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# Article Effect of Stator Blades on the Startup Dynamics of a Vertical **Axis Wind Turbine**

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Abstract: Vertical Axis Wind Turbines (VAWTs) are omni-directional, low-cost, low-efficiency wind 11 power extractors. A conventional drag-based VAWT consists of multiple thin rotor blades with a 12 typical peak Tip Speed Ratio ( $\lambda$ ) of < 1. Their lower cut-in speed and maintenance cost make them 13 ideal for power generation in urban environments. Numerous studies have been carried out ana-14 lysing steady operation of VAWTs and quantifying their performance characteristics, however, 15 minimal attention has been paid to their start-up dynamics. There are a few recent studies in which 16 start-up dynamics of lift-based VAWTs have been analysed but such studies for drag-based VAWTs 17 are severely limited. In this study, start-up dynamics of a conventional multi-blade drag-based 18 VAWT have been numerically investigated using a time-dependant Computational Fluid Dynamics 19 (CFD) solver. In order to enhance the start-up characteristics of the drag-based VAWT, a stator has 20 been integrated in the design assembly. The numerical results obtained in this study indicate that 21 an appropriately designed stator can significantly enhance the start-up of a VAWT by directing the 22 flow towards the rotor blades, leading to higher rotational velocity ( $\omega$ ) and  $\lambda$ . With the addition of 23 a stator, the flow fields downstream the VAWT becomes more uniform. 24

Keywords: Vertical Axis Wind Turbine (VAWT); Computational Fluid Dynamics (CFD); Start-Up 25 Dynamics; Dynamic Meshing; Tip Speed Ratio 26

#### 1. Introduction

In order to reduce dependency on fossil fuels and to tackle climate change, the use 29 of renewable energy sources, especially wind energy, has become very popular in the past 30 few decades. Wind energy is harnessed using wind turbines by capturing the kinetic en-31 ergy from the wind. This is achieved through wind induced rotation to the wind turbine 32 blades, and the generator shaft, to produce electricity. Depending on the axis of rotation, 33 wind turbines can be classified into Vertical Axis Wind Turbines (VAWTs) and Horizon-34 tal Axis Wind Turbines (HAWTs) [1,2]. HAWTs are a popular choice for harnessing wind 35 energy in mid to large scale wind farms [3], which are built on vast, open locations where 36 the turbines can get sustained wind velocities above turbines' cut-in speed. They also need 37 a yaw control mechanism so that the turbine is always facing the direction of the wind for 38 maximum power extraction [4]. Therefore, they have a high utilisation coefficient. In ur-39 ban locations however, the wind speed and direction is highly non-uniform, making 40VAWTs advantageous in these locations, as they have a much lower cut-in speed and are 41 omni-directional, thus, not requiring a yaw control mechanism [5]. Apart from the sim-42 plified design of VAWTs, research studies [6-8] have shown that they can be installed 43 much closer to each other since they can also extract power from the wake generated by 44 the turbines upstream. 45

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#### 1.1 Rationale to carry out this study

Irrespective of their cut-in speeds, the start-up speed of wind turbines, both HAWTs 47 and VAWTs, is a function of turbine size, shape and wind speed. VAWTs are generally 48 considered to have lower start-up speeds, not particularly because of their shape, rather 49 their smaller size compared to HAWTs. Thus, for bigger VAWTs with potentially higher 50 power generation, their start-up speeds become an important contributing factor towards 51 their commercial viability. It is a well-known fact that VAWT's peak Tip Speed Ratio ( $\lambda$ ) 52 is much smaller than HAWTs [9], thus, for same size and wind speed, VAWTs' rotational 53 velocity ( $\omega$ ) is lower than HAWTs'. This is probably the primary limiting factor VAWTs' 54 industry is currently experiencing in terms of upscaling. In order to address these indus-55 trial challenges, i) the start-up dynamics of VAWTs must be fully understood and ana-56 lysed, and ii) technologies must be developed to decrease VAWTs' start-up speeds (or 57 increase their  $\omega$  at the same start-up speeds). We aim to address both these challenges in 58 the present study however, we must first review the published literature that deal with 59 the start-up dynamics of wind turbines, focusing on VAWTs. 60

#### 1.2 Rationale for choosing Drag-based VAWTs

There are two main types of VAWTs i.e. lift-based and drag-based, commonly known 63 as Darrieus and Savonius VAWTs [10]. Most of the published literature deals with lift-64 based VAWTs, however, the present study analyses drag-based VAWTs. The primary 65 reason for choosing drag-based VAWTs over more commonly studied lift-based VAWTs 66 is the fact that the manufacturing and disposal of lift-based VAWTs' rotor blades is a sig-67 nificant challenge. These are the same challenges faced by the HAWT industry. Thus, the 68 rotor blades of lift-based VAWTs are quite expensive (because of their NACA profile) and 69 there are very limited number of blade manufacturers (e.g. Siemens, GE, Bureau Veritas) 70 that can deliver mid to large scale blades. Moreover, currently these blades are disposed 71 in landfills, causing potentially considerable environmental impact. With 40,000 non-re-72 cyclable composite blades going to be disposed of in the next 10 years in Europe alone, 73 the authors believe a more viable option for technology development is drag-based 74 VAWTs, where the blades are made out of inexpensive recyclable materials (e.g. steel, 75 aluminium) and their shape is much simpler (cup shaped) compared to lift-based VAWTs. 76 It is however noteworthy that most of the published literature deals with the start-up dy-77 namics of lift-based VAWTs, a review of which is presented below. 78

# 1.3 Start-up Dynamics of Lift-based VAWTs

Hill et al [11], to the authors' best knowledge, were the first to report the start-up 81 dynamics of VAWTs. They carried out experimental investigations on a H-rotor lift-based 82 VAWT model in an open jet configuration wind tunnel. The results obtained show that 83 there are four stages of start-up process. The first stage is the linear acceleration where the 84 turbine rotor accelerates from rest to a  $\lambda$  value of just above 1. The second stage is called 85 the first plateau where the turbine almost stops accelerating and operates at roughly a 86 constant speed ( $\lambda \sim 1.2$ ). The third stage of start-up is the rapid acceleration to  $\lambda \sim 3$ . The 87 last stage of the start-up is the second plateau, or steady operation of the VAWT at  $\lambda \sim 3$ . 88 The plateaus have been referred to as 'dead band' by some researchers [12-13], during 89 which the turbine cannot accelerate because of negative torque [14]. However, Hill et al's 90 [11] results show that if  $\lambda$  can increase gradually to ~ 1.5, then the VAWT can accelerate 91 rapidly to its maximum steady operation. Asr et al [15] conducted numerical investiga-92 tions to study the start-up characteristics of a H-rotor lift-based VAWT, having 93 NACA0018 airfoils for the blades, using flow induced rotation approach. The numerically 94 predicted results have been validated against the published experimental data of Rainbird 95 [16] and Hill et al [11]. Asr et al [15] have also noticed four stages in the start-up of the 96 turbine, however, carrying further investigations, they found out that the turbines with 97 NACA0015 and NACA0022 airfoils were not able to accelerate past the dead band region. 98

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They also investigated the effect of changing the pitch angle and found that inward pitch 99 angles worsened the performance of NACA0018 airfoils, but applying small outward 100 pitch angles  $(1.5^{\circ} - 3^{\circ})$  improved the start-up performance, resulting in a decrease of 23% 101 - 32% in start-up time. Using higher outward pitch angles eliminated the plateau region 102 but increased the time the turbine took to reach its peak  $\omega$ . 103

Zhu et al [17-18] and Sun et al [19] carried out a number of numerical investigations 104 on the start-up dynamics of H-rotor lift-based VAWT. It has been reported that fluctua-105 tions in the wind velocity influences the start-up of VAWTs [17]. The fluctuation fre-106 quency is found to have negligible effect of the start-up, while large fluctuation ampli-107 tudes can degrade the start-up performance of VAWTs. Zhu et al [18] further investigated 108 the effects of rotational friction ratio on the power extraction capability of the VAWT. It 109 has been found that smaller rotational friction ratio can benefit the power extraction at 110 lower wind speeds, while larger rotational friction ratio is beneficial at higher wind 111 speeds. Sun et al [19] then investigated the effects of number of blades (3 and 5) and blade 112 pitch angles (-4° to +4°) on the start-up of VAWTs. It has been found out that both these 113 parameters have negligible effects on the start-up of the VAWTs at lower wind speeds, 114 while at higher wind speeds, a greater number of blades and larger blade pitch angles aid 115 the start-up process. Zhu et al [20] also agree with the findings of Sun et al [19], with the 116 addition that the blade solidity also plays an important part in the start-up of the VAWTs. 117 It has been reported that with lower solidity, more power extraction is possible however, 118 the start-up is delayed, and vice versa. 119

Some researchers have reported that VAWTs face difficulty in self-starting without 120 any external assistance [1,21]. Solidity of a VAWT is an important factor that strongly af-121 fects its self-starting ability, optimal  $\lambda$  and maximum Power Coefficient (C<sub>P</sub>) [22]. Hill et 122 al's [11] study shows that the turbine with solidity of 0.33 can self-start under a wind speed 123 of 6 m/s. Dominy et al [23] have found that under lightly loaded conditions, the ability of 124 two-bladed lift-based turbines to self-start depends on the angular orientation of the 125 blades but a three-bladed turbine can start from any orientation. It can thus be concluded 126 that the start-up of lift-based VAWTs are dependent on the initial orientation of the rotor 127 blades. Moreover, Tigabu et al [24] numerically analysed the effects of turbine inertia on 128 the start-up characteristics of a H-rotor hydrokinetic turbine. They have noticed that if the 129 turbine starts from rest with no load condition, it overshoots steady  $\lambda$  point before coming 130 back to the steady operation of the VAWT i.e. steady  $\lambda$ . Goude and Lundin [25] found that 131 as the turbine inertia reduces, the start-up time decreases, however, at higher inertia, the 132 overshoot decreases and start-up goes through the four stage start-up process described 133 by Hill et al [11]. 134

#### 1.4 Start-up Dynamics of Drag-based Turbines

Zhao et al [26] investigated the start-up process of a two-bladed drag-based hydro-137 kinetic turbine using experimentation and numerical investigations, considering the ef-138 fects of initial azimuthal angles. It has been found that the static torque coefficient changes 139 significantly with the azimuthal angle and therefore, the starting position of the rotor is 140important for successful start-up of the turbine. Apart from the dead band azimuthal po-141 sitions where the turbine experiences a negative torque, it has been found that some posi-142 tions, where the static torque on the rotor is positive, the rotor could not start successfully. 143 The start-up process of drag-based turbines has been divided into two stages i.e. an initial 144 acceleration and the transition stages. At the end of the transition stage, the angular ve-145 locity fluctuates periodically over a small range representing a stable operation of the ro-146 tor. Kang et al [27] have carried out further investigations with the aim to improve the 147 start-up of the hydrokinetic turbine considered by Zhao et al [28]. They developed a two-148 stage hydrokinetic turbine (top and bottom configuration) and noticed a similar pattern 149 of start-up to single stage turbine. It has been reported that the angular speed of the rotor 150 accelerated smoothly to the stable state and the fluctuations in the stable state reduced 151

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considerably. The start-up time for the two-stage rotor has been shown to be lower than 152 the single stage rotor. Mu et al [29] have carried out experimental and numerical investi-153 gations on the start-up and steady performance of two-bladed drag-based VAWTs, with 154 conventional (straight) and spiral blades. It has been reported that for certain azimuthal 155 angles, the conventional drag-based VAWT could not start, but the spiral blade VAWT 156 did, at all azimuthal angles. Like Zhao et al [26] and Kang et al [27], rapid initial accelera-157 tion and then transition to steady operation of the VAWT have been reported, with peak 158  $\omega$  increasing as wind speed increases. It has also been recorded that the spiral blade 159 VAWT generated significantly higher torque. After conducting a review of the published 160 literature regarding the start-up dynamics of VAWTs, the review of research studies on 161 the performance characterisation of VAWTs, with and without stator blades, is presented 162 below. 163

# 1.5 Performance Characterisation of VAWTs with and without Stator

Altan and Atilgan [30] placed a curtain arrangement as a simple wind deflector in 166 front of a drag-based rotor. It has been found that at the curtain's optimum angle, CP in-167 creases by 38 %, however, the use of the curtain reduced the operational range of the rotor. 168 Golecha et al [31] showed that a properly placed deflector plate can increase VATW's CP 169 by 50 % compared to a rotor with no deflector. The reason for the improvement in perfor-170 mance of the VAWT due to curtain is that shielding the returning blade with a deflector 171 plate reduces the flow resistance experienced by the blade, which reduces negative torque. 172 The same principle was utilised by Alizadeh et al [32] to increase the performance of a 173 VAWT where they shielded the returning blade of the turbine rotor by a quadrant barrier. 174Pope et al [5] investigated the effects of vanes on CP of a Zephyr VAWT which includes 175 stator vanes with reverse winglets around the rotor. This design has a high solidity, due 176 to which, it performs well at low  $\lambda$  but at higher wind speeds, the peak performance is 177 limited. Hayashi et al [33] showed that using guide vanes in front of a drag-based rotor 178 increases the average Torque Coefficient ( $C_T$ ) at low  $\lambda$ . They also tested a stacked config-179 uration of the turbine with three stages. Golecha et al [31] conducted experiments on sim-180 ilarly stacked two stage and three stage configurations, resulting in a decrease in C<sub>P</sub> as 181 compared to single stage rotor with the same aspect ratio. However, for multi-stage tur-182 bines, the performance improved when a deflector plate is placed in front of the rotors. 183

Su et al [34] investigated the effect of V-shaped blades to improve the aerodynamic 184 performance of a conventional straight bladed lift-based turbine. High-fidelity 3D Com-185 putational Fluid Dynamics (CFD) based simulations were carried out and the results ob-186 tained indicate 20 % increase in C<sub>P</sub> when V-shaped blades are used. Reduction in torque 187 fluctuation has also been observed, resulting in more stable performance of the VAWT. 188 Jiang et al [35] numerically investigated the effects of blade tip shape and supporting strut 189 on a large turbine model with the aim to reduce blade tip vortex and to optimise the struc-190 ture of the turbine for enhanced performance. Roy and Saha [36] developed a new blade 191 profile for a drag-based turbine from a series of experimental studies on Bach and Benesh 192 type turbines by modifying geometric arcs, overlapping distances and dimensions of 193 blade profiles. Their design show improved performance as compared to other blade pro-194 file shapes such as semi-circular and semi-elliptical. It is clear that the use of a deflector 195 (or stator) improves the performance characteristics of VAWTs however, as most of these 196 studies have been carried out using prescribed  $\omega$ , flow induced rotation of the turbines 197 has not been investigated [37], which may have a significant effect on the performance of 198 the turbines. We aim to numerically investigate flow induced rotation of drag-based 199 VAWT, with and without stator, focusing on the start-up dynamics of the VAWTs. 200

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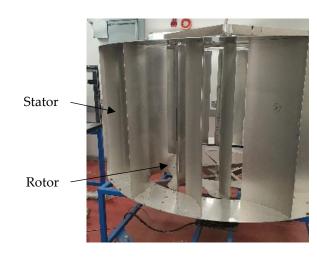
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# 2. Numerical Modelling of the Drag-based VAWT

This section details the various steps involved in developing the numerical model of 204 the drag-based VAWT, with and without the stator. 205

#### 2.1. Geometry of the VAWT and the Flow Domain

The VAWT considered in the present study comprises of 12 rotor blades. This is the 208 same VAWT that Gareth [38] initially developed and analysed, and then the authors carried out a number of empirical and numerical investigations on it over a number of years 210 [39-44]. The full-scale VAWT (with stator) is shown in figure 1(a). A Computer Aided 211 Design (CAD) model of the same VAWT has been created, as shown in figure 1(b) (not 212 including the stator). The inner and outer diameters of the rotor are 1 m and 1.4 m respectively, while the height of the VAWT is 1 m. 214



(a)

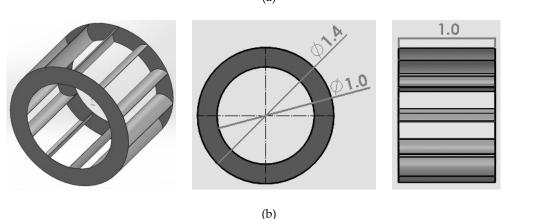


Figure 1. (a) Real-world VAWT model (with stator) (b) CAD model of the VAWT without stator 220

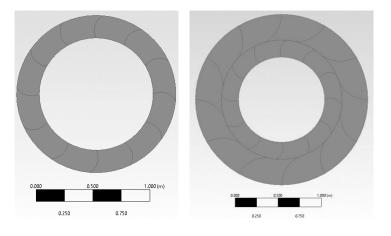
For the purpose of carrying out Computational Fluid Dynamics (CFD) investigations 221 on the start-up dynamics of the VAWT, the CAD model has been converted into the nu-222 merical model using a widely used commercial CFD package Ansys® Fluent. It is im-223 portant to point out here that although the CAD model of the VAWT shown in figure 1(b) 224 is 3D, we have extracted only a 2D VAWT model from it for the purpose of numerical 225 investigations. The justification for not considering the 3D VAWT model is its computa-226 tional expense and the availability of computational resources. A typical 3D VAWT 227 model, for steady-state CFD investigations, comprises of  $> 5 \times 10^6$  mesh elements [45]. For 228 time-dependant CFD investigations (as in this study), where the aim is to capture the start-229

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up dynamics of the VAWT, the number of mesh elements required is significantly higher. 230 Additionally, instead of running the numerical simulations for a few thousand iterations, 231 the simulations should be performed over a few million iterations (details in sections below), which is computationally prohibitive; each simulation in the present work took 4 months. Other computational-based research studies carried out on the start-up dynamics have also considered 2D models [15,17-20,24-25]. Thus, the choice of considering a 2D model is justified. 230 231 232 233 234 235

The 2D numerical models of both the VAWTs considered in the present study i.e. 237 with and without the stator, are shown in figure 2(a). Note: the stator also comprises of 12 238 blades and has an outer diameter of 2 m [38]. These VAWT models are placed inside a 239 rectangular flow domain, having a length of 15 m and a height of 6 m, as shown in figure 240 2(b). In order to generate a controlled mesh in the near-wall and wake regions of the 241 VAWTs, where the flow is expected to be more complex in nature and changing fast w.r.t. 242 time, an inner domain has been created, having a length of 10 m and a height of 2.5 m. 243 Asim et al [10] have previously shown that using two domains, having sizes specified 244 here, for multi-blade drag-based VAWTs, help in capturing the complex flow features ac-245 curately. 246





(b)

Figure 2. (a) 2D models of the VAWTs with and without stator (b) Flow domain of the VAWTs

# 2.2 Meshing of the Flow Domains

In order to generate an appropriate mesh within the flow domain, three different 254 sizing have been specified. For the outer domain, a mesh size of 40 mm has been generated. For inner domain and the core region of the VAWT, a sizing of 20 mm has been 256 specified, while for the rotor (and stator) region, mesh elements of size 10 mm have been 257

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generated. The resulting mesh within the flow domain is shown in figure 3. Asim et al [10] 258 specified size of 100 mm to the outer domain, 30 mm to the inner domain and 15 mm to 259 the blades region for conducting steady performance analysis of this VAWT (with stator). We have further decreased the size because we have to capture more intrinsic flow details during the start-up of the VAWTs. Moreover, the mesh elements generated in the present study are all quadrilaterals, with enhanced mesh structure. The resulting number of mesh 263 elements is ~ 2 x 10<sup>5</sup>. 264

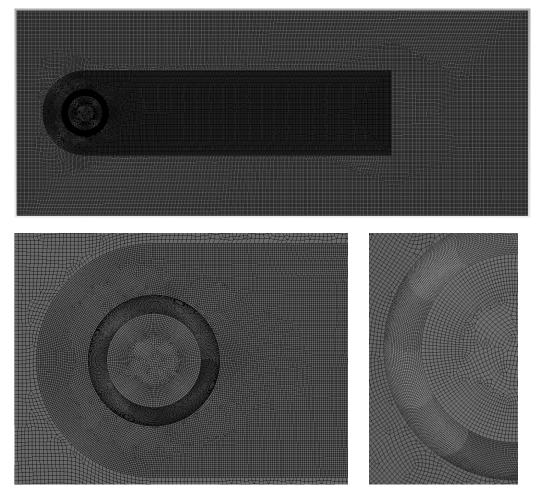


Figure 3. Spatial discretisation of the flow domains

It is a well-known fact that in any CFD based study, the appropriateness of the mesh 268 being used is crucial for the accuracy of the results obtained [46]. There is quite a straight-269 forward approach used for this when analysing steady-state performance of tur-270 bomachines, which is known as mesh independence study [47]. This approach, however, 271 is not suitable for time-dependant solutions, especially where the length scales of energy 272 carrying eddies are expected to change significantly during the numerical calculations. 273 The suitability of a mesh is dependent on how well it can capture the smallest length scale 274 in the flow domain. As wall effects are dominant in case of VAWTs, we need to ensure 275 that near-wall mesh density is adequate. Thus, we define a parameter that relates cell-base 276 length scale to its height as a Custom Field Function (CFF). Equation (1) defines the ratio 277 of cell-base length to the cell's height ( $\varphi$ ) as [10]: 278

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$$=\frac{\sqrt{A}}{2 y} \tag{1} 280$$

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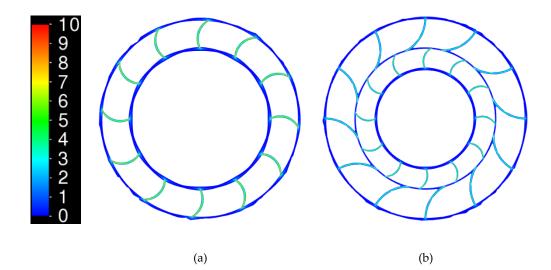
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where A is the element's face area (in m<sup>2</sup>) and y is the wall distance (in m). Low values of 282  $\varphi$  are considered appropriate to capture small eddies in near-wall regions (typically < 10). 283 Figure 4 depicts the spatial variations in  $\varphi$  at time 0.0382s i.e. very early on in the start-up 284 of the VAWT. It can be seen that  $\varphi$  values are higher in the near-wall regions, as expected, 285 but less than 10. In comparison with the stator blades, the rotor blades have slightly higher 286  $\varphi$ , due to their rotation. Hence, it can be concluded that the mesh considered in the present 287



study is appropriate to capture the small-scale eddies accurately.

**Figure 4**. Variations of  $\varphi$  in the near-wall regions of the VAWT (a) without stator (b) with stator 289

#### 2.3 Boundary Conditions and Flow Governing Equations

The boundary conditions specified in the present study are summarised in Table 1. The UK annual average wind speed of 4.2 m/s has been considered as the inlet flow velocity. The outlet of the flow domain has been specified with atmospheric pressure condi-293 tion. The top and bottom edges of the outer flow domain have been specified as symmetries, to avoid wall effects [24]. Stator blades are stationary, as expected, while the rotor blades are moving. The modelling of rotor blades' rotational motion is discussed in the 296 next section.

Table 1. Boundary conditions specified to the numerical model

Boundary	Туре	Value
Inlet	Velocity	4.2 m/s
Outlet	Pressure	0 Pa,g
Top and Bottom edges of the Outer Domain	Symmetry	
All edges of the Inner Domain	Sliding Interfaces	
Stator Blades	Stationary Wall	No-slip
Rotor Blades	6DOF	No-slip

Based on the inlet flow velocity of 4.2 m/s, air density and dynamic viscosity of 1.225 kg/m<sup>3</sup> and  $1.789 \times 10^{-5}$  Pa.s, the Reynolds number (Re) has been calculated to be  $4 \times 10^{5}$  for 301 the VAWT without stator (and  $5.7 \times 10^5$  for the VAWT with stator). Hence, the flow is 302 turbulent, and the corresponding mass conservation equation is: 303

$$\frac{\partial \mathbf{u}_{i}}{\partial \mathbf{x}_{i}} = \mathbf{0} \tag{2} \quad 305$$

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For momentum conservation, we employ the conventional Reynolds Averaged Na-307 vier-Stokes equation (URANS), which can be expressed as [48]: 308

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$$\frac{\partial \overline{u}_{i}}{\partial t} + \frac{\partial}{x_{j}} \left( \overline{u_{1}} \overline{u_{j}} \right) = -\frac{1}{\rho} \frac{\partial \overline{p}}{\partial x_{i}} + \frac{\partial}{x_{j}} \left[ \upsilon \left( \frac{\partial \overline{u}_{i}}{\partial x_{j}} + \frac{\partial \overline{u}_{j}}{\partial x_{i}} \right) \right] + \frac{\partial}{x_{j}} \left( -\overline{\dot{u}_{1}} \dot{u}_{j} \right)$$
(3) 310  
311

where u<sub>i</sub> and u<sub>j</sub> are velocity vectors, ' represents fluctuations, <sup>-</sup> represent mean, p is the 312 pressure and v is the turbulent kinematic viscosity (or eddy viscosity).  $-\dot{u}_1\dot{u}_1$  is the Reyn-313 olds stress. In order to relate the eddy viscosity (v) to Reynolds stress, we employ the 314 Spalart-Allmaras turbulence model that comprises of an eddy viscosity transport equa-315 tion, in order to close equation (3). Sun et al [19] also employed the Spalart-Allmaras 316 model while investigating start-up dynamics of lift-based VAWTs, as it is widely used in 317 aerodynamics and turbomachinary applications. The transport equation for the turbulent 318 kinematic viscosity in the Spalart-Allmaras model is expressed as [49]: 319

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$$\frac{\partial}{\partial t}(\rho \upsilon) + \frac{\partial}{\partial x_{i}}(\rho \upsilon u_{i}) = A_{\upsilon} + \frac{1}{\sigma_{\upsilon}} \left[ \frac{\partial}{\partial x_{j}} \left\{ (\mu + \rho \upsilon) \frac{\partial \upsilon}{\partial x_{j}} \right\} + B\rho \left( \frac{\partial \upsilon}{\partial x_{j}} \right) \right] - C_{\upsilon}$$
(4)

where  $A_{\nu}$  is the production and  $C_{\nu}$  is the destruction of  $\nu$ ,  $\sigma_{\nu}$  and B are the constants.

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#### 2.4 Six Degree of Freedom (6DOF) Solver

For modelling flow induced rotation of the rotor blades, a Six Degree of Freedom (6DOF) solver has been employed in the present study. In 6DOF solver, the aerodynamic forces and moments applied by the wind on the rotor blades are computed numerically, and then used to determine (and apply) the resulting angular rotation to the rotor blades, along their centre of gravity. Based on Newton's second law, we know that the torque applied (T) is expressed as:

$$\Gamma = I\alpha \tag{5} 333$$

where I is the mass moment of inertia of the rotor (in kg.m<sup>2</sup>) and  $\alpha$  is its angular accelera-335 tion (in rad/s<sup>2</sup>). The real-world VAWT shown in figure 1(a) is made out of cold rolled 336 sheets of aluminium, having a wall thickness of 0.5 mm. Based on VAWT's dimensions, I 337 of the rotor has been calculated as 12.93 kg.m<sup>2</sup>, and specified in the 6DOF solver. In the 338 present study, we have not considered the generator load on the VAWT, thus, the angular 339 velocity ( $\omega$ ) of the rotor keeps on increasing ( $\alpha > 0$ ) till average T $\rightarrow 0$ . There will still be 340 fluctuations in T across 0 after the VAWT has reached steady operation due to the cyclic 341 variations. Once T has been computed at a particular time stamp (t), the angular position 342 of the rotor ( $\theta$ ) is updated according to [19]: 343

$$\theta^{i+1} = 2\theta^i - \theta^{i-1} + \frac{\omega}{\tau} \Delta t^2 \tag{6} 345$$

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# 2.5 Temporal Resolution

An important parameter in equation (6) is the change in time stamps ( $\Delta t$ ), which is 348 more commonly known as the time step size. We know that during the start-up of 349 VAWTs, as t increases,  $\omega$  also increases, till the steady operation of the VAWT is achieved. 350 Published literature on the start-up dynamics of VAWTs [18-19,24] use a fixed value of  $\Delta t$ 351 in seconds. The authors believe that this is not an appropriate approach because  $\omega$  keeps 352 on increasing during the start-up. Using a fixed  $\Delta t$  in seconds can potentially not capture 353 the small-scale eddies, leading to inaccurate flow fields and thus, the torque applied on 354 the rotor (T) and eventually, the angular position of the blades ( $\theta$ ). Thus, a more appro-355 priate methodology needs to be adopted for time advancement, which can take into ac-356 count the variations in the flow field. In the present study, we use the Courant Number 357 (CL) for this purpose, which can be expressed as [50]: 358

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$$C_{\rm L} = |\mathbf{U}| \frac{\Delta t}{\Delta x} \tag{7} \quad 360$$

where |U| is the flow velocity magnitude (in m/s) and  $\Delta x$  is the distance between neighbouring mesh elements (in m).  $C_L$  is a dimensionless number that basically represents the amount of time a flow particle stays within a mesh element.  $C_L$  should ideally be < 1, but that is computationally very expensive. In the present study, we use  $C_L = 1$ .

Specification of an appropriate CL does not mean that further checks on its suitability 366 are not required, especially during the start-up phase. Thus, in the present study, we define a CFF as: 368

$$\tau = \frac{V^{\frac{1}{3}}}{|U|} \tag{8} 369$$

where V is the cell volume (in m<sup>3</sup>). We need to ensure that  $\Delta t$  we obtain by using  $C_L = 1$  is 371 less than  $\tau$ . At t = 0.0382s (very early on during the start-up of VAWTs),  $\Delta t$  for VAWT 372 without stator is 0.000082s and for the VAWT with stator is 0.000047s. We set these as the 373 lower limit and show the variations in  $\tau$  in figure 5. It can be seen that there are no data 374 gaps (unfilled areas) in both the cases, confirming that the choice of  $C_L$  is appropriately 375 capturing small-scale eddies. 376

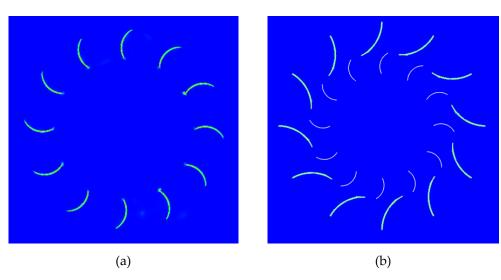


Figure 5. Variations of  $\tau$  in the vicinity of VAWT (a) without stator (b) with stator

#### 3. Results and Discussions

# 3.1 Start-up Dynamics of VAWT without Stator

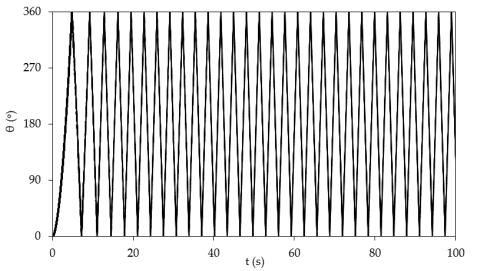
In this section, we present detailed analysis of flow induced rotation of the first 381 VAWT model considered in the present study i.e. VAWT without Stator. Firstly, the timedependant rotation of the VAWT has been discussed, followed by the variations in the 383 flow fields (pressure and velocity) in the vicinity of the VAWT. Note: both the VAWT 384 models have been run for 100s to ensure that steady rotational speed is obtained. In terms 385 of time steps, this means that both the simulations have been run for more than 6x10<sup>5</sup> time 386 steps. 387

Figure 6 depicts the time-dependant rotation of the VAWT without Stator, where figure 6(a) shows the angular rotation ( $\theta$  in °) while figure 6(b) shows the rotational speed of the VAWT ( $\omega$  in rpm). It can be seen in figure 6(a) that, starting from t = 0s, as incident air exerts force on the rotor blades, they start to rotate about their central axis. As time progresses, the rotation of the VAWT increases till 360° or one revolution. Rotating further, it can be seen that  $\theta$  decreases from 360° to 0°, which doesn't mean that the VAWT rotates in the opposite direction; it just completes another full revolution in normal 394

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direction of operation. The first revolution of the VAWT is completed in 4.86s, while the 395 second revolution takes 2.38s, which is less than half the time taken to complete the first 396 revolution (because the VAWT is constantly picking up speed). The next two revolutions 397 take 2s and 1.84s respectively. The time taken to complete a full revolution keeps on de-398 creasing as the VAWT's rotational speed keeps on increasing. During the first 100s of op-399 eration, the VAWT undergoes ~ 60 full revolutions. It is important to note here that figure 400 6(a) provides information on the angular rotation of the VAWT but doesn't provide ex-401plicit information on when the VAWT reaches a steady/constant rotational speed. More-402 over, the information provided in figure 6(a) remain the same (qualitatively) for all differ-403 ent types of turbomachines. 404

Figure 6(b) depicts time-dependant variations in the rotational speed of the VAWT. 405 There are three distinct observable regions in figure 6(b). It can be seen that starting from 406 initial position at t = 0s, the VAWT's rotational speed starts to increase almost linearly till 407 9.4s. By this time, the rotational speed of the VAWT reaches 32 rpm. From 9.4s till 23s, the 408rate of increase in the rotational speed of the VAWT decreases. By 23s, the VAWT has 409 achieved its maximum rotational speed of 37.8 rpm. From 23s onwards, the rotational 410 speed of the VAWT remains constant. There are a number of interesting observations and 411 statistics that we need to analyse here in order to better understand the start-up dynamics 412 of the VAWT. Figure 6(b) clearly demonstrates the differences between drag-based and 413 lift-based VAWTs. For lift-based VAWTs, it has been observed in a number of recent re-414 search studies [11,15-16,19-20] that there are four distinct zones associated with the start-415 up of the VAWT. These are i) linear acceleration, ii) first plateau (constant  $\omega$ ), iii) second 416 linear acceleration, and iv) second plateau. Please note that here we have not considered 417 the transition zones from linear accelerations to the plateaus; if we include these then there 418 are six start-up zones. For the drag-based VAWT considered in the present study, we ob-419 served i) linear acceleration, and ii) plateau only (again disregarding the transition; for 420 better comparison with lift-based VAWTs). We believe that we have run the simulations 421 for long enough time to ensure that if a second acceleration existed, it would have been 422 appropriately captured. Moreover, in comparison with the work carried out by Tigabu et 423 al [24] on lift-based hydro turbines, where they noticed an additional zone i.e. offshoot, 424 which was a continuation from linear acceleration past the maximum rotational speed, we 425 did not experience any such offshoot in case of drag-based turbine. In the present case, 426 the VAWT completed 12 full rotations before reaching its maximum/peak rotational speed 427 of 37.8 rpm in 23s). It would be also useful to know the local  $\omega$  values at certain key time 428 stamps. After 5s of operation,  $\omega$  = 22.3, which is 59 % of the maximum  $\omega$ . After 10s of 429 operation,  $\omega$  = 32.6 (86 % of maximum) and at t = 20s,  $\omega$  = 27s (98 % of maximum), which 430 is in-line with our categorisation of the zones associated with the start-up of the VAWT. 431



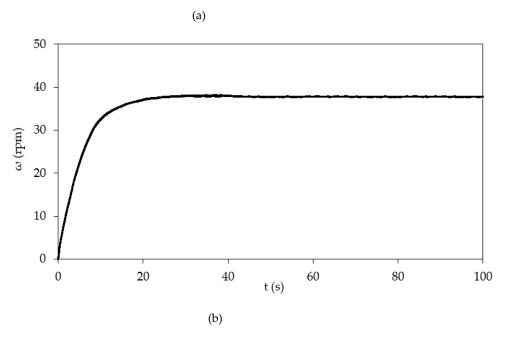


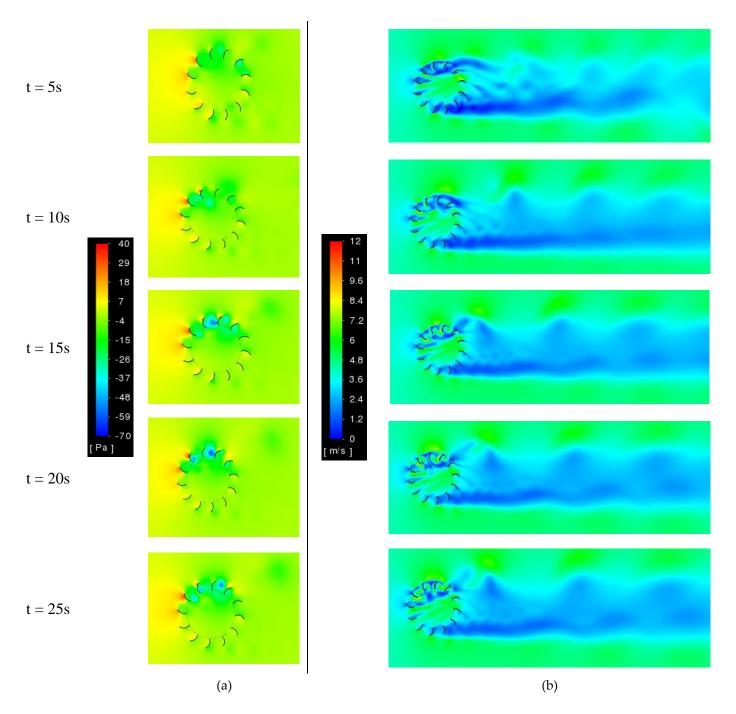
Figure 6. Rotational characteristics of the VAWT without stator

After analysing the rotational behaviour of the VAWT during its start-up, it is beneficial 438 to study the flow fields in the vicinity of the VAWT. Thus, figure 7(a) depicts the static 439 gauge pressure (in Pa), while figure 7(b) depicts the flow velocity magnitude (in m/s) 440 within the flow domain. Please note that in all the flow fields presented in the present 441 study, the flow of air is from left to right in the figures. The flow fields have been obtained 442 at t = 5s, 10s, 15s, 20s and 25s respectively, where the first two time stamps (t = 5s and 10s) 443 represent the linear acceleration zone, the next two (t = 15s and 20s) represent the transi-444 tion zone, while the last time stamp (t = 25s) represent steady operation of the VAWT at 445 its maximum  $\omega$ . For effective comparison purposes, the scale of the variations, both in 446 pressure and in velocity, have been kept the same for all the different time stamps consid-447 ered here. It can be seen in figure 7(a) that the pressure is higher on the windward side 448 (upstream), while it is relatively lower on the leeward side (downstream) of the VAWT, 449 as expected [40-45]. This trend remains the same irrespective of the time stamps consid-450 ered. Upon impinging the rotor blades upstream, the air pressure increases up to 40 Pa,g, 451 (on the pressure side of the blades) which remains roughly the same for all the time 452 stamps. On the suction side of these blades, the pressure is weakly negative. The flow 453 detachment from the top and bottom of the VAWT (in the figure) is visible, generating 454 local flow recirculation zones (possibly vortices). Due to the different orientation of the 455 blades at the top and bottom, the frequency and size of these recirculating zones is differ-456 ent; on the top side of the VAWT, because the blades' curvature is in the direction of flow, 457 the recirculation zones are fewer in number but bigger in size, while on the bottom side, 458 the number of these local flow structures is more and they are smaller in size, leading to 459 asymmetric flow on either side of the VAWT. In between the top blades, low-pressure 460 zones start to develop during the linear acceleration phase of the start-up (t = 5s and 10s). 461 The pressure in these zones drop further during the transition phase (t = 15s and 20s), and 462 is then maintained during the steady operation of the VAWT (t = 25s). 463

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**Figure 7.** (a) Static gauge pressure and (b) flow velocity magnitude variations in the vicinity of the VAWT without stator

While the pressure field provides valuable information in the near-wall (blades) re-467 gions of the VAWT, the velocity field shown in figure 7(b) highlights the local flow char-468 acteristics downstream the VAWT, including the wake region. The asymmetry discussed 469 above on either side of the VAWT is clearly visible here. It is important to point out over 470 here that such local time-based flow field analysis of VAWTs, where rotation is induced 471 by the flow, is very scarce in the published literature and thus, comparative analysis with, 472 for example lift-based VAWTs, is very difficult. The velocity field maps provided by Asr 473 et al [15] for a lift-based VAWT are inconclusive, stating that complex flow patterns are 474 observed. Some other studies have presented vorticity maps. We are skeptical about these 475 as these studies are based on 2D modelling (similar to the present study) and representing 476 an inherently 3D phenomena (vorticity) in 2D can be very misleading, and hence, we have 477

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# 3.2 Start-up Dynamics of VAWT with Stator

to stabilise during the transition phase.

The time-dependant rotational characteristics of the VAWT with the stator blades 485 have been presented in figure 8, where figure 8(a) depicts the angular rotation ( $\theta$ ), while 486 figure 8(b) depicts the rotational speed ( $\omega$ ). It can be seen be seen in both figures 8(a and 487 b) that the start-up dynamics of the VAWT with the stator are qualitatively very similar 488 to the VAWT without the stator. For the first 100s of operation, the VAWT with stator 489 undergoes ~ 133 full revolutions, compared to 60 revolutions for the VAWT without the 490 stator, indicating that the rotational speed of the VAWT with stator is significantly higher. 491 In figure 8(b), it can be clearly seen that the maximum  $\omega$  of the VAWT reaches 85.3 rpm, 492 which is more than twice the maximum  $\omega$  for the VAWT without the stator. Hence, it can 493 be concluded that by adding an appropriately designed stator to a drag-based VAWT, the 494 steady rotational speed of the VAWT increases, and thus, the power output of the VAWT 495 also increases. 496

not included any vorticity maps here. It is however evident from figure 7(b) that the flow

characteristics in the wake region of the VAWT are highly non-uniform and extend mul-

tiple VAWT diameters downstream. During the linear acceleration phase, the velocity

field seems to be a strong function of time and angular position of the blades, and seems

In terms of the start-up dynamics of the VAWT with the stator blades, we observe 497 the same three zones that we observed earlier in case of VAWT without the stator i.e. 498 linear acceleration, transition and plateau region. The time taken by the VAWT to reach 499 peak  $\omega$  is 49s, which is slightly more than double the time taken by the VAWT without 500 the stator (23s). Similarly, the VAWT with the stator undergoes 61 full revolutions before 501 achieving peak  $\omega$ , while the VAWT without the stator underwent 12 full revolutions. This 502 demonstrates that it takes longer for the VAWT with stator to achieve steady rotation, 503 because it has to reach a much higher  $\omega$ . Further analysing the start-up dynamics of the 504 VAWT with stator with the aim to investigate the reasons behind this delay in reaching 505 steady operation, we observe that  $\omega$  = 50.8 (59 % of peak) after 5s of operation. Interest-506 ingly, the VAWT without the stator also reached 59 % of peak  $\omega$  after first 5s of operation. 507 At t = 10s,  $\omega$  = 67.6 (79 % of peak) for the VAWT with stator, while at the same time stamp, 508the VAWT without stator reached 86 % of peak  $\omega$ . Further progressing in time, at t = 20s, 509  $\omega$  = 79.3 (93 % of peak) for the VAWT with stator, while the VAWT without stator reached 510 98 % of peak  $\omega$ . Up to this point in time, the difference in the ratio of instantaneous-to-511 peak  $\omega$  between the two VAWT models considered has been < 10 %. At t = 30s,  $\omega$  = 83.1 512 (97 % of peak), and at t = 40s,  $\omega$  = 84.5 (99 % of peak) in case of VAWT with stator. Thus, 513 it is clear that although the time spent by both the VAWT models during linear accelera-514 tion is roughly the same, the time spent on the transition to steady operation is signifi-515 cantly different, with VAWT with stator spending 39s, while the VAWT without stator 516 spending just 13s to complete this transition. This will be further analysed later in the 517 study while discussing figure 10. 518

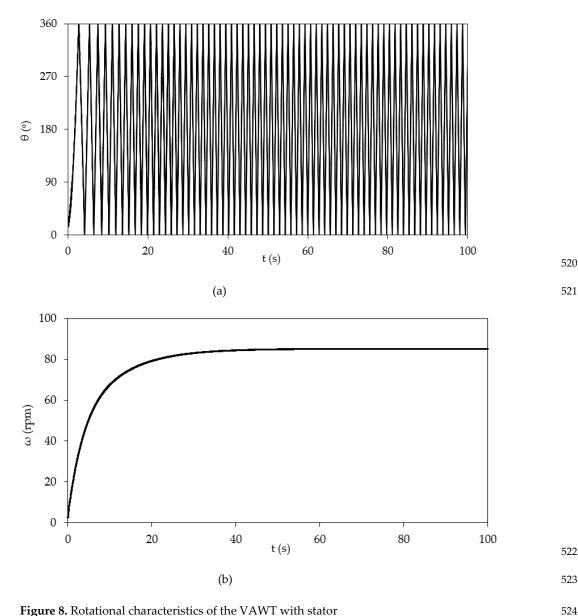
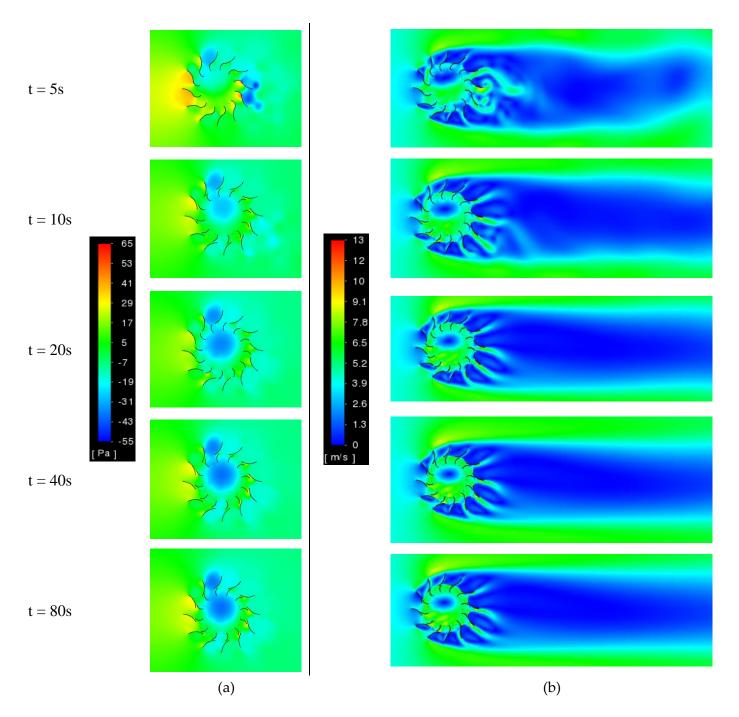


Figure 8. Rotational characteristics of the VAWT with stator

The flow fields associated with the VAWT model considered here (having stator 525 blades) are shown in figure 9. It can be seen in figure 9(a), in comparison with figure 7(a) 526 regarding the VAWT without the stator, that although the scale of static gauge pressure 527 variations is similar i.e. ~ 110 Pa,g (due to the same wind velocity), it has shifted towards 528 the positive pressure region (higher positive and lower negative pressures). The flow pat-529 terns observed in case of VAWT with stator are quite different to the VAWT without sta-530 tor. It is observed that there exist large areas of negative pressure inside the core region 531 and between top stator blades of the VAWT, which initially start to develop while the 532 VAWT is undergoing linear acceleration. The pressure further drops during the transition 533 phase until it reaches its maximum value at steady operation of the VAWT. The asym-534 metry between the top and bottom blades is visible again, due to the orientation of the 535 stator blades, with the difference that the stator blades are non-rotating. 536



**Figure 9.** (a) Static gauge pressure and (b) flow velocity magnitude variations in the vicinity of the VAWT with stator

Analysing the flow velocity magnitude field shown in figure 9(b), it can be seen that 540 the flow is significantly more uniform downstream the VAWT, in comparison with the 541 VAWT without the stator blades (figure 7(b)). Hence, the addition of non-rotating stator 542 blades leads towards enhanced structural stability of the VAWT, based on flow uni-543 formity. This must not be confused with higher vibrations in the structure of the VAWT 544 with stator blades due to higher rotational speed (physically observed in lab-based exper-545 iments [38-39]). With regards to the start-up dynamics of the VAWT with stator, it is evi-546 dent that during the linear acceleration phase, as the flow field is constantly developing, 547 the flow is quite non-uniform in the wake region of the VAWT. The flow then stabilises 548 during the transition phase, and eventually reaches a steady condition. In terms of the 549

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scale of flow velocity variations, there is negligible difference between the two VAWT models considered in the present study (maximum flow velocity of 12 m/s).

#### 3.3. Start-up Performance Characteristics of VAWTs with and without Stator

After conducting a detailed analysis of the rotational characteristics and the flow 554 fields associated with the two VAWT models considered, it is prudent to investigate timedependant variations in the conventional performance characteristics of the VAWTs, with 556 special attention to the start-up phases. Thus, figure 10 depicts the variations in Tip Speed 557 Ratio ( $\lambda$ ) for both the VAWT models. As  $\lambda$  is dependent on the rotational velocity ( $\omega$ ) of 558 the VAWT (equation 9), the trends are identical to  $\omega$  variations presented above. 559

$$\lambda = \frac{\omega R}{V} \tag{9} 561$$

where R is the outer radius of the rotor blades (m) and V is the wind velocity (m/s). The 563 reason for presenting the variations of  $\lambda$  here is to identify the operational  $\lambda$  of the VAWT 564 at that particular wind velocity. This information is of particular interest because in con-565 ventional CFD based studies on VAWTs, where  $\omega$  is prescribed as a boundary condition, 566 it is a common practise to estimate the operational  $\lambda$ , which corresponds to peak power 567 generation. This estimation is based on numerical investigations over a wide range of pre-568 scribed  $\omega$ . This process can lead towards significant over or under estimation of opera-569 tional  $\lambda$ . For example, Park et al [51] have carried out steady CFD investigations on the 570 VAWT with stator (same as considered in the present study) and has prescribed  $0.1 \le \omega \le$ 571 0.9. The operational  $\lambda$  estimated based on the numerical results is 0.5. Gareth [38] has also 572 considered operational  $\lambda$  of 0.5 for the same VAWT. In figure 10 however, it can be clearly 573 seen that the operational  $\lambda$  for the VAWT with stator is 1.5. It is important to note here 574 that both Park et al [51] and Gareth [38] have considered wind speed of 4 m/s, which is 575 quite similar to the wind speed of 4.2 m/s considered in the present study. For the VAWT 576 without the stator, the operational  $\lambda$  has been calculated as 0.7, which is ~ half of the op-577 erational  $\lambda$  for VAWT with stator (like  $\omega$ ). 578

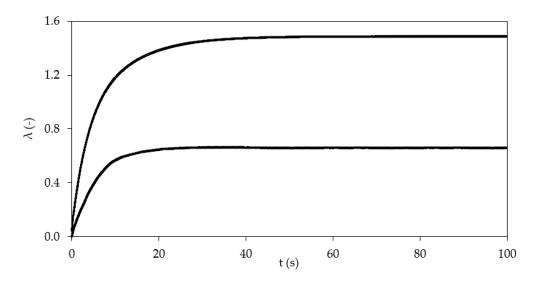


Figure 10. Variations in the Tip Speed Ratio of the VAWTs with and without stator

Concluding the discussions on the start-up dynamics of VAWTs, with and without 582 stator, it is important to highlight the main limitation of the present work. We have not 583 (numerically) considered the generator load, which is applied to the turbine shaft in the 584 real-world, and which potentially affect the operational  $\lambda$  of the VAWT. Without applying 585 the generator load, we get zero average torque (and thus power) output from the VAWT, 586

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which is shown in figure 11. It can be seen that, during the steady operation of the VAWTs 587 (plateau region), although we get +ive and -ive instantaneous torque values from both the 588 VAWT models, their average is zero. This has been highlighted by Tigabu et al [24] as 589 well. The instantaneous torque values for the VAWT with stator are considerably higher 590 than for the VAWT with stator (+16.5 Nm compared to +2.1 Nm max), indicating that there 591 is potentially more power generating capability in the former, which can be extracted 592 through the application of brake torque. 593

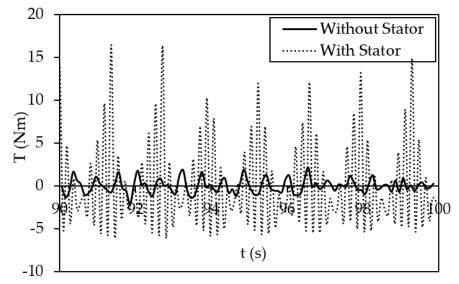


Figure 11. Variations in the torque generated by the VAWTs with and without Stator

# 4. Conclusions

The start-up dynamics of 12-bladed drag-based VAWT models, with and without the 597 stator blades, have been investigated in the present study. The start-ups of the VAWTs 598 have been numerically simulated, at average UK wind speed, through the use of 6DOF 599 solver in a commercial Computational Fluid Dynamics solver, mimicking flow induced 600 rotation to the VAWTs. We capture time-dependant performance characteristics and local 601 flow features in the vicinity of both the VAWT considered, not only during the start-up 602 phase, but extending to the steady operation of the VAWTs. Based on the detailed results 603 presented, it can be concluded that the addition of appropriately designed stator blades 604 increases the rotational speed, and thus, tip speed ratio and potential power output from 605 the VAWT.

In terms of the start-up, we observe three distinct zones i.e. linear acceleration, transition and steady operation of the VAWTs, irrespective of whether there are stator blades or not. The linear acceleration phase takes roughly the same time, from starting, in both the VAWTs, while the transition phase in case of VAWT with stator is prolonged, as the peak  $\omega$  is more than double compared to the VAWT without stator. The flow fields asso-611 ciated with rotor only VAWT are significantly more asymmetric, both due to the orienta-612 tion and the rotation of the blades, leading to pockets of recirculation zones being shed 613 into the wake region of the VAWT. 614

For the VAWT with stator, as the stator blades are stationary, the intensity of these 615 recirculation zones is less and thus, we observe more uniform flow downstream the 616 VAWT. In either case, the complex flow patterns in the near-wall regions start to develop 617 during the linear acceleration phase, get strengthened during the transition phase, before 618 reaching their maximum values (of corresponding flow parameters) during steady oper-619 ation of the VAWTs. The operational tip speed ratio of the VAWT with stator, like the 620 rotational speed, is more than double than the VAWT with stator, reaching a value of 1.5. 621

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It is envisaged that this study will pave the way for further scientific investigations on the start-up dynamics of drag-based VAWTs.

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Conflicts of Interest: The authors declare no conflict of interest.

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