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PREFACE

The 6 th International conference "Contemporary Trends and Innovations in the Textile Industry" CT&ITI 2023, is co-organized by the Union of Engineers and Textile Technicians of Serbia, the Union of Engineers and Technicians of Serbia, the Faculty of Technology and Metallurgy in Belgrade, the University of Faculty of Technology, Shtip, North of Macedonia, Society for Robotics of Bosnia i Herzegovina and Balkan Society Of Textile Engineering-BASTE of Greece.

The Ministry of Science, Technological Development and Innovation of the Republic of Serbia of the Republic of Serbia recognized the importance of this Conference, and thus, supported it.

The aim of this Conference is to consider current technical, technological, economic, ecological, R&D, legal and other issues related to the textile industry, then the application of contemporary achievements and the introduction of technical and technological innovations in the production process of fiber, textile, clothing and technical textile by applying scientific solutions in order to improve the business and increase the competitive advantages of the textile industry on the domestic and global market.

Leading scientists and experts from the Balkans and other countries, working at faculties, textile colleges and institutes, but also individuals who professionally deal with the issues at hand are taking part in this Conference.

The Conference program involves papers dedicated to the scientific and practical aspects of the following topics: Textile and Textile Technology, Textile Design, Management and Marketing in the Textile Industry and Ecology and Sustainable Development in the Textile Industry. The Conference program includes 54 papers, and a total of 132 participants from 16 countries: Albania, Bosnia and Herzegovina, Bulgaria, Croatia, Germany, Greece, India, Latvia, North of Macedonia, Portugal, Russia, Serbia, Spain, Slovenia, Turkey and Ukraina.

Therefore, this Conference is an opportunity for establishing scientific, educational and economic cooperation of our country with other countries. Certain number of papers by domestic authors present the project results dealing with fundamental research and technological development, financed by the Ministry of Education, Science and Technological Development of the Republic of Serbia.

I would like to thank all those who have made it possible to organize the conference Contemporary Trends and Innovations in the Textile Industry and make it a success. First, I would like to thank the Scientific and Organizing Committee for working hard, spending countless hours and finding the best solutions for numerous organizational aspects of our Conference. Also, I would like to express my gratitude to all sponsors who believed in the importance of this Conference and co-financed it. I also thank all the other institutions that supported the Conference in various ways, because without their support, the Conference could not have been organized. Last but not least, I would like to thank plenary lecturers, all authors and co-authors and guests for their participation in the Conference.

On behalf of the Organizing Committee

Prof. dr Snežana Urošević, president



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TABLE OF CONTENTS

PLENARY LECTURES	1
Radomir Folić, Damir Zenunović TEXTILE REINFORCED CONCRETE	3
Marija Vukčević, Biljana Pejić, Marina Maletić, Katarina Trivunac, Mirjana Ko CONTRIBUTION TO THE CIRCULAR ECONOMY THROUGH THE UTILIZATION OF FIBROUS TEXTILE WASTE AS BIOSORBENTS FOR WATER PURIFICATION	stić
Andreja Rudolf, Zoran Stjepanović, Tadeja Penko DIGITAL VERSUS REAL DRAPING OF TEXTILES	27
Iva Ćurić, Doris Curiš, Leonarda Bambić, Davor Dolar ENZYMATIC TREATMENT OF DYED COTTON KNITTED FABRIC WITH REUSED TEXTILE WASTEWATER	37
Ineta Nemeša, Marija Pešić THE MAIN CHARACTERISTICS OF SMART SEWING MACHINES	45
Kosana Vićentijević SUSTAINABILITY REPORTING IN THE TEXTILE INDUSTRY	62
CONFERENCE PAPERS SESSION I	 71
Irem Çelik, Gizem Karakan Günaydın EVALUATION OF YARN PROPERTIES PRODUCED FROM MECHANICAI COMPACT AND CONVENTIONAL RING SPINNING SYSTEMS	73
Matea Korica, Mirjana Kostić THE INFLUENCE OF LONG-TERM AGING ON THE PROPERTIES OF TE OXIDIZED COTTON	 MPO 82
Hülya Kesici Güler, Funda Cengiz Çallioğlu EFFECT OF PHASE RATIO ON CORE-SHEATH COMPOSITE NANOFIBER MORPHOLOGY	 92
Ana Santiago, Cláudia Pinheiro, Nuno Belino DEVELOPMENT OF NEW TEXTILE APPLICATIONS USING NETTLE FIBRE	101



Emilja Tosnikj, igor Jordanov
CHITOSAN-BASED DIFFERENTLY OXIDIZED COTTON YARNS ANALYZED BY FTIR-ATR 108
Koviljka Asanovic, Nadiia Bukhonka, Tatjana Mihailovic, Isidora Cvijetic, Mirjana Reljic, Mirjana Kostic
INFLUENCE OF PILLING ON THE STRUCTURAL CHARACTERISTICS OF FLAX PLAIN WEFT-KNITTED FABRICS 115
Jovana Stepanović, Tatjana Šarac, Nataša Stamenković, Nenad Ćirković, Dušan Trajković, Jovan Stepanović
THE INFLUENCE OF TENSILE FORCE ON THE MORPHOLOGICAL CHARACTERISTICS OF TEXTURED POLYESTER MULTIFILAMENT
YARNS 125
Hülya Kesici Güler, Funda Cengiz Çallioğlu
PCL BASED NANOFIBROUS WOUND DRESSING DESIGN 130
Mokina Anna Y, Ulme AndraMODERN ECOTEXTILE IN FASHION AND INTERIOR138
Damjana Celcar, Tanja Pilar
ACOUSTIC TEXTILES FOR INTERIOR 140
Liudmyla Melnyk, Olena Kyzymchuk
PROSPECTS OF FUNCTIONAL TEXTILE PRODUCTS FOR REHABILITATION AFTER LIMB AMPUTATION 156
Lazaros Firtikiadis, Prodromos Minaoglou, Nikolaos Efkolidis, Panagiotis Kyratsis
SLIPPER-SOLE COMPUTATIONAL DESIGN: A CUSTOMER-BASED162APPROACH162
Nikola Ilanković, Dragan Živanić, Mirko Katona ANALYSIS OF THE CONVEYOR BELT TENSILE PROPERTIES 173
Marija Stoicovska, Frosina Vielickova, Sara Srebrenkoska, Svetlana Risteska, Sanja Risteski, Vineta Srebrenkoska
UTILIZING VACUUM BAGGING PROCESS TO PRIPARE CARBON/EPOXY COMPOSITE LAMINATES 181
Teodora Gvoka, Gojko Vladić, Nemanja Kašiković, Katarina Maričić, Gordana Bošnjaković, Savka Adamović
THE UTILIZATION OF WASTE TEXTILE MATERIALS IN THE PACKAGING PRODUCTION 186



Maja Jankoska, Ruzica Stevkovska-Stojanovska APPLICATION OF FUTURISTIC DESIGN IN WOMEN'S CLOTHING MODELING

CONFERENCE PAPERS SESSION II

203

194

Gordana Kokeza, Sonja Josipović, Snežana Urošević POSSIBILITIES OF APLICATION OF THE GREEN ECONOMY CONCEPT IN THE BUSINESS OF THE TEXTILE INDUSTRY 205

Dragan Dimitrijević, Natalija Dimitrijević, Jasmina Gligorijević E-BUSINESS IMPLEMENTATION TRENDS IN SMALL AND MEDIUM-SIZED ENTERPRISES OF THE TEXTILE AND CLOTHING INDUSTRY 216

Bruno Završnik THE IMPORTANCE OF INFLUENCER MARKETING IN FASHION CLOTHING BRANDS 229

Njegoš Dragović, Snežana Urošević, Milovan Vuković, Dragoljub Đorđević, Zoran Stević DECRISING POLUTION IN TEXTILE INDUSTRY WITH RECYCLED AND 236

NEW MATERIALS

Biljana Popović, Adela Medović Baralić, Tijana Adžić, Ljiljana Sretković ECOLOGICAL PROCEDURES AND TECHNOLOGIES IN THE TEXTILE **INDUSTRY**

Gordana Čolović, Nikola Maksimović, Aleksandar Marković APPLICATION OF ERGONOMIC STANDARDS TO INCREASE SAFETY AND HEALTH IN THE GARMENT INDUSTRY 260

Ivana Mladenović-Ranisavljević, Violeta Stefanović, Aleksandar Savić, Snežana Urošević, Milovan Vuković

IMPLEMENTATION OF THE 5S METHOD IN OPTIMIZING THE WORKPLACE - A TEXTILE TESTING LAB

Gordana Čolović, Danijela Paunović, Nikola Maksimović, Mina Paunović WORKPLACE INNOVATION

Olga Stojanović, Kristina Savić, Marija Savić Pojužina, Jelica Simeunović DETERMINING THE OPTIMUM SERIES SIZES FOR PRODUCTION IN THE CLOTHING INDUSTRY 276

272

264

247



Bruno Završnik FEMALE SHOPPING BEHAVIOR TOWARDS FASHION HANDBAGS			
Andrea Dobrosavljević, Snežana Urošević, Đorđe Nikolić BUSINESS PROCESS ORIENTATION AS A BASE FOR ORGANIZATIONAL DEVELOPMENT OF TEXTILE INDUSTRY SMEs	292		
Sanja Risteski, Silvana Zhezhova, Saska Golomeova Longurova, Sonja Jordeva FASHION IN SPAIN AS AN INSPIRATION FOR DESIGN AND CONSTRUCT OF CONTEMPORARY CHILDREN'S CLOTHING	'ION 299		
Dragana Frfulanović, Milena Savić			
ART OF THE 20TH CENTURY AS INSPIRATION IN FASHION DESIGN	308		
Marina Jovanović WOMEN IN THE FUNCTION OF ORGANIZATIONAL COMMITMENT IN T MANUFACTURE OF PIROT RUGS	Ъ ЭНЕ Э15		
Miličić Ljiljana, Mina Paunović			
SUSTAINABILITY IN FASHION-MODULAR CLOTHING INSPIRED BY TH NATIONAL COSTUMES OF DISTANT NATIONS	E 325		
Maja Jankoska, Ruzica Stevkovska-Stojanovska MODELING OF WOMEN'S PANTS INSPIRED BY 60s AND 70s	332		
Kristina Savić, Olga Stojanović, Marija Savić Pojužina, Jelica Simeunović DEVELOPMENT TENDENCIES OF UNIFORMS FOR POLICE OFFICERS O MOTORCYCLES	N 341		
Vasilije Petrović, Dragan Đorđić, Samir Pačavar, Anita Milosavljević, Predrag Pecev, Marija Petrović			
DUAL MODEL OF TRAINING TEXTILE ENGINEERS FOR THE NEEDS OF THE 4.0 INDUSTRY	350		
CONFERENCE PAPERS SESSION III	359		
Emilija Toshikj UNCONVENTIONAL YARN STRUCTURE	361		
Dušan Nešić APPLICATION OF SELF-ADHESIVE POLYETHYLENE FOAM (PE) AS A SUBSTRATE OR PROTECTION FOR CONDUCTORS	370		



Ivanka Ristić, Aleksandra Mičić, Nebojša Ristić	
CHEMICAL CATIONIZATION OF COTTON FABRIC FOR IMPROVED ABSORPTION OF DIRECT DYES	374
Mehmet Cağlayan, Banu Hatice Gürcüm	
DEMATEL METHOD FROM MULTI-CRITERIA DECISION MAKING	
METHODS AND EXAMPLES IN TEXTILE FIELD	383
Marina Maletić, Marija Vukčević, Biljana Pejić, Mirjana Kostić, Aleksandra	
Perić Grujić	
ORGANIC DYES ADSORPTION ON CARBON ADSORBENTS DERIVED F	ROM
WASTE COTTON AND COTTON/POLYESTER YARN	394
Svitlana Arabuli, Anastasiia Truba, Arsenii Arabuli, Liudmyla Hanushchak-	
OPTICAL PROPERTIES OF MODIFIED FABRICS INTENDED FOR	
PROTECTION AGAINST UV RADIATION	403
Đorđić Dragan, Petrović Vasilije, Samir Pačavar, Enis Kolić, Anita Milosavljev ANALYSIS OF MECHANICAL CHARACTERISTICS AND WATER VAPOF PEDMEABILITY OF FABRICS USED FOR MAKING THE LINING OF	ic {
GARMENTS	409
Silvana Zhezhova, Sonja Jordeva, Sashka Golomeova Longurova, Sanja	
Risteska, Vanga Dimitrijeva Kuzmanoska	
FINAL CONTROL OF MEN'S SHIRTS	415
Julieta Ilieva, Mariya Koleva	
FROM NATURE TO FASHION – DIGITALIZED ELEMENTS AS INSPIRAT FOR PERSONALIZED CLOTHING AND ACCESSORIES	TON 424
Subrata Das, Amitabh Wahi, Nivetha Sharathi S	
COLOUR MATCHING OF TEXTILES USING DEEP LEARNING	
CONVOLUTIONAL NEURAL NETWORKS	434
Marija Petrović, Samir Pačavar, Vasilije Petrović, Dragan Đorđić, Anita	
Milosavijevic, Predrag Pecev	OF
CLOTHING PRODUCERS	441
 Dimitrios Chaidas, Tatjana Spahiu, John Kechagias	
AN INVESTIGATION OF 3D PRINTING ON THLLE TEXTILE CHANGING	
THE PLATFORM LEVELLING	451



Adela Medović Baralić, Biljana Popović, Ana Krkobabić, Ljiljana Sretković NEW TRENDS IN THE TEXTILE INDUSTRY		
Tatjana Spahiu, Andrea Ehrmann, Henrique Almeida, Antonio Jimeno-		
Morenilla, Panagiotis Kyratsis		
FOOTWEAR PRODUCTS AND THE ROLE OF INDUSTRY 4.0	463	
FOR SUSTAINABLE PRODUCTION		
Author index	473	
FRIENDS OF THE CONFERENCE	477	





PLENARY LECTURES





VI International conference "Contemporary trends and innovations in the textile industry" 14-15th September, 2023, Belgrade, Serbia



UTILIZING VACUUM BAGGING PROCESS TO PRIPARE CARBON/EPOXY COMPOSITE LAMINATES

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ABSTRACT: In the frame of this work, composite laminates based on unidirectional carbon/epoxy prepreg were produced by using of by Automated Tape Laying (ATL)/ Automated Fiber Placement (AFP) and vaccum bagging processes. This research based on unidirectional carbon fiber/epoxy composites shows the effect of fiber architecture on mechanical properties using a automated tape/fiber laying procedure followed by the vacuum bagging process method. The properties of the prepreg material have been tested and some mechanical properties of the obtained composite laminates has been performed. The vacuum bagging method showed improvement in tensile strength and modulus.

Keywords: automation, layup, vaccum bagging, prepreg, laminates.

KORIŠĆENJE PROCESA VAKUMSKOG PAKOVANJA ZA PRIPREMU UGLJENIK/EPOKSI KOMPOZITNIH LAMINATA

APSTRAKT: U okviru ovog rada proizvedeni su kompozitni laminati na bazi jednosmernog ugljenik/epoksidnog preprega primenom procesa automatskog postavljanja traka (ATL)/ automatsko postavljanje vlakna (AFP) i vakumskog pakovanja. Ovo istrazivanje zasnovano na jednosmernim karbonskim/epoksidnim kompozitima pokazuje efekt arhitekture vlakana na mehanička svojstva korišćenjem automatizovane procedure polaganja trake/vlakana praćene metodom procesa vakumskog pakovanja. Ispitane su osobine prepreg materijala i određene mehaničke osobine dobijenih kompozitnih laminata. Metoda vakumskog pakovanja pokazala je poboljšanje zatezne čvrstoće i modula.

Ključne reči: automatizacija, polaganje, vakumsko pakuvanje, prepreg, laminati.

1. INTRODUCTION

In the automotive, aerospace and ship industry, carbon/epoxy composites are used extensively since the mechanical strength, chemical resistance and service temperatures requirements are fulfilled apart from higher fatigue strength and higher corrosion



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resistance [1,2]. In recent years, prepreg material, which is an intermediate material has gained its importance in manufacturing structural composite components as these materials allow to adjust reinforcement positioning, thickness, number of layer and their orientations in the molding. Composites made by hand lay-up process can be seen in several works as shown in the literature [3]. However, composites fabricated by prepregs utilize hand lay-up process that leads to non-uniform deposition of epoxy. Parts manufactured by using prepregs are clean and offer ease in manufacturing of complex geometries with almost 60% fiber volume fraction. Hence, in this work, our objective was to make prepreg based composites with optimized curing cycle [4].

The structures made of advanced composites have been majorly manufactured by hand layup of prepreg tapes to produce composite parts that are finalized by a consolidation and a curing process in an autoclave. However, widespread use of composite materials for aerospace, automotive and other applications has been limited due to high manufacturing costs, long processing times and size limitations of an autoclave. To achieve both desired quality and lower costs, for manufacturing high performance composite structures, the applying of automated manufacturing process with advanced prepreg systems has to be done [5-7]. This method rely on the robotized layup of bands of unidirectional reinforcement material allowing more accurate and more repeatable manufacturing process [6-9]. In this group, two main methods can be identified Automated Tape Layup (ATL) and Automated Fiber Placement (AFP). ATL makes use of a robotic arm to layup tapes (up to 150mm wide) of unidirectional prepeg and benefits from high productivity for large and simple flat parts. Therefore, the manufacturing of more complicated parts can be handled by AFP but with a lower productivity than ATL. For ATL and AFP processes, one of the manufacturing issues is the determination of successive courses trajectories [10-13].

In this paper, study of the influence of fiber orientation on the structural and mechanical properties of laminate manufactured with help of ATL technology and vacuum bagging process is presented.

2. EXPERIMENTAL

The material used in this paper was unidirectional (UD150 IM7 12K/M21) carbon fiber/epoxy prepreg with a 34 wt.% nominal resin content (Hexcel, France). Prepreg is a semiproduct consisting of reinforcing fibers and thermosetting or thermoplastic polymer matrix. This material can be further processed at a certain temperature depending on the polymer matrix and the appropriate pressure for forming a composite structure with certain strength characteristics.

The composite laminates were produced by using a laser assisted tape layng head, manufactured by Mikrosam, Macedonia. Head is attached to a robot arm, as it is shown in Specimens unidirectional laminates were produced with 8/14 layers of UD prepreg and they were processed at appropriate temperatures with vacuum bagging technology (show at Figure 1).





Figure 1: Vacuum Bagging of composite laminates/ Curing curve for vacuum bagging: time – temperature

Manufactured composite laminates had different angles on UD prepreg placement. Codes of unidirectional laminates manufactured with different design (laying angle) are shown in table 1.

Properties of manufactured prepreg					
Volatile content (%)	e content (%) <1				
Mass resin content (%)	30-36				
PAW (g/m ²)	300				
Prepared unidirectional laminates					
Codes of composite	Parameter level				
samples	technology of placement	laying angle			
1	ATL	0			
2	ATL	90			
3	AFP	0			
4	ΔED	90			

Table 1: Properties of prepreg and prepared unidirectional laminates

Tensile properties of the composites such as tensile strength and the modulus (ASTM D 3039) were determined. The properties of manufactured samples were determined with test in accordance with the procedure described in standard. For that purpose computer controlled universal testing machine (UTM) Hydraulic press, SCHENCK- Hidrauls PSB with maximal load of 250 kN, constant crosshead speed of 5 mm/min was used. Load and displacement were recorded by an automatic data acquisition system for each sample. Minimum four reproducible tests were conducted for each sample at room temperature.

3. RESULT AND DISCUSSION

Manufactured composite samples were clamped and tensile tests were performed. The tests were closely monitored and the load at which completed fracture of the specimen occurred has been accepted as breakage load. Load-displacement curves were plotted for every sample and values for stress, strain and module of elasticity were calculated as





VI International conference "**Contemporary trends and innovations in the textile industry"** 14-15th September, 2023, Belgrade, Serbia

average. The flexural stress (σ) in the outer surface of the test specimens occurred at the midpoint. These stresses were determined from the relation [16]:

$$\sigma = \frac{F}{A} \tag{1}$$

Where, σ is the Tesnsile strength (MPa), *F* is the load (N), A is the area (mm²), *b* is the width of the specimen (mm), and *h* is the thickness of the specimen (mm). Tensile modulus of elasticity (*E*) and strain (ε) of the composite specimens were determinate using equations according standard [16].

Table 3 shows a summary of the flexural properties for composite laminates.

Received results from performed tests on laminated composite samples are given in table 2, where maximal flexural strength of 1925.85 MPa for sample N°3 and minimal flexural strength of 7.33 MPa for sample N°6 can be observed.

Comparison between results of specimens manufactured at same technological parameters, but different fiber orientation can give a notice that all samples tested at UD direction (0^0) had performed better mechanical properties in comparison to the samples tested at CD direction (90^0) . This means that fiber direction directly affects Tensile strength of laminated composite samples up to 99%. Received average results from tensile tests are shown in table 2.

	Tensile tests			
Samples	technology of placement	laying angle	Strength (MPa)	Modulus (GPa)
1	ATL	0	1546.21	17.33
2	ATL	90	147.31	7,11
3	AFP	0	1831.08	21.78
4	AFP	90	221.86	8,46

Table 2: Tensile properties for composite laminates

Comparison between results of specimens manufactured with different fiber orientation can give a notice that all samples tested at UD direction (0°) had performed better mechanical properties in comparison to the samples tested at CD direction (90⁰). This means that fiber direction directly affects tensile strength of laminated composite samples.

4. CONCLUSION

Experimental measurements of the tensile strengths of laminated samples for determined ranges of parameters have been carried. The results for tensile strength were analyzed as a function of the technology of placement, polymerization technology and laying angle. The experimental procedure described in the present work is sufficient to show the influence of the fiber architecture on the tensile properties of laminated samples . The test results indicated that the change of the laying angle cause a variation in the final mechanical results, whereas the influence of the technology of placement (AFP/ATL) is much lower, and the vacuum bagging method has showed improvement in tensile strength and modulus.





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