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# A Novel Spherical Permanent Magnet Actuator with Three Degrees-of-Freedom 

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#### Abstract

The paper describes a new version of spherical actuator, which is capable of three degrees-of-freedom and a high specific torque. The three-dimensional magnetic field distribution is established using an analytical technique formulated in spherical co-ordinates, and enables the torque vector and back-emf to be derived in closed forms. This facilitates the characterisation of the actuator, and provides the foundation for design optimisation, actuator dynamic modelling and servo control development.


## I. Introduction

There is a growing need for programmable servocontrolled high-speed actuation in multiple axes, for applications as diverse as robotic manipulators, infra-red and laser tracking systems, and automated manufacturing. At present, multi-degree-of-freedom motions are realised almost exclusively by using a separate motor/actuator for each axis, which results in relatively complicated and heavy transmission systems. This inevitably compromises the dynamic performance and servo-tracking accuracy, due to the combined effects of inertia, backlash, non-linear friction, and elastic deformation of gears. Actuators with multiple degrees-of-freedom should alleviate these problems, whilst being lighter and more efficient. However, although they have been the subject of research for several decades, and numerous concepts have been proposed [1]-[4], their application potential has not been realised, probably due to the complexity of their structure and related difficulties in modelling their electromagnetic behaviour and optimising their design.

Recently, a new form of spherical actuator, which is capable of two degrees-of-freedom, has been described, systematically analysed and experimentally demonstrated [5], [6]. This paper describes a more advanced version, shown schematically in Fig.1(a), which is capable of three degrees-of-freedom. The actuator has a four-pole spherical permanent magnet rotor, which is formed from two pairs of parallel magnetised quarter-spheres, as shown in Fig. 1(b), and a simple stator winding arrangement. The spherical rare-earth magnet rotor is housed within the spherical stator on a low friction surface coating. Accommodated on the stator are four sets of windings, as shown Fig. 1(c), which are arranged so that three independently controllable torque components can be developed by energising the appropriate windings. The stator can be either air-cored or iron-cored by enclosing the windings with a spherical iron shell so as to increase the torque capability.
II. Magnetic Field Distribution

A knowledge of the magnetic field distribution produced


Fig. 1 Schematic of a 3 DOF spherical actuator. (a) Schematic. (b) rotor magnetisation pattern. (c) winding arrangement (rear view).
by the spherical permanent magnet rotor is fundamental to establishing an accurate model of the actuator, for design optimisation and dynamic modelling. Without loss of generality, an air-cored actuator is considered. Thus, the entire magnetic field region can be divided into two subregions, the outer airspace/winding region in which the permeability is $\mu_{0}$, and the magnet region in which the permeability is $\mu_{0} \mu_{r}$. Therefore:

$$
\boldsymbol{B}=\left\{\begin{array}{cl}
\mu_{0} \boldsymbol{H} & \text { in the airspace/windings }  \tag{1}\\
\mu_{0} \mu_{r} \boldsymbol{H}+\mu_{0} \boldsymbol{M} & \text { in the magnet }
\end{array}\right.
$$

where $\mu_{r}$ is the relative recoil permeability of the magnet and $M$ is its remanent magnetization. For a permanent magnet having a linear demagnetization characteristic, $\mu_{r}$ is constant and $\boldsymbol{M}$ is related to the remanence, $\boldsymbol{B}_{r e m}$, by $\boldsymbol{M}=\boldsymbol{B}_{r e m} / \mu_{\theta}$. It is convenient to formulate the field distribution in terms of a scalar potential, $\varphi$ defined as $\boldsymbol{H}=-\nabla \varphi$, and the spherical coordinate system shown in Fig. 2. This leads to the following field equations:


Fig. 2 Spherical Co-ordinate system

$$
\begin{cases}\nabla^{2} \varphi_{I}=0 &  \tag{2}\\ \nabla^{2} \varphi_{I I}=\nabla \cdot \boldsymbol{M} / \mu_{r} & \\ \text { in the airspace/windings } \\ \text { in thagnet }\end{cases}
$$

The components of the magnetisation vector $\boldsymbol{M}$ shown in Fig. l(b) may be expressed as:
$\left\{\begin{array}{l}M_{r}=M_{c}[\operatorname{sgn}(\tan \theta) \sin \theta \sin \alpha+\operatorname{sgn}(\sin \alpha) \cos \theta] \\ M_{\theta}=M_{c}[\operatorname{sgn}(\tan \theta) \cos \theta \sin \alpha-\operatorname{sgn}(\sin \alpha) \sin \theta] \\ M_{\alpha}=M_{c}[\operatorname{sgn}(\tan \theta) \cos \alpha]\end{array}\right.$
where $\operatorname{sgn}(\cdot)$ denotes the sign function and $M_{c}=B_{r e m} / \mu_{0} \sqrt{2}$ It can be shown that $\nabla \cdot \boldsymbol{M} \equiv 0$. However, the contribution of the magnets will appear in the following interface boundary conditions:

$$
\left\{\begin{array}{l}
-\left.\mu_{r} \frac{\partial \varphi_{l l}}{\partial r}\right|_{r=R_{m}}+M_{r}=-\left.\frac{\partial \varphi_{I}}{\partial r}\right|_{r=R_{m}}  \tag{4}\\
-\left.\frac{1}{r} \frac{\partial \varphi_{I I}}{\partial \theta}\right|_{r=R_{m}}=-\left.\frac{1}{r} \frac{\partial \varphi_{I}}{\partial \theta}\right|_{r=R_{m}} \\
-\left.\frac{1}{r \sin \theta} \frac{\partial \varphi_{I I}}{\partial \alpha}\right|_{r=R_{m}}=-\left.\frac{1}{r \sin \theta} \frac{\partial \varphi_{I}}{\partial \alpha}\right|_{r=R_{m}}
\end{array}\right.
$$

where $R_{m}$ is the radius of the spherical rotor. The radial component, $M_{r}$, may be expanded into spherical harmonics of the following form:

$$
\begin{equation*}
M_{r}=\sum_{l=2,4}^{\infty} \sum_{m=1,3, \ldots}^{l} M_{l m} P_{l}^{m}(\cos \theta) \sin \alpha \tag{5}
\end{equation*}
$$

where $P_{l}^{m}(\cdot)$ denotes the associated Legendre polynomial of degree $l$ and order $m$, and $M_{l m}$ is given by:

$$
\begin{equation*}
M_{l m}=\frac{4(2 l+1)(l-m)!}{\pi m(l+m)!} M_{c} \int_{0}^{1} P_{l}^{m}(x) x d x \tag{6}
\end{equation*}
$$

Solving for equation (2) with the boundary conditions of equation (4) yields the following expressions for the flux density distribution in the airspace/winding region:

$$
\begin{align*}
& B_{l r}=\sum_{l=2,4, \ldots}^{\infty} \sum_{m=1,3, \ldots}^{l} C_{l m}(l+1) r^{-(l+2)} P_{l}^{m}(\cos \theta) \sin \alpha \\
& B_{l \theta}=\sum_{l=2,4, \ldots, m=1,3, \ldots}^{\infty} \sum_{l m}^{l} C_{l m} r^{-(l+2)} \frac{d}{d \theta} P_{l}^{m}(\cos \theta) \sin \alpha \tag{7-a}
\end{align*}
$$



Fig. 3 Radial component of flux density as a function of $\alpha$ at $r=0.045(\mathrm{~m}), \theta$

$$
=60^{\circ} \text { for } R_{m}=0.040(\mathrm{~m})
$$

$$
\begin{equation*}
B_{I \alpha}=\sum_{l=2,4, \ldots}^{\infty} \sum_{m=1,3, \ldots}^{l} \frac{m C_{l m}}{\sin \theta} r^{-(l+2)} P_{l}^{m}(\cos \theta) \cos \alpha \tag{7-b}
\end{equation*}
$$

where $C_{l m}$ is given by:

$$
C_{l m}=-\mu_{0} R_{m}^{l+2} M_{l m} /\left[\left(1+l\left(1+\mu_{r}\right)\right]\right.
$$

This result has been validated by finite element analysis, as shown in Fig. 3, the differences between the two predictions being attributable to the effect of discretization in the finite element model.

## III Torque and EMF Prediction

The torque exerted on the rotor, resulting from the interaction between the current in a stator winding and the rotor magnetic field, is given by:

$$
\begin{equation*}
\boldsymbol{T}=-\int_{V} \boldsymbol{r} \times(\boldsymbol{J} \times \boldsymbol{B}) d V \tag{8}
\end{equation*}
$$

where $J$ denotes the current density vector in the winding region $V$. Each winding comprises a number of circular turns distributed on the spherical stator, and occupying an area bounded by $r=R_{0}, r=R_{s}$, and $\delta=\delta_{0}, \delta=\delta_{l}$, as shown in Fig. 4. Considering a single turn with infinitesimal cross-section $d s=r d r d \delta$ and an enclosing circular contour $C$, the total torque produced by the winding may be obtained from the following integration:


Fig. 4 Winding distribution

$$
\begin{equation*}
T=-2 J \int_{R_{0}}^{R_{s}} \int_{\delta_{0}}^{\delta_{1}}\left\{\int r B_{I r}(r, \theta) d l\right\} r d r d \delta \tag{9}
\end{equation*}
$$

It can be shown that $B_{I r}$ is dominated by its fundamental component, viz. $l=2, m=1$. Therefore, the result of equation (9), neglecting high order harmonics, is given by:

$$
T=\left[\begin{array}{c}
T_{x}  \tag{10}\\
T_{y} \\
T_{z}
\end{array}\right]=T_{m}\left[\begin{array}{c}
v_{y}^{2}-v_{z}^{2} \\
-v_{x} v_{y} \\
v_{x} v_{z}
\end{array}\right]
$$

where $v=\left[\begin{array}{lll}v_{x} & v_{y} & v_{z}\end{array}\right]^{\mathrm{T}}$ is the direction cosines of the winding, viz: $\quad v_{x}=\sin \theta \cos \alpha ; \quad v_{y}=\sin \theta \sin \alpha ; \quad v_{z}=\cos \theta$. The torque magnitude, $T_{m}$, is given by: $T_{m}=15 \pi \sqrt{2} B_{r e m} J R_{m}^{4} \ln \left(R_{s} / R_{0}\right)\left(\sin ^{3} \delta_{1}-\sin ^{3} \delta_{0}\right) / 8\left(2 \mu_{r}+3\right)$.
As can be seen, $T_{m}$ is dependent upon $B_{r e m}$ and $J$ as well as the geometrical parameters of the rotor and the winding. If the airgap between the rotor and winding is $G$, then for given $B_{r e m}, J, \delta_{0}, \delta_{1}$ and $R_{s}, T_{m}$ is a function of $R_{m} / R_{s}$. This relation is plotted in Fig. 5 assuming $B_{r e m}=1.2 \mathrm{~T}, \mu_{r}=1.15, J=$ $2.0 \mathrm{~A} / \mathrm{mm}^{2}, \delta_{0}=5^{0}, \delta_{l}=30^{\circ}, G=0.002 \mathrm{~m}$ and $R_{s}=0.042 \mathrm{~m}$. It is evident that there exists an optimal ratio of $R_{m} / R_{s}$ viz. 0.738 , that yields maximum torque.

The torque, $\boldsymbol{T}_{c}$, produced by a pair of windings carrying current $i$ and having direction cosines $\left[v_{x} v_{y} v_{z}\right]^{\mathrm{T}}$ and $\left[v_{x}^{\prime} v_{y}^{\prime}{ }_{y}\right.$ $\left.v_{z}^{\prime}\right]^{\mathrm{T}}$, respectively, is given by:

$$
\boldsymbol{T}_{c}=\left[\begin{array}{c}
T_{c x}  \tag{11}\\
T_{c y} \\
T_{c z}
\end{array}\right]=K_{T} i\left[\begin{array}{c}
v_{y}^{2}-v_{z}^{2}+v_{y}^{\prime 2}-v_{z}^{\prime 2} \\
-v_{x} v_{y}-v_{x}^{\prime} v_{y}^{\prime} \\
v_{x} v_{z}+v_{x}^{\prime} v_{z}^{\prime}
\end{array}\right]
$$

where

$$
\begin{equation*}
K_{T}=T_{m} N /\left[J\left(R_{s}^{2}-R_{0}^{2}\right)\left(\delta_{1}-\delta_{0}\right)\right] \tag{12}
\end{equation*}
$$

is defined as the winding torque constant, and $N$ is the number of turns for the winding pair. The total torque, $T_{e m}$, produced by four sets of identical windings, $A, B, C$, and $D$, having direction cosines $\left[v_{x j} v_{y j} v_{z j}\right]^{\mathrm{T}}$ and $\left[v_{x j}^{\prime} v_{y j}^{\prime} v_{z j}\right]^{\mathrm{T}}(\mathrm{j}=\mathrm{A}, \ldots, \mathrm{D})$ is, therefore, given by:


Fig. 5 Torque as a function of $R_{m} / R_{s}$

$$
\begin{equation*}
T_{e m}=\sum_{j=A, \ldots, D} T_{c j}=K_{T M} i \tag{13}
\end{equation*}
$$

where $\boldsymbol{i}=\left[\begin{array}{lll}i_{A} & i_{B} & i_{C} \\ i_{D}\end{array}\right]^{\mathrm{T}}$ is the winding current vector, and $\boldsymbol{K}_{r_{M}}$, defined as the torque matrix of the actuator, is given by:

$$
\begin{align*}
& \boldsymbol{K}_{T M}=K_{T}\left[\begin{array}{cc}
v_{y A}^{2}-v_{z A}^{2}+v^{\prime 2}-v_{y A}^{\prime 2} & v_{z B}^{2}-v_{z B}^{2}+v^{\prime}{ }_{y B}-v_{z B}^{\prime 2} \\
-v_{x A} v_{y A}-v_{x A}^{\prime} v^{\prime}{ }_{y A} & -v_{x B} v_{y B}-v_{x B}^{\prime} v^{\prime}{ }_{y B} \\
v_{x A} v_{z A}+v_{x A}^{\prime} v_{z A}^{\prime} & v_{x B} v_{z B}+v_{x B}^{\prime} v_{z B}^{\prime}
\end{array}\right. \\
& \left.v_{y C}^{2}-v_{z C}^{2}+v_{y C}^{\prime 2}-v_{z C}^{2} \quad v_{y D}^{2}-v_{z D}^{2}+v_{y D}^{\prime 2}-v_{z D}^{\prime 2}\right] \\
& -v_{x C} v_{y C}-v_{x C}^{\prime} v_{y C}^{\prime} \quad-v_{x D} v_{y D}-v_{x D}^{\prime} v_{y D}^{\prime} \\
& v_{x C} v_{z C}+v_{x C}^{\prime} v_{z C}^{\prime} \quad v_{x D} v_{z D}+v_{x D}^{\prime} v_{z D}^{\prime} \tag{14}
\end{align*}
$$

It can be shown that the rank of $\boldsymbol{K}_{T M}$ is three within the working envelope of the actuator, implying that the actuator is able to deliver motion in three degrees-of-freedom, viz. $\pm 45^{\circ}$. pan-tilt excursions and continuous rotation.

By a similar integration procedure, the flux-linkage of a pair of windings due to the rotor magnetic field is given by:

$$
\begin{equation*}
\Psi_{w}=K_{E}\left(v_{y} v_{z}+v_{y}^{\prime} v_{z}^{\prime}\right) \tag{15}
\end{equation*}
$$

where $K_{E}$, defined as the back-emf constant of the winding pair, is identical to $K_{T}$. The back-emfs of four identical windings $\mathrm{A}, \mathrm{B}, \mathrm{C}$, and D are, therefore, given by:

$$
\begin{equation*}
e_{j}=-K_{T} \frac{d}{d t}\left(v_{y j} v_{z j}+v_{y j}^{\prime} v_{z j}^{\prime}\right) \quad j=A, \ldots, D \tag{16}
\end{equation*}
$$

## IV. CONCLUSION

A new version of spherical actuator which is capable of three degrees-of-freedom has been described and analyzed. It has a four-pole spherical magnet rotor and four sets of stator windings which are arranged to produce three independent torque components. The magnetic field distribution, torque vector and back-emf have been analytically derived. The results provide a basis for design optimisation, system dynamic modelling and closed-loop control law development.

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