

AOCS for future multi-satellite geodesy missions

Andreas Leipner¹, Stefanie Bremer¹, Meike List¹, Benny Rievers²

¹DLR Institute for Satellite Geodesy and Inertial Sensing

Relativistic Modelling

²ZARM University of Bremen



Knowledge for Tomorrow



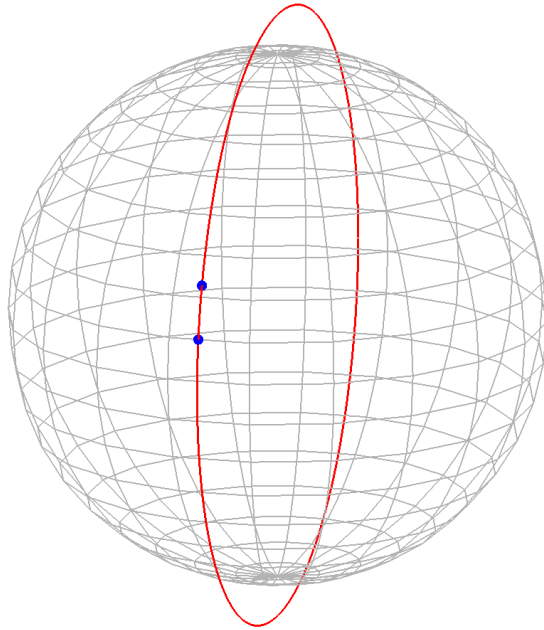
Outline

- Striping with GRACE data in gravity field recovery
- Pendulum orbits reduce striping
- Formation and Clusters for reduced AOCS requirements
- Simulating satellite formations
- Challenges with formations and gravity field recovery



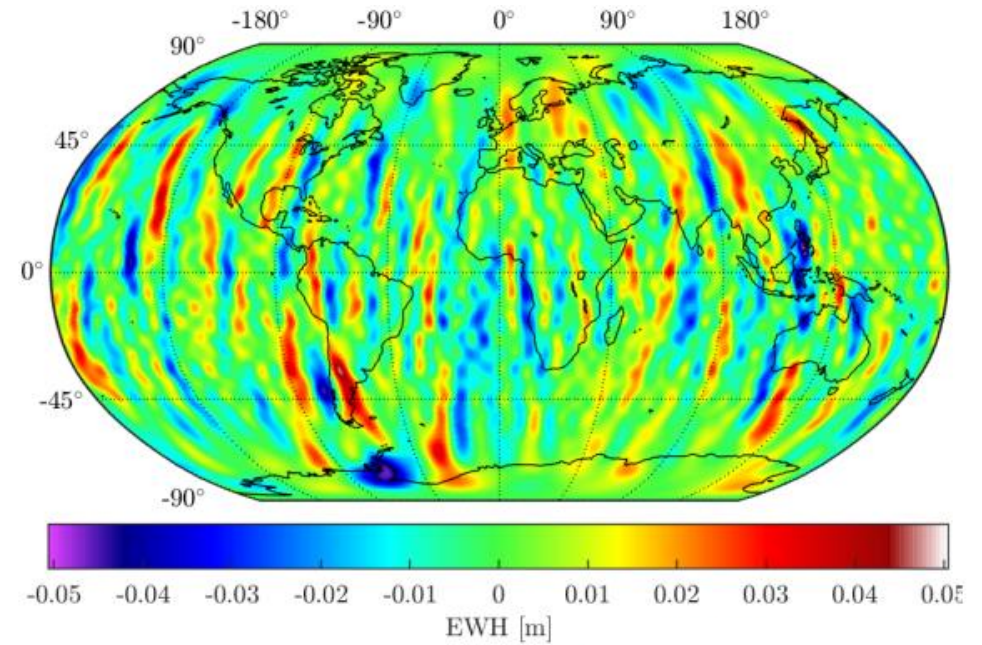
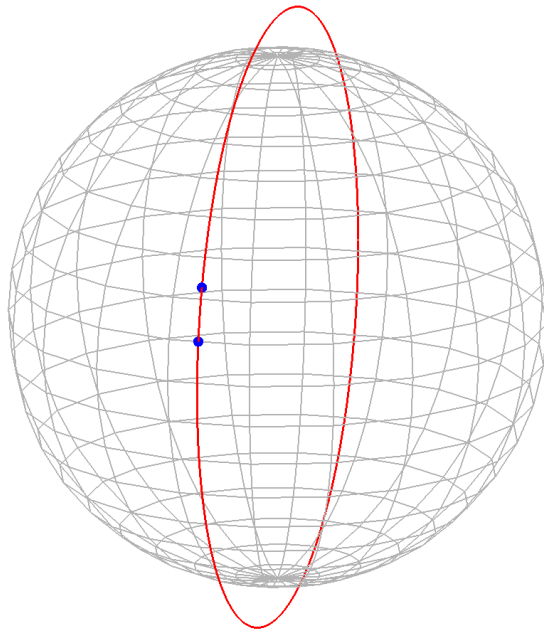
The Problem with GRACE like Missions

- GRACE like mission trajectory



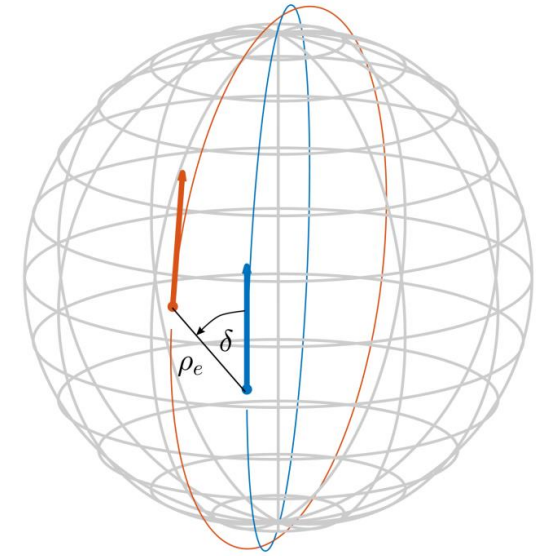
The Problem with GRACE like Missions

- GRACE measurement sensitivity is directional



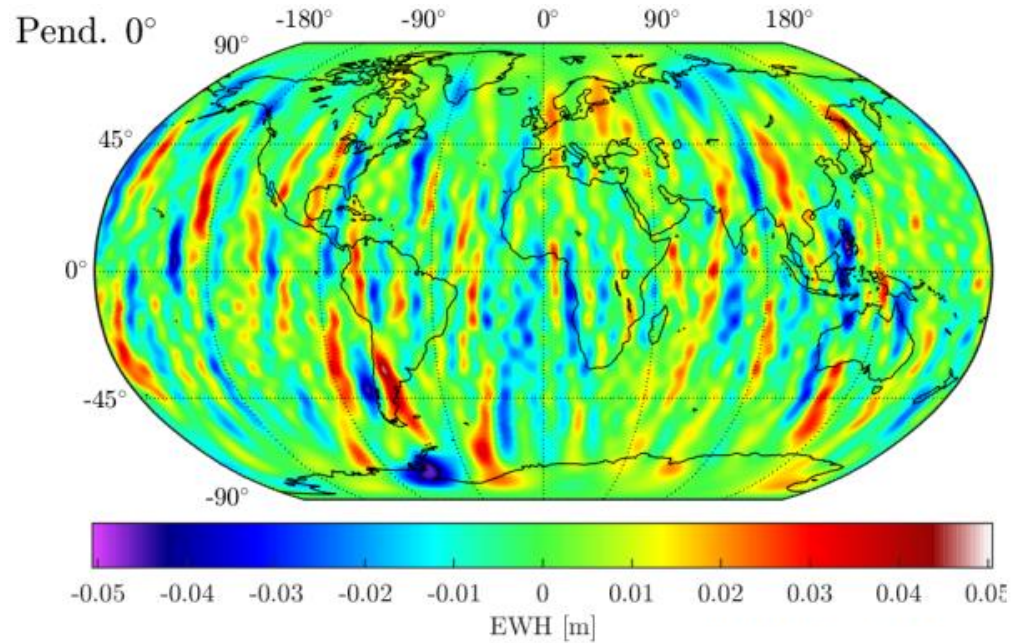
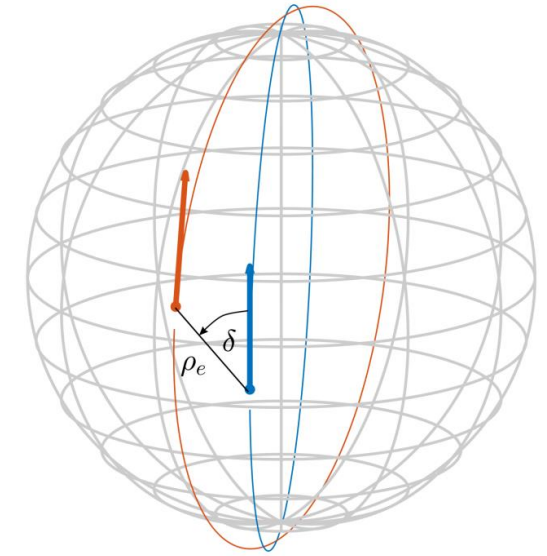
Pendulum Orbits - A Solution to the “GRACE Problem”

- Add additional measurement direction

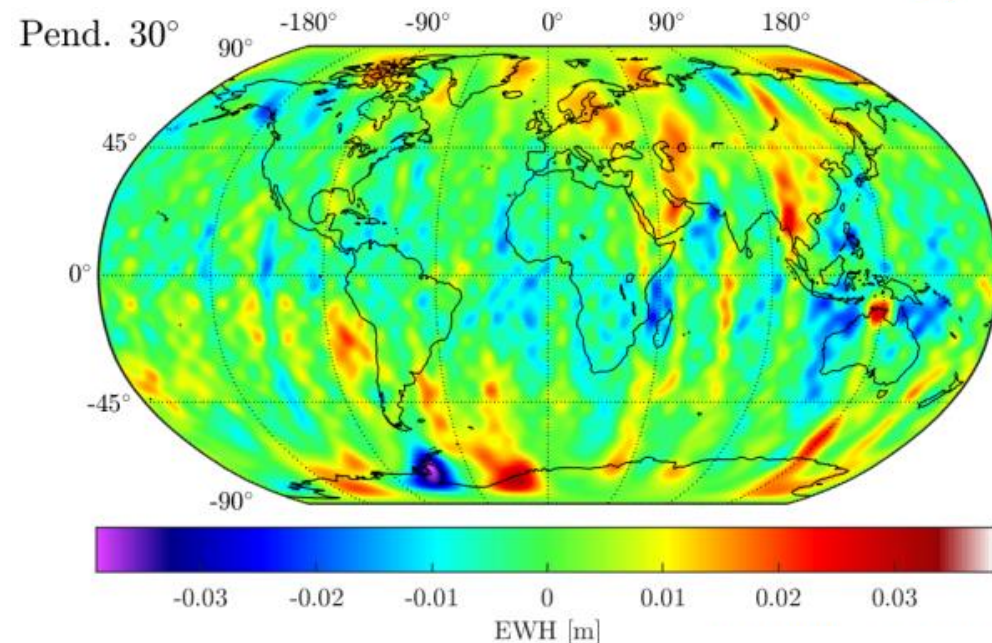


Pendulum Orbits - A Solution to the “GRACE Problem”

- Add additional measurement direction
 - Pendulum orbit would introduce another measurement direction



GRACE like

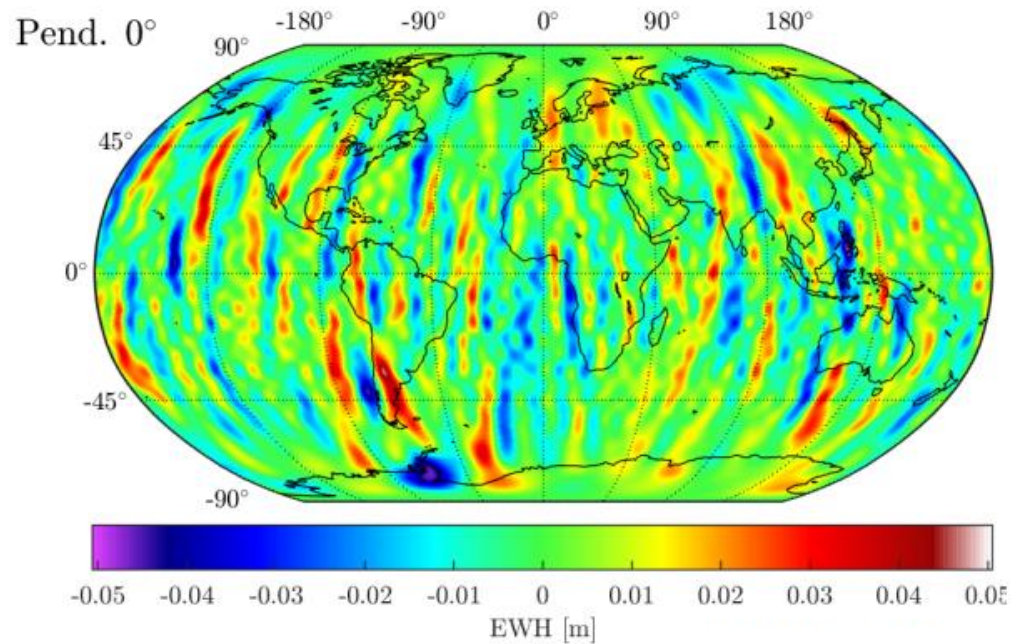
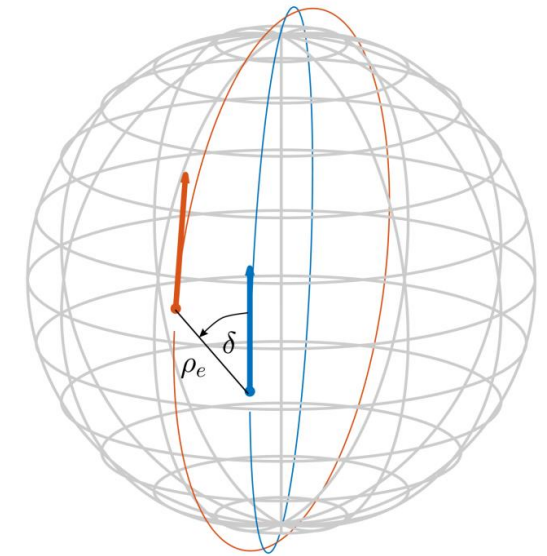


Pendulum Orbit

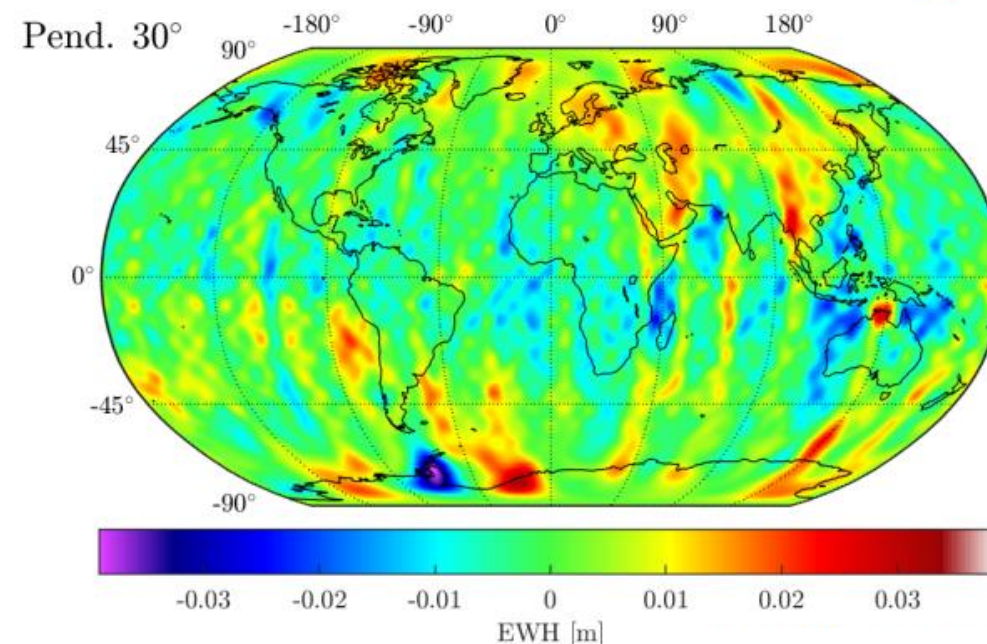


Pendulum Orbits - A Solution to the “GRACE Problem”

- Add additional measurement direction
 - Pendulum orbit would introduce another measurement direction
 - Pendulum orbits have high requirements on AOCS



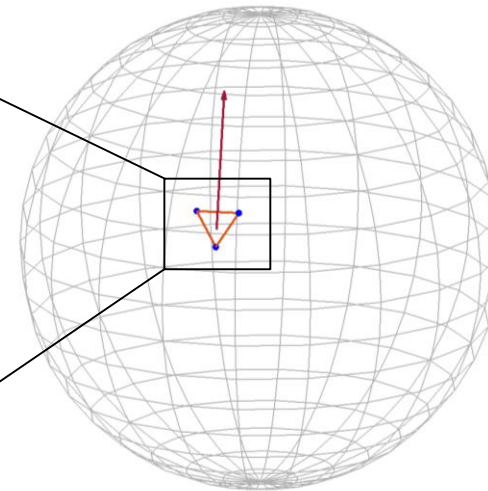
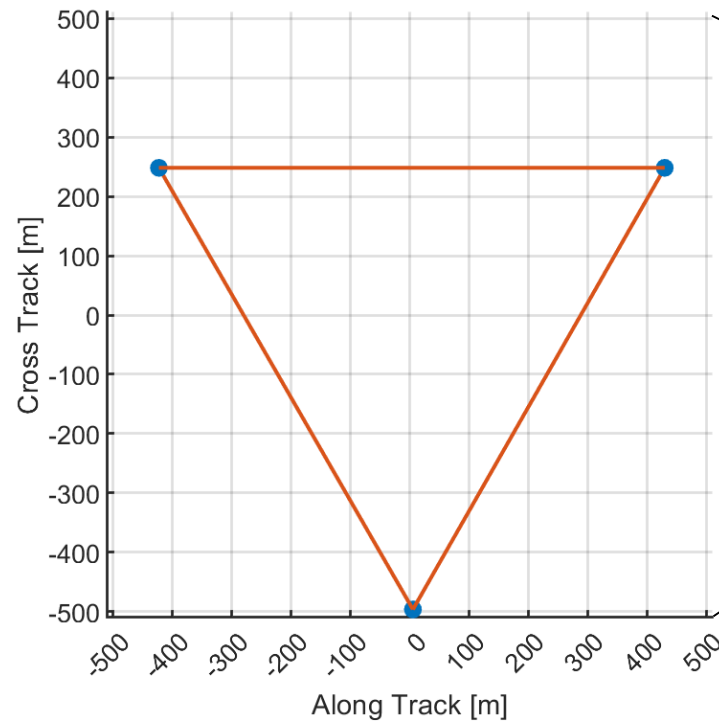
GRACE like



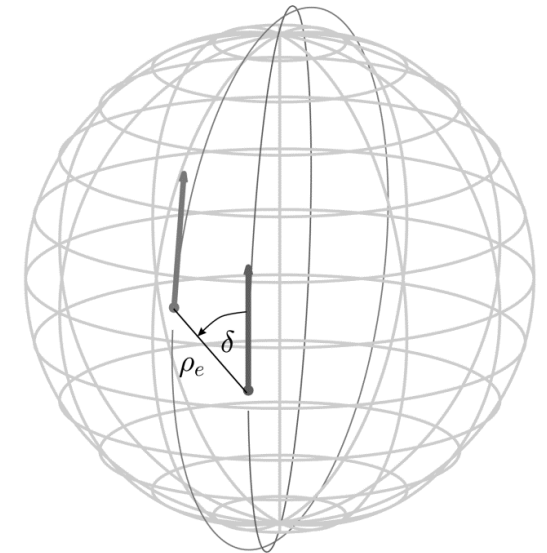
Pendulum Orbit

Cluster/Formation flying

- Add additional measurement direction with a triangle formation



AOCS Requirements ↓



AOCS Requirements ↑



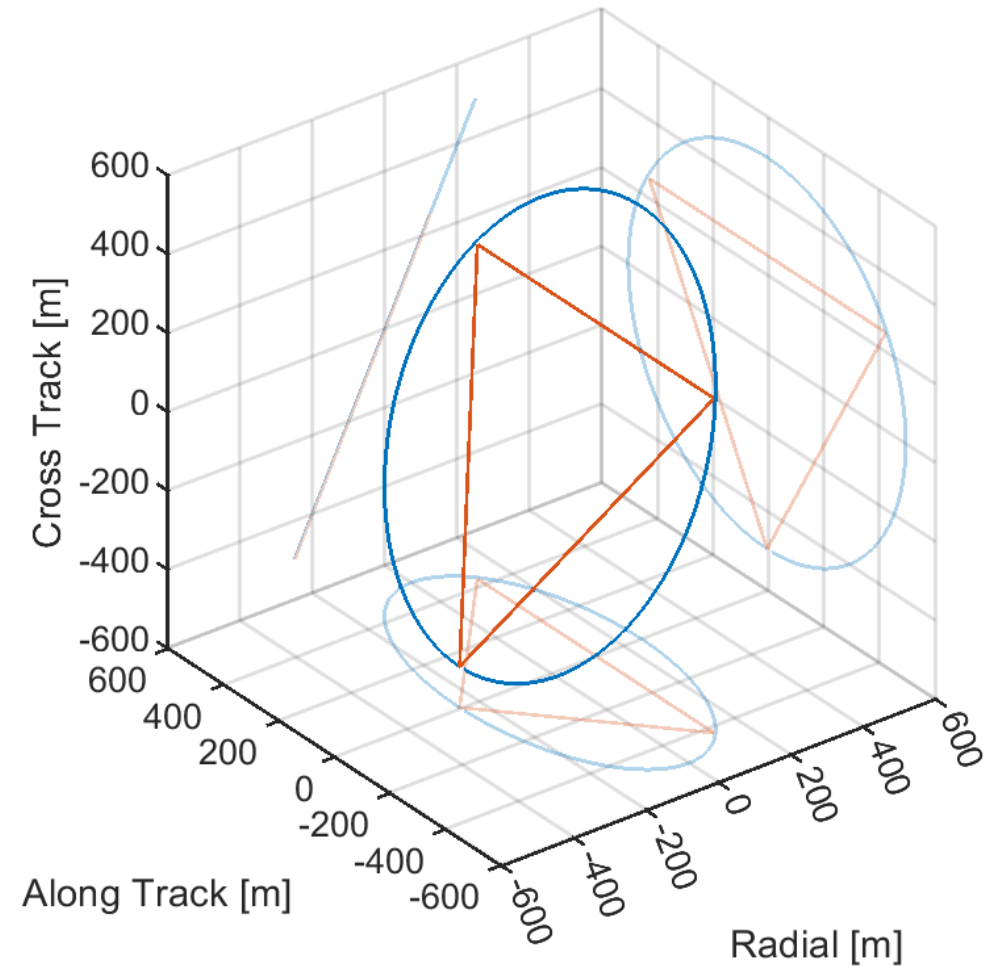
Constraints for Cluster/Formation *free flying*

Hill's equations¹:

$$x = \rho \sin(\omega t + \theta) + a$$

$$y = 2\rho \cos(\omega t + \theta) - \frac{3\omega}{2}at + b$$

$$z = m\rho \sin(\omega t + \theta) + 2n\rho \cos(\omega t + \theta)$$



¹ [Hsi-Han Yeh and A. Sparks, "Geometry and control of satellite formations," *Proceedings of the 2000 American Control Conference. ACC (IEEE Cat. No.00CH36334)*, 2000, pp. 384-388 vol.1, doi: 10.1109/ACC.2000.878926.]



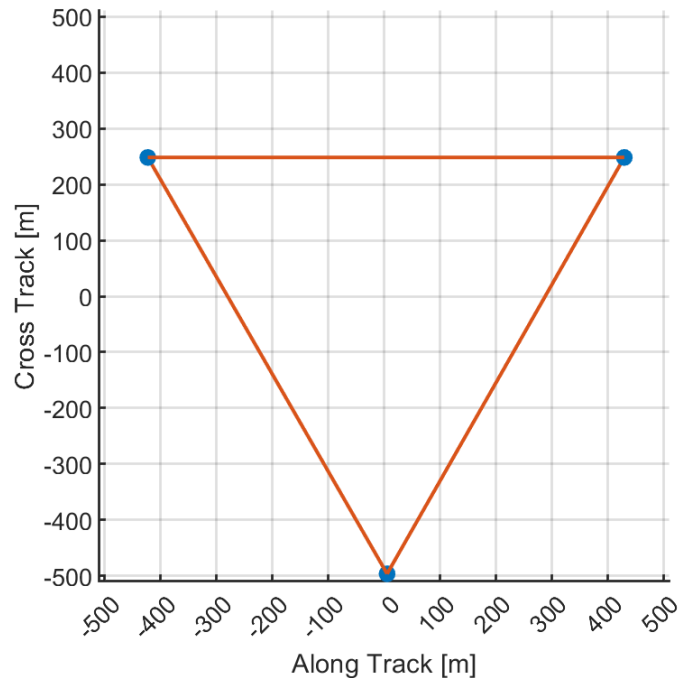
Constraints for Cluster/Formation *free flying*

Hill's equations¹:

$$x = \rho \sin(\omega t + \theta) + a$$

$$y = 2\rho \cos(\omega t + \theta) - \frac{3\omega}{2}at + b$$

$$z = m\rho \sin(\omega t + \theta) + 2n\rho \cos(\omega t + \theta)$$



¹[Hsi-Han Yeh and A. Sparks, "Geometry and control of satellite formations," *Proceedings of the 2000 American Control Conference. ACC (IEEE Cat. No.00CH36334)*, 2000, pp. 384-388 vol.1, doi: 10.1109/ACC.2000.878926.]



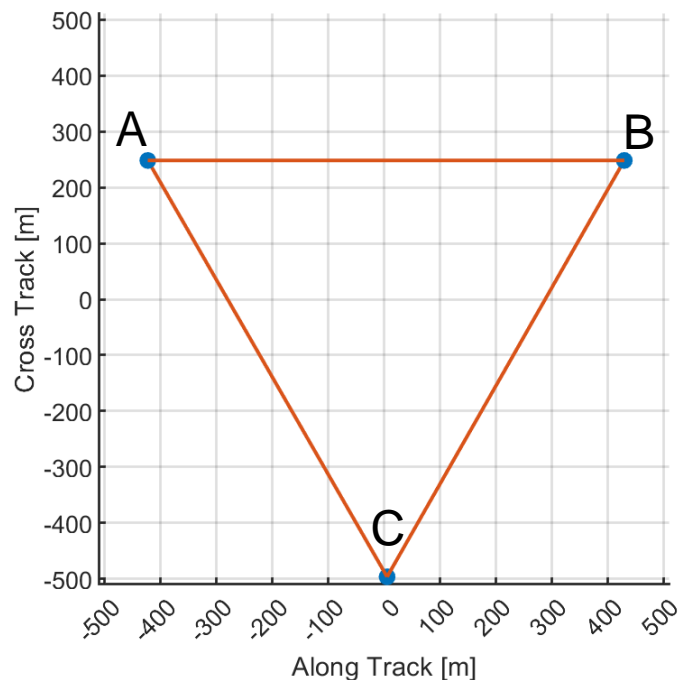
Constraints for Cluster/Formation *free flying*

Hill's equations¹:

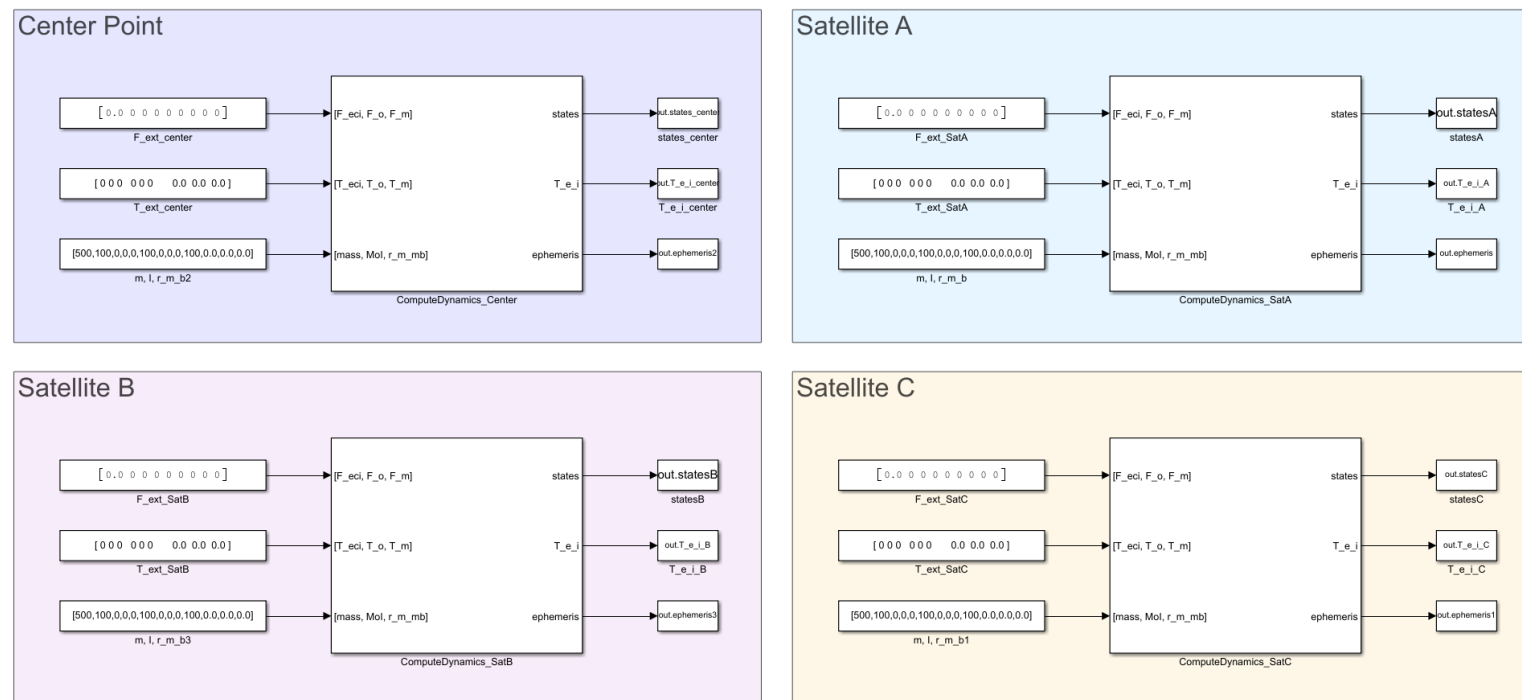
$$x = \rho \sin(\omega t + \theta) + a$$

$$y = 2\rho \cos(\omega t + \theta) - \frac{3\omega}{2}at + b$$

$$z = m\rho \sin(\omega t + \theta) + 2n\rho \cos(\omega t + \theta)$$



Simulink XHPS Simulation Setup



¹[Hsi-Han Yeh and A. Sparks, "Geometry and control of satellite formations," *Proceedings of the 2000 American Control Conference. ACC (IEEE Cat. No.00CH36334)*, 2000, pp. 384-388 vol.1, doi: 10.1109/ACC.2000.878926.]



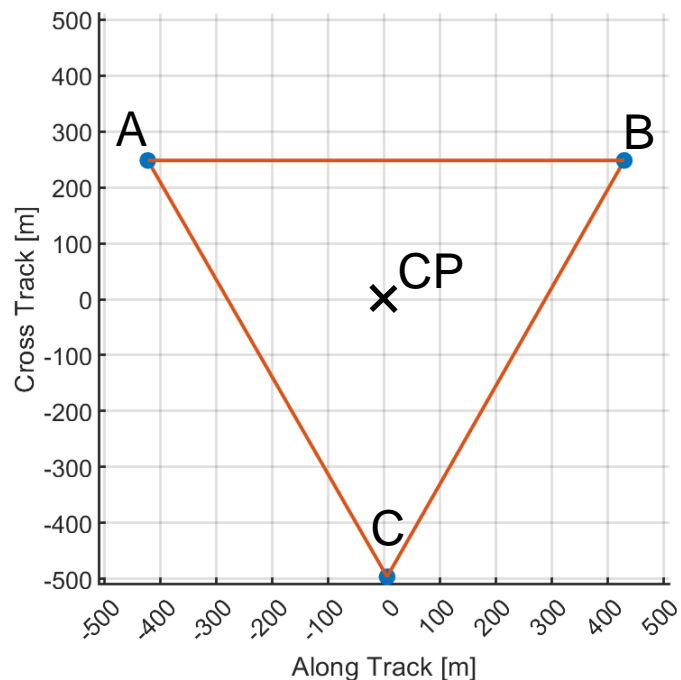
Constraints for Cluster/Formation *free flying*

Hill's equations¹:

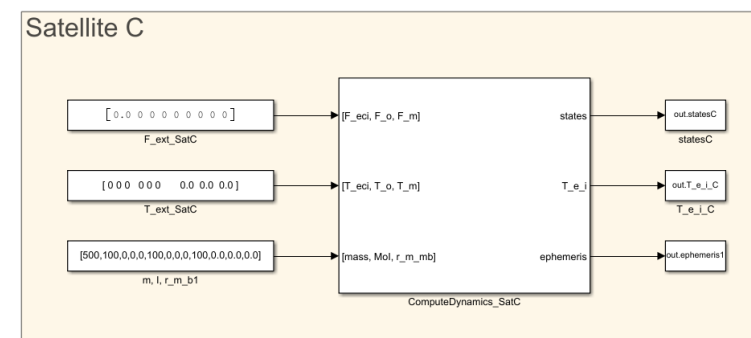
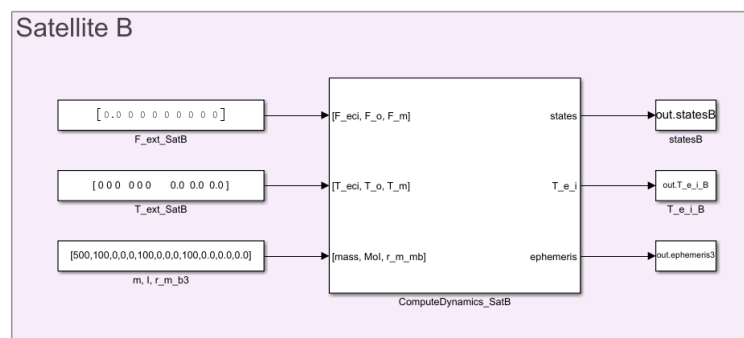
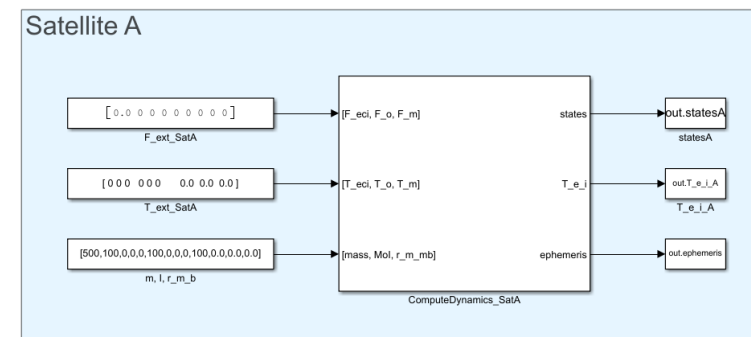
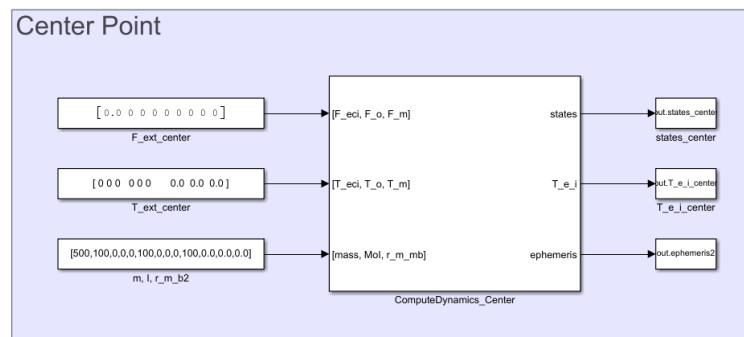
$$x = \rho \sin(\omega t + \theta) + a$$

$$y = 2\rho \cos(\omega t + \theta) - \frac{3\omega}{2}at + b$$

$$z = m\rho \sin(\omega t + \theta) + 2n\rho \cos(\omega t + \theta)$$



Simulink XHPS Simulation Setup

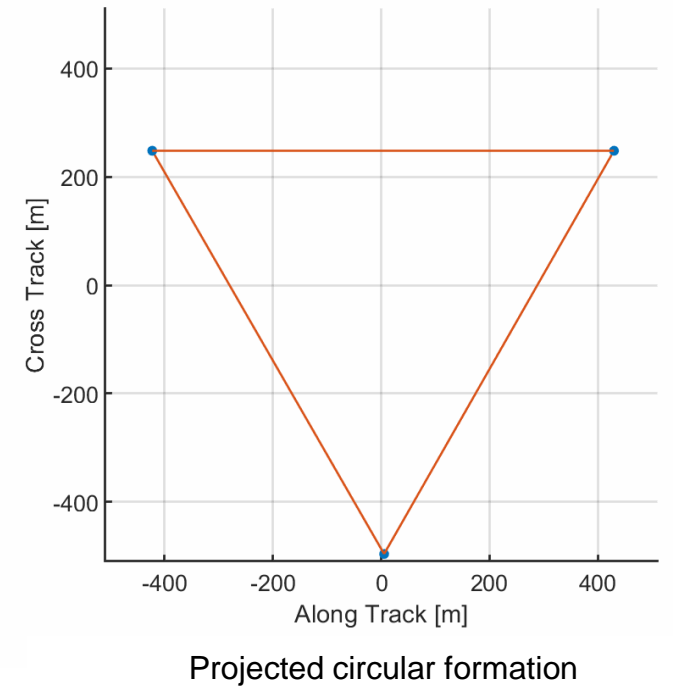
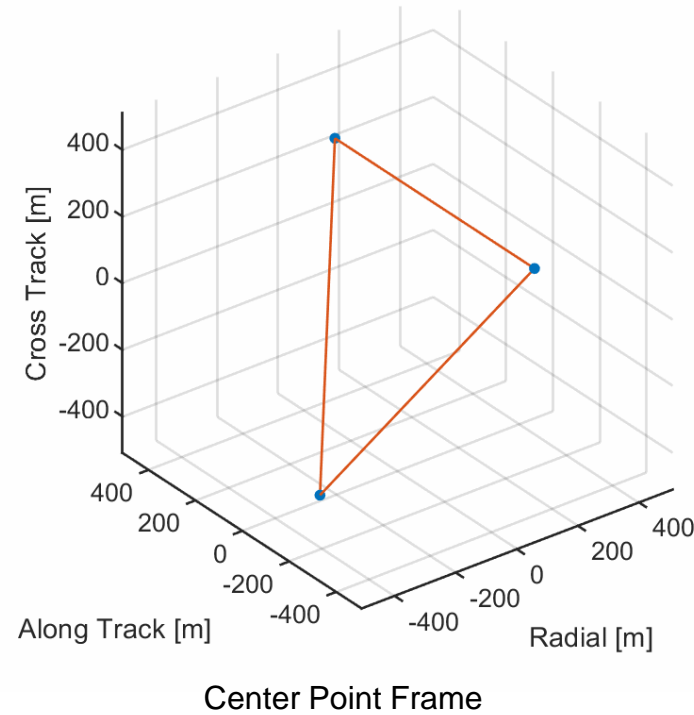
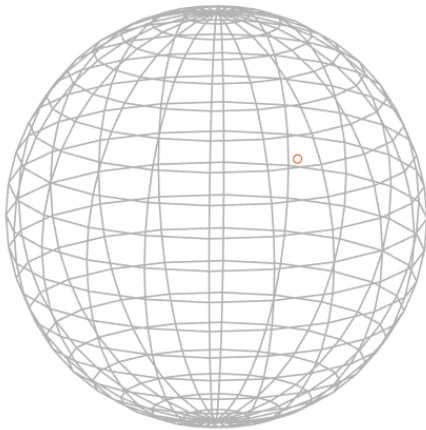


¹ [Hsi-Han Yeh and A. Sparks, "Geometry and control of satellite formations," *Proceedings of the 2000 American Control Conference. ACC (IEEE Cat. No.00CH36334)*, 2000, pp. 384-388 vol.1, doi: 10.1109/ACC.2000.878926.]



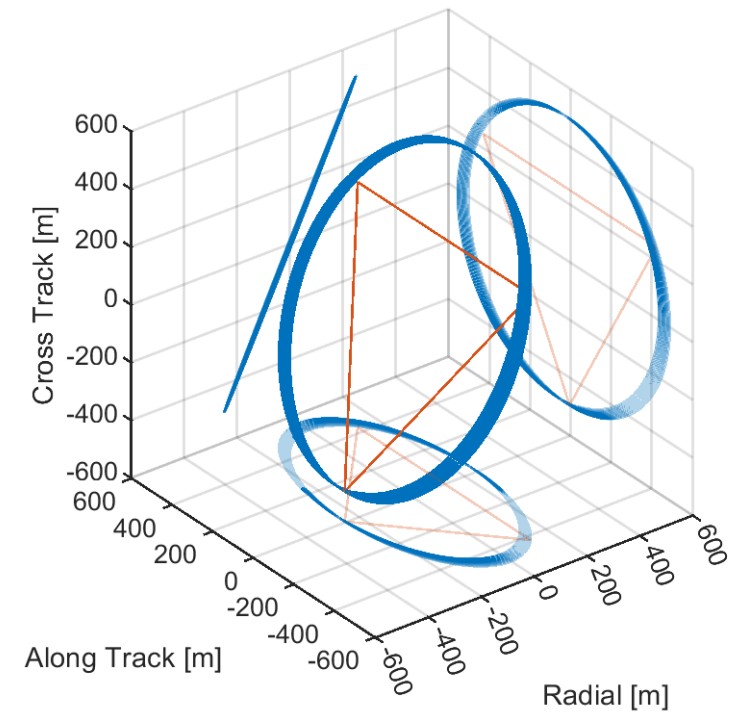
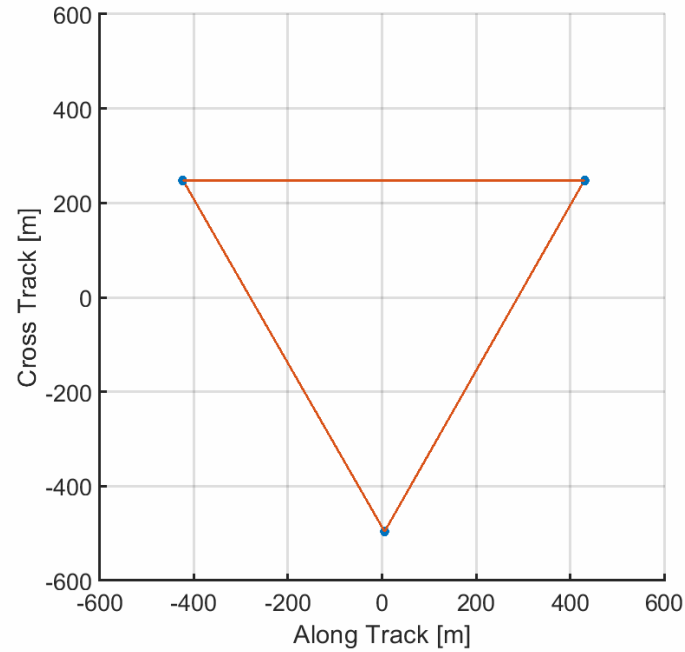
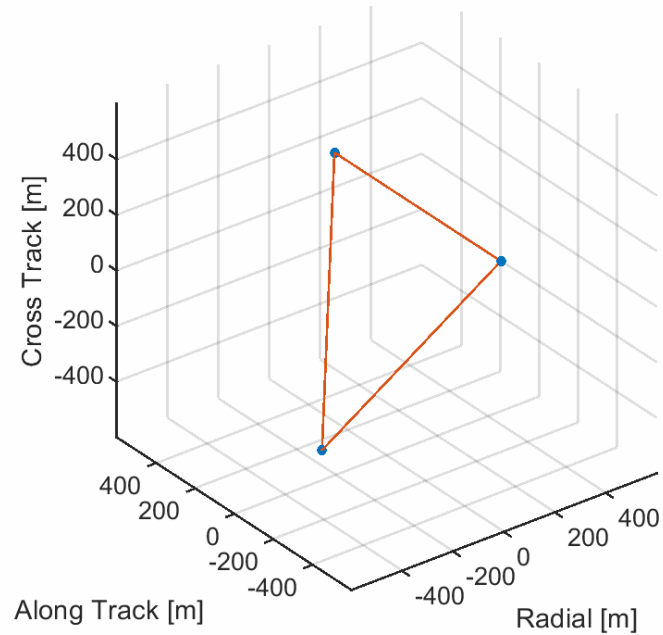
Undisturbed flying Clusters/Formations

- Free fly orbits for a triangular formation



Disturbed flying Clusters/Formations

- Formation will break up when disturbed by a non-spherical gravitational field



Possible AOCS Solutions for NGGM & Future Work

Formation maintenance

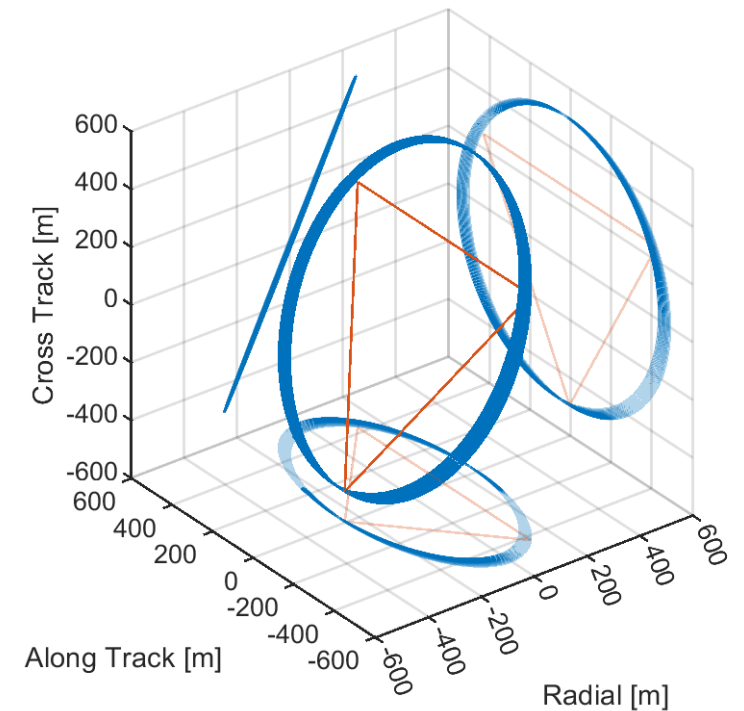
- Continuous error correction
- Error threshold correction

Energy consumption vs Measurement accuracy

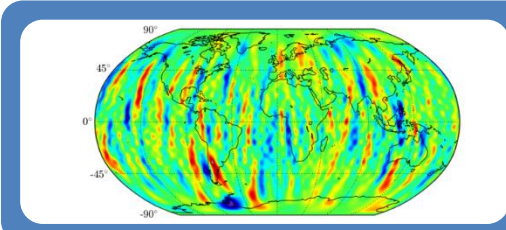
- How does the formation maintenance influence the measurement?

Challenges

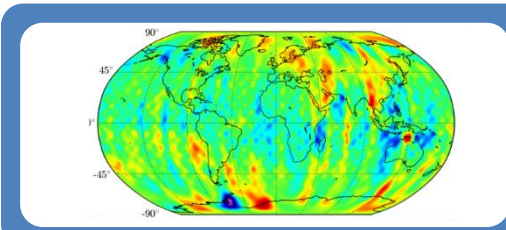
- Distance measurement between satellites
- Angles inside triangle formation
- Satellite form (Cube-Sats)



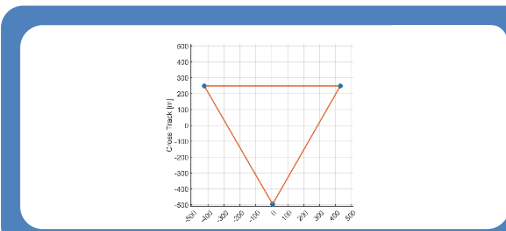
Summary



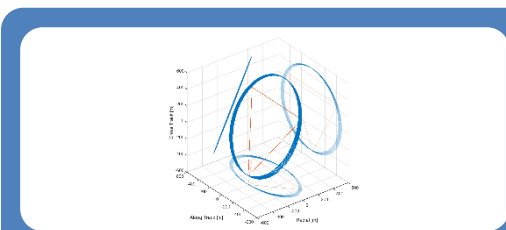
GRACE like missions sensitivity is directional, which yields to striping in the gravity field recovery



Introducing another measurement direction reduces the striping



Formations introduce multiple measuring directions and have low AOCS requirements



AOCS is required to avoid break up of formation



Thank you for your attention

This work was supported by DFG (CRC 1464 TerraQ).



Gefördert durch
DFG Deutsche
Forschungsgemeinschaft

