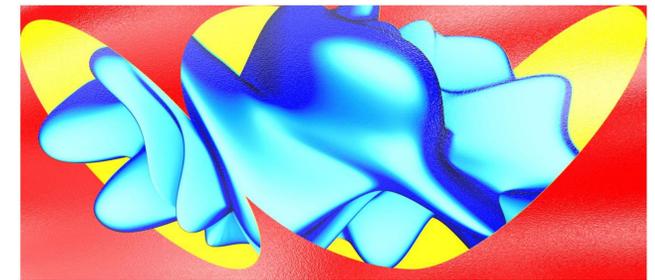


# Simultaneous Simulation of Microgravity and Ionizing Radiation in a Laboratory Environment



Student Research Symposium  
Merrill-Cazier Library 04.12.13 | 10:00-3:00  
Research  
UtahStateUniversity.

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## INTRODUCTION

- Astronauts suffer significant cellular damage during spaceflight.
- Main causes-
  - Simultaneous exposure to microgravity, and
  - Ionizing radiation
- Goal-** Design a mini-rotary cell culture system (mRCCS) that can-
  - Simulate microgravity
  - Be used in combination with a radiation source, and
  - Is biologically compatible and autoclavable

## RADIATION EXPOSURE

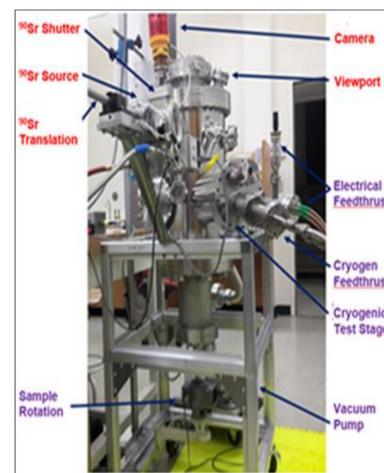


Fig.1 Space Survivability Test Chamber

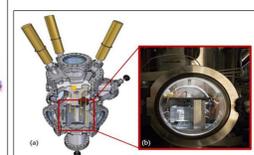


Fig. 2 (a) Cut away view of SST chamber experimental setup. (b) Image of the SST chamber with the mini-RCCS seated within the beam path of the <sup>90</sup>Sr source

## METHODS

- Reduced gravity conditions for 200 μm microcarrier beads suspended in a media were achieved when-
  - Beads reached terminal settling velocity  $v_s$ , as the centripetal force ( $F_c$ ) is balanced when they fall with near zero net forces from gravity ( $F_g$ ), buoyancy ( $F_b$ ), viscous drag ( $F_v$ ).
  - $v_s = 2g(\rho_b - \rho)(R_{bead})^2/9\mu$  where  $\rho_b$  = density of the beads,  $\rho$  = density of the media,  $R_{bead}$  = radius of the beads,  $\mu$  = dynamic viscosity of the media.
- For rotation calibration, different density liquids were chosen and vessels were rotated from 1 rpm to 55 rpm.
- Another measure of an effective mRCCS — low radial acceleration.

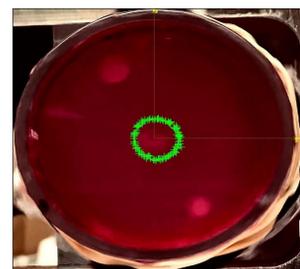


Fig.3 Microbeads cluster tracking inside the media.

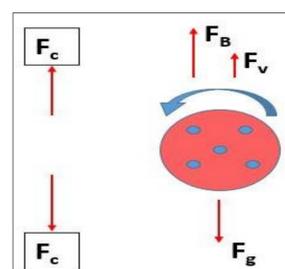


Fig.4 Balance of external forces from Stoke's law

## RESULTS

- The range of rotational speeds for microbeads undergoing microgravity was confirmed for the media.
  - Lower limit = 6 rpm
  - Upper limit = 42 rpm
- Reduced gravity environments were reached from  $\sim 1 \times 10^{-5}$  g to  $\sim 2 \times 10^{-2}$  g
- The combined mini RCCS and SST chamber system can provide average effective dose rates for the cells controlled over a broad range 900 X).

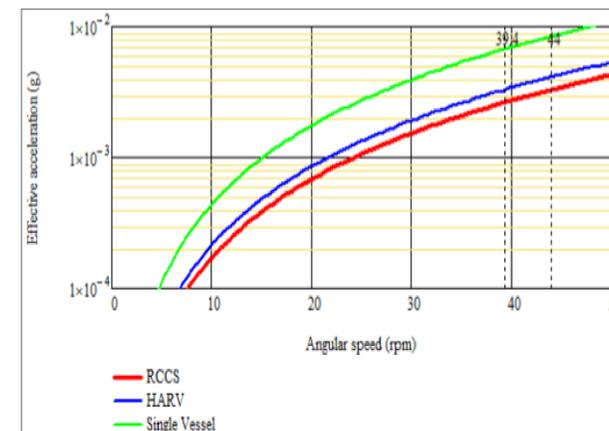


Fig. 5 Comparison of mRCCS effective gravitational acceleration of with commercial systems

## CONCLUSIONS AND FUTURE WORK

- A novel, versatile, cost-effective system is developed that can-
  - Model cellular damage from microgravity and ionizing radiation
  - Provide stable, simultaneous space-like radiation and reduced gravity environments
  - Be used to model microgravity at the same level as the commercially available systems.
- Further analysis of rotation calibration will be done for salt water, IPA etc.

	Four vessel RCCS (Synthecon)	Single vessel RCCS (Synthecon)	Mini-RCCS	ISS	Moon	Mars
Vessel radius (mm)	21.5	44.1	17.1	-	-	-
Rotational speed (rpm)	39.4	27.8	44.4*	5.3*	311*	470*
Effective gravitational acceleration (g)	$3.3 \times 10^{-3}$	$3.4 \times 10^{-3}$	$3.4 \times 10^{-3}$	$4.84 \times 10^{-3}$	0.165	0.376

\* Rotation to achieve effective gravitational acceleration.

Table 1 Comparison of effective gravitational force of mRCCS to gravitational force on the ISS, Moon, and Mars

Fluid	Density (g/cm <sup>3</sup> )	Rotation Speeds (rad/s)	Average Radius (cm)	Radius Amplitude (cm)
Water	0.99	4.35	0.550	0.0343
Isopropyl Alcohol	0.79	4.79	0.317	0.0241
Salt Water	1.18	4.42	1.15	0.0013
Media	0.98	4.33	0.312	0.0351

Table 2 Rotation speeds of different density liquids.

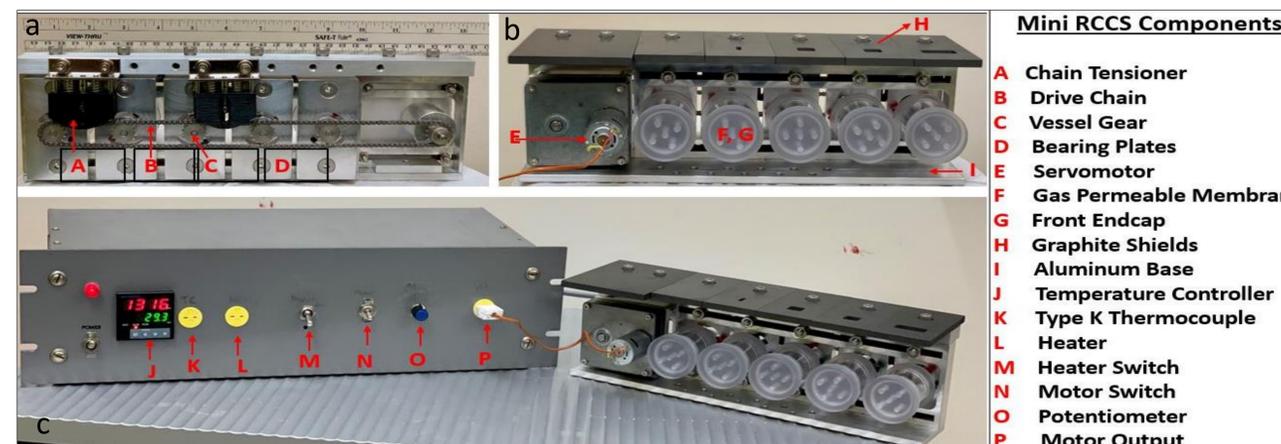


Fig. 6(a) Rear- view of the mRCCS. (b) Front view of the mRCCS with variable slit widths of graphite plates. (c) mRCCS with controller.

### Mini RCCS Components

- A Chain Tensioner
- B Drive Chain
- C Vessel Gear
- D Bearing Plates
- E Servomotor
- F Gas Permeable Membrane
- G Front Endcap
- H Graphite Shields
- I Aluminum Base
- J Temperature Controller
- K Type K Thermocouple
- L Heater
- M Heater Switch
- N Motor Switch
- O Potentiometer
- P Motor Output

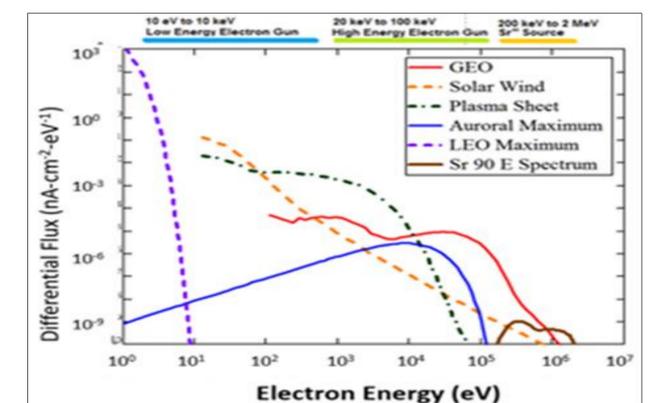


Fig. 7 Differential electron flux for five typical space environments. <sup>90</sup>Sr source electron emission spectrum is also shown..

