

# FIRST ASSESSMENTS REGARDING THE USE OF SPLISS IN VARIOUS CFD- RELATED APPLICATIONS

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# Intro: CFD and related simulations at DLR



- High Fidelity Computational Fluid Dynamics requires the use of **large computational resources in parallel**
- At least implicit methods require to (approximately) **solve large linear equation systems**
- There are different CFD codes (even within DLR) for different flow regimes, which can benefit from a shared development:
- Common library for (approximatively) solving a linear equation system with characteristics from aeronautical CFD
  - More focus on low-level performance and hardware technologies
  - May adapt to specific technologies more easily due to its comparably limited functional range



# Key features of a linear solver for aeronautical CFD



## Sparse block matrices

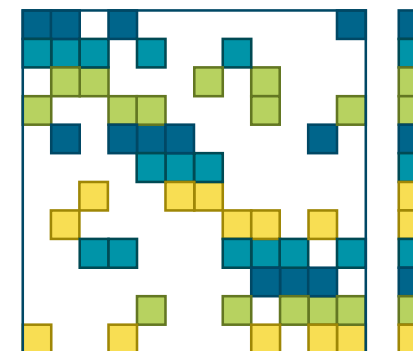
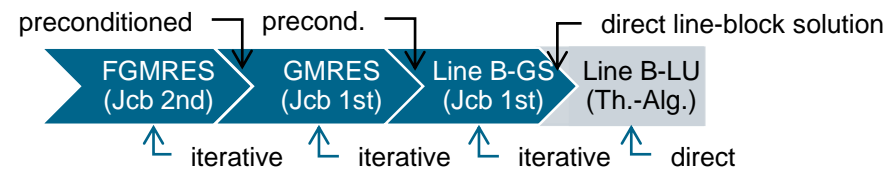
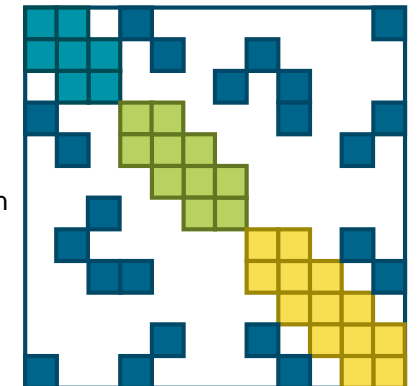
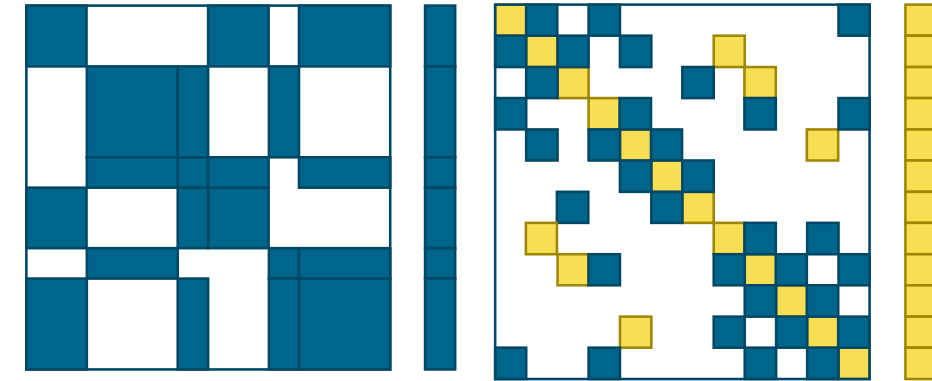
- Dense blocks with a fixed block size or variable block sizes
- Mixed data types: e.g. some entries are complex, others real, some multiscalars

## Solver

- Different components should be combinable (as preconditioner)
- Robust methods for stiff CFD problems:
  - Direct inversion of (generalized) diagonal blocks (LU/Thomas-Algorithm)
  - Jacobi, Gauss-Seidel, GMRES, linear multigrid, ...

## Efficient parallelization for HPC

- Distributed memory (e.g. MPI)
- Shared memory (Threading)
- GPU support
- Vector instructions (SIMD)
- Reduced memory footprint





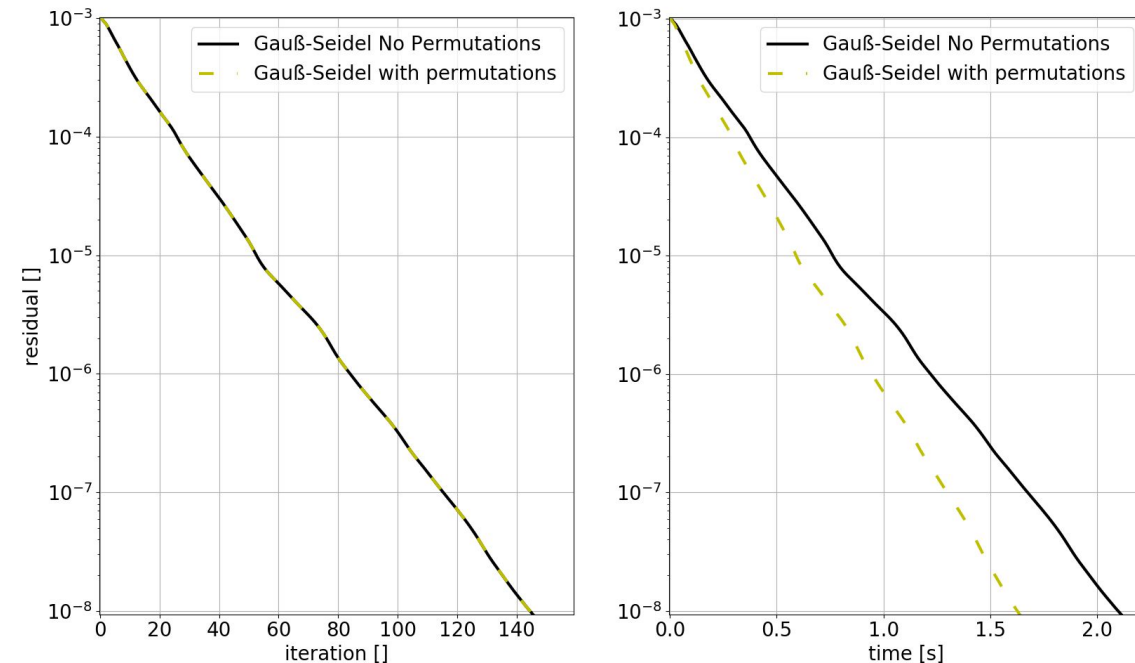
# ALGORITHMIC FEATURES

## Coloring

- Used for (otherwise sequential) solver components like Gauss-Seidel or ILU
- Allows **parallelism** while keeping results **independent from partitioning**

## Permutation

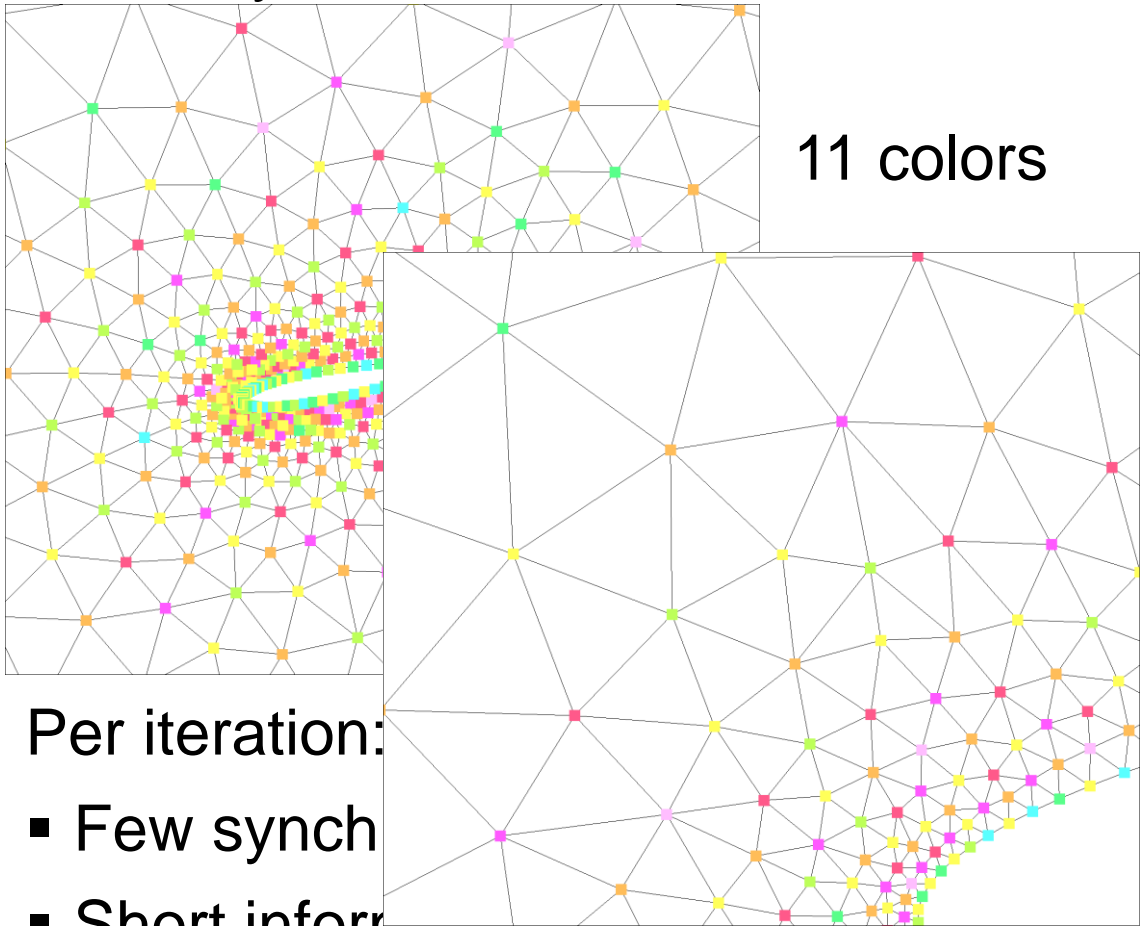
- For cache optimization, it can be beneficial to permute the matrix entries
- ➔ Use a permutation according to colors, when multi-colored algorithms are used



TRACE matrix, results from Alexander Bleh

# Comparison of different color selections for multi-color algorithms

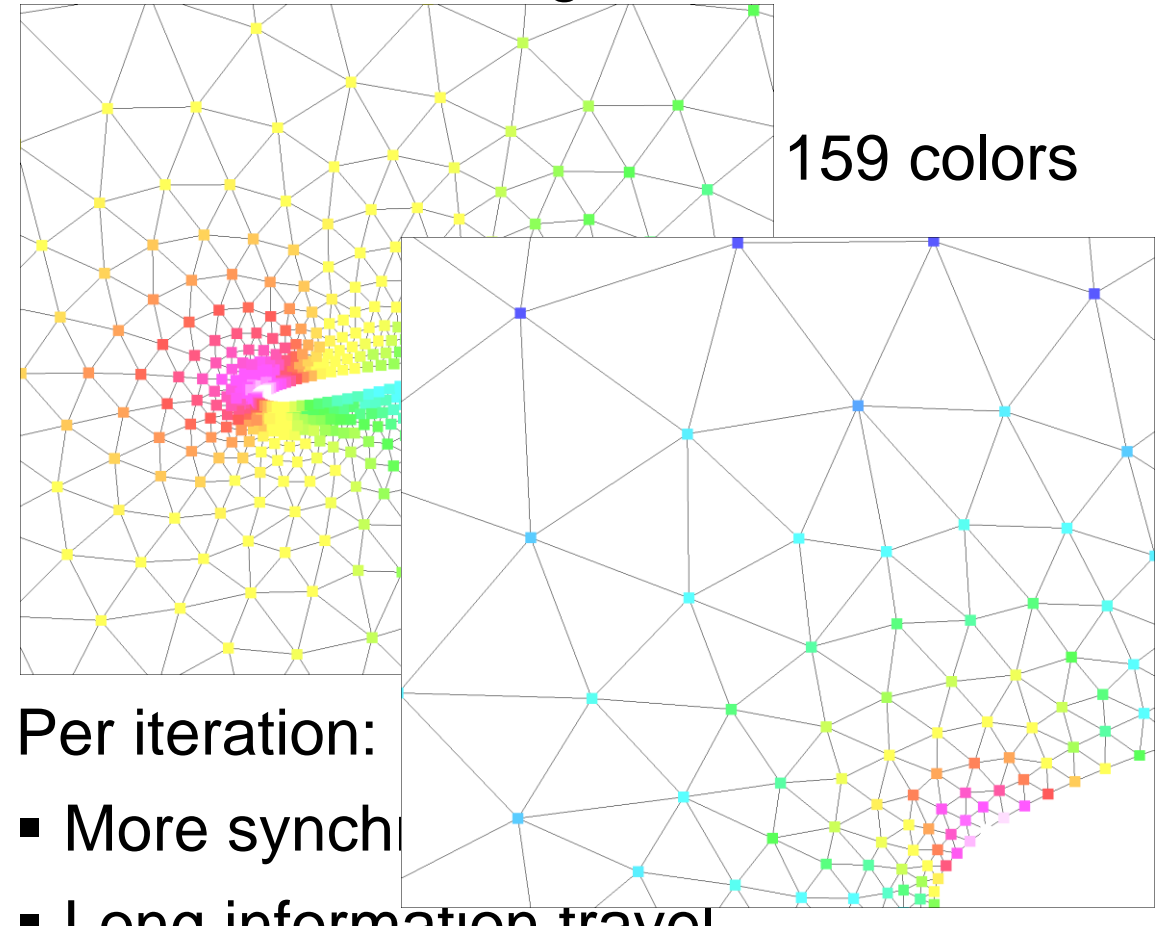
## ■ Greedy minimal number of colors



Per iteration:

- Few synchronizations
- Short information travel

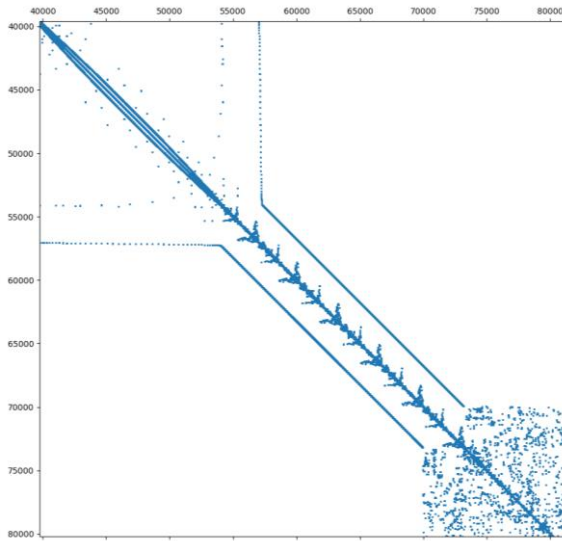
## ■ Reverse marching front



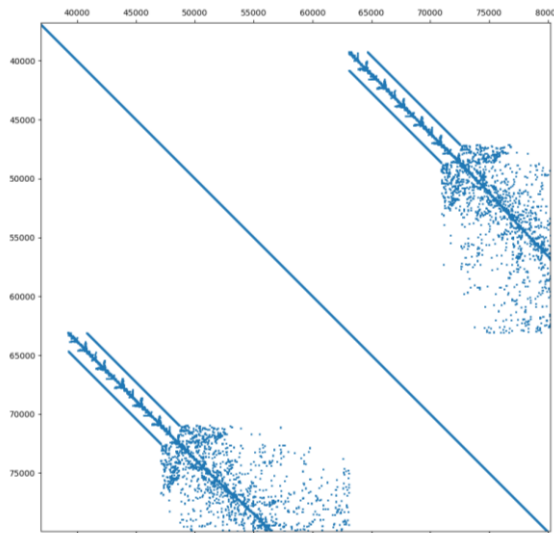
Per iteration:

- More synchronizations
- Long information travel

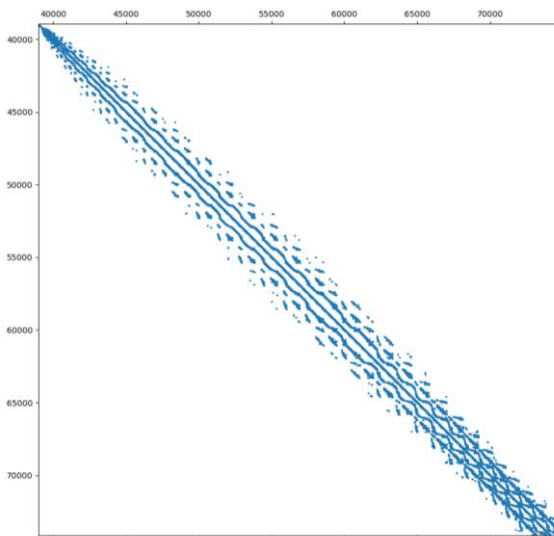
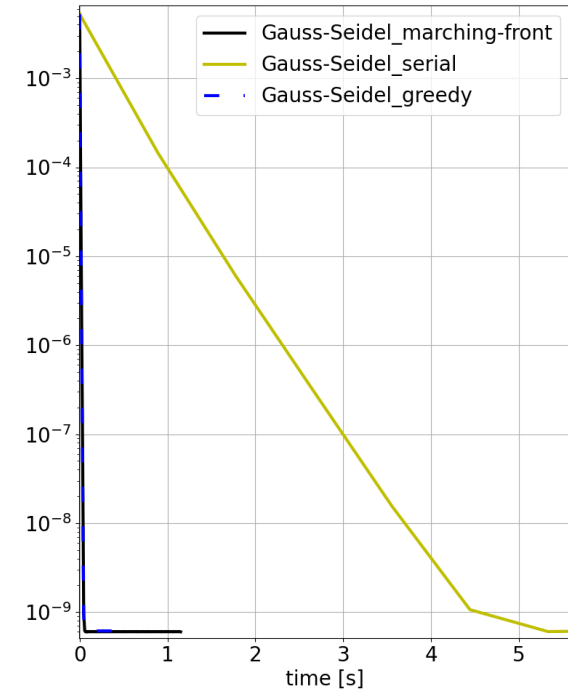
# TRACE: Different colors for symmetrical Gauss-Seidel



Serial



Greedy



Reverse marching front

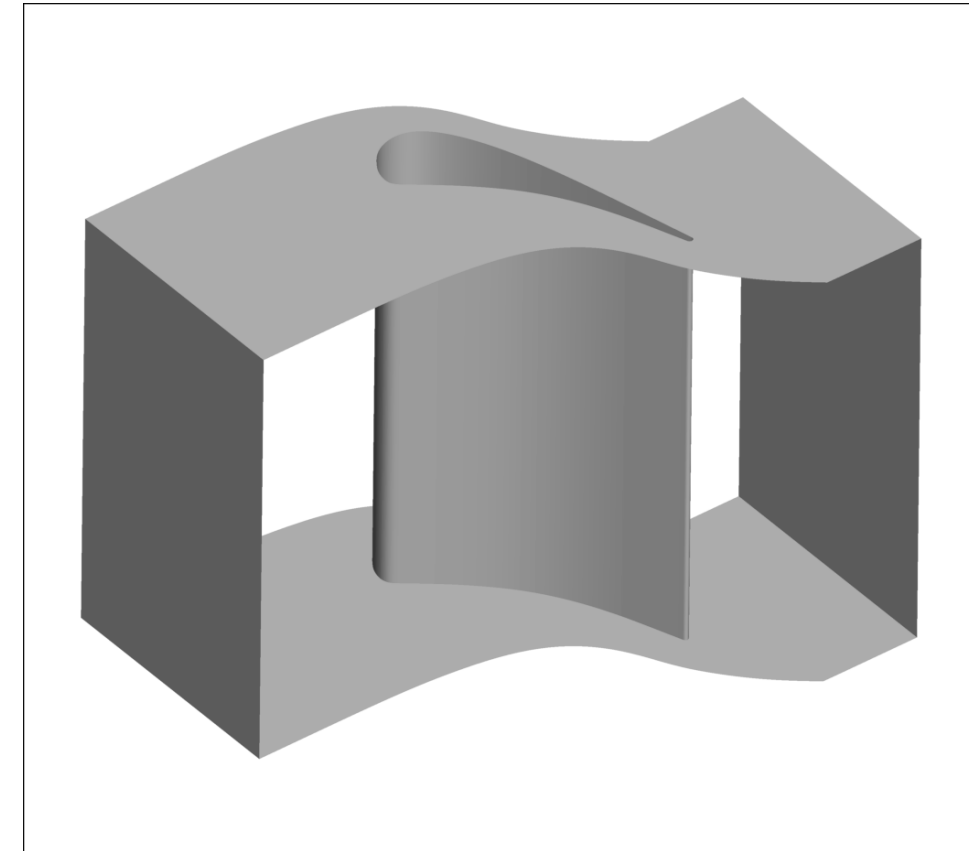
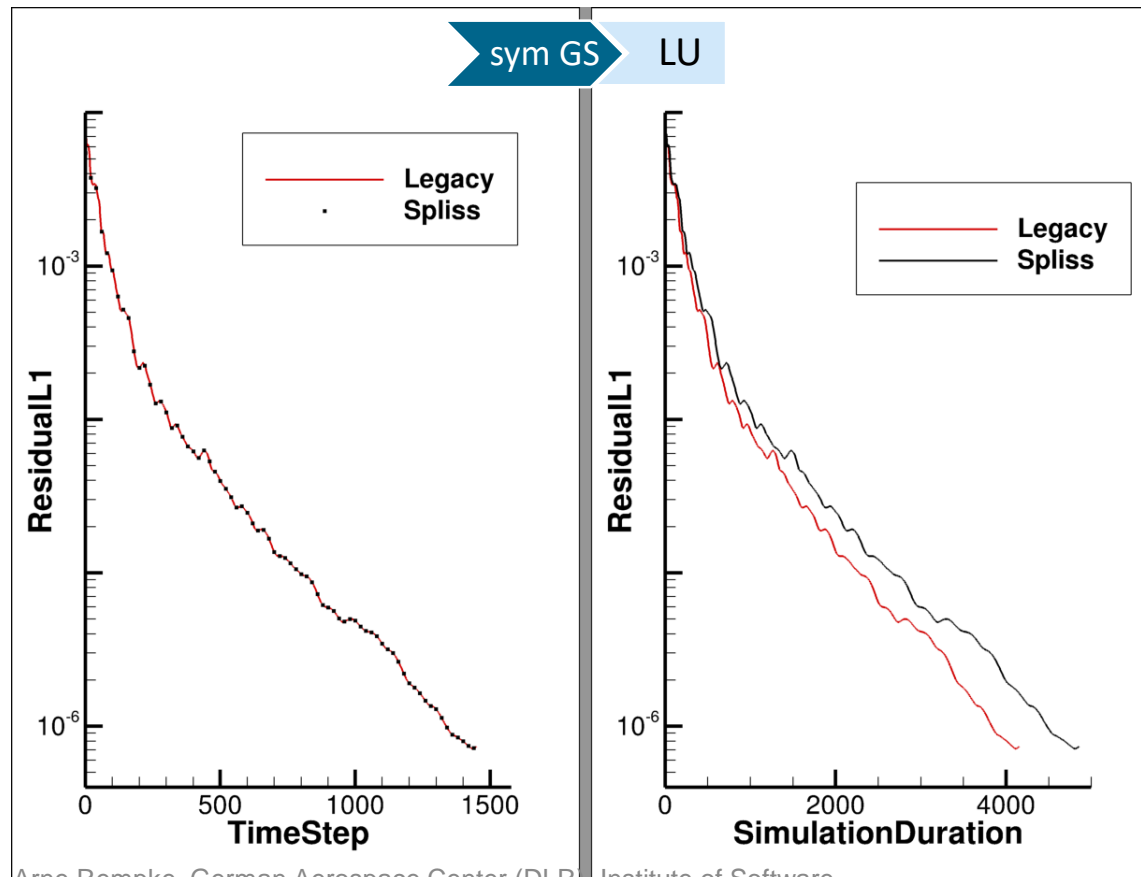
sym GS → BI

- TRACE case of transsonic bump
- hybrid unstructured/structured mesh
- 13e3 hexahedrons, prisms & tetraeder
- Euler 2nd order FV
- „Serial“ implementation uses a huge amount of colors

# TRACE VKI-LS89 Comparison to Legacy



- VKI-LS89 single blade configuration
- RANS- $k\omega$ ,  $Ma=0.92$ ,  $Re=2.1e6$
- Grid with  $2e6$  elements, FV scheme



- Focus on **replicating the algorithm** from Legacy to get equal results
  - Currently Spliss is 10% slower to TRACE Legacy
- Computation & results from Alexander Bleh & Jan Backhaus





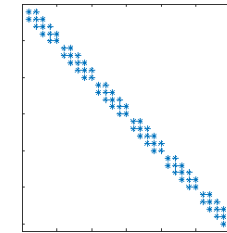
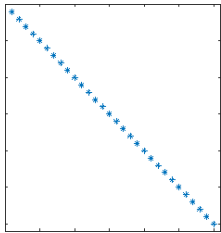
# CFD SPECIFIC ALGORITHMS

# Lines Inversion / Thomas Algorithm

- Jacobi-method uses a diagonal inversion:

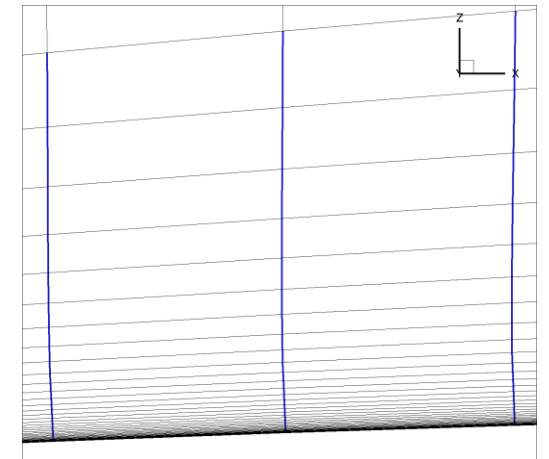
$$x^{(i+1)} := x^{(i)} + D^{-1}(b - Ax^{(i)})$$

where



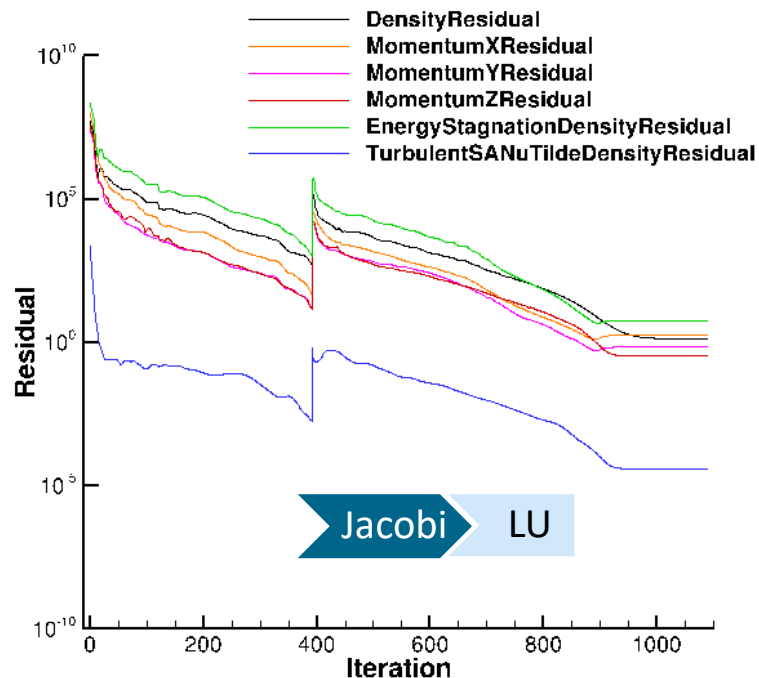
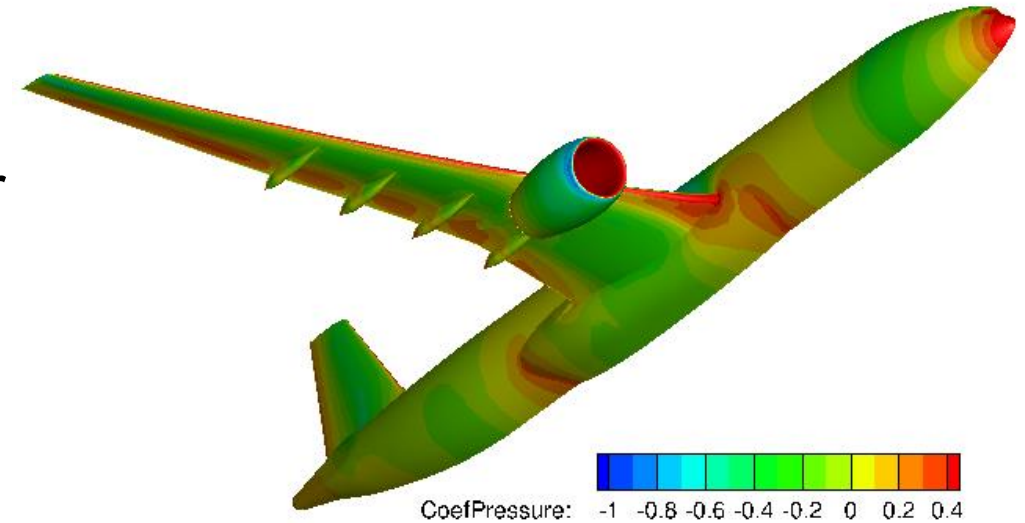
- $D := \text{diag}(A)$  (point-implicit)      or      ▪  $D := \text{tridiag}(A)$  (lines-implicit)

- Especially favourable/needed when mesh has very anisotropic cells, aspect ratios  $\geq 5000:1$

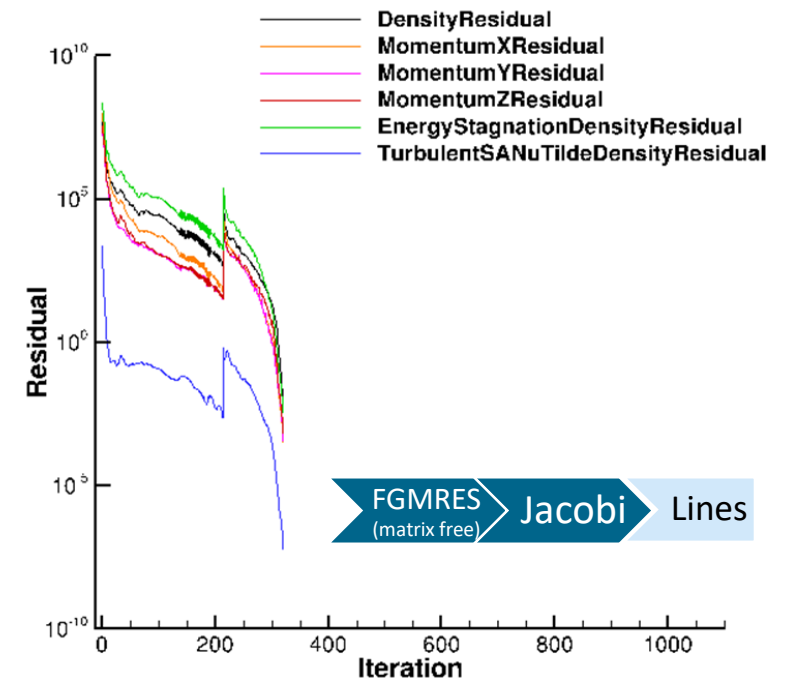


# CODA XRF1-V4 FV

- Airbus XRF1-V4 flow-through-nacelle (power off) configuration
- RANS-SAneg,  $Ma=0.86$ ,  $Re=25e6$
- Grid with  $32e6$  elements, FV scheme



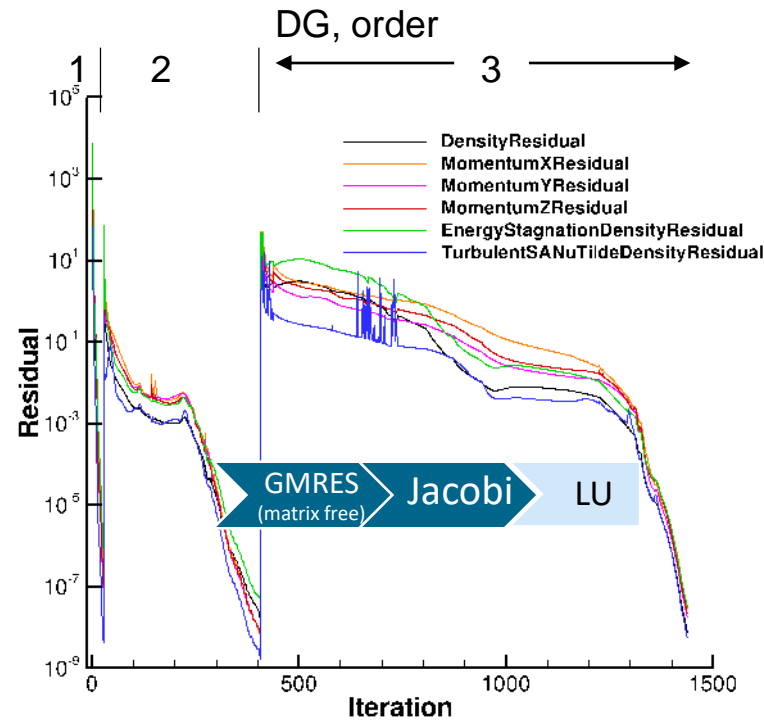
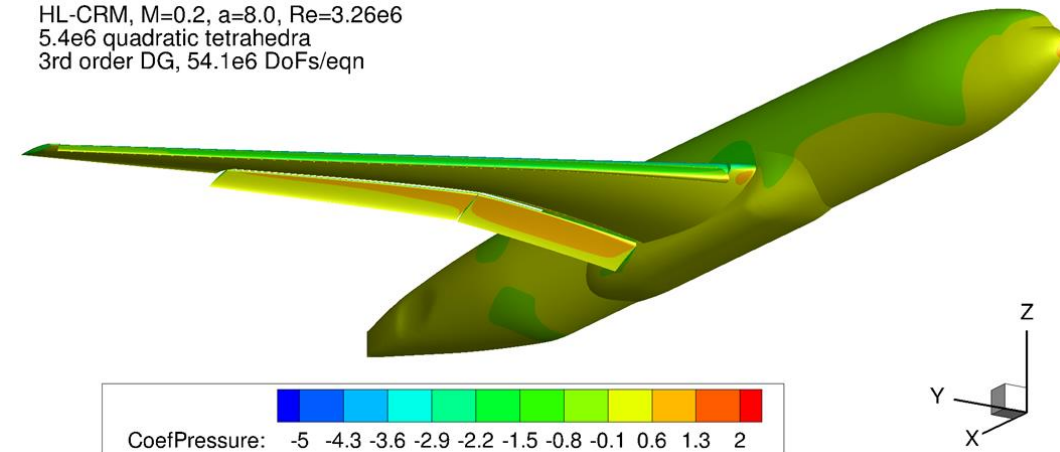
Reduction of time to solution by 73% on 256 cores



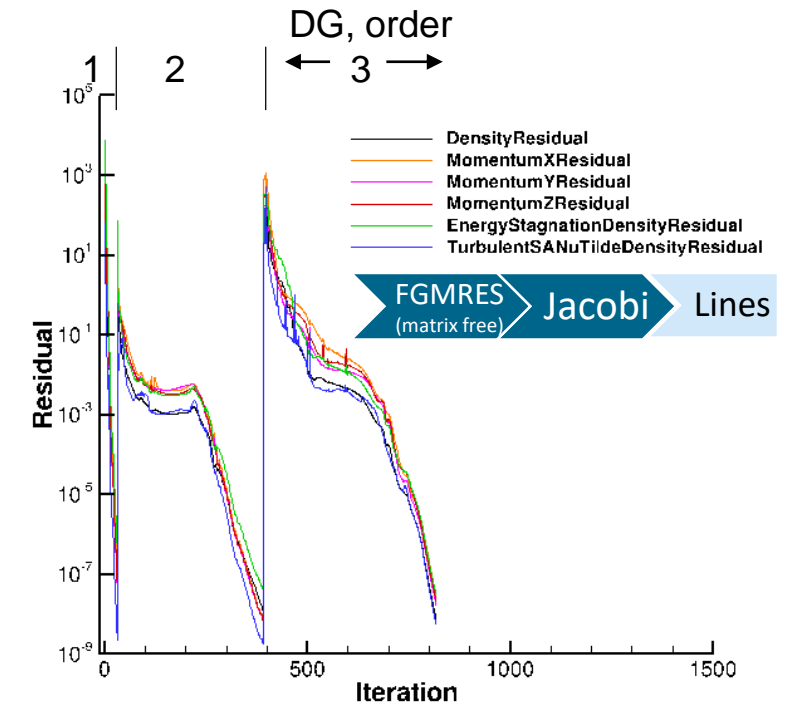
# CODA High-Lift CRM DG

- Geometry from High Lift Prediction Workshop 3
- RANS-SAneg, Ma=0.2, Re=3.26e6
- Curved Grid with 5.4e6 quadratic tetrahedra, DG scheme (3<sup>rd</sup> order: 54.1e6 DoFs/eqn)

HL-CRM, M=0.2, a=8.0, Re=3.26e6  
 5.4e6 quadratic tetrahedra  
 3rd order DG, 54.1e6 DoFs/eqn



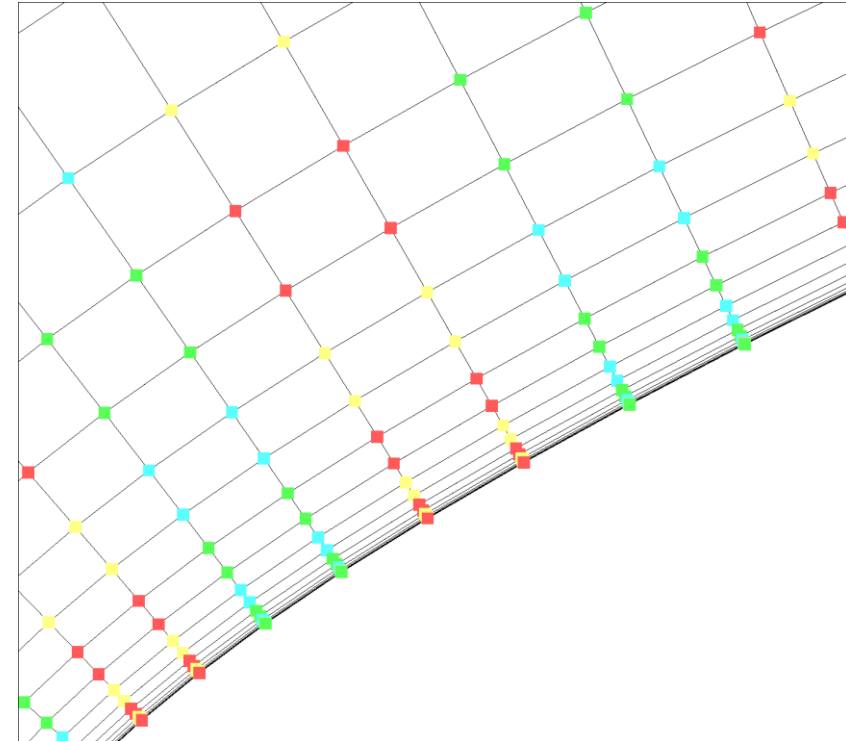
Reduction of time to solution by 56%



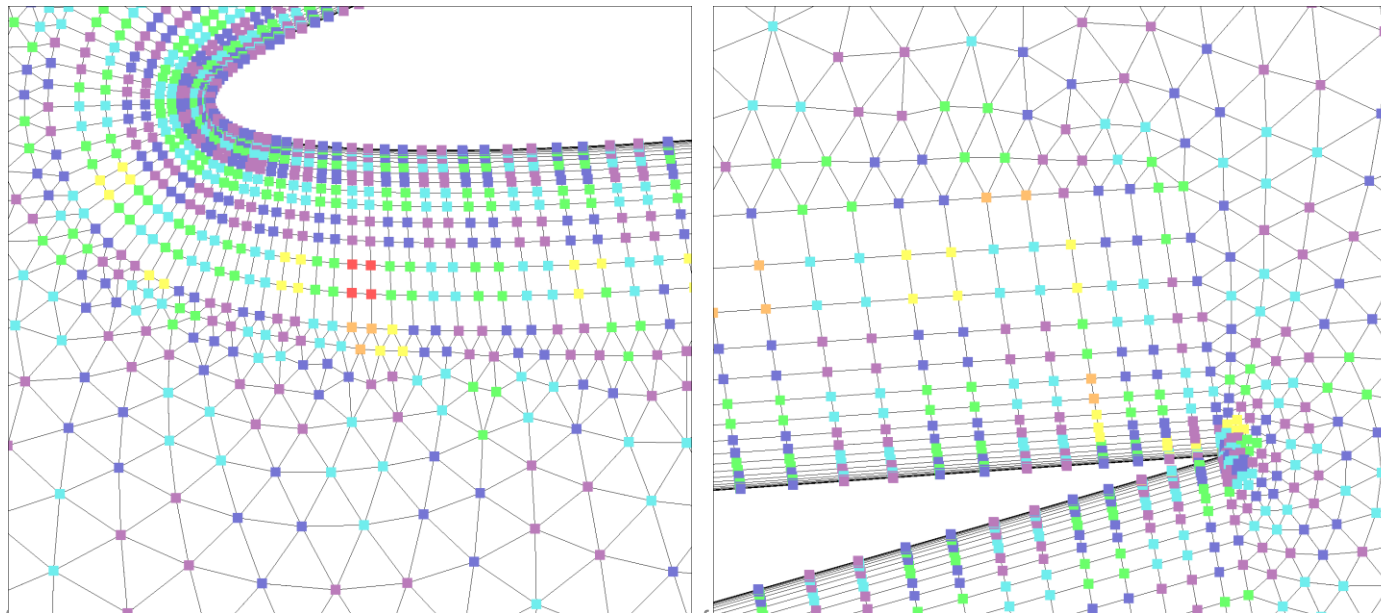
Computation & results from Ralf Hartmann

# Algebraic agglomerations visualized

- Agglomerations are computed simply by inspecting the matrix connectivity, not the values
- When the matrix blocks correspond to geometrical elements/vertices, the agglomerates can be visualized in the original mesh

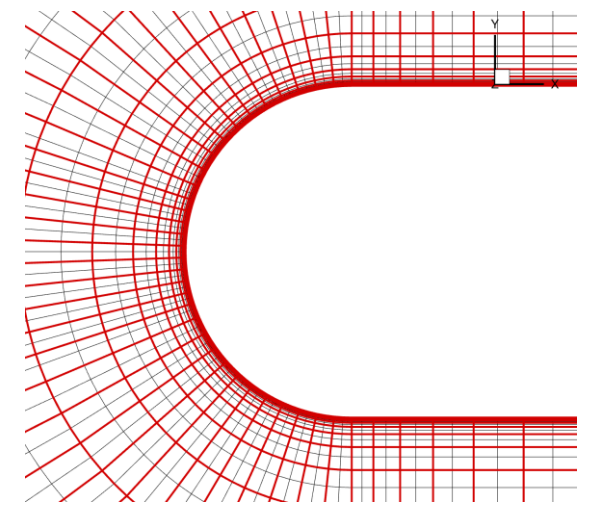
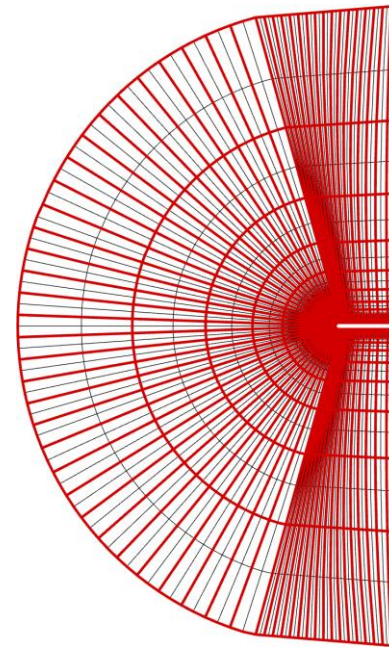
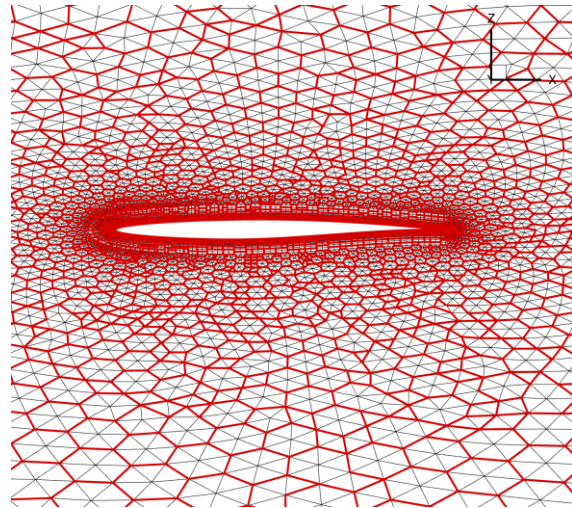
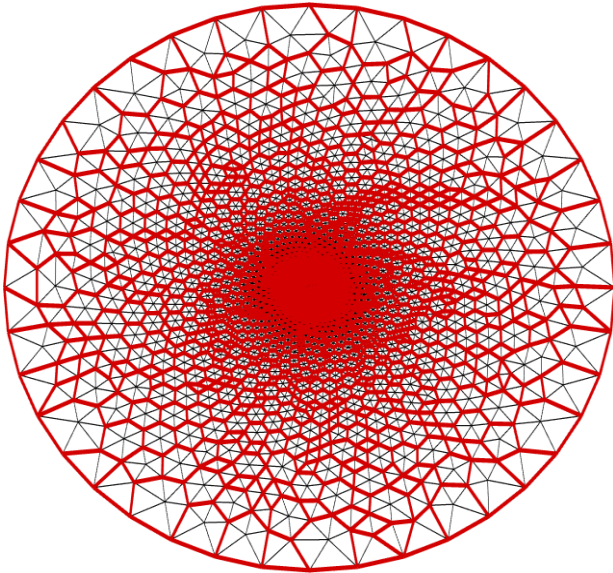


- First level agglomerates for a vertex-based discretization



# Algebraic agglomerations visualized

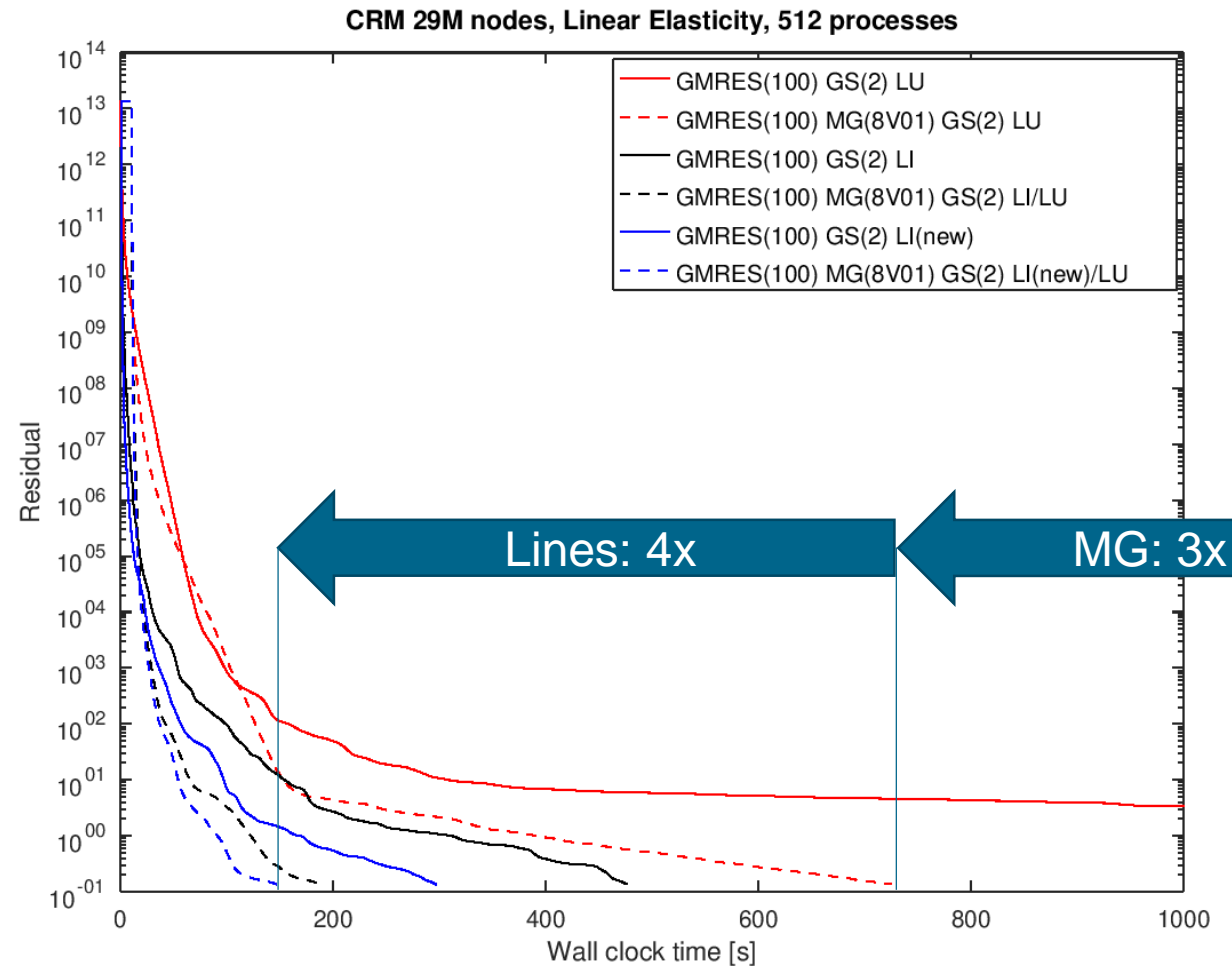
- First level agglomerates for a volume-based discretization (CODA FV)



# FSMeshDeformation: Efficiency of the tailored solver components



- Red solid curve is a „standard linear solver“
- Multigrid gives speedup of 2-3 (dashed)
- LinesInversion gives additional speedup of 3-4 (black/blue)

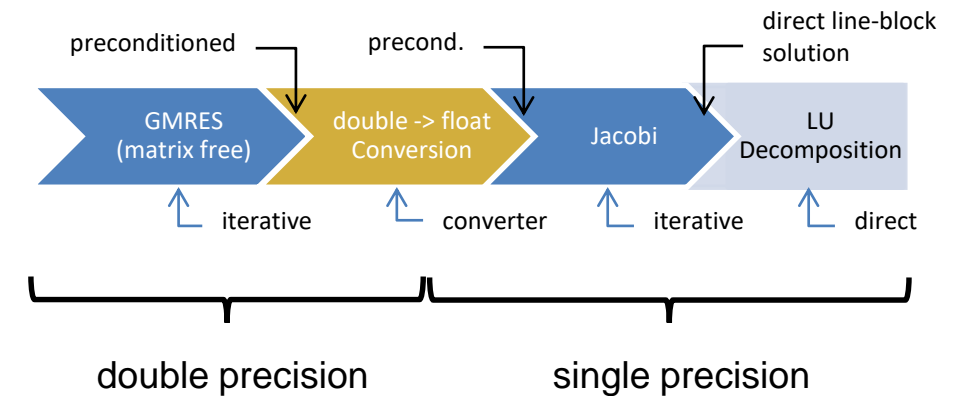
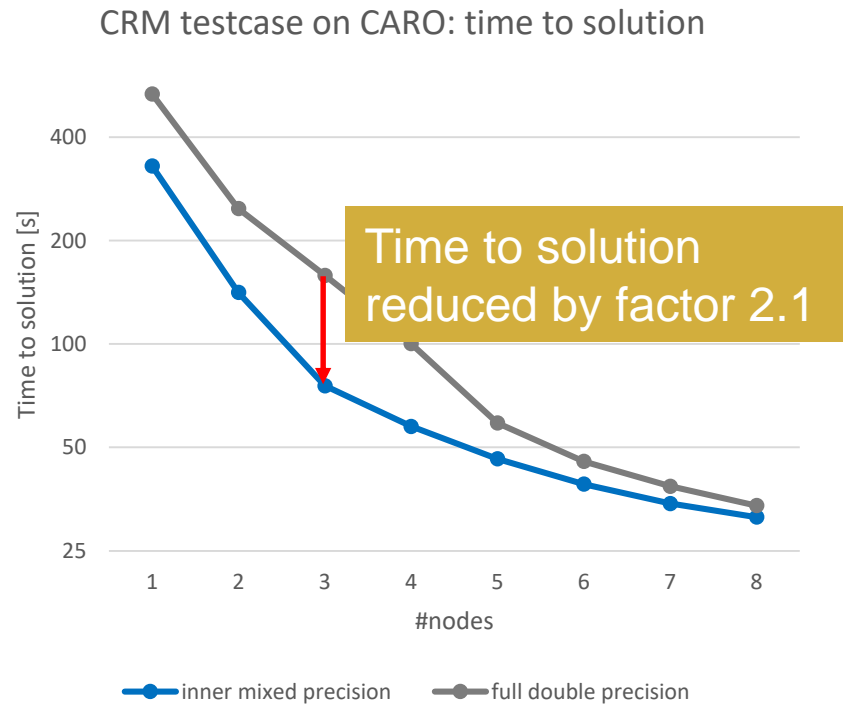


# PERFORMANCE, SCALING & ACCELERATORS



# Mixed precision

- Idea: Reduce memory footprint of inner hot loops since performance is memory bound

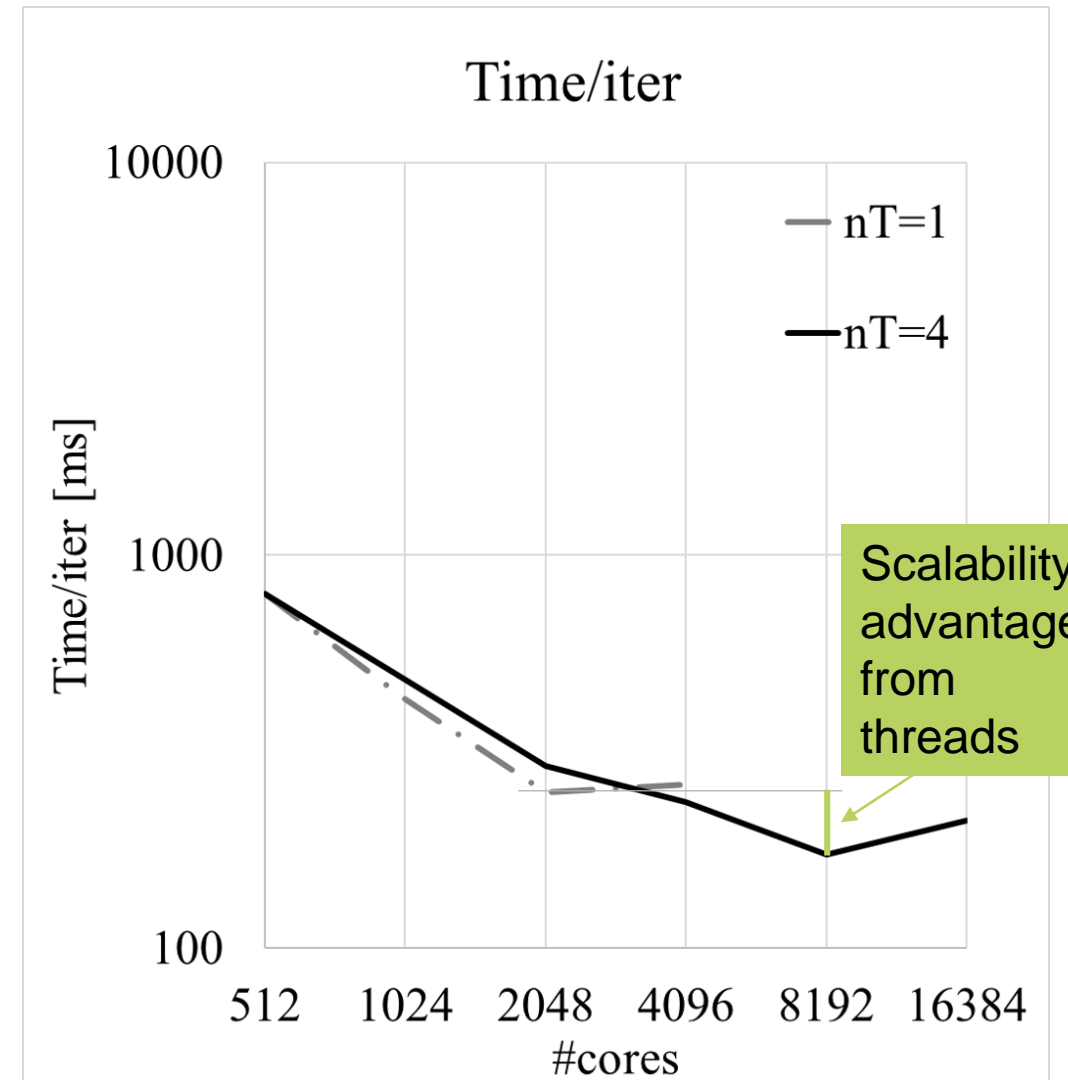


- User still provides matrix / input vectors and receives solution vector in double precision
- Inner Spliss solver components operate in float precision

# FSMeshDeformation: Strong scaling and multi-threading



- XRF1 test case, 31M nodes
- Linear elasticity mesh deformation
- CARO: 2xAMD EPYC 7702 («Rome», 64 cores, 2,0 GHz)
- GMRES Multigrid GaussSeidel configuration
- When using more than 2048 ranks, scaling becomes difficult (very much communication during solving, initialization/partitioner takes very long)
  - But: 2048 ranks can still employ more cores when using threads



# GPU Development

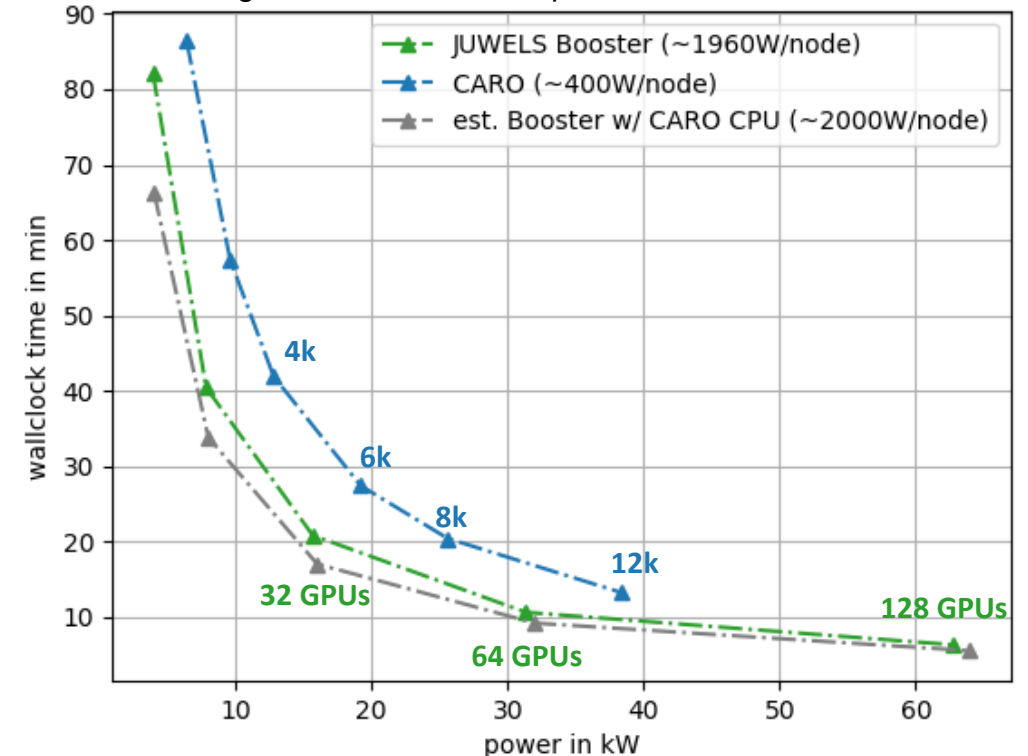
## Next gen GPUs



### Juwels Booster (Jülich)

- 4x Nvidia Tesla A100 per node
- Time to solution: speedup of 8-9 for same number of nodes on Juwels
  - Rather unfair, since on Juwels every process uses a GPU **in addition** to the CPU
- Energy comparison (seconds per used Watt): speedup of 1.6-1.9 on Juwels
- Hypothetical Juwels Booster node with CARO CPU: 1.8-2.3 speedup (energy-wise)

**Runtime** | CARO (AMD Rome) vs. Juwels (4x Nvidia A100)  
M6 wing, 69.2M elements, implicit Euler, Jacobi + Block Inv.



Results from Michael Wagner & Jasmin Mohnke

# Conclusion



- Spliss in use in CODA, TRACE, HYDRA, FSMeshDeformation
- Demanded features are supported and successfully demonstrated
- Regular exchange with users in order to still develop further and improve

**QUESTIONS?**