

Numerical simulation of turbulent duct flows with constant power input

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(Received 1 December 2013; revised 30 March 2014; accepted 8 May 2014;
first published online 2 June 2014)

The numerical simulation of a flow through a duct requires an externally specified forcing that makes the fluid flow against viscous friction. To this end, it is customary to enforce a constant value for either the flow rate (CFR) or the pressure gradient (CPG). When comparing a laminar duct flow before and after a geometrical modification that induces a change of the viscous drag, both approaches lead to a change of the power input across the comparison. Similarly, when carrying out direct numerical simulation or large-eddy simulation of unsteady turbulent flows, the power input is not constant over time. Carrying out a simulation at constant power input (CPI) is thus a further physically sound option, that becomes particularly appealing in the context of flow control, where a comparison between control-on and control-off conditions has to be made. We describe how to carry out a CPI simulation, and start with defining a new power-related Reynolds number, whose velocity scale is the bulk flow that can be attained with a given pumping power in the laminar regime. Under the CPI condition, we derive a relation that is equivalent to the Fukagata–Iwamoto–Kasagi relation valid for CFR (and to its extension valid for CPG), that presents the additional advantage of naturally including the required control power. The implementation of the CPI approach is then exemplified in the standard case of a plane turbulent channel flow, and then further applied to a flow control case, where a spanwise-oscillating wall is used for skin-friction drag reduction. For this low-Reynolds-number flow, using 90% of the available power for the pumping system and the remaining 10% for the control system is found to be the optimum share that yields the largest increase of the flow rate above the reference case where 100% of the power goes to the pump.

Key words: drag reduction, Navier–Stokes equations, turbulence simulation

1. Introduction

Fluids flow through a duct upon external actions (pressure differences, forces). For the numerical simulation of a duct flow, one needs to mimic these physical effects by

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supplementing the governing equations with a forcing that drives the flow through the duct against viscous drag, which in the case of a straight duct only arises from friction. In practice, unless the wall moves and produces a shear-driven flow, this forcing is imposed by selecting a constant (in time) value for either the pressure gradient or the flow rate. We define the former the constant pressure gradient (CPG) and the latter the constant flow rate (CFR) approach. The two approaches are trivially identical in the laminar flow regime, where flow rate and pressure gradient are uniquely related.

For steady laminar flows, however, the difference between CFR and CPG becomes visible whenever a comparison between two flows is done. If, for example, an obstacle is placed in a duct flow (like an orifice in a hydraulic pipeline system or a stenosis growing in an artery), the changed flow state can be compared to the original one under the assumption that both possess the same flow rate or the same pressure gradient. The corresponding result would of course differ, since the additional flow resistance would lead to an increase of required pressure gradient in the case of CFR or to a decrease of achieved flow rate in the case of CPG. A consequence of this difference is sketched in figure 1, where it is schematically shown how the length of the recirculation bubble after the obstacle obtained by CFR is larger than that obtained by CPG.

When direct numerical simulation (DNS) of the Navier–Stokes equations, or large-eddy simulation (LES), is used to numerically simulate an unsteady turbulent flow even in a simple straight duct, CPG results in a flow rate that fluctuates in time, with a well-defined time average under equilibrium conditions. Similarly, CFR leads to temporal fluctuations of the space-averaged pressure gradient which, after appropriate time averaging, becomes constant under equilibrium conditions. Such temporal fluctuations are known to depend on the computational domain size, and become smaller for larger domains (Lozano-Durán & Jiménez 2014). In any case, CFR and CPG result in the same time-averaged values for pressure gradient and flow rate respectively, provided the integration time is long enough. However, an important difference between the two approaches exists in terms of power input. The power input into the system (which physically might come from the action of a pump) is given by the product of pressure gradient and flow rate. When either pressure gradient or flow rate change in time, the power input is not instantaneously constant. Although this can be regarded as a minor issue in DNS, the instantaneous fluctuations of the power input under CFR and CPG are larger in LES owing to the larger time step size, and might be of some concern if a constant filter width is used in the definition of the subgrid model. Indeed, the fluctuations in the energy flux are reflected in corresponding fluctuations of the filter length scale expressed in viscous (Kolmogorov) units, while the filter width is usually set statically, once and for all, in relation to the mesh size.

In the context of flow control, where changes in the flow resistance are introduced by design, CFR has been more often employed than CPG, since evaluating control performance with the resultant drag reduction is more intuitive and the initial transient period to reach an equilibrium state after the onset of control is shorter for CFR than for CPG. From a fundamental viewpoint, the choice between CFR and CPG becomes interesting when the question arises of how to meaningfully compare the controlled flow field to the reference flow without control. Owing to the action of control, one of the several Reynolds numbers by which the flow can be characterized is modified: using for example a skin-friction drag-reduction technique under CFR has the effect of reducing drag (hence reducing the value of the friction-based Reynolds number, Re_τ , at constant bulk velocity Reynolds number, Re_b); however, using the

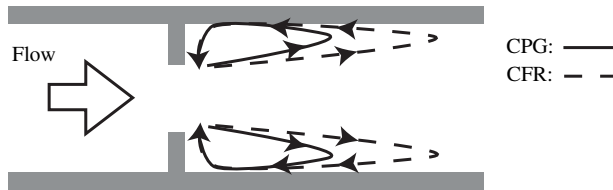


FIGURE 1. Example sketch of the difference between CFR and CPG when comparing the steady laminar flow in a straight reference pipe to the flow in the same pipe with an obstacle. This might be for example the study of a stenotic flow in an artery. Depending on whether the mass flow or the pressure gradient of the reference flow is kept constant, the details of the recirculating bubble after the obstacle are different; in particular the recirculating bubble is longer with CFR.

same technique under CPG cannot result in drag reduction, since the drag is constant by definition and the drag-reducing effect manifests itself in an increased flow rate (larger value of Re_b at constant Re_τ). In this case the energy flux (power input) is not constant even in the time-averaged sense: for the controlled flow under CFR the energy flux decreases (pressure gradient decreases and flow rate is unchanged), whereas under CPG the energy flux increases (pressure gradient is unchanged and flow rate increases). It therefore becomes difficult to make any definite statement about the behaviour of energy dissipation, i.e. the mechanism by which we lose energy, since the sign of the global change of dissipation is simply prescribed by the simulation strategy (Ricco *et al.* 2012). In the end, the physical interpretation of results can be quite different depending on which numerical condition is employed to run the simulation. One example is given in figure 2, which shows how the drag-reduction technique known as the spanwise-oscillating wall (Jung, Mangiavacchi & Akhavan 1992) affects the root-mean-square (r.m.s.) intensity of the streamwise velocity fluctuations in a turbulent channel flow. It can be seen that, depending on whether the simulation is run under CPG or CFR (and, in the latter case, depending on whether the friction velocity of the controlled or uncontrolled flow is used to plot the data in viscous units), the global picture of how the oscillations of the wall affect turbulence near the wall can be significantly different: one might correctly conclude that turbulence intensity is strongly suppressed by the oscillating wall, or that it is essentially unchanged, with perhaps only the profile being slightly shifted outwards.

In this paper we propose a strategy to carry out the numerical simulation of duct flow with the power input, i.e. the energy flux through the system, kept constant. We call this strategy the constant power input (CPI) approach, an alternative to CFR and CPG. This alternative strategy naturally arises from the money-versus-time plane, a methodology for assessing flow control techniques for skin-friction drag reduction, which we have recently proposed (Frohnäpfel, Hasegawa & Quadrio 2012). In the money-versus-time plane, the drag reduction problem is generally formulated as finding a compromise between energy consumption and the time required to transport a given amount of fluid over a unit distance, i.e. the inverse of the flow rate. These two essential quantities are plotted on the vertical and horizontal axes of the plane as shown in figure 3. The goal of flow control is to modify a flow state towards the origin of the plane (less energy consumption and more convenience) by manipulating turbulence. By plotting first in figure 3(a) the pumping energy versus the inverse of the bulk velocity, an uncontrolled fully turbulent flow state is shown to be shifted by

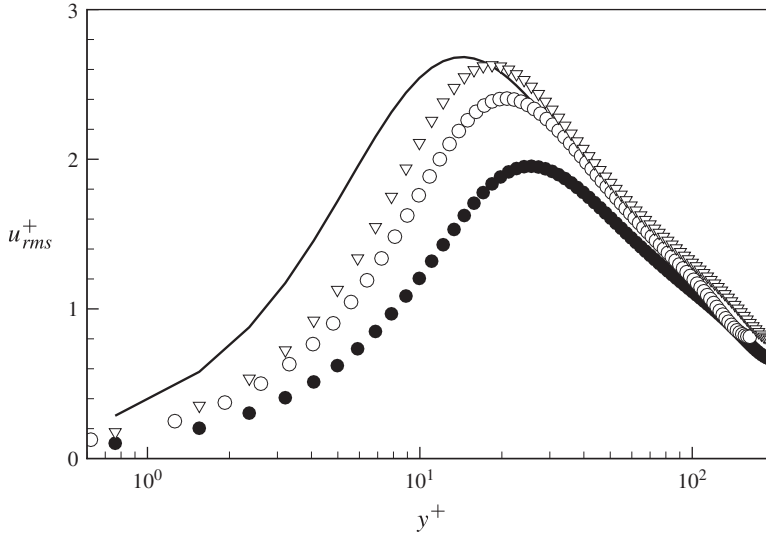


FIGURE 2. Wall-normal distribution of the r.m.s. value of the streamwise velocity fluctuations: effect of a spanwise-oscillating wall. Data from a turbulent channel flow at $Re_\tau = 200$ ($Re_b = 3173$) taken from Quadrio & Ricco (2011). Continuous line is the reference flow, and symbols correspond to the spanwise-oscillating wall operating with oscillation amplitude $W_0^+ = 12$ and nearly optimal oscillating period $T^+ = 100$: triangles, CPG calculation; circles, CFR calculation (for filled circles the friction velocity of the reference flow has been used for non-dimensionalization, while for open circles the actual friction velocity in the drag-reduced state was used).

successful control to two different end points (both somewhat nearer to the laminar state) under CFR and CPG. Note that the pumping energy per unit fluid volume is proportional to the pressure gradient (Frohnapfel *et al.* 2012), and consequently the flow states under CPG move along a horizontal line as shown in figure 3(a). Since in this plane any straight line starting from the origin connects points with the same pumping power input, control under CPI would therefore move from the uncontrolled flow state along a straight line towards the origin into a third, different final state. In a next step, sketched in figure 3(b), the pumping energy is replaced by the sum of pumping and control energy on the vertical axis, in order to account for the energetic cost of active flow control techniques. Again, it is shown how flow control applied under CFR and CPG shifts the operating point. The straight line between a turbulent reference point and the origin now connects all points with constant total power input (CtPI). In order to account for the difference of these two planes represented in figure 3 we distinguish between constant pumping power input (CpPI) and CtPI in the following. The latter (CtPI) is a specific approach for the evaluation of active flow control techniques; it can be further generalized to account for costs that are related to introducing changes into the fluid system, like geometry or roughness changes. The former approach (CpPI) does not consider active flow control. It is an alternative simulation strategy to the conventional CFR and CPG approaches, which might be considered more adequate for specific problem formulations. For example, a flow state realized with a certain pump is described by a characteristic curve which depends on the principles of the pump employed and also its control scheme. Therefore, the possible trace of the flow state satisfies neither CFR, CPG nor CPI

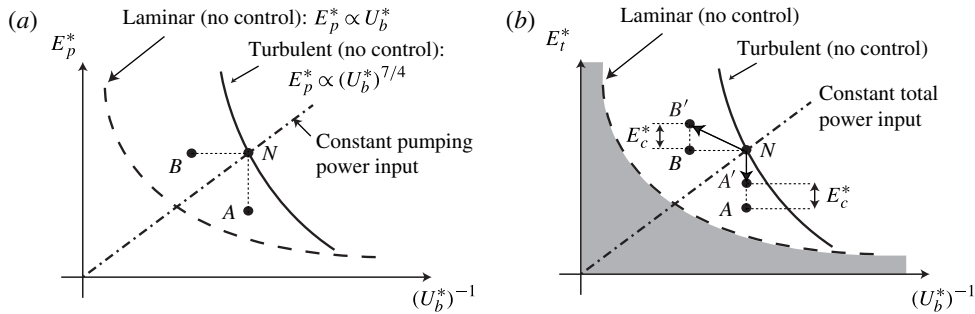


FIGURE 3. Pumping energy, E_p^* (a), and total energy, $E_t^* = E_p^* + E_c^*$ (b), plotted versus the inverse of the dimensional bulk mean velocity, U_b^* , for fully developed turbulent flow (solid line) and laminar flow (dashed line). If flow control for skin-friction drag reduction is applied at a turbulent reference point, N , the resulting shift in the depicted plane depends on the operating condition of the numerical simulation. The shift to A or A' corresponds to CFR and the one to B or B' to CPG. The vertical distances between A and A' or B and B' in (b) correspond to the energy consumption, E_c^* , of the control technique. The dash-dot lines connecting N with the origin of the coordinate system represents states of CpPI in (a) and CtPI in (b), respectively. Adapted from Frohnepfel *et al.* (2012).

in general. The characteristic curves of most existing turbopumps typically show a decreasing pressure head with increasing flow rate, while the change in the shaft power is quite small in a wide range of flow conditions. For such a case, CPI might be considered the most reasonable numerical representation.

Our paper begins by first showing in § 2 how the CPI approach can be introduced by identifying a velocity scale, and thus a related Reynolds number, based on the power consumption. The CPI approach is then discussed in the context of the so-called FIK identity (Fukagata, Iwamoto & Kasagi 2002) in § 3, where it is shown how an insightful exact relationship between wall friction (or flow rate) and the turbulent shear stresses can be obtained under CFR, CPG and CPI, and how the latter approach allows a natural consideration of the energetic cost of the control. Next, in § 4 it is shown how to carry out a numerical simulation under CpPI, whose implementation in a computer code is briefly summarized. Last, the implementation of the CtPI concept is shown in § 5 to be able to identify, for the simple example of the spanwise-oscillating wall, which percentage of the available power should be used to run the control in order to achieve the maximum flow rate.

In this paper, all dimensional quantities are indicated with an asterisk. Non-dimensionalization is always based on the velocity scale used in the definition of the relevant Reynolds number. In some instances, viscous (inner) scaling is explicitly denoted with the conventional '+' superscript. The geometrically simple setup of plane channel flow that will be considered in the following is sketched in figure 4; the streamwise, wall-normal and spanwise directions are indicated with x , y and z , with u , v and w being the respective velocity components. The static pressure is p . Instantaneous wall-parallel averages of fluid quantities are denoted, for example, by $[u]$, $[p]$ and so forth, which are in general a function of y . The volume average is denoted by $\{\cdot\}$, so that $\{u\}$ corresponds to the instantaneous bulk velocity. The time average is applied to statistically steady flows. The combination of spatial and temporal averaging is indicated by angular brackets: $\langle u \rangle$.

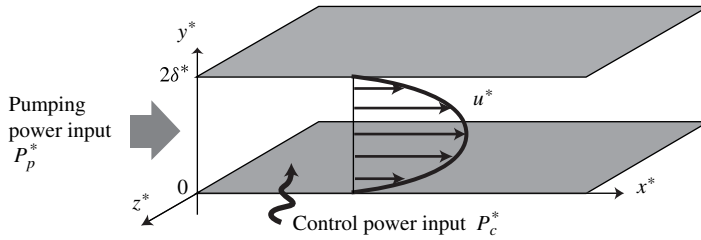


FIGURE 4. Sketch of the plane channel flow geometry. The two walls are located at $y^* = 0$ and $y^* = 2\delta^*$. The sketch shows the power P_p^* that enters the system through pumping and the power P_c^* required by a flow control device, if present.

2. The concept of CPI

We consider one of the simplest canonical flows, namely the fully developed plane channel flow, sketched in figure 4, bounded by two infinite parallel walls, located $2\delta^*$ apart. The fluid has density ρ^* , dynamic viscosity μ^* and kinematic viscosity ν^* . In the laminar regime, a given pumping power P_p^* produces a parabolic streamwise velocity profile with a well-defined corresponding mass flow rate. The pumping power per unit wetted area is given by

$$P_p^* = -\frac{dp^*}{dx^*} \delta^* U_b^*, \tag{2.1}$$

where $-dp^*/dx^*$ is the streamwise pressure gradient and U_b^* is the bulk velocity, i.e. the volume-averaged streamwise velocity $\{u^*\}$. The value of the bulk velocity can be easily deduced by integrating in the wall-normal direction the expression for the laminar parabolic velocity profile:

$$U_b^* = \frac{1}{2\delta^*} \int_0^{2\delta^*} u^*(y^*) dy^* = \frac{1}{3\mu^*} \left(-\frac{dp^*}{dx^*}\right) \delta^{*2}. \tag{2.2}$$

In a fully developed flow, the power input to the flow balances the dissipation rate ϵ^* of the kinetic energy of the flow. The parabolic form of the velocity profile allows ϵ^* per unit wetted area to be computed as:

$$\epsilon^* = \int_0^{\delta^*} \mu^* \left(\frac{du^*}{dy^*}\right)^2 dy^* = \frac{3\mu^* U_b^{*2}}{\delta^*}. \tag{2.3}$$

By equating the dissipation rate (2.3) with the pumping power P_p^* , the following relation between the flow rate and the pumping power is found:

$$U_b^* = \sqrt{\frac{P_p^* \delta^*}{3\mu^*}}. \tag{2.4}$$

This equation identifies a velocity scale, that we indicate with the symbol U_Π^* in the following, which represents the flow rate (per unit width) achieved by a given pumping power P_p^* in the laminar regime. The flow rate achieved by the same pumping power is of course less in the turbulent regime, where U_Π^* becomes an upper bound. Moreover, theoretical arguments exist (Bewley 2009; Fukagata, Sugiyama & Kasagi 2009) showing that U_Π^* is the maximum achievable flow rate for controlled

flows if the power consumption of control is considered. The velocity scale U_{Π}^* , defined as

$$U_{\Pi}^* = \sqrt{\frac{P_p^* \delta^*}{3\mu^*}} \quad (2.5)$$

is therefore the appropriate reference velocity for the CPI approach.

Once the velocity scale U_{Π}^* is identified, it is straightforward to normalize all quantities by U_{Π}^* and the channel half-width δ^* in order to obtain governing equations for the CPI concept. In this dimensionless form the incompressible continuity and Navier–Stokes equations read:

$$\frac{\partial u_i}{\partial x_i} = 0, \quad (2.6)$$

$$\frac{\partial u_i}{\partial t} + \frac{\partial (u_j u_i)}{\partial x_j} = -\frac{\partial p}{\partial x_i} + \frac{1}{Re_{\Pi}} \frac{\partial^2 x_i}{\partial x_j \partial x_j}, \quad (2.7)$$

where the Reynolds number is defined as:

$$Re_{\Pi} = \frac{U_{\Pi}^* \delta^*}{\nu^*}. \quad (2.8)$$

Accordingly, by using (2.5) the dimensionless power input is given by:

$$P_p = \frac{P_p^*}{\rho^* U_{\Pi}^{*3}} = \frac{3}{Re_{\Pi}}. \quad (2.9)$$

Therefore, Re_{Π} defines the power input to the system and is the natural choice for a CPI simulation, just as Re_b defines the flow rate and is conveniently used in CFR, and Re_{τ} defines the pressure gradient and is used in CPG. The relation between the different Reynolds numbers and corresponding velocity scales is summarized in appendix A.

In the case of active control where the total power input to the fluid system consists of pumping and control power,

$$P_t^* = P_p^* + P_c^*, \quad (2.10)$$

(2.9) becomes

$$P_t = \frac{3}{Re_{\Pi}} \quad (2.11)$$

and states that the numerical value of Re_{Π} sets the total power input. Depending on the particular control, the relative importance of the terms P_p^* and P_c^* , i.e. the amount of the available power that is actually used for pumping and that which goes into the flow control device, is described by the quantity

$$\gamma = \frac{P_c^*}{P_t^*} = 1 - \frac{P_p}{P_t} \quad (2.12)$$

which reflects the percentage of the total power that is used for control. Accordingly,

$$P_p = P_t (1 - \gamma) = \frac{3(1 - \gamma)}{Re_{\Pi}}. \quad (2.13)$$

3. Flow control under CFR, CPG and CPI

A few years ago Fukagata *et al.* (2002) derived the so-called FIK identity, which relates the skin-friction coefficient in a fully developed channel flow to its dynamical

contributions under CFR. Since it nicely shows how the wall-normal distribution of the Reynolds shear stress contributes to the wall friction quantitatively, the FIK identity has been widely used for analysing different drag-reduction mechanisms and also for developing novel control strategies. This idea was later been extended to CPG by Marusic, Joseph & Mahesh (2007). After briefly recalling these two results, we present the corresponding formulation for the CPI case for which we also obtain the result that the success of a flow control technique in a turbulent channel flow can be quantified by the achieved reduction of the Reynolds shear stress.

3.1. CFR

The original FIK identity is valid under CFR, hence the relevant Reynolds number is $Re_b \equiv U_b^* \delta^* / \nu^*$, and velocities are scaled with the bulk velocity U_b^* . It reads:

$$C_f = \frac{6}{Re_b} + 3 \int_0^2 (1-y) \langle -u'v' \rangle dy. \quad (3.1)$$

The first term on the right-hand side corresponds to the laminar drag at the same flow rate, whereas the second term represents the turbulence contribution. Typically, since $\langle -u'v' \rangle > 0$, the turbulent shear stress enhances the friction coefficient C_f .

3.2. CPG

The corresponding relationship for CPG has been derived by Marusic *et al.* (2007). The relevant Reynolds number is $Re_\tau \equiv u_\tau^* \delta^* / \nu^*$, and velocities are scaled with friction velocity u_τ^* . It reads:

$$U_b = \frac{Re_\tau}{3} - \frac{Re_\tau}{2} \int_0^2 (1-y) \langle -u'v' \rangle dy. \quad (3.2)$$

In analogy with (3.1), the bulk mean velocity under CPG is composed of two terms, i.e. laminar and turbulence contributions. The first term on the right-hand side is equal to the laminar flow rate under the same pressure gradient. The second term represents the momentum loss due to the turbulent shear stress leading to a lower flow rate in turbulent flows. This trend is enhanced at higher Reynolds number.

3.3. CPI

Under CPI conditions the relevant Reynolds number is $Re_\Pi \equiv U_\Pi^* \delta^* / \nu^*$, and velocities are scaled with velocity U_Π^* . The averaged streamwise component of the momentum equation can be written as:

$$0 = -\frac{d \langle u'v' \rangle}{dy} - \left\langle \frac{dp}{dx} \right\rangle + \frac{1}{Re_\Pi} \frac{d^2 \langle u \rangle}{dy^2}. \quad (3.3)$$

Triple integration yields:

$$U_b = -\left\langle \frac{dp}{dx} \right\rangle \frac{Re_\Pi}{3} - \frac{Re_\Pi}{2} \int_0^2 (1-y) \langle -u'v' \rangle dy. \quad (3.4)$$

This expression for the bulk mean velocity again consists of two terms on the right-hand side: a laminar contribution and a turbulence contribution. However, in contrast to (3.1) and (3.2) the first term on the right-hand side is not defined solely by the prescribed Reynolds number but also by the resulting mean pressure gradient term. Hence, it needs to be considered further in order to obtain an expression that only contains the relevant Reynolds number and the Reynolds shear stress.

The mean pressure gradient is given by the pumping power per unit wetted area divided by the bulk velocity; using (2.13) it reads:

$$-\left\langle \frac{dp}{dx} \right\rangle = \frac{P_p}{U_b} = \frac{3(1-\gamma)}{Re_{\Pi} U_b}. \quad (3.5)$$

Notice how γ has already been introduced into the equation at this stage, since one of the CPI advantages is to be able to discriminate between pumping power and total power. Using this relationship to replace the mean pressure gradient term in (3.4) and multiplying by U_b (for $U_b \neq 0$) results in a quadratic equation for U_b :

$$U_b^2 + T U_b - (1-\gamma) = 0, \quad (3.6)$$

where the coefficient T is defined as

$$T = \frac{Re_{\Pi}}{2} \int_0^2 (1-y) \langle -u'v' \rangle dy \quad (3.7)$$

and represents the contribution of turbulence to the resulting flow rate. The solution to (3.6) is given by

$$U_b = \frac{-T \pm \sqrt{T^2 + 4(1-\gamma)}}{2} \quad (3.8)$$

and can be regarded as the FIK identity for the CPI condition.

It should be noted that, in the CPI condition, once $0 < \gamma < 1$ is given, the pumping power is fixed, whereas the flow direction is not prescribed. The pumping power being, from (2.1), the product of pressure gradient and bulk mean velocity explains why two solutions appear in (3.8): one corresponds to the case where both the bulk mean velocity and the pressure gradient, are positive, and the other to the case where they are both negative. However, choosing the x axis such that the mean flow is oriented in its positive direction, as figure 4 suggests, implies that one of the two solutions in (3.8) is selected, namely that with the positive sign, to satisfy (3.6).

It is evident that for a standard laminar flow, where the power input for control and the turbulence contribution vanish, namely $\gamma = 0$ and $T = 0$, (3.8) reduces to $U_b = 1$ or $U_b^* = U_{\Pi}^*$, which is the upper limit for the bulk velocity under the CPI condition. When the flow is turbulent, the Reynolds shear stresses are typically negative, i.e. $\langle -u'v' \rangle > 0$ so that $T > 0$, thus leading to $U_b < 1$. In the limiting case where the turbulence contribution becomes infinitely large, i.e. $T \rightarrow \infty$, U_b tends to zero. Another limiting behaviour is the case with $\gamma = 1$, i.e. all available power goes into the control system. In this case, the quadratic equation (3.6) becomes singular, so that it reduces to the original linear equation (3.4), which possesses the unique solution $U_b = -T$. Depending on the particular control strategy, some flow rate may be produced by control through $T \neq 0$, like for example the forcing scheme devised by Min *et al.* (2006), where a viscous streaming mechanism provides the equivalent of a pumping action.

4. Implementing CPI into a DNS code: CpPI

We consider first the no-control case, where the power input to the system is entirely given by the pumping power, i.e. $\gamma = 0$, and describe how to implement the CpPI concept into a DNS or LES code in the simple geometry of plane channel flow.

In a CPI simulation the pumping power P_p^* is prescribed, and set by the value of Re_Π . Compared to a laminar simulation, both the space-averaged instantaneous streamwise pressure gradient $-\langle \partial p / \partial x \rangle$ and the instantaneous flow rate $\{u\}$ may fluctuate around their mean values G and U_b , but their product sets the instantaneous value of the pumping power and must consequently remain constant to satisfy (2.9). Both $-\langle \partial p / \partial x \rangle$ and $\{u\}$ are thus computed as time-dependent quantities during the simulation.

However, it is not easy to satisfy the CpPI condition exactly at every time step, since the flow state is obtained as a result of the computation. Hence, we follow the simplest approach in a time-discrete setting, where the equations of motion are advanced by a typically very small computational time step. To advance the solution from time step n to time step $n + 1$, the pressure gradient is determined as:

$$-\left[\frac{\partial p}{\partial x}\right]^{(n+1)} = \frac{P_p}{\{u\}^n} = \frac{3}{Re_\Pi \{u\}^n} \tag{4.1}$$

and this enables advancing the solution by one time step to compute $\{u\}^{n+1}$. Note that the above scheme is first-order accurate in time. In the present study, this is sufficient, since the estimation error of $\{u\}^{n+1}$ is commonly of the order of 10^{-8} . This indicates that the CPI condition is satisfied within the same order. The order of accuracy can be increased in a straightforward way by using past variables. For second-order accuracy, $u^{n+1} \approx 2u^n - u^{n-1}$ can be used. We confirmed that the increased accuracy of the numerical scheme does not affect the present results.

When the flow reaches an equilibrium state, the mean pressure gradient and the wall friction must balance:

$$-\left\langle \frac{dp}{dx} \right\rangle = G = \frac{1}{Re_\Pi} \left. \frac{d\langle u \rangle}{dy} \right|_{y=0}. \tag{4.2}$$

This leads to

$$P_p = GU_b = \frac{U_b}{Re_\Pi} \left. \frac{d\langle u \rangle}{dy} \right|_{y=0}. \tag{4.3}$$

Here, it is assumed that the temporal correlation between G and U_b is negligibly small. This is generally valid in a sufficiently large computational domain except for the special case where the pressure gradient is actively varied in time, so that the correlation between G and U_b becomes significant.

Since $P_p = 3/Re_\Pi$ from (2.9), the following relationship is obtained:

$$U_b \left. \frac{d\langle u \rangle}{dy} \right|_{y=0} = 3 \tag{4.4}$$

which is exact and can be used to verify that the time averaging is sufficient.

The friction coefficient C_f can be defined in terms of pumping power by:

$$C_f = \frac{P_p^*}{\frac{1}{2}\rho^* U_b^{*3}} \tag{4.5}$$

and by using (2.4) it can be expressed as:

$$C_f = \frac{6\mu^* U_\Pi^{*2}}{\rho^* \delta^* U_b^{*3}} = \frac{6}{Re_\Pi} \left(\frac{U_\Pi^*}{U_b^*} \right)^3. \tag{4.6}$$

When the flow is laminar, $U_b^* = U_\Pi^*$ and thus $Re_b = Re_\Pi$; in this case the above expression reduces to the conventional laminar formula $C_f = 6/Re_b$.

Example

We now present the results of a sample CpPI calculation of (uncontrolled) turbulent channel flow. The statistics are the result of a DNS, carried out by adapting an existing DNS code to the CPI procedure. The code (Luchini & Quadrio 2006) solves the incompressible Navier–Stokes equations, written in terms of wall-normal velocity and wall-normal vorticity components, by a pseudo-spectral method, where Fourier expansions are employed in the streamwise and spanwise directions, and compact fourth-order explicit finite-difference schemes are used in the wall-normal direction. Full details can be found in the original paper. The code changes that we had to implement in order to enable the simulation to run under CPI are extremely limited (five lines of source code).

The streamwise and spanwise dimensions of the computational domain are set at the standard values of $L_x^* = 4\pi\delta^*$ and $L_z^* = 2\pi\delta^*$, as in Kim, Moin & Moser (1987). We set the value of the Reynolds number based on the total power input at $Re_\Pi = 6500$, which yields $Re_\tau \approx 200$. The number of modes employed in the streamwise, wall-normal and spanwise directions are $(N_x, N_y, N_z) = (256, 128, 256)$, respectively. If expressed in wall units, the spatial resolution of $\Delta x^+ = 9.8$, $\Delta z^+ = 4.9$ and $\Delta y^+ = 0.9\text{--}4.9$ is within the usually accepted values (Hoyas & Jiménez 2008). Time integration is carried out via a classic partially implicit three-stage Crank–Nicholson Runge–Kutta scheme; integration time is $2000\delta/U_\Pi$, corresponding to more than 12 000 viscous time units.

Figure 5 demonstrates the temporal behaviour of the flow rate and the pressure gradient. They are normalized with their time-averaged values U_b and G . The figure clearly shows that the product of the two quantities is actually kept constant at every time instant during the simulation. The product of the corresponding bulk velocity and wall shear stress is 2.99914, which indicates that (4.4) is satisfied to a very high degree. Additionally, it can be also appreciated that the time scale of variations of both quantities is $O(100)$ and thus far larger than the time step of the simulation, which is of the order of 10^{-2} .

The resultant bulk and friction Reynolds numbers are found to be $Re_b = 3176$ and $Re_\tau = 199.7$, respectively. The mean velocity and r.m.s. values of velocity fluctuations obtained, normalized by U_Π^* , are shown in figure 6. The right-hand vertical axis shows the same quantities made dimensionless with the conventional viscous wall units, to highlight the scale factor involved.

5. Constant total power input (CtPI)

Active flow control techniques require power to operate. As described in § 2, the additional control power P_c^* is a further contribution to the total power input P_t^* besides the pumping power P_p^* .

The procedure previously sketched in § 4 to adjust flow rate and pressure gradient during the simulation must be adapted as follows. Once the bulk velocity $\{u\}^n$ and the control power P_c^n are known at the time step n , the pumping power at time step $n + 1$ is determined as $P_p^{n+1} = P_t - P_c^n$, and using (2.13), the pressure gradient is determined as

$$-\left[\frac{\partial p}{\partial x}\right]^{(n+1)} = \frac{P_p^{n+1}}{\{u\}^n} = \frac{P_t - P_c^n}{\{u\}^n} = \frac{3(1 - \gamma)}{Re_\Pi \{u\}^n}. \quad (5.1)$$

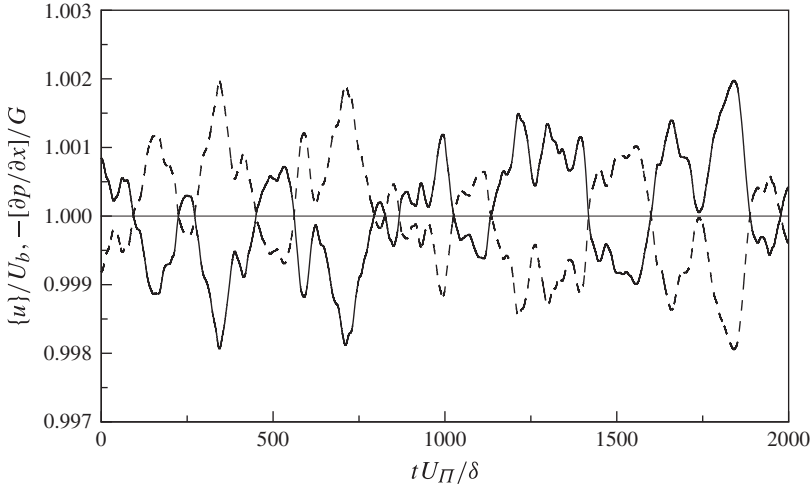


FIGURE 5. Temporal behaviour of the space-mean instantaneous bulk velocity $\{u\}$ (thick solid line) and the space-mean instantaneous pressure gradient $-\partial p/\partial x$ (dashed line), normalized with their time-average values U_b and G . The thin horizontal line is the product of the two quantities, proportional to the instantaneous power input, which remains constant during the simulation.

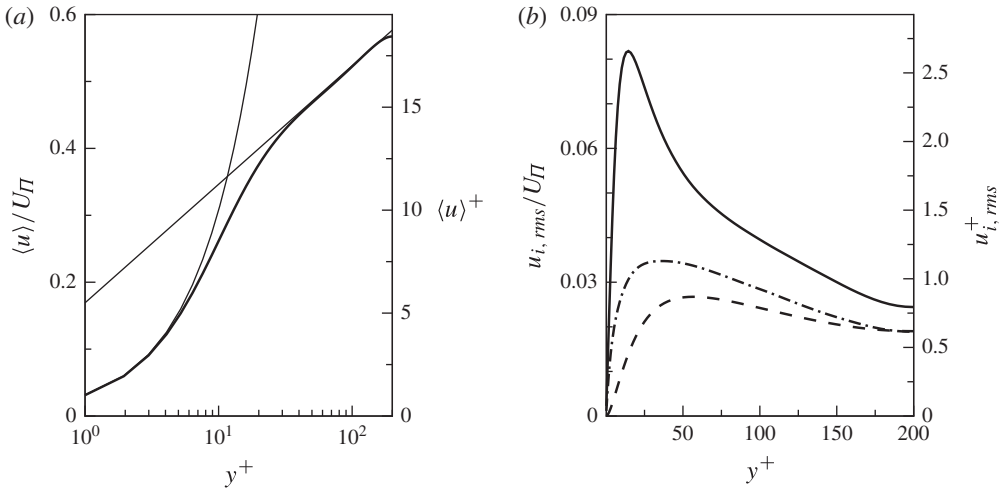


FIGURE 6. (a) Mean velocity profile $\langle u \rangle$ and (b) wall-normal distribution of the r.m.s. values of velocity fluctuations $u_{i,rms}$ for the DNS of a turbulent channel flow under CpPI. In each plot, the left-hand axis is ‘power’ units, and the right-hand axis is the conventional viscous (plus) units. In (a), the thin lines are the linear and logarithmic laws given by $\langle u \rangle^+ = y^+$ and $\langle u \rangle^+ = 2.5 \ln y^+ + 5.5$. In (b), the streamwise component is plotted with a continuous line, the wall-normal component with a dashed line, and the spanwise component with a dash-dotted line.

Equation (4.3) is still valid (in equilibrium conditions). Hence,

$$P_t (1 - \gamma) = \frac{U_b}{Re_{\Pi}} \left. \frac{d \langle u \rangle}{dy} \right|_{y=0}. \tag{5.2}$$

By using (2.11) one obtains

$$U_b \left. \frac{d \langle u \rangle}{dy} \right|_{y=0} = 3(1 - \gamma) \quad (5.3)$$

which is an exact relation, like (4.4).

The friction coefficient is still defined in terms of P_p^* :

$$C_f = \frac{P_p^*}{\frac{1}{2} \rho^* U_b^{*3}} = \frac{2P_t^*(1 - \gamma)}{\rho^* U_b^{*3}} = \frac{6}{Re_\Pi} \left(\frac{U_\Pi^*}{U_b^*} \right)^3 (1 - \gamma). \quad (5.4)$$

When no control is applied, i.e. $\gamma = 0$, and the flow is laminar, the conventional laminar relationship $C_f = 6/Re_b$ is recovered. Frohnapfel *et al.* (2012) introduced the notion of an effective wall friction τ_w^{e*} (and an effective friction coefficient C_f^e) to account for the total power consumption:

$$\tau_w^{e*} = \frac{P_t^*}{U_b^*} = \tau_w^* + \frac{P_c^*}{U_b^*}. \quad (5.5)$$

By writing it as $\tau_w^{e*} = \tau_w^*/(1 - \gamma)$, we obtain:

$$C_f^e = \frac{C_f}{1 - \gamma} = \frac{6}{Re_\Pi} \left(\frac{U_\Pi^*}{U_b^*} \right)^3, \quad (5.6)$$

indicating that achieving $U_b^* = U_\Pi^*$ is equivalent to reducing C_f^e to the laminar value.

Example

To exemplify the application of the CPI concept in the context of flow control, we consider the problem of reducing the turbulent friction drag, and employ to this purpose the well-known spanwise-oscillating-wall technique (Jung *et al.* 1992). The choice of this particular technique is motivated by the amount of information already available, as well as by the limited number of control parameters involved. The forcing scheme consists of harmonically moving the wall in the spanwise direction according to:

$$w(x, y = 0, z, t) = W_0 \sin(\omega t) \quad (5.7)$$

where the only two parameters are W_0 , the oscillation amplitude, and $\omega = 2\pi/T$, its frequency (T is the oscillation period). The oscillating wall obviously requires power for its actuation, and thus the question we are asking in the CPI setting is: for a given available total power, what is the optimal share of power between the pump and the control device that maximizes the flow rate? In other words, for a given value of Re_Π , we seek the optimal value of γ for which the oscillating-wall technique provides the largest increase of the flow rate U_b above the value $U_{b,0}$ corresponding to $\gamma = 0$, the uncontrolled flow where all of the power is used for pumping. It is known (Baron & Quadrio 1996) that, for relatively small values of the forcing amplitude, the oscillating-wall technique can achieve small positive net savings, i.e. combinations of parameters exist where control power is smaller than pumping power saved thanks to the control action. We thus expect U_b to be maximized by some optimal non-zero value of $\gamma > 0$.

A general optimization procedure would of course involve looping over different values of γ in order to identify the best. Moreover, a single value of γ corresponds to

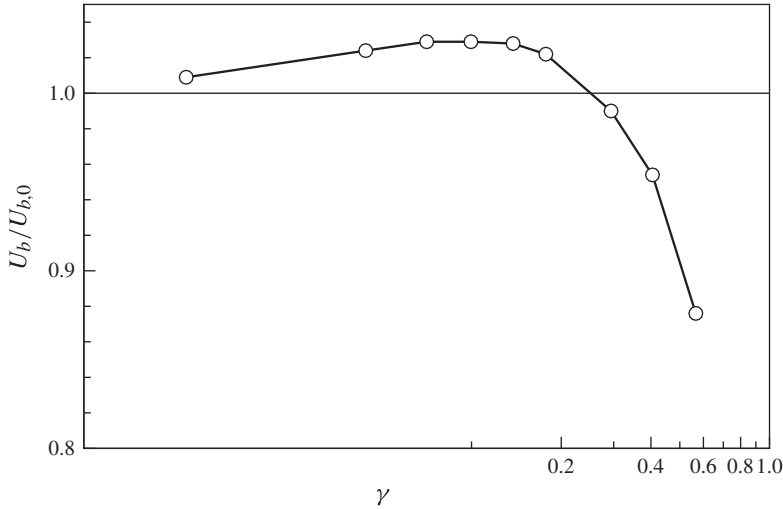


FIGURE 7. Ratio between the actual flow rate U_b and the flow rate $U_{b,0}$ of the uncontrolled case, as a function of the share γ of the available power between pumping power and control power provided to the oscillating walls.

several different combinations of the control parameters, the optimal set of which has to be identified. An additional difficulty is that, given a set of control parameters, the value of γ , i.e. the power cost of the control technique in that particular configuration, is in general unknown beforehand, thus calling for an iterative procedure.

However, in the particular case of the oscillating wall, previous knowledge can be exploited, and available information (Quadrio & Ricco 2004) allows us to determine the range of parameters where best performance in terms of net saved power is expected. Moreover, it is known (Quadrio & Sibilla 2000) that the oscillating spanwise boundary layer created by the oscillations obeys the laminar solution of the Stokes second problem in the range of forcing conditions of interest: it can thus be anticipated that the control power increases with the square of the oscillation amplitude and with the square root of the oscillation frequency. The values of γ corresponding to a given pair of W_0 and ω can thus be first-guessed *a priori*, and then simply verified or slightly adjusted *a posteriori*.

Figure 7 summarizes the small parametric study carried out for the oscillating wall in the present work. In these simulations, we have chosen the same value $Re_{\Pi} = 6500$ previously used in § 4, as it corresponds to the uncontrolled flow at $Re_{\tau} = 200$ which is the case for most of the available drag-reduction data for the oscillating wall. The same computational parameters mentioned before are employed here. In calculating time-average quantities, the initial transient is carefully excluded. We note, in passing, that the duration of this transient is one of the outcomes that is most affected by the simulation strategy, with CFR presenting the shortest transient. However, as discussed by Ricco *et al.* (2012), the longer transient shown by CPG is roughly compensated by the smaller fluctuation of the quantity to be averaged in time (flow rate in CPG and wall friction in CFR), so that the two approaches are roughly equivalent in terms of computational cost. CPI presents an intermediate behaviour in this respect.

The figure shows how the ratio $U_b/U_{b,0}$ changes with γ . The oscillating wall is confirmed to possess a parameters range where the optimal share of power between pumping and control power yields an increase in the flow rate. Best performances

are observed near $\gamma \approx 0.1$, indicating that the optimal use of the available power (for the investigated Reynolds number) is to employ 90% for the pumping system and the remaining 10% for the control system. The maximum control-induced increase of the flow rate is rather small, of the order of 3%, and this is in agreement with previous information (Quadrio & Ricco 2004) that the maximum net power saving is only 7%. Of course better-performing drag-reduction techniques are expected to yield larger increase of the flow rate.

The calculations at $\gamma > 0.25$ do not provide a positive yield for the oscillating wall. When such a control input is applied in CFR or CPG simulation, the resultant drag-reduction rate (CFR) or flow rate (CPG) is enhanced. However, the control power input increases more rapidly. This important point is directly visible for the CPI condition, where fixed total power input results in a decrease of the flow rate.

In the limit $\gamma \rightarrow 1$, all the energy is used up by the control system, and in the present case the flow rate decreases to zero. As mentioned in § 3.3, however, this is not necessarily true in general.

6. Conclusion

An original approach to carry out numerical simulations of duct flows, that is different from the well-established options of keeping constant the flow rate (CFR) or the pressure gradient (CPG), is introduced. The approach, that we name CPI, consists of keeping the power input to the system strictly constant in time, thus ensuring a constant energy flux. CPI represents a third alternative to the well-established CFR and CPG techniques.

The paper describes in some detail how to implement the CPI concept into a DNS or LES computer code, using the simple case of a plane channel flow. The idea is general though, as it is based on the identification of a velocity scale U_Π that expresses the flow rate achievable in the laminar regime with a given power input, and leads to the definition of a new power-based type of Reynolds number Re_Π , which is naturally suited for CPI calculations, much as Re_b is flow-rate based and suited for CFR, and Re_τ is wall-friction based and suited for CPG.

The application example of CPI given in the paper concerns the problem of turbulent skin-friction drag reduction through the spanwise-oscillating wall. Drag reduction is often motivated by the need to save power. The CPI concept naturally leads to setting up a power-constrained problem where the optimal management of the power budget is identified by finding the optimal share of power between the pump and the flow control device that provides the largest flow rate. A plot of the gain (the flow rate) against the share of the available power between pumping and control provides a simple and effective tool to assess different drag-reduction techniques and thus determine which ones should be studied and developed further.

We believe that the potential of the CPI concept goes beyond the applications presented in the paper. For turbulent drag reduction, the CPI concept has the potential to highlight general properties of flows with drag reduction. It might be expected that drag-reduced turbulent flows with different control strategies share some particular and yet universal feature, which must possess a statistical footprint. However, identifying such a footprint requires comparing flows with and without drag reduction, and the terms of such comparison must be clearly defined. We believe that the CPI concept is a key property of this definition. In the Richardson–Kolmogorov theory of homogeneous isotropic turbulence it is well known that what happens in the inertial range of scales is determined, at statistical equilibrium, by the rate at which

energy enters the system. This corresponds to the rate at which, on average, energy is transferred to smaller scales, and to the mean dissipation rate of turbulent kinetic energy that takes place at the Kolmogorov scale. In the more complicated setting of inhomogeneous, anisotropic turbulence, the CPI approach will allow us to analyse and compare controlled flows where the rate at which energy enters the system is fixed, thus allowing a meaningful analysis of how energy is distributed in the system.

From a more general standpoint, the CPI concept could play an interesting role whenever a comparison has to be carried out between two (laminar or turbulent) flows where a modification is introduced. If added to the comparison of two flow fields shown in figure 1, CPI represents a third alternative (where e.g. the size of the resulting recirculation bubble is in between those for CPG and CFR) whose meaning must be evaluated on the basis of the specific problem. An extension of the presented CPI concept to unsteady flow conditions, e.g. in analogy to varying flow rates in arteries, can be considered.

As a final note, we would like to suggest that whenever fundamental studies on laminar to turbulent transition are concerned it can be envisaged that the way simulations are run (CFR, CPG or CPI) directly influences the result. CPI might therefore be an interesting option to open up the parameter space in this context.

Acknowledgements

The authors would like to thank Anna Slotosch for providing the idea to figure 1, and Marco Carini for useful discussions. Y.H. acknowledges the support by the Ministry of Education, Culture, Sports, Science and Technology of Japan (MEXT) through the Grant-in-Aid for Scientific Research (B) (No. 25289037). B.F. acknowledges the support of project FR2823/2-1 of the German Research Foundation (DFG).

Appendix A. Relation between Re_{Π} , Re_b , Re_{τ}

The relation between the newly introduced Re_{Π} for CPI and the existing Reynolds numbers customarily employed for a turbulent channel flow, i.e. the friction and the bulk Reynolds number, is discussed in the following.

For a fully developed turbulent channel flow, the friction coefficient can be estimated by the so-called Dean's formula (Dean 1978) as:

$$C_f \equiv \frac{\tau_w^*}{\frac{1}{2}\rho^*U_b^{*2}} = \alpha Re_b^{-1/4}, \quad (\text{A } 1)$$

where the coefficient α takes the value of $\alpha = 0.0614$.

Based on this well-established empirical correlation, the relationship between Re_{τ} and Re_b is given by:

$$Re_{\tau} = \sqrt{\frac{\alpha}{2}} Re_b^{7/8} = 0.1752 Re_b^{7/8}. \quad (\text{A } 2)$$

From (4.5), the pumping power P_p^* is related to C_f by:

$$P_p^* = \frac{1}{2}\rho^*U_b^{*3}C_f. \quad (\text{A } 3)$$

Re_τ	Re_b	Re_Π	U_b^*/u_τ^*	U_Π^*/u_τ^*	U_Π^*/U_b^*
100	1 413	2 171	14.4	21.7	1.54
150	2 247	4 105	15.0	27.4	1.83
200	3 121	6 451	15.6	32.3	2.07
300	4 961	12 199	16.5	40.7	2.46
450	7 885	23 070	17.5	51.3	2.93
600	10 954	36 256	18.3	60.4	3.31

TABLE 1. Relationship between different Reynolds numbers obtained by Dean's formula (A 1).

Substituting P_p^* by (2.9) yields:

$$\frac{U_\Pi^*}{U_b^*} = \frac{Re_\Pi}{Re_b} = \sqrt{\frac{1}{6} Re_b C_f} = 0.1012 Re_b^{3/8}, \quad (\text{A } 4)$$

and therefore

$$Re_\Pi = 0.1012 Re_b^{11/8}. \quad (\text{A } 5)$$

Equation (A 3) can also be written as:

$$P_p^* = \rho^* u_\tau^{*2} U_b^*. \quad (\text{A } 6)$$

Combining (A 6) and (2.9), the following relationship is obtained:

$$\frac{U_\Pi^*}{u_\tau^*} = \frac{Re_\Pi}{Re_\tau} = \sqrt{\frac{Re_b}{3}} = 1.5621 Re_\tau^{4/7}. \quad (\text{A } 7)$$

Hence,

$$Re_\Pi = 1.5621 Re_\tau^{11/7}. \quad (\text{A } 8)$$

The relationship between the different Reynolds numbers and the corresponding velocity scales in a fully developed turbulent channel flow at relatively low Reynolds number, at which DNS are typically carried out, are summarized in table 1. Comparison with the DNS described in § 4 where $Re_\tau = 199.7$, $Re_b = 3176$ and $Re_\Pi = 6500$ reveals the known small inaccuracy of the Dean's correlation at low values of Re .

The last column of the table gives the ratio between the maximum achievable velocity U_Π^* for the fixed power input (realized by the laminar flow state) and the actual turbulent bulk velocity U_b^* according to (A 4). This quantity is of particular interest in order to illustrate an important difference between CPI and CPG. The corresponding ratio between laminar and turbulent flow conditions at CPG is given by:

$$\frac{(U_{b,lam}^*)^{CPG}}{U_b^*} = 0.0102 Re_b^{3/4}. \quad (\text{A } 9)$$

Two curves, corresponding to (A 4) (where $U_{b,lam}^* = U_\Pi^*$) and (A 9), are plotted in figure 8. For CPG a drastic increase in flow rate is revealed if the flow is laminarized: for example, at the still relatively modest value of $Re_b \approx 80\,000$, the flow rate is almost 50 times larger for laminar flow conditions under CPG. In contrast, at the same value

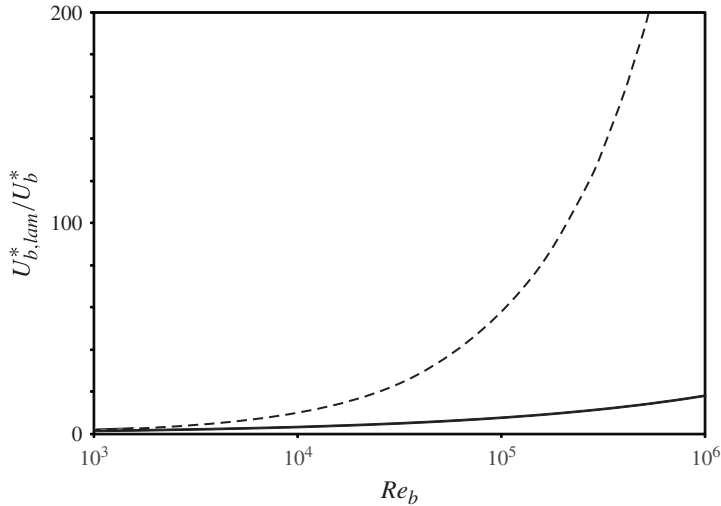


FIGURE 8. The ratio of maximum achievable bulk flow velocity (i.e. that in laminar flow) to the actual turbulent bulk flow velocity as a function of bulk Reynolds number. Continuous line is the CPI case, (A 4), and dashed line is the CPG case, (A 9).

of Re , CPI results in a potential flow rate increase of a factor seven only. It should be noted that the laminar flow under CPG is driven by a power input that exceeds the one for the turbulent flow by factor 50. Therefore, the impressive effect of relaminarization found for CPG conditions cannot be expected in an engineering system if, as is often the case, the system is limited by the available power. Marusic *et al.* (2007) note that the CPG condition can be linked to an example given by Kolmogorov in which he addresses the increase in flow speed if the Volga river were laminar. The river – in contrast to most technical applications – is driven by an infinite reservoir of energy (if we assume an infinite reservoir of water that is feeding the river) such that the required increase in power input is not a limiting factor.

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