

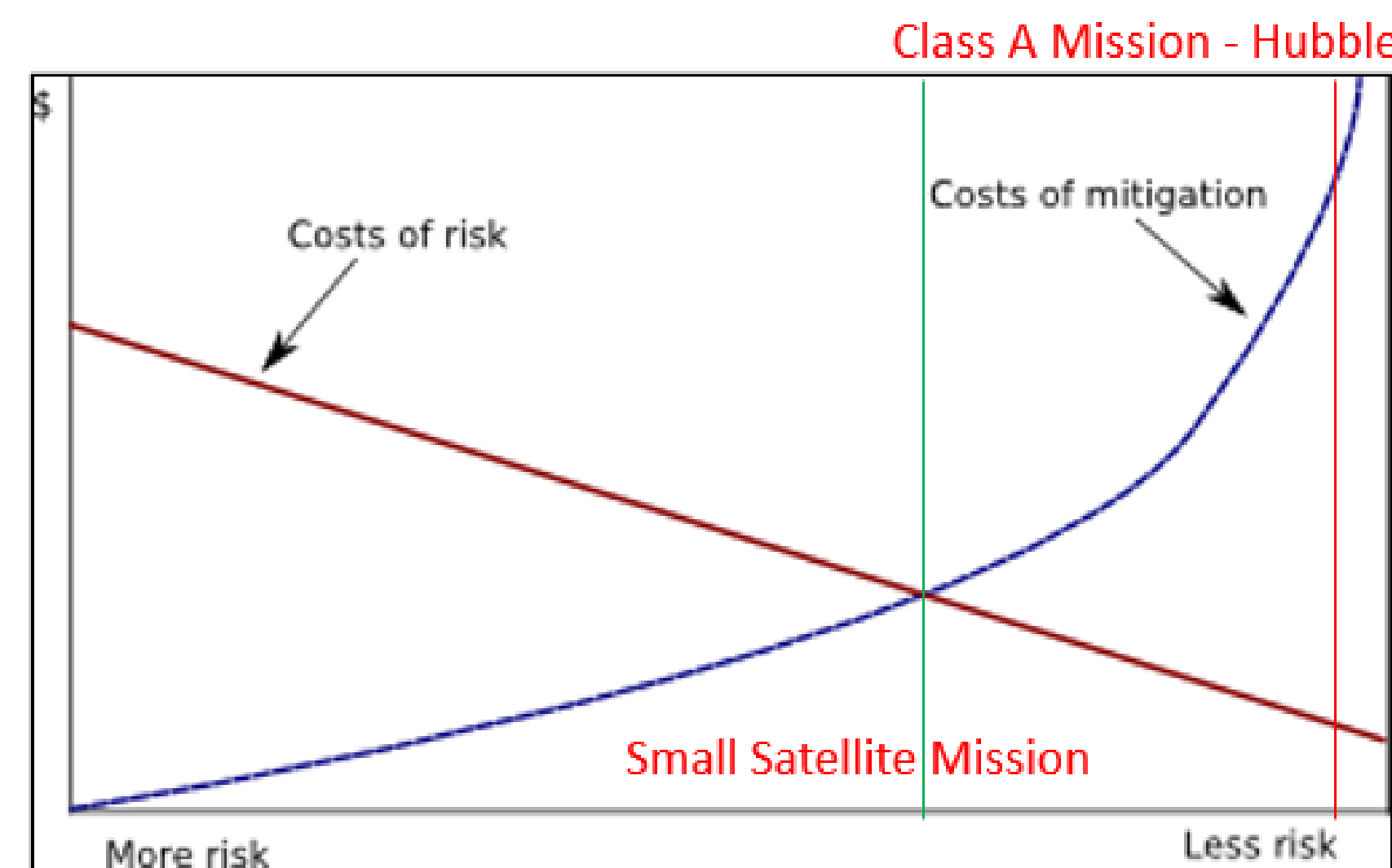
Small Satellite Payload Calibration

Collaborators: Deron Scott, Cole Miller, Eli Salay, Shane Canfield, Scott Hansen, Matt Berghout, Jan Kunzler, Duane Miles

This project focused on developing an efficient and cost-effective method for calibrating optical payloads that streamlines setup, measurement, and analysis time while staying within a SmallSat budget. To develop and test the concept, the team identified key calibration parameters and performed a demonstration on a surrogate payload using spatial, spectral, and radiometric calibration methods. Calibration results were derived from the demonstration and are detailed below.

A Small Satellite Calibration on a Budget

SmallSat missions have cost, time, and scale limitations. Calibration methods can be optimized to enable mission success at a low cost by providing customers the best “bang for their buck.”



Objective 1: Identify key payload calibration parameters that offer the best chance of mission success.

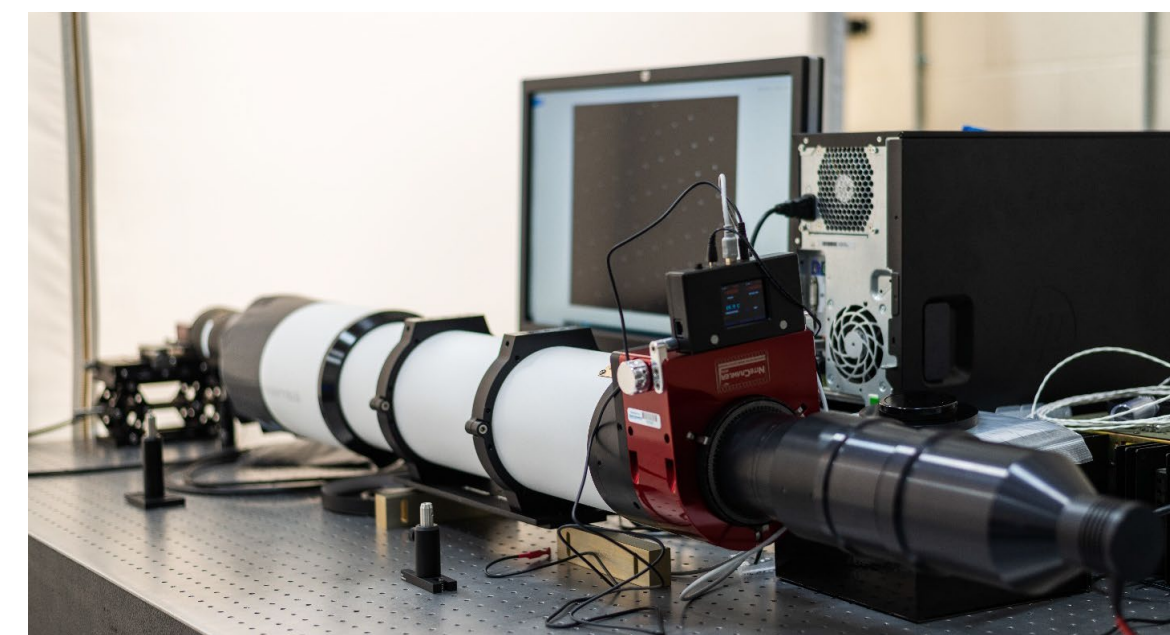
Objective 2: Develop a working calibration method to provide spatial, spectral, and radiometric calibration.

Objective 3: Demonstrate that the calibration method improves performance for a surrogate payload at a low cost.

B Calibration Methods

Spatial Calibration

- Determine best focus over field of view (FOV)
- Determine unit under test (UUT) current focus relative to test system focus (truth)
- Adjust UUT focus to match system focus
- Re-test to verify that UUT is in focus



Target and Collimator

Spectral Calibration

- Characterize payload response as a function of incident wavelength
- Payload views scene with known spectral channels
- Response is normalized to the integrated spectral radiance and to the maximum response

Radiometric Calibration

Determine peak radiative responsivity (PRR), non-uniformity correction (NUC), and system noise response

- PRR: Conversion from measured response (corrected counts) to engineering units ($W/m^2\text{-sr}$)
- NUC: Pixel-by-pixel correction to adjust response to be homogeneous for uniform source
- System noise: UUT response to lowest signal level

C Small Satellite Surrogate Payload

Unit Under Test

- Camera: 1380 p x 1038 p visible focal plane array (FPA)
- Telescope: Stellarvue SVX152T



Surrogate Payload

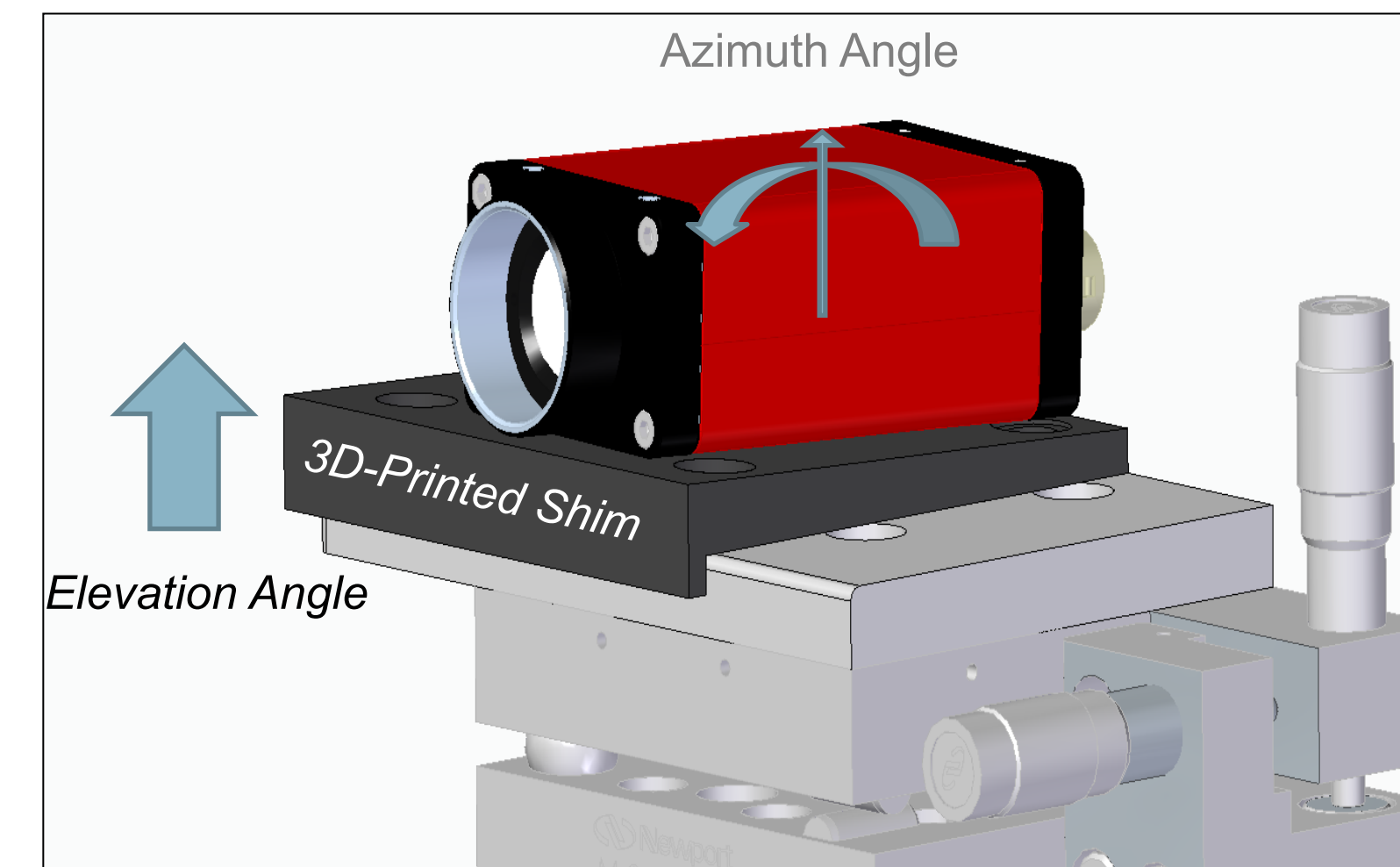
D Small Satellite Calibration Demonstration

To perform the demonstration, the team simulated a real-world scenario by intentionally placing the surrogate payload out of focus.

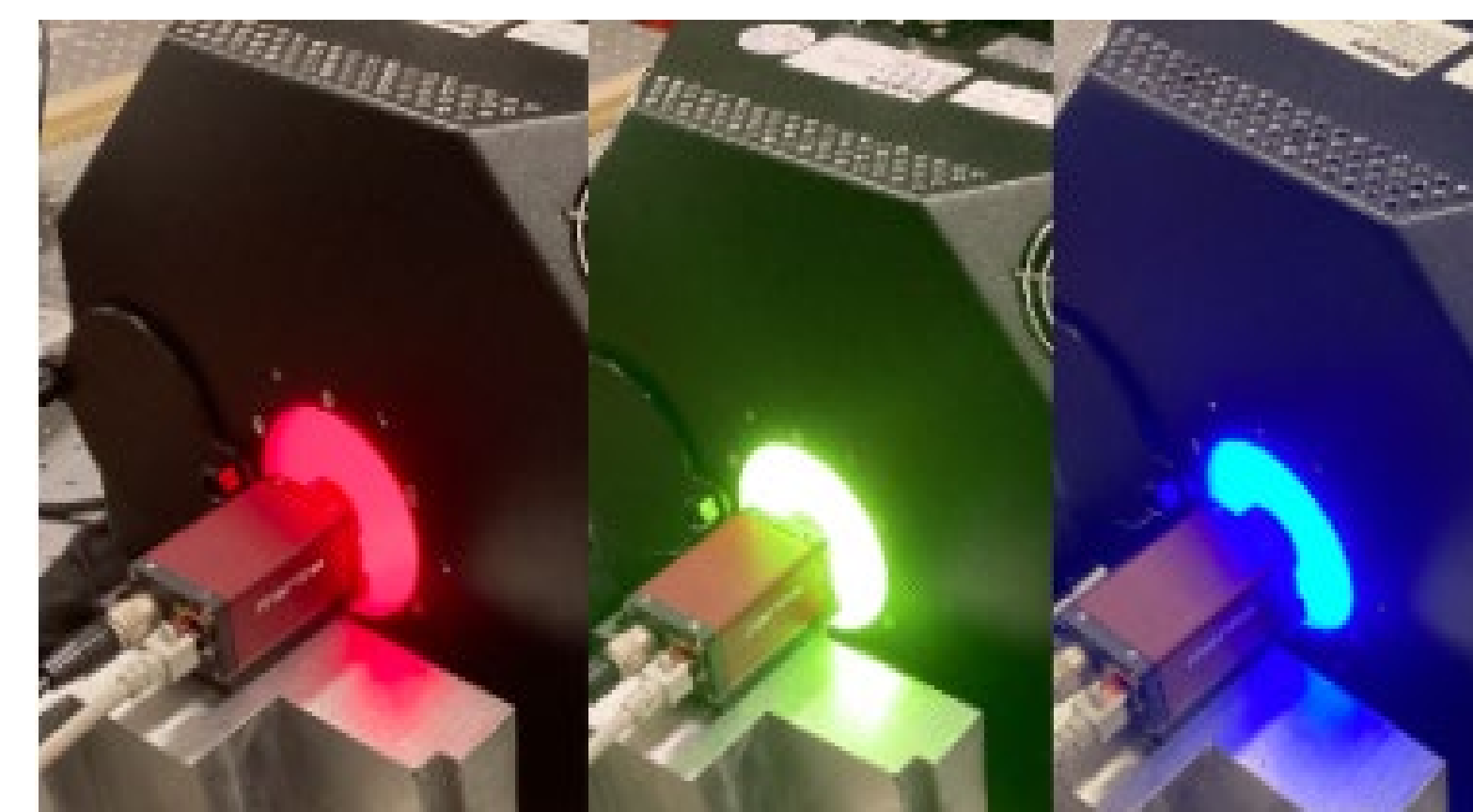
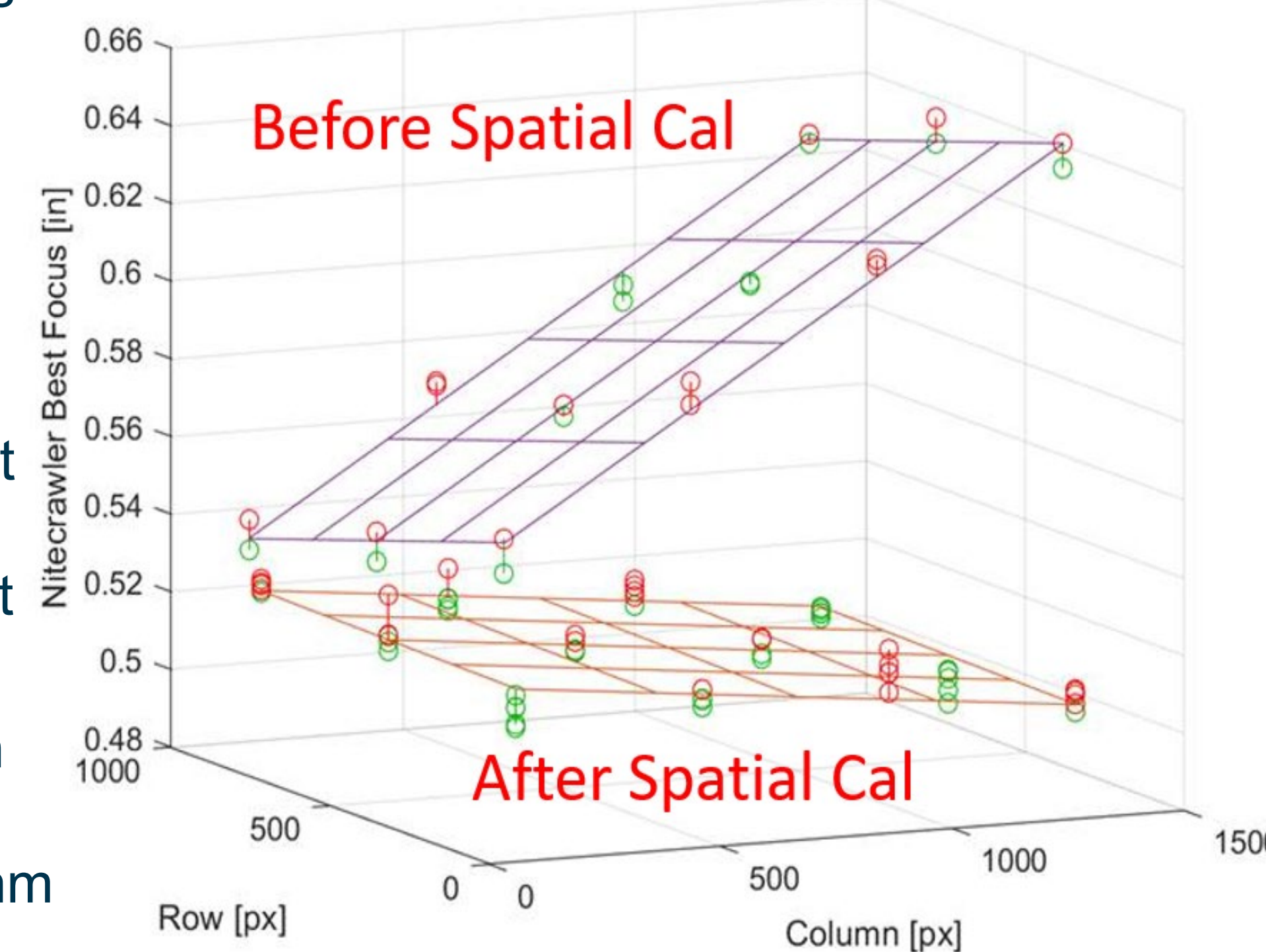
Spatial Calibration

Key Parameters

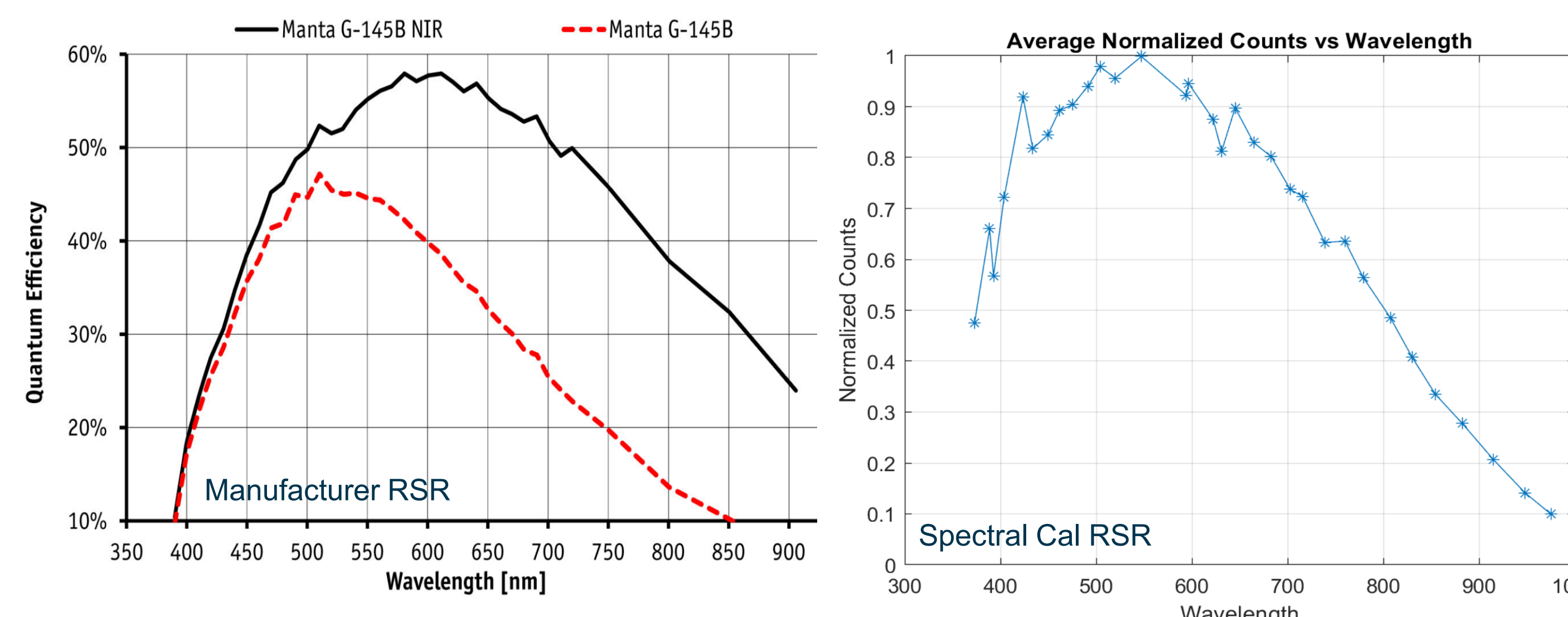
- Camera spatial adjustment
- Point response function (PRF)
- Distortion can be calculated
- Surrogate payload looks into multi-point source



Best Focus Plane of Best Fit Comparison



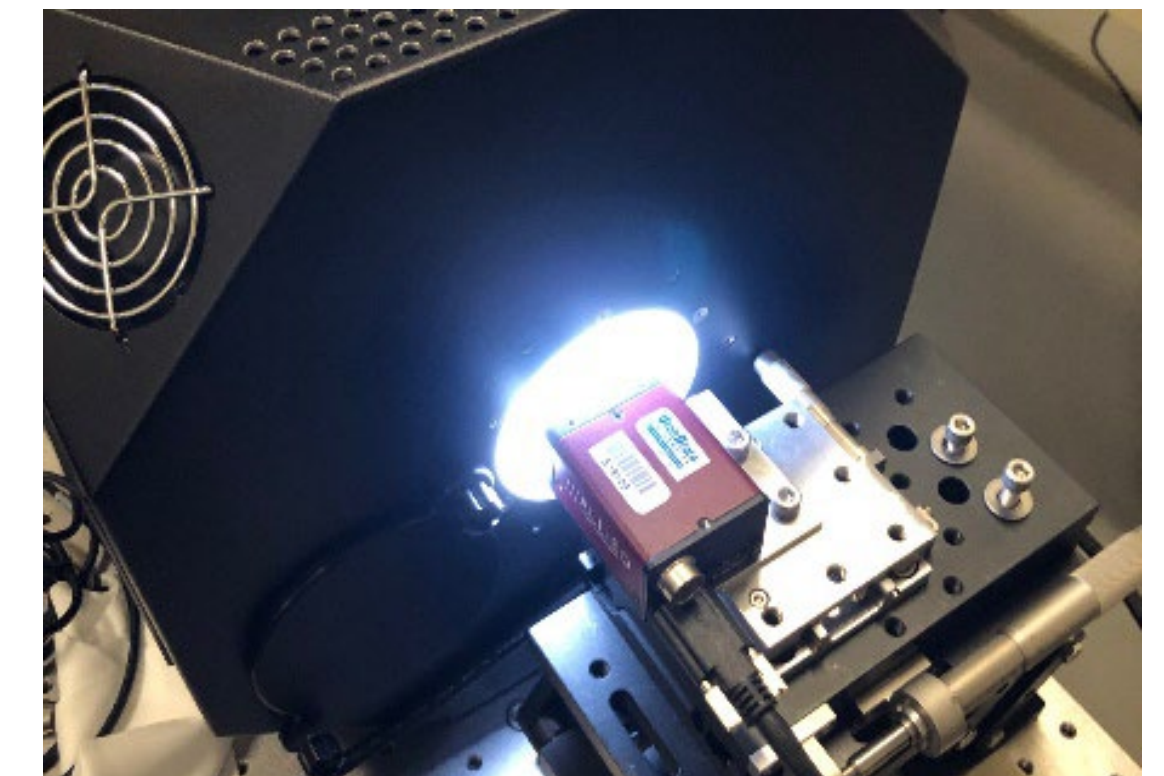
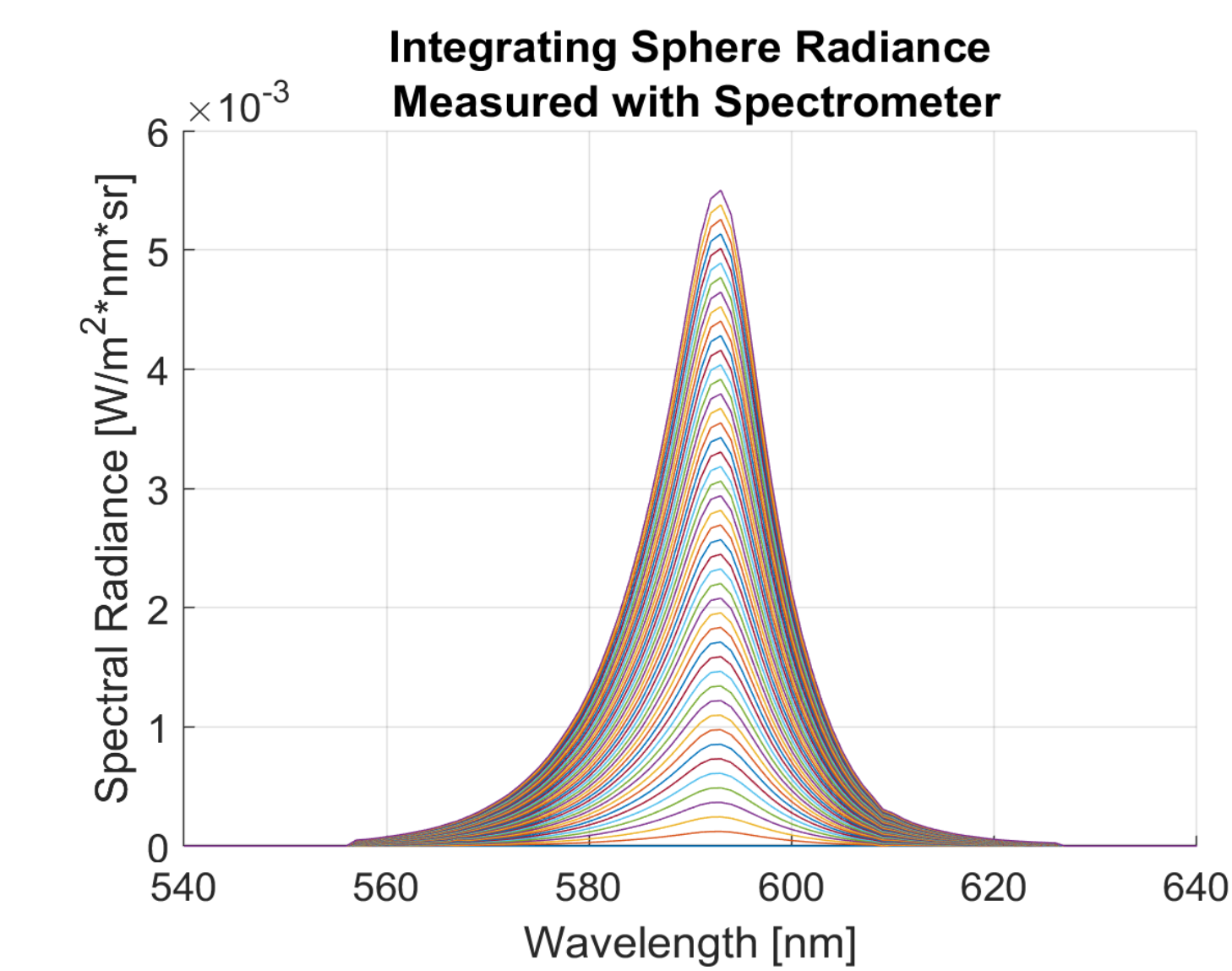
Surrogate Payload Exposed to Various LED Wavelengths for RSR



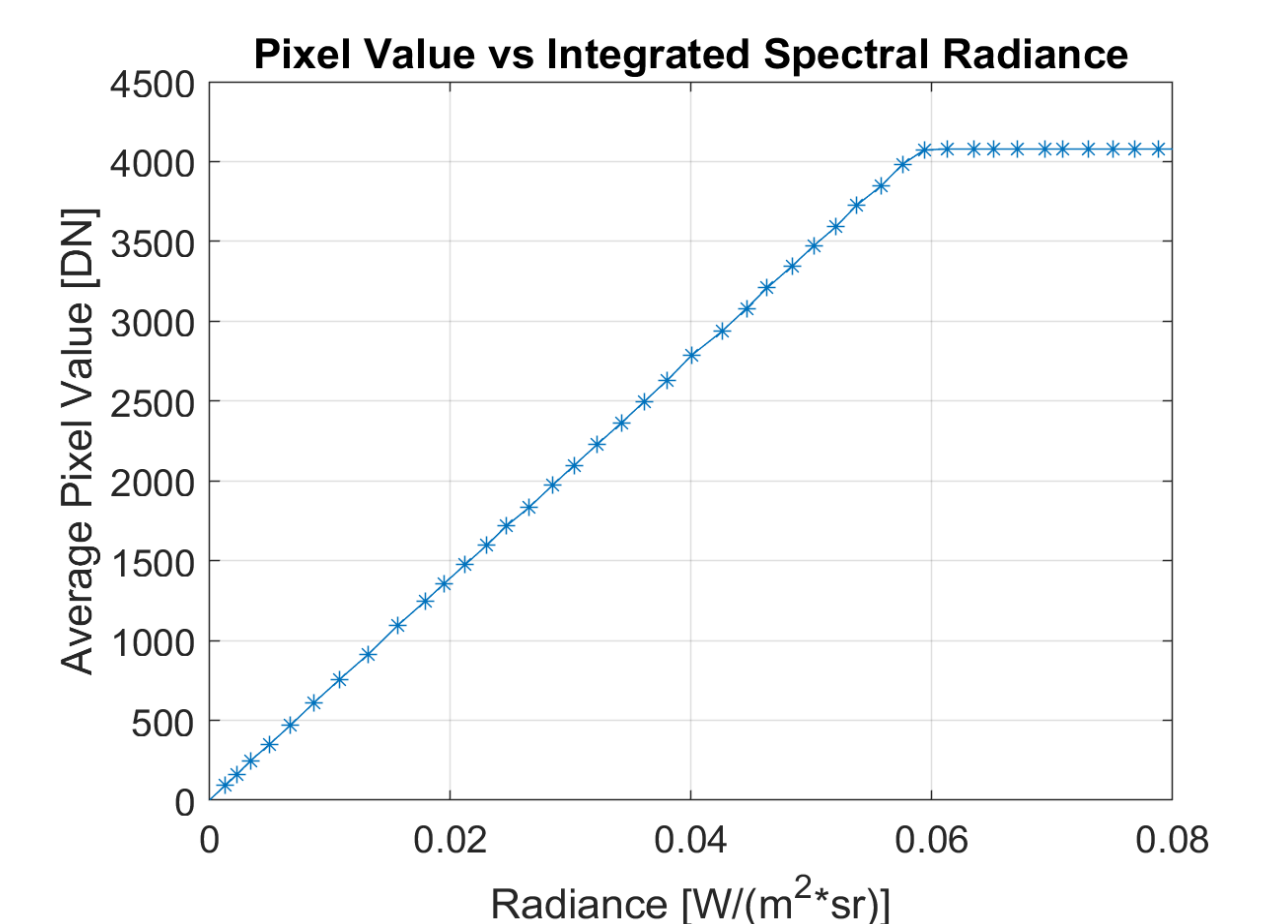
Radiometric Calibration

Key Parameters

- Peak radiative responsivity
- Non-uniformity correction
- Rudimentary linearity correction can be derived from peak radiative response

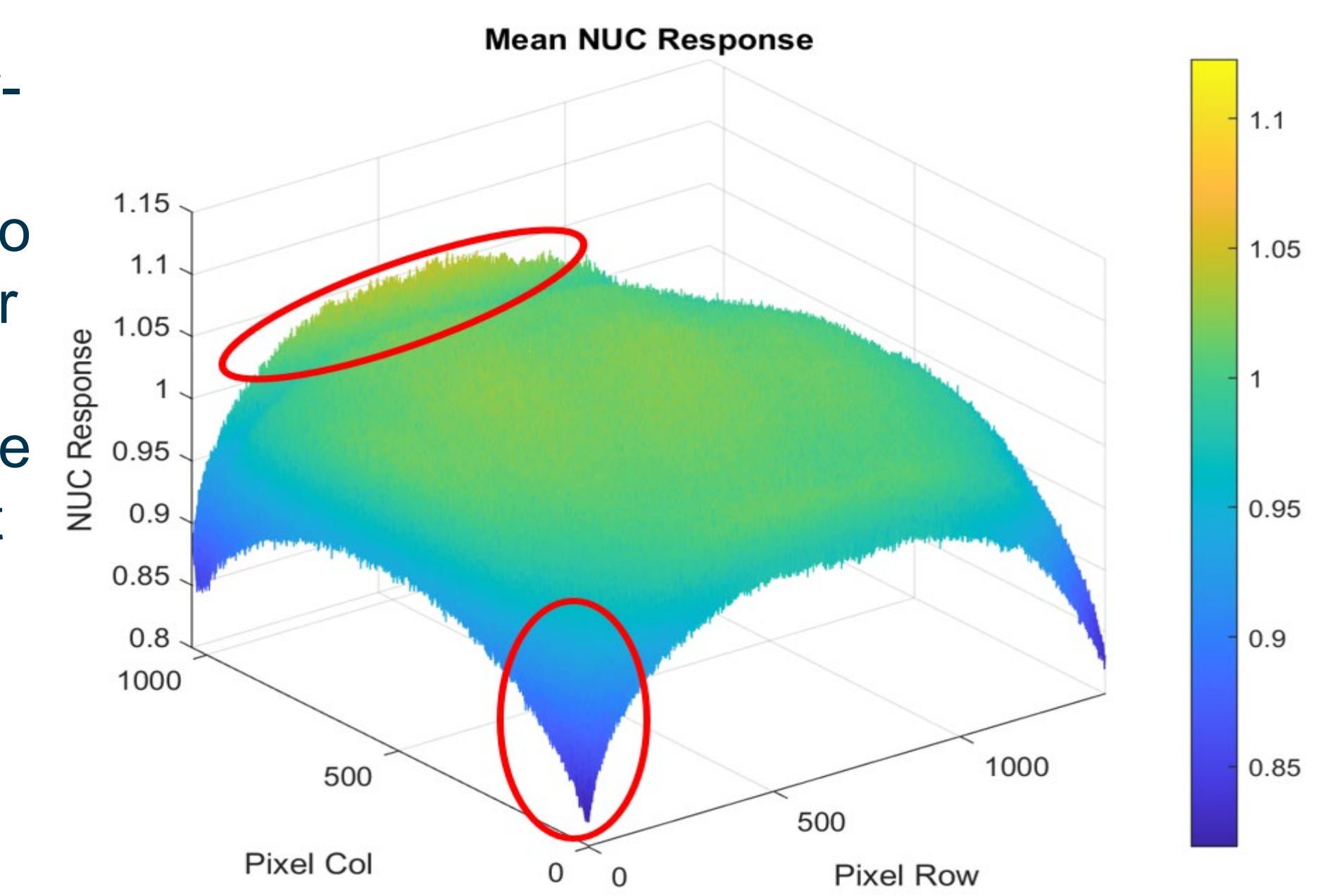


Integrating Sphere and Payload



Non-Uniformity Correction

- Regions with response in pixels >1 are due to pixel noise
- Regions with response <1 are due to vignetting in corners of detector
- Determined pixel-by-pixel correction
- Adjusted response to be homogeneous for uniform source
- Multiple images were collected at different illumination levels
- Surface plot shows average variation in pixel-to-pixel response



E Summary

Calibration Results

- Improved camera spatial focus to within ~ 25 μm of best focus
- Verified manufacturer RSR and identified variations
- Calculated distortion, PRR, NUC, and linearity correction

The calibration regiment developed during this effort provides the following benefits to a SmallSat mission:

- **Quick:** Fewer tests and more automation lead to faster results.
- **Flexible:** Calibration can be performed on a variety of payloads.
- **Mobile:** Less hardware allows for calibration to be performed on site or at customer locations.
- **Reliable:** Less test hardware introduces fewer uncertainty contributions.
- **Cost Effective:** This method can meet the technical demands of a mission and maintain an acceptable level of risk on a SmallSat budget.