

## Preparation and toughening mechanism of Al<sub>2</sub>O<sub>3</sub> composite ceramics toughened by B<sub>4</sub>C@TiB<sub>2</sub> core–shell units

Yingjie Shi<sup>a</sup>, Weixing Li<sup>a</sup>, Xiaorong Zhang<sup>a,b,\*</sup>, Jiachao Jin<sup>a</sup>, Jilin Wang<sup>c</sup>, Yu Dong<sup>d</sup>,  
Jingbo Mu<sup>a</sup>, Guangsuo Wang<sup>a</sup>, Xiaoliang Zhang<sup>a</sup>, Zhixiao Zhang<sup>a,\*</sup>

<sup>a</sup>Key Laboratory of New Inorganic Nonmetallic Composites of Handan, College of Materials Science and Engineering, Hebei University of Engineering, Handan 056038, China

<sup>b</sup>Key Laboratory of Intelligent Industrial Equipment Technology of Hebei Province, College of Mechanical and Equipment Engineering, Hebei University of Engineering, Handan 056038, China

<sup>c</sup>College of Materials Science and Engineering, Guilin University of Technology, Guilin 532100, China

<sup>d</sup>School of Civil and Mechanical Engineering, Curtin University, Perth 6845, Australia

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**Abstract:** In this paper, the concept of incorporating core–shell structured units as secondary phases to toughen Al<sub>2</sub>O<sub>3</sub> ceramics is proposed. Al<sub>2</sub>O<sub>3</sub> composite ceramics toughened by B<sub>4</sub>C@TiB<sub>2</sub> core–shell units are successfully synthesized using a combination of molten salt methodology and spark plasma sintering. The synthesis of B<sub>4</sub>C@TiB<sub>2</sub> core–shell toughening units stems from the prior production of core–shell structural B<sub>4</sub>C@TiB<sub>2</sub> powders, and this core–shell structure is effectively preserved within the Al<sub>2</sub>O<sub>3</sub> matrix after sintering. The B<sub>4</sub>C@TiB<sub>2</sub> core–shell toughening unit consists of a micron-sized B<sub>4</sub>C core enclosed by a shell approximately 500 nm in thickness, composed of numerous nanosized TiB<sub>2</sub> grains. The regions surrounding these core–shell units exhibit distinct geometric structures and encompass multidimensional variations in phase composition, grain dimensions, and thermal expansion coefficients. Consequently, intricate stress distributions emerge, fostering the propagation of cracks in multiple dimensions. This behavior consumes a considerable amount of crack propagation energy, thereby enhancing the fracture toughness of the Al<sub>2</sub>O<sub>3</sub> matrix. The resulting Al<sub>2</sub>O<sub>3</sub> composite ceramics display relative density of 99.7%±0.2%, Vickers hardness of 21.5±0.8 GPa, and fracture toughness 6.92±0.22 MPa·m<sup>1/2</sup>.

**Keywords:** Al<sub>2</sub>O<sub>3</sub> composite ceramics; microstructure design; core–shell structure; toughening mechanism; spark plasma sintering (SPS)

\* Corresponding authors.

E-mail: X. Zhang, [xrzhang611@163.com](mailto:xrzhang611@163.com);

Z. Zhang, [zhixiao351@hebeu.edu.cn](mailto:zhixiao351@hebeu.edu.cn)

## 1 Introduction

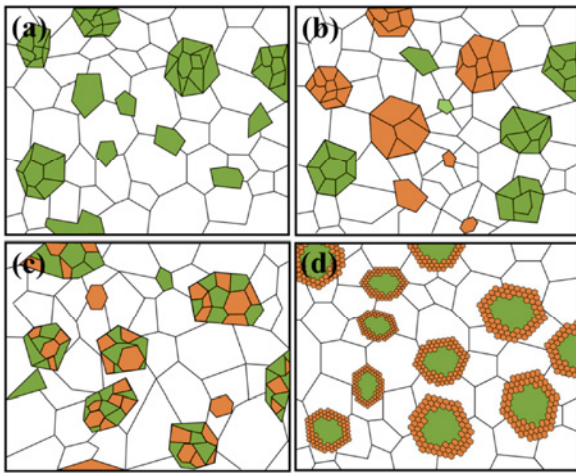
Alumina ( $\text{Al}_2\text{O}_3$ ), widely used advanced ceramics, has applications in armor, aerospace, biomedicine, machinery, and chemicals due to its favorable properties, such as cost effectiveness, exceptional thermal resistance, chemical inertness, biocompatibility, high hardness, and wear resistance [1–6]. However, its inherent low fracture toughness hampers its overall mechanical performance and restricts its broader utility. Numerous endeavors have aimed to enhance the fracture toughness of  $\text{Al}_2\text{O}_3$  ceramics, including techniques, such as refining grain size through specialized powder preparation methods and advanced sintering technologies [7–9], and incorporating diverse secondary phases, such as  $\text{TiB}_2$ ,  $\text{B}_4\text{C}$ ,  $\text{SiC}$ ,  $\text{ZrO}_2$ ,  $\text{BN}$ , and  $\text{C}$  with the forms of particles, whiskers, fibers, tubes, or plate-like materials [10–16]. Despite varying degrees of success attained by these toughening methods, the improvements are often marginal, accompanied by elevated production costs or challenges in large-scale manufacturing. Therefore, the continued exploration of novel methodologies for cost-effective, high-toughness  $\text{Al}_2\text{O}_3$  ceramic production is imperative.

The integration of high-hardness ceramic particles, such as  $\text{TiB}_2$ ,  $\text{B}_4\text{C}$ , and  $\text{SiC}$ , as secondary phases within the  $\text{Al}_2\text{O}_3$  matrix, holds promise because these additives enhance fracture toughness while preserving the hardness, strength, and elastic modulus of the  $\text{Al}_2\text{O}_3$  matrix. The commonly used preparation methods of these  $\text{Al}_2\text{O}_3$  composite ceramics are directly mixing  $\text{Al}_2\text{O}_3$  and secondary phase powders, and then hot pressing sintering or spark plasma sintering (SPS). For example, Parchoviansky *et al.* [17] prepared  $\text{Al}_2\text{O}_3$ – $\text{SiC}$  composite ceramics by hot pressing at 1740 °C using  $\text{Al}_2\text{O}_3$  and  $\text{SiC}$  powders as raw materials. The fracture toughness of the  $\text{Al}_2\text{O}_3$ – $\text{SiC}$  composites containing different volume fractions of  $\text{SiC}$  is distributed in the range from 5.0 to 6.1  $\text{MPa}\cdot\text{m}^{1/2}$ . Deng *et al.* [18] prepared  $\text{Al}_2\text{O}_3$ – $\text{TiB}_2$  composite ceramics by mixing  $\text{Al}_2\text{O}_3$  and  $\text{TiB}_2$  powders, and subsequent hot pressing sintering at 1650–1800 °C. The optimal fracture toughness, 5.2  $\text{MPa}\cdot\text{m}^{1/2}$ , is realized when the  $\text{TiB}_2$  content is 30 vol%. Currently, despite the preparation of  $\text{Al}_2\text{O}_3$  composite ceramics toughened by various ceramic particles, the toughening effects of these ceramic particles have yet to yield substantial breakthroughs; the achieved fracture toughness for  $\text{Al}_2\text{O}_3$  composite ceramics range from

3.8 to 6.2  $\text{MPa}\cdot\text{m}^{1/2}$  over years [10–19]. The primary reason for this limitation lies in the prevailing toughening structure shared by various  $\text{Al}_2\text{O}_3$  composite ceramics toughened, wherein the secondary phases are independently dispersed within the  $\text{Al}_2\text{O}_3$  matrix, as shown in Fig. 1(a). Despite variations in the types of secondary phases and sintering methods employed, the underlying toughening structure remains constant, leading to marginal differences in fracture toughness. The reported three-phase  $\text{Al}_2\text{O}_3$  composite ceramics contain two types of secondary phases [20–23], but the different secondary phases are segregated from each other and independently dispersed within the  $\text{Al}_2\text{O}_3$  matrix, as shown in Fig. 1(b). Consequently, the fundamental toughening structure in Figs. 1(a) and 1(b) remains analogous, limiting the enhancement of fracture toughness compared with two-phase  $\text{Al}_2\text{O}_3$  composites. Therefore, the development of a novel toughening structure is essential to improve the fracture toughness of  $\text{Al}_2\text{O}_3$  composite ceramics further.

In our previous studies [24,25],  $\text{B}_4\text{C}$  composite ceramics featuring  $\text{TiB}_2$ – $\text{SiC}$  agglomerates as a secondary phase were synthesized, as shown in Fig. 1(c). Unlike conventional setups where  $\text{TiB}_2$  and  $\text{SiC}$  remain isolated and individually dispersed, the secondary phase in these composites comprises interlocked  $\text{TiB}_2$  and  $\text{SiC}$  entities. The existence of such  $\text{TiB}_2$ – $\text{SiC}$  agglomerates remarkably elevates the toughness of  $\text{B}_4\text{C}$ – $\text{TiB}_2$ – $\text{SiC}$  composites from 4.6 to 6.5  $\text{MPa}\cdot\text{m}^{1/2}$ . The methodology used for preparation results in randomly formed  $\text{TiB}_2$ – $\text{SiC}$  agglomerates within the  $\text{B}_4\text{C}$  matrix, rendering their size, structure, and content uncontrollable. However, these findings indicate that composite structural units as secondary phases can achieve better toughening effect compared with isolated, independently dispersed secondary phases because the multi-interface collaborative toughening behavior of the composite structural units can induce the multiple deflection of cracks, thus consuming a large amount of crack propagation energy.

Drawing inspiration from this work, adopting a toughening structure using composite structural units as secondary phases offers an effective approach to overcome the current toughening bottleneck faced by  $\text{Al}_2\text{O}_3$  composite ceramics. The  $\text{Al}_2\text{O}_3$  composite ceramics with composite structural units as secondary phase can be prepared by sintering using the mixture of  $\text{Al}_2\text{O}_3$  powders and the powders possessing composite structure as raw materials. In this method, the controlled



**Fig. 1** Schematic diagram of microstructures of Al<sub>2</sub>O<sub>3</sub> composite ceramics toughened by (a) single secondary phase, (b) two isolated and individually dispersed secondary phases, (c) agglomerates of two types of secondary phases, and (d) core-shell composite structural units.

advance preparation of the powders possessing the composite structure is key. Currently, the technology of preparing core-shell structural composite powders by the molten salt method has been widely reported with success [26,27]. In terms of the phase composition for the composite powders, B<sub>4</sub>C and TiB<sub>2</sub> have ultrahigh hardness, relatively low density, and chemical inertness to Al<sub>2</sub>O<sub>3</sub>, and are ideal toughening phases for the Al<sub>2</sub>O<sub>3</sub> matrix [28,29]. Therefore, core-shell structural powders consisting of B<sub>4</sub>C and TiB<sub>2</sub> is a suitable candidate that can be mixed with Al<sub>2</sub>O<sub>3</sub> powders to prepare the mixture for subsequent sintering.

In this work, the innovative notion of adopting core-shell composite structural units as secondary phases to enhance the toughness of the Al<sub>2</sub>O<sub>3</sub> composite ceramics is introduced, as shown in Fig. 1(d). This breakthrough is achieved by crafting B<sub>4</sub>C@TiB<sub>2</sub> core-shell units encapsulated within an Al<sub>2</sub>O<sub>3</sub> matrix, employing a combined approach of molten salt

methodology and SPS. First, core-shell B<sub>4</sub>C@TiB<sub>2</sub> powders are synthesized using B<sub>4</sub>C and TiH<sub>2</sub> as precursors through the molten salt technique. Subsequently, the Al<sub>2</sub>O<sub>3</sub> composite ceramics are fabricated by SPS using Al<sub>2</sub>O<sub>3</sub> powders and synthesized B<sub>4</sub>C@TiB<sub>2</sub> powders as starting materials. The microstructure and synthesis mechanism of the B<sub>4</sub>C@TiB<sub>2</sub> powders are analyzed. The microstructure, compositions, and interfaces of the B<sub>4</sub>C@TiB<sub>2</sub> core-shell units inside the Al<sub>2</sub>O<sub>3</sub> matrix are studied. The mechanical properties and toughening mechanism of the Al<sub>2</sub>O<sub>3</sub> composite ceramics toughened by B<sub>4</sub>C@TiB<sub>2</sub> core-shell units are also discussed.

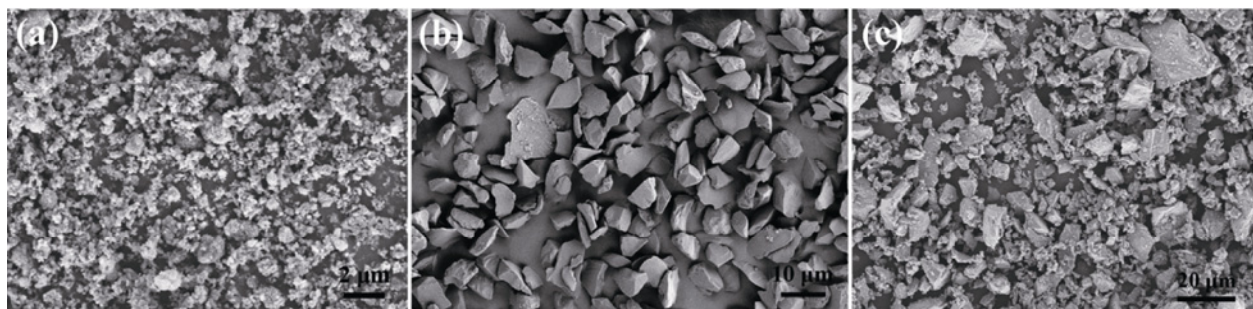
## 2 Materials and methods

### 2.1 Materials

Al<sub>2</sub>O<sub>3</sub> powders with purity of  $\geq 99.99\%$  and average particle size of 0.2  $\mu\text{m}$  (Shanghai Aladdin Biochemical Technology Co., Ltd., China), B<sub>4</sub>C powders with purity of  $\geq 97\%$  and average particle size of 5.0  $\mu\text{m}$  (Jingangzuan Boron Carbide Co., Ltd., China), and TiH<sub>2</sub> powders with purity of  $\geq 99\%$  and particle size of  $\leq 45 \mu\text{m}$  (Shanghai Aladdin Biochemical Technology Co., Ltd., China) were adopted as raw materials. NaCl and KCl powders with purity of  $\geq 99\%$  (Shanghai Aladdin Biochemical Technology Co., Ltd., China) were adopted as molten salt medium. The characteristics of these raw materials are shown in Fig. 2.

### 2.2 Preparation of core-shell structured B<sub>4</sub>C@TiB<sub>2</sub> powders

A mixture of B<sub>4</sub>C, TiH<sub>2</sub>, NaCl, and KCl powders at mass ratios of 7 : 6 for B<sub>4</sub>C and TiH<sub>2</sub>, 1 : 1 for NaCl and KCl, and 1 : 4 for (B<sub>4</sub>C+TiH<sub>2</sub>) and (NaCl+KCl) was created. This blend was subjected to magnetic stirring in ethanol solution for 24 h, followed by drying



**Fig. 2** SEM images of raw materials: (a) Al<sub>2</sub>O<sub>3</sub> powders, (b) B<sub>4</sub>C powders, and (c) TiH<sub>2</sub> powders.



using rotary evaporation and vacuum oven techniques, yielding a powdered mixture. This mixture was loaded into a corundum crucible equipped with a cap and subjected to calcination within a tube furnace under a flowing argon atmosphere. The process involved heating at a rate of 2 °C/min to a calcination temperature of 1400 °C and dwelling time of 2 h. Following cooling to the ambient temperature, the calcined mixture underwent a triple washing sequence using water, hydrochloric acid, and water, and is subsequently dried in a vacuum oven for 24 h, resulting in the prepared B<sub>4</sub>C@TiB<sub>2</sub> powders.

### 2.3 Preparation of Al<sub>2</sub>O<sub>3</sub>–B<sub>4</sub>C@TiB<sub>2</sub> composite ceramics

The Al<sub>2</sub>O<sub>3</sub> composite ceramics containing 10 wt% B<sub>4</sub>C@TiB<sub>2</sub> core–shell units were designed. First, Al<sub>2</sub>O<sub>3</sub> and synthesized B<sub>4</sub>C@TiB<sub>2</sub> powders, at a mass ratio of 9 : 1, were mixed in ethanol via magnetic stirring for 24 h. After drying and passing through a 200-mesh sieve, a powder mixture of Al<sub>2</sub>O<sub>3</sub> and B<sub>4</sub>C@TiB<sub>2</sub> was obtained. This mixture was loaded into a cylindrical graphite die with an inner diameter of 30 mm. The interior surfaces of the die and the punches were covered with a thin graphite foil, and the exterior of the die was encased in an 8 mm thick graphite felt. Sintering was conducted within a spark plasma sintering furnace (HPD 60/0, FCT Systeme GmbH, Germany). The samples were rapidly heated to 600 °C within 2 min and then 100 °C/min up to 1600 °C, dwelling for 5 min. An uniaxial pressure of 30 MPa was applied at 600 °C, maintained through 800 °C, and sustained throughout sintering. Then, the samples were cooled, and the pressure was released by turning off the power.

### 2.4 Characterizations

The phase components of powders and ceramics were characterized using an X-ray diffractometer (XRD; X'Pert PRO, PANalytical, the Netherlands) with Cu K $\alpha$  radiation (40 kV, 40 mA). The microstructures of powders and ceramics were observed using a scanning electron microscope (SEM; S4800, Hitachi, Japan) coupled with an energy-dispersive X-ray analyzer system (EDS). Ceramic composite density was determined using Archimedes method, and the relative density is calculated by dividing the actual density by the theoretical density. Vickers hardness was evaluated with 10 × 9.8 N load applied for 15 s to the polished surface. Fracture toughness was calculated from the

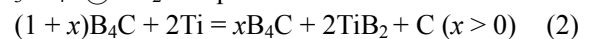
length of the cracks of Vickers indentations according to Eq. (1) [30].

$$K_{IC} = \chi \left( \frac{E}{H_V} \right)^{1/2} \frac{P}{c^{3/2}} \quad (1)$$

where  $E$  is the Young's modulus,  $H_V$  is the Vickers hardness,  $c$  is the average of all radial crack length measured from the center of the indent,  $P$  is the applied load, and  $\chi$  is an empirical calibration constant, which is 0.016±0.004. The reported density, fracture toughness, and Vickers hardness values represent averages from 5, 7, and 3 measurements, respectively.

## 3 Results and discussion

A schematic of the formation processes for the Al<sub>2</sub>O<sub>3</sub> composite ceramics toughened by B<sub>4</sub>C@TiB<sub>2</sub> core–shell units is shown in Fig. 3. The initial step involves uniform mixing of B<sub>4</sub>C, TiH<sub>2</sub>, NaCl, and KCl powders, which undergo heating. As the temperature increases, TiH<sub>2</sub> breaks down into nanosized Ti powders and H<sub>2</sub> at approximately 430 °C [31], and subsequently, NaCl and KCl powders liquefy at approximately 658 °C [32]. The nanosized Ti and micron-sized B<sub>4</sub>C particles are rearranged, uniformly dispersing within the liquid medium. Upon reaching the reaction temperature of B<sub>4</sub>C and Ti, the nanosized Ti particles commence reacting on the surface of micron-sized B<sub>4</sub>C particles, engendering the formation of a TiB<sub>2</sub> shell along with a minor quantity of C based on Reaction (2). With extended reaction duration, the thickness of the TiB<sub>2</sub> shell progressively increases until the total consumption of Ti. The surplus B<sub>4</sub>C in the core of B<sub>4</sub>C particles remains unreacted due to the excess B<sub>4</sub>C powder in the raw materials, culminating in the formation of the core–shell structured B<sub>4</sub>C@TiB<sub>2</sub> powders. Subsequently, the synthesized B<sub>4</sub>C@TiB<sub>2</sub> powders are mixed with the Al<sub>2</sub>O<sub>3</sub> powders, generating the initial sintering powders. Finally, SPS is performed to fabricate the targeted Al<sub>2</sub>O<sub>3</sub>–B<sub>4</sub>C@TiB<sub>2</sub> composite ceramics.



### 3.1 B<sub>4</sub>C@TiB<sub>2</sub> powder characteristics

The progressive development of the B<sub>4</sub>C@TiB<sub>2</sub> powders is depicted in Fig. 3. Micron-sized B<sub>4</sub>C powders are deliberately employed in excess to ensure B<sub>4</sub>C within the core of the B<sub>4</sub>C particles remains intact, and a controlled scarcity of TiH<sub>2</sub>, which can decompose

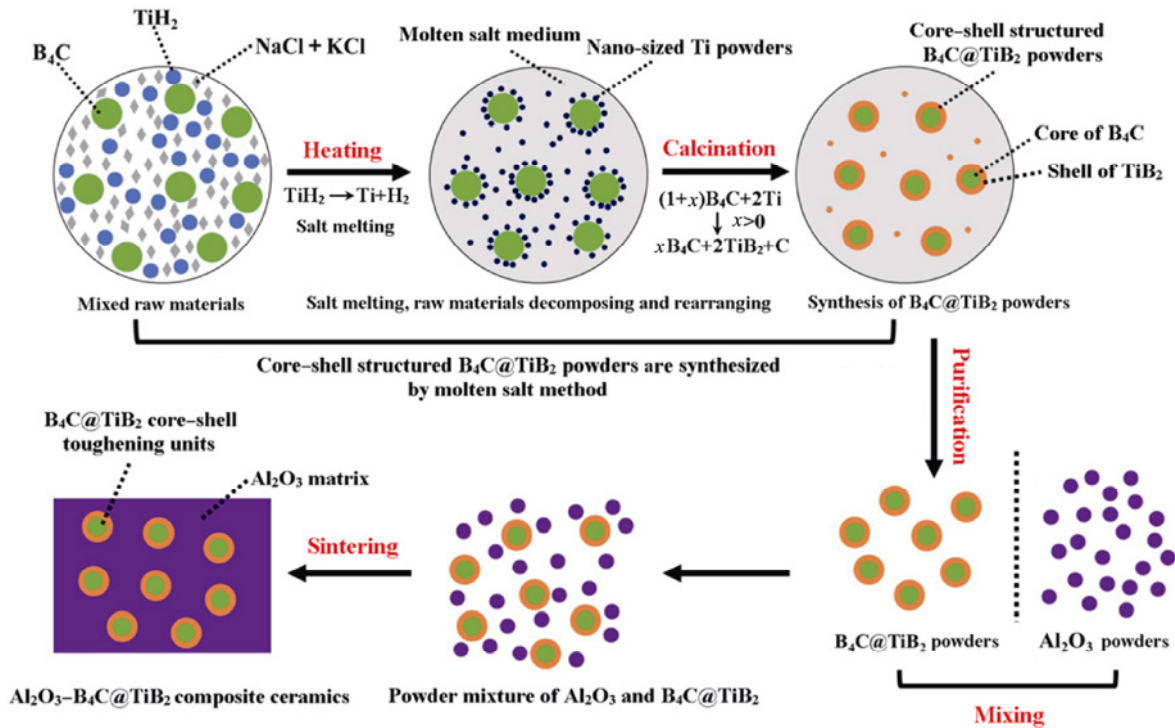


Fig. 3 Schematic illustration for formation processes of Al<sub>2</sub>O<sub>3</sub> composite ceramics toughened by B<sub>4</sub>C@TiB<sub>2</sub> core-shell units.

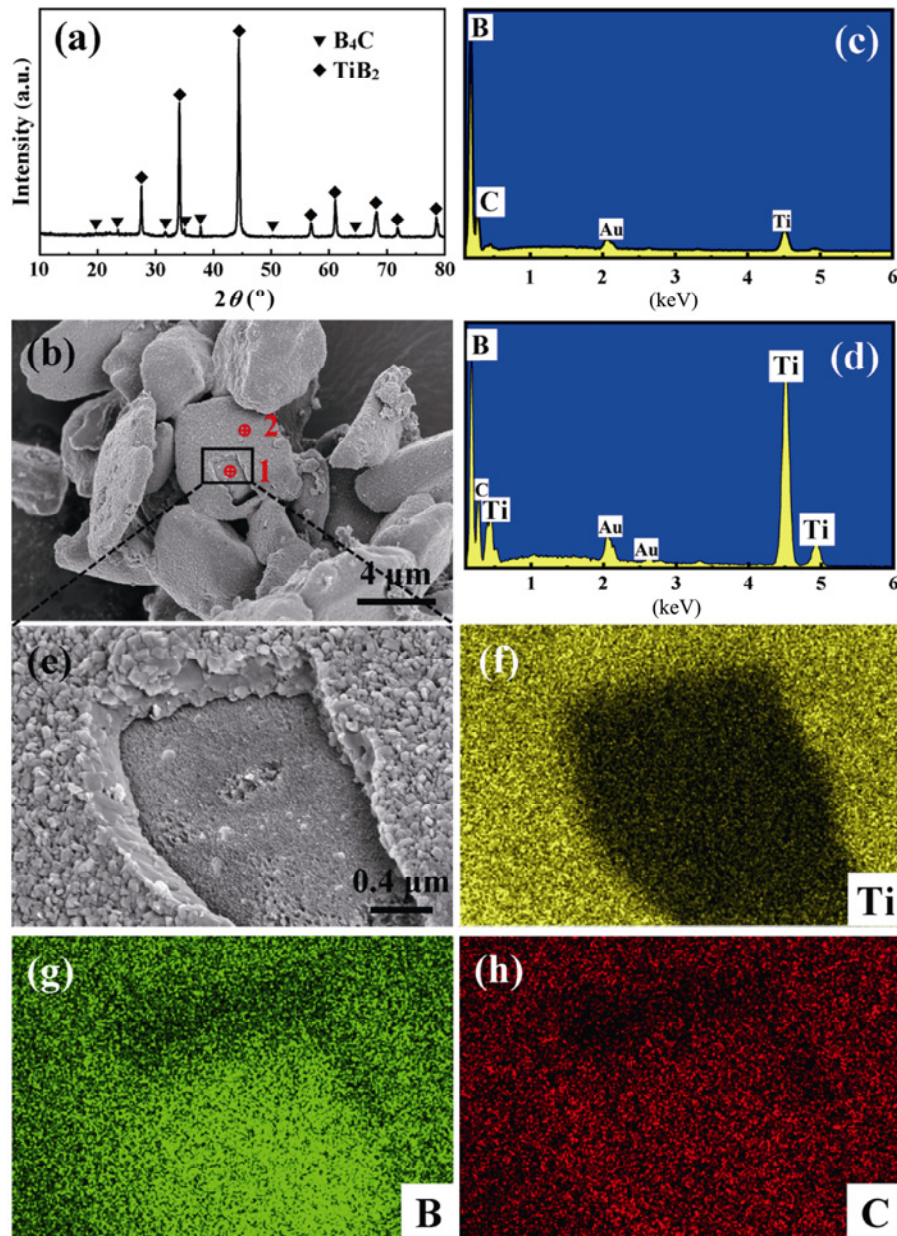
into nanosized Ti powders, is adopted to ensure nanosized Ti particles react on the surface of the micron-sized B<sub>4</sub>C particles, ultimately forming the TiB<sub>2</sub> shell. The thickness of the TiB<sub>2</sub> shell can be fine-tuned by adjusting the ratio of B<sub>4</sub>C to TiH<sub>2</sub>.

Figure 4(a) shows the XRD pattern of the B<sub>4</sub>C@TiB<sub>2</sub> powders. Only peaks corresponding to B<sub>4</sub>C and TiB<sub>2</sub> are detected, indicating the composite powders' B<sub>4</sub>C and TiB<sub>2</sub> composition. The absence of a peak for C is attributed to its negligible quantity and low atomic weight [26]. The microstructure of the B<sub>4</sub>C@TiB<sub>2</sub> powders, as indicated in Fig. 4(b), illustrates the surface of B<sub>4</sub>C@TiB<sub>2</sub> particles becomes textured. However, their shape and size distribution remain largely unchanged compared with the original B<sub>4</sub>C particle powders. Importantly, fractured particles revealing a partial shell detachment, as highlighted in the rectangular area of Fig. 4(b), reaffirm the presence of a distinct core-shell configuration. EDS analysis in Figs. 4(c) and 4(d) indicates point 1, selected on the core, primarily comprises B and C, and point 2, chosen from the shell, mainly consists of Ti and B. This analysis, coupled with the XRD outcomes, confirms the core as B<sub>4</sub>C and the shell as TiB<sub>2</sub>. The rectangular region of Fig. 4(b) is magnified in Fig. 4(e) to gain deeper insight into B<sub>4</sub>C@TiB<sub>2</sub> particle's microstructure. The

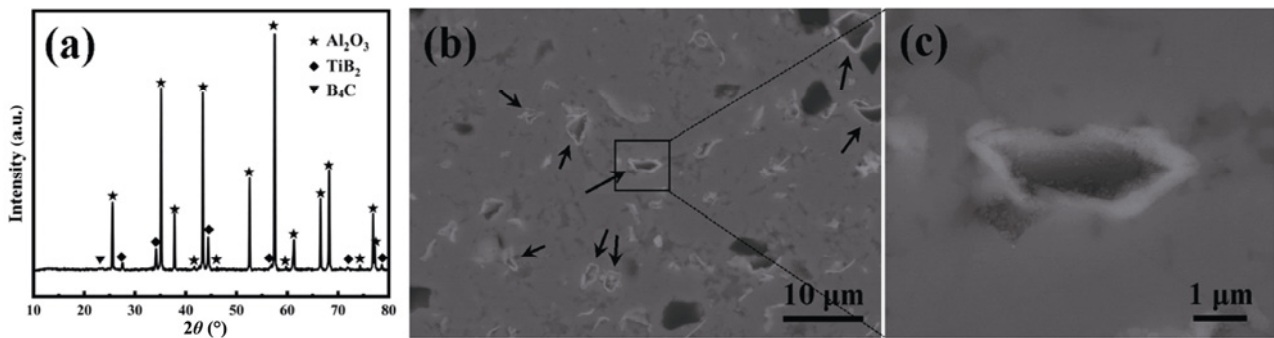
core's surface displays rugged features, marked by gullies, distinctly differing from the smooth surface of raw B<sub>4</sub>C powders. This alteration substantiates the *in situ* reaction between B<sub>4</sub>C and Ti on the particle's surface. The shell encompasses numerous nanosized TiB<sub>2</sub> grains spanning 10–100 nm, with an approximate thickness of 500 nm. The EDS element mapping images in Figs. 4(f)–4(h) for this region further validate the core as B<sub>4</sub>C and the shell as TiB<sub>2</sub>. These analyses affirm the successful synthesis of core-shell structured B<sub>4</sub>C@TiB<sub>2</sub> powders, with the core composed of micron-sized B<sub>4</sub>C particles and the shell comprising numerous nanoscale TiB<sub>2</sub> particles.

### 3. 2 Microstructure and mechanical properties of Al<sub>2</sub>O<sub>3</sub>-B<sub>4</sub>C@TiB<sub>2</sub> composite ceramics

The phase constituents of the Al<sub>2</sub>O<sub>3</sub>-B<sub>4</sub>C@TiB<sub>2</sub> composite ceramics are characterized through XRD patterns, as shown in Fig. 5(a). The XRD analysis demonstrates the presence of Al<sub>2</sub>O<sub>3</sub>, B<sub>4</sub>C, and TiB<sub>2</sub>, affirming the retention of B<sub>4</sub>C@TiB<sub>2</sub> powder components within the Al<sub>2</sub>O<sub>3</sub> matrix after sintering. No new phases emerge, indicating the preservation of initial components during sintering. The backscattered electron (BSE) image of the polished surface of the composite ceramics in Fig. 5(b) delineates three distinct phase



**Fig. 4** (a) XRD pattern and (b) SEM image of  $B_4C@TiB_2$  powders. (c, d) EDS analysis for points 1 and 2 in (b), respectively. (e) Magnification for rectangular in (b). (f–h) EDS element mapping images for (e).



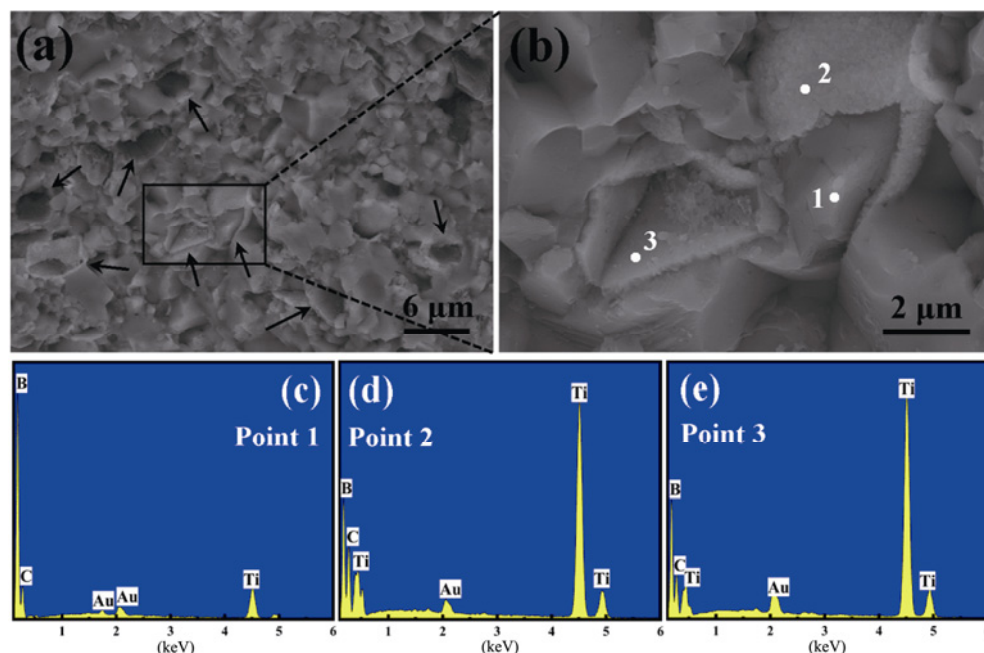
**Fig. 5** (a) XRD pattern and (b) BSE image of polished surface of  $Al_2O_3-B_4C@TiB_2$  composite ceramics. (c) Magnification for rectangular in (b).



components. The glossy white regions correspond to  $\text{TiB}_2$ , the dark gray regions indicate  $\text{B}_4\text{C}$ , and the remaining gray regions denote  $\text{Al}_2\text{O}_3$ . The surface hardly exhibits pores, demonstrating the ceramics' high density. Intact  $\text{B}_4\text{C}@ \text{TiB}_2$  core-shell units are discernible (indicated by arrows). An amplified view of a typical  $\text{B}_4\text{C}@ \text{TiB}_2$  core-shell unit in Fig. 5(c) offers further insight into the composite structure units. Within this unit, the  $\text{B}_4\text{C}$  is completely encapsulated by the  $\text{TiB}_2$ , and the  $\text{TiB}_2$  shell is approximately 500 nm in thickness, consistent with the shell thickness in the  $\text{B}_4\text{C}@ \text{TiB}_2$  powders in Fig. 4(e). Interfaces between  $\text{B}_4\text{C}-\text{TiB}_2$  and  $\text{TiB}_2-\text{Al}_2\text{O}_3$  exhibit strong interfacial compatibility. The analysis indicates the preservation of the  $\text{B}_4\text{C}@ \text{TiB}_2$  core-shell structure within the  $\text{Al}_2\text{O}_3$  matrix after sintering, thereby accomplishing the targeted  $\text{Al}_2\text{O}_3$  composite ceramics toughened by  $\text{B}_4\text{C}@ \text{TiB}_2$  core-shell units. Independently dispersed  $\text{TiB}_2$  and  $\text{B}_4\text{C}$  phases also exist within the  $\text{Al}_2\text{O}_3$  matrix due to the disruption of the core-shell structure of  $\text{B}_4\text{C}@ \text{TiB}_2$  powders prior to sintering. Mechanical impacts during the powder handling of removal from the corundum crucible, water washing, sieving, and mechanical mixing with  $\text{Al}_2\text{O}_3$  powders may have damaged the core-shell structure. This outcome is corroborated by Figs. 4(b) and 4(e), verifying the structure's disruption.  $\text{TiB}_2$  detaches from  $\text{B}_4\text{C}@ \text{TiB}_2$  powders, and the remaining  $\text{B}_4\text{C}$  disperses within the

starting sintering powders. This result leads to the formation of independently dispersed  $\text{TiB}_2$  and  $\text{B}_4\text{C}$  within the  $\text{Al}_2\text{O}_3$  matrix after sintering. Although the diffusion of  $\text{TiB}_2$  during sintering can also contribute to this phenomenon, the sintering temperature of 1600 °C is relatively low for  $\text{B}_4\text{C}$  and  $\text{TiB}_2$ , whose typical sintering temperature range is 1900–2200 °C [33,34]. The fast heating rate and short sintering time adopted in this paper minimize diffusion effects [35–38]. Therefore, the diffusion is not the main reason for core-shell structure disruption.

Figure 6(a) presents the SEM image of the composite ceramics' fracture surface. The fracture surface displays a rugged topography characterized by undulations and depressions. Numerous micron-sized bulged grains and sunken cavities are encompassed by a covering layer, as marked by arrows. A representative region is magnified, as shown in Fig. 6(b), which shows a bulging micron-sized grain (marked by point 1) enclosed by a shell approximately 500 nm in thickness, composed of numerous nanosized grains (marked by point 2). A shell without a core is also observed (marked by point 3), which is the relic that the core is pulled out. EDS analysis conducted at point 1 confirms the micron-sized grain is  $\text{B}_4\text{C}$ , and the results from points 2 and 3 verify the shell is  $\text{TiB}_2$ . The above observations show the typical core-shell structure of the toughening units and their fracture paths during the



**Fig. 6** (a) SEM image of fracture surface of  $\text{Al}_2\text{O}_3-\text{B}_4\text{C}@ \text{TiB}_2$  composite ceramics. (b) Magnification for rectangular in (a). (c–e) EDS analysis for points 1, 2, and 3 in (b), respectively.

fracture of the composite ceramics. In the core–shell toughening units, the core consists of micron-sized  $B_4C$  grains, and the shell comprises numerous nanosized  $TiB_2$  grains, a composition identical to the microstructure of the  $B_4C@TiB_2$  powders. The size of the  $TiB_2$  grains in the toughening units experiences minimal growth compared with those in the  $B_4C@TiB_2$  powders because the grain growth temperature for  $TiB_2$  is approximately 1700 °C [39], higher than the sintering temperature of 1600 °C. Multidimensional fracture modes are evident within this localized region. Intergranular fracture emerges along the interface of the micron-sized  $Al_2O_3$  grain and the outer side of the  $TiB_2$  shell layer, and the micron-sized  $B_4C$  grain and the inner side of the  $TiB_2$  shell layer. In addition, the  $TiB_2$  shell layer itself fractures, forming a generated cross section that bridges the interfaces on both sides of the shell layer. Furthermore, within the interior of the shell layer, intergranular fracture occurs amidst the nanosized  $TiB_2$  grains, consistent with the fracture mode observed in  $TiB_2$  ceramics [40]. This amalgamation of multidimensional fracture modes contributes to the enhancement of composite ceramics' fracture toughness.

The relative density, Vickers hardness, and fracture toughness of the  $Al_2O_3-B_4C@TiB_2$  composite ceramics, along with other referenced  $Al_2O_3$  composite ceramics [10,11,16,17,21], are listed in Table 1. The relative density of the  $Al_2O_3-B_4C@TiB_2$  composite ceramics is  $99.7\% \pm 0.2\%$ , indicating approximately complete density. The result agrees with the SEM image in Fig. 5(b). The high relative density is pivotal to ensuring excellent mechanical properties. The Vickers hardness of the ceramics achieved in this paper is  $21.5 \pm 0.8$  GPa, surpassing that of other  $Al_2O_3$  composite ceramics. This heightened hardness in the  $Al_2O_3-B_4C@TiB_2$  composite ceramics arises from two key factors: the high relative density of the composite ceramics itself, which is the basic guarantee for high

hardness, and the ultrahigh inherent hardness of  $B_4C$  and  $TiB_2$  (29–41 GPa [36,41,42] and 25–35 GPa [39,40,43], respectively), which can elevate the hardness of the  $Al_2O_3$  matrix. Notably, the fracture toughness of the  $Al_2O_3-B_4C@TiB_2$  composite ceramics reaches an impressive  $6.92 \pm 0.22$   $MPa \cdot m^{1/2}$ , outperforming other  $Al_2O_3$  composite ceramics. This advancement can be attributed to the formation of  $B_4C@TiB_2$  core–shell toughening units, which induce a multidimensional fracture mode within the composite ceramics. This mode effectively absorbs substantial crack propagation energy compared with the approach of independently dispersed toughening phases.

### 3. 3 Toughening mechanism of $Al_2O_3-B_4C@TiB_2$ composite ceramics

Within the vicinity of the core–shell toughening units, micron-sized  $Al_2O_3$  matrix grains, micron-sized  $B_4C$  grains, and nanosized  $TiB_2$  grains coexist. Simultaneously, the micron-sized  $B_4C$  core is completely enclosed by a shell layer consisting of nanosized  $TiB_2$  grains. The thermal expansion coefficients of  $Al_2O_3$  ( $7.2 \times 10^{-6} K^{-1}$  [44]),  $B_4C$  ( $4.5 \times 10^{-6} K^{-1}$  [24]), and  $TiB_2$  ( $8.1 \times 10^{-6} K^{-1}$  [24,45]) differ remarkably. This localized region features a distinctive geometric configuration that amalgamates multidimensional variations in phase types, grain size, and thermal expansion coefficients. Consequently, an intricate stress distribution is established around the core–shell units. This complexity underscores the potential influence of  $B_4C@TiB_2$  core–shell units on the toughening mechanism when introduced as a composite toughening phase within the  $Al_2O_3$  composite ceramics. This mechanism may deviate from the approach observed in  $Al_2O_3$  composite ceramics, which are toughened by isolated, independently distributed secondary phases.

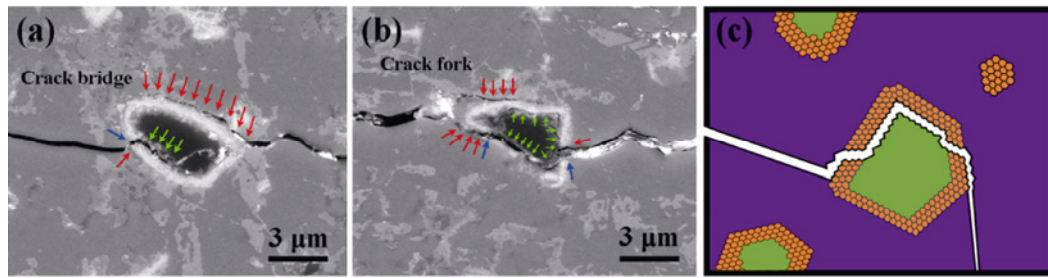
The propagation of cracks within the  $Al_2O_3-B_4C@TiB_2$  composite ceramics is presented in Figs. 7(a) and 7(b), highlighting the BSE images of polished

**Table 1 Relative density and mechanical properties of  $Al_2O_3$  composite ceramics**

Sample	Relative density (%)	Vickers hardness (GPa)	Fracture toughness ( $MPa \cdot m^{1/2}$ )	Preparation method	Ref.
$Al_2O_3-B_4C@TiB_2$	$99.7 \pm 0.2$	$21.5 \pm 0.8$	$6.92 \pm 0.22$ (IND)	SPS	This work
$Al_2O_3-TiC$	—	17.1–19.4	4.98–6.14 (IND)	Hot pressing	[10]
$Al_2O_3-TiB_2$	—	18.5–24.1	4.27–5.40 (CN)	SPS	[11]
$Al_2O_3-Ti_3SiC_2$	96.0–98.0	18.6–22.6	5.1–6.2 (IND)	Hot pressing	[16]
$Al_2O_3-SiC$	98.7–99.5	18.4–20.3	5.0–6.1 (IND)	Hot pressing	[17]
$Al_2O_3-TiB_2$	98.5–99.9	19.6–21.3	3.7–5.2 (IND)	Hot pressing	[18]
$Al_2O_3-TiB_2-TiSi_2$	—	~16.9	~4.7 (IND)	Hot pressing	[21]

Note: IND refers to indentation method; CN refers to Chevron notch.





**Fig. 7** (a, b) BSE images of polished surface with cracks of  $\text{Al}_2\text{O}_3\text{-B}_4\text{C@TiB}_2$  composite ceramics. (c) Schematic diagram for crack propagation in regions surrounding core-shell units.

surfaces bearing cracks. The modes of crack propagation can be analyzed from two perspectives. First, the  $\text{B}_4\text{C@TiB}_2$  core-shell unit is regarded as a unified entity. When cracks traverse these units, they generate crack bridges and forks around them, which is similar to the crack propagation pattern observed in  $\text{Al}_2\text{O}_3$  ceramics toughened by independently distributed  $\text{B}_4\text{C}$  or  $\text{TiB}_2$  phases [4,11]. However, further analysis of crack propagation within the interior of these core-shell units is warranted upon considering the  $\text{B}_4\text{C@TiB}_2$  core-shell units as composite structures. As the crack propagates to the external interface of the core-shell units, the direction of propagation deflects along the interface between the  $\text{Al}_2\text{O}_3$  matrix and  $\text{TiB}_2$  shell layer, as indicated by the red arrows. Subsequently, the crack deflects into the shell layer consisting of nanosized  $\text{TiB}_2$  grains and traverses this layer by propagating along the interface of these nanosized  $\text{TiB}_2$  grains, allowing it to enter the interior of the core-shell units, as indicated by the blue arrows. The subsequent change in crack propagation direction occurs along the interface between the shell layer of  $\text{TiB}_2$  and the core of the  $\text{B}_4\text{C}$  grain, as indicated by the green arrows. Subsequently, the cracks traverse once more through the shell layer consisting of nanosized  $\text{TiB}_2$  grains and then continue to propagate along the interface between the shell layer of  $\text{TiB}_2$  and the  $\text{Al}_2\text{O}_3$  matrix. Finally, the cracks exit the core-shell units and enter the  $\text{Al}_2\text{O}_3$  matrix. During the above process, the crack may halt at any location where the energy required for crack propagation is completely consumed. The schematic portraying the deflection of cracks in the region around core-shell units is presented in Fig. 7(c). Therefore, the existence of  $\text{B}_4\text{C@TiB}_2$  core-shell units within the  $\text{Al}_2\text{O}_3$  matrix generates a complex stress distribution that induces cracks to deflect in multiple dimensions, remarkably consuming crack propagation energy. This behavior leads to an enhancement in the fracture toughness of  $\text{Al}_2\text{O}_3$  ceramics.

## 4 Conclusions

In this paper, an  $\text{Al}_2\text{O}_3$  composite ceramics toughened by  $\text{B}_4\text{C@TiB}_2$  core-shell units are successfully fabricated through a combination of the molten salt method and SPS. The synthesis of  $\text{B}_4\text{C@TiB}_2$  core-shell toughening units stems from the prior production of core-shell structural  $\text{B}_4\text{C@TiB}_2$  powders, and this core-shell structure is effectively preserved within the  $\text{Al}_2\text{O}_3$  matrix after sintering. The  $\text{B}_4\text{C@TiB}_2$  core-shell unit consists of a micron-sized  $\text{B}_4\text{C}$  core enclosed by a shell approximately 500 nm in thickness, composed of numerous nanosized  $\text{TiB}_2$  grains. The area encompassing the core-shell unit exhibits a distinct geometric arrangement amalgamating multidimensional differences in phase types, grain sizes, and thermal expansion coefficients. Consequently, an intricate stress distribution ensues. When viewing the  $\text{B}_4\text{C@TiB}_2$  core-shell unit as an entirety, crack bridges and crack forks are generated around the units. Alternatively, the crack path sequentially traverses along the interface of the  $\text{Al}_2\text{O}_3$  matrix and  $\text{TiB}_2$  shell layer when interpreting the  $\text{B}_4\text{C@TiB}_2$  core-shell units as composite structures. Then, it proceeds through the internal interface of the nanosized  $\text{TiB}_2$  shell layer, followed by propagation along the interface of the  $\text{TiB}_2$  shell layer and the  $\text{B}_4\text{C}$  core, and subsequently along the internal interface of the nanosized  $\text{TiB}_2$  shell layer. Ultimately, the crack proceeds along the interface of the  $\text{TiB}_2$  shell layer and the  $\text{Al}_2\text{O}_3$  matrix. The crack along the internal interface of the nanosized  $\text{TiB}_2$  shell layer bridges the cracks on either side of the shell layer. Therefore, the existence of core-shell units induces the multidimensional deflection of cracks, necessitating a considerable amount of crack propagation energy. This phenomenon ultimately enhances the fracture toughness of the  $\text{Al}_2\text{O}_3$  matrix. The resulting  $\text{Al}_2\text{O}_3\text{-B}_4\text{C@TiB}_2$  displays relative density of  $99.7\% \pm 0.2\%$ , Vickers hardness of  $21.5 \pm 0.8$  GPa, and fracture toughness  $6.92 \pm 0.22$   $\text{MPa} \cdot \text{m}^{1/2}$ .

This novel concept, the preparation methodology, and the elucidated toughening mechanism of utilizing core-shell units as a secondary phase to enhance ceramic matrix toughness offer a new perspective and theoretical foundation for the toughening of other structural ceramics.

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### Declaration of competing interest

The authors have no competing interests to declare that are relevant to the content of this article.

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