$6^{\text {th }}$ International Conference on Large Scale Applications and Radiation Hardness of Semiconductor Detectors

Firenze, September 29th - October 15t, 2003

## The PAMELA silicon tracker

Samuele Straulino
INFN and University, Firenze

## Outline

## I will talk about:

$>$ scientific objectives of the PAMELA experiment
$>$ tracking system's characteristics
$>$ development of the detectors
$>$ expected capabilities
$>$ current status of the apparatus





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## Assembling of detector planes in Firenze



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## Front-end and read-out electronics

$>8 \mathrm{VA} 1$ chips per ladder view:

- low power consumption: $1 \mathrm{~mW} /$ channel
- low noise: $\mathrm{ENC}=230 \mathrm{e}^{-}+13 \cdot \mathrm{C}_{\mathrm{L}} \sim 500 \mathrm{e}^{-}$in our detector configuration
$>1024$ strips are read-out sequentially (at 0.5 MHz frequency) and the analog signals are digitized by 36 12-bit ADCs;
$>12$ DSPs control the tracker acquisition and perform on-line data compression


## Radiation hardness tests

$>$ Total dose: cumulative effect after long-term irradiation

- 1 krad expected in three years in orbit
- 30 krad dose reached for about 100 chips; 5 showed a failure
$>$ SEE (single event effect): produced by a single interacting particle
- not negligible in spite of the very low flux of heavy ions in orbit
- test for upset and latch-up of FPGA chips on high-Z beam at GSI
- test for latch-up of DC/DC converters at JINR
- protection circuits (against latch-ups) and hot-cold redundancies foreseen in all the electronic boards


## Performances of the detectors

observed on a test beam at Cern SPS: September 15-22, 2003


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## Tracker alignment

In order to reconstruct the trajectories of charged particles inside the spectrometer, the mutual positions of all the planes need to be known. An alignment procedure is thus necessary.
$>$ preliminary alignment using data gathered in the last test beam;
$>$ on-orbit check after the launch of the satellite.

## Spatial resolution




| View | S/N ratio | Resolution | MDR |
| :---: | :---: | :---: | :---: |
| Junction | $\sim 50$ | $3 \mu \mathrm{~m}$ | $>740$ GV/c |
| Ohmic | $\sim 25$ | $12 \mu \mathrm{~m}$ |  |

Note: $M D R=$ Maximum Detectable Rigidity; rigidity = momentum/charge

## Detector simulation

GEANT-based code developed: it describes generation and collection on the strips of the charge carriers when ionizing particles cross the silicon detector.


Track by a MIP in which an energetic secondary electron is produced

## Output of the simulation



## Output of the simulation



## Output of the simulation



$\sim 150 \mu \mathrm{~m}$ shift !

## Data analysis: the $\eta$ algorithm


$f(\eta)$ cumulative pdf of $\eta$, estimated as:

$$
f(\eta)=\frac{\int_{0}^{\eta}\left(d N / d \eta^{\prime}\right) d \eta^{\prime}}{\int_{0}^{1}\left(d N / d \eta^{\prime}\right) d \eta^{\prime}}
$$

References for the $\eta$ algorithm:
R. Turchetta, Nucl. Instr. and Meth. A335, 44 (1993)
E. Belau, Nucl. Instr. and Meth. A214, 253 (1983)

## Distribution of residuals, junction side



It is not Gaussian!
Accurate fit by means of a Lorentz distribution:

$$
L=p_{1} \cdot\left[1+\left(\frac{2 \cdot\left(x-p_{2}\right)}{p_{3}}\right)^{2}\right]^{p_{4}}
$$

Tails of the distribution are related to the spillover !

## The spillover ...

... indicates the particle background in antiparticle measurements, due to a wrong reconstructed charge sign for small curvature, high momentum particles.

Since particles are much more abundant than antiparticles
( $\sim 10^{4}$ times), this can give rise to a mistaken flux determination.

The detailed knowledge of the distribution of spatial residuals is essential to correctly estimate the spillover background. In this case simulation provides a powerful instrument for data analysis.

## Conclusions

$>$ The flight model of the spectrometer has been already completed; the whole PAMELA apparatus will be delivered to the Russian space agency within the end of November;
$>$ the performances of the spectrometer are very satisfying: they match the requirements of the design, needed to perform antiparticle measurements in orbit;
$>$ the next steps of data analysis will concern the alignment of the tracker using curved tracks by test data;
$>$ another important subject will be the simulation of the whole tracker in order to evaluate the systematic errors in the alignment procedure and to estimate the spillover background in orbit.

