

CEAS-ASC Workshop 12.-13.10.2023, Budapest
“Aeroacoustics of Electrically Driven Air Vehicles: Towards a Green and Quiet Aviation”

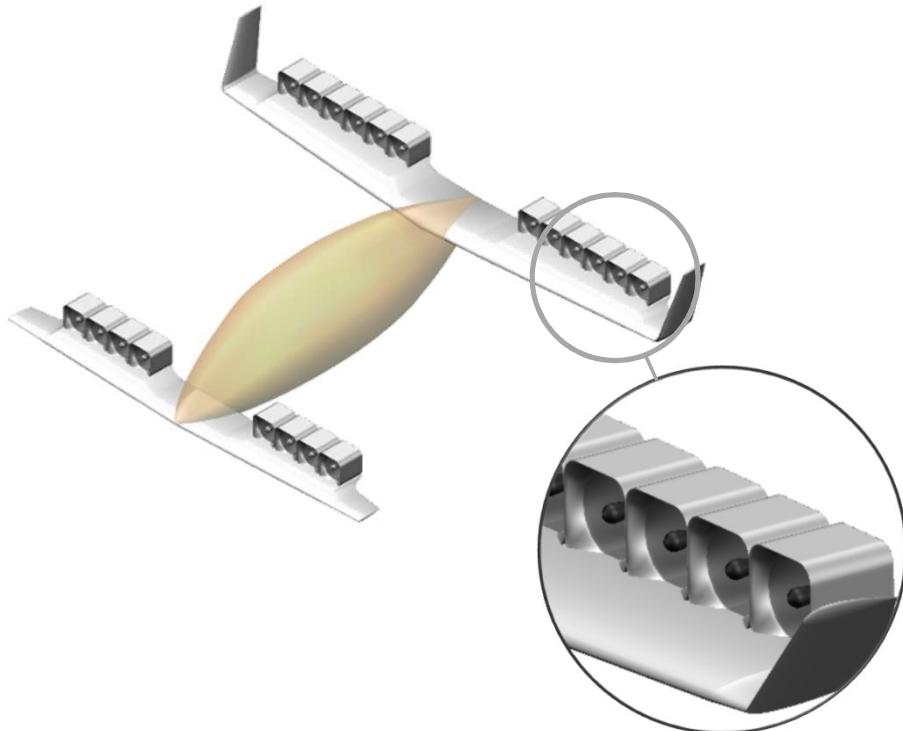
CRAFT TEST RIG FOR ASSESSING THE AEROACOUSTIC IMPACT OF FANS OF ELECTRICALLY POWERED URBAN/REGIONAL AIRCRAFT

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Noise emission from fans is a crucial point in the acceptance of UAM & RAM vehicles

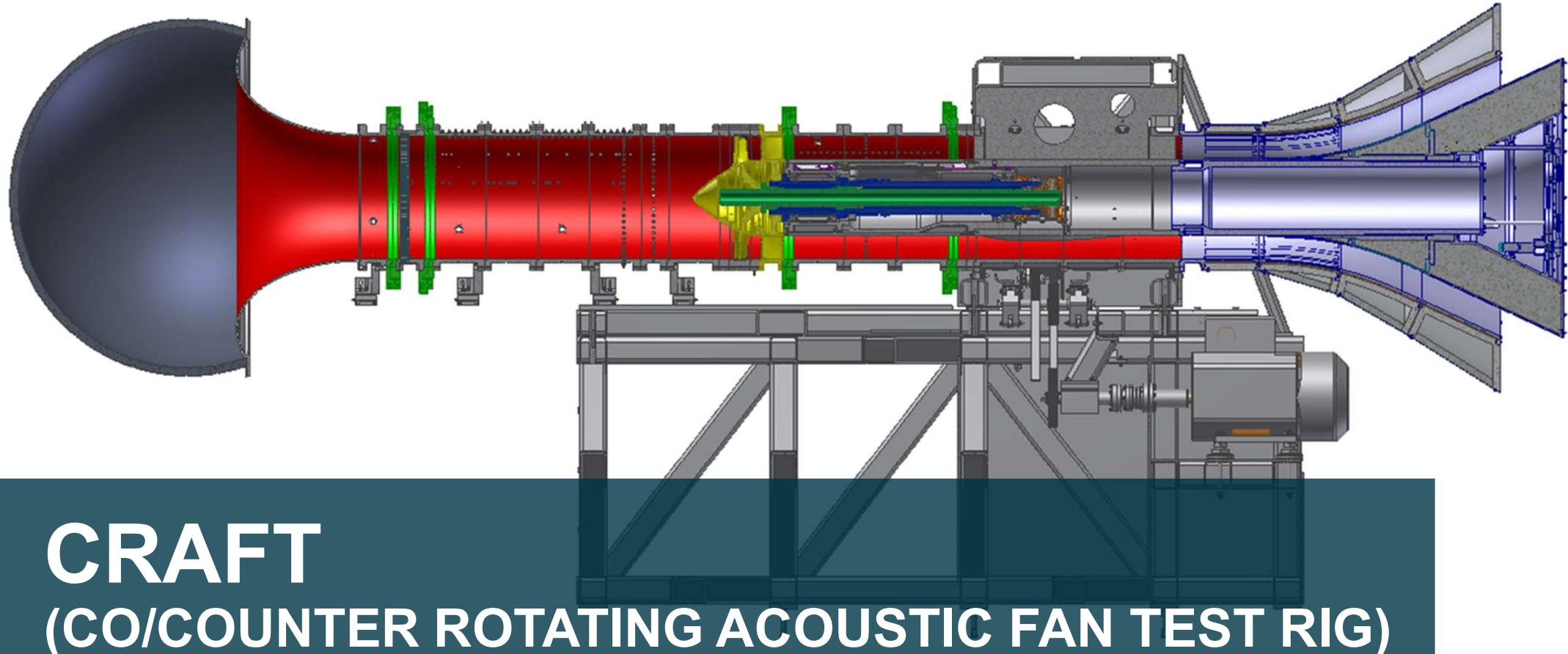


Sketch of vehicle configuration
with simplified fan housings contour
(credits: P. Ratei, SL-AES)

DLR project VIRLWINT (2023-2026)
objective: evaluation of noise perception

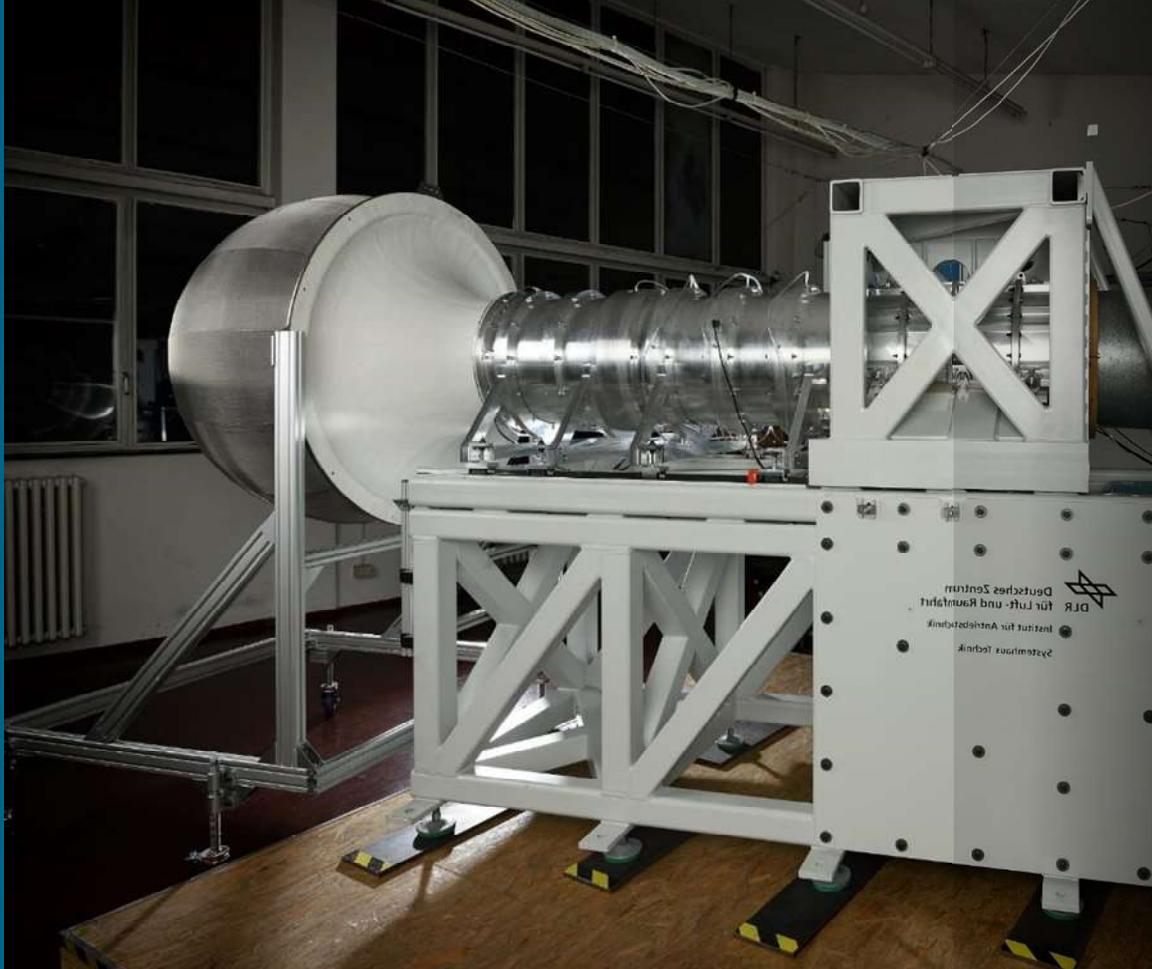
Experimental assessment of fan noise

- fan test bench
- fans with different acoustic signatures
- impact of inflow distortions due to propulsor installation

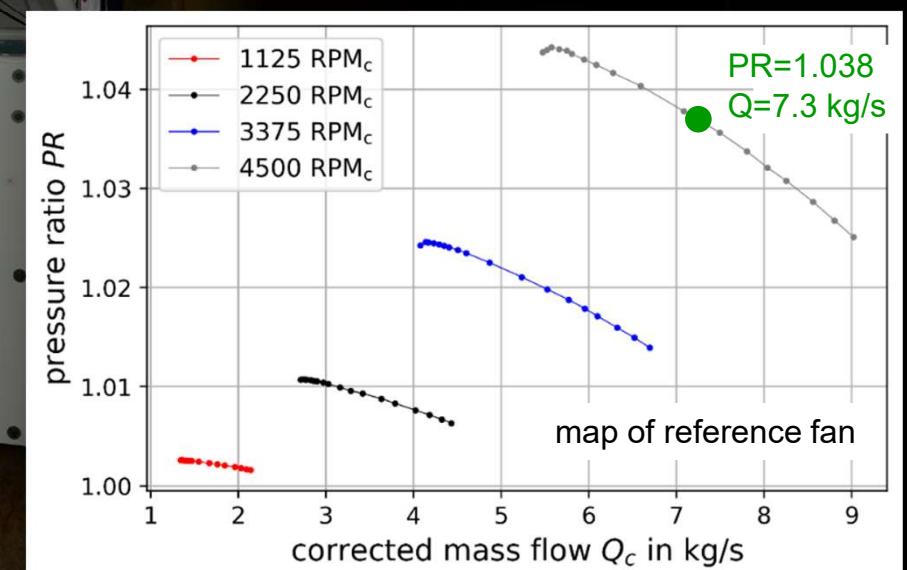


CRAFT (CO/COUNTER ROTATING ACOUSTIC FAN TEST RIG)

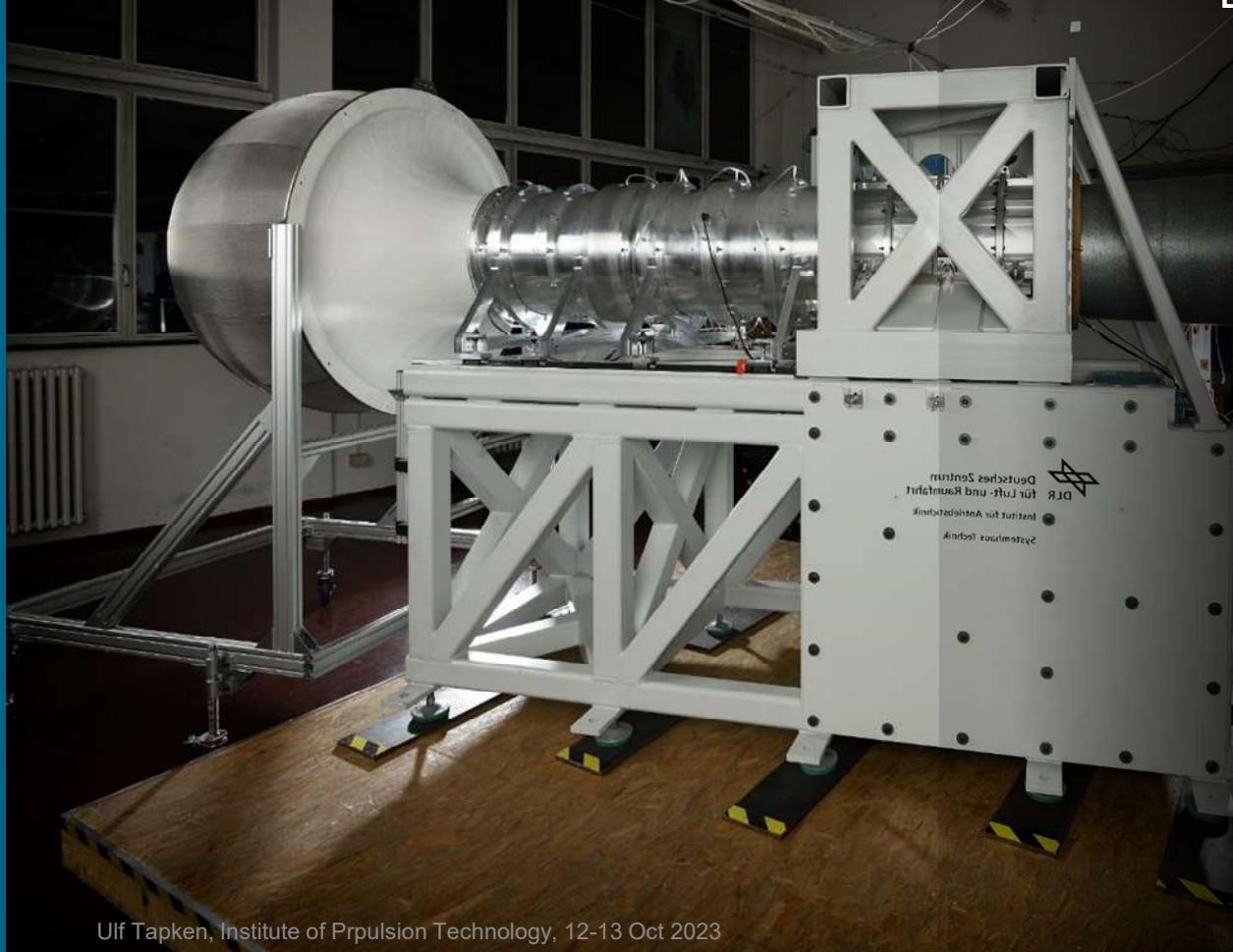
Test conditions representative for ducted fans of UAM / RAM



- fan diameter 453 mm
- shaft power 25 kW / 25 kW
- fan speed 1000 ... 4500 rpm
- max. blade tip Mach number 0.31
- tip clearance 0.6 mm

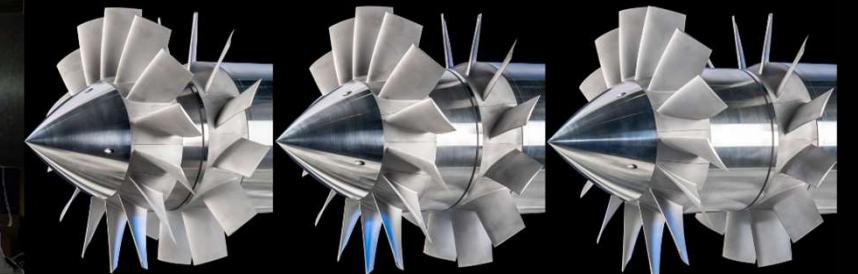


Modular and flexible setup

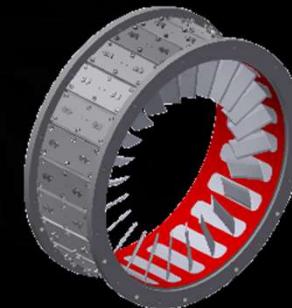


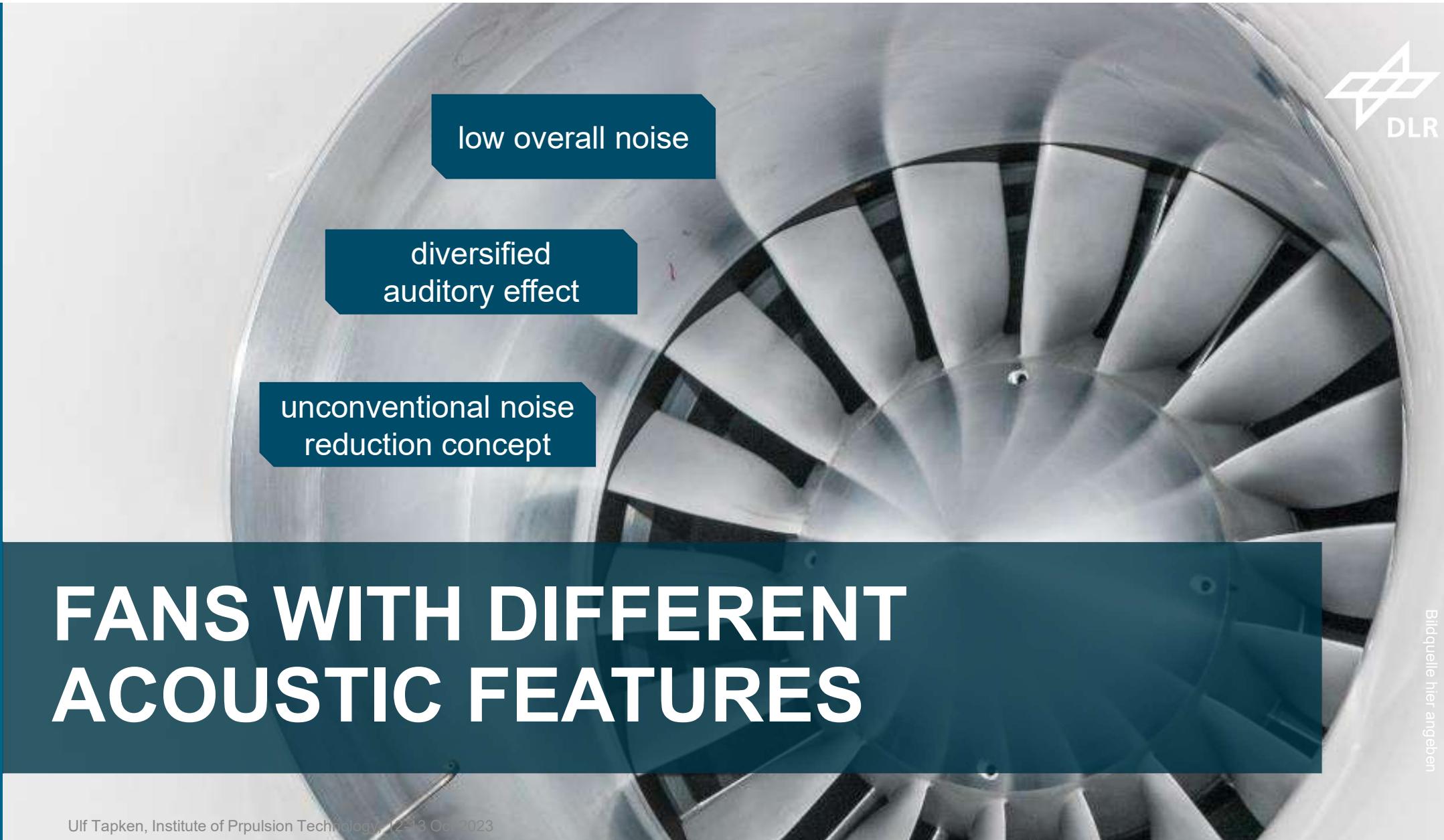
Discrimination of source mechanisms by

- distance variation
- isolated rotor operation



• Stator casing with exchangeable vanes





low overall noise

diversified
auditory effect

unconventional noise
reduction concept

FANS WITH DIFFERENT ACOUSTIC FEATURES

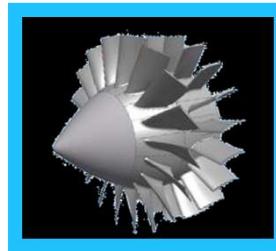
Bildquelle hier angeben

Derivation of low noise designs



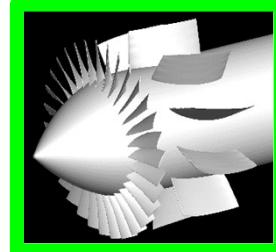
Reference fan

18 rotor blades
21 stator vanes



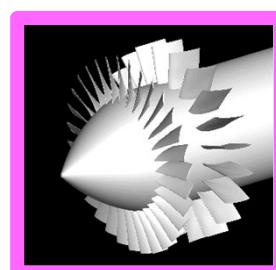
Low noise fan 1

31 rotor blades
10 stator vanes
BPF weak

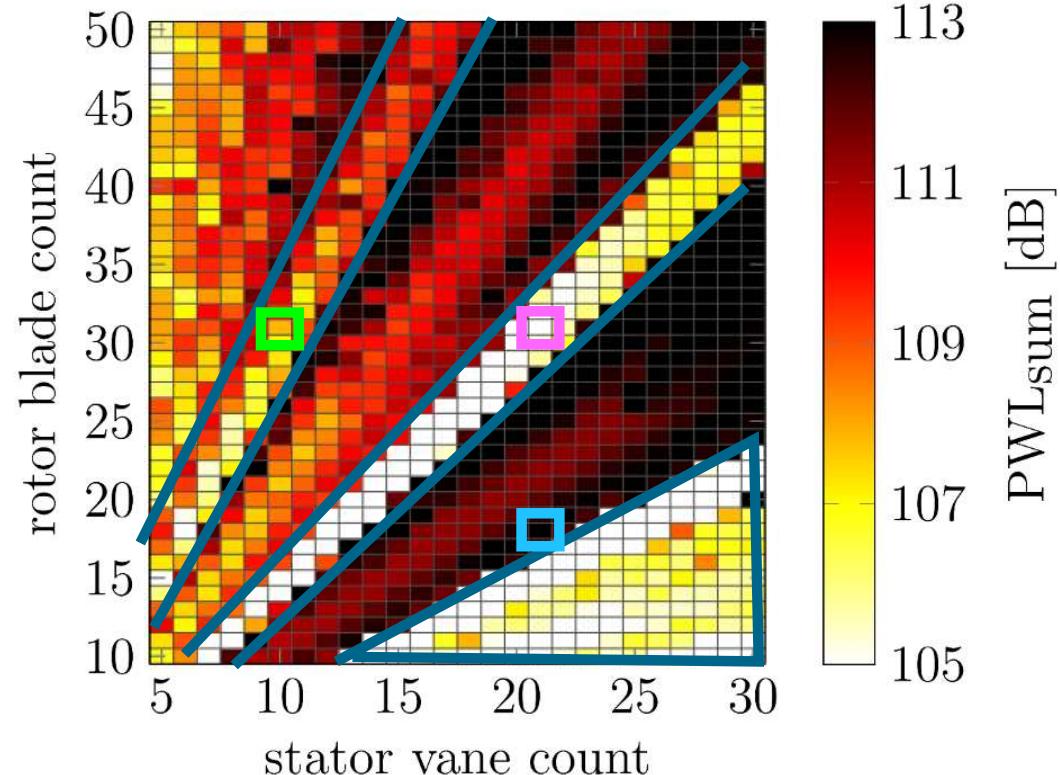


Low noise fan 2

31 rotor blades
21 stator vanes
BPF cut-off



Sound power level cumulated for 3 OP over RSI tonal, RSI BBN



Characteristics with regard to noise perception



Fan	lowest blade tone	tonal dominance upstream vs downstream	relation tonal to broadband noise radiation
Reference B=18 V=21 	BPF 1.35 kHz	<p>PWL [dB]</p> <p>x [m]</p> <p>Rotor</p> <p>Stator</p> <ul style="list-style-type: none"> Tonal sum BPF1 BPF2 Tonal sum refl. BPF1 refl. BPF2 refl. 	
Low Noise 1 B=31 V=10 	BPF 2.325 kHz	<p>PWL [dB]</p> <p>x [m]</p> <p>Rotor</p> <p>Stator</p> <ul style="list-style-type: none"> Tonal sum BPF1 BPF2 Tonal sum refl. BPF1 refl. BPF2 refl. 	<p>angle [deg]</p> <p>SPL [dB]</p> <p>0° 30° 60° 90° 120° 150° 180°</p> <p>downstream upstream</p> <ul style="list-style-type: none"> bbn-TE-rotor bbn-TE-stator RSI-bbn RSI-tonal
Low Noise 2 B=31 V=21 	2 BPF 4.65 kHz	<p>PWL [dB]</p> <p>x [m]</p> <p>Rotor</p> <p>Stator</p> <ul style="list-style-type: none"> Tonal sum BPF1 BPF2 Tonal sum refl. BPF1 refl. BPF2 refl. 	<p>angle [deg]</p> <p>SPL [dB]</p> <p>0° 30° 60° 90° 120° 150° 180°</p> <p>downstream upstream</p> <ul style="list-style-type: none"> bbn-TE-rotor bbn-TE-stator RSI-bbn RSI-tonal

MEASUREMENT INSTRUMENTATION

Aerodynamic instrumentation



radial and azimuthal traverse



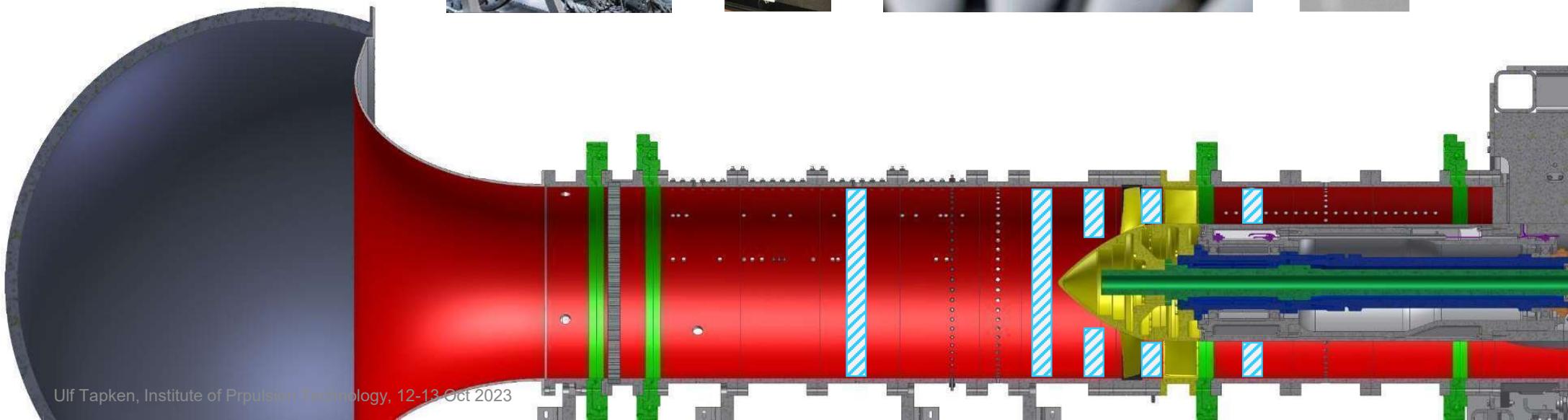
total pressure rakes



hotwire x-probes



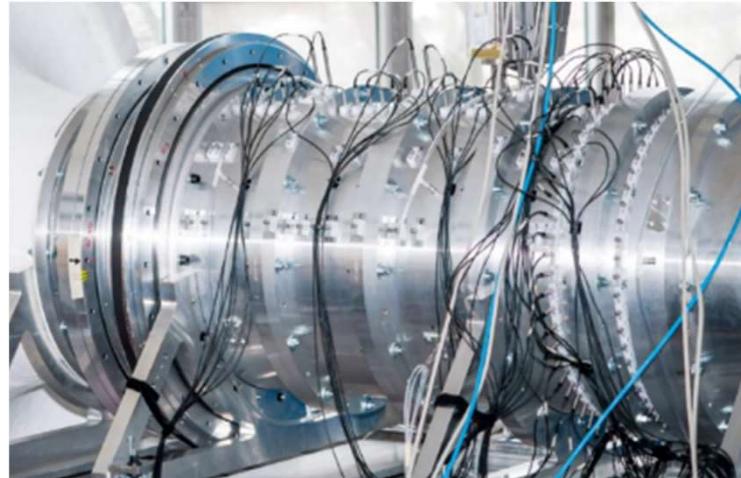
5-hole probe



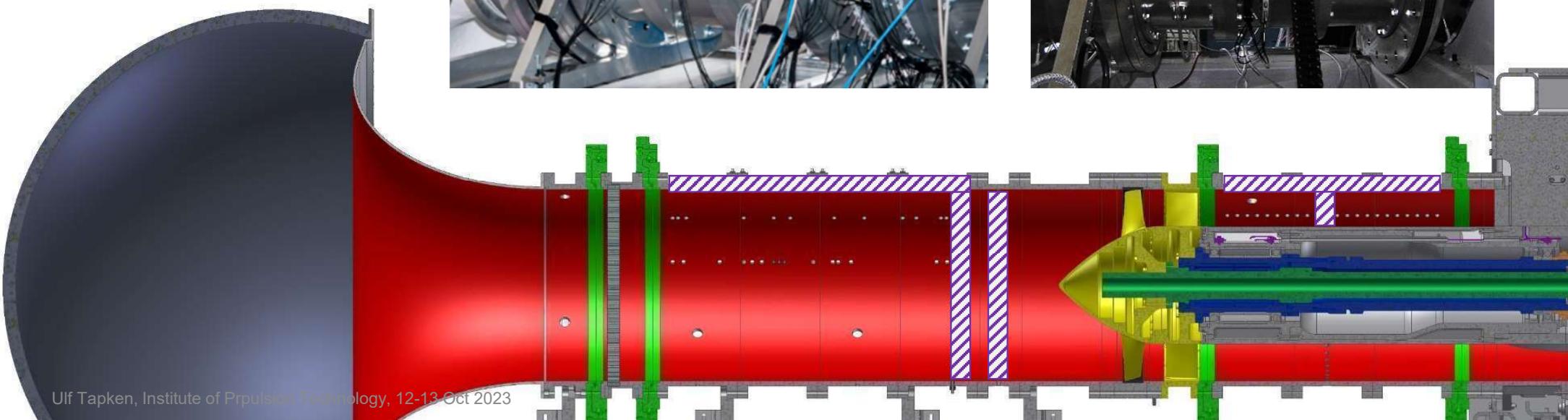
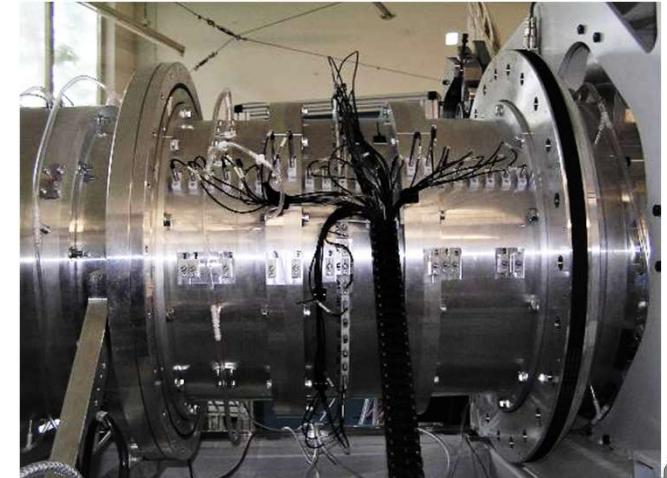
Acoustic instrumentation



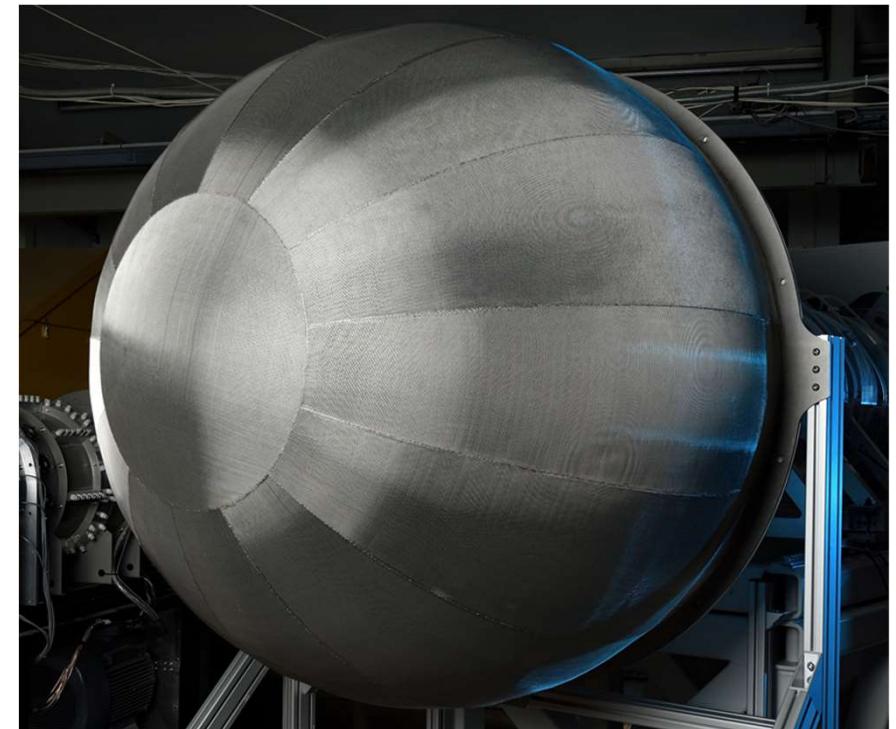
upstream mode detection arrays



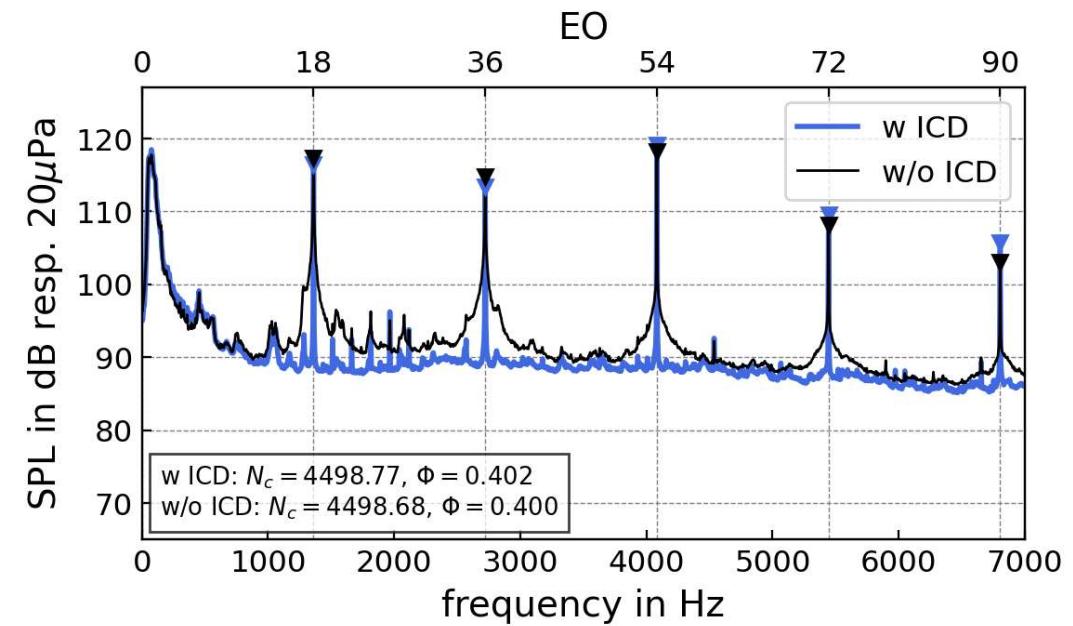
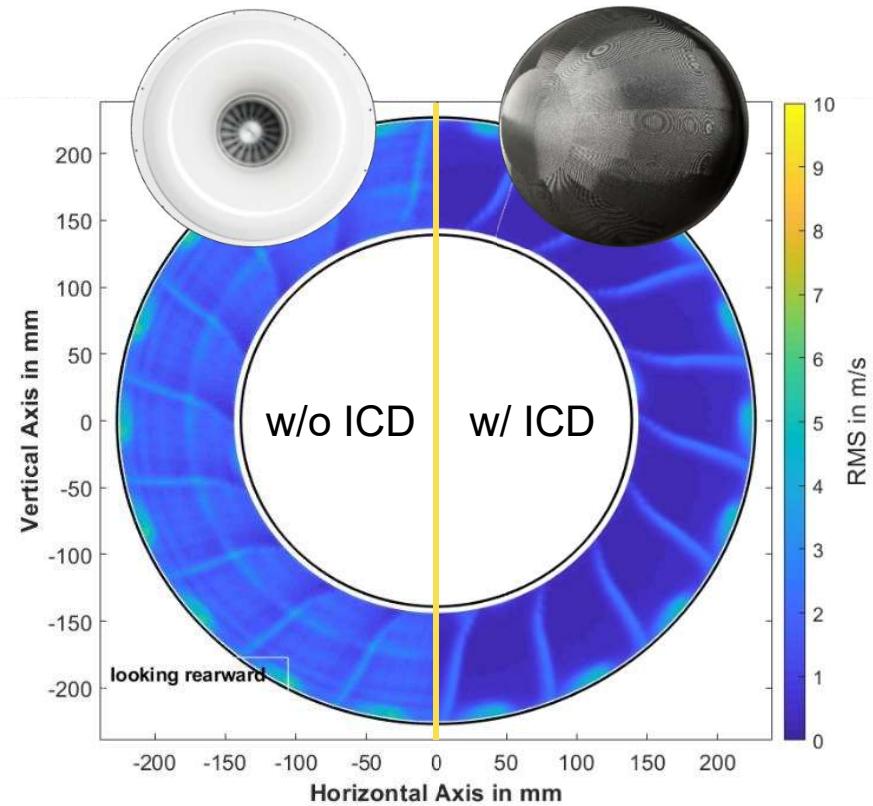
downstream mode detection arrays



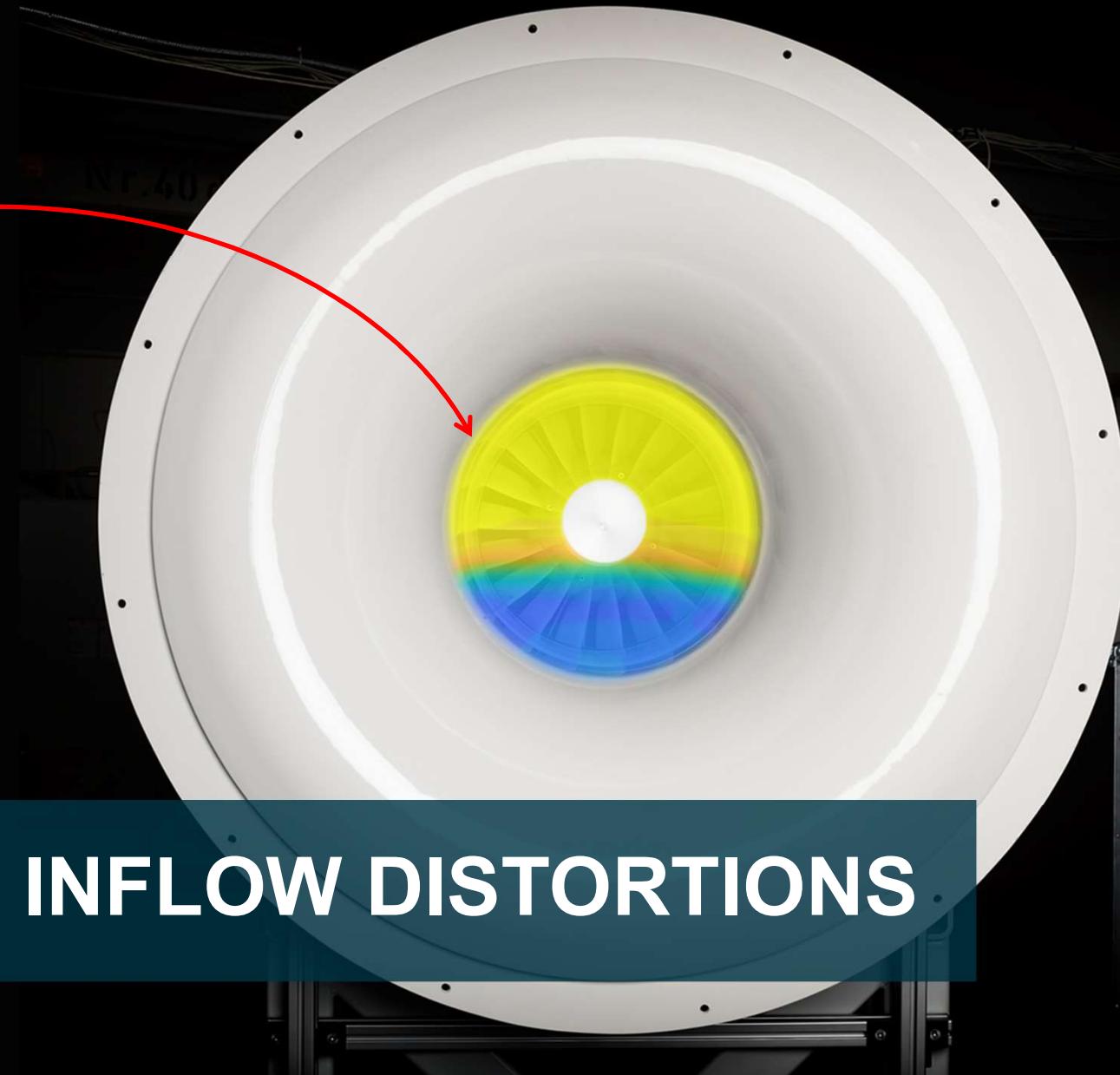
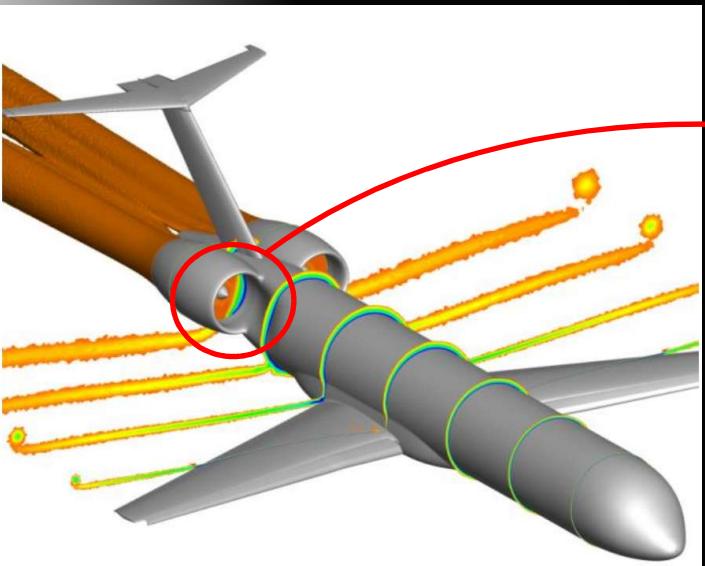
Inflow control device (ICD) establishing clean base inflow



ICD impact on flow field and fan noise generation



- reduction of non-representative noise components
- stabilization of rotor-coherent source mechanisms

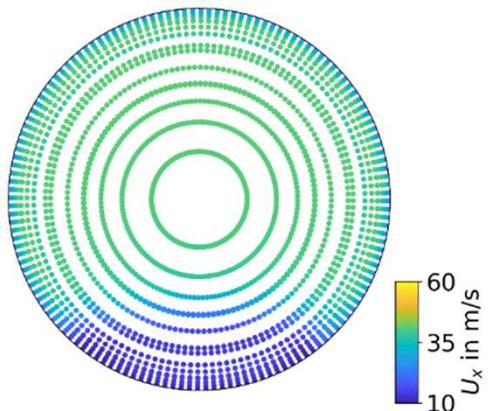


GENERATION OF INFLOW DISTORTIONS

First tests with perforated distortion fences

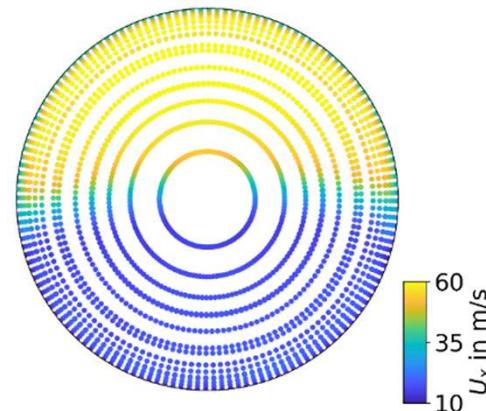
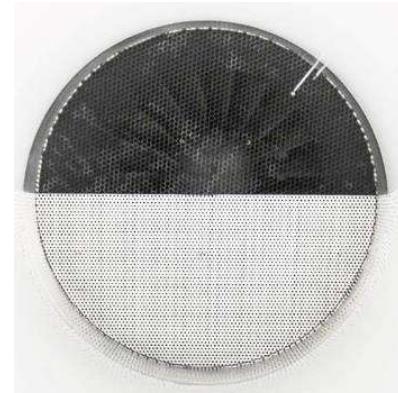


fence height 20%
opening ratio 33%

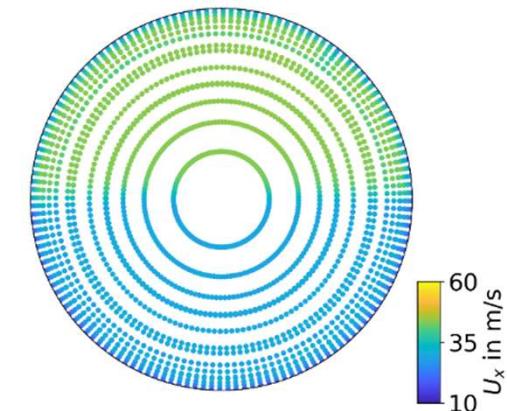
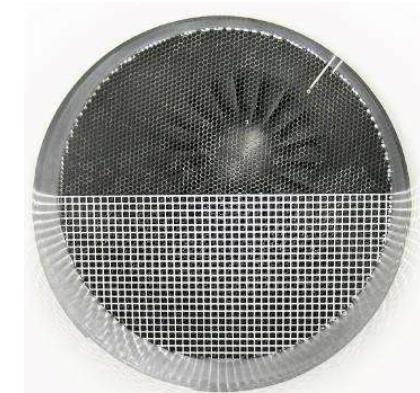


Ulf Tapken, Institute of Propulsion Technology, 12-13 Oct 2023

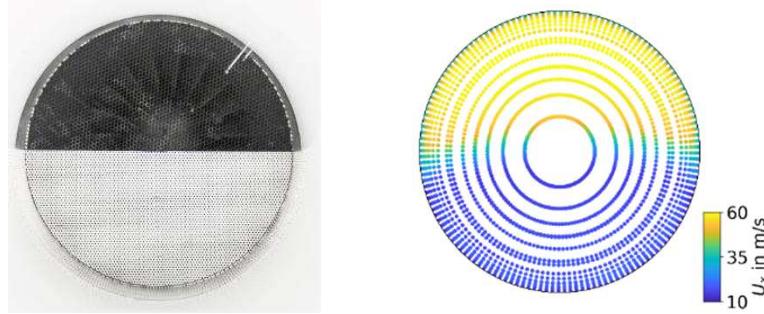
fence height 50%
opening ratio 33%



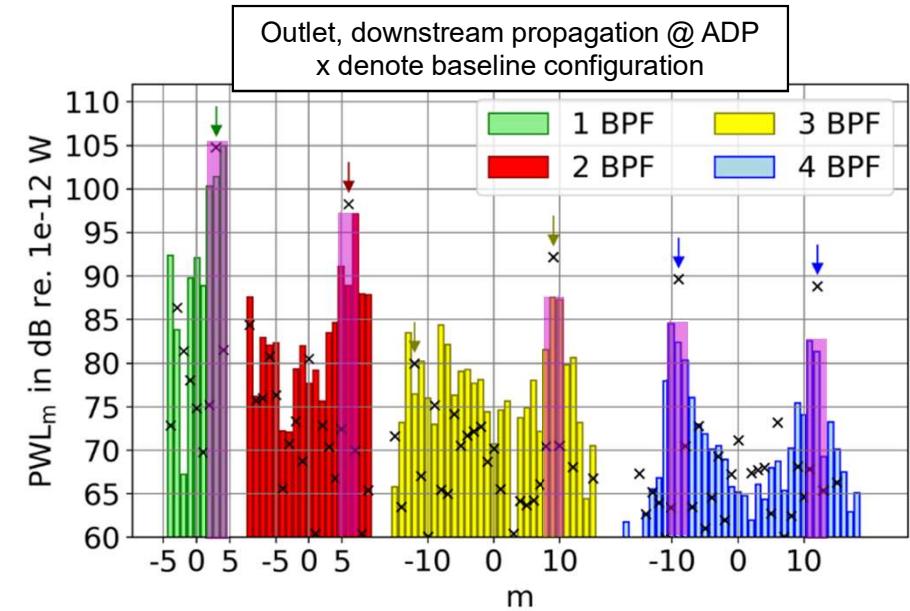
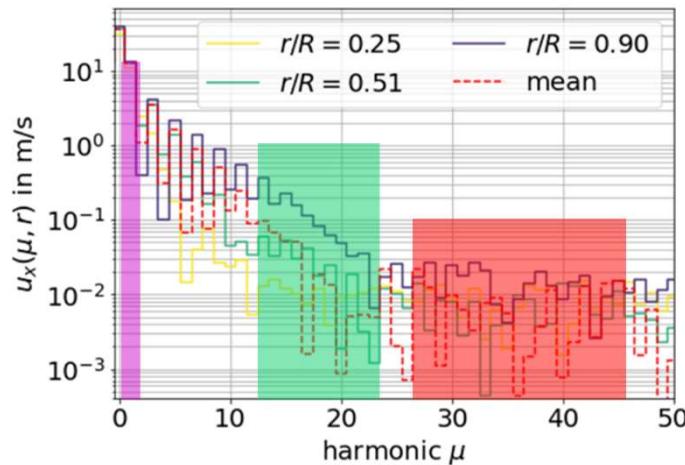
fence height 50%
opening ratio 64%



Impact of disturbed mean flow profile on fan tones



circumferential harmonics of the flow disturbance @ ADP

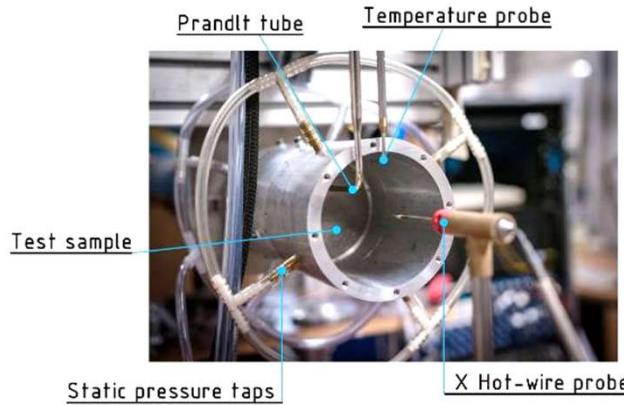


- rotor/stator interaction is modified
 $m_{RSI,\mu} = (-)hB + kV \pm \mu$
- disturbed-inflow/rotor interaction results in
 $m_{BLI} = (-)hB \pm \mu$

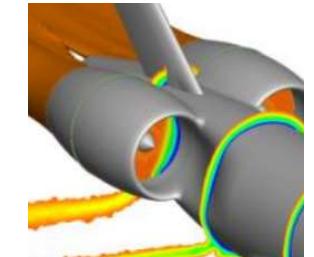
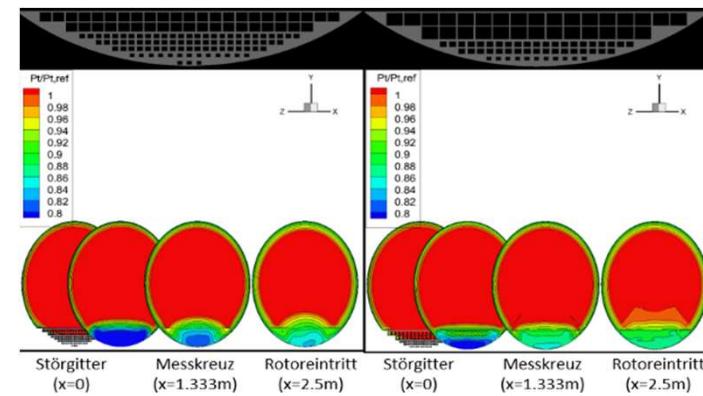
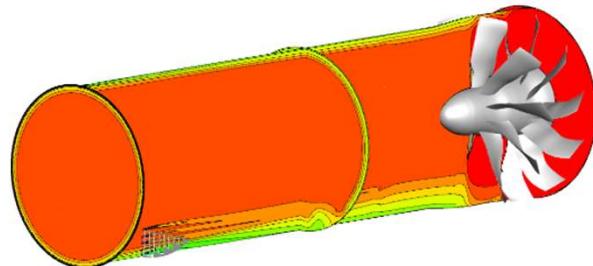
Development of devices for BLI representative mean inflows



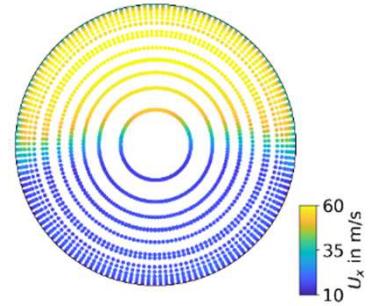
Experimental study of different concepts



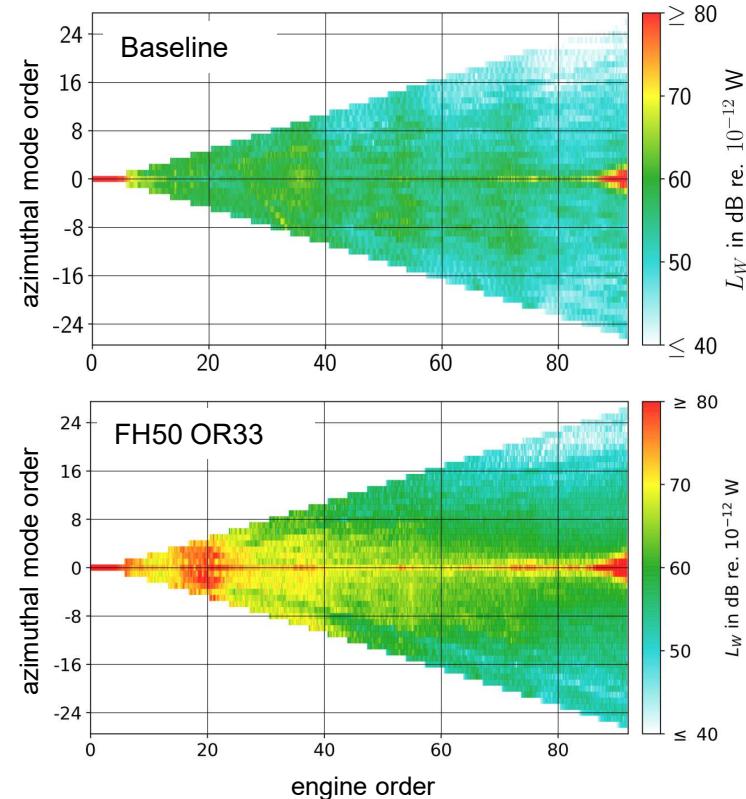
CFD optimization



Impact of increased turbulent inflow on fan broadband noise

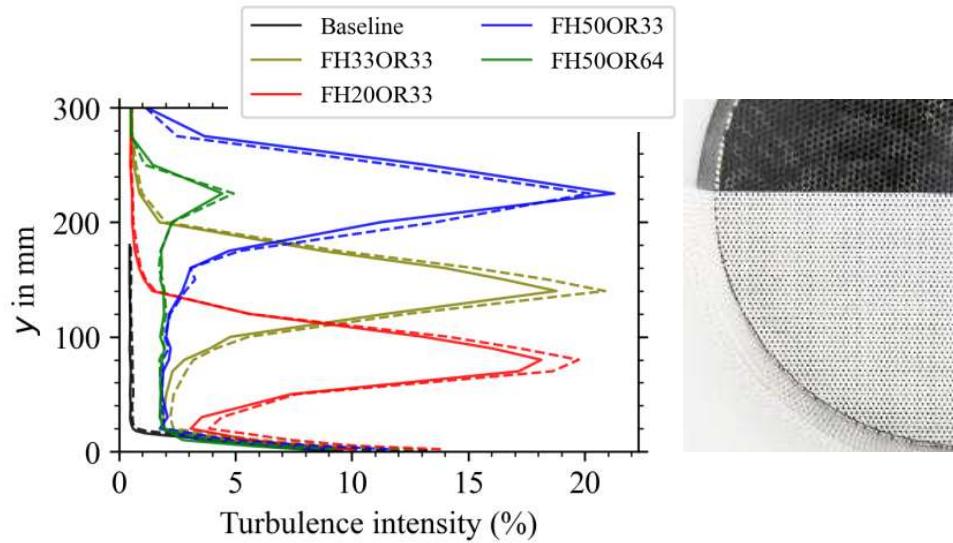


sound power of upstream modes in inlet @ ADP



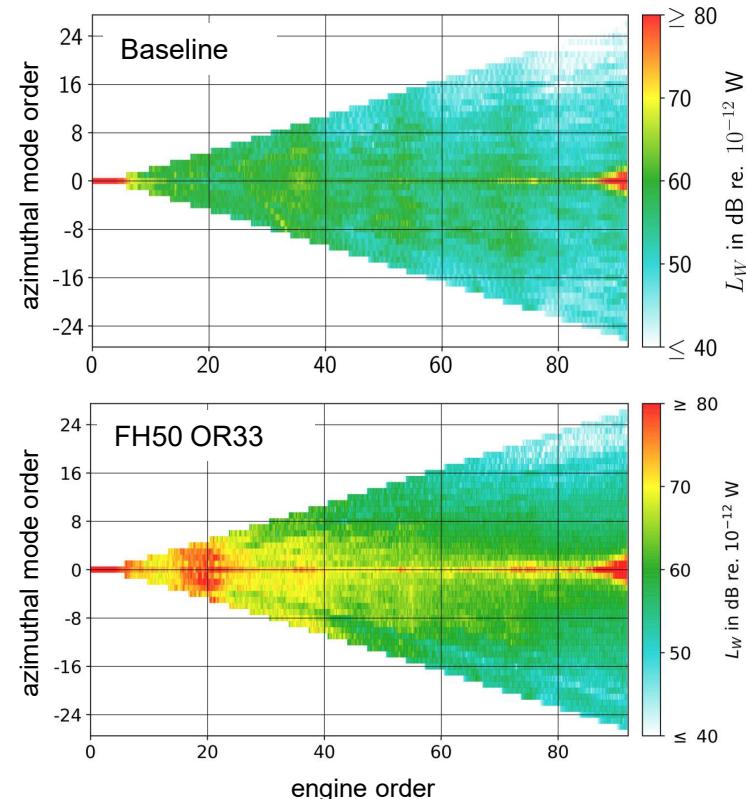
- turbulent-inflow/rotor interaction results in broad sound power increase
- modified rotor/stator interaction plus rotor shielding leads to increase of co-rotating modes

Impact of increased turbulent inflow on fan broadband noise



- turbulent-inflow/rotor interaction results in broad sound power increase
- modified rotor/stator interaction plus rotor shielding leads to increase of co-rotating modes

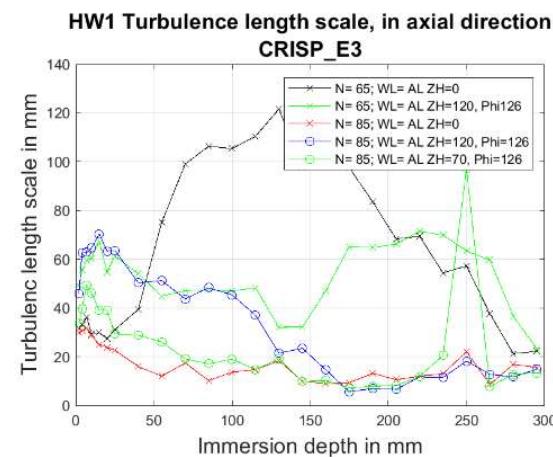
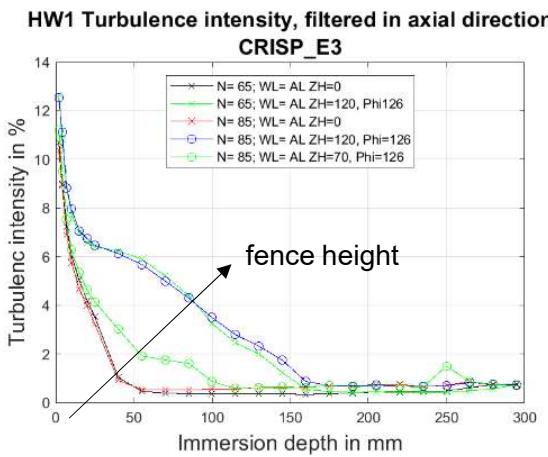
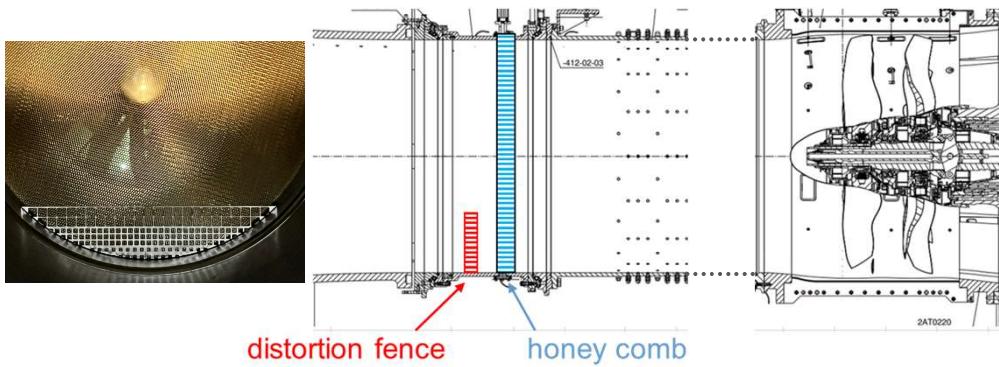
sound power of upstream modes in inlet @ ADP



Challenge: Generation of specific turbulence characteristics



Objective: realistic turbulence profile



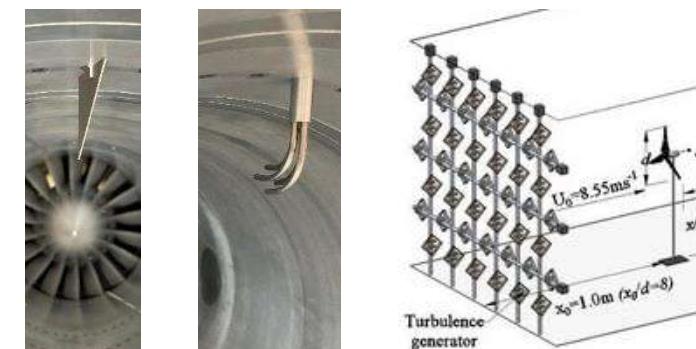
Objective: Controlled generation of turbulence

- absolute length scales
- relation of lenght scales (anisotropy)
- spectral shape

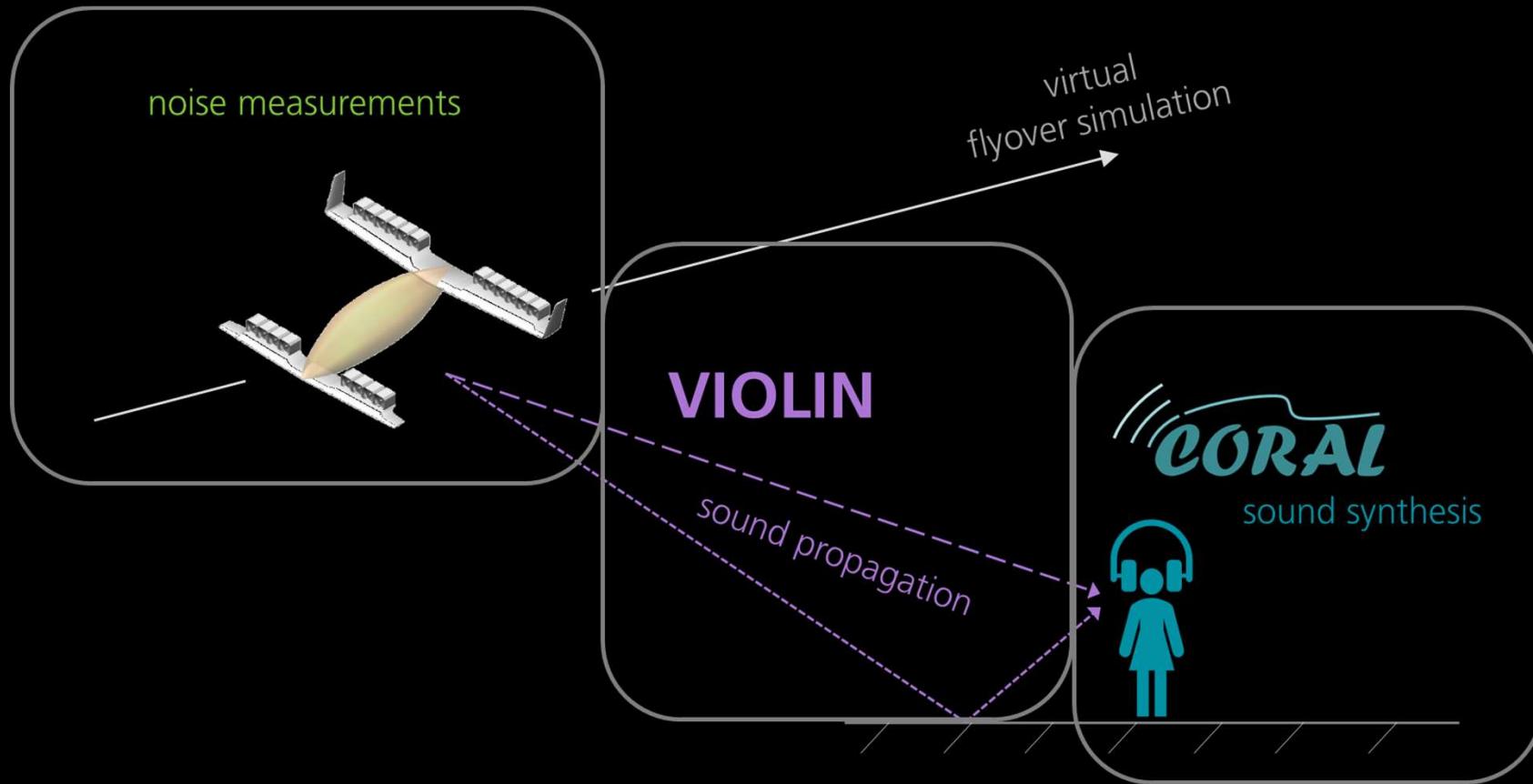
to investigate impact on noise

- spectral coherence
- broadband ... narrowband components
- excited noise levels

→ develop and test suitable devices

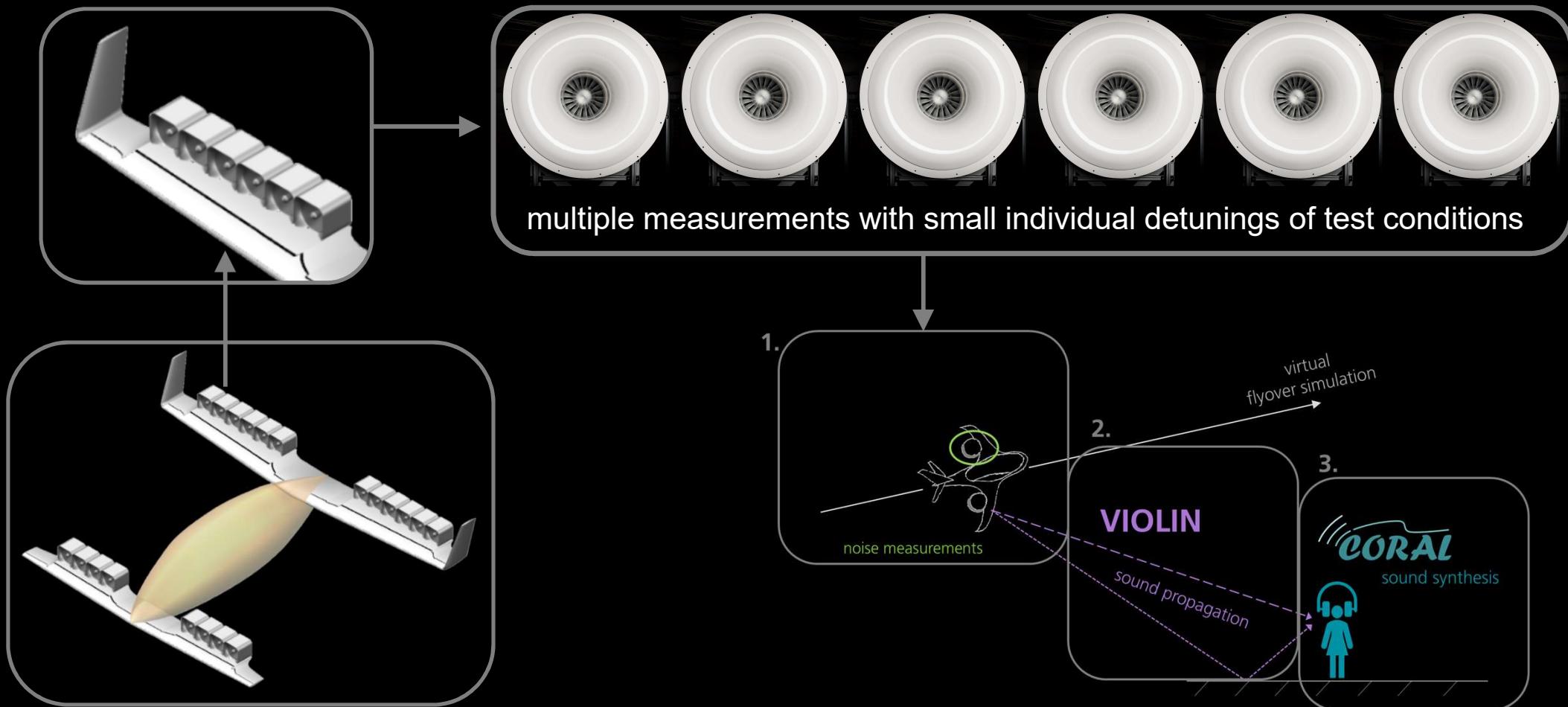


source: Y. Jin, DOI: 0.3390/en9100830



AURALIZATION OF ENGINE NOISE

Simulation of distributed fan configuration as input for auralization



SUMMARY

Opportunities offered by test bench CRAFT

- Test conditions representative for fans of UAM / RAM
- Fans with different acoustic/auditory signatures
- Measurement of fan aero-acoustics in high resolution and with highlighting specific noise sources by test set-up variations
- Inflow distortion devices to evaluate engine integration effects
 - Challenge: generation of specific turbulence characteristics
- Data for validations and for psychoacoustic assessments
- Investigation of noise reduction concepts e.g. by flow control

Contact: ulf.tapken@dlr.de



- [1] Klähn, Lukas und Caldas, Luciano und Meyer, Robert und Moreau, Antoine und Tapken, Ulf (2023) Experimental Investigation of Inflow-Distortion-Induced Tonal Noise in a Sub-Sonic Fan Test Rig. In: 29th International Congress on Sound and Vibration, ICSV 2023. 29th International Congress on Sound and Vibration, 09.-13. Jul. 2023, Prag, Tschechische Republik. ISBN 978-801103423-8. ISSN 2329-3675.
- [2] L. Caldas, S. Kruck, L. Klähn, A. Rudolphi, R. Meyer, L. Enghardt and U. Tapken. (2022) *Construction and Assessment of an Inflow-Control-Device for a Low-Speed Aeroacoustic Fan Rig*. AIAA Journal 2022 60:9, 5299-5312, DOI: [10.2514/1.J061729](https://doi.org/10.2514/1.J061729)
- [3] U. Tapken, R. Meyer, L. Caldas, L. Klähn, M. Behn, A. Rudolphi, R. Jaron, A. Moreau, and L. Enghardt. (2021) *New test facility to investigate the noise generated by shrouded fans for small aircraft applications*. DICUAM 2021 - Delft International Conference on Urban Air-Mobility, 15.-17. März 2021, Delft, Niederlande.
- [4] U. Tapken, L. Caldas, R. Meyer, M. Behn, L. Klähn, R. Jaron and A. Rudolphi. (2021) *Fan test rig for detailed investigation of noise generation mechanisms due to inflow disturbances*. AIAA 2021-2314. AIAA AVIATION 2021 FORUM. August 2021. DOI: [10.2514/6.2021-2314](https://doi.org/10.2514/6.2021-2314)
- [5] L. Caldas, L. Klähn, R. Meyer and U. Tapken. *Measurements of fan inflow distortion noise generation in a low speed fan - part I: aerodynamic analyses*, AIAA 2022-2944. 28th AIAA/CEAS Aeroacoustics 2022 Conference. June 2022. DOI: [10.2514/6.2022-2944](https://doi.org/10.2514/6.2022-2944)
- [6] L. Klähn, L. Caldas and U. Tapken. *Measurements of fan inflow distortion noise generation in a low speed fan - part II: acoustic analyses*, AIAA 2022-2945. 28th AIAA/CEAS Aeroacoustics 2022 Conference. June 2022. DOI: [10.2514/6.2022-2945](https://doi.org/10.2514/6.2022-2945)
- [7] L. Klähn, L. Caldas, R. Meyer, A. Rudolphi, A. Moreau, und U. Tapken. (2022): *Experimentelle Untersuchung der Auswirkung von Einlaufstörungen auf die Aeroakustik einer Rotor-Stator-Konfiguration am Fan-Prüfstand CRAFT*. Deutsche Gesellschaft für Luft- und Raumfahrt - Lilienthal-Oberth e.V. DOI: [10.25967/570161](https://doi.org/10.25967/570161)
- [8] L. Klähn, A. Moreau, L. Caldas, R. Meyer, und U. Tapken. (2021) *Interpretation von Breitbandgeräusch-Messungen am Fan-Prüfstand CRAFT mithilfe analytischer Modelle*. In: Fortschritte der Akustik - DAGA 2021, Wien, Österreich. ISBN 978-3-939296-18-8.
- [9] L. Klähn, A. Moreau, L. Caldas, R. Jaron and U. Tapken. *Advanced analysis of fan noise measurements supported by theoretical source models*. International Journal of Aeroacoustics. 2022; 21(3-4): 239-259. DOI: [10.1177/1475472X221093703](https://doi.org/10.1177/1475472X221093703)
- [10] L. Klähn, A. Moreau, L. Caldas and U. Tapken. (2022): Assessment of In-Duct Fan Broadband Noise Measurements in a Modern Low-Speed Test Rig. FAN 2022 – International Conference on Fan Noise, Aerodynamics, Applications and Systems, Senlis, Frankreich, DOI: [10.26083/tuprints-00021709](https://doi.org/10.26083/tuprints-00021709)
- [11] S. Schade, R. Jaron, A. Moreau: Smart blade count selection to align modal propagation angle with stator stagger angle for low-noise fan stage designs, DLRK 2023