

1 **Design, Dimensional Synthesis and**
2 **Evaluation of a Novel 2-DOF Spherical RCM**
3 **Mechanism for Minimally Invasive Surgery**

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26
27 **ABSTRACT**
28

29 *With the development of minimally invasive surgery (MIS) technology, higher requirements are put forward*
30 *for the performance of remote center of motion (RCM) manipulator. This paper presents the conceptual*
31 *design of a novel two degrees of freedom (2-DOF) spherical RCM mechanism, whose axes of all revolute joints*
32 *share the same RCM. Compared with the existing design, the proposed mechanism indicates a compact*
33 *design and high structure stability, and the same scissor-like linkage makes it easy to realize modular design.*
34 *It also has the advantages of singularity free and motion decoupling in its workspace, which simplifies the*

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35 *implementation and control of the manipulator. In addition, compared with the traditional spherical scissor*
36 *linkage mechanism, the proposed mechanism adds a rotation constraint on the output shaft to provide*
37 *better operating performance. In this paper, the kinematics and singularities of different cases are deduced*
38 *and compared, and the kinematic model of the best case is established. According to the workspace and*
39 *constraints in MIS, the optimal structural parameters of the mechanism are determined by dimensional*
40 *synthesis with the goal of optimal global operation performance. Furthermore, a prototype is assembled to*
41 *verify the performance of the proposed mechanism. The experimental results show that the 2-DOF prototype*
42 *can provide a reliable RCM point. The compact design makes the manipulator have potential application*
43 *prospects in MIS.*

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45
46

1 INTRODUCTION

47 The basic operation concept of MIS is to insert surgical instruments such as a
48 laparoscope into a patient's body through a small incision, so as to carry out surgical
49 operation inside the patient's body [1]. Robot-assisted MIS is widely used in clinical
50 surgery because of its great advantages over traditional open surgery [2]. Limited by the
51 constraints of the incision point on the patient, the motion of the surgical instruments is
52 limited to 1-DOF translation along the entry axis and 3-DOF rotations around the entry
53 point called *the remote center of motion (RCM)*. In recent decades, RCM mechanisms have
54 attracted extensive interests from researchers in the fields of mechanisms and robotics.

55 Compared with controlling redundant joints and adding passive joints, an RCM
56 mechanism constructed by mechanical constraint method has the characteristics of high
57 safety and simple control algorithms [3, 4]. According to the structural characteristics, this
58 type of mechanism can be divided into parallelogram RCM mechanism, arc RCM
59 mechanism, spherical RCM mechanism, and parallel RCM mechanism etc. Li et al. [5]

60 proposed a type synthesis method for constructing 2-DOF planar RCM mechanisms with
61 a virtual double parallelogram structure. Huang et al. [6] proposed the design of 2-DOF
62 planar RCM mechanisms based on closed-loop coupling-cable-driven strategy. Through
63 the analysis and determination of the transmission ratio, it provided a large workspace
64 but low collision risk for MIS robot. Kuo et al. [7] proposed a novel 4-DOF parallel RCM
65 robot, and the 4 DOFs are fully decoupled. However, due to the large space occupied by
66 the robot, it is almost impossible for multiple instruments to work together. Chen et al.
67 [8] proposed a spatial dual-arm parallel manipulator with 3R1T (where R denotes rotation,
68 and T stands for translation) motion capability, it provides better flexibility but has the
69 disadvantage of occupying too much space. In order to reduce the space occupied by the
70 robot and prevent the collision among the multi-robot system, Chen and his colleagues
71 [9, 10] developed a new RCM mechanism based on double-triangular linkage.

72 In some operations, such as the abdominal surgery, three to four RCM operating
73 arms are often required to work simultaneously. This means higher requirements for the
74 compactness of the robots, especially in the retracted state. Hence, researchers are
75 continuously looking for better mechanisms and solutions.

76 The axes of a spherical mechanism intersect at one point, and the links all move
77 on the concentric spherical surfaces, and the intersection of the axes is the remote center
78 of motion (RCM). Spherical mechanism has been applied to portable MIS robot [11], force
79 reflection robot MC²E [12] and the others. On the other hand, scissor-like mechanisms
80 have good expansibility, high volume expansion rate and are easy to fabricate and
81 assemble. Such mechanisms have wide applications in aerospace, novel buildings, and

82 other fields [13]. The spherical scissor-like linkage mechanisms combine the advantages
83 of the spherical and scissor-like mechanisms, and have found applications in the fields of
84 shoulder rehabilitation [14] and architecture design [15]. Kocabas [16] designed a
85 spherical gripper using a network of spherical parallelogram mechanisms, the gripper has
86 only 1-DOF for grasping many shapes. Castro et al. [17] applied the spherical scissor-like
87 linkages to the shoulder mechanism, and simplified the design of a spatial spherical
88 mechanism. In this work, all the joints are passive and the positioning accuracy of the
89 mechanism has not been verified. Afshar et al. [18] optimized the spherical scissor-like-
90 linkage RCM robot for the task of ultrasonic scanning. In which, the singularity occurs
91 when all the linkages collapse to the same plane in the limit position state, and the output
92 shaft of the robot rotates along with the end link, which is not conducive to the control
93 of the operating instruments.

94 According to the requirements of the MIS, a novel spherical RCM mechanism is
95 proposed in this paper. The proposed mechanism indicates a compact design and high
96 structure stability. In the retracted state, the compact structure effectively avoids the
97 collision with the surrounding mechanical arms. It also has the advantages of singularity
98 free and motion decoupling in its workspace. The same scissor-like linkage makes it easy
99 to realize modular design, which has potential to adapt to different angles of workspace
100 in different operations through changing the number of motion units. In addition, based
101 on the traditional spherical scissor-like linkage RCM mechanism, the rotation constraint
102 on the output shaft is added to provide a stable rotating base for the surgical instrument.

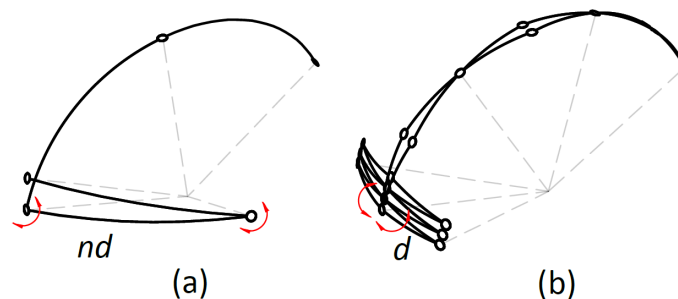
103 The rest of this paper is arranged as follows. In Sec. 2, the alternative cases of drive
 104 unit are selected through the kinematics and singularity analysis results. In Sec. 3, the
 105 characteristics of the proposed two output units are compared and analyzed. Then, in Sec.
 106 4, based on the analysis of the first two sections, the series structure composed of several
 107 identical spherical four-bar units combined with planar constraint branch is selected as
 108 the final solution, and the dimensional synthesis is carried out with the goal of optimal
 109 global operation performance. In Sec. 5, a physical prototype is design and developed
 110 according to the optimal structural parameters, and the performance is tested through
 111 experiments. Conclusions are addressed in Sec. 6

112

113 **2 Conceptual Design of the Drive Unit**

114

115 In MIS operation, the surgical instrument holder needs to carry the surgical
 116 instrument to adjust the pitch and yaw attitude angle. As shown in Fig. 1(a), the traditional
 117 spherical RCM mechanisms normally use 2-DOF rotations through two arc links in series
 118 to achieve the desired function. Large-scale pitching workspace will lead to the excessive
 119 size of the links, and the width of the whole mechanism will be very large in the retracted
 120 state.



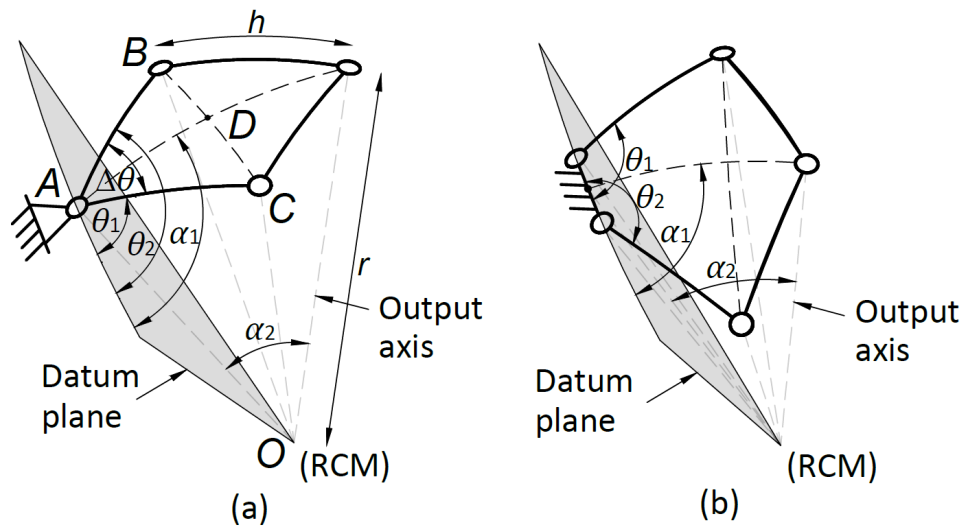
121

122 Fig. 1 Comparison of two kinds of spherical RCM mechanisms. (a) traditional spherical
 123 mechanism, and (b) the proposed spherical scissor-like linkage mechanism

124

125 As shown in Fig. 1(b), in order to ensure a small volume space ratio, n spherical
 126 scissor-like units are considered to be connected in series. Under the condition of the
 127 same workspace, compared with the traditional spherical mechanism, the size of the links
 128 in the scissors-like linkage mechanism will be reduced by n times, and the overall
 129 structure will be more compact in the retracted state. On the other hand, the hybrid
 130 structure allows the integration of the motors in the fixed platform.

131 In order to drive a series of scissor-like link units, two drive links are required to
 132 connect them to the base. According to the location of the connection points, two
 133 solutions can be distinguished as shown in Fig. 2: (1) drive unit with zero-length ground
 134 link, and (2) drive unit with non-zero-length ground link.



135

136 Fig. 2 Design of drive unit. (a) Case 1: drive unit with zero-length ground link, and (b)
 137 Case 2: drive unit with non-zero-length ground link

138

139 2.1 Analysis of Case 1

140

141 As shown in Fig. 2(a), drive unit with zero-length ground link consists of four curved links
 142 with the same angle h . All links keep moving on a spherical surface with radius r . The
 143 drive angles of the active links AC and AB are θ_1 and θ_2 respectively. The output axis
 144 is located on the symmetrical plane of the unit and its attitude angles are α_1 and α_2
 145 respectively. To prevent singularity, let $h \in (0, \pi/2)$, $\alpha_2 \in (0, \pi)$.

146 2.1.1 Forward Kinematics

147

148 The pitch angle α_2 is uniquely determined by $\Delta\theta = (\theta_2 - \theta_1)$, let $\Delta\theta \in (0, \pi)$. According to
 149 the spherical cosine theorem and spherical Pythagorean theorem

$$150 \quad \alpha_2 = 2\angle AOD = 2\cos^{-1}(\cos h / \cos(\angle BOC / 2)) \quad (1)$$

151 where $\cos \angle BOC = \cos^2 h + \sin^2 h \cos \Delta\theta$.

152 The yaw angle α_1 is determined by the angle bisector of the two drive angles. Thus, the
 153 forward kinematics equation of the mechanism is obtained as follows.

$$154 \quad \begin{cases} \alpha_1 = 0.5(\theta_1 + \theta_2) \\ \alpha_2 = 2\cos^{-1}(\cos h / \cos(\angle BOC / 2)) \end{cases} \quad (2)$$

155 If transformations $\theta = (\theta_1 + \theta_2)$ and $\Delta\theta = (\theta_2 - \theta_1)$ are applied, it can be considered that
 156 the synthesized 2-DOF RCM motion is decoupled.

157 2.1.2 Inverse Kinematics

158

159 The inverse kinematics can be obtained by deducing the forward kinematics in reverse.

$$160 \quad \begin{cases} \theta_1 = \alpha_1 - 0.5 \cos^{-1} \left(\left(\cos \left(2 \cos^{-1} \left(\cos h / \cos 0.5 \alpha_2 \right) \right) - \cos^2 h \right) / \sin^2 h \right) \\ \theta_2 = \alpha_1 + 0.5 \cos^{-1} \left(\left(\cos \left(2 \cos^{-1} \left(\cos h / \cos 0.5 \alpha_2 \right) \right) - \cos^2 h \right) / \sin^2 h \right) \end{cases} \quad (3)$$

161 *2.1.3 Singularity*

162

163 The relationship between joint velocity and end velocity can be described by the Jacobian
164 matrix.

$$165 \quad \mathbf{v} = \mathbf{J} \dot{\boldsymbol{\theta}} \quad (4)$$

$$166 \quad \mathbf{J} = \begin{bmatrix} \frac{\partial \alpha_1}{\partial \theta_1} & \frac{\partial \alpha_1}{\partial \theta_2} \\ \frac{\partial \alpha_2}{\partial \theta_1} & \frac{\partial \alpha_2}{\partial \theta_2} \end{bmatrix} = \begin{bmatrix} 0.5 & 0.5 \\ -\frac{\sin^2 h \cdot \cosh \cdot \sin M \cdot \sin \Delta \theta}{\sqrt{1 - \cos^2 2M} \cdot \sin 0.5 \alpha_2 \cdot \cos^2 M} & \frac{\sin^2 h \cdot \cosh \cdot \sin M \cdot \sin \Delta \theta}{\sqrt{1 - \cos^2 2M} \cdot \sin 0.5 \alpha_2 \cdot \cos^2 M} \end{bmatrix}$$

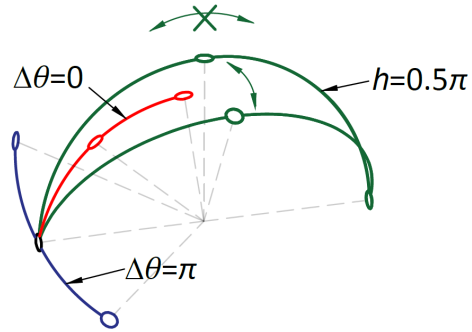
$$167 \quad (5)$$

168 where $M = 0.5 \cos^{-1} (\cos^2 h + \sin^2 h \cdot \cos \Delta \theta)$.169 Singularity occurs when the value of Jacobian determinant is 0, and the solution is $\Delta \theta = 0$ 170 or π , and $h = \pi$ or 0.5π . As shown in Fig. 3, when $\Delta \theta = \pi$, the drive unit is fully

171 retracted, all links overlap into one plane, and a forward kinematics singularity occurs.

172 When $\Delta \theta = 0$, the drive unit is fully expanded, resulting in an uncontrollable degree of173 freedom. When $h = 0.5\pi$, the drive unit loses its ability to extend. Similarly, $h = \pi$ is also

174 a singular state.



175

176

Fig. 3 Singularity of drive unit with zero-length ground link

177

178 **2.2 Analysis of Case 2**

179

180 As shown in Fig. 4, drive unit with non-zero-length ground link consists of five curved links.

181 Link BD is the base of the motion unit, $\angle AOB = \angle AOD = h_1$. The drive angles of the active

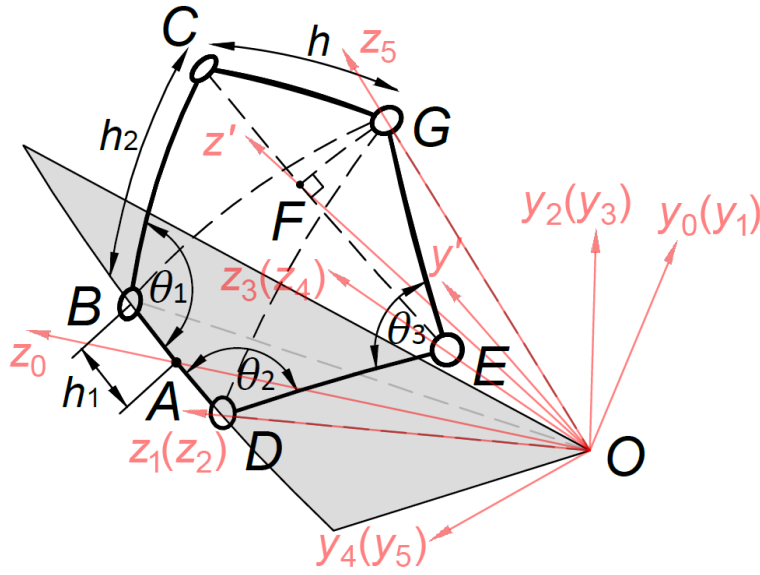
182 links BC and DE are θ_1 and θ_2 respectively, and they are both in the range of $(0, \pi)$.

183 It is known that, $\angle BOC = \angle DOE = h_2$, $\angle COG = \angle GOE = h$, and the angles of all links are less

184 than $\pi / 2$. The attitude angles of output axis α_1 and α_2 are shown in Fig. 2(b).

185 **2.2.1 Forward Kinematics**

186



187

188

Fig. 4 Drive unit with non-zero-length ground link

189

190 As shown in Fig. 4, establish the reference coordinate systems at each joint and let the z
 191 -axes pass through the joint axes. The y -axes are perpendicular to the planes
 192 determined by the links and point to the center of the unit, and the x -axes are
 193 determined by the right-hand screw rule. The Z_0 -axis of the base coordinate system
 194 passes through point A .

195 Then vector OC and vector OE can be expressed as

196
$$OC = R_y(-h_1) \cdot R_z(\theta_1 - \pi) \cdot R_y(-h_2) \cdot z, \quad OE = R_y(h_1) \cdot R_z(\pi - \theta_2) \cdot R_y(h_2) \cdot z \quad (6)$$

197 Where R_y and R_z represent the rotation matrix around the Y -axis and z -axis

198 respectively, $R_y(\chi) = \begin{bmatrix} \cos \chi & 0 & \sin \chi \\ 0 & 1 & 0 \\ -\sin \chi & 0 & \cos \chi \end{bmatrix}$, $R_z(\phi) = \begin{bmatrix} \cos \phi & -\sin \phi & 0 \\ \sin \phi & \cos \phi & 0 \\ 0 & 0 & 1 \end{bmatrix}$.

199 Establish an intermediate coordinate system through the \mathbf{OF} axis, and the direction of
 200 each coordinate axis is

$$201 \quad \mathbf{z}' = \mathbf{OF} = \mathbf{OC} + \mathbf{OE}, \mathbf{y}' = \mathbf{EC} = \mathbf{OC} - \mathbf{OE}, \mathbf{x}' = \mathbf{y}' \times \mathbf{z}' \quad (7)$$

202 Thus, the rotation matrix of the intermediate coordinate system can be obtained

$$203 \quad \mathbf{R}_1 = \begin{bmatrix} \frac{\mathbf{x}'}{|\mathbf{x}'|} & \frac{\mathbf{y}'}{|\mathbf{y}'|} & \frac{\mathbf{z}'}{|\mathbf{z}'|} \end{bmatrix} \quad (8)$$

204 $\angle FOG$ can be expressed as

$$205 \quad \angle FOG = \cos^{-1} \left(\cosh / \cos(\cos^{-1}(\mathbf{OC} \cdot \mathbf{OE}) / 2) \right) \quad (9)$$

206 Finally, the vector of the output axis can be obtained

$$207 \quad \mathbf{OG} = \mathbf{R}_1 \cdot \mathbf{R}_y(\angle FOG) \cdot \mathbf{z} \quad (10)$$

208 According to the above analysis, it can be seen that the forward kinematics of drive unit
 209 with non-zero-length ground link is extremely complex. Substituting Eqs. (6-9) into Eq. (10)
 210 will get a lengthy analytical equation, which severely limits the efficiency of kinematics
 211 calculation and is not appropriate for real-time control of the robot in practical
 212 applications.

213 2.2.2 Inverse Kinematics

214

215 The inverse kinematics needs to use the given output axis vector \mathbf{OG} to solve the drive
 216 angles θ_1 and θ_2 of two active links.

$$217 \quad \angle BOG = \cos^{-1}(\mathbf{OB} \cdot \mathbf{OG}), \angle DOG = \cos^{-1}(\mathbf{OD} \cdot \mathbf{OG}) \quad (11)$$

218 Apply cosine theorem on the spherical $\triangle BCG$, $\triangle BDG$ and $\triangle DEG$.

$$\begin{aligned}
\angle CBG &= \cos^{-1} \left(\frac{\cos h - \cos h_2 \cos \angle BOG}{\sin h_2 \sin \angle BOG} \right) \\
\angle EDG &= \cos^{-1} \left(\frac{\cos h - \cos h_2 \cos \angle DOG}{\sin h_2 \sin \angle DOG} \right) \\
\angle GBD &= \cos^{-1} \left(\frac{\cos \angle DOG - \cos 2h_1 \cos \angle BOG}{\sin 2h_1 \sin \angle BOG} \right) \\
\angle GDB &= \cos^{-1} \left(\frac{\cos \angle BOG - \cos 2h_1 \cos \angle DOG}{\sin 2h_1 \sin \angle DOG} \right)
\end{aligned} \tag{12}$$

Hence, referring to Fig. 4, the two driving angles θ_1 and θ_2 can be obtained as,

$$\begin{cases} \theta_1 = \angle CBG + \angle GBD \\ \theta_2 = \angle EDG + \angle GDB \end{cases} \tag{13}$$

2.2.3 Singularity

223

224 The forward kinematics of drive unit with non-zero-length ground link is complex, and it
 225 is difficult to obtain Jacobian matrix. Consider dividing the drive unit and discuss the
 226 singularities of each part separately. Firstly, the singularity caused by link CG and link
 227 GE is similar to zero-length ground link drive unit. As shown in Fig. 5 (a), CG and GE
 228 have two collinear states. When overlapping collinear, there is an uncontrollable degree
 229 of freedom. When straightening collinear, it is a forward kinematics singularity.

230 The singularity caused by link DE and link EG can be solved by D-H parameter method.

231 Suppose $\angle DEG$ is θ_3 , the forward kinematics equation can be expressed as

$${}^0_5\mathbf{R} = {}^0_1\mathbf{R} \cdot {}^1_2\mathbf{R} \cdot {}^2_3\mathbf{R} \cdot {}^3_4\mathbf{R} \cdot {}^4_5\mathbf{R} \tag{14}$$

233 where ${}^A_B\mathbf{R}$ represents the rotation matrix of B coordinate system relative to A
 234 coordinate system, ${}^0_1\mathbf{R} = \mathbf{R}_y(h_1)$, ${}^1_2\mathbf{R} = \mathbf{R}_z(\pi - \theta_2)$, ${}^2_3\mathbf{R} = \mathbf{R}_y(h_2)$, ${}^3_4\mathbf{R} = \mathbf{R}_z(\pi - \theta_3)$,
 235 ${}^4_5\mathbf{R} = \mathbf{R}_y(h)$.

236 Thus,

$$237 \quad {}^2_5\mathbf{R} = {}^2_3\mathbf{R} \cdot {}^3_4\mathbf{R} \cdot {}^4_5\mathbf{R} = \begin{bmatrix} -ch_2c\theta_3ch - sh_2sh & ch_2s\theta_3 & ch_2c\theta_3sh - sh_2ch \\ -s\theta_3ch & -c\theta_3 & s\theta_3sh \\ -sh_2c\theta_3ch + ch_2sh & sh_2s\theta_3 & sh_2c\theta_3sh + ch_2ch \end{bmatrix}, {}^4_5\mathbf{R} = \begin{bmatrix} ch & 0 & -sh \\ 0 & 1 & 0 \\ sh & 0 & ch \end{bmatrix} \quad (15)$$

238 According to the principle of differential transformation

$$239 \quad {}^T\boldsymbol{\omega} = \begin{bmatrix} {}^TJ_2 & {}^TJ_4 \end{bmatrix} \begin{bmatrix} \dot{\theta}_2 \\ \dot{\theta}_3 \end{bmatrix} \quad (16)$$

240 where ${}^T\boldsymbol{\omega} = [{}^T\omega_x \quad {}^T\omega_y \quad {}^T\omega_z]^T$, indicating the angular velocity of the output axis relative
 241 to the end coordinate system. TJ_i represents the angular velocity caused by the unit joint
 242 velocity of joint i ,

$$243 \quad {}^TJ_2 = {}^2_5\mathbf{R}^T \begin{bmatrix} 0 \\ 0 \\ 1 \end{bmatrix} = \begin{bmatrix} -sh_2c\theta_3ch + ch_2sh \\ sh_2s\theta_3 \\ sh_2c\theta_3sh + ch_2ch \end{bmatrix}, {}^TJ_4 = {}^4_5\mathbf{R}^T \begin{bmatrix} 0 \\ 0 \\ 1 \end{bmatrix} = \begin{bmatrix} sh \\ 0 \\ ch \end{bmatrix} \quad (17)$$

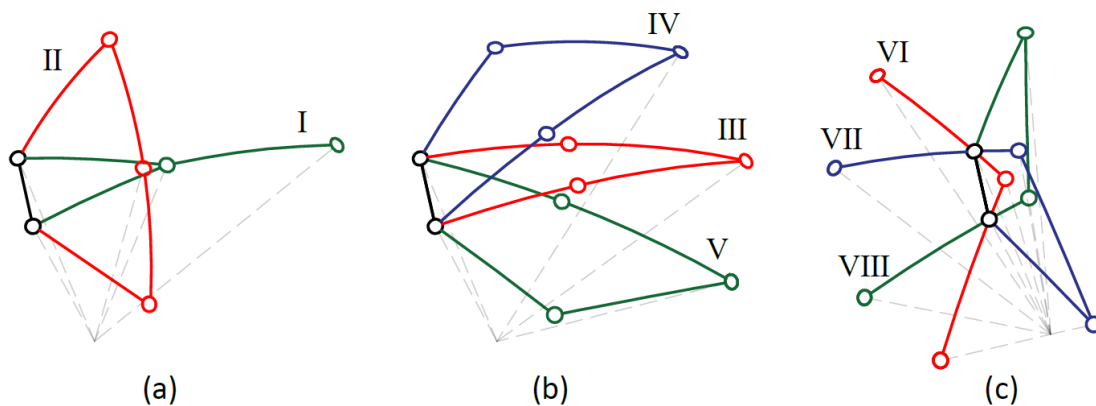
244 Ignoring the rotation of the end coordinate system around the z -axis, the Jacobian matrix
 245 can be expressed as

$$246 \quad {}^TJ = \begin{bmatrix} -sh_2c\theta_3ch + ch_2sh & sh \\ sh_2s\theta_3 & 0 \end{bmatrix} \quad (18)$$

247 Let the value of determinant $\sin h \sin h_2 \sin \theta_3 = 0$, and the solution is $\sin \theta_3 = 0$, which
 248 means that when DE and EG are collinear, the inverse kinematics singularity occurs.

249 Similarly, when BC is collinear with CG , it is also a singular state. Two collinear states
 250 are shown in Fig. 5 (b) and (c), respectively.

251 It is noteworthy that when $h > h_1 + h_2$, situation II will not occur; When $h < h_1 + h_2$,
 252 situation VI will not occur; When $h = h_1 + h_2$, situations II and VI occur at the same time,
 253 and the five links are located on the same plane.

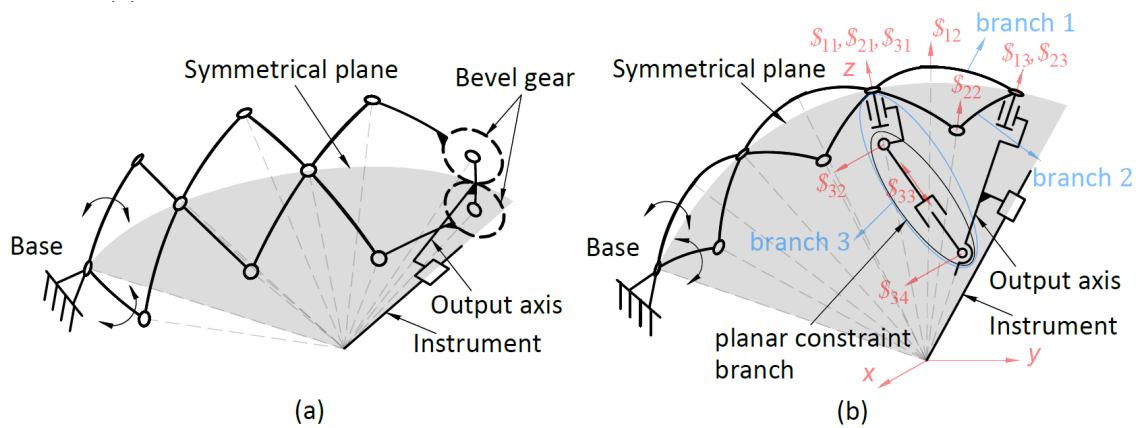


254 (a)
 255 Fig. 5 Singularity of drive unit with non-zero-length ground link. (a) collinear of CG and
 256 GE , (b) straightening collinear, and (c) overlapping collinear

257 3 Conceptual Design of the Output Unit

258
 259
 260 According to the analysis results in Sec. 2, it can be seen that it is difficult to solve
 261 the forward kinematics of drive unit with non-zero-length ground link, and the singular
 262 configurations are complex. In addition, non-zero-length ground link drive unit cannot
 263 realize 360° rotation of yaw DOF like zero-length ground link drive unit. Hence, drive unit
 264 with zero-length ground link is selected as the drive for the proposed RCM mechanism.

265 To meet the operation requirements of surgical instruments, it is necessary to
 266 limit the rotation of the output shaft around its own axis, so that the instrument can be
 267 kept on the symmetrical plane of the mechanism. For the two solutions given in Fig. 6, an
 268 arc linkage with bevel gear constraint and a planar constraint branch are added on the
 269 basis of the spherical four-bar unit respectively.



270 (a) (b)
 271 Fig. 6 Design of output unit. (a) Case 1: bevel gear constraint, and (b) Case 2: planar
 272 branch constraint

273 3.1 Analysis of Case 1

274 As shown in Fig. 6 (a), an arc linkage is added on the basis of the four-bar unit. To eliminate
 275 an extra degree of freedom, a gear constraint is added to form a symmetrical spherical
 276 five-bar unit, and the output axis is limited to the symmetrical plane of the spherical
 277 mechanism.

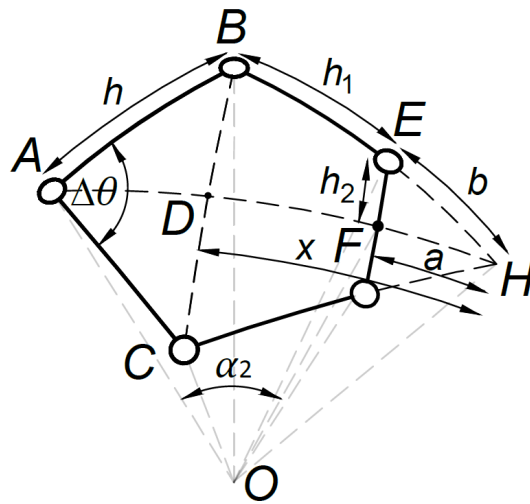
278 There are $n = 6$ links and $j = 7$ joints (including 6 revolute joints and 1 bevel gear joint) in
 279 the output unit of case 1. According to the Grübler-Kutzbach formula [19], the DOF of the
 280 mechanism is
 281
 282

283
$$F = d(n - j - 1) + \sum_{i=1}^j f_i + \nu - \xi = 3(6 - 7 - 1) + 6 + 2 = 2 \quad (19)$$

284 where $d = 6 - \lambda$ is the dimension of the space in which the mechanism is presented,
 285 $d = 6$ for spatial mechanism and $d = 3$ for planar and spherical mechanisms. Each joint
 286 has f_i degrees of freedom. The mechanism has λ common constraints, ν redundant
 287 constraints, and ξ local degrees of freedom. Through the test in the model, it is found
 288 that once any joint constraint is removed, the original motion cannot be realized, so there
 289 is no redundant constraint. All the components in the mechanism have no local motion
 290 so there is no local degree of freedom either.

291 *3.1.1 Forward Kinematics*

292



293

294 Fig. 7 Symmetrical five-bar spherical unit

295

296 The shape of the unit is only affected by $\Delta\theta$. As shown in Fig. 7, extend arc BE and arc
 297 CF intersect at point H , assuming that $\angle EOH = b$, $\angle FOH = a$, $\angle DOH = x$. It is

298 known that, $\angle BOE = h_1$, $\angle EOF = h_2$, and the angles of all links are less than $\pi / 2$. The
 299 calculation method of $\angle AOD$ has been given in Sec. 2.1, and the calculation of $\angle DOF$ is
 300 carried out below.

301 Apply sine theorem on the spherical $\triangle EHF$ and $\triangle BHD$

$$302 \quad \begin{cases} \sin b = \sin h_2 / \sin \angle EHF \\ \sin(h_1 + b) = \sin \angle BOD / \sin \angle EHF \end{cases} \quad (20)$$

303 Thus,

$$304 \quad b = \begin{cases} \tan^{-1}(\sin h_2 \sin h_1 / (\sin \angle BOD - \sin h_2 \cos h_1)) & \angle BEF \geq 90^\circ \\ \tan^{-1}(\sin h_2 \sin h_1 / (\sin \angle BOD - \sin h_2 \cos h_1)) + \pi & \angle BEF < 90^\circ \end{cases} \quad (21)$$

305 Apply cosine theorem on the spherical $\triangle EHF$ and $\triangle BHD$

$$306 \quad \begin{cases} a = \cos^{-1}(\cos b / \cos h_2) \\ x = \cos^{-1}(\cos(h_1 + b) / \cos \angle BOD) \end{cases} \quad (22)$$

307 The pitch angle of output axis can be obtained by substituting Eq. (1) and Eqs. (21-22) into
 308 the following formula.

$$309 \quad \alpha_2 = \angle AOD + x - a \quad (23)$$

310 The calculation of yaw angle follows the same principle as Eq. (2).

311 3.1.2 Inverse Kinematics

312

313 Given α_2 , $\angle AOE$ can be obtained by the spherical Pythagorean theorem. Apply cosine

314 theorem on the spherical $\triangle EAF$ and $\triangle BAE$ to get

$$\begin{aligned}
315 \quad \angle EAF &= \cos^{-1} \left(\frac{\cos h_2 - \cos \alpha_2 \cos \angle AOE}{\sin \alpha_2 \sin \angle AOE} \right) \\
\angle BAE &= \cos^{-1} \left(\frac{\cos h_1 - \cos h \cos \angle AOE}{\sin h \sin \angle AOE} \right)
\end{aligned} \tag{24}$$

316 where $\angle AOE = \cos^{-1}(\cos h_2 \cos \alpha_2)$.

317 Thus, the driving angle

$$318 \quad \Delta\theta = 2(\angle BAE + \angle EAF) \tag{25}$$

319 3.1.3 Singularity

320

321 Differentiating Eq. (23) yields

$$322 \quad \frac{d\alpha_2}{dD} = -\frac{\cos h \cdot \sin D}{\cos D \sqrt{\cos^2 D - \cos^2 h}} - \frac{\cos D \cdot \sin H \cdot N + \sin D \cdot \cos H}{\cos D \sqrt{\cos^2 D - \cos^2 H}} + \frac{\sin b \cdot N}{\sqrt{\cos^2 h_2 - \cos^2 b}}$$

323 (26)

$$324 \quad \frac{dD}{d\Delta\theta} = \frac{\sin^2 h \cdot \sin \Delta\theta}{2\sqrt{1 - (\cos^2 h + \sin^2 h \cdot \cos \Delta\theta)^2}} \tag{27}$$

$$325 \quad J = \frac{d\alpha_2}{dD} \cdot \frac{dD}{d\Delta\theta} \tag{28}$$

326 where $N = \cos D \sin h_1 \sin h_2 / \left((\sin D - \sin h_2 \cos h_1)^2 + \sin^2 h_1 \sin^2 h_2 \right)$, D represents

327 $\angle BOD$, and H represents $\angle BOH$. Then discuss the singularity in the range of

328 $\Delta\theta \in (0, 180^\circ)$ based on the above results.

329 (1) When $\Delta\theta = 0$, a triangle is formed at the end, resulting in an uncontrollable degree of

330 freedom.

331 (2) When AB coincides with BE , $\sin D = \sin h \sin h_2 / \sin(h + h_1)$, and $b = \pi - (h + h_1)$.

332 Substitute them into Eq. (28) and the numerator is 0, indicating that it is an inverse
333 kinematics singularity.

334 (3) When BE coincides with EF , $\angle BOD = h_1 + h_2$, and $b = h_2$. At this time, the
335 denominator of Eq. (28) is 0, indicating that it is a forward kinematics singularity. Note
336 that this condition occurs only when $h > h_1 + h_2$.

337

338 3.2 Analysis of Case 2

339

340 The second solution, as shown in Fig. 6 (b), is by eliminating the rotation constraint of the
341 output axis and another axis on the symmetry plane, and adding a planar constraint
342 branch between two axes to limit the rotation of the output shaft.

343 According to the screw theory, the motion-screw systems of the three branches are:

$$\begin{aligned}
 \{\mathcal{S}_{11}\} &= \left\{ \begin{array}{l} \mathcal{S}_{11} = (0, 0, 1, 0, 0, 0)^T \\ \mathcal{S}_{12} = (a_2, b_2, c_2, 0, 0, 0)^T \\ \mathcal{S}_{13} = (0, b_3, c_3, 0, 0, 0)^T \end{array} \right\}, \{\mathcal{S}_{12}\} = \left\{ \begin{array}{l} \mathcal{S}_{21} = (0, 0, 1, 0, 0, 0)^T \\ \mathcal{S}_{22} = (-a_2, b_2, c_2, 0, 0, 0)^T \\ \mathcal{S}_{23} = (0, b_3, c_3, 0, 0, 0)^T \end{array} \right\}, \\
 \{\mathcal{S}_{13}\} &= \left\{ \begin{array}{l} \mathcal{S}_{31} = (0, 0, 1, 0, 0, 0)^T \\ \mathcal{S}_{32} = (1, 0, 0, 0, 1, 0)^T \\ \mathcal{S}_{33} = (0, 0, 0, 0, e_3, f_3)^T \\ \mathcal{S}_{34} = (1, 0, 0, 0, e_4, f_4)^T \end{array} \right\} \quad (29)
 \end{aligned}$$

345 where a_i, b_i, c_i, e_i , and f_i are parameters determined by the position of the screw.

346 According to the product of reciprocity is zero, each branch constraint-screw system is

347 easily calculated,

$$348 \quad \{\mathbf{S}_{li}^r\} = \left\{ \begin{array}{l} \mathbf{S}_{i1}^r = (1,0,0,0,0,0)^T \\ \mathbf{S}_{i2}^r = (0,1,0,0,0,0)^T \\ \mathbf{S}_{i3}^r = (0,0,1,0,0,0)^T \end{array} \right\}, (i=1,2), \{\mathbf{S}_{l3}^r\} = \left\{ \begin{array}{l} \mathbf{S}_{31}^r = (1,0,0,0,0,0)^T \\ \mathbf{S}_{32}^r = (0,0,0,0,1,0)^T \end{array} \right\} \quad (30)$$

349 The constraint-screw multiset of the output unit combines the three basis sets,

$$350 \quad \langle \mathbf{S}^r \rangle = \{\mathbf{S}_{l1}^r\} + \{\mathbf{S}_{l2}^r\} + \{\mathbf{S}_{l3}^r\} \quad (31)$$

351 where $\text{card} \langle \mathbf{S}^r \rangle = 8$. However, $\langle \mathbf{S}^r \rangle$ only contains four linearly independent screws, so

352 a nonunique basis for the subspace \mathbf{S}^r can be selected as

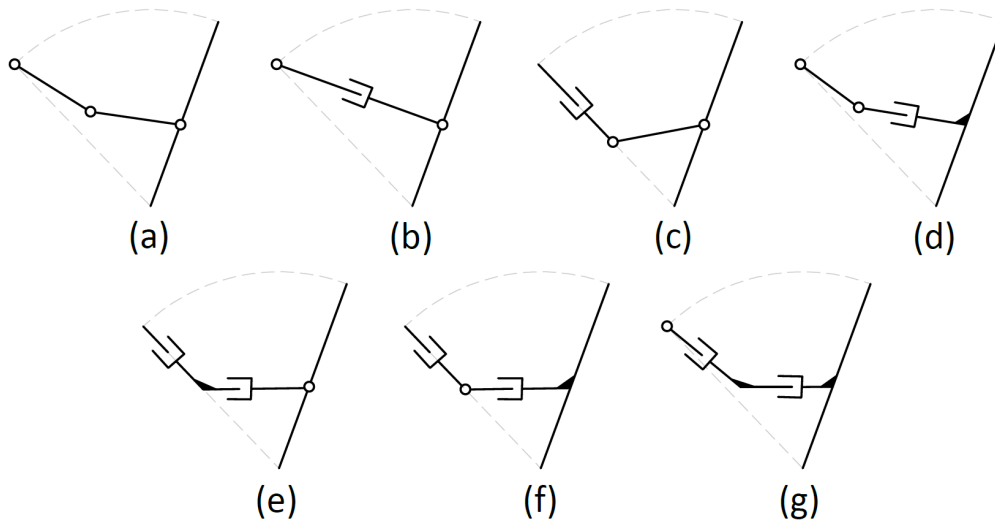
$$353 \quad \{\mathbf{S}^r\} = \left\{ \begin{array}{l} \mathbf{S}_{11}^r = (1,0,0,0,0,0)^T \\ \mathbf{S}_{12}^r = (0,1,0,0,0,0)^T \\ \mathbf{S}_{13}^r = (0,0,1,0,0,0)^T \\ \mathbf{S}_{33}^r = (0,0,0,0,1,0)^T \end{array} \right\} \quad (32)$$

354 Taking the reciprocal of \mathbf{S}^r gives the motion-screw system \mathbf{S}_f with the basis

$$355 \quad \{\mathbf{S}_f\} = \left\{ \begin{array}{l} \mathbf{S}_{f1} = (1,0,0,0,0,0)^T \\ \mathbf{S}_{f2} = (0,0,1,0,0,0)^T \end{array} \right\} \quad (33)$$

356 This shows that the output unit in case 2 has two rotational freedoms along the x -axis

357 and z -axis respectively.



358

359 Fig. 8 Different configurations of planar constrain branch. (a) RRR branch, (b) RPR
 360 branch, (c) PRR branch, (d) RRP branch, (e) PPR branch, (f) PRP branch, and (g) RPP
 361 branch

362

363 It is observed that any planar branch containing three or more revolute or prismatic joints
 364 (and containing at least one revolute joint) in series with the revolute joint along the z-
 365 axis will produce a constraint equivalent to branch 3. According to the number and
 366 position of prismatic joints, the seven layouts in Fig. 8 meet the requirements. After
 367 verification in modeling software, it can be found that, the two prismatic joints in cases
 368 (e)-(g) are prone to interference. Comparing cases (b)-(d), case (b) has the advantage of
 369 compact structure. When the two links are collinear, case (a) will be singular. In conclusion,
 370 case (b) is the best choice.

371

372 4 Kinematic Model and Dimension Synthesis

373

374 According to the analysis result in Sec. 3, the forward kinematics of case 1 needs
 375 to be discussed separately. When it is used in series with spherical four-bar unit, the
 376 solution of inverse kinematics will also become very complex. In addition, the gear
 377 clearance may bring angle error to the end instrument.

378

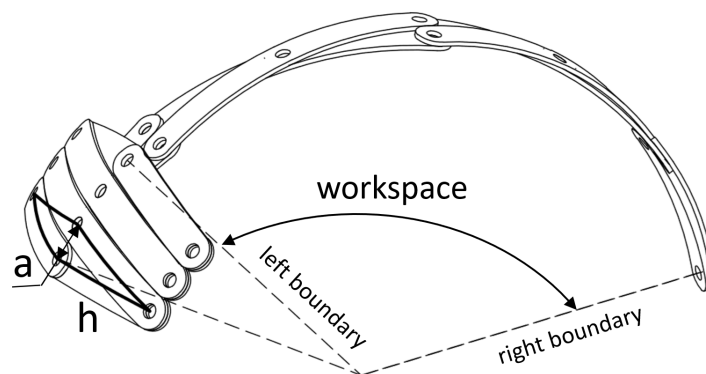
379 4.1 Kinematic Model and Workspace of the spherical four-bar unit based RCM 380 Mechanism

381

382 Based on the above analysis, the series structure composed of n spherical four-bar units
 383 combined with planar constraint branch in Fig. 6(b) is selected as the final solution. Thus,
 384 the kinematics model of the proposed RCM mechanism has an n -fold relationship with
 385 Eq. (2).

$$386 \quad \begin{cases} \alpha_1 = 0.5(\theta_1 + \theta_2) \\ \alpha_2 = 2n \cos^{-1} \left(\cos h / \cos \left(0.5 \cos^{-1} \left(\cos^2 h + \sin^2 h \cdot \cos(\theta_2 - \theta_1) \right) \right) \right) \end{cases} \quad (34)$$

387 In the analysis of the workspace, the critical state in which interference occurs needs to
 388 be considered. For yaw direction, 360° rotation can be achieved in any state. For pitch
 389 direction, due to the existence of solid material at each joint shaft, the linkages will collide
 390 near the singular position, which also makes the robot successfully avoid the singularity.



391

392 Fig. 9 Boundaries on both sides of the workspace

393

394 As shown in Fig. 9, assume that the width of the linkage is a . The arc length is
 395 approximately equal to the chord length when the angle value is small. Thus, the left
 396 boundary $\alpha_{2\min}$, the right boundary $\alpha_{2\max}$, and the total actual workspace α_{2all} can be
 397 approximately expressed as:

$$398 \quad \alpha_{2\min} = na / r, \quad \alpha_{2\max} = 2n \cos^{-1}(\cosh / \cos(a / 2r)), \quad \alpha_{2all} = \alpha_{2\max} - \alpha_{2\min} \quad (35)$$

399 Therefore, the actual workspace is not only affected by the number of units n and the
 400 angle of linkages h , but also related to a and r . It can be seen from Eq. (35) that the
 401 smaller the rod width a , the larger the spherical radius r , the smaller the loss angle and
 402 the larger the total working space. In this mechanism, the minimum value of a is 30mm.

403 Hence, the optimum dimensional synthesis of the mechanism can be summarized as:

404 Given the workspace angle α_{2all} , determine n , h , and r such that the optimality of the
 405 global performance can be achieved.

406

407 **4.2 Optimization and Dimension Synthesis**

408

409 *4.2.1 Object function*

410

411 Due to the precise operation in the process of MIS, the robots need to have good
 412 operational flexibility for the needs of surgical safety. Simultaneously, in order to avoid
 413 interference among the multi-robot system, the occupied volume of the robot should also
 414 be considered. Therefore, in the process of dimension synthesis, two functions of
 415 operation performance index and compactness index should be considered.

416 The condition number κ of Jacobian matrix describes the uniformity of transformation in
 417 all directions [20-22]. The smaller the condition number, the closer the robot's movement
 418 ability in all directions.

419 The singular values of Jacobian matrix can be determined by solving the characteristic
 420 equation $\det(\sigma^2 E - J \cdot J^T) = 0$. The results are $\sqrt{0.5}$ and $|\sqrt{2Q}|$ respectively, where

$$421 \quad Q = \frac{n \sin^2 h \cdot \cosh \cdot \sin M \cdot \sin \Delta \theta}{\sqrt{1 - \cos^2 2M} \cdot \sin(\alpha_2 / 2n) \cdot \cos^2 M}, \quad M \text{ is given in Eq. (5). It can be seen that the}$$

422 flexibility of the yaw direction is a constant, and the flexibility of the pitch direction
 423 changes with $\Delta \theta$.

$$424 \quad \kappa = \frac{\sigma_2}{\sigma_1} = \begin{cases} |1/2Q| & \sqrt{0.5} \geq |\sqrt{2Q}| \\ |2Q| & \sqrt{0.5} < |\sqrt{2Q}| \end{cases} \quad (36)$$

425 Where σ_1 and σ_2 represent the minimum and maximum singular values respectively.

426 Considering that κ varies with the configuration of the robot, $\bar{\eta}$ and $\tilde{\eta}$ are used to
 427 describe the average value and fluctuation degree of global operational performance,
 428 respectively. The former index is similar to that proposed by Gosselin and Angeles [23].

$$429 \quad \bar{\eta} = \int \kappa dW / \int dW = \frac{1}{\alpha_{2all}} \int_{\alpha_{2min}}^{\alpha_{2max}} \kappa d\theta \quad (37)$$

$$430 \quad \tilde{\eta} = \max(\kappa) / \min(\kappa) \quad (38)$$

431 Where $\min(\kappa)$ and $\max(\kappa)$ represent the minimum and maximum values of κ in
 432 working space W . Combining the mean value and the degree of fluctuation, the following
 433 global comprehensive performance index η can be constructed.

434
$$\eta = \sqrt{\bar{\eta}^2 + (\omega_{\eta} \tilde{\eta})^2} \quad (39)$$

435 Where ω_{η} is the weight being placed upon the ratio of $\bar{\eta}$ to $\tilde{\eta}$.

436 In addition, the area occupied in the retracted state is taken as the compactness
437 evaluation index.

438
$$s = b \times 2r \sin(\alpha_{2\min} / 2) \quad (40)$$

439 *4.2.2 Constrains*

440

441 According to the operation requirements of MIS, given the design objective: pitch
442 workspace angle $\alpha_{2all} = 120^\circ$. The constraints that the robot needs to meet are discussed
443 below.

444 Firstly, in order to avoid interference between the instruments and patient's body in the
445 preoperative adjustment process under specific posture, the minimum angle $\alpha_{2\min}$ in the
446 retraction state needs to meet

447
$$\alpha_{2\min} \leq \theta_{\max} \quad (41)$$

448 Secondly, the spherical radius r should be sufficient to accommodate the end
449 translational joint.

450
$$r \geq r_{\min} \quad (42)$$

451 When the robot retracts to the smallest angle, excessive width can easily lead to
452 interference in the operation of multi-robot systems. So given the constraint

453
$$b = 2r \sin(\cos^{-1}(\cosh / \cos(a / 2r))) \leq b_{\max} \quad (43)$$

454 Finally, an excessive number of motion units may lead to greater transmission error and
 455 increase assembly difficulty. Hence, a constraint associated with the number of motion
 456 units n should also be set such that

$$457 \quad n_{\min} \leq n \leq n_{\max} \quad (44)$$

458 459 4.3 Implementation and Discussion

460
461 The optimum dimensional synthesis of the 2-DOF spherical mechanism can be regarded
 462 as the following constrained nonlinear programming problem:

$$463 \quad \eta(x) \rightarrow \min_{x \in R^3} \quad (45)$$

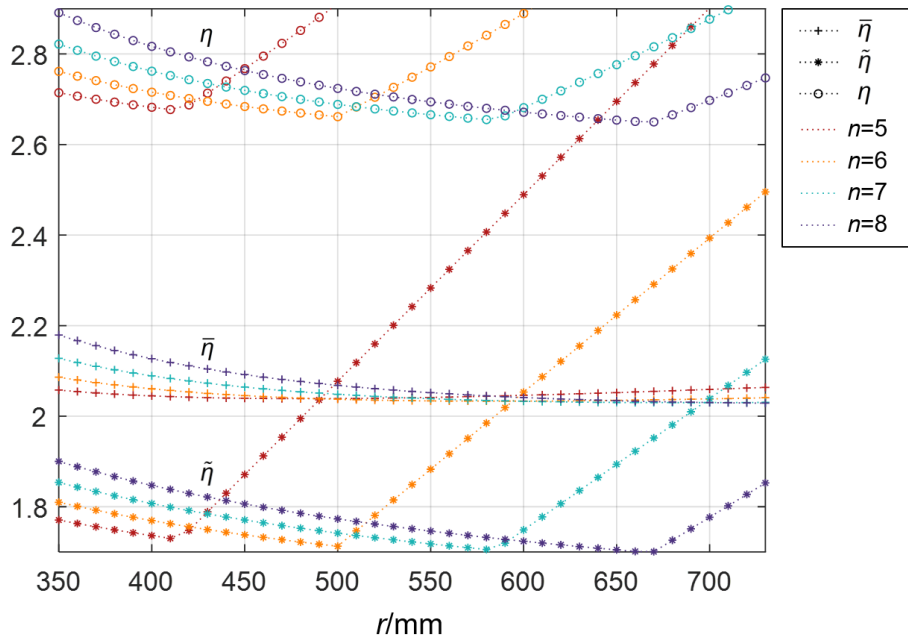
464 subject to the constraints in Eqs. (41) throughout (44), where $x = (n \ r \ h)^T$. Given
 465 $\omega_\eta = 0.6$, $\theta_{\max} = 30^\circ$, $r_{\min} = 460\text{mm}$, $b_{\max} = 300\text{mm}$, $n_{\min} = 4$, $n_{\max} = 8$, and calculate the κ
 466 values in the workspace through bisection node method.

467 Figure 10 shows the variation of $\bar{\eta}$, $\tilde{\eta}$ and η with n and r in the range of $n = 5 \sim 8$ (When
 468 $n = 4$, the constraint conditions cannot be satisfied.) and $r = 350 \sim 750\text{mm}$. As shown in
 469 Fig. 10, both $\bar{\eta}$ and $\tilde{\eta}$ reduces firstly and increases afterward with the increase of r , but
 470 the position of the minimum value is different. Since the change of $\tilde{\eta}$ is more significant,
 471 the trend of η depends more on the change of $\tilde{\eta}$. In addition, the larger the number of
 472 units, the higher the flexibility that the mechanism can achieve, and the larger the optimal
 473 radius, indicating that improving the operating performance must be at the expense of
 474 increasing the volume.

475 When the constraints are met, draw η and s corresponding to different n into the
 476 broken line diagram in Fig. 11. and construct a comprehensive index of compactness and
 477 operability.

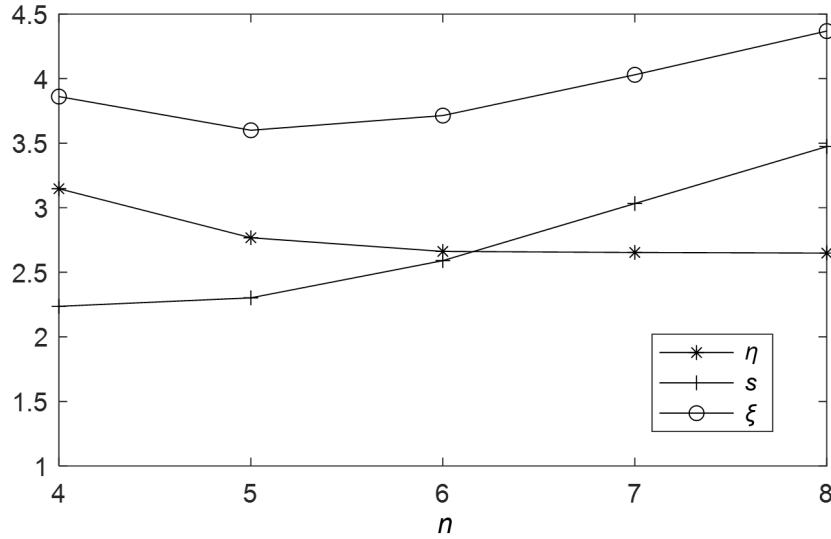
478
$$\xi = \sqrt{\eta^2 + (\omega_s s)^2} \tag{46}$$

479 Where ω_s is the weight being placed upon the ratio of η to s . In order to make η and s
 480 have equivalent values, let $\omega_s = 1/16000$ based on the ratio of the means of the two
 481 indexes. It can be seen that when $n=5$ and $r=460mm$, the comprehensive
 482 performance of compactness and operability is the best.



483
 484 Fig. 10 Variations of $\bar{\eta}$, $\tilde{\eta}$ and η vs. spherical radius r and the number of units n , where
 485 different color curves correspond to different n values.

486



487

Fig. 11 Variations of η , s and ξ vs. the number of units n

488

489

Substituting $\alpha_{2all} = 120^\circ$ into Eq. (35), and all the optimized structural parameters in Tab.

490

1 can be obtained.

491

Tab. 1 Optimized structural parameters

Parameters	n	r	h
Values	5	460mm	14°

492

493

494

5 Prototype and Error Evaluation

495

496

In order to verify the motion accuracy of the proposed RCM mechanism, a

497

prototype as shown in Fig. 12 is design, fabricated and assembled according to the

498

optimization results. When the robot is extended, the output linkage is far from the fixed

499

base, in order to ensure the motion stability and structural stiffness, the first three groups

500

of linkages near the fixed platform are made of stainless steel, and the other parts are

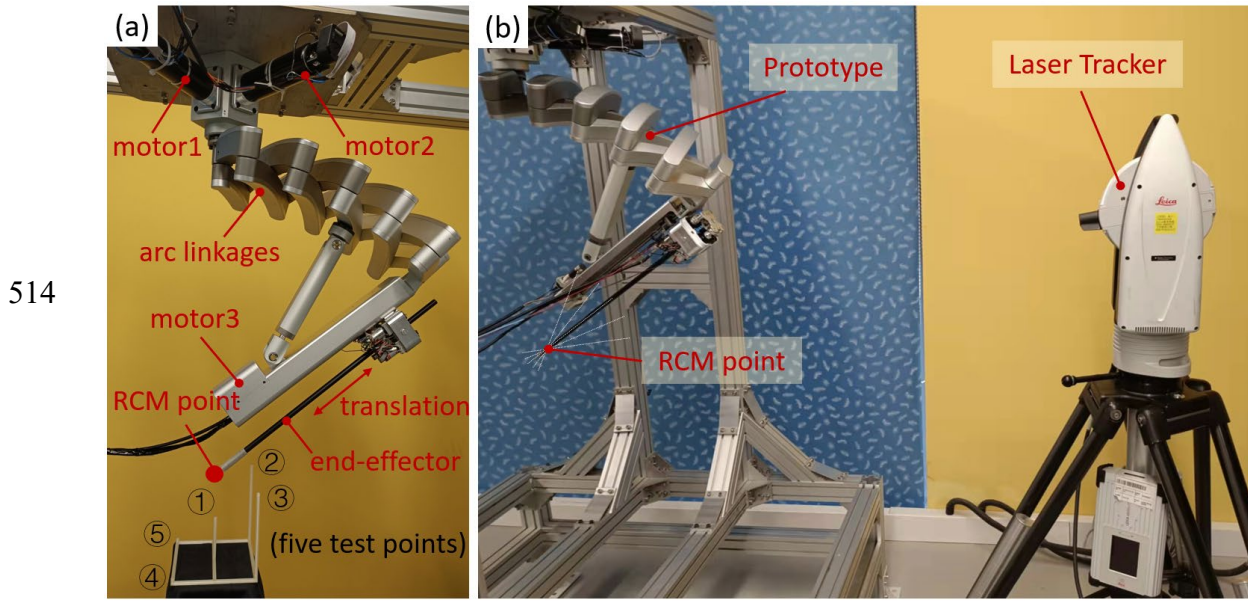
501 made of aluminum alloy to reduce the impact of weight. On the other hand, the clearance
 502 of each revolute joint needs to be concerned. Increasing the contact thickness of the
 503 linkage relative to the diameter of the shaft hole will contribute to reducing the impact
 504 of joint clearance. Reference to the diameter of the hole 26 mm, a relatively larger 30mm
 505 was chosen as the thickness value here. Two servo motors are installed on the fixed
 506 platform in a compact manner, and the motion is transmitted through the bevel gear set
 507 with a reduction ratio of 2. And a translational joint is added at the end to realize one
 508 degree of freedom translation along the instrument axis. The parameters and variables
 509 of the prototype are shown in Tab. 2. The actual position error is measured by Leica
 510 AT960-MR absolute laser tracker of Hexagon Manufacturing Intelligence Company. The
 511 measurement setup is shown in Fig. 12(b).

512

Tab. 2 Parameters and variables of the prototype

Parameters	Values
Minimum pitch angle	$\alpha_{2\min} = 25.2^\circ$
Maximum pitch angle	$\alpha_{2\max} = 145.4^\circ$
Minimum radius of prototype	$R_{\min} = 460mm$
Maximum radius of prototype	$R_{\max} = 532mm$
Maximum width of prototype	$B_{\max} = 284mm$

513

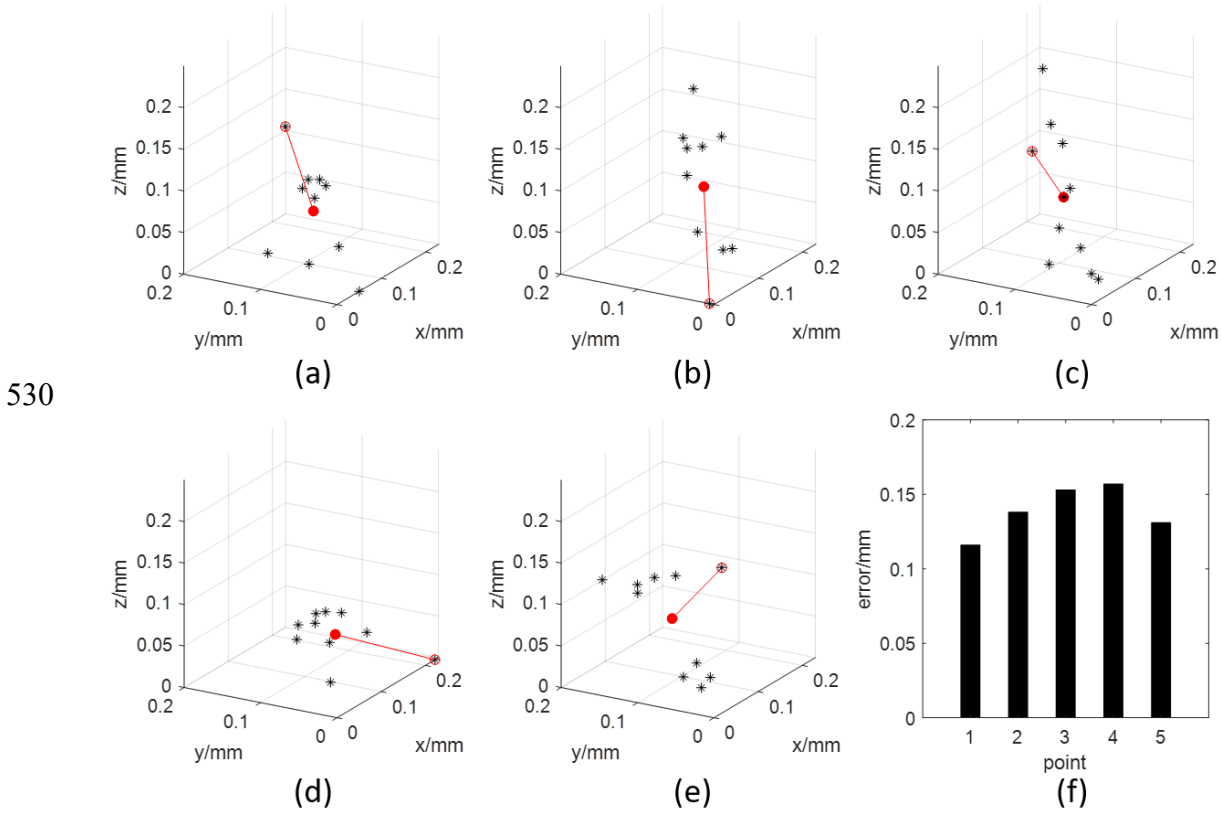


515 Fig. 12 Prototype of the proposed RCM mechanism. (a) composition of prototype, and
 516 (b) experimental measurement setup

517

518 The repeated positioning accuracy of the robot was first measured. As shown in
 519 Fig. 12(a), the parameters of the three motors are adjusted to control the 2R1T motion of
 520 the robot so that the end of the instrument reaches the five test points in turn. Setting
 521 the velocity of the end point as 5cm/s, the control program is cycled 10 times and the co-
 522 ordinate values of the end point of the instrument are recorded by the laser tracker. The
 523 experimental data obtained are given in Fig. 13(a)-(e), where the solid red dots are the
 524 average of each set of data. The distance between the average point and the furthest data
 525 point is defined as the repeated positioning error (length of the red line segment in the
 526 figure), and the calculations are summarized in Fig. 13(f). The repeated positioning errors
 527 at each point are 0.12mm, 0.14mm, 0.15mm, 0.16mm and 0.13mm respectively. The

528 repeated positioning accuracy of this robot is relatively high, so it can be inferred that the
 529 return error caused by joint clearance is very small.



530

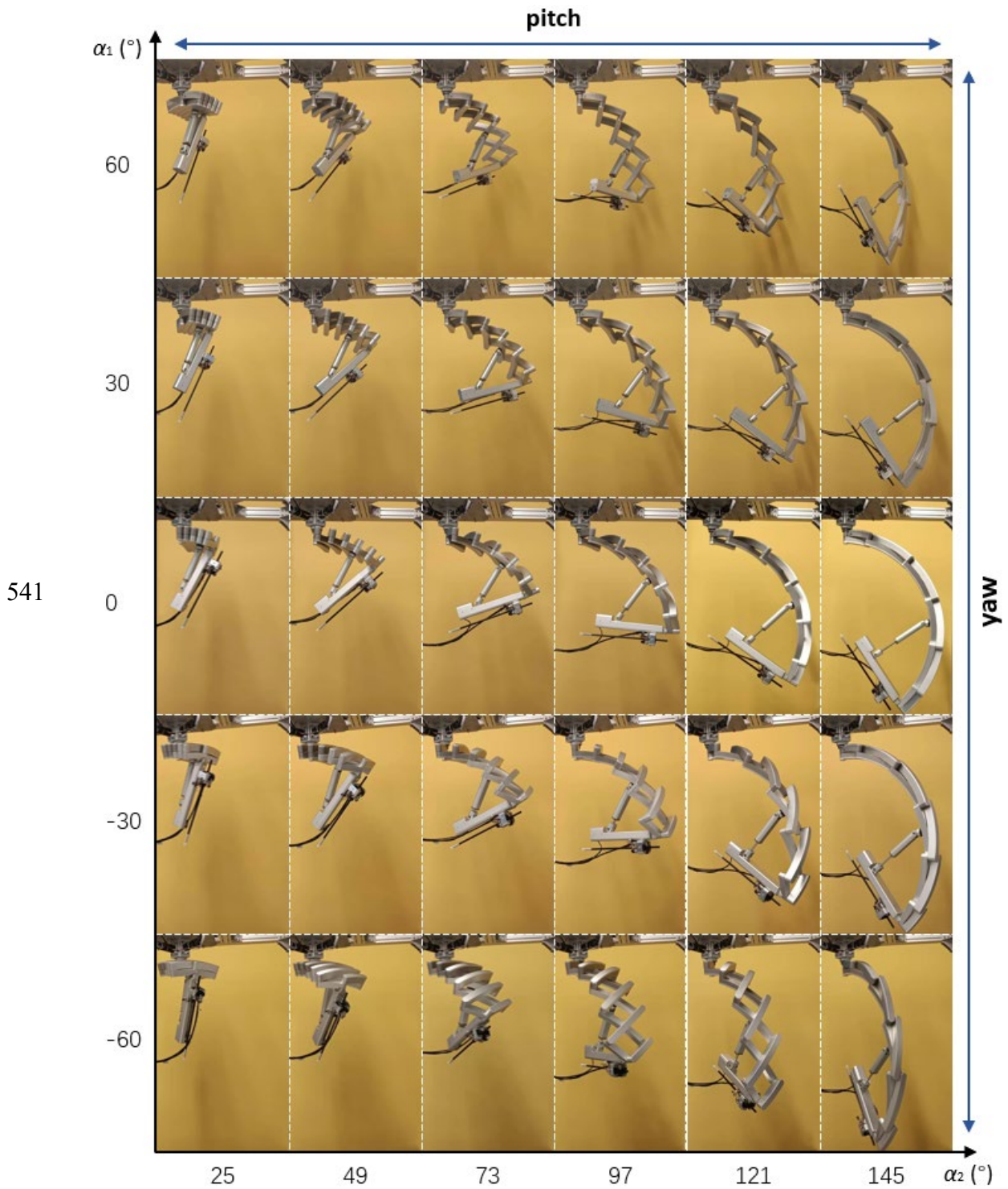
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Fig. 13 The results of repeated positioning accuracy

532

533 Next, the misalignment of the end instrument is measured to verify the RCM
 534 characteristics of the prototype. The end point of the instrument is made to coincide with
 535 the RCM point in the axial direction, and then the 2-DOF rotations are adjusted to drive
 536 the robot to each of the $5 \times 6 = 30$ states in Fig. 14. The pitch angle is between 25° and
 537 145° and the yaw angle is between -60° and 60° . Figure 15 shows the distance error
 538 between the end point of the instrument and the standard RCM point in all acquisition

539 attitudes. The maximum and mean values of these errors are 1.06 mm and 0.56 mm
540 respectively.

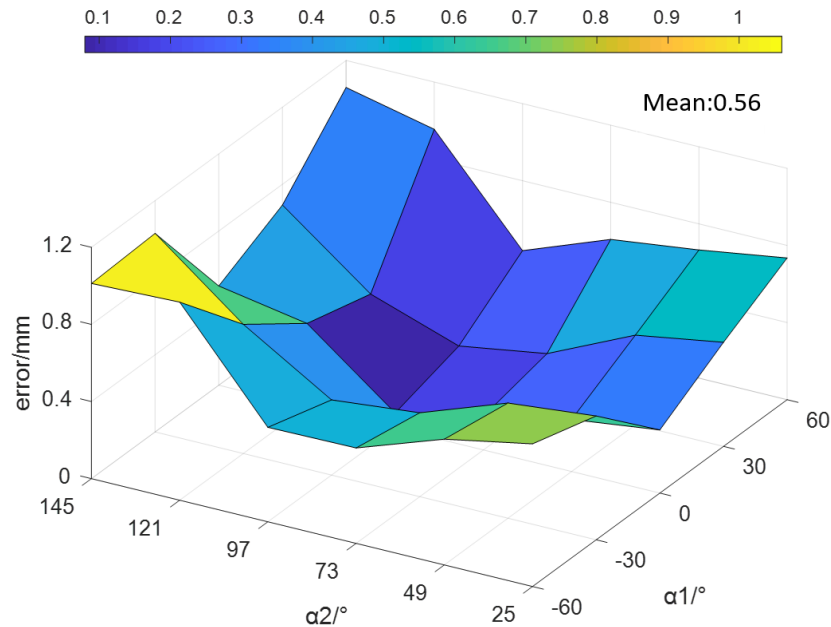


542

Fig. 14 Workspace division of the developed RCM prototype

543

544



545

Fig. 15 The position error of the RCM point

546

547 It is noteworthy that the position error of general MIS robot should be within 2
 548 mm [24]. Therefore, it can be considered that the developed RCM prototype can provide
 549 a stable remote center for surgical tasks. When α_1 deviates from 0° , the position error of
 550 the developed prototype will increase accordingly. When α_2 deviates from 90° , the
 551 position error increases at a more significant rate. This indicates that the yaw accuracy is
 552 slightly higher than the pitch accuracy. Considering the randomness of gear clearance,
 553 manufacturing error and assembly error, the experimental results are completely
 554 acceptable. In the future work, the accuracy of the prototype can be further improved by
 555 using high-precision manufacturing and assembly technology, initial configuration

556 calibration technology and structural optimization. Therefore, it is believed that the
557 proposed RCM robot with unique structure has potential application in MIS robot.

558

559 **6 Conclusions**

560

561 In conclusion, the mechanism proposed in this paper utilizes the spherical unit to
562 keep all links on the spherical surface with fixed radius, which greatly improves the
563 compactness of the mechanism. It is a major improvement of the existing surgical robot.
564 The mechanism has high volume expansion rate, easy to realize modular design, and can
565 meet the needs of workspace in different situations by increasing or reducing the number
566 of units. For the pitch direction, the existence of solid material at the joints makes the
567 mechanism free from singularity. For yaw direction, 360° rotation can be achieved in any
568 state. For different workspace and constraints, the optimal structural parameters of the
569 mechanism can be determined by dimensional synthesis with the goal of optimal global
570 operation performance. Moreover, an experimental prototype was developed to verify
571 the feasibility of the proposed RCM mechanism. The results show that the repetitive
572 positioning accuracy of the mechanism is within 0.2mm, and the RCM point accuracy is
573 within 1.1mm. Therefore, the proposed 2-DOF RCM mechanism can be used as a precision
574 manipulator for MIS completely.

575

576

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578

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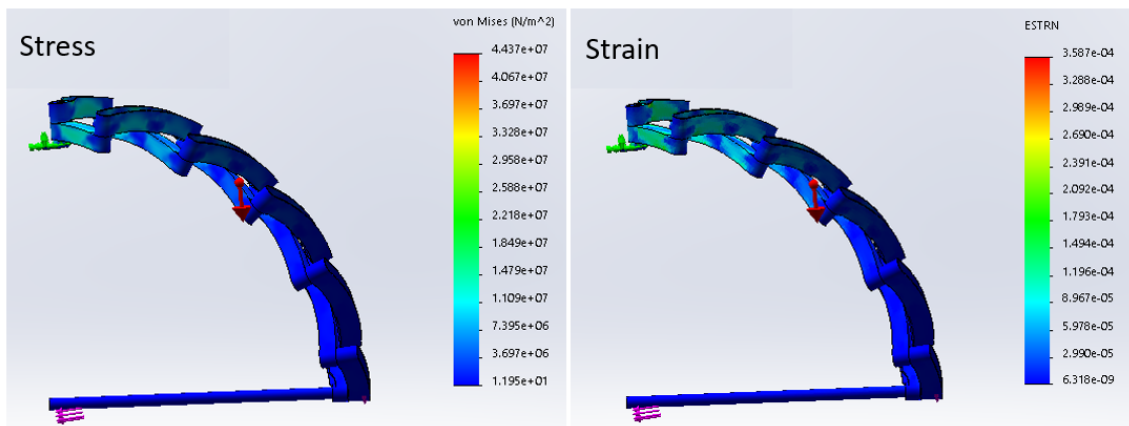
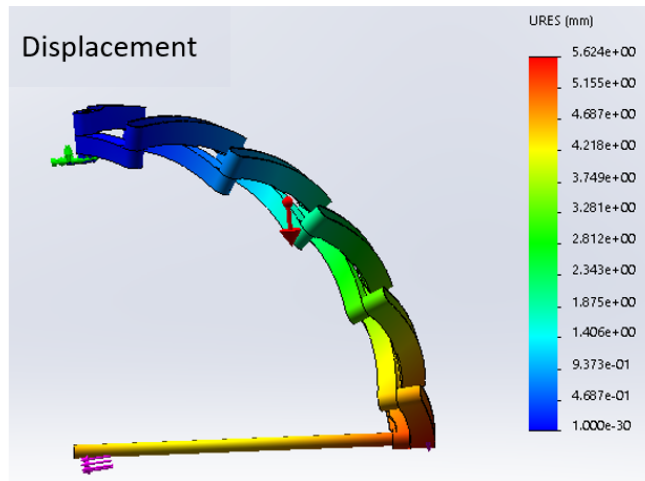
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583

584 **APPENDIX**

585

586 Before manufacturing the prototype, finite element simulation is performed to adjust the
587 material and structural details. A 20 N load force and a 5 N tissue operating force are
588 applied to the end of the curved linkage and the end of the instrument, respectively, and
589 a ground gravitational force is applied. It can be seen that the optimized model has more
590 uniform stress and strain and less deformation.



591

592

Fig. A1 Finite element simulation results before optimization

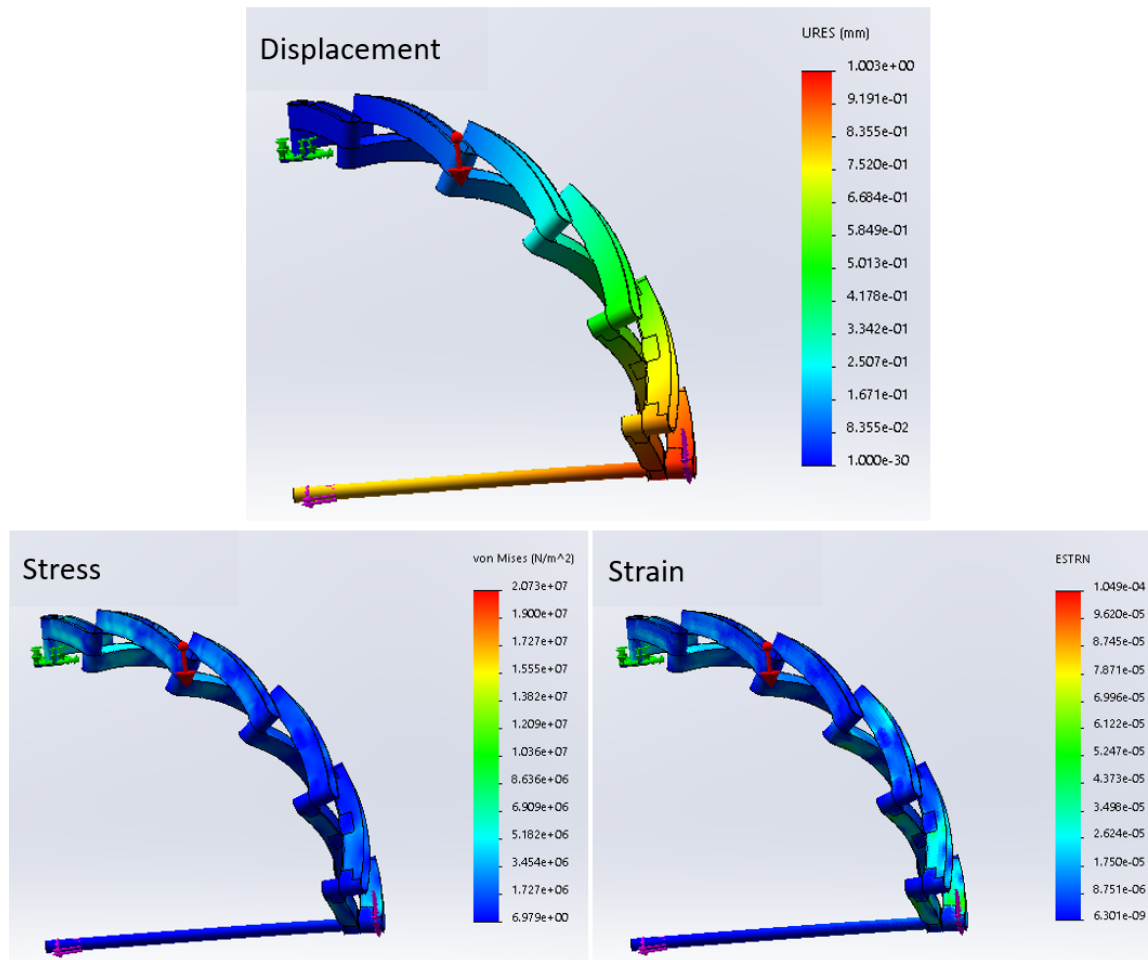
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Fig. A2 Finite element simulation results after optimization

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702**Figure Captions List**

- Fig. 1 Comparison of two kinds of spherical RCM mechanisms. (a) traditional spherical mechanism, and (b) the proposed spherical scissor-like linkage mechanism
- Fig. 2 Design of drive unit. (a) Case 1: drive unit with zero-length ground link, and (b) Case 2: drive unit with non-zero-length ground link
- Fig. 3 Singularity of drive unit with zero-length ground link
- Fig. 4 Drive unit with non-zero-length ground link
- Fig. 5 Singularity of drive unit with non-zero-length ground link. (a) collinear of CG and GE , (b) straightening collinear, and (c) overlapping collinear
- Fig. 6 Design of output unit. (a) Case 1: bevel gear constraint, and (b) Case 2: planar branch constraint
- Fig. 7 Symmetrical five-bar spherical unit
- Fig. 8 Different configurations of planar constrain branch. (a) RRR branch, (b) RPR branch, (c) PRR branch, (d) RRP branch, (e) PPR branch, (f) PRP branch, and (g) RPP branch
- Fig. 9 Boundaries on both sides of the workspace
- Fig. 10 Variations of $\bar{\eta}$, $\tilde{\eta}$ and η vs. spherical radius r and the number of units n , where different color curves correspond to different n values.
- Fig. 11 Variations of η , s and ξ vs. the number of units n
- Fig. 12 Prototype of the proposed RCM mechanism. (a) composition of prototype, and (b) experimental measurement setup
- Fig. 13 The results of repeated positioning accuracy
- Fig. 14 Workspace division of the developed RCM prototype
- Fig. 15 The position error of the RCM point

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Table Caption List

Table 1	Optimized structural parameters
Table 2	Parameters and variables of the prototype

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