

Time-delayed characteristics of turbulence in pulsatile pipe flow

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(Received xx; revised xx; accepted xx)

Direct numerical simulations are performed to study temporal variations of the wall shear stresses and flow dynamics in the turbulent pulsatile pipe flow. The mechanisms, responsible for the paradoxical phenomenon for which the amplitude of the oscillating wall shear stress in the turbulent flow is smaller than that in the laminar flow for the same pulsation conditions, are investigated. It is shown that the delayed response of turbulence in the buffer layer generates a large magnitude of the radial gradient of the Reynolds shear stress near the wall, which counteracts the effect of the oscillating pressure gradient on the change of the streamwise velocity and hence reduces the amplitude of the wall shear stress. Such a delayed response consists of two processes: the delayed development of near-wall streaks and the subsequent energy redistribution from the streamwise velocity fluctuation to the other two co-existing components. This is a dynamical manifestation of the viscoelasticity of turbulent eddies. As the frequency is reduced, the variation of the friction Reynolds number results in a phase-wise variation of the time scale and intensity of the turbulence response, causing the hysteresis of the wall shear stress. Such a phase asymmetry is amplified by the increase of the pulsation amplitude. An examination of the energy spectra reveals that the near-wall streaks are stretched in the streamwise direction during the acceleration phase, and then break up into small-scale structures in the deceleration phase, accompanied by the enhanced dissipation that transforms the turbulent kinetic energy into heat.

Key words: wall shear stress, pulsatile pipe flow, wave-turbulence interaction

1. Introduction

Pulsatile turbulent pipe flows are widely encountered in engineering applications and biological systems, such as the turbomachinery and blood flow in aortic arteries. The oscillating nature of the pulsatile flow leads to a high-level fluctuation of the wall friction. Understanding the variation of the wall shear stress is of great practical significance to, e.g., pipeline leak detections (Colombo *et al.* (2009)) and blood-vessel problems

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(Cunningham & Gotlieb (2005)), which require precise knowledge of flow dynamic behaviours. This paper is concerned with the wall shear stresses in pulsatile turbulent pipe flows and the corresponding fluid dynamics.

The pulsation process consists of continuous accelerating and decelerating phases. In the past decades, several researchers have focused on the flow responses in the accelerating turbulent pipe or channel flow, in which the flow rate varies in a step- or ramp-changing manner (Maruyama *et al.* (1976); Greenblatt & Moss (1999); He & Jackson (2000); Greenblatt & Moss (2004)). It is well-established, in both scenarios, that the flow initially evolves in a laminar-like way, accompanied by the streamwise stretching of the near-wall streaks (stage 1). Then, the elongated streaks break up, leading to a formation and subsequent merging of turbulent spots (stage 2). Finally, the turbulence reaches its new fully-developed state (stage 3) (He *et al.* (2011); Seddighi *et al.* (2014); He *et al.* (2016); Jung & Chung (2012); He & Seddighi (2015)). Specifically, this three-stage flow evolution resembles closely the bypass transition in the boundary layer induced by the free-stream-turbulence (He & Seddighi (2013)). Mathur *et al.* (2018) interpreted this process in a different way by regarding the preexisting turbulence as a perturbation that leads to the instability of the temporally developing laminar boundary layer from the wall. In any case, the laminar-like flow behaviours at the early stage allow an unsteady friction model to be established to predict the wall shear stress. Based on the assumption that the turbulence in stage 1 is nearly 'frozen', He & Ariyaratne (2011) derived a laminar-flow formulation to describe the wall shear stress at that stage. The acquired results are in good agreement with experimental or computational outcomes as further consolidated in He *et al.* (2011). He & Seddighi (2015) and Jung & Kim (2017) discussed the effects of the ratio of the final to initial Reynolds number and the effects of the acceleration rate on the transition, respectively. They both showed that the turbulence evolves progressively for a low Reynolds number ratio and low acceleration rate, which is in contrast to the aforementioned bypass transition. Guerrero *et al.* (2021) investigated the transient dynamics of the accelerating turbulent pipe flow in detail. By utilizing the FIK identity which is an exact expression developed by Fukagata *et al.* (2002) to quantify the friction coefficient for wall-bounded flows, they were able to quantify the different contributions to the wall friction during the transient. Moreover, Sundstrom & Cervantes (2018*a*) showed that the flow responses during the accelerating phase of the pulsatile flow are similar to those in the first two stages of the uniformly accelerating flow. As for the decelerating flow which is characterized by a decay of the preexisting turbulence (Mathur (2016)), Sundstrom & Cervantes (2018*c*) also demonstrated its laminar similarity to the accelerating flow at the initial stage.

Different from the one-way flow excursion reviewed above, the pulsatile turbulent flow exhibits strong wave-turbulence interactions due to the shear wave generated near the wall. The laminar Stokes thickness $l_s^+ = l_s \bar{u}_\tau / \nu = \sqrt{2\nu/\omega}$, where ν and ω are the kinematic viscosity and the angular pulsatile frequency, respectively, is generally used to characterize the wall-normal length scale of such a near-wall shear wave. In the present paper, the superscript + denotes normalization using the mean friction velocity \bar{u}_τ and the kinematic viscosity ν . To take into account the diffusion effect of turbulence, Scotti & Piomelli (2001) proposed a turbulent Stokes thickness l_t , based on the eddy-viscosity theory, as the scaling parameter. There have been several experimental studies that focused on the pulsatile flows (Ronneberger & Ahrens (1977); Gerrard (1971); Ramaprian & Tu (1980, 1983); Tu & Ramaprian (1983); Mao & Hanratty (1986); Lodahl *et al.* (1998); Shemer & Kit (1984); Shemer *et al.* (1985); Tardu & Binder (1993); Tardu *et al.* (1994); He & Jackson (2009)), covering a wide range of pulsation parameters. The wave-turbulence interactions exhibit a strong frequency dependence. When the pulsation

89 frequency is high ($0.02 \lesssim \omega^+ \lesssim 0.04$), the Stokes thickness is small such that the shear
 90 wave is only confined to the very narrow near-wall region. In this case, the inner shear
 91 wave and outer turbulence are weakly coupled (quasi-laminar state). There is a 45°
 92 phase lag between the centreline velocity and the wall shear stress, which coincides
 93 with the laminar Stokes solution. When the frequency falls in a low-frequency range
 94 ($\omega^+ \lesssim 0.005$), the flow variation is slow enough to allow the turbulence to react and
 95 settle down. Hence, the instantaneous flow field resembles that of the steady flow at
 96 the corresponding Reynolds number (quasi-steady state). Generally speaking, if the
 97 pulsation amplitude is not large enough to induce a reversal flow, the time-averaged
 98 flow quantities remain nearly unchanged from their values in the steady flow (Brereton
 99 *et al.* (1990)). However, Tu & Ramaprian (1983) showed a deviation of the mean velocity
 100 profile from the steady one when the frequency is close to or larger than the turbulent
 101 bursting frequency in the turbulent pipe flow, i.e., a very-high frequency range ($\omega^+ \gtrsim 0.04$).
 102 This can be possibly attributed to a resonance effect with the closeness between the
 103 pulsation frequency and the characteristic frequency of the near-wall coherent structure.
 104 On the contrary, Tardu *et al.* (1994) and Scotti & Piomelli (2001) did not report
 105 such a deviation in turbulent channel flows with different pulsation parameters. This
 106 might be due to the dependency of the bursting frequency on the Reynolds number
 107 and geometry such that a resonance condition is not easily satisfied. Sundstrom *et al.*
 108 (2016) performed experimental research on a double-frequency pulsatile turbulent pipe
 109 flow and showed that the time-averaged flow quantities are also unaffected by the double-
 110 frequency pulsation. For a large-amplitude oscillation (usually refers to a situation when
 111 the oscillatory to mean velocity ratio A is larger than one), Manna *et al.* (2012) reported
 112 a drag-reducing effect which manifests as an upward shift of the mean velocity profile
 113 in the logarithmic region. This is consistent with Mao & Hanratty (1994) and Manna &
 114 Vacca (2005) in which a reduction of wall shear stresses was both reported.

115 The laminar-turbulent transition is also an important phenomenon that occurs in
 116 pulsatile flows. Turbulence can be completely relaminarized in the pulsatile pipe flow
 117 with a non-zero mean flow under certain parameters (Lodahl *et al.* (1998)), and can
 118 also appear intermittently in a purely oscillatory flow (Feldmann & Wagner (2012)),
 119 which belongs to a subcritical transition scenario (Feldmann & Wagner (2016*b*)). Xu
 120 *et al.* (2017) conducted an experimental study on the transition in a pulsatile pipe at
 121 amplitudes $A \leq 0.7$. Based on the transition theory in the steady pipe flow (Avila *et al.*
 122 (2011, 2010); Hof *et al.* (2006)), they summarized the effects of pulsation frequency and
 123 further elaborated in Xu & Avila (2018) with the aid of the direct numerical simulation
 124 (DNS). For a large-amplitude pulsation with $A > 0.7$, Xu *et al.* (2021, 2020) reported a
 125 helical instability mechanism that induces the burst of turbulence in a pulsatile pipe
 126 flow. In particular, this helical disturbance is triggered during the decelerating phase
 127 and disappears in the accelerating phase, indicating a strong phase asymmetry in the
 128 pulsatile pipe flow (Feldmann & Wagner (2016*a*)). Further, Morón *et al.* (2022) linked this
 129 helical instability to the linear stability of the corresponding laminar flow and discussed
 130 the effect of the pulsation wave form on the turbulence transition. Similarly, Feldmann
 131 *et al.* (2020) investigated the spatio-temporal intermittency associated with a competition
 132 between the helical structures and puffs. This intermittency in the pulsatile pipe flow is
 133 qualitatively similar to the gas-liquid slug pipe flow reported in Padrino *et al.* (2023).

134 A paradoxical phenomenon occurs in the intermediate frequency range ($0.005 \lesssim \omega^+ \lesssim 0.02$).
 135 In this frequency range, the amplitude of the oscillating wall shear stress in a turbulent
 136 flow ($A_{\tau_w, t}$) is smaller than that in a laminar flow with the same pulsation conditions
 137 ($A_{\tau_w, s}$). This suggests a turbulence-induced drag reduction that is opposite to that in
 138 the steady flow where turbulence generally produces a larger drag than a laminar flow

(Mao & Hanratty (1986); Tardu *et al.* (1994); Sundstrom & Cervantes (2018a)). To deal with a non-closure problem in the governing equation, several researchers have established theoretical models for the Reynolds shear stress based on the concept of eddy-viscosity. Most of these models in general fail to describe accurately the paradoxical phenomenon due to the inherent assumption that the Reynolds stress is in phase with the imposed oscillation (Weng *et al.* (2016)). Weng *et al.* (2013, 2016) introduced the time-dependency of the Reynolds shear stress into the standard eddy-viscosity model, which brings a phase lag between the Reynolds shear stress and the oscillating shear strain rate. This improved model is shown to be able to predict correctly the paradoxical phenomenon. However, a detailed explanation on why and how the turbulence reduces the wall shear stress is still lacking. Sundstrom & Cervantes (2018b) provided a new interpretation of this paradoxical phenomenon by decomposing the total wall shear stress into the contributions from the oscillating pressure gradient (τ_p) and the Reynolds shear stress (τ_s). It is shown that a phase shift between τ_p and τ_s results in a mutual cancellation that leads to the reduction of the total wall shear stress. Nevertheless, in their experimental study, τ_s is calculated by subtracting τ_p from the measured total wall shear stress while τ_p is calculated by the laminar Stokes solution. Hence, this interpretation still cannot explain the underlying mechanisms; for instance, it does not explain what causes the phase shift. Based on these, we choose to perform direct numerical simulations in turbulent pulsatile pipe flows, in the hope of extending the numerical database of pulsatile flows, elucidating the physical mechanisms that cause the paradoxical phenomenon and revealing the corresponding turbulence dynamics in detail.

This paper is organized as follows. A computational set-up is introduced in §2. Some basic statistics and the properties of the varying wall shear stress are given in §3. §4 explores the causes of the paradoxical phenomenon, and §5 discusses the hysteresis phenomenon of the wall shear stress. The effects of the pulsation amplitude are examined in §6. §7 further examines the phase-wise variation of spectra, and §8 summarizes the main findings of this paper.

2. Computational set-up

For pulsatile flows, it is common to introduce a triple decomposition of the flow quantity $f(\mathbf{x}, t)$ (Hussain & Reynolds (1970); Sundstrom & Cervantes (2018b); Weng *et al.* (2016)):

$$f(\mathbf{x}, t) = \langle f \rangle(\mathbf{x}, \varphi) + f'(\mathbf{x}, t) = \bar{f}(\mathbf{x}) + \tilde{f}(\mathbf{x}, \varphi) + f'(\mathbf{x}, t), \quad (2.1)$$

where \bar{f} and $\langle f \rangle$ are the time-averaged and phased-averaged values, \tilde{f} is the oscillating component, f' is the turbulent fluctuation and φ is the phase. By further including the spatial average in statistically homogeneous directions, the time and phase averages can be defined as:

$$\langle f \rangle(y, \varphi) = \lim_{M \rightarrow \infty} \frac{1}{2\pi ML} \sum_{n=1}^M \int_0^L \int_0^{2\pi} f(\mathbf{x}, t + \frac{2\pi n}{\omega}) dx d\theta, \quad (2.2a)$$

$$\bar{f}(y) = \int_0^{2\pi} \langle f \rangle(y, \varphi) d\varphi, \quad (2.2b)$$

where $x, y=R-r, \theta$ denote the streamwise, wall-normal and azimuthal directions, respectively, with the corresponding velocity components being $u, -v$ and w . The pipe radius is R , L is the pipe length, and M is the total number of periods.

In the present study, the pulsation is achieved by imposing a sinusoidally varying

180 component to the constant streamwise pressure gradient that drives the steady turbulent
181 pipe flow:

$$\frac{\partial p}{\partial x}(t) = \frac{\partial \bar{p}}{\partial x}(1 + \beta \sin(\omega t)). \quad (2.3)$$

182 The constant mean pressure gradient $\partial \bar{p} / \partial x$ is set to aim for a steady turbulent flow
183 of $Re_\tau = \bar{u}_\tau R / \nu = 180$. Throughout the paper, the quantity φ refers to a phase of the
184 streamwise pressure gradient $\partial p / \partial x$. We use the ratio of the amplitude of the oscillating
185 velocity to the mean velocity at the pipe centreline, i.e. $A = A_{\tilde{u}_{cl}} / \bar{u}_{cl}$ to characterize the
186 pulsation amplitude. Using (2.1), the governing equation for the oscillating component
187 of the streamwise velocity reads:

$$\frac{\partial \tilde{u}}{\partial t} = -\frac{\partial \tilde{p}}{\partial x} + \nu \frac{1}{r} \frac{\partial \tilde{u}}{\partial r} + \nu \frac{\partial^2 \tilde{u}}{\partial r^2} - \underbrace{\frac{\partial r u' v'}{r \partial r}}_{\varphi}. \quad (2.4)$$

188 By evaluating (2.4) at the pipe centreline, we have:

$$A = -\frac{\partial \bar{p}}{\partial x} \frac{\beta}{\omega} \quad (2.5)$$

189 Hence, β can be predetermined based on the desired pulsation amplitude A .

190 To enable a direct comparison, the pulsation parameters selected in this paper, which
191 are given in table 1, are similar to those in Weng *et al.* (2016) and Sundstrom & Cervantes
192 (2018b) (Figure 1a). Five frequencies are chosen for $A=0.1$ (cases 1 to 5), and the resulting
193 Womersley numbers $W = R\sqrt{\omega/\nu}$ ranges from 12.7 to 63.6. In addition, simulations of
194 cases 6 and 7 are performed at a higher amplitude of $A=0.4$ to evaluate the effect of
195 pulsation amplitude. To provide a more comprehensive comparison with previous studies,
196 each case is plotted in the Re_b - Re_w plane (Lodahl *et al.* (1998)) together with available
197 literature data (figure 1b). Here, $Re_b = U_b D / \nu$ is the bulk Reynolds number based on the
198 bulk velocity U_b and the pipe diameter D , and $Re_w = A_{\tilde{u}_{cl}}^2 / \omega \nu$ is the oscillatory Reynolds
199 number. It is clear that all cases fall in the turbulent regime, as demonstrated in figure 3
200 where the instantaneous streamwise velocity at the meridional plane for case 7 is shown.
201 Albeit there is a large coincidence in the parameter space between the current study and
202 Cheng *et al.* (2020), we note that our goal is not to carry out an investigation in the
203 unexplored parameter space but to further elucidate the mechanisms responsible for the
204 aforementioned paradoxical phenomenon based on existing parameters in the literature.
205 Further, all the simulation cases in Cheng *et al.* (2020) are conducted at the fixed β while
206 the pulsation amplitude A is fixed in the present study. Therefore, our cases are in fact
207 different from those in Cheng *et al.* (2020) according to equation (2.5). For each case,
208 the pulsation is imposed at a single flow field of the steady turbulent pipe flow. After
209 discarding the transient effect, the phase-averaged data are collected over more than 23
210 periods to obtain the final statistics.

211 A cylindrical-coordinate spectral element-Fourier DNS solver *Semtex* is employed to
212 conduct the simulations (Blackburn & Sherwin (2004); Blackburn *et al.* (2019)). The
213 computational mesh is the same as that in Liu *et al.* (2022), where a 50×10 two-
214 dimensional spectral element mesh is deployed to discretize the meridional semi-plane.
215 192 Fourier expansion planes are used in the azimuthal direction to represent the three-
216 dimensional computational domain. The pipe length is set to be $L = 6\pi R$. In order to
217 ensure that this mesh configuration is appropriate to resolve the precise turbulence
218 dynamics, we choose cases 3 and 7 to conduct the mesh independence test. The choice
219 of these two cases is based on the fact that the former is the case for which the ratio
220 $A_{\tau_w, t} / A_{\tau_w, s}$ reaches its minimum and the latter is a high-amplitude case in which the

Case	A	ω^+	l_s^+	W	Re_b	Re_w	$A_{\tau_w,t}^-/A_{\tau_w,s}^-$
1	0.1	0.125	4	63.6	5265	30	1
2	0.1	0.016	11	23.1	5273	229	0.94
3	0.1	0.01	14	18.2	5277	416	0.65
4	0.1	0.007	17	15.0	5280	924	0.86
5	0.1	0.005	20	12.7	5260	1325	1.41
6	0.4	0.01	14	18.2	5654	6230	0.80
7	0.4	0.007	17	15.0	5796	12559	0.59

TABLE 1. Cases with different pulsation parameters, namely the pulsation amplitude A , the frequency ω^+ . l_s^+ is the laminar Stokes thickness, W is the Womersley number. $A_{\tau_w,t}^-/A_{\tau_w,s}^-$ is the amplitude of the wall shear stress normalized by its laminar Stokes value. $Re_b=U_b D/\nu$ and $Re_w=A_{\bar{u}_{cl}}^2/\omega\nu$ are the bulk and oscillatory Reynolds numbers, respectively.

Case	P	N	H	Δx^+	Δy^+	$\Delta(r\theta)_{wall}^+$
Baseline	11	192	50	[2.24, 10.0]	[0.18, 5.91]	5.89
3-1	10	192	50	[2.73, 11.2]	[0.21, 6.61]	5.89
3-2	12	192	50	[1.87, 9.26]	[0.15, 5.46]	5.89
3-3	11	240	50	[2.24, 10.0]	[0.18, 5.91]	4.71
7-1	11	240	50	[2.24, 10.0]	[0.18, 5.91]	4.71
7-2	12	192	50	[1.87, 9.26]	[0.15, 5.46]	5.89
7-3	11	192	65	[1.72, 7.71]	[0.18, 5.91]	5.89

TABLE 2. Summary of the grid information. Δx^+ , Δy^+ are the normalized streamwise, wall-normal grid resolutions. $\Delta(r\theta)_{wall}^+$ is the normalized circumferential grid resolution at the wall. P is the number of Lagrange knot points along the side of each element, corresponding to a polynomial order of $P-1$. N represents the number of Fourier expansion planes in the circumferential direction. H denotes the number of elements in the streamwise direction, which is associated with the streamwise h -refinement strategy in the spectral element method.

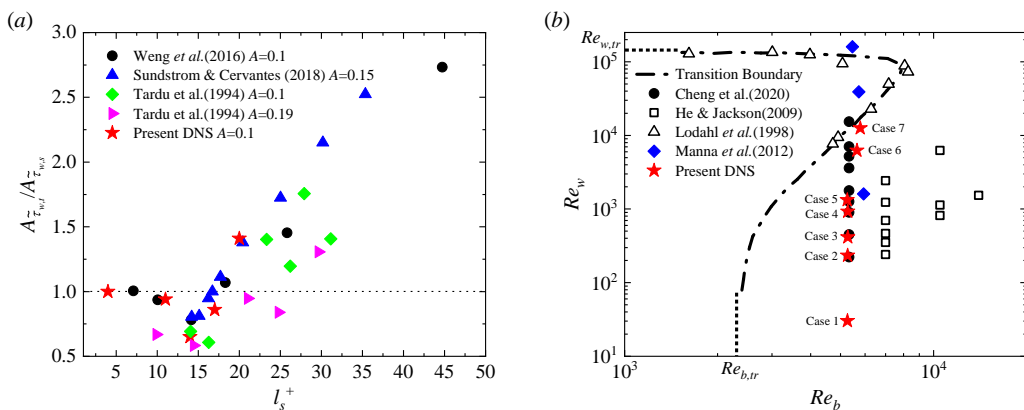


FIGURE 1. (a) Variation of the amplitude of the wall shear stress $A_{\tau_w,t}^-$ normalized by the laminar Stokes amplitude $A_{\tau_w,s}^-$ with respect to the laminar Stokes thickness l_s^+ . (b) Laminar-turbulent transition boundary (chain-dotted line, Lodahl et al. (1998)) and parameter combinations considered by previous studies. The critical Reynolds numbers $Re_{b,tr}$ and $Re_{w,tr}$ are denoted by the dashed lines.

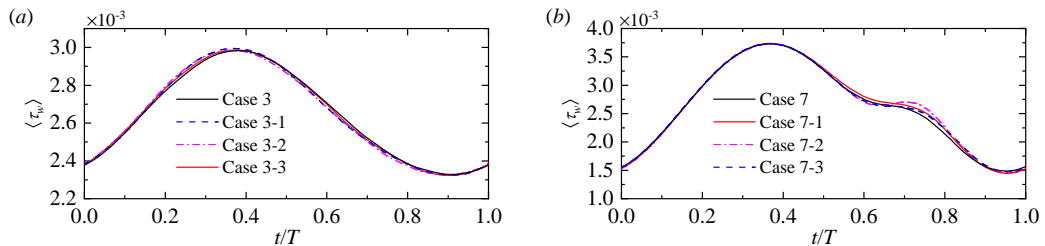


FIGURE 2. Comparisons of the phase-averaged wall shear stress $\langle \tau_w \rangle$ between different mesh-independence validation cases. (a) Case 3 ($A=0.1, l_s^+=14$), (b) Case 7 ($A=0.4, l_s^+=17$).

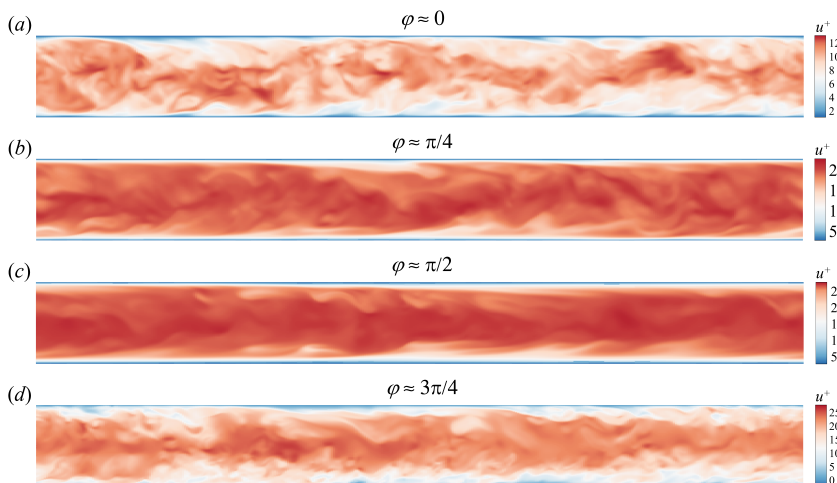


FIGURE 3. Contours of the instantaneous inner-scaled streamwise velocity u^+ in the meridional plane for case 7 ($A=0.4, l_s^+=17$) at (a) $\varphi \approx 0$, (b) $\varphi \approx \pi/4$, (c) $\varphi \approx \pi/2$, (d) $\varphi \approx 3\pi/4$.

221 variation of turbulence dynamics is more intense. Details of the spatial grid resolution are
 222 given in table 2. The hp -refinement strategy for the spectral element method is employed.
 223 Our baseline grid resolutions are comparable with the regular and high resolutions
 224 reported in Zahtila *et al.* (2023). Figure 2 shows the phase-averaged wall shear stress
 225 obtained from various grids. It can be found that all the curves for case 3 overlap well
 226 and the kink variation trend in case 7 can be observed for all test cases, indicating that
 227 our phase-averaged statistics are grid-independent. Figure 1(a) compares the present
 228 results of $A_{\widetilde{\tau}_w, t} / A_{\widetilde{\tau}_w, s}$ for $A=0.1$ with previous studies. It is shown that the paradoxical
 229 phenomenon can be clearly reproduced and that the variation trend of $A_{\widetilde{\tau}_w, t} / A_{\widetilde{\tau}_w, s}$
 230 with respect to l_s^+ is also consistent with that in the literature. Note that the temporal
 231 variation of the wall shear stress is not necessarily a pure sine function; thus, here the
 232 amplitude corresponds to the amplitude of the fundamental mode calculated from the
 233 Fourier analysis. Quantitatively, a reasonable agreement can also be found except that
 234 the minimum of $A_{\widetilde{\tau}_w, t} / A_{\widetilde{\tau}_w, s}$ is 0.65 at $l_s^+=14$, which is smaller compared with that in
 235 Weng *et al.* (2016) and Sundstrom & Cervantes (2018b) but in agreement with Tardu
 236 *et al.* (1994). This can probably be attributed to the different Reynolds numbers used in
 237 these studies. Moreover, the instantaneous fields in figure 3 exhibit visual smoothness,
 238 and no mesh imprints can be found. Hence, these results give us confidence in accuracy
 239 of the DNS data used in this paper.

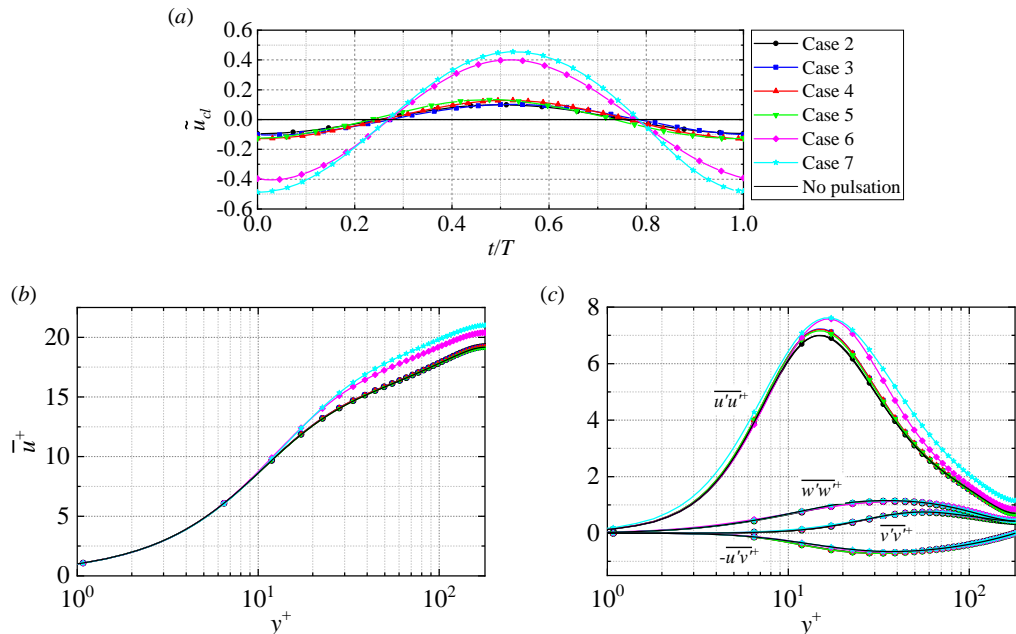


FIGURE 4. (a) Time evolution of the oscillating component of the streamwise velocity \tilde{u}_{cl} at the pipe centreline. (b) Wall-normal profiles of the mean streamwise velocity \bar{u}^+ . (c) Wall-normal profiles of the components of the Reynolds stress tensor. Note that the curves for steady pipe (no pulsation) overlap completely with those for case 2 in (b)(c).

3. Fundamental characteristics of pulsatile flow

First, we present some basic properties of the pulsatile pipe flow. Figure 4(a) shows the oscillating component of the streamwise velocity at the pipe centreline \tilde{u}_{cl} . As expected, the oscillating amplitudes for high-frequency cases (cases 2 and 3) are exactly $A=0.1$, while it is slightly larger than 0.1 for low-frequency cases (cases 4 and 5). This is due to the strong coupling of the near-wall shear layer and the central region for low-frequency cases (Weng *et al.* (2016)). A similar phenomenon can also be found for higher amplitude cases where the centreline velocity oscillates at an amplitude larger than 0.4 for case 7. Figure 4(b)(c) shows the wall-normal profiles of the normalized mean velocity \bar{u} and the components of the Reynolds stress tensor. The mean velocity is insensitive to the frequency at a low amplitude of $A=0.1$, but an increase in amplitude leads to the elevation of \bar{u} in the log region. This is consistent with previous studies (Scotti & Piomelli (2001); Manna *et al.* (2012)). Besides, the mean velocity seems to be more sensitive to the frequency for high-amplitude cases. For the presented Reynolds stresses, all the curves collapse well except for the $\overline{u'u'}$. The increase of amplitude produces a larger magnitude of $\overline{u'u'}$ beyond $y^+ \approx 10$ and $\overline{u'u'}$ is insensitive to the frequency for low-amplitude cases. Furthermore, the increase of $\overline{u'u'}$ is accompanied by the wall-normal location of the maximum $\overline{u'u'}$ moving away from the wall, which is similar to the situation where a transverse Stokes layer is generated by the wall oscillation (Quadrio & Sibilla (2000); Liu *et al.* (2022)).

The near-wall flow dynamics are directly reflected by the wall shear stress. Figure 5 compares the phase-wise variation of the wall shear stress $\widetilde{\tau}_w$ (black solid lines) with their corresponding laminar Stokes values (dashed blue lines) for cases with $A=0.1$, with the latter calculated by evaluating the radial derivative of the laminar Stokes solution at the

264 wall (see Manna *et al.* (2012)):

$$\tilde{u}(r, t) = A \cdot \operatorname{Re} \left[i \left(\frac{J_0(i^{3/2}\sqrt{2}r/l_s)}{J_0(i^{3/2}\sqrt{2}R/l_s)} - 1 \right) e^{i\omega(t-T/4)} \right], \quad (3.1)$$

265 where J_0 is the Bessel function of the first kind of order zero, $\operatorname{Re}[\cdot]$ represents the real part
 266 of the argument, i is the imaginary unit and T is the pulsation period. Also included is
 267 the corresponding fundamental mode obtained from the Fourier analysis (red solid lines),
 268 which allows us to evaluate qualitatively the extent of the nonlinear effect due to the
 269 turbulence. For a better presentation of the variation tendency, the data are duplicated
 270 and then spliced such that two periods are shown.

271 For case 1, the three curves overlap completely, indicating a quasi-laminar flow state.
 272 The phase-wise variation of the wall shear stress follows a purely sinusoidal pattern (figure
 273 1a). A mild increase in frequency leads to the subtle departure from the laminar Stokes
 274 value and the decrease of $A_{\widetilde{\tau}_w, t} / A_{\widetilde{\tau}_w, s}$ from unity for case 2 (figure 1b). For case 3,
 275 it is clear that the amplitude of the wall shear stress is significantly smaller than its
 276 laminar value but they still synchronize in phase. The differences between the wall shear
 277 stress and its fundamental mode are subtle, suggesting that the phase symmetry still
 278 holds. Considerable changes occur for case 4. As seen, the wall shear stress deviates
 279 significantly from its fundamental mode. The drag-increasing phase occupies for a longer
 280 portion of the cycle than the drag-decreasing phase. That is, a hysteresis occurs during
 281 the oscillation cycle, indicating the destruction of the phase symmetry. Interestingly,
 282 when approaching the maxima, the increasing rate decreases, leading to a stage of
 283 the high-level wall shear stress with a slow growth. Similar phenomena have also been
 284 reported experimentally by Sundstrom & Cervantes (2018b), where up to 500 cycles
 285 of measurements have been performed to obtain the phase-averaged wall shear stress
 286 at a higher amplitude and Reynolds number. Nevertheless, their data still suffer from
 287 fluctuations due to measurement uncertainties. Chen *et al.* (2014) also reported the same
 288 tendency for similar pulsation parameters, but their data were obtained only from the
 289 final period. In the present study, the sufficient number of averaging periods and the
 290 spatial average in homogeneous directions ensure the smoothness of the phase-averaged
 291 statistics. In addition, the wall shear stress lags behind the laminar value, which does
 292 not occur in former cases. This discrepancy and the physical meaning of this phase
 293 lag will be discussed in section 5. For the smallest frequency considered (case 5), the
 294 deviations from the fundamental mode are still observable, and the hysteresis is less
 295 evident but discernable. A phase lag with respect to the laminar value can also be
 296 observed. Theoretically, if the frequency is small enough to reach the quasi-steady state,
 297 all three curves should be in phase with the sinusoidally varying pressure gradient, and no
 298 hysteresis occurs. Hence, it can be inferred that as the frequency increases from zero, the
 299 aforementioned phase lag and hysteresis emerge initially and then disappear gradually.

300 Figure 6 shows the phase-wise variation of the wall shear stress for cases with $A=0.4$.
 301 A reduction of the amplitudes compared with the corresponding laminar value is still
 302 observable. Specifically, $A_{\widetilde{\tau}_w, t} / A_{\widetilde{\tau}_w, s}$ increases from 0.65 to 0.8 as the amplitude increases
 303 from 0.1 to 0.4 for $l_s^+ = 14$ while it decreases from 0.86 to 0.59 for $l_s^+ = 17$. For case
 304 6, a distinct deviation from the fundamental mode occurs when the wall shear stress
 305 reaches its minimum (negative peak of $\widetilde{\tau}_w$), which implies a strong nonlinear effect of
 306 turbulence at that phase. Same as that for $A=0.1$, the wall shear stress is still in phase
 307 with the laminar value. For case 7, a distinct kink can be clearly observed at around
 308 the phase of $t=3T/4$. We note that both Tu & Ramaprian (1983) and Scotti & Piomelli
 309 (2001) reported such kink, but the causes are still unclear. Such kink also implies special

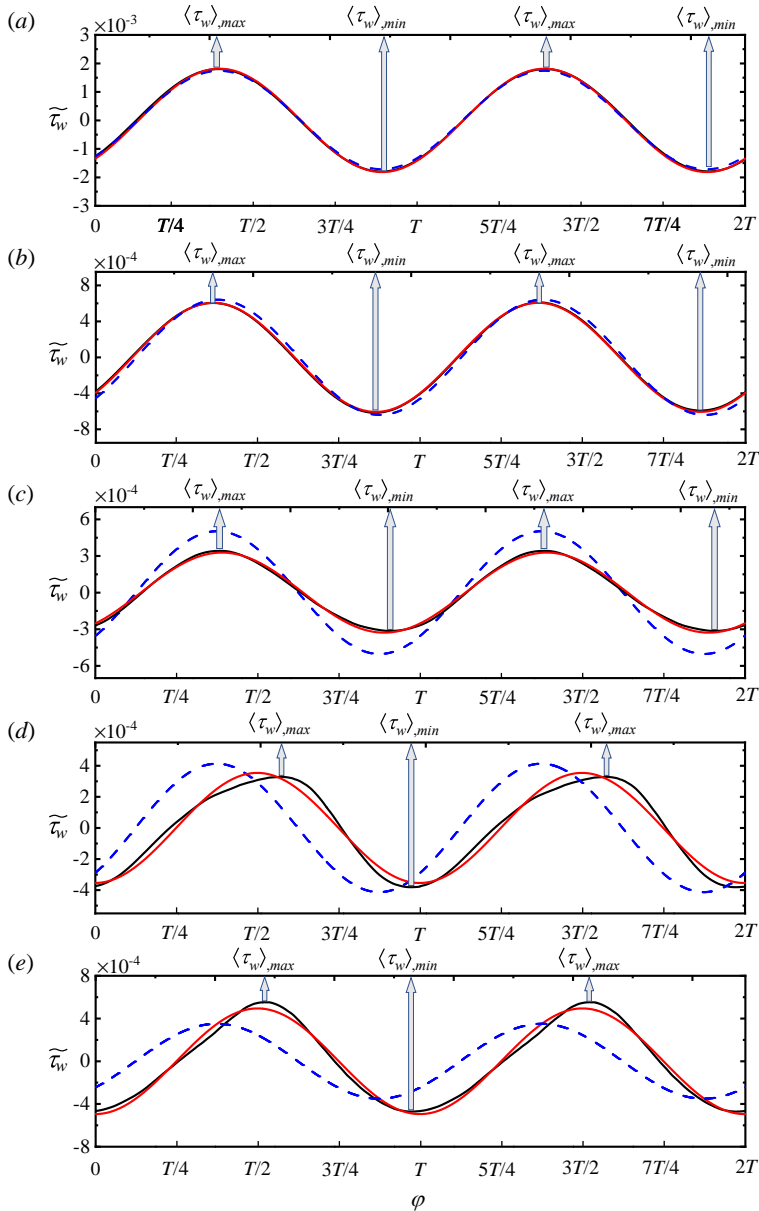


FIGURE 5. Phase-wise variations of the oscillating component of the wall shear stress $\tilde{\tau}_w$ for $A=0.1$ (the black lines), with the corresponding fundamental Fourier mode represented by the red solid lines. The blue dashed lines correspond to the laminar Stokes solution. The vertical arrows denote the phases where the phase-averaged wall shear stress $\langle \tau_w \rangle$ reaches its maximum or minimum. (a) Case 1. (b) Case 2. (c) Case 3. (d) Case 4. (e) Case 5.

310 turbulence dynamics and will be discussed in detail in section 6. The fundamental mode
 311 lags behind the laminar value, consistent with that for $A=0.1$. It is worth noting that we
 312 only consider a higher amplitude at two intermediate frequencies. In fact, increasing the
 313 amplitude will not change the quasi-laminar state for high-frequency cases (e.g. cases 1
 314 and 2), hence their wall shear stresses are expected to still follow the laminar solution;
 315 while for a low-frequency case where the quasi-steady state is reached, a higher pulsation

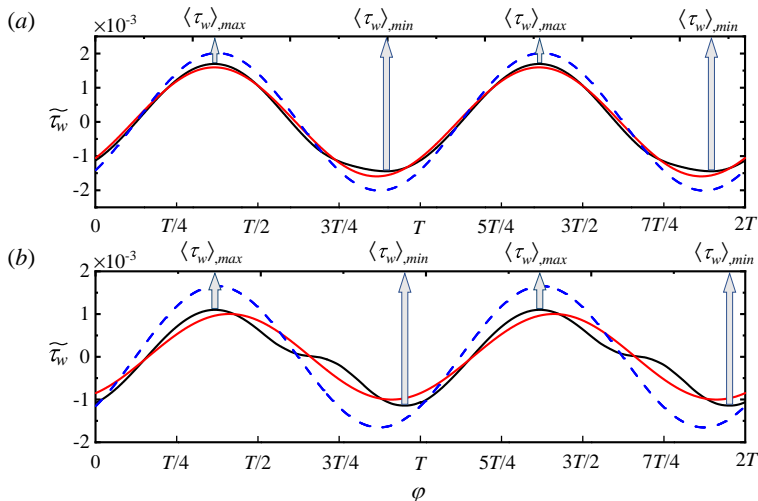


FIGURE 6. Phase-wise variations of the oscillating component of the wall shear stress $\tilde{\tau}_w$ for $A=0.4$ (the black lines), with the corresponding fundamental Fourier mode represented by the red solid lines. The blue dashed lines correspond to the laminar Stokes solution. The vertical arrows denote the phases where the phase-averaged wall shear stress $\langle \tau_w \rangle$ reaches its maximum or minimum. (a) Case 6. (b) Case 7.

316 amplitude leads to a higher amplitude of the wall shear stress, but the variation trend
 317 would still follow a sinusoidal manner. Case 5 can be considered as a transition case for
 318 which the flow state is closer to the quasi-steady state. The hysteresis phenomenon at the
 319 frequency of case 5 with a higher amplitude is probably not evident. Thus, behaviours
 320 of the wall shear stress for cases with high or low frequencies are somewhat predictable,
 321 which is the reason for considering only these two frequencies with a higher amplitude.

322 In this section, we provide a general description of the pulsatile pipe flow, with a
 323 focus on the phase-wise variation of the wall shear stress. Different combinations of the
 324 pulsation parameters lead to significantly different behaviours of $\tilde{\tau}_w$. In the following
 325 sections, we aim to explore the causes of such differences and the corresponding flow
 326 dynamics based on (2.4) since $\tilde{\tau}_w$ correlates closely to \tilde{u} near the wall.

327 4. Damping mechanisms of the oscillating wall shear stress

328 We focus on case 3, for which the ratio $A_{\tilde{\tau}_w,t}/A_{\tilde{\tau}_w,s}$ reaches its minimum, to elucidate
 329 the mechanisms that lead to the amplitude reduction of the oscillating wall shear stress
 330 ($\tilde{\tau}_w$) in the turbulent state compared with its corresponding laminar value.

331 The phase-wise variation of \tilde{u} for case 3 is shown in figure 7(b), together with its laminar
 332 value calculated from (3.1) shown in figure 7(a). As seen, the most dramatic difference
 333 resides in the near-wall region; namely, below $y^+ \approx 20$, the laminar contour appears to
 334 be more distorted than that in the turbulent case in that region. By examining the
 335 contour values, it can be observed that the amplitude of the oscillating \tilde{u} is smaller in
 336 the turbulent case, corresponding to the smaller amplitude of $\tilde{\tau}_w$ shown in figure 5(c).
 337 In the outer layer, the laminar and turbulent cases behave almost the same. According
 338 to (2.4), the key term that the laminar equation lacks is the weighted radial derivative
 339 of the Reynolds shear stress $\widetilde{u'v'}$ (Sundstrom & Cervantes (2018c)), which we denote
 340 as φ . This term can be further decomposed into the derivative term $-\partial \widetilde{u'v'}/\partial r$ and the
 341 curvature term $-\widetilde{u'v'}/r$, for which the latter is generally negligible compared with the

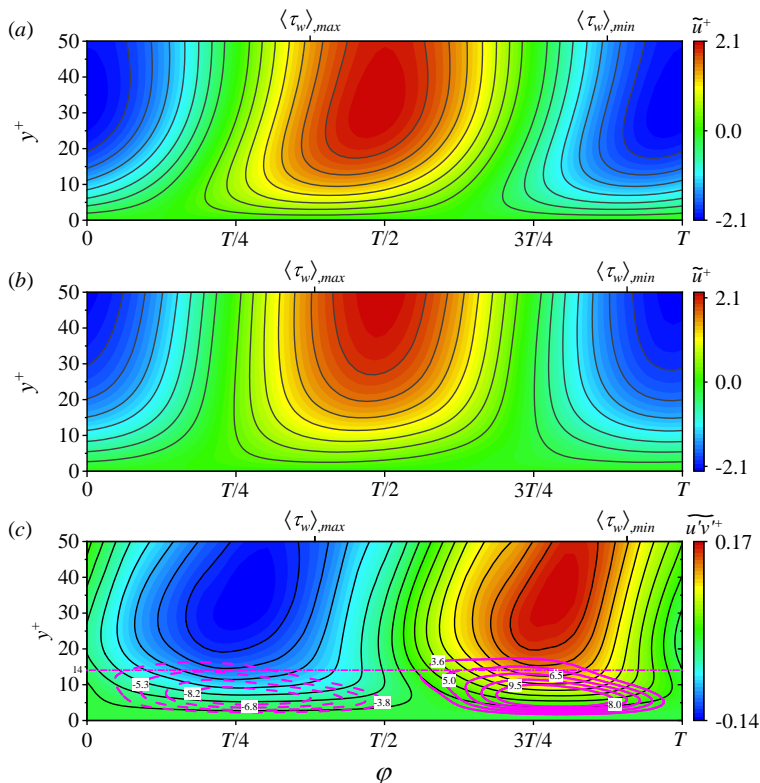


FIGURE 7. Phase-wise variations of (a) the laminar Stokes velocity \tilde{u}^+ , (b) the turbulent streamwise velocity \tilde{u}^+ , (c) the Reynolds shear stress $\tilde{u}'v'^+$ for case 3 ($A=0.1$, $l_s^+=14$). The magenta contours in (c) represent the term φ , with the values (divided by 10^{-3}) marked on the contour lines. The magenta chain-dotted straight line indicates the laminar Stokes thickness of $l_s^+=14$.

342 former in the near-wall region. Hence, we examine the phase-wise variation of $\tilde{u}'v'$ in
 343 figure 7(c). Importantly, there is a phase lag between $\tilde{u}'v'$ and \tilde{u} such that the positive
 344 and negative peaks of $\tilde{u}'v'$ are reached when $|\partial\tilde{u}/\partial t|$ reaches its maximum (the regions
 345 between the blue and red contours). Evidently, the term φ is prominent below the blue
 346 and red contours, as shown by the magenta contours in figure 7(c). For negative $\tilde{u}'v'$
 347 (blue region, at $t \approx T/4$), the radial derivative of $\tilde{u}'v'$ near the wall is positive, hence the
 348 term φ acts as a sink term that leads to a reduction of the positive acceleration $\partial\tilde{u}/\partial t$.
 349 Similarly for positive $\tilde{u}'v'$ (red region, at $t \approx 3T/4$), the term φ contributes positively to
 350 the negative $\partial\tilde{u}/\partial t$. Thus, in the turbulent case, the acceleration and deceleration of \tilde{u}
 351 in the near-wall region are both damped by term φ compared with its laminar value,
 352 therefore causing the reduction of the amplitude of \tilde{u} and then the wall shear stress. It
 353 is noted that, in the near-wall region of a steady fully-developed turbulent pipe flow, $-$
 354 $d\tilde{u}'v'/dr$ acts as a gain term in the \tilde{u} transport equation and hence contributes positively
 355 to the mean velocity ($-\tilde{u}'v'/r$ is comparatively negligible) (Wu & Moin (2008)). Thus,
 356 the effect of turbulence is reversed by the imposed unsteadiness. It should also be noted
 357 that the temporal variation of $\tilde{u}'v'$ is not strictly sinusoidal, i.e. the time span of the
 358 blue contour is not strictly the same as that of the red contour, and the same goes for

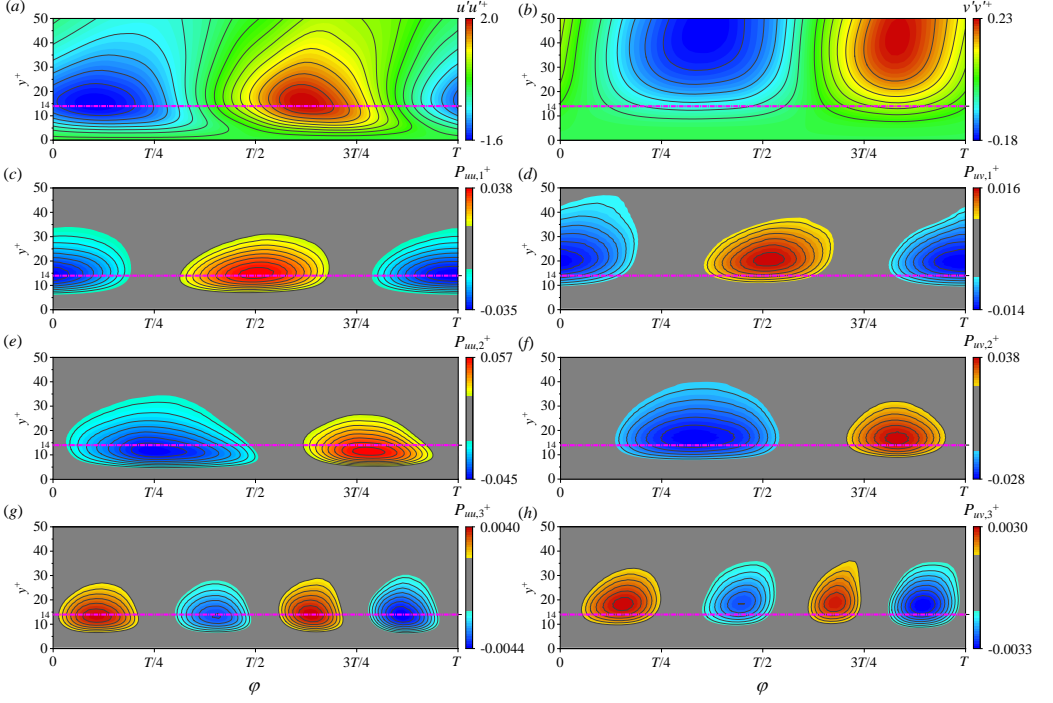


FIGURE 8. Phase-wise variations of (a) the streamwise Reynolds normal stress $\widetilde{u'u'^+}$, (b) the radial Reynolds normal stress $\widetilde{v'v'^+}$ for case 3 ($A=0.1$, $l_s^+=14$). Phase-wise variations of the production terms in (4.1) for case 3 ($A=0.1$, $l_s^+=14$): (c) $P_{uu,1}^+$, (e) $P_{uu,2}^+$, (g) $P_{uu,3}^+$, (d) $P_{uv,1}^+$, (f) $P_{uv,2}^+$, (h) $P_{uv,3}^+$. The magenta chain-dotted lines indicate the laminar Stokes thickness of $l_s^+=14$.

359 the following contours. This coincides with the fact that the wall shear stress deviates
 360 slightly from its fundamental mode as shown in figure 5(c).

361 Apparently, it is the phase lag between \widetilde{u} and $\widetilde{u'v'}$ that causes the nearly antiphase
 362 variation between the term φ and $\partial\widetilde{u}/\partial t$, leading to the subsequent reduction of the
 363 amplitude of $\widetilde{\tau}_w$. To further understand the origin of such phase lag, we examine the
 364 phase-wise variations of $\widetilde{u'u'}$, $\widetilde{v'v'}$ and the production terms of $\widetilde{u'u'}$ and $\widetilde{u'v'}$ (shown in
 365 figure 8):

$$P_{\widetilde{u'u'}} = \underbrace{-\overline{u'v'} \frac{\partial \widetilde{u}}{\partial r}}_{P_{uu,1}} - \underbrace{\widetilde{u'v'} \frac{\partial \bar{u}}{\partial r}}_{P_{uu,2}} - \underbrace{(\widetilde{u'v'} \frac{\partial \widetilde{u}}{\partial r} - \overline{u'v'} \frac{\partial \widetilde{u}}{\partial r})}_{P_{uu,3}}, \quad (4.1a)$$

366

$$P_{\widetilde{u'v'}} = \underbrace{-\overline{v'v'} \frac{\partial \widetilde{u}}{\partial r}}_{P_{uv,1}} - \underbrace{\widetilde{v'v'} \frac{\partial \bar{u}}{\partial r}}_{P_{uv,2}} - \underbrace{(\widetilde{v'v'} \frac{\partial \widetilde{u}}{\partial r} - \overline{v'v'} \frac{\partial \widetilde{u}}{\partial r})}_{P_{uv,3}}. \quad (4.1b)$$

367 These quantities are closely correlated as $\widetilde{v'v'}$ dictates the production terms of $\widetilde{u'v'}$; $\widetilde{u'u'}$
 368 dictates the production terms of $\widetilde{u'u'}$, and the energy redistribution from $\widetilde{u'u'}$ directly
 369 feeds energy into $\widetilde{v'v'}$. In addition, the laminar Stokes thickness $l_s^+=14$ (magenta chain-
 370 dotted line) is included to mark the edge of the laminar Stokes layer.

371 The phase-wise variation of $\widetilde{u'u'}$ is well correlated with \widetilde{u} , with the latter leading

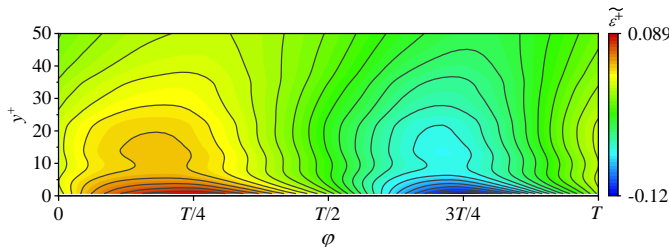


FIGURE 9. Phase-wise variation of the dissipation rate $\widetilde{\varepsilon}^+$ of the TKE $k=\overline{u'_i u'_i}$ ($i=1,2,3$) for case 3 ($A=0.1$, $l_s^+=14$). Note that the time-averaged dissipation rate $\overline{\varepsilon}^+$ is negative, thus a negative value of the oscillating component indicates the enhanced dissipation rate. For the definition of $\widetilde{\varepsilon}^+$, we refer the reader to Eggels *et al.* (1994) for further details.

372 by a small phase margin. Since the streamwise velocity fluctuation is directly linked
 373 to the near-wall streak, this phase lag reflects the inertial effect of the turbulence
 374 structure. Here, the contour only gives the information regarding the intensity of near-
 375 wall streaks. It will be shown later by the spectrum analysis that the increase of $\widetilde{u'u'}$
 376 is also accompanied by the elongation of near-wall streaks. The shear-strain-oscillated
 377 production $P_{uu,1}$ leads $\widetilde{u'u'}$ in phase and the wall-normal location of the maximum
 378 coincides with that of $\widetilde{u'u'}$. The Reynolds-stress-oscillated production $P_{uu,2}$ peaks inside
 379 the Stokes layer, and its magnitude is larger than $P_{uu,1}$. Thus, the main portion of the
 380 near-wall streak is dominated by the oscillating shear strain rate through $P_{uu,1}$, with the
 381 phase lag corresponding to the delayed development of the streak. $P_{uu,2}$ is in phase with
 382 $\widetilde{u'v'}$ and therefore responsible for the distortion of $\widetilde{u'u'}$ contour within the Stokes layer.
 383 The phase-wise variation of $\widetilde{v'v'}$ further lags behind $\widetilde{u'u'}$; it reaches a positive (negative)
 384 peak during the deceleration (acceleration) phase of \widetilde{u} . The production terms $P_{uv,1}$ and
 385 $P_{uv,2}$ both peak outside the Stokes layer, but $P_{uv,2}$ is more correlated with $\widetilde{u'v'}$ and
 386 its magnitude is larger than $P_{uv,1}$. Thus, it can be inferred that the variation of $\widetilde{u'v'}$ is
 387 dominated by $\widetilde{v'v'}$ rather than the oscillating shear strain rate, but the existence of $P_{uv,1}$
 388 makes the $\widetilde{u'v'}$ lead $\widetilde{v'v'}$ by a small phase margin. The phase lag between $\widetilde{v'v'}$ and $\widetilde{u'u'}$
 389 indicates the energy redistribution from $\widetilde{u'u'}$ to $\widetilde{v'v'}$ (The same goes for $\widetilde{w'w'}$, not shown
 390 here. The variation of $\widetilde{w'w'}$ is in phase with $\widetilde{v'v'}$). Moreover, the magnitude of $\widetilde{u'u'}$ is
 391 one order of magnitude larger than $\widetilde{v'v'}$ and $\widetilde{w'w'}$. The reduction of the energy of $\widetilde{u'u'}$
 392 does not match the total energy gain of $\widetilde{v'v'}$ and $\widetilde{w'w'}$, which implies that there must
 393 be significant dissipation of the turbulent kinetic energy (TKE) during the deceleration
 394 phase of \widetilde{u} as confirmed in figure 9. It should be noted that the nonlinear production
 395 terms $P_{uu,3}$ and $P_{uv,3}$ are both negligible compared with the other linear production
 396 terms.

397 Based on the information given, we can summarize the flow evolution with a schematic
 398 in figure 10. There are two circulations representing the time evolution of $\widetilde{u'u'}$ and the
 399 bulk velocity U_b , respectively. The inside of the U_b loop corresponds roughly to the inner
 400 Stokes layer, while the outside is associated with the region outside the Stokes layer. As
 401 shown in figure 8(a), the majority of $\widetilde{u'u'}$ contours are outside the Stokes layer, hence
 402 here the $\widetilde{u'u'}$ loop encloses the U_b loop. Two main points deserve to be highlighted. One
 403 is the time delays that are associated with the development of near-wall streaks and the
 404 energy redistribution process. The former corresponds to a phase lag between \widetilde{u} and $\widetilde{u'u'}$,
 405 and the latter gives rise to a phase lag between $\widetilde{u'u'}$ and $\widetilde{v'v'}$. These time delays together

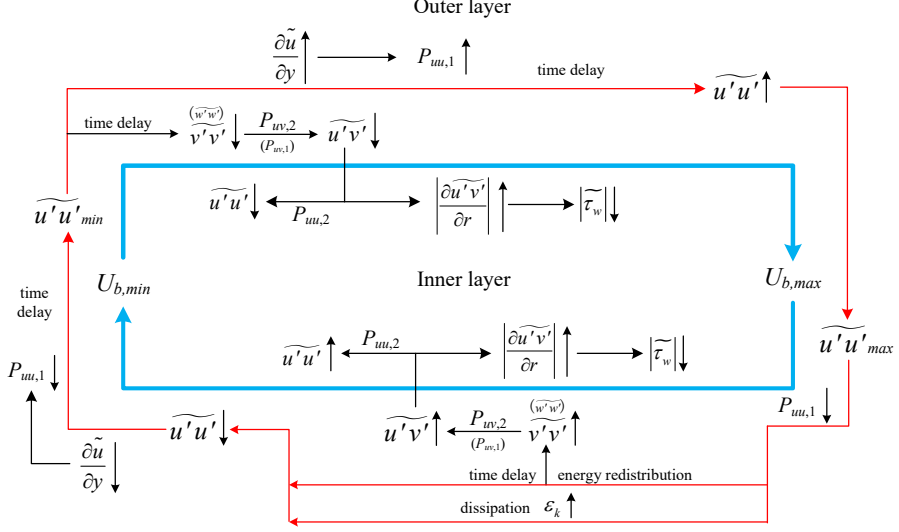


FIGURE 10. Schematic of the flow dynamics leading to the damping of the wall shear stress. The vertical arrows indicate whether the quantities increase or decrease. The vertical lines denote the absolute values.

406 result in a phase lag between $\tilde{u}'v'$ and $-\partial\tilde{u}/\partial r$ (figure 11a), which is $\Delta\phi=0.52\pi$ at $y^+=15.8$
 407 where the overall absolute value of $\tilde{u}'\tilde{u}'$ reaches its maximum. This value is quite close to
 408 that reported in Weng *et al.* (2016) (figure 21 in their paper). Another point is the phase
 409 asymmetry of the turbulence activity. At the initial stage of the acceleration phase, the
 410 level of turbulence intensity is low, hence the whole phase is occupied by the smooth
 411 intensification of turbulent activity. For the deceleration phase, the situation is more
 412 complicated; the energy redistribution process is accompanied by a high-level dissipation
 413 rate that transfers partial TKE into heat. We note that Feldmann & Wagner (2016a) also
 414 reported phase asymmetries in the oscillatory pipe flow (zero mean bulk flow), but their
 415 focus was on the laminar-turbulent transition during the reciprocal cycles of the bulk
 416 flow, therefore different from the present study where the phase asymmetry is reflected
 417 in the evolution of sustained turbulence. Xu *et al.* (2020) also reported strong phase
 418 asymmetry in a pulsatile pipe with a non-zero mean flow. They showed, in the case of
 419 a large pulsation amplitude, that helical flow structures occur in the deceleration phase
 420 and decay in the acceleration phase. In the present study, the pulsation amplitude is
 421 small such that the helical instability cannot be triggered, but the energy transfer from
 422 $\tilde{u}'\tilde{u}'$ to the other two co-existing components bears some similarity with the occurrence
 423 of the helical structure since the latter is characterized by a large circumferential velocity
 424 whose energy probably comes from the streamwise velocity.

425 However, the distinct phase asymmetry in the evolution of turbulence does not result
 426 in a remarkable phase asymmetry in $\tilde{\tau}_w$. As shown before, the fact that $\tilde{u}'\tilde{u}'$ is mainly
 427 dominated by $P_{uu,1}$ suggests that the strength of the main portion of the near-wall streak
 428 directly follows from the magnitude of the shear strain rate. In the deceleration phase,
 429 $\tilde{v}'\tilde{v}'$ drains energy from $\tilde{u}'\tilde{u}'$ and feeds energy into $\tilde{u}'\tilde{v}'$ through $P_{uv,2}$ simultaneously,
 430 yielding a large magnitude of $P_{uu,2}$. Nevertheless, $P_{uu,2}$ only affects $\tilde{u}'\tilde{u}'$ in the very
 431 near-wall region (within the Stokes layer), which means that the main portion of the
 432 near-wall streak is unaffected by the variation of $\tilde{u}'\tilde{v}'$. That is, the energy flow from

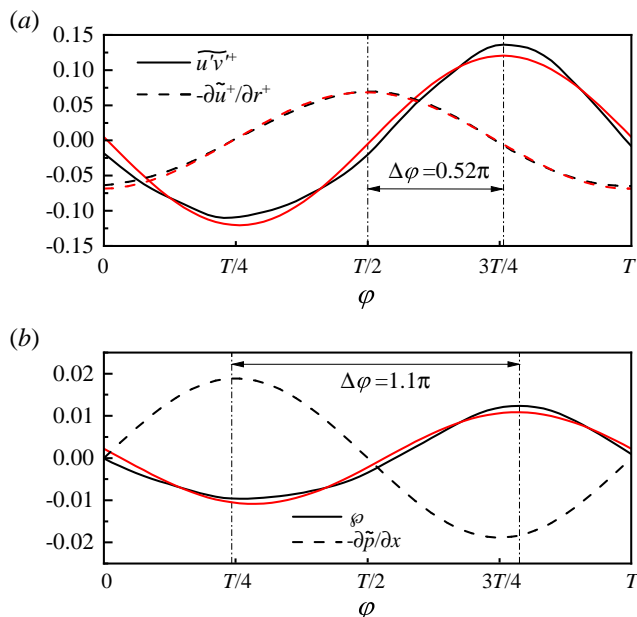


FIGURE 11. (a) Phase-wise variations of the Reynolds shear stress $\widetilde{u'v'}$ and the shear strain rate $-\partial\widetilde{u^+}/\partial r^+$ at $y^+=15.8$ where the overall absolute value of $\widetilde{u'u'}$ reaches its maximum for case 3 ($A=0.1$, $l_s^+=14$). (b) Phase-wise variations of the term φ and the oscillating pressure gradient $-\partial\widetilde{p}/\partial x$ at $y^+=6.5$ where the overall absolute value of term φ reaches its maximum for case 3 ($A=0.1$, $l_s^+=14$). The red solid and dashed lines represent the respective fundamental Fourier modes that are used to calculate the phase difference.

433 $\widetilde{u'u'}$ to $\widetilde{v'v'}$ to $\widetilde{u'v'}$ is somewhat one-way coupled. In the acceleration phase, a negative
 434 $\widetilde{u'u'}$ results in a negative $\widetilde{v'v'}$, $\widetilde{u'v'}$, and $P_{uu,2}$. On the other hand, due to the high-
 435 frequency oscillation, the Reynolds number variation does not influence markedly the
 436 turbulence responding time scale. Therefore, the phase-wise variation of $\widetilde{u'v'}$ follows a
 437 nearly sinusoidally varying manner (figure 11a), and the term φ follows suit. Besides,
 438 the delayed response of turbulence leads to an antiphase variation pattern between the term
 439 φ and the oscillating pressure gradient (figure 11b), indicating the coincidence between
 440 the phase lag between $-\partial\widetilde{u}/\partial r$ and $\widetilde{u'v'}$ and the quarter of pulsation period. The resulting
 441 phase-averaged wall shear stress is equivalent to that in the laminar flow with a reduced
 442 amplitude of the oscillating pressure gradient. Hence, the above-mentioned factors make
 443 the phase-averaged wall shear stress oscillate at a lower amplitude without losing its
 444 phase asymmetry.

445 It is noted that similar discussions regarding the production process of Reynolds
 446 stresses have also been made by Weng *et al.* (2016). Their focus was mainly on the
 447 effects of frequency on the production process and the wall-normal propagation of the
 448 shear wave. Here, the flow dynamics presented in figure 10 are broadly consistent with
 449 them, but we further elaborate the production process focusing on only one case. The
 450 differences are that: we identify the relative importance of the production terms in (4.1)
 451 and their effective region during the cycle; we give an explanation of why the turbulence
 452 damps the oscillating wall shear stress based on the production process; we explain why
 453 the phase asymmetry of the evolution of turbulence does not induce a remarkable phase
 454 asymmetry in the variation of wall shear stress.

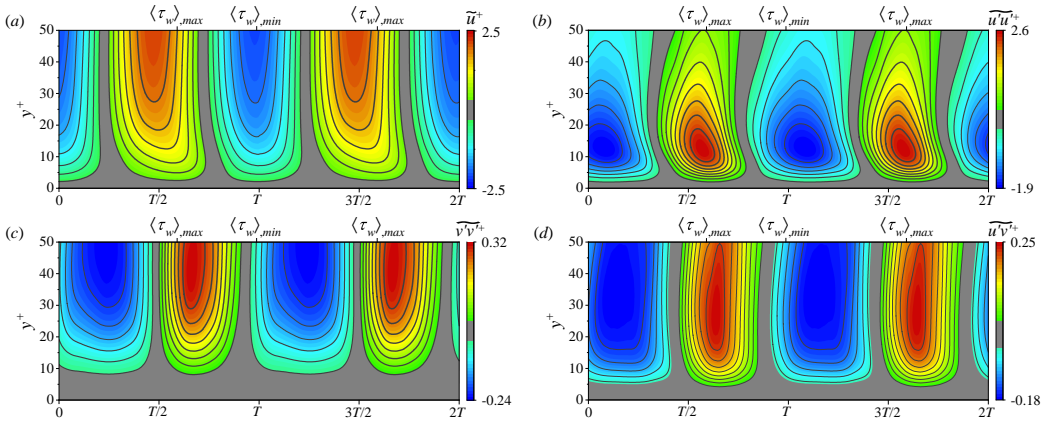


FIGURE 12. Phase-wise variations of (a) the turbulent streamwise velocity \tilde{u}^+ , (b) the Reynolds normal stress $\widetilde{u' u'}$, (c) the Reynolds normal stress $\widetilde{v' v'}$, and (d) the Reynolds shear stress $\widetilde{u' v'}$ for case 4 ($A=0.1$, $l_s^+=17$). The data are duplicated and then spliced such that two periods are presented.

5. Hysteresis phenomenon in the wall shear stress

An interesting issue to be addressed next is the distinct hysteresis in the phase-wise variation of $\widetilde{\tau}_w$ for case 4, where the drag-increasing phase occupies a longer portion of the cycle than the drag-decreasing phase. Figure 12 shows the phase-wise variations of \tilde{u} , $\widetilde{u' u'}$, $\widetilde{v' v'}$ and $\widetilde{u' v'}$ for case 4. Again, the data are duplicated to two periods for a better presentation of the hysteresis. It is observed that the phase asymmetry is only remarkable in the near-wall region (approximately below $y^+=20$) for \tilde{u} , while it shows good symmetry in the outer layer. This is because the outer turbulence intensity is low such that the flow in that region behaves almost in a laminar-like manner. The hysteresis is brought out well by the contours of $\widetilde{u' u'}$, $\widetilde{v' v'}$ and $\widetilde{u' v'}$. They all exhibit similar patterns, that is, the low-magnitude phase of the stress progresses over a longer portion of the cycle than the high-magnitude phase. Again, at $y^+\approx 12$ where the mean streak intensity reaches its maximum, $\widetilde{u' u'}$ lags behind \tilde{u} and $\widetilde{v' v'}$ further lags behind $\widetilde{u' u'}$, while $\widetilde{v' v'}$ is almost in phase with $\widetilde{u' v'}$. This suggests the same evolution process of turbulence as that in case 3 discussed above.

The question addressed next is why the hysteresis occurs in case 4 rather than in case 3 given the same turbulence evolution process. Based on the discussion in section 4, the term \wp and the oscillating pressure gradient are the key factors that determine the evolution of the near-wall streamwise velocity \tilde{u} . Hence, we found the wall-normal location where the overall absolute value of term \wp reaches its maximum and plotted the phase-wise variation of \tilde{u} , $-\partial\tilde{p}/\partial x$, term \wp and the sum of the last two terms at that location in figure 13(a). It can be observed that the phase-wise variation of term \wp is apparently non-sinusoidal. Specifically, the term \wp rises to a positive peak with a larger absolute value than the negative peak; this leads to a sharp increase of the positive sum (black chain dotted-dotted line) of the term \wp and $-\partial\tilde{p}/\partial x$, which counteracts the slow-down effect from the viscous force. Hence, \tilde{u} keeps increasing at a lower rate even though the oscillating pressure gradient has already changed its direction, leading to the short lingering of the high-level wall shear stress and thereby the hysteresis. In the region away from the wall at $y^+=12.2$ (figure 13b), for instance, the magnitude of term \wp is

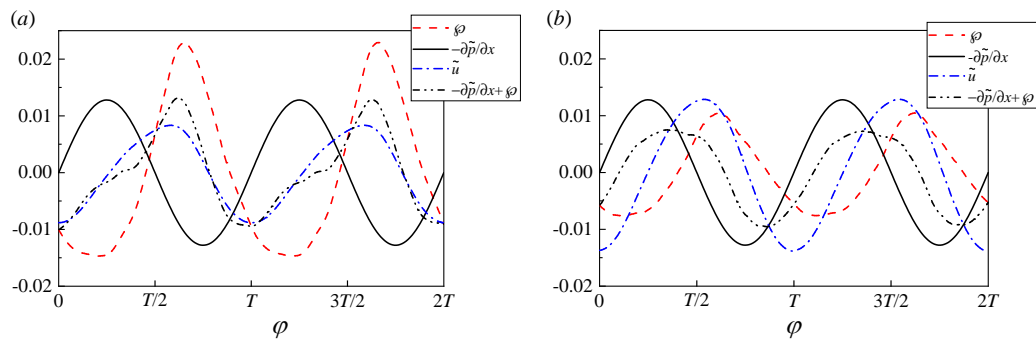


FIGURE 13. Phase-wise variations of the term φ , the oscillating pressure gradient $-\partial\tilde{p}/\partial x$, the streamwise velocity \tilde{u} (divided by a scale factor of 5), and the quantity $-\partial\tilde{p}/\partial x+\varphi$ at (a) $y^+=6.1$ where the overall absolute value of term φ reaches its maximum, and at (b) $y^+=12.2$ where the overall absolute value of $\tilde{u}'u'$ reaches its maximum for case 4 ($A=0.1$, $l_s^+=17$).

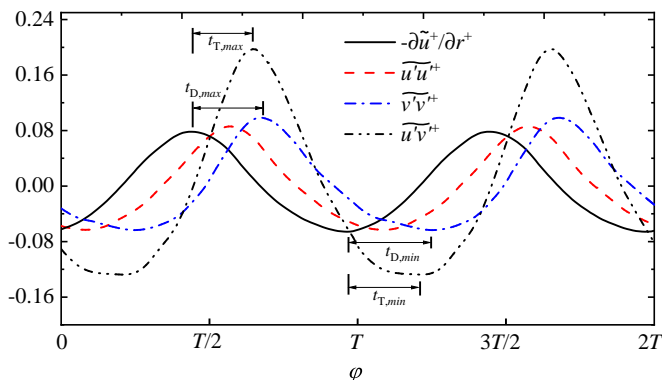


FIGURE 14. Definitions of the quantities associated with the time scale of turbulence response: $t_{T,max}$, $t_{T,min}$, $t_{D,max}$, $t_{D,min}$. The data of the curves are taken from case 4 ($A=0.1$, $l_s^+=17$) at $y^+=12.2$. The values of $\tilde{u}'u'^+$ are divided by a scale factor of 30.

484 smaller such that it cannot effectively influence the evolution of the streamwise velocity,
 485 and therefore the hysteresis is less evident.

486 According to the discussions above, it can be found that the magnitude of term φ and
 487 its phase relation to the oscillating pressure gradient in the vicinity of the wall are the
 488 crucial factors that affect the wall shear stress. Taking the wall-normal location where
 489 the overall absolute value of term φ reaches its maximum as a reference, these two terms
 490 vary in antiphase to each other in case 3 (the phase lag is approximately π (figure 11b));
 491 while in case 4, the phase lag is less than π if we take the positive peak of term φ as the
 492 reference (figure 13a). Besides, the magnitude of term φ in case 4 is larger than that in
 493 case 3. The combination of these facts leads to the significantly different behaviours of
 494 $\tilde{\tau}_w$. The oscillating pressure gradient is predetermined based on the pulsation parameters.
 495 As for term φ , its magnitude and phase near the wall are determined by the outer $\tilde{u}'v'$;
 496 while the outer $\tilde{u}'v'$ is closely associated with $\tilde{v}'v'$ through the production terms, with
 497 the latter deriving from the energy redistribution process. Thus, this is a top-down effect
 498 that reflects the high wall-normal inhomogeneity of the wall-bounded turbulent flow,
 499 which is different from that in the homogeneous turbulence case (Yu & Girimaji (2006)).
 500 Hereafter, the outer region corresponds roughly to the wall-normal location where the

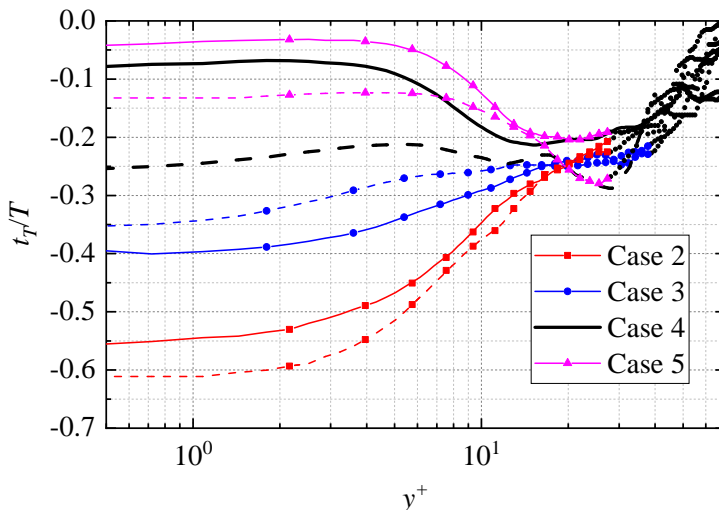


FIGURE 15. Wall-normal profiles of t_T normalized by the pulsation period T . Solid lines: $t_{T,max}$; dashed lines: $t_{T,min}$. Note that $-\partial\tilde{u}/\partial r$ approaches zero close to the centreline and numerical errors occur when determining the t_T . The regions where the magnitude of local maximum $|\partial\tilde{u}/\partial r|$ drops below 10% of the overall maximum are considered to be accompanied by certain numerical errors, which are marked by the black dots.

501 overall absolute value of $\widetilde{u'u'}$ reaches its maximum and the inner region is associated with
 502 the wall-normal location where the overall absolute value of term φ reaches its maximum.

503 In regard to the phase lag between the term φ and the oscillating pressure gradient,
 504 it is natural to focus on the time scale that the turbulence reacts to the varying shear
 505 strain rate, i.e., the time delay between $-\partial\tilde{u}/\partial r$ and $\widetilde{u'v'}$, which we denote as t_T . In the
 506 outer region, $\widetilde{u'v'}$ lags behind $-\partial\tilde{u}/\partial r$ due to the above-mentioned two processes. In the
 507 meantime, as the wall-normal location moves away from the wall, the phase lag between
 508 $-\partial\tilde{u}/\partial r$ and the oscillating pressure gradient gradually approach $\pi/2$ (figure 12a). Thus,
 509 we can use a quarter of the pulsation period ($T/4$) as a benchmark to measure the phase
 510 lag between the term φ and the oscillating pressure gradient. If t_T is equal to $T/4$ in the
 511 outer region, then the term φ would be in antiphase to the oscillating pressure gradient
 512 in the inner region, which is the situation in case 3. To quantify t_T and take into account
 513 the hysteresis, we denote the time delay between the maximum $-\partial\tilde{u}/\partial r$ and maximum
 514 $\widetilde{u'v'}$ as $t_{T,max}$; while for the minimum, it is denoted as $t_{T,min}$ (see figure 14). Figure 15
 515 shows the wall-normal distribution of t_T normalized by the respective pulsation period
 516 (T) for cases 2 to 5. It is first observed that both $t_{T,max}$ and $t_{T,min}$ in the buffer layer
 517 are exactly a quarter of the pulsation period for cases 2 and 3. For cases 4 and 5, $t_{T,max}$
 518 is smaller than $T/4$, in accordance with the aforementioned reduction of the phase lag
 519 between φ and $-\partial\tilde{p}/\partial x$ from π . It is noted that the wall-normal variation tendency of t_T
 520 presented in figure 15 agrees qualitatively well with that reported in Weng *et al.* (2016)
 521 where they showed the wall-normal profiles of the phase lag between $-\widetilde{u'v'}$ and $\partial\tilde{u}/\partial y$. A
 522 remarkable feature is the significantly reduced phase lag below $y^+=10$ for $\omega^+\leq 0.006$ of
 523 their cases. In the present study, both $t_{T,max}$ and $t_{T,min}$ for cases 4 ($\omega^+=0.007$) and 5
 524 ($\omega^+=0.005$) exhibit such a feature. The cause is that the delayed generation of $\widetilde{u'v'}$ in
 525 the outer region produces a large radial gradient of $\widetilde{u'v'}$ in the inner region, promoting

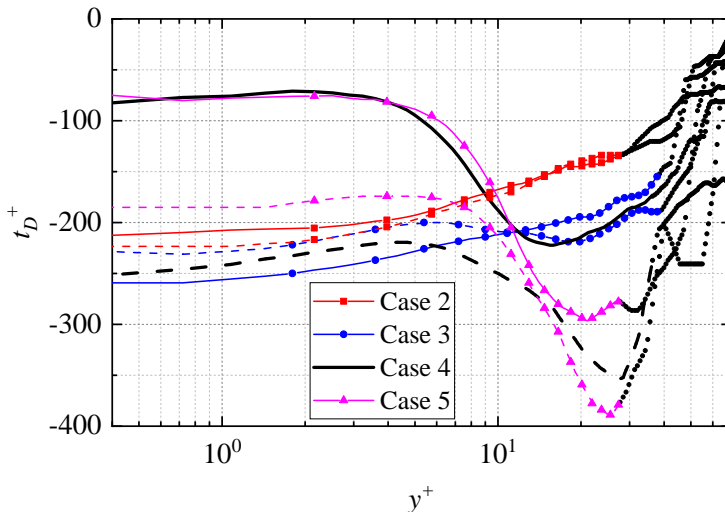


FIGURE 16. Wall-normal profiles of inner-scaled t_D^+ . Solid lines: $t_{D,max}$; dashed lines: $t_{D,min}$. The meaning of the black dots is the same as that in figure 15.

526 the continuous increase of \tilde{u} and then shifting the overall phase of \tilde{u} in the inner region.
 527 Therefore, the phase lag between $-\partial\tilde{u}/\partial r$ and $\tilde{u}'v'$ is reduced, and this also causes the
 528 phase lag of the wall shear stress between the turbulent and laminar cases mentioned in
 529 section 3.

530 Another important information conveyed by figure 15 is the inequality between $t_{T,max}$
 531 and $t_{T,min}$, especially below $y^+=10$. In the model proposed by Weng *et al.* (2016), a
 532 constant turbulent relaxation time is assumed such that the turbulent eddies can be
 533 considered as viscoelastic; this turbulent relaxation time together with the pulsation fre-
 534 quency determines the phase lag between $\tilde{u}'v'$ and $\partial\tilde{u}/\partial y$. Furthermore, they attempted
 535 to improve the turbulence model by considering the wall-normal variation of turbulent
 536 relaxation time. However, it turns out that the improvement in predicting the wall shear
 537 stress is not satisfactory. The reason might be that the turbulent relaxation time varies
 538 not only in space but also in phase. As shown in figure 15, the difference between $t_{T,min}$
 539 and $t_{T,max}$ increases as the wall is approached ($y^+<10$), which means that the assumption
 540 of constant turbulent relaxation is inappropriate; the cause might be attributed to the
 541 significant viscous effect near the wall such that the temporal variation of the friction
 542 Reynolds number leads to the distinct temporal variation of turbulence relaxation time
 543 scale t_T in that region. Besides, such difference is more prominent in low-frequency cases
 544 (cases 4 and 5) where $t_{T,max}$ and $t_{T,min}$ differ significantly not only below $y^+=10$ but
 545 also in the buffer layer. This might explain to a certain extent why considering the wall-
 546 normal variation of turbulent relaxation time can help to improve the prediction of $\tilde{\tau}_w$
 547 for cases with $\omega^+\geq 0.01$ but fails for low-frequency cases (Weng *et al.* (2016)).

548 We further examine the time delay between $-\partial\tilde{u}/\partial r$ and $\tilde{v}'v'$, denoted as t_D , in figure
 549 16 since $\tilde{u}'v'$ leads $\tilde{v}'v'$ by a small phase margin due to $P_{uv,1}$, and t_D can describe a
 550 more complete turbulent reaction process. Again, the time delays between the respective
 551 maximum and minimum values are denoted as $t_{D,max}$ and $t_{D,min}$ (see figure 14). For
 552 cases 2 and 3, $t_{D,max}$ and $t_{D,min}$ differ slightly in the near-wall region, in accordance
 553 with the phase symmetry shown in previous contours. For cases 4 and 5, $t_{D,max}$ and

554 $t_{D,min}$ follow a similar wall-normal distribution pattern, with the overall magnitude of
 555 $t_{D,min}$ being larger than $t_{D,max}$. Besides, in the buffer layer where the near-wall streak
 556 populates, t_D increases as the frequency decreases, indicating that the turbulence reacts
 557 more slowly in low-frequency cases. As suggested by Weng *et al.* (2016), the delayed
 558 response of turbulence is a manifestation of the viscoelasticity of turbulent eddies, then
 559 the above-mentioned features of t_D reflect the variation of the viscoelasticity with respect
 560 to the pulsation frequency. For high-frequency cases (cases 2 and 3), the frequency is large
 561 such that the turbulent eddies exhibit phase-wise-invariant viscoelastic property, that is,
 562 the relative importance between the elasticity and viscosity remains nearly constant in
 563 phase (in Maxwell's viscoelastic model, the ratio between the viscosity and the elasticity
 564 dictates the relaxation time), and this is reflected by the closeness between $t_{D,max}$ and
 565 $t_{D,min}$; while for low-frequency cases (cases 4 and 5), the viscoelasticity varies in phase.
 566 Specifically, for a high-level shear strain rate (positive $-\partial\tilde{u}/\partial r$), the instantaneous friction
 567 Reynolds number is large, hence it can be regarded that the elasticity is enhanced such
 568 that the turbulent responding time $t_{D,max}$ is small; while for the low shear strain rate
 569 phase (negative $-\partial\tilde{u}/\partial r$), the viscosity dominates, leading to a larger $t_{D,min}$ which is
 570 also characteristic of low-Reynolds-number flows. Therefore, this highlights again the
 571 importance of considering the phase-wise variation of turbulent relaxation time when
 572 employing the viscoelastic model to predict the wall shear stress in pulsatile wall-bounded
 573 flows. Moreover, the effects of the varying Reynolds number on the turbulence response
 574 time bears qualitative resemblance with that reported in Xu *et al.* (2017) where the
 575 transition in the pulsatile pipe flow is studied. In the high-frequency regime, the transition
 576 threshold is unaffected due to the too-fast variation of the flow rate, corresponding to
 577 the equivalence between $t_{D,max}$ and $t_{D,min}$ here. When the frequency is reduced, the
 578 Reynolds number effect sets in: for the transition problem, it is reflected by the fact
 579 that the entrance of a low Reynolds number interval significantly elevates the transition
 580 threshold; while in the present study, the turbulence responds quickly in the high-
 581 Reynolds-number interval but slows down in the low-Reynolds-number interval.

582 6. Effects of the pulsation amplitude

583 In this section, the effects of pulsation amplitude are examined. We focus on $l_s^+ = 14$ and
 584 17 with the amplitude being $A = 0.4$, which corresponds to cases 6 and 7, respectively. As
 585 shown in figure 6(a), the wall shear stress decreases at a lower rate when it is close to its
 586 minimum in case 6, leading to a short lingering low-level wall shear stress at that phase.
 587 This behaviour is similar to that in case 4 except for that the lingering stage occurs
 588 when the wall shear stress is close to its maximum and the extent is more distinct.
 589 According to the discussion on case 3, the phase of the low-level wall shear stress is
 590 accompanied by a high-level Reynolds shear stress $\widetilde{u'v'}$ which results from the delayed
 591 turbulence response. It is reasonable to expect a more intense generation of $\widetilde{u'v'}$ for a
 592 larger pulsation amplitude. This is confirmed in figure 17 where the instantaneous spatial-
 593 averaged wall shear stress over all the collected periods for $A = 0.1$ and 0.4 are shown.
 594 As seen, for low-amplitude cases (cases 3 and 4), all the curves are evenly dispersed
 595 (figure 17a,b); while for high-amplitude cases (cases 6 and 7), evident local scattering of
 596 the curves can be observed around the phase of $t = 3T/4$ even for a larger vertical axis
 597 limit (figure 17c,d). Note that the energy redistribution from $\widetilde{u'u'}$ to $\widetilde{v'v'}$ and the rise
 598 of $\widetilde{u'v'}$ occurs at this phase. It should also be noted that apparent scattering can be
 599 still observed for case 4 during the lingering phase (figure 17b). Hence, such a localized
 600 scattering of the wall shear stress curves indicates that the delayed response of turbulence

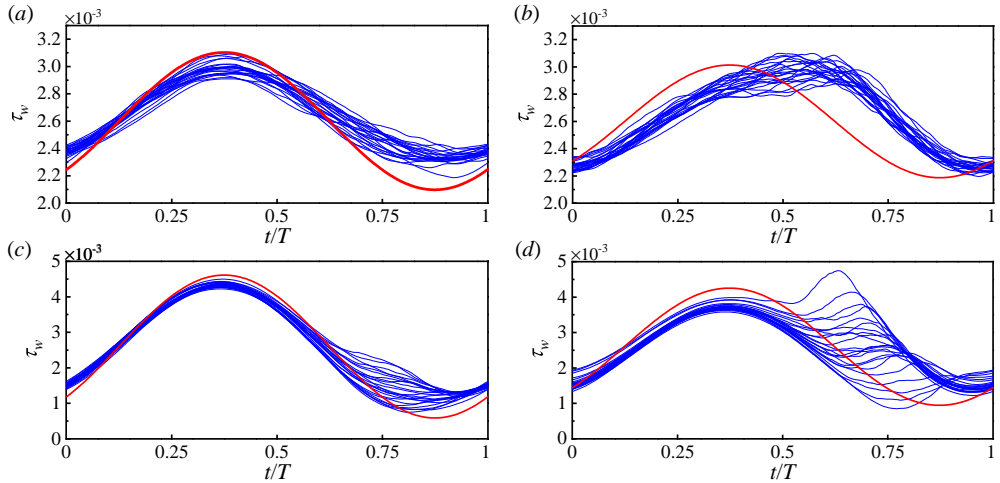


FIGURE 17. Temporal evolution of instantaneous space-averaged wall shear stress τ_w over all the pulsation periods that are used to perform the phase average. The red solid lines represent the corresponding laminar Stokes values. (a) case 3; (b) case 4; (c) case 6; (d) case 7.

601 in the drag-reducing phase is an intense event with a high degree of randomness. The
 602 higher the pulsation amplitude, the more intense the turbulence response. In addition,
 603 the corresponding laminar Stokes values are also included for comparison. Although the
 604 amplitude of the phase-averaged wall shear stress for all the cases shown in figure 17
 605 are lower than their laminar values, the instantaneous value could possibly exceed the
 606 variation range of laminar value for the cases with a relatively low frequency (figure
 607 17*b,d*), especially when the pulsation amplitude is large (figure 17*d*).

608 Figure 18 shows the phase-wise variations of \tilde{u} , $\overline{u'u'}$, $\overline{v'v'}$ and $\overline{u'v'}$ for cases 6 and
 609 7, with the dashed magenta lines marking the edge of the laminar Stokes layer. The
 610 exhibited phase lag indicates again the delayed response of turbulence. A first observation
 611 is that the positive $\overline{u'u'}$ peaks inside the laminar Stokes layer for case 7 (figure 18*d*).
 612 As discussed in section 4, the increase of $\overline{u'u'}$ within the Stokes layer results from the
 613 generated $\overline{u'v'}$ through $P_{uu,2}$. This indicates the significantly large magnitude of $\overline{u'v'}$ such
 614 that it produces higher streamwise velocity fluctuations near the wall. Second, distinct
 615 hysteresis can be observed for case 6. The blue contour occupies a longer portion of the
 616 cycle than that by the red contour, especially for $\overline{v'v'}$ and $\overline{u'v'}$. Note that this hysteresis is
 617 indiscernible in case 3. Therefore, the increase of pulsation amplitude amplifies the phase
 618 asymmetry of turbulence activity, leading to hysteresis in the wall shear stress (figure 6*a*)
 619 that is reversed compared with case 4, i.e., the drag-reducing phase is longer than the
 620 drag-increasing phase. The same goes for case 7, the amplification of the phase asymmetry
 621 is reflected by the more distinct hysteresis conveyed by the contour plots (figure 18*d,f,h*).
 622 Furthermore, in comparison with case 6, the larger magnitude of all turbulence quantities
 623 in case 7 indicates the higher intensity of turbulence activity, which accords well with
 624 the fact that the instantaneous curves of the space-averaged wall shear stress are more
 625 dispersed at that specific phase (figure 17*b,d*).

626 Next, we evaluate the relative importance of the term φ and the oscillating pressure
 627 gradient ($-\partial\tilde{p}/\partial x$) in figure 19 to explore the cause of the behaviour of $\overline{\tau_w}$ for $A=0.4$.
 628 The wall-normal locations are selected where the overall maximum absolute value of
 629 term φ is reached. As seen, the variation of term φ exhibits a distinct phase asymmetry

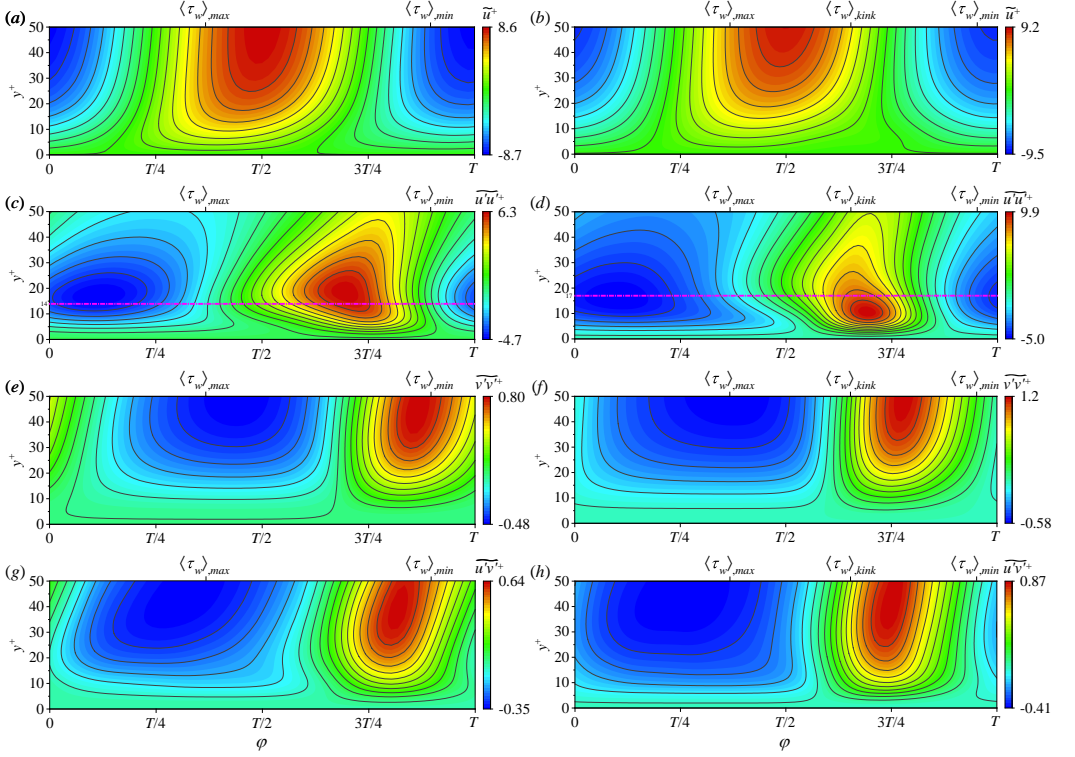


FIGURE 18. Phase-wise variations of the streamwise velocity \tilde{u}^+ , the Reynolds stresses $\widetilde{u'u'^+}$, $\widetilde{v'v'^+}$ and $\widetilde{u'v'^+}$ for $A=0.4$. (a,c,e,g) case 6 ($A=0.4$, $l_s^+=14$); (b,d,f,h) case 7 ($A=0.4$, $l_s^+=17$).

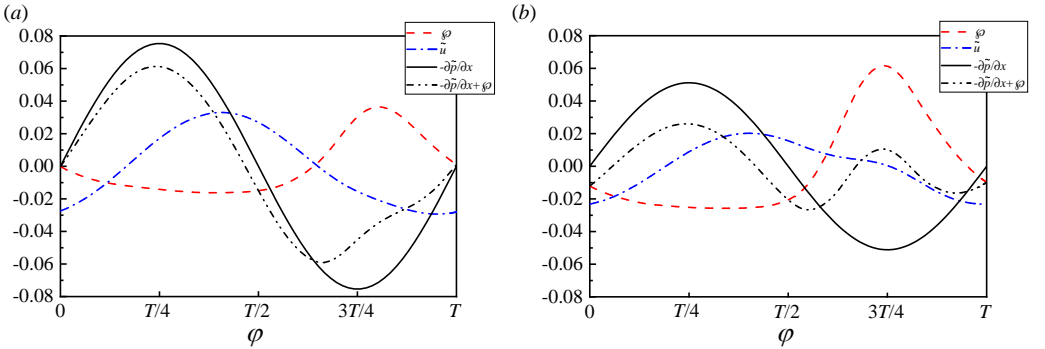


FIGURE 19. Phase-wise variations of the term φ , the oscillating pressure gradient $-\partial\tilde{p}/\partial x$, the streamwise velocity \tilde{u} (divided by a scale factor of 5), and the quantity of $-\partial\tilde{p}/\partial x + \varphi$ at (a) $y^+=6.5$ for case 6 ($A=0.4$, $l_s^+=14$), (b) $y^+=5.8$ for case 7 ($A=0.4$, $l_s^+=17$) where the over absolute value of term φ reaches its maximum.

630 for both cases; it declines very slowly and smoothly during the acceleration phase of \tilde{u}
 631 but increases sharply in the deceleration phase. Importantly, the sharp increase of term
 632 φ leads to a significant change of the sum (black chain-dotted-dotted lines), especially
 633 for case 7 where the sum even bounces back to a positive value at around the phase of
 634 $t=3T/4$, causing the kink of the \tilde{u} curve and hence the wall shear stress; while for case

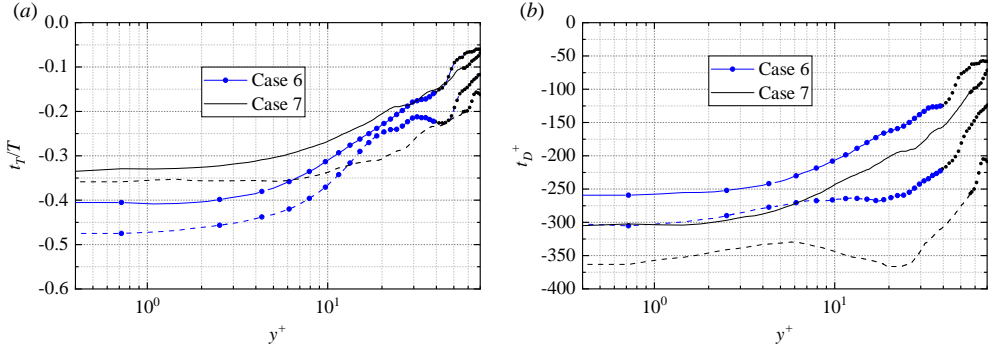


FIGURE 20. (a) Wall-normal profiles of t_T normalized by the pulsation period T . (b) Wall-normal profiles of the inner-scaled t_D . Solid lines: $t_{T,max}$ and $t_{D,max}$; dashed lines: $t_{T,min}$ and $t_{D,min}$. The meaning of the black dots is the same as that in figure 15.

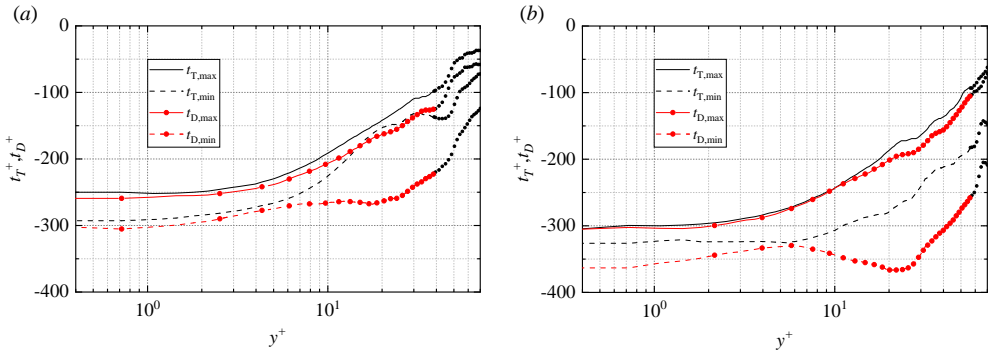


FIGURE 21. Wall-normal profiles of the inner-scaled t_T and t_D . (a) Case 6 ($A=0.4$, $l_s^+=14$); (b) Case 7 ($A=0.4$, $l_s^+=17$). The meaning of the black dots is the same as that in figure 15.

635 6, the \tilde{u} descends at a lower rate instead. This discrepancy results from the fact that the
 636 value of the positive peak of term φ in case 7 is larger than that in case 6, while the
 637 amplitude of the oscillating pressure gradient is smaller in the former case (according to
 638 (2.5), β is proportional to the frequency ω for fixed A). The large pulsation amplitude
 639 tends to enhance the phase asymmetry and produces a large magnitude of term φ . In the
 640 meantime, the amplitude of the oscillating pressure gradient also varies with respect to
 641 the pulsation amplitude. Hence, it can be inferred that there would be diverse forms of
 642 the phase-wise variation of $\tilde{\tau}_w$ for different combinations of pulsation parameters, which
 643 requires more cases to summarize the general law and is therefore beyond the scope of the
 644 present paper. It also highlights the complexity of pulsatile flows which possess multiple
 645 predefined parameters.

646 We further examine the quantities t_T and t_D in figure 20. A first observation is that
 647 the significantly reduced time scale below $y^+ \approx 10$ in case 4 disappears for case 7. This is
 648 easily understood since the outer delayed variation of $u'v'$ does not induce a remarkable
 649 phase shift of the inner \tilde{u} (figure 18b). The general trend is that these two quantities both
 650 gradually increase as the wall is approached; this can be attributed to the strong viscous
 651 effect near the wall that shifts the phase of \tilde{u} forward, which can be clearly conveyed by
 652 the distorted laminar contour in figure 7(a) as an example, and thereby enlarging the time
 653 delay between the shear strain rate and the Reynolds stresses in that region. Compared

654 to $A=0.1$, t_T does not change significantly in the buffer layer; both $t_{T,max}$ and $t_{T,min}$ are
655 nearly a quarter of the period in case 6, and $t_{T,min}$ is larger than $t_{T,max}$ in case 7. For
656 t_D , a remarkable observation is the enlarged difference between $t_{D,max}$ and $t_{D,min}$ in the
657 buffer layer for both cases. This accords well with the hysteresis exhibited in figure 18.
658 However, there are obvious differences between t_T and t_D . For case 6, $t_{T,min}$ and $t_{T,max}$
659 are nearly the same in the buffer layer, while $t_{D,min}$ and $t_{D,max}$ differ significantly in
660 that region. Similarly, the difference between $t_{D,min}$ and $t_{D,max}$ is also apparently larger
661 than t_T for case 7. Note that such discrepancy between t_T and t_D is not evident for
662 cases with $A=0.1$. To explore the physical meaning, we compare the magnitude of t_T
663 and t_D in figure 21. Generally, t_T is smaller than t_D since $\widetilde{u'v'}$ leads $\widetilde{v'v'}$ by a small phase
664 margin due to the shear-strain-oscillated production term $P_{uv,1}$. It can be observed in
665 figure 21 that the values of $t_{T,max}$ and $t_{D,max}$ are close while $t_{T,min}$ and $t_{D,min}$ differ
666 significantly for both cases. This discrepancy highlights the importance of the shear-
667 strain-oscillated production term $P_{uv,1}$, that is, the variation of $\widetilde{u'v'}$ is governed both by
668 the shear strain rate and $\widetilde{v'v'}$. In the deceleration phase, the generation of $\widetilde{v'v'}$ is intense
669 such that the production process of $\widetilde{u'v'}$ is dominated by $\widetilde{v'v'}$, thus the variation of $\widetilde{u'v'}$
670 synchronizes with $\widetilde{v'v'}$ and the values of $t_{T,max}$ and $t_{D,max}$ are very close; while in the
671 acceleration phase, the magnitude of $\widetilde{v'v'}$ is relatively small, then the effect of the shear-
672 strain-oscillated production term $P_{uv,1}$ becomes prominent, leading to a large phase lag
673 between $\widetilde{u'v'}$ and $\widetilde{v'v'}$ and also the large difference between $t_{T,min}$ and $t_{D,min}$. For the
674 small amplitude of $A=0.1$, the magnitude of the varying shear strain rate is small such
675 that the variation of $\widetilde{u'v'}$ is mainly dominated by $\widetilde{v'v'}$ throughout the cycle, hence the
676 difference between t_T and t_D is small.

677 7. Energy spectra

678 We finally examine the streamwise and circumferential one-dimensional spectra to
679 provide some information about the variation of scales of turbulence structures during
680 the pulsating process, as shown in figure 22 and 23. These spectra data are collected
681 at the wall-normal locations where $\widetilde{u'u'}$ reaches its maximum. Taking case 4 as an
682 example (figure 22c), the energy spectrum of large wavelengths (roughly $\lambda_x^+ > 1000$)
683 increases during the acceleration phase of \widetilde{u} (approximately $0 \sim T/2$, see figure 12a),
684 accompanied by an initially mild decrease and the subsequent increase of the energy of
685 small wavelengths ($\lambda_x^+ < 1000$); in the deceleration phase ($T/2 \sim T$), the increasing energy
686 spectrum of small wavelengths reaches its maximum quickly and then decreases, while
687 the energy of large wavelengths keep decreasing. Notably, the phase of the peak energy
688 spectrum of small wavelengths coincides with that of the peak of $\widetilde{v'v'}$. Hence, a rough
689 scenario can be depicted: the increase of the bulk flow yields a large magnitude of the
690 near-wall shear strain, stretching the near-wall streaks in the streamwise direction and
691 suppressing the turbulence motions that are of small scales; when the bulk flow starts
692 to decrease, the existing streamwise-stretched long streaks break up into many small-
693 scale structures, accompanied by the energy redistribution from $\widetilde{u'u'}$ to the other two
694 components and high intensity of dissipation; subsequently, the overall TKE drops to a
695 low level and the flow state goes back to the beginning and repeats. This scenario is more
696 clear in large-amplitude cases, as shown in figure 22(e)(f), where the stretching and the
697 breaking up processes are separated distinctly. For low-amplitude cases, the increase of
698 the energy spectrum of small wavelengths becomes less evident as the frequency increases,
699 it almost disappears for case 2 (figure 22a); besides, the streamwise stretching is also
700 weak. Such changes result from two factors: one is that the pulsation is too fast for the

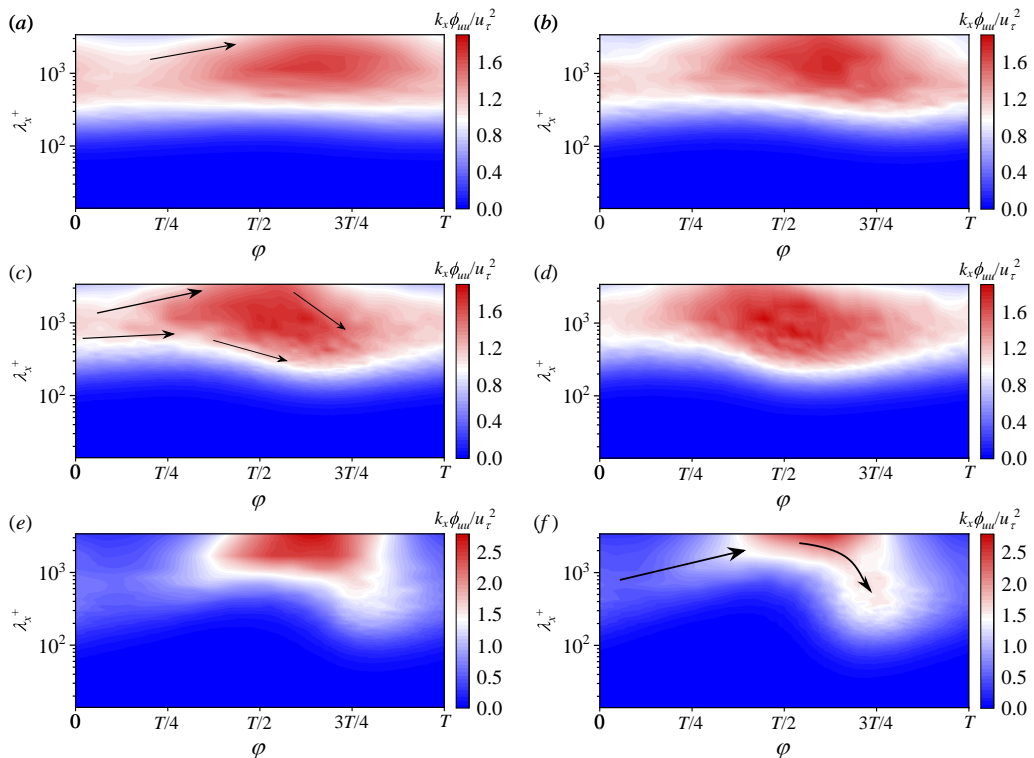


FIGURE 22. Normalized streamwise pre-multiplied spectra $k_x \phi_{uu}$ of streamwise velocity fluctuations, where $\lambda_x = 2\pi/k_x$ is the wavelength. The contour levels are set to be the same for the fixed amplitude. The wall-normal location, where the spectra data are taken from, is at $y^+ = 14.7$ for (a) Case 2, (b) Case 3, (c) Case 4, (d) Case 5, and at $y^+ = 16.9$ for (e) Case 6, (f) Case 7. The black arrows indicate qualitatively the variation trend of the energy spectrum.

701 turbulence structures to respond, and the other is the low thickness of the Stokes layer in
 702 high-frequency cases such that only a small portion of the near-wall streaks are affected
 703 by the varying shear strain rate. Nevertheless, there is still an evident increase of the peak
 704 energy spectrum of $\lambda_x^+ \approx 1000$, which is the commonly-accepted averaged length scale in
 705 a steady flow, outside the laminar Stokes layer. This can be attributed to the turbulent
 706 diffusion that diffuses "upward ejected" or "downward sweeping" fluids with higher wall-
 707 normal velocity due to the bottom Stokes layer away or toward the wall, leading to a
 708 larger deficit or excess of the streamwise velocity outside the Stokes layer and hence the
 709 higher energy spectrum.

710 For circumferential spectra, all the contours are centered around $\lambda_\theta^+ = 100$ which is
 711 the averaged circumferential length scale in the steady pipe. This means that the
 712 pulsation does not change the dominated circumferential length scale. Nevertheless,
 713 the streamwise stretching of the streaks is accompanied by a slight enlarging of their
 714 circumferential space, and the subsequent breaking up leads to the increase of energy of
 715 small circumferential wavelengths. It is noted that the results reported above resemble
 716 closely those in He & Seddighi (2013) where the turbulence in a channel with a step-
 717 increase of the bulk flow was investigated, both of which are characterized by the initial
 718 stretching of near-wall streaks and the subsequent breaking up into small-scale structures.
 719 However, the breaking up of the elongated streaks in He & Seddighi (2013) occurs in a
 720 circumstance that the bulk flow has already settled down, thus it can be considered

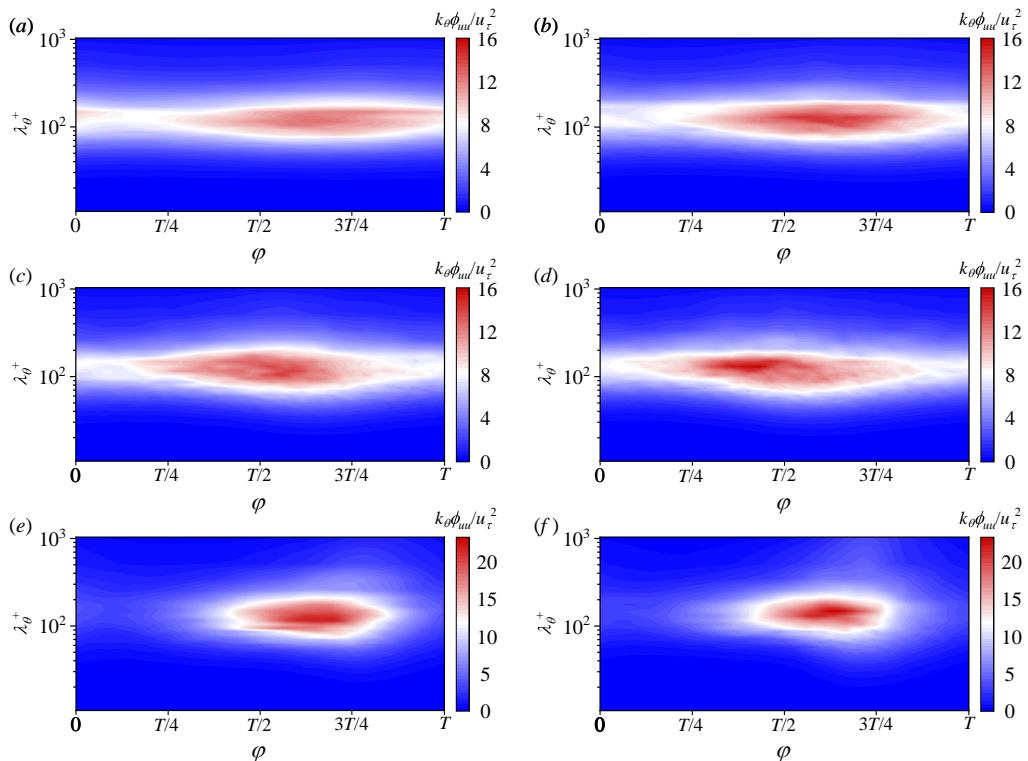


FIGURE 23. Normalized circumferential pre-multiplied spectra $k_\theta \phi_{uu}$ of streamwise velocity fluctuations, where $\lambda_\theta = 2\pi/k_\theta$ is the wavelength. The contour levels are set to be the same for the fixed amplitude. The wall-normal locations, where the spectra data are collected, are the same as those in figure 22. (a) Case 2; (b) Case 3; (c) Case 4; (d) Case 5; (e) Case 6; (f) Case 7.

721 as a spontaneous and gradual event because the turbulence structures have to change
 722 its scales to accommodate to the new larger Reynolds number; while in the pulsatile
 723 flow, the elongated streaks break up in the deceleration phase of the bulk flow, that
 724 is, the decelerating bulk flow cannot accommodate the existing high-energy turbulence
 725 structures such that they are forced to break up, hence it is a non-spontaneous, externally-
 726 forced and transient process that is different from that in He & Seddighi (2013).

727 8. Summary

728 The phase-wise variations of the wall shear stress in turbulent pulsatile pipe flow with
 729 low-amplitude oscillations at $Re_\tau=180$ have been investigated using DNS. We focus on
 730 the paradoxical phenomenon reported in previous studies, that is, the amplitude of the
 731 oscillating wall shear stress in the turbulent flow is smaller than that in the laminar flow
 732 for the same pressure gradient in the intermediate frequency range. This implies that the
 733 turbulence reduces the wall shear stress. It is shown that the phase-wise variation of the
 734 wall shear stress exhibits a strong dependence on the frequency at the fixed amplitude of
 735 $A=0.1$. For high-frequency cases, the wall shear stress synchronizes with its corresponding
 736 laminar Stokes value, displaying an evident phase symmetry. As the frequency is reduced,
 737 a distinct hysteresis occurs, i.e., the time occupied by the drag-reducing phase differs
 738 significantly from that by the drag-increasing phase, accompanied by a phase shift with

739 respect to its laminar Stokes value. This hysteresis would disappear if the frequency
740 further reduces to reach a quasi-steady state.

741 The cause of the paradoxical phenomenon can be attributed to the delayed response
742 of the turbulence. Specifically, the delayed generation of the Reynolds shear stress $\overline{u'v'}$
743 in the buffer layer gives rise to a large magnitude of the radial gradient of $\overline{u'v'}$ near the
744 wall whose contribution to the oscillating streamwise velocity \tilde{u} is opposite to that of the
745 oscillating pressure gradient, thus damping the variation of \tilde{u} near the wall and leading
746 to a lower amplitude of the wall shear stress. This is a top-down effect that reflects the
747 wall-normal inhomogeneity of wall-bounded turbulent flows. The delayed generation of
748 $\overline{u'v'}$ derives from two processes: the delayed development of near-wall streaks and the
749 subsequent energy redistribution from streamwise velocity fluctuations to the other two
750 coexisting components. This is an interpretation of the viscoelasticity of turbulent eddies
751 from the perspective of flow dynamics.

752 The hysteresis in the variation of the wall shear stress that occurs in low-frequency
753 cases is caused by the phase asymmetry of turbulence response. In the deceleration phase
754 of the bulk flow, the turbulence response is intense such that a large magnitude of $\overline{u'v'}$ is
755 generated in the buffer layer, yielding a large magnitude of the radial derivative of $\overline{u'v'}$
756 near the wall that is comparable to the oscillating pressure gradient and thus deviating
757 the variation of \tilde{u} and wall shear stress from the sinusoidally varying manner; while in
758 the acceleration phase, the turbulence response is mild, thereby the variation of near-wall
759 streamwise velocity is dominated by the oscillating pressure gradient and hence follows a
760 sinusoidal manner. Such a phase asymmetry causes the hysteresis of the wall shear stress
761 and also a phase shift from its laminar Stokes value. The intensity of turbulence response
762 and the magnitude of the oscillating pressure gradient are both closely related to the
763 pulsation parameters. Thus, there would be diverse forms of phase-wise variations of the
764 wall shear stress given the different combinations of pulsation amplitude and frequency,
765 highlighting the complexity of pulsatile flows. Further, a quantitative examination of the
766 turbulence responding time scale reveals that the viscoelastic model proposed by Weng
767 *et al.* (2016) should not only consider the wall-normal variation of turbulent relaxation
768 time but also take into account its phase-wise variation to acquire a better performance.

769 For larger amplitude cases, the phase asymmetry of the turbulence response is amplified
770 due to the larger variation range of the Reynolds number. The flow evolution can be
771 clearly separated into two stages. In the acceleration phase of bulk flow, the near-wall
772 streaks are stretched in the streamwise direction, accompanied by the suppression of
773 small-scale turbulent motions. When the bulk flow starts to decrease, the existing long
774 streaks break up into small-scale structures, together with a high dissipation rate that
775 transforms the turbulent kinetic energy into heat. This process is of a high degree of
776 randomness that leads to a more intense fluctuation of the instantaneous wall shear stress.
777 Moreover, the importance of the shear-strain-oscillated production term of $\overline{u'v'}$ increases
778 for large-amplitude cases, reflected by the enlarged phase lag between the minimum $\overline{u'v'}$
779 and $\overline{v'v'}$ compared with that in low-amplitude cases.

780 Declaration of interests

781 The authors report no conflict of interest.

782 Acknowledgment

783 The authors gratefully acknowledge the financial support by the National Natural
784 Science Foundation of China (nos. 12372270, 52122110, 52101322, 52371285) and the

785 Program for Intergovernmental International S&T Cooperation Projects of Shanghai
 786 Municipality (no. 22160710200). The numerical calculations in this study were partially
 787 carried out on the ORISE Supercomputer.

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