

Flow Around Curved Tandem Cylinders

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ABSTRACT

1
2 *The flow around curved tandem cylinders of equal diameter has been investigated*
3 *for the first time, by means of direct numerical simulations. A convex configuration was*
4 *used. The nominal gap ratio was $L/D = 3.0$ and a Reynolds number of 500 was chosen.*
5 *Due to the change in effective gap ratio along the cylinder axis, there is a variation of*
6 *tandem flow regimes, from alternating overshoot/reattachment, via stable reattachment, to*
7 *co-shedding, in this case called gap shedding. The combination of reattachment and gap*

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8 *shedding gives near-zero drag and vertical forces for the downstream cylinder, whereas the*
9 *corresponding forces on the upstream cylinder are comparable to single curved cylinders.*
10 *Meanwhile, the opposite is true for the lift forces. A low-frequency variation of horizontal*
11 *and vertical forces is seen, and this is attributed to a slow variation of the position where*
12 *gap shedding commences. Finally, the concept of a critical angle is proposed to describe*
13 *the transition to gap shedding, for a given combination of nominal gap ratio and Reynolds*
14 *number.*

INTRODUCTION

15 Curved cylinders are of key interest in offshore applications, in particular when it comes to ris-
16 ers and pipelines. For single curved cylinders, the flow is highly dependent on the inflow direction
17 with respect to the plane of curvature, as shown in several studies [1–5]. Directing the incoming
18 flow towards the outer face of the cylinder (convex configuration) results in a completely different
19 flow topology than directing it towards the inner face (concave configuration). The concave config-
20 uration has been studied by several [4, 6–9], and the effect of oblique inflow angles was the subject
21 of a recent study [5]. Vortex induced vibration of curved cylinders, an important engineering topic,
22 has also been the subject of a number of studies in recent years [10–13]. As the current investiga-
23 tion is concerned with tandem curved cylinders in the parallel, convex configuration, we shall limit
24 ourselves to describing the main contributions related to convex curved cylinders.

25 A pioneering work investigated flow around a cylinder at Reynolds numbers ($Re = U_0 D / \nu$,
26 where ν is the kinematic viscosity, D is the cylinder diameter, and U_0 is the free-stream velocity)
27 100 and 500 [2–4], by means of numerical simulations. The geometry consisted of a quarter-
28 segment of a ring, with a radius of curvature of $r_c = 12.5D$. A horizontal extension of $L_h = 10D$
29 was used in the wake, but there was no vertical extension. An important result from this study was
30 that there is a single vortex shedding-frequency along the entire cylinder, driven by the shedding at
31 the top part, which is nearly normal to the incoming flow. This discovery challenges the so-called
32 independence principle, where it is assumed that two-dimensional sections of a curved cylinder
33 can be analyzed independently. Using this method, the shedding frequency would have varied
34 along the span, according to the variation in local Reynolds number. Due to the axial curvature,
35 there were strong vertical flow components, and it was estimated that approximately one third of
36 the incoming flow rate was deflected downwards [4].

37 Later, the Reynolds number was extended to the subcritical range [14–17], with $Re = 3900$.
38 Direct numerical simulations (DNS) were used. The initial study [14] employed the same geometry
39 as [4]. It was later discovered that the free-slip condition on the top boundary of the computational
40 domain suppressed the vertical velocity component, unless a straight vertical extension $L_v = 6D$
41 was added to the geometry [15]. Because the vortex shedding frequency of the curved cylinder
42 differed from that of the straight vertical extension, splitting of the spanwise vortices occurred near

43 the interface between the two parts [16]. This manifested itself in a low-frequency variation of the
44 velocity time traces, and was confirmed visually in the velocity field plots.

45 An experimental study investigated the effect of radius of curvature [6]. It was found that r_c
46 impacts the Strouhal number ($St = fD/U_0$, where f is the vortex shedding frequency), as well as
47 the shedding angle of the spanwise vortices, but the influence decreases with increasing Reynolds
48 number.

49 It is a challenge for the present study that there are few experimental investigations of single
50 curved cylinders available in the literature. However, DNS is widely considered a high-fidelity
51 method, and several of the aforementioned investigations use well-resolved DNS. Nonetheless,
52 experimental results would be beneficial to further advance this field of research, and will hopefully
53 be carried out in the future.

54 Flow around straight tandem cylinders is governed by the Reynolds number and the spacing
55 between the cylinders, called the gap ratio. For tandem cylinders of equal diameter, the gap ratio
56 is defined as L/D where L is the center-to-center distance. As the gap ratio is increased, the
57 flow regime develops from overshoot, where the shear layers from the upstream cylinder bypass
58 the downstream cylinder and roll up in the wake, through alternating and steady reattachment
59 of the upstream shear layers onto the downstream cylinder, to co-shedding, where large-scale
60 vortices are shed from both cylinders. The spacing at which co-shedding starts is called the
61 critical spacing, L_c . It is well known that the transition from one tandem regime to another is
62 strongly dependent on the Reynolds number [18], which makes it challenging to predict the exact
63 gap ratio at which transition will occur. Nonetheless, the following classification is conventionally
64 adopted: Overshoot $1.0 \leq L/D \leq 1.2 - 1.8$, reattachment $1.2 - 1.8 \leq L/D \leq 3.4 - 3.8$, and
65 co-shedding $3.4 - 3.8 \leq L/D$ [19]. Within the reattachment regime, there is suction in the gap, so
66 that the downstream cylinder experiences thrust instead of drag [20]. For this reason, the critical
67 spacing is sometimes called the drag-inversion spacing, i.e. the spacing at which the downstream
68 cylinder drag coefficient switches sign.

69 Thus far, there is only one study on the subject of two curved cylinders [21]. A side-by-side,
70 convex configuration is used, with a Reynolds number of 500. To the knowledge of the authors,
71 there are no studies that deal with tandem curved cylinders. However, a study of a symmetrically
72 curved circular cylinder with variable span ratio (G/r_c where G is the distance between the cylin-
73 der ends) shares similarities with tandem cylinders when the span ratio is small [22]. The Reynolds
74 number used was 100. An interesting result from that study is that the axial flow along the curved
75 surface is influenced by the wake interference effects. While the side-by-side scenario is perhaps
76 more common in the offshore industry, tandem configurations occur, and any challenges related
77 to these must be clarified.

COMPUTATIONAL ASPECTS78 **Numerical method**

79 In the present study, the full Navier-Stokes equations for incompressible flow are solved through
80 DNS.

$$\frac{\partial u_j}{\partial x_j} = 0, \quad (1)$$

$$\frac{\partial u_i}{\partial t} + u_j \frac{\partial u_i}{\partial x_j} = -\frac{1}{\rho} \frac{\partial P}{\partial x_i} + \frac{\partial}{\partial x_j} \left(\nu \left[\frac{\partial u_i}{\partial x_j} + \frac{\partial u_j}{\partial x_i} \right] \right), \quad i, j = 1, 2, 3 \quad (2)$$

81 All simulations were carried out using the MGLET flow solver. MGLET is based on a finite
82 volume formulation of the incompressible Navier-Stokes equations, and uses a staggered Carte-
83 sian grid [23]. Solid bodies are introduced through an immersed boundary method [24], where the
84 boundary is discretized using a cut-cell approach. A third-order low-storage explicit Runge-Kutta
85 time integration scheme is used for time stepping, and the Poisson equation is solved using an
86 iterative, strongly implicit procedure (SIP). MGLET has previously been used for convex [16] and
87 concave [8, 9] curved cylinder studies.

88 Free-slip boundary conditions are used on all domain boundaries except the inlet and out-
89 let. Uniform inflow is imposed at the inlet, and a Neumann condition is imposed on the velocity
90 components at the outlet.

91 **Computational domain, geometry and definitions**

92 In the present study, the geometry consists of two curved tandem cylinders of equal diameter,
93 with the plane of curvature parallel to the uniform inflow. The convex configuration is used, and
94 the gap ratio is $L/D = 3.0$. The Reynolds number is 500, and this combination of gap ratio and
95 Reynolds number is expected to fall under the reattachment regime for straight tandem cylinders.
96 A Reynolds number of 500 is low for engineering purposes. However, given the novelty of this
97 investigation, it is important to develop a thorough understanding of the basic flow physics before
98 embarking on the more complex case of higher Reynolds numbers. Moreover, it allows more
99 cases to compare with, as $Re = 500$ is used by several single curved cylinder studies.

100 The computational domain and geometry are depicted in figure 1a. The total domain size is
101 $L_x \times L_y \times L_z = 43D \times 20D \times 33D$.

102 The curved part of the cylinder is a quarter-segment of ring with a radius of curvature r_c . The
103 upstream cylinder has a radius of curvature of $r_{cu} = 12.5D$. In order to ensure a constant gap ratio

104 along the entire geometry, a stronger curvature of $r_{cd} = 9.5D$ is used for the downstream cylinder.
 105 However, one of the challenges of a curved tandem cylinder setup is that, regardless of the inflow
 106 direction, the effective gap ratio will vary along the curved part of the cylinders. This is because
 107 the inflow cannot be normal to the local curvature at every point along the cylinder axis. Along
 108 the straight vertical extensions, the gap ratio is constantly $L/D = 3.0$, but along the curved part it
 109 increases with β . In accordance with previous results for a single curved cylinder [15], the curved
 110 tandem cylinders were fitted with straight vertical extensions of $L_v = 7D$, as well as horizontal
 111 extensions of $L_h = 15D$, in order to avoid influence of the computational domain boundaries.

112 Herein, the x direction is referred to as streamwise and y direction as crossflow. The z direction
 113 is referred to as vertical, and vortical structures that align with this direction are dubbed spanwise.
 114 The time-averaged base pressure coefficient is given as $\overline{C}_{pb} = \overline{P} - \overline{P}_0 / \overline{P}_s - \overline{P}_0$. Here, \overline{P}_0 is
 115 the free-stream pressure and \overline{P}_s is the stagnation pressure. Force coefficients are defined as
 116 $\overline{C}_F = 2\overline{F} / \rho U_0 A$, where F is the force component in question, ρ is the fluid density and A is the
 117 projected frontal area. Subscripts D and L denote drag and lift, respectively, and subscript z
 118 denotes vertical force. Note that "lift" implies crossflow (i.e. y) direction in the present study. To
 119 separate the upstream and downstream cylinder coefficients, lower case u and d are used. The
 120 Strouhal numbers listed herein are based on spectral analysis of crossflow velocity time traces in
 121 the wake.

122 Grid independency and validation

123 Because there are no other tandem curved cylinder works to compare with, an initial study
 124 was carried out with two single curved cylinders of $r_c = 9.5D$ and $12.5D$. Four different grids were
 125 tested, but since this was merely used as a starting point for the tandem cylinder grid convergence
 126 study, only the results from the finest grid is shown herein. In table 1, the results are compared
 127 with the available literature. There is something of a spread in the values from different studies,
 128 but even so there is reasonable agreement with the present results.

129 A number of different grids were tested for the curved tandem cylinders, independently varying
 130 the refinement level on upstream and downstream cylinder, as well as in the gap. The flow is sen-
 131 sitive to the grid close to the upstream cylinder and in the gap region. This is no surprise, as these
 132 govern the inflow to the downstream cylinder, and hence are instrumental to the development of
 133 the wake. Refinement of the downstream cylinder influences its force coefficients, naturally, but
 134 also has some influence of the fluctuating lift of the upstream cylinder, through the interaction
 135 between the gap and wake flow.

136 Results for the four main grids are given in table 2. Here, all grids have equal element size on
 137 the upstream and downstream cylinders. For grids t1 and t2, the element size was the same for
 138 the cylinders and the gap region. For t3, the element size in the gap was twice that of the cylinder
 139 surface. For t4, the curved part of the gap was refined to the same element size as the cylinder
 140 surface. The remainder of the gap of t4 had grid cells twice that size, as illustrated in figure 1b.

141 The fluctuating lift and vertical forces, as well as the downstream cylinder drag coefficient,
142 are most sensitive to grid refinement. There is a monotonic decrease of the upstream cylinder
143 drag coefficient as the element size near the surface of the solid bodies is decreased. In addition,
144 refinement of the gap region (from grid t3 to grid t4), further decreases the drag. However, the
145 change in C_{Du} from the coarsest to the finest grid resolution is a mere 1.9 percent. The change in
146 C_{Lrms} for the upstream and downstream cylinder from t2 to t4 is in the order of 2.0 percent. Figure
147 2 shows the effect of grid resolution on the time-averaged velocity field. The differences are in the
148 order of 5 - 10 percent maximum if each of the profiles from grid t1 to t3 are compared with that of
149 grid t4. With this in mind, the resolution of t2 may have been sufficient. Large differences between
150 grids in the downstream cylinder drag, however, indicated that further refinement was needed.

151 The grid convergence study was complicated by the fact that there is long term variation of
152 drag and vertical forces for both cylinders. For t1 and t2, statistics were sampled for 290 time
153 units, tU_0/D , which correspond to approximately 40 vortex shedding cycles. Sampling started
154 after 60 time units. For straight circular cylinder statistics, 40 cycles is ample, but in the present
155 study, a longer sampling time is required. Therefore, statistics were sampled for 550 time units
156 for t3, and 710 time units for t4, which amounts to approximately 81 and 106 large-scale vortex
157 shedding cycles, respectively. In all simulations, the time step was adjusted by means of a built-in
158 procedure in MGLET, in order to reach a target Courant number of 0.8. For t3 and t4, time step
159 adjustment was carried out for 250 time units, after which sampling of statistics commenced.

160 In the end, grid t4 was chosen, due to the strong gradients in the curved gap region. The total
161 number of elements was 529 million, and while this may seem excessive for a Reynolds number of
162 500, there is certainly enough uncharted territory in the present study to warrant careful treatment.

RESULTS

163 Flow topology

164 The instantaneous flow field is depicted in figure 3. We see that, similar to a single curved cylin-
165 der at this Reynolds number, there is shedding of slightly backwards-slanted, large-scale vortices
166 in the wake. This is reminiscent of the flow topology in the wake of a yawed circular cylinder [25].
167 Because the effective gap ratio varies along the span of the cylinders, there is a variation of tandem
168 flow regimes, from alternating overshoot/reattachment along the straight the vertical extensions,
169 via stable reattachment in the upper part of the curved gap, to gap shedding, the equivalent of
170 co-shedding, in the lower curved part. The approximate extent of the instantaneous reattachment
171 zone is marked in figure 3a.

172 Due to the axial curvature, the flow is highly three-dimensional. At a Reynolds number of 500,
173 the flow is expected to display a mode B instability of the wake [26], with streamwise structures
174 of spanwise wavelength $\lambda \approx 1D$ bridging the von Kármán vortices. Evidence of this type of
175 organization is seen throughout the instantaneous flowfield in figure 3b. However, the presence

176 of the downstream cylinder, with some contribution from the gap shedding, causes bending and
 177 tilting of the vortices, so that a much more complex picture emerges.

178 The time-averaged streamwise and vertical velocity fields are shown in figures 4a and b, re-
 179 spectively. Recirculation zones, characterized by negative streamwise velocity, are clearly seen
 180 in figure 4a. These encompass the entire straight vertical gap and near wake, as well as ap-
 181 proximately half of the curved part. Gap shedding, defined as $U/U_0 \leq 0$ at the front face of the
 182 downstream cylinder, commences at $\beta \approx 34.3^\circ$.

183 The vertical velocity plot in figure 4b shows that there is a strong downdraft induced by the
 184 cylinder curvature. However, there are also zones of upwelling in the gap and near wake. Up-
 185 welling in the near wake is previously reported for a single curved cylinder, with a maximum veloc-
 186 ity of $0.08U_0$ [16]. In the present study, the maximum upwelling velocity is approximately $0.093U_0$.
 187 Moreover, relatively high values of upwelling seem to occur along a larger portion of the vertical
 188 extension than for a single curved cylinder.

189 The strongest vertical flow occurs along the stagnation face of the upstream cylinder (see
 190 figure 4b), where the downdraft reaches $-0.44U_0$. However, the values in the gap are also quite
 191 significant. The maximum downdraft at the downstream cylinder stagnation face is $-0.37U_0$.

192 Previous studies of single curved cylinders have found that the axial flow suppresses vortex
 193 formation in the near wake, below $\beta \approx 45^\circ$. A similar result is seen for the downstream cylinder,
 194 marked both in figure 3a and 4a. Recirculation in the wake, defined as $U/U_0 \leq 0$ along the back
 195 face of the downstream cylinder, is suppressed at approximately 36° .

196 Because the axial velocity at the back face of the upstream cylinder is smaller than for the
 197 downstream cylinder (shown indirectly by the vertical velocity plot in figure 4b), it does not suppress
 198 the gap vortex shedding. In fact, gap shedding commences approximately in the region where
 199 recirculation is suppressed on the downstream cylinder. Meanwhile, the axial flow influences the
 200 orientation of the vortical structures. The gap vortices start out closely aligned with the cylinder
 201 curvature and the axial velocity, and appear to rotate so that they become almost normal to the
 202 axial velocity in the horizontal part of the gap, corresponding to the large-scale spanwise vortices in
 203 the wake. Figure 3 indicates that the gap shedding is in phase with the large-scale wake shedding.

204 Forces and frequencies

205 The forces experienced by the two cylinders are strikingly different from each other. As shown
 206 in table 3, the drag forces on the upstream cylinder are significantly larger than on the downstream
 207 cylinder. Moreover, the downstream cylinder experiences negative drag, i.e. a thrust force, albeit
 208 very small. This is consistent with the reattachment regime of straight tandem cylinders, where,
 209 we recall, recirculation in the gap causes a negative \overline{C}_{Dd} . If we separate the pressure forces and
 210 viscous forces, we see that they are similar in magnitude, with $\overline{C}_{Ddp} \approx -0.088$ and $\overline{C}_{Ddv} \approx 0.077$.
 211 In the current study, we have not quantified the force contributions from the straight extensions, but
 212 it is a likely hypothesis that the horizontal extension is responsible for the majority of the viscous

213 drag. Conversely, there is recirculation along the entire gap between the straight vertical cylinders,
 214 as well as along nearly half of the curved gap, which results in negative pressure drag.

215 The main statistics for the curved tandem cylinder case show reasonable agreement with their
 216 straight counterparts in some parameters, shown in table 3. Note that the base pressure co-
 217 efficients, as well as the separation and reattachment angles, are computed as the z direction
 218 average along the straight vertical extension, for the present study. This is to better facilitate com-
 219 parison with straight tandem cylinders, since the values along the curved part vary considerably
 220 with the local curvature.

221 The upstream drag coefficient compares well with previous studies, although perhaps best
 222 with $Re = 1000$. The same is true for the fluctuating lift. The Strouhal number does not depart
 223 significantly from the value expected for straight tandem cylinders. For tandem cylinders, St is
 224 identical for the upstream and downstream cylinders, due to a "lock-in" effect [18]. This proves to
 225 be the case for curved tandem cylinders as well, although, as will be addressed in the discussion,
 226 there are small differences between the lower gap and the rest of the flow. The separation and
 227 reattachment angles, based on the zero shear stress criterion, also compare reasonably well,
 228 though there are few studies that provide this data.

229 The downstream force coefficients differ substantially from straight tandem cylinders. The
 230 reason for the discrepancy is twofold. Firstly, there is a significant positive contribution to the drag
 231 from the horizontal extensions, as well as from the part of the curved cylinder where there is gap
 232 shedding. These nearly balance the negative pressure drag from the reattachment region. All
 233 other studies in table 3 fall within the reattachment regime, and thus have negative C_{Dd} . Secondly,
 234 the vortex shedding strength of the downstream curved cylinder is weakened by the axial flow,
 235 causing smaller fluctuating lift, and possibly influencing the drag as well.

236 The value of the net vertical force is $\bar{C}_{zu} = 0.1854$ and $\bar{C}_{zd} = -0.0293$ for the upstream and
 237 downstream cylinder, respectively. \bar{C}_{zu} corresponds well with single curved cylinder results, as
 238 shown in table 1, although it is somewhat smaller in magnitude. At first glance, the observation that
 239 \bar{C}_{zd} should be negative appears somewhat peculiar. For a single curved cylinder, we assume that
 240 there are two main factors that ensure a positive net vertical force: backwards slanted spanwise
 241 vortices that give a vertical component, and the upwelling in the near wake. However, the upwelling
 242 velocities are very small and contribute primarily to the viscous forces. Thus, their contribution is
 243 expected to be nearly negligible. The negative vertical force component is mainly created by the
 244 induced downdraft.

245 For the curved tandem cylinders, in the part of the gap that falls under the reattachment regime,
 246 there is formation of quasi-steady vortices, similar to those reported by previous straight tandem
 247 cylinder studies [27–29]. These, as well as the shed gap vortices, align with the axial curvature of
 248 the cylinders, giving a net vertical force in the curved part of the gap. From figure 4c, we see that
 249 the vortices create a suction zone in the lower part of the gap, whose magnitude and extent are
 250 larger than those of the suction zone in the wake. The resulting pressure gradient contributes to a

251 positive vertical force for the upstream cylinder, and a negative vertical force for the downstream
252 cylinder.

253 The lift force on the upstream cylinder is very low compared to that on the downstream cylinder.
254 In figure 5a, it is difficult to discern the large-scale wake shedding frequency, f_v , from other peaks,
255 due to its low energy. This is probably due to the low energy of the quasi-steady gap vortices.

256 A portion of the upstream drag coefficient time trace is shown in figure 6a. There are low-
257 frequency undulations in the drag forces, with a period of some 60 time units. The same type of
258 time variation is visible in the downstream drag, in figure 6b, as well the vertical forces of both
259 cylinders (not shown). The undulations are in-phase for the cylinders. In a previous study, low-
260 frequency variation of the forces was related to the splitting and merging of spanwise vortices
261 in the wake [16], which resulted in two dominant vortex shedding frequencies. Two dominant
262 frequencies are not found in the present study, although spanwise vortex dislocations do occur
263 quite frequently, as shown in figure 6c. However, there appears to be no obvious connection be-
264 tween these and the low-frequency undulations in the drag coefficient. A possible explanation is
265 a low-frequency variation of the location of gap shedding inception. For straight tandem cylinders,
266 co-shedding is associated with larger drag for the upstream cylinder and positive drag coefficient
267 for the downstream cylinder. An upwards movement of the gap vortex shedding, with a corre-
268 sponding shortening of the reattachment range, would intuitively cause a surge in drag for both
269 cylinders.

DISCUSSION

270 The critical spacing is somewhat hard to define for this geometry, as both the effective gap ratio
271 and the cross-sectional geometry changes with the curvature. In any given z/D plane along the
272 curved section, the cross-sectional geometry is no longer cylindrical, but elliptical, with different
273 streamwise lengths for the upstream and downstream cylinder. This implies that the classical
274 definition of the gap ratio is no longer sensible. It would perhaps be more fruitful to characterize
275 the advent of gap-shedding/co-shedding by a critical angle, for a given Reynolds number and
276 nominal gap ratio. For straight tandem cylinders, the critical gap ratio decreases with increasing
277 Reynolds number. Given that the inflow is parallel to the straight horizontal extensions, the critical
278 angle is expected to decrease as the Reynolds number increases.

279 For straight tandem cylinders, transition between reattachment and co-shedding is associated
280 with bistable flow, where co-shedding occurs intermittently [27]. For a symmetrically curved cylin-
281 der, it was found that intermittent transition occurred for $Re = 100$, at spacing ratios corresponding
282 to a gap ratio in the range $3.76 \leq L/D \leq 4.53$ [22]. It was suggested that the induced axial flow
283 was the perturbation that caused the switch, which was associated with a non-dimensional fre-
284 quency of 0.0061. With this in mind, and given that the flow varies from overshoot/reattachment
285 to co-shedding, it seems logical that bi-stability may occur in the present study, although we have

286 not observed this directly. For straight tandem cylinders, bi-stability manifests itself in a secondary
287 peak in the velocity spectra, with a frequency close to the single cylinder St , for a given Reynolds
288 number [18, 27]. Such a peak is not immediately apparent in the velocity spectra in the present
289 study. However, in the curved part of the gap, the dominant frequency changes to 0.162. This
290 frequency is also in evidence in the vertical velocity component spectra for probes in the straight
291 part of the gap, as exemplified in figure 7b. It is likely that the gap shedding frequency is influenced
292 by the quasi-steady gap vortices, which share St with the wake shedding. This would explain why
293 the gap shedding does not jump to the St of a single curved cylinder, the way co-shedding causes
294 a jump in St for straight tandem cylinders. We note that the spectra from the symmetrically curved
295 cylinder study ([22]-figure 21) also lack a secondary peak, although bi-stability was confirmed
296 visually.

297 There is a possibility that the low-frequency variation of the forces, which we have already
298 attributed to a change in the position of the gap shedding, can be linked to the bi-stability phe-
299 nomenon. In previous studies, two bi-stable modes were found for straight tandem cylinders: one
300 of short duration, and one where the duration was "very long" [27]. The duration of the bi-stable
301 flow patterns was seen to increase when the critical spacing was approached. This means that
302 for straight tandem cylinders near the critical spacing, the secondary peak in the velocity spectra
303 should have a magnitude close to that of the dominant peak. In the present study, the velocity time
304 traces display the same low-frequency variation as the forces, and the surges in the force com-
305 ponents are associated with surges in vertical gap velocities as high up as $z/D = 2.0$ (see figure
306 7a). Spectra of the vertical component do display a secondary peak near $fU_0/D = 0.162$, which
307 increases in magnitude as we move further into the curved gap. This supports the hypothesis that
308 the low-frequency variation is related to bi-stability.

309 Despite the $7D$ vertical extension, there is some influence from the top boundary, which is
310 visible as an unphysical bump in the $U/U_0 = 0$ contour in the upper part of figure 4a. This
311 indicates that further studies of curved tandem cylinders require an investigation into the effect of
312 the vertical extension length.

SUMMARY AND CONCLUSIONS

313 The flow around curved tandem cylinders has been investigated for the first time, using DNS.
314 The inflow was parallel to the plane of curvature, and the convex configuration was chosen. The
315 Reynolds number was 500. Similar to single curved cylinders, there are significant negative ver-
316 tical velocities due to the curvature, and the wake is highly three-dimensional. The wake vortex
317 shedding is in-phase along the span, with a slight backwards slanting of the vortex lines.

318 It was found that, due to the gradual change in effective gap ratio along the curved part of the
319 cylinder, several tandem flow regimes co-exist in the flow. Along the straight vertical extension,
320 there is alternating overshoot/reattachment, which changes to stable reattachment and, finally, gap

321 shedding in the curved gap. Because recirculation is suppressed for the downstream cylinder in
322 the region where vortices are shed from the upstream cylinder, the term gap shedding is adopted
323 instead of co-shedding, which would be inaccurate.

324 Due to reattachment, there is suction in a large portion of the gap, which leads to a state of
325 near-zero drag for the downstream cylinder. Conversely, the lift forces on the upstream cylinder
326 are very small, due to the weakness of the quasi-steady gap vortices.

327 There is a significant positive vertical force on the upstream cylinder, which is comparable to
328 the values for single curved cylinders at the same Reynolds numbers. Meanwhile, the downstream
329 cylinder experiences very small, but negative, vertical forces. This is because positive contribu-
330 tions to the vertical forces by the slanted wake vortices are balanced by the negative contribution
331 from the vortices in the gap, as well as downdraft induced by the axial curvature.

332 Because the effective gap ratio, and hence the tandem flow regime, varies along the cylinder
333 axis, we suggest that instead of using the term critical spacing to describe the transition to shed-
334 ding in the gap, the concept of a critical angle should be used. Based on the behavior of straight
335 tandem cylinder flow, the critical angle is expected to decrease with increasing Reynolds number,
336 for a given nominal gap ratio.

337 A low-frequency variation of the drag and vertical force is observed, and this is attributed to
338 a slow variation of the gap shedding inception angle. A smaller angle, resulting in shedding in a
339 larger portion of the gap, is associated with increased drag for the upstream cylinder, and drag
340 inversion for the downstream cylinder. We believe that this slow variation of the gap shedding is
341 related to bi-stability of the flow near the critical angle.

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FIGURES

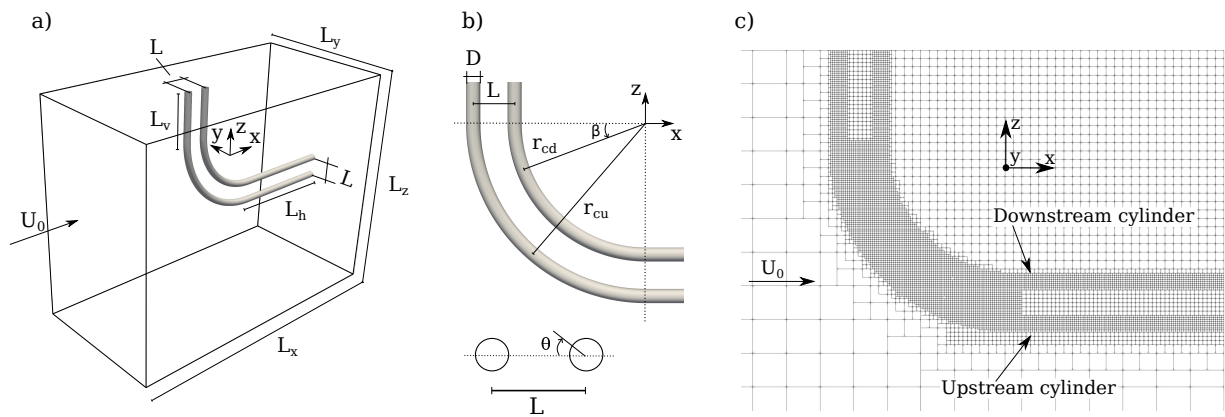


Fig. 1. a) Computational domain and geometry b) coordinate system and definitions c) schematic of the refinement regions of the computational grid. The origin is placed at the center of curvature of the cylinders.

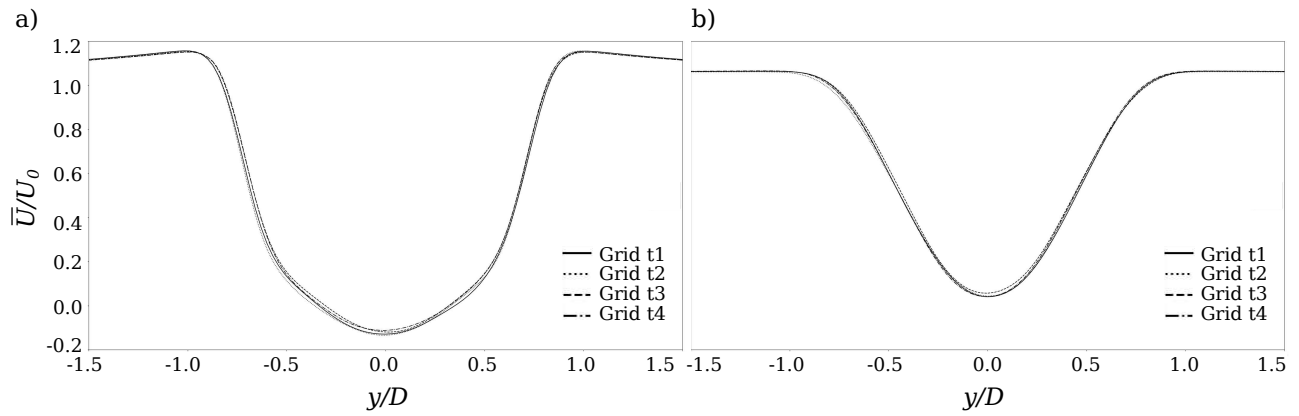


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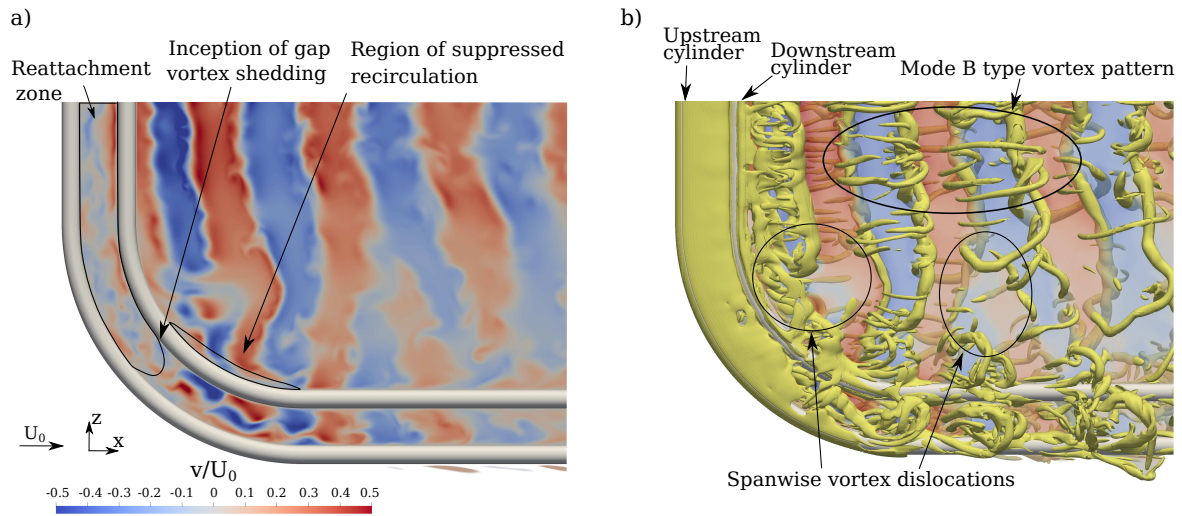


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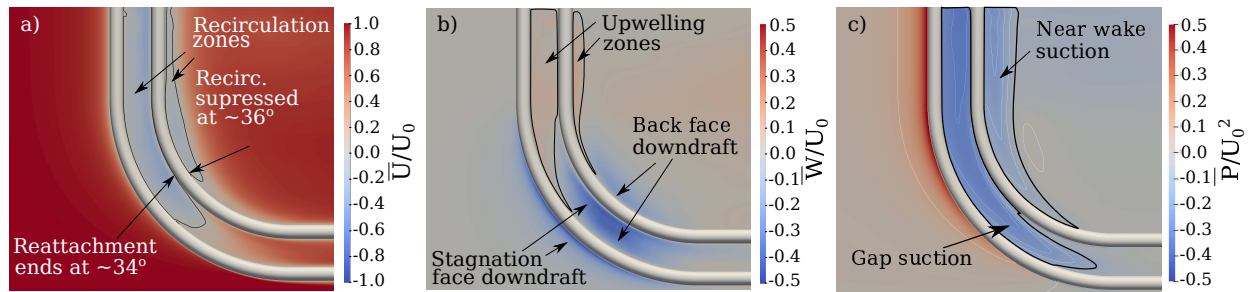


Fig. 4. Time-averaged flow field. a) streamwise and b) vertical velocities, and c) pressure.

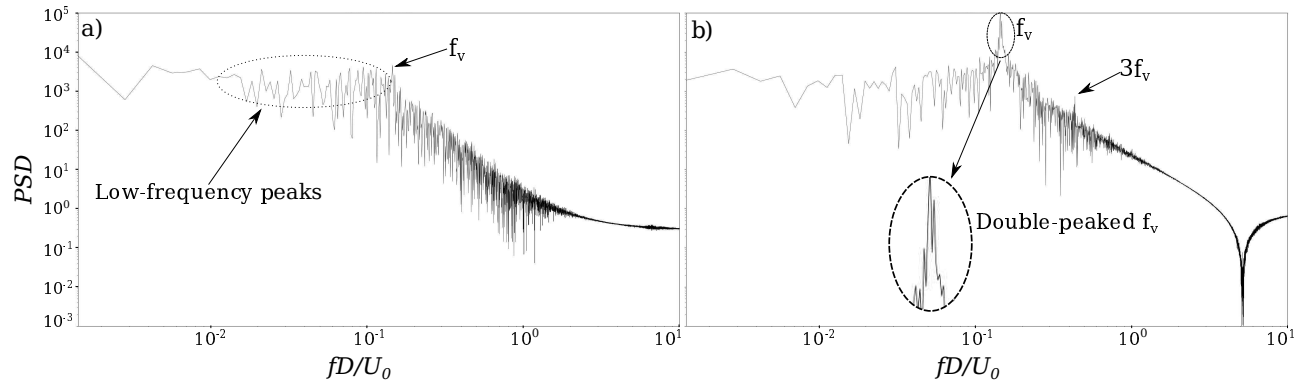


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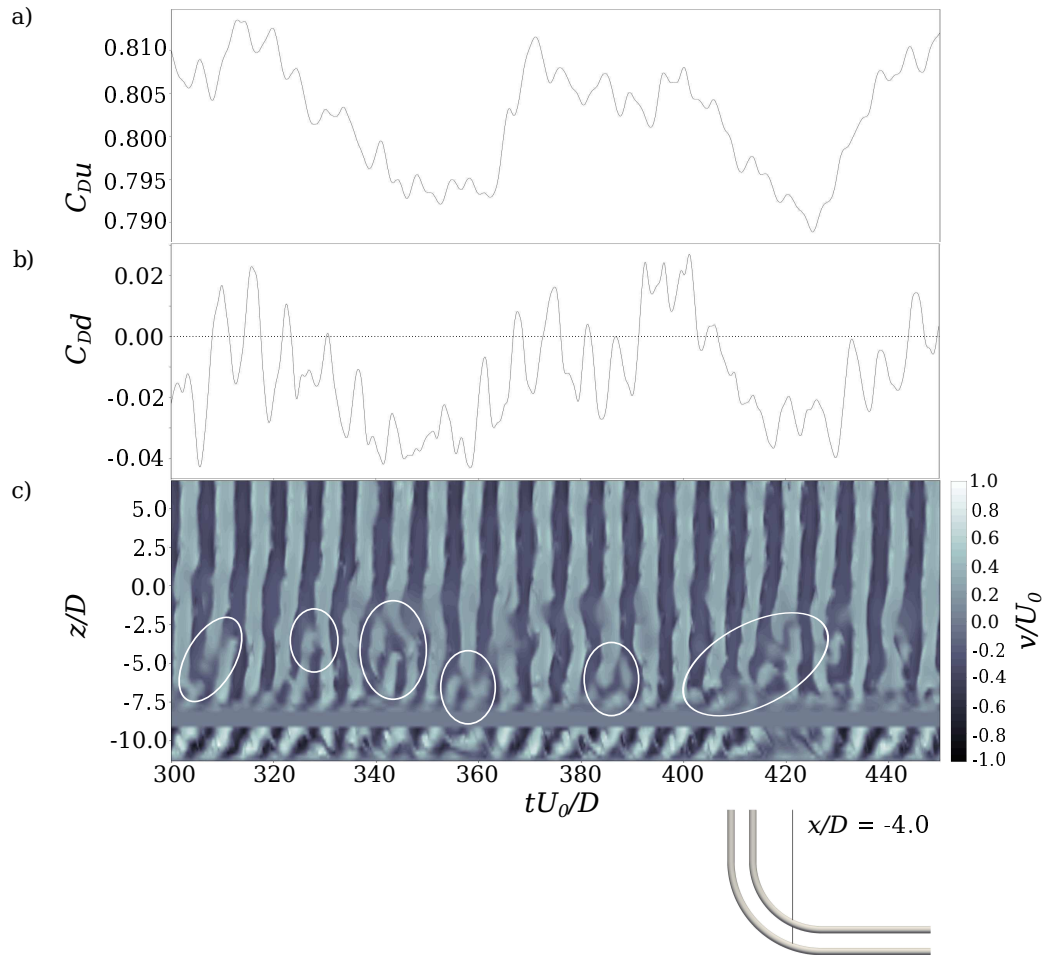


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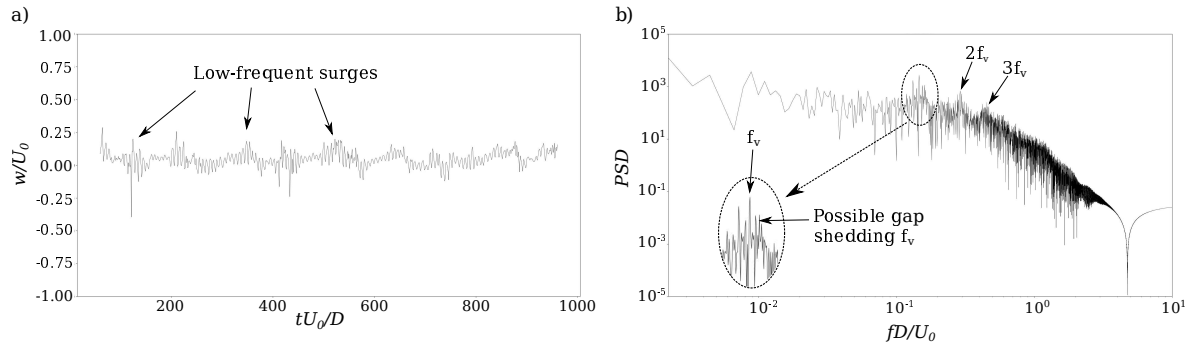


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TABLES

	Δ_{min}/L	no. elem.	r_c/D	Re	\bar{C}_D	C_{Lrms}	\bar{C}_z	St	Method
r9.5	0.0125	99M	9.5	500	1.001	0.130	0.233	0.213	DNS
r12.5	0.0125	106M	12.5	500	1.001	0.149	0.247	0.208	
[21]*			12.5	500	1.310	0.549		0.204	DNS
[16]			12.5	3900	0.742	0.017	0.182	0.213/0.223	DNS
[5]			12.5	500	0.872		0.343		LES
[4]			12.5	500	0.92		0.380		
[6]			19.1	458				0.155	Exp.

Table 1. Results from the single curved cylinder grid study, compared with convex cases in the literature. M denotes million. *Data from a single curved cylinder validation case.

Grid	Δ_{min}/L	no. elem.	\overline{C}_D	C_{Lrms}	\overline{C}_z	St
<i>Upstream cylinder</i>						
t1	0.015	104 M	0.8144	0.0192	0.1897	0.148
t2	0.0125	216 M	0.8079	0.0205	0.1830	0.152
t3	0.0075	364 M	0.8026	0.0180	0.1896	0.148
t4	0.0075	529 M	0.7995	0.0209	0.1854	0.152
<i>Downstream cylinder</i>						
t1	0.015	104 M	-0.01359	0.1603	-0.0311	0.148
t2	0.0125	216 M	-0.0166	0.1561	-0.0304	0.152
t3	0.0075	364 M	-0.0113	0.1556	-0.0331	0.148
t4	0.0075	529 M	-0.0112	0.1545	-0.0293	0.152

Table 2. Main statistics from curved tandem cylinder grid study

	Upstream cylinder				Downstream cylinder					St
	\overline{C}_{Du}	C_{Lrmsu}	$-\overline{C}_{pbu}$	θ_u [deg]	\overline{C}_{Dd}	C_{Lrmsd}	$-\overline{C}_{pbd}$	θ_d [deg]	θ_r [deg]	
present study	0.7995	0.0209	0.69	98.05	-0.0112	0.1545	0.49	126.87	68.9	0.152
<i>Straight tandem cyl. studies:</i>										
$Re = 500, L/D = 2.5$ [30]	0.958				-0.142					0.150
$Re = 500, L/D = 3.5$ [30]	0.894				-0.126					0.144
$Re = 500, L/D = 3.0$ [18]										0.168
$Re = 500, L/D = 3.0$ [31]	1.12				-0.25					
$Re = 1000, L/D = 3.0$ [32]	0.88	0.03	0.63	92.5	-0.15	0.34	0.42	125	67	0.149
$Re = 2.2 \times 10^4, L/D = 3.0$ [29]	0.80	0.02	0.6		-0.20	0.3	0.4		70	0.155
$Re = 4.0 \times 10^4, L/D = 3.2$ [33]							0.45	120	67.2	0.144
$Re = 1.57 \times 10^5, L/D = 2.0$ [34]		0.1	0.9			0.7	0.6		60	
$Re = 1.57 \times 10^5, L/D = 3.0$ [34]		0.02	0.75			0.48	0.49		none	

Table 3. Main statistics for curved tandem cylinders compared with straight tandem cylinders from the literature. θ_u and θ_d denote the primary separation angle of the upstream and downstream cylinders, respectively, and θ_r denotes the reattachment angle.