

Citation

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1 Stress Wave Mitigation Properties of Dual-meta Panels against Blast Loads

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3 Abstract

4 A dual-meta panel functioning as a sacrificial cladding is proposed and its blast mitigation
5 capacity is investigated in this study. The proposed panel possesses the potential to generate
6 bandgaps that target at a specific range of frequencies to stop stress wave propagating through
7 the panel, leading to the favourable stress wave mitigation for structural protection. Aside from
8 the unique stress wave manipulation capability, more energy can be absorbed by a combination
9 of plastic deformation and local resonance. The effectiveness of the proposed panel is validated
10 through numerical simulations. An analytical solution of wave propagation in an ideal meta
11 truss bar is derived to validate the numerical model with good agreement. It is found that the
12 proposed dual-meta panel exhibits an increase in energy absorption, a reduction in transmitted
13 reaction force (up to 30%), and the back plate central displacements (up to 20%) compared to
14 other conventional sandwich panels, e.g. sandwich panel with hollow trusses and solid trusses,
15 in resisting blast loadings. In pursuit of optimizing the performance of the proposed panel,
16 parametric investigations are also conducted to examine the influences of the plate thickness,

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17 boundary condition, and the blast load profiles including duration and intensity on the transient
18 response of the proposed dual-meta panel.

19 **Keywords:** Metastructure; Dual-meta panel; Sacrificial sandwich panel; Stress wave
20 mitigation; Wave manipulation; Dual-resonator; Blast-resistant structures; Blast loading.

21 **1. Introduction**

22 With the increasing risk of extreme incidents (e.g. explosive and ballistic attacks) worldwide,
23 there are escalating demands for more robust protective structures. Whereas solid monolithic
24 structures [1] and porous materials [2] are currently popular candidates for protective structures,
25 the underlying drivers to refrain from using these types of structures are that they are likely to
26 be cumbersome and bulky. It is, therefore, essential to promote and apply sandwich panels for
27 blast-resistants [3, 4]. The use of sandwich panels attached to main structures as sacrificial
28 cladding were investigated by many researchers, e.g., Hanssen et al. [5]. The role of the panel
29 is to deform in such a way that it absorbs energies from the incident loadings, therefore,
30 minimizing transmitted energy to the protected structure.

31 Sandwich panels consist of two plates referred as front and back plates separated by a core,
32 have provided promising solutions for energy absorptions from blast loadings. The core
33 comprises materials categorized as cellular foam or lattice type while the plates are often made
34 of thin metals or composite laminates. Generally, sandwich panels can be classified into two
35 categories by the core topology including cellular material cores, e.g. foams [6], kirigami folded
36 [7], honeycombs [8]; and periodic lattice cores, e.g. tetrahedral hollow trusses [9], pyramidal
37 solid trusses [10]. Their means of energy absorption to mitigate dynamic damage rely
38 significantly on plastic deformation mechanisms [7, 11, 12]. For instance, substantial energy
39 can be absorbed by aluminium foams through plastic dissipation [6], thus demonstrating
40 promising potentials against blast loadings. Recently, indebted to the proliferation of the
41 fabrication technology, the core topology developments of sacrificial cladding in blast resistant
42 structures has attracted many researchers. The experimental investigation was carried out to
43 examine the response of the sandwich panel with layered pyramidal truss cores subjected to
44 blast loadings by Wadley et al. [13]. The blast and impact resistance of the sandwich panels
45 was comprehensively presented in a review by Yuen et al. [14]. It was found that the sandwich

46 panels outperform solid plates of the same material and the same mass [15], indicating the
47 significant advantages of the sandwich panels over monolithic plates in blast-resistant
48 functions.

49 Apart from solely applying the deformation mechanism for blast loading effect mitigation,
50 researchers have approached the problem differently by filtering blast-induced stress wave
51 using the localized resonance mechanism, thus resulting in the loading mitigation. These
52 structures are called metastructures [16], in which the prefix “meta” comes from the Greek
53 preposition and means “beyond”, indicating that the characteristics of these structures are
54 beyond what can be seen in nature [17]. The primary concept of these structures is to utilize
55 artificially designed and fabricated structural units to achieve the designed properties and
56 functionalities. In 2016, Li and Tan [18] proposed a meta-lattice truss which is a proportional
57 elastic wave filter based on the unique local resonance of elastic metamaterial to achieve an
58 asymmetric low-frequency bandgap. Subsequently, Li et al. [19] proposed meta-lattice
59 sandwich panels with single-resonators, which show the impact/blast attenuation and higher
60 energy absorption owing to the local resonance of the internal resonator with soft coating.
61 Regarding the dynamic resistance, sandwich structures with lattice cores show better
62 performance compared to the conventional honeycomb sandwich structures [20, 21]. Besides,
63 the application of the metamaterial concept for blast protection has also been found in
64 developing metaconcrete by Jin et al. [22], and Xu et al. [23]. However, despite all these recent
65 works, the relevant research to the application of metamaterials for wave manipulation is still
66 very limited, especially on the comprehensive investigations of the performance of
67 metastructures under blast loadings. Therefore, further studies on this topic are deemed
68 necessary.

69 This study proposes a new meta sandwich panel with dual-cores as a sacrificial cladding (Fig.
70 1) by adopting the coupled mechanisms of absorbing strain energy through plastic deformation

71 and local resonance. In this study, the transient responses of the proposed panel against blast
72 loadings are investigated by numerical simulation utilizing LS-DYNA. The bandgap frequency
73 ranges obtained from the numerical simulation are compared with the analytical solution for
74 model validation. For comparison, the responses of the conventional panels – namely solid and
75 hollow truss panel are also simulated to evaluate their blast mitigation capacity compared to the
76 proposed panel. The central displacements of the plates, peak reaction forces, and energy
77 absorption are utilized to assess the performance of panels with different configurations.
78 Parametric studies on the proposed panel are also performed to examine the effects of plate
79 thickness, boundary condition, and blast loading profiles on its transient responses. The results
80 prove that the proposed dual-meta panel possesses superior characteristics that enhance its
81 protective effectiveness against blast loadings compared to its conventional counterparts.
82 Although the physical phenomenon for mitigating effect of the dual-meta panel under blast
83 loading were clearly demonstrated in this study through numerical and analytical analysis, the
84 results also lead to a number of interesting observations, some of which may pave the way for
85 future work through experimental study to comprehensively understand and demonstrate the
86 performance of the dual-meta panel.

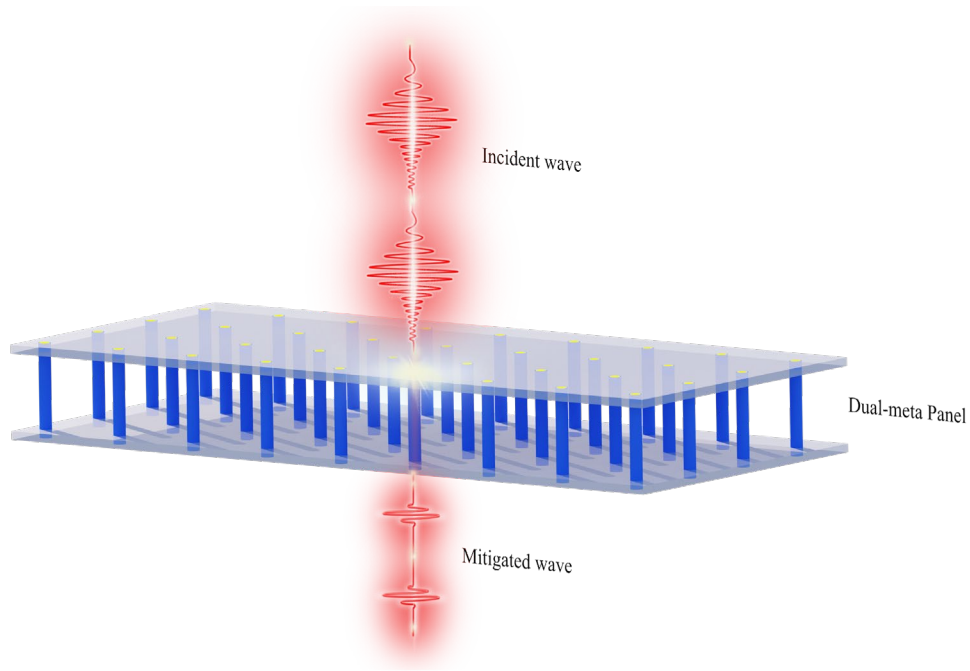


Fig. 1. Schematic diagram of the Dual-meta panel.

87 **2. Design of the Dual-meta panel**

88 Without loss of generality, the proposed dual-meta panel consisting of two thin skins is
 89 connected to the meta-lattice trusses as shown in Fig. 2(a). The meta-lattice truss element
 90 considered in this study comprises 7 unit cells (Fig. 2(c)). Each unit cell has five parts including
 91 the outer tube, two soft coats, and two resonators as shown in Fig. 2(b). The compositions and
 92 dimensions of each unit cell are presented in Figs. 2(b) and 2(c), respectively. Aluminium and
 93 lead are respectively selected for the tube and the resonators, while the two soft coatings are
 94 made from Polyurethane (PU) which can deform elastically to large strain. The two plates are
 95 also made of aluminium and connected rigidly to the trusses. As a sacrificial cladding, the
 96 perimeter of the back plate is clamped whereas there is no boundary condition imposed on the
 97 front plate. All material properties are summarized in Tables 1 and also used in the numerical
 98 model in this study.

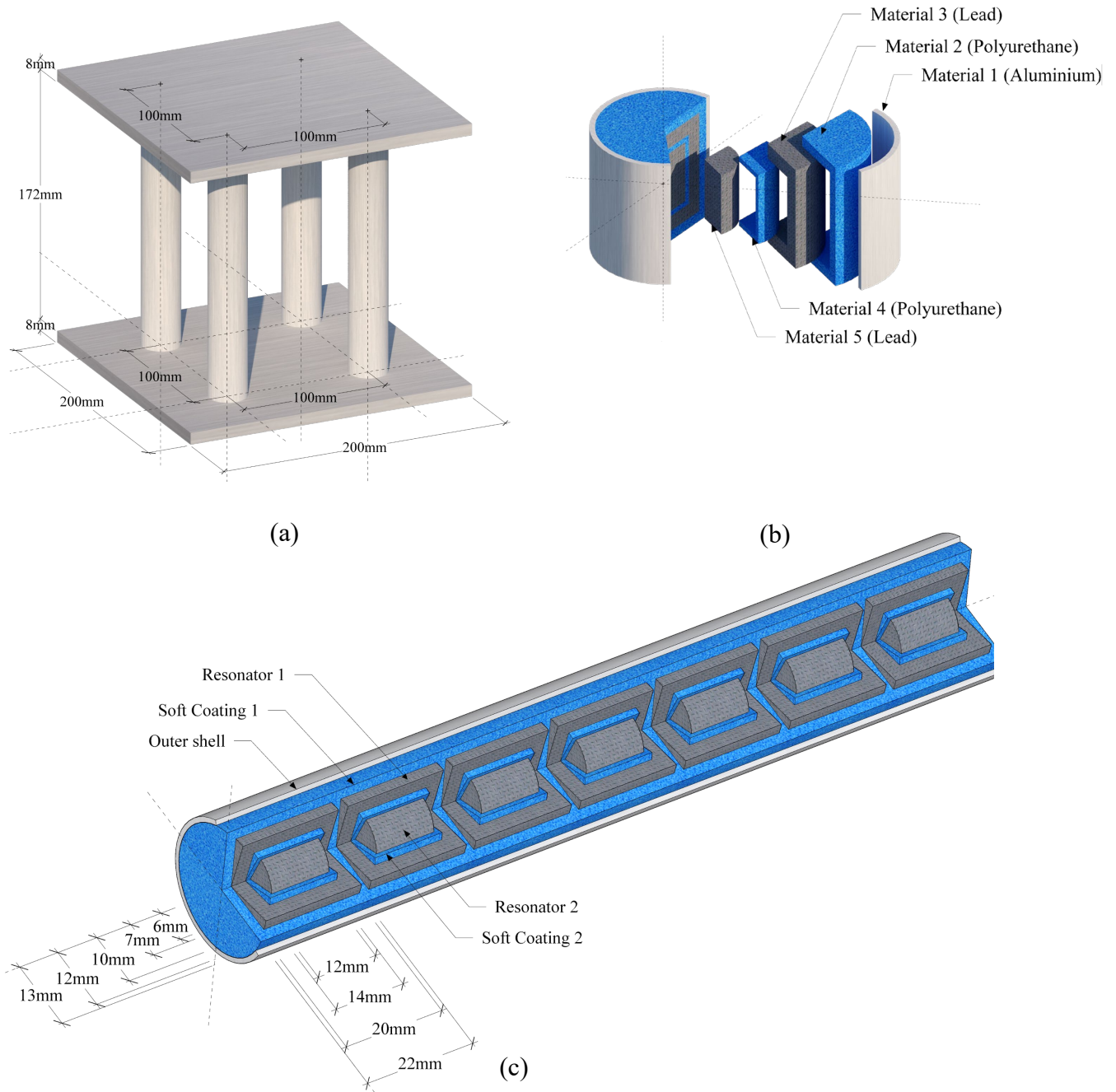


Fig. 2. (a) Schematic view of the dual-meta panel, (b) Unit cell, and c) Meta-lattice truss.

99

Table 1. Elastic material properties used in the numerical simulation [19], [22]

Properties	Material 1	Materials 2 & 4	Materials 3&5
	Aluminium	Polyurethane	Lead
Density ρ (kg/m ³)	2770	900	11400

Young's modulus E (Pa)	70×10^9	1.47×10^8	16×10^9
Poisson's ratio ν	0.33	0.42	0.44

100 **3. Analytical method**

101 A mass-in-mass model can be utilized to analytically describe the diatomic unit cell as depicted
 102 in Fig. 3 (a). In the model, the matrix is represented by material 1, i.e., the aluminium truss bar
 103 while the two masses of m_1 and m_2 represent the external and internal masses made of material
 104 3 and material 5, respectively. The outer soft coating made of material 2 is modelled by two
 105 springs including the outside shear spring k_2 connecting the resonator with the outer truss bar
 106 and the axial spring k_1 connecting the adjacent resonators. Similarly, two springs k_3 and k_4 are
 107 respectively introduced to describe the axial and shear springs of material 4 connecting the
 108 internal mass and external mass.

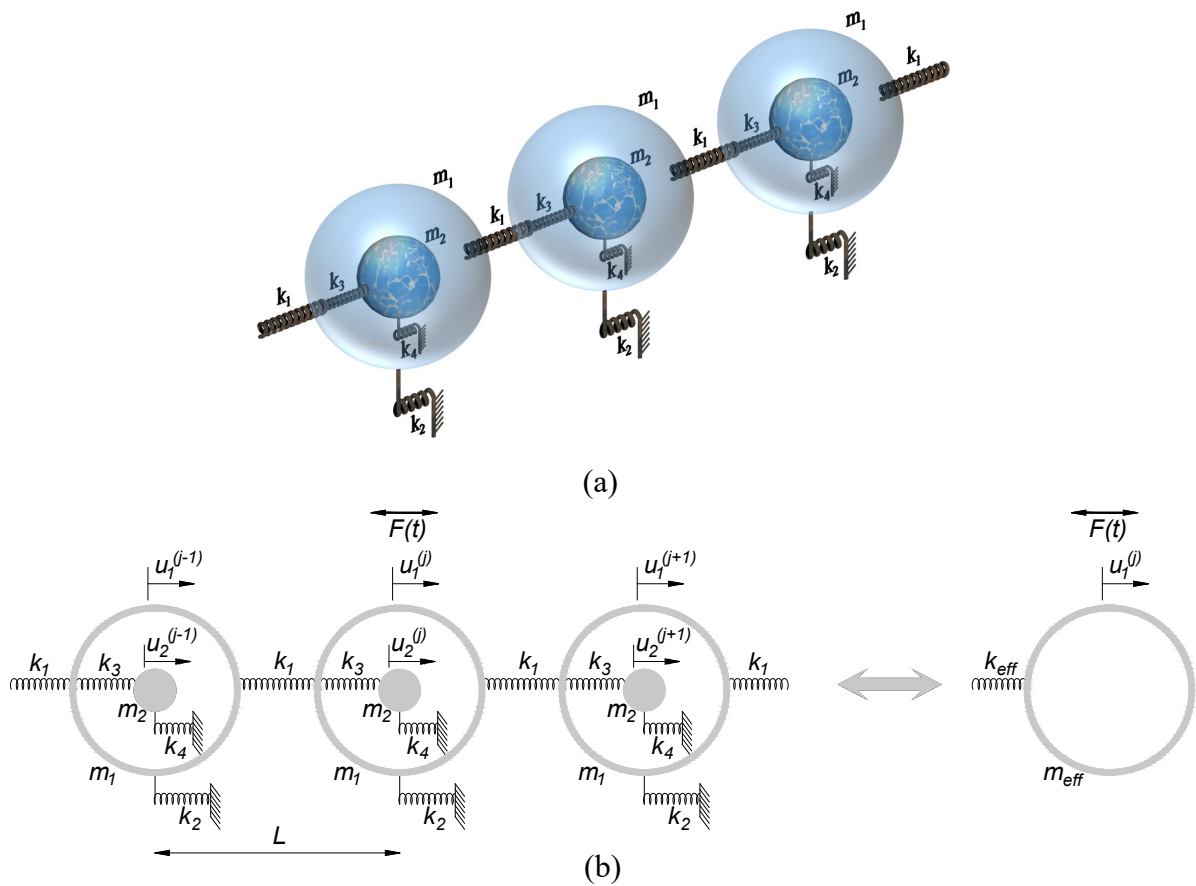


Fig. 3. (a) Schematic microstructure of infinite dual-core metamaterials, (b) Equivalent effective mass-spring model.

109 The approximate values of the inner mass and outer mass can be calculated as

$$m_\alpha = \rho_\alpha V_\alpha = \rho_\alpha \pi r_\alpha^2 l_\alpha \quad \alpha = 1, 2 \quad (1)$$

110 where ρ_α and V_α are the material density and volume of the α^{th} material while the length and
 111 radius of the α^{th} unit are denoted by l_α and r_α , respectively.

112 Besides, the stiffness of equivalent spring can be calculated as follows:

$$k_1 = \frac{E_3 A_1}{l_2}, \quad k_2 = \frac{G_3 A_2}{l_1}, \quad G_3 = \frac{E_3}{2(1+\nu_3)} \quad (2)$$

$$k_3 = \frac{E_3 A_3}{l_3}, \quad k_4 = \frac{G_3 A_4}{l_4}$$

113 where Young's modulus and shear modulus of the soft material are denoted as E and G ,
 114 respectively. Due to the shape complexity, the nominal cross-sections of the distinct segments
 115 of the soft layer A_i ($i=1,2,3,4$) presented in the appendix are obtained by FEA. The relevant
 116 estimations of the equivalent mass and stiffness are computed as $m_1 = 4.71 \times 10^{-2}$ kg, $m_2 =$
 117 1.55×10^{-2} kg, $k_1 = 57,375$ kN/m, $k_2 = 35,498$ kN/m, $k_3 = 40,760$ kN/m, and $k_4 = 24,802$ kN/m.

118 For an infinite lattice system in which u_1 and u_2 represent the internal and external mass
 119 displacements (as shown in Fig. 3(b)). The equations of motion for the j^{th} unit cell can be
 120 expressed as:

$$m_1 \frac{d^2 u_1^{(j)}}{dt^2} + k_1 (2u_1^{(j)} - u_1^{(j+1)} - u_1^{(j-1)}) + k_3 (u_1^{(j)} - u_2^{(j)}) + k_2 u_1^{(j)} = 0 \quad (3)$$

$$m_2 \frac{d^2 u_2^{(j)}}{dt^2} + k_3 (u_2^{(j)} - u_1^{(j)}) + k_4 u_2^{(j)} = 0 \quad (4)$$

121 For harmonic wave solution based on the theory of Floquet-Bloch [24], the displacement of the
 122 j^{th} unit cells are given as follows:

$$\mathbf{u}^{(j)} = Ue^{i(jqL - \omega t)} \quad (5)$$

123 where the displacement amplitude and the wavenumber are denoted by U and q , respectively
 124 while ω is the angular frequency and L is the length of the unit cell.

125 The dispersion relation can be obtained by applying the identity $e^{-iqL} + e^{iqL} = 2\cos(qL)$ and
 126 substitute Eq. (5) into Eqs. (3) and (4) as

$$\cos qL = 1 - \frac{m_1\omega^2 - (k_2 + k_3) + \frac{k_3^2}{(k_3 + k_4) - m_2\omega^2}}{2k_1} \quad (6)$$

127 The mass-in-mass system can be simplified by a mass-spring system comprising effective mass
 128 m_{eff} connecting each other by effective stiffness k_{eff} (Fig. 3(b)). Based on dispersion relation
 129 derivation from Eq. (6), the effective mass (m_{eff}) and effective stiffness (k_{eff}) of the equivalent
 130 system can be derived as [25]

$$m_{\text{eff}} = m_1 - \frac{k_2 + k_3}{\omega^2} + \frac{k_3^2}{(k_3 + k_4)\omega^2 - m_2\omega^4} \quad (7)$$

$$k_{\text{eff}} = k_1 + \frac{1}{4}(k_2 + k_3) - \frac{1}{4} \left(m_1\omega^2 + \frac{k_3^2}{(k_3 + k_4) - m_2\omega^2} \right) \quad (8)$$

131 Applying the transmission equations of the unit cells, the transmission coefficients of the
 132 system, T , can be calculated as follows:

$$T = \left| \prod_{j=1}^N \frac{\mathbf{u}^{(j)}}{\mathbf{u}^{(j-1)}} \right| = \left| \prod_{j=1}^N T^{(j)} \right| \quad (9)$$

133 where the wave transmission of the j^{th} and N^{th} unit cells can be expressed as

$$T^{(j)} = \frac{k_1}{k_1(2 - T^{(j+1)}) - \omega^2 m_{eff}}, \quad j \in [1, N-1] \quad (10)$$

$$T^{(N)} = \frac{k_1}{k_1 - \omega^2 m_{eff}}, \quad j=N \quad (11)$$

134 Based on Eqs. (7), (8), and (9), the analytical dispersion curve of the meta-lattice truss can be
 135 calculated and it is depicted in Fig. 4(a), while Figs. 4(b) and 4(c) show the corresponding
 136 effective mass and effective stiffness with respect to frequencies, respectively. It is evident that
 137 the first and the third bandgaps which are at [0-5] kHz and [13.5-50] kHz are independently
 138 formed when the effective mass and the effective stiffness become negative, respectively (see
 139 Figs. 4(b) and (c)). Whereas the negativity of both of them collaboratively constitutes the
 140 second bandgap which is at [9.3-11.5] kHz (Figs. 4(b) and (c)). It is worth mentioning that the
 141 interested frequency range in this study is only up to 50 kHz, covering the frequency band of
 142 common blast loads acting on structures [26].

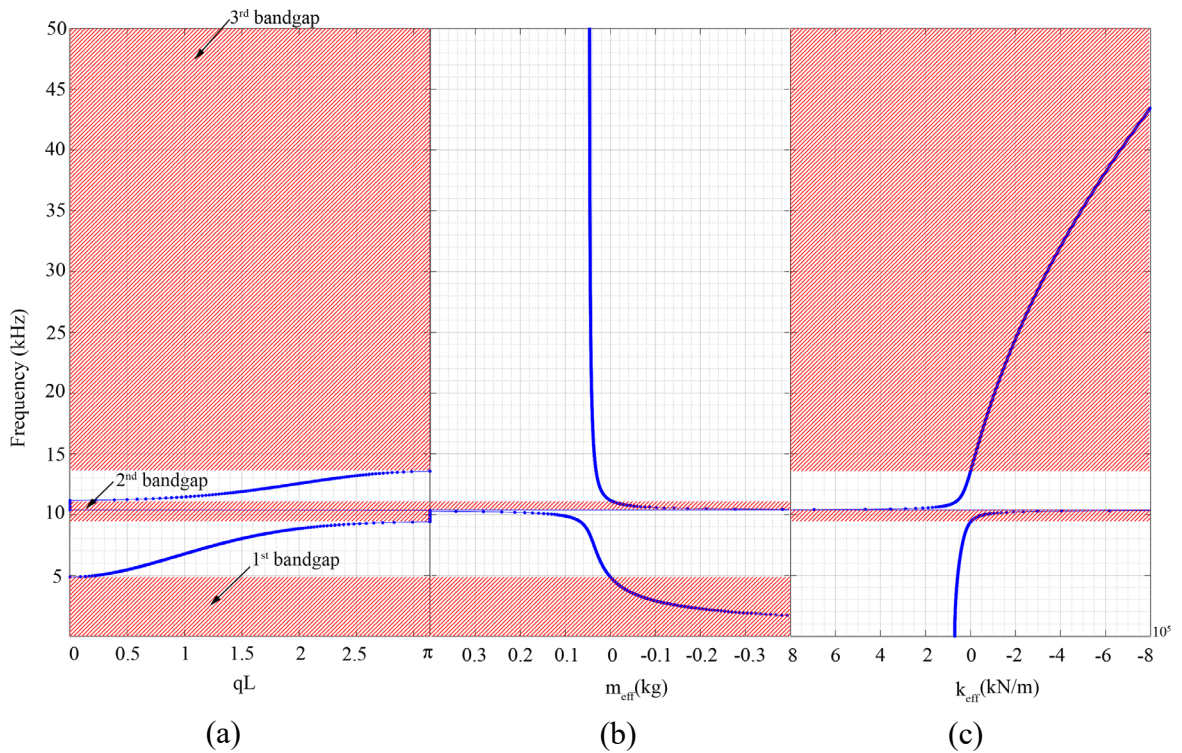


Fig. 4. Analytical solution of the bandgaps range for meta-lattice truss (a) Dispersion curve,
(b) Effective mass, and (c) Effective stiffness.

143 **3. Numerical approach**

144 Owing to the complexity, the infinite unit cells and single harmonic wave assumptions have
145 been applied to analytically solve the Eigen frequency and calculate the bandgaps. Since no
146 study of structural responses and stress wave propagations in the dual-meta panel against the
147 blast loadings has been reported yet, and it is not straightforward to derive such responses
148 analytically, especially when the combined effect of material plastic deformation and meta-
149 lattice truss bandgaps on wave energy dissipation and absorption is considered. The above
150 derivations based on idealized conditions are used in the numerical model to implicitly verify
151 the accuracy of the model. The design of the proposed dual-meta panel and its dimension were
152 presented in Section 2 and shown in Fig. 2.

153 **3.1 Model development**

154 In this study, commercial software LS-DYNA is employed to investigate the characteristics of
155 the dual-meta panel. Constitutive material models, contact definition, initial conditions, element
156 sizes, and blast load modeling are also presented in this section.

157 *3.1.1 Constitutive material models*

158 The *MAT_JOHNSON_COOK material (Mat_15) is adopted to capture the behaviour of
159 aluminium while the dynamic behaviour of polyurethane elements is simulated by
160 *MAT_ELASTIC material model due to their distinguished properties [19]. The elastic and
161 plastic material properties are summarized in Tables 1 and 2, respectively. To initialize the
162 thermodynamic state of the material, the Johnson-Cook material model requires an equation of
163 state [27] which is defined by the card *EOS_LINEAR_POLYNOMIAL in which the

164 pressure and initial relative volume are denoted by coefficients C0-C6 and V0, respectively and
 165 is presented in Table 3. Furthermore, for simulation of the lead cores, the model
 166 *MAT_PLASTIC_KINEMATIC is used and the material properties are given in Table 4 [28].
 167 The Johnson-Cook material model can be expressed as [29]

$$\sigma = \left[A + B(\varepsilon^p)^n \right] (1 + C \ln \dot{\varepsilon}^*) (1 - T^{*m}) \quad (22)$$

168 where the dynamic yield stress and the equivalent plastic strain are represented by σ and ε^p ,
 169 respectively while $\dot{\varepsilon}^* = \dot{\varepsilon} / \dot{\varepsilon}_0$ is the dimensionless plastic strain rate, where $\dot{\varepsilon}_0$ is a reference
 170 strain rate which is generally set to 1.0 s^{-1} . Regarding the temperature, $T^* = (T - T_r) / (T_m - T_r)$
 171 is defined as the homologous temperature, in which T_r and T_m are the material reference and the
 172 melting temperature, respectively. In this study, the room temperature ($T_r = 20 \text{ }^\circ\text{C}$) is applied
 173 as the reference temperature [27]. In Eq. (22), there are five material constants including the
 174 yield stress determined by the quasi-static compressive strain-stress data represented by A , the
 175 influences of strain hardening B and n , the effect of thermal softening m , and the strain rate
 176 effect which is represented by C .

177 Table 2. Johnson-cook material parameters for aluminium [19]

Density (kg/m ³)	Poisson's ratio	Young's Modulus (GPa)	A (Pa)	B (Pa)	C	m	n	T _m	$\dot{\varepsilon}_0$ (1/s)
2770	0.33	70	0.369	0.675	0.007	1.5	0.7	800	1.0

178 Table 3. Equation of state for aluminium used in the numerical simulation [27]

C ₀ (Pa)	C ₁ (Pa)	C ₂ (Pa)	C ₃ (Pa)	C ₄	C ₅	C ₆	E ₀ (Pa)	V ₀ (m ³ /m ³)
0	74.2x10 ⁹	60.5x10 ⁹	36.5x10 ⁹	1.96	0	0	0	1

179

Table 4. Plastic kinematic material parameters for lead [28]

Density (kg/m ³)	Poisson's ratio	Young's Modulus (GPa)	SIGY (MPa)	ETAN (MPa)	BETA	SRC	SRP	FS	VP (1/s)
11400	0.44	16	20	50	10 ⁹	10 ⁹	1	0	1

180 3.1.2 *Constraint and initial conditions*

181 The *BOUNDARY_SPC_SET option in LS-DYNA was adopted to account for the fully
182 clamped boundary along the perimeter of the back plate. The contact between the metals and
183 polyurethane is defined by the keyword *TIED_SURFACE_TO_SURFACE and the keyword
184 *CONTACT_INTERIOR option was utilized for polyurethane to model the slippage and
185 contact failure between materials. Besides, the contact between the outer truss bar and the two
186 plates is defined by the keyword *TIED_NODE_TO_SURFACE to make rigid connections. In
187 this study, all the elements are modeled by the solid hexahedron element (SOLID 164), and the
188 minimum meshing size of 1 mm for all elements is chosen after performing a mesh convergence
189 test, as will be detailed later.

190 3.1.3 *Blast load modeling*

191 The keyword *LOAD_BLAST_ENHANCED is widely utilized in LS-DYNA to generate blast
192 load [3, 4] via the CONWEP feature, which takes into consideration the reflection of the blast
193 wave from the surface of the panels. In this study, the blast load on the front plate of the dual-
194 meta panel which considers the enhancement of the reflected waves is defined by this function.
195 The definition of the loading area on the front plate is determined by the keyword
196 *LOAD_BLAST_SEGMENT whereas the function *DATABASE_BINARY_BLSTFOR is
197 utilized to compute the blast pressure data. The transient blast pressure on the dual-meta panel
198 is determined by the amount of Trinitrotoluene (TNT), the stand-off distance, and the angle of
199 incidence. The blast pressure is computed by the following equation [30]

$$P(t) = P_r \cos^2 \theta + P_i (1 + \cos^2 \theta - 2 \cos \theta) \quad (23)$$

200 where θ is the angle of incidence. The incident pressure and the reflected one are denoted by
 201 P_i and P_r , respectively. These peak pressures are calculated by the scaled distance, $Z = R / W^{1/3}$
 202 , in which R and W are the stand-off distance and the amount of TNT, respectively. In this study,
 203 0.15 kg TNT is detonated at a distance of 0.35 m above the front plate of the dual-meta panel,
 204 which corresponds to the scaled distance of $0.65 \text{ m/kg}^{1/3}$. The reflected pressure time history at
 205 the center point of the front plate and the corresponding FFT spectrum are illustrated in Fig. 5.
 206 As shown, the peak reflected pressure is approximately 13.5 MPa and the dominant blast
 207 loading energy distributes in the frequency band up to 50 kHz.

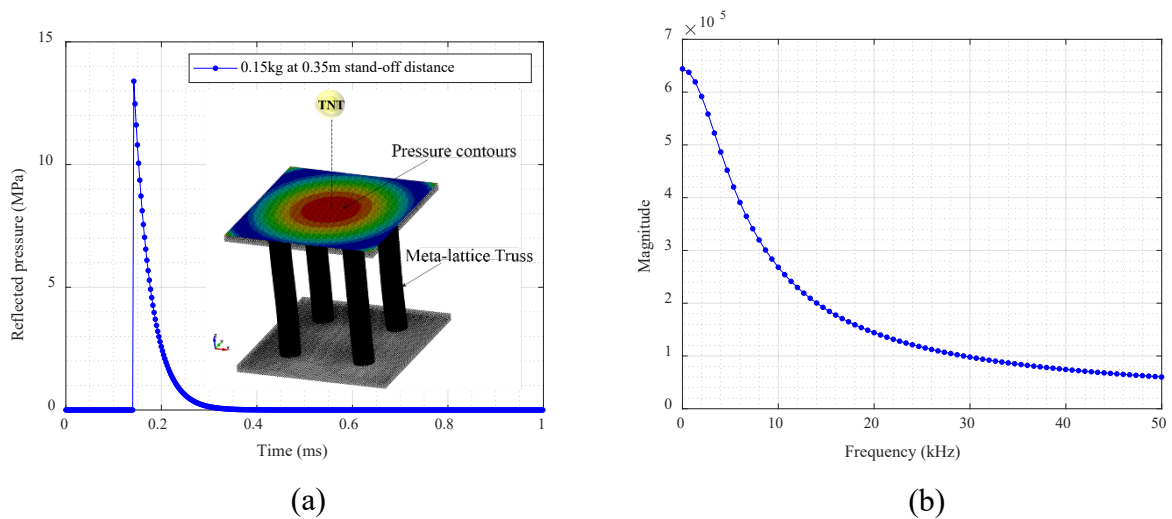


Fig. 5. Peak reflected pressure profile (a) Time history, and (b) FFT spectrum.

208 3.2 Mesh convergence test

209 A convergence test is necessary to be carried out to determine the size of elements in finite
 210 element modeling for computational accuracy and efficiency. To obtain the optimal solution,
 211 different mesh sizes comprise 3 mm, 2 mm, 1 mm, 0.5 mm, and 0.25 mm representing coarse,
 212 medium, and fine meshes are considered in the convergence test. The calculated displacements
 213 at the central point of the back plate of the dual-meta panel corresponding to the various mesh
 214 sizes are shown in Fig. 6. It can be seen from the figure that the displacement becomes

215 converged when the mesh size is 1 mm. Further reducing the mesh size does not considerably
216 affect the predicted displacement but increases significantly the computational cost. The mesh
217 size of 1 mm is, therefore, utilized in the subsequent investigations.

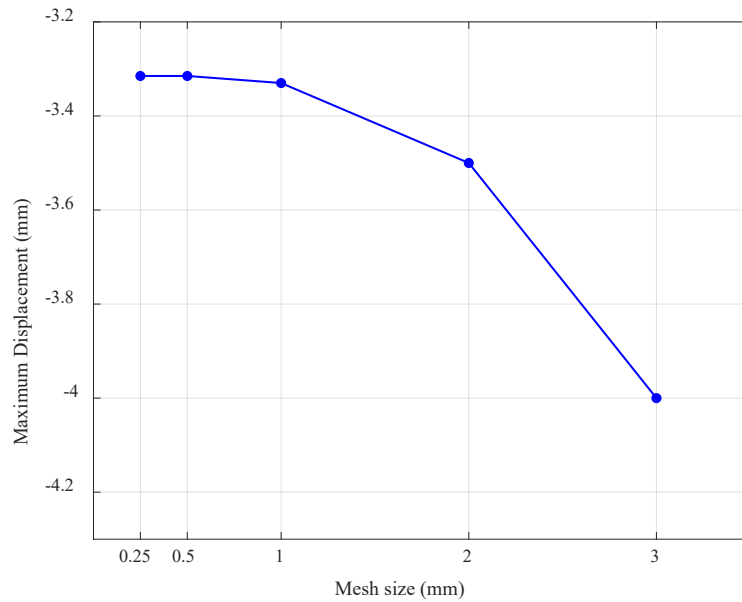


Fig. 6. Effect of mesh sensitivity on the maximum displacement of the back plate.

218 3.3 Model validation

219 To validate the numerical simulation, the transmission coefficient of a single truss bar
220 calculated by the numerical simulation is compared with that obtained by the above analytical
221 derivation. The transmission coefficient is the ratio between the output and the input signals of
222 the structure. For the numerical simulation, the input signal defined by a sweep frequency
223 ranging from 0 – 50 kHz is applied at one end of the meta-lattice truss, and the displacement
224 response at the other end is calculated to derive the transmission coefficient. The numerical
225 transmission coefficient of the meta-lattice truss is shown in Fig. 7 along with the analytical
226 result. The numerical simulation shows that the meta-lattice truss possesses three bandgaps at
227 the frequency ranges of [0-5] kHz for the 1st bandgap, [8.1-11.8] kHz for the 2nd bandgap, and
228 [13.3-50] kHz for the 3rd bandgap, while the corresponding ranges from the analytical solution

229 are [0-5] kHz, [9.3-11.5] kHz, and [13.5-50] kHz as presented above. These results indicate that
 230 the numerical results agree closely with the theoretical transmission coefficient, implying the
 231 validity of the model. The slight variations in the bandgaps between the analytical and the
 232 oscillations of the numerical results are because the meta-lattice truss is assumed continuous
 233 with an infinite number of unit cells connected by springs in the analytical derivation, while the
 234 numerical meta-lattice truss has a finite length with 7 unit cells only, and each component is
 235 modelled with its respective elastic material property and density instead of the lumped mass
 236 connected with idealized springs.

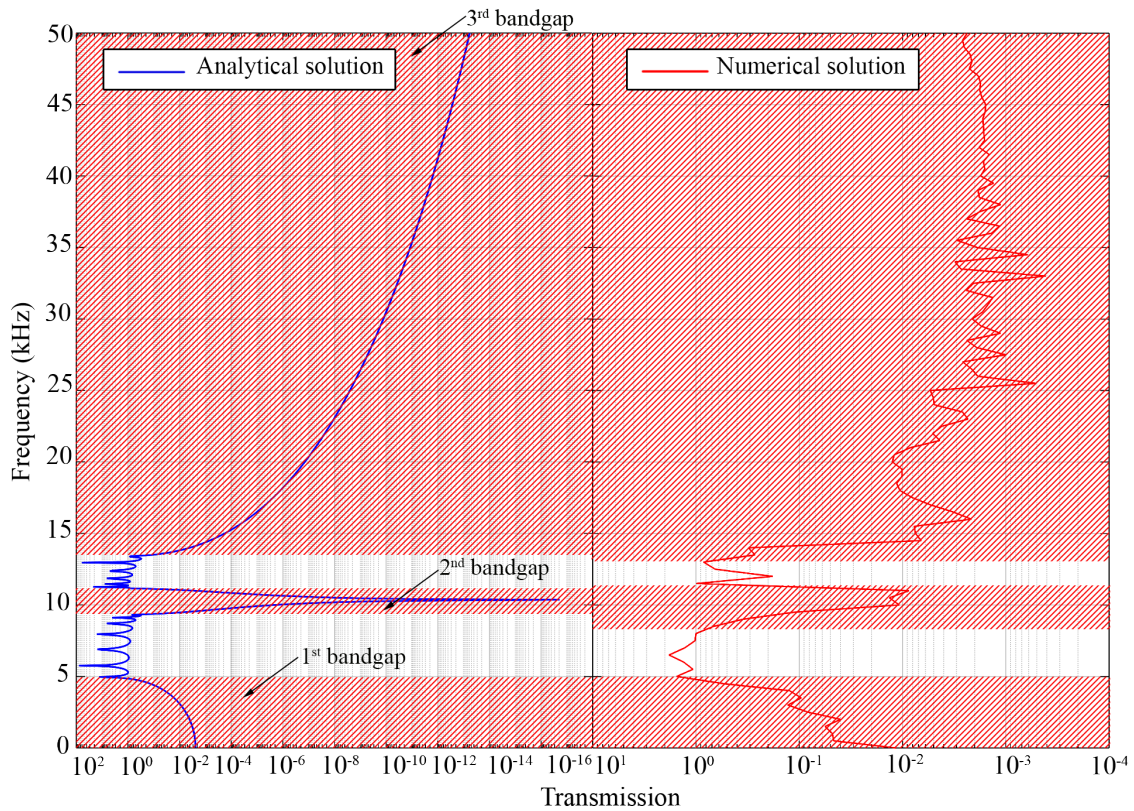


Fig. 7. Transmittance profiles of meta-lattice truss under sweep frequency input: analytical analysis vs numerical simulation.

237 To further testify the frequency suppression capacity of the meta-lattice truss, an excitation is
 238 generated by a prescribed displacement time history with multi-frequency components [31] as

$$u(t) = 10^{-4} [\sin(2\pi f_1 t) + \sin(2\pi f_2 t) + \sin(2\pi f_3 t)] H(t) \quad (12)$$

239 where the unit-step function $H(t)$ is defined as

$$H(t) = \begin{cases} 1, & t \geq 0 \\ 0, & t < 0 \end{cases} \quad (13)$$

240 and $f_1=2$ kHz, $f_2=7$ kHz, and $f_3=10$ kHz. This excitation is applied at one end of the meta-lattice
 241 truss to calculate the response at its other end. Fig. 8 shows the displacement time history at the
 242 two ends of the meta-lattice truss (i.e. the input and the output, respectively). It is worth
 243 mentioning that f_1 and f_3 are intentionally designed to fall within the first and the second
 244 bandgap, respectively, while f_2 is within its passband range. Theoretically, only the signal with
 245 f_2 can pass while other signals will be stopped by the metacores. The FFT spectrum of the input
 246 and output signal are shown in Fig. 9. As shown, only one input signal with the frequency of 7
 247 kHz can pass through the meta-lattice truss while the other two signals at frequencies 2 kHz
 248 and 10 kHz are suppressed by the meta-lattice truss.

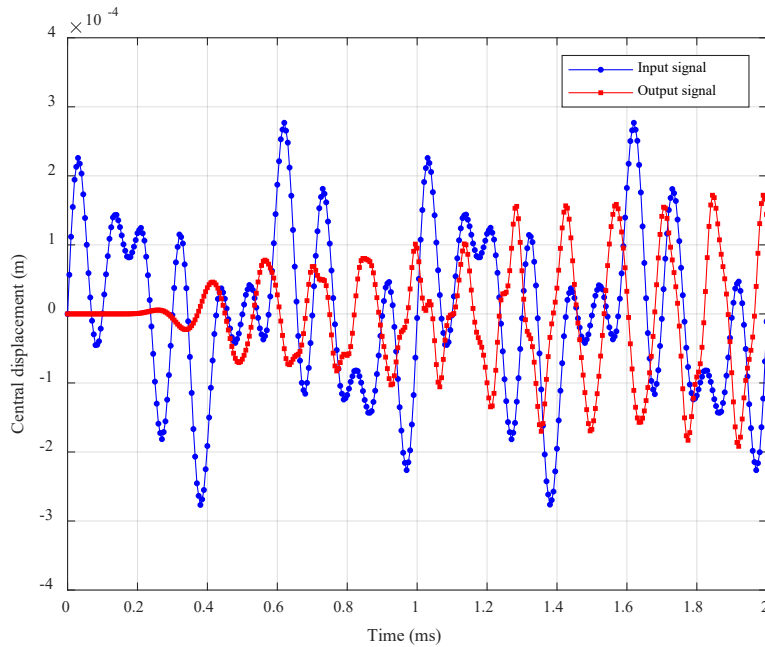


Fig. 8. Input and Output displacement time histories at the center points of two ends of the meta-lattice truss.

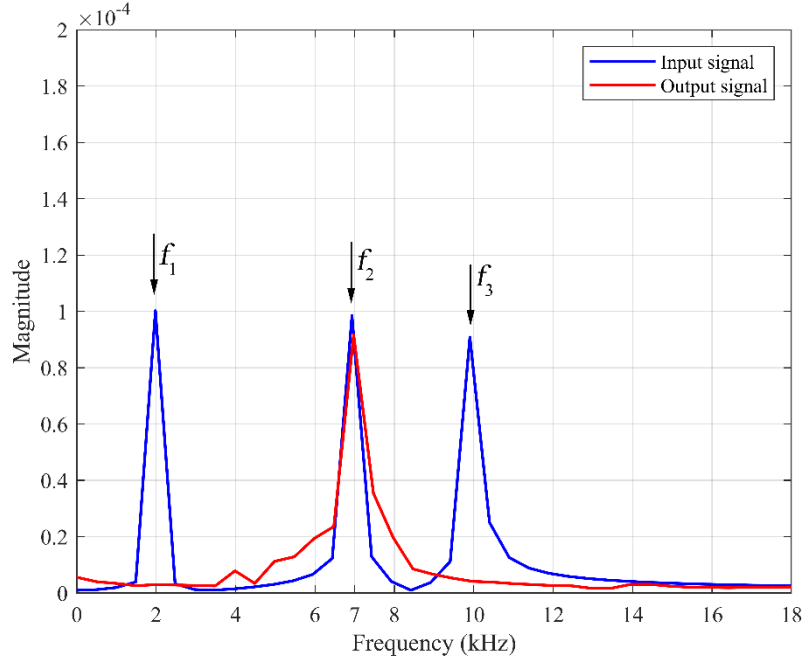


Fig. 9. The Fourier spectrum (FFT) of the input and output displacements at the center points of two ends of the meta-lattice truss.

249 3.4 Results and discussions

250 To further demonstrate the extraordinary characteristics of the dual-meta panel in resisting blast
 251 load, responses of the proposed dual-meta panel consisting of four meta-lattice truss bars
 252 (shown in Fig. 2) subjected to the blast loads defined in Fig. 5 are calculated. For comparison,
 253 two conventional sandwich panels with solid trusses and hollow trusses as shown in Fig. 10(a)
 254 and 10(b), respectively, are also modelled. As shown in Fig. 10, the two conventional sandwich
 255 panels have the same geometries and dimensions as the proposed dual-meta panel. The only
 256 difference among the three panels is the truss bars connecting the two plates. The diameter of
 257 the solid truss bar is the same as the meta-lattice truss bar, and the hollow truss bar is the same
 258 as the outer hollow tube of the meta-lattice truss bar. It should be noted that the total weight of

259 these three structures are not the same. To make the total weight of the three structures the same,
260 the size of the solid and hollow truss bars need be adjusted. Since the primary objective of this
261 study is to investigate the performance of the meta-lattice truss bar in mitigating the blast
262 loading effect, the size of the truss bars are kept the same instead of making the weight the same
263 in the analysis. It is because to keep the mass constant in the study, the thickness and/or diameter
264 of the hollow truss and the solid truss need be adjusted, which affects the stiffness of the core
265 and hence the deformation and energy absorption of the structure. Specifically, the wall of the
266 hollow truss will be thicker or its diameter larger compared to the current referenced hollow
267 truss because the mass has to be increased to match the mass of the soft coats and the lead cores.
268 Similarly, the diameter of the solid truss has to be increased because the density of the lead core
269 is higher than the aluminium tube. This would increase the stiffness of the core and decrease
270 the deformation and the energy absorption of these panels. This phenomenon can be seen from
271 the results that the hollow truss panel outperforms the solid truss panel as a sacrificial cladding
272 for blast resistance due to its higher energy absorption capacity. However, it should be noted
273 that increasing the mass enhances the inertial resistance of the structure, hence the structural
274 capacity to resist the blast load. Although the primary design targets of a sacrificial panel are
275 energy absorption and load transferred to the protected structure, instead of the loading
276 resistance capacity of the sacrificial structure itself, it would be interesting to also compare the
277 performance of the proposed dual-meta panel with reference panels having the same mass.
278 Nonetheless the scenario of the three panels having the same mass is not considered in this
279 investigation, but it is believed that increasing the mass of the traditional panels with hollow
280 truss and solid truss bars would reduce their energy absorption capacity and increase the loading
281 amplitude acting on the protected structures because of the increased stiffness of the core.
282 Responses of the three panels subjected to the same blast loads, including displacement

283 response at the center point of the back plate, total energy absorption, and the boundary reaction
284 force are compared to examine the effectiveness of the proposed panel on structure protections.

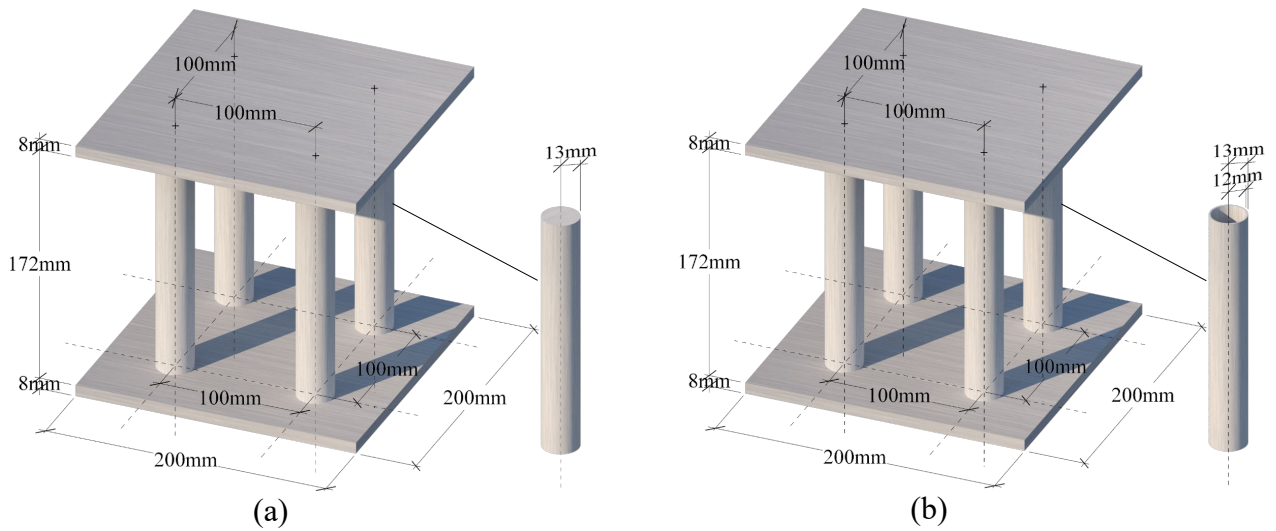


Fig. 10. Schematic view of the sandwich panels with (a) Solid trusses and (b) Hollow trusses.

285 Fig. 11 compares the displacement time histories at the center point of the back plate of the
286 three panels. It is seen that the panel with solid trusses has a higher maximum displacement (i.e.
287 4 mm), followed by the panel with the hollow trusses (i.e. 3.81 mm). The corresponding
288 maximum displacement of the dual-meta panel is 3.36 mm, i.e., 13.5%, and 20.0% lower than
289 that of the hollow truss panel and solid truss panel, respectively. It is also noted that there is a
290 substantial reduction in the second negative peak displacement in comparison between the dual-
291 meta panel with the panels with solid truss (i.e. 40.9%) and hollow truss (i.e. 52.0%). These can
292 be attributed to the fact that the effect of metacores results in lower impulse transfer to the lower
293 plate of the panel. Placing the metacores inside the truss bars of the panel results in a
294 considerable reduction of the maximum peak central displacement of the back plate compared
295 to the conventional panel, indicating the dual-meta panel has better protective performances.

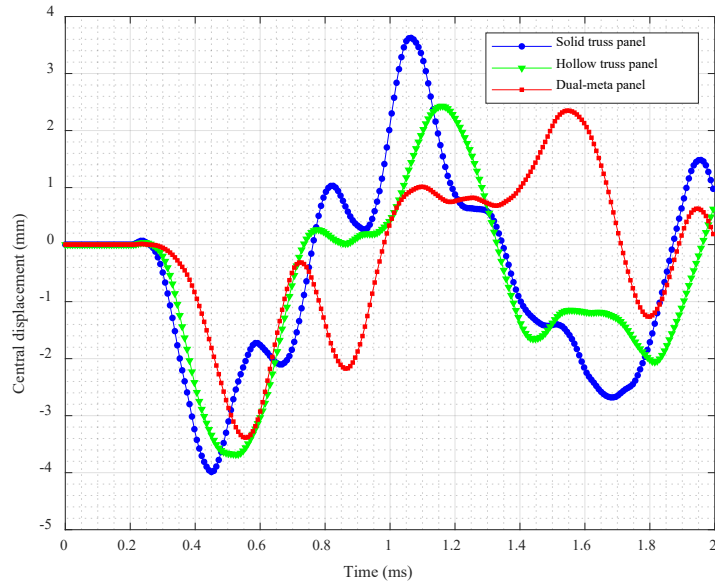


Fig. 11. Time histories of central point displacement of the back plate of the three panels.

296 To gain a comprehensive insight into blast response mitigation, investigations on the energy
 297 absorption of the dual-meta panel are carried out. The total energy (E_t), the kinetic energy (E_k),
 298 and the internal energy (E_i) absorbed by each component of the dual-meta panel subjected to
 299 blast loading are shown in Figs. 12(a), (b), (c), respectively. It should be noted that, since the
 300 energy fluctuates in the time histories, the estimated energy in this study is its mean values. It
 301 is observed that the amount of energy absorbed by the metacores and the soft coating is
 302 generally higher than that of the outer hollow truss bars. The energy absorption by the hollow
 303 truss bars is mainly associated with its plastic deformation, while the energy absorption by the
 304 metacores and soft coatings is primarily caused by local vibrations of the cores. These results
 305 indicate the damage to the truss bars by the blast load is reduced because of the local vibrations
 306 of the metacores. As shown, the outer hollow tubes of the metatruss bars experience plastic
 307 deformation which also consumes energy imparted to the panel (Fig. 12 (c)), whereas the
 308 relative movement of the metacores contributes mainly to kinetic energy (Fig. 12 (b)) and partly
 309 to the internal energy due to the deformation of the coatings. These results indicate the dual-

310 meta panel possesses the high energy absorption capability through the local vibration of the
 311 metacores, which not only protects the back plate of the sandwich structure but also reduces the
 312 damage to the outer hollow tube of the metatruss bars. Fig. 12 (d) shows the movements of each
 313 component in the panel including the two plates, the outer tube, the soft coatings, and the cores.
 314 It is worth noting that there are out-of-phase motions between the metacores and the outer tube
 315 due to the existence of the soft coatings, which effectively mitigate the blast loading effect on
 316 the back plate.

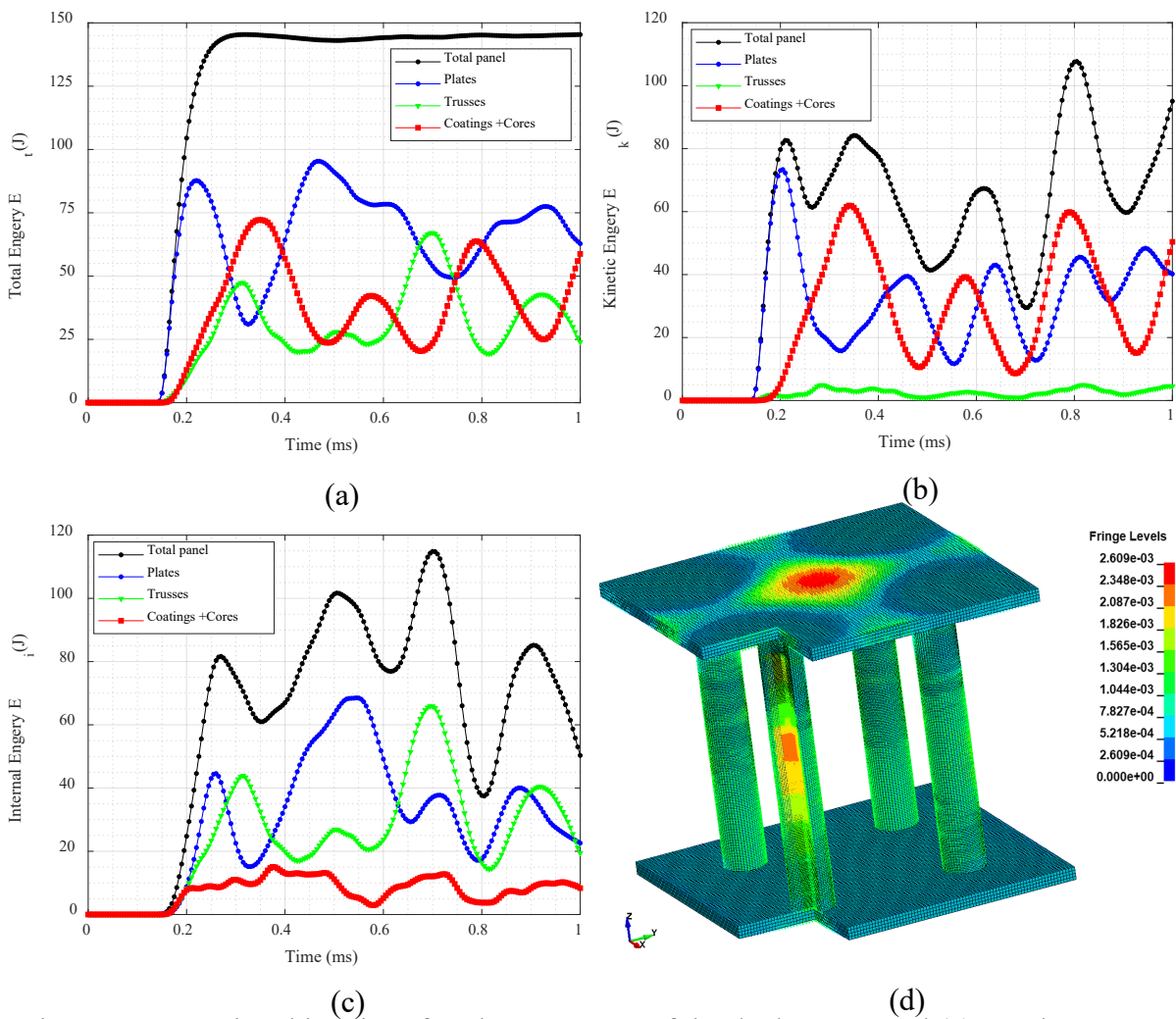


Fig. 12. Energy time histories of each component of the dual-meta panel (a) Total energy, (b) Kinetic energy, (c) Internal energy, and (d) Displacement contour of each component of the dual-meta panel.

317 For comparison, the energy absorptions of the two reference panels are also calculated. The
 318 total energy absorption of the whole panel and each component of the three panels are shown
 319 in Fig. 13. The dual-meta panel shows the highest total energy absorption. Among the three
 320 panels, the panel with solid truss bars absorbs the least amount of energy, and in which the most
 321 energy absorption is due to the plastic deformation of the plates, indicating the least protective
 322 effectiveness. The panel with hollow truss bars absorbs energy through plastic deformation of
 323 the plates and the truss bars. The energy absorbed by the hollow truss bars is the largest
 324 compared to the solid and meta-lattice truss bars, implying the largest plastic deformation of
 325 the hollow truss bars. The energies absorbed by the plates and the truss bars of the dual-meta
 326 panel are the smallest among the three panels although the dual-meta panel absorbs more energy
 327 than the two reference panels, indicating the smallest plastic deformation and hence the
 328 mitigation of damages to plates and outer hollow truss bars. These results further demonstrate
 329 the good performance of the proposed dual-meta panel.

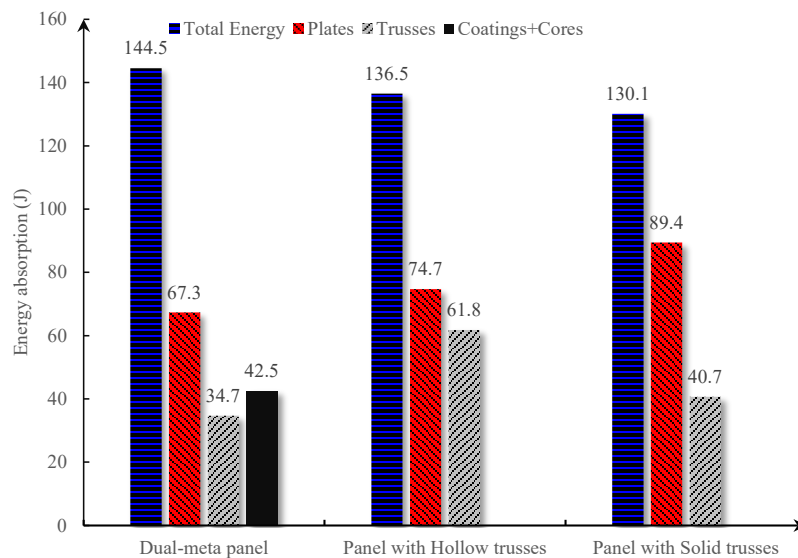


Fig. 13. Energy absorption of the three panels.

330 The reaction force in Z-direction (F_z) along the boundary of the back panel is also a key factor
 331 for the assessment of the protective effectiveness of the sacrificial panels. To evaluate this

332 factor, the reaction force time histories of the three panels are shown in Fig. 14. It is worth
 333 mentioning that the investigated reaction force is the sum of the reaction forces distributed
 334 around the boundary. As observed, the dual-meta panel is effective in reducing the reaction
 335 force of the sandwich panel. The maximum reaction force of the dual-meta panel is 18.2% and
 336 30.1% less compared to that of the hollow truss and solid truss panels, respectively. The reaction
 337 force of the dual-meta panel almost stabilizes (25 kN) after the first positive peak at 2 ms while
 338 the second positive peaks of the reaction force of the other two panels are still large (90~115
 339 kN which is comparable to the first peak). The second positive peak of the reaction force of the
 340 dual-meta panel reduces by 72% and 78% as compared to that of the panel with the hollow truss
 341 and solid truss core, respectively. This is because the metacores filter out the stress from the
 342 blast loading due to the relative movement of the metacore and the soft coating, thus less stress
 343 from the blast load is transferred to the back plate and then the supports. The reaction force at
 344 the supports, therefore, reduces which in turn relieves the demand on support designs of the
 345 sandwich panel and loading on the protected structure.

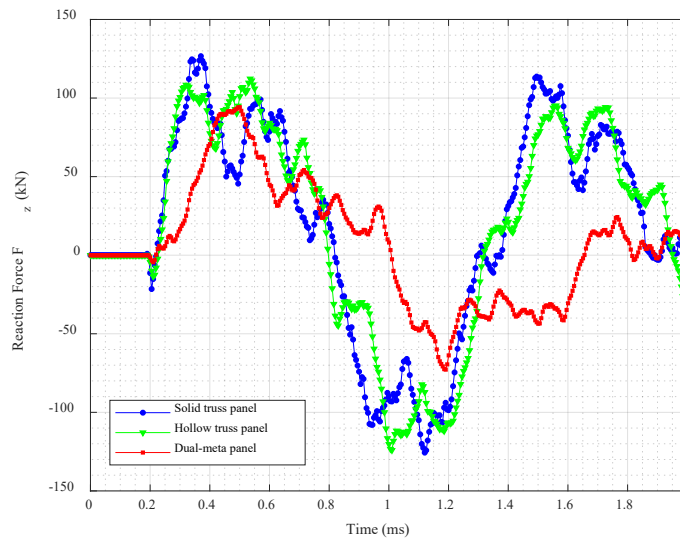


Fig. 14. Comparison of the reaction force time histories of the three panels under blast loading.

346 The effective performance of the dual-meta panel is further evaluated by analyzing the von
347 Mises stress distribution of the back plate. Fig. 15 shows the stress contours at the back plates
348 of the dual-meta panel, solid truss panel, and hollow truss panel, respectively. As shown in the
349 figure, the back plate exhibits the stress concentration at the connections between the truss bars
350 and the back plate since the blast loading generates the stress wave propagating through the
351 truss bars. The results clearly show that the von Mises stresses in the back plate of the dual-
352 meta panel is the smallest among the three panels, while that of the solid truss panel is the
353 largest, indicating again the effectiveness of the stress wave mitigation capability of the dual-
354 meta panel.

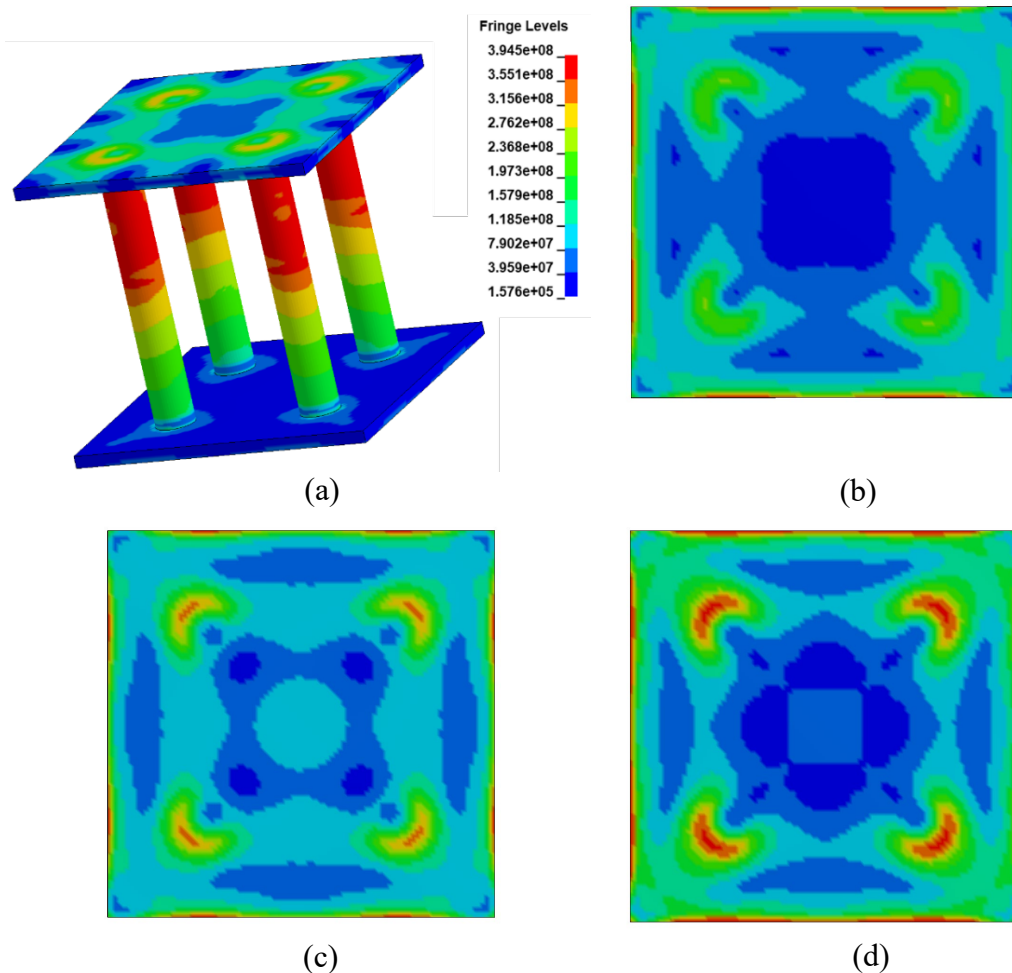


Fig. 15. (a) Stress contours of 3D dual-meta panel and stress contours at the back plate of
(b) Dual-meta panel, (c) Hollow truss panel, and (d) Solid truss panel.

355 To further compare the blast resistance of the panels with the same mass, two other conventional
356 panels with hollow truss bars and solid truss bars are also considered. The masses of these
357 panels are kept the same as that of the dual-meta panel and thus the diameter of the truss bars
358 of these three panels are different. Geometries of these panels and the blast loading are kept the
359 same as described in Section 3.4 except for the diameter of the solid truss bar and the thickness
360 of the hollow truss bar, which are adjusted to have the same mass as the meta-truss bar. The
361 solid truss bar has the radius of 25.5 mm and the hollow truss bar has the outer and inner radii
362 of 28 mm and 12 mm, respectively. The results show that the energy absorption of these panels
363 (i.e. 74.1 J and 69.5 J for panels with hollow trusses and solid trusses, respectively) are
364 significantly smaller than that of the dual-meta panel (144.5 J). Therefore, it demonstrates again
365 that the dual-meta panel outperforms the same mass conventional panels.

366 In summary, the proposed dual-meta panel reduces the maximum displacement of the back
367 plate (up to 20.0% for the first peak and 52.0% for the second peak) and the reaction forces (up
368 to 30.0% for the first peak and 78.0% for the second peak), and absorbs more energy compared
369 to the conventional panel with solid and hollow truss bars. The local vibration of the metacores
370 also reduces the stress and plastic deformation of the truss bars and the back plates of the
371 sandwich panel, therefore mitigates the damages to these components of the panel. These results
372 demonstrate the better performance of the dual-meta panel as a sacrificial cladding to resist
373 blast loading than the conventional sandwich panels with solid and hollow truss bars.

374 **3.5 Parametric investigations**

375 In this section, the influences of critical parameters such as the thickness of the plate, boundary
376 condition, blast load duration and intensity on the performance of the dual-meta panel are
377 numerically investigated. This section is carried out to gain further insights into the performance
378 of the dual-meta panel subjected to confined blast loading as a sacrificial cladding.

379 3.5.1 *Effect of the thicknesses of the plates*

380 Herein, the transient response of the dual-meta panel is examined with varying front plate
381 thickness while keeping the back plate thickness unchanged and vice versa. Three thicknesses,
382 i.e. 4 mm, 6 mm, 12 mm, are taken into consideration. Therefore, six panels with different
383 combinations of thicknesses of front plate and back plate are considered in this section including
384 4 mm (F) + 8 mm (B), 6 mm (F) + 8 mm (B), 12 mm (F) + 8 mm (B), 8 mm (F) + 4 mm (B), 8
385 mm (F) + 6 mm (B), and 8 mm (F) + 12 mm (B). Figs. 16 (a) and (b) depict the central deflection
386 of the back and front plates with varying plate thicknesses. It should be noted that the above
387 plate configurations are determined to obtain a more comprehensive and valid comparison on
388 protective effectiveness, i.e, the panels experience different levels of deformation without
389 failure. This predetermined condition also assumes these panels after deformation would not
390 touch the main structure and only transfer the load to the main structure through their supports.
391 As expected, the deflections of both the front and back plates decrease with the increase of their
392 thicknesses. Drastic reduction in displacements by increasing the plate thickness demonstrates
393 its significance in suppressing the blast loading of the dual-meta panel. It is noted that in most
394 cases, the displacement at the central point of the front plate is smaller than that of the back
395 plate due to their boundary conditions. The four edges of the back plate are restrained in all
396 directions while the edges of the front plate are free. The displacement of the overhanging
397 portion of the front plate would counteract its central point displacement resulting in a reduction
398 in the displacement amplitude. As shown in Figs. 16 (c) and (d), with an increase in the
399 thickness of the front plate from 4 mm to 12 mm, there is an increase in the reaction force and
400 a substantial reduction of the total energy absorption. This phenomenon happens mainly
401 because the less deflection of the plate means less energy absorption through its plastic
402 deformation. In brief, the reaction force is highly sensitive to the front plate thickness and it is
403 not beneficial to use a thick front plate in the design of sacrificial panels.

404 Regarding the influence of the thickness of the back plate, when changing its thickness, the first
405 peak displacement of the front plate is the same but the second peak displacement and the
406 subsequent displacement responses vary. This is attributed to the stress waves generated by the
407 blast load transfer from the front plate to the back plate. Regardless of the thickness (thus
408 stiffness) of the back plate, the front plate will be the first component to resist the blast load, its
409 first peak displacement, therefore, is not sensitive to the thickness of the back plate. However,
410 its second peak displacement is affected by the stiffness of the back plate owing to the reflected
411 stress and deformation of the back plate. It is observed that the second peak displacement occurs
412 when the panel rebounds from its first peak and it moves back in the opposite direction.
413 Meanwhile, the reaction force and the total energy absorption are not sensitive to the thickness
414 of the back plate. In brief, the displacement of the plate is highly sensitive to the stiffness of the
415 plate, due to the correlation between the stiffness and the displacement. While the thickness of
416 the back plate only affects its own displacement, reducing the front plate thickness results in
417 smaller reaction forces and absorbing more energy. From Fig. 16, it can be seen that the two
418 good combinations are 4 mm (F) + 8 mm (B) and 8 mm (F) + 4 mm (B) since they absorb more
419 energy compared to its total amount of materials. However, the combination of 8 mm (F) + 4
420 mm (B) exhibits a much higher reaction force than that of 4 mm (F) + 8 mm (B). Hence, the
421 optimal design of the dual-meta panel as a sacrificial cladding should have a fairly thin front
422 plate and a thick back plate to fully manifest its protective performance such as high energy
423 absorption and less deflection of the back plate. It is noted that the effect of the plate's
424 thicknesses on the blast mitigation of the dual-meta panel is similar to that of other blast-
425 resistant sandwich panels [3].

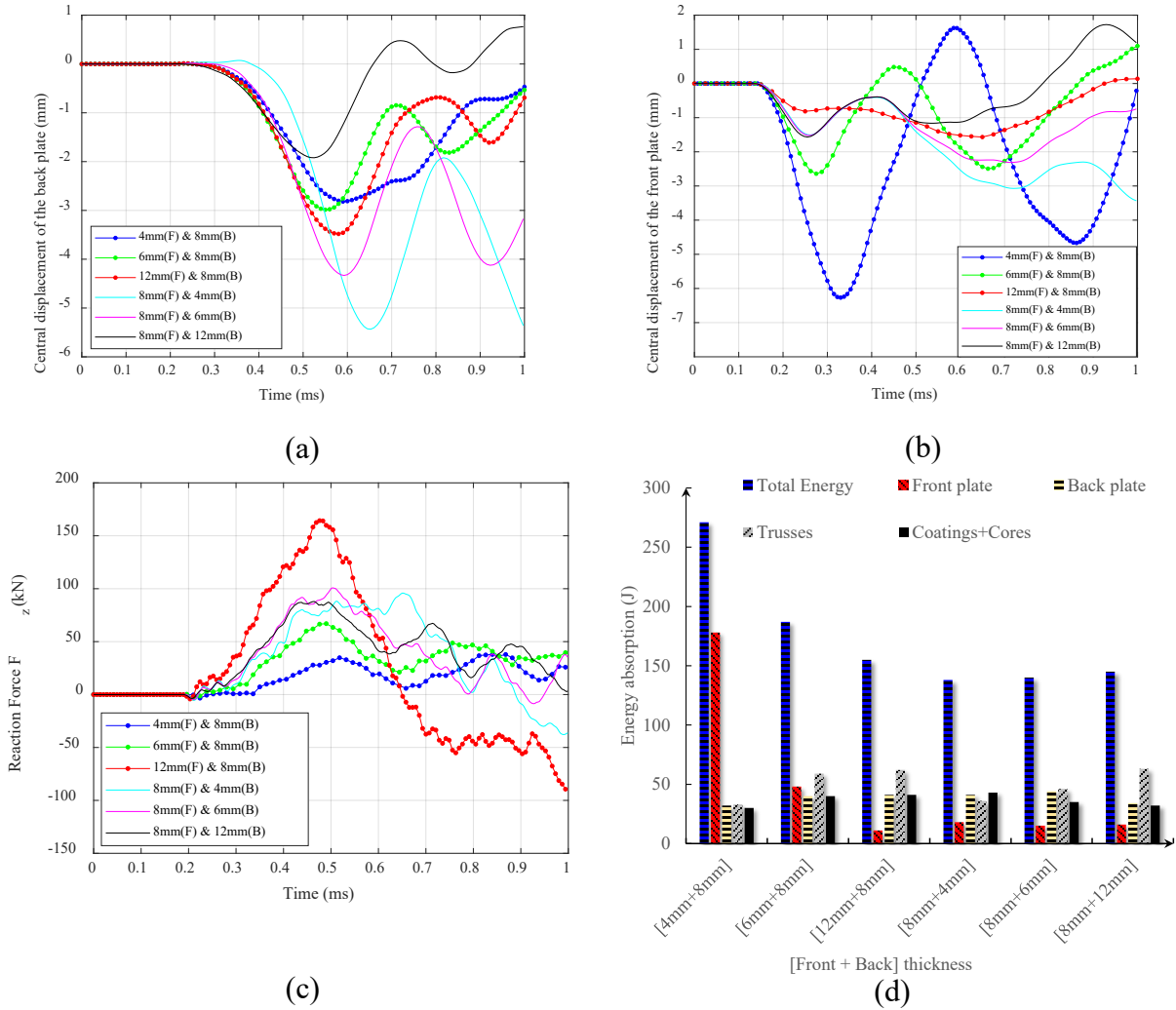


Fig. 16. Effects of the plate thickness on dual-meta panel dynamic response time histories (a) Central displacement of the back plate, (b) Central displacement of the front plate, (c) Reaction force, and (d) Total energy absorption of various components.

426 3.5.2 Effect of boundary conditions

427 The boundary condition determination relies on how sacrificial claddings can be utilized in
 428 structural protection [32]. There are various ways that the protective panel can be attached to
 429 the main structure, namely, clamped or pinned at the edges allowing some clear space between
 430 the panel and the protected structure, or directly fixed against the system without a gap. In this
 431 study, these attachment methods are considered with three boundary conditions including all
 432 perimeter is clamped, simply pinned, and all the surface of the back plate is fixed, i.e., directly

433 attaching the panel on the protected structure. The transient responses of the dual-meta panel
434 with these selected boundary conditions subjected to the same blast loading (defined in Fig. 5)
435 are compared in Fig. 17 and Table 5. As shown in Fig. 17, the central displacement of the back
436 plate of the panel with pinned boundary is 13.1% larger than that with clamped boundary
437 condition. Meanwhile, the plates and the metacores of the pinned panel witness a decrease in
438 energy absorption compared to the clamped panel by 16.7 % and 20.7 %, respectively.
439 However, the energy absorbed by the truss bars of the clamped panel is lower than that of the
440 pinned panel. Therefore, the total energy absorption of the panel with the pinned and fixed
441 boundary conditions differs by only 1 %. As for the case with the fixed back surface, the energy
442 absorbed by the trusses is higher compared to that of the other two boundary conditions because
443 the constraints of the lower plate result in more deformation of the trusses leading to more
444 energy absorption. The total energy absorption of the fully fixed panel is comparable to that of
445 the panel with clamped boundary, implying the amount of energy absorbed by the back plate
446 deformation is compensated by the larger plastic deformation of the outer tube of the truss bars.
447 The reaction force in the Z-direction of the panel with the fully fixed back surface is two times
448 higher than those of the panel with other boundary conditions, therefore, it is not recommended
449 to apply the fixed back surface in practice. Moreover, the reaction force with the clamped
450 condition is slightly higher than that of the pinned condition. In summary, the displacement and
451 the energy absorption by various parts of the panel are significantly affected by the boundary
452 conditions while it exerts less influence on the total energy absorption of the panel. This
453 conclusion is in good agreement with other blast-resistant sandwich panels such as aluminium
454 foam-cored sandwich panels [33]. For practical applications, the dual-meta panel will perform
455 better as a sacrificial cladding if there is a gap between it and the protected structure, with less
456 blast force transferred to the protected structure, but concentrated at the supports. Directly

457 attaching the panel on the protected structure also leads to larger plastic deformations of the
 458 outer tube of the truss bars, making the metacores less effective in absorbing blast energy.

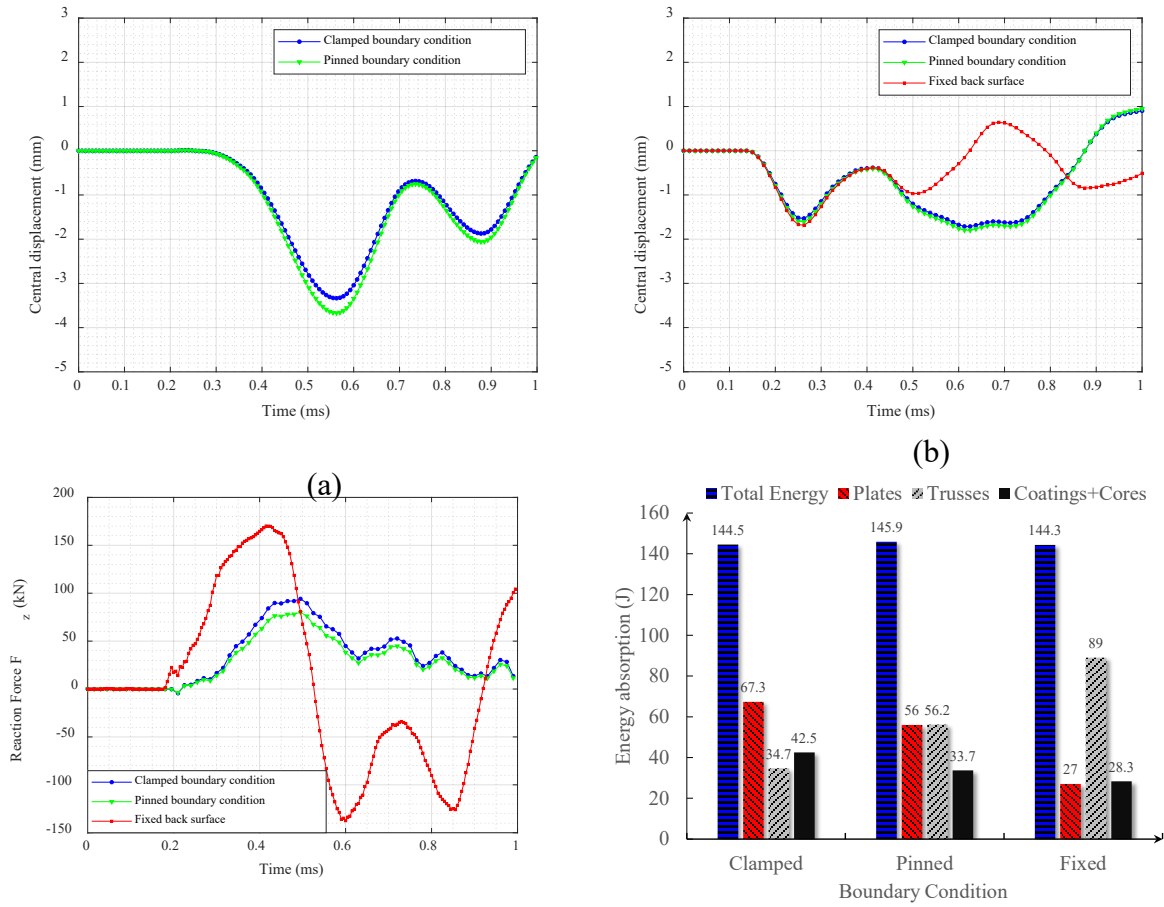


Fig. 17. Effects of the boundary conditions on dual-meta panel dynamic response time histories (a) Central displacement of the back plate, (b) Central displacement of the front plate, (c) Reaction force, and (d) Total energy absorption of various components.

459 Table 5. Effects of boundary conditions on displacements, reaction force, and energy
 460 absorptions.

Boundary condition	Displacement (mm)				Reaction force (kN)			Energy absorption (J)		
	Front plate	Reduction	Back plate	Reduction	F _z	Reduction	Plate	Trusses	Coatings + Cores	Total
Clamped	1.55	N/A	3.33	N/A	91.5	N/A	67.3	34.7	42.5	144.5
Pinned	1.57	-1%	3.81	-13.1%	82.7	10.6%	56.0	56.2	33.7	145.9
Fixed	1.74	-12%	N/A	N/A	170.1	-85.9%	27.0	89.0	28.3	144.3

461 3.5.3 *Effects of blast loading duration and intensity*

462 To comprehend the influence of different levels of blast loading duration on a given dual-meta
463 panel, four blast loading profiles (shown in Fig. 18) with different duration but the same
464 amplitude are considered. The blast loading profile is defined by using the modified
465 Friedlander's equation [34]:

$$F = F_{\max} \left(1 - \frac{t - t_0}{t_d} \right) e^{-\frac{t - t_0}{t_d}}, t_0 \leq t < t_0 + t_d \quad (14)$$

$$F = 0, t < t_0 \text{ or } t \geq t_0 + t_d$$

466 where F_{\max} is the amplitude while the time constants t_0 and t_d are the blast initial time and blast
467 duration, respectively. In this study, the negative phase in the blast loading profile is neglected
468 in the analysis [35]. For different loading regimes, durations t_d of 0.1 ms, 0.2 ms, 0.3 ms, and
469 0.4 ms are chosen in the analyses. It should be noted that the blast loading duration is purposely
470 chosen relatively short to generate a wider loading frequency band for evaluating the
471 performance of the dual-meta panel in mitigating the blast loading effect. In an explosion case,
472 such short loading duration could be associated to contact and very close-in explosions. With
473 the amplitude F_{\max} of 13.5 MPa, the corresponding impulses are 530.8 Ns, 1027.2 Ns, 1523.8
474 Ns, and 2020.4 Ns, respectively with the four different duration. It is obvious when varying the
475 duration of the blast loading, the dominant frequency band of the blast loading would change
476 accordingly. The corresponding blast loading energy in the three bandgaps of the current meta-
477 lattice trusses can be calculated by the area (A_{bandgap}) enclosed by the FFT spectrum of the blast
478 loading in each bandgap as illustrated in Fig. 18(b). The portion of the blast loading energy
479 corresponding to each bandgap is calculated by dividing the energy in each bandgap by the total
480 blast loading energy (A_{total}), and are given in Table 6. As shown, more proportion of energy
481 from blast load with longer duration falls into the bandgaps of the dual-meta panel, i.e., 77.0%,
482 81.0%, 82.4%, and 83.4%, respectively for the four considered loading cases, implying the

483 dual-meta panel is more effective in mitigating the blast loading effect from load with the
 484 longest considered duration in this study.

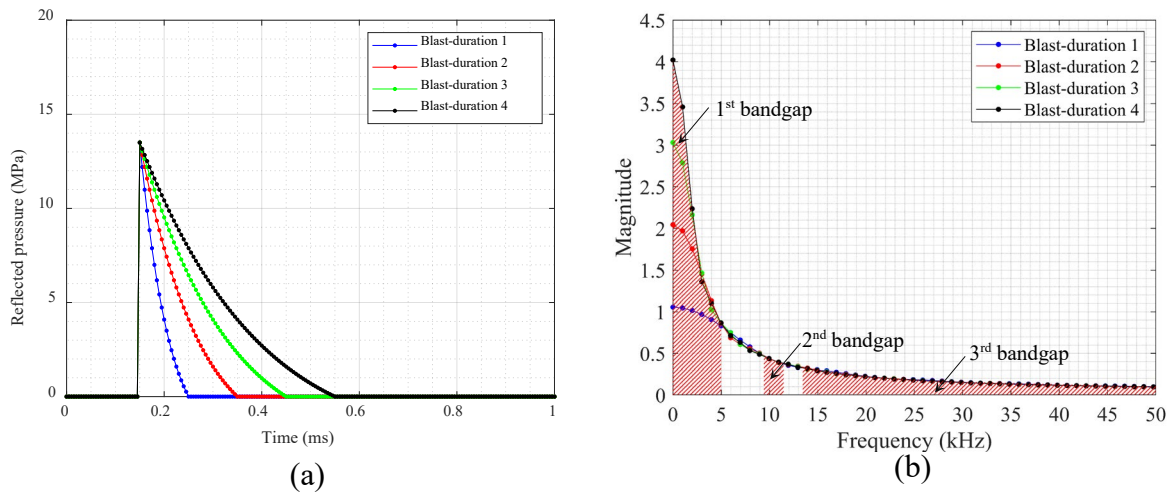


Fig. 18. Blast loading profiles with different duration, (a) Time histories, and (b) FFT spectra.

485 Table 7 gives the dynamic response of the dual-meta panel subjected to the assumed blast loads
 486 with different duration uniformly applied to the front plate of the panel. The results indicate
 487 that the dynamic responses of the dual-meta panel rely heavily on the blast loading impulse.
 488 The plate deflections, the reaction forces, and the energy absorption of the panel increase with
 489 the loading impulse. It is obvious that the effectiveness of the dual-meta panel in blast load
 490 mitigation depends on the frequency band of the blast loading, therefore, aside from the total
 491 energy absorption increases from loading case 1 to case 4 due to the increased blast loading
 492 energy imparted to the structure, the largest percentage of energy absorption by the coatings
 493 and the cores of the meta-lattice trusses corresponds to the loading case 4, which is
 494 $392/1188=32.9\%$ as shown in Table 7, followed by 31.9%, 29.6%, and 28.7%, respectively for
 495 the loading cases 3 to 1. This is because the proportion of the blast energy of the loading cases
 496 considered in the analyses reduces from case 4 to case 1, implying the meta-lattice truss can
 497 stop more blast loading energy transmission as shown in Fig. 18(b) when more proportion of
 498 the blast loading energy falls into the bandgaps. It should be noted that the percentage of energy

499 absorption of the coatings and the cores in the dual-meta panel calculated from Table 7 is
500 smaller than the corresponding values in Table 6. It is because the energy absorption of the
501 dual-meta panel is constituted by four components, i.e., the plates, the truss bars, the coatings,
502 and the cores. Only the metacores including the coatings and the cores have the bandgap-related
503 mitigating capability, while the plates and trusses absorb energy through plastic deformation.

504 Table 6. Proportion of blast loading energy with different duration falling in the bandgaps of
505 the single meta-lattice truss.

Blast loadings	1 st bandgap		2 nd bandgap		3 rd bandgap		Total %
	$\frac{A_{\text{bandgap}}}{A_{\text{total}}}$	%	$\frac{A_{\text{bandgap}}}{A_{\text{total}}}$	%	$\frac{A_{\text{bandgap}}}{A_{\text{total}}}$	%	
	Blast-duration 1	$\frac{4875}{15340}$	31.7%	$\frac{927}{15340}$	6.0%	$\frac{6039}{15340}$	
Blast-duration 2	$\frac{7746}{17976}$	43.1%	$\frac{929}{17976}$	5.1%	$\frac{5903}{17976}$	32.8%	81.0%
Blast-duration 3	$\frac{9364}{19580}$	47.8%	$\frac{929}{19580}$	4.7%	$\frac{5855}{19580}$	29.9%	82.4%
Blast-duration 4	$\frac{10570}{20740}$	50.9%	$\frac{926}{20740}$	4.4%	$\frac{5835}{20740}$	28.1%	83.4%

506 Table 7. Effect of blast loading duration on displacements, reaction force, and energy
507 absorption.

Blast loadings	Displacement (mm)		Reaction force (kN)	Energy absorption (J)			Total
	Front plate	Back plate	F _z	Plates	Trusses	Coatings + Cores	
Blast-duration 1	1.7	4.1	120	95	46	57	198
Blast-duration 2	2.6	7.3	190	253	136	164	553
Blast-duration 3	3.1	9.7	230	392	212	283	887
Blast-duration 4	3.5	11.2	270	528	268	392	1188

508 To evaluate the effectiveness of the dual-meta panel subjected to blast load with different peaks
509 but the same impulse, the responses of the dual-meta panel subjected to blast load with the
510 impulse of 530.8 Ns but varying the peak pressure and duration. The blast loading duration of

511 0.1 ms, 0.2 ms, 0.3 ms, and 0.4 ms with the corresponding peak pressure of 13.5 MPa, 6.75
 512 MPa, 4.5 MPa, 3.375 MPa, respectively are considered. Figure 19 shows the blast loading time
 513 histories and the corresponding FFT spectra of these blast loading profiles. The portions of the
 514 blast loading energy in each bandgap of the meta-lattice trusses are given in Table 8. The
 515 corresponding percentages of the blast loading energy falling into the bandgaps of the panel are
 516 77.0%, 80.9%, 82.2%, and 83.4%, respectively for Blast loading case 1-4 as shown in Table 8.

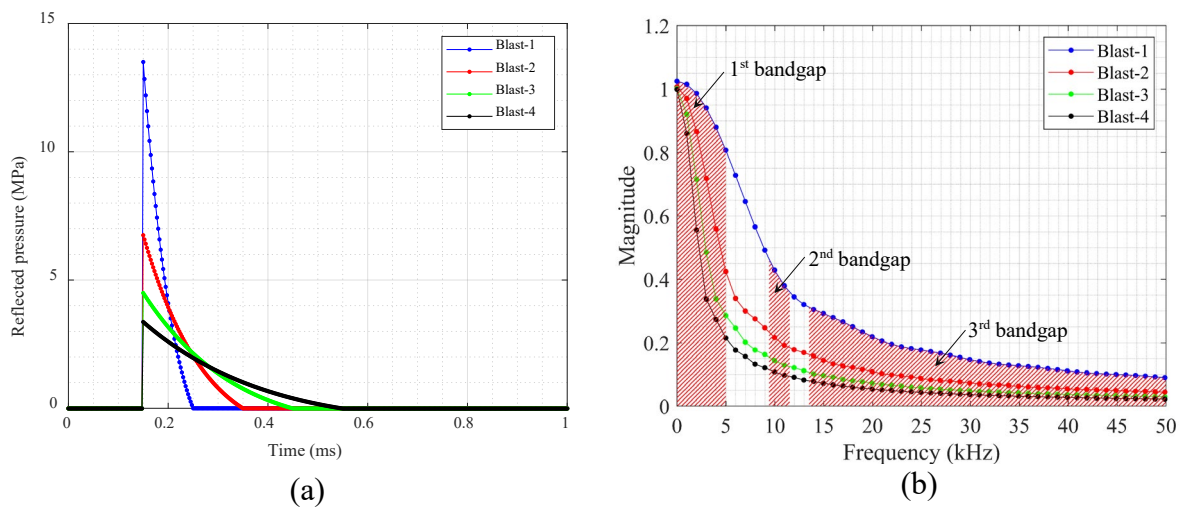


Fig. 19. Blast loading profiles with different duration and intensities (a) Time histories, and
 (b) FFT spectra.

517 As shown in Table 9, the plate displacements and energy absorption of the dual-meta panel
 518 decrease with the reduction of the peak blast load given the same impulse. Also, the highest
 519 peak reaction force corresponds to the loading Blast-1, which is reasonable since it is associated
 520 with the highest peak blast load. As given in Table 9, although the total energy absorption
 521 increases from the loading Blast-4 to Blast-1 due to the increase of the peak blast load, the
 522 largest percentage of energy absorption by the coatings and the cores corresponds to Blast-4,
 523 which is $26/76=34.2\%$, followed by 32.5%, 30.1%, and 28.7%, for Blast-3 to Blast-1,
 524 respectively. It is again attributed to the proportion of the blast loading energy falling into the
 525 bandgaps given in Table 8. These results demonstrate that the transient responses of the dual-
 526 meta panel correlate with the peak blast load and its capacity to absorb energy in the bandgap

527 ranges. Therefore, the proposed dual-meta panel can be designed to maximize its blast load
 528 mitigation efficiency for an expected blasting scenario.

529 Table 8. Proportion of blast loading energy with different duration and intensities falling in
 530 the bandgaps of the single meta-lattice truss.

Blast loadings	1 st bandgap		2 nd bandgap		3 rd bandgap		Total %
	$\frac{A_{\text{bandgap}}}{A_{\text{total}}}$	%	$\frac{A_{\text{bandgap}}}{A_{\text{total}}}$	%	$\frac{A_{\text{bandgap}}}{A_{\text{total}}}$	%	
	Blast-1	$\frac{4875}{15340}$	31.7%	$\frac{927}{15340}$	6.0%	$\frac{6039}{15340}$	
Blast-2	$\frac{3877}{9010}$	43.1%	$\frac{465}{9010}$	5.1%	$\frac{2958}{9010}$	32.8%	81.0%
Blast-3	$\frac{3125}{6543}$	47.8%	$\frac{310}{6543}$	4.7%	$\frac{1956}{6543}$	29.9%	82.4%
Blast-4	$\frac{2646}{5198}$	50.9%	$\frac{232}{5198}$	4.4%	$\frac{1462}{5198}$	28.1%	83.4%

531 Table 9. Effects of blast loading duration and intensities on displacements, reaction force, and
 532 energy absorption.

Blast loadings	Displacement (mm)		Reaction force (kN)	Energy absorption (J)			Total
	Front plate	Back plate	F_z	Plates	Trusses	Coatings + Cores	
	Blast- 1	1.7	4.1	120	95.0	46.0	
Blast- 2	1.3	3.6	109	63.2	34.0	41.8	139
Blast- 3	0.9	3.2	97	42.5	25.0	32.5	100
Blast- 4	0.8	2.8	87	30.8	19.2	26.0	76

533 4. Conclusions

534 The capability of the proposed dual-meta panel to attenuate blast loading effect is examined in
 535 this study. Theoretical derivations and numerical simulations are carried out to investigate the
 536 mechanism and responses of the dual-meta panel against blast load. The proposed dual-meta
 537 panel is aimed to increase the blast resistance capacity, whilst maintaining a low base reaction
 538 force. The key points found from the study can be enumerated as follows:

539 1. Compared to the conventional sandwich panel with solid and hollow truss core, the panel
540 with dual-meta truss core has smaller central peak deflections of the back plate (up to 20% for
541 the first peak and 52% for the second peak), smaller reaction force (up to 30% for the first peak
542 and 78% for the second peak), and absorbs more blast loading energy, demonstrating that the
543 dual-meta panel has the potential for significantly enhancing the dynamic performance of the
544 cladding and outperforms its conventional counterparts.

545 2. The performance of the dual-meta panel on blast loading mitigation depends on the structural
546 configurations. A relatively weak front plate and stronger back plate, and separating the
547 sacrificial dual-meta panel from the protected structure with a small gap lead to better protective
548 effectiveness of the panel in terms of energy absorption and the level of load transmitted to the
549 protected structure.

550 3. The performance of the dual-meta panel also depends on the blast loading profile and energy
551 distribution. The dual-meta panel with bandgaps consistent with the primary blast loading
552 energy distribution in the frequency domain is more effective in mitigating the blast loading
553 effect.

554 The study proves that the dual-meta panel holds great potential for extensive applications in
555 various engineering fields requiring blast load mitigation.

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692

693 **Appendix**

694 With an attempt to estimate the accurate values of the spring stiffness k_i ($i=1,2,3,4$), the
 695 commercial software COMSOL MULTIPHYSICS was leveraged to conduct the numerical
 696 simulation. A constant force F which is depicted in Fig. 20 (a) is applied to the model to
 697 calculate the value of shear spring stiffness k_2 of the internal core and while two constant force
 698 F was put in two directions of the model to estimate the values of k_1 shown in Fig. 20(b).
 699 Similarly, the calculation of value k_4 and k_3 is carried out with the same procedure but different
 700 dimensions. As seen in Fig. 20 (a) and 20 (b), the average displacements monitored at the
 701 surfaces are denoted as u_i ($i=1,2,3,4$) and captured by commercial software. The boundary
 702 condition for all edges of the outer shell is clamped. The relation between stiffness and
 703 displacement of the unit model which is shown in will be achieved as following [19]:

$$k_1(u_1 + u_2) + k_2u_1 = F \tag{30}$$

$$k_2u_3 = F$$

$$k_3(u_4 + u_5) + k_4u_4 = F$$

$$k_4u_6 = F$$

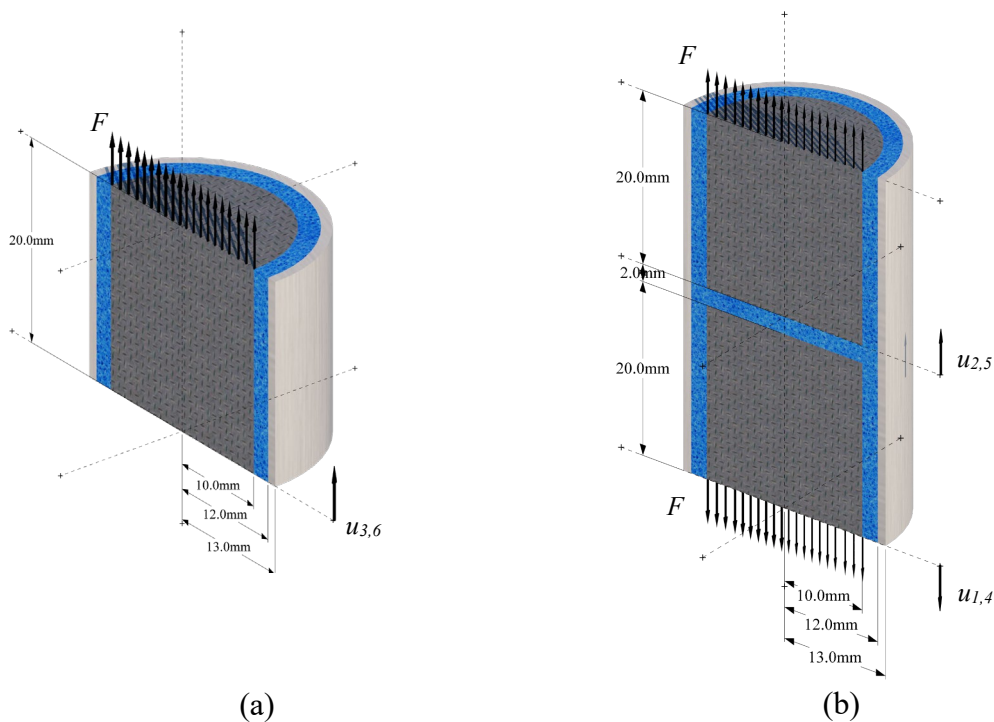


Fig. 20. Outline model utilized for the calculation of (a) k_2 and k_4 , and (b) k_1 and k_3 .

