

# Innovative Desalination System Driven by a Solar Micro Gas Turbine for Off-Grid Applications

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## Abstract

Past work by the authors has suggested that Solar micro Gas Turbines (SmGTs) can be used cost-effectively to produce electric power and heat for freshwater production through desalination, mainly in off-grid locations. This is further studied in this work, presenting a detailed description of system performance at design and part-load conditions, as well as the characteristics of the components of the SmGT and the desalination unit. To this end, the SmGT is assessed first, considering techniques that achieve a greater off-design performance such as incorporating Variable Inlet Guide Vanes (VIGVs) at the compressor inlet, and including the sensitivity to control strategies and ambient conditions, exploring their expectedly very negative impact on the SmGT performance. Water treatment system is comprised of two elements. A Reverse Osmosis desalination unit is driven by the electric power produced by the SmGT. This produces brine with high salt concentration to be partially treated further in a Zero Liquid Discharge (ZLD) unit driven by the exhaust gases of the microturbine (at about 250-300°C), where the sensible heat of this stream is harvested by the ZLD unit to “dry” and concentrate the effluent. Finally, the potential and the operational limitations of the ZLD system are discussed, supplemented by an experimental proof of concept where its feasibility was verified.

*Keywords:* SOLMIDIFF, Solar Micro Gas Turbine, Microturbine, Desalination, ZLD

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## 1 Nomenclature

2	$\Delta_z$	Rotor Axial Length	[m]
3	$\dot{C}_i^*$	Ratio of Heat Capacities	[-]
4	$\dot{m}$	Mass Flow Rate	[kg/s]
5	$\eta$	Efficiency	[-]
6	$\nu$	Kinematic Viscosity	[m <sup>2</sup> /s]
7	$\Omega$	Total Pressure Loss Coefficient	[-]
8	$\omega_\lambda$	Abrupt Expansion Losses	[-]
9	$\omega_{BL}$	Blades Losses	[-]
10	$\omega_{CH}$	Aerodynamic Blockage Losses	[-]
11	$\omega_{cl}$	Clearance Losses	[-]
12	$\omega_{cr}$	Supercritical Mach Number Losses	[-]

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13	$\omega_{dif}$	Entrance Diffusion Losses	[-]
14	$\omega_{HS}$	Hub-to-Shroud Loading Losses	[-]
15	$\omega_{inc}$	Incidence Losses	[-]
16	$\omega_{mix}$	Mixing Losses	[-]
17	$\omega_{sf}$	Skin Friction Losses	[-]
18	$\phi$	Flow Coefficient	[-]
19	$\Pi$	Osmotic Pressure	[bar]
20	$\psi$	Work Coefficient	[-]
21	$\varepsilon$	Effectiveness	[-]
22	$b_{in}$	Rotor inlet passage width	[m]
23	$C_d$	Discharge Coefficient	[-]
24	$d$	Solar Receiver Nozzle Diameter	[m]
25	$D_{cav}$	Cavity Diameter	[m]
26	$F$	Feed	[-]
27	$I$	Work Input Coefficient	[-]
28	$I_{bl}$	Blade Loading Losses	[-]
29	$I_{df}$	Disk Friction Losses	[-]
30	$I_{leaf}$	Leakage Losses	[-]
31	$I_{recir}$	Recirculation Losses	[-]
32	$j$	Solar Receiver Nozzle rows	[-]
33	$k_{wf}$	Thermal Conductivity of the Working Fluid	[W/(m · K)]
34	$m_{bi}$	Brine Inlet Mass Flow Rate	[kg/s]
35	$m_{bo}$	Brine Outlet Mass Flow Rate	[kg/s]
36	$m_{gi}$	Gases Inlet Mass Flow Rate	[kg/s]
37	$N$	Total Number of sub-Heat Exchangers	[-]
38	$n$	Total number of Solar Receiver nozzles	[-]
39	$n_s$	Specific Speed	[-]
40	$P$	Permeate	[-]
41	$p_t$	Total Pressure	[-]
42	$P_{jet}$	Jet Pitch	[m]
43	$q_v$	Volumetric Flow Rate	[m <sup>3</sup> /s]
44	$r_V$	Recovery Rate	[-]

45	$S$	Salinity	[kg/kg]
46	$S_{bi}$	Brine Inlet Salinity	[kg/kg]
47	$S_{bo}$	Brine Outlet Salinity	[kg/kg]
48	$T_{bi}$	Brine Inlet Temperature	[K]
49	$T_{bo}$	Brine Outlet Temperature	[K]
50	$T_{gi}$	Gases Inlet Temperature	[K]
51	$U$	Global Heat Exchange Coefficient	[W/(m <sup>2</sup> · K)]
52	$v_s$	Total-to-Total Velocity Ratio	[-]
53	$Y_{BL}$	Blade Loading Losses	[-]
54	$Y_{cl}$	Clearance Losses	[-]
55	$Y_{HS}$	Hub-to-Shroud Losses	[-]
56	$Y_{inc}$	Incidence Losses	[-]
57	$Y_p$	Profile Losses	[-]
58	$Y_Q$	Moisture Losses	[-]
59	$Y_{rot}$	Rotor Loss Coefficient	[-]
60	BB	Billion Barrels	[-]
61	BCF	Billion Cubic Feet	[-]
62	BD	BlowDown	[-]
63	BP	Booster Pump	[-]
64	CHP	Combined Heat and Power	[-]
65	CSP	Concentrated Solar Power	[-]
66	DNI	Direct Normal Irradiance	[W/m <sup>2</sup> ]
67	ERD	Energy Recovery Device	[-]
68	GDP	Gross domestic product	[-]
69	HDH	Humidification-DeHumidification	[-]
70	HPP	High Pressure Pump	[-]
71	ICE	Internal Combustion Engine	[-]
72	ICH	Isobaric Chamber	[-]
73	MD	Membrane Distillation	[-]
74	MED	Multi-Effect Distillation	[-]
75	mGT	Micro Gas Turbine	[-]
76	MSF	Multi-Stage Flash	[-]

77	ORC	Organic Rankine Cycle	[-]
78	PR	Pressure Ratio, Performance Ratio	[-]
79	Pr	Prandtl Number	[-]
80	PSHE	Primary Surface Heat Exchanger	[-]
81	PTC	Parabolic-Trough Collector	[-]
82	PV	PhotoVoltaic	[-]
83	RO	Reverse Osmosis	[-]
84	SmGT	Solar Micro Gas Turbine	[-]
85	SSC	Stationary Solar Collector	[-]
86	SWRO	SeaWater Reverse Osmosis	[-]
87	TDS	Total Dissolved Solids	[ppm]
88	TIT	Turbine Inlet Temperature	[K]
89	TOT	Turbine Outlet Temperature	[K]
90	TVC	Thermal Vapour Compression	[-]
91	UA	Conductance	[W/K]
92	VIGV	Variable Inlet Guide Vanes	[-]
93	VNGV	Variable Nozzle Guide Vanes	[-]
94	ZLD	Zero Liquid Discharge	[-]

## 95 1. Introduction

### 96 1.1. Background

97 This paper presents the technical details of an innovative integrated system aimed at enabling a large reduction  
 98 of the carbon footprint of water desalination in areas currently subjected to extreme water scarcity, in particular the  
 99 Middle East and Northern Africa. In these regions, the characteristic boundary conditions include a very high available  
 100 solar resource and also large regions where the grid is not available. Therefore, the conditions to integrate solar power  
 101 generation and water desalination are met, in an attempt to tackle not only water scarcity but also a heavy reliance on  
 102 fossil fuel for electricity supply.

103 The countries in the Middle East have almost 50% of the world reserves of crude oil and they can be categorised  
 104 in two main groups. On the one hand, certain countries in the Middle-East with very high state and individual  
 105 wealth (gross GDP and GDP per capita), mostly affiliated with the Gulf Cooperation Council (Saudi Arabia, Kuwait,  
 106 Qatar, United Arab Emirates and, to a lesser extent, Oman and Bahrain) and accounting for over 30% of the world  
 107 reserves of crude oil; these countries have seen a remarkable economic growth, as a consequence of which the energy  
 108 consumption per capita has rocketed not only for the increased demand but also for certain policies discouraging an  
 109 efficient use of energy (for instance, through subsidised energy prices). On the other hand, certain countries also in  
 110 the Middle East (Iran, Iraq) and in Northern Africa (Libya, Algeria) which have not seen a similar improvement of the  
 111 individual wealth despite their large share of oil reserves (15-20%). When natural gas and not oil is considered, the  
 112 situation is somewhat similar even if the numbers add differently. Overall, regardless of the different characteristics  
 113 of these two groups of countries, they share a strong dependence on fossil fuels to cover the power needs (>95% of  
 114 the electricity comes from natural gas or coal, 100% in several cases) [1, 2]. All this information is shown in Table 1.

Country	Consumption of Electricity [kWh/person-year]	Fossil Fuels Energy mix [%]	Water Availability [m <sup>3</sup> /person-year]	Oil Reserves		GDP per capita [USD <sub>2010</sub> ]	Natural Gas Reserves	
				[BB]	[%]		[BCF]	[%]
Algeria	1363	99	272.3	12.2	0.69	4710.6	159054	2.3
Bahrain	19597	99.7	2.679	0.09	0.01	20913	3250	0.05
Egypt	1683	90	10.25	3.3	0.19	3008.8	77200	1.1
Iran	3022	92.7	1583	155.6	8.74	5627	1201382	17.3
Iraq	1328	95.5	919.7	145.02	8.15	4837.4	111522	1.6
Kuwait	15591	100	0	101.5	5.70	32697.3	63500	0.9
Libya	1811	100	109.8	48.36	2.72	8122.2	58183	0.8
Morocco	904	84.6	811.4	0	0.00	3396.1	51	0
Oman	6446	100	302	5.37	0.30	14992.6	24910	0.4
Qatar	14792	100	21.22	25.24	1.42	62021.1	871585	12.5
Saud Arabia	9401	100	72.86	267.03	15.00	20542.2	294205	4.2
UAE	11088	100	15.96	97.8	5.50	41420.5	215098	3.1
Yemen	220	98	74.34	3	0.17	0	16900	0.2

Table 1: Energy, economic and water related indicators of selected countries in the MENA region [1, 2]. Oil and Natural Gas reserves are given in billion barrels [BB] and billion cubic feet [BCF] (2019 data).

The aforescribed abundance of fossil fuels is in contrast with a low availability of water resources, in most cases well below the water poverty or absolute water poverty lines, 1000 and 500 m<sup>3</sup>/(person· year) respectively, below which water becomes a primary constraint for human life. In some extreme cases, water availability is even below 100 m<sup>3</sup>/(person· year), a limit that poses a very large and pressing threat for the sustainability of future generations. This is also shown in Table 1.

In the light of the situation above, it is now agreed that this environmental emergency must be tackled simultaneously for water and energy, exploiting the evident synergies between these two human needs and the abundant solar resource in the region. The current landscape for large-scale water technologies in this part of the world is characterised by large-capacity thermal desalination plants using technologies based on Multi-Effect Distillation (MED) or Multi-Stage Flash (MSF) integrated into thermal power plants, installed in the 20<sup>th</sup> century for the most part but already obsolete in present times as a result of the superior performance of Reverse Osmosis (RO). At the smaller-scale and for off-grid installations, RO powered by photovoltaic panels seems to be the preferred option despite its disadvantages, since these systems typically require: i) backup generators burning fossil fuels to enable operation in the absence of solar energy, and ii) batteries to run the system overnight if needed. Such redundancy is, of course, less reliable and has higher capital and O&M costs than a single, flexible system.

This work introduces a technical solution which can possible tackle the flaws of PV-RO systems mentioned in the foregoing paragraph, producing freshwater with a single system able to work on solar energy and on backup fossil fuels whilst also reducing the liquid effluent from the water production process (i.e., lower environmental impact). This system is based on the coupling of a solar dish collector and a micro gas turbine, downstream of which water is produced in an RO unit driven by the electric power produced by the gas turbine system. A Zero Liquid Discharge (ZLD) system running on the available thermal energy of the exhaust of the gas turbine evaporates water from the concentrate, hence ensuring lower water content in the latter and, accordingly, lower environmental impact. Although no commercial systems have been installed so far, the techno-economic viability of Solar micro Gas Turbines (SmGT) has been addressed in literature, finding better-than-expected results in all aspects [3–7] and with the experimental support of a demonstrator installed at ENEA Research Centre (Italy) within the framework of the OMSoP project [8], funded by the European Commission. This system demonstrated the technical feasibility of the coupling of a solar dish collector and a micro gas turbine.

## 1.2. Scope of work and objectives

Micro Gas Turbines (mGT) are compact engines for on-site power and heat generation able to achieve electrical efficiencies between 15% (small engines of ~3 kWe [9]) and 40% (larger engines with unit output of ~400 kWe [10]) at full load and an overall energy utilisation of more than 90% in Combined Heat and Power (CHP) configuration [9]. A mGT includes the following components: compressor, turbine/expander, combustor, recuperator, fuel supply

147 system, bearings (most commonly with an associated lube oil system), electric generator, power conditioning and  
148 control unit, enclosure and balance-of-plant. Some of these components might not be present in a particular mGT,  
149 as this depends on the type of application such as the solar micro gas turbines where a solar receiver replaces the  
150 combustor.

151 Although market for decentralised, micro energy systems is currently dominated by reciprocating Internal Com-  
152 bustion Engines (ICE), micro gas turbines are reportedly competitive against ICEs in applications requiring interme-  
153 diate to high-grade heat and high heat-to-power ratio [11]. The micro gas turbines in these applications exhibit some  
154 variations with respect to large gas turbines, the utilisation of radial machinery being the most relevant feature. In  
155 micro gas turbines, single-stage radial compressors are standard and this limits the pressure ratio of the engine to  
156 values not higher than 4 (approximately). On the other hand, the utilisation of radial expanders prevents the use of  
157 blade cooling and this, in turn, limits Turbine Inlet Temperature (TIT) to values lower than 950°C. Both pressure ratio  
158 and turbine inlet temperature are significantly lower than in large gas turbines, what brings about a dramatic efficiency  
159 drop if simple Brayton cycles are used. Accordingly, micro gas turbines typically adopt recuperative cycle layouts in  
160 order to increase thermal efficiency to more competitive values [12].

161 Further to the aforementioned potential of micro gas turbines for decentralised power, in particular applied to renewable  
162 energy sources, and based on the past experience of the authors, a solar desalination system based on SmGTs with  
163 the techno-economic features already presented in [13–15] seems to be a cost-effective solution with a promising  
164 future against other popular renewable desalination options integrating PV panels. This has already been explored  
165 for isolated units [16] but, according to recent literature on the topic, it might also apply to larger scales by coupling  
166 several SmGTs in parallel.

167 The main features of the solar desalination arrangement proposed are:

- 168 • The high temperature available at the turbine exhaust has the potential to drive thermal desalination processes  
169 based on phase change. In this case, the permeate of an upstream RO process could be blended with the distillate  
170 produced by distillation in order to increase the quality of the fresh water produced. Another option would be  
171 to direct the concentrate of the RO unit to a thermal ZLD system were the total water content of the effluent  
172 could be reduced, thereby minimising the environmental impact of water desalination.
- 173 • Biofuels can be used as a back-up energy source to maintain the system operating at full capacity 24/7 and to  
174 compensate for cloudy/hazy sky conditions. This avoids the usual oversizing of desalination systems typically  
175 required in solar applications in order to achieve reasonable capacity factors. Moreover, this oversizing brings  
176 about higher costs that challenge the economics of the systems and stem as the main economic hurdle for the  
177 development of solar desalination technologies.
- 178 • In parabolic dish systems, the integrated concentrator-receiver system can reach efficiencies as high as 80%.  
179 Then, the complete assembly with a mGT can potentially attain annual solar-to-electric efficiencies of 18.3%,  
180 when turbine inlet temperature is in the order of 850°C. [17]. This is very close to the rated efficiency of approx  
181 20%, or even more if a higher turbine inlet temperature is considered [5], which is similar to the efficiency  
182 of conventional, larger CSP plants. This ensures interesting market prospects of the technology when fully  
183 developed [7].

184 In order to justify the interest of this work, Fig. 1 presents a comparison of the expected efficiency achieved  
185 by different solar desalination technologies in terms of estimated solar energy required [kWh] to produce 1 m<sup>3</sup> of  
186 freshwater. The following technologies are considered:

- 187 • Five collector technologies: i) Stationary Solar Collectors (SSC), ii) Parabolic Trough Collectors (PTC), iii)  
188 heliostats, iv) Dish collectors and v) PV panels.
- 189 • Two thermo-mechanical energy conversion technologies: Organic Rankine Cycle systems (ORC) and micro  
190 Gas Turbines (mGT).
- 191 • Four desalination technologies based on phase-change: i) Membrane Distillation (MD), ii) Multi-Effect Distilla-  
192 tion (MED), iii) MED with Thermal Vapour Compression (MED-TVC) and iv) Humidification-DeHumidification  
193 (HDH) with values of Performance Ratios (PR) taken from recent bibliography on the topic [18, 19].
- 194 • Desalination based on membrane technology: SWRO desalination with a specific energy consumption of 2.2  
195 kWh/m<sup>3</sup> (as reported in recent literature on the topic [20]).

196 From the comparison in Fig. 1, the interest of the proposed solution in terms of specific energy consumption  
 197 becomes evident. Therefore, a more detailed, in depth analysis is needed to verify if the performance reported in the  
 198 char is actually accurate.

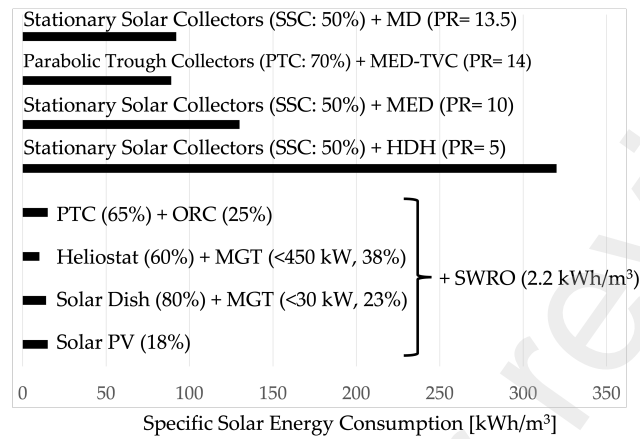


Figure 1: Comparison of solar desalination based on reverse osmosis and phase change desalination. Estimates of energy efficiency of solar energy technologies are given in brackets along with that of power cycles [20].

## 199 2. System Description and Methodology

200 The system integrating the solar mGT and water treatment systems proposed, namely reverse osmosis desalina-  
 201 tion and ZLD, has been developed conceptually in the SOLMIDEFF project. This project, entitled *SOLar Micro gas*  
 202 *turbine-driven Desalination for Environmental oFF-grid applications*, was developed by the research team at Univer-  
 203 sity of Seville, Spain, with funding from the National R&D programme of the Ministry of Science and Innovation.  
 204 The reference site for the techno-economic assessment of the technology is in the Canary Islands.

205 The SOLMIDEFF system is based on a 10 kWe SmGT producing electric power and heat to run a desalination unit  
 206 in off-grid applications. The desalination unit is comprised of a SWRO unit, driven by the electric power produced by  
 207 the engine, and the Zero Liquid Discharge making use of the available gas in the exhaust from the SmGT. A schematic  
 208 of the SOLMIDEFF system is presented in Fig. 2, where the three main subsystems (SmGT, SWRO and ZLD) are  
 209 easily identified. A parabolic dish collects and concentrates solar energy onto a solar receiver where a large fraction of  
 210 this concentrated solar power is absorbed by the working fluid of an open-cycle recuperated mGT, achieving a solar-  
 211 to-electric efficiency of almost 20%. In this micro gas turbine, atmospheric air is compressed to 3.5 bar by a radial  
 212 compressor and then heated up to 500°C by a recuperative heat exchanger. The energy source for this temperature  
 213 rise is the stream of hot gases flowing out from the turbine. After pressurisation and preheating, air flows into a solar  
 214 receiver where it achieves a turbine inlet temperature of 850°C thanks to the concentrated solar energy absorbed. If  
 215 this were not possible, because of a lack of solar energy supply (hazy sky or, maybe, operation at night), a combustor  
 216 in series or parallel with the receiver would be able to burn fuel in order to ensure power and water supply. Regardless  
 217 of the energy source used to take the hot air to the target turbine inlet temperature, the hot pressurised gas is then  
 218 expanded across the turbine which, in turns, produces power to drive the compressor and electric generator.

219 The electric power produced by the SmGT is used by a Reverse Osmosis plant, whose feed stream is seawater,  
 220 brackish water or industrial wastewater. The particular arrangement of pumps and pressure exchange devices is shown  
 221 in Fig. 2. The feed stream is split into two flows. One of them is sent to the high-pressure pump (HPP) where pressure  
 222 is increased to 55 bar approximately. The remainder of the feed flow is sent to an isobaric chamber (ICH) where  
 223 pressure is also increased thanks to the high pressure available at the outlet concentrate stream of the RO unit. In  
 224 an energy recovery device (ERD) such as an ICH, volumetric flow rates on both sides are similar and the feed outlet  
 225 pressure (stream 14) is slightly lower than the concentrate inlet pressure (stream 18) [21]. This pressure loss comes  
 226 about because the ERD is not ideal and there are energy losses preventing a perfect match of the energy exchange

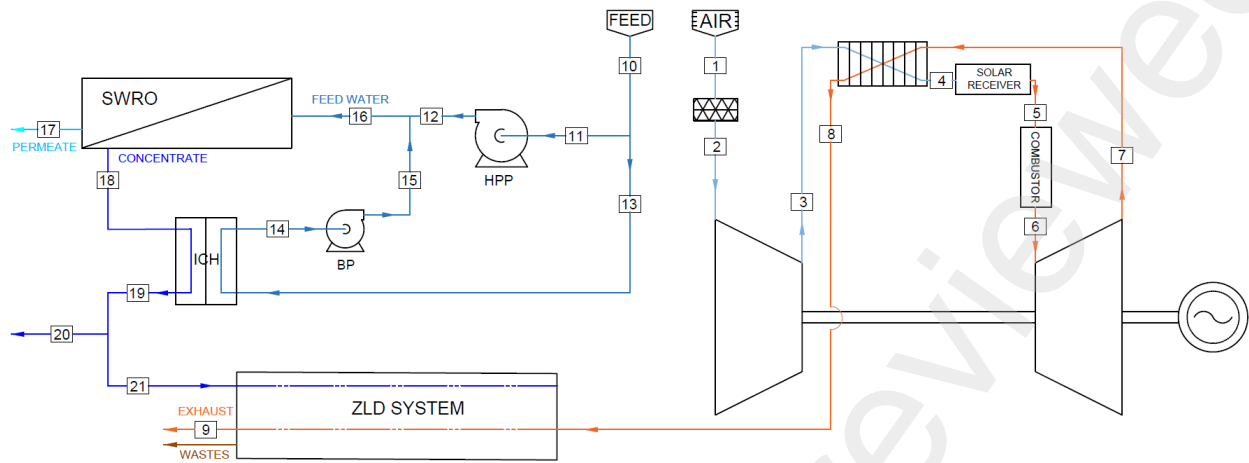


Figure 2: Schematic of the SmGT-driven desalination system (SOLMIDEFF system)

227 across the interacting streams; accordingly, a booster pump is needed downstream of the ERD (feed flow side) to  
 228 make up for the pressure difference between the outlet from the ERD and the outlet from the HPP (inlet to the RO  
 229 unit), as well as for the pressure drop through the membrane rack.

230 The concentrate flows across the energy recovery device, where pressure decreases to almost atmospheric. A  
 231 fraction of this stream is then sent to the ZLD unit, where part of the water content evaporates thanks to the thermal  
 232 energy available in the exhaust gases of the micro gas turbine. The ZLD unit is a direct-contact heat exchanger whose  
 233 purpose is to reduce the amount of water released for disposal [16].

234 The methodology followed to model the design and performance of the SOLMIDEFF system is based on an inte-  
 235 grated design and simulation platform which combines the design and off-design performance of each SOLMIDEFF  
 236 component. The individual models have been implemented in Matlab environment [22] and are solved through a mod-  
 237 ular approach which relies on the conservation of mass, energy and momentum and on set correlations to estimate  
 238 energy losses and irreversibility of the individual components.

239 The working fluid of the SmGT is air (dry real gas) whose thermodynamic properties are obtained from Coolprop  
 240 [23]. Once the simple recuperated Brayton cycle is resolved for a given set of boundary conditions (i.e., the pressure  
 241 ratio yielding maximum net efficiency is calculated), one dimensional (1-D) models are used to produce efficient  
 242 and flexible turbomachinery designs (compressor and turbine) whilst non-dimensional (0-D) approaches are used to  
 243 design the recuperator and solar receiver. The final step to completely model the SmGT is to evaluate the off-design  
 244 performance of the engine; this is driven by compressor-turbine matching, which is implemented by a proprietary  
 245 code making use of the methodology described by Razak [24]. In addition, several control strategies are embedded to  
 246 ensure the safe operation of the SmGT in different load ranges and operating conditions.

247 The performance of the RO system for the treatment of industrial effluents and brackish or seawater has been mod-  
 248 elled through well-known equations from literature [25], compiled in a block model calibrated against the software  
 249  $Q+$  developed by LG Chem [26]. finally, still within the water treatment subsystem, the proof-of-concept ZLD unit  
 250 has been experimentally demonstrated for this application, and the associated model has already been published by  
 251 the authors [16].

### 252 3. System Modelling and Performance

253 This section introduces the main features of the models used to characterise each main component of the SOLMID-  
 254 EFF system. This includes the designs models as well as the models used to estimate off-design performance. The  
 255 following subsystems are included: i) solar Micro Gas Turbine (compressor, solar subsystem, turbine/expander and  
 256 recuperator), ii) SWRO system, and iii) ZLD system. Comments on the validation or verification of each of these  
 257 models are also provided.



258 3.1. Solar Micro Gas Turbine

259 3.1.1. Compressor

260 The design and performance model of the centrifugal compressor is of the one-dimensional, mean-line type and  
 261 it is based on Aungier’s works [27]. The design process begins from the work ( $\psi$ ) and flow ( $\phi$ ) coefficients, target  
 262 pressure ratio and mass flow rate, along with the specific ambient conditions of the reference locations. This standard  
 263 approach, widely used in literature, is based on a stepwise methodology where the compressor elements are designed  
 264 independently through non-dimensional coefficients (pressure losses and work input) along the meridional coordinate:  
 265 impeller, diffuser and volute. Since the outlet conditions from the machine are required to estimate these coefficients,  
 266 this methodology renders an iterative process whose design flowcharts are shown in Figs. 3-4. In the latter flowchart,  
 267 blade loading ( $BL$ ) is defined by Eq. 1.

$$BL = \frac{2 \Delta W}{W_1 + W_2} \quad (1)$$

268 where  $\Delta W$  represents the average blade velocity difference, used to estimate the number of blades of the impeller,  
 269 and  $W_1$  and  $W_2$  is the relative flow velocity at impeller inlet and outlet.

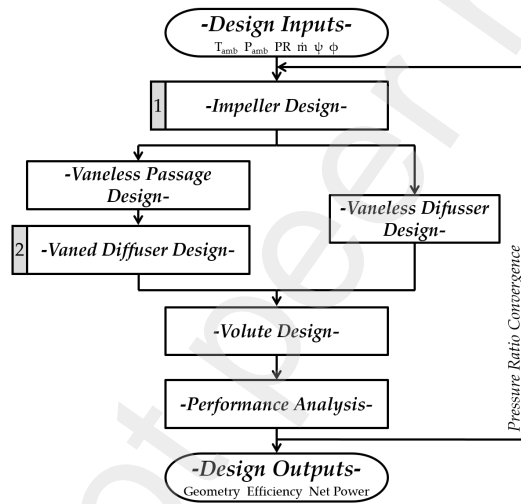


Figure 3: Compressor design flowchart.

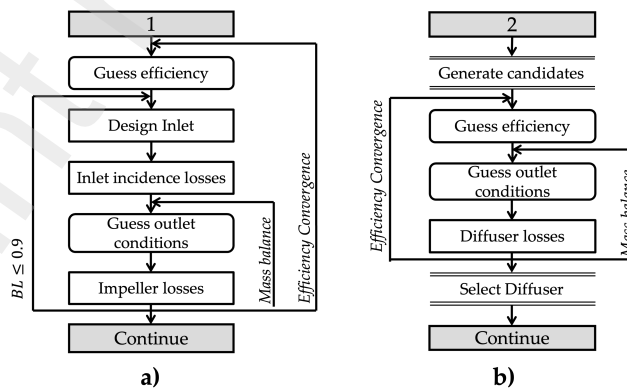


Figure 4: a) Impeller design flowchart and b) Vaned diffuser design flowchart.

270 Single stage compressors are considered by default, given the low pressure ratio of this type of engines (owing  
 271 to the recuperative cycle layout), most often with full-inducer impellers. The blades are three-dimensional, based on

272 straight-line surface elements that connect hub and shroud. The flow at the inlet is assumed axial with uniform flow  
 273 distribution, with the variable inlet guide vanes (VIGV) set in the axial direction at the design point<sup>1</sup>. The meridional  
 274 plane at the hub is constructed with the largest circular-arc possible (including linear extensions for the hub and shroud  
 275 contours if needed) by specifying the slopes and coordinates of the impeller eye and tip. The contour of the shroud is  
 276 defined on the other hand by a power-law linking the radial and axial coordinates.

277 The performance of the compressor is estimated from correlations modelling the internal flow physics and main  
 278 loss mechanisms. For both, variations and corrections with respect to the original work by Aungier [27] have been  
 279 introduced with the aim to obtain more accurate estimates. In particular: i) the slip factor is calculated from Wiesner  
 280 correlation [28]; ii) the pressure loss coefficient across the Variable Inlet Guide Vanes is obtained as described by  
 281 Japikse [29]; iii) leakage flows are calculated according to the work by Egli [30]; and iv) disk friction losses at the  
 282 endwalls are estimated as suggested by Daily and Nece [31]. Incorporating all these losses, the total pressure and  
 283 enthalpy at impeller outlet is determined by Eqs.(2-5):

$$P'_{t,2} = \frac{P'_{t,2,ideal}}{1 + \frac{P'_{t,1} - P_1}{P'_{t,1}} \cdot \Omega_{rot}} \quad (2)$$

$$h_{t,2} = h_{t,1} + I \cdot U_2^2 \quad (3)$$

$$\Omega_{rot} = \sum \omega_{rot} = \omega_{inc} + \omega_{dif} + \omega_{CH} + \omega_{sf} + \omega_{BL} + \omega_{HS} + \omega_{\lambda} + \omega_{mix} + \omega_{cl} + \omega_{cr} \quad (4)$$

$$I = I_{bl} + I_{df} + I_{leak} + I_{recir} \quad (5)$$

284 The total pressure loss coefficient of the impeller,  $\Omega_{rot}$ , accounts for incidence losses, entrance diffusion losses,  
 285 aerodynamic blockage, skin friction losses, blade and hub-to-shroud loading losses, abrupt expansion losses, mixing  
 286 losses, pressure losses given by blade-to-blade pressure difference (in open impellers), clearance losses and super-  
 287 critical Mach number losses. The work input coefficient  $I$  on the other hand includes loading losses, friction losses,  
 288 leakage losses and recirculation losses.

289 The diffuser model is based on total enthalpy conservation (i.e., the diffuser is considered to be adiabatic) and  
 290 pressure loss coefficients, following a similar approach to that of the impeller. In addition, the model considers  
 291 flow deviation and blockage factors and is applicable to either vaned or vaneless diffuser designs (accounting for the  
 292 vaneless space between impeller and diffuser). The vaneless diffuser is modelled through mass and angular momentum  
 293 conservation (accounting for friction and diffusion losses) whilst the geometry of the vaned diffuser is defined by the  
 294 area ratio, divergence angle and number of blades (varied in a predefined design space suggested by Aungier for  
 295 preliminary design). In this work, vaneless diffusers are preferred in order to obtain a wider operating range, even  
 296 though this implies a slightly lower rated (peak) efficiency; it is to note that this flexibility is essential for SmGTs [32].

297 A compressor map showing the effect of incorporating variable inlet guide vanes (VIGVs) is shown in Fig. 5.  
 298 For each shaft speed, two lines are overlaid. The solid constant-speed lines correspond to a compressor without  
 299 VIGVs while the dashed lines represent the same compressor with a row of VIGVs (whose stagger angle adapt to  
 300 the corresponding operating conditions to yield maximum efficiency). For the compressor incorporating VIGVs,  
 301 each constant-speed lines is comprised of three segments, corresponding to the different control strategies applied.  
 302 Starting from compressor choke (rightmost operating point), the following sections are identified when mass flow rate  
 303 is reduced.

- 304 • From compressor choke to peak efficiency, the constant speed line is virtually the same as for the compressor  
 305 without VIGVs.
- 306 • In a second section, in blue, pressure ratio is slightly lower than if no VIGVs were incorporated. Mass flow  
 307 rate is controlled through the positioning of VIGVs at different stagger angles in order to achieve peak impeller  
 308 efficiency for each mass flow rate. This operating strategy is applied until the VIGVs are closed 50° with respect  
 309 to the axial direction, at which angle the high incidence losses at the inlet to the VIGVs increase exponentially  
 310 [29].

<sup>1</sup>The utilisation of VIGVs is not mandatory. This is a case-specific design choice for enhanced flexibility.

311 • In the third and last section, shown in red with cross markers, mass flow is reduced further but the position of  
 312 VIGVs remains constant at 50°. This causes a quick drop in efficiency.

313 It is observed that the utilisation of VIGVs enables a wider operating map on the left, around to the surge line,  
 314 which is instrumental to maximise the annual production of electricity/water (i.e., increased capacity factor).

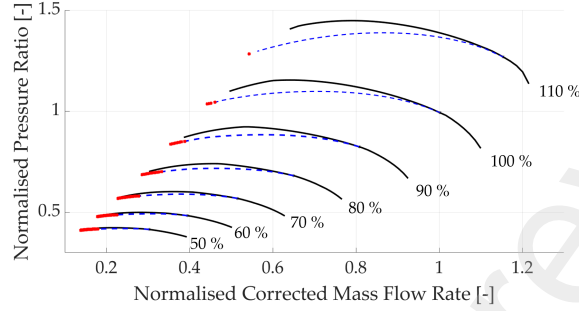


Figure 5: Combined normalised compressor performance map which shows the differences of using or not VIGVs at the impeller inlet.

### 315 3.1.2. Solar subsystem

316 The solar receiver is a cylindrical cavity with an hemispherical bottom cap whose geometry is presented in Fig.  
 317 6. This indirectly illuminated receiver has already been demonstrated in Europe [8] and it is one of the most widely  
 318 considered receiver designs in parabolic dish Stirling systems [33], despite the potential occurrence of hot spots  
 319 and also the potentially large temperature difference between receiver wall and working fluid, which may trigger  
 320 mechanical issues as highlighted in literature [34].

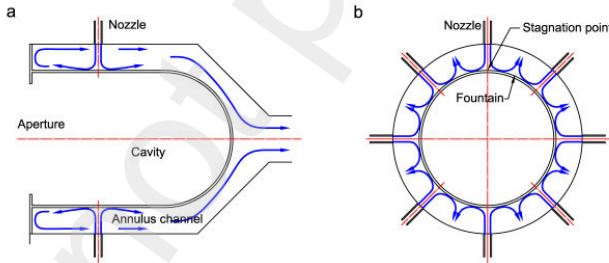


Figure 6: Axial (a) and radial (b) cross section schematics of the impinging solar receiver [35].

321 The solar dish is modelled under the assumption of a Gaussian distribution of the solar flux, corrected as proposed  
 322 by Aichmayer *et al.* [36], and considering a focus/diameter ratio of about 0.6. This is the optimal shape of the parabola  
 323 as demonstrated in the OMSoP Project [37], yielding the highest concentration ratio and highest temperatures.

324 The methodology adopted to design the receiver is based on the work by Wang [38]. It is an inverse design method  
 325 [39] taking into consideration the following requirements: interaction with the impinging jet, minimum heat transfer  
 326 coefficient (heat loss) and pressure drop. The main dimensions of the receiver are set following the recommendations  
 327 provided by Wang *et al.* [40], considering three degrees of freedom: i) diameter of the cavity,  $D_{cav}$ , ii) nozzle diameter,  
 328  $d$ , and iii) total number of nozzles,  $n$  (see Fig. 6). The interaction between jets is negligible for high values of the ratio  
 329 from the curvilinear distance between nozzle-centre and jet-centre to the diameter of the nozzle, but the undesired  
 330 interference will increase as this ratio decreases. The first requirement yields the following equation:

$$P_{jet}/d = \pi \cdot j \cdot (D_{cav} + 2t)/(n \cdot d) \geq 8 \quad (6)$$

331 where  $P_{jet}$  stands for jet pitch,  $t$  is wall thickness, and  $j$  is the number of nozzle rows in the axial direction.

332 The local Nusselt number on the outer receiver wall decreases far from the axis of the impinging jet. Therefore,  
 333 heat transfer 'valleys' are located between the stagnation areas (impingement of jet on receiver wall) and the 'fountain'  
 334 region (halfway between adjacent jets), whereas the maximum heat transfer coefficient is located in the 'fountain'  
 335 region between two impinging jets (see Fig. 6). In order to reach the targeted wall temperature, the heat transfer  
 336 coefficient at the 'fountain' regions (region between two impinging jets) must be higher than the required peak heat  
 337 transfer coefficient  $h_{req,max}$ . This second requirement can be expressed as follows:

$$h_{mid} = \frac{0.156 \cdot k_{wf} \cdot \dot{m}^{0.75} \cdot Pr^{0.42} \cdot e^{-0.025 \cdot (\pi \cdot j \cdot (D_{cav} + 2t) / (2n \cdot d))^2}}{(n \cdot \pi \cdot \rho \cdot \nu)^{0.75} \cdot d^{1.75}} \geq h_{req,max} \quad (7)$$

338 where  $\nu$  is kinematic viscosity,  $\dot{m}$  is mass flow rate,  $Pr$  is Prandtl number,  $\rho$  is air density and  $k_{wf}$  is thermal  
 339 conductivity of the working fluid.

340 Finally, an upper bound of pressure losses is set to  $\Delta P/P = 3\%$ , in order to avoid a strong negative impact on  
 341 turbine performance. Accordingly, the last requirement of the impinging receiver design can be expressed as follows,  
 342 according to the discussion presented in [38]:

$$n \cdot d^2 \geq 4 \cdot \dot{m} / (\pi \cdot C_d \cdot \sqrt{0.06 \cdot \rho \cdot P}) \quad (8)$$

343 where  $C_d$  is the discharge coefficient of the nozzle, which depends on the configuration of the nozzle.

344 Two main differences between the receiver design code employed in the present paper and that developed in the  
 345 reference work by Wang [38] are worth noting. First, while Wang's code starts from a 2D axisymmetric model and  
 346 then identifies the required distribution of heat transfer coefficients on the inner wall yielding optimum performance,  
 347 the present code aims to achieve the smallest receiver design possible, always fulfilling the three aforelisted require-  
 348 ments. This general difference in the overall structure of the code has a minimum impact on the results obtained.  
 349 Second, heat losses due to natural convection are accounted for in this work whereas they are neglected by Wang. To  
 350 calculate these, the *Nusselt* number on the outer wall of the receiver is estimated with Eq.(9), taken from [41]. The  
 351 approach adopted in this research is therefore deemed to provide a more conservative (safer) approach to receiver  
 352 design than that used in the reference documents.

$$Nu = 0.0027 \cdot Gr^{0.54} \cdot \left(\frac{T_w}{T_\infty}\right)^{0.47} \cdot (2 + 1.8 \cdot \cos^3 \theta)^{-3.62} \quad (9)$$

353 In Eq.(9),  $T_w$  and  $T_\infty$  are the temperatures of cavity wall and ambient air in Kelvin.

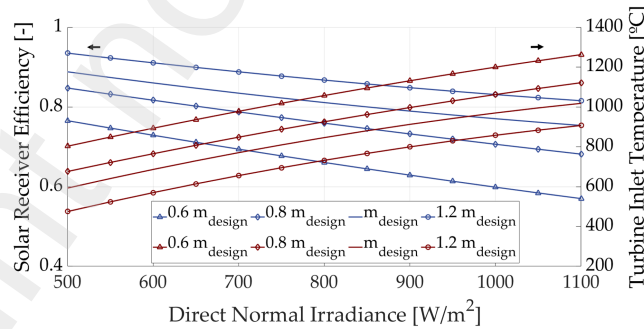


Figure 7: Impact of operating conditions on receiver performance.

354 The impact of operating temperature on receiver performance is shown in Fig. 7. It is observed that a lower  
 355 irradiance on the collector yields lower operating temperatures (at set values of air mass flow rate), caused by both  
 356 the lower thermal input and the lower thermal energy recovered on the high-pressure side of the recuperative heat  
 357 exchanger. Moreover, as expected, a decrease of the operating temperature brings about higher receiver efficiencies  
 358 thanks to reduced heat losses. Furthermore, the receiver tends to operate at higher efficiencies when the circulating  
 359 mass flow rate is also higher, which suggests scaling up the receiver. Nevertheless, as highlighted by Gavagnin *et al.*

360 [4], this has a strong negative effect on the economics of the solar collector whose cost increases exponentially with  
 361 size.

### 362 3.1.3. Recuperator

363 The recuperator is a critical component of a micro gas turbine given that it contributes some 30% of the total cost  
 364 of the engine [42], and it cannot be considered a secondary element relative to turbomachinery. This sets the target to  
 365 achieve as high recuperator efficiency as possible whilst, at the same time, ensuring small footprint, low cost and high  
 366 reliability amongst other specifications [43].

367 Figure 8 shows estimates of effectiveness and specific weight of different types of recuperators gathered from  
 368 literature [44, 45]. The specific weight of the recuperator matrix (excluding headers) is defined as the ratio of matrix  
 369 weight to gas flow rate so, under the same effectiveness, the smaller the specific weight, the lighter and more com-  
 370 pact the recuperator. This explains why primary surface heat exchangers (PSHEs) are the most common choice in  
 371 microturbine applications [46].

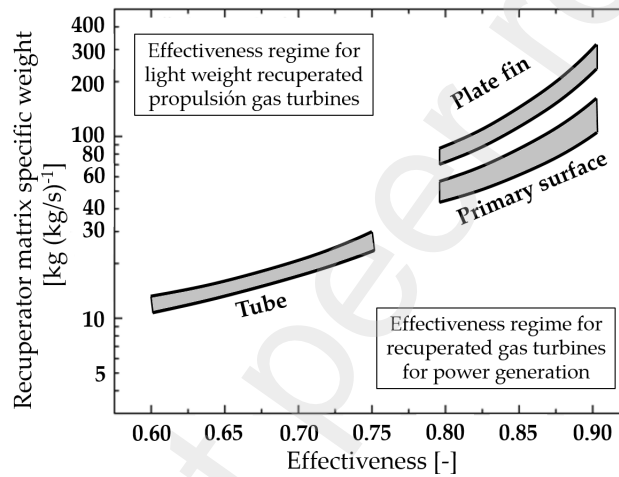


Figure 8: Comparisons of effectiveness and specific weight of the recuperator matrix (adapted from [44, 45])

372 The corresponding model employed to size (design) and model the performance of the recuperator is based on  
 373 splitting the component into several (sub) heat exchangers of equal heat duty,  $i$ , following the methodology described  
 374 by Nellis & Klein (2008) [47] to evaluate the overall conductance ( $UA$ ). This calculation relies on the  $\varepsilon - NTU$   
 375 approach presented in Eqs.(10-12).

$$\varepsilon = \frac{\dot{Q}_i}{\dot{C}_{min,i} \cdot (T_{H,i} - T_{C,i+1})} \quad \text{for } i = 1 \dots N \quad (10)$$

$$NTU_i = \frac{1}{1 - \dot{C}_i^*} \cdot \log \frac{1 - \varepsilon_i \cdot \dot{C}_i^*}{1 - \varepsilon_i} \quad \text{for } i = 1 \dots N \quad (11)$$

$$UA_i = NTU_i \cdot \dot{C}_{min,i} \quad \text{for } i = 1 \dots N \quad (12)$$

376 Where  $N$  is the total number of (sub) heat exchangers,  $U$  is the global heat exchange coefficient and  $\dot{C}_i^*$  the ratio  
 377 of heat capacities  $\frac{\dot{C}_{min,i}}{\dot{C}_{max,i}}$ .

378 The off-design performance of the recuperator is modelled through the conductance ratio method, a methodology  
 379 already presented and discussed in literature by some of the authors and other researchers [48–51]. Given the inlet  
 380 conditions and mass flow rates at both sides of the recuperator, the outlet conditions can be calculated with Eqs. (13-  
 381 15), where the rules to scale the rated value of conductance and to compensate for the variations of thermodynamic

382 properties throughout the recuperator have been validated experimentally by the authors in [49], for fluids with both  
 383 ideal and non-ideal gas behaviour.

$$\Delta P = \Delta P_{des} \cdot \frac{\dot{m}^2}{\dot{m}_{des}^2} \cdot \frac{\rho_{des}}{\rho} \quad (13)$$

$$\frac{1}{UA_{des}} = \frac{1}{hA_{hot,des}} + \frac{1}{hA_{cold,des}} \quad (14)$$

$$hA = hA_{des} \cdot \left(\frac{k}{k_{des}}\right) \cdot \left(\frac{\dot{m}\mu_{des}}{\dot{m}_{des}\mu}\right)^{0.8} \cdot \left(\frac{Pr}{Pr_{des}}\right)^y \quad (15)$$

384 Where  $y = 0.3$  for the hot side and  $y = 0.4$  for the cold side.

### 385 3.1.4. Turbine

386 The model of the radial-inflow turbine is based on the work by Aungier again [52] and it is therefore similar to that  
 387 used for the compressor. This is seen in Fig. 9, presenting the indirect process used to design the machine as well as  
 388 the most relevant design parameters: specifications and boundary conditions set by the thermodynamic cycle, specific  
 389 speed, total-to-total velocity ratio, and other first-guess values used to initialise the calculation process. In contrast  
 390 with the compressor model, there have been minor modifications with respect to the original model by Aungier and  
 391 the loss correlations selected. A detailed discussion of the expander design model, complementary to the information  
 392 presented in this work, can be found in [3].

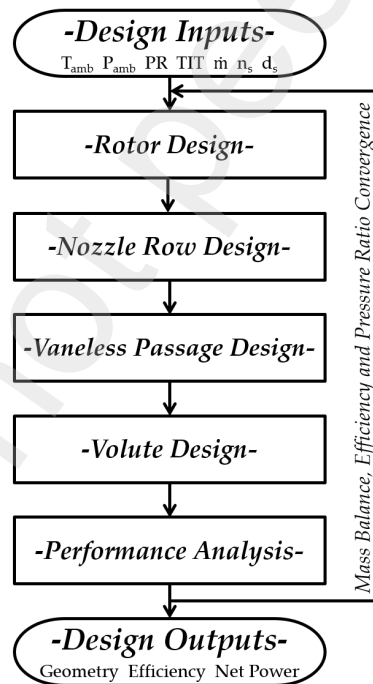


Figure 9: Turbine design flowchart.

393 Two constraints are used to size the rotor: i) the minimum hub radius compatible with the torque of the shaft  
 394 and the materials used ( $r_{out,shroud} < 0.9 \cdot r_{in}$ ), and ii) the lower limit of the axial length of the rotor ( $\Delta z_r \geq 1.5 \cdot b_{in}$ ).  
 395 Additionally, for mechanical integrity and to enable higher shaft speeds, the inlet flow at the design point is assumed  
 396 to be radial ( $\beta_{in} = 90^\circ$ ) [3]. Considering profile losses, incidence losses, loss coefficients due to blade loading and  
 397 hub-to-shroud loading effects, clearance losses and moisture losses (if any), the actual relative total pressure at the  
 398 outlet is given by the rotor loss coefficient,  $Y_{rot}$ , as expressed in Eqs. (16 - 17).

$$Y_{rot} = \sum Y_{i,rot} = Y_p + Y_{inc} + Y_{BL} + Y_{HS} + Y_{cl} + Y_Q \quad (16)$$

$$p'_{t,out} = \frac{p'_{t,out,id} + Y_{rot} \cdot p_{out}}{1 + Y_{rot}} \quad (17)$$

399 In order to size the nozzle row, the normalised blade profile is calculated first. To this end, the vanes are geometri-  
 400 cally set so as to match the required rotor inlet conditions and, then, the stagger angle yielding the required throat area  
 401 is determined. At the same time, velocity distributions within the blade passage are checked to avoid inconsistencies  
 402 and the nozzle throat area is fine tuned iteratively to avoid supersonic flow, often occurring at the outlet from the  
 403 nozzles. The performance map of an exemplary radial-inflow turbine is presented in Fig. 10 showing normalised  
 404 non-dimensional mass flow rate, pressure ratio and shaft speed.

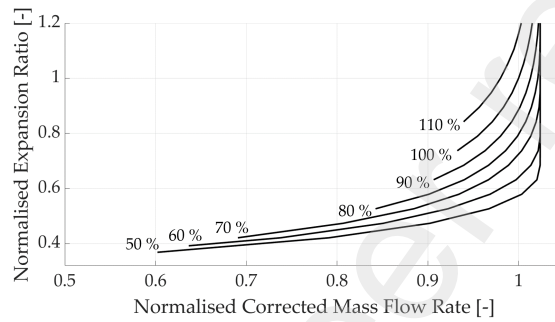


Figure 10: Normalised performance map of the turbine.

### 405 3.2. Reverse Osmosis

406 A schematic of the Reverse Osmosis unit is presented in Fig. 11, whose flow arrangement and equipment have  
 407 already been described in a previous section. Three main unitary processes can be identified:

408 1. Pressurisation of feedwater from ambient pressure to operating pressure, compliant with the following require-  
 409 ment:

$$\Pi_{BD} - \Pi_P < p_F - p_P - \Delta p_{lossF-BD} \quad (18)$$

- 410 2. Solvent separation through the reverse osmosis membrane.  
 411 3. Pressure recovery from the outlet flow of concentrate.

412 It is further assumed that the outlet flows of freshwater and brine ( $p_p$  and  $p_{BD,ERDout}$ ) are at atmospheric pressure,  
 413  $p_E$ .

414 The Specific Energy Consumption (SEC) of the process shown in Fig. 11 depends on the electric power consump-  
 415 tion of the high-pressure and booster pumps (HPP and BP) and can be calculated through the corresponding flow  
 416 rates,  $q_v$ , pressure,  $p$ , and pumps and engines efficiencies,  $\eta$ , as shown in Eq. (19):

$$SEC = \frac{\sum P_{W,HPP} + \sum P_{W,BP}}{q_{v,P}} = \frac{q_{v,HPP}}{q_{v,P}} \cdot \frac{p_{F,HPP} - p_F}{\eta_{HPP} \cdot \eta_{eng,HPP}} + \frac{q_{v,ERD}}{q_{v,P}} \cdot \frac{p_{F,HPP} - p_{F,ERDout}}{\eta_{BPP} \cdot \eta_{eng,BP}} \quad (19)$$

417 Then, the energy recovery process taking place in the isobaric chamber is characterised by Eqs. (20-21), whereas  
 418 the ratio of volumetric flow rates through each stream is defined by Eqs. (22-24):

$$\eta_{ICH} = \frac{q_{v,ERD} \cdot p_{F,ERDout} + q_{v,BD} \cdot p_{BD,ERDout}}{q_{v,ERD} \cdot p_{F,ERDin} + q_{v,BD} \cdot p_{BD,ERDin}} \quad (20)$$

$$\varepsilon_{ICH} = \frac{p_{BD,ERDout} - \eta_{ICH} \cdot p_{BD,ERDin}}{\eta_{ICH} \cdot p_{F,ERDin} - p_{F,ERDout}} \quad (21)$$





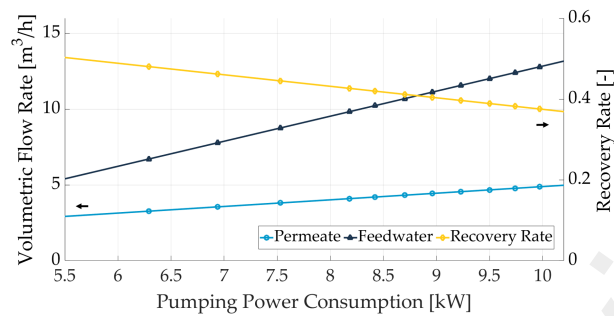


Figure 12: Off-design performance of the SWRO unit.

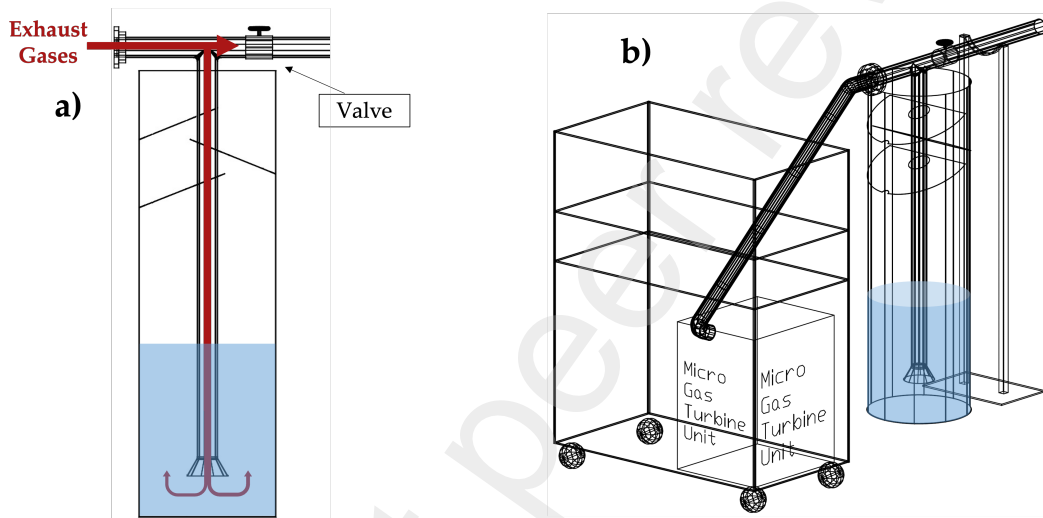


Figure 13: Configuration of the low-cost ZLD prototype. a) Cross-sectional view of the ZLD unit. b) Schematic showing global arrangement.

444 The thermal energy carried by the gas stream is used to evaporate water in the brine, and this increases the  
 445 concentration of salt in the well. At full load (hence peak mass flow rate of gases into the well), water drift becomes  
 446 a concern. Hence, in order to minimise this, baffles are installed inside the well above the brine level: droplets carried  
 447 over by the gaseous stream impinge on the baffles, returning to the well for evaporation.

448 The performance of the prototype has been assessed for operation in batches and is illustrated in Fig. 14. Temper-  
 449 ature measurements, logged with infrared imagery and thermocouples, indicate that the temperature of exhaust gases  
 450 flowing across the ZLD drops from  $\sim 450^{\circ}\text{C}$  to  $\sim 70^{\circ}\text{C}$ . A picture of the ZLD in operation without the upper seal is  
 451 also shown in Fig. 14, revealing the plume coming out from the well (high steam content), alongside a picture of the  
 452 resulting solid waste (solid salt).

### 453 3.3.2. Performance modelling of the ZLD system

454 As already discussed in earlier sections of the paper, Zero-Liquid-Discharge concepts have recently gained impor-  
 455 tance, out of the growing desalination capacity worldwide and of the associated concerns about the impact that this  
 456 highly-concentrated effluents have on the environment. The term brine is typically used to define the resulting by-  
 457 product (waste) of desalination systems, whose salt concentration typically exceeds 55 mg/L [54]. Based on different  
 458 technologies, ZLD systems tackle this problem by trying to increase the concentration of the brine further with the  
 459 ideal objective to completely remove the water content of it; accordingly, the resulting waste would be in solid state  
 460 and would therefore represent a much more manageable product, easier to store and transport and with a lower poten-  
 461 tial to harm the environment. For these reasons, ZLD systems are increasingly being seen as mandatory requirements

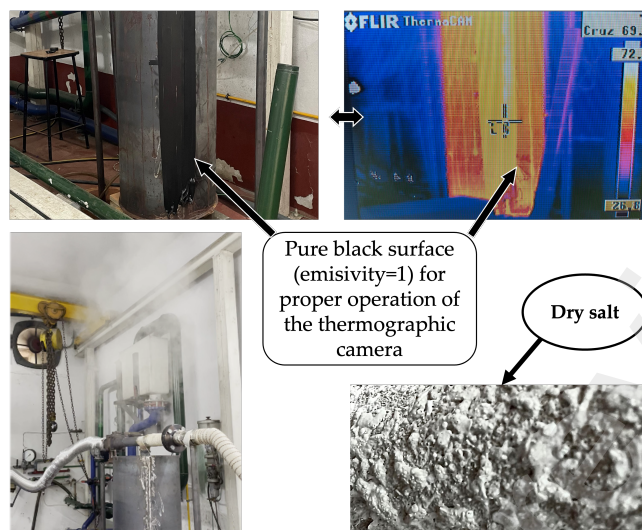


Figure 14: Graphical information of the proof-of-concept ZLD.

462 in some industrial processes [55].

463 Indeed, reducing the amount of energy needed to minimise the discharge of liquid, highly-saline effluents is behind  
 464 the growing interest in coupling membrane-based RO technology with ZLD systems, given the energy-intensive nature  
 465 of this process [56]. In this regard, it is also to note that RO cannot be used for the further concentration of effluents  
 466 given that these membranes can work with feedwaters whose salinity is not higher than 80 g/L, in order to avoid  
 467 membrane fouling and scaling. For brines with higher concentrations, a phase-change ZLD system is needed [57–59].

468 In the actual SOLMIDEFF system, the ZLD unit works very similarly to the proof-of-concept unit tested at the  
 469 lab (Fig. 14): the concentrate resulting from the RO process is fed into an evaporation chamber, which is nothing but  
 470 a direct contact heat exchanger where the thermal energy of the exhaust gases from the SmGT (at 225-325°C) is used  
 471 to evaporate water in the brine (therefore raising the concentration of this stream). Moreover, in order to enhance the  
 472 performance of the ZLD unit, the brine flowing into the evaporation chamber is preheated in another chamber fed by  
 473 the stack gases at a lower temperature of ~60-100°C (given that a large portion of their sensible heat has already been  
 474 harvested in the evaporation process). This is shown schematically in Fig. 15.

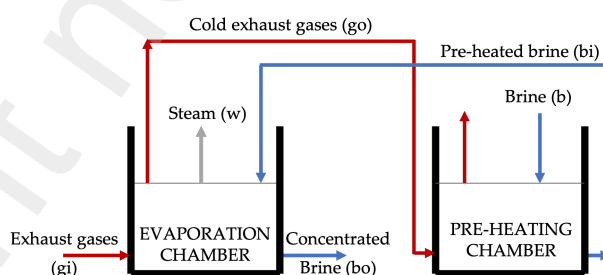


Figure 15: Conceptual representation of the Zero Liquid Discharge unit.

475 The ZLD system is modelled through the methodology already presented and discussed by some of the authors  
 476 in [16]. The ratios of mass flow  $m_{bo}/m_{bi}$  and salinity  $S_{bo}/S_{bi}$  are estimated for operating conditions different to the  
 477 rated values, noting that there is an operational limit of the inlet mass flow ratio ( $m_{gi}/m_{bi}$ ). In particular, the mass flow  
 478 rate of exhaust gases from the turbine must be, at least, 10 times higher than the mass flow rate of brine inasmuch as,  
 479 otherwise, the stability and sustainability of the evaporation process would be compromised [60].

480 Figures 16(a) and 16(c) show trends of the ratio from inlet to outlet salinity ratio of the brine ( $S_{bi} / S_{bo}$ ), for  
 481 values of inlet brine salinity of  $S_{bi} = 0.07$  kg/kg and  $S_{bi} = 0.12$  kg/kg respectively. This ratio increases for higher

482 temperatures of either the exhaust gases from the turbine ( $T_{gi}$ ) or the inlet brine ( $T_{bi}$ ) until saturation is reached, thus  
 483 setting the limit up to which brine could be concentrated. Pre-heating the brine to the lowest temperature ( $T_{bi} = 60^\circ\text{C}$ )  
 484 enables to reach the saturation point if the inlet temperature of the exhaust gases inlet temperature ( $T_{gi}$ ) is  $305^\circ\text{C}$ , and  
 485 this latter temperature decreases to no more than  $290^\circ\text{C}$  when the brine is pre-heated to  $100^\circ\text{C}$ . The inlet to outlet mass  
 486 flow rates of the brine ( $m_{bi} / m_{bo}$ ) is shown on the right charts, Figs. 16(b) and 16(d). As either temperature increases  
 487 (brine or gases), more water is evaporated and salinity becomes higher. Moreover, linking all charts, it is also observed  
 488 that the mass flow ratio lines for different brine temperatures and concentrations change curvature and tend to collapse  
 489 into a single line which corresponds with saturated brine conditions. This is an interesting result though it must be  
 490 acknowledged that curves past saturation point are merely hypothetical.

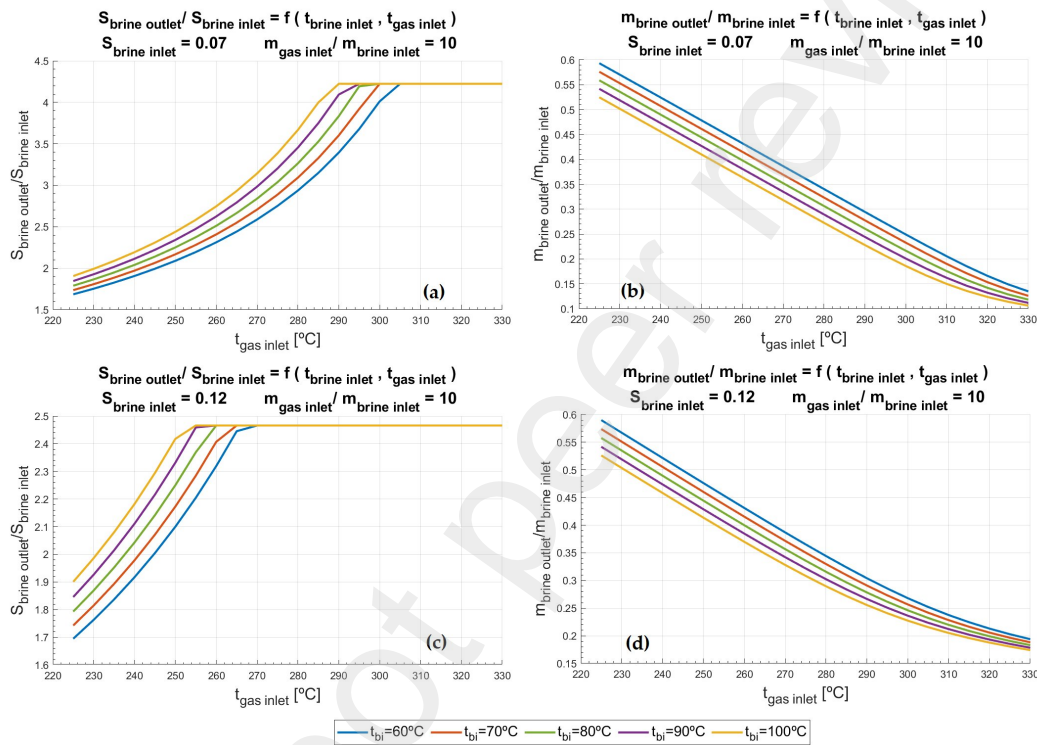


Figure 16: (a) Ratio between outlet and inlet brine salinities as a function of temperatures of inlet brine and inlet exhaust gases from the turbine in evaporation chamber, for  $S_{bi}=0.07$  kg/kg and  $m_g/m_b=10$ ; (b) Ratio between outlet and inlet brine mass flow rates as a function of temperatures of inlet brine and inlet exhaust gases from the turbine in evaporation chamber, for  $S_{bi}=0.07$  kg/kg and  $m_g/m_b=10$ ; (c) Ratio between outlet and inlet brine salinities as a function of temperatures of inlet brine and inlet exhaust gases from the turbine in evaporation chamber, for  $S_{bi}=0.12$  kg/kg and  $m_g/m_b=10$ ; (d) Ratio between outlet and inlet brine mass flow rates as a function of temperatures of inlet brine and inlet exhaust gases from the turbine in evaporation chamber, for  $S_{bi}=0.12$  kg/kg and  $m_g/m_b=10$  [16]

#### 491 4. Validation

492 Given the lack of available SmGT experimental data in literature, it is not possible to validate the system presented  
 493 in this work directly. Nevertheless, the accuracy of the model developed for each component of the system can be  
 494 verified individually, either theoretically or experimentally, in order to ensure that the results obtained are trustful:

- 495 • **Turbomachinery:** the modelling approach of compressor and turbine are widely known in literature and have  
 496 been validated extensively against a considerable set of experimental data [27, 52], in addition to being used in  
 497 past works by the authors [5].
- 498 • **Solar subsystem:** the solar dish has been modelled according to the work carried out by Aichmayer *et al.*  
 499 for the OMSoP project [36], whereas the methodology applied to model the impinging solar receiver follows

500 the inverse design method proposed by Wang [39]. Both models were validated through experimentally at the  
501 Royal Institute of Technology in Stockholm (KTH) [38].  
502 • **Recuperator:** the recuperator is modelled following the well known method described by Nellis & Klein (2008)  
503 [47], whilst the conductance ratio method used to estimate the off-design performance of the recuperator was  
504 previously validated experimentally by the authors [49].  
505 • **Reverse Osmosis:** the validation of the design and off-design models of performance of the RO system is  
506 supported on the software Q+ [26], whose accuracy is widely acknowledged within the industrial and scientific  
507 communities.  
508 • **Zero Liquid Discharge:** the reduction of the water content of the concentrate in the ZLD unit has been verified  
509 in the previous section, although further experimental validation for continuous operation is needed.

## 510 5. Results

511 The description provided so far in this paper depicts a tightly integrated system with a larger potential for overall  
512 energy utilisation than water desalination systems based on solar photovoltaic or wind power. Nevertheless, in order  
513 to verify and quantify this, it is mandatory to assess the performance of the system. This is presented now.

514 The general boundary conditions considered in this assessment correspond to the Canary Islands, given their need  
515 for freshwater and high availability of solar energy; the applicability to neighbouring coastal locations in Northern  
516 Africa is an additional feature making this location interesting and representative of a larger area. In particular, ambient  
517 conditions have been generated with METEONORM v.7 [61], gathering records from the last 25 years and establishing  
518 95 percentile values, whereas seawater composition and temperature have been collected from the guidebook by Wilf  
519 and Awerbuch [62]:

- 520 • Ambient temperature: 26.4 °C.
- 521 • Direct Normal Irradiance: 796 W/m<sup>2</sup>.
- 522 • Water temperature: 22.0 °C.
- 523 • Total Dissolved Solids (TDS) of feedwater: 38739 ppm.

### 524 5.1. Rated Performance

525 The specifications of major equipment are set to the values listed in Table 2 for a preliminary assessment, according  
526 to the discussions presented in foregoing sections of the paper.

527 The application of the design and performance models shown in earlier sections of the paper to the specifications  
528 in Table 2 yields the performance summarised in Table 3 for a system with the layout shown in Fig. 2. The net  
529 efficiency of the SmGT is around 26%, a value that is in line with the performance reported by the main Original  
530 Equipment Manufacturers of this type of engines [63]. The rated electric power is set to 10 kWe, which is mostly  
531 consumed by the pumps in the RO module -producing 4.68 m<sup>3</sup>/h of fresh water-; the remaining power is consumed  
532 by the auxiliary systems needed.

<b>Parameter</b>	<b>Units</b>	<b>Value</b>
<i>Solar Energy Collection</i>		<i>Calculated</i>
Dish aperture area	[m <sup>2</sup> ]	65.87
Solar energy input	[kWt]	52.44
Collector efficiency	[%]	89.87
Receiver efficiency	[%]	81.87
<i>Gas Turbine</i>		<i>Calculated</i>
Isentropic compressor efficiency	[%]	78.77
Isentropic turbine efficiency	[%]	81.58
Receiver pressure drop	[%]	1.37
		<i>Set</i>
Recuperator effectiveness	[%]	85.0
Mechanical efficiency	[%]	98.0
Generator efficiency	[%]	98.0
Combustor pressure drop	[%]	2.0
Recuperator hot pressure drop	[%]	2.5
Recuperator cold pressure drop	[%]	2.5
Air filter pressure drop	[%]	2.0
<i>Reverse Osmosis</i>		<i>Set</i>
Isobaric chamber (ERD) efficiency	[%]	97.0
High pressure pump efficiency	[%]	87.0
Booster pump efficiency	[%]	80.0

Table 2: Specifications of major equipment in SOLMIDEFF.

533 Table 4 presents the heat and mass balance of the overall SOLMIDEFF system, supporting the results in 3. Streams  
534 1 to 9 correspond to the SmGT whereas the water treatment subsystem is defined by streams 10 to 21.

Parameter	Units	Value
<i>Micro Gas Turbine</i>		
Power output	[kWe]	10.0
net efficiency	[%]	26.04
Solar-electric efficiency	[%]	19.15
Air mass flow rate	[kg/s]	0.121
Pressure ratio	[-]	3.5
Turbine inlet temperature	[K]	1123.1
Exhaust temperature	[K]	531.1
Shaft speed	[rpm]	134,294
<i>Reverse Osmosis</i>		
Freshwater production	[m <sup>3</sup> /h]	4.68
Recovery rate	[%]	38.7
Operating pressure	[bar]	59.6
Specific Energy Consumption	[kWh/m <sup>3</sup> ]	2.03
TDS of concentrate	[ppm]	59,500
Pumping power	[kWe]	9.5
Membrane area	[m <sup>2</sup> ]	286
<i>Zero Liquid Discharge</i>		
Mass flow rate	[kg/s]	0.011
Salinity of outflow	[kg/kg]	0.286

Table 3: Performance of SOLMIDEFF. Component specifications are reported in Table 2

<i>Micro Gas Turbine</i>			
Stream	p [bar]	T [K]	h [kJ/kg]
1	1.013	299.5	425.84
2	0.993	299.5	425.85
3	3.464	461.9	590.09
4	3.377	842.9	995.90
5	3.331	1123	1314.6
6	3.264	1123	1314.5
7	1.071	907.1	1067.3
8	1.045	531.1	661.52
9	1.013	531.1	661.52
<i>Sea Water Reverse Osmosis</i>			
Stream	p [bar]	$\dot{m}$ [m <sup>3</sup> /h]	TDS [ppm]
10	1.621	12.1	38,739
11	1.621	4.77	38,739
12	59.60	4.77	38,739
13	1.621	7.33	38,739
14	57.75	7.33	38,739
15	59.60	7.33	38,739
16	59.60	12.1	38,739
17	1.013	4.68	116.52
18	58.45	7.42	62,564
19	1.013	7.42	62,564
20	1.013	7.42	62,564
21	1.013	-	62,564

Table 4: Heat and mass balance of the reference case for the SOLMIDEFF system without ZLD operation.

535 The recovery rate of the RO module in SOLMIDEFF at design conditions is 39%, although this value increases  
536 as pump consumption decreases, sustaining the maximum pressure as described in the previous section. This recov-  
537 ery rate is equivalent to a Specific Energy Consumption of 2.03 kWh/m<sup>3</sup>, accounting for the RO module only (i.e.,  
538 excluding any water steam produced by the ZLD unit), which translates into a Solar Specific Energy Consumption of  
539 10.6 kWh/m<sup>3</sup> when the efficiency of the topping SmGT system is accounted for.

540 The ambient temperature play an important role in the turbomachinery design process since the air density is  
 541 significantly dependent on the temperature. A different ambient temperature supposes that, for the same mass flow  
 542 rate, a higher volumetric flow rate must go through the SmGT, and of course a higher sizing of turbomachinery must be  
 543 considered. Besides, the pressure ratio which maximised the net efficiency of the simple recuperated open cycle also  
 544 rely on the ambient temperature, introducing a twofold influence on the SmGT. Table 5 lists some of the parameters  
 545 that are influenced by ambient temperature.

Parameter	Units	Value	
<b>Ambient temperature</b>	<b>[K]</b>	<b>293</b>	<b>313</b>
<i>Compressor &amp; Turbine</i>			
Mass Flow Rate	[kg/s]	0.120	0.141
Pressure Ratio	[-]	3.5	3.3
Shaft Speed	[rpm]	135,594	121,722
<i>Compressor</i>			
Total-to-Total Efficiency	[%]	78.75	78.34
Rotor Inlet Hub Radius	[mm]	2.20	2.40
Rotor Inlet Shroud Radius	[mm]	17.6	19.8
Rotor Outlet Radius	[mm]	35.7	39.9
<i>Turbine</i>			
Total-to-Total Efficiency	[%]	80.73	81.28
Rotor Inlet Radius	[mm]	36.8	36.7
Rotor Outlet Shroud Radius	[mm]	25.9	27.1
Rotor Outlet Hub Radius	[mm]	3.0	3.1

Table 5: Sensitivity analysis of the sMGT turbomachinery design against ambient temperature

## 546 5.2. Off Design Performance

547 This section is aimed at presenting the sizing process of the SOLMIDIFF system according to the environmental  
 548 conditions of the site (Direct Normal Irradiance and ambient temperature) and, also, to assess how a particular system  
 549 performs under different daily conditions in Gran Canaria (Canary Islands); to this latter end, an example of days  
 550 with clear and hazy skies are considered. As a summary of some of these aspects, the performance map of this sizing  
 551 process for every combination of DNI and ambient temperature is presented.

552 Compressor-turbine matching in the solar micro gas turbine is adapted from Razak for a simple, open-cycle  
 553 industrial gas turbine [24], incorporating the effect of pressure drops introduced by the recuperative heat exchanger.  
 554 The iterative process in the cited reference is nevertheless modified in order to have it controlled by the required turbine  
 555 inlet temperature  $TIT$  and direct normal irradiance  $DNI$ , rather than shaft speed power output setting. Furthermore,  
 556 for simplicity, inlet and exhaust losses are neglected ( $P_{intake} = P_{exhaust}$ ) and the gas constant  $R$  and isentropic index  $\gamma$   
 557 are determined under the assumption of dry air. A graphical summary of changes from the original method is shown  
 558 in Fig. 17.

559 Engine operation relies on three main control strategies, aimed at preventing overheating and/or overspeeding of  
 560 the engine under very high insolation. In particular, when direct normal irradiance  $DNI$  increase:

- 561 • Shaft speed of turbomachinery can be increased, yielding higher throughput and pressure ratio. This yields  
 562 lower turbine outlet temperature, therefore reducing the recuperative potential of the engine (i.e., lower temper-  
 563 ature at receiver inlet). The cumulative effect of a higher mass flow rate and lower temperature at receiver inlet  
 564 increases the amount of concentrated solar energy absorbed for the same turbine inlet temperature.
- 565 • If this is not enough to manage the large solar input, turbine inlet temperature is allowed to rise to 25°C higher  
 566 than the design value. This strategy can nevertheless not be used for long periods of time (given the severe  
 567 reduction of the useful life of the turbine).
- 568 • Lastly, in case of continued operation under this undesired conditions (lower than 1% yearly as estimated by  
 569 the METEONORM database [61]), the solar concentrator would have to be defocused. This is, unfortunately,  
 570 not practical given that defocusing the concentrator implies increasing the spillage at the receiver window, thus

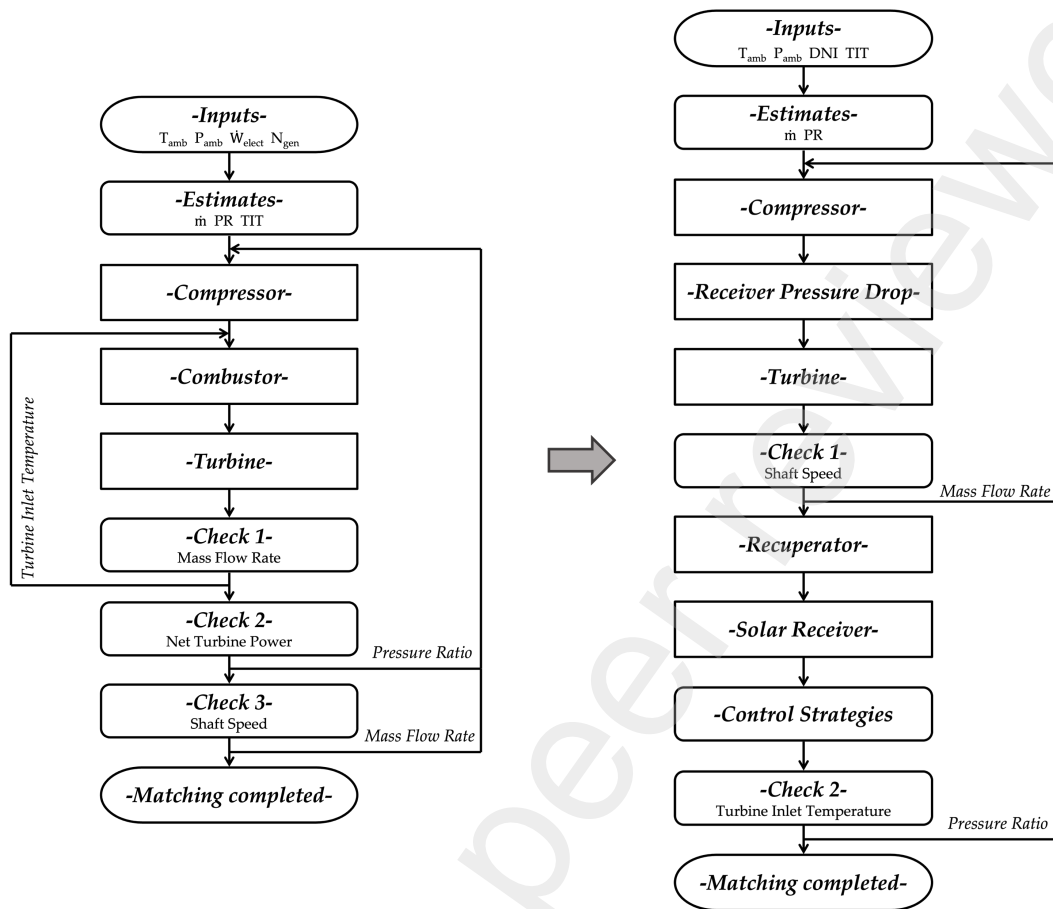


Figure 17: Comparison of the reference compressor-turbine matching process in [24] (a) and the process followed in this work (b)

571 compromising the mechanical integrity of this component. In other words, if defocusing were needed, the  
 572 system would have to be shut down.

573 On the other end of the operating range, reducing the heat input to the system implies either of the following  
 574 alternatives: turbine inlet temperature is reduced for somewhat constant mass flow rate, or mass flow rate is reduced  
 575 whilst keeping turbine inlet temperature at the rated value. In either case, the pressure ratio of the engine is reduced,  
 576 hence lifting turbine exhaust temperature for constant turbine inlet temperature, and this poses the risk to overheat  
 577 the recuperator. Therefore, a limit on turbine exhaust temperature is set to 650°C, assuming that the recuperator  
 578 is manufactured from Stainless Steel without specific treatments (e.g. ceramic coatings) [46]. The impact of this  
 579 constraint on part-load efficiency is illustrated in Fig. 18.

580 As discussed previously, the presence of VIGVs helps enhance the performance of the machine for operating  
 581 conditions closer to the surge line, but the running line of a standard engine without Variable Nozzle Guide Vanes  
 582 (VNGVs) is far from this region of the compressor map. Indeed, at reduced loads, even if mass flow rate and/or  
 583 turbine inlet temperature are reduced, so is pressure ratio, hence moving the operating conditions to lower regions of  
 584 the map where the presence of VIGVs is not providing any added flexibility. Based on this, the utilisation of variable  
 585 compressor geometry (i.e., VIGV) comes together with the adoption of variable nozzle guide vanes in the expander  
 586 (VNGV). This combination is required to avoid derating the part-load performance of the engine through the adoption  
 587 of a lower turbine inlet temperature once the inlet temperature to low-pressure side of the recuperator (turbine exhaust  
 588 temperature) achieves the maximum value allowed by materials. In particular, the incorporation of Variable Nozzle  
 589 Guide Vanes in the expander enables reducing the inlet area to the nozzles at part load, thus keeping pressure ratio at  
 590 or close to the rated value. This higher pressure ratio than in the standard engine (without VNGVs) also yields a larger



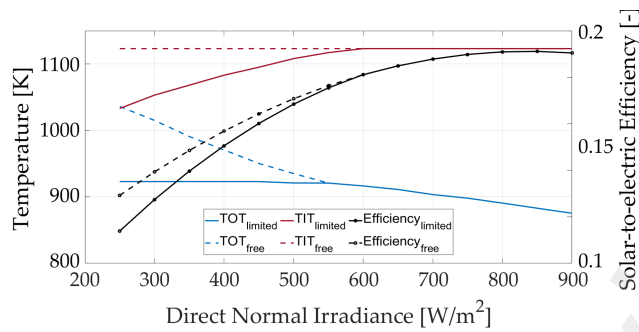


Figure 18: Part-load efficiency for a system with recuperator temperature limited to 650°C.

591 temperature drop across the turbine, thus enabling operation at low partial loads without the need to reduce turbine  
 592 inlet temperature. This has a twofold beneficial effect. From a system standpoint, it means higher part-load efficiency  
 593 due to the higher pressure ratio and turbine inlet temperature. For the compressor, it means higher pressure ratios at  
 594 low mass flow rates, thus moving the running line closer to surge and to the higher efficiency region of the compressor  
 595 map. Accordingly, the utilisation of VIGVs in the compressor, whose impact on compressor inlet flow was described in  
 596 in the previous section, is instrumental to enable such operation.

597 The strong influence of incorporating VNGVs on system performance is better understood by observing the run-  
 598 ning lines of compressor and turbine on their respective performance maps when VNGVs are or are not used. Figure  
 599 19 shows this information for a representative solar micro gas turbine where DNI ranges from 250 W/m<sup>2</sup> to 900 W/m<sup>2</sup>  
 600 at constant ambient temperature (26.4°C). The blue and red solid lines correspond to compressor and turbine with any  
 601 variable geometry feature, whilst the blue and red dotted lines are those of the compressor and turbine incorporating  
 602 VIGVs and VNGVs respectively; the low mass flow end of these latter running lines are set by the minimum TIT drop  
 603 compliant with a surge margin of at least 10% (note that mass flow rate is reported in corrected values to account for  
 604 the effect of inlet pressure and temperature). For the sake of clarity, turbine maps for three different VNGVs angles  
 605 only are shown in Fig. 19: design point (fully open), 75% closed and 55% closed.

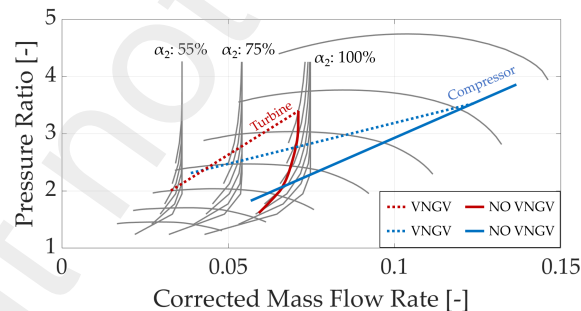


Figure 19: Running lines of the SmGT. Blue and red lines correspond to compressor and turbine. Solid and dotted correspond to machines with fixed geometry and machines with variable geometry (VIGVs for compressor and VNGVs for turbines) respectively.

606 The global impact of variable geometry on system performance is shown on Fig. 20, showing the pressure ratio  
 607 and efficiency (solar-to-electric efficiency) of solar micro gas turbines with and without variable geometry across the  
 608 entire load range, for the same rated output and efficiency. Two main features are observed. First of all, the superior  
 609 performance (efficiency) of the solar micro gas turbine incorporating variable geometry compressor and turbine is  
 610 shown. Indeed, despite having the same rated efficiency, the part-load efficiency curve of the variable geometry  
 611 SMGT is much more resistant to load reductions thanks to the higher pressure ratio (and turbine inlet temperature).  
 612 Second, the wider operating range without having to shut the engine down is also visible. This confirms the interest  
 613 of adopting variable geometry turbomachinery in spite of the higher cost and complexity.

614 The impact of the different performance, illustrated in Fig. 20, on the production of electricity of the SMGT for

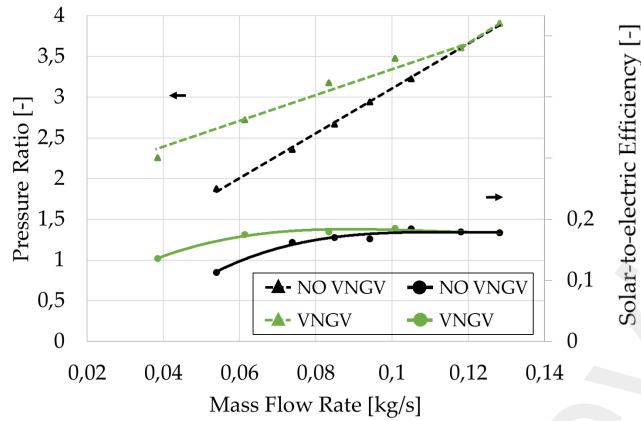


Figure 20: Part-load performance of solar micro gas turbines with (black) and without (green) variable geometry turbomachinery. Pressure ratio (dashed) and solar-to-electric efficiency (solid) curves are shown.

615 two different sets of boundary conditions -sunny and cloudy days- is summarised in Figs. 21 and 22 and in Tables  
 616 6 and 7. Both of these tables provide detailed information about the performance of the operating parameters of the  
 617 standard engine without variable geometry and those of the system incorporating variable geometry. The benefits  
 618 of a stronger resistance of pressure ratio and turbine inlet temperature of the variable geometry engine when load  
 619 is reduced (i.e., higher pressure ratio and turbine inlet temperature) translate into higher efficiency and, therefore,  
 620 production of electricity. As expected, this is more visible in the case of a cloudy day.

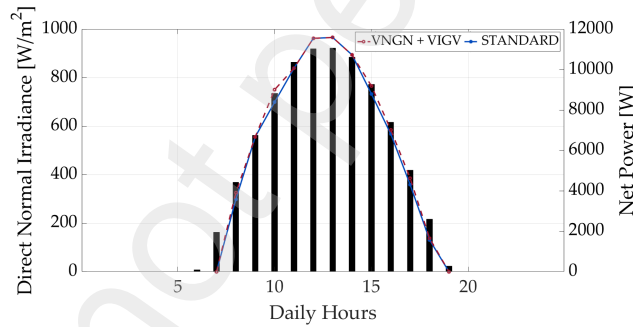


Figure 21: Net power production of the SmGT during a sunny day

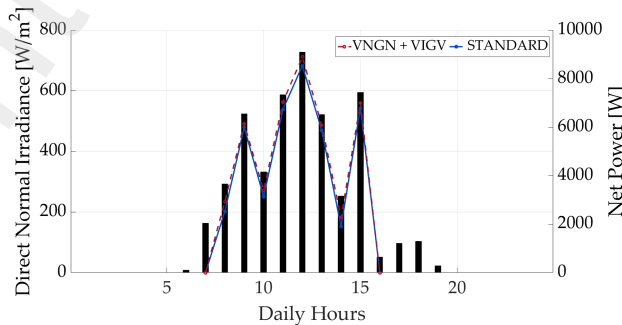


Figure 22: Net power production of the SmGT during a cloudy day

Hour	$T_{amb}$	$\eta_{comp}[\%]$		$\eta_{turb}[\%]$		$\eta_{sol-elec}[\%]$		Pressure Ratio [-]		TIT [K]	
		VIGV VNGV	Standard	VIGV VNGV	Standard	VIGV VNGV	Standard	VIGV VNGV	Standard	VIGV VNGV	Standard
8:00	28.7°C	72.93	78.72	86.26	81.41	16.09	14.74	2.68	2.25	1,103	1,053
9:00	29.9°C	75.82	79.54	83.58	81.77	18.01	17.99	3.31	2.82	1,123	1,098
10:00	31.0°C	78.15	78.59	81.62	79.65	18.59	17.32	3.60	3.45	1,123	1,108
11:00	32.1°C	77.71	77.71	81.05	81.05	17.68	17.68	3.85	3.85	1,113	1,113
12:00	32.9°C	77.07	77.07	83.25	83.25	19.07	19.07	4.14	4.14	1,138	1,138
13:00	33.6°C	77.04	77.04	83.23	83.23	19.08	19.08	4.13	4.13	1,143	1,143
14:00	34.1°C	77.14	77.14	83.62	83.62	18.40	18.40	4.10	4.10	1,113	1,113
15:00	33.2°C	78.20	78.16	81.16	80.33	18.04	17.27	3.61	3.57	1,123	1,108
16:00	33.2°C	76.43	78.87	82.14	79.61	17.30	16.77	3.42	3.05	1,123	1,103
17:00	33.2°C	76.73	78.55	84.26	81.79	16.75	15.58	2.70	2.41	1,098	1,068
18:00	33.7°C	73.50	75.02	86.60	83.78	11.58	11.00	2.01	2.01	1,038	1,013

Table 6: Variation of key performance parameters of engines with and without variable geometry, in a sunny day

Hour	$T_{amb}$	$\eta_{comp}[\%]$		$\eta_{turb}[\%]$		$\eta_{sol-elec}[\%]$		Pressure Ratio [-]		TIT [K]	
		VIGV VNGV	Standard	VIGV VNGV	Standard	VIGV VNGV	Standard	VIGV VNGV	Standard	VIGV VNGV	Standard
8:00	21.2°C	72.36	79.12	87.10	80.67	15.18	13.09	2.45	2.01	1,073	1,028
9:00	22.2°C	72.92	78.55	85.00	80.32	17.79	17.36	3.25	2.72	1,123	1,093
10:00	23.2°C	72.43	79.28	86.43	81.21	15.52	14.35	2.52	2.13	1,083	1,043
11:00	24.3°C	75.52	78.32	83.41	79.89	18.20	17.41	3.43	2.93	1,113	1,103
12:00	25.5°C	78.11	78.73	80.79	79.28	18.57	17.82	3.56	3.42	1,123	1,108
13:00	26.0°C	74.02	78.58	85.01	80.45	17.70	17.07	3.29	2.72	1,108	1,093
14:00	26.1°C	72.58	78.61	87.05	81.33	13.61	11.43	2.24	1.88	1,063	1,018
15:00	26.3°C	75.02	78.58	83.43	79.80	17.85	17.33	3.45	2.96	1,118	1,103

Table 7: Variation of key performance parameters of engines with and without variable geometry, in a partly cloudy day

## 6. CONCLUSIONS

A novel solar desalination system driven by a SmGT for off-grid applications is assessed in this work, with a specific focus on the modelling of each subsystem. A complete set of detailed models enables characterising the design and off-design performance of the SOLMIDEFF system, which is comprised of the following components: i) a solar-to-electric power conversion unit based on dish-mGT, ii) a RO unit for desalination or industrial water treatment and iii) a ZLD system for the treatment of desalination or industrial process effluents. The computational tool enables to size the components and to evaluate their off-design performance for different design configurations, including the incorporation of VIGV in the compressor.

The following relevant conclusions are drawn:

- Incorporating variable inlet guide vanes in the compressor has the effect of extending the operating range of the engine, enabling operation at low mass flow rates and high pressure ratios (close to the surge). Since using VNGVs in the turbine enables operation at high pressure ratio even when mass flow rate is reduced at part load, their combination with VIGVs (Variable Inlet Guide Vanes) at compressor inlet is crucial to increase the off-design performance of the SmGT proposed in this work.
- Incorporating VIGVs in the compressor without considering a radial turbine with VNGVs (Variable Nozzle Guide Vanes) seems to be of no interest, since the running line of the compressor does not approach the surge line.
- Since recuperators made of stainless steel cannot operate at temperatures higher than 650°C, operation at part-load using standard (constant geometry) SmGT implies that turbine inlet temperature be decreased what, in turn,

640 reduces turbine outlet temperature. This, unfortunately, brings about a faster efficiency decay when running the  
641 system in partial load.

- 642 • The ZLD system is capable of saturating the concentrate of the RO unit, enabling zero liquid discharge. This  
643 has been proved experimentally for operation in batches, where the concentrate has been reduced to solid (dry)  
644 salts.
- 645 • If higher flexibility at partial load is needed by a RO system, a moderate recovery rate at rated conditions is  
646 recommended.

647 Globally, the system presented in this work shows interesting features, mostly a very high flexibility, robustness  
648 and easy integration with backup fuels of different types. These are all promising characteristics for applications  
649 where a robust and reliable polygeneration system (power, heat and fresh water) is needed, able to ensure the security  
650 of supply of these primary services for off-grid communities.

651 Future work by the authors will include the analysis of transient operation (currently supported by a research  
652 project funded by the Regional Council for Research of the Government of Andalusia, Spain), due to the dissimilar  
653 characteristics of the solar gas turbine and reverse osmosis in this regard, as well as an economic assessment in  
654 different latitudes. This is a fundamental step towards assessing the commercial feasibility of the system proposed.

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