A DESIGN MODEL FOR STRAIN-SOFTENING AND STRAIN-HARDENING FIBER REINFORCED

ELEMENTS REINFORCED LONGITUDINALLY WITH STEEL AND FRP BARS

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Abstract

A close form solution is developed for the prediction of the moment-curvature relationship of cross sections of fiber

reinforced concrete (FRC) elements failing in bending, and reinforced longitudinally with steel and fiber reinforced

polymer (FRP) bars. The FRP bars are installed with the largest possible internal arm, e.g. with the minimum concrete

cover that assures the bond conditions for a sound stress transfer from FRC to the FRP bars. The model is also able of

simulating the flexural strengthening contribution provided by FRP bars installed according to the Near Surface Mounted

(NSM) technique. To have good protection conditions against corrosion, the steel bars are applied with a relatively thick

FRC cover. Since steel stirrups are the reinforcement with the smaller concrete cover thickness, they are the most

susceptible to corrosion. In the reinforcement concept to be developed in the present research program, steel stirrups are

replaced with discrete fibers. This hybrid reinforcement aims to develop high durable pre-fabricated elements that fail in

bending. The proposed analytical formulation can simulate FRC with strain softening or strain hardening behavior. In the

present work, the formulation is described and its predictive performance is appraised.

Keyword: A. Discontinuous reinforcement; A. Fibres; A. Hybrid; C. Analytical modeling; flexural reinforcement

1. INTRODUCTION

Corrosion of the steel reinforcement (SR) of concrete structures is the main material pathology responsible for the huge amounts of resources spent in the rehabilitation of the built infrastructure. The corrosion of SR can be caused by a deficient concrete quality, a too small concrete cover, inadequate construction practices, design inaccuracies, an improper maintenance of the buildings and an unexpected high level of aggressiveness of environmental agents. As a consequence of the SR corrosion, the sound cross sectional area decreases, the concrete cover spalls, the concrete cross sectional area diminishes and the concrete strength decreases. This leads to a decrease of the load carrying capacity of the reinforced concrete (RC) member, which can affect the structural safety. Therefore, all the strategies that can avoid the occurrence of corrosion of the reinforcement systems applied in concrete structural elements are of paramount importance in terms of decreasing the maintenance costs of these structures. This contributes decisively for the development of a new generation of high durable and sustainable RC buildings.

The present work is part of a research project aiming to contribute to the development of a reinforcement system that combines Fiber Reinforced Polymers (FRP) and steel bars in an optimized arrangement of the bars, using the potentialities that each material can provide. In fact, non-corrodible FRP bars of high tensile strength, but with brittle behavior, are near surface mounted, while ductile steel bars have a concrete cover thickness that assures a high protection from the effects of corrosive agents. The steel reinforcement ratio is also designed in order to guarantee the safety of the structure in case of a fire occurrence and the consequent loss of FRP reinforcing capacities.

A fiber reinforced concrete (FRC) of high shear strength is used to suppress the necessity of using steel stirrups, to improve the bond behavior of concrete-FRP and concrete-steel bars, and to increase protection to the steel bars from corrosive agents.

The replacement of conventional steel reinforcement by FRP bars has been investigated [1-3] to prevent the corrosion problem and to improve the durability of concrete structures near marine environments, near the ground, in chemical and other industrial plants, in places where good quality concrete cannot be achieved, and in thin structural elements. In fact, in comparison with steel, FRP materials have higher resistance to corrosion, and higher strength-to-weight ratio [4]. Furthermore, they are non-electrical conductive and non-magnetic materials. However, the major obstacles of the application of FRP bars as a reinforcing material for concrete structures are the relatively high initial costs, low modulus of elasticity, lack of ductility (linear stress-strain diagram up to rupture with no discernible yield point), and absence of well-consolidated design guidelines [2-4]. Concrete members reinforced with FRP and subjected to bending moments behave linearly up to cracking, and almost linearly after cracking with a significant lower stiffness [5, 9]. Deflections and

strains of concrete members reinforced with FRP bars are generally larger than the homologous members reinforced with steel bars. This is due to the low modulus of elasticity and the different bond characteristics of the FRP reinforcements [3, 6, 7]. In addition, as a result of a larger crack width and a smaller compressive stress blocks when using FRP bars for the flexural reinforcement, the shear capacity of FRP-reinforced concrete beams is smaller than in the case of steel-reinforced concrete beams of same reinforcement ratio [4].

In an attempt to overcome these drawbacks, Aiello and Ombres [8] proposed a combination of FRP and steel reinforcements for concrete beams. With this combination of the reinforcement materials, and considering the minor concrete cover required for FRP, an effective solution in terms of durability is obtained by placing the FRP bars near the outer surface of the tensile zone and steel bars at the inner level of the tensile zone. In terms of structural behavior of the concrete member, the presence of the steel bars in the above mentioned hybrid reinforcement system contributes significantly to ductility and stiffness.

The few tests that were executed confirmed this idea. In fact, Tian and Yuan [7] concluded that the deflection of concrete beams reinforced with FRP and steel bars was smaller than that of beams reinforced with GFRP. Aiello and Ombres [8] verified that, in comparison with beams exclusively reinforced with FRP bars, the presence of steel bars reduces crack width and crack spacing values. These studies also indicate that the hybrid reinforcement system (steel and FRP) is a competitive solution when the long term costs with repairing activities are also taken into account.

Available research [10-14] evidences that steel fibers can replace steel stirrups, especially when high strength concrete is used, and when beams are relatively shallow [12, 13]. Steel fibers also reduce the width of shear cracks, thus also contributing to improve concrete durability [14]. Research also shows that up to a maximum crack width of 0.5 mm, steel fibers are not affected by corrosion [15]. Furthermore, advances in the manufacture technology of synthetic fibers show that these non-corrodible fibers have high potentialities for the shear resistance of RC elements [16].

In the present work, a close-form solution is developed for the design of cross section elements of FRC reinforced with longitudinal steel and FRP bars that fail in bending. This model is an extension of the one proposed by Soranakom [17] for FRC beams failing in bending and reinforced with steel bars. This Author has shown that this reinforcing strategy can be cost effective. In the present extended model, the contribution of FRP bars is simulated, since the favorable benefits of using, simultaneously, steel and FRP bars can contribute for the development of cost competitive and durable structural concrete elements. In Soranakom's model a strain softening (SS) FRC was simulated when applied simultaneously with tensile steel bars [17]. In the model to be presented hereafter, the FRC can have strain softening or strain hardening (SH) post-cracking behavior [18], and can be reinforced with steel bars, FRP bars or both reinforcements. The predictive

performance of the model is assessed by using a cross section layer model that takes into account the constitutive laws of the materials [19]. Using the moment-curvature relationship derived from this model and applying the concepts of the stiffness method, the force-deflection relationship of statically determinate elements are determined. This numerical strategy was applied to assess its potentiality for the evaluation of the force-deflection response recorded experimentally in non-fibrous concrete beams reinforced with longitudinal steel bars and flexurally strengthened with carbon fibre reinforced polymer (CFRP) laminate strips applied according to the near surface mounted (NSM) technique [19].

2. CONSTITUTIVE LAWS FOR THE INTERVINING MATERIALS

2.1 Tensile and compressive behavior of FRC

The stress-strain (σ - ε) response for the FRC is based on the model proposed by Soranakom and Mobasher, represented in Fig. 1 [20]. These diagrams are based on an idealized model indicated by Lim *et al.* [21] for steel fiber reinforced concrete. As shown in Fig. 1a, the linear portion of an elastic–perfectly plastic compressive stress–strain response ends at the pseudo-yield point (ε_{cy} , σ_{cy}) and remains constant at compressive "yield" stress (σ_{cy}) until the ultimate compressive strain (ε_{cu}). After this strain limit the concrete strength capacity is assumed null. The concrete stiffness factor (γ) represents the ratio between the Young's modulus of the FRC in compression (E_c) and in tension (E). In this model ω is the normalized compressive "yield" stress, and λ_{cu} is the normalized ultimate compressive strain. As shown in Fig. 1b, the tensile behavior is described by a trilinear diagram with an elastic range defined by the tensile modulus (E), followed by a branch characterized by the post-cracking modulus (E_{cr}) that can be obtained by using a post-crack modulus parameter (η). By attributing to η a negative or a positive value, the same model can be used to simulate strain softening (SS) or strain hardening (SH) FRC, respectively. At the third region of tensile response, tensile strength remains constant and equal to σ_{cst} . The β_{tu} is the normalized ultimate tensile strain, after which the concrete tensile strength capacity is assumed null. The stress–strain relationship for compression and tension can be expressed as:

$$\sigma_{t}(\varepsilon_{t}) = \begin{cases} E\varepsilon_{t} & 0 \leq \varepsilon_{t} \leq \varepsilon_{cr} \\ E\varepsilon_{cr} + E_{cr}(\varepsilon_{t} - \varepsilon_{cr}) & \varepsilon_{cr} \leq \varepsilon_{t} \leq \varepsilon_{trn} \\ \mu E\varepsilon_{cr} & \varepsilon_{trn} \leq \varepsilon_{t} \leq \varepsilon_{tu} \\ 0 & \varepsilon_{t} > \varepsilon_{tu} \end{cases}$$
(1)

$$\sigma_{c}(\varepsilon_{c}) = \begin{cases} E_{c}\varepsilon_{c} & 0 \leq \varepsilon_{c} \leq \varepsilon_{cy} \\ E_{c}\varepsilon_{cy} & \varepsilon_{cy} \leq \varepsilon_{c} \leq \varepsilon_{cu} \\ 0 & \varepsilon_{c} > \varepsilon_{cu} \end{cases}$$
 (2)

where σ_c , σ_t , ε_c and ε_t are the compressive and tensile stresses and strains, respectively. In the following equations, the model is represented in normalized format by utilizing the first cracking tensile strain (ε_{cr}) and tensile modulus (E) as two inherent parameters of the material:

$$\frac{\sigma_{t}(\beta)}{E \varepsilon_{cr}} = \begin{cases}
\beta & 0 \le \beta \le 1 \\
1 + \eta(\beta - 1) & 1 \le \beta \le \alpha \\
\mu & \alpha \le \beta \le \beta_{tu} \\
0 & \beta > \beta_{tu}
\end{cases} \tag{3}$$

$$\frac{\sigma_c(\lambda)}{E\varepsilon_{cr}} = \begin{cases} \gamma\lambda & 0 \le \lambda \le \omega \\ \gamma\omega & \omega \le \lambda \le \lambda_{cu} \\ 0 & \lambda > \lambda_{cu} \end{cases}$$
(4)

where the dimensionless parameters have the following meaning (Fig. 1):

$$\omega = \frac{\varepsilon_{cy}}{\varepsilon_{cr}} \tag{5}$$

$$\lambda_{cu} = \frac{\varepsilon_{cu}}{\varepsilon_{cr}} \tag{6}$$

$$\gamma = \frac{E_c}{E} \tag{7}$$

$$\alpha = \frac{\varepsilon_{trn}}{\varepsilon_{cr}} \tag{8}$$

$$\beta_{tu} = \frac{\varepsilon_{tu}}{\varepsilon_{rr}} \tag{9}$$

$$\eta = \frac{E_{cr}}{E} \tag{10}$$

$$\mu = \frac{\sigma_{cst}}{E\varepsilon_{CT}} \tag{11}$$

The normalized tensile strain at the bottom fiber (β) and the normalized compressive strain at the top fiber (λ) are defined as:

$$\beta = \frac{\varepsilon_{tbot}}{\varepsilon_{cr}} \tag{12}$$

$$\lambda = \frac{\varepsilon_{ctop}}{\varepsilon_{cr}} \tag{13}$$

Due to linear variation of strain along the height of the section, β and λ have the following dependence:

$$\lambda = \frac{k}{1-k}\beta\tag{14}$$

where k is the normalized depth of the neutral axis, being determined by dividing the depth of the neutral axis by the depth of section, d (Fig. 4).

2.2 Tensile behavior of steel and FRP bars

The FRC cross section (Fig. 4), of width b and height d, can be reinforced with steel and FRP bars, being $\rho_s = A_s/(bd_s)$ and $\rho_f = A_f/(bd_f)$ the reinforcement ratio of steel and FRP bars, respectively, where A_s and A_f are the area of steel and FRP bars cross section, and $d_s = d - C_s$ and $d_f = d - C_f$ are the internal arms of steel and FRP bars, respectively, being C_s and C_f the concrete cover for the steel and FRP bars, respectively.

As Fig. 2 illustrates, the tensile behavior of the steel bars is simulated by a bilinear stress-strain diagram, with a linear-elastic branch up to the yield strain (ε_{sy}) , followed by a perfectly plastic branch up to attain the ultimate tensile strain (ε_{su}) , after which the steel tensile strength capacity is assumed null. The steel modulus of elasticity (E_s) is defined from the concrete tensile modulus of elasticity (E) by using the steel stiffness factor (γ_s) . The ultimate strain ε_{su} is defined as a multiple of the concrete strain at crack initiation (ε_{cr}) by using the ψ_{su} factor (Fig. 2). The stress–strain relationship to simulate the tensile behavior of steel bars can be expressed as:

$$\frac{\sigma_{S}(\psi)}{E\varepsilon_{cr}} = \begin{cases} \gamma_{S}\psi & 0 \leq \psi \leq \zeta \\ \gamma_{S}\psi & \zeta \leq \psi \leq \psi_{cu} \\ 0 & \psi > \psi_{cu} \end{cases}$$
(15)

where the dimensionless parameters are represented by the following equations:

$$\zeta = \frac{\varepsilon_{sy}}{\varepsilon_{cr}} \tag{16}$$

$$\psi_{su} = \frac{\varepsilon_{su}}{\varepsilon_{cr}} \tag{17}$$

$$\gamma_s = \frac{E_s}{E} \tag{18}$$

$$\psi = \frac{\varepsilon_{\rm S}}{\varepsilon_{\rm cr}} \tag{19}$$

Due to linear variation of strain along the height of the cross section, normalized steel strain (ψ) can be obtained from the following equation:

$$\psi = \frac{1 - k - \Delta_{\rm S}}{1 - k} \beta \tag{20}$$

where Δ_s is the normalized concrete cover thickness for the steel bars (Fig. 4).

Fig. 3 represents the stress-strain linear-elastic diagram for modeling the FRP tensile behavior. When attaining the ultimate tensile strain (ε_{fu}) the FRP bar fails. The modulus of elasticity of the FRP (E_f) can be defined as being a multiple of the Young's modulus of the FRC in tension (E_f), by using the FRP tensile stiffness factor (γ_f). The stress-strain relationship for the tensile behavior of FRP bars can be expressed as:

$$\frac{\sigma_f(v)}{E\varepsilon_{cr}} = v\gamma_f \tag{21}$$

$$\nu_{fu} = \frac{\varepsilon_{fu}}{\varepsilon_{cr}} \tag{22}$$

$$\gamma_f = \frac{E_f}{E} \tag{23}$$

where

$$\nu = \frac{\varepsilon_f}{\varepsilon_{CT}} \tag{24}$$

Due to linear variation of strain along the depth of the cross section, FRP normalized strain (ν) can be obtained from the following equation:

$$\nu = \frac{1 - k - \Delta_{\rm f}}{1 - k} \beta \tag{25}$$

where Δ_f is the normalized concrete cover thickness for the FRP bars (Fig. 4).

3. CLOSE FORM SOLUTION FOR THE EVALUATION OF THE MOMENT-CURVATURE OF A HYBRID REINFORCED SECTION

Assuming that a plane section remains plane after bending, and shear deformation of the section can be ignored, the moment-curvature $(M-\chi)$ diagram of a rectangular cross section, composed of FRC reinforced by tensile steel and FRP bars, can be obtained in accordance with Fig. 5.

The distribution of tensile and compressive stresses in this figure results from the stress-strain constitutive laws assumed for the intervening materials. The three possible stages for the tensile strain at bottom fiber are represented in Fig. 5a $(0 \le \beta \le 1)$, 5b $(1 < \beta \le \alpha)$, and 5c $(\alpha < \beta \le \beta_{tu})$. Each of stage 2 and 3 has four possible conditions due to the value of compressive strain at top fiber in either elastic $(0 < \lambda \le \omega)$ or plastic $(\omega < \lambda \le \lambda_{cu})$ behavior in compression, and also due to the tensile strain of steel bars in either elastic $(0 < \psi \le \zeta)$ or plastic $(\zeta < \psi \le \psi_{su})$ behavior, and also due to the tensile strain of FRP bars in elastic behavior $(0 < v \le \nu_{fu})$. The details of each stage are summarized in Table 1.

The concrete force components and their lines of action, shown in Fig. 5, are obtained from the area and the centroid of the stress block in each zone. The force components for the steel and FRP bars are obtained from the cross sectional area and the stress in these bars.

At each stage of applied normalized concrete tensile strain at the bottom fiber, β , the net force is obtained as the difference between the tensile and compressive forces equated to zero for internal equilibrium and solved to obtain the neutral axis depth ratio k. The equations for the evaluation of the k values for the considered stages are presented in Table 2. After achieving the correct value of k for the corresponding stage, internal moment is obtained by operating on the force

components and their distance from the neutral axis. The corresponding curvature is determined as the ratio between the concrete compressive strain at top fiber and the depth of the neutral axis.

At the generic loading stage i, the moment and curvature (M_i, ϕ_i) are obtained from the following equations:

$$M_i = M_i' M_{cr} \tag{26}$$

$$\phi_i = \phi_i' \phi_{cr} \tag{27}$$

where the M_i' and ϕ_i' the normalized moment and normalized curvature at stage i, respectively, obtained from the equations included in Table 3. In equations (26) and (27) M_{cr} and ϕ_{cr} are the cracking moment and cracking curvature, respectively, which for a rectangular cross section are obtained from:

$$M_{cr} = \frac{1}{6}bd^2(E\varepsilon_{cr}) \tag{28}$$

$$\phi_{cr} = \frac{2\varepsilon_{cr}}{d} \tag{29}$$

The derivation of the expressions in Tables 2 and 3 can be found elsewhere [22].

4. ALGORITHM TO PREDICT THE MOMENT-CURVATURE RESPONSE OF FRC HYBRID REINFORCED CROSS SECTION

To calculate the moment-curvature relationship of a rectangular FRC hybrid reinforced cross section, an algorithm was implemented in a computer program. In this algorithm the following materials parameters are input data and they should be obtained by experimental tests or from bibliographical sources: ω , λ_{cu} , γ , E, ε_{cr} , α , β_{tu} , μ for the concrete; ζ , ψ_{su} , γ_s for the steel bars; ν_{fu} , γ_f for the FRP bars. According to this algorithm, the normalized ultimate FRC tensile strain (β_{tu}) is divided in normalized strain increment:

$$\Delta \beta = \frac{\beta_{tu}}{N} \tag{30}$$

where N is the number of increments. The normalized concrete tensile strain at the bottom fiber (β) is increased from zero to β_{tu} in increments of $\Delta\beta$. For each β the corresponding moment and its curvature are determined from the calculation process represented in the flowchart of Fig. 6. During this process, if the ultimate strain of the intervening materials is attained, the analysis is ended, and the corresponding failure mode is reported: concrete crushing when $\varepsilon_c = \varepsilon_{cu}$; concrete tensile failure when $\varepsilon_t = \varepsilon_{tu}$; steel tensile failure when $\varepsilon_s = \varepsilon_{su}$; FRP tensile failure when $\varepsilon_f = \varepsilon_{fu}$.

5. MODEL APPRAISAL

To evaluate the accuracy of the proposed formulation, the results obtained from the implemented algorithm are compared with those determined from the DOCROS software [23]. According to the model implemented in DOCROS, a cross section is discretized in layers that can have distinct constitutive laws for the characterization of the behavior of the materials that constitute these layers. A cross section can be composed of plain concrete and FRC layers, and can include steel and FRP bars. The cross section can be subjected to axial load and increasing curvature. A detailed description of DOCROS can be found elsewhere [23].

In Table 4 are indicated the values of the geometrical and material parameters for the examples adopted for the assessment of the predictive performance of the developed formulation.

Two types of analysis were performed in order to show the relevance of adopting a strain softening (SS) or a strain hardening (SH) FRC. The moment-curvature relationships obtained from the present formulation and from DOCROS, for the case of using a SS-FRC and a SH-FRC, are compared in Figs. 7 and 8, respectively, from which it can be concluded that the proposed formulation predicts the M- χ with high accuracy. In both analyses the calculation process ended due to the failing of FRP bars (the ultimate tensile strain was attained). Comparing the diagrams of these both figures it is verified that an increase of μ from 0.33 to 1.00 has provided an increase of 60% in the maximum moment.

To appraise the capability of the proposed formulation for the case of structural repairing and strengthening applications, the results of model are compared with those registered in a previous experimental investigation about the use of CFRP laminate strips applied according to the NSM technique for the flexural strengthening of RC beams [19]. Four series of beams reinforced with different percentages of tensile longitudinal steel bars with the mean concrete cover of 25 mm, and also flexurally strengthened with variable number of CFRP strips were tested in a four point loading configuration, see Fig. 9. This figure also represents the geometry of the beams, the reinforcement arrangement and the number and position of the CFRP laminates, as well as the loading and the support conditions. The failure mode of the strengthened beams was characterized by the detachment of a concrete cover that includes the CFRP laminates. Up to near the failure of the beams, sliding of CFRP was marginal. To simulate the flexural capacity of these beams using the developed model, the CFRP strips were assumed perfectly bonded to surrounding concrete, and were replaced by equivalent CFRP bars in tensile face of the section, with a concrete cover of 6.0 mm. Based on available information, the mechanical properties adopted in the numerical simulations for the intervening materials are indicated in Table 5. Note that an almost null value was assumed for the normalized post-cracking residual strength (μ) to neglect the residual strength of plain concrete in tension after the linear softening branch (for strain levels above ε_{trn}). Small differences exist between the values for the properties of the

intervening materials since each series was casted in distinct periods and steel bars of different diameters had distinct properties.

Utilizing the moment-curvature relationships obtained from the proposed model, the force-deflection response of beams failing in bending can be determined using the algorithm described in Fig. 10 [24]. Comparison between experimental and numerical load-middle span deflection relationships are depicted in figures 11 to 14. The points corresponding to CFRP sliding in the experimental tests are also indicated. As it is observed, the developed numerical strategy is capable of fitting with enough accuracy the recorded experimental load-middle span deflection curves of each tested beam up to the CFRP sliding point. After this deflection level, the developed model is no more capable of fitting the experimental response, since a localized failure mode (rip-off) was formed in the experimentally tested beams, consisting of the detachment of the concrete cover layer that includes the CFRP laminates [19]. In fact, the numerical curve increases almost linearly, while the experimental curve shows a gradual decrease of the stiffness of the beam up to its collapse. The deviation of the numerical curve from the experimental one in its last part (when sliding comes to be significant and rip-off failure mode starts its formation) increases with the percentage of longitudinal steel and FRP elements, as it would be expected based on the conclusions already pointed out elsewhere [25]. The maximum strains experimentally registered in the laminates for the V1R1, V2R2, V3R2 and V4R3 beams at ultimate load were 0.016, 0.013, 0.013 and 0.011, respectively, while the model predicted the following values: 0.017, 0.015, 0.014, and 0.011, which proves the validity of the FRP-concrete perfect bond assumption for this type of applications.

6. CONCLUSIONS

The simultaneous use of steel and FRP longitudinal bars for the flexural reinforcement of pre-fabricated fiber reinforced concrete (FRC) beams is a hybrid reinforcement concept that can contribute to the development of new generation of more durable concrete structures. The steel and FRP bars are positioned in order to optimize their reinforcing potentialities, to assure the necessary bond conditions for the FRP bars, and to provide extra protection against corrosion to the steel bars. Steel stirrups are replaced by discrete fibers, applying a FRC with a post-cracking residual strength that can provide the required shear resistance. To analyze the flexural potentialities of this concept, an analytical model was developed, and the equations for the determination of the moment-curvature relationship for a rectangular cross section reinforced according to this concept are presented in this work. The good predictive performance of the developed model was appraised using a computer program based on the cross section layer model. The proposed formulation was also utilized in order to simulate the load-middle deflection response of plain concrete beams reinforced with steel bars and flexurally strengthened with

carbon fiber reinforced polymer laminate strips according to the Near Surface Mounted technique. The moment-curvature relationship was used as the entity capable of supplying the tangential or the secant flexural stiffness of small elements that discretize a statically determinate beam, in order to predict its force-deflection response. This numerical strategy was capable of simulating with high accuracy the force-deflection response of this hybrid reinforced beams. This model can also be used to perform parametric studies for the optimization of the reinforcing systems for given types of beams.

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Notation

b = cross section width d = cross section depth

E = tensile modulus of elasticity of FRC

 E_c = compressive modulus of elasticity of FRC

 E_s = modulus of elasticity of steel bars

 E_f = modulus of elasticity of FRP bars

k = the neutral axis depth ratio

 $M_{i}^{'}$ = normalized bending moment (M/M_{cr}) in stage i

 M_{cr} = bending moment at FRC crack initiation

 α = normalized transition strain

 β = normalized tensile strain at bottom fiber

 β_{tu} = normalized ultimate tensile strain

 γ = normalized compressive modulus of elasticity of FRC

 γ_s = normalized modulus of elasticity of steel bars

 γ_f = normalized modulus of elasticity of FRP bars

 ε_{cy} = compressive yield strain of FRC

 ε_{cu} = ultimate compressive strain of FRC

 ε_{sv} = tensile yield strain of steel bars

 ε_{su} = ultimate tensile strain of steel bars

 ε_{fu} = ultimate tensile strain of FRP bars

 ε_{cr} = tensile strain at crack initiation of FRC

 ε_{trn} = tensile strain at transition point of FRC

 ε_{tu} = ultimate tensile strain of FRC

 ζ = normalized transition tensile strain of steel bars

η = normalized post-crack modulus of FRC

 λ = normalized compressive strain at the FRC top fiber

 λ_{cu} = normalized ultimate compressive strain of FRC

 μ = Normalized post-crack residual strength of FRC

 ρ_{s} = reinforcement ratio of longitudinal steel bars

 ρ_f = reinforcement ratio of longitudinal FRP bars

 σ_c = compressive stress of FRC

 σ_t = tensile stress of FRC

 σ_s = tensile stress of the steel bars

 σ_f = tensile stress of the FRP bars

 σ_{cst} = residual tensile stress of FRC

 ϕ_{cr} = curvature at crack initiation of FRC

 $\phi_i^{'}$ = normalized curvature ϕ/ϕ_{cr}

 ψ = normalized tensile strain of steel bars

 ψ_{su} = normalized ultimate steel tensile strain of steel bars

 ω = normalized compressive yield strain of FRC

 Δ_s = normalized cover thickness of steel bars

 $\Delta_{\rm f}$ = normalized cover thickness of FRP bars

v = normalized tensile strain of FRP bars

 v_{fu} = normalized ultimate tensile strain of FRP bars

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- Table 5: Components properties used in numerical simulation.

Table 1: Variations of normalized strain parameters of the intervening materials in the possible stages.

Stage	C	oncrete	Steel	FRP		
Stage	Tension	Compression	Steel	7.14		
1	$0 \le \beta \le 1$	$0 \le \lambda \le \omega$	$0 \le \psi \le \zeta$	$0 \le v \le v_{fu}$		
2.1.1.1	$1<\beta\leq\alpha$	$0 \le \lambda \le \omega$	$0 \leq \psi \leq \zeta$	$0 \leq \nu \leq \nu_{fu}$		
2.1.2.1	$1<\beta\leq\alpha$	$0 \le \lambda \le \omega$	$\zeta < \psi \leq \psi_{su}$	$0 \leq \nu \leq \nu_{fu}$		
2.2.1.1	$1<\beta\leq\alpha$	$\omega < \lambda \leq \lambda_{cu}$	$0 \le \psi \le \zeta$	$0 \leq \nu \leq \nu_{fu}$		
2.2.2.1	$1<\beta\leq\alpha$	$\omega < \lambda \leq \lambda_{cu}$	$\zeta < \psi \leq \psi_{su}$	$0 \leq \nu \leq \nu_{fu}$		
3.1.1.1	$\beta > \alpha$	$0 \le \lambda \le \omega$	$0 \le \psi \le \zeta$	$0 \leq \nu \leq \nu_{fu}$		
3.1.2.1	$\beta > \alpha$	$0 \le \lambda \le \omega$	$\zeta < \psi \leq \psi_{su}$	$0 \leq \nu \leq \nu_{fu}$		
3.2.1.1	$\beta > \alpha$	$\omega < \lambda \leq \lambda_{cu}$	$0 \le \psi \le \zeta$	$0 \leq \nu \leq \nu_{fu}$		
3.2.2.1	$\beta > \alpha$	$\omega < \lambda \le \lambda_{cu}$	$\zeta < \psi \leq \psi_{su}$	$0 \leq \nu \leq \nu_{fu}$		

Table 2: Equations for the evaluation of the depth of the neutral axis parameter, k, for each stage [22].

Stage	k
1	$k_1 = \begin{cases} \frac{2\gamma_{\rm S}\rho_{\rm S}(1-\Delta_{\rm S}) + 2\gamma_{\rm f}\rho_{\rm f}(1-\Delta_{\rm f}) + 1}{2\sqrt{\rm D_1}} & for \ \gamma = 1\\ \frac{-(1+\rho_{\rm S}\gamma_{\rm S} + \gamma_{\rm f}\rho_{\rm f}) + \sqrt{\rm D_1}}{(\gamma-1)} & for \ \gamma < 1 \ or \ \gamma > 1 \end{cases}$
	$D_1 = 2\gamma_s \rho_s (0.5\rho_s \gamma_s + \gamma_f \rho_f + \gamma - \gamma \Delta_s + \Delta_s) + \gamma + 2\gamma_f \rho_f (0.5\rho_f \gamma_f + \gamma - \gamma \Delta_f + \Delta_f)$
2.1.1.1	$k_{2111} = \frac{D_{2111} + \gamma_{\rm S} \beta^2 \rho_{\rm S} + \gamma_{\rm f} \beta^2 \rho_{\rm f} - \sqrt{\left[(D_{2111} + \gamma_{\rm S} \beta^2 \rho_{\rm S} + \gamma_{\rm f} \beta^2 \rho_{\rm f})^2 - (D_{2111} - \beta^2 \gamma) (D_{2111} + 2\gamma_{\rm S} \beta^2 \rho_{\rm S} - 2\gamma_{\rm S} \beta^2 \rho_{\rm S} \Delta_{\rm S} + 2\gamma_{\rm f} \beta^2 \rho_{\rm f} - 2\gamma_{\rm f} \beta^2 \rho_{\rm f} \Delta_{\rm f}) \right]}}{D_{2111} - \beta^2 \gamma}$
$D_{1} = \begin{bmatrix} & & & & & & \\ 2.1.1.1 & & & & & \\ & & & & & \\ & & & & & \\ 2.1.2.1 & & & & \\ & & & & & \\ & & & & & \\ & & & & & \\ & & & & & \\ & & & & & \\ & & & & & \\ & & & & & \\ & & & & & \\ & & & & & \\ & & & & \\ & & & & \\ & & & & \\ & & & & \\ & & & & \\ & & & & \\ & & & & \\ & & & & \\ & & \\ & & & \\ & & \\ & & & \\ & & $	$D_{2111} = \eta(\beta^2 - 2\beta + 1) + 2\beta - 1$
2.1.2.1	$k_{2121} = \frac{D_{2121} + \zeta\gamma_{\rm S}\beta\rho_{\rm S} + \gamma_{\rm f}\beta^2\rho_{\rm f} - \sqrt{\left[(D_{2121} + \zeta\gamma_{\rm S}\beta\rho_{\rm S} + \gamma_{\rm f}\beta^2\rho_{\rm f})^2 - (D_{2121} - \beta^2\gamma)(D_{2121} + 2\zeta\gamma_{\rm S}\beta\rho_{\rm S} + 2\gamma_{\rm f}\beta^2\rho_{\rm f} - 2\gamma_{\rm f}\beta^2\rho_{\rm f}\Delta_{\rm f})\right]}{D_{2121} - \beta^2\gamma}$
	$D_{212} = \eta(\beta^2 - 2\beta + 1) + 2\beta - 1$
2.2.1.1	$k_{2211} = \frac{D_{2211} + \omega \gamma \beta + \gamma_S \beta^2 \rho_S + \gamma_f \beta^2 \rho_f - \sqrt{\left[(D_{2211} + \omega \gamma \beta + \gamma_S \beta^2 \rho_S " + \gamma_f \beta^2 \rho_f)^2 - (D_{2211} + 2\omega \gamma \beta) (D_{2211} + 2\gamma_S \beta^2 \rho_S - 2\gamma_S \beta^2 \rho_S \Delta_S + 2\gamma_f \beta^2 \rho_f - 2\gamma_f \beta^2 \rho_f \Delta_f) \right]}{D_{2211} + 2\omega \gamma \beta}$
	$D_{2211} = \eta(\beta^2 - 2\beta + 1) + 2\beta - 1 + \omega^2 \gamma$
2.2.2.1	$k_{2221} = \frac{{}^{D_{2221} + \beta\zeta\gamma_{S}\rho_{S} + \gamma_{\mathrm{f}}\beta^{2}\rho_{\mathrm{f}} + \omega\gamma\beta - \sqrt{[(\mathrm{D}_{2221} + \omega\gamma\beta + \gamma_{S}\beta\zeta\rho_{\mathrm{S}} + \gamma_{\mathrm{f}}\beta^{2}\rho_{\mathrm{f}})^{2} - (\mathrm{D}_{2221} + 2\omega\gamma\beta)(\mathrm{D}_{2221} + 2\gamma_{S}\beta\zeta\rho_{\mathrm{S}} + 2\gamma_{\mathrm{f}}\beta^{2}\rho_{\mathrm{f}} - 2\gamma_{\mathrm{f}}\beta^{2}\rho_{\mathrm{f}}\Delta_{\mathrm{f}})]}}{{}^{D_{2221} + 2\omega\gamma\beta}}$
	$D_{2221} = \eta(\beta^2 - 2\beta + 1) + 2\beta - 1 + \omega^2 \gamma$
3.1.1.1	$k_{3111} = \frac{D_{3111} + \gamma_{\text{S}} \beta^2 \rho_{\text{S}} + \gamma_{\text{f}} \beta^2 \rho_{\text{f}} - \sqrt{\left[(D_{3111} + \gamma_{\text{S}} \beta^2 \rho_{\text{S}} + \gamma_{\text{f}} \beta^2 \rho_{\text{f}})^2 - (D_{3111} - \beta^2 \gamma)(D_{3111} + 2\gamma_{\text{S}} \beta^2 \rho_{\text{S}} - 2\gamma_{\text{S}} \beta^2 \rho_{\text{S}} \Delta_{\text{S}} + 2\gamma_{\text{f}} \beta^2 \rho_{\text{f}} - 2\gamma_{\text{f}} \beta^2 \rho_{\text{f}} \Delta_{\text{f}}) \right]}}{D_{3111} - \beta^2 \gamma}$
	$D_{311} = \eta(\alpha^2 - 2\alpha + 1) + 2\alpha - 1 + 2\mu\beta - 2\mu\alpha$
3121	$k_{3121} = \frac{D_{3121} + \zeta\gamma_{\rm S}\beta\rho_{\rm S} + \gamma_{\rm f}\beta^2\rho_{\rm f} - \sqrt{\left[(D_{3121} + \zeta\gamma_{\rm S}\beta\rho_{\rm S} + \gamma_{\rm f}\beta^2\rho_{\rm f})^2 - (D_{3121} - \beta^2\gamma)(D_{3121} + 2\zeta\gamma_{\rm S}\beta\rho_{\rm S} + 2\gamma_{\rm f}\beta^2\rho_{\rm f} - 2\gamma_{\rm f}\beta^2\rho_{\rm f}\Delta_{\rm f})\right]}{D_{3121} - \beta^2\gamma}$
3.1.2.1	$D_{3121} = \eta(\alpha^2 - 2\alpha + 1) + 2\alpha - 1 + 2\mu\beta - 2\mu\alpha$
	$k_{3211} = \frac{D_{3211} + \omega \gamma \beta + \gamma_S \beta^2 \rho_S + \gamma_f \beta^2 \rho_f - \sqrt{\left[(D_{3211} + \omega \gamma \beta + \gamma_S \beta^2 \rho_S + \gamma_f \beta^2 \rho_f)^2 - (D_{3211} + 2\omega \gamma \beta)(D_{3211} + 2\gamma_S \beta^2 \rho_S - 2\gamma_S \beta^2 \rho_S \Delta_S + 2\gamma_f \beta^2 \rho_f - 2\gamma_f \beta^2 \rho_f \Delta_f) \right]}{D_{3211} + 2\omega \gamma \beta}$
3.2.1.1	-3211:/F
	$D_{3211} = \eta(\alpha^2 - 2\alpha + 1) + 2\alpha - 1 + \omega^2 \gamma + 2\mu\beta - 2\mu\alpha$
3.2.2.1	$k_{3221} = \frac{D_{3221} + \beta \zeta \gamma_S \rho_S + \gamma_f \beta^2 \rho_f + \omega \gamma \beta - \sqrt{\left[(D_{3221} + \zeta \gamma_S \beta \rho_S + \gamma_f \beta^2 \rho_f + \omega \gamma \beta)^2 - (D_{3221} + 2\omega \gamma \beta) (D_{3221} + 2\zeta \gamma_S \beta \rho_S + 2\gamma_f \beta^2 \rho_f - 2\gamma_f \beta^2 \rho_f \Delta_f) \right]}{D_{3221} + 2\omega \gamma \beta}$
,-	$D_{321} = \eta(\alpha^2 - 2\alpha + 1) + 2\alpha - 1 + \omega^2 \gamma + 2\mu\beta - 2\mu\alpha$

Table 3: Equations for the evaluation of the normalized moment, $M^{'}$, and normalized curvature, $\phi^{'}$, for each stage [22].

Stage	<i>M′</i>	φ̈́
1	$M_{1}^{'} = \frac{2\beta \left[(\gamma-1)K_{1}^{3} + \left(3\gamma_{S}\rho_{S} + 3\gamma_{f}\rho_{f} + 3 \right)K_{1}^{2} + \left(6\gamma_{S}\rho_{S}(\Delta_{S}-1) + 6\gamma_{f}\rho_{f}(\Delta_{f}-1) - 3 \right)K_{1} + 3\gamma_{S}\rho_{S}(\Delta_{S}-1)^{2} + 3\gamma_{f}\rho_{f}(\Delta_{f}-1)^{2} + 1 \right]}{(1-K_{1})}$	$\phi_1' = \frac{\beta}{2(1-K_1)}$
2.1.1.1	$M_{2111}^{'} = \frac{(2\beta\gamma + C_{2111})K_{2111}^{3} + \left(6\gamma_{s}\beta\rho_{s} + 6\gamma_{f}\beta\rho_{f} - 3C_{2111}\right)K_{2111}^{2} + (3C_{2111} + 12\gamma_{s}\beta\rho_{s}(\Delta_{s} - 1) + 12\gamma_{f}\beta\rho_{f}(\Delta_{f} - 1))K_{2111} + 6\gamma_{s}\beta\rho_{s}(\Delta_{s} - 1)^{2} + 6\gamma_{f}\beta\rho_{f}(\Delta_{f} - 1)^{2} - C_{2111}}{(1 - K_{2111})}$ $C_{211} = \frac{-2\eta\beta^{3} + 3\eta\beta^{2} - 3\beta^{2} - \eta + 1}{\beta^{2}}$	$\phi_{2111}' = \frac{\beta}{2(1 - K_{2111})}$
2.1.2.1	$M_{2121}^{'} = \frac{(2\beta\gamma + C_{2121})K_{2121}^{3} + 3\left(2\gamma_{5}\zeta\rho_{5} + 2\gamma_{f}\beta\rho_{f} - C_{2121}\right)K_{2121}^{2} + 3\left(C_{2121} + 2\gamma_{5}\zeta\rho_{5}(\Delta_{5} - 2) + 4\gamma_{f}\beta\rho_{f}(\Delta_{f} - 1)\right)K_{2121} + 6\gamma_{5}\zeta\rho_{5}(1 - \Delta_{5}) + 6\gamma_{f}\beta\rho_{f}(\Delta_{f} - 1)^{2} - C_{2121}}{(1 - K_{2121})}$ $C_{2121} = \frac{-2\eta\beta^{3} + 3\eta\beta^{2} - 3\beta^{2} - \eta + 1}{\beta^{2}}$	$\phi_{2121}^{'} = \frac{\beta}{2(1 - K_{2121})}$
2.2.1.1	$M_{2211}^{'} = \frac{-(3\omega\gamma + C_{2211})K_{2211}^{3} + 3(\omega\gamma + C_{2211} + 2\gamma_{s}\beta\rho_{s} + 2\gamma_{f}\beta\rho_{f})K_{2211}^{2} + 3(4\gamma_{s}\beta\rho_{s}(\Delta_{s} - 1) + 4\gamma_{f}\beta\rho_{f}(\Delta_{f} - 1) - C_{2211})K_{2211} + 6\gamma_{s}\beta\rho_{s}(\Delta_{s} - 1)^{2} + 6\gamma_{f}\beta\rho_{f}(\Delta_{f} - 1)^{2} + C_{2211}}{(1 - K_{2211})}$ $C_{2211} = \frac{2\eta\beta^{3} - 3\eta\beta^{2} + 3\beta^{2} - \omega^{3}\gamma + \eta - 1}{\beta^{2}}$	$\phi_{2211}' = \frac{\beta}{2(1 - K_{2211})}$
2.2.2.1	$\begin{split} M_{2221}^{'} &= \\ &\frac{-(3\omega\gamma + C_{2221})K_{2221}^{3} + \left(3\omega\gamma + 2C_{22221} + 6\gamma_{s}\zeta\rho_{s} + 6\gamma_{f}\beta\rho_{f}\right)K_{2221}^{2} - \left[6(\gamma_{s}\zeta\rho_{s} - \gamma_{s}\zeta\rho_{s}\Delta_{s} + \gamma_{s}\zeta\rho_{s}) + 3C_{2221} - 12\gamma_{f}\beta\rho_{f}(\Delta_{f} - 1)\right]K_{2221} + 6\gamma_{s}\zeta\rho_{s} - 6\gamma_{s}\zeta\rho_{s}\Delta_{s} + 6\gamma_{f}\beta\rho_{f}(\Delta_{f} - 1)^{2} + C_{2221}}{(1 - K_{2221})} \\ C_{2221} &= \frac{2\eta\beta^{3} - 3\eta\beta^{2} + 3\beta^{2} - \omega^{3}\gamma + \eta - 1}{\beta^{2}} \end{split}$	$\phi_{2221}^{'} = \frac{\beta}{2(1 - K_{2221})}$
3.1.1.1	$M_{3111}^{'} = \frac{(c_{3111} - 2\beta\gamma)\kappa_{3111}^{3} - 3(2\gamma_{s}\beta\rho_{s} + 2\gamma_{f}\beta\rho_{f} + c_{3111})\kappa_{3111}^{2} + 3(c_{3111} - 4\gamma_{s}\beta\rho_{s}(\Delta_{s} - 1) - 4\gamma_{f}\beta\rho_{f}(\Delta_{f} - 1))\kappa_{3111} - (6\gamma_{s}\beta\rho_{s}(\Delta_{s} - 1)^{2} + 6\gamma_{f}\beta\rho_{f}(\Delta_{f} - 1)^{2} + c_{3111})}{(K_{3111} - 1)}$ $C_{3111} = \frac{3(\mu\beta^{2} - \mu\alpha^{2} - \eta\alpha^{2} + \alpha^{2}) + 2\eta\alpha^{3} + \eta - 1}{\beta^{2}}$	$\phi_{3111}' = \frac{\beta}{2(1 - K_{3111})}$
3.1.2.1	$M_{3121}^{'} = \frac{(C_{3121} - 2\beta\gamma)K_{3121}^{3} - 3\left(2\gamma_{5}\zeta\rho_{5} + 2\gamma_{f}\beta\rho_{f} + C_{3121}\right)K_{3121}^{2} + 3\left(C_{3121} + 2\gamma_{5}\zeta\rho_{5}(2 - \Delta_{5}) - 4\gamma_{f}\beta\rho_{f}(\Delta_{f} - 1)\right)K_{3121} + 6\gamma_{5}\zeta\rho_{5}(\Delta_{5} - 1) - 6\gamma_{f}\beta\rho_{f}(\Delta_{f} - 1)^{2} - C_{3121}}{(K_{3121} - 1)}$ $C_{3121} = \frac{3\left(\mu\beta^{2} - \mu\alpha^{2} - \eta\alpha^{2} + \alpha^{2}\right) + 2\eta\alpha^{3} + \eta - 1}{\beta^{2}}$	$\phi_{3121}^{'} = \frac{\beta}{2(1 - K_{3121})}$
3.2.1.1	$M_{3211}^{'} = \frac{-(3\omega\gamma + C_{3211})K_{3211}^{3} + 3(\omega\gamma + C_{3211} + 2\gamma_{s}\beta\rho_{s} + 2\gamma_{f}\beta\rho_{f})K_{3211}^{2} + 3(4\gamma_{s}\beta\rho_{s}(\Delta_{s} - 1) + 4\gamma_{f}\beta\rho_{f}(\Delta_{f} - 1) - C_{3211})K_{3211} + 6\gamma_{s}\beta\rho_{s}(\Delta_{s} - 1)^{2} + 6\gamma_{f}\beta\rho_{f}(\Delta_{f} - 1)^{2} + C_{3211}}{(1 - K_{3211})}$ $C_{3211} = \frac{3(\mu\beta^{2} - \mu\alpha^{2} - \eta\alpha^{2} + \alpha^{2}) + 2\eta\alpha^{3} - \omega^{3}\gamma + \eta - 1}{\beta^{2}}$	$\phi_{3211}^{'} = \frac{\beta}{2(1 - K_{3211})}$
3.2.2.1	$M_{3221}^{'} = \frac{-(3\omega\gamma + C_{3221})K_{3221}^{3} + (3\omega\gamma + 3C_{3221} + 6\gamma_{5}\zeta\rho_{5} + 6\gamma_{f}\beta\rho_{f})K_{3221}^{2} - 3(2\gamma_{5}\zeta\rho_{5} + 2\gamma_{5}\zeta\rho_{5}(1 - \Delta_{5}) - 4\gamma_{f}\beta\rho_{f}(\Delta_{f} - 1) + C_{3221})K_{3221} + 6\gamma_{5}\zeta\rho_{5}(1 - \Delta_{5}) + 6\gamma_{f}\beta\rho_{f}(\Delta_{f} - 1)^{2} + C_{3221}}{(1 - K_{3221})}$ $C_{322} = \frac{3(\mu\beta^{2} - \mu\alpha^{2} - \eta\alpha^{2} + \alpha^{2}) + 2\eta\alpha^{3} - \omega^{3}\gamma + \eta - 1}{\beta^{2}}$	$\phi_{3221}^{'} = \frac{\beta}{2(1 - K_{3221})}$

Table 4: Data for the model parameters used in the examples for the assessment of the predictive performance of the developed model.

Geo	metric		Mechanical parameters															
	nmeter nm)		Concrete (tension)								Concrete (compression)			Steel (tension)			FRP (tension)	
b d	250 500	ε _{cr} (%)	E (GPa)	α	eta_{tu}	μ^*	μ^{**}	η^*	η^{**}	ω	γ	λ_{cu}	ζ	γ_s	ψ_{su}	ν_{fu}	γ_f	
C_s C_f	80 50	1e-4	30	10	150	1.00	0.33	-0.074	0.0	10	1	40	12	10	120	120	1	

^{*} For strain softening FRC ** For strain hardening FRC

Table 5: Components properties used in numerical simulation.

Beam	Tensile concrete					Compressive concrete			Lo	ngitudin	al steel	CFRP laminates			
series	ε _{cr} (%)	E (GPa)	α	μ	eta_{tu}	ω	γ	λ_{cu}	ρ _s (%)	ζ	γ_s	ψ_{su}	ρ _f (%)	ν_{fu}	γ_f
V1R1	0.01	33.35	30	1e-9	320	175	1.00	350	0.33	38.25	6.00	450	0.084	180	4.76
V2R2	0.01	36.50	25	1e-9	300	175	1.00	350	0.48	40.25	5.48	450	0.160	170	3.70
V3R2	0.01	34.89	30	1e-9	340	175	1.00	350	0.61	34.00	5.73	800	0.160	170	4.55
V3R4	0.01	35.86	15	1e-9	320	175	1.00	350	0.84	31.00	5.48	1500	0.240	170	3.20

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- Fig. 2: Tensile stress-strain relationship for steel bars.
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- Fig. 5: Stress-strain diagram at different stages of normalized tensile strain at the bottom fiber (β) .
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- Fig. 14: Experimental versus numerical load-middle span deflection curve of Beam V4R3.

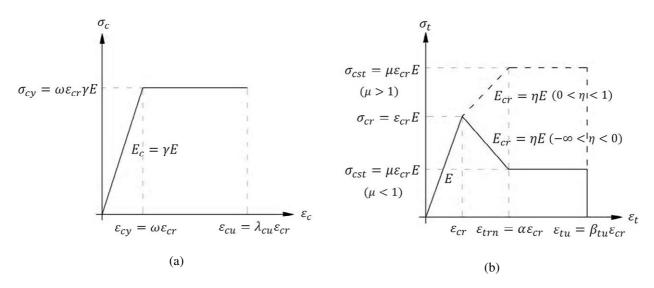


Fig. 1: Stress-strain diagrams for modeling the: a) compressive and b) tensile behavior of fiber reinforced concrete with softening or hardening character [20].

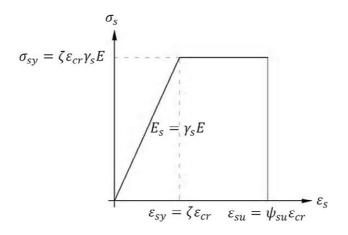


Fig. 2: Tensile stress-strain relationship for steel bars.

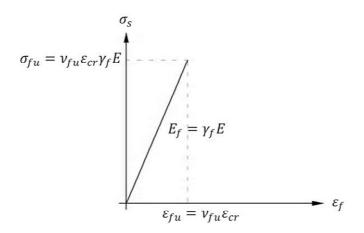


Fig. 3: Tensile stress-strain relationship for FRP bars.

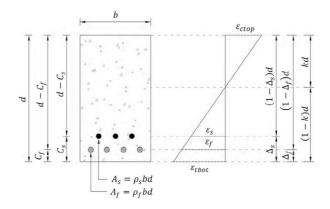


Fig. 4: Cross section reinforced with steel and FRP bars: variables involved in the analytical model.

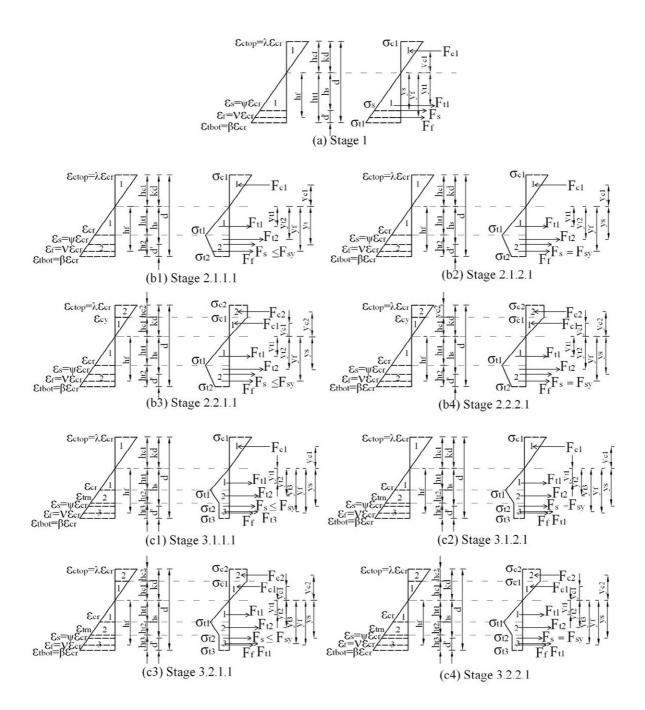


Fig. 5: Stress-strain diagram at different stages of normalized tensile strain at the bottom fiber (β) .

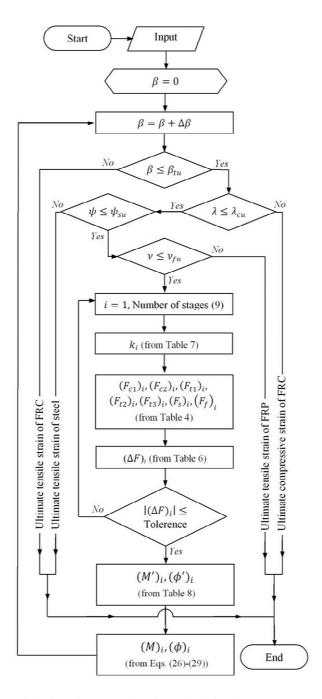


Fig. 6: Flow-chart of the iterative procedure for calculation the moment-curvature response

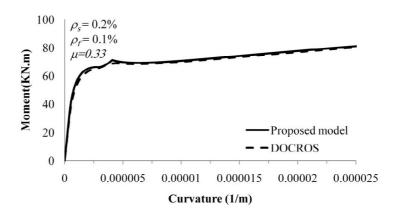


Fig. 7: Moment-curvature response of strain softening behavior of FRC.

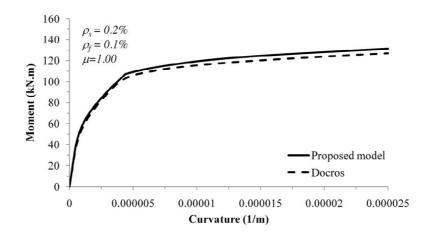


Fig. 8: Moment-curvature response of strain hardening behavior of FRC.

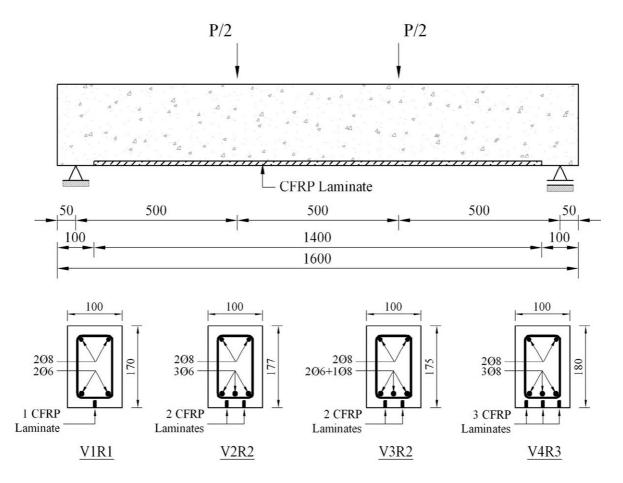


Fig. 9: Series of experimentally tested beams [19]

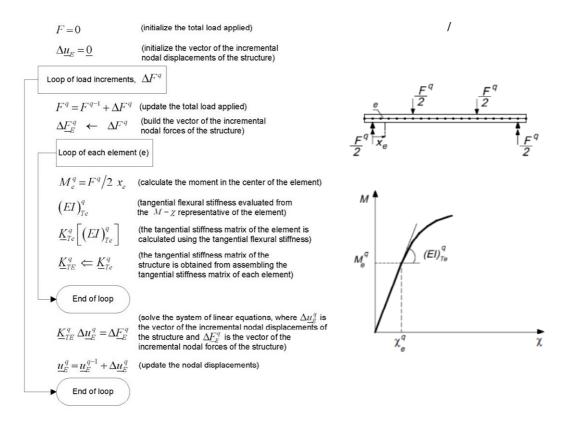


Fig. 10: Numerical approach to simulate the deformational behaviour of structural elements failing in bending [24].

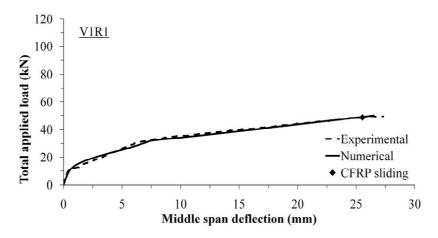


Fig. 11: Experimental versus numerical load-middle span deflection curve of beam V1R1.

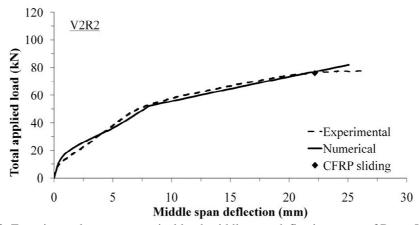


Fig. 12: Experimental versus numerical load-middle span deflection curve of Beam V2R2.

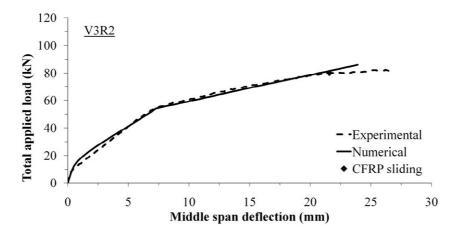


Fig. 13: Experimental versus numerical load-middle span deflection curve of Beam V3R2.

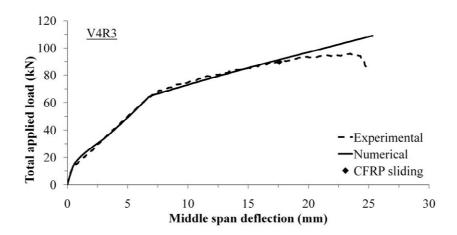


Fig. 14: Experimental versus numerical load-middle span deflection curve of Beam V4R3.