

## Contribution to the mechanics of machining

***Citation for published version (APA):***

Veenstra, P. C. (1965). *Contribution to the mechanics of machining*. (TH Eindhoven. Afd. Werktuigbouwkunde, Laboratorium voor mechanische technologie en werkplaatstechniek : WT rapporten; Vol. WT0139). Technische Hogeschool Eindhoven.

***Document status and date:***

Published: 01/01/1965

***Document Version:***

Publisher's PDF, also known as Version of Record (includes final page, issue and volume numbers)

***Please check the document version of this publication:***

- A submitted manuscript is the version of the article upon submission and before peer-review. There can be important differences between the submitted version and the official published version of record. People interested in the research are advised to contact the author for the final version of the publication, or visit the DOI to the publisher's website.
- The final author version and the galley proof are versions of the publication after peer review.
- The final published version features the final layout of the paper including the volume, issue and page numbers.

[Link to publication](#)

***General rights***

Copyright and moral rights for the publications made accessible in the public portal are retained by the authors and/or other copyright owners and it is a condition of accessing publications that users recognise and abide by the legal requirements associated with these rights.

- Users may download and print one copy of any publication from the public portal for the purpose of private study or research.
- You may not further distribute the material or use it for any profit-making activity or commercial gain
- You may freely distribute the URL identifying the publication in the public portal.

If the publication is distributed under the terms of Article 25fa of the Dutch Copyright Act, indicated by the "Taverne" license above, please follow below link for the End User Agreement:

[www.tue.nl/taverne](http://www.tue.nl/taverne)

***Take down policy***

If you believe that this document breaches copyright please contact us at:

[openaccess@tue.nl](mailto:openaccess@tue.nl)

providing details and we will investigate your claim.



technische hogeschool eindhoven

laboratorium voor mechanische technologie en werkplaatstechniek

rapport van de sectie: Verspaningsonderzoek

titel: Contribution to the Mechanics of Machining

auteur(s): Prof.dr. P.C. Veenstra

sectieleider: Chr. Bus

hoogleraar: Prof.dr. P.C. Veenstra

**samenvatting**

Uitgaande van het afschuifmodel van Merchant en aannemend dat er mechanisch evenwicht bestaat tussen gemiddelde waarden van de spanningen in een toestand van vlakke spanning, wordt een hoofdvergelijking afgeleid door te aanvaarden dat de richting van de maximale rek van het materiaal de richting is van de maximale hoofdspanning.

Er wordt aangetoond dat naast de Huber-Hencky voorwaarde voor plastische vloeï geen verdere energie voorwaarde noodzakelijk is. De oplossing van de hoofdvergelijking wordt vastgelegd door de heersende spanningstoestand, die kan worden bepaald door de verhouding tussen de waarde van de maximale schuifspanning en de plasticiteitskonstante van het materiaal. Dit geldt ook als vervormingsversteving optreedt.

De theorie wordt geconfronteerd met experimentele resultaten en een ware spannings-rek kromme, gebaseerd op de theorie, wordt afgeleid.

**prognose**

blz. 1 van 49 blz.

rapport nr. 0139

codering:

P.7.a.1

trefwoord:

Verspanings-  
leer.

datum:

1 juni 1965

aantal blz.

49

geschikt voor  
publicatie in:

C.I.R.P.-  
Annalen.

### Summary

Based on the Merchant shear plane model and assuming global mechanical equilibrium between average values of stress in a state of plane stress, a shear angle relation is derived by identifying the direction of maximum strain with the direction of maximum principal stress.

It is shown that but the von Mises plasticity condition no particular assumption as to minimum work has to be introduced. The shear angle solution is fixed by the prevalent state of stress, which can be expressed in terms of the ratio between the average value of the maximum shear stress and the plasticity constant of the material machined, which also holds when strain-hardening occurs.

A comparison is made with experimental results and a true strain-stress curve of the work-piece material, as obtained from the present theory is given.

### Resumé

Basé sur le modèle de cisaillement de Merchant et supposé qu'une équilibre globale existera entre valeurs moyennes des tensions dans un cas de tension plan, une relation d'angle de cisaillement est déduite par identifier la direction du allongement maximum contre la direction de la tension maximum principale.

Il est démontré que outre la Mises-Huber-Hencky condition de plasticité aucune supposition quelconque sera besoin d'introduire. La résolution de l'angle de cisaillement est complètement fixée par l'état de tension prépondérant, étant calé par l'idée de relation de la valeur moyenne de la tension de cisaillement maximum et la constante de plasticité du matériel travaillé. Aussi dans le domaine de tremper ce théorème reste valable.

Enfin un parallèle est tiré entre les résultats expérimentals et une courbe allongement-tension vrai du matériel de la pièce à travailler. obtenue de la théorie présente est montrée.

### Zusammenfassung

Gegründet auf dem Merchant'schen Modell des Schervorgangs bei der Zerspannung und mit der Annahme dass ein Gleichgewicht zwischen mittlere Werte der Spannungen in **einem ebenen Spannungszustand** bestehe, wird mittels Identifizierung der Richtung der Maximaldehnung mit derjenige der maximalen Hauptspannung im System eine Scherwinkelgleichung abgeleitet.

Es wird gezeigt dass ausser die von Mises-Huber-Hencky Bedingung keine weitere Voraussetzung bezüglich die Minimalarbeit notwendig ist.

Die Lösungen der Scherwinkelgleichung werden völlig bestimmt von dem herrschenden **Spannungszustand** wie festgelegt durch das Verhältnis zwischen den Wert der maximalen Scherspannung und die Plastizitätskonstante des Materials. Auch in Gebiete der Dehnungsverfestigung bewährt die Theorie seine Gültigkeit.

Die Voraussage der Theorie wird verglichen mit Experimentalergebnisse und eine Dehnungs-Spannungskurve für das bearbeitete Material, wie aus der Theorie hervor geht, wird dargestellt.

Nomenclature and units

0			
5	$\sigma_1, \sigma_3$	average principal stresses in shear zone	Nm <sup>-2</sup>
	$\sigma_x, \sigma_y$	average normal stresses in shear zone	Nm <sup>-2</sup>
10	$\tau_s$	average shear stress in shear plane	Nm <sup>-2</sup>
	$\tau_{max}$	average maximum shear stress in shear zone	Nm <sup>-2</sup>
15	$\varphi$	shear angle	
	$\beta$	friction angle	
	$\alpha$	rake angle	
20	$\psi$	direction of maximum crystal elongation with respect	
	$\Omega$	direction of maximum principal stress to the shear	
		plane	
25	$\gamma_s$	shear strain, $\tan \gamma_s = \tan(\varphi - \alpha) + \cot \varphi$	
30	$k$	plasticity constant	Nm <sup>-2</sup>
	$\sigma_e$	true tensile stress = $k\sqrt{3}$	Nm <sup>-2</sup>
	$g$	$\frac{2\sigma}{\sigma_y - \sigma_x}$ stress parameter	
35	$f$	$\frac{\tau_{max}}{k}$ ratio factor	
	$t$	feed	m/rev
40	$d$	depth of cut	m
	$r_c$	chip thickness ratio	
45	$v$	cutting speed	ms <sup>-1</sup>
	$e$	true strain	
50	$\epsilon$	natural strain = $Pn(1 + \epsilon)$	

## Contribution to the Mechanics of Machining.

### I. Introduction.

During the past decades a number of theories on the mechanics of machining has been published. Some of them investigate the entire state of stress, while otherwise equilibrium between average values of stress is assumed to be present in a geometric model of the cutting process. All theories are directed towards the formulation of a shear angle relation, which is an accessible equation between measurable quantities predicting an unique steady-state configuration for tool rake and friction angle. A hypothesis of minimum work is generally introduced in order to secure the uniqueness of the shear angle solution.

It even has been shown (1) that the search for uniqueness might considered being fruitless, as a range of steady-state solutions of the Merchant shear-plane type (2) is to be expected within permissible regions of the characteristic angles describing the geometry and the mechanics of the cutting process.

The present author reconsidered extant theories based on the assumption of global equilibrium between average values of stress.

It will be shown that when identifying the direction of maximum strain with the direction of maximum principal stress a shear angle relation can be formulated.

As to this it is not required to introduce any energy condition.

However when aiming at a shear angle solution an additional assumption has to be made with regard to the prevalent state of stress, which will prove to be equivalent to assuming a value of the maximum shear stress in the system in the case that materials behaving according to the von Mises condition of plasticity are being machined.

As a matter of fact the introduction of the von Mises condition implies accepting an energy condition. The latter however, regards exclusively the deformation of the workpiece material and does not refer to the cutting process as a whole.

A treatment of the problem along these lines will prove to be able to account for the strain-hardening properties of the material.

2. The direction of maximum strain (3), and a shear angle relation.

In the present theory the Merchant shear plane geometric model according to fig. 1 is accepted. The problem will be treated as a case of plane stress. From fig. 1 follows the geometric condition:

$$\sigma_y = \tau_s \tan(\varphi + \beta - \alpha) \quad \dots\dots (1)$$

and hence can be deduced from the Mohr equilibrium condition as represented in fig. 2:

$$\tan(\varphi + \beta - \alpha) = g \cot 2\Omega \quad (4) \quad \dots\dots (2)$$

where the parameter  $g$  defines the state of stress. As is clear from the figure this parameter can be expressed in terms of the prevalent stresses:

$$g = \frac{OP}{MP} = \frac{2\sigma_y}{\sigma_y - \sigma_x} \quad \dots\dots (3)$$

Thus  $g = 1$  defines a state of pure shear.

Merchant introduces the angle  $\Psi$  as the direction of the maximum value of the crystal elongation in the chip with respect to the shear plane, which can be interpreted as the direction of the maximum value of strain and hence in mechanical respect as the direction of the maximum (tensile) principal stress in the system.

This is expressed by:

$$\Psi = \Omega \quad \dots\dots (4)$$

Now, as shown in fig. 3 an element  $AF$  of the workpiece material will be transformed by the cutting process into the state  $AF'$ .

Its original position is fixed by an angle  $p$  relative to the coordinate system shown in the figure, the position after deformation is defined by the angle  $q$ .

The strain resulting from the deformation amounts:

$$\epsilon = \frac{AF' - AF}{AF} = \frac{\cos p}{\cos q} - 1 \quad \dots\dots (5)$$

Furthermore follows from fig. 3:

$$\tan q = \tan \gamma_s + \tan p \quad \dots\dots (6)$$



and hence:

$$\cos p = \left[ \frac{1}{1 + (\tan q - \tan \gamma_s)^2} \right]^{\frac{1}{2}} \dots (7)$$

Combining eqs. 5 and 7:

$$e = \frac{1}{\cos q} \left[ \frac{1}{1 + (\tan q - \tan \gamma_s)^2} \right]^{\frac{1}{2}} - 1 \dots (8)$$

The direction of the maximum strain in terms of the angle q by now follows from:

$$\frac{de}{dq} = 0$$

which renders:

$$\begin{aligned} \tan q_{e.\max} &= \cot \Psi = \\ &= \frac{1}{2} \tan \gamma_s + \left[ \frac{1}{4} \tan^2 \gamma_s + 1 \right]^{\frac{1}{2}} \dots (9) \end{aligned}$$

from which easily can be derived:

$$\cot 2\Psi = \frac{1}{2} \tan \gamma_s \dots (10)$$

Using the eqs. 4 and 2 :

$$\tan(\varphi + \beta - \alpha) = \frac{1}{2} g \tan \gamma_s \dots (11)$$

Substitution of the explicite expression for the shear strain in terms of  $\varphi$  and  $\alpha$  according to Merchant results in:

$$\tan(\varphi + \beta - \alpha) = \frac{1}{2} g \left[ \tan(\varphi - \alpha) + \cot \varphi \right] \dots (12)$$

which is a shear angle relation valid in a state of stress defined by the parameter g.

The value of the maximum strain follows from eqs. 9 and 8:

$$\begin{aligned} e_{\max} &= \left[ 1 + \tan \gamma_s \left\{ \frac{1}{2} \tan \gamma_s + (1 + \frac{1}{4} \tan^2 \gamma_s)^{\frac{1}{2}} \right\} \right]^{\frac{1}{2}} - 1 \\ \text{Hence: } \xi &= \frac{1}{2} \ln \left[ 1 + \tan \gamma_s \left\{ \frac{1}{2} \tan \gamma_s + (1 + \frac{1}{4} \tan^2 \gamma_s)^{\frac{1}{2}} \right\} \right] \dots (13) \end{aligned}$$

By now it is possible to derive the shear angle relation eq.11 in a direct way from the Mohr equilibrium diagram fig.2.

According to eqs.4 and 10 the equality holds:

$$\frac{MP}{PQ} = \cot 2\Omega = \frac{1}{2} \tan \gamma_s$$

and as:  $MP = \frac{1}{2} PR$

follows:  $\angle PQR = \gamma_s$

by which a graphical interpretation is obtained of the relation between the shear strain and the characteristic angle ( $\varphi + \beta - \alpha$ ).

From this follows the shear angle relation:

$$\tan(\varphi + \beta - \alpha) = \frac{OP}{MP} \frac{1}{2} \tan \gamma_s$$

\* and hence:

$$\epsilon = \frac{OP}{MP}$$

as already has been defined in eq. 3.

Finally is remarked that the shear strain which in origin has been defined merely as a geometric quantity can be expressed in terms of stress, as also can be concluded from fig.2:

$$\tan \gamma_s = \frac{\sigma_x - \sigma_y}{\tau_s}$$

### 3. The stress parameter $g$ .

In the case that the value of the stress parameter  $g$  is known, the shear angle relation eq. 12 allows for a shear angle solution, i.e. the determination of the shear angle in dependence of the friction angle, with the rake angle as a parameter.

As eq. 13 predicts that the strain in the material can be expressed merely in terms of the shear strain, and thus in terms of the shear angle  $\varphi$ , this means that an analytical formulation will be obtained accounting for the interaction between the friction on the rake of the tool - whatsoever the physical background of this particular process might be - and the deformation of the workpiece material in the shear zone. Thus it is important to investigate the physical meaning of the stress parameter  $g$ , apart from its definition eq. 3.

The general von Mises plasticity condition reduces to:

$$\sigma_x^2 + \sigma_y^2 - \sigma_x \sigma_y + 3\tau_s^2 = 3k^2 \quad \dots (14)$$

in the state of plane stress.

The plasticity constant  $k$  is considered being a function of the strain  $\epsilon$ , and hence eq. 14 remains valid when strain-hardening occurs.

This means that the plasticity ellipse, when transferred to the coordinate system of principal stresses:

$$\sigma_1^2 + \sigma_3^2 - \sigma_1 \sigma_3 = 3k^2 \quad \dots (15)$$

shows semi-axes of variable magnitude in dependence of the state of strain at a given strain rate.

The equilibrium condition according to fig.2 requires:

$$\sigma_x = \sigma_y - 2\tau_s \cot 2\Omega \quad \dots (16)$$

The geometric condition as to the stresses has been formulated in eq. 1.

Now the solution of eqs. 1, 14 and 16 refers to a state of stress satisfying simultaneously the geometric condition prescribed by the shear plane model, the condition of global equilibrium and finally the condition of plasticity at the given state of strain and strain rate.

The solution is:

$$\tau_E = \frac{k\sqrt{3}}{\left[ \tan^2(\varphi + \beta - \alpha) - 2\tan(\varphi + \beta - \alpha)\cot 2\Omega + 4\cot^2 2\Omega + 3 \right]^{1/2}} \dots (17)$$

$$\sigma_y = \frac{k\sqrt{3} \tan(\varphi + \beta - \alpha)}{\left[ \tan^2(\varphi + \beta - \alpha) - 2\tan(\varphi + \beta - \alpha)\cot 2\Omega + 4\cot^2 2\Omega + 3 \right]^{1/2}}$$

while  $\sigma_x$  can be solved from eq. 14.

The counterpart of eq. 16 also goes from fig. 2:

$$\tau_E = -\tau_{\max} \sin 2\Omega \dots (18)$$

and:

$$2\tau_{\max} = \sigma_1 - \sigma_3$$

Now two different extreme situations of stress may occur:

1) a state of linear stress:

$$\sigma_1 \neq 0 \quad \text{or} \quad \sigma_1 = 0$$

$$\sigma_3 = 0 \quad \sigma_3 \neq 0$$

In this case follows from eqs. 15 and 18 :

$$\tau_{\max} = \frac{1}{2} k\sqrt{3}$$

2) a state of pure shear :

$$\sigma_3 = -\sigma_1$$

where follows from the same equations :

$$\tau_{\max} = k$$

In general thus can be put:

$$\tau_{\max} = f.k$$

and :

$$\tau_E = -fk \sin 2\Omega$$

where :

$$\frac{1}{2}\sqrt{3} \leq f \leq 1 \dots (19)$$

From this it is clear that any a priori assumption with regard to the value of the maximum shear stress in terms of the plasticity constant defines a state of stress.

In particular the condition  $\tau_{\max} = k$ , which is quite common in extant theories, defines a state of pure shear.

Substitution of eq. 19 into eq. 17 and using eqs. 4 and 10 again leads to a shear angle relation:

$$\tan(\varphi + \beta - \alpha) = \left[ 1 \pm \left\{ 3 \left( 1 + \frac{4}{\tan^2 \gamma_s} \right) \left( \frac{1}{f^2} - 1 \right) \right\}^{\frac{1}{2}} \right]^{\frac{1}{2}} \tan \gamma_s \dots (20)$$

Comparison with eq. 11 shows that holds:

$$g = 1 \pm \left[ 3 \left( 1 + \frac{4}{\tan^2 \gamma_s} \right) \left( \frac{1}{f^2} - 1 \right) \right]^{\frac{1}{2}} \dots (21)$$

from which it is obvious that the state of stress defined in terms of the parameter  $g$  at a given state of strain has its physical origin in the ratio  $f$  between the average value of the maximum shear stress and the plasticity constant of the material.

The positive sign in eq. 21 implies:

$$|\sigma_y| > \sigma_x$$

the negative sign means:

$$|\sigma_y| < \sigma_x$$

The condition  $f = 1$  is compatible with  $g = 1$ , and defines a state of pure shear, as is shown before.

#### 4. Shear angle solutions.

##### 1. The case of pure shear.

From the foregoing it will be clear that the shear angle relation eq. 20 or 22 reduces to :

$$\tan(\varphi + \beta - \alpha) = \frac{1}{2} \left[ \tan(\varphi - \alpha) + \cot \varphi \right] \quad \dots(22)$$

from which  $\varphi$  can be solved as a function of  $\beta$ , for given values of the rake angle  $\alpha$ .

The solution has been plotted in fig. 4, where in the usual way of representation the shear angle  $\varphi$  appears as a function of the angle  $\beta - \alpha$ .

A remarkable fact is that in the present theory the rake angle operates as a parameter which definitely influences the solution obtained.

This is shown for the values  $\alpha = \pm 30^\circ$  and  $\alpha = 0^\circ$ .

As a comparison also the Merchant and Lee and Shaffer (5) solutions have been plotted.

The present theory proves to arrive at values intermediate between those predicted by the theories mentioned, as it should do whenever it would have a chance to cover reality. It is observed that in the interval  $0 < \varphi < \frac{1}{2}\pi$ , the theory apparently does not allow for unique solutions.

As to deal with this it is sufficient to remark that the shear strain passes through a minimum value as a function of the shear angle  $\varphi$  :

$$\frac{d \tan \gamma_s}{d \varphi} = \frac{1}{\cos^2(\varphi - \alpha)} - \frac{1}{\sin^2 \varphi} = 0$$

Hence the minimum value of the shear strain is reached at:

$$\varphi = \frac{1}{2}\pi - \frac{1}{2}\alpha \quad \dots(23)$$

where the friction angle  $\beta$  has the value zero as can be checked by substitution of eq.23 into eq. 22.

In this state the cutting process dissipates energy only by deformation of the workpiece material in absence of friction on the rake of the tool.

It seems obvious that this never can be a physical reality and thus the uniqueness of the solution of eq. 22 is secured by:

$$\begin{aligned} \beta &> 0 \\ \varphi &< \frac{1}{2}\pi + \frac{1}{2}\alpha \end{aligned} \quad \dots(24)$$

In the figure 4 the region of physical significance is restricted to :

$$\varphi = 30^\circ \text{ for } \alpha = -30^\circ, \text{ to } \varphi = 45^\circ \text{ for } \alpha = 0^\circ \text{ and to } \\ \varphi = 60^\circ \text{ for } \alpha = +30^\circ.$$

The solutions again prove to be unique.

## 2. The case of a general state of stress.

As discussed before the state of general stress prevails when:

$$f = \frac{\tau_{\max}}{k} < 1$$

is

The system governed by the shear angle relation eq.20, from which after expressing the shear strain in terms of shear angle and rake angle, shear angle solutions can be obtained with both  $f$  and  $\alpha$  as parameters.

As is shown in fig. 5 the ratio  $f$  has a very strong influence on the course of the shear angle solution, and so it does in particular in the region close to  $f = 1$ .

When reading fig. 5 it should be kept in mind that every value of the parameter  $f$  gives to two different shear angle solutions, corresponding to the choice of the sign in the eqs. 20 and 21, and hence dependent on the modulus of the ratio between the principal stresses, which can be expressed in terms of  $g \gg 1$ , as shown before.

When it is accepted that the average value of the maximum shear stress as a resultant of a hypothetic stress distribution might differ up to about 2% from the plasticity constant of the material machined, quite a number of the observations published in current literature is covered by the present theory.

It even might be that the extreme sensitiveness of the shear angle solution with respect to the state of stress suggests a lack of unique solutions of the problem.

A more complete picture gives fig. 6 where the effect of both of the two parameters is shown simultaneously under the condition  $g \gg 1$ , as appears to be usual in a majority of the practical cases investigated.

It is observed that the influence of the rake angle decreases rapidly as the value of  $f$  decreases, i.e. when the average behaviour of the system moves out of the state of pure shear.

In conclusion is shown the figure 7 where experimental data as used by Oxley (6) as an example are compared with the present theory.

## 5. Experimental results.

A major difficulty in verifying shear-angle solutions arises from measuring the shear-angle  $\phi$  in an accuracy comparable with which can be obtained when measuring the friction angle  $\beta$  by means of dynamometry. As a rule cutting forces will be recorded during a considerable length of time and hence an average value of the friction angle can be determined with fair precision.

On the contrary determination of the shear-angle depends on measuring the chip-ratio from samples of the chip. A vast number of samples should be taken in order to arrive at an accuracy comparable with the one obtained by dynamometry.

\* Now, in a program of investigation of cutting temperatures, an extensive study has been made of the behaviour of the chip contact length in relation to the cutting conditions (7).

When machining obliquely an annealed steel C 45 with a carbide tool of the grade S 2 (1 S 0 - P 20) a definite relation between feed, speed and chip ratio proves to exist:

$$r_c = \frac{0,615 t}{0,205 \cdot 10^{-3} + 0,850 t - 0,029 \cdot 10^{-3} v} \quad \dots (25)$$

in the speed range  $1 < v < 5 \text{ ms}^{-1}$ , in the feed range  $0,2 \cdot 10^{-3} < t < 1,0 \cdot 10^{-3} \text{ m/rev.}$ , and at the depth of cut of  $d = 3 \cdot 10^{-3} \text{ m}$ .

As eq.25 has been obtained from the study of the average behaviour of the chip contact length as recorded in a natural way in the wear pattern on the rake of the tool, the accuracy in determining the shear-angle from it proves to be about the same in determining the friction-angle from recordings obtained with a sensitive strain-gage dynamometer (8).

Statistical evaluation shows a relative error of 2% in the shear-angle and a relative error of 2,5% in the angle  $\beta - \alpha$ .

The experimental results have been plotted in fig. 8, where both values of the shear-angle obtained by use of eq. 25 and those obtained by direct measurement of the chip ratio have been used. The presence of a systematic error is evident.

The agreement with the present shear-angle relation eq. 22 is pretty good, from which it might be concluded that the material is machined in an average state of pure shear, and probably behaves according to the von Mises condition of plasticity.



A second series of experiments has been performed with a negative value of the rake-angle. The results are shown in fig. 9 from which the conclusion might be the same.

In conclusion it is remarked that eq. 17 when used in connection with dynamometric experiments allows for investigation into the plastic behaviour of the workpiece material under machining conditions when assuming validity of the von Mises condition.

When using:  $\cot 2\Omega = \frac{1}{2} \tan \gamma_s$ ,

as has been derived earlier, the eq. 17 can be written like:

$$\sigma_\epsilon = k\sqrt{3} = \tau_s \left[ \tan^2(\varphi + \beta - \alpha) - \tan(\varphi + \beta - \alpha) \tan \gamma_s + \tan^2 \gamma_s + 3 \right]^{\frac{1}{2}} \quad \dots (26)$$

by which the true stress  $\sigma_\epsilon$  is expressed in dynamometric quantities and hence can be calculated from numerical experimental values. The amount of computing work is considerably reduced when it is known that the material is machined in a state of pure shear, as in this case eq. 17 reduces to  $\tau_s = k \sin 2\Omega$ .

As derived in eq. 13 the true strain can be calculated from the prevalent value of the shear strain and thus a stress-strain relation in the region of machining conditions can be plotted.

This is shown in fig. 10 as based on the measurements of fig. 8 when machining an annealed steel C 45. The mechanical properties of the material are illustrated in fig. 11, which represents the true stress-strain relation as obtained from a step by step interrupted tensile test.

The yield point of the material is reached at a true stress of  $3.42 \cdot 10^8 \text{ Nm}^{-2}$  ( $34.2 \text{ kgf/mm}^2$ ), and fracture occurs at a value of the stress close to  $10^9 \text{ Nm}^{-2}$  ( $100 \text{ kgf/mm}^2$ ).

In the region between, the stress-strain curve behaves almost perfectly in accordance with the power function:

$$\sigma_\epsilon = 1.18 \cdot 10^9 \epsilon^{0.22}$$

From fig. 10 it may be concluded that in the region of high strain and strain rate, as typical for conditions of machining, approximately holds:

$$\sigma_\epsilon = 1.56 \cdot 10^9 \epsilon^{1.32}$$

A simultaneous representation of both of the two relations is given in fig. 12.

0  
5  
10  
15  
20  
25  
30  
35  
40  
45  
50

This preliminary investigation into the plastic behaviour of the material under conditions of machining does not conclude to a significant influence of the cutting speed and hence of the strain rate on the strain hardening. By far the most important factor in strain hardening, referring to the particular material studied, appears to be the value of the strain.

So far, however, no experiments have been performed in the region of strain which links the ultimate values of the quasi-static tensile test with the minimum values achievable in machining, in order to investigate whether some continuous transition from the one region to the other might exist.

References

1. Hill, R. J. Mech. Phys. Solids 3, (1954) 47
2. Merchant, M.E. J. Appl. Phys. 11, (1940) 3 230  
 J. Appl. Phys. 16, (1945) 5 267  
 J. Appl. Phys. 16, (1945) 6 318
3. Zweckhorst, E.T.W. Metaalbewerking 29, (1964) 22 471
4. Colding, B.N. Thesis, Stockholm 1959, p. 39
5. Lee, F.H. and Shaffer, B.W. J. Appl. Mech. 73, (1951) 405
6. \* Oxley, P.L.B. Int. J. Mach. Tool Des. Res. 2 (1962) 219  
 Prod. Eng. 43 (1964) 609  
 Oxley, P.L.B. and Hatton, A.P. Int. J. Mech. Sci. 5 (1963) 41
7. Hulst, A.P.A.J. Lab. research report WT 129, WT 135  
 Eindhoven (1965)
8. Ten Horn, B.L. Metaalbewerking 24 (1958) 3,39  
 and Schürmann, R.A. Metaalbewerking 24 (1958) 5,85

Fig. 1.

The Merchant shear plane model and the geometric stress condition:

$$\sigma_y = \tau_s \tan(\varphi + \beta - \alpha)$$



TECHNISCHE HOGESCHOOL EINDHOVEN

LABORATORIUM VOOR MECHANISCHE TECHNOLOGIE  
EN WERKPLAATSTECHNIEK

fig.1.

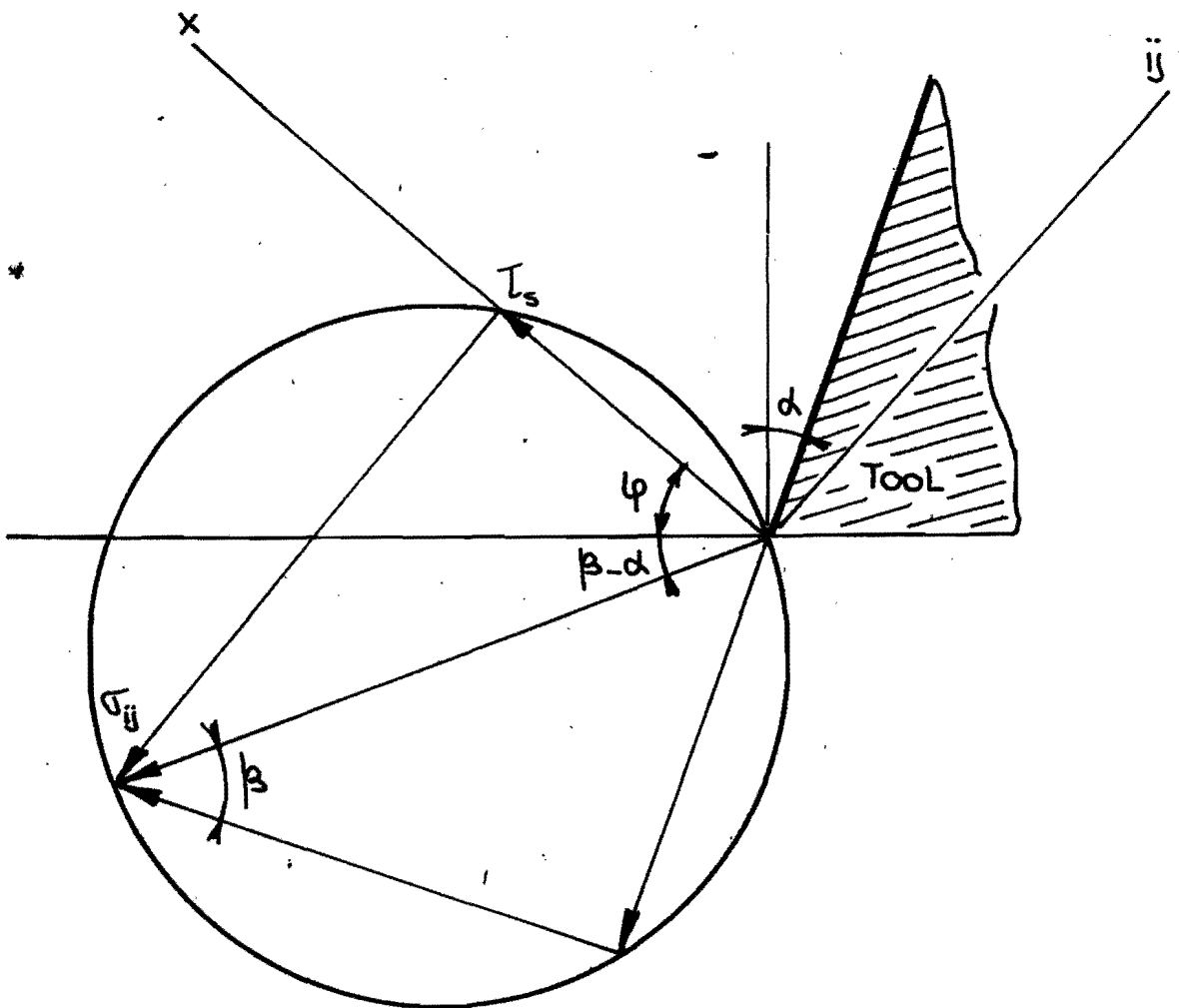


Fig. 2

The Mohr equilibrium condition and the stress parameter

$$g = \frac{OP}{MP} = \frac{2\sigma_y}{\sigma_y - \sigma_x} \quad , \quad \text{from which is derived:}$$

$$\tan(\varphi + \beta - \alpha) = g \cot 2\Omega$$

If the direction  $\Omega$  of the maximum principal stress is identified with the direction of maximum strain, it can be shown that holds:

$$\cot 2\Omega = \frac{1}{2} \tan \gamma_s$$

from which follows:

$$\angle PQR = \gamma_s$$

and hence the shear angle, equation:

$$\tan(\varphi + \beta - \alpha) = \frac{1}{2} g \tan \gamma_s = \frac{1}{2} g \left[ \tan(\varphi - \alpha) + \cot \varphi \right] .$$



TECHNISCHE HOGESCHOOL EINDHOVEN  
 LABORATORIUM VOOR MECHANISCHE TECHNOLOGIE  
 EN WERKPLAATSTECHNIEK

fig. 2

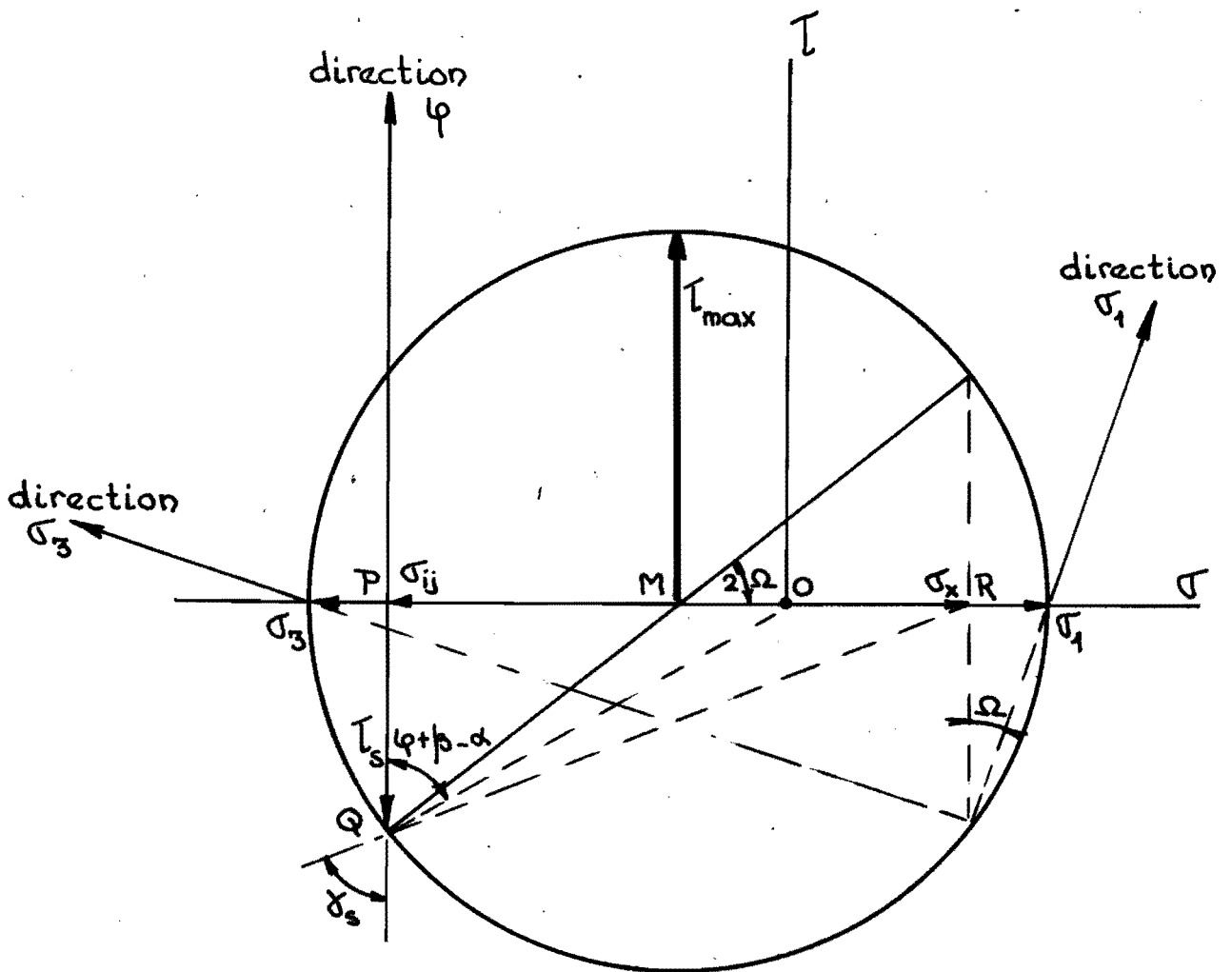
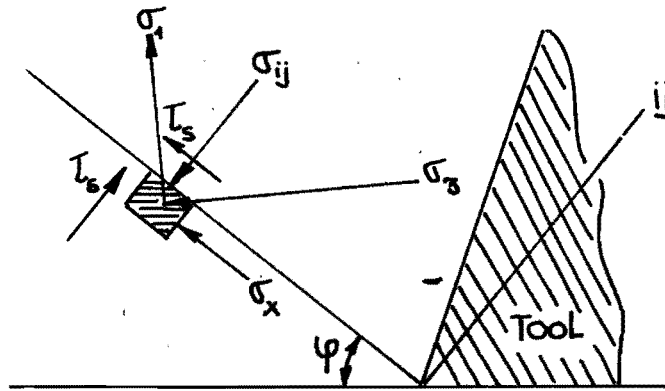


Fig. 3

Determination of the direction  $q = \frac{\pi}{2} - \psi = \frac{\pi}{2} - \Omega$

of maximum strain.

An element of material AF is deformed by the cutting process into the state AF' and is thus strained to the amount

$$\epsilon = \frac{AF'}{AF} - 1 = \frac{\cos p'}{\cos q} - 1$$

From the condition  $\frac{d\epsilon}{dq} = 0$ , can be derived the direction of maximum strain:

$$\cot 2\Omega = \frac{1}{2} \tan \gamma_s = \cot 2\psi$$





Fig. 4

The shear angle solution eq. 20 in the state of stress of pure shear, as defined by the condition  $\sigma_3 = -\sigma_1$  and hence by  $\tau_{\max} = k$  or  $f = g$ .

Shown is the effect of the rake angle as a parameter.

A comparison is made with both the Merchant and the Lee and Schaffer solutions.



TECHNISCHE HOGESCHOOL EINDHOVEN

LABORATORIUM VOOR MECHANISCHE TECHNOLOGIE  
EN WERKPLAATSTECHNIEK

fig. 4

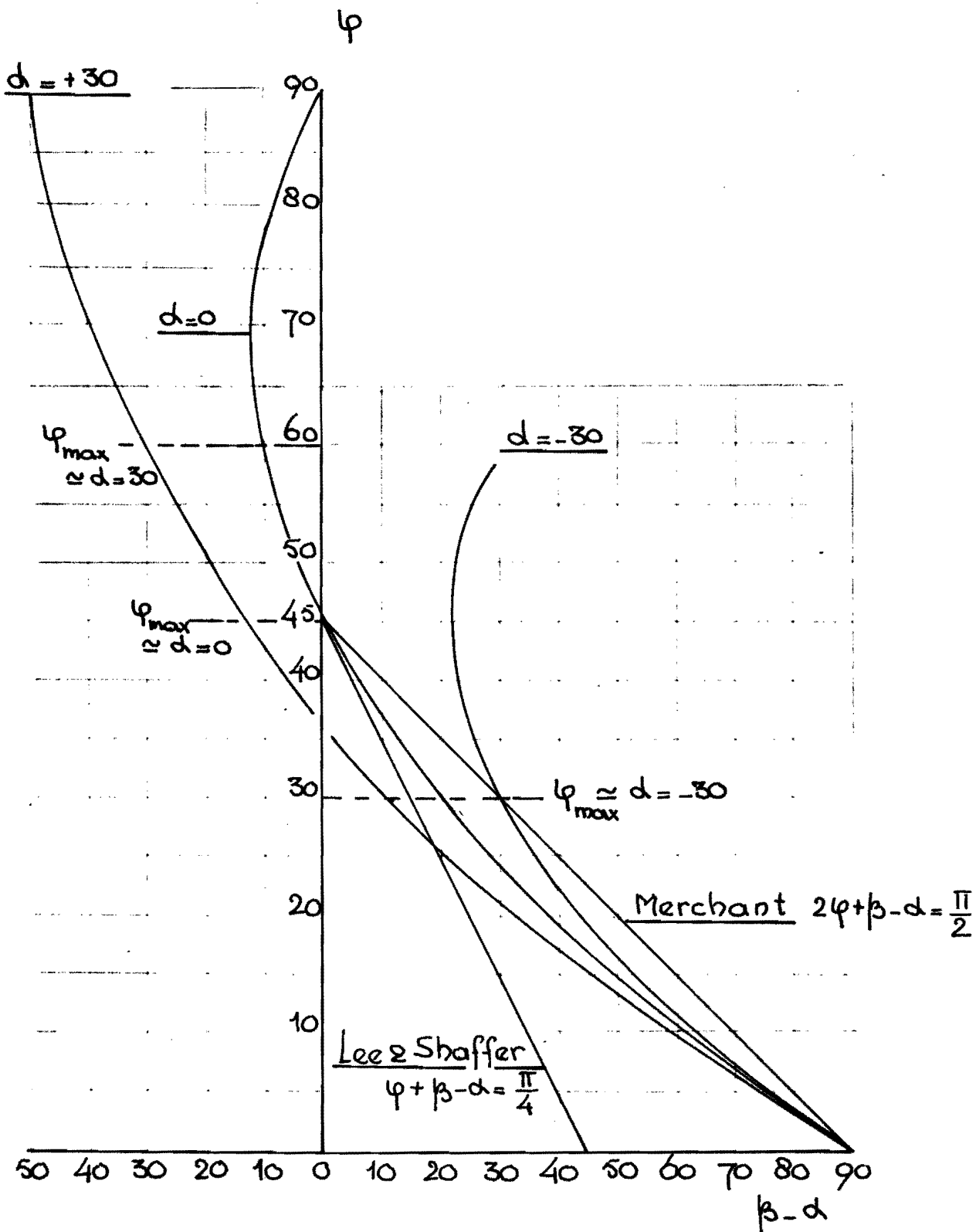


Fig. 5

The shear angle relation eq. 20 for the value of the rake angle  $\alpha = 0$  and different values of the ratio

$$f = \frac{\tau_{\max}}{k}$$

Shown is the sensitiveness of the solution with respect to minor changes in  $f$  in the region close to  $f = 1$ . Both the possible solutions have been plotted according to the value  $f = 0,99$ , corresponding with the two possible different states of average stress.



TECHNISCHE HOGESCHOOL EINDHOVEN  
 LABORATORIUM VOOR MECHANISCHE TECHNOLOGIE  
 EN WERKPLAATSTECHNIEK

fig. 5

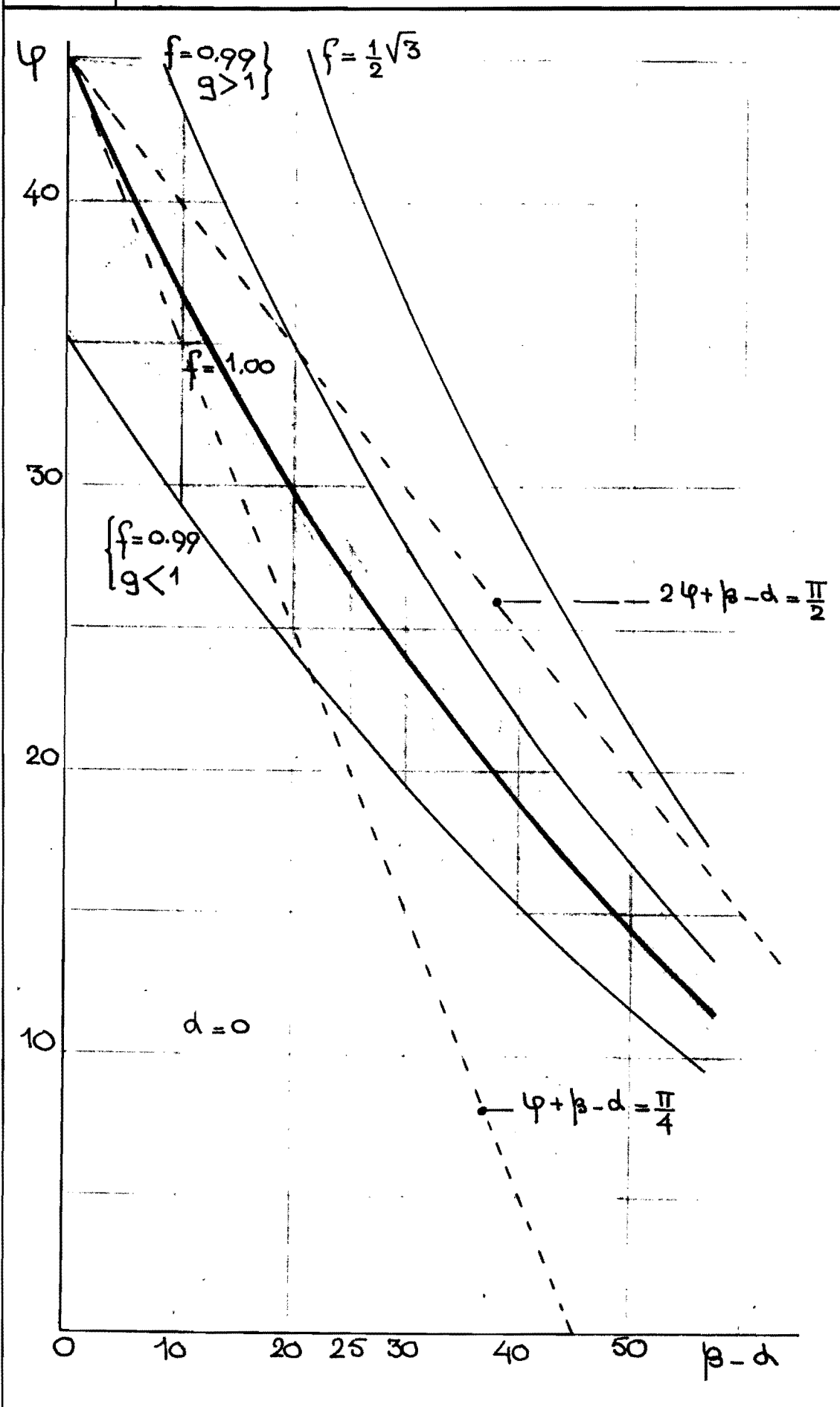


Fig. 6

The shear angle relation eq.20 for different values of both the rake angle  $\alpha$  and the ratio  $f$ .

Only the solutions corresponding with the positive sign in the eqs. 20 and 21 have been plotted, as will refer to the majority of the practical cases.

To be observed is the decreasing importance of the rake angle as the ratio  $f$  decreases due to the moving of the system out of a state of pure shear.



TECHNISCHE HOGESCHOOL EINDHOVEN  
 LABORATORIUM VOOR MECHANISCHE TECHNOLOGIE  
 EN WERKPLAATSTECHNIEK

fig.6

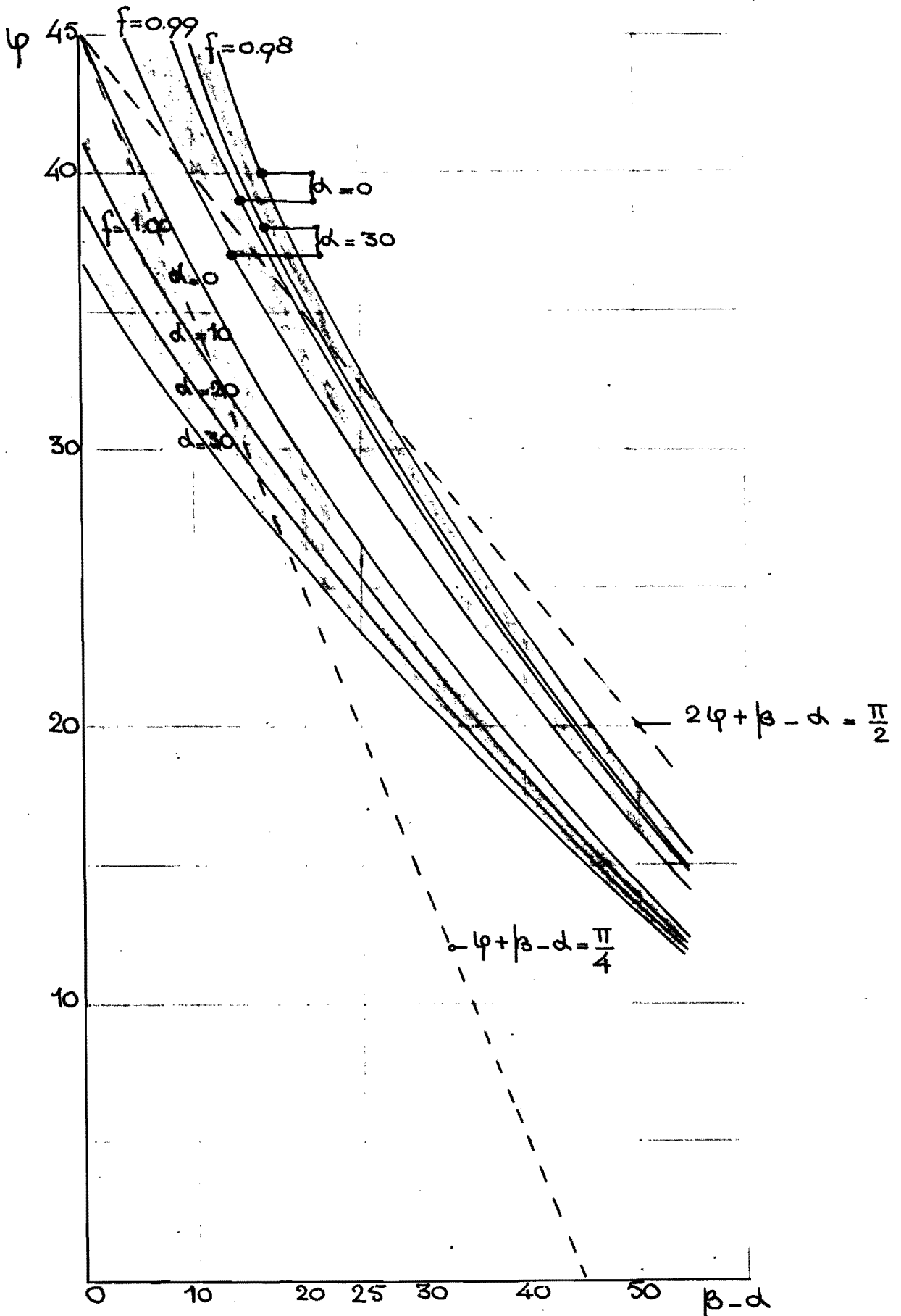


Fig. 7

Comparison of a number of shear angle values as used by Oxley (6) with the predictions of the present theory according to eq. 20.

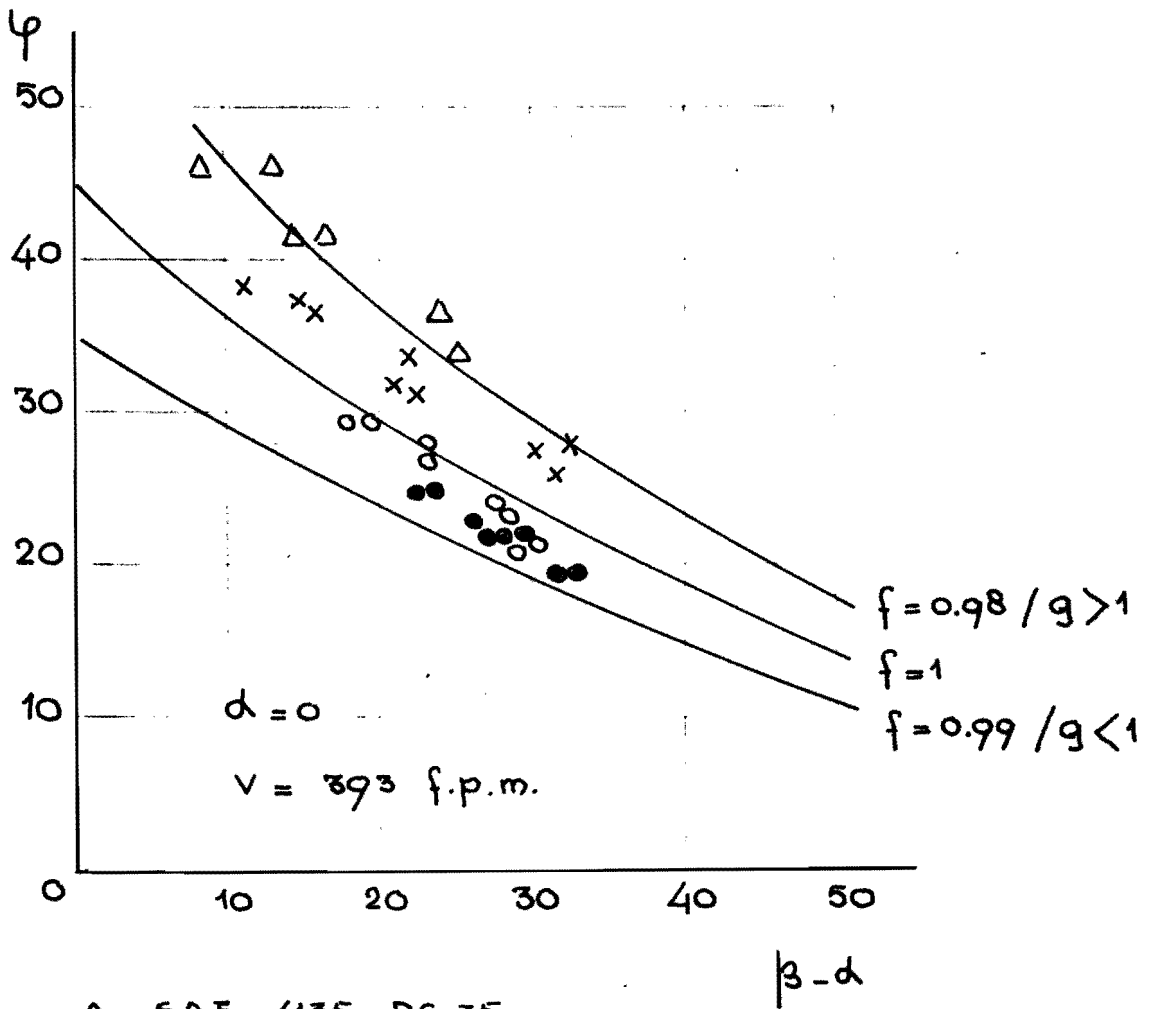




TECHNISCHE HOGESCHOOL EINDHOVEN

LABORATORIUM VOOR MECHANISCHE TECHNOLOGIE  
EN WERKPLAATSTECHNIEK

fig. 7



- $\Delta$  SAE 4135 RC-35
- x SAE 4135 RC-26
- o SAE 4135 - as received
- SAE 4135 - annealed

0  
5  
10  
15  
20  
25  
30  
35  
40  
45  
50Fig. 8 and fig. 9

Comparison of experimental results with the predictions of the present theory, eq. 22 when machining an annealed steel C 45.

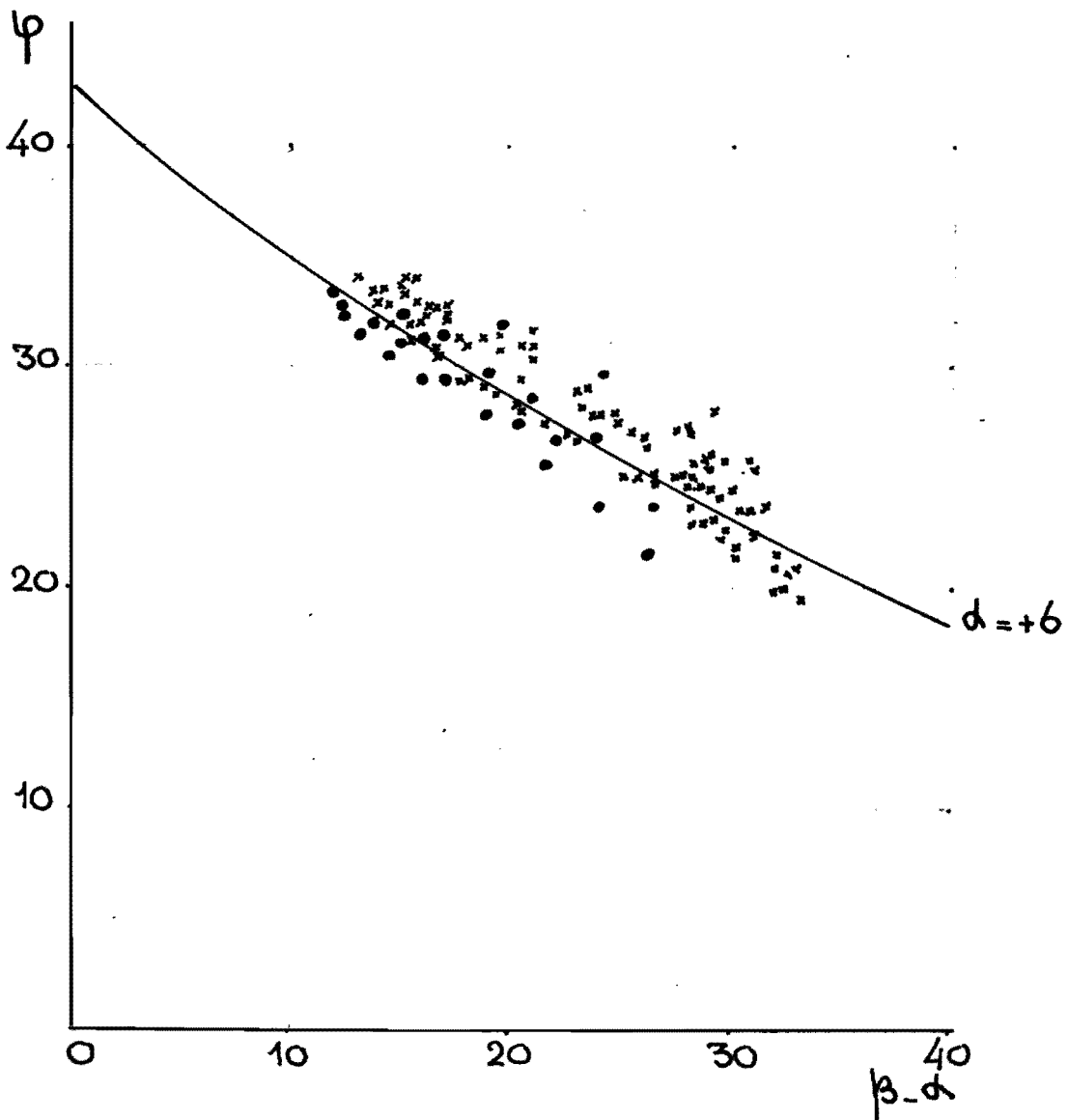
Speed range  $1 \div 5 \text{ ms}^{-1}$ , feed range  $0,2 \div 1,0 \text{ mm/rev}$ , depth of cut 3 mm.

● = determined indirectly from chip ratio relation eq. 25  
x = measured directly from chip ratio by sampling.



TECHNISCHE HOGESCHOOL EINDHOVEN  
LABORATORIUM VOOR MECHANISCHE TECHNOLOGIE  
EN WERKPLAATSTECHNIEK

fig. 8





TECHNISCHE HOGESCHOOL EINDHOVEN

LABORATORIUM VOOR MECHANISCHE TECHNOLOGIE  
EN WERKPLAATSTECHNIEK

*fig. 9*

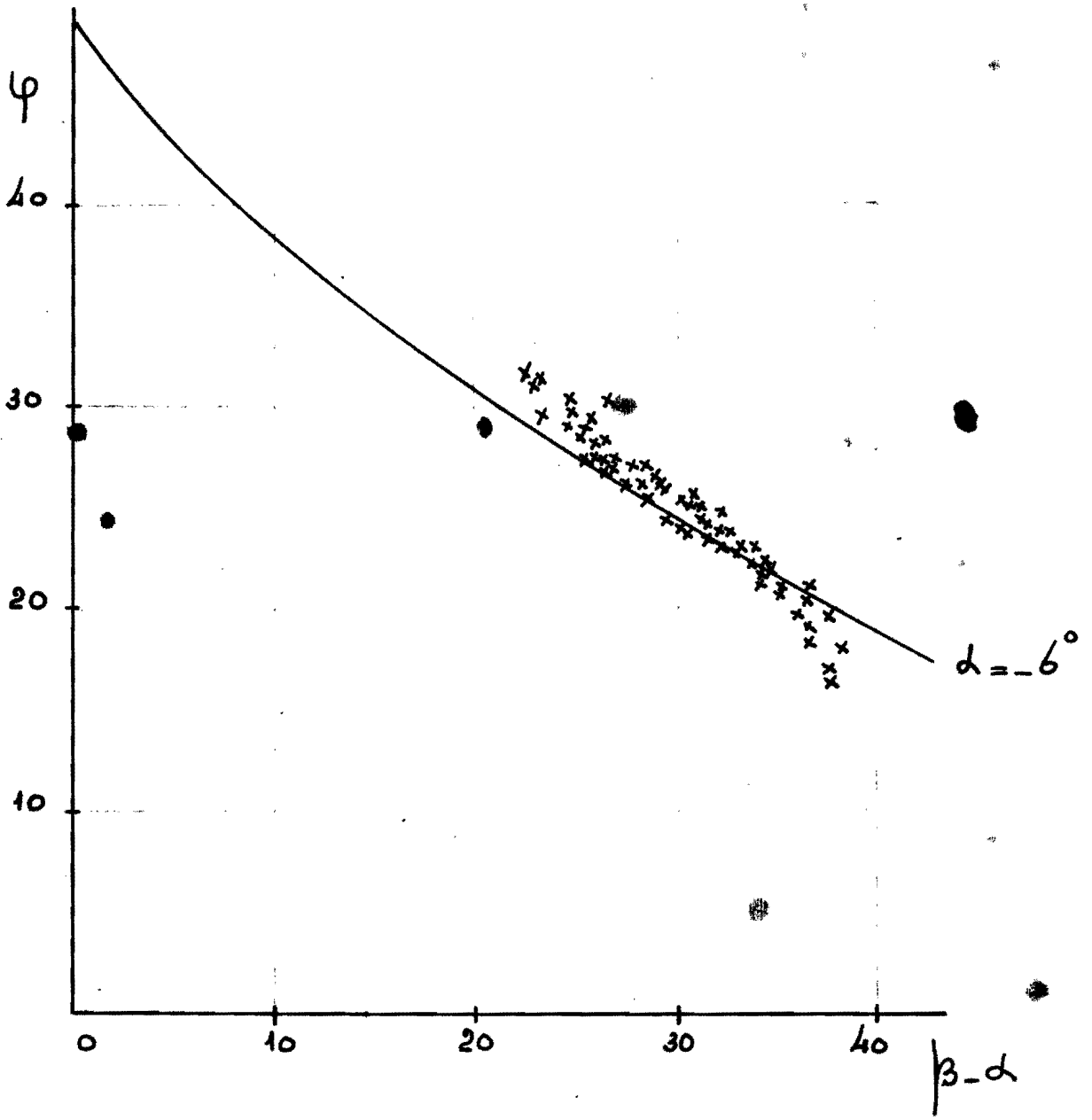


Fig. 10

The stress-strain relation of an annealed steel C 45 in metal cutting, according to eqs. 13 and 26 and based on the measurements of fig. 8.



TECHNISCHE HOGESCHOOL EINDHOVEN

LABORATORIUM VOOR MECHANISCHE TECHNOLOGIE  
EN WERKPLAATSTECHNIEK

fig.10

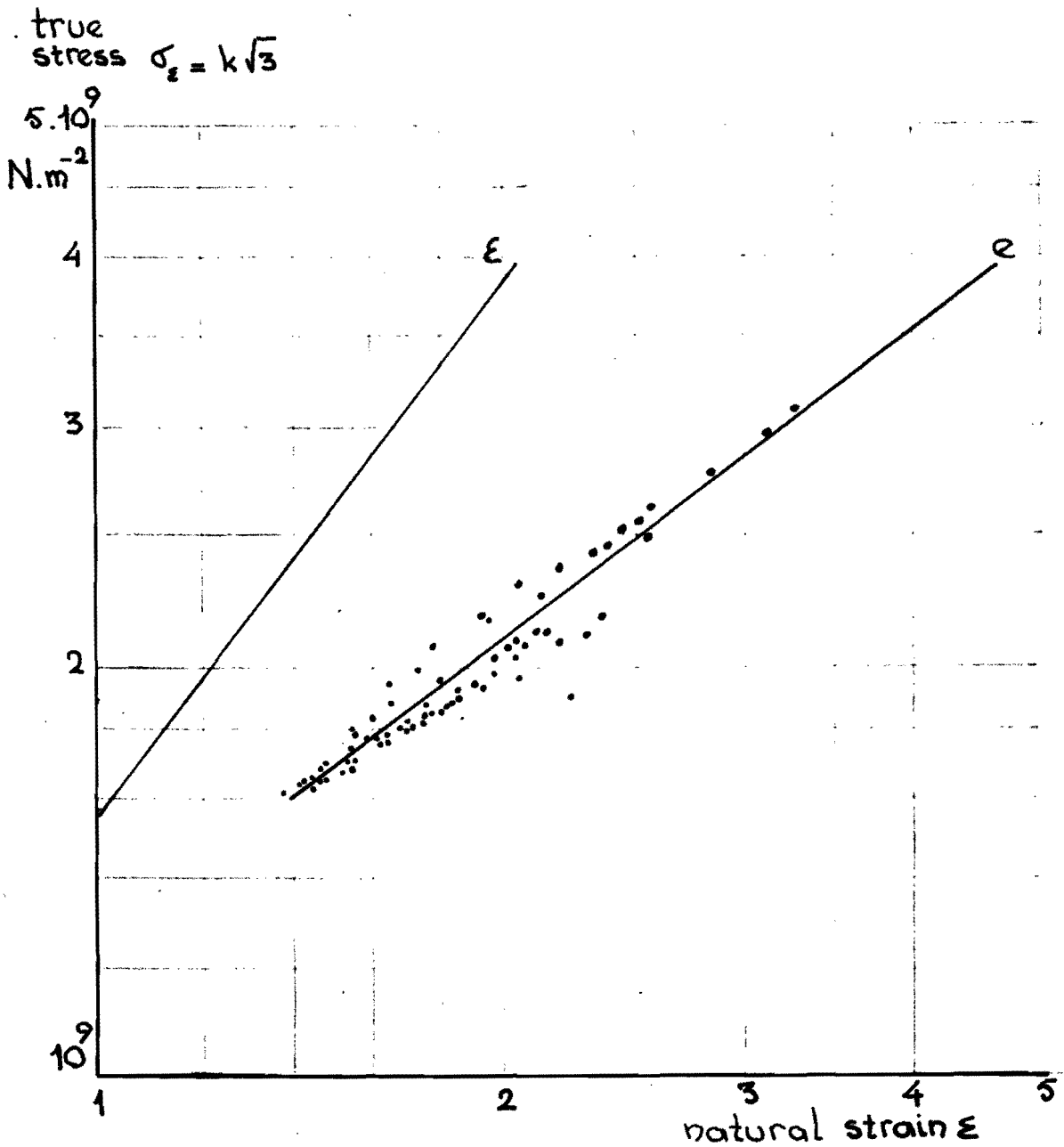


Fig. 11

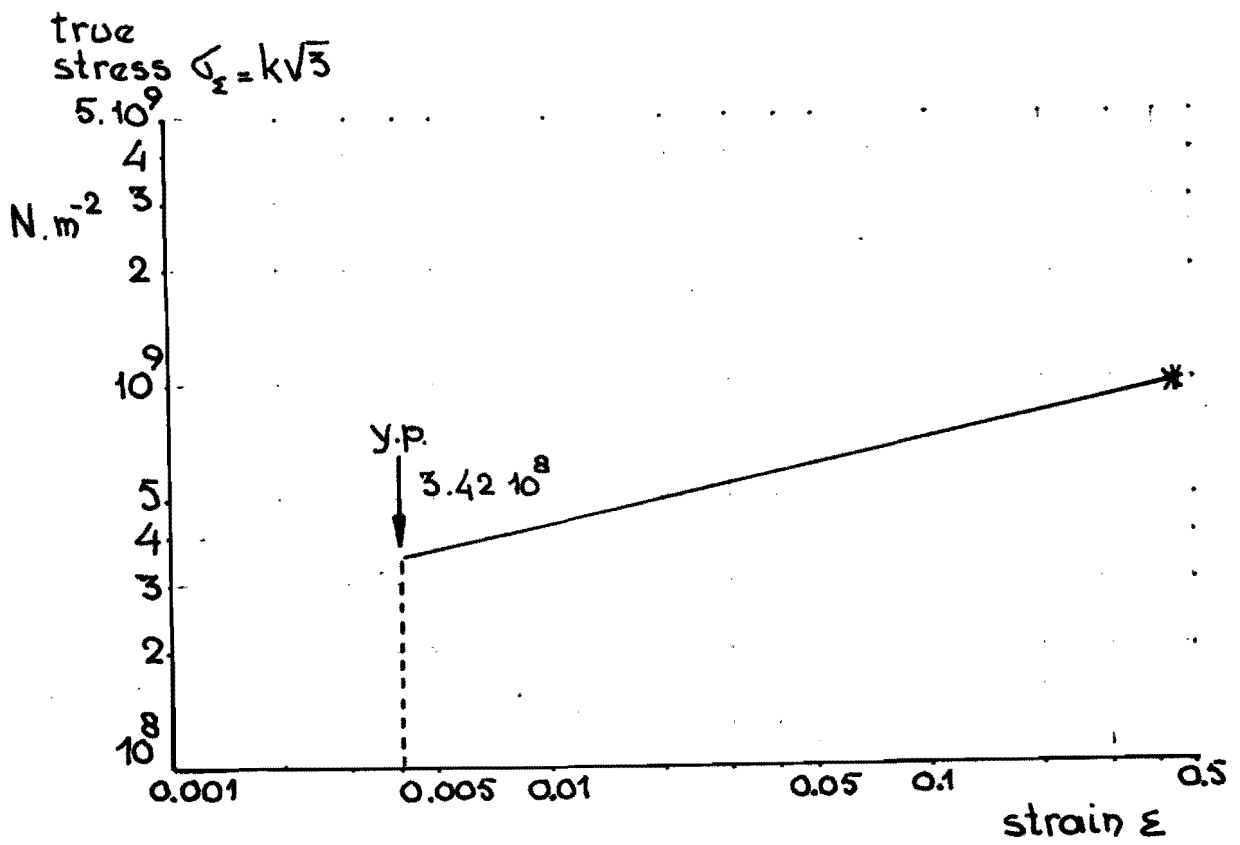
The stress-strain relation of an annealed steel C 45 as obtained from a step by step interrupted tensile test. Yield point  $3.42 \cdot 10^8 \text{ Nm}^{-2}$  ( $34.2 \text{ kgf/mm}^2$ ), fracture  $10^9 \text{ Nm}^{-2}$  ( $100 \text{ kgf/mm}^2$ ).



TECHNISCHE HOGESCHOOL EINDHOVEN

LABORATORIUM VOOR MECHANISCHE TECHNOLOGIE  
EN WERKPLAATSTECHNIEK

fig. 11





0  
5  
10  
15  
20  
25  
30  
35  
40  
45  
50

Fig. 12

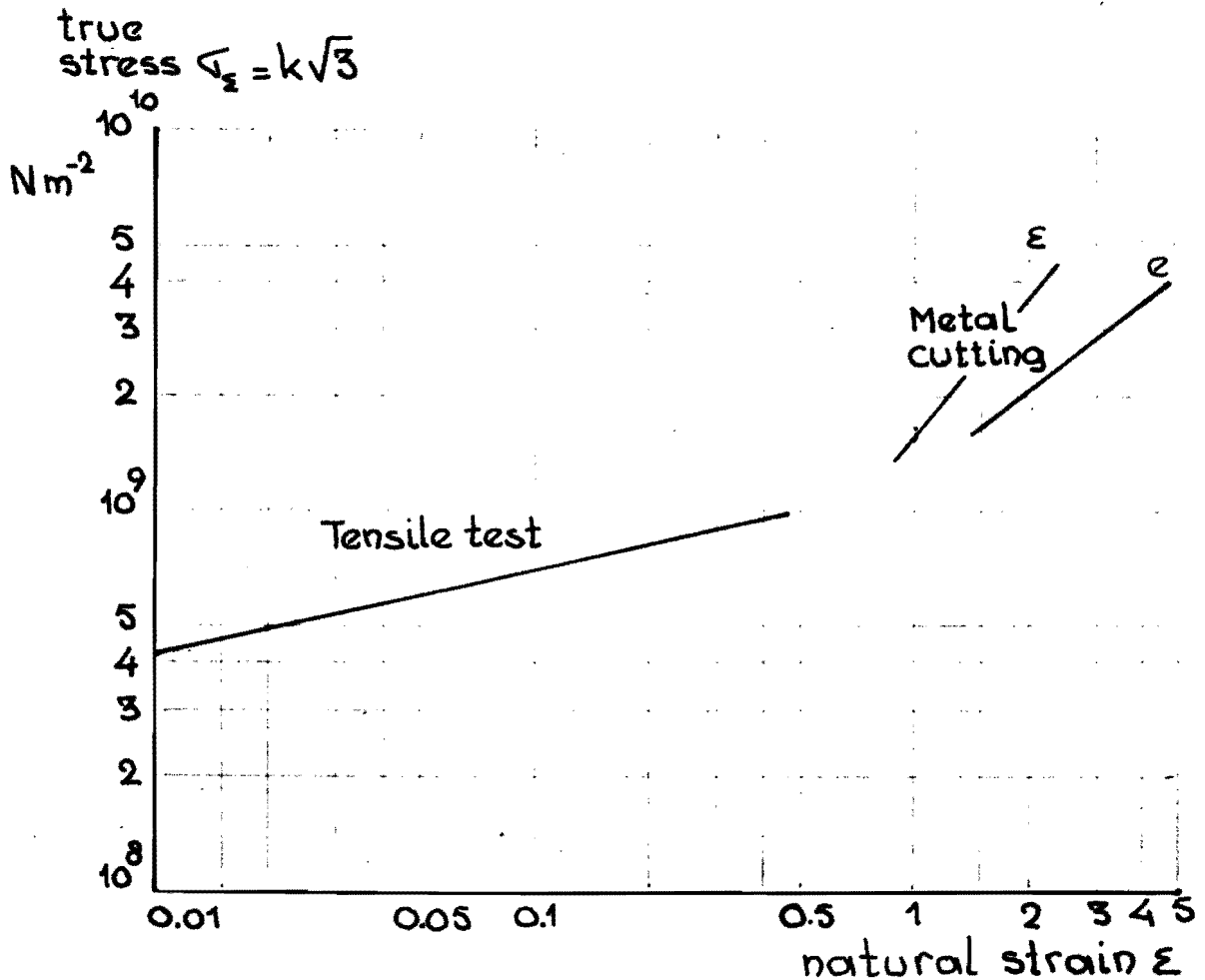
Comparison of strain hardening in a tensile test and in the process of metal cutting.



TECHNISCHE HOGESCHOOL EINDHOVEN

LABORATORIUM VOOR MECHANISCHE TECHNOLOGIE  
EN WERKPLAATSTECHNIEK

fig.12



Appendix

concerning numerical values obtained by dynamometry when machining steel S 45 (annealed) with a carbide tool grade S2(P20)

In these reports the following symbols are used:

- v = cutting speed
- a = feed
- $P_v$  = main cutting force
- $P_A$  = force in direction of feed, thrust force
- $\tau_s$  = average shear stress in shear plane
- $\lambda$  = inverse value of chip thickness ratio
- $\gamma$  = shear strain
- $\varphi$  = shear angle
- $\beta$  = friction angle
- $\alpha$  = rake angle
- $\epsilon$  = true strain
- $\mu$  = coefficient of friction
- $\sigma_\epsilon$  = true tensile stress

The rake angle is defined in a plane passing through the direction of the cutting speed vector and the direction of the normal to the plane machined.

Table A <sub>1</sub>	rake angle + 6° clearance angle 5° side cutting edge angle 15° nose radius 1,2 mm depth of cut 3 mm
Table A <sub>2</sub>	results of calculations referring to observations A <sub>1</sub>
Table B <sub>1</sub>	rake angle + 6° clearance angle 5° side cutting edge angle 0° nose radius 1,2 mm depth of cut 3 mm
Table B <sub>2</sub>	results of calculations referring to observations B <sub>1</sub>

0	Table C <sub>1</sub>	rake angle	- 6°
		clearance angle	17°
		side cutting edge angle	0°
5		nose radius	1,2 mm
		depth of cut	3 mm

Table C<sub>2</sub> results of calculations referring to observations C<sub>1</sub>

The machine tool used is a lathe, type A.I.DR 200-special  
input power 60/80 kW.

Number of revs. 0 ÷ 5000, continuous control  
range of feeds 0.0025 ÷ 40 mm/rev., continuous  
control, max. cutting force 10<sup>4</sup>N (1000 kgf).

The measurements have been performed by the metal cutting  
research team under the direction of Chr. Bus, ing.

Detailed information is given in the laboratory report  
WT 138 by A.G. Strous and H. Munnecom.

	0,10		0,13		0,16		0,20		0,25		0,32		0,40		0,50		0,63		0,79	
1,00	125	410	130	346	137	287	150	260	182	240	215	231	247	212	287	200	340	186	402	169
	77	830	80	743	85	711	90	655	97	685	104	660	109	640	110	630	113	630	115	625
1,12	122	400	127	330	133	281	155	255	177	232	210	217	240	206	282	192	340	178	407	169
	75	825	78	749	82	694	90	690	95	670	98	665	103	630	104	631	110	640	115	638
1,26	110	370	125	315	128	263	148	240	175	224	202	212	245	200	287	192	335	177	412	164
	70	780	80	748	78	694	83	680	88	685	92	645	100	662	101	652	102	640	110	646
1,41	113	370	120	292	132	256	148	235	172	220	212	209	237	195	275	186	335	175	400	169
	75	775	75	755	78	730	83	690	85	685	92	695	92	656	92	636	97	650	102	639
1,58	112	350	121	292	125	250	144	225	170	212	205	200	232	188	277	176	330	163	410	155
	70	825	73	768	74	696	78	685	82	690	86	680	88	647	89	661	91	655	100	673
1,78	110	320	115	269	123	231	140	205	157	200	190	187	222	185	277	177	327	165	398	154
	65	860	65	774	68	721	70	705	70	660	73	670	75	634	82	673	85	655	90	664
2,00	95	300	102	246	115	231	135	215	157	200	190	187	230	177	273	164	325	160	405	153
	58	770	60	706	63	678	67	675	70	660	72	670	79	671	78	676	82	660	90	678
2,24	90	250	105	231	120	213	140	200	163	192	190	187	230	173	275	170	330	159	405	155
	55	790	60	749	65	727	66	720	68	705	70	670	77	676	80	673	83	670	88	678
2,51	90	270	105	230	118	213	135	195	160	184	200	178	225	175	265	172	337	163	400	158
	55	770	60	749	60	732	63	705	65	705	73	715	70	672	73	658	80	690	85	673
2,82	90	240	105	216	120	200	140	185	162	180	192	172	220	170	277	166	337	159	400	158
	53	810	59	768	60	758	63	750	64	725	69	720	69	661	75	696	82	690	87	669
3,16	85	230	100	215	112	188	132	175	160	184	190	163	225	160	271	160	325	160	400	157
	51	780	55	738	57	710	59	710	65	705	68	695	68	690	71	691	75	670	85	674
3,55	83	240	99	208	115	194	135	180	155	165	194	159	225	153	263	148	323	150		
	47	760	52	752	54	750	55	745	55	730	67	720	65	700	69	673	75	670		
3,98	85	220	100	208	117	187	140	180	160	160	194	159	220	153	267	158	330	148		
	50	790	53	754	55	760	57	775	60	740	65	725	67	676	72	674	78	690		
4,47	83	200	95	185	112	181	130	170	155	172	192	157	230	160	275	154			R <sup>1</sup>	λ <sup>1</sup>
	48	795	49	741	50	753	53	730	55	725	68	710	71	704	77	690			R <sup>3</sup>	Z <sub>s</sub> <sup>4</sup>
5,01	85	210	100	200	122	194	140	195	157	184	190	175	240	168						
	51	795	52	767	57	801	64	740	64	695	66	695	76	721						A <sub>1</sub>

rap-  
port  
0139

blz.  
49

m/sec	0.10		0.13		0.16		0.20		0.25		0.32		0.40		0.50		0.63		0.79	
	1	2	1	2	1	2	1	2	1	2	1	2	1	2	1	2	1	2	1	2
100	4.14	0.77	3.54	0.77	3.02	0.78	2.78	0.75	2.61	0.68	2.54	0.62	2.39	0.57	2.30	0.51	2.19	0.45	2.08	0.40
	3.40	3.10	2.80	2.66	2.34	2.29	2.10	2.12	1.96	1.97	1.89	1.90	1.75	1.83	1.68	1.75	1.58	1.69	1.49	1.65
	14.10	31.40	16.30	31.40	19.50	31.50	21.50	31.00	23.30	28.05	24.20	25.50	26.20	23.50	27.50	21.00	29.30	18.20	32.10	16.00
112	4.05	0.77	3.40	0.77	2.96	0.77	2.74	0.73	2.55	0.68	2.43	0.60	2.34	0.56	2.24	0.49	2.14	0.44	2.08	0.40
	3.30	3.10	2.67	2.55	2.27	2.25	2.07	2.08	1.90	1.94	1.80	1.84	1.71	1.80	1.63	1.75	1.54	1.69	1.49	1.65
	14.20	31.40	17.20	31.35	20.15	31.40	22.10	30.10	24.10	28.15	25.50	25.00	27.00	23.20	28.50	20.20	30.45	18.00	32.10	15.45
126	3.77	0.80	3.27	0.80	2.81	0.76	2.62	0.71	2.59	0.64	2.39	0.59	2.30	0.53	2.24	0.47	2.13	0.42	2.05	0.40
	3.03	2.90	2.55	2.50	2.13	2.17	1.96	2.02	1.93	1.97	1.78	1.84	1.68	1.77	1.63	1.75	1.53	1.66	1.46	1.62
	15.30	32.30	18.10	32.40	21.30	31.25	23.30	29.20	25.10	26.40	26.20	24.30	27.40	22.10	28.50	19.25	30.55	17.00	33.00	16.00
141	3.77	0.91	3.06	0.78	2.75	0.74	2.58	0.71	2.45	0.63	2.36	0.57	2.26	0.51	2.20	0.46	2.12	0.41	2.08	0.37
	3.03	2.90	2.36	2.37	2.07	2.13	1.92	2.02	1.81	1.91	1.73	1.82	1.64	1.77	1.59	1.73	1.52	1.66	1.49	1.65
	15.30	36.10	19.30	32.00	22.05	30.35	24.00	29.20	25.30	26.20	26.40	23.30	28.20	21.10	29.40	18.30	31.10	16.10	32.10	14.20
158	3.58	0.78	3.06	0.76	2.70	0.74	2.49	0.69	2.39	0.62	2.30	0.55	2.24	0.51	2.13	0.44	2.04	0.39	2.00	0.36
	2.83	2.78	2.36	2.37	2.03	2.13	1.84	1.95	1.76	1.88	1.68	1.79	1.60	1.75	1.53	1.68	1.45	1.64	1.42	1.60
	16.20	32.00	19.30	31.10	22.40	30.40	25.00	28.30	26.20	25.45	27.40	22.50	29.20	20.50	31.00	17.50	33.10	15.30	34.40	13.40
178	3.31	0.74	2.86	0.71	2.54	0.70	2.33	0.64	2.30	0.58	2.20	0.51	2.19	0.46	2.13	0.41	2.06	0.38	1.99	0.34
	2.60	2.62	2.18	2.28	1.90	2.03	1.71	1.89	1.68	1.82	1.59	1.76	1.58	1.72	1.53	1.68	1.47	1.64	1.41	1.60
	17.50	30.40	21.05	19.30	24.20	29.00	27.10	26.30	27.45	24.00	29.20	21.00	29.45	18.40	30.50	16.30	32.50	14.40	34.45	12.50
200	3.13	0.77	2.67	0.74	2.54	0.69	2.41	0.63	2.30	0.58	2.20	0.51	2.13	0.47	2.05	0.40	2.02	0.37	1.98	0.33
	2.43	2.52	2.00	2.17	1.90	2.03	1.77	1.92	1.68	1.82	1.59	1.76	1.53	1.72	1.46	1.65	1.43	1.64	1.40	1.60
	19.00	31.30	23.00	30.30	24.20	28.40	26.40	26.20	27.45	24.00	29.20	20.50	30.55	19.00	33.00	16.00	33.40	14.10	35.00	12.30
224	2.71	0.77	2.54	0.72	2.40	0.69	2.30	0.61	2.24	0.55	2.20	0.49	2.14	0.47	2.09	0.41	2.02	0.37	2.00	0.33
	2.05	2.31	1.90	2.10	1.76	1.97	1.68	1.86	1.62	1.82	1.59	1.76	1.51	1.70	1.50	1.68	1.43	1.64	1.42	1.60
	23.05	31.30	24.20	29.45	26.10	28.30	27.45	25.15	28.50	22.40	29.20	20.20	31.30	19.00	32.00	16.75	33.50	14.10	34.40	12.20
251	2.87	0.77	2.54	0.72	2.40	0.65	2.26	0.60	2.18	0.53	2.14	0.49	2.12	0.43	2.10	0.39	2.00	0.35	2.01	0.32
	2.18	2.39	1.90	2.10	1.76	1.97	1.65	1.86	1.58	1.80	1.54	1.73	1.52	1.70	1.50	1.68	1.42	1.62	1.43	1.60
	21.00	31.30	24.30	29.45	26.10	27.00	28.20	25.00	29.55	22.10	30.40	20.05	31.10	17.20	31.20	15.20	33.20	13.20	34.00	12.00
282	2.62	0.74	2.42	0.71	2.30	0.64	2.18	0.58	2.15	0.52	2.10	0.48	2.09	0.43	2.06	0.39	2.02	0.36	2.01	0.33
	1.96	2.23	1.79	2.03	1.68	1.94	1.58	1.84	1.55	1.77	1.50	1.74	1.50	1.70	1.46	1.65	1.43	1.64	1.43	1.60
	24.05	30.40	25.50	29.20	27.45	26.35	29.45	24.15	30.30	21.35	31.40	19.50	32.00	17.25	32.40	15.10	33.50	13.40	34.00	12.20
316	2.53	0.75	2.41	0.70	2.24	0.65	2.12	0.58	2.18	0.53	2.04	0.48	2.02	0.42	2.02	0.38	2.03	0.34	2.01	0.32
	1.88	2.20	1.79	2.03	1.60	1.88	1.52	1.78	1.58	1.80	1.45	1.69	1.43	1.67	1.43	1.65	1.44	1.64	1.43	1.60
	24.25	31.00	26.00	28.50	29.20	27.00	31.10	24.10	29.55	22.10	33.10	19.40	33.40	16.50	33.40	14.40	33.40	13.00	34.10	12.00
355	2.62	0.72	2.36	0.67	2.25	0.60	2.15	0.53	2.05	0.48	2.02	0.46	1.98	0.41	1.95	0.38	1.97	0.35		
	1.96	2.23	1.74	2.00	1.68	1.94	1.55	1.84	1.46	1.72	1.44	1.69	1.40	1.65	1.37	1.61	1.39	1.62		
	24.05	29.35	26.50	27.40	28.30	25.10	30.30	22.10	33.00	19.30	33.50	19.00	35.00	16.10	36.00	14.40	35.30	13.10		
398	2.45	0.74	2.36	0.67	2.24	0.60	2.15	0.53	2.02	0.50	2.02	0.46	1.98	0.42	2.04	0.39	1.96	0.35		
	1.84	2.16	1.74	2.00	1.60	1.88	1.55	1.81	1.43	1.72	1.44	1.69	1.40	1.65	1.43	1.63	1.38	1.62		
	25.30	30.40	26.50	28.00	29.30	25.10	30.30	22.10	33.40	20.30	33.50	18.30	35.00	17.00	34.00	15.10	35.55	13.20		
447	2.30	0.73	2.19	0.66	2.16	0.58	2.09	0.53	2.10	0.48	2.04	0.48	2.02	0.43	1.99	0.40			tang	u
	1.68	2.06	1.58	1.94	1.56	1.85	1.50	1.78	1.50	1.74	1.43	1.66	1.43	1.67	1.40	1.63			E	u
	27.40	30.00	29.45	27.20	30.20	24.00	32.00	22.10	31.40	19.30	34.10	19.30	33.40	17.10	34.50	15.40			φ	β-s
501	2.37	0.75	2.30	0.66	2.25	0.59	2.26	0.59	2.18	0.53	2.12	0.47	2.08	0.44						A <sub>2</sub>
	1.74	2.10	1.68	1.97	1.68	1.91	1.65	1.86	1.58	1.80	1.52	1.71	1.49	1.70						
	26.30	31.00	27.45	27.30	28.30	24.40	28.20	24.35	29.55	22.10	31.10	19.10	32.20	17.40						

blz. 45

	0,10		0,13		0,16		0,20		0,25		0,32		0,40		0,50		0,63		0,79	
100	110	380	117	316	135	282	153	265	180	248	214	225	248	210	295	198	350	184	420	171
	68	770	75	699	83	707	90	667	95	667	101	665	103	657	108	657	110	655	115	658
112	110	380	120	308	135	288	153	260	180	240	214	222	245	208	288	192	343	179	412	171
	70	764	75	734	84	688	89	674	95	676	99	671	100	651	100	655	102	694	105	657
126	108	350	111	285	131	275	150	255	173	228	205	216	241	205	283	186	340	176	405	165
	71	786	74	687	79	697	85	674	87	673	90	662	93	655	94	658	98	657	100	655
141	115	380	115	300	130	269	150	245	177	224	203	209	270	235	280	188	333	175	398	162
	71	802	71	717	75	704	82	693	88	693	85	673	125	663	90	654	90	652	94	643
158	113	340	115	270	132	250	150	230	180	212	205	200	242	195	285	176	337	171	420	165
	75	836	73	748	80	729	83	706	90	717	89	679	90	675	92	676	94	659	107	672
178	123	400	134	316	130	250	155	240	177	208	203	194	247	187	280	174	335	159	410	163
	80	825	80	817	76	727	85	718	88	710	85	688	101	679	90	667	94	667	103	660
200	111	380	127	293	125	250	146	230	175	200	203	203	250	182	280	172	335	157	407	162
	74	765	78	802	72	703	78	697	83	719	83	681	99	692	88	672	91	675	100	661
224	100	310	124	262	123	231	145	225	170	196	200	194	250	180	280	170	330	154	407	162
	59	803	76	825	69	692	75	701	78	711	79	687	97	702	85	680	90	663	100	661
251	95	300	115	238	122	231	145	225	167	188	202	200	250	177	278	168	350	163	412	158
	60	764	71	790	67	688	72	710	77	708	75	695	95	709	84	675	95	698	95	682
282	90	290	106	208	120	231	145	220	168	188	199	197	250	175	277	168	345	159	415	158
	56	739	65	754	66	675	70	728	76	715	75	687	94	712	81	683	95	690	96	686
316	91	290	106	193	120	212	145	225	168	184	197	191	238	180	285	166	348	159		
	55	755	60	790	67	715	71	722	73	728	73	686	83	688	85	698	95	694		
355	90	250	105	208	120	219	137	195	167	184	192	178	238	175	282	166	348	162		
	53	800	62	761	66	715	57	746	73	733	59	717	81	697	85	686	93	697		
398	90	230	106	208	119	206	135	190	167	188	190	191	240	175	285	168				
	52	834	60	783	65	730	57	733	72	721	60	702	83	699	85	698				
447	92	250	105	216	120	206	135	190	165	184	189	191	225	175					$P_v^c \lambda^2$	
	53	816	59	770	65	735	57	733	72	716	63	686	70	676					$P_A^3 \tau_s^2$	
501	92	250	105	208	115	206	135	195	170	184	195	187	225	175						
	55	807	58	781	60	705	58	735	81	713	68	645	72	673						<b>B 1</b>

	0.10		0.13		0.16		0.20		0.25		0.32		0.40		0.50		0.63		0.79	
100	386 311 1510	078 <sup>2</sup> 295 3450	327 256 4810	080 250 3240	297 230 2010	077 209 3140	282 216 2120	074 212 3030	268 203 2250	067 201 2750	249 185 2500	061 190 2520	237 174 2630	054 183 2230	228 166 2800	049 175 2010	218 156 2950	043 171 1730	209 148 3150	039 165 1520
112	386 311 1510	0.80 295 3230	320 250 1830	078 245 320	303 235 1950	078 214 3150	278 212 2150	073 212 3010	262 197 2330	067 197 2750	247 183 2520	060 187 2450	236 173 2650	054 180 2210	224 162 2850	047 175 1910	215 154 3030	041 168 1630	209 148 3150	037 165 1420
126	358 285 1620	082 277 3330	300 232 2000	083 236 3340	291 224 2040	076 189 3100	274 208 2210	071 209 2930	252 188 2440	064 194 2640	242 179 2550	057 185 2340	234 171 2710	051 180 2410	219 159 2930	046 172 1830	213 152 3100	041 168 1610	206 146 3250	036 163 1350
141	386 311 1510	077 295 3140	313 244 1900	077 241 3140	286 219 2110	073 206 3000	266 201 2300	069 206 2870	248 184 2500	064 192 2630	237 174 2640	055 185 2240	257 193 2400	060 192 2450	221 160 2920	044 172 1750	212 150 3110	039 166 1510	204 144 3320	035 163 1320
158	349 277 1650	0.83 277 3340	287 220 2100	079 237 3220	270 204 2240	076 196 3110	253 197 2420	070 199 2900	239 176 2620	064 188 2640	230 168 2750	057 180 2330	226 164 2820	050 177 2020	213 152 3100	045 170 1800	209 148 3150	040 166 1520	206 146 3250	037 163 1420
178	405 328 1420	081 309 3300	327 256 1805	075 250 3050	270 204 2240	073 196 3020	261 189 2330	069 206 2850	236 173 2650	063 185 2630	225 163 2830	055 180 2250	220 159 2920	054 175 2220	211 167 3120	044 167 1750	202 142 3350	040 163 1520	204 144 3310	037 163 1410
200	386 311 1505	0.83 295 3340	386 238 1930	083 241 3135	270 204 2240	072 196 2950	253 189 2420	068 199 2810	230 168 2750	060 184 2530	232 170 2720	054 182 2220	217 156 3010	052 171 2120	210 149 3140	044 167 1730	200 140 3410	039 161 1510	204 144 3320	036 163 1350
224	322 252 1825	0.74 256 3030	280 214 2140	077 224 3130	253 189 2430	071 189 2920	249 185 2500	066 199 2730	228 166 2820	059 184 2440	225 163 2830	052 180 2130	216 155 3030	051 171 2410	209 148 3200	042 167 1700	199 138 3450	039 161 1520	204 144 3320	036 163 1350
251	313 244 1900	0.79 251 3220	260 195 2340	077 217 3140	253 189 2430	070 189 2850	244 185 2500	062 199 2540	220 159 2940	058 180 2440	230 168 2750	049 180 2020	214 153 3050	051 171 2050	207 146 3210	042 167 1700	204 144 3310	039 163 1510	201 140 3400	034 160 1300
282	304 235 1940	0.78 243 3155	236 173 2650	077 200 3135	253 189 2430	070 189 2850	244 180 2500	062 195 2540	220 159 2940	058 180 2440	228 166 2810	050 180 2040	212 151 3110	050 169 2040	207 146 3210	041 167 1620	202 142 3350	039 163 1520	201 140 34	034 160 13
316	304 235 1940	0.76 243 3110	225 163 2840	071 197 2930	237 174 2630	071 183 2920	249 185 2500	063 199 2610	218 156 2950	056 178 2520	223 161 2850	049 176 2020	216 155 3030	047 171 1920	206 145 3240	042 165 1640	202 142 3350	039 163 1520		
355	271 205 2305	0.74 230 3030	236 173 2650	074 200 3030	245 181 2530	070 187 2850	226 164 2810	054 187 2220	218 156 2950	057 178 2330	214 153 3050	043 174 1710	212 151 3110	046 169 1850	206 145 3240	042 165 1700	203 143 3320	038 163 1500		
398	253 188 2425	0.73 219 3000	236 173 2650	071 200 2930	234 171 2700	070 180 2840	222 160 2900	055 183 2250	221 160 2920	056 180 2320	223 161 2850	044 180 1730	212 151 3110	047 169 1920	207 146 3210	042 167 1640				
447	271 205 2305	0.72 230 2955	242 178 2550	071 210 2920	234 171 2700	068 180 2820	222 160 2900	055 183 2250	218 156 2950	056 178 2320	223 161 2850	045 180 1830	212 151 3110	043 169 1720						
501	271 205 2305	0.75 230 3050	236 173 2650	070 200 2900	234 171 2700	066 180 2730	226 164 2820	056 187 2320	218 156 2920	060 178 2530	220 158 2930	047 176 1920	212 151 3110	044 169 1740						

6000  
 E  
 4  
 5  
 B-5  
 B2



	0,10	0,13	0,16	0,20	0,25	0,32	0,40	0,50	0,63	0,79									
1,00	135 <sup>(2)</sup> 4,40	152 3,54	158 3,32	170 2,80	206 2,60	239 2,38	277 2,15	332 2,02	410 1,84	-	-								
	110 <sup>(3)</sup> 77,9	118 78,3	120 69,0	125 65,4	138 67,5	150 64,5	159 64,0	178 64,0	202 66,0	-	-								
1,12	132 4,20	140 3,39	150 3,15	165 2,70	195 2,48	233 2,28	272 2,08	328 1,88	409 1,76	-	-								
	105 78,4	110 73,5	115 66,8	119 64,3	131 65,0	145 64,0	154 64,0	172 65,0	198 66,6	-	-								
1,26	125 3,90	130 3,23	143 2,94	164 2,55	191 2,36	228 2,19	269 2,03	325 1,86	405 1,73	-	-								
	98 79,0	104 69,8	105 67,2	115 66,2	126 65,0	138 64,5	147 64,0	165 65,0	190 66,8	-	-								
1,41	115 3,60	123 3,00	137 2,75	162 2,45	190 2,28	224 2,09	266 1,95	326 1,80	400 1,62	-	-								
	90 76,1	96 69,0	102 66,3	112 67,0	122 65,5	132 64,5	142 64,5	159 66,5	210 64,0	-	-								
1,58	106 3,30	120 2,84	135 2,56	161 2,35	187 2,20	220 2,03	263 1,85	315 1,74	400 1,58	-	-								
	84 73,3	95 68,6	98 67,0	105 68,8	117 66,5	126 64,5	137 65,5	150 65,3	209 65,5	-	-								
1,78	101 3,10	118 2,69	132 2,50	154 2,25	182 2,12	218 1,97	261 1,77	305 1,68	392 1,53	-	-								
	79 72,0	90 69,7	92 67,4	100 67,0	104 67,8	120 65,5	128 66,5	145 63,7	204 63,0	-	-								
2,00	116 3,30	115 2,62	135 2,44	155 2,20	179 2,08	215 1,94	257 1,75	305 1,66	392 1,53	-	-								
	88 81,5	82 70,5	90 70,5	99 71,0	102 66,9	113 65,6	125 66,0	141 64,5	203 63,0	-	-								
2,24	106 3,10	115 2,54	130 2,31	154 2,15	175 2,04	210 1,91	255 1,75	303 1,66	380 1,54	-	-								
	81 77,0	81 71,8	88 69,0	95 70,0	99 65,8	109 64,8	122 66,0	140 65,6	160 66,5	-	-								
2,51	105 2,90	110 2,46	129 2,25	153 2,10	174 1,96	210 1,87	254 1,70	310 1,64	380 1,56	-	-								
	79 79,0	76 70,0	85 70,0	92 70,0	96 64,8	106 65,8	120 66,5	139 66,5	162 66,0	-	-								
2,82	92 2,70	108 2,31	127 2,19	150 2,05	173 1,92	210 1,84	253 1,70	306 1,62	390 1,57	-	-								
	67 72,0	73 70,6	81 70,0	89 69,5	94 66,8	105 66,2	119 66,5	137 65,6	165 67,5	-	-								
3,16	90 2,70	106 2,23	126 2,13	147 2,00	174 1,88	210 1,84	253 1,65	296 1,64	-	-									
	64 71,0	70 70,6	80 70,0	87 69,0	94 67,8	103 66,6	116 67,5	126 64,2	-	-									
3,55	89 2,60	106 2,15	125 2,06	149 1,95	172 1,88	209 1,84	256 1,65	310 1,66											
	62 71,5	69 71,8	78 70,5	86 70,5	88 68,1	99 67,5	118 68,0	131 67,4											
3,98	89 2,60	106 2,15	124 2,06	143 1,95	170 1,84	210 1,82	260 1,68	-	-										
	60 73,0	68 72,1	77 70,0	76 69,5	87 67,6	103 67,0	120 68,5	-	-										
4,47	89 2,50	108 2,33	125 2,06	142 1,85	170 1,84	215 1,78	-	-											$P_1^{(2)}$ $\lambda^{(2)}$
	61 73,0	69 72,0	78 70,5	76 70,0	88 67,5	106 68,5	-	-											$P_1^{(3)}$ $Z_5^{(3)}$
5,01	86 2,50	110 2,33	126 2,06	150 2,00	-	-	-	-											$C_1$
	59 71,0	71 73,0	78 71,0	90 69,0	-	-	-	-											

Report  
0139  
-  
blz.  
48

$\frac{V}{m/sec.}$	0,10	0,13	0,16	0,20	0,25	0,32	0,40	0,50	0,63	0,79											
1,00	4,83 4,08 12°30'	1,01 3,46 39°28'	4,03 3,27 15°20'	0,96 2,80 37°50'	3,83 3,08 16°20'	0,94 2,62 37°10'	3,37 2,64 19°	0,91 2,33 36°10'	3,19 2,48 20°20'	0,83 2,20 33°50'	3,01 2,30 22°	0,79 2,08 32°10'	2,83 2,15 24°	0,72 1,96 29°50'	2,72 2,05 25°10'	0,68 1,90 28°10'	2,59 1,93 27°15'	0,63 1,83 26°15'	—	—	
1,12	4,64 3,88 13°05'	0,99 3,24 38°40'	3,89 3,13 16°	0,97 2,72 38°10'	3,68 2,94 17°10'	0,95 2,54 37°40'	3,28 2,56 19°40'	0,89 2,28 35°40'	3,09 2,39 21°10'	0,84 2,15 34°	2,93 2,24 22°50'	0,78 2,04 31°50'	2,77 2,09 24°40'	0,71 1,95 29°30'	2,62 1,96 26°50'	0,67 1,85 27°40'	2,54 1,89 28°15'	0,62 1,81 25°50'	—	—	
1,26	4,36 3,60 14°	0,97 3,08 38°	3,75 3,00 16°40'	0,99 2,65 38°40'	3,49 2,76 18°10'	0,91 2,44 36°10'	3,15 2,45 20°40'	0,87 2,21 35°	2,99 2,30 22°10'	0,82 2,10 33°50'	2,86 2,17 23°30'	0,76 2,00 31°10'	2,73 2,06 25°10'	0,69 1,93 28°40'	2,61 1,95 27°	0,65 1,85 27°	2,52 1,87 28°40'	0,60 1,80 25°10'	—	—	
1,41	4,09 3,35 15°10'	0,97 2,94 38°	3,54 2,80 17°50'	0,96 2,52 38°	3,32 2,60 19°20'	0,92 2,35 36°30'	3,07 2,36 21°25'	0,86 2,17 34°40'	2,85 2,17 23°05'	0,80 2,04 32°40'	2,78 2,10 24°30'	0,74 1,97 30°30'	2,67 2,01 26°	0,68 1,90 28°10'	2,57 1,92 27°45'	0,63 1,83 26°	2,45 1,81 30°10'	0,67 1,77 27°40'	—	—	
1,58	3,71 2,96 16°20'	0,98 2,72 38°20'	3,40 2,67 18°50'	0,98 2,46 38°20'	3,16 2,44 20°40'	0,91 2,26 36°10'	2,99 2,29 22°10'	0,82 2,14 33°	2,87 2,18 23°30'	0,78 2,04 32°	2,73 2,05 25°10'	0,72 1,94 29°50'	2,60 1,94 27°10'	0,66 1,87 27°30'	2,53 1,88 28°50'	0,61 1,82 25°30'	2,43 1,79 30°50'	0,66 1,76 27°30'	—	—	
1,78	3,63 2,88 17°20'	0,97 2,67 38°	3,27 2,55 19°40'	0,94 2,38 37°20'	3,11 2,40 21°	0,87 2,24 35°	2,90 2,21 23°	0,81 2,08 33°	2,80 2,12 24°10'	0,71 2,01 29°20'	2,69 2,02 25°50'	0,70 1,92 28°50'	2,55 1,90 28°10'	0,63 1,84 26°10'	2,49 1,85 29°20'	0,61 1,80 25°30'	2,40 1,76 31°30'	0,66 1,75 27°30'	—	—	
2,00	3,81 3,06 16°20'	0,94 2,83 37°10'	3,21 2,50 20°10'	0,88 2,36 35°30'	3,06 2,35 21°30'	0,83 2,21 33°50'	2,87 2,18 23°50'	0,73 2,12 30°10'	2,77 2,10 24°35'	0,72 2,00 29°40'	2,67 2,00 26°10'	0,67 1,91 27°45'	2,53 1,88 28°20'	0,62 1,84 25°50'	2,48 1,83 29°40'	0,60 1,79 24°50'	2,40 1,76 31°30'	0,66 1,75 27°30'	—	—	
2,24	3,63 2,88 17°20'	0,94 2,67 37°10'	3,14 2,42 20°40'	0,87 2,32 35°10'	2,95 2,26 22°30'	0,84 2,16 34°10'	2,83 2,15 24°	0,76 2,06 31°10'	2,74 2,06 25°	0,72 1,98 29°50'	2,65 1,98 26°30'	0,66 1,90 27°20'	2,53 1,88 28°20'	0,62 1,84 25°40'	2,48 1,83 29°40'	0,59 1,79 24°30'	2,40 1,76 31°20'	0,66 1,75 27°20'	—	—	
2,51	3,45 2,71 18°20'	0,93 2,58 36°50'	3,08 2,38 21°20'	0,86 2,29 34°40'	2,90 2,21 23°	0,82 2,14 33°50'	2,79 2,11 24°30'	0,75 2,03 31°	2,68 2,02 25°50'	0,70 1,95 28°50'	2,62 1,96 26°50'	0,65 1,89 26°50'	2,50 1,85 29°	0,62 1,82 25°30'	2,46 1,82 29°50'	0,58 1,78 24°10'	2,41 1,77 31°	0,66 1,76 23°	—	—	
2,82	3,28 2,56 19°40'	0,91 2,47 36°10'	2,95 2,25 22°30'	0,84 2,22 34°	2,86 2,18 23°30'	0,80 2,10 32°40'	2,75 2,07 25°	0,74 2,01 30°30'	2,65 1,99 26°20'	0,69 1,93 28°30'	2,60 1,94 27°15'	0,64 1,88 26°40'	2,50 1,85 29°	0,60 1,82 25°10'	2,45 1,81 30°10'	0,58 1,78 24°10'	2,42 1,78 30°50'	0,66 1,76 23°	—	—	
3,16	3,28 2,56 19°40'	0,91 2,47 35°20'	2,89 2,20 23°10'	0,82 2,18 33°30'	2,81 2,13 24°10'	0,79 2,09 32°30'	2,71 2,04 25°30'	0,74 2,00 30°30'	2,63 1,97 26°50'	0,68 1,93 28°20'	2,60 1,94 27°15'	0,63 1,88 26°10'	2,47 1,83 29°40'	0,59 1,81 24°40'	2,46 1,82 29°50'	0,56 1,78 23°10'	—	—	—	—	
3,55	3,19 2,48 20°20'	0,87 2,42 35°	2,83 2,16 24°	0,81 2,15 33°	2,76 2,08 24°50'	0,77 2,06 31°50'	2,67 2,00 26°	0,71 1,98 29°50'	2,63 1,97 26°50'	0,66 1,93 27°10'	2,60 1,94 27°15'	0,61 1,88 25°30'	2,47 1,83 29°40'	0,59 1,81 24°40'	2,48 1,83 29°40'	0,55 1,79 23°	—	—	—	—	
3,98	3,19 2,48 20°20'	0,83 2,42 35°50'	2,83 2,16 24°	0,80 2,15 32°40'	2,76 2,08 24°50'	0,77 2,06 31°50'	2,67 2,00 26°	0,67 1,98 28°	2,60 1,94 27°20'	0,66 1,91 27°10'	2,58 1,92 27°40'	0,63 1,87 26°10'	2,49 1,85 29°20'	0,59 1,81 24°40'	—	—	—	—	—	—	
4,47	3,11 2,41 21°	0,85 2,38 34°20'	2,97 2,27 22°20'	0,80 2,23 32°50'	2,76 2,08 24°50'	0,77 2,06 31°50'	2,60 1,94 27°10'	0,68 1,94 28°10'	2,60 1,94 27°20'	0,66 1,91 27°20'	2,55 1,90 28°	0,63 1,86 26°20'	—	—	—	—	—	—	—	—	
5,01	3,11 2,41 21°	0,86 2,38 34°40'	2,97 2,27 22°20'	0,81 2,23 32°50'	2,76 2,08 24°50'	0,77 2,06 31°50'	2,60 1,94 27°30'	0,75 2,01 31°	—	—	—	—	—	—	—	—	—	—	—	—	—

Report  
0139  
—  
blz.  
49

Long M  
E 0 0 0  
0 0 0-3

C 2