% Properties
PROOF BURST BURST kPa PRESSENSOR_DB1 = {
% Units kPa 344.738 861.845 'Ser 48-0025',
% PRESSENSOR_DB1 = { 'Ser 48-0025',
% 100 0.011 20.0 0.25 2000 2.0 0.0 95.0 5.0 ; % 1 100E+06
% 'Ser 48-0050',
% 100 0.011 20.0 0.25 2000 2.0 0.0 95.0 5.0 ; % 2 100E+06
% 'Ser 48-0100',
% 100 0.011 20.0 0.25 2000 2.0 0.0 95.0 5.0 ; % 3 100E+06
% 'Ser 48-0200',
% 100 0.011 20.0 0.25 2000 2.0 0.0 95.0 5.0 ; % 4 100E+06
% 'Ser 48-1000',
% 100 0.011 20.0 0.25 2000 2.0 0.0 95.0 5.0 ; % 5 100E+06
% '19(C,U)-003P',
% 100 0.011 10.0 0.60 2000 3.0 0.0 0.0 0.0 ; % 6 1E+06
% '19(C,U)-005P',
% 100 0.011 10.0 0.60 2000 3.0 0.0 0.0 0.0 ; % 7 1E+06
% '19(C,U)-010P',
% 100 0.011 10.0 0.60 2000 3.0 0.0 0.0 0.0 ; % 8 1E+06
% 103.421 172.369
% 206.843 344.738
}

PERFORMANCE_CHARACTERISTICS @ 77
ACCURACY STABILITY OVER PRESSURE BURST PRESSURE LIFE CYCLE PRESSURE RANGE
VIB SHOCK FREQ @ 1 yr @ Mx Rated @ Mx Rated Rated # Low kPa High
% g's Low Hz High Low %RH High ]
0.5 0.25 20 2000 2.0 0.0 95.0 5.0 ; % 1 100E+06
20.0 20 2000 2.0 0.0 95.0 5.0 ; % 2 100E+06
0.5 0.25 20 2000 2.0 0.0 95.0 5.0 ; % 3 100E+06
0.5 0.25 20 2000 2.0 0.0 95.0 5.0 ; % 4 100E+06
0.5 0.25 20 2000 2.0 0.0 95.0 5.0 ; % 5 100E+06
2.0 20.0 0.60 2000 3.0 0.0 0.0 0.0 ; % 6 1E+06
2.0 10.0 0.60 2000 3.0 0.0 0.0 0.0 ; % 7 1E+06
2.0 10.0 0.60 2000 3.0 0.0 0.0 0.0 ; % 8 1E+06
}

}

```

310.264	517.107	'19(C,U)-015P'	2.0	0.60	3.0	0.0	5.0	1E+06	0.0	103.421
100	0.011	10.0	20	2000	0.0	0.0	0.0	% 9	0.0	
'19(C,U)-030P'	2.0	0.60	3.0	2000	3.0	5.0	1E+06	0.0	206.843	
100	0.011	10.0	20	2000	0.0	0.0	0.0	% 10	0.0	
'19(C,U)-050P'	2.0	0.60	3.0	2000	3.0	5.0	1E+06	0.0	344.738	
100	0.011	10.0	20	2000	0.0	0.0	0.0	% 11	0.0	
'19(C,U)-100P'	2.0	0.60	3.0	2000	3.0	5.0	1E+06	0.0	689.476	
100	0.011	10.0	20	2000	0.0	0.0	0.0	% 12	0.0	
'19(C,U)-200P'	2.0	0.60	3.0	2000	3.0	5.0	1E+06	0.0	1378.950	
100	0.011	10.0	20	2000	0.0	0.0	0.0	% 13	0.0	
'19(C,U)-300P'	2.0	0.60	3.0	2000	3.0	5.0	1E+06	0.0	2068.430	
100	0.011	10.0	20	2000	0.0	0.0	0.0	% 14	0.0	
'19(C,U)-500P'	2.0	0.60	2.4	2000	2.4	4.8	1E+06	0.0	3447.380	
100	0.011	10.0	20	2000	0.0	0.0	0.0	% 15	0.0	
'OMRON E8Y'	3.0	0.0	0.0	0.0	0.0	25.00	0	0.0	2.00	
30	0.0	10.0	10	150	25.0	85.0	% 16	0	0.0	
'OMRON E8Y'	3.0	0.0	0.0	150	0.0	10.00	0	0.0	5.00	
30	0.0	10.0	10	150	25.0	85.0	% 17	0	0.0	
'13(C,U)0500P'	2.0	0.60	3.0	2000	3.0	5.0	1E+06	0.0	3447.380	
100	0.011	10.0	20	2000	0.0	0.0	0.0	% 18	0.0	
'13(C,U)1000P'	2.0	0.60	3.0	2000	3.0	5.0	1E+06	0.0	6894.760	
100	0.011	10.0	20	2000	0.0	0.0	0.0	% 19	0.0	
'13(C,U)2000P'	2.0	0.60	3.0	2000	3.0	5.0	1E+06	0.0	13789.500	
100	0.011	10.0	20	2000	0.0	0.0	0.0	% 20	0.0	
'13(C,U)3000P'	2.0	0.60	3.0	2000	3.0	3.333	1E+06	0.0	20684.300	
100	0.011	10.0	20	2000	0.0	0.0	0.0	% 21	0.0	
'13(C,U)5000P'	2.0	0.60	2.0	2000	2.0	2.0	1E+06	0.0	34473.800	
100	0.011	10.0	20	2000	0.0	0.0	0.0	% 22	0.0	
'PDCR 3500'	0.1	0.30	0.0	0	2.0	6.0	0	5.0	6894.76	
0	0.0	0.0	0	0	0.0	0.0	0	0	0	
'PDCR 3500'	0.1	0.30	0.0	0	2.0	4.0	0	5.0	68947.60	
0	0.0	0.0	0	0	0.0	0.0	0	0.0	0	
'2403SAT'	0.20	0.10	0.10	0	2.0	3.0	0	0.0	34473.79	
30	0.011	25.0	0	0	0.0	0.0	0	0.0	0.0	

```

};
[
  % Properties
  DIMENSIONS (mm)
  % Units
  Length Width Height
  PRESSSENSOR_DB2 = {
    -40.0 104.44 -51.1 121.11 121.11 5.0 5.0 deg C ]
    0.0 -53.9 121.1 121.1 1.1 1.1 ]
    -40.0 104.44 -51.1 121.11 121.11 5.0 5.0 ]
    0.0 -53.9 121.1 121.1 1.1 1.1 ]
    -40.0 104.44 -51.1 121.11 121.11 5.0 5.0 ]
    0.0 -53.9 121.1 121.1 1.1 1.1 ]
    -40.0 104.44 -51.1 121.11 121.11 5.0 5.0 ]
    0.0 -53.9 121.1 121.1 1.1 1.1 ]
    -40.0 125.00 -40.0 125.00 0 5.0 2.0 ]
    0.0 ]
  }
  BANDWIDTH ]
  SENSOR ]
  MATERIAL ]
  CASE ]

```

```

85.47 19.05 19.05 -40.0 125.00 -40.0 125.00 5.0 2.0 0 'Stainless Steel' 'Stainless Steel'
0.0 125.00 -40.0 125.00 0 5.0 0 0 0 'Stainless Steel' 'Stainless Steel'
85.47 19.05 19.05 0.0 125.00 -40.0 125.00 5.0 2.0 0 'Stainless Steel' 'Stainless Steel'
0.0 125.00 -40.0 125.00 0 5.0 0 0 0 'Stainless Steel' 'Stainless Steel'
85.47 19.05 19.05 -40.0 125.00 -40.0 125.00 5.0 2.0 0 'Stainless Steel' 'Stainless Steel'
0.0 125.00 -40.0 125.00 0 5.0 0 0 0 'Stainless Steel' 'Stainless Steel'
85.47 19.05 19.05 0.0 125.00 -40.0 125.00 5.0 2.0 0 'Stainless Steel' 'Stainless Steel'
0.0 125.00 -40.0 125.00 0 5.0 0 0 0 'Stainless Steel' 'Stainless Steel'
85.47 19.05 19.05 -40.0 125.00 -40.0 125.00 5.0 2.0 0 'Stainless Steel' 'Stainless Steel'
0.0 125.00 -40.0 125.00 0 5.0 0 0 0 'Stainless Steel' 'Stainless Steel'
85.47 19.05 19.05 0.0 125.00 -40.0 125.00 5.0 2.0 0 'Stainless Steel' 'Stainless Steel'
0.0 125.00 -40.0 125.00 0 5.0 0 0 0 'Stainless Steel' 'Stainless Steel'
85.47 19.05 19.05 -40.0 125.00 -40.0 125.00 5.0 2.0 0 'Stainless Steel' 'Stainless Steel'
0.0 125.00 -40.0 125.00 0 5.0 0 0 0 'Stainless Steel' 'Stainless Steel'
85.47 19.05 19.05 0.0 125.00 -40.0 125.00 5.0 2.0 0 'Stainless Steel' 'Stainless Steel'
0.0 125.00 -40.0 125.00 0 5.0 0 0 0 'Stainless Steel' 'Stainless Steel'
31.00 30.00 30.00 79.38 -10.0 55.00 -25.0 30.0 50.0 0 'Resin pipe/Zinc' 'PBT'
79.38 -10.0 55.00 -25.0 30.0 50.0 0 0 0 'Resin pipe/Zinc' 'PBT'
81.03 19.05 19.05 -40.0 125.00 -40.0 125.00 15.0 2.0 0 'Stainless Steel' 'Stainless Steel'
0.0 125.00 -40.0 125.00 0 15.0 0 0 0 'Stainless Steel' 'Stainless Steel'
81.03 19.05 19.05 0.0 125.00 -40.0 125.00 15.0 2.0 0 'Stainless Steel' 'Stainless Steel'
0.0 125.00 -40.0 125.00 0 15.0 0 0 0 'Stainless Steel' 'Stainless Steel'
81.03 19.05 19.05 -40.0 125.00 -40.0 125.00 15.0 2.0 0 'Stainless Steel' 'Stainless Steel'
0.0 125.00 -40.0 125.00 0 15.0 0 0 0 'Stainless Steel' 'Stainless Steel'
81.03 19.05 19.05 0.0 125.00 -40.0 125.00 15.0 2.0 0 'Stainless Steel' 'Stainless Steel'
0.0 125.00 -40.0 125.00 0 15.0 0 0 0 'Stainless Steel' 'Stainless Steel'
92.71 25.40 25.40 -53.9 121.11 -53.9 125.00 5.0 0.0 0 'Stainless Steel' 'Stainless Steel'
-53.9 121.11 -53.9 125.00 15.0 2.0 0 0 0 'Stainless Steel' 'Stainless Steel'
102.87 25.40 25.40 -54.0 121.00 -54.0 121.00 36.0 0.0 0 'Stainless Steel' 'Stainless Steel'
88.90 31.75 31.75 170.0 0 0 0 0 0 0 'Stainless Steel' 'Stainless Steel'
};
PressUnitLabelStr1 = { 'Type' '%', '% @ 1 yr' 'N/A' 'N/A' 'Cycles' 'kPa' 'kPa' 'kPa' 'g's' 'sec'
'g's' 'Hz' 'Hz' '% RH' '% RH' };
PressUnitLabelStr2 = { 'deg C' 'deg C' 'deg C' 'V' 'mA' 'dB' 'N/A' 'mm' 'mm' 'gr'
'deg C' 'deg C' 'deg C' };
end
if strcmpl(UNITS,'British') == 1
%
ENVIRONMENTAL DATA
% Properties
PROOF BURST SHOCK IMPULSE SHOCK IMPULSE VIB SHOCK FREQUENCY RANGE HUMIDITY RANGE BURST PRESSURE LIFE CYCLE PRESSURE RANGE
PERFORMANCE CHARACTERISTICS @ 77
}

```

% psig	Units psig	[ g's	TYPE sec	g's	%	Low Hz	% @ 1 yr High	@ Mx Rated Low %RH	High	@ Mx Rated %RH	Rated #	Low psig	High
50.0	125.0	100	'Ser 48-0025'	20.0	0.5	20	0.25	2.0	95.0	5.0	100E+06	0.0	25.0
100.0	250.0	100	'Ser 48-0050'	20.0	0.5	20	0.25	2.0	95.0	5.0	100E+06	0.0	50.0
200.0	500.0	100	'Ser 48-0100'	20.0	0.5	20	0.25	2.0	95.0	5.0	100E+06	0.0	100.0
400.0	1000.0	100	'Ser 48-0200'	20.0	0.5	20	0.25	2.0	95.0	5.0	100E+06	0.0	200.0
2000.0	4000.0	100	'Ser 48-1000'	20.0	0.5	20	0.25	2.0	95.0	5.0	100E+06	0.0	1000.0
9.0	15.0	100	'19(C,U)-003P'	10.0	2.0	20	0.60	3.0	0.0	5.0	1E+06	0.0	3.0
15.0	25.0	100	'19(C,U)-005P'	10.0	2.0	20	0.60	3.0	0.0	5.0	1E+06	0.0	5.0
30.0	50.0	100	'19(C,U)-010P'	10.0	2.0	20	0.60	3.0	0.0	5.0	1E+06	0.0	10.0
45.0	75.0	100	'19(C,U)-015P'	10.0	2.0	20	0.60	3.0	0.0	5.0	1E+06	0.0	15.0
90.0	150.0	100	'19(C,U)-030P'	10.0	2.0	20	0.60	3.0	0.0	5.0	1E+06	0.0	30.0
150.0	250.0	100	'19(C,U)-050P'	10.0	2.0	20	0.60	3.0	0.0	5.0	1E+06	0.0	50.0
300.0	500.0	100	'19(C,U)-100P'	10.0	2.0	20	0.60	3.0	0.0	5.0	1E+06	0.0	100.0
600.0	1000.0	100	'19(C,U)-200P'	10.0	2.0	20	0.60	3.0	0.0	5.0	1E+06	0.0	200.0
900.0	1500.0	100	'19(C,U)-300P'	10.0	2.0	20	0.60	3.0	0.0	5.0	1E+06	0.0	300.0
1200.0	2400.0	100	'19(C,U)-500P'	10.0	2.0	20	0.60	2.4	0.0	4.8	1E+06	0.0	500.0
0.0	7.25	30	'OMRON E8Y'	10.0	3.0	10	0.0	0.0	0.0	25.14	0	0.0	0.29
0.0	7.25	30	'OMRON E8Y'	10.0	3.0	10	0.0	0.0	85.0	9.93	0	0.0	0.73
1500.0	2500.0	100	'13(C,U)0500P'	10.0	2.0	20	0.60	3.0	0.0	5.0	1E+06	0.0	500.0
3000.0	5000.0	100	'13(C,U)1000P'	10.0	2.0	20	0.60	3.0	0.0	5.0	1E+06	0.0	1000.0
6000.0	10000.0	100	'13(C,U)2000P'	10.0	2.0	20	0.60	3.0	0.0	5.0	1E+06	0.0	2000.0
9000.0	10000.0	100	'13(C,U)3000P'	10.0	2.0	20	0.60	3.0	0.0	3.333	1E+06	0.0	3000.0
10000.0	10000.0	100	'13(C,U)5000P'	10.0	2.0	20	0.60	2.0	0.0	2.0	1E+06	0.0	5000.0
2000.0	3000.0	0	'PDCR 3500'	0.0	0.1	0	0.30	2.0	0.0	6.0	0	5.0	1000.0
10000.0	10000.0	0	'PDCR 3500'	0.0	0.1	0	0.30	2.0	0.0	4.0	0	5.0	10000.0
10000.0	15000.0	30	'2403SAT'	0.20	25.0	0	0.10	2.0	0.0	3.0	0	0.0	5000.0





```

PRESSESENSOR_DB{21,34} = 'Low cost. Rugged Isolated Stainless Steel Design. Calibrated and temperature compensated. Oil-
isolated housing. Measurement in hostile environments.';
PRESSESENSOR_DB{22,34} = 'Low cost. Rugged Isolated Stainless Steel Design. Calibrated and temperature compensated. Oil-
isolated housing. Measurement in hostile environments.';
PRESSESENSOR_DB{23,34} = 'Digitally enhanced, high accuracy output, coupled with high analog bandwidth.';
PRESSESENSOR_DB{24,34} = 'Digitally enhanced, high accuracy output, coupled with high analog bandwidth.';
PRESSESENSOR_DB{25,34} = 'TABER Industries. Flight Proven. Model extends 2403 design. Flown on ChinaStar, Clementine, EOS,
Landsat, Wakeshield, Seastar, Indostar, Lockheed Astro A2100, 7000, NASA-JPL Deep Space One mission.';

% Build Up Database of Sensors Based On Operational Range
c = 1;
for p = 1:size(PRESSESENSOR_DB,1)
    if (PLPRESS <= (PRESSESENSOR_DB{p,8})*Press_Cnvt))
        for a = 1:size(PRESSESENSOR_DB,2)
            P_MATRIX{c,a} = PRESSESENSOR_DB{p,a};
        end
        c = c+1;
    end
end

if exist('P_MATRIX') == 0
    fprintf('\n\n *** WARNING ***');
    fprintf('\n\n NO VIABLE PRESSURE SENSOR EXISTS IN DATABASE!');
    fprintf('\n\n Other methods to determine Planetary Pressure must be used.');
```

```

for t = 1:size(P_MATRIX2,1)
    if (P_PRESS2 <= P_MATRIX2{t,8})
        for a = 1:size(P_MATRIX2,2)
            P_MATRIX3{c,a} = P_MATRIX2{t,a};
        end
        c = c+1;
    end
end

end

% Copies P_MATRIX2 to P_MATRIX3 if all Sensors Eliminated
if exist('P_MATRIX3') == 0
    P_MATRIX3 = P_MATRIX2;
end

if size(P_MATRIX3,1) > 1
    for c = 1:size(P_MATRIX3,1)
        Min_Des(c) = P_MATRIX3{c,8};
    end

    Min_Sens = min(Min_Des);
    k = 1;
    for c = 1:size(P_MATRIX3,1)
        if P_MATRIX3{c,8} == Min_Sens
            for j = 1:size(P_MATRIX3,2)
                P_MATRIX4{k,j} = P_MATRIX3{c,j};
            end
            k = k + 1;
        end
    end

end

% Copies P_MATRIX3 to P_MATRIX4 if all Sensors Eliminated
if exist('P_MATRIX4') == 0
    P_MATRIX4 = P_MATRIX3;
end

if size(P_MATRIX4,1) > 1
    % Create Mass Volume Scaling Factor Optimization
    for v = 1:size(P_MATRIX4,1)
        SCALE_FAC(v) = P_MATRIX4{v,27} * P_MATRIX4{v,28} * P_MATRIX4{v,29} * P_MATRIX4{v,30};
    end
    if (P_MATRIX4{v,27} == 0) || (P_MATRIX4{v,28} == 0) || (P_MATRIX4{v,29} == 0) || (P_MATRIX4{v,30} == 0)
        SCALE_FAC(v) = 9.99E99;
    end
end

[ MIN_SCL, INDEX ] = min(SCALE_FAC);
% Build Sensor Matrix Based on Scale Factors
k = 1;

```



```

for c = 1:length(SCALE_FAC)
    if SCALE_FAC(c) == MIN_SCI
        for j = 1:size(P_MATRIX4,2)
            P_MATRIX5{k,j} = P_MATRIX4{c,j};
        end
        k = k + 1;
    end
end
end

% Copies P_MATRIX4 to P_MATRIX5 if all Sensors Eliminated
if exist('P_MATRIX5') == 0
    P_MATRIX5 = P_MATRIX4;
end

% Down Select a Single Sensor if Multiple Options Exist
if size(P_MATRIX5,1) > 1
    % Create Variable Array of Sensor Accuracy
    for a = 1:size(P_MATRIX5,1)
        ACCURACY(a) = P_MATRIX5{a,2};
    end

    % Find the Sensors with the Tightest Accuracy Tolerances
    MIN_ACC = min(ACCURACY);

    % Build Sensor Matrix Based on Min Accuracy
    k = 1;
    for c = 1:size(P_MATRIX5,1)
        if P_MATRIX5{c,2} == MIN_ACC
            for j = 1:size(P_MATRIX5,2)
                P_MATRIX6{k,j} = P_MATRIX5{c,j};
            end
            k = k + 1;
        end
    end
end

% Copies P_MATRIX5 to P_MATRIX6 if all Sensors Eliminated
if exist('P_MATRIX6') == 0
    P_MATRIX6 = P_MATRIX5;
end

%
%
% Add Seasonal tolerances to Planetary Atmosphere and select Lowest Successful Sensor
PLPRESS2 = PLPRESS + 0.25*PLPRESS;
c = 1;
for t = 1:size(P_MATRIX4,1)
    if (PLPRESS2 <= P_MATRIX4{t,8})
        for a = 1:size(P_MATRIX,2)
            P_MATRIX5{c,a} = P_MATRIX4{t,a};
        end
    end
end

```







```

fprintf(Fout, '\n Input Voltage: %s', P_MATRIX6{1,22}, PressUnitLabelStr{22} );
fprintf(Fout, '\n Input Amperage: %s', P_MATRIX6{1,23}, PressUnitLabelStr{23} );
fprintf(Fout, '\n Sensing Bandwidth: %s', P_MATRIX6{1,24}, PressUnitLabelStr{24} );

fprintf(Fout, '\n\n Physical Properties');
fprintf(Fout, '\n Sensor Material: %s', P_MATRIX6{1,25});
fprintf(Fout, '\n Case Material: %s', P_MATRIX6{1,26});
fprintf(Fout, '\n Dimensions - Length: %10.3f %s', P_MATRIX6{1,27}, PressUnitLabelStr{27} );
fprintf(Fout, '\n Width: %10.3f %s', P_MATRIX6{1,28}, PressUnitLabelStr{28} );
fprintf(Fout, '\n Height: %10.3f %s', P_MATRIX6{1,29}, PressUnitLabelStr{29} );
fprintf(Fout, '\n Sensor Mass: %10.3f %s', P_MATRIX6{1,30}, PressUnitLabelStr{30} );

% Temperature Sensing Range
fprintf(Fout, '\n\n Temperature Sensing Range');
fprintf(Fout, '\n Lower Sensing Limit: %10.3f %s', P_MATRIX6{1,31}, PressUnitLabelStr{31} );
fprintf(Fout, '\n Upper Sensing Limit: %10.3f %s', P_MATRIX6{1,32}, PressUnitLabelStr{32} );
fprintf(Fout, '\n Accuracy: %10.3f %s', P_MATRIX6{1,33}, PressUnitLabelStr{33} );

fprintf(Fout, '\n\n Sensor Comments:');
fprintf(Fout, '\n %s', P_MATRIX6{1,34});

fclose(Fout);

```

## D15. Radiation Sensors

```

% function RADIATION_SENSORS(RAD_TYPE,UNITS,Print)
% Radiation Sensor Database
%
% Developed by: Keith Schreck
% Mechanical and Aerospace Engineering
% San Jose State University
% Date: Fall 2007
%
% This file contains Radiation Sensor values for the ISSPO program.
% Data here is loaded into the main program to determine the proper sensor
% to use for Missions based on the sensor requirements.
% Values are assigned to a variable name and saved to a Matlab '.mat' file
% and loaded when calling the ISSPO program.
%
% Some of the values are based on the Unit system chosen and are loaded
% in separate sections.
%
% References
% http://www.psicorp.com/products/lpd-detector.shtml
% http://www.psicorp.com/products/lpd-spectrometer.shtml
% http://www.blackcatsystems.com/GM/pages.html
% http://www.seintl.com/products/monitor_4.html
% http://www.pnwx.com/Equipment/Test/SurveyMeters/ASM-990/
% http://www.evproducts.com/pdf/capture_plus.pdf

```









```

133Xe, 40K, 226Ra, 232Th, 238U, ' ;
'Identifies: 233U, 235U, 237Np, Pu, 67Ga, 51Cr, 75Se, 99mTc, 103Pd, 111In, 123I, 131I, 201Tl,
, ;
57Co, 60Co, 133Ba, 137Cs, 192Ir, 204Tl, 226Ra, 241Am'
};

```

```

RAD_MATRIX = RADIATIONSENSOR_DB;

```

```

% Down Select Sensors Based on Radiation Types

```

```

for loc = 1:length(RAD_TYPE)
    STR1 = RAD_TYPE{loc};
    if strcmpi(STR1, 'Charged Particle') == 1
        C = 1;
        FLAG = 0;
        for r = 1:size(RAD_MATRIX,1)
            for pos = 1:length(RAD_MATRIX{r,2})
                STR2 = RAD_MATRIX{r,2}{pos};
                if strcmpi(STR1,STR2) == 1
                    for s = 1:size(RAD_MATRIX,2)
                        RAD_MATRIX_TMP{c,s} = RAD_MATRIX{r,s};
                    end
                end
                FLAG = 1;
            end
        end
    if FLAG == 1
        C = C + 1;
    end
    FLAG = 0;
end

if exist('RAD_MATRIX_TMP') == 0
    if strcmpi(Print,'Y') == 1
        fprintf('\n\n *** ERROR: NULL DATABASE ***');
        fprintf('\n\n No Sensor Exist in Database which Meet Radiation Requirements!');
        fprintf('\n\n Radiation Type: %s', STR1);
        RAD_MATRIX_TMP = RAD_MATRIX;
    end
end

clear RAD_MATRIX;
RAD_MATRIX = RAD_MATRIX_TMP;
clear RAD_MATRIX_TMP;

end

if strcmpi(STR1, 'Alpha') == 1
    C = 1;
    FLAG = 0;
    for r = 1:size(RAD_MATRIX,1)
        for pos = 1:length(RAD_MATRIX{r,2})
            STR2 = RAD_MATRIX{r,2}{pos};
            if strcmpi(STR1,STR2) == 1
                for s = 1:size(RAD_MATRIX,2)
                    RAD_MATRIX_TMP{c,s} = RAD_MATRIX{r,s};
                end
            end
        end
    end
end

```

```

FLAG = 1;
end
end
if FLAG == 1
    C = C + 1;
end
FLAG = 0;
end

if exist('RAD_MATRIX_TMP') == 0
    if strcmpi(Print,'Y') == 1
        fprintf('\n\n *** ERROR: NULL DATABASE ***');
        fprintf('\n\n No Sensor Exist in Database which Meet Radiation Requirements!');
        fprintf('\n Image Type: %s', STR1);
        RAD_MATRIX_TMP = RAD_MATRIX;
    end
end

clear RAD_MATRIX;
RAD_MATRIX = RAD_MATRIX_TMP;
clear RAD_MATRIX_TMP;

end

if strcmpi(STR1, 'Beta') == 1
    C = 1;
    FLAG = 0;
    for r = 1:size(RAD_MATRIX,1)
        for pos = 1:length(RAD_MATRIX{r,2})
            STR2 = RAD_MATRIX{r,2}{pos};
            if strcmpi(STR1,STR2) == 1
                for s = 1:size(v,2)
                    RAD_MATRIX_TMP{c,s} = RAD_MATRIX{r,s};
                end
            end
            FLAG = 1;
        end
    end
    if FLAG == 1
        C = C + 1;
    end
end
FLAG = 0;
end

if exist('RAD_MATRIX_TMP') == 0
    if strcmpi(Print,'Y') == 1
        fprintf('\n\n *** ERROR: NULL DATABASE ***');
        fprintf('\n\n No Sensor Exist in Database which Meet Radiation Requirements!');
        fprintf('\n Image Type: %s', STR1);
        RAD_MATRIX_TMP = RAD_MATRIX;
    end
end

clear RAD_MATRIX;
RAD_MATRIX = RAD_MATRIX_TMP;
clear RAD_MATRIX_TMP;

```

```

end
if strcmpi(STR1, 'Gamma') == 1
    C = 1;
    FLAG = 0;
    for r = 1:size(RAD_MATRIX,1)
        for pos = 1:length(RAD_MATRIX{r,2})
            STR2 = RAD_MATRIX{r,2}{pos};
            if strcmpi(STR1,STR2) == 1
                for s = 1:size(RAD_MATRIX,2)
                    RAD_MATRIX_TMP{c,s} = RAD_MATRIX{r,s};
                end
            end
            FLAG = 1;
        end
    end
    if FLAG == 1
        C = C + 1;
    end
    FLAG = 0;
end
if exist('RAD_MATRIX_TMP') == 0
    if strcmpi(print,'Y') == 1
        fprintf('\n\n *** ERROR: NULL DATABASE ***');
        fprintf('\n\n No Sensor Exist in Database which Meet Radiation Requirements!');
        fprintf('\n\n Image Type: %s', STR1);
        RAD_MATRIX_TMP = RAD_MATRIX;
    end
end
clear RAD_MATRIX;
RAD_MATRIX = RAD_MATRIX_TMP;
clear RAD_MATRIX_TMP;
end
if strcmpi(STR1, 'X-Ray') == 1
    C = 1;
    FLAG = 0;
    for r = 1:size(RAD_MATRIX,1)
        for pos = 1:length(RAD_MATRIX{r,2})
            STR2 = RAD_MATRIX{r,2}{pos};
            if strcmpi(STR1,STR2) == 1
                for s = 1:size(RAD_MATRIX,2)
                    RAD_MATRIX_TMP{c,s} = RAD_MATRIX{r,s};
                end
            end
            FLAG = 1;
        end
    end
    if FLAG == 1
        C = C + 1;
    end
    FLAG = 0;
end

```



```

end
if strcmpi(RAD_TYPE{rad},'Beta')==1
% Add up Alpha Particle Energy Range
for b = 1:size(RAD_MATRIX,1)
SUM_RAD_ARRAY(b) = SUM_RAD_ARRAY(b) + RAD_MATRIX{b,16} - RAD_MATRIX{b,15};
end
end
if strcmpi(RAD_TYPE{rad},'Gamma')==1
% Add up Alpha Particle Energy Range
for g = 1:size(RAD_MATRIX,1)
SUM_RAD_ARRAY(g) = SUM_RAD_ARRAY(g) + RAD_MATRIX{g,19} - RAD_MATRIX{g,18};
end
end
if strcmpi(RAD_TYPE{rad},'X-Ray')==1
% Add up Alpha Particle Energy Range
for a = 1:size(RAD_MATRIX,1)
SUM_RAD_ARRAY(a) = SUM_RAD_ARRAY(a) + RAD_MATRIX{a,22} - RAD_MATRIX{a,21};
end
end
end
end
% Find the Maximum Operational Range sensor
[ RAD_MAX_SCL, INDEX ] = max(SUM_RAD_ARRAY);
% For Multiple Sensors with the Same Range Reduce Sensor Matrix to Those
% with the Greatest Range
pos = 1;
for r = 1:length(SUM_RAD_ARRAY)
if SUM_RAD_ARRAY(r) == RAD_MAX_SCL
for j = 1:size(RAD_MATRIX,2)
RAD_MATRIX2{pos,j} = RAD_MATRIX{r,j};
end
pos = pos + 1;
end
end
end
if exist('RAD_MATRIX2')==0
RAD_MATRIX2 = RAD_MATRIX;
end
if size(RAD_MATRIX2,1) > 1
% Create Volume Scaling Factor Optimization
for v = 1:size(RAD_MATRIX2,1)
RAD_SCALE_FAC(v) = RAD_MATRIX2{v,27} * RAD_MATRIX2{v,28} * RAD_MATRIX2{v,29};
if (RAD_MATRIX2{v,27} == 0) || (RAD_MATRIX2{v,28} == 0) || (RAD_MATRIX2{v,29} == 0)

```





```

fprintf(Fout, '\n Radiation Detected:
');
for r = 1:length(RAD_MATRIX3{1,2}){r}
    fprintf(Fout, ' %s', RAD_MATRIX3{1,2}{r} );
end
fprintf(Fout, '\n Proton Energy Range - Low: %11.3f %s', RAD_MATRIX3{1,3}, RadUnitLabelStr{3} );
fprintf(Fout, '\n High: %11.3f %s', RAD_MATRIX3{1,4}, RadUnitLabelStr{4} );
fprintf(Fout, '\n Number of Detection Bins: %11d %s', RAD_MATRIX3{1,5}, RadUnitLabelStr{5} );
fprintf(Fout, '\n Electron Energy Range - Low: %11.3f %s', RAD_MATRIX3{1,6}, RadUnitLabelStr{6} );
fprintf(Fout, '\n High: %11.3f %s', RAD_MATRIX3{1,7}, RadUnitLabelStr{7} );
fprintf(Fout, '\n Number of Detection Bins: %11d %s', RAD_MATRIX3{1,8}, RadUnitLabelStr{8} );
fprintf(Fout, '\n Alpha Particle Range - Low: %11.3f %s', RAD_MATRIX3{1,9}, RadUnitLabelStr{9} );
fprintf(Fout, '\n High: %11.3f %s', RAD_MATRIX3{1,10}, RadUnitLabelStr{10} );
fprintf(Fout, '\n Number of Detection Bins: %11d %s', RAD_MATRIX3{1,11}, RadUnitLabelStr{11} );
fprintf(Fout, '\n Nucleon/Heavy Ion Range - Low: %11.3f %s', RAD_MATRIX3{1,12}, RadUnitLabelStr{12} );
fprintf(Fout, '\n High: %11.3f %s', RAD_MATRIX3{1,13}, RadUnitLabelStr{13} );
fprintf(Fout, '\n Number of Detection Bins: %11d %s', RAD_MATRIX3{1,14}, RadUnitLabelStr{14} );
fprintf(Fout, '\n Beta Particle Range - Low: %11.3f %s', RAD_MATRIX3{1,15}, RadUnitLabelStr{15} );
fprintf(Fout, '\n High: %11.3f %s', RAD_MATRIX3{1,16}, RadUnitLabelStr{16} );
fprintf(Fout, '\n Number of Detection Bins: %11d %s', RAD_MATRIX3{1,17}, RadUnitLabelStr{17} );
fprintf(Fout, '\n Gamma Particle Range - Low: %11.3f %s', RAD_MATRIX3{1,18}, RadUnitLabelStr{18} );
fprintf(Fout, '\n High: %11.3f %s', RAD_MATRIX3{1,19}, RadUnitLabelStr{19} );
fprintf(Fout, '\n Number of Detection Bins: %11d %s', RAD_MATRIX3{1,20}, RadUnitLabelStr{20} );
fprintf(Fout, '\n X-Ray Energy Range - Low: %11.3f %s', RAD_MATRIX3{1,21}, RadUnitLabelStr{21} );
fprintf(Fout, '\n High: %11.3f %s', RAD_MATRIX3{1,22}, RadUnitLabelStr{22} );
fprintf(Fout, '\n Number of Detection Bins: %11d %s', RAD_MATRIX3{1,23}, RadUnitLabelStr{23} );
fprintf(Fout, '\n G-Factor: %11.3f %s', RAD_MATRIX3{1,24}, RadUnitLabelStr{24} );
for g = 1:length(RAD_MATRIX3{1,25})
    fprintf(Fout, ' %s', RAD_MATRIX3{1,25}{g} );
end
fprintf(Fout, ' %s', RadUnitLabelStr{25} );

fprintf(Fout, '\n\n Physical Properties');
fprintf(Fout, '\n Sensor Mass: %11.3f %s', RAD_MATRIX3{1,26}, RadUnitLabelStr{26} );
fprintf(Fout, '\n Dimensions - Length: %11.3f %s', RAD_MATRIX3{1,27}, RadUnitLabelStr{27} );
fprintf(Fout, '\n Width: %11.3f %s', RAD_MATRIX3{1,28}, RadUnitLabelStr{28} );
fprintf(Fout, '\n Height: %11.3f %s', RAD_MATRIX3{1,29}, RadUnitLabelStr{29} );
fprintf(Fout, '\n Power Requirement: %11.3f %s', RAD_MATRIX3{1,30}, RadUnitLabelStr{30} );

fprintf(Fout, '\n\n Sensor Comments: ');
for r = 1:length(RAD_MATRIX3{1,31})
    fprintf(Fout, '\n %s', RAD_MATRIX3{1,31}{r} );
end
fclose(Fout);

```



## D16. Temperature Sensors

```

% function TEMP_SENSORS(TEMP_SENS_TYPE,PLTEMP,UNITS,PLNAME,Print)
% Temperature Sensor Database
%
% Developed by: Keith Schreck
% Mechanical and Aerospace Engineering
% San Jose State University
% Date: Fall 2007
%
% This file contains Temperature Sensor values for the ISSPO program.
% Data here is loaded into the main program to determine the proper sensor
% to use for Missions based on the planetary Atmosphere composition.
% Values are assigned to a variable name and saved to a MatLAB '.mat' file
% and loaded when calling the ISSPO program.
%
% Some of the values are based on the Unit system chosen and are loaded
% in separate sections.
%
% References
% http://www.omega.com/guides/thermocouples.html
% http://www.lakeshore.com/temp/sen/prtdts.html
% http://www.sensors.goodrich.com/literature/lit_pdfs/4025_Sensor_146MD.pdf
% http://www.rdfcorp.com/products/aerospace/aerospace.shtml

function TEMP_SENSORS(TEMP_SENS_TYPE,PLTEMP,UNITS,PLNAME,Print)
% Define Internal Program Variables
Poor = 1;
Fair = 2;
Good = 3;
Excellent = 4;

% Load Constants values into Local Program
load('Constants_DB.mat');
load('Planet_Data.mat');

%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%
%% Temperature Sensor Data Base Created Based on un Unit System
%% 2 - D Matrix TEMPSSENSOR_DB(T_Type, Property)
%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%

if strcmp(UNITS,'SI')== 1
% Properties [ SENSOR COLD LIMIT HOT LIMIT POWER REQ MASS REQ
ERROR TOL ERROR TOL TRL TYPE Alloy Combination EMF over Max Trange OPERATING
% Units deg C LEVEL + Lead deg C deg C W - TBD g - TBD Type
TBD % Units deg C LEVEL + Lead deg C deg C mV mV
TEMPSENSOR_DB = {
0.5 0.0 0 0 'Pt-30Rh' 'Pt-6Rh' 0.000 1700.000 0.000 0.000
1.0 4.5 0 0 'W-5Re' 'W-26Re' 0.000 2320.000 0.000 0.000
};

%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%
%% PLANETS USED [
%% Length Width Height
%% N/A ]
%% PLANETS USED [
%% Length Width Height
%% N/A ]
%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%

```

```

1.0 4.5 0 'D' 0.000 2320.000 0.000 39.506 0.000 'Voltage' 0.00 0.00 0.00 0.000
'W-3%Re' 'W-25%Re' 'E' -200.000 900.000 'Voltage' [ ] ;
0.000 0.5 1.7 0 'Ni-Cr' 'Cu-Ni' 'Cu-Ni' 76.373 0.000 'Voltage' [ ] ;
0.000 1.0 4.5 0 'W' 'W-26%Re' 'W-26%Re' 38.564 0.000 'Voltage' [ ] ;
0.000 0.75 2.2 0 'Fe' 'Cu-Ni' 'Cu-Ni' 69.553 0.000 'Voltage' [ ] ;
0.000 0.75 2.2 0 'Ni-Cr' 'Ni-Al' 'Ni-Al' 54.886 0.000 'Voltage' [ ] ;
0.000 0.75 2.2 0 'Ni-Cr-Si' 'Ni-Si-Mg' 'Ni-Si-Mg' 47.513 0.000 'Voltage' [ ] ;
0.000 0.25 1.5 0 'Pt-13%Rh' 'Pt' 'Pt' 21.101 0.000 'Voltage' [ ] ;
0.000 0.25 1.5 0 'Pt-10%Rh' 'Pt' 'Pt' 18.693 0.000 'Voltage' [ ] ;
0.000 0.75 1.0 0 'Cu' 'Cu-Ni' 'Cu-Ni' 20.872 0.000 'Voltage' [ ] ;
0.56 4.5 0 'Platinum' 'Platinum' 'Platinum' 0.000 0.250 'Resistance' [ ] ;
0.10 0.26 0 'Platinum' 'Platinum' 'Platinum' 0.060 0.350 'Resistance' [ ] ;
0.000 0.26 0 'Platinum' 'Platinum' 'Platinum' 0.000 0.000 'Resistance' [ ] ;

TempUnitLabelStr = { 'Type' 'deg C' 'deg C' 'W' 'g' 'mm' 'mm' 'mm' 'mm' 'deg C' 'TRL'
'+Lead' '-Lead' 'mV' 'mV' 'N/A' } ;

```

end

```

if strcmpi(UNITS,'British')==1
% Properties [ SENSOR
ERROR TOL ERROR TOL TRL Alloy Combination COLD LIMIT HOT LIMIT POWER REQ MASS REQ OPERATING PLANETS USED ] PRESSURE
% Units deg F LEVEL + Lead deg F deg F BTU/sec - TBD oz - TBD Type Height psi - TBD
% TEMPSSENSOR_DB = { deg F deg F deg F deg F mV mV Type Height Length Width Height Length Width Height Length Width Height
0.5 0.0 0 'B' 32.000 3092.000 0.000 13.280 'Voltage' 0.00 0.00 0.00 0.000 0.00 0.00 0.00 0.000
'C' 32.000 4208.000 0.000 37.066 'Voltage' 0.00 0.00 0.00 0.000 0.00 0.00 0.00 0.000
1.0 40.1 0 'W-5%Re' 'W-26%Re' 'W-26%Re' 39.506 0.000 'Voltage' 0.00 0.00 0.00 0.000 0.00 0.00 0.00 0.000
'D' 32.000 4208.000 0.000 1652.000 0.000 'Voltage' 0.000 0.000 0.000 0.000 0.00 0.00 0.00 0.000
1.0 40.1 0 'W-3%Re' 'E' -328.000 32.000 3092.000 76.373 0.000 'Voltage' [ ] ;
0.000 0.5 35.06 0 'Ni-Cr' 'Cu-Ni' 'Cu-Ni' 4208.000 0.000 4208.000 38.564 0.000 'Voltage' [ ] ;
0.000 1.0 40.1 0 'W' 'W-26%Re' 'W-26%Re' 1382.000 0.000 1382.000 69.553 0.000 'Voltage' [ ] ;
0.000 0.75 35.96 0 'Fe' 'Cu-Ni' 'Cu-Ni' 2282.000 0.000 2282.000 54.886 0.000 'Voltage' [ ] ;
0.000 0.75 35.96 0 'Ni-Cr' 'Ni-Al' 'Ni-Al' 2372.000 0.000 2372.000 47.513 0.000 'Voltage' [ ] ;
0.000 0.75 35.96 0 'Ni-Cr-Si' 'Ni-Si-Mg' 'Ni-Si-Mg' 2642.000 0.000 2642.000 21.101 0.000 'Voltage' [ ] ;
0.000 0.25 34.7 0 'Pt-13%Rh' 'Pt' 'Pt' 2642.000 0.000 2642.000 18.693 0.000 'Voltage' [ ] ;
0.000 0.25 34.7 0 'Pt-10%Rh' 'Pt' 'Pt' -0.236 0.000 0.000 0.000 0.000 'Voltage' [ ] ;

```

```

0.000 0.75 33.8 0 'T' -418.000 662.000 0.000 0.000 0.00 0.00 0.00 0.00
'PT-102' 0 'Cu' 'Cu-Ni' -6.258 20.872 'Voltage' [ 0.000
0.056 8.10 0 'Platinum' 'Platinum' 599.850 0.0001 0.0088 1.225 0.079 0.079 ; 0.000
'M-0146MD' -269.000 400.000 0.000 0.000 'Resistance' [ ;
0.10 0.47 0 'Platinum' 'Platinum' 0.000 0.0600 0.0123 1.80 0.060 0.060 0.000
'Resistance' [ ] ;

TempUnitLabelStr = { 'Type' 'deg F' 'deg F' 'BTU/sec' 'oz' 'in' 'in' 'psi' '%' 'deg F' 'TRL'
'+Lead' '-Lead' 'mv' 'N/A' };
end

% Comments On the Bare Wire Environment - Use of the Thermocouple Material
TEMPSENSOR_DB{ 1,19} = 'Oxidizing or Inert. Do Not Insert in Metal Tubes. Beware of Contamination. High Temp. Common Use in
Glass Industry.' ;
TEMPSENSOR_DB{ 2,19} = 'Vacuum, Inert, Hydrogen. Beware of Embrittlement. Not Practical Below 399°C (750°F). Not for
Oxidizing Atmosphere.' ;
TEMPSENSOR_DB{ 3,19} = 'Vacuum, Inert, Hydrogen. Beware of Embrittlement. Not Practical Below 399°C (750°F). Not for
Oxidizing Atmosphere.' ;
TEMPSENSOR_DB{ 4,19} = 'Oxidizing or Inert. Limited Use in Vacuum or Reducing. Highest EMF Change Per Degree.' ;
TEMPSENSOR_DB{ 5,19} = 'Vacuum, Inert, Hydrogen. Beware of Embrittlement. Not Practical Below 399°C (750°F). Not for
Oxidizing Atmosphere.' ;
TEMPSENSOR_DB{ 6,19} = 'Reducing, Vacuum, Inert. Limited Use in Oxidizing at High Temperatures. Not Recommended for Low
Temperatures.' ;
TEMPSENSOR_DB{ 7,19} = 'Clean Oxidizing and Inert. Limited Use in Vacuum or Reducing. Wide Temperature Range, Most Popular
Calibration.' ;
TEMPSENSOR_DB{ 8,19} = 'Alternative to Type K. More Stable at High Temps.' ;
TEMPSENSOR_DB{ 9,19} = 'Oxidizing or Inert. Do Not Insert in Metal Tubes. Beware of Contamination. High Temperature.' ;
TEMPSENSOR_DB{ 10,19} = 'Oxidizing or Inert. Do Not Insert in Metal Tubes. Beware of Contamination. High Temperature.' ;
TEMPSENSOR_DB{ 11,19} = 'Mild Oxidizing, Reducing Vacuum or Inert. Good Where Moisture Is Present. Low Temperature &
Cryogenic Applications.' ;
TEMPSENSOR_DB{ 12,19} = 'Resistance type Thermal Sensor.' ;
TEMPSENSOR_DB{ 13,19} = 'Resistance type Thermal Sensor. High Reliability, Linear Output, Miniature Size, Wide Temperature
Range, Recommended for Space Vehicles.' ;

%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%
%% Thermocouple Wire Insulator Data Base Created Based on un Unit System
%% 2 - D Matrix TCWIns_DB(Ins_Type, Property)

if strcmpi(UNITS,'SI')==1
% Properties { INSULATION COLD LIMIT HOT LIMIT
% Units TYPE deg C deg C
TCWIns_DB1 = {
'PP' -40.000 105.000 ;
'FF' -200.000 200.000 ;
'TT' -267.000 260.000 ;
'KK' -267.000 316.000 ;
'TG' -73.000 260.000 ;
'GG' -73.000 482.000 ;
'HH' -73.000 871.000 ;
'XR' -73.000 871.000 ;
'XC' -73.000 1204.000 ;
'XS' -73.000 1038.000 ;
'TFE' -267.000 260.000 ;
};

```

```

WinsUnitLabelStr = { 'Type' 'deg C' 'deg C' };
end

if strcmpi(UNITS,'British')== 1
% Properties [ INSULATION COLD LIMIT HOT LIMIT
% Units TYPE deg F deg F ]
TCWIns_DB1 = {
'PP' -40.000 221.000 ;
'FF' -338.000 392.000 ;
'TT' -450.000 500.000 ;
'KK' -450.000 600.000 ;
'TG' -100.000 500.000 ;
'GG' -100.000 900.000 ;
'HH' -100.000 1300.000 ;
'XR' -100.000 1600.000 ;
'XC' -100.000 2200.000 ;
'XS' -100.000 1990.000 ;
'TFE' -450.000 500.000 ;
};

WinsUnitLabelStr = { 'Type' 'deg F' 'deg F' };
end

% Adds Additional Environmental Insulation Data to the Thermal Limit Properties
% Material Construction [ Abrasion Flexibility Water Resistance To:
% Humidity Overall Resistance Conductors Submersion Solvent Acid Base Flame
TCWIns_DB2 = {
'PVC' 'Good' 'Excellent' 'Good' 'Fair' 'Good' 'Good' 'Good'
'Excellent' 'PEP Teflon' 'Good' 'Excellent' 'Excellent' 'Excellent' 'Excellent'
'Excellent' 'PFA Teflon' 'Good' 'Excellent' 'Excellent' 'Excellent' 'Excellent'
'Excellent' 'Kapton' 'Good' 'Good' 'Good' 'Good' 'Good'
'Excellent' 'Glass Braid' 'Good' 'Good' 'Excellent' 'Excellent' 'Excellent'
'Glass Braid' 'Glass Braid' 'GG' Type 'Poor' 'Excellent' 'Excellent' 'Fair'
'HT Glass Braid' 'HT Glass Braid' 'HH' Type 'Poor' 'Excellent' 'Excellent' 'Fair'
'Refrasil Braid' 'Refrasil Braid' 'XR' Type 'Poor' 'Excellent' 'Excellent' 'Poor'
'Nextel Braid' 'Nextel Braid' 'XC' Type 'Poor' 'Excellent' 'Excellent' 'Fair'
'Silica' 'Silica' 'XS' Type 'Poor' 'Excellent' 'Excellent' 'Fair'
'Excellent' 'TFE Teflon' 'TFE Teflon' 'Excellent' 'Excellent' 'Excellent'
};
% Combine Arrays along the second dimension
TCWIns_DB = cat(2,TCWIns_DB1,TCWIns_DB2);

tmp = 1;
for t = 1:size(TEMPSENSOR_DB,1)

```

```

if strcmpi(TEMPSENSOR_DB{t,17},TEMP_SENS_TYPE) == 1
for s = 1:size(TEMPSENSOR_DB,2)
MATRIX{tmp,s} = TEMPSENSOR_DB{t,s};
end
tmp = tmp + 1;
end
end

if strcmpi(TEMP_SENS_TYPE, 'VOLTAGE') == 1
c = 1;
% Build Up Database of Possible Thermocouple Sensors Based on Temperature Range
for t = 1:size(MATRIX,1)
if (PLTEMP >= MATRIX{t,2}) && (PLTEMP <= MATRIX{t,3})
for a = 1:size(MATRIX,2)
MATRIX2{c,a} = MATRIX{t,a};
end
c = c+1;
end
end

if exist('MATRIX2') == 0
fprintf('\n\n *** WARNING ***');
fprintf('\n\n NO VIABLE TEMPERATURE SENSOR EXISTS IN DATABASE!');
fprintf('\n\n Other methods to determine Planetary Temperature must be used. ');
end

for c = 1:size(MATRIX2,1)
RESULT = findstr(MATRIX2{c,19}, 'Most Popular');

if size(RESULT) == 0
SENS_NUM = c;
break;
end
end

if strcmpi(PLNAME, 'PLUTO') == 1
for c = 1:size(MATRIX2,1)
RESULT = findstr(MATRIX2{c,19}, 'Cryogenic');

if size(RESULT) == 0
else
SENS_NUM = c;
break;
end
end
end
end

```

```

k = 1;
% Build Up Database of Possible Thermocouple Insulation Material Based on Temperature Range
for insul = 1:size(TCWIns_DB,1)
    if (PLTEMP >= TCWIns_DB{insul,2}) && (PLTEMP <= TCWIns_DB{insul,3})
        for a = 1:size(TCWIns_DB,2)
            INS_MATRIX{k,a} = TCWIns_DB{insul,a};
        end
        k = k+1;
    end
end

SUM_ARRAY = [];
total = 0;
% Determine Optimal Insulation Material
for z = 1:size(INS_MATRIX,1)
    for mat = 4:11
        total = 0;
        switch (INS_MATRIX{z,mat})
            case 'POOR'
                total = total + Poor;
            case 'Fair'
                total = total + Fair;
            case 'Good'
                total = total + Good;
            case 'Excellent'
                total = total + Excellent;
        end
    end
    SUM_ARRAY(z) = total;
end

[ SUM, Max_INS ] = max(SUM_ARRAY);

% Save Temperature Sensor Properties to Data File
save('TEMP_SENSOR_Data.mat');

save('TEMP_SENSOR_Final.mat', 'MATRIX2', 'TempUnitLabelStr');

%*****
%
% *** PRINT SUMMARY Output Section for 'Print' = 'Y'

if strcmpi(Print, 'Y') == 1
    fprintf('\n\n *** T E M P E R A T U R E S E N S O R S U M M A R Y R E S U L T S ***');
    fprintf('\n\n *** I N P U T S ***');
    fprintf('\n Planetary Surface Temperature: %7.2f %s', PLTEMP, TempUnit);
end

```







```

end

if strcmpi(TEMP_SENS_TYPE, 'RESISTANCE') == 1
    if size(MATRIX,1) > 1
        C = 1;
        % Build Up Database of Possible Thermocouple Sensors Based on Temperature Range
        for t = 1:size(MATRIX,1)
            if (PLTEMP >= MATRIX{t,2}) && (PLTEMP <= MATRIX{t,3})
                for a = 1:size(MATRIX,2)
                    MATRIX2{c,a} = MATRIX{t,a};
                end
                C = C+1;
            end
        end
        if exist('MATRIX2') == 0
            fprintf('\n\n *** WARNING ***');
            fprintf('\n\n NO VIABLE TEMPERATURE SENSOR EXISTS IN DATABASE!');
            fprintf('\n\n Other methods to determine Planetary Temperature must be used. ');
            MATRIX2 = MATRIX;
        end
    end
end

if size(MATRIX2,1) > 1
    % Sort Sensors By Smallest Error Tolerance Percent
    for err = 1:size(MATRIX2,1)
        TEMP_ERR(err) = MATRIX2{err,10};
    end;
    [ MIN_ERR_SCL, ER_INDEX ] = min(TEMP_ERR);
    % Build Sensor Matrix Based on Min ERROR Tolerances
    k = 1;
    for c = 1:length(TEMP_ERR)
        if TEMP_ERR(c) == MIN_ERR_SCL
            for j = 1:size(MATRIX2,2)
                MATRIX3{k,j} = MATRIX2{c,j};
            end
            k = k + 1;
        end
    end
end

% Copies MATRIX2 to MATRIX3 if all Sensors Eliminated
if exist('MATRIX3') == 0
    MATRIX3 = MATRIX2;
end

clear MATRIX2;
MATRIX2 = MATRIX3;

```









```

'N/A' WIND_VELOCITY_SENSOR_DB2 = { 0.227 180.34 76.20 20.32 124.0 70.0 41.0 0
'N/A' 0 0 ; 0.227 180.34 76.20 20.32 180.3 25.4 25.4 0
'N/A' 0 0 ; 0.355 185.00 78.00 38.00 280.0 12.0 12.0 0
'N/A' 0 0 ; 0 150.00 81.00 30.00 178.0 70.0 70.0 0
'N/A' 0 0 ; 0.227 180.00 76.00 20.00 180.0 70.0 70.0 0
'N/A' 0 0 ; 0.227 180.00 76.00 20.00 180.0 70.0 70.0 0
'N/A' 0 0 ; 1.400 181.00 70.00 35.00 181.0 73.0 73.0 0
'N/A' 0 0 ; 0 183.00 76.00 45.00 0 0 0 0
'N/A' 0 0 ; 0.220 200.00 68.00 30.00 600.0 13.0 13.0 0
'N/A' 0 0 ; 0.210 181.00 76.00 45.00 0 0 0 0
'N/A' 0 0 ; 0.500 0 0 0 162.0 50.0 50.0 7.0
'DC' 14.0 100.0 };

WVelocityUnitLabelStr1 = { 'N/A', 'm/s', 'mm', 'mm', '% FS', 'mm', 'mm', 'N/A', 'deg C', 'deg C', 'mA', 'deg C', 'N/A', 'Type' };
WVelocityUnitLabelStr2 = { 'kg', 'mm', 'mm', 'mm', '% FS', 'mm', 'mm', 'mm', 'V', 'N/A', 'mm', 'deg C', 'deg C', 'N/A', 'Type' };

```

```
end
```

```

if strcmp(UNITS,'British')==1
% Properties
INST_TEMP_RANGE POWER_REQUIREMENT ] SENSOR
% Units { TYPE ]
Low deg F High Number WIND_VELOCITY_SENSOR_DB1 = { 'HH-30A'
122.0 22.0 'AA' 'HHF300A'
32.0 122.0 2 'AA' 'HHF42'
32.0 122.0 6 'AAA' 'HHF710'
32.0 212.0 0 'NiCad w/AC' '
32.0 122.0 2 'AA' 'HHF750'
32.0 122.0 2 'AA' 'HHF753'
-10.0 122.0 1 '9 Volt' 'HHF91'
-10.0 122.0 1 '9 Volt' 'HHF92A'
32.0 122.0 1 'REED YK-80AS' '
14.0 122.0 1 'REED 8903' '

```



```

WIND_VELOCITY_SENSOR_DB3{10,1} = 'Combination Velocity/Temperature Probe. Min/Max/Ave Velocity Measurements. 2 hour
Continuous Ave Velocity. RS232 Signal Output Format. Integrated Probe/Electronics configuration.';
WIND_VELOCITY_SENSOR_DB3{11,1} = 'High accuracy wind speed and direction sensing. WIND DIRECTION MEASUREMENT. Compact,
unobtrusive solid-state design with no moving parts';

```

```

WIND_VELOCITY_SENSOR_DB= cat(2,WIND_VELOCITY_SENSOR_DB1,WIND_VELOCITY_SENSOR_DB2,WIND_VELOCITY_SENSOR_DB3);
WVelocityUnitLabelStr = cat(2,WVelocityUnitLabelStr1,WVelocityUnitLabelStr2);

```

```

% Determine Optimal Sensor if Planetary Atmosphere Exists
ATM_FLG = 0;
% Search Planetary Atmosphere for Components
if strcmpi(PlanAtm_Prop{1,1}, 'N/A') == 0
    ATM_FLG = 1;
end

```

```

if ATM_FLG == 0
    if strcmpi(PlanAtm_MEPProp{1,1}, 'N/A') == 0
        ATM_FLG = 1;
    end
end

```

```

% Determines the WIND_VELOCITY Sensor required if Atmosphere is Defined
if ATM_FLG == 1

```

```

    % Find the Max Operational Range
    for w = 1:size(WIND_VELOCITY_SENSOR_DB,1)
        W_DELTA(w) = WIND_VELOCITY_SENSOR_DB{w,3} - WIND_VELOCITY_SENSOR_DB{w,2};
    end

```

```

    % Find the Maximum Wind Velocity Range of the Sensor
    MAX_DELTA = max(W_DELTA);

```

```

    % For Multiple Sensors with the Same Range Reduce Sensor Matrix to Those
    % with the Greatest Range
    pos = 1;
    for r = 1:length(W_DELTA)
        if W_DELTA(r) == MAX_DELTA
            for j = 1:size(WIND_VELOCITY_SENSOR_DB,2)
                W_MATRIX{pos,j} = WIND_VELOCITY_SENSOR_DB{r,j};
            end
            pos = pos + 1;
        end
    end

```

```

end

```

```

% Continue Selection Process if More than one Sensor have Same Max Range
if size(W_MATRIX,1) > 1

```

```

    % Down Select Remaining Sensors Based on Mass and Dimensions
    % Create Mass / Volume Scaling Factor Optimization % Probe Dimensions
    for mv = 1:size(W_MATRIX,1)

```

```

    Dimensions

```

```

    Instrument

```



```

* W_MATRIX{mv,17} * W_MATRIX{mv,18}));
WV_SCALE_FAC(mv) = W_MATRIX{mv,12} * ( (W_MATRIX{mv,13} * W_MATRIX{mv,14} * W_MATRIX{mv,15}) + (W_MATRIX{mv,16}
* W_MATRIX{mv,17} * W_MATRIX{mv,18}));
if (W_MATRIX{mv,12} == 0) || (W_MATRIX{mv,13} == 0) || (W_MATRIX{mv,14} == 0) || (W_MATRIX{mv,15} == 0)
WV_SCALE_FAC(mv) = 9.99E99;
end
end
end

[ WV_MIN_SCL, WV_INDEX ] = min(WV_SCALE_FAC);
% Build Sensor Matrix Based on Scale Factors
k = 1;
    for c = 1:length(WV_SCALE_FAC)
        if WV_SCALE_FAC(c) == WV_MIN_SCL
            for j = 1:size(W_MATRIX,2)
                W_MATRIX2{k,j} = W_MATRIX(c,j);
            end
            k = k + 1;
        end
    end
end

% Copies W_MATRIX to W_MATRIX2 if all Sensors Eliminated
if exist('W_MATRIX2') == 0
    W_MATRIX2 = W_MATRIX;
end

if size(W_MATRIX2,1) > 1
    % Sort Sensors by Accuracy
    % Find the Min Wind Velocity Range of the Sensor
    [ MIN_ACC, ACC_INDEX ] = min(W_MATRIX2);
    W_MATRIX3 = W_MATRIX2{ACC_INDEX,:};
end

% Copies W_MATRIX2 to W_MATRIX if all Sensors Eliminated
if exist('W_MATRIX3') == 0
    W_MATRIX3 = W_MATRIX2;
end
end

end
if exist('W_MATRIX3') == 0
    W_MATRIX3 = W_MATRIX;
end

% Closes IF Loop If Atmosphere is Defined
end

% Determines if a sensor is determined from database
if exist('W_MATRIX3') == 0
    for wv = 1:size(WIND_VELOCITY_SENSOR_DB,2)
        W_MATRIX3{1,w} = 0.0;
    end
end

```





```

fprintf(Fout, '\n %s', W_MATRIX3{1,23});
fclose(Fout);

end

% Begins Selection Section For Doppler Type Wind Velocity Sensors
if strcmpi(WV_Type, 'DOPPLER') == 1
    % Wind Velocity Sensor Data Base Created Based on un Unit System
    % 2 - D Matrix WIND_VELOCITY_SENSOR_DB(WV_Type, Property)

    % Electrical Specifications
    [ Properties
      AGING DATA ACCURACY STABILITY WARM UP TIME VOLTAGE TYPE OUTPUT LEVEL SPECTRAL PURITY
      % Units @ 25 C @ 100 sec min N/A Volts
      Monthly WIND_VELOCITY_SENSOR_DB1 = { '8130A' 10.0 7.0 1.50 -30.0
      5.0E-11 1.0E-09 1.0E-11 3.0E-12 14.0 10.0 'DC' [ 22 32 ] ;
      3.3E-09 1.0E-06 0.00 5.0E-12 0.0 'DC' [ 15 ] };

      WVelocityUnitLabelStr1 = { 'N/A' 'MHz' 'dBm' 'dBm' 'dBc' 'Hz/month' '10 yrs' 'Hz @ 25 deg C' 'Hz @
      100 sec' 'min' 'Type' 'V' };

      if strcmpi(UNITS, 'SI') == 1
          % Properties
          PRESSURE SENS WARM-UP MAX POWER STEADY STATE POWER INPUT POWER QUIESCENT WEIGHT TEMP SENSITIVITY ORIENTATION SENS
          LIFETIME ]
          % Units W @28 V W @ 28 V High W @ 28 V @ 25 deg C High Kg Op Temp Range Hz
          Years ] W @28 V W @ 28 V W @ 28 V @ 25 deg C 85.0 0.900 3.0E-10 10.26 7.41 5.0E-11
          1.0E-13 35.0 22.0 -40.0 68.0 12.0 -62.0 85.0 0.900 10.26 7.41 7.28 20
          ;
          0.00 12.0 4.0 -40.0 70.0 3.0 -55.0 85.0 0.159 7.62 4.24 0.00 4.06 0
          };

          WVelocityUnitLabelStr2 = { 'deg C' 'deg C' 'deg C' 'Hz' 'Hz/mbar' 'W' 'W' 'W' 'Kg' 'cm'
          'cm' 'years' };
          end

          if strcmpi(UNITS, 'British') == 1
              % Properties
              PRESSURE SENS WARM-UP MAX POWER STEADY STATE POWER INPUT POWER QUIESCENT WEIGHT TEMP SENSITIVITY ORIENTATION SENS
              LIFETIME ]
              % Units BTUs/hr @28 V BTUs/hr @ 28 V BTUs/hr @ 28 V @ 77 deg F Low deg F High Op Temp Range Hz
              Hz/psi BTUs/hr @28 V BTUs/hr @ 28 V BTUs/hr @ 28 V lbm Op Temp Range Length Width Height
              years ]
          end
    end
  end

```

```

6.89E-12 WIND_VELOCITY_SENSOR_DB2 = {
20 119.351 -40.0 154.4 -79.6 185.0 1.984 3.0E-10 2.92 5.0E-11
; 75.134 40.982 4.04 2.87
0.00 40.982 -40.0 158.0 -67.0 185.0 0.351 1.0E-08 1.67 0.00
0 13.661 10.245 3.00 1.60
};

WVelocityUnitLabelStr2 = { 'deg F' 'deg F' 'deg F' 'Hz' 'Hz' 'Hz/psi' 'BTUs/hr' 'BTUs/hr' 'BTUs/hr'
'lbm' 'in' 'in' 'years' };
end

WIND_VELOCITY_SENSOR_DB3{1,1} = { 'Modern Militarized Design' ;
'5 and 10 MHz Sinewave Outputs' ;
'RS-232 Digital Control and Monitoring' ;
'Fuggedized High Performance Rb Physics Package' ;
'Meets many Mil-Spec Standards' ;
'Data for Single Unit' ;
'Two Units required for Doppler Tracking' ;
'Space Qualified Hardware Production capability' };

WIND_VELOCITY_SENSOR_DB3{2,1} = { 'Modern Militarized Ovenized SC Design' ;
'10 MHz Sinewave Output' ;
'RS-232 Digital Control and Monitoring' ;
'Low g Sensitivity Military OCXO' ;
'Meets many Mil-Spec Standards' ;
'Data for Single Unit' ;
'Two Units required for Doppler Tracking' ;
'Space Qualified Hardware Production capability' };

% Combine Data Arrays into a Single Data Matrix
WIND_VELOCITY_SENSOR_DB = cat(2,WIND_VELOCITY_SENSOR_DB1,WIND_VELOCITY_SENSOR_DB2,WIND_VELOCITY_SENSOR_DB3);
WVelocityUnitLabelStr = cat(2,WVelocityUnitLabelStr1,WVelocityUnitLabelStr2);

% Determine Optimal Sensor if Planetary Atmosphere Exists
ATM_FLG = 0;
% Search Planetary Atmosphere for Components
if strcmpi(PlanAtm_Prop{1,1},'N/A') == 0
ATM_FLG = 1;
end

if ATM_FLG == 0
if strcmpi(PlanAtm_MEPprop{1,1},'N/A') == 0
ATM_FLG = 1;
end
end

% Determines the WIND VELOCITY Sensor required if Atmosphere is Defined
if ATM_FLG == 1
% Find the Max Operational Output Range
for w = 1:size(WIND_VELOCITY_SENSOR_DB,1)
W_DELTA(w) = WIND_VELOCITY_SENSOR_DB{w,3};
end
end

% Find the Maximum Output Range of the Sensor

```

```

MAX_DELTA = max(W_DELTA);
% For Multiple Sensors with the Same Range Reduce Sensor Matrix to Those
% with the Greatest Range
pos = 1;
for r = 1:length(W_DELTA)
    if W_DELTA(r) == MAX_DELTA
        for j = 1:size(WIND_VELOCITY_SENSOR_DB,2)
            W_MATRIX{pos,j} = WIND_VELOCITY_SENSOR_DB{r,j};
        end
        pos = pos + 1;
    end
end
end

% Continue Selection Process if More than one Sensor have Same Max Range
if size(W_MATRIX,1) > 1
end

% Find the Min Spectral Purity Value
for w = 1:size(W_MATRIX,1)
    W_DELTA2(w) = abs(W_MATRIX{w,5});
end

% Find the Min Spectral Purity Value Range of the Sensor
MIN_DELTA2 = min(W_DELTA2);

% For Multiple Sensors with the Same Range Reduce Sensor Matrix to Those
% with the Greatest Range
pos = 1;
for r = 1:length(W_DELTA2)
    if W_DELTA2(r) == MIN_DELTA2
        for j = 1:size(W_MATRIX,2)
            W_MATRIX2{pos,j} = W_MATRIX{r,j};
        end
        pos = pos + 1;
    end
end

% Copies W_MATRIX to W_MATRIX2 if all Sensors Eliminated
if exist('W_MATRIX2') == 0
    W_MATRIX2 = W_MATRIX;
end

if size(W_MATRIX2,1) > 1
    % Down Select Remaining Sensors Based on Mass and Dimensions
    % Create Sensitivity Scaling Factor Optimization
    for sv = 1:size(W_MATRIX2,1)
        W_SCALE_FAC(sv) = W_MATRIX2{sv,18} * W_MATRIX2{sv,18} * W_MATRIX2{sv,20};
        if (W_MATRIX2{sv,18} == 0) || (W_MATRIX2{sv,19} == 0) || (W_MATRIX2{mv,20} == 0)
            W_SCALE_FAC(sv) = 9.99E99;
        end
    end
end

```

```

end
[ WV_MIN_SCL, WV_INDEX ] = min(WV_SCALE_FAC);
% Build Sensor Matrix Based on Scale Factors
k = 1;
    for c = 1:length(WV_SCALE_FAC)
        if WV_SCALE_FAC(c) == WV_MIN_SCL
            for j = 1:size(W_MATRIX2,2)
                W_MATRIX3{k,j} = W_MATRIX2(c,j);
            end
            k = k + 1;
        end
    end
end
end
% Copies W_MATRIX2 to W_MATRIX3 if all Sensors Eliminated
if exist('W_MATRIX3') == 0
    W_MATRIX3 = W_MATRIX2;
end
end
end
if exist('W_MATRIX3') == 0
    W_MATRIX3 = W_MATRIX;
end
end
% Closes IF Loop If Atmosphere is Defined
end
% Determines if a sensor is determined from database
if exist('W_MATRIX3') == 0
    for wv = 1:size(WIND_VELOCITY_SENSOR_DB,2)
        W_MATRIX3{1,w} = 0.0;
    end
    W_MATRIX3{1,1} = 'N/A';
    W_MATRIX3{1,12} = 'N/A';
    W_MATRIX3{1,29} = 'N/A';
end
% Save Wind Velocity Sensor Properties to Data File
save('WIND_VELOCITY_SENSOR_Data.mat');
save('WIND_VELOCITY_SENSOR_Final.mat', 'W_MATRIX3', 'WVelocityUnitLabelStr', 'WV_TYPE');
% *****
% *** PRINT SUMMARY Output Section for 'Print' = 'y'
if strcmpi(Print, 'y') == 1
    fprintf('\n\n *** WIND VELOCITY SENSOR SUMMARY RESULTS ***');
    fprintf('\n\n *** INPUTS ***');

```







```

fprintf(Fout, '\n\n Sensor Comments:');
for r = 1:length(W_MATRIX3{1,29})
    fprintf(Fout, '\n %s', W_MATRIX3{1,29}{r} );
end
fclose(Fout);

```

end

## D18. Refraction Sensors

```

% function REFRACTION_SENSORS(UNITS,Print)
% Refraction Sensor Database
%
% Developed by: Keith Schreck
% Mechanical and Aerospace Engineering
% San Jose State University
% Date: Fall 2007
%
% This file contains Refraction Sensor values for the ISSPO program.
% Data here is loaded into the main program to determine the proper sensor
% to use for Missions based on the sensor requirements.
% Values are assigned to a variable name and saved to a Matlab '.mat' file
% and loaded when calling the ISSPO program.
%
% Some of the values are based on the Unit system chosen and are loaded
% in separate sections.
%
% References
% http://www.sensata.com/products/sensors/spreeta-r.htm
% http://www.fiso.com/index.php?module=CMS&id=26
%
function REFRACTION_SENSORS(UNITS,Print)

% Load Constants values into Local Program
load('Constants_DB.mat');
load('Planet_Data.mat');

%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%%
%% Refraction Sensor Data Base Created Based on un Unit System
%% 2 - D Matrix REFRACTION_DB(N_Type, Property)
if strcmpi(UNITS,'SI')==1
% Properties [ Sensor REFRACTION INDEX RANGE RESOLUTION DRIFT ACCURACY WEIGHT VOLTAGE TYPE
CURRENT POWER DIMENSIONS - cm ]

```

```

% Units mA Length [ Width Height ] Low N/A High RIU RIU % g V N/A
REFRACTION_DB1 = { 'TSPR2KXY-R', 1.333 1.500 5.0E-06 0 1.00 0 5 'DC'
100 500 3.00 0.70 1.50 ;
'REF-NP', 1.000 1.700 1.0E-04 0 0.071 0 0 'N/A'
0 0 22.80 0.97 0.97 ;
'REF-Huy', 1.250 1.450 1.0E-03 0 1.00 0 15 'DC'
6.66 10 10.00 10.00 11.61 ;

% Properties [ OPERATING TEMPERATURE ]
% Units [ Low deg C High deg C ]
REFRACTION_DB2 = { -10.0 70.0 ;
0.0 100.0 ;
-40.0 65.0 } ;

RefUnitLabelStr1 = { 'sensor', 'N/A', 'N/A', 'RIU', 'RIU', '%', 'g', 'V', 'Type', 'mA', 'mW', 'cm', 'cm' };
RefUnitLabelStr2 = { 'deg C', 'deg C', 'deg C', 'deg C', 'deg C', 'deg C', 'deg C', 'deg C', 'deg C', 'deg C', 'deg C', 'deg C', 'deg C', 'deg C', 'deg C' };

```

```
end
```

```

if strcmp(UNITS, 'BRITISH') == 1
% Properties [ Sensor DIMENSIONS - in ] REFRACTION INDEX RANGE Resolution DRIFT Accuracy Weight Voltage Type
CURRENT POWER [ Length Width Height ] Low N/A High RIU RIU % oz V N/A
% Units [ BTU's/hr Length Width Height ]
REFRACTION_DB1 = { 'TSPR2KXY-R', 1.333 1.50 5.0E-06 0 1.00 0 5 'DC'
100 1.708 1.170 0.265 0.599 ;
'REF-NP', 1.000 1.700 1.0E-04 0 0.071 0 0 'N/A'
0 0 8.980 0.382 0.382 ;
'REF-Huy', 1.250 1.450 1.0E-03 0 1.00 0 15 'DC'
6.66 0.034 3.940 3.940 4.570 } ;

% Properties [ OPERATING TEMPERATURE ]
% Units [ Low deg C High deg C ]
REFRACTION_DB2 = { 14.0 158.0 ;
32.0 212.0 ;
-40.0 149.0 } ;

RefUnitLabelStr1 = { 'sensor', 'N/A', 'N/A', 'RIU', 'RIU', '%', 'oz', 'V', 'Type', 'mA', 'BTUs/hr', 'in', 'in' };
RefUnitLabelStr2 = { 'deg F', 'deg F', 'deg F', 'deg F', 'deg F', 'deg F', 'deg F', 'deg F', 'deg F', 'deg F', 'deg F', 'deg F', 'deg F', 'deg F', 'deg F' };

```

```
end
```

```

%comments on sensors
REFRACTION_DB3{ 1,1} = { 'Optical based sensing technology for real-time, liquid quality and/or concentration analysis';
'High performance and low cost measurement systems';
'High performance, Highly quantitative, Customized software available' ;
'Robust, Lightweight, Small size, Low power' };
REFRACTION_DB3{ 2,1} = { 'Ideal for refractive index measurement of fluids in industrial, chemical and food processing industry';
'Rugged stainless steel package' ;
'Intrinsically safe, immune to EMI/RFI' ;
};

```

```

    'In-situ measurement' };
    REFRACCTION_DB3{ 3,1} = { 'Unique Design Solution - Combination NMOS Linear Image Sensor and Sapphire Refraction Prism';
    'Self-Scanning Photodiode Array' ;
    'Linear critical-angle refractometer.' ;
    'Flight Proven Hardware configuration - Huygens Lander SSP.' };

% Combine Refraction Sensor Arrays
REFRACTION_DB = cat(2,REFRACTION_DB1,REFRACTION_DB2,REFRACTION_DB3);
RefUnitLabelStr = cat(2,RefUnitLabelStr1,RefUnitLabelStr2);

% Find the Max Operational Temp Range
for r = 1:size(REFRACTION_DB,1)
    REF_DELTA(r) = REFRACCTION_DB{r,16} - REFRACCTION_DB{r,15};
end

% Find the Maximum Refraction Operating Range of the Sensor
MAX_REF_DELTA = max(REF_DELTA);

% For Multiple Sensors with the Same Temp Range Reduce Sensor Matrix to Those
% with the Greatest Range
k = 1;
for c = 1:length(REF_DELTA)
    if REF_DELTA(c) == MAX_REF_DELTA
        for j = 1:size(REFRACTION_DB,2)
            REF_MATRIX{k,j} = REFRACCTION_DB{c,j};
        end
        k = k + 1;
    end
end

if exist('REF_MATRIX') == 0
    REF_MATRIX = REFRACCTION_DB;
end

% Select Sensor Based on Refraction Range if More than One Sensor Exists In Database
if size(REF_MATRIX,1) > 1
    % Find the Max Operational Range
    for n = 1:size(REF_MATRIX,1)
        REF_DELTA_RNG(n) = REF_MATRIX{n,3} - REF_MATRIX{n,2};
    end

    % Find the Maximum Refraction Operating Range of the Sensor
    MAX_REF_DELTA_RNG = max(REF_DELTA_RNG);

    % For Multiple Sensors with the Same Range Reduce Sensor Matrix to Those
    % with the Greatest Range
    k = 1;
    for c = 1:length(REF_DELTA_RNG)
        if REF_DELTA_RNG(c) == MAX_REF_DELTA_RNG
            for j = 1:size(REF_MATRIX,2)
                REF_MATRIX{k,j} = REF_MATRIX{c,j};
            end
        end
    end
end

```

```

        k = k + 1;
    end
    end
end

if exist('REF_MATRIX2') == 0
    REF_MATRIX2 = REF_MATRIX;
end

% Select Sensor Based on Minimum Volume if More than One Sensor Exists In Database
if size(REF_MATRIX2,1) > 1
    % Create Volume Scaling Factor Optimization
    for v = 1:size(REF_MATRIX2,1)
        REF_SCALE_FAC(v) = REF_MATRIX2{v,9} * REF_MATRIX2{v,10} * REF_MATRIX2{v,11} * REF_MATRIX2{v,6};
        if (REF_MATRIX2{v,9} == 0) || (REF_MATRIX2{v,10} == 0) || (REF_MATRIX2{v,11} == 0) || (REF_MATRIX2{v,6} == 0)
            REF_SCALE_FAC(v) = 9.99E99;
        end
    end
end

{ REF_MIN_SCL, INDEX } = min(REF_SCALE_FAC);

% Build Sensor Matrix Based on Scale Factors
k = 1;
for c = 1:length(REF_SCALE_FAC)
    if REF_SCALE_FAC(c) == REF_MIN_SCL
        for j = 1:size(REF_MATRIX2,2)
            REF_MATRIX3{k,j} = REF_MATRIX2{c,j};
        end
        k = k + 1;
    end
end

% Copies REF_MATRIX2 to REF_MATRIX3 if all Sensors Eliminated
if exist('REF_MATRIX3') == 0
    REF_MATRIX3 = REF_MATRIX2;
end

% Save Refraction Sensor Properties to Data File
save('REFRACTION_SENSOR_Data.mat');

save('REFRACTION_SENSOR_Final.mat', 'REF_MATRIX3', 'RefUnitLabelStr');

%*****
%
% *** PRINT SUMMARY Output Section for 'Print' = 'Y'
if strcmpi(Print,'Y') == 1

```

```

fprintf('\n\n\n *** REFRACTION SENSOR SUMMARY RESULTS ****');
fprintf('\n\n\n INPUTS ****');
fprintf('\n Unit System:      %s', UNITS);

fprintf('\n\n\n **** REFRACTION SENSOR SUMMARY RESULTS ****\n\n');

% Refraction Sensor Properties
fprintf('\n\n\n Sensor Properties **');
fprintf('\n Refraction Sensor Type:      %s', REF_MATRIX3{1,1});
fprintf('\n Sensing Refraction Range - Min:   %12.5f %s', REF_MATRIX3{1,2}, RefUnitLabelStr{2});
fprintf('\n                               Middle: %12.5f %s', REF_MATRIX3{1,3}, RefUnitLabelStr{3});
fprintf('\n Sensing Resolution:                 %12.4E %s', REF_MATRIX3{1,4}, RefUnitLabelStr{4});
fprintf('\n Sensing Drift Rate:                 %12.5f %s', REF_MATRIX3{1,5}, RefUnitLabelStr{5});
fprintf('\n Sensor Accuracy:                   %12.5f %s', REF_MATRIX3{1,6}, RefUnitLabelStr{6});

fprintf('\n\n\n Power Requirements');
fprintf('\n Input Voltage:                      %12.3f %s', REF_MATRIX3{1,8}, RefUnitLabelStr{8});
fprintf('\n Voltage Type:                        %s', REF_MATRIX3{1,9}, RefUnitLabelStr{9});
fprintf('\n Input Current:                       %12.3f %s', REF_MATRIX3{1,10}, RefUnitLabelStr{10});
fprintf('\n Input Power:                          %12.3f %s', REF_MATRIX3{1,11}, RefUnitLabelStr{11});

fprintf('\n\n\n Physical Properties');
fprintf('\n Sensor Mass:                         %12.2f %s', REF_MATRIX3{1,7}, RefUnitLabelStr{7});
fprintf('\n Sensor Dimensions - Length:          %12.2f %s', REF_MATRIX3{1,12}, RefUnitLabelStr{12});
fprintf('\n                               Width:  %12.2f %s', REF_MATRIX3{1,13}, RefUnitLabelStr{13});
fprintf('\n                               Height:  %12.2f %s', REF_MATRIX3{1,14}, RefUnitLabelStr{14});

% Environmental Properties
fprintf('\n\n\n Environmental Properties');
fprintf('\n Temperature Range - Low:            %12.2f %s', REF_MATRIX3{1,15}, RefUnitLabelStr{15});
fprintf('\n                               High:   %12.2f %s', REF_MATRIX3{1,16}, RefUnitLabelStr{16});

fprintf('\n\n\n Sensor Comments:');
for r = 1:length(REF_MATRIX3{1,17})
    fprintf('\n %s', REF_MATRIX3{1,17}{r});
end

end

%*****
%*** Writes a text file summary of the Planetary Properties

Fout = fopen('REFRACTION_SENSOR_Summary.txt','w+');

fprintf(Fout, '\n\n\n *** REFRACTION SENSOR SUMMARY RESULTS ****');
fprintf(Fout, '\n\n\n INPUTS ****');
fprintf(Fout, '\n Unit System:      %s', UNITS);

fprintf(Fout, '\n\n\n **** REFRACTION SENSOR SUMMARY RESULTS ****\n\n');

```

```

% Refraction Sensor Properties
fprintf(Fout, '\n\n** Sensor Properties **');
fprintf(Fout, '\n Refraction Sensor Type:
fprintf(Fout, '\n Sensing Refraction Range - Min:
fprintf(Fout, '\n Sensing Refraction Range - Middle:
fprintf(Fout, '\n Sensing Resolution:
fprintf(Fout, '\n Sensing Drift Rate:
fprintf(Fout, '\n Sensing Accuracy:

%12.5f %s', REF_MATRIX3{1,1});
%12.5f %s', REF_MATRIX3{1,2}, RefUnitLabelStr{2} );
%12.5f %s', REF_MATRIX3{1,3}, RefUnitLabelStr{3} );
%12.4E %s', REF_MATRIX3{1,4}, RefUnitLabelStr{4} );
%12.5f %s', REF_MATRIX3{1,5}, RefUnitLabelStr{5} );
%12.5f %s', REF_MATRIX3{1,6}, RefUnitLabelStr{6} );

fprintf(Fout, '\n\n Power Requirements');
fprintf(Fout, '\n Input Voltage:
fprintf(Fout, '\n Voltage Type:
fprintf(Fout, '\n Input Current:
fprintf(Fout, '\n Input Power:

%12.3f %s', REF_MATRIX3{1,8}, RefUnitLabelStr{8} );
%12.3f %s %s', REF_MATRIX3{1,9}, RefUnitLabelStr{9} );
%12.3f %s', REF_MATRIX3{1,10}, RefUnitLabelStr{10} );
%12.3f %s', REF_MATRIX3{1,11}, RefUnitLabelStr{11} );

fprintf(Fout, '\n\n Physical Properties');
fprintf(Fout, '\n Sensor Mass:
fprintf(Fout, '\n Sensor Dimensions - Length:
fprintf(Fout, '\n Width:
fprintf(Fout, '\n Height:

%12.2f %s', REF_MATRIX3{1,7}, RefUnitLabelStr{7} );
%12.2f %s', REF_MATRIX3{1,12}, RefUnitLabelStr{12} );
%12.2f %s', REF_MATRIX3{1,13}, RefUnitLabelStr{13} );
%12.2f %s', REF_MATRIX3{1,14}, RefUnitLabelStr{14} );

% Environmental Properties
fprintf(Fout, '\n\n Environmental Properties');
fprintf(Fout, '\n Temperature Range - Low:
fprintf(Fout, '\n Temperature Range - High:

%12.2f %s', REF_MATRIX3{1,15}, RefUnitLabelStr{15} );
%12.2f %s', REF_MATRIX3{1,16}, RefUnitLabelStr{16} );

fprintf(Fout, '\n\n Sensor Comments:');
for r = 1:length(REF_MATRIX3{1,17})
    fprintf(Fout, '\n %s', REF_MATRIX3{1,17}{r} );
end
fclose(Fout);

```

## D19. Digital Signal Processing

```

% function DIGITAL_SIGNAL_PROCESSING(DIG_SIG_TYPE, UNITS, Print)
% Digital Signal Processing Database
%
% Developed by: Keith Schreck
% Mechanical and Aerospace Engineering
% San Jose State University
% Date: Fall 2007
%

```





```

if strcmpi(UNITS,'BRITISH')== 1
% Properties [ Sensor
COMPUTATIONAL UNITS [ CYCLE TIME
% Units [ #
Locs Number of Units ns
3 DIGSIGPROC_DB1 = { 'ADSP-2100', 4
'ADSP-2115',
3 40.0 'ADSP-2185', 0
3 13.3 'ADSP-2185', 6
};

% Properties [ OPERATING TEMPERATURE STORAGE TEMPERATURE CPU POWER REQ WEIGHT VOLTAGE
TYPE CURRENT CPU DIMENSIONS - in ]
% Units [ Low deg F High deg F Low deg F High deg F BTU'S/hr lb Min V Max
N/A Length Width Height ]
DIGSIGPROC_DB2 = { -67.0 257.0 -85.0 302.0 2.698 0 -0.3 7.0
'DC' 100 1.332 1.332 0.359 257.0 ; -67.0 257.0 ; -85.0 302.0 0.700 0 -0.3 7.0
'DC' 38 1.110 1.110 0.344 ; -40.0 185.0 ; -85.0 302.0 0.376 0 -0.5 4.0
'DC' 38 0.638 0.638 0.063 } ;

DSPUnitLabelStr1 = { 'sensor' 'N/A' 'MHZ' '#' 'K-Words' '#' 'K-Words' '#' 'ns' '#' 'in' 'in' 'in'
DSPUnitLabelStr2 = { 'deg F' 'deg F' 'deg F' 'deg F' 'BTUs/hr' 'lbm' 'v' 'v' 'N/A' 'mA' 'in' 'in'
};

end

%comments on sensors
DIGSIGPROC_DB3{ 1,1} = {
'Dual Purpose Program Memory for Both Instruction and Data Storage';
'Three Independent Computational Units: ALU, Multiplier/Accumulator and Barrel Shifter' ;
'Two Independent Data Address Shifters' ;
'Powerful Program Sequencer' ;
'Internal Instruction Cache' ;
'Provisions for Multiprecision Computation and Saturation Logic' ;
'Single-Cycle Instruction Execution' ;
'Multifunction Instructions' ;
'APPLICATIONS' ;
'- Optimized for DSP Algorithms including: Digital Filtering, Fast Fourier Transforms' ;
'- Image Processing' ;
'- Radar, Sonar' ;
'- Speech Processing' ;
'- Telecommunications' ;
'- 16-Bit Fixed-Point DSP Microprocessors with On-Chip Memory' ;
'Enhanced Harvard Architecture for Three-Bus Performance: Instruction Bus & Dual Data Buses' ;
'Independent Computational Units: ALU, Multiplier/Accumulator, and Shifter' ;
'Single-Cycle Instruction Execution & Multifunction Instructions' ;
'On-Chip Program Memory RAM or ROM & Data Memory RAM' ;
'Separate On-Chip Buses for Program and Data Memory' ;
'Program Memory Stores Both Instructions and Data (Three-Bus Performance)' ;
'Dual Data Address Generators with Modulo and Bit-Reverse Addressing' ;
};

```

```

Operation' ;
'Dual-Purpose Program Sequencing with Zero-Overhead Looping: Single-Cycle Loop Setup' ;
'Automatic Booting of On-Chip Program Memory from Byte-Wide External Memory (e.g., EPROM)' ;
'Double-Buffered Serial Ports with Commanding Hardware, Automatic Data Buffering, and Multichannel
Operation' ;
'Three Edge- or Level-Sensitive Interrupts' ;
'Low Power Idle Instruction' ;
'MIL-STD-883B Versions Available' ;
'DIGSIGPROC_DB3{ 3,1} = {
'Single-Cycle Instruction Execution' ;
'Single-Cycle Context Switch' ;
'3-Bus Architecture Allows Dual Operand Fetches in Every Instruction Cycle' ;
'Multifunction Instructions' ;
'Power-Down Mode Featuring Low CMOS Standby Power Dissipation with 200 CLKIN Cycle Recovery from
Power-Down Condition' ;
'Low Power Dissipation in Idle Mode' ;
'80K Bytes of On-Chip RAM, Configured as 16K Words Program Memory RAM 16K Words Data Memory RAM' ;
'Dual-Purpose Program Memory for Both Instruction and Data Storage' ;
'Independent ALU, Multiplier/Accumulator, and Barrel Shifter Computational Units' ;
'Two Independent Data Address Generators' ;
'Powerful Program Sequencer Provides Zero Overhead Looping Conditional Instruction Execution' ;
'Programmable 16-Bit Interval Timer with Prescaler' ;
};

% Combine Refraction Sensor Arrays
DIGSIGPROC_DB = cat(2,DIGSIGPROC_DB1,DIGSIGPROC_DB2,DIGSIGPROC_DB3);
DSPUnitLabelStr = cat(2,DSPUnitLabelStr1,DSPUnitLabelStr2);

tmp = 1;
for DSP = 1:size(DIGSIGPROC_DB,1)
    if strcmpi(DIGSIGPROC_DB{DSP,2},DIG_SIG_TYPE) == 1
        for s = 1:size(DIGSIGPROC_DB,2)
            DSP_MATRIX{tmp,s} = DIGSIGPROC_DB{DSP,s};
        end
        tmp = tmp + 1;
    end
end

% Select Processor Based on Temp Range if More than One Sensor Exists In Database
if size(DSP_MATRIX,1) > 1
    % Find the Max Operational Range
    for n = 1:size(DSP_MATRIX,1)
        DSP_DELTA_RNG(n) = DSP_MATRIX{n,13} - DSP_MATRIX{n,12};
    end
    % Find the Maximum Refraction Operating Range of the Sensor
    MAX_DSP_DELTA_RNG = max(DSP_DELTA_RNG);
    % For Multiple Sensors with the Same Range Reduce Sensor Matrix to Those
    % with the Greatest Range
    k = 1;
    for c = 1:length(DSP_DELTA_RNG)
        if DSP_DELTA_RNG(c) == MAX_DSP_DELTA_RNG

```





```

fprintf(Fout, '\n          Max:
fprintf(Fout, '\n Voltage Type:
fprintf(Fout, '\n Input Current:

fprintf(Fout, '\n\n Physical Properties');
fprintf(Fout, '\n Processor Core Mass:
fprintf(Fout, '\n Sensor Dimensions - Length:
fprintf(Fout, '\n Width:
fprintf(Fout, '\n Height:

% Environmental Properties
fprintf(Fout, '\n\n Environmental Properties');
fprintf(Fout, '\n Operating Temperature Range - Low:
fprintf(Fout, '\n High:
fprintf(Fout, '\n Storage Temperature Range - Low:
fprintf(Fout, '\n High:

fprintf(Fout, '\n\n Sensor Comments:');
for r = 1:length(DSP_MATRIX2{1,25})
    fprintf(Fout, '\n %s', DSP_MATRIX2{1,25}{r} );
end
fclose(Fout);

%12.3f %s', DSP_MATRIX2{1,19},DSPUnitLabelStr{19} );
%12.3f %s %s' DSP_MATRIX2{1,20},DSPUnitLabelStr{20} );
%12.3f %s', DSP_MATRIX2{1,21},DSPUnitLabelStr{21} );

%12.2f %s', DSP_MATRIX2{1,17},DSPUnitLabelStr{17} );
%12.3f %s', DSP_MATRIX2{1,22},DSPUnitLabelStr{22} );
%12.3f %s', DSP_MATRIX2{1,23},DSPUnitLabelStr{23} );
%12.3f %s', DSP_MATRIX2{1,24},DSPUnitLabelStr{24} );

%12.2f %s', DSP_MATRIX2{1,12},DSPUnitLabelStr{12} );
%12.2f %s', DSP_MATRIX2{1,13},DSPUnitLabelStr{13} );
%12.2f %s', DSP_MATRIX2{1,14},DSPUnitLabelStr{14} );
%12.2f %s', DSP_MATRIX2{1,15},DSPUnitLabelStr{15} );

```

## D20. Error Program

```
% function ERROR_PRG(ERR_Code)
%   ERROR Code Handling Program
%
%   Developed by: Keith Schreck
%                 Mechanical and Aerospace Engineering
%                 San Jose State University
%   Date:        Fall 2007
%
%   This file contains ERROR code handles for the ISSPO program.
%   When an ERROR case flag is detected within the main ISSPO program
%   this program is called with the appropriate ERR_Code. A message as
%   to the error that occurred and the point in the program at which it
%   occurred is printed to the screen and the program either waits for
%   corrected input or simply exits. Along with the error message is a
%   possible cause of the problem or a method to fix it. In most cases
%   the error is the result of an unexpected input not matching a pre-
%   defined match in the program. This is usually the result of a
%   erroneous input to the program and must be remedied by the user.
%
function ERROR_PRG(ERR_Code)

% DISPLAY ERROR CODE NUMBER
fprintf('\n\n *** ERROR CODE:  %d ***',ERR_Code);

% Display Error Code Statement
switch(ERR_Code)
%   CASE X  ERRORS - PROGRAMATIC ERRORS DATA FILES DO NOT EXIST
case 1      % ERROR 1 - INPUT FILE DOES NOT EXIST
    disp('ERROR:  Declared INPUT FILE does not exist!');
    disp('  Verify INPUT FILE exists before executing ISSPO!');
    disp('***  TERMINATING ISSPO PROGRAM  ***');

case 2      % ERROR 2 - INPUTDECK FILE DID NOT LOAD SUCCESSFULLY
    disp('ERROR:  Loaded InputDeck File does not exist!');
    disp('  Program erred off in loading InputDeck file. ');
    disp('  Verify INPUT FILE is successfully loaded into InputDeck. ');
    disp('***  TERMINATING ISSPO PROGRAM  ***');
    % Return to Main Directory
    cd ..\..\..;

case 3      % ERROR 3 - CONSTANTS DATABASE FILE FAILED TO BE SUCCESSFULLY CREATED
    disp('ERROR:  Constants Database File does not exist!');
    disp('  Program erred off in creating Constants database file. ');
    disp('  Verify Constants program is correctly creating the database file. ');
    disp('***  TERMINATING ISSPO PROGRAM  ***');
    % Return to Main Directory
    cd ..\..\..;

case 4      % ERROR 4 - PLANETARY DATABASE FILE FAILED TO BE SUCCESSFULLY CREATED
    disp('ERROR:  Planetary Database File does not exist!');
    disp('  Program erred off in creating Planetary database file. ');
    disp('  Verify Planetary Database program is correctly creating the database
file. ');
    disp('***  TERMINATING ISSPO PROGRAM  ***');
    % Return to Main Directory
    cd ..\..\..;

%   CASE 5X ERRORS UNKNOWN VARIABLES LOADED INTO PROGRAM
case 50     % CASE 50 ERROR UNIDENTIFIED UNIT TYPE IN INPUT FILE
    disp('ERROR:  UNKNOWN UNIT TYPE IN INPUT FILE!');
    disp('  Unknown Unit Type entered into Input File. ');
    disp('  Verify UNIT is 'SI' or 'British' in Input File. ');
    disp('***  TERMINATING ISSPO PROGRAM  ***');
    % Return to Main Directory
    cd ..\..\..;

case 51     % CASE 51 ERROR UNIDENTIFIED PRINTFLG IN INPUT FILE
    disp('ERROR:  UNKNOWN PRINT FLAG IN INPUT FILE!');
    disp('  Unknown PRINTFLG variable entered into Input File. ');
    disp('  Verify PRINTFLG is 'Y' or 'N' in Input File. ');
    disp('***  TERMINATING ISSPO PROGRAM  ***');
    % Return to Main Directory
    cd ..\..\..;

case 52     % CASE 52 ERROR UNIDENTIFIED DATA TYPE IN SENSOR_DATA
    disp('ERROR:  UNKNOWN SENSOR_DATA ARRAY DATA TYPE!');
```

```

disp(' Program erred off in DataTypeVerifier Program. ');
disp(' Verify only allowed Data Types are entered in Input File. ');
disp(' Data Input Types are case sensitive. Values must be entered in ALL
CAPS. ');
disp('*** TERMINATING ISSPO PROGRAM ***');
% Return to Main Directory
cd ..\..\..\..;
case 53 % CASE 53 ERROR MASS_LIMIT VARIABLE UNDEFINED IN SENSOR_DATA
disp('ERROR: UNDEFINED MASS_LIMIT VARIABLE IN INPUT FILE!');
disp(' Program erred off in DataTypeVerifier Program. ');
disp(' Verify MASS_LIMIT Variable is defined in Input File. ');
disp('*** TERMINATING ISSPO PROGRAM ***');
% Return to Main Directory
cd ..\..\..\..;
case 54 % CASE 54 ERROR POWER_LIMIT VARIABLE UNDEFINED IN SENSOR_DATA
disp('ERROR: UNDEFINED POWER_LIMIT VARIABLE IN INPUT FILE!');
disp(' Program erred off in DataTypeVerifier Program. ');
disp(' Verify POWER_LIMIT Variable is defined in Input File. ');
disp('*** TERMINATING ISSPO PROGRAM ***');
% Return to Main Directory
cd ..\..\..\..;
case 55 % CASE 55 ERROR VOLUME_LIMIT VARIABLE UNDEFINED IN SENSOR_DATA
disp('ERROR: UNDEFINED VOLUME_LIMIT VARIABLE IN INPUT FILE!');
disp(' Program erred off in DataTypeVerifier Program. ');
disp(' Verify VOLUME_LIMIT Variable is defined in Input File. ');
disp('*** TERMINATING ISSPO PROGRAM ***');
% Return to Main Directory
cd ..\..\..\..;
case 56 % CASE 56 ERROR SENSOR OPTION REQUIRES ADDITIONAL DATA IN SENSOR_DATA
disp('ERROR: SENSOR OPTION REQUIRES ADDITIONAL DATA IN SENSOR_DATA ARRAY!');
disp(' Program erred off in DataTypeVerifier Program. ');
disp(' Selected Sensor Option Requires Additional Design Information. ');
disp(' Verify Proper Data Input Format for each Sensor Type. ');
disp('*** TERMINATING ISSPO PROGRAM ***');
% Return to Main Directory
cd ..\..\..\..;
case 57 % CASE 57 ERROR SENSOR OPTION OPTICS REQUIRES ADDITIONAL DATA IN
SENSOR_DATA
disp('ERROR: SENSOR OPTION OPTICS REQUIRES ADDITIONAL DATA IN SENSOR_DATA
ARRAY!');
disp(' Program erred off in DataTypeVerifier Program. ');
disp(' Selected Sensor Option 'OPTICS' Requires Additional Design
Information. ');
disp(' Verify Proper Data Input Format for each Sensor Type. ');
disp('*** TERMINATING ISSPO PROGRAM ***');
% Return to Main Directory
cd ..\..\..\..;
case 58 % CASE 58 ERROR SENSOR OPTION OPTICS REQUIRES ADDITIONAL DATA IN
SENSOR_DATA
disp('ERROR: SENSOR OPTION OPTICS REQUIRES ADDITIONAL DATA IN SENSOR_DATA
ARRAY!');
disp(' Program erred off in DataTypeVerifier Program. ');
disp(' Selected Sensor Option 'OPTICS' Requires Additional Design
Information. ');
disp(' Verify Proper Data Input Format for each Sensor Type. ');
disp('*** TERMINATING ISSPO PROGRAM ***');
% Return to Main Directory
cd ..\..\..\..;
case 59 % CASE 59 ERROR SENSOR OPTION OPTICS REQUIRES SYSTEM TYPE IN SENSOR_DATA
disp('ERROR: SENSOR OPTION OPTICS REQUIRES SYSTEM TYPE IN SENSOR_DATA ARRAY!');
disp(' Program erred off in DataTypeVerifier Program. ');
disp(' Selected Sensor Option 'OPTICS' Requires Additional Design
Information. ');
disp(' Verify Optical Sensor System Type is defined in 2nd Column position. ');
disp(' Specify Resolution as: 'CAMERA' 'ARRAY' 'LINEAR' ');
disp('*** TERMINATING ISSPO PROGRAM ***');
% Return to Main Directory
cd ..\..\..\..;
case 60 % CASE 60 ERROR SENSOR OPTION OPTICS REQUIRES RESOLUTION DATA IN
SENSOR_DATA
disp('ERROR: SENSOR OPTION OPTICS REQUIRES RESOLUTION DATA IN SENSOR_DATA
ARRAY!');
disp(' Program erred off in DataTypeVerifier Program. ');
disp(' Selected Sensor Option 'OPTICS' Requires Additional Design
Information. ');
disp(' Verify Optical Sensor Resolution is defined in 3rd Column position. ');

```

```

disp(' Specify Resolution as: 'LOW' 'MEDIUM' 'HIGH' ');
disp('*** TERMINATING ISSPO PROGRAM ***');
% Return to Main Directory
cd ../../..;
case 61 % CASE 61 ERROR SENSOR OPTION OPTICS UNDEFINED IMAGE TYPE IN SENSOR DATA
disp('ERROR: SENSOR OPTION OPTICS UNDEFINED IMAGE TYPE IN SENSOR_DATA ARRAY!');
disp(' Program erred off in DataTypeVerifier Program. ');
disp(' Selected Sensor Option 'OPTICS' Requires Additional Design
Information. ');
disp(' Undefined Image Option Type in Optical Sensor Data. ');
disp(' Specify Image Type as: 'X-RAY' 'VISUAL' 'UV' 'NIR'
'MICRO' ');
disp('*** TERMINATING ISSPO PROGRAM ***');
% Return to Main Directory
cd ../../..;
case 62 % CASE 62 ERROR SENSOR OPTION RADIATION REQUIRES ADDITIONAL DATA IN
SENSOR_DATA
disp('ERROR: SENSOR OPTION RADIATION REQUIRES ADDITIONAL DATA IN SENSOR_DATA
ARRAY!');
disp(' Program erred off in DataTypeVerifier Program. ');
disp(' Selected Sensor Option 'RADIATION' Requires Additional Design
Information. ');
disp(' Verify Proper Data Input Format for each Sensor Type. ');
disp('*** TERMINATING ISSPO PROGRAM ***');
% Return to Main Directory
cd ../../..;
case 63 % CASE 63 ERROR SENSOR OPTION OPTICS UNDEFINED RADIATION TYPE IN
SENSOR_DATA
disp('ERROR: SENSOR OPTION OPTICS UNDEFINED RADIATION TYPE IN SENSOR_DATA
ARRAY!');
disp(' Program erred off in DataTypeVerifier Program. ');
disp(' Selected Sensor Option 'RADIATION' Requires Additional Design
Information. ');
disp(' Undefined RADIATION Option Type in Radiation Sensor Data. ');
disp(' Specify Image Type as: 'Charged Particle' 'Alpha' 'Beta'
'Gamma' 'X-Ray' ');
disp('*** TERMINATING ISSPO PROGRAM ***');
% Return to Main Directory
cd ../../..;
case 64 % CASE 64 ERROR SENSOR OPTION INCLINATION REQUIRES ADDITIONAL DATA IN
SENSOR_DATA
disp('ERROR: SENSOR OPTION INCLINATION REQUIRES ADDITIONAL DATA IN SENSOR_DATA
ARRAY!');
disp(' Program erred off in DataTypeVerifier Program. ');
disp(' Selected Sensor Option 'INCLINATION' Requires Additional Design
Information. ');
disp(' Verify Proper Data Input Format for each Sensor Type. ');
disp('*** TERMINATING ISSPO PROGRAM ***');
% Return to Main Directory
cd ../../..;
case 65 % CASE 65 ERROR SENSOR OPTION INCLINATION REQUIRES RESOLUTION DATA IN
SENSOR_DATA
disp('ERROR: SENSOR OPTION INCLINATION REQUIRES ANGLE DATA IN SENSOR_DATA
ARRAY!');
disp(' Program erred off in DataTypeVerifier Program. ');
disp(' Selected Sensor Option 'INCLINATION' Requires Additional Design
Information. ');
disp(' Verify Inclination Sensor Range is defined in 2nd Column position. ');
disp(' Specify Range as: 'LOW' 'MEDIUM' 'HIGH' ');
disp('*** TERMINATING ISSPO PROGRAM ***');
% Return to Main Directory
cd ../../..;
case 66 % CASE 66 ERROR SENSOR OPTION ACCELERATION REQUIRES ADDITIONAL DATA IN
SENSOR_DATA
disp('ERROR: SENSOR OPTION ACCELERATION REQUIRES ADDITIONAL DATA IN SENSOR_DATA
ARRAY!');
disp(' Program erred off in DataTypeVerifier Program. ');
disp(' Selected Sensor Option 'ACCELERATION' Requires Additional Design
Information. ');
disp(' Verify Proper Data Input Format for each Sensor Type. ');
disp('*** TERMINATING ISSPO PROGRAM ***');
% Return to Main Directory
cd ../../..;
case 67 % CASE 67 ERROR SENSOR OPTION ACCELERATION REQUIRES RANGE DATA IN
SENSOR_DATA

```



```

        disp('ERROR:  SENSOR OPTION ACCELERATION REQUIRES ACCELERATION DATA IN
SENSOR_DATA ARRAY!');
        disp('  Program erred off in DataTypeVerifier Program. ');
        disp('  Selected Sensor Option 'ACCELERATION' Requires Additional Design
Information.');
```

```

        disp('  Verify Acceleration Sensor Range is defined in 2nd Column position.');
```

```

        disp('  Specify Range as:  'SOFT'  'MEDIUM'  'HIGH'  'IMPACT'
'BALLISTIC'  ');
        disp('***  TERMINATING ISSPO PROGRAM  ***');
```

```

        % Return to Main Directory
        cd ..\..\..;
        case 68 % CASE 68 ERROR SENSOR OPTION WIND VELOCITY REQUIRES ADDITIONAL DATA IN
SENSOR_DATA
        disp('ERROR:  SENSOR OPTION ATMOSPHERE REQUIRES ADDITIONAL DATA IN SENSOR_DATA
ARRAY!');
```

```

        disp('  Program erred off in DataTypeVerifier Program. ');
        disp('  Selected Sensor Option 'WIND VELOCITY' Requires Additional Design
Information.');
```

```

        disp('  Verify Proper Data Input Format for each Sensor Type.');
```

```

        disp('***  TERMINATING ISSPO PROGRAM  ***');
```

```

        % Return to Main Directory
        cd ..\..\..;
        case 69 % CASE 69 ERROR SENSOR OPTION WIND VELOCITY REQUIRES ADDITIONAL DATA IN
SENSOR_DATA
        disp('ERROR:  SENSOR OPTION ATMOSPHERE REQUIRES ADDITIONAL DATA IN SENSOR_DATA
ARRAY!');
```

```

        disp('  Program erred off in DataTypeVerifier Program. ');
        disp('  Selected Sensor Option 'WIND VELOCITY' Requires Additional Design
Information.');
```

```

        disp('  Verify Wind Velocity Sensor type is defined in the next Column
position.');
```

```

        disp('  Specify Type as:  'ANEMOMETER' or 'DOPPLER'  ');
        disp('***  TERMINATING ISSPO PROGRAM  ***');
```

```

        % Return to Main Directory
        cd ..\..\..;
        case 70 % CASE 70 ERROR SENSOR OPTION GAS ANALYSIS REQUIRES ADDITIONAL DATA IN
SENSOR_DATA
        disp('ERROR:  SENSOR OPTION GAS ANALYSIS REQUIRES ADDITIONAL DATA IN SENSOR_DATA
ARRAY!');
```

```

        disp('  Program erred off in DataTypeVerifier Program. ');
        disp('  Selected Sensor Option 'GAS ANALYSIS' Requires Additional Design
Information.');
```

```

        disp('  Verify GCMS Sensor type is defined in the 2nd Column position.');
```

```

        disp('  Specify Type as:  'WAVELENGTH' or 'MASS-CHARGE'  ');
        disp('***  TERMINATING ISSPO PROGRAM  ***');
```

```

        % Return to Main Directory
        cd ..\..\..;
        case 71 % CASE 71 ERROR SENSOR OPTION GAS ANALYSIS REQUIRES ADDITIONAL DATA IN
SENSOR_DATA
        disp('ERROR:  SENSOR OPTION GAS ANALYSIS REQUIRES ADDITIONAL DATA IN SENSOR_DATA
ARRAY!');
```

```

        disp('  Program erred off in DataTypeVerifier Program. ');
        disp('  Selected Sensor Option 'GAS ANALYSIS' Requires Additional Design
Information.');
```

```

        disp('  Selected Sensing Type 'MASS-CHARGE' Option Requires input in the 3rd
Column.');
```

```

        disp('  Verify Proper Data Input Format for each Sensor Type.');
```

```

        disp('***  TERMINATING ISSPO PROGRAM  ***');
```

```

        % Return to Main Directory
        cd ..\..\..;
        case 72 % CASE 72 ERROR SENSOR OPTION GAS ANALYSIS REQUIRES ADDITIONAL DATA IN
SENSOR_DATA
        disp('ERROR:  SENSOR OPTION GAS ANALYSIS REQUIRES ADDITIONAL DATA IN SENSOR_DATA
ARRAY!');
```

```

        disp('  Program erred off in DataTypeVerifier Program. ');
        disp('  Selected Sensor Option 'GAS ANALYSIS' Requires Additional Design
Information.');
```

```

        disp('  Selected Sensing Type 'MASS-CHARGE' Option Requires input in the 3rd
Column.');
```

```

        disp('  Verify Gas Analysis Sensor Range is defined in 3rd Column position.');
```

```

        disp('  Specify Range as:  'LOW'  'MEDIUM'  'HIGH'  ');
        disp('***  TERMINATING ISSPO PROGRAM  ***');
```

```

        % Return to Main Directory
        cd ..\..\..;
        case 73 % CASE 73 ERROR SENSOR OPTION ACCELERATION REQUIRES ADDITIONAL DATA IN
SENSOR_DATA

```

```

disp('ERROR:  SENSOR OPTION ACCELERATION REQUIRES ADDITIONAL DATA IN SENSOR_DATA
ARRAY!');
disp('  Program erred off in DataTypeVerifier Program. ');
disp('  Selected Sensor Option ''ACCELERATION'' Requires Additional Design
Information.');
```

```

disp('  Verify Acceleration Range in 2nd Column Position, Power Type in 3rd
Column.');
```

```

disp('  Verify Proper Data Input Format for each Sensor Type.');
```

```

disp('*** TERMINATING ISSPO PROGRAM ***');
```

```

% Return to Main Directory
cd ..\..\..\..;
case 74 % CASE 74 ERROR SENSOR OPTION ACCELERATION REQUIRES POWER TYPE DATA IN
SENSOR_DATA
disp('ERROR:  SENSOR OPTION ACCELERATION REQUIRES ACCELERATION DATA IN
SENSOR_DATA ARRAY!');
```

```

disp('  Program erred off in DataTypeVerifier Program. ');
disp('  Selected Sensor Option ''ACCELERATION'' Requires Additional Design
Information.');
```

```

disp('  Verify Acceleration Sensor Power Type is defined in 3rd Column
position.');
```

```

disp('  Specify Type as:  ''Voltage''  ''Const Current''  ''Self Generating''
');
```

```

disp('*** TERMINATING ISSPO PROGRAM ***');
```

```

% Return to Main Directory
cd ..\..\..\..\..;
case 75 % CASE 75 ERROR SENSOR OPTION TEMPERATURE REQUIRES ADDITIONAL DATA IN
SENSOR_DATA
disp('ERROR:  SENSOR OPTION ATMOSPHERE REQUIRES ADDITIONAL DATA IN SENSOR_DATA
ARRAY!');
```

```

disp('  Program erred off in DataTypeVerifier Program. ');
disp('  Selected Sensor Option ''TEMPERATURE'' Requires Additional Design
Information.');
```

```

disp('  Verify Proper Data Input Format for each Sensor Type.');
```

```

disp('*** TERMINATING ISSPO PROGRAM ***');
```

```

% Return to Main Directory
cd ..\..\..\..\..;
case 76 % CASE 76 ERROR SENSOR OPTION TEMPERATURE REQUIRES ADDITIONAL DATA IN
SENSOR_DATA
disp('ERROR:  SENSOR OPTION TEMPERATURE REQUIRES ADDITIONAL DATA IN SENSOR_DATA
ARRAY!');
```

```

disp('  Program erred off in DataTypeVerifier Program. ');
disp('  Selected Sensor Option ''TEMPERATURE'' Requires Additional Design
Information.');
```

```

disp('  Verify Temperature Sensor type is defined in the next Column
position.');
```

```

disp('  Specify Type as:  ''VOLTAGE'' or ''RESISTANCE''  ');
disp('*** TERMINATING ISSPO PROGRAM ***');
```

```

% Return to Main Directory
cd ..\..\..\..\..;
case 77 % CASE 77 ERROR SENSOR OPTION ACOUSTICS REQUIRES ADDITIONAL DATA IN
SENSOR_DATA
disp('ERROR:  SENSOR OPTION ACOUSTICS REQUIRES ADDITIONAL DATA IN SENSOR_DATA
ARRAY!');
```

```

disp('  Program erred off in DataTypeVerifier Program. ');
disp('  Selected Sensor Option ''ACOUSTICS'' Requires Additional Design
Information.');
```

```

disp('  Verify Proper Data Input Format for each Sensor Type.');
```

```

disp('*** TERMINATING ISSPO PROGRAM ***');
```

```

% Return to Main Directory
cd ..\..\..\..\..;
case 78 % CASE 78 ERROR SENSOR OPTION ACOUSTICS REQUIRES ADDITIONAL DATA IN
SENSOR_DATA
disp('ERROR:  SENSOR OPTION ACOUSTICS REQUIRES ADDITIONAL DATA IN SENSOR_DATA
ARRAY!');
```

```

disp('  Program erred off in DataTypeVerifier Program. ');
disp('  Selected Sensor Option ''ACOUSTICS'' Requires Additional Design
Information.');
```

```

disp('  Verify Acoustic Sensor type is defined in the next 2nd position.');
```

```

disp('  Specify Type as:  ''SENSOR'' or ''ARRAY''  ');
disp('*** TERMINATING ISSPO PROGRAM ***');
```

```

% Return to Main Directory
cd ..\..\..\..\..;
case 79 % CASE 79 ERROR SENSOR OPTION DENSITY REQUIRES ADDITIONAL DATA IN
SENSOR_DATA
disp('ERROR:  SENSOR OPTION ATMOSPHERE REQUIRES ADDITIONAL DATA IN SENSOR_DATA
ARRAY!');
```

```

disp(' Program erred off in DataTypeVerifier Program. ');
disp(' Selected Sensor Option 'DENSITY' Requires Additional Design
Information. ');
disp(' Verify Proper Data Input Format for each Sensor Type. ');
disp('*** TERMINATING ISSPO PROGRAM ***');
% Return to Main Directory
cd ..\..\..\..;
case 80 % CASE 80 ERROR SENSOR OPTION SENSITY REQUIRES ADDITIONAL DATA IN
SENSOR_DATA
disp('ERROR: SENSOR OPTION DENSITY REQUIRES ADDITIONAL DATA IN SENSOR_DATA
ARRAY! ');
disp(' Program erred off in DataTypeVerifier Program. ');
disp(' Selected Sensor Option 'DENSITY' Requires Additional Design
Information. ');
disp(' Verify Density Sensor type is defined in the next Column position. ');
disp(' Specify Type as: 'GAS' or 'LIQUID' ');
disp('*** TERMINATING ISSPO PROGRAM ***');
% Return to Main Directory
cd ..\..\..\..\..;
case 81 % CASE 81 ERROR OPTION DIGITAL SIGNAL PROCESSING REQUIRES ADDITIONAL
DATA IN SENSOR_DATA
disp('ERROR: OPTION DATA PROCESSING REQUIRES ADDITIONAL DATA IN SENSOR_DATA
ARRAY! ');
disp(' Program erred off in DataTypeVerifier Program. ');
disp(' Selected Sensor Option 'DATA PROCESSING' Requires Additional Design
Information. ');
disp(' Verify Proper Data Input Format for each Sensor Type. ');
disp('*** TERMINATING ISSPO PROGRAM ***');
% Return to Main Directory
cd ..\..\..\..\..;
case 82 % CASE 82 ERROR OPTION DIGITAL SIGNAL PROCESSING REQUIRES ADDITIONAL
DATA IN SENSOR_DATA
disp('ERROR: OPTION DATA PROCESSING REQUIRES ADDITIONAL DATA IN SENSOR_DATA
ARRAY! ');
disp(' Program erred off in DataTypeVerifier Program. ');
disp(' Selected Sensor Option 'DATA PROCESSING' Requires Additional Design
Information. ');
disp(' Verify CPU SPEED in 2nd Column Position, NUMBER OF DATA INPUTS in 3rd
Column. ');
disp(' Verify Proper Data Input Format for each Sensor Type. ');
disp('*** TERMINATING ISSPO PROGRAM ***');
% Return to Main Directory
cd ..\..\..\..\..;
case 83 % CASE 83 ERROR OPTION DIGITAL SIGNAL PROCESSING REQUIRES ADDITIONAL
DATA IN SENSOR_DATA
disp('ERROR: OPTION DATA PROCESSING REQUIRES ADDITIONAL DATA IN SENSOR_DATA
ARRAY! ');
disp(' Program erred off in DataTypeVerifier Program. ');
disp(' Selected Sensor Option 'DATA PROCESSING' Requires Additional Design
Information. ');
disp(' Verify CPU SPEED is defined in 2nd Column position. ');
disp(' Specify Range as: 'LOW' 'MEDIUM' 'HIGH' ');
disp('*** TERMINATING ISSPO PROGRAM ***');
% Return to Main Directory
cd ..\..\..\..\..;
case 84 % CASE 84 ERROR OPTION DIGITAL SIGNAL PROCESSING REQUIRES ADDITIONAL
DATA IN SENSOR_DATA
disp('ERROR: OPTION DATA PROCESSING REQUIRES ADDITIONAL DATA IN SENSOR_DATA
ARRAY! ');
disp(' Program erred off in DataTypeVerifier Program. ');
disp(' Selected Sensor Option 'DATA PROCESSING' Requires Additional Design
Information. ');
disp(' Non-Numeric Number detected in SENSOR DATA Array 3rd Column. ');
disp(' Verify NUMBER OF DATA INPUTS expected is defined in 3rd Column
position. ');
disp('*** TERMINATING ISSPO PROGRAM ***');
% Return to Main Directory
cd ..\..\..\..\..;

% CASE 1XX SENSOR RESULT DATA FILES NOT CREATED PROPERLY
case 100 % ERROR 100 - TEMPERATURE SENSOR SUMMARY DATA FILE FAILED TO BE CREATED
disp('ERROR: TEMP SENSOR Database File Failed to be Created! ');
disp(' Program erred off in TEMP_SENSORS subroutine. ');
disp(' Verify TEMP_SENSORS program is correctly creating the database file. ');
disp('*** TERMINATING ISSPO PROGRAM ***');
% Return to Main Directory

```

```

        cd ..\..\..\..;
    case 101 % ERROR 101 - PRESSURE SENSOR SUMMARY DATA FILE FAILED TO BE CREATED
    disp('ERROR: PRESSURE SENSOR Database File Failed to be Created!');
    disp(' Program erred off in PRESSURE_SENSORS subroutine. ');
    disp(' Verify PRESSURE_SENSORS program is correctly creating the database
file. ');
    disp('*** TERMINATING ISSPO PROGRAM ***');
    % Return to Main Directory
    cd ..\..\..\..;
    case 102 % ERROR 102 - NEPHELOMETER SENSOR SUMMARY DATA FILE FAILED TO BE CREATED
    disp('ERROR: NEPHELOMETER SENSOR Database File Failed to be Created!');
    disp(' Program erred off in NEPHELOMETER_SENSORS subroutine. ');
    disp(' Verify NEPHELOMETER_SENSORS program is correctly creating the database
file. ');
    disp('*** TERMINATING ISSPO PROGRAM ***');
    % Return to Main Directory
    cd ..\..\..\..;
    case 103 % ERROR 103 - HUMIDITY SENSOR SUMMARY DATA FILE FAILED TO BE CREATED
    disp('ERROR: HUMIDITY SENSOR Database File Failed to be Created!');
    disp(' Program erred off in HUMIDITY_SENSORS subroutine. ');
    disp(' Verify HUMIDITY_SENSORS program is correctly creating the database
file. ');
    disp('*** TERMINATING ISSPO PROGRAM ***');
    % Return to Main Directory
    cd ..\..\..\..;
    case 104 % ERROR 104 - DENSITY SENSOR SUMMARY DATA FILE FAILED TO BE CREATED
    disp('ERROR: DENSITY SENSOR Database File Failed to be Created!');
    disp(' Program erred off in DENSITY_SENSORS subroutine. ');
    disp(' Verify DENSITY_SENSORS program is correctly creating the database
file. ');
    disp('*** TERMINATING ISSPO PROGRAM ***');
    % Return to Main Directory
    cd ..\..\..\..;
    case 105 % ERROR 105 - ACOUSTIC SENSOR SUMMARY DATA FILE FAILED TO BE CREATED
    disp('ERROR: ACOUSTIC SENSOR Database File Failed to be Created!');
    disp(' Program erred off in ACOUSTIC_SENSORS subroutine. ');
    disp(' Verify ACOUSTIC_SENSORS program is correctly creating the database
file. ');
    disp('*** TERMINATING ISSPO PROGRAM ***');
    % Return to Main Directory
    cd ..\..\..\..;
    case 106 % ERROR 106 - AC FIELD SENSOR SUMMARY DATA FILE FAILED TO BE CREATED
    disp('ERROR: AC FIELD SENSOR Database File Failed to be Created!');
    disp(' Program erred off in AC_FIELD_SENSORS subroutine. ');
    disp(' Verify AC_FIELD_SENSORS program is correctly creating the database
file. ');
    disp('*** TERMINATING ISSPO PROGRAM ***');
    % Return to Main Directory
    cd ..\..\..\..;
    case 107 % ERROR 107 - WIND VELOCITY SENSOR SUMMARY DATA FILE FAILED TO BE
CREATED
    disp('ERROR: WIND VELOCITY SENSOR Database File Failed to be Created!');
    disp(' Program erred off in WIND_VELOCITY_SENSORS subroutine. ');
    disp(' Verify WIND_VELOCITY_SENSORS program is correctly creating the database
file. ');
    disp('*** TERMINATING ISSPO PROGRAM ***');
    % Return to Main Directory
    cd ..\..\..\..;
    case 108 % ERROR 108 - ACCELEROMETER SENSOR SUMMARY DATA FILE FAILED TO BE
CREATED
    disp('ERROR: ACCELEROMETER SENSOR Database File Failed to be Created!');
    disp(' Program erred off in ACCCELEROMETER_SENSORS subroutine. ');
    disp(' Verify ACCCELEROMETER_SENSORS program is correctly creating the database
file. ');
    disp('*** TERMINATING ISSPO PROGRAM ***');
    % Return to Main Directory
    cd ..\..\..\..;
    case 109 % ERROR 109 - GCMS SENSOR SUMMARY DATA FILE FAILED TO BE CREATED
    disp('ERROR: GCMS SENSOR Database File Failed to be Created!');
    disp(' Program erred off in GCMS_SENSORS subroutine. ');
    disp(' Verify GCMS_SENSORS program is correctly creating the database file. ');
    disp('*** TERMINATING ISSPO PROGRAM ***');
    % Return to Main Directory
    cd ..\..\..\..;
    case 110 % ERROR 110 - RADIATION SENSOR SUMMARY DATA FILE FAILED TO BE CREATED
    disp('ERROR: RADIATION SENSOR Database File Failed to be Created!');

```

```

disp('  Program erred off in RADIATION_SENSORS subroutine. ');
disp('  Verify RADIATION_SENSORS program is correctly creating the database
file.');
```

```

disp('*** TERMINATING ISSPO PROGRAM ***');
% Return to Main Directory
cd ..\..\..\..;
case 111 % ERROR 111 - INCLINOMETER SENSOR SUMMARY DATA FILE FAILED TO BE CREATED
disp('ERROR:  INCLINOMETER SENSOR Database File Failed to be Created!');
disp('  Program erred off in INCLINOMETER_SENSORS subroutine. ');
disp('  Verify INCLINOMETER_SENSORS program is correctly creating the database
file.');
```

```

disp('*** TERMINATING ISSPO PROGRAM ***');
% Return to Main Directory
cd ..\..\..\..;
case 112 % ERROR 112 - IMAGING SENSOR SUMMARY DATA FILE FAILED TO BE CREATED
disp('ERROR:  IMAGING SENSOR Database File Failed to be Created!');
disp('  Program erred off in IMAGING_SENSORS subroutine. ');
disp('  Verify IMAGING_SENSORS program is correctly creating the database
file.');
```

```

disp('*** TERMINATING ISSPO PROGRAM ***');
% Return to Main Directory
cd ..\..\..\..;
case 113 % ERROR 113 - REFRACTION SENSOR SUMMARY DATA FILE FAILED TO BE CREATED
disp('ERROR:  REFRACTION SENSOR Database File Failed to be Created!');
disp('  Program erred off in REFRACTION_SENSORS subroutine. ');
disp('  Verify REFRACTION_SENSORS program is correctly creating the database
file.');
```

```

disp('*** TERMINATING ISSPO PROGRAM ***');
% Return to Main Directory
cd ..\..\..\..;
case 114 % ERROR 114 - DATA PROCESSING SUMMARY DATA FILE FAILED TO BE CREATED
disp('ERROR:  DATA PROCESSING Database File Failed to be Created!');
disp('  Program erred off in DIGITAL_SIGNAL_PROCESSING subroutine. ');
disp('  Verify DIGITAL_SIGNAL_PROCESSING program is correctly creating the
database file.');
```

```

disp('*** TERMINATING ISSPO PROGRAM ***');
% Return to Main Directory
cd ..\..\..\..;

% CASE 2XX SENSOR RESULT SUMMARY FILES NOT CREATED
case 200 % ERROR 200 - TEMPERATURE SENSOR RESULT SUMMARY FILE FAILED TO BE
CREATED
disp('ERROR:  TEMP SENSOR SUMMARY File Failed to be Created!');
disp('  Program erred off in writing summary file in TEMP_SENSORS subroutine.
');
disp('  Verify TEMP_SENSORS program is correctly creating the summary file.');
```

```

disp('*** TERMINATING ISSPO PROGRAM ***');
% Return to Main Directory
cd ..\..\..\..;
case 201 % ERROR 201 - PRESSURE SENSOR RESULT SUMMARY FILE FAILED TO BE CREATED
disp('ERROR:  PRESSURE SENSOR SUMMARY File Failed to be Created!');
disp('  Program erred off in writing summary file in PRESSURE_SENSORS
subroutine. ');
disp('  Verify PRESSURE_SENSORS program is correctly creating the summary
file.');
```

```

disp('*** TERMINATING ISSPO PROGRAM ***');
% Return to Main Directory
cd ..\..\..\..;
case 202 % ERROR 202 - NEPHELOMETER SENSOR RESULT SUMMARY FILE FAILED TO BE
CREATED
disp('ERROR:  NEPHELOMETER SENSOR SUMMARY File Failed to be Created!');
disp('  Program erred off in writing summary file in NEPHELOMETER_SENSORS
subroutine. ');
disp('  Verify NEPHELOMETER_SENSORS program is correctly creating the summary
file.');
```

```

disp('*** TERMINATING ISSPO PROGRAM ***');
% Return to Main Directory
cd ..\..\..\..;
case 203 % ERROR 203 - HUMIDITY SENSOR RESULT SUMMARY FILE FAILED TO BE CREATED
disp('ERROR:  HUMIDITY SENSOR SUMMARY File Failed to be Created!');
disp('  Program erred off in writing summary file in HUMIDITY_SENSORS
subroutine. ');
disp('  Verify HUMIDITY_SENSORS program is correctly creating the summary
file.');
```

```

disp('*** TERMINATING ISSPO PROGRAM ***');
% Return to Main Directory
```

```

        cd ..\..\..\..;
    case 204 % ERROR 204 - DENSITY SENSOR RESULT SUMMARY FILE FAILED TO BE CREATED
    disp('ERROR: DENSITY SENSOR SUMMARY File Failed to be Created!');
    disp(' Program erred off in writing summary file in DENSITY_SENSORS subroutine.
');
    disp(' Verify DENSITY_SENSORS program is correctly creating the summary
file. ');
    disp('*** TERMINATING ISSPO PROGRAM ***');
    % Return to Main Directory
    cd ..\..\..\..;
    case 205 % ERROR 205 - ACOUSTIC SENSOR RESULT SUMMARY FILE FAILED TO BE CREATED
    disp('ERROR: ACOUSTIC SENSOR SUMMARY File Failed to be Created!');
    disp(' Program erred off in writing summary file in ACOUSTIC_SENSORS
subroutine. ');
    disp(' Verify ACOUSTIC_SENSORS program is correctly creating the summary
file. ');
    disp('*** TERMINATING ISSPO PROGRAM ***');
    % Return to Main Directory
    cd ..\..\..\..;
    case 206 % ERROR 206 - AC FIELD SENSOR RESULT SUMMARY FILE FAILED TO BE CREATED
    disp('ERROR: AC FIELD SENSOR SUMMARY File Failed to be Created!');
    disp(' Program erred off in writing summary file in AC_FIELD_SENSORS
subroutine. ');
    disp(' Verify AC_FIELD_SENSORS program is correctly creating the summary
file. ');
    disp('*** TERMINATING ISSPO PROGRAM ***');
    % Return to Main Directory
    cd ..\..\..\..;
    case 207 % ERROR 207 - WIND VELOCITY SENSOR RESULT SUMMARY FILE FAILED TO BE
CREATED
    disp('ERROR: WIND VELOCITY SENSOR SUMMARY File Failed to be Created!');
    disp(' Program erred off in writing summary file in WIND_VELOCITY_SENSORS
subroutine. ');
    disp(' Verify WIND_VELOCITY_SENSORS program is correctly creating the summary
file. ');
    disp('*** TERMINATING ISSPO PROGRAM ***');
    % Return to Main Directory
    cd ..\..\..\..;
    case 208 % ERROR 208 - ACCELEROMETER SENSOR RESULT SUMMARY FILE FAILED TO BE
CREATED
    disp('ERROR: ACCELEFOMETER SENSOR SUMMARY File Failed to be Created!');
    disp(' Program erred off in writing summary file in ACCELEROMETER_SENSORS
subroutine. ');
    disp(' Verify ACCELEROMETER_SENSORS program is correctly creating the summary
file. ');
    disp('*** TERMINATING ISSPO PROGRAM ***');
    % Return to Main Directory
    cd ..\..\..\..;
    case 209 % ERROR 209 - GCMS SENSOR RESULT SUMMARY FILE FAILED TO BE CREATED
    disp('ERROR: GCMS SENSOR SUMMARY File Failed to be Created!');
    disp(' Program erred off in writing summary file in GCMS_SENSORS subroutine.
');
    disp(' Verify GCMS_SENSORS program is correctly creating the summary file. ');
    disp('*** TERMINATING ISSPO PROGRAM ***');
    % Return to Main Directory
    cd ..\..\..\..;
    case 210 % ERROR 210 - RADIATION SENSOR RESULT SUMMARY FILE FAILED TO BE CREATED
    disp('ERROR: RADIATION SENSOR SUMMARY File Failed to be Created!');
    disp(' Program erred off in writing summary file in RADIATION_SENSORS
subroutine. ');
    disp(' Verify RADIATION_SENSORS program is correctly creating the summary
file. ');
    disp('*** TERMINATING ISSPO PROGRAM ***');
    % Return to Main Directory
    cd ..\..\..\..;
    case 211 % ERROR 211 - INCLINOMETER SENSOR RESULT SUMMARY FILE FAILED TO BE
CREATED
    disp('ERROR: INCLINOMETER SENSOR SUMMARY File Failed to be Created!');
    disp(' Program erred off in writing summary file in INCLINOMETER_SENSORS
subroutine. ');
    disp(' Verify INCLINOMETER_SENSORS program is correctly creating the summary
file. ');
    disp('*** TERMINATING ISSPO PROGRAM ***');
    % Return to Main Directory
    cd ..\..\..\..;
    case 212 % ERROR 212 - IMAGING SENSOR RESULT SUMMARY FILE FAILED TO BE CREATED

```

```

disp('ERROR: IMAGING SENSOR SUMMARY File Failed to be Created!');
disp(' Program erred off in writing summary file in IMAGING_SENSORS subroutine.
');
disp(' Verify IMAGING_SENSORS program is correctly creating the summary
file.');
```

```

disp('*** TERMINATING ISSPO PROGRAM ***');
% Return to Main Directory
cd ..\..\..;
case 213 % ERROR 213 - REFRACTION SENSOR RESULT SUMMARY FILE FAILED TO BE CREATED
disp('ERROR: REFRACTION SENSOR SUMMARY File Failed to be Created!');
disp(' Program erred off in writing summary file in REFRACTION_SENSORS
subroutine. ');
disp(' Verify REFRACTION_SENSORS program is correctly creating the summary
file.');
```

```

disp('*** TERMINATING ISSPO PROGRAM ***');
% Return to Main Directory
cd ..\..\..;
case 214 % ERROR 215 - DATA PROCESSING RESULT SUMMARY FILE FAILED TO BE CREATED
disp('ERROR: DIGITAL SIGNAL PROCESSING SUMMARY File Failed to be Created!');
disp(' Program erred off in writing summary file in DIGITAL_SIGNAL_PROCESSING
subroutine. ');
disp(' Verify DIGITAL_SIGNAL_PROCESSING program is correctly creating the
summary file.');
```

```

disp('*** TERMINATING ISSPO PROGRAM ***');
% Return to Main Directory
cd ..\..\..;
case 215 % ERROR 216 - MASS PROPERTY SUMMARY DATA FILE FAILED TO BE CREATED
disp('ERROR: MASS PROPERTIES SUMMARY File Failed to be Created!');
disp(' Program erred off in MASS_PROPERTIES subroutine. ');
disp(' Verify MASS_PROPERTIES program is correctly creating the summary
variables file.');
```

```

disp('*** TERMINATING ISSPO PROGRAM ***');
% Return to Main Directory
cd ..\..\..;
case 216 % ERROR 217 - POWER PROPERTY SUMMARY DATA FILE FAILED TO BE CREATED
disp('ERROR: POWER PROPERTIES SUMMARY File Failed to be Created!');
disp(' Program erred off in POWER_PROPERTIES subroutine. ');
disp(' Verify POWER_PROPERTIES program is correctly creating the summary
variables file.');
```

```

disp('*** TERMINATING ISSPO PROGRAM ***');
% Return to Main Directory
cd ..\..\..;
otherwise
disp('*** UNKNOWN ERROR DETECTED ***');
disp('Report ERROR and Case Configuration to Developer for Resolution!');
fprintf(' ERROR CODE: %3d', ERR_Code);
cd ..\..\..;
end

% Plays Error Sound Warning
%SoundPath = [ '\PRGM_FILES\SOUNDS' ]
if ERR_Code ~= 1
cd PRGM_FILES;
end
%
% Return to Main Directory
% cd ..\..\..;

```

## D21. Mass Properties Program

```
% function MASS_PROPERTIES(DataArray, UNITS)
% Wind Velocity_Sensor Database
%
% Developed by: Keith Schreck
% Mechanical and Aerospace Engineering
% San Jose State University
% Date: Fall 2007
%
% This file contains the MASS PROPERTY calculation for the ISSPO program.
% Data here is based on the sensors selected in the ISSPO Case Study and loads
% the mass property values for weights and dimensions into a Mass Property
% Array and calculates total values for all the components in the program.
% Values are assigned to a variable name and saved to a Matlab '.mat' file
% and loaded when calling the ISSPO program.

function MASS_PROPERTIES(DataArray, UNITS)

% Load Data files in to program for use
load('Constants_DB.mat');

% Load Final sensor Matrix values into Main Program
for i = 1:size(DataArray,1)
    switch DataArray{i,1}
        case 'ATMOSPHERE'
            load('TEMP_SENSOR_Final.mat')
            load('PRESSURE_SENSOR_Final.mat')
            load('WIND_VELOCITY_SENSOR_Final.mat')
            load('HUMIDITY_SENSOR_Final.mat')
            load('DENSITY_SENSOR_Final.mat')
        case 'ACCELERATION'
            load('ACCELEROMETER_SENSOR_Final.mat')
        case 'ACOUSTICS'
            load('ACOUSTIC_SENSOR_Final.mat')
        case 'EM_FIELD'
            load('AC_FIELD_SENSOR_Final.mat')
        case 'GAS_ANALYSIS'
            load('GCMS_SENSOR_Final.mat')
        case 'INCLINATION'
            load('INCLINOMETER_SENSOR_Final.mat')
        case 'NEPHELOMETRY'
            load('NEPHELOMETER_SENSOR_Final.mat')
        case 'OPTICS'
            load('IMAGING_SENSOR_Final.mat')
        case 'RADIATION'
            load('RADIATION_SENSOR_Final.mat')
        case 'REFRACTION'
            load('REFRACTION_SENSOR_Final.mat')
        case 'DATA_PROCESSING'
            load('DIGITAL_SIGNAL_PROCESSING_Final.mat')
        otherwise
    end
end
```





```

k = k + 1;
end
if strcmpi(UNITS,'British') == 1
    MASS_PROPS(k,1:4) = [ ACC_MATRIX6{1,20}/oz2lbm ACC_MATRIX6{1,21}/in2ft ACC_MATRIX6{1,22}/in2ft
    ACC_MATRIX6{1,23}/in2ft ];
    k = k + 1;
end
case 'ACOUSTICS'
    if strcmpi(UNITS,'SI') == 1
        MASS_PROPS(k,1:4) = [ ACS_MATRIX4{1,28}/gr2kg ACS_MATRIX4{1,25}/mm2m ACS_MATRIX4{1,26}/mm2m
        ACS_MATRIX4{1,27}/mm2m ];
        k = k + 1;
    end
    if strcmpi(UNITS,'British') == 1
        MASS_PROPS(k,1:4) = [ ACS_MATRIX4{1,28}/oz2lbm ACS_MATRIX4{1,25}/in2ft ACS_MATRIX4{1,26}/in2ft
        ACS_MATRIX4{1,27}/in2ft ];
        k = k + 1;
    end
end
case 'EM FIELD'
    FIELD Sensor
    if strcmpi(UNITS,'SI') == 1
        MASS_PROPS(k,1:4) = [ AC_MATRIX5{1,23}/gr2kg AC_MATRIX5{1,17}/cm2m AC_MATRIX5{1,18}/cm2m
        MASS_PROPS(k,5) = [ AC_MATRIX5{1,24}/gr2kg AC_MATRIX5{1,20}/cm2m AC_MATRIX5{1,21}/cm2m
        AC_MATRIX5{1,22}/cm2m ];
        k = k + 1;
    end
    if strcmpi(UNITS,'British') == 1
        MASS_PROPS(k,1:4) = [ AC_MATRIX5{1,23}/oz2lbm AC_MATRIX5{1,17}/in2ft AC_MATRIX5{1,18}/in2ft
        MASS_PROPS(k,5) = [ AC_MATRIX5{1,24}/oz2lbm AC_MATRIX5{1,20}/in2ft AC_MATRIX5{1,21}/in2ft
        AC_MATRIX5{1,22}/in2ft ];
        k = k + 1;
    end
end
case 'GAS ANALYSIS'
    if strcmpi(UNITS,'SI') == 1
        if strcmpi(GCMS_TYPE,'WAVELENGTH') == 1
            MASS_PROPS(k,1:4) = [ GCMS_MATRIX3{1,8} GCMS_MATRIX3{1,9}/mm2m GCMS_MATRIX3{1,10}/mm2m
            GCMS_MATRIX3{1,11}/mm2m ];
            k = k + 1;
        end
        if strcmpi(GCMS_TYPE,'MASS-CHARGE') == 1
            MASS_PROPS(k,1:4) = [ GCMS_MATRIX3{1,33} GCMS_MATRIX3{1,25}/cm2m GCMS_MATRIX3{1,26}/cm2m
            GCMS_MATRIX3{1,27}/cm2m ];
            k = k + 1;
        end
    end
    if strcmpi(UNITS,'British') == 1
        if strcmpi(GCMS_TYPE,'WAVELENGTH') == 1
            MASS_PROPS(k,1:4) = [ GCMS_MATRIX3{1,8} GCMS_MATRIX3{1,9}/in2ft GCMS_MATRIX3{1,10}/in2ft
            GCMS_MATRIX3{1,11}/in2ft ];
            k = k + 1;
        end
        if strcmpi(GCMS_TYPE,'MASS-CHARGE') == 1
            MASS_PROPS(k,1:4) = [ GCMS_MATRIX3{1,33} GCMS_MATRIX3{1,25}/in2ft GCMS_MATRIX3{1,26}/in2ft
            GCMS_MATRIX3{1,27}/in2ft ];
            k = k + 1;
        end
    end
end

```

```

GCMS_MATRIX3{1,27}/in2ft ];
      MASS_PROPS(k ,1:4) = [ GCMS_MATRIX3{1,33}   GCMS_MATRIX3{1,25}/in2ft   GCMS_MATRIX3{1,26}/in2ft
      k = k + 1;
    end
  end
  case 'INCLINATION'
    if strcmpi(UNITS,'SI')==1
      MASS_PROPS(k ,1:4) = [ INCL_MATRIX5{1,19}/gr2kg   INCL_MATRIX5{1,16}/mm2m   INCL_MATRIX5{1,17}/mm2m
      INCL_MATRIX5{1,18}/mm2m ];
      k = k + 1;
    end
    if strcmpi(UNITS,'British')==1
      MASS_PROPS(k ,1:4) = [ INCL_MATRIX5{1,19}/oz2lbm   INCL_MATRIX5{1,16}/in2ft   INCL_MATRIX5{1,17}/in2ft
      INCL_MATRIX5{1,18}/in2ft ];
      k = k + 1;
    end
  case 'NEPHELOMETRY'
    if strcmpi(UNITS,'SI')==1
      MASS_PROPS(k ,1:4) = [ N_MATRIX3{1,6}   N_MATRIX3{1,11}/mm2m   N_MATRIX3{1,9}/mm2m   N_MATRIX3{1,10}/mm2m
      ];
    end
    if strcmpi(UNITS,'British')==1
      MASS_PROPS(k ,1:4) = [ N_MATRIX3{1,6}   N_MATRIX3{1,11}/in2ft   N_MATRIX3{1,9}/in2ft   N_MATRIX3{1,10}/in2ft
      ];
    end
  case 'OPTICS'
    if strcmpi(UNITS,'SI')==1
      MASS_PROPS(k ,1:4) = [ IMG_MATRIX5{1,26}   IMG_MATRIX5{1,23}/mm2m   IMG_MATRIX5{1,24}/mm2m
      IMG_MATRIX5{1,25}/mm2m ];
      k = k + 1;
    end
    if strcmpi(UNITS,'British')==1
      MASS_PROPS(k ,1:4) = [ IMG_MATRIX5{1,26}   IMG_MATRIX5{1,23}/in2ft   IMG_MATRIX5{1,24}/in2ft
      IMG_MATRIX5{1,25}/in2ft ];
      k = k + 1;
    end
  case 'RADIATION'
    if strcmpi(UNITS,'SI')==1
      MASS_PROPS(k ,1:4) = [ RAD_MATRIX3{1,26}   RAD_MATRIX3{1,27}/mm2m   RAD_MATRIX3{1,28}/mm2m
      RAD_MATRIX3{1,29}/mm2m ];
      k = k + 1;
    end
    if strcmpi(UNITS,'British')==1
      MASS_PROPS(k ,1:4) = [ RAD_MATRIX3{1,26}/oz2lbm   RAD_MATRIX3{1,27}/in2ft   RAD_MATRIX3{1,28}/in2ft
      RAD_MATRIX3{1,29}/in2ft ];
      k = k + 1;
    end
  case 'REFRACTION'
    if strcmpi(UNITS,'SI')==1
      MASS_PROPS(k ,1:4) = [ REF_MATRIX3{1,7}/gr2kg   REF_MATRIX3{1,12}/cm2m   REF_MATRIX3{1,13}/cm2m
      REF_MATRIX3{1,14}/cm2m ];
      k = k + 1;
    end
  end
end

```

```

end
if strcmpi(UNITS, 'British') == 1
    MASS_PROPS(k, 1:4) = [ REF_MATRIX3{1,7}/oz2lbf REF_MATRIX3{1,12}/in2ft REF_MATRIX3{1,13}/in2ft
REF_MATRIX3{1,14}/in2ft ];
    k = k + 1;
end
case 'DATA PROCESSING'
if strcmpi(UNITS, 'SI') == 1
    MASS_PROPS(k, 1:4) = [ DSP_MATRIX2{1,17} DSP_MATRIX2{1,22}/mm2m DSP_MATRIX2{1,23}/mm2m
DSP_MATRIX2{1,24}/mm2m ];
    k = k + 1;
end
if strcmpi(UNITS, 'British') == 1
    MASS_PROPS(k, 1:4) = [ DSP_MATRIX2{1,17} DSP_MATRIX2{1,22}/in2ft DSP_MATRIX2{1,23}/in2ft
DSP_MATRIX2{1,24}/in2ft ];
    k = k + 1;
end
otherwise
end
end
save('MASS_PROPERTIES_Data.mat');
end

```

## D22. Power Properties Program

```

% function POWER_PROPERTIES(DataArray, UNITS)
% Sensor Power Property Database
%
% Developed by: Keith Schreck
%               Mechanical and Aerospace Engineering
%               San Jose State University
% Date:         Fall 2007
%
% This file contains the POWER PROPERTY calculation for the ISSPO program.
% Data here is based on the sensors selected in the ISSPO Case Study and loads
% the power requirement data into a sensor power array and records values
% for all the sensor components chosen in the Sensor Data file for the main program.
% Values are assigned to a variable name and saved to a Matlab '.mat' file
% and loaded when calling the ISSPO program.
%
% Cases where Power data is not available and Voltage - Current information is
% available, Power requirements are calculated based on Ohm's Law:
%
%  $P = V * I$  in Watts
%
function POWER_PROPERTIES(DataArray, UNITS)

```

```

% Load Data files in to program for use
load('Constants_DB.mat');

% Load Final sensor Matrix values into Main Program
for i = 1:size(DataArray,1)
    switch DataArray{i,1}
        case 'ATMOSPHERE'
            load('TEMP_SENSOR_Final.mat')
            load('PRESSURE_SENSOR_Final.mat')
            load('WIND_VELOCITY_SENSOR_Final.mat')
            load('HUMIDITY_SENSOR_Final.mat')
            load('DENSITY_SENSOR_Final.mat')
        case 'ACCELERATION'
            load('ACCELEROMETER_SENSOR_Final.mat')
        case 'ACOUSTICS'
            load('ACOUSTIC_SENSOR_Final.mat')
        case 'EM_FIELD'
            load('AC_FIELD_SENSOR_Final.mat')
        case 'GAS_ANALYSIS'
            load('GCMS_SENSOR_Final.mat')
        case 'INCLINATION'
            load('INCLINOMETER_SENSOR_Final.mat')
        case 'NEPHELOMETRY'
            load('NEPHELOMETER_SENSOR_Final.mat')
        case 'OPTICS'
            load('IMAGING_SENSOR_Final.mat')
        case 'RADIATION'
            load('RADIATION_SENSOR_Final.mat')
        case 'REFRACTION'
            load('REFRACTION_SENSOR_Final.mat')
        case 'DATA_PROCESSING'
            load('DIGITAL_SIGNAL_PROCESSING_Final.mat')
        otherwise
            end
    end
end

% Load Final sensor Matrix values into Main Program
k = 1;
for i = 1:size(DataArray,1)
    switch DataArray{i,1}
        case 'ATMOSPHERE'
            % Loads and Converts POWER Properties to SI Standard Units For Comparison to Case Inputs
            if strcmpi(UNITS,'SI') == j
                POWER_PROPS(k,:) = [ MATRIX2{1,4} ];
                POWER_PROPS(k+1,1) = [ P_MATRIX6{1,22} * P_MATRIX6{1,23}/m^2A ]; % Pressure Sensor - converts V & mA to W
                if strcmpi(WV_TYPE,'ANEMOMETER') == 1
                    POWER_PROPS(k+2,1) = [ 0 ]; % Probe Dimensions - Voltage Only no Current Data
                    POWER_PROPS2(1,1) = [ 0 0 ]; % Handheld Dimensions
                end
            if strcmpi(WV_TYPE,'DOPPLER') == 1
                POWER_PROPS(k+2,1) = [ W_MATRIX3{1,21} ];
            end
        end
    end
end

```

```

end
POWER_PROPS(k+3,1) = [ D_MATRIX6{1,14} 0 ]; % Humidity Sensor - No Voltage or Current Data
POWER_PROPS(k+4,1) = [ D_MATRIX6{1,15}/mA2A ]; % - converts V & mA to W
k = k + 5;
end
if strcmpi(UNITS,'British') == 1
POWER_PROPS(k,1) = [ MATRIX2{1,4} ];
POWER_PROPS(k+1,1) = [ P_MATRIX6{1,22} * P_MATRIX6{1,23}/mA2A * BTU_hr2W ]; % Pressure Sensor- converts V
& mA to W to BTU/hr
if strcmpi(WV_TYPE,'ANEMOMETER') == 1
POWER_PROPS(k+2,1) = [ 0 ]; % Probe Dimensions - Voltage Only no Current Data
POWER_PROPS2(1,1) = [ 0 ]; % Handheld Dimensions
end
if strcmpi(WV_TYPE,'DOPPLER') == 1
POWER_PROPS(k+2,1) = [ W_MATRIX3{1,21} ];
end
POWER_PROPS(k+3,1) = [ 0 ]; % Humidity Sensor - No Voltage or Current Data
POWER_PROPS(k+4,1) = [ D_MATRIX6{1,14} * D_MATRIX6{1,15}/mA2A * BTU_hr2W ]; % - converts V & mA to W to
BTU/hr
k = k + 5;
end
case 'ACCELERATION'
if strcmpi(UNITS,'SI') == 1
if ACC_MATRIX6{1,25} ~= 0
POWER_PROPS(k,1) = [ ACC_MATRIX6{1,25} ]; % W
else
POWER_PROPS(k,1) = [ ACC_MATRIX6{1,15} * ACC_MATRIX6{1,14}/mA2A ]; % - converts V & mA to W
end
k = k + 1;
end
if strcmpi(UNITS,'British') == 1
if ACC_MATRIX6{1,25} ~= 0
POWER_PROPS(k,1) = [ ACC_MATRIX6{1,25} ]; % BTU's/hr
else
POWER_PROPS(k,1) = [ ACC_MATRIX6{1,15} * ACC_MATRIX6{1,14}/mA2A * BTU_hr2W ]; % - converts V & mA to W
end
k = k + 1;
end
case 'ACOUSTICS'
if strcmpi(UNITS,'SI') == 1
POWER_PROPS(k,1) = [ ACS_MATRIX4{1,13} * ACS_MATRIX4{1,14}/mA2A ]; % - converts V & mA to W
k = k + 1;
end
if strcmpi(UNITS,'British') == 1
POWER_PROPS(k,1) = [ ACS_MATRIX4{1,13} * ACS_MATRIX4{1,14}/mA2A * BTU_hr2W ]; % - converts V & mA to W to
BTU/hr
end
k = k + 1;
end
case 'EM_FIELD'
if strcmpi(UNITS,'SI') == 1
POWER_PROPS(k,1) = [ 0 ]; % Probe Dimensions - Voltage Only no Current Data
POWER_PROPS3(1,1) = [ 0 ]; % Handheld Dimensions
k = k + 1;
end

```

```

end
if strcmpi(UNITS,'British')== 1
    POWER_PROPS(k ,1) = [
    POWER_PROPS3(1 ,1) = [
    k = k + 1;
]; % Probe Dimensions
]; % Handheld Dimensions
end
case 'GAS ANALYSIS'
if strcmpi(UNITS,'SI')== 1
    if strcmpi(GCMS_TYPE,'WAVELENGTH')== 1
        POWER_PROPS(k ,1) = [
        k = k + 1;
]; % Voltage Only no Current Data
    end
    if strcmpi(GCMS_TYPE,'MASS-CHARGE')== 1
        POWER_PROPS(k ,1) = [ GCMS_MATRIX3{1,37} ];
        k = k + 1;
    end
end
if strcmpi(UNITS,'British')== 1
    if strcmpi(GCMS_TYPE,'WAVELENGTH')== 1
        POWER_PROPS(k ,1) = [
        k = k + 1;
]; % Voltage Only no Current Data
    end
end
if strcmpi(GCMS_TYPE,'MASS-CHARGE')== 1
    POWER_PROPS(k ,1) = [ GCMS_MATRIX3{1,37} ];
    k = k + 1;
end
end
case 'INCLINATION'
if strcmpi(UNITS,'SI')== 1
    POWER_PROPS(k ,1) = [
    k = k + 1;
]; % Voltage Only no Current Data
end
if strcmpi(UNITS,'British')== 1
    POWER_PROPS(k ,1) = [
    k = k + 1;
]; % Voltage Only no Current Data
end
case 'NEPHELOMETRY'
if strcmpi(UNITS,'SI')== 1
    POWER_PROPS(k ,1) = [
    k = k + 1;
]; % Watts
end
if strcmpi(UNITS,'British')== 1
    POWER_PROPS(k ,1) = [
    k = k + 1;
]; % BTUS/hr
end
case 'OPTICS'
if strcmpi(UNITS,'SI')== 1
    POWER_PROPS(k ,1) = [
    k = k + 1;
]; % mW to W
end
if strcmpi(UNITS,'British')== 1
    POWER_PROPS(k ,1) = [
    k = k + 1;
]; %
end

```

```

case 'RADIATION'
    if strcmpi(UNITS,'SI')==1
        POWER_PROPS(k ,1) = [ RAD_MATRIX3{1,30} ]; % Watts
        k = k + 1;
    end
    if strcmpi(UNITS,'British')==1
        POWER_PROPS(k ,1) = [ RAD_MATRIX3{1,30} ]; % BTUs/hr
        k = k + 1;
    end
case 'REFRACTION'
    if strcmpi(UNITS,'SI')==1
        POWER_PROPS(k ,1) = [ REF_MATRIX3{1,11} ]; %
        k = k + 1;
    end
    if strcmpi(UNITS,'British')==1
        POWER_PROPS(k ,1) = [ REF_MATRIX3{1,11} ]; %
        k = k + 1;
    end
case 'DATA PROCESSING'
    if strcmpi(UNITS,'SI')==1
        POWER_PROPS(k ,1) = [ DSP_MATRIX2{1,16}/mw2W ]; %
        k = k + 1;
    end
    if strcmpi(UNITS,'British')==1
        POWER_PROPS(k ,1) = [ DSP_MATRIX2{1,16} ]; %
        k = k + 1;
    end
    otherwise
        end
end
save('POWER_PROPERTIES_Data.mat');

```



## APPENDIX E

### ISSPO Program Error Codes

The following table details the errors that can occur during the execution of the ISSPO tool. The ERROR\_PRG is triggered when an error is encountered with a single corresponding error code. The codes detail the nature of the error and a possible remedy to the problem.

#### ISSPO Program ERROR Codes

<b>PROGRAM OPERATION ERRORS</b>		
<b>ERROR CODE</b>	<b>SOURCE MODULE</b>	<b>ACTION</b>
1	ISSPO	INPUT_FILE does not exist. Verify correct input file name.
2	ISSPO	InputDeck file does not exist. InputDeck file not successfully created.
3	CONSTANTS	CONSTANTS database file not created. Verify Constants database is being created.
4	PLANETARY_DATABASE	Planetary database file does not exist. Verify the Planetary Database program is correctly making the database file.
<b>UNKNOWN PROGRAM VARIABLES</b>		
<b>ERROR CODE</b>	<b>SOURCE MODULE</b>	<b>ACTION</b>
50	ISSPO	Invalid UNITS variable entered into input file. Verify UNITS is 'SI' or 'British'.
51	ISSPO	Invalid Print Option variable entered into input file. Verify PRINTFLG is 'Y' or 'N'. Prints out individual program results to main program window.
52	DATA_TYPE_VERIFIER	Invalid type of science data entered into SENSOR_DATA Array. Verify sensor data types entered into array are valid science data options.
53	ISSPO	Undefined Mass Limit variable entered into input file. Verify MASS_LIMIT variable is set and defined in user input file.
54	ISSPO	Undefined Power Limit variable entered into input file. Verify POWER_LIMIT variable is set and defined in user input file.
55	ISSPO	Undefined Volume Limit variable entered into input file. Verify VOLUME_LIMIT variable is set and defined in user input file.
56	DATA_TYPE_VERIFIER	Sensor Option Requires additional Input parameters in SENSOR_DATA Variable in Input file. Verify proper data format for selected sensor.

57	DATA_TYPE_ VERIFIER	Sensor Option 'OPTICS' Requires additional Input parameters in SENSOR_DATA Variable in Input file. Verify proper data format for selected sensor. Size of SENSOR_DATA = 2
58	DATA_TYPE_ VERIFIER	Sensor Option 'OPTICS' Requires additional Input parameters in SENSOR_DATA Variable in Input file. Verify proper data format for selected sensor. Size of SENSOR_DATA = 3
59	DATA_TYPE_ VERIFIER	Invalid value in sensor option 'OPTICS'. Requires Optical Sensor System Type in 2 <sup>nd</sup> column position in SENSOR_DATA variable in Input file. Valid Options are 'CAMERA', 'ARRAY', 'LINEAR'. Verify proper system format for selected sensor.
60	DATA_TYPE_ VERIFIER	Invalid value in sensor option 'OPTICS'. Requires Optical Sensor Resolution in 3 <sup>rd</sup> column position in SENSOR_DATA variable in Input file. Valid Resolutions are 'LOW', 'MEDIUM', 'HIGH'. Verify proper resolution format for selected sensor.
61	DATA_TYPE_ VERIFIER	Invalid value in sensor option 'OPTICS'. Requires Optical Sensor Type in 4 <sup>th</sup> column position in SENSOR_DATA variable in Input file. Valid image types are 'X-RAY', 'VISUAL', 'UV', 'NIR', and 'MICRO'. Verify proper image format for selected sensor. Multiple Image types are allowed. Enter additional image types in subsequent columns.
62	DATA_TYPE_ VERIFIER	Sensor Option 'RADIATION' Requires additional Input parameters in SENSOR_DATA Variable in Input file. Verify proper data format for selected sensor.
63	DATA_TYPE_ VERIFIER	Invalid value in sensor option 'RADIATION'. Requires Radiation Sensor Type in 3 <sup>rd</sup> column position in SENSOR_DATA variable in Input file. Valid radiation types are 'Charged Particle', 'Alpha', 'Beta', 'Gamma', and 'X-Ray'. Verify proper radiation type for selected sensor. Multiple Radiation types are allowed. Enter additional radiation types in subsequent columns.
64	DATA_TYPE_ VERIFIER	Sensor Option 'INCLINATION' Requires additional Input parameters in SENSOR_DATA Variable in Input file. Verify proper data format for selected sensor.
65	DATA_TYPE	Invalid value in sensor option

	VERIFIER	'INCLINATION'. Requires Inclination Sensor Range in 2 <sup>nd</sup> column position in SENSOR_DATA variable in Input file. Valid Ranges are 'LOW', 'MEDIUM', 'HIGH'. Verify proper range format for selected sensor.
66	DATA_TYPE_VERIFIER	Sensor Option 'ACCELERATION' Requires additional Input parameters in SENSOR_DATA Variable in Input file. Verify proper data format for selected sensor.
67	DATA_TYPE_VERIFIER	Invalid value in sensor option 'ACCELERATION'. Requires Acceleration Sensor Range in 2 <sup>nd</sup> column position in SENSOR_DATA variable in Input file. Valid Ranges are 'SOFT', 'MEDIUM', 'HIGH', 'IMPACT', 'BALLISTIC'. Verify proper range format for selected sensor.
68	DATA_TYPE_VERIFIER	Sensor Option 'ATMOSPHERE' -- 'WIND VELOCITY' Requires additional Input parameters in SENSOR_DATA Variable in Input file. Verify proper data format for selected sensor.
69	DATA_TYPE_VERIFIER	Invalid value in sensor option 'WIND VELOCITY'. Requires Sensor Configuration in next column position in SENSOR_DATA variable in Input file. Valid Types are 'ANEMOMETER', 'DOPPLER'. Verify proper system format for selected sensor.
70	DATA_TYPE_VERIFIER	Invalid value in sensor option 'GAS ANALYSIS'. Requires Sensor Configuration in next column position in SENSOR_DATA variable in Input file. Valid Types are 'WAVELENGTH', 'MASS-CHARGE'. Verify proper system format for selected sensor.
71	DATA_TYPE_VERIFIER	Invalid value in sensor option 'GAS ANALYSIS' -- 'MASS-CHARGE' Option. Requires Sensing Mass Range in 3rd column position in SENSOR_DATA variable in Input file. Verify proper system format for selected sensor.
72	DATA_TYPE_VERIFIER	Invalid value in sensor option 'GAS ANALYSIS' -- 'MASS-CHARGE' Option. Requires Sensing Mass Range in 3rd column position in SENSOR_DATA variable in Input file. Valid Ranges are 'LOW', 'MEDIUM', 'HIGH'. Verify proper system format for selected sensor.
73	DATA_TYPE_VERIFIER	Sensor Option 'ACCELERATION' Requires additional Input parameters in

		SENSOR_DATA Variable in Input file. Sensor Power Type value required in 3 <sup>rd</sup> Column. Verify proper data format for selected sensor.
74	DATA_TYPE_VERIFIER	Invalid value in sensor option 'ACCELERATION'. Requires Acceleration Sensor Power Type in 3 <sup>rd</sup> column position in SENSOR_DATA variable in Input file. Valid types are 'Voltage', 'Const Currant', 'Self Generating'. Verify proper range format for selected sensor.
75	DATA_TYPE_VERIFIER	Sensor Option 'ATMOSPHERE' – 'TEMPERATURE' Requires additional Input parameters in SENSOR_DATA Variable in Input file. Sensor Power Type value required in next Column. Verify proper data format for selected sensor.
76	DATA_TYPE_VERIFIER	Invalid value in sensor option 'TEMPERATURE. Requires Temperature Sensor Power Type in next column position in SENSOR_DATA variable in Input file. Valid types are 'VOLTAGE' or 'RESISTANCE'. Verify proper range format for selected sensor.
77	DATA_TYPE_VERIFIER	Sensor Option 'ACOUSTICS' Requires additional Input parameters in SENSOR_DATA Variable in Input file. Verify proper data format for selected sensor.
78	DATA_TYPE_VERIFIER	Invalid value in sensor option 'ACOUSTICS'. Requires Acoustic Sensor Type in 2 <sup>nd</sup> column position in SENSOR_DATA variable in Input file. Valid Ranges are 'SENSOR', or 'ARRAY'. Verify proper type format for selected sensor.
79	DATA_TYPE_VERIFIER	Sensor Option 'ATMOSPHERE' – 'DENSITY' Requires additional Input parameters in SENSOR_DATA Variable in Input file. Sensor Medium Type value required in next Column. Verify proper data format for selected sensor.
80	DATA_TYPE_VERIFIER	Invalid value in sensor option 'DENSITY'. Requires Density Sensor Medium in next column position in SENSOR_DATA variable in Input file. Valid types are 'GAS' or 'LIQUID'. Verify proper operating medium for selected sensor.
81	DATA_TYPE_VERIFIER	Sensor Option 'DATA PROCESSING' Requires additional Input parameters in SENSOR_DATA Variable in Input file. Verify proper data format for selected

		sensor.
82	DATA_TYPE_ VERIFIER	Sensor Option 'DATA PROCESSING' Requires additional Input parameters in SENSOR_DATA Variable in Input file. Requires Microprocessor Cycle speed in 2 <sup>nd</sup> column position and number of connected devices in 3 <sup>rd</sup> column position. Verify proper data format for selected sensor.
83	DATA_TYPE_ VERIFIER	Invalid value in sensor option 'DATA PROCESSING'. Requires Microprocessor Cycle speed in 2 <sup>nd</sup> column position in SENSOR_DATA variable in Input file. Valid types are 'LOW', 'MEDIUM', or 'HIGH'. Verify proper operating medium for selected sensor.
84	DATA_TYPE_ VERIFIER	Invalid value in sensor option 'DATA PROCESSING'. Required number of connected devices in SENSOR_DATA variable in Input file. Non-numeric value detected in 3 <sup>rd</sup> column position. Verify proper operating medium for selected sensor.
<b>SENSOR DATA FILES</b>		
<b>ERROR CODE</b>	<b>SOURCE MODULE</b>	<b>ACTION</b>
100	TEMP_SENSORS	TEMP_SENSORS Database file was not created successfully. Program erred off somewhere in module.
101	PRESSURE_ SENSORS	PRESSURE_SENSORS Database file was not created successfully. Program erred off somewhere in module.
102	NEPHELOMETER_ SENSOR	NEPHELOMETER_SENSORS Database file was not created successfully. Program erred off somewhere in module.
103	HUMIDITY_ SENSORS	HUMIDITY_SENSORS Database file was not created successfully. Program erred off somewhere in module.
104	DENSITY_ SENSORS	DENSITY_SENSORS Database file was not created successfully. Program erred off somewhere in module.
105	ACOUSTIC_ SENSORS	ACOUSTIC_SENSORS Database file was not created successfully. Program erred off somewhere in module.
106	AC_FIELD_ SENSORS	AC_FIELD_SENSORS Database file was not created successfully. Program erred off somewhere in module.
107	WIND_VELOCITY_ SENSORS	WIND_VELOCITY_SENSORS Database file was not created successfully. Program erred off somewhere in module.
108	ACCELEROMETER	ACCELEROMETER_SENSORS Database

	SENSORS	file was not created successfully. Program erred off somewhere in module.
109	GCMS_SENSORS	GCMS_SENSORS Database file was not created successfully. Program erred off somewhere in module.
110	RADITATION_SENSORS	RADIATION_SENSORS Database file was not created successfully. Program erred off somewhere in module.
111	INCLINOMETER_SENSORS	INCLINOMETER_SENSORS Database file was not created successfully. Program erred off somewhere in module.
112	IMAGING_SENSORS	IMAGING_SENSORS Database file was not created successfully. Program erred off somewhere in module.
113	REFRACTION_SENSORS	REFRACTION_SENSORS Database file was not created successfully. Program erred off somewhere in module.
114	DIGITAL_SIGNAL_PROCESSING	DIGITAL_SIGNAL_PROCESSING Database file was not created successfully. Program erred off somewhere in module.
<b>SENSOR SUMMARY FILES</b>		
<b>ERROR CODE</b>	<b>SOURCE MODULE</b>	<b>ACTION</b>
200	TEMP_SENSORS	TEMP_SENSORS Summary file was not created successfully. Program erred off somewhere in module.
201	PRESSURE_SENSORS	PRESSURE_SENSORS Summary file was not created successfully. Program erred off somewhere in module.
202	NEPHELOMETER_SENSOR	NEPHELOMETER_SENSORS Summary file was not created successfully. Program erred off somewhere in module.
203	HUMIDITY_SENSORS	HUMIDITY_SENSORS Summary file was not created successfully. Program erred off somewhere in module.
204	DENSITY_SENSORS	DENSITY_SENSORS Summary file was not created successfully. Program erred off somewhere in module.
205	ACOUSTIC_SENSORS	ACOUSTIC_SENSORS Summary file was not created successfully. Program erred off somewhere in module.
206	AC_FIELD_SENSORS	AC_FIELD_SENSORS Summary file was not created successfully. Program erred off somewhere in module.
207	WIND_VELOCITY_SENSORS	WIND_VELOCITY_SENSORS Summary file was not created successfully. Program erred off somewhere in module.
208	ACCELEROMETER_SENSORS	ACCELEROMETER_SENSORS Summary file was not created successfully. Program erred off somewhere in module.

209	GCMS_ SENSORS	GCMS_SENSORS Summary file was not created successfully. Program erred off somewhere in module.
210	RADITATION_ SENSORS	RADIATION_SENSORS Summary file was not created successfully. Program erred off somewhere in module.
211	INCLINOMETER_ SENSORS	INCLINOMETER_SENSORS Summary file was not created successfully. Program erred off somewhere in module.
212	IMAGING_ SENSORS	IMAGING_SENSORS Summary file was not created successfully. Program erred off somewhere in module.
213	REFRACTION_ SENSORS	REFRACTION_SENSORS Summary file was not created successfully. Program erred off somewhere in module.
214	DIGITAL_SIGNAL_ PROCESSING	DIGITAL_SIGNAL_PROCESSING Summary file was not created successfully. Program erred off somewhere in module.
215	MASS_ PROPERTIES	MASS_PROPERTIES SUMMARY file was not created successfully. Program erred off somewhere in module.
216	POWER_ PROPERTIES	POWER_PROPERTIES SUMMARY file was not created successfully. Program erred off somewhere in module.