

Generic behaviour of grazing impact oscillators

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Generic Behavior of Grazing Impact Oscillators

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We give experimental evidence for a new bifurcation structure that arises when smooth dynamical systems cross a boundary. Our experiment concerns a driven impacting leaf-spring oscillator with a very precise control of the driving amplitude. The results are in surprisingly good agreement with the predictions of a simple nonlinear mapping that is valid near grazing impact (i.e., impact with zero velocity). The agreement is surprising because a multitude of vibration modes of the spring is excited upon impact whereas the mapping is two dimensional. These findings point to the universality of the observed bifurcation structure. [S0031-9007(96)00276-1]

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Impact oscillators have an oscillating mass that impacts with a fixed boundary when its amplitude is large enough. In between impacts the dynamics is smooth and often linear, but it derives a strong nonlinearity from the mere presence of the impact. These systems are known to exhibit a richness of bifurcation phenomena [1,2], but a strong guiding principle has, so far, been lacking. Impact oscillators are prototypical for nonlinear phenomena in engineering systems that are often designed with loose fittings. A timely application of impact oscillators is in atomic force microscopy [3].

A grazing impact is a boundary impact with zero velocity. In order to describe the events that occur when an orbit evolves to grazing impact and beyond, let us define ρ as the bifurcation parameter. When the driving amplitude is increased from zero, impacts first occur at $\rho = 0$. These impacts may be with a relatively large velocity and the transition that has taken place may be hysteretic; i.e., impacts may remain when ρ is subsequently smoothly decreased and may vanish only at a negative value of ρ . When the impacts are about to disappear, the orbit is closest to grazing. In a recent paper Nordmark [4] has reduced the dynamics of impact oscillators to a simple nonlinear mapping that is valid for orbits close to grazing ones. A striking phenomenon in this mapping is the emergence of a square-root singularity.

An extensive bifurcation analysis of this mapping has recently been reported by Chin *et al.* [5]. Among the found characteristic phenomena are an infinite series of period addings for the overdamped oscillator and an infinite series of period-1 (p_1) to period-*m* (p_m), $m =$ 3, 4, . . . transitions for the underdamped case. These phenomena are a direct consequence of the square-root singularity of the mapping and are independent of the details of the dynamics in between collisions. They are, therefore, predicted to be *universal*. In this Letter we present for the first time experimental evidence for one of these universal phenomena, namely, the infinite series of p_1 to p_m transitions.

The experiment is sketched in Fig. 1. The oscillator consists of a 13 cm long brass leaf spring that is mounted on a large electromagnetic exciter which oscillates horizontally. The beam is weakly damped by strips of damping material that are glued to it. It is U shaped to suppress torsional modes but the excitation of higher-order bending modes is unavoidable. Collisions occur between a ceramic ball that is attached to the end of the beam and a hardened steel plate that is attached to the exciter. The motion of the beam is registered by reflecting a laser beam off the beam onto a position sensitive photodiode. The measurement and control of the excitation amplitude is a crucial aspect of the experiment. The excitation amplitude (typically 2 mm) is measured using a laser interferometer and fringe counter and is digitally controlled with a long-term stability of 0.5 μ m [6].

FIG. 1. A U-shaped leaf spring is brought into oscillation by horizontally oscillating its support. At a large enough forcing amplitude *F*, the attached mass impacts with a stop. Collisions take place between a hard ceramic ball and a hardened steel plate. The amplitude of the exciter is measured interferometrically. The deflection of the leaf spring is registered by reflecting a laser beam off the spring onto a position sensitive diode.

The experiment is approximately described by the differential equation (overdots denote time derivatives)

$$
\ddot{\xi} + \mu \dot{\xi} + \Omega^2 \xi = F \sin(\omega t), \qquad (1)
$$

where Ω is the eigenfrequency of the free oscillator, μ represents the damping, and the excitation force has amplitude *F* and frequency ω . When the position $\xi(t)$ comes to the boundary at, say, t_0 the velocity reflects as $\dot{\xi}(t_{0^+}) = -\tau \dot{\xi}(t_{0^-}),$ where τ is the restitution coefficient. Equation (1) is approximate in that small oscillation amplitudes are assumed and, more importantly, higherorder modes of the beam are ignored. For near-grazing orbits in a broad class of second-order dynamical systems [one of which would be Eq. (1)] the following mapping can be derived [4] (see also [5]):

$$
x_{n+1} = \alpha x_n + y_n + \rho \qquad \text{for } x_n \le 0 \qquad (2)
$$

$$
y_{n+1} = -\gamma x_n
$$

$$
x_{n+1} = -\sqrt{x_n} + y_n + \rho \quad \text{for } x_n > 0,
$$
 (3)

where x_n and y_n are transformed coordinates of the $(\xi, \dot{\xi})$ space at stroboscopic times $t_n = n2\pi/\omega$, and where ρ is the bifurcation parameter that measures the distance to the point of grazing impact. If no collision occurs between t_n and t_{n+1} , the linear map Eq. (2) applies, whereas Eq. (3) describes the dynamics in the case that an impact will occur on $[t_n, t_{n+1}]$. For oscillators described by Eq. (1), the parameters of the mapping can be expressed explicitly in those of the differential equation, $\gamma = \exp(-\mu/f)$, $\alpha = 2\gamma^{1/2} \cos(2\pi/fT)$ [5], and $\rho = (F/F_g - 1)[1 - \alpha + \gamma]/2\gamma(fT)^2(1 +$ τ ² sin²(2 π /*fT*), where *f* = ω /2 π is the excitation frequency, $T = 2\pi/[\Omega^2 - (\mu/2)^2]^{1/2}$ is the period of the free damped oscillations of the beam, and F_g is the excitation amplitude at grazing impact. However, we emphasize again that Eqs. (2) and (3) apply to more general impacting oscillators than those described by Eq. (1). The presence of the square root in Eq. (3) is a key aspect of the mapping; it causes the Jacobian to be singular [7] at $x_n = 0$, and it gives rise to the characteristic *bordercollision bifurcations* [8] that are reported in this Letter. The nature of the square root is simply the relation between elapsed time and traveled distance in systems with constant acceleration but the precise derivation of the mapping is highly nontrivial.

Our experiment corresponds to the underdamped case Ω^2 > (μ /2)² and analysis of the mapping predicts an infinite series of $p_m(m=3,\ldots,\infty)$, saddle-node bifurcations at $\rho = \rho_m \leq 0$ as the parameter α is varied from 0 to $2\sqrt{\gamma}$ [9]. The stable period *m* orbit has a single impact per period and is called a *maximal* periodic orbit [5]. At $\rho = 0$ the unstable *m* saddle collides with the p_1 orbit. Because the p_m orbit is born away from the p_1 orbit, the transition $p_1 \leftrightarrow p_m$ *appears* to be hysteretic: Whereas the transition $p_1 \leftrightarrow p_m$ takes place at $\rho = 0$, the transition $p_m \leftrightarrow p_1$ in a smooth downward scan of ρ takes place at $\rho_m \leq 0$.

A measured series of transitions $p_1 \leftrightarrow p_m$ for $m =$ 3 to $m = 10$ is shown in Fig. 2. In our experiments the period-1 orbit is the trivial periodic motion of the noncolliding beam. The highest observed period (p_{10}) is at the limit of our experimental resolution and stability. The shown series of bifurcations accurately follows the prediction of the mapping.

The $p_1 \leftrightarrow p_m$ transitions occur in bands in the (α, γ) parameter plane of the mapping [5] that are crossed approximately transversely by scanning the excitation frequency in our experiment. At the upper boundary of the *m* region, the p_m saddle node is born exactly at $\rho = 0$ and there is no hysteresis. The hysteresis increases towards the upper boundary of the lower-lying $m - 1$ boundary. Figure 3 compares the predicted hysteresis $F_m/F_g - 1$ with that found experimentally, where F_m is the excitation amplitude at which the period-*m* orbit vanishes in a smooth downward scan of *F*. All experimental resonances are found in place, but the measured hysteresis differs from the prediction. We believe that the discrepancy is due to higher-order vibrational modes of the beam.

Excitation amplitude (mm)

FIG. 2. Experimental bifurcation diagrams of $p_1 \leftrightarrow p_m$ transitions with $m = 3, 4, 5, 6, 8, 10$ for (a)–(f), respectively. The displacement of the beam is measured at stroboscopic times $t_n = n2\pi/\omega$. The closed circles are for the upward scan of the excitation amplitude; the open circles are for the downward scan. The period of the free swinging beam is $T =$ 40.80 ± 0.02 ms. The excitation frequencies are $f = 20.90$, 21.50, 22.20, 22.60, 23.05, and 23.40 Hz, for (a)–(f), respectively. In all cases the observed periodicity agrees with the prediction from the Nordmark map [Eqs. (2) and (3)].

FIG. 3. Hysteresis of "maximal" *pm* orbits as a function of excitation frequency. Full lines: as computed from the Nordmark map Eqs. (2) and (3). Dotted lines: stability boundaries of the p_m orbits as predicted by the Nordmark map; the corresponding periods *m* are indicated. Open circles connected by dashed lines: experimental results. In a few cases p_m orbits continue to exist in the lower-lying p_{m-1} regions. This coexistence is actually predicted by the mapping Eqs. (2) and (3).

The excitation of higher-order modes upon impact is a characteristic feature of impacting in continuous systems. The energy in these modes is quickly dissipated, for example, by the radiation of sound. As a consequence, the collision is highly inelastic. This is accounted for by a nonunity restitution coefficient τ . The presence of a multitude of higher-order modes on impact is demonstrated in Fig. 4, which shows the measured deflection as a function of time for a period-4 orbit. It therefore appears that the dynamics in our experiment is much more complex than described by Eq. (1) and the very favorable agreement with the mapping becomes a surprise. We will show that it is, perhaps paradoxically, precisely one of the nonidealities of the experiment that makes the mapping more appropriate.

It is important to realize that the mapping is derived for orbits that have a collision that is near grazing and the predicted bifurcations occur when the orbit near a particular grazing collision point evolves from near grazing to nonimpacting and back again when a parameter is smoothly varied. The proximity of an orbit to one containing a grazing collision is, therefore, the key parameter that determines the applicability of the mapping. Obviously, a p_m orbit that results from a $p_1 \rightarrow$ p_m bifurcation with a large hysteresis will at $\rho = 0$ be far away from a grazing collision; the distance increases as the hysteresis becomes larger.

The magnitude of the hysteresis $|\rho_m|$ decreases with increasing energy loss per collision. Therefore, for impacting systems with a large energy loss at each impact, the mapping [Eqs. (2) and (3)] is applicable in larger areas of parameter space than for impacting systems with smaller energy loss. In our experiment the restitution coefficient is

FIG. 4. Time trace of a period-4 orbit at $\rho = F/F_g - 1 =$ 0.013. At impact a multitude of high-order modes is excited in the beam.

estimated to be $\tau \approx 0.1$ and the $p_1 \rightarrow p_m$ bifurcations could be found in a fairly large region of the predicted stability band in the (α, γ) plane. Outside this region, i.e., towards the upper stability boundary of the p_{m-1} orbit, other transitions than the predicted $p_1 \rightarrow p_m$ bifurcation occur; these may even be to chaotic states. In these cases the p_m stable periodic orbit is found only at some negative value of ρ and the $p_m \rightarrow p_1$ bifurcation is encountered when ρ is decreased further. These other transitions are caused by secondary bifurcations on $\rho_m < \rho < 0$ that are due to second impacts. In the mapping these secondary bifurcations always occur at $\rho > 0$. In Eq. (1) secondary bifurcations may occur at values $\rho < 0$ that become smaller if collisions become more elastic.

The breakdown of the mapping near $\rho = 0$ is illustrated in Fig. 5, which shows an experimental bifurcation diagram at parameter values where the mapping predicts the p_3 orbit to be stable. At $\rho = 0$ the trivial noncolliding p_1 orbit gives way to a period-6 orbit. The stable periodic *p*³ orbit is found only at the very end of the downward scan of the driving amplitude. Incidentally, Eq. (1) predicts for this case a chaotic state at $\rho = -0.0707$ that is not connected to the state that is born at $\rho = 0$. Clearly, the experimental result is also strongly influenced by the excitation of higher-order modes. The prediction of the mapping applies only when the orbit is closest to grazing.

The observed bifurcation phenomena are a consequence of the square-root singularity of the mapping. This singularity may be smoothed by the excitation of higherorder modes that introduce a time delay at impact. A crude model results if the impacts of the system described by Eq. (1) are imaged to occur with a soft wall such that at impact the spring constant changes by a factor $K \gg 1$ [10]. We have studied numerically the influence of a finite value of *K* on the observed bifurcation phenomena [11]. A large value of *K* affects the bifurcations close

FIG. 5. Experimental bifurcation diagram measured in the region where the mapping predicts the $p_1 \leftrightarrow p_3$ transition. For positive values of ρ finer excitation amplitude steps are taken. At each amplitude setting the system is allowed to equilibrate for $\approx 10^4$ cycles after which 40 stroboscopic points are drawn. The excitation frequency is $f = 19.65$ Hz, the measured period of the freely swinging beam is $T = 41.2 \pm 0.1$ ms, and the damping constant $\mu = 3.56 \pm 0.1 \text{ s}^{-1}$.

to $\rho = 0$ in a characteristic manner. The influence of the wall softness extends to larger values of ρ as K decreases. As we have not observed these secondary bifurcations in our experiment, we conclude that the excitation of higher order modes does affect the energy loss of a collision, but does not effectively smooth the singularity. The regularization of the singularity is an interesting problem and will be discussed in a forthcoming paper.

In conclusion, we have given experimental evidence for a new class of universal bifurcations that arises when smooth dynamical systems cross a boundary. These bifurcations are predicted by a mapping that reduces the dynamics to its essentials: namely, the occurrence of a singularity. Surprisingly, our evidence has been aided by the increased damping in the experiment that is a consequence of the excitation of higher-order modes. We expect that these phenomena should be observed in a wide class of experimental boundary crossing systems. The energy loss and time delay at impact are two parameters to effectively represent higher-order modes in a singlemode description. Obviously, when impacts are hard and these modes become an essential part of the dynamics, the analysis presented here no longer applies.

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- [10] A very similar model, but not for grazing collisions, was studied in [2].
- [11] An efficient numerical scheme for solving Eq. (1) in the presence of grazing impacts uses the analytical solutions for the motion between impacts. The (exact) positions $\xi(t)$ are computed in a small number of discrete points t_1, \ldots, t_k in each drive period. It is crucial not to miss boundary crossings of $\xi(t)$. These crossings are detected both directly by checking $\xi(t_1), \ldots, \xi(t_k)$ and by computing the position $\xi(t)$ at the turning points of the velocity $\dot{\xi}$.