

## The Argon Power Cycle

***Citation for published version (APA):***

Diepstraten, N., Somers, B., & van Oijen, J. (2022). *The Argon Power Cycle: Exploring DI-H2 and DI-O2 injection strategies using CFD*. Poster session presented at 12th Biennial Conference on Thermo-and Fluid Dynamics of Clean Propulsion Powerplants, THIESEL 2022, Valencia, Spain.

***Document status and date:***

Published: 13/09/2022

***Document Version:***

Publisher's PDF, also known as Version of Record (includes final page, issue and volume numbers)

***Please check the document version of this publication:***

- A submitted manuscript is the version of the article upon submission and before peer-review. There can be important differences between the submitted version and the official published version of record. People interested in the research are advised to contact the author for the final version of the publication, or visit the DOI to the publisher's website.
- The final author version and the galley proof are versions of the publication after peer review.
- The final published version features the final layout of the paper including the volume, issue and page numbers.

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## Exploring DI-H<sub>2</sub> and DI-O<sub>2</sub> injection strategies using CFD

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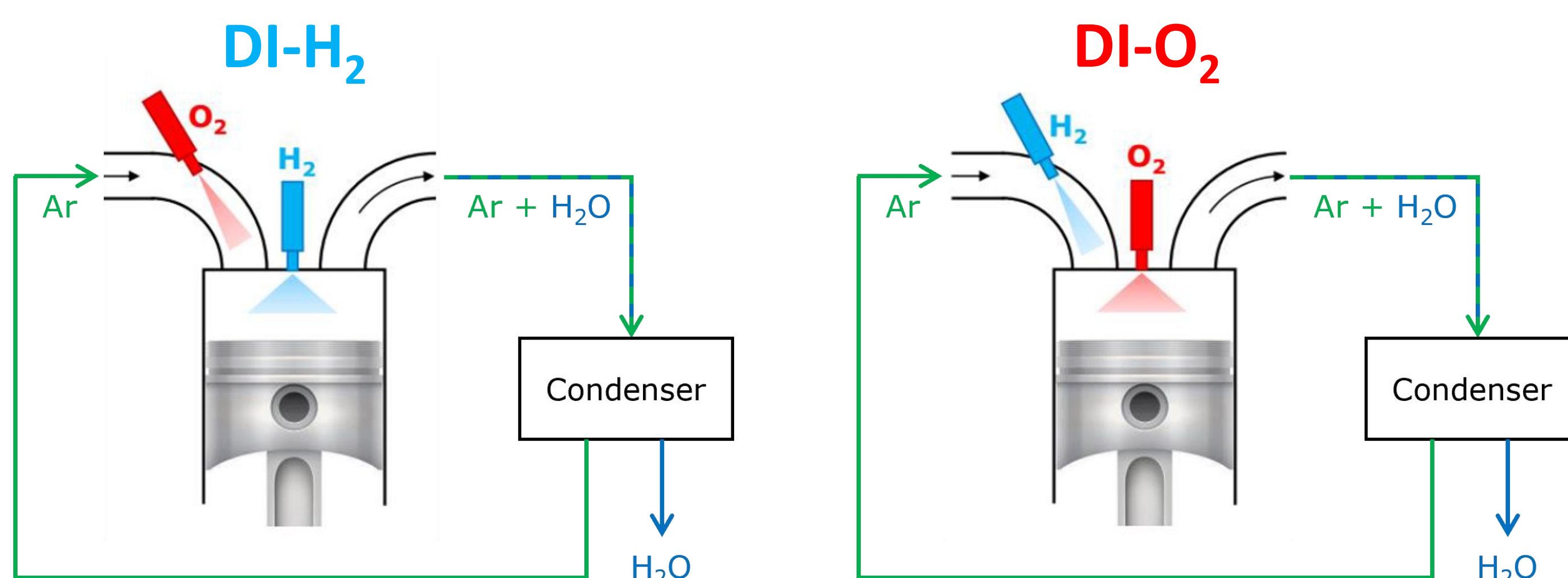
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### Why Argon Power Cycle?

Internal Combustion Engines (ICEs) can counteract the intermittent nature of renewable energy sources thanks to their ability to dispatch power rapidly. Challenges for ICE development are their relatively low efficiencies (<55%) compared to fuel cells, emissions, and pollutants. The Argon Power Cycle (APC) can overcome these challenges:

- Using argon as working fluid: **~25% efficiency gain, no NO<sub>x</sub>**;
- When hydrogen fuelled: **no carbonaceous emissions**;
- Closed-loop configuration: **stoichiometric consumption, recyclability of Argon and unburned O<sub>2</sub> and/or H<sub>2</sub>**.

The APC allows to explore multiple injection strategies, since both hydrogen and oxygen need to be injected. This work presents the first tries of the DI-H<sub>2</sub> (left) and DI-O<sub>2</sub> (right) strategies.



### CFD environment setup

- Software program CONVERGE
- Closed-valve cycle
- 60° sector mesh
- Grid aligned with injection direction
- Sonic injection through inflow boundary, resulting in an under-expanded jet.
- Equal fuelMEP and  $\lambda$  requires different nozzle diameters in both cases due to oxygen-to-hydrogen ratio.

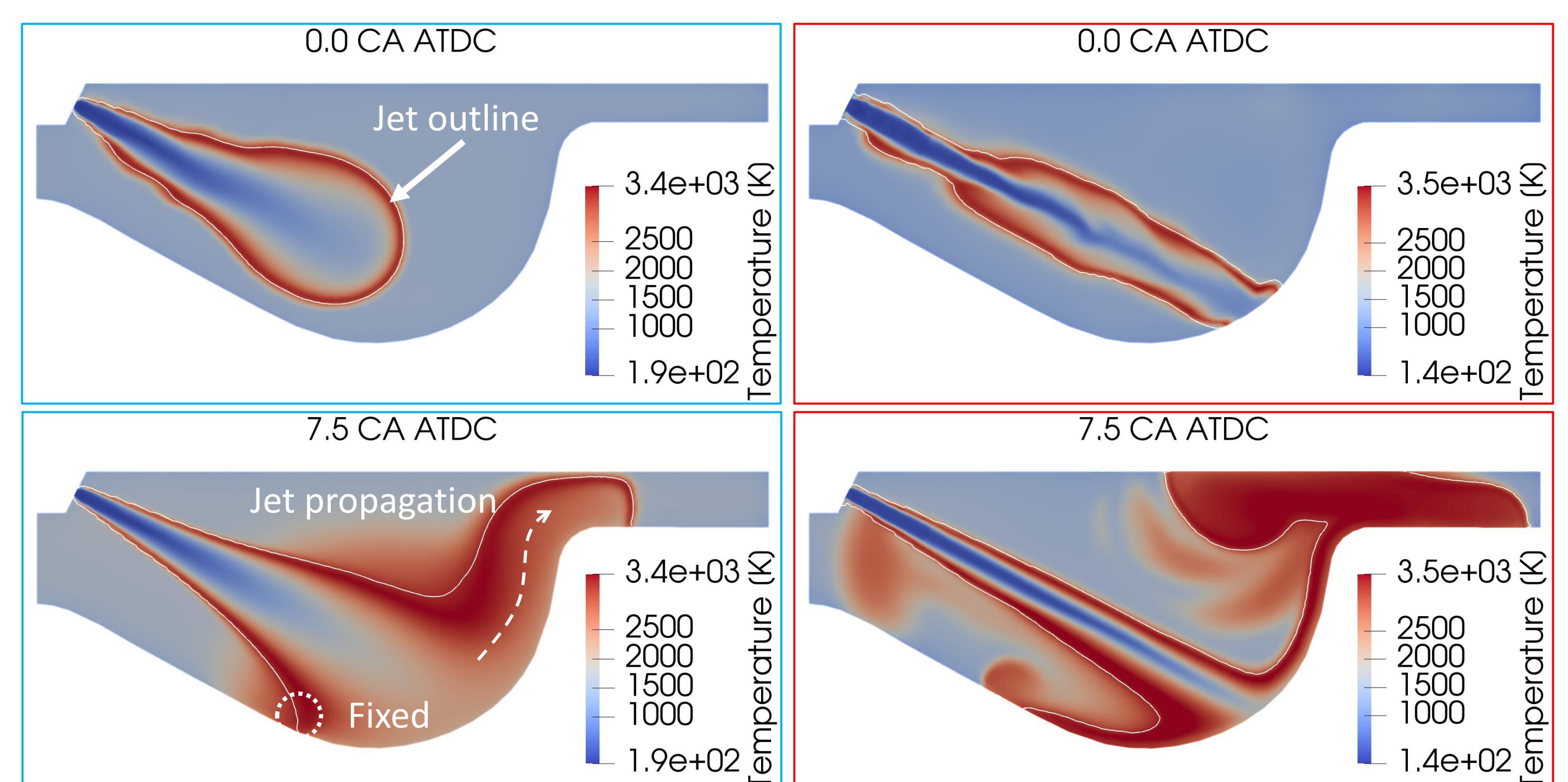
#### Engine specifications

Swept volume per cylinder	1.47 L
Bore / Stroke	112 / 149 mm
Compression ratio	16.1 : 1
Number of injector nozzles	6
Injection angle (w.r.t. vertical)	62°
Engine speed	1800 RPM
fuelMEP	19 bar

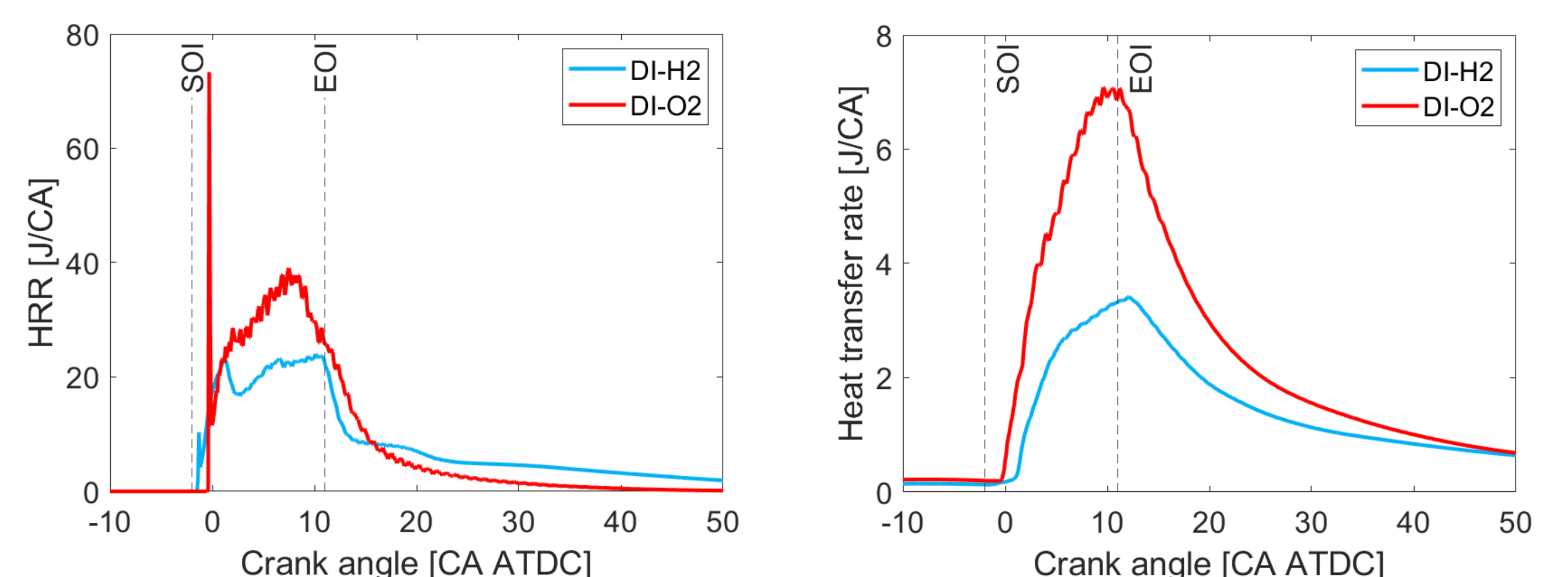
### DI-H<sub>2</sub>

300	Injection pressure [bar]	300
1228	Sonic velocity [m/s]	308
3.6	Injection rate [mg/s]	28.2
-2 / 11	SOI / EOI [CA ATDC]	-2 / 11
0.50	Nozzle diameter [mm]	0.65

### DI-O<sub>2</sub>



Snapshots of DI-H<sub>2</sub> (left) and DI-O<sub>2</sub> (right) simulations.



Chemical heat release rate and heat transfer rate through the boundaries.

### Conclusions

- Low momentum of H<sub>2</sub> jet requires revised combustion chamber design which allows better free-jet turbulent combustion.
- About 25% of injected hydrogen is trapped in squish volume resulting in a long combustion tail.
- DI-O<sub>2</sub> has a predominant global mixing combustion phase, similar to diesel combustion, thanks to the high jet momentum.
- DI-O<sub>2</sub> has more heat loss due to properties (viscosity and conductivity) of the premixed H<sub>2</sub>-Ar charge (as opposed to the O<sub>2</sub>-Ar charge in the other case).

### Future work

- Revise combustion chamber design for DI-H<sub>2</sub>.
- Quantitative analysis of energy flows and thermal efficiency.
- System analysis of APC using 1D CFD to find suitable operating conditions.

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