

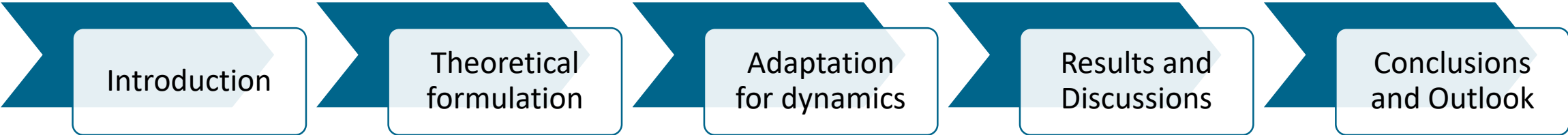
A Momentum Subspace-based Model Order Reduction for Finite Element Models in Nonlinear Dynamic Analyses

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graph LR; A[Introduction] --> B[Theoretical formulation]; B --> C[Adaptation for dynamics]; C --> D[Results and Discussions]; D --> E[Conclusions and Outlook]
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Introduction

Theoretical
formulation

Adaptation
for dynamics

Results and
Discussions

Conclusions
and Outlook

Introduction



- Structural nonlinearities is of interest in several engineering fields.
- Full finite element (FE) solution can be computationally expensive.
- Reduced order models (ROM) provide an efficient solution to such problems.
- The momentum subspace ROM, discussed here, is an adaptation of the Koiter-Newton reduction technique (K. Liang et al., 2013).
- Extended to dynamics (Sinha et al., 2020) with focus on panel structures.
- Current studies on cantilevers.

Theoretical Formulation



- The equilibrium equations (statics) are expanded up to the third order in Taylor series.

$$f(\mathbf{u}) = \mathcal{L}(\mathbf{u}) + Q(\mathbf{u}, \mathbf{u}) + C(\mathbf{u}, \mathbf{u}, \mathbf{u}) = \mathbf{f}_{\text{ext}} = \mathbf{F}\boldsymbol{\phi} \quad (1)$$

- The equilibrium displacement \mathbf{u} is parametrised by generalized displacements ξ .

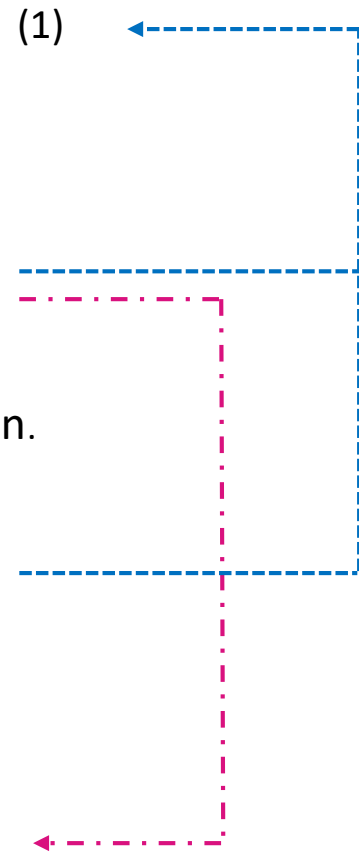
$$\mathbf{u}(\xi) = \mathbf{u}_{\alpha}\xi_{\alpha} + \mathbf{u}_{\alpha\beta}\xi_{\alpha}\xi_{\beta} \quad (2)$$

- In the reduced subspace, a similar assumption is made for the equilibrium equation.

$$\bar{\mathcal{L}}(\xi) + \bar{Q}(\xi, \xi) + \bar{C}(\xi, \xi, \xi) = \phi \quad (3)$$

- Work equivalence to fix the parametrisation.

$$(\mathbf{F}\boldsymbol{\phi})' \cdot \delta\mathbf{u} = \boldsymbol{\phi}' \cdot \delta\xi \quad (4)$$



Theoretical Formulation

- By regrouping the coefficients of ξ , a set of ROM equations are obtained.

$$\begin{bmatrix} \mathbf{L} & -\mathbf{F} \\ -\mathbf{F} & \mathbf{0} \end{bmatrix} \begin{Bmatrix} \mathbf{u}_\alpha \\ \bar{\mathbf{L}}_\alpha \end{Bmatrix} = \begin{Bmatrix} \mathbf{0} \\ \mathbf{E}_\alpha \end{Bmatrix} \quad \mathbf{u}(\xi) = \mathbf{u}_\alpha \xi_\alpha + \mathbf{u}_{\alpha\beta} \xi_\alpha \xi_\beta$$
$$\begin{bmatrix} \mathbf{L} & -\mathbf{F} \\ -\mathbf{F} & \mathbf{0} \end{bmatrix} \begin{Bmatrix} \mathbf{u}_{\alpha\beta} \\ \bar{\mathbf{Q}}_{\alpha\beta} \end{Bmatrix} = \begin{Bmatrix} -\mathbf{Q}(\mathbf{u}_\alpha, \mathbf{u}_\beta) \\ \mathbf{0} \end{Bmatrix}$$

$$\bar{\mathbf{C}}_{\alpha\beta\gamma\delta} = \mathbf{C}(\mathbf{u}_\alpha, \mathbf{u}_\beta, \mathbf{u}_\gamma, \mathbf{u}_\delta) - \frac{2}{3} [\mathbf{u}^t_{\alpha\beta} \mathbf{L}(\mathbf{u}_{\delta\gamma}) + \mathbf{u}^t_{\beta\gamma} \mathbf{L}(\mathbf{u}_{\delta\alpha}) + \mathbf{u}^t_{\gamma\alpha} \mathbf{L}(\mathbf{u}_{\delta\beta})]$$

- The stiffness tensors are obtained as higher order derivatives of strain energy.

Adaptations for dynamics



- Full FE equations described by

$$\mathbf{M}\ddot{\mathbf{u}} + \mathbf{D}\dot{\mathbf{u}} + \mathbf{L}\mathbf{u} + \mathbf{Q}\mathbf{u}\mathbf{u} + \mathbf{C}\mathbf{u}\mathbf{u}\mathbf{u} = \mathbf{F}(t)$$

- 1st order differential equations in order to perform parametric continuation (AUTO, Doedel, 2007)
- Hamiltonian formulation to derive the equations of motion.

$$H(\mathbf{u}, \mathbf{p}) = T(\mathbf{u}, \mathbf{p}) + V(\mathbf{u})$$

- Conservative system - damping and external force excluded initially.
- An assumption is made for the momentum:

$$\mathbf{p} = \mathbf{P}\boldsymbol{\pi}, \quad \mathbf{P} = \mathbf{M}\boldsymbol{\Phi}$$

where \mathbf{P} is the basis matrix, $\boldsymbol{\pi}$ is a vector of amplitudes for the momentum vectors.

Adaptations for dynamics



- Potential energy in the reduced subspace:

$$\bar{V} = \frac{1}{2} \bar{L}_{\alpha\beta} \xi_{\alpha} \xi_{\beta} + \frac{1}{3} \bar{Q}_{\alpha\beta\gamma} \xi_{\alpha} \xi_{\beta} \xi_{\gamma} + \frac{1}{4} \bar{C}_{\alpha\beta\gamma\delta} \xi_{\alpha} \xi_{\beta} \xi_{\gamma} \xi_{\delta}$$

- Kinetic energy in the reduced subspace:

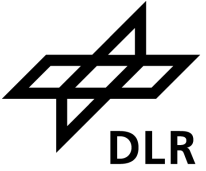
$$\bar{T} = \frac{1}{2} \pi' \underbrace{(\Phi' \mathbf{M} \Phi)}_{\bar{\mathbf{M}}^{-1}} \pi$$

- For a conservative system,

$$\dot{\xi} = \frac{\partial \bar{H}}{\partial \pi} = \bar{\mathbf{M}}^{-1} \pi$$

$$\dot{\pi} = \frac{\partial \bar{H}}{\partial \xi} = -\{\bar{\mathbf{L}}\xi + \bar{\mathbf{Q}}\xi\xi + \bar{\mathbf{C}}\xi\xi\xi\}$$

Adaptations for dynamics



- Potential energy in the reduced subspace:

$$\bar{V} = \frac{1}{2} \bar{L}_{\alpha\beta} \xi_{\alpha} \xi_{\beta} + \frac{1}{3} \bar{Q}_{\alpha\beta\gamma} \xi_{\alpha} \xi_{\beta} \xi_{\gamma} + \frac{1}{4} \bar{C}_{\alpha\beta\gamma\delta} \xi_{\alpha} \xi_{\beta} \xi_{\gamma} \xi_{\delta}$$

- Kinetic energy in the reduced subspace:

$$\bar{T} = \frac{1}{2} \pi' \underbrace{(\Phi' \mathbf{M} \Phi)}_{\bar{\mathbf{M}}^{-1}} \pi$$

- For a non-conservative system,

$$\dot{\xi} = \frac{\partial \bar{H}}{\partial \pi} = \bar{\mathbf{M}}^{-1} \pi$$

$$\dot{\pi} = - \frac{\partial \bar{H}}{\partial \xi} = -\{\bar{\mathbf{L}}\xi + \bar{\mathbf{Q}}\xi\xi + \bar{\mathbf{C}}\xi\xi\xi\} - \bar{\mathbf{D}}\bar{\mathbf{M}}^{-1}\pi + \bar{\boldsymbol{\phi}}(t)$$

Rayleigh damping

Quadratic damping model

Discussion of results



- Test case description
 1. Test case 1 : Simply supported square plate (M. Amabili, 2004).
 2. Test case 2 : Stiffened plate with free boundary conditions (Sinha et al, 2020).
 3. Test case 3 : Ongoing studies, cantilever beam.

Discussion of results

Test case 1: Rectangular plate, simply supported (M. Amabili, 2004)

Analysis parameters:

$$l = b = 0.3 \text{ m}$$

$$t = 0.001 \text{ m}$$

$$\text{Damping ratio } \zeta = 0.065$$

$$\text{Applied force } f_{ext} = 1.74 \text{ N (centre of the plate)}$$

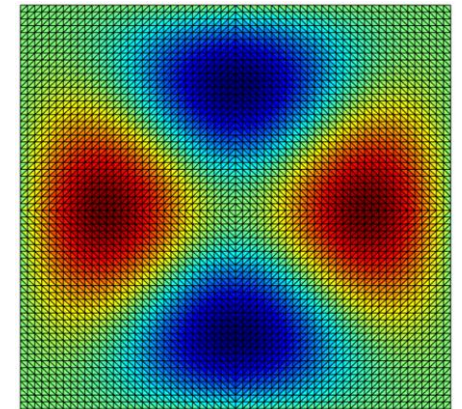
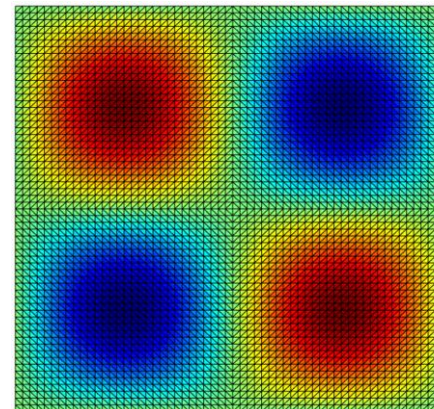
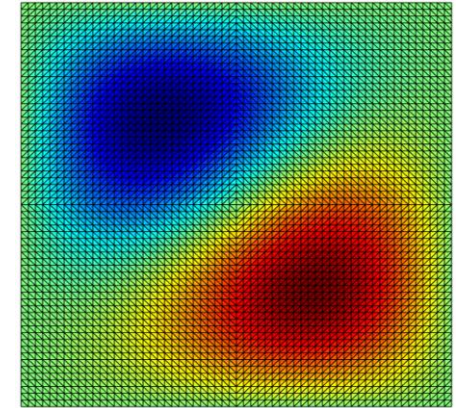
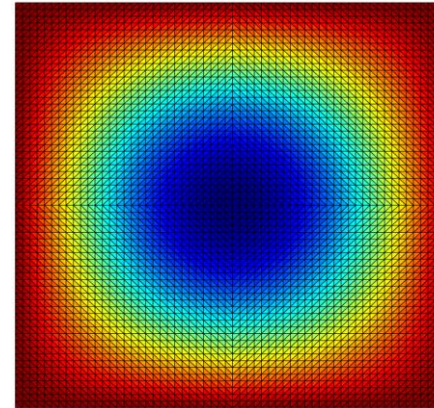
Mesh size = 60 x 60

Material parameters:

$$E = 70 \text{ GPa}$$

$$\rho = 2778 \text{ kg/m}^3$$

Pre-processing: Linear modes analysis



Discussion of results

Test case 1: Rectangular plate, simply supported (M. Amabili, 2004)

Comparison to full FE solution in time domain simulation

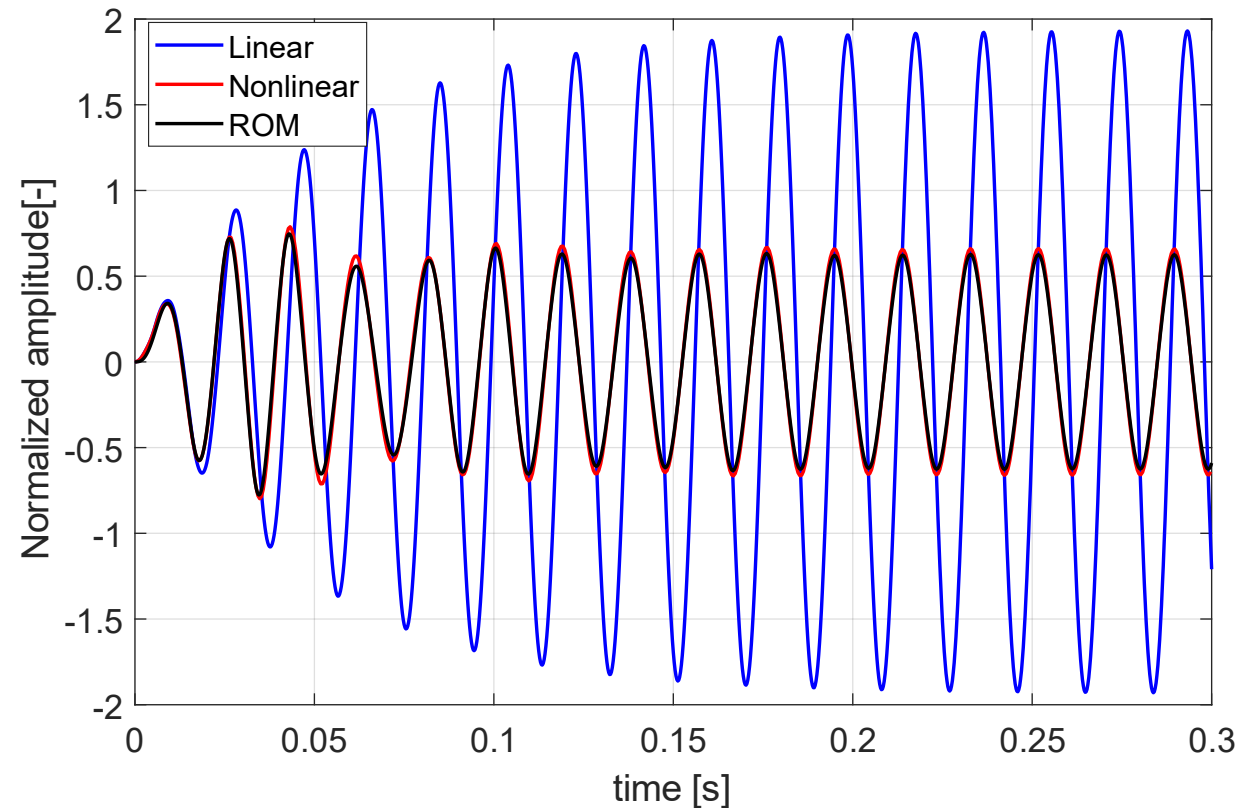
Excitation at $\sim 0.997 \cdot \omega_1$

Full FE solution time = 274.6 sec

ROM solution time = 3.9 sec

ROM pre-processing = 8.4 sec
includes formulation of ROM parameters,
stiffness tensors and modal eigenvalue
analysis

Total code run-time = 12.3 sec



Discussion of results

Test case 1: Rectangular plate, simply supported (M. Amabili, 2004)

1-DOF model

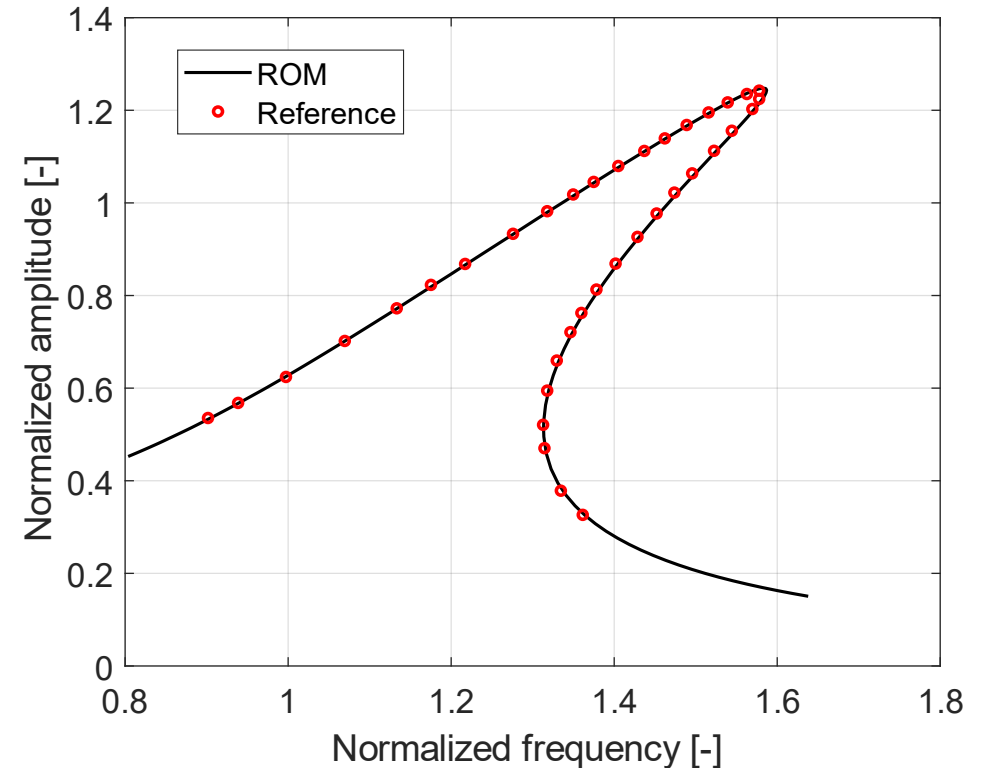
Analysis using the software AUTO (Doedel, 2007)

Linear modal analysis (pre-processing) = 0.65 sec

ROM formulation time = 0.99 sec

AUTO analysis = 5.59 sec (1560 data points along the solution curve)

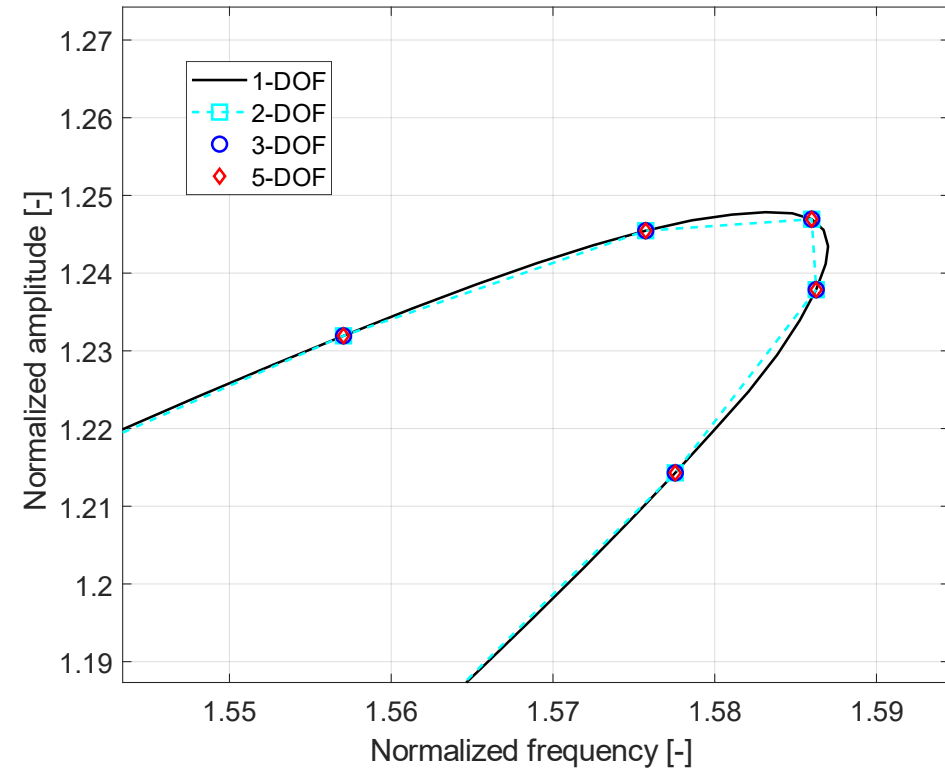
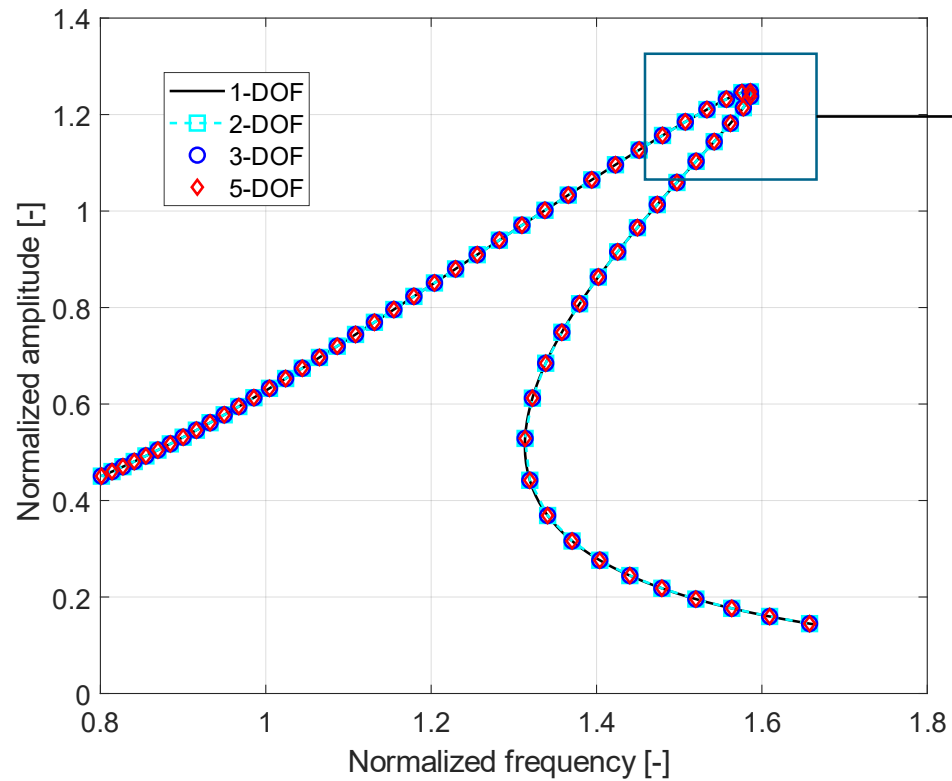
Difference from reference solution (Amabili, 2004)
= 0.43 %



Discussion of results

Test case 1: Rectangular plate, simply supported (M. Amabili, 2004)

Convergence study - increase the number of modes in the reduction subspace.



Discussion of results

Test case 2: Stiffened plate with free boundary conditions (Sinha et al, 2020).

Analysis parameters:

$l = 0.5 \text{ m}$, $b = 0.4 \text{ m}$, $t = 0.002 \text{ m}$

$l_s = 0.4 \text{ m}$, $b_s = 0.008 \text{ m}$, $t_s = 0.005 \text{ m}$ (stiffener)

Damping ratio $\zeta = 0.0012$ (initial guess)

Applied force $f_{ext} = 0.2 - 1 \text{ N}$

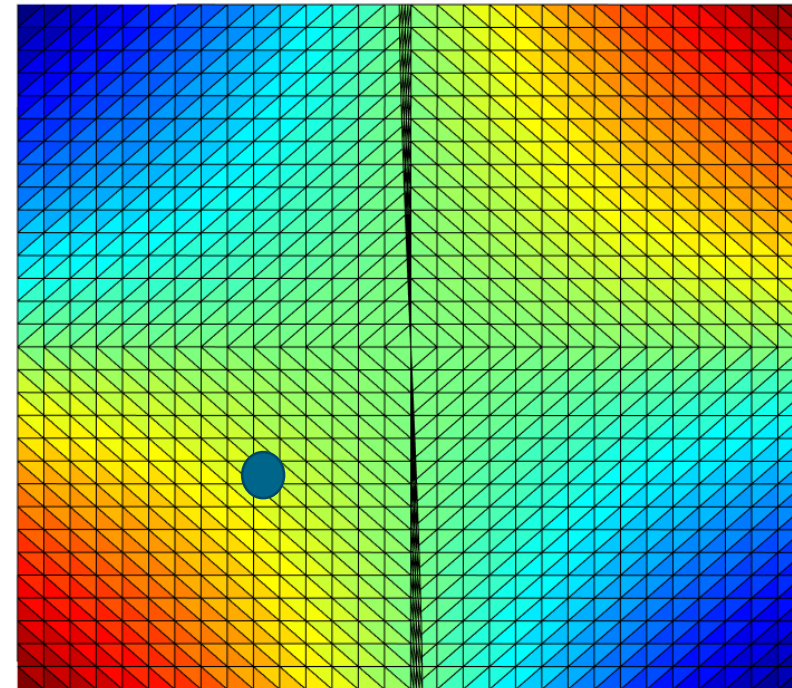
Excitation frequency = 32 – 40 Hz (sweep)

Material parameters:

$E = 70 \text{ GPa}$

$\rho = 2660 \text{ kg/m}^3$

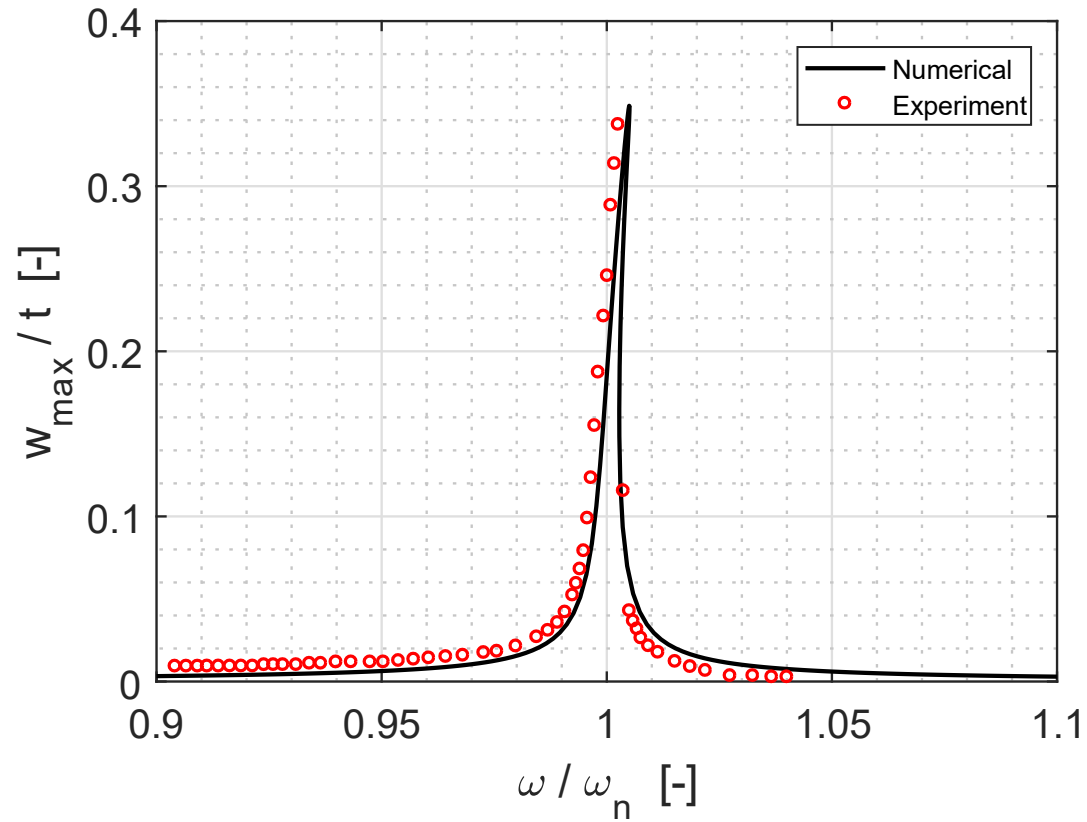
1st elastic mode at 36.78 Hz



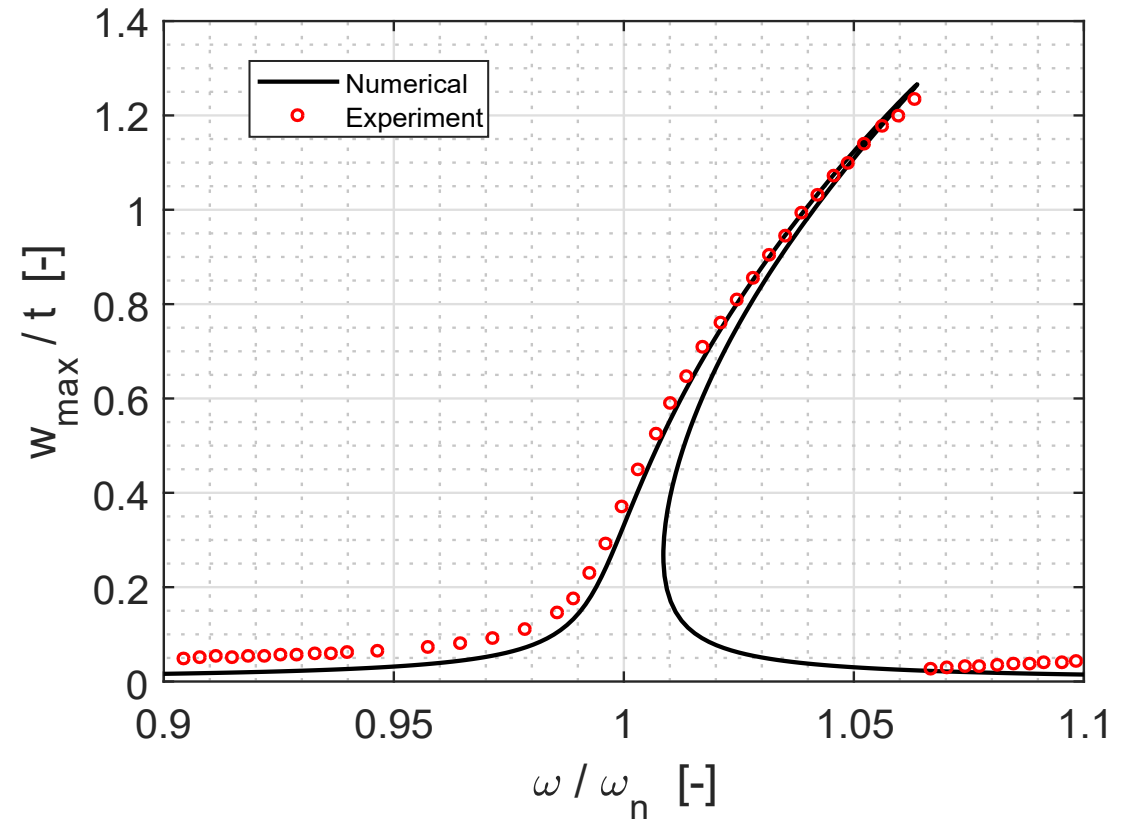
Excitation at $(x, y) = (0.2, 0.16 \text{ m})$

Discussion of results

Test case 2: Stiffened plate with free boundary conditions (Sinha et al, 2020).



0.2 N



1.2 N

Discussion of results

Test case 3: Ongoing studies, cantilever beam (Pany and Rao, 2002)

Analysis parameters:

$l = 0.693166$ m

$t = 0.001$ m

Damping ratio $\zeta = 0.0467$

Applied force $f_{ext} = 0.2$ N

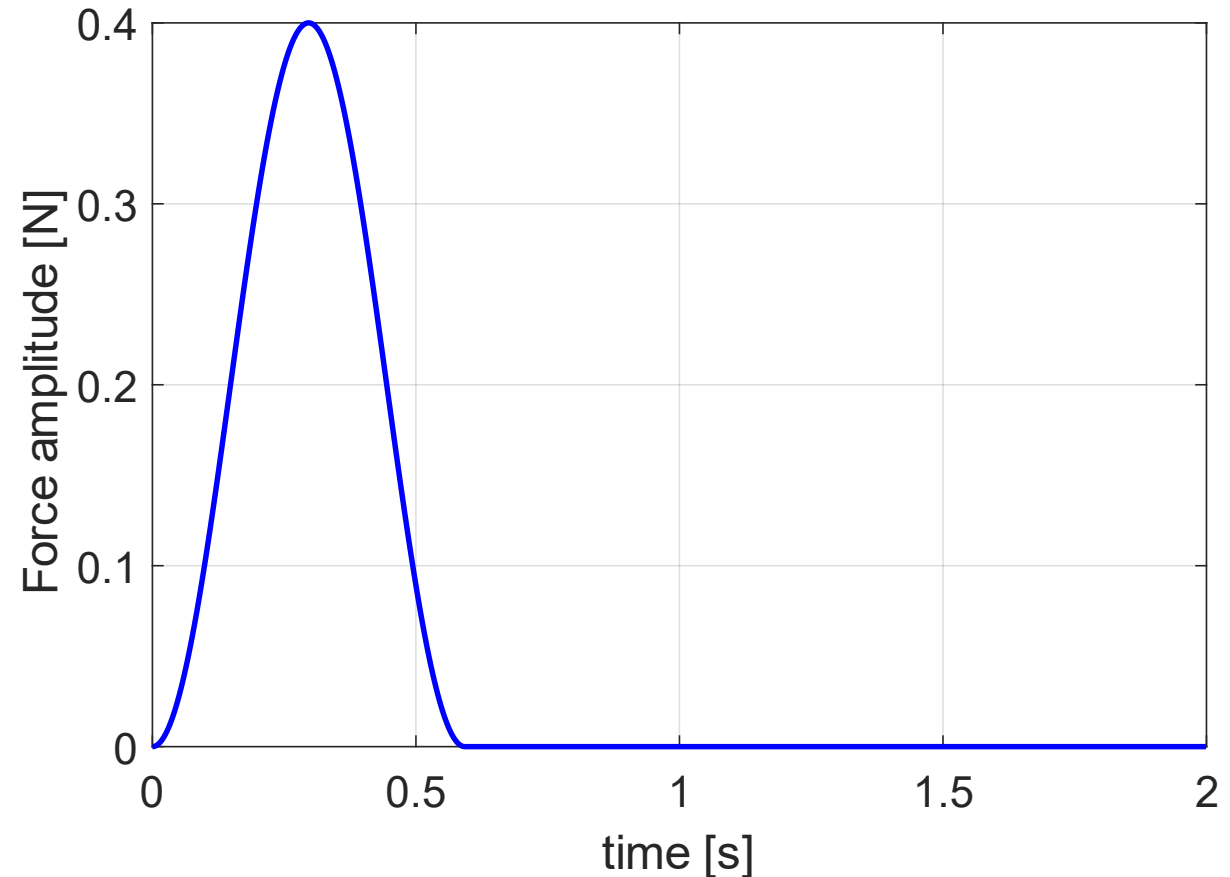
Excitation frequency = 10.6 rad/s

Material parameters:

$E = 200$ GPa

$\rho = 7800$ kg/m³

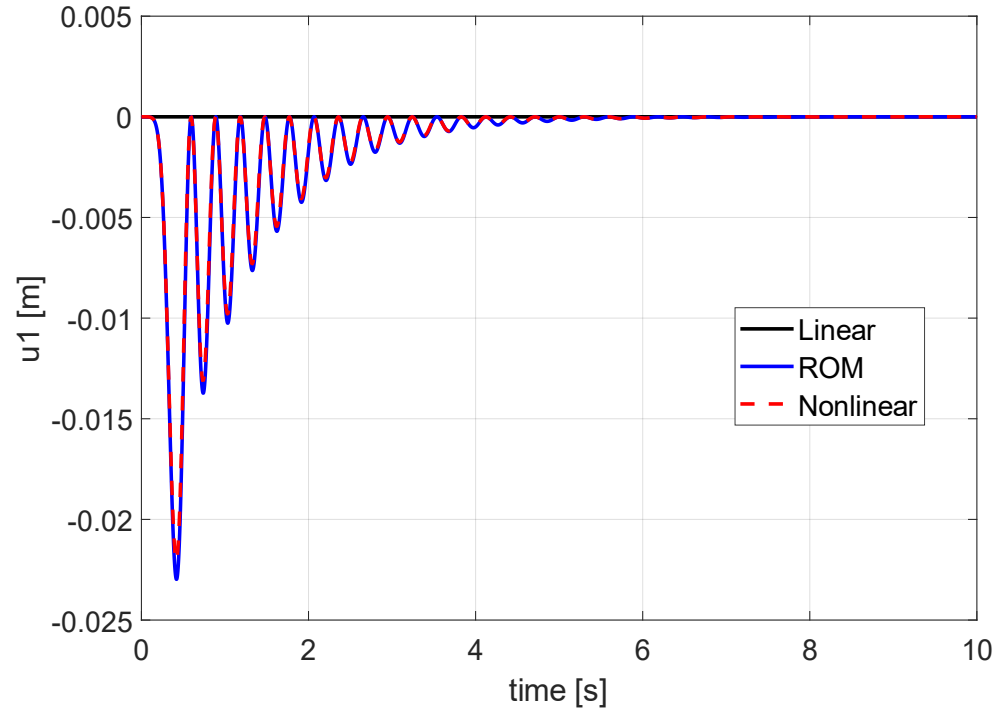
1-cosine profile



Discussion of results

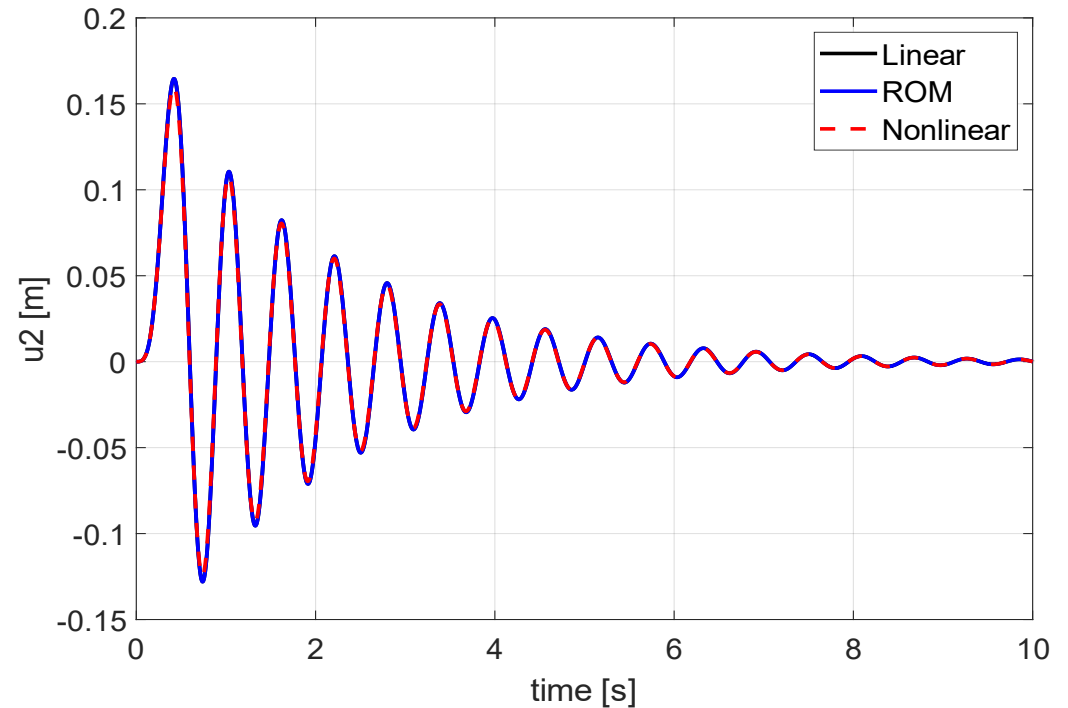
Test case 3: Ongoing studies, cantilever beam (Pany and Rao, 2002)

In-plane (IP) displacement



3.2 % (of length) maximum IP deflection

Out-of-plane (OOP) displacement



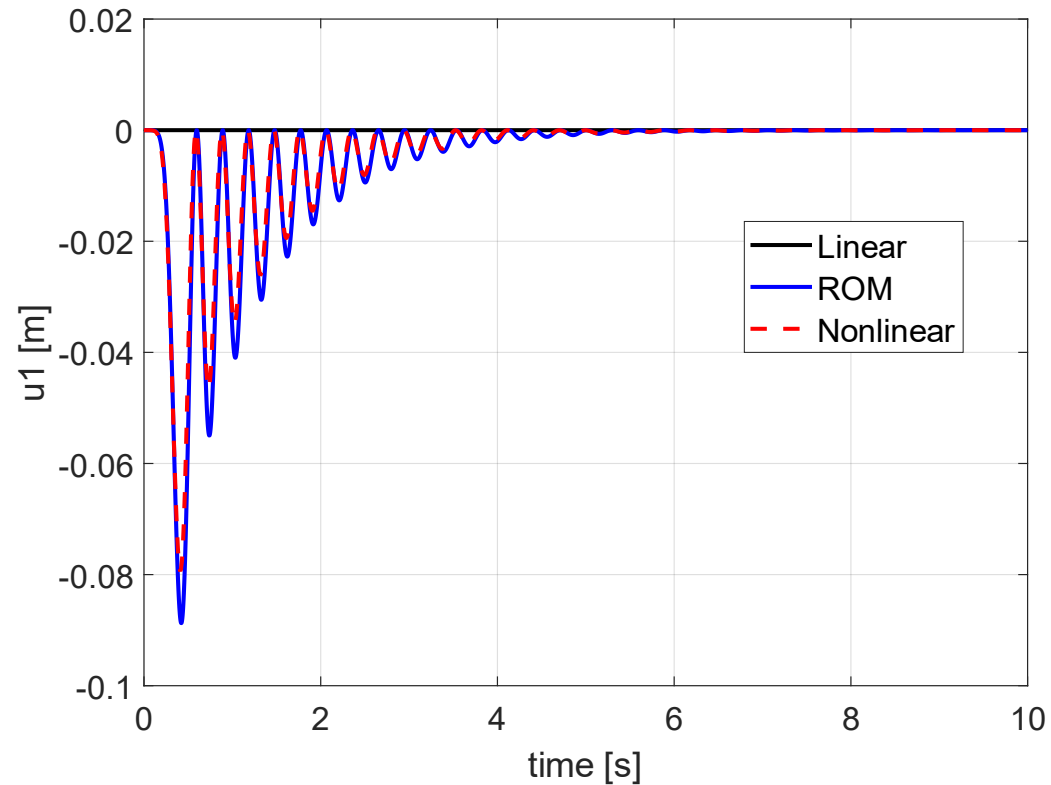
23.5 % (of length) maximum OOP deflection

Simulation time (4-DOF ROM) = 2.4 sec, simulation time (full Nonlinear) = 223 sec

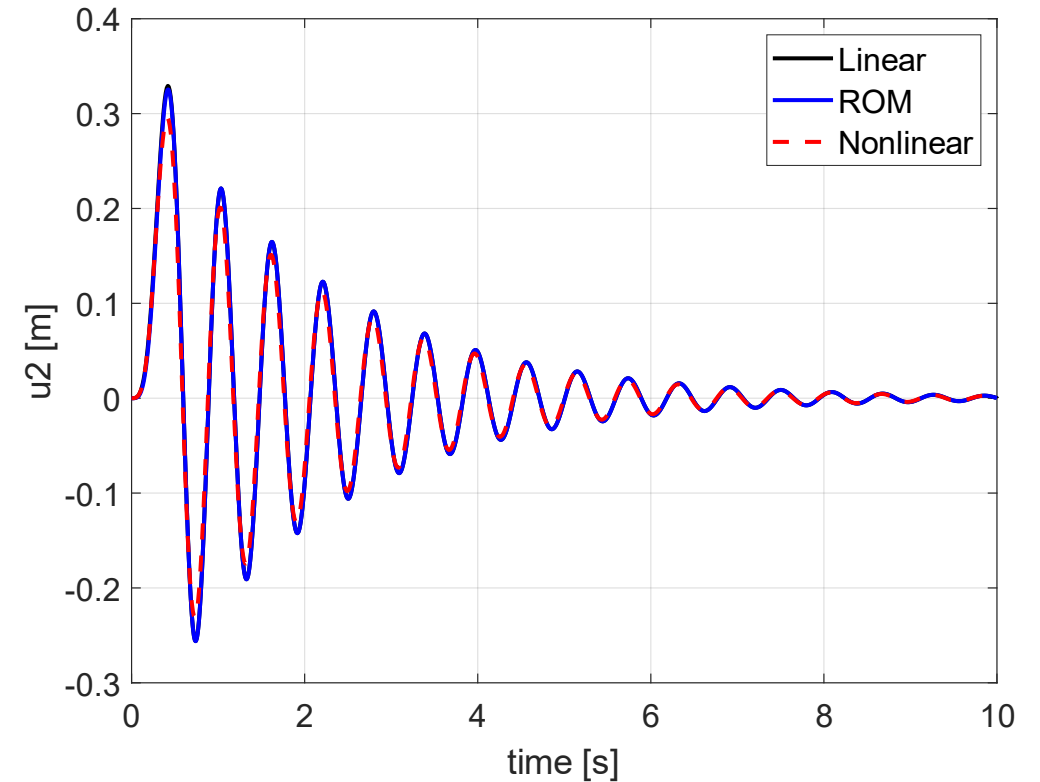
Discussion of results

Test case 3: Ongoing studies, cantilever beam (Pany and Rao, 2002)

Double the force amplitude

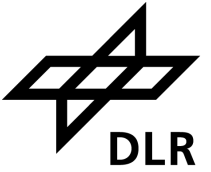


11.5 % (of length) maximum IP deflection



42.7 % (of length) maximum OOP deflection

Outlook and Conclusions



- ROM works well for various boundary conditions.
- Experiments show us a need of nonlinear damping model.
- Limited region of ROM validity.
- Ongoing studies aim to extend the limit of validity for larger deflections, specially in cantilevers.
- Intended application towards large scale model reduction of generic FE models.

Thank you for your attention !

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- Orthogonality conditions derived from constraint equations:

$$\begin{aligned}f'_{\alpha} u_{\beta} &= \delta_{\alpha\beta} \\f'_{\alpha} u_{\beta\gamma} &= 0\end{aligned}$$

- Deriving the reduced force from conditions of work equivalence:

$$(\mathbf{F}\boldsymbol{\phi})' \cdot \delta\mathbf{u} = \boldsymbol{\phi}' \cdot \delta\xi$$

Substitute for u:

$$\mathbf{u}(\xi) = \mathbf{u}_{\alpha}\xi_{\alpha} + \mathbf{u}_{\alpha\beta}\xi_{\alpha}\xi_{\beta}$$

With use of the orthogonality constraints we get, $\boldsymbol{\phi} = \mathbf{f}_{\text{ext}} \mathbf{u}_{\alpha}$

- Dissipation energy

$$E_d = \frac{1}{2} \dot{\mathbf{u}}' \mathbf{D} \dot{\mathbf{u}} = \frac{1}{2} (\mathbf{M}^{-1} \mathbf{P} \boldsymbol{\pi})' \mathbf{D} (\mathbf{M}^{-1} \mathbf{P} \boldsymbol{\pi})$$

$$E_d = \frac{1}{2} \dot{\boldsymbol{\xi}} (\bar{\mathbf{M}} \mathbf{P}' \mathbf{M}^{-1} \mathbf{D} \mathbf{M}^{-1} \mathbf{P} \bar{\mathbf{M}}) \dot{\boldsymbol{\xi}}$$

- von Karman Strain, beam element

$$\epsilon = u_{,x} + \frac{1}{2} (u_{,x}^2 + w_{,x}^2)$$

$$\chi = w_{,xx}$$

Discussion of results

Test case 3: Ongoing studies, cantilever beam (Pany and Rao, 2002)

Analysis parameters:

$l = 0.693166 \text{ m}$

$t = 0.001 \text{ m}$

Damping ratio $\zeta = 0.0467$

Applied force $f_{ext} = 0.2 \text{ N}$

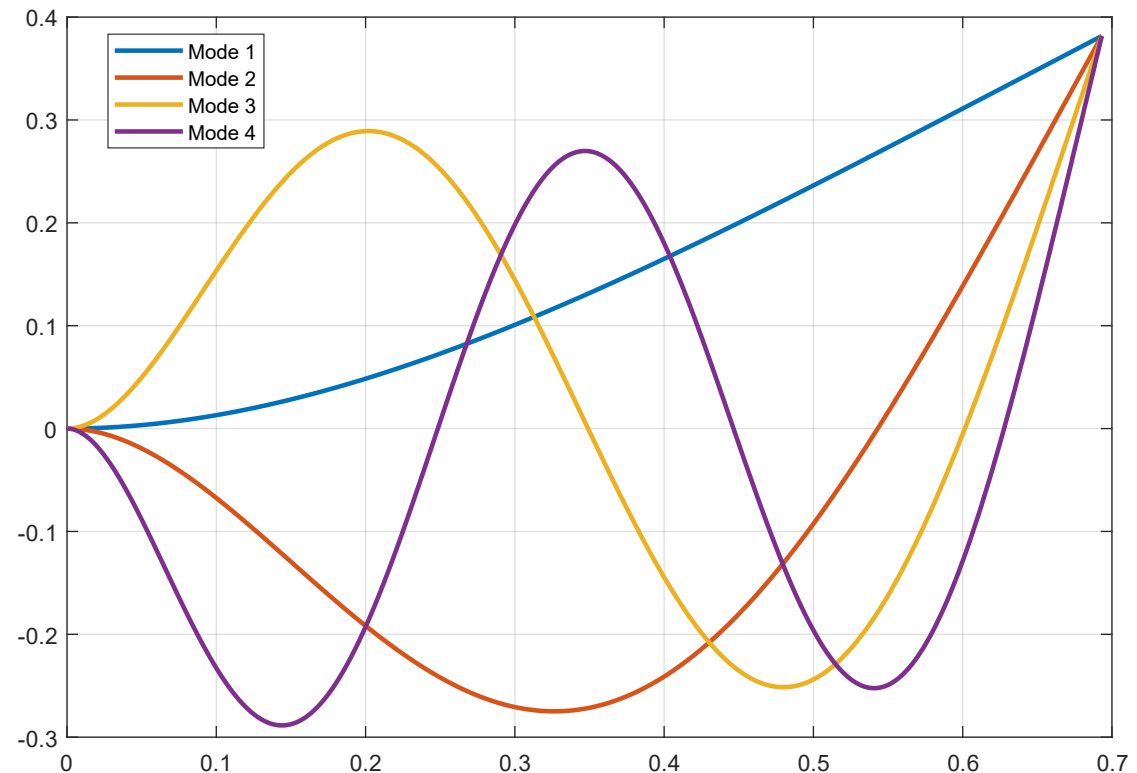
Excitation frequency = 10.6 rad/s

Material parameters:

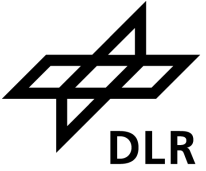
$E = 200 \text{ GPa}$

$\rho = 7800 \text{ kg/m}^3$

Pre-processing: Linear modes analysis



Discussion of results



Test case 3: Ongoing studies, cantilever beam (Pany and Rao, 2002)

Convergence analysis

Number of modes	u_1 [m]	u_2 [m]	Simulation time [sec]
1	-0.0222	0.1627	0.83
2	-0.0229	0.1643	1.65
3	-0.0229	0.1645	2.26
4	-0.0230	0.1645	2.40
5	-0.0230	0.1646	2.74
8	-0.0230	0.1646	4.68
10	-0.0230	0.1646	6.58