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## 2 **On the Global Kinematic Positioning Variations during the September 2017**

3

### **Solar Flare Events**

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### 22 **Key Points**

- 23 • The SID induced by the SF is controlled by the intensity, location and duration  
24 time of the SF.
- 25 • The SID triggered by the X9.3 SF deteriorate the detection of the cycle slips,  
26 thus degrade the accuracy of the kinematic PPP solution.
- 27 • The X8.2 SF that occurred during the recovery phase of the geomagnetic  
28 storm can reduce the precision of GPS measurements.

29 **Abstract**

30 Several X-class solar flares (SFs) with different intensities and locations on the solar  
31 disk occurred in September, 2017. Among them, the X9.3 SF on September 6, 2017  
32 was the most intensive SF in the 24<sup>th</sup> solar cycle. In this study, we investigated and  
33 compared the ionospheric response to the different X-class SFs and their impacts on  
34 the Global Positioning System (GPS) kinematic precise point positioning (PPP)  
35 solutions. We aim to study the mechanism behind the positioning degradation from  
36 the perspective of the impacts of the SF-induced ionosphere disturbance on the GPS  
37 data processing. By comparing the sudden ionospheric disturbance (SID) induced by  
38 the SFs, we observed that the SID is controlled by the intensity, the location, and the  
39 duration time of the SF. We found that the SID induced by the SF may deteriorate the  
40 cycle slip (CS) detection algorithms seriously. The threshold of the CS detection  
41 observables is sensitive to the SID, making the CS easier to be falsely detected. On  
42 the other hand, for the SF that occurred during the recovery phase of the geomagnetic  
43 storm, as the case of the X8.2 SF, the effects of the SF can reduce the precision of the  
44 GPS measurements, thus affecting the positioning accuracy. These mechanisms are  
45 essential and significant for the accuracy and stability of the kinematic PPP solution  
46 obtained with GPS during the SF events.

47 **Plain Language Summary**

48 Solar flares (SFs) are one kind of extreme space weather events, which disturb the  
49 radio communication and degrade the precision of the Global Positioning System  
50 (GPS). Much efforts have been made to distinguish the different ionosphere responses  
51 to the SFs with different intensity and locations on the Sun. However, the positioning  
52 performances during the different SFs are less discussed, let alone the mechanism  
53 behind the degraded positioning accuracy. In this study, we observed that the sudden  
54 ionosphere disturbance (SID) induced by the SF is not only dependent on the intensity  
55 and location on the solar disk, but also on the duration time of the SF. We found that the  
56 cycle slip (CS) detection algorithms are sensitive to the SID, making the CS easier to  
57 be falsely detected. Fortunately, most of the SID effects on the dual-frequency

58 positioning can be mitigated by optimizing the CS detection threshold. On the other  
59 hand, the effects of the SF that occurred during the recovery phase of the geomagnetic  
60 storm can reduce the precision of the GPS measurements, thus degrading the accuracy  
61 of the kinematic precise point positioning solution.

## 62 **1 Introduction**

63 Solar flares (SFs) are one of the severest solar events, when the Sun releases  
64 high-energy protons, electrons and intense radiation in all wavelengths, affecting not  
65 only the Earth's upper atmosphere but also propagation of radio waves. The high-level  
66 radiations of X-ray and of extreme ultraviolet (EUV) radiation results in ionization in  
67 the ionosphere on the sunlit side of the Earth. Intense X-ray emission causes absorption  
68 in the lower ionospheric D layer, which results in degradation or complete absorption  
69 of high-frequency signals. Solar EUV radiation has a decisive impact on the  
70 ionospheric heights from 120 to 200 km, and a sudden increase of the EUV emission  
71 during SF causes an abrupt enhancement of the ionization that can last from minutes to  
72 hours (Donnelly, 1976; Mitra, 1974; Prölss, 2012). The enhanced X-ray photons and  
73 the EUV disturbs radio communications and degrades the precision of Global  
74 Navigation Satellite System (GNSS) measurements, even could damage the  
75 Earth-orbiting satellites and reduce their lifetime (Afraimovich et al., 2008; Cerruti et  
76 al., 2006; Chen et al., 2005; Cheng et al., 2018; Demyanov & Yasyukevich, 2021;  
77 Desai & Shah, 2020; Sato et al., 2019; Sreeja et al., 2014; Yasyukevich et al., 2018).  
78 Therefore, the ionospheric responses to severe SFs are a key topic of study in the space  
79 weather community and considered as an important factor of improving the accuracy  
80 and stability of GNSS positioning.

81 The SF effects on the ionosphere have been studied for many decades, especially with  
82 the advent of GNSS (Donnelly, 1976; Hernández-Pajares et al., 1998; Lei et al., 2018;  
83 Li et al., 2018; Liu et al., 2011; Mitra, 1974; Tsurutani et al., 2009; Xiong et al., 2011).  
84 Tsurutani et al., (2006) discussed the extreme SFs of Oct 28, Oct 29, Nov 4 of 2003  
85 and July 14, 2000 (Bastille Day event) and their photoionization effects on the  
86 dayside ionosphere. The largest increase on the dayside ionosphere occurred on the 28

87 October 2003, where a SF (X17) peaked to 25 TECU (1 TECU equals  $10^{16}$  electron  
88  $\text{m}^{-2}$ , and corresponds to 0.1623 m for the Global Positioning System (GPS) L1 band  
89 signal), and not the more intense X-ray flare of 4 November 2003 (X28). The latter  
90 event caused only a moderate increase of 5 to 7 TECU. Since the 4 November SF  
91 occurred near the limb of the Sun, and the 28 October SF EUV peak flux increase  
92 doubled that of the 4 November SF. Then, it was suspected that the solar EUV flux  
93 was primarily responsible for the increased total electron content (TEC) in the  
94 ionospheric E and F regions during and immediately after SFs. However, other  
95 possible contributions may exist.

96 Many researchers have been devoted to discover the factors that impact the  
97 ionospheric response to the SFs in the past decade. Qian et al. (2010) addressed how  
98 the location of SFs on the solar disk affects the thermosphere and the corresponding  
99 response of the ionosphere. It was found that the X-ray ultraviolet (XUV) radiation  
100 dominated ionization in the lower thermosphere whereas the EUV dominated  
101 ionization in the upper thermosphere. Thus, the SF location had a minor effect on the  
102 E region and lower thermosphere but had a large effect on the F region ionosphere  
103 and upper thermosphere. In fact, the magnitude of sudden ionospheric disturbances  
104 (SIDs) increased with the decrease of central meridian distance. Besides, the solar  
105 zenith angle was found to be an important factor to determine the distribution of  
106 ionospheric disturbances, and case studies as well as statistical studies had reported  
107 that the distribution of SIDs presented almost a linear relationship with the cosine  
108 value of solar zenith angle (Le et al., 2013; Le et al., 2016; Zhang et al., 2002; Zhang  
109 et al., 2011). In addition, Qian et al., (2011) found that for SFs with similar  
110 magnitudes and the same location on the solar disk, the thermosphere and ionosphere  
111 responses showed large variability. The risetime and the decay time following a SF  
112 were also important factors in determining the responses.

113 As for the impacts of the SF-induced ionosphere disturbances on GNSS positioning,  
114 numerous studies have reported that the L-band signals of the GPS can be affected,  
115 thus degrading the positioning accuracy (Afraimovich et al., 2008; Berdermann et al.,

116 2018; Carrano et al., 2009; Cheng et al., 2018; Demyanov & Yasyukevich, 2021;  
117 Desai & Shah, 2020; Dey et al., 2020; Linty et al., 2018; Sato et al., 2019; Sreeja et al.,  
118 2014; Yasyukevich et al., 2018). Afraimovich et al., (2008) investigated failures in the  
119 GPS performance produced by solar radio bursts with unprecedented radio flux  
120 density during the X6.5 and X3.4 SFs on 6 and 13 December 2006, respectively.  
121 Significant experimental evidence was found that high-precision GPS positioning on  
122 the Earth's entire sunlit side was partially disrupted for 10 to 15 min on 6 and 13  
123 December 2006. A great number of losses-of-lock (LoL) and carrier-phase cycle slips  
124 (CS) resulted from the wideband solar radio noise emission. Berdermann et al. (2018)  
125 reported the ionospheric response to the X9.3 flare on September 6, 2017 as well as  
126 its impacts on the navigation services over Europe. It was found that the EUV peaks  
127 caused strong dynamics in the bottom-side ionosphere, challenging GNSS receivers to  
128 maintain signal tracking and, in most cases, to produce satellite LoLs. For the  
129 dual-frequency precise point positioning (PPP), deviations of the estimated position in  
130 East, North, and Up directions reached 1 to 2 m. Aiming to address the impacts of the  
131 X-class SF on September 6, 2017 on the GNSS, Yasyukevich et al., (2018) reported  
132 that the X2.2 and X9.3 SFs did not cause LoLs in the GPS, GLONASS, or Galileo  
133 systems, while the positioning errors increased by a factor three in the PPP solution  
134 with GPS. By using higher time and frequency resolution of the GNSS and solar radio  
135 burst data sets, Sato et al., (2019) found that the rapid EUV and associated ionization  
136 enhancement primarily caused the LoLs of the GPS signals. The impact of the radio  
137 burst/EUV flare on GNSS positioning was studied using dual and single-frequency  
138 methods, the results of which showed that the dual-frequency solution was more  
139 affected by the solar activity than the L1 single-frequency solution. In summary,  
140 many efforts have been made to assess the GNSS positioning performances during the  
141 SF events and it has been reported that the GNSS signals as well as the positioning  
142 accuracies are degraded by SF events.

143 Recently, Yang et al., (2020) made a comprehensive study of the storm-induced  
144 ionospheric disturbance on kinematic PPP solutions using globally distributed GNSS

145 stations during the 2015 St. Patrick's Day storm. Motivated by Yang et al., (2020), we  
146 would like to investigate the SF-induced ionospheric disturbance on the global GPS  
147 kinematic positioning. Though the ionosphere as well as the thermosphere responses  
148 to different SFs have been analyzed from cases to cases, the corresponding  
149 positioning performances are deserved to be further studied, especially under the SFs  
150 with different intensities and locations on the solar disk. In September 2017, four  
151 X-class eruptions emerged by the Active Region AR2673 with different intensity and  
152 locations. Among them, the X9.3 SF on September 6, 2017 was the strongest SF of  
153 the 24<sup>th</sup> solar cycle. A considerable number of researchers have investigated the  
154 ionospheric disturbances as well as the GPS positioning performance associated with  
155 this SF (Alfonsi et al., 2021; Berdermann et al., 2018; Blagoveshchensky & Sergeeva,  
156 2019; Desai & Shah, 2020; Fagundes et al., 2020; Imtiaz et al., 2020; Kumar and  
157 Kumar, 2020; Lei et al., 2018; Li et al., 2018; Linty et al., 2018; Liu et al., 2020; Liu  
158 et al., 2021; Nishimura et al., 2021; Owolabi et al., 2020; Qian et al., 2019; Sato et al.,  
159 2019; Yamauchi et al., 2018; Yasyukevich et al., 2018; Zakharenkova & Cherniak,  
160 2021; Zhang et al., 2019). The novelty of the present study is, on the one hand, to  
161 assess and compare the different positioning performances using globally distributed  
162 GPS stations during the SFs with different ionosphere responses. On the other hand,  
163 we try to discover the mechanism behind the degraded positioning performances from  
164 the point view of GPS data processing by considering the ionosphere disturbance  
165 effect on GPS signals. This mechanism is essential to mitigate the SF effects on the  
166 GNSS positioning.

## 167 **2 Data and Methodology**

### 168 2.1 GPS Data

169 GPS data from the International GNSS Service (IGS) and Crustal Movement  
170 Observation Network of China (CMONOC) were processed for the SF effects on GPS  
171 kinematic positioning solutions in September 2017. The number of the stations from  
172 IGS was about 520 while that from CMONOC was about 250; therefore, a total number

173 of 700+ stations around the world were selected with a sampling rate of 30 s. The  
174 distribution of the GPS stations is presented in Figure 1. The twelve yellow pentagrams  
175 represent the selected stations to display the ionosphere responses along the zenith of  
176 the stations in the Experiment and Results section. It is noted that though most of the  
177 stations may also track GLONASS, even Galileo or BDS signals, we only use the GPS  
178 data to study the positioning solutions.

## 179 2.2 Solar flare and ionospheric disturbance index

180 In the present study, X-ray flux was used to represent the characteristics of SFs in  
181 September 2017. The 1-min averaged X-ray data in the wavelength of 0.1–0.8 nm was  
182 observed by the Geostationary Operational Environmental Satellite (GOES) 13 and 15.  
183 Besides, the solar radio flux at 10.7 cm, as the F10.7 index, was also used to depict the  
184 solar space weather condition. In order to evaluate the level of magnetic activity when  
185 the SFs erupted, the magnetic storm index, including the Interplanetary Magnetic Field  
186 (IMF) components, the longitudinally symmetric disturbances index in the horizontal  
187 direction H (SYM-H) and Auroral Electrojet indices, were obtained from the Goddard  
188 Space Flight Center (GSFC) server.

189 To measure ionospheric disturbances at the same stations presented in Figure 1, we  
190 used the Rate of TEC (ROT), the vertical ROT (vROT) and the ROT Index (ROTI)  
191 metrics derived from GPS dual-frequency carrier-phase measurements. By ROT, we  
192 mean the variation of the TEC along the line-of-sight from the receiver to the satellite.  
193 To describe the ROT along the zenith of the stations, we used the vROT metric. The  
194 vROT is weighted by the inverse great-circle distance between the ionospheric pierce  
195 points (IPPs) of the satellites and the zenith location of the stations. Defined as the  
196 standard deviation of the rate of TEC, the ROTI is an indicator generally used for  
197 quantifying small-scale ionospheric plasma irregularities (Cherniak et al., 2014; Pi et  
198 al., 1997; Yang & Liu, 2016). These disturbances are expected to generate significant  
199 scintillation effects on GPS signals, in such a way that ROTI values can be related to  
200 the impacts on kinematic PPP solutions throughout our analysis. We computed ROTIs



201 every 5 min with data sampled every 30 s. It was mapped onto IPPs assuming a thin  
 202 shell model at an altitude of 350 km.

### 203 2.3 GPS kinematic positioning performance

204 The GPS carrier phase and pseudorange measurements were processed as kinematic  
 205 PPP solutions by utilizing routines from the Real-Time Kinematic Library (RTKLIB)  
 206 (Takasu, 2013). In our study, the kinematic positions of the globally distributed GPS  
 207 receivers were produced every 30 s. An elevation cut-off angle of  $7^\circ$  was used to reduce  
 208 the multipath effects on the position calculation while that of  $20^\circ$  was adopted to avoid  
 209 the geometrical effects as well as the multipath effects on the ROTI calculations (Juan  
 210 et al., 2018; Li et al., 2015; Pi et al., 1997). The detailed data processing techniques and  
 211 implementations for GPS dual-frequency PPP are summarized in Table 1.

212

213 **Table 1** Summary of data processing techniques and implementations for GPS  
 214 dual-frequency PPP

Item	techniques and implementations
Observations	GPS dual-frequency code and phase measurements
Processing mode	Forward
Sampling interval	30 s
Elevation mask angle	$7^\circ$
Cycle-slip detection	Phase Geometry-free and Hatch-Melbourne-Wubbena combination (Blewitt, 1990)
Satellite orbit	Fixed with the final products from IGS with an interval of 15 minutes
Satellite clock	Fixed with the final products from IGS with an interval of 5 minutes
Phase center offset	igs08.atx
Ionospheric delay	Ionosphere-free model: First-order effect eliminated by ionospheric-free linear combination

Differential Code Bias	Corrected by P1C1 DCBs from Center for Orbit determination in Europe
Tropospheric delay	The Saastamoinen model for the initial zenith dry and wet delay (Saastamoinen, 1972), along with the GMF projection function (Boehm et al., 2006), and the wet delay is estimated as constant every 2 h
Solid earth tide, ocean tide loading and pole tide	IERS Conventions 2010(Petit and Luzum, 2010); FES2004 (Lyard et al., 2006) for ocean tides
Relativity effect	IERS Conventions 2010

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215

216 Under normal conditions, the kinematic PPP can achieve decimeter-level accuracy for  
 217 the receiver with the above processing strategies. In our study, the kinematic PPP  
 218 solutions were evaluated by a comparison with their daily solutions in static PPP mode  
 219 on September 3, 2017. Their position errors were further examined in association with  
 220 the storm-induced ionospheric disturbances. In addition, we define the positioning  
 221 convergence by the time that the positioning error converges to 20 cm for the first time  
 222 and lasts for 10 epochs.

#### 223 2.4 Cycle slip detection algorithm

224 Since the CS detection algorithm is critical for the positioning accuracy and stability  
 225 when extreme space weather events occur, the CS detection algorithm is introduced in  
 226 the following. In fact, although many CS detection algorithms have been proposed  
 227 over the past decades, the TurboEdit algorithm is maybe the most widely used in  
 228 GNSS data processing (Blewitt, 1990). Popular GNSS data processing software, such  
 229 as GIPSY, Bernese, PANDA as well as the RTKLIB package used in this study, adopt  
 230 the TurboEdit algorithm or algorithms based on TurboEdit (Bertiger et al., 2020;  
 231 Dach et al., 2015; Liu & Ge, 2003). The TurboEdit algorithm is based on the Hatch–  
 232 Melbourne-Wübbena (HMW) and Phase-Geometry-Free (PGF) combination

233 observables (Hatch, 1983; Melbourne, 1985; Wubben, 1985). The two basic detection  
 234 observables can be written as follows:

$$\begin{aligned}
 \Delta N_{HMW} &= \Delta\varphi_1 - \Delta\varphi_2 - \frac{f_1\Delta P_1 + f_2\Delta P_2}{\lambda_{HMW}(f_1 + f_2)} + \Delta\xi_{HMW} \\
 \Delta\varphi_{PGF} &= \lambda_1\Delta\varphi_1 - \lambda_2\Delta\varphi_2 = \lambda_1\Delta N_1 - \lambda_2\Delta N_2 + (\gamma - 1)\Delta I + \Delta\xi_{PGF}
 \end{aligned}
 \tag{1}$$

236 where  $\Delta$  is the difference operator between two consecutive epochs which are  
 237 separated 30 s in the present study and  $\lambda_{HMW}$  is the wide-lane wavelength of 86 cm;  $p_j$   
 238 in the unit of meter and  $\varphi_j$  in the unit of cycle are the pseudorange and carrier phase  
 239 at frequency  $f_j$ , respectively;  $I$  is the line-of-sight ionospheric delay at frequency  $f_1$   
 240 in unit of meter;  $\gamma$  is the frequency-dependent multiplier factor ( $\gamma = (f_1 / f_2)^2$ ); it is  
 241 noted that for the ionosphere-free (IF) combination, the factor is nearly zero;  $\lambda_j$  is  
 242 the carrier wavelength at frequency  $f_j$  in unit of meter;  $N_j$  is the phase ambiguity in  
 243 unit of cycle;  $\xi$  is the combination of measurement noise and multipath error for  
 244 pseudorange and carrier phase observations in unit of meter.

245 To detect the CS by the TurboEdit algorithm, the commonly used threshold for the  
 246 HMW observable is from 1 to 2 cycles of  $\lambda_{HMW}$  (i.e. 0.86 to 1.72 m), while that for the  
 247 PGF observable is from 0.05 to 0.15 m (Zhang et al., 2014). The RTKLIB package used  
 248 in the present study, sets as default threshold of the PGF observable to 0.05 m (Takasu,  
 249 2013). We will show in the Experiment and Results section that, with the default  
 250 thresholds, the kinematic PPP can achieve accurate and stable performance under  
 251 ionospheric quiet conditions.

### 252 **3 Experiment and Results**

#### 253 3.1 Solar flare conditions in September 2017

254 September 2017 was an active space weather period in which many SFs and two  
 255 geomagnetic storms occurred. The stormy conditions were driven by the Active

256 Region AR2673, from which four X-class eruptions emerged. Especially, the period  
 257 from September 3 to September 13, 2017 gave us an insight into solar-terrestrial  
 258 interaction and allowed us to study its influence on GPS positioning in more detail. In  
 259 the following, we briefly describe such an interesting solar activity period.

260 Figure 2a depicts the X-ray measurements recorded by the primary (G15) and  
 261 secondary (G13) GOES satellites. We can distinguish between the two channels 0.1–  
 262 0.8 nm (G15: brown; G13: blue), indicating the well-known flare size in the X-ray  
 263 range, and 0.05–0.4 nm (G15: light brown; G13: light blue). Figure 2b depicts that the  
 264 F10.7 measurements from September 3 to September 10 were all larger than 90 sfu;  
 265 therefore, the solar space weather condition was quite active during the period. The  
 266 occurrence information of the X-class SFs during the September 6-10,2017 space  
 267 weather event is summarized in Table 2. It is noted that the X9.3 class SF was the  
 268 strongest eruption in more than a decade.

269 **Table 2** Summary of the SF occurrence information during the September 6-10,2017  
 270 space weather event.

Class	Date 2017	Start (UT)	Max (UT)	End (UT)	Risetime (min)	Decay time(min)	Duration (min)
X2.2	Sep. 6	08:57	09:10	09:17	13	7	20
X9.3	Sep. 6	11:53	12:02	12:10	9	8	17
X1.3	Sep. 7	14:20	14:36	14:55	16	19	35
X8.2	Sep. 10	15:35	16:06	16:31	31	25	56

271 Besides, the strongest X9.3 flare on September 6 triggered also a geoeffective CME  
 272 event, which triggered disturbed magnetospheric and ionospheric conditions, as  
 273 depicted in Figure 2c. Two sequential geomagnetic storms peaked on September 8,  
 274 the first with a SYM-H minimum value of  $-146$  nT, observed at 01:08 UT, and the  
 275 second with a SYM-H minimum of  $-112$  nT at 17:08 UT. The sequence of these  
 276 events was in accordance with the time line of the SF effects as described in Tsurutani  
 277 et al. (2009).

278 In the following, we focused on the main event of this solar activity phase, the X-class  
279 SF on September 6 and September 10, 2017, and we follow their effects on the  
280 ionosphere down to application examples in the kinematic positioning domain.

### 281 3.2 Ionospheric response on September 6 and Kinematic PPP Error

282 We study the response of the ionosphere to the SFs on September 6 from the point view  
283 of the aforementioned ROTI, in Figure 3. To complement the time series of ROTI, we  
284 include the vROT along the zenith of each station. To distinguish the SF effects on the  
285 sunlit side of the Earth, we selected and displayed the ionosphere responses at twelve  
286 stations according to their longitude distribution. The distribution of the twelve stations  
287 are depicted as yellow pentagrams in Figure 1.

288 From Figure 3, we can see fluctuations of the vROT series at around 9:00 UT and  
289 12:00 UT, the time of which was consistent with the peak eruption of the X2.2 and  
290 X9.3 SF on the day. The vROT fluctuation is a typical phenomenon of the SID due to  
291 the SF. In detail, the amplitude of the vROT fluctuation at around 9:00 UT was about  
292 0.2 TECU/min while that at around 12:00 UT was about 1.5 TECU/min. The  
293 difference of the fluctuation amplitude revealed the effects of the different magnitude  
294 of the X-class SF on the ionosphere. Besides, we can find that not all of the selected  
295 stations showed the vROT fluctuations during the SF events. Taking the vROT  
296 fluctuation at around 12:00 UT as an example, the apparent fluctuations occurred for  
297 stations whose longitudes were from 71.9 W to 70.2 E. Combing the distribution of  
298 the twelve stations within the earth terminator in Figure 4b, we deduced that the  
299 stations with apparent vROT fluctuation were experiencing the sunlit period.

300 Figure 4 presents the global temporal-spatial variations of ROTI during the X2.2 and  
301 X9.3 SF on September 6, 2017. Specifically, we plotted the Earth terminator,  
302 separating the day side and night side. The twelve stations in Figure 3 are also  
303 presented in Figure 4.

304 Figure 4a depicts the variations of ROTI from 08:55 UT to 09:20 UT on a global scale,  
305 when the X2.2 SF occurred. Since the ROTI was calculated at a 30 s rate over 5 min,

306 the time title above the subfigures identifies the start time of the window computing the  
307 ROTI. From Figure 4a, we can see that the global distribution of the ROTI was around  
308 0.1 TECU/min most of the time. From the point of the spatial variations, ROTIs at  
309 high-latitude were larger than that at any other area. Numerically, Figure 4a shows that  
310 the magnitude of the ROTI at high latitude in the northern hemisphere was around 0.2  
311 TECU/min. This can be explained by the complex particle precipitation in the  
312 high-latitude area (Juan et al., 2018). In addition, we can see a quite apparent increase  
313 of the ROTI, due to the almost instant photoionization, to the magnitude of 0.25  
314 TECU/min in the snapshot of 09:00 UT and 09:05 UT during the dayside when the  
315 X2.2 SF erupted.

316 Correspondingly, Figure 4b depicts the variations of the global ROTI from 11:45 UT  
317 to 12:10 UT, during which time the X9.3 SF occurred. The largest difference of the  
318 ionospheric response to the X2.2 SF in Figure 4a and X9.3 SF in Figure 4b was the  
319 magnitude of the ROTI increase. From Figure 4b, we can see that the ROTI increased  
320 apparently to the magnitude of more than 0.5 TECU/min from 11:55 UT and decayed  
321 at 12:05 UT. And we can find that the variations of the ROTI in Figure 4 were  
322 consistent with the vROT fluctuations in Figure 3, indicating the different level of the  
323 ionospheric responses to the different X-class SF.

324 We now turn our attention to the kinematic PPP solutions producing positioning  
325 errors during the SF events depicted in Figure 5. Since the sampling rate of the GPS  
326 data was 30 s, we computed the root mean square (RMS) of the positioning errors  
327 every 5 minutes to match the ROTI analysis. The time description above the  
328 subfigures in Figure 5 indicates the start time of the 5 minutes window to compute the  
329 3-Dimension (3D) RMS of the positioning errors.

330 From Figure 5a, we can see that the positioning errors during the selected time were  
331 quite stable. The 3D RMS positioning errors of most of the stations ranged from 0.10  
332 m to 0.20 m. By comparing with Figure 4a, when the ionospheric response at 09:00  
333 UT on the dayside increased, the positioning errors did not experience large variations  
334 in Figure 5a. On the contrary, in Figure 5b, positioning errors increase from 11:55 UT,

335 the time of which was consistent with the vROT fluctuation and ROTI variations in  
336 Figure 3 and Figure 4. From Figure 5b at the snapshot of 11:45 UT and 11:50 UT, we  
337 can see that the positioning errors were at the same magnitude as that in Figure 5a.  
338 However, when the ROTI varied starting from 11:55 UT, the positioning performance  
339 severely deteriorated. The 3D RMS positioning errors for most of the stations on the  
340 dayside increased to more than 0.5 m, while those in the night side did not vary, as  
341 expected.

342 Therefore, the positioning errors in Figure 5 indicated the effect of the different level  
343 X-class SF on the positioning accuracy. The positioning errors in Figure 5 may  
344 suggest that the kinematic PPP solution can cope with the effect of the X2.2 SF rather  
345 than the X9.3 SF. It is worth noting that the SF classes are given in logarithmic units;  
346 therefore, the difference between the X2.2 and X9.3 SF is quite huge. In the  
347 Discussion Section, we tried to analyze the reasons behind the positioning accuracy  
348 reduction during the two X-class SFs.

### 349 3.3 Ionospheric response on September 10 and Kinematic PPP Error

350 We now proceed with the analysis of September 10, 2017. In the same way as in  
351 Figure 3, Figure 6 depicts the time series of the vROT along the zenith of the stations  
352 on September 10, 2017 for the same twelve stations.

353 Figure 6 depicts that most of the time the vROT varied below 0.1 TECU/min. At  
354 around 16:00 UT, when the X8.2 SF erupted, the magnitude of the vROT reached 0.2  
355 TECU/min for stations from about 149.6 W to 18.4 E. It is interesting to note that the  
356 magnitude of the vROT fluctuation during the X8.2 SF on September 10 was at the  
357 same level as that during the X2.2 SF on September 6. The small fluctuation of the ROT  
358 during the X8.2 SF lead to the inconspicuous ROTI variations at the level of about 0.1  
359 TECU/min, as can be seen from Figure 7.

360 The 3D positioning errors of the kinematic PPP solutions during the X8.2 SF are  
361 depicted in Figure 8. Different from the inconspicuous ROTI variations in Figure 7,  
362 we can see apparent positioning accuracy degradation from 15:50 UT to 16:15 UT on

363 the dayside in Figure 8. Numerically, the 3D RMS of the positioning errors varied  
364 from 0.1 m to 0.5 m during the occurrence of the X8.2 SF.

365 Comparing the positioning errors during the X9.3 SF in Figure 5b with that during the  
366 X8.2 SF in Figure 8, we can find that the pattern and distribution of the PPP errors  
367 were different. In the case of the X8.2 SF, the positioning accuracy degradation  
368 occurred the whole interval from 15:50 UT to 16:15 UT. When the eruption of the  
369 X8.2 SF reached the maximum intensity at 16:06 UT, the positioning errors did not  
370 increase. Furthermore, the distribution of the magnitude of the positioning errors  
371 during the X8.2 SF was not as even as that during the X9.3 SF. For example, at 16:05  
372 UT in Figure 8, most of the 3D RMS of the positioning errors in Europe were below  
373 0.1 m while that in north and South America ranged from 0.2 m to 0.45 m. In contrast,  
374 at 12:00 UT in Figure 5b, most of the 3D RMS of the positioning errors in Europe,  
375 Africa, as well as south America were over 0.5 m.

## 376 **4 Discussion**

### 377 4.1 The factors that impact the ionospheric response to the SF

378 The different ionospheric responses to the X2.2 and X9.3 SFs that are depicted in  
379 Figure 3 and Figure 4 indicate that the greater magnitude of the SF is, the greater the  
380 impact on the ionosphere. However, this is not the case for the ionospheric responses  
381 to the X8.2 SF on September 10, 2017, in comparison to the X2.2 SF and X9.3 SF on  
382 September 6, 2017. We expected that the vROT fluctuation and ROTI variation  
383 during the X8.2 SF should be almost the same level as those during the X9.3 SF. In  
384 fact, the results in Figure 6 and Figure 7 show that the magnitude level of the vROT  
385 fluctuation during the X8.2 SF was comparable to that during the X2.2 SF while the  
386 ROTI variation during the X8.2 SF was not even stronger than that during the X2.2  
387 SF. Therefore, there should be some other factors that impact the ionosphere.

388 We assessed the differences among the three X-class SFs; although originated from  
389 the same active regions of AR2673, the locations of the SFs were different as depicted



390 in Figure 9. The two largest flares with similar magnitudes were a disk flare (X9.3 SF  
391 on 6 September 2017) and a limb flare (X8.2 SF on 10 September 2017), respectively.

392 The response of the Thermosphere-Ionosphere system during SFs is controlled by  
393 many factors, including the location of a SF on the Sun (Qian et al., 2010; Zhang et al.,  
394 2011). Although soft X-ray enhancement is not affected by the SF location, EUV  
395 enhancement in disk flares is much larger than in limb flares of the same magnitude,  
396 especially in the wavelengths longer than 30 nm. This larger enhancement in the EUV  
397 for a disk flare was related to the difference in the optical thickness of the soft X-rays  
398 and EUV in the solar atmosphere.

399 Interplanetary Coronal Mass Ejection (ICME), which dominated the soft X-ray  
400 spectrum, is generally optically thin in the solar atmosphere. The amount of the soft  
401 X-ray that is absorbed when it travels through the solar atmosphere does not change  
402 much when the location of a SF changes from the center to the limb. Chromospheric  
403 emissions, which dominated the EUV spectrum, are often optically thick in the solar  
404 atmosphere. Absorption of optically thick emissions was greater if a SF occurs on the  
405 limb, due to the longer optical path length. Therefore, limb flares imply less  
406 enhancement at EUV wavelengths.

407 Figure 10b depicts the EUV observations from SOHO in the wavelengths of 26-34  
408 nm and 0.1-50 nm. As it can be seen, both EUV fluxes during the X8.2 limb flare on  
409 September 10, 2017 were much larger than those during the 9.3 disk flare on  
410 September 6, 2017. Therefore, in the case of the X8.2 SF, the intensity of the EUV  
411 radiation might not be attributed to the low ROTI variations of the ionosphere  
412 responses.

413 In addition, according to the model simulations in Qian et al. (2011), SFs with similar  
414 magnitudes and Sun locations generated variable responses in the thermosphere and  
415 ionosphere. In particular, the risetime and decay time of SFs was an important factor  
416 in determining the responses. Specifically, increasing the decay time of the control SF  
417 had a large effect in enhancing the thermosphere and ionosphere responses, whereas

418 reducing the risetime of the control SF had a relatively small effect in weakening the  
419 responses.

420 In the present study, the occurrence time of the SFs is presented in Table 2. The  
421 duration and decay time of the X8.2 SF lasted for 56 minutes and 25 minutes, whereas  
422 those of the X9.3 SF were about 17 minutes and 8 minutes. Although the decay time  
423 of the X8.2 SF was longer than that of the X9.3 SF, the ionospheric response to the  
424 X8.2 SF was much weaker than that to the X9.3 SF, which is different to the  
425 simulation results in Qian et al. (2011). Therefore, we deduced that the SF energy  
426 may be liberated with a longer duration and decay time, leading to the weak  
427 ionospheric response to the X8.2 SF.

428 Qian et al. (2019) discussed the SF and geomagnetic storm effects on the  
429 thermosphere and Ionosphere during 6-11 September 2017. It was found that, from  
430 the point of the TEC variation, the maximum TEC increases during the X9.3 SF and  
431 X8.2 SF were very similar, reaching a magnitude of about 4 TECU. Thus, the  
432 ionosphere responses to the X8.2 SF presented in Qian et al. (2019) and the present  
433 study are quite different. This discrepancy might be attributed to the metrics used. The  
434 TEC metric was adopted by Qian et al. (2019) while the vROT and ROTI metrics  
435 were used in this study. The TEC data was the standard vertical TEC data product  
436 binned into 1° latitude by 1° longitude with a 5-min resolution from the Coupling,  
437 Energetics, and Dynamics of Atmospheric Regions (CEDAR) Madrigal database at  
438 <http://cedar.openmadrigal.org>. On the one hand, the slant TEC data was retrieved by  
439 levelling the large-noise pseudorange measurements with the high-accuracy  
440 carrier-phase measurements (Ciraolo et al., 2007; Nie et al., 2018a). On the other hand,  
441 the vertical TEC data was projected from the slant TEC data (Li, et al., 2018).  
442 Therefore, the accuracy of the vertical TEC data was reduced by both the levelling  
443 and projecting errors. In comparison, the relative slant TEC data, as ROT in the  
444 present study, only used the high-precision carrier-phase measurements; therefore, the  
445 vROT and ROTI metrics were more sensitive to the ionosphere responses during the

446 SF. But still, the exact factors that affect the X8.2 limb SF on September 10, 2017  
447 remain to be discovered.

#### 448 4.2 The impacts of SFs on the kinematic PPP error

449 One of the key features of the SF effects is the SID. Taking the X9.3 SF in Figure 3 as  
450 an example, the  $vROT$  fluctuation increased from around 0.1 TECU/min before the  
451 eruption to around 1.5 TECU/min when the eruption reached the maximum. The  
452 impact of the SID on the positioning depends on the number of frequencies involved in  
453 the computation. On the one hand, single frequency positioning will suffer seriously  
454 because the SID cannot be modelled precisely by broadcast models such as the  
455 Klobuchar model (Klobuchar, 1987), or the post-processing models such as the global  
456 ionosphere maps computed by the IGS (Hernández-Pajares et al., 2009; Nie et al.,  
457 2018b; Ren et al., 2016). On the other hand, the refraction effects of the ionospheric  
458 delay error (including the SID), can be mitigated by up to 99.9% through the  
459 dual-frequency IF combination (Bassiri & Hajj, 1993; Petrie et al., 2011). As shown in  
460 Table 1, the kinematic PPP solution in this study adopted the IF combination, thus the  
461 degraded positioning accuracy presented in Figure 5 cannot be attributed to the  
462 SF-induced TEC increase directly.

463 The SID may have a serious impact on the pre-processing performance of the GPS  
464 observables, especially for the CS detectors. Those detectors ensure that the  
465 carrier-phase measurements do not experience discontinuities or jumps. As shown in  
466 Equation (1), the HMW combination does not contain any ionosphere delay and it is  
467 efficient at detecting large CSs despite using pseudorange observables with large-noise.  
468 The PGF combination aims at detecting small CSs as it only uses high-precision carrier  
469 phase observables. However, there exists ionospheric delay effects in the PGF  
470 combination. The premise of the CS detection based on the PGF combination is that the  
471 ionosphere varies little in adjacent epochs, thus most of the ionospheric delay effects in  
472 the PGF combination can be mitigated.

473 Under the default threshold, that is 0.05 m (0.48 TECU) for the PGF combination (1  
474 meter corresponds to 9.52 TECU for the PGF combination), the kinematic PPP  
475 performances presented in Figure 5 and Figure 8, result in typical positioning accuracy  
476 at the level of 0.1 m to 0.2 m even during the X2.2 SF. However, the positioning  
477 accuracy deteriorated during the eruption of the X9.3 SF. The reason may be that the  
478 SID rate was faster than the threshold of PGF combination divided by the sampling  
479 interval (i.e. 0.48 TECU/ 30 s equals 0.96 TECU/min), leading to a false of the CS  
480 detection. When the carrier-phase observables are flagged as CSs, we have to  
481 re-estimate the ambiguities as unknown parameters by increasing their noise. When  
482 multiple CSs are detected simultaneously, the number of the unknown parameters  
483 increases. Hence, the redundancy of observations decreases, ultimately weakening the  
484 estimation of all unknowns and deteriorating the positioning accuracy and stability (Nie  
485 et al., 2022a; Nie et al., 2022b). To verify such hypothesis, we loosen the CS threshold  
486 of the PGF combination from 0.05 m (0.48 TECU) to 0.20 m (1.92 TECU), and the  
487 results during the X2.2, X9.3 and the X8.2 SFs were depicted from Figure 11 to Figure  
488 14.

489 Figure 11 depicts that the kinematic PPP errors during the X2.2 and X9.3 SFs  
490 obtained with the CS threshold of the PGF combination enlarged to 0.20 m (1.92  
491 TECU). We can see that during the X9.3 SF, the kinematic PPP errors decreased from  
492 above 0.50 m to the typical values comprised from 0.10 m to 0.20 m.

493 To validate the relationship between the positioning errors and the CS threshold in  
494 detail, we selected VILL as a representative station from IGS network. The kinematic  
495 PPP solution for station VILL on September 6, 2017 with different CS threshold are  
496 presented in Figure 12. The left and middle columns in Figure 12 depicted the  
497 kinematic PPP analysis when the PGF CS threshold was 0.05 m (0.48 TECU) and 0.20  
498 m (1.92 TECU), respectively. Since data resolution plays an important role in the PPP  
499 errors (Bahadur, B., & Nohutcu, M., 2021), the kinematic PPP solution for VILL with  
500 high-rate data sampling of 1-second is presented in the right column of Figure 12.  
501 Beneficial from the high-rate sampling data, we computed the amplitude index (S4)

502 according to the suggested Mrak et al., (2020) and Luo et al., (2020), using the  
503 signal-to-noise ratio (SNR) observations.

504 Figure 12a depicts a notable positioning degradation during the X9.3 SF at about  
505 12:00, simultaneous with large ROTI values. A previous weak ROTI variation occurs  
506 at about 09:00 during the X2.2 SF, but the accuracy of the kinematic PPP maintains  
507 the typical accuracy at the level of 0.1 m. The positioning errors are consistent with  
508 those presented in Figure 5.

509 The ROT fluctuations of each tracked satellite from VILL are presented in Figure 12d.  
510 Satellite ROT fluctuations are below 0.1 TECU/min most of the time, except during  
511 the X2.2 and X9.3 SFs, that increase to 0.2 TECU/min and more than 1.5 TECU/min,  
512 respectively. The satellite ROT fluctuations in Figure 12d are consistent with the  
513 vROT fluctuations for the station VILL in Figure 3. Figure 12j depicts the number of  
514 tracked satellites, the used satellites, the flagged CS satellites and the excluded  
515 satellites in the kinematic PPP solution.

516 From Figure 12j, we can see that the receiver of VILL can track 8 to 10 satellites most  
517 of the time and most of the tracked satellites were used in the kinematic PPP solution.  
518 The position dilution of precision (PDOP) around 2 indicates a good geometry.  
519 Besides, we can see that there always existed 1 to 2 satellites that were flagged as CS  
520 satellites. However, during the X9.3 SF at around 12:00, we can observe that most of  
521 the used satellites were flagged as CS satellites. For example, at 11:57, all of the used  
522 6 satellites were detected as CS satellites. Correspondingly, the positioning errors  
523 increased to more than 1.0 m.

524 From the ROT fluctuations in Figure 12d, we deduce the epoch-wise ionosphere  
525 variation  $\Delta I$  (i.e., ROT) was around 0.20 TECU/min and 1.50 TECU/min (i.e., 0.10  
526 TECU and 0.75 TECU every two consecutive epochs of 30 seconds) during the X2.2  
527 SF and X9.3 SF. Therefore, the threshold of the PGF observable  $\Delta\phi_{\text{PGF}}$  should be  
528 larger than 0.10 TECU and 0.75 TECU to cope with the rapid ionosphere variations  
529 under the SF condition. Otherwise, signals continuously tracked are falsely signaled

530 with CS. In this regard, the commonly used PGF threshold of 0.05 m (0.48 TECU)  
531 works with the ionosphere variations during the peak eruption of the X2.2 SF but not  
532 with that of the X9.3 SF.

533 In comparison, Figure 12b depicts that the positioning errors did not increase during  
534 the X9.3 SF when the CS threshold of the PGF combination was 0.20 m (1.92 TECU).  
535 Especially, comparing Figure 12j with Figure 12k, the most remarkable difference  
536 between them is the number of the satellites flagged with CS. Indeed, in Figure 12k,  
537 most of the satellites during the whole day were not flagged as CS satellites. In  
538 particular, at 11:57, all of the used 6 satellites were not flagged as CS satellites. From  
539 the positioning performance in Figure 12a and Figure 12b, we can deduce that most of  
540 such flagged CS satellites were falsely detected due to the tight CS threshold of the  
541 PGF combination.

542 From Figure 12c, the positioning errors become normal using the 1-second high-rate  
543 data, despite the CS threshold of the PGF combination was 0.05 m (0.48 TECU). We  
544 also notice that the distribution of the amplitude scintillation index S4 is below 0.2 rad  
545 most of the time, including the period of the X2.2 and X9.3 SF, which is different  
546 from the ROTI. It indicates that the amplitude scintillation does not play the role in  
547 the increased PPP errors when using the 30-second sampling data. Comparing the  
548 results in the left with those in right column, we can find that the detected CSs are  
549 much reduced when using high-rate sampling data, indicating that it is much easier to  
550 eliminate erroneous CSs in higher resolution data. Therefore, we conclude that the  
551 kinematic PPP solution is very sensitive to CSs which are difficult to detect in  
552 30-second data. On the other hand, the CS detection using the high-rate sampling data  
553 is more robust for the kinematic PPP solution.

554 From Figure 11 and Figure 12, we deduce that most of the SID effect on the  
555 dual-frequency positioning can be mitigated quite well by optimizing the threshold of  
556 the CS detection observables, especially for the PGF combination. However, for the  
557 positioning accuracy degradation during the X8.2 SF on September 10, 2017, the  
558 optimization of the CS threshold of the PGF combination did not produce positive

559 results, as presented in Figure 13 and Figure 14. Comparing Figure 13 with Figure 8,  
560 we can see that the magnitude and distribution of the PPP errors during the X8.2 SF  
561 presented minor differences under the different thresholds of the CS detection  
562 observables.

563 In order to analyze the mechanism behind the positioning degradation, we selected  
564 two representative stations, PALM and FLRS, from the IGS network. The kinematic  
565 PPP for station PALM (left column) and FLRS (middle and right column) on  
566 September 10, 2017 under the threshold of 0.20 m (1.92 TECU) for the PGF  
567 combination were presented in Figure 14. The data sampling in left and middle  
568 column is 30-second while that in the right column is 1-second. Similarly, the  
569 amplitude index S4 for station FLRS was computed using the high-rate sampling data.

570 From Figure 14a and Figure 14b, we can see that the positioning errors of both stations  
571 increased to more than 0.5 m during the X8.2 SF. However, no apparent ROTI  
572 variations are observed. Indeed, ROT fluctuations of each tracked satellite, as  
573 presented in Figure 14d and Figure 14e, remained below 0.1 TECU/min most of the  
574 time, which is consistent with the results presented in Figure 6. This may explain the  
575 reason that the positioning errors did not change with different thresholds of 0.05 m  
576 and 0.20 m for the PGF combination. In the case of the X8.2 SF event, the  
577 ionospheric effect, as represented by the ROT fluctuations of each satellites, was  
578 small at the same level of the X2.2 SF due to the limb-effect of the X8.2 SF.  
579 Therefore, the customary threshold of 0.05 m was adequate to cope the SID rate of  
580 both the X2.2 and X8.2 SF, let alone the threshold of 0.20 m.

581 As for the mechanism behind the positioning degradation during the X8.2 SF, we  
582 assessed the 5-min Root Mean Square (RMS) of the carrier-phase and pseudorange  
583 residuals depicted in Figure 14g and Figure 14h. Since the precision of the GPS  
584 carrier-phase measurement is about 100 times higher than that of the pseudorange  
585 measurement, the residuals of the carrier-phase measurement is more critical to the  
586 accuracy of the kinematic PPP solution. From Figure 14g and 14h, we notice the  
587 significant increase of the RMS of the carrier-phase residuals, simultaneously with

588 large positioning errors. In comparison, the RMS of the carrier-phase residuals were  
589 stable during the X2.2 and X9.3 SF as presented in Figure 12g and Figure 12h. In fact,  
590 the carrier-phase and pseudorange residuals of the PPP solutions can reflect the  
591 precision of the corresponding GPS measurements (Juan et al., 2017). Therefore, the  
592 increase of the carrier-phase residuals may suggest that the accuracy of the GPS  
593 carrier-phase measurements were affected during the X8.2 SF. The accuracy  
594 degradation of the GPS carrier-phase measurements can also be revealed from Figure  
595 14j and Figure 14k, from which we can see that some satellites are excluded from the  
596 solution. The reason for such exclusion is for satellites with carrier-phase residuals  
597 larger than 4 times of its variance. This satellite exclusion occurs frequently at station  
598 PALM and FLRS during the X8.2 SF. It is noted that though the data sampling is  
599 increased from 30-second to 1-second, the positioning accuracy during the X8.2 SF can  
600 still be affected, as shown in Figure 14c. Since the degraded accuracy of the GNSS  
601 measurement cannot be mitigated by the improvement of the data resolution, the  
602 positioning errors still exist.

603 As presented in Figure 2, the X8.2 SF on September 10, 2017 occurred during the  
604 recovery phase of the geomagnetic storm. The combined effects of the magnetic field  
605 disturbance and the X8.2 SF may cause the accuracy degradation of the GPS  
606 carrier-phase measurements and, consequently, the positioning accuracy.

## 607 **5 Conclusions**

608 The present study focused on the ionospheric responses during different X-class SFs,  
609 including the strongest X9.3 SF in Solar Cycle 24 on September 6-10, 2017. The  
610  $v$ ROT and ROTI metrics were exploited as representative indicators of the SID  
611 induced by the SFs. We discussed different factors that affected the ionosphere  
612 disturbances, and then on the accuracy of the kinematic PPP positioning. For such  
613 purpose, we evaluated the estimated coordinate variations of 700+ GPS stations  
614 available worldwide. The major findings from the analysis are summarized as  
615 follows:



- 616 (1) We confirm that the ionospheric response to the SF is related to the intensity level,  
617 location and duration time of the three X-class SFs occurred in September, 2017.  
618 ROTIs for the stations on the dayside can reach up to 0.2 TECU/min and 1.5  
619 TECU/min during the maximum eruption of the different magnitude level of X2.2  
620 and X9.3 SF, respectively. In contrast, the maximum ROTI variations during the  
621 eruption of the X8.2 SF, only reach 0.2 TECU/min. Although the magnitude of  
622 the X8.2 SF was only a little weaker than that of the X9.3 SF, the X8.2 SF  
623 occurred on the limb side whereas the X9.3 SF occurred near the center of the  
624 solar disk. In addition, the longer duration and decay time may also be attributed  
625 to the inconspicuous ionosphere responses to the X8.2 SF on September 10, 2017.
- 626 (2) The SID induced by the SF can degrade the accuracy of the kinematic PPP  
627 solution on the dayside seriously. This kind of accuracy degradation can be related  
628 to the CS algorithms or the threshold of the CS detection observable. When the  
629 CS threshold of a combination containing the ionosphere effect is too tight,  
630 satellites can be falsely flagged as CS, re-initializing the ambiguity estimation in  
631 the PPP solution and deteriorating the positioning accuracy. When loosen the CS  
632 threshold of the PGF combination from 0.05 m (0.48 TECU) to 0.20 m (1.92  
633 TECU), the positioning accuracy of the PPP solutions on the dayside during the  
634 X9.3 SF improved from larger than 0.50 m to the typical level of 0.10 to 0.20 m.
- 635 (3) The combined effects of the magnetic field disturbance and the SF, as shown in  
636 the case of the X8.2 SF, can also degrade the accuracy of the kinematic PPP  
637 solution on the dayside. But the pattern of the PPP errors induced by the combined  
638 effect was different to those induced by the SID. In this case, the optimization of  
639 the CS threshold did not improve the positioning accuracy. The combined effect  
640 of the SF may deteriorate the precision of the GPS measurements, especially the  
641 carrier-phase measurement. Under these conditions, the accuracy of the  
642 positioning decreases.

643 The impacts of the different SFs on the ionosphere and the kinematic PPP solutions  
644 are not monotonous. For the SID effect of the different SFs on the positioning, it is  
645 suggested that the CS algorithms should be carefully developed further. Especially,  
646 the threshold should be set adaptively according to the intensity of the SF. For the  
647 combined effect of the magnetic field disturbance and the SF on the positioning, more  
648 studies are needed to improve the positioning accuracy and stability in the future.

649

### 650 **Open Research**

651 The raw GPS data is from <ftp://igs.ign.fr> as well as <https://cddis.nasa.gov/archive/>.  
652 The 1-min averaged X-ray data in the wavelength of 0.1–0.8 nm are observed by the  
653 Geostationary Operational Environmental Satellite (GOES) 13 as well as 15 at  
654 <http://www.swpc.noaa.gov/products/goes-x-ray-flux>. The EUV flux are from the  
655 Solar Heliospheric Observatory (SOHO) observations at  
656 <https://dornsifecms.usc.edu/space-sciences-center/download-sem-data/>. The ACE  
657 solar wind and IMF data are provided from CDAWeb at <https://cdaweb.gsfc.nasa.gov/>.  
658 The F10.7 measurement, longitudinally symmetric disturbances index in the  
659 horizontal direction H (SYM-H) and Auroral Electrojet indices, are provided by the  
660 Goddard Space Flight Center (GSFC) from <https://omniweb.gsfc.nasa.gov/>. The  
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662 <http://www.rtklib.com/>.

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675

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934 **Figure captions**

935 **Figure 1.** Worldwide distribution of the used GPS stations. The twelve yellow  
936 pentagrams represent the selected stations to display the ionosphere responses along  
937 the zenith of the stations in the Experiment and Results section. The black solid line  
938 indicates the magnetic equator.

939 **Figure 2.** The time series of (a) the X-ray flux from GOES satellites; (b) F10.7 solar  
940 flux;(c) IMF Bx, By and Bz components;(d) the geomagnetic SYM-H index and (e)  
941 auroral electrojet AE/AL/AU indices during September 3 to September 13, 2017. The  
942 shaded areas highlight September 6 and September 10 when the X-class SF events  
943 occurred.

944 **Figure 3.** The time series of vROT along the zenith of the stations on September 6,  
945 2017. Twelve stations around the world were selected and displayed according to their  
946 longitude distribution. The shaded areas represent the occurrence time of the X2.2 and  
947 X9.3 SF.

948 **Figure 4.** Snapshots of global temporal-spatial variations of ROTI during the (a) X2.2  
949 SF and (b) X9.3 SF on September 6, 2017. The gray shade indicates the night side.  
950 The black solid line represents the magnetic equator. The twelve yellow pentagrams  
951 represent the selected stations to display the ionosphere responses along the zenith of  
952 the stations in Figure 3. The CS threshold of the PGF observable is 0.2 m when  
953 computing the ROTI.

954 **Figure 5.** Snapshots of 3D kinematic PPP errors during (a) X2.2 SF and (b) X9.3 SF  
955 on September 6, 2017. The gray shading indicates nightside. The black solid line  
956 represents the magnetic equator.

957 **Figure 6.** The time series of vROT along the zenith of the stations on September 10,  
958 2017. The twelve stations are the same as those in Figure 3. The shaded area  
959 represents the occurrence time of the X8.2 SF.

960 **Figure 7.** Snapshots of 5 min ROTI during the X8.2 SF on September 10, 2017. The

961 gray shading indicates night side. The black solid line represents the magnetic equator.  
962 The twelve yellow pentagrams represent the selected stations to display the  
963 ionosphere responses along the zenith of the stations in Figure 3 and Figure 6. The CS  
964 threshold of the PGF observable is 0.2 m when computing the ROTI.

965 **Figure 8.** Snapshots of 3D kinematic PPP errors during the X8.2 SF on September 10,  
966 2017. The gray shading indicates nightside. The black solid line represents the  
967 magnetic equator.

968 **Figure 9.** Location of SFs on the solar disk: (a) the X2.2 SF on September 6, 2017, (b)  
969 the X9.3 SF on September 6, 2017, (c) the X8.2 SF on September 10, 2017 (Courtesy  
970 of NASA/SDO and the AIA, EVE, and HMI science teams from  
971 <https://sdo.gsfc.nasa.gov/>).

972 **Figure 10.** SF indexes from September 3, 2017 to September 13, 2017. (a) the X-ray  
973 flux from the GOES observations, (b) the EUV from the Solar Heliospheric  
974 Observatory (SOHO) observations  
975 (<https://dornsifecms.usc.edu/space-sciences-center/download-sem-data/>). The shaded  
976 areas highlight the day of September 6 and September 10 when the X-class SF events  
977 occurred.

978 **Figure 11.** Snapshots of 3D kinematic PPP errors during (a) X2.2 SF and (b) X9.3 SF  
979 on September 6, 2017. The gray shading indicates night side. The black solid line  
980 represents the magnetic equator. In this case, the CS threshold of the PGF observable  
981 is 0.2 m (1.92 TECU).

982 **Figure 12.** The kinematic PPP for station VILL on September 6, 2017 with different  
983 CS threshold and data sampling. The results in the left column correspond to the  
984 threshold of 0.05 m (0.48 TECU) while those in the middle column correspond to the  
985 threshold of 0.20 m (1.92 TECU) with data sampling of 30-second. The results in the  
986 right column represent the threshold of 0.05 m with high-rate data sampling of  
987 1-second. The two shaded area represents the eruption time of the X2.2 and X9.3 SF.  
988 (a) to (c) present the 3D-RMS of the positioning accuracy and the ROTI or S4

989 variations; (d) to (f) present the ROT fluctuation for the different satellites separated  
990 by the Pseudorandom Number (PRN); (g) to (i) present the 5-min RMS of  
991 carrier-phase and pseudorange residuals of the PPP solutions; (j) to (l) present the  
992 number satellites tracked, used in the solution and flagged with CSs, as well as the  
993 PDOP information of the PPP solution. The track satellites represent the number of all  
994 the satellites that tracked by the receiver, the used satellites represent the number of  
995 the satellites used in the solution, the slip satellites represent the number of the  
996 detected CS satellites and the excluded satellites represent the number of satellites that  
997 excluded from the solution due to the large residuals.

998 **Figure 13.** Snapshots of 3D kinematic PPP errors during the X8.2 SF on September  
999 10, 2017. The gray shade indicates the night side. The black solid line represents the  
1000 magnetic equator. In this case, the CS threshold of the PGF observable is 0.2 m (1.92  
1001 TECU).

1002 **Figure 14.** The kinematic PPP for station PALM (left panel) and FLRS (middle and  
1003 right panel) on September 10, 2017 under the threshold of 0.20 m (1.92 TECU) for  
1004 the PGF combination. The data sampling in left and middle column is 30-second  
1005 while that in the right column is 1-second. The shaded area represents the time of the  
1006 X8.2 SF. (a) to (c) present the 3D-RMS of the positioning accuracy and the ROTI or  
1007 S4 variations; (d) to (f) present the ROT fluctuation for the different satellites  
1008 separated by the PRN; (g) to (i) present the 5-min RMS of the carrier-phase and  
1009 pseudorange residuals of the PPP solutions; (j) to (l) present the number of the tracked  
1010 satellites, the used satellites in the solution, the satellites with CSs and the satellites  
1011 that excluded from the PPP solution, as well as the PDOP information of the PPP  
1012 solution, which have the same meaning as those in Figure 12.

1013

Figure 1.

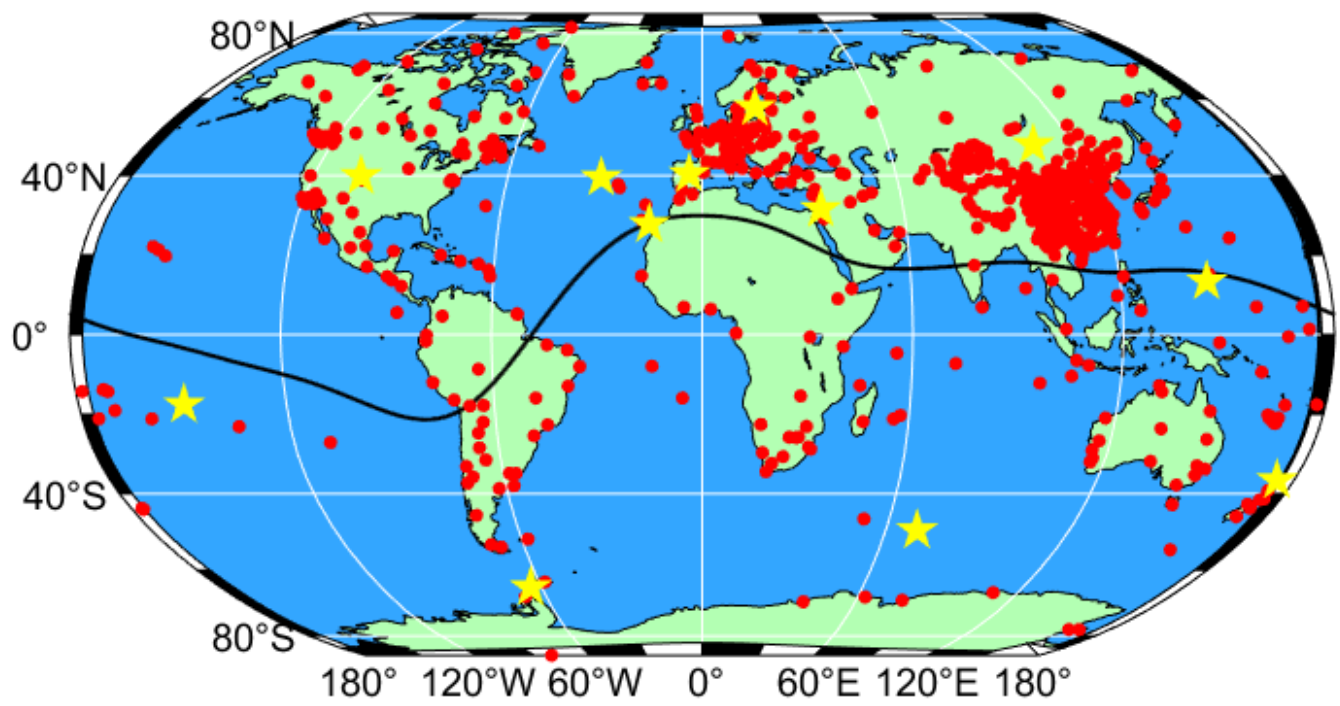




Figure 2.

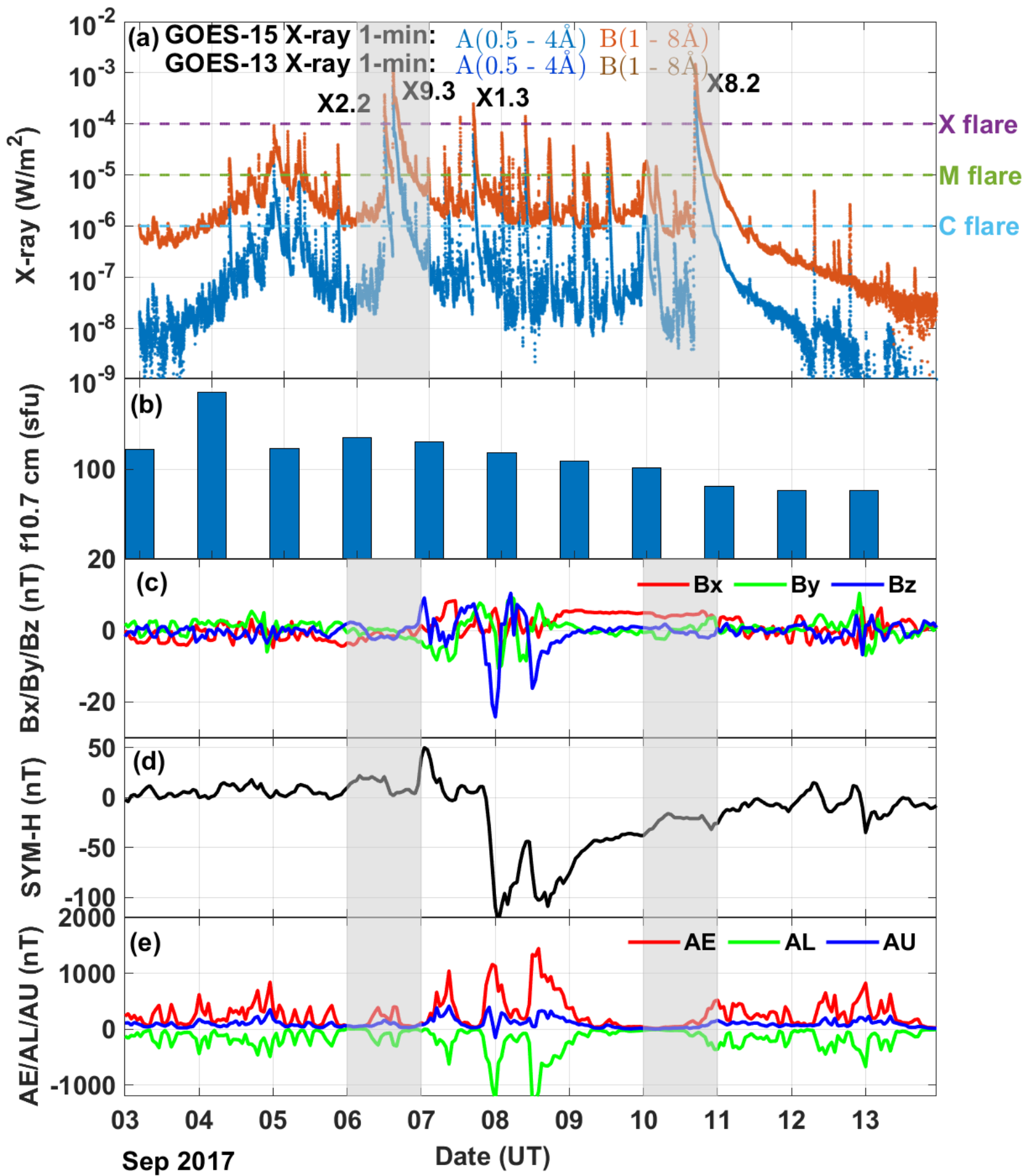


Figure 3.

vROT (TECU/min)

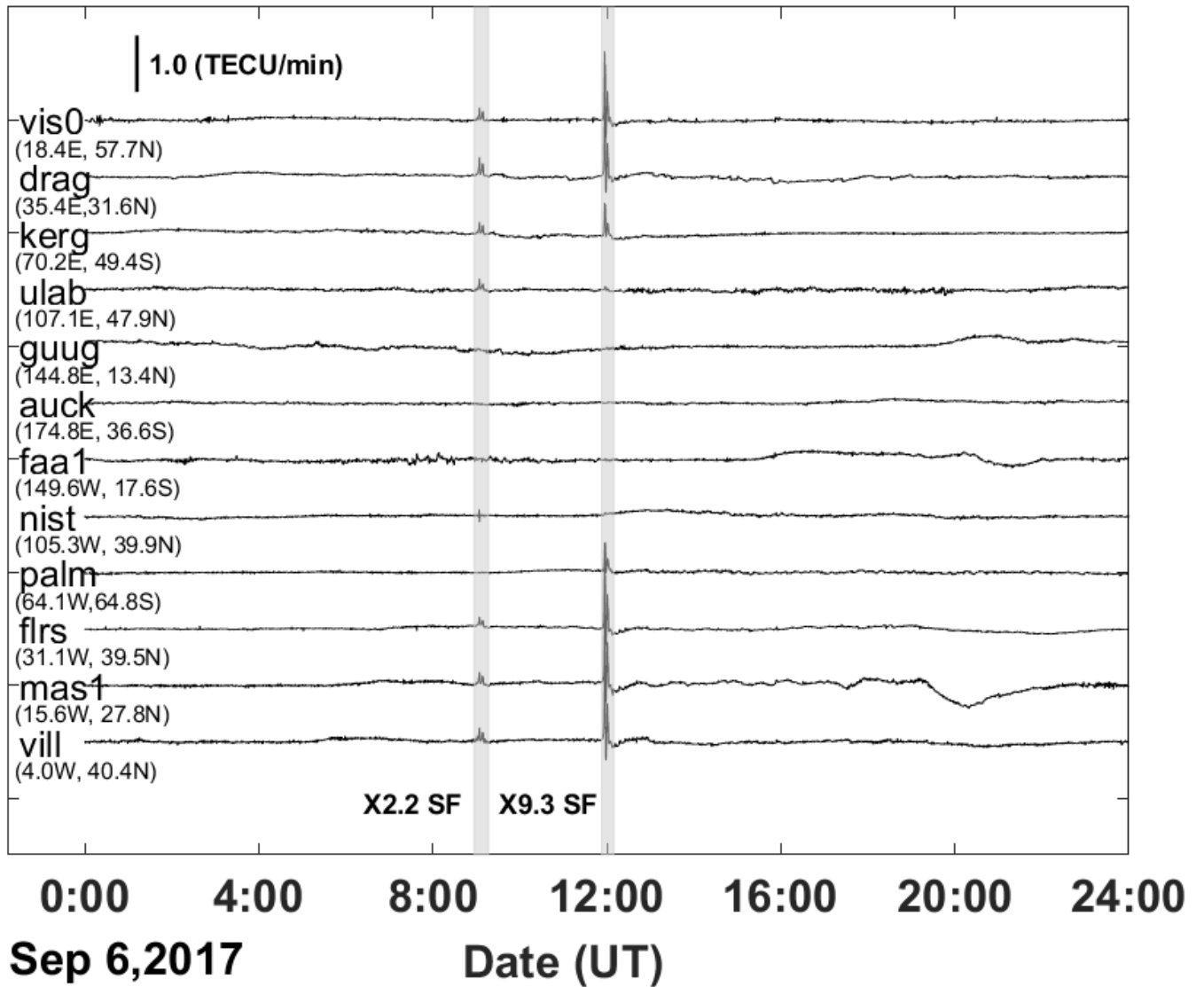


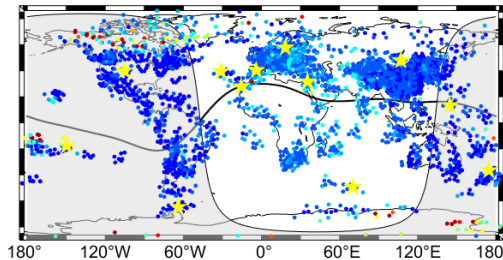
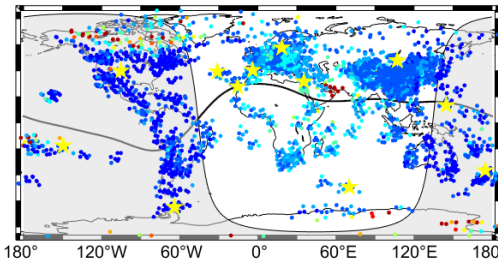
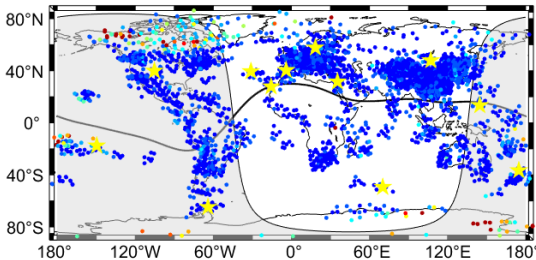
Figure 4.

(a) X2.2 ROTI

2017/09/06 08:55 UTC

2017/09/06 09:00 UTC

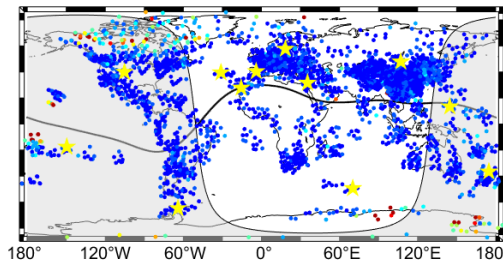
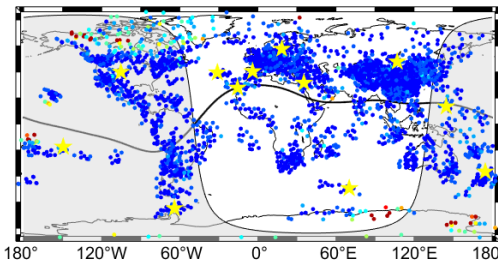
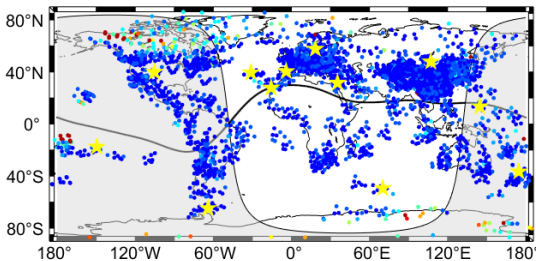
2017/09/06 09:05 UTC



2017/09/06 09:10 UTC

2017/09/06 09:15 UTC

2017/09/06 09:20 UTC

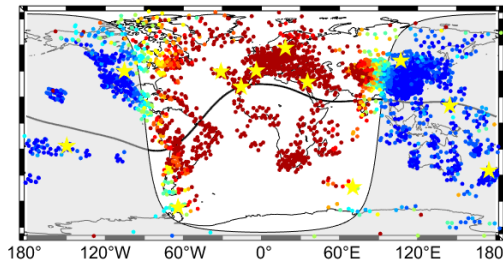
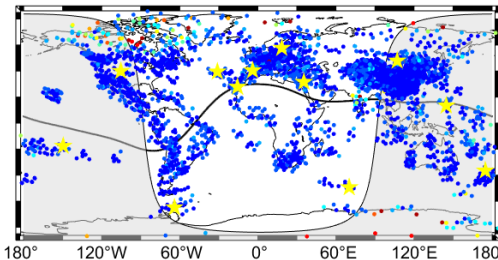
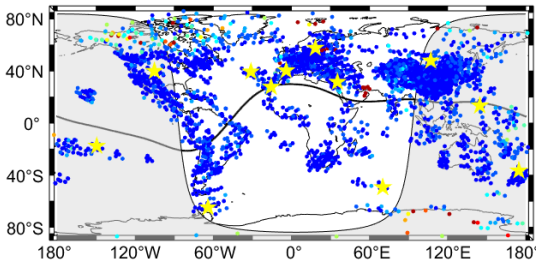


(b) X9.3 ROTI

2017/09/06 11:45 UTC

2017/09/06 11:50 UTC

2017/09/06 11:55 UTC



2017/09/06 12:00 UTC

2017/09/06 12:05 UTC

2017/09/06 12:10 UTC

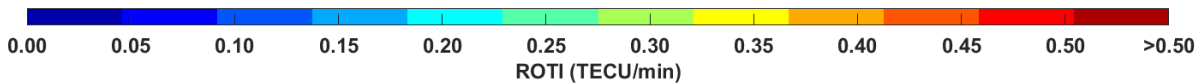
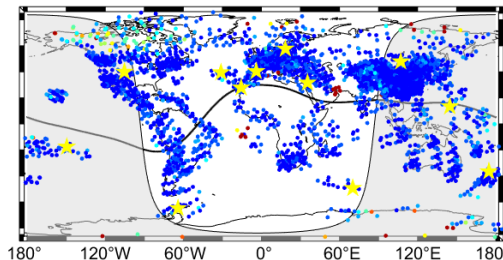
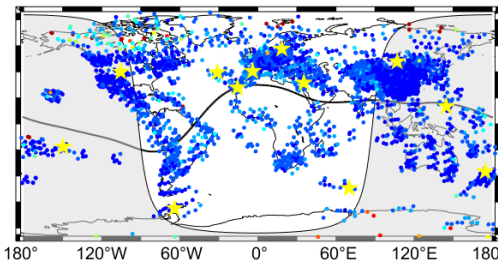
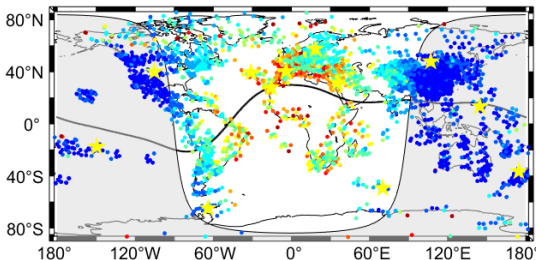
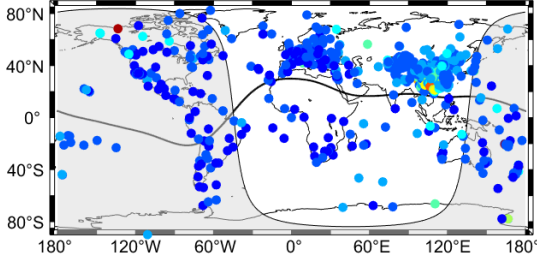


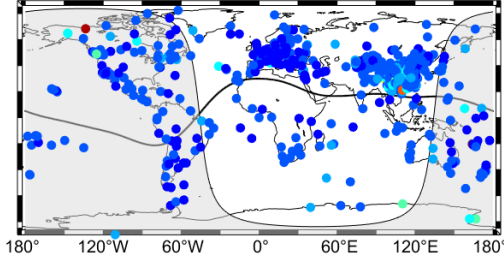
Figure 5.

(a) X2.2 PPP Error

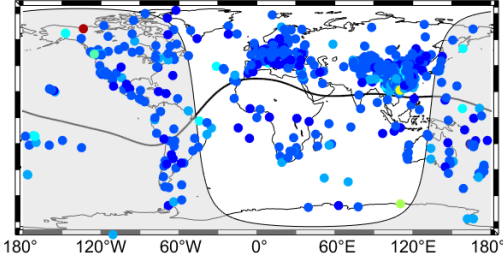
2017/09/06 08:55 UTC



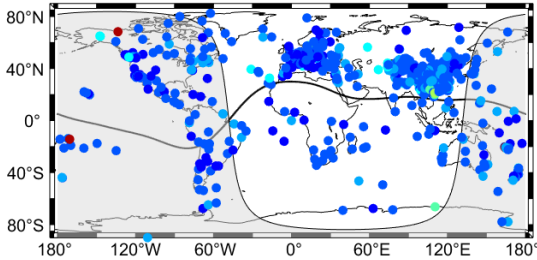
2017/09/06 09:00 UTC



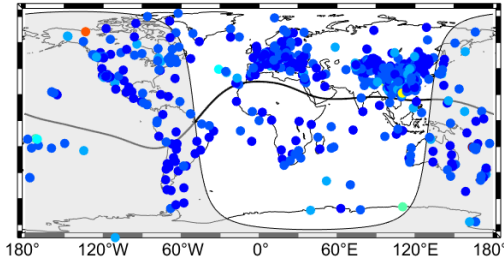
2017/09/06 09:05 UTC



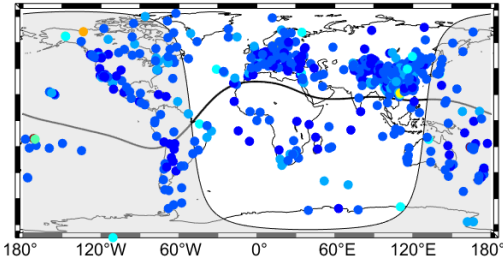
2017/09/06 09:10 UTC



2017/09/06 09:15 UTC

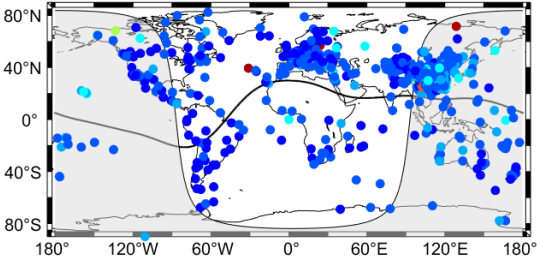


2017/09/06 09:20 UTC

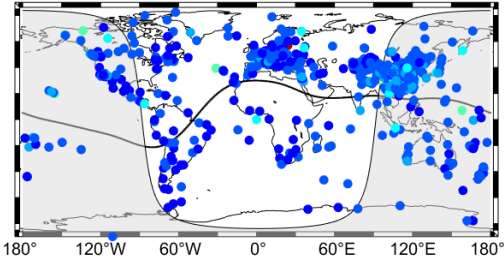


(b) X9.3 PPP Error

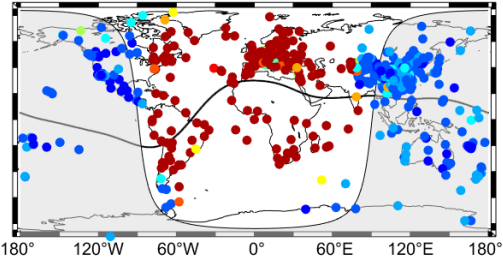
2017/09/06 11:45 UTC



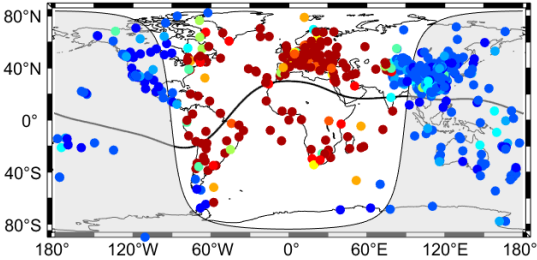
2017/09/06 11:50 UTC



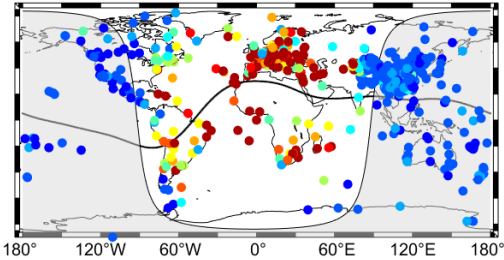
2017/09/06 11:55 UTC



2017/09/06 12:00 UTC



2017/09/06 12:05 UTC



2017/09/06 12:10 UTC

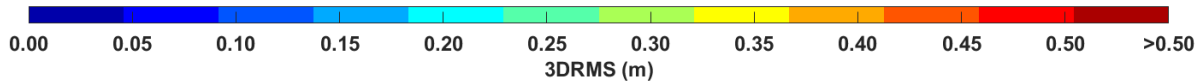
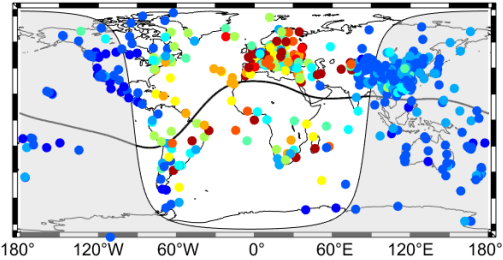




Figure 6.

vROT (TECU/min)

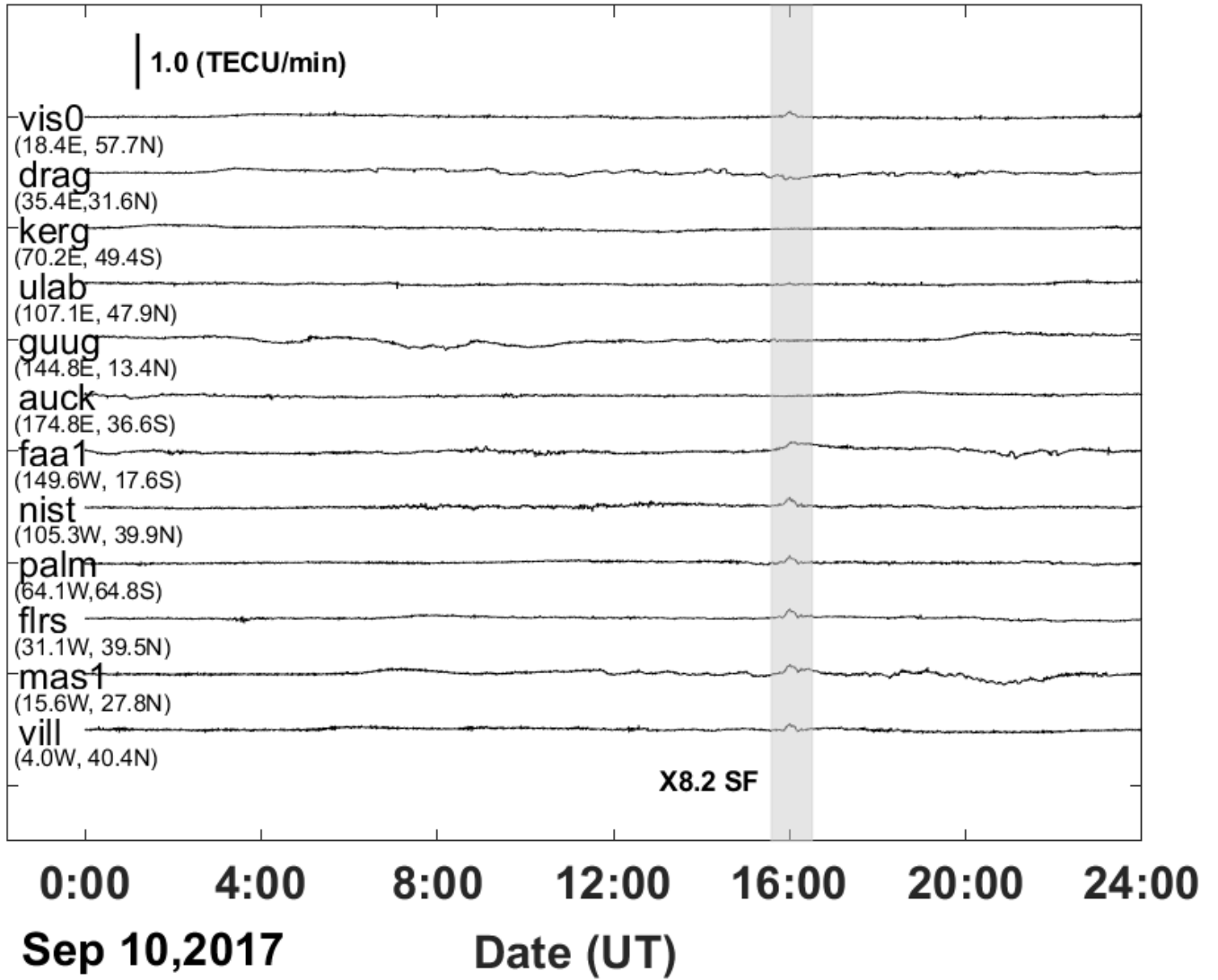
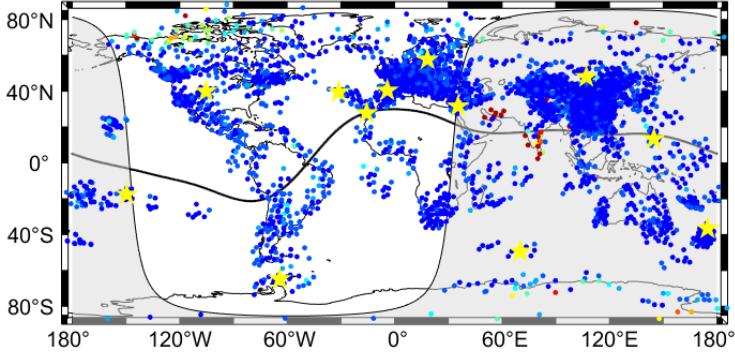
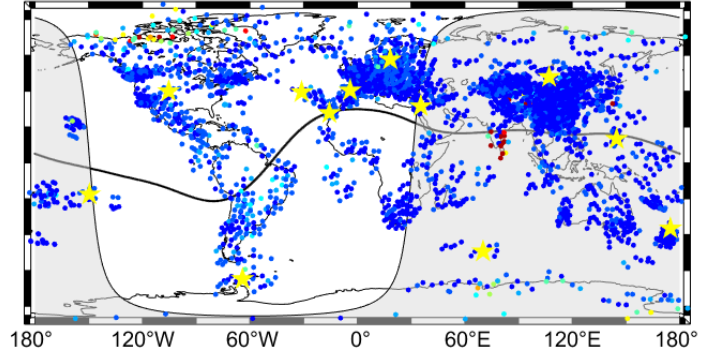


Figure 7.

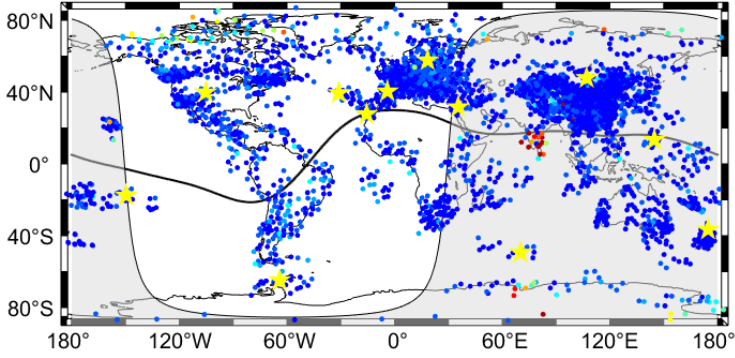
**2017/09/10 15:50 UTC**



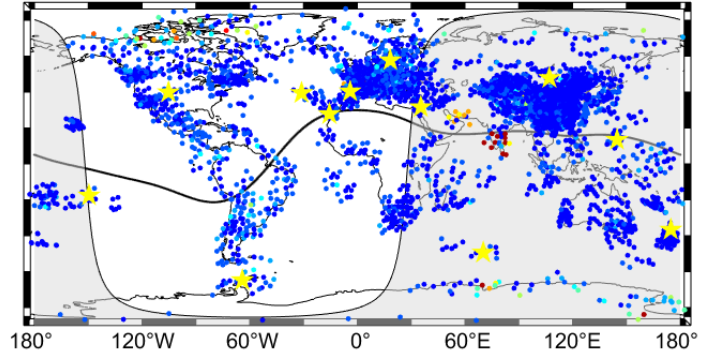
**2017/09/10 15:55 UTC**



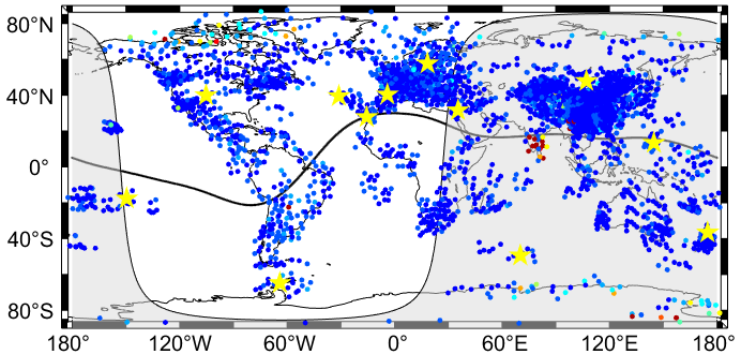
**2017/09/10 16:00 UTC**



**2017/09/10 16:05 UTC**



**2017/09/10 16:10 UTC**



**2017/09/10 16:15 UTC**

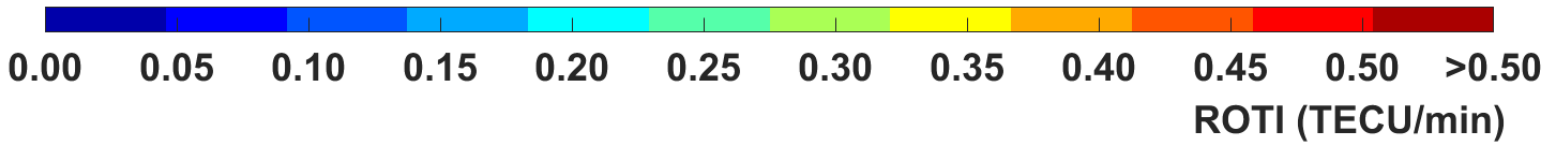
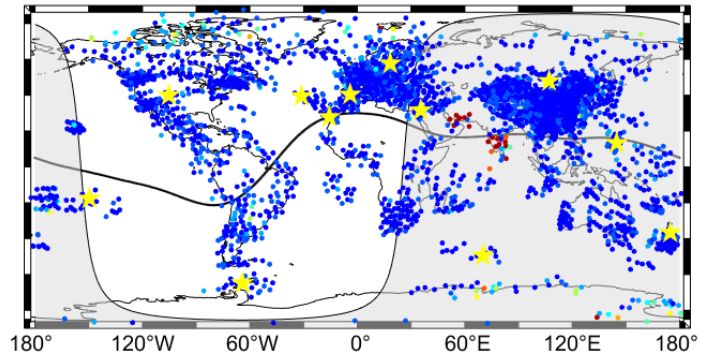
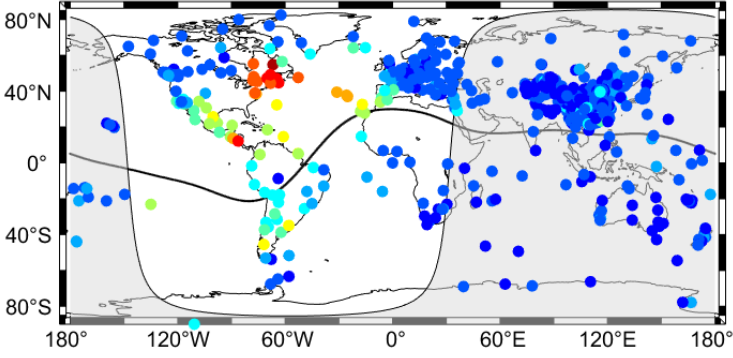
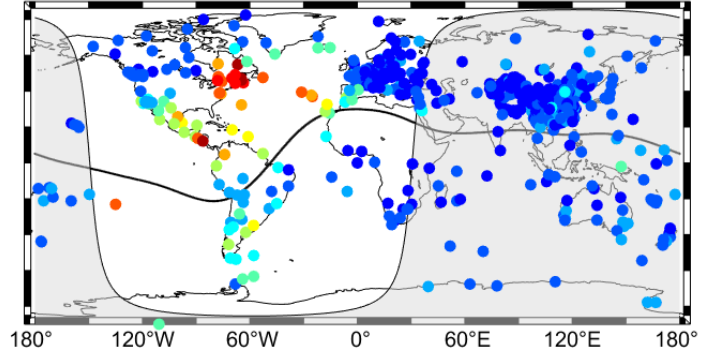


Figure 8.

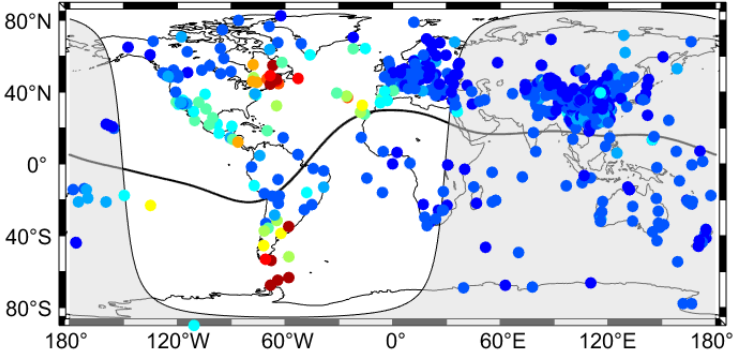
2017/09/10 15:50 UTC



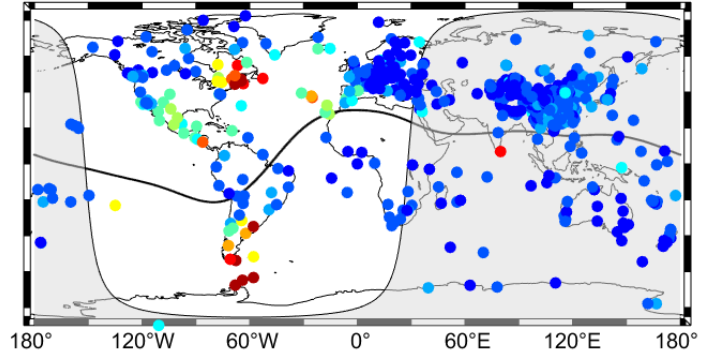
2017/09/10 15:55 UTC



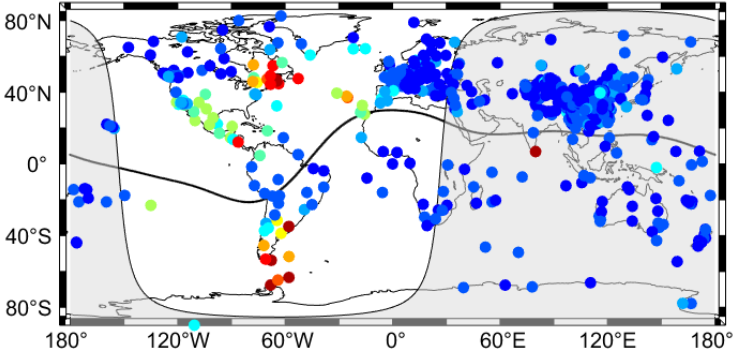
2017/09/10 16:00 UTC



2017/09/10 16:05 UTC



2017/09/10 16:10 UTC



2017/09/10 16:15 UTC

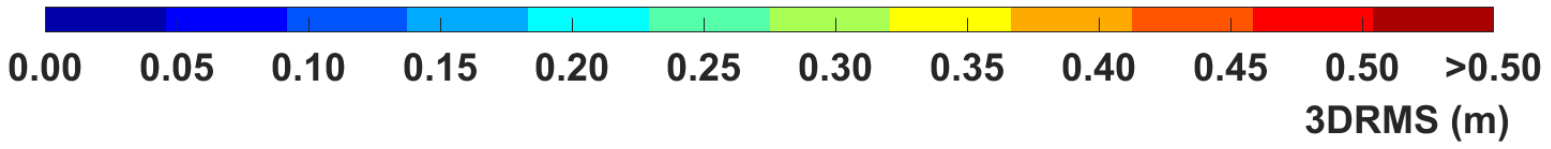
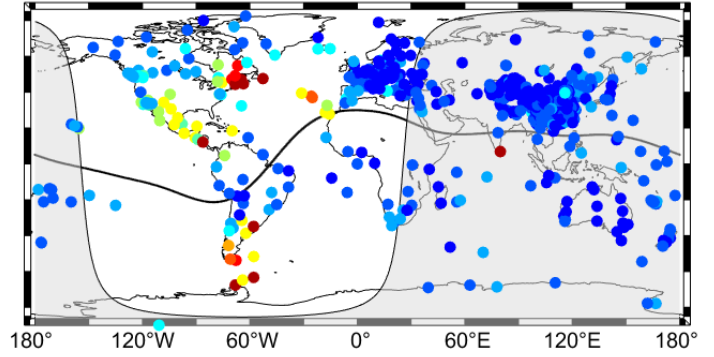


Figure 9.



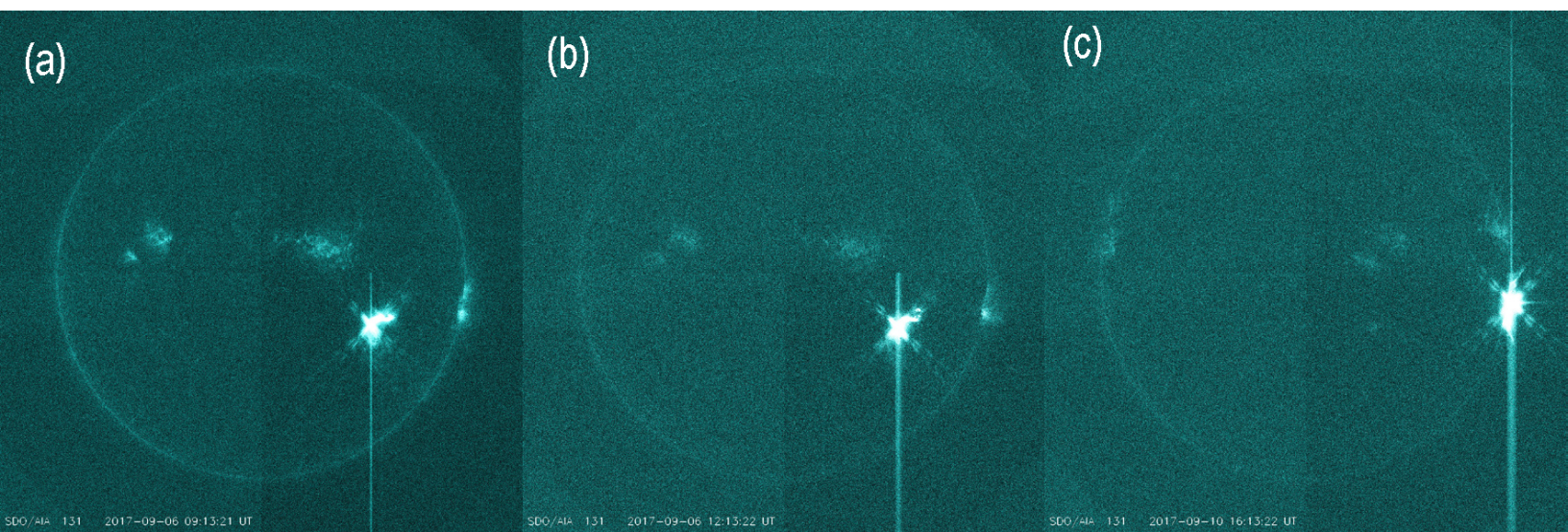




Figure 10.

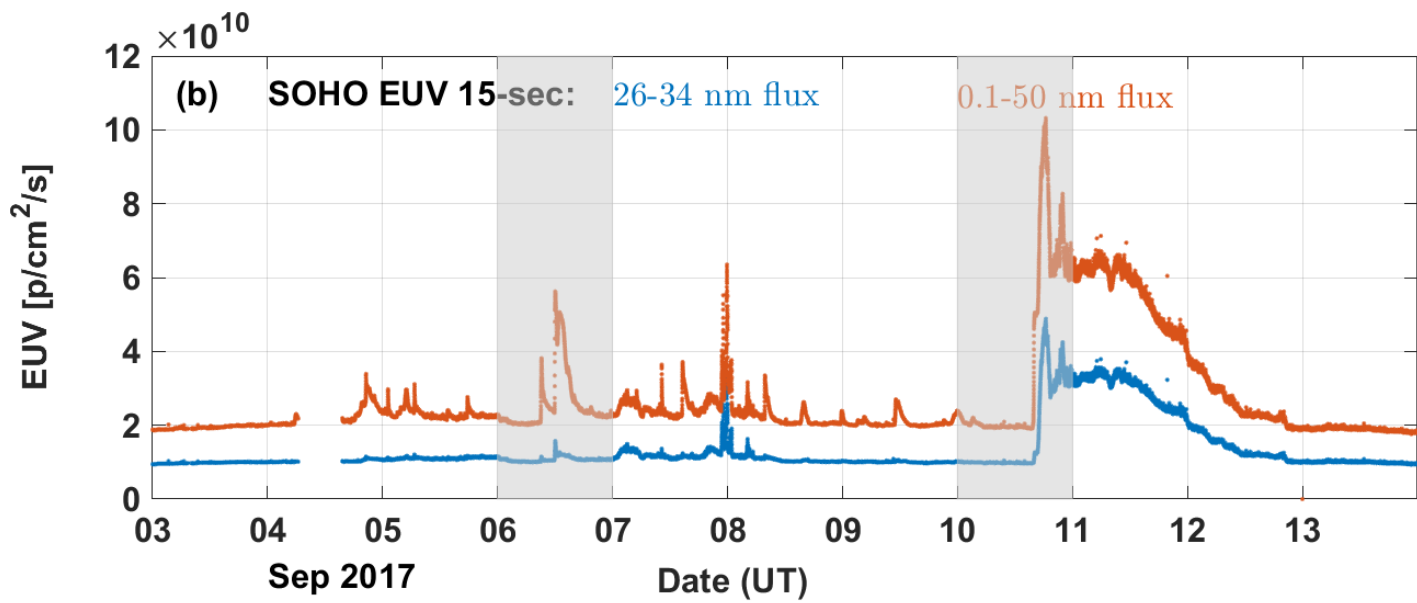
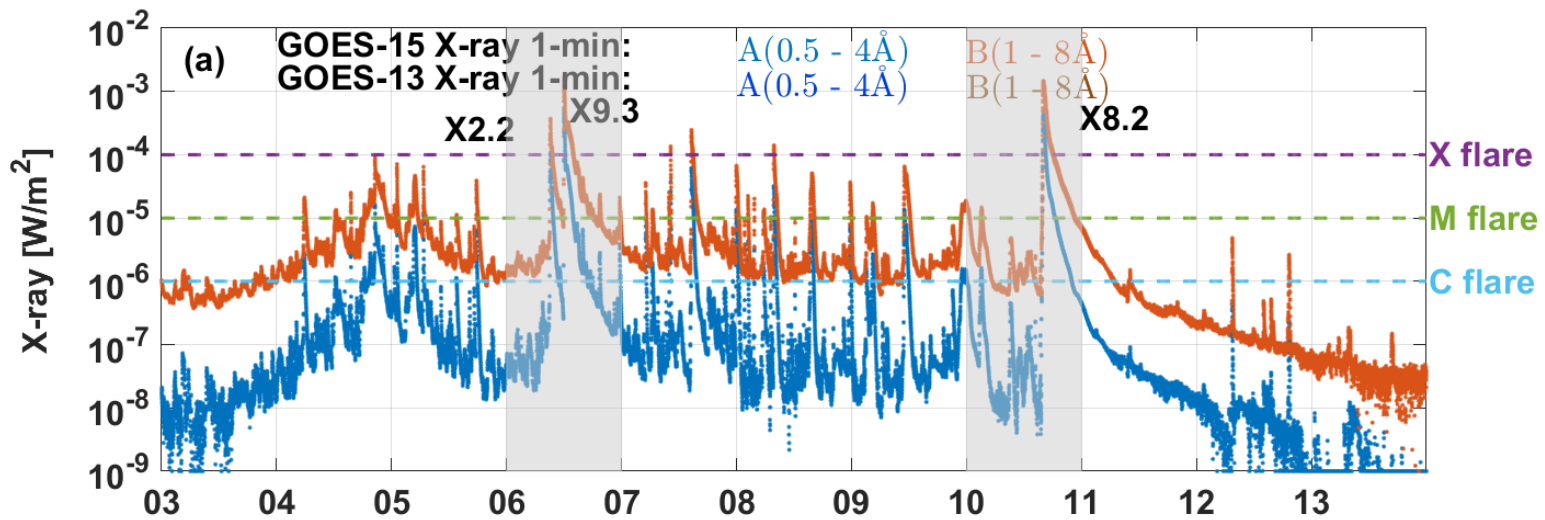
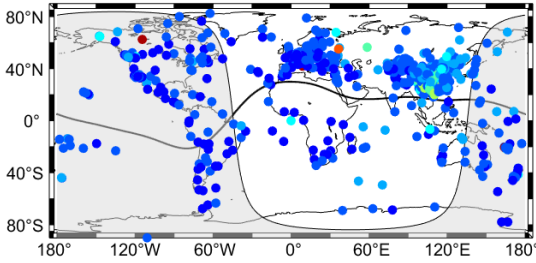


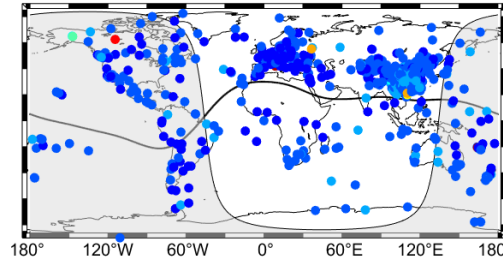
Figure 11.

(a) X2.2 PPP Error

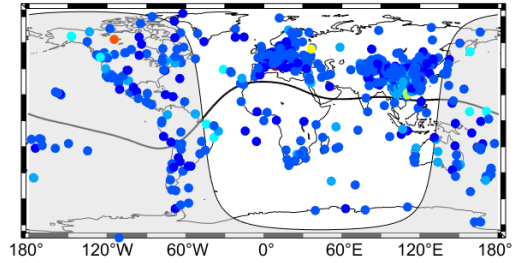
2017/09/06 08:55 UTC



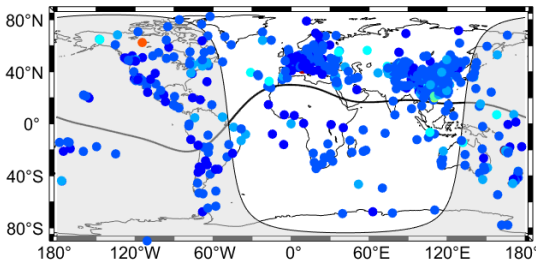
2017/09/06 09:00 UTC



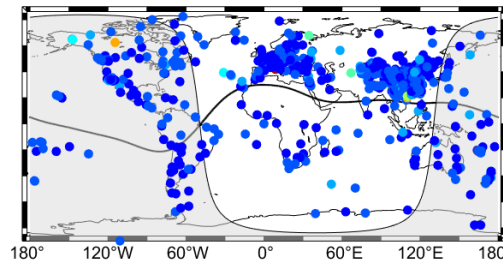
2017/09/06 09:05 UTC



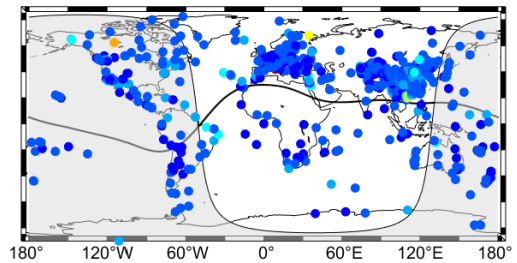
2017/09/06 09:10 UTC



2017/09/06 09:15 UTC

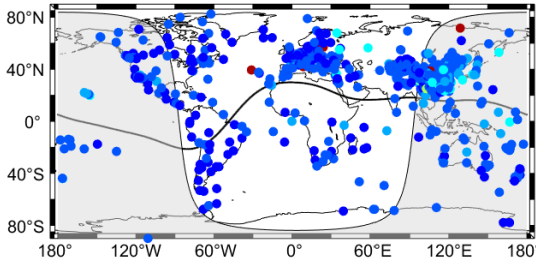


2017/09/06 09:20 UTC

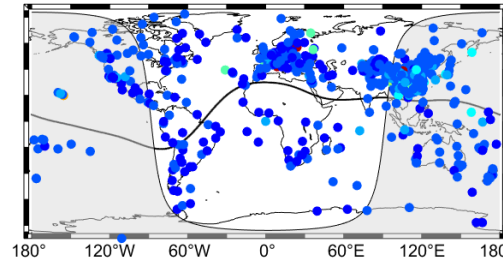


(b) X9.3 PPP Error

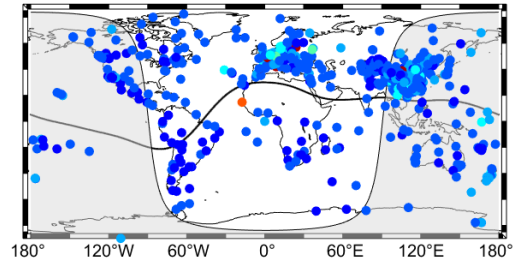
2017/09/06 11:45 UTC



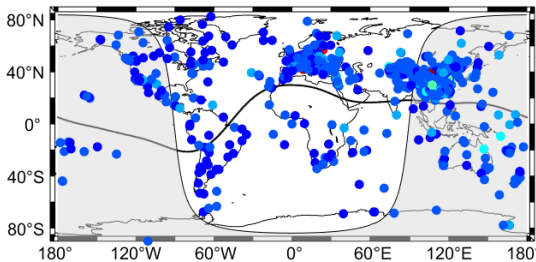
2017/09/06 11:50 UTC



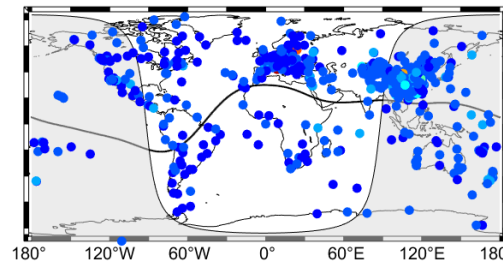
2017/09/06 11:55 UTC



2017/09/06 12:00 UTC



2017/09/06 12:05 UTC



2017/09/06 12:10 UTC

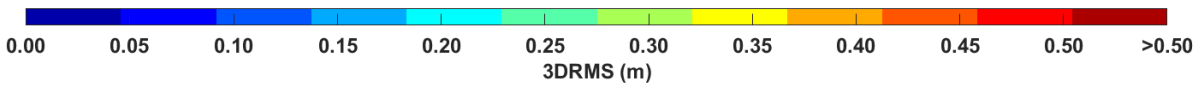
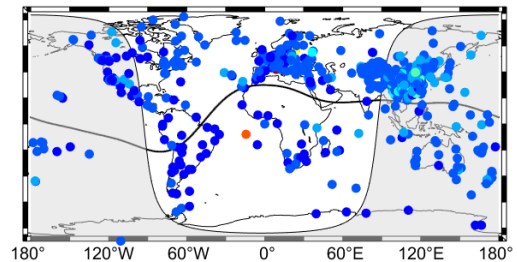
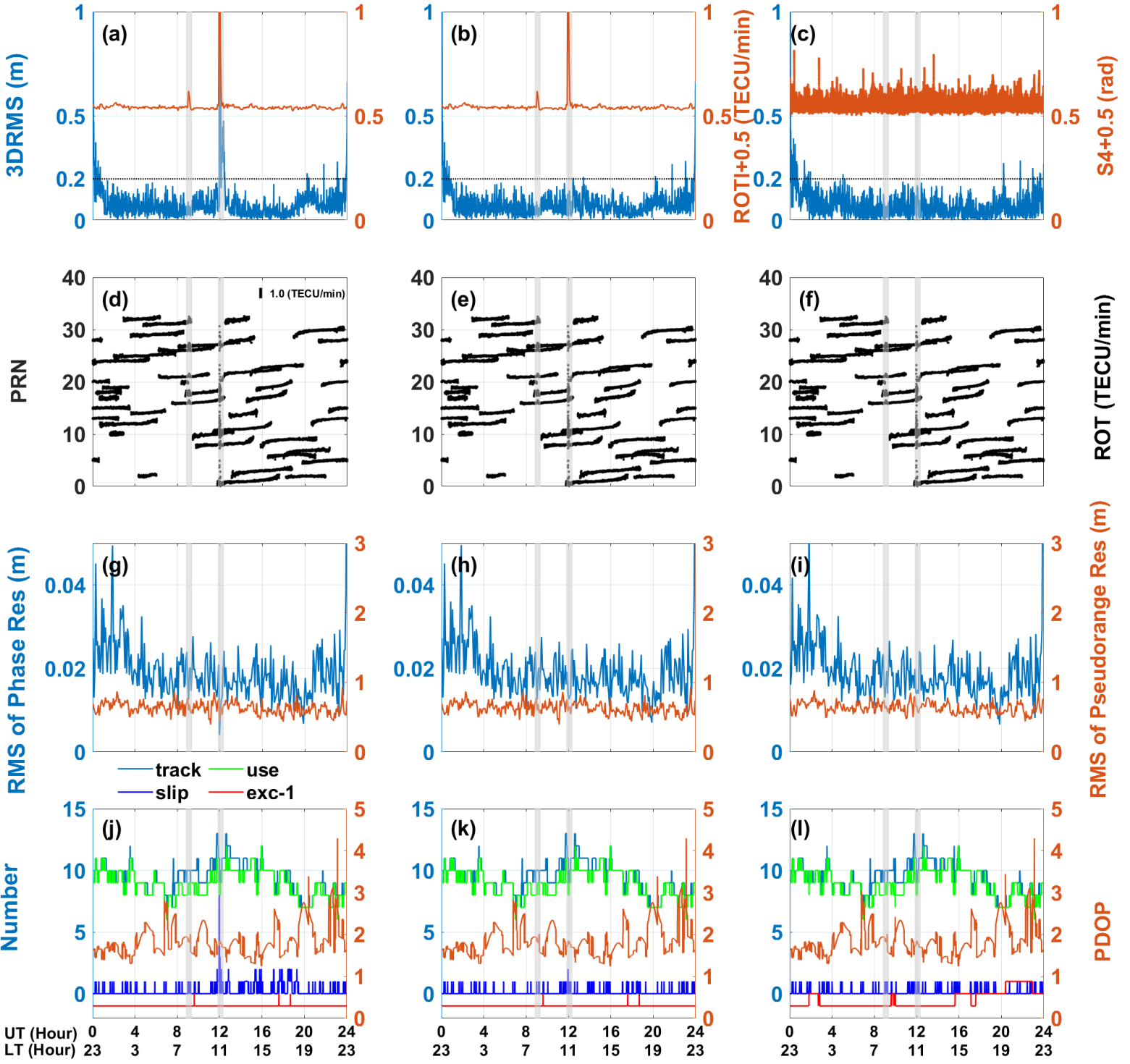


Figure 12.

VILL(40.4N,4.0W; 43.1 MLAT; 30s; PGF=0.05m)

VILL(30s; PGF=0.20m)

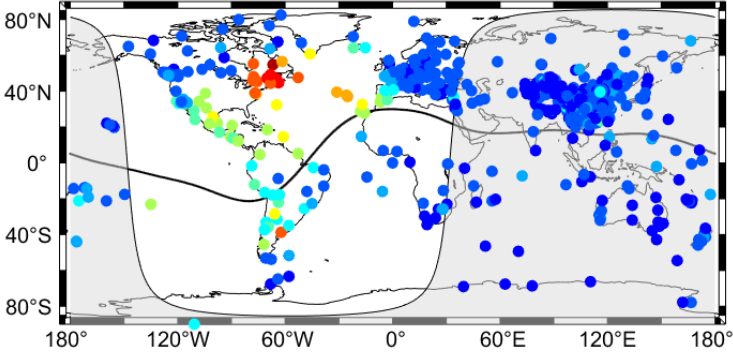
VILL(1s; PGF=0.05m)



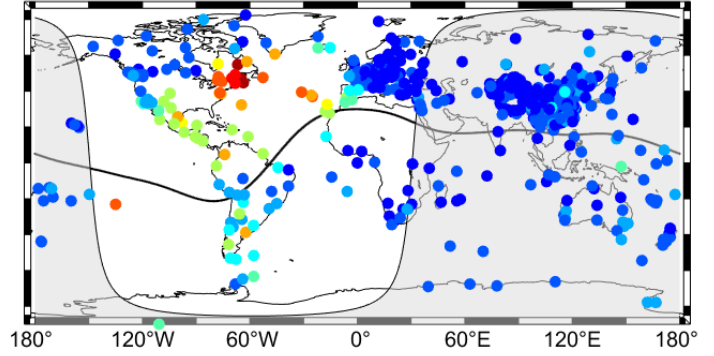
September 6, 2017

Figure 13.

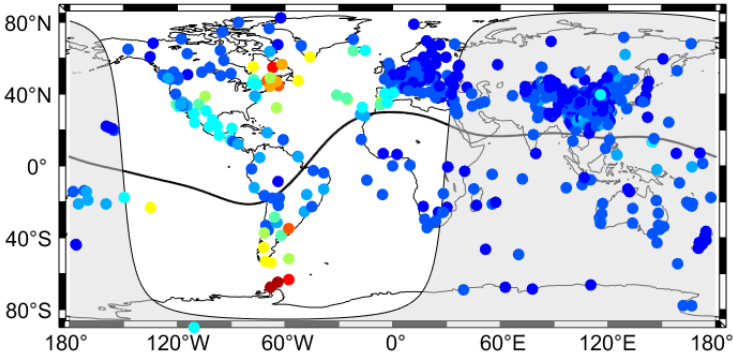
**2017/09/10 15:50 UTC**



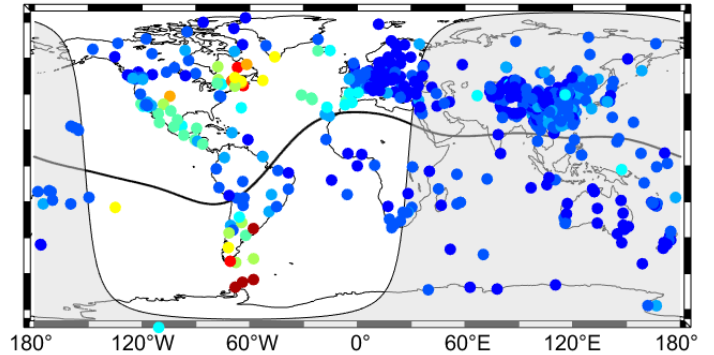
**2017/09/10 15:55 UTC**



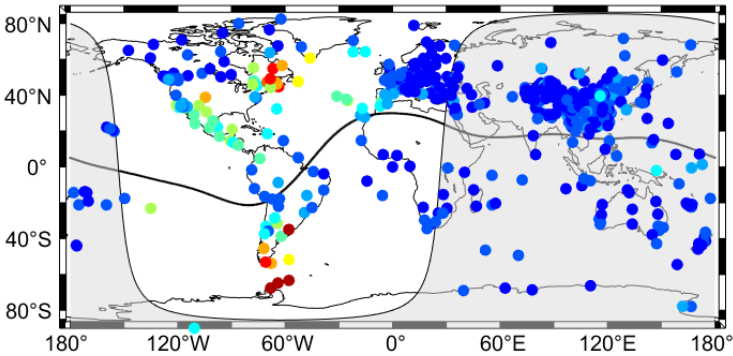
**2017/09/10 16:00 UTC**



**2017/09/10 16:05 UTC**



**2017/09/10 16:10 UTC**



**2017/09/10 16:15 UTC**

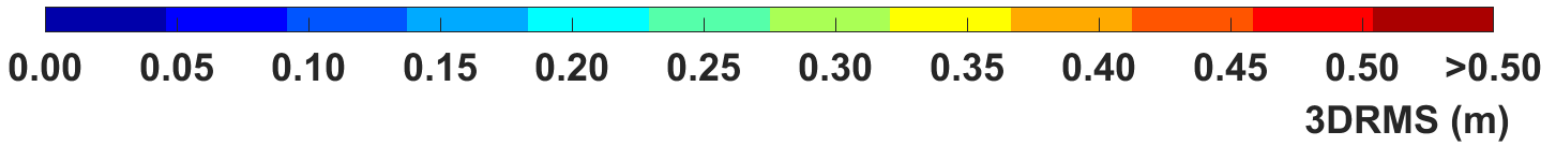
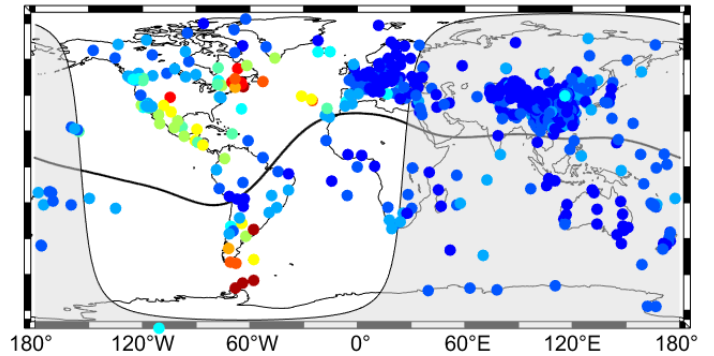
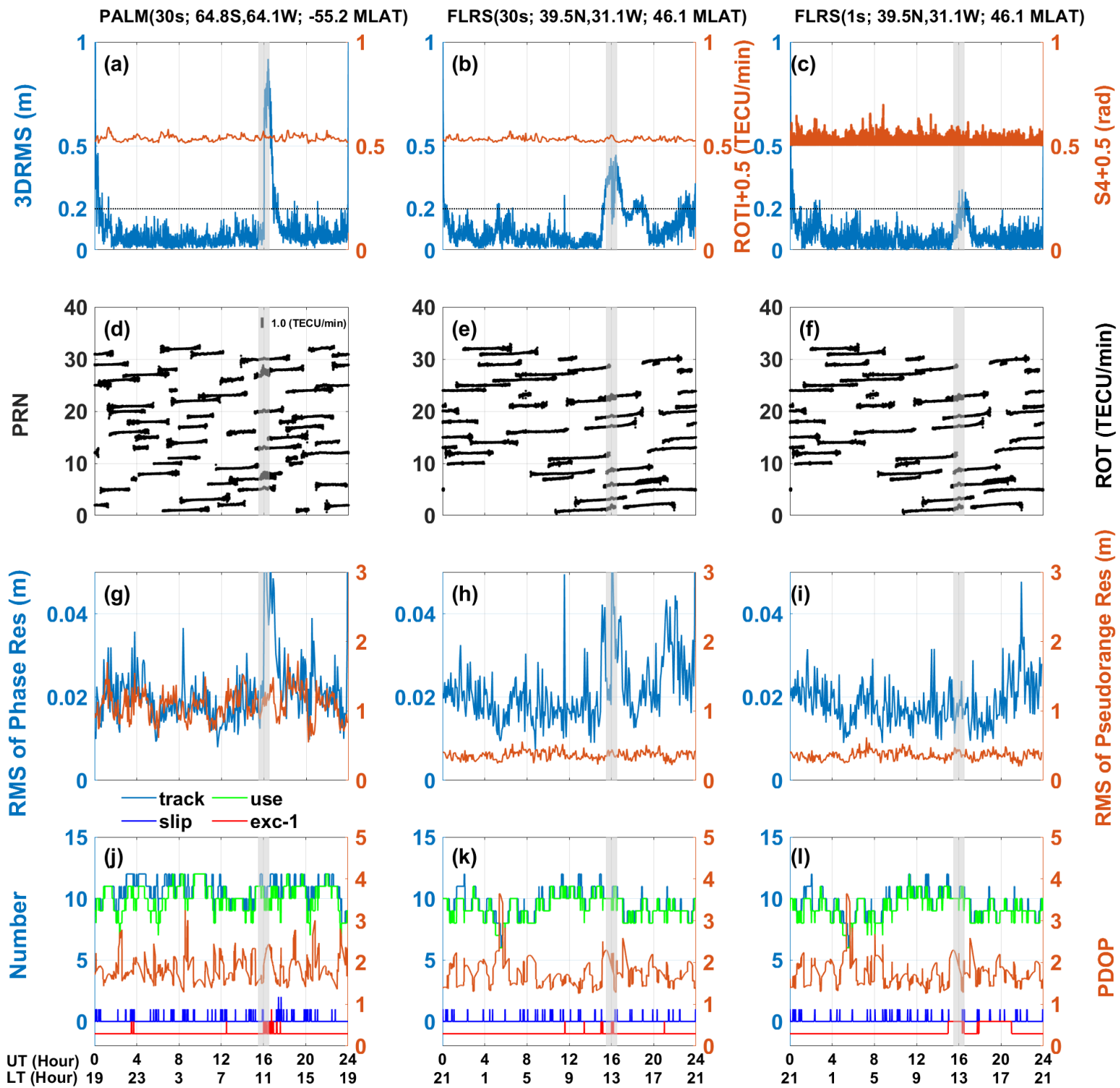




Figure 14.



September 10, 2017