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Review



## **Karlovitz Numbers and Premixed Turbulent Combustion Regimes for Complex-Chemistry Flames**

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**Abstract**: The structure of premixed turbulent flames and governing physical mechanisms of the influence of turbulence on premixed burning are often discussed by invoking combustion regime diagrams. In the majority of such diagrams, boundaries of three combustion regimes associated with (i) flame preheat zones broadened locally by turbulent eddies, (ii) reaction zones broadened locally by turbulent eddies, and (iii) local extinction are based on a Karlovitz number *Ka*, with differently defined *Ka* being used to demarcate different combustion regimes. The present paper aims to overview different definitions of *Ka*, comparing them, and suggesting the most appropriate choice of *Ka* for each combustion regime boundary. Moreover, since certain Karlovitz numbers involve a laminar flame thickness, the influence of complex combustion chemistry on the thickness and, hence, on various *Ka* and relations between them is explored based on results of complex-chemistry simulations of unperturbed (stationary, planar, and one-dimensional) laminar premixed flames, obtained for various fuels, equivalence ratios, pressures, and unburned gas temperatures.

Keywords: turbulent flame; combustion regime diagram; Karlovitz number; complex chemistry

### 1. Introduction

Premixed turbulent combustion is an interplay of turbulent fluid motion, molecular heat and mass transfer, hundreds of chemical reactions between thousands of species, and intensive localized heat release. It is a highly non-linear phenomenon that involves processes whose length or time scales can differ by several orders of magnitude. Therefore, the development of a rigorous theory of turbulent flames is not feasible, whereas direct numerical simulations of them are affordable for the simplest flame configurations only.

In such a situation, governing physical mechanisms of premixed turbulent combustion and turbulent flame structure are often discussed by comparing velocity, length, and time scales of the unperturbed laminar flame, i.e., a planar and one-dimensional flame, which propagates at a constant speed and has a constant thickness, with the counterpart scales of turbulence. More specifically, a laminar premixed flame is commonly characterized by its speed  $S_L$ , thickness  $\delta_L$ , and time scale  $\tau_F = \delta_L/S_L$ . Since turbulence involves a wide spectrum of eddies of significantly different scales, the simplest and widely accepted approach to characterizing the turbulence spectrum consists in adopting both scales of the largest eddies and scales of the smallest eddies. The former includes the rms turbulent velocity u', an integral length scale L, and a time scale  $\tau_T = L/u'$ . The latter includes Kolmogorov velocity  $u_K$ , length scale  $\eta_K$ , and time scale  $\tau_K \approx \tau_T Re_T^{-1/2} \ll \tau_T$ , and  $\eta_K \propto LRe_T^{-3/4} \ll L$ , where  $Re_T = u'L/v_u \gg 1$  is the turbulent

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**Copyright:** © 2022 by the authors. Licensee MDPI, Basel, Switzerland. This article is an open access article distributed under the terms and conditions of the Creative Commons Attribution (CC BY) license (https://creativecommons.org/license s/by/4.0/). Reynolds number and  $v_u$  is the kinematic viscosity of unburned reactants. Note that the three scaling laws written above contain unity–order constants, which will be discussed later. Moreover, the entire inertial range of Kolmogorov turbulence is characterized by a single scalar quantity, the mean rate of viscous dissipation of turbulent kinetic energy  $\varepsilon = 2v\overline{S_{ij}S_{ij}} \propto u'^3/L$  [1–4]. Here,  $S_{ij} = 0.5(\partial u_i/\partial x_j + \partial u_j/\partial x_i)$  is the rate of the strain tensor,  $u_i$  is the *i*-th component of the velocity vector **u**, and the summation convention applies to repeated indexes.

Depending on the relations between the aforementioned flame and turbulence scales, different scenarios of the influence of turbulence on premixed combustion can be assumed. For example, in their classical papers, Damköhler [5] and Shelkin [6] have hypothesized that the influence of turbulence on combustion can be reduced to a wrinkling flame surface (or flame sheet) if the turbulence length scales are much larger than the thickness  $\delta_L$ , i.e.,  $\delta_L \ll \eta_K \ll L$  in contemporary terms. Under these conditions, the local flame structure was considered to be similar to the structure of the unperturbed laminar flame, with the bulk burning rate being increased due to wrinkling of the flame surface by turbulent eddies. Today, such a premixed turbulent combustion regime is well established and is called "flamelet regime", where the word "flamelet" refers to a thin, inherently laminar flame wrinkled and strained by turbulent eddies.

In the opposite limit case of  $\eta_K \ll L \ll \delta_L$ , the influence of turbulence on combustion was hypothesized to be reduced to intensifying mixing [5] within a flame broadened significantly by turbulent eddies [6,7]. The existence of this regime is still under discussion, and this issue will be briefly addressed later.

While the pioneering papers by Damköhler [5] and Shelkin [6] highlighted relations between length scales, over (velocity and time) scales were later used to specify boundaries of the two limit combustion regimes discussed above and to explore other combustion regimes. That research resulted in the appearance of premixed turbulent combustion regime diagrams. Today, such diagrams are widely used in the literature. The first combustion regime diagrams were invented by Bray [8], Barrere [9], Borghi [10,11], Williams [12–14], and Peters [15], with a number of modified diagrams being introduced later, e.g., see Refs. [16–22]. The goal of such diagrams is to speculate what physical mechanisms of the influence of turbulence on combustion are of importance under specific conditions and what structure local reaction zones have under such conditions. The following discussion is restricted to premixed combustion in a turbulent flow characterized by a low Mach number, and almost all such combustion regime diagrams are restricted to considering the influence of turbulence on an adiabatic, single-stepchemistry (i.e., thousands of reactions between hundreds of species are modeled with a single reaction between two reactants), and equidiffusive (i.e., molecular diffusivities of all reactants and products are equal to molecular heat diffusivity  $\kappa$  of the mixture) flames. Important effects that stem from the influence of combustion-induced thermal expansion on turbulence [23–27], differences in molecular transport coefficients [28], heat losses, and complex combustion chemistry are often beyond the scope of such diagrams. Thus, a typical combustion regime diagram deals with a constant-density single-reaction wave, which passively propagates in turbulence described by the Kolmogorov theory [1-4].

Moreover, in the classical combustion regime diagrams [8–15], the following simplifications were invoked: (i) all constants in the aforementioned scaling laws were set equal to unity, i.e.,  $u_K = (v\varepsilon)^{1/4} = u'Re_T^{-1/4}$ ,  $\tau_K = (v/\varepsilon)^{1/2} = \tau_T Re_T^{-1/2}$ ,  $\eta_K = (v^3/\varepsilon)^{1/4} = LRe_T^{-3/4}$ , and  $\varepsilon = u'^3/L$ , (ii) all molecular transport coefficients were set equal to the kinematic viscosity v of the mixture, and (iii) the laminar flame thickness was evaluated as follows  $\delta_L = v_u/S_L$ . Under these simplifications,

$$Ka = \frac{\tau_F}{\tau_K} = \left(\frac{\delta_L}{\eta_K}\right)^2 = \left(\frac{u_K}{S_L}\right)^2 = \left(\frac{u'}{S_L}\right)^{3/2} \left(\frac{L}{\delta_L}\right)^{-1/2} = \left(\frac{\varepsilon\delta_L}{S_L^3}\right)^{1/2} = \left(\frac{u'}{S_L}\right)^2 Re_T^{-1/2} = \frac{Re_T^{1/2}}{Da},$$
(1)

where the Karlovitz number Ka characterizes a ratio of the flame and Kolmogorov time scales, whereas the Damköhler number  $Da = \tau_T/\tau_F$  involves the time scale  $\tau_T$  of the largest turbulent eddies. Here, a single symbol Ka is adopted to designate various numbers, which are different in a general case, e.g., if  $\delta_L \neq v_u/S_L$ . Differences in these numbers will be discussed in Section 3. According to Equation (1), among various nondimensional characteristics of flame–turbulence interaction, e.g.,  $Da, Ka, Re_T, \delta_L/L, \delta_L/\eta_K, S_L/u', S_L/u_K$ , etc., only two characteristics are independent. Therefore, typical combustion regime diagrams are 2D planes with different coordinate axes, e.g.,  $Re_T$  and  $u'/S_L$  [8],  $L/\delta_L$  and  $u'/S_L$  [9,11,15],  $\eta_K/\delta_L$  and  $u'/S_L$  [12], or  $Re_T$ and Da [14].

In all the combustion regime diagrams [8–22], lines of constant Karlovitz numbers are of particular importance, while different definitions of these numbers are adopted in different diagrams. Equation (1) shows that the same Ka can be obtained by invoking different non-dimensional characteristics of the influence of turbulence on premixed combustion. However, if (i) differences between unity and the aforementioned constants adopted to link small-scale and large-scale turbulence characteristics are taken into account, (ii) Prandtl number  $Pr \equiv \nu/\kappa \neq 1$ , (iii) the thickness  $\delta_L$  is defined using a molecular transport coefficient evaluated inside the flame, i.e., at a higher temperature  $T > T_u$ , etc., none of the equalities in Equation (1) are met. Moreover, in this case, physical mechanisms that are hypothesized to play an important role under conditions of either  $\tau_K < \tau_F$  or  $\eta_K < \delta_L$ , should be associated with different areas of a combustion regime diagram, whereas these areas coincide if Equation (1) holds. While these ambiguities are rarely emphasized in the literature, they should be borne in mind when analyzing published data and, especially, when comparing results reported by different research groups. Accordingly, the present paper aims to summarize the use of differently defined Karlovitz numbers for demarcating combustion regimes. Moreover, the paper aims to discuss the influence of combustion chemistry on the thickness  $\delta_L$  and, consequently, on differences between differently defined Karlovitz numbers. It is worth stressing, however, that the focus of the following discussion will be restricted to the ambiguity of the term "Karlovitz number", whereas consideration of various regimes of premixed turbulent combustion or boundaries of such regimes is beyond the major scope of the present article, while these issues will briefly be addressed in the following.

In the next section, physical mechanisms whose importance is commonly assessed by invoking criteria of Ka = const are discussed. A summary of differently used in the literature is given in Section 3. Effects of complex combustion chemistry on thicknesses of different zones in a laminar premixed flame and, hence, on differently defined Karlovitz numbers are addressed in Section 4, followed by conclusions summarized in Section 5.

#### 2. Combustion Regime Boundaries and Karlovitz Numbers: A Historical Overview

To the best of the authors' knowledge, Kovasznay [29] was the first who proposed to use a criterion of  $\tau_F u'/\lambda = O(1)$  to characterize the transition from (i) combustion localized to thin inherently laminar flames wrinkled by turbulent eddies (at low values of  $\tau_F u'/\lambda$ ) to (ii) chemical reactions distributed in wide zones (at high values of  $\tau_F u'/\lambda$ ). Here,  $\lambda$  designates the transverse Taylor microscale of turbulence and, under the simplifications invoked to arrive at Equation (1),  $\tau_K = \lambda/u'$ . Kovasznay [29] attributed the transition to the local break-up of laminar flames by turbulence-induced velocity gradients and, following the Kolmogorov theory, adopted  $u'/\lambda$  to characterize the magnitude of the largest velocity gradients created by the smallest-scale turbulent eddies [2–4]. Earlier, the quenching of laminar premixed flames by an external velocity gradient was predicted by Karlovitz et al. [30], and another Karlovitz number  $K = \tau_F dU/dy$  was used as a criterion of such quenching in the laminar combustion literature [31]. Here, dU/dy is the transverse gradient of the *x*-component U(y) of the velocity of a laminar flame upstream of a flame. Later, by theoretically studying the response of twin laminar premixed flames to a strain rate created by two identical counter-flows, Klimov [32] obtained a number of seminal results. In particular, (i) the flame speed can be negative if the strain rate is sufficiently high, and (ii) the flame can be quenched by a higher strain rate, but (iii) the quenching process takes time. Based on these findings, Klimov [32] hypothesized three combustion regimes: (i) combustion localized to thin inherently laminar flames wrinkled by turbulent eddies (at low values of  $\tau_F u'/\lambda$ ), (ii) combustion localized to thin reaction zones strongly perturbed by turbulent eddies, with the probability of local combustion extinction being low (if  $\tau_F u'/\lambda$  is of unity order), and (iii) intermittency of such reaction zones and reacting (self-igniting) hot volumes appearing due to local combustion quenching by turbulent eddies (if  $\tau_F u'/\lambda$  is much larger than unity). The third regime is similar to a turbulent combustion regime hypothesized earlier by Shetinkov [33] and discussed subsequently in his book [34] in a more detailed manner. The reader interested in Shetinkov's concept is also referred to [35].

Based on the theoretical results obtained by Klimov [32], Williams [36] highlighted the appearance of negative flame speeds in a turbulent flow, which could cause the annihilation of hot spots of burned gas. Due to this mechanism, combustion was hypothesized to be localized to thin inherently laminar flames wrinkled by turbulent eddies only if  $\tau_F u'/\lambda \ll 1$  and a criterion of  $\tau_F u'/\lambda = O(1)$  was adapted [36,37], i.e., Williams [36,37] wrote this criterion in a form of  $\delta_L \gtrsim \eta_K$  by invoking the assumptions taken to arrive at Equation (1). Subsequently, by introducing a premixed turbulent combustion regime diagram, Williams [13] (i) placed the focus of consideration on the limit cases of  $\delta_L \ll \eta_K$  and  $\delta_L \gg L$ , (ii) did not highlight a constraint of  $\delta_L = \eta_K$  at high  $Re_T$ , but (iii) noted that an intermediate regime or more than one intermediate regime could exist if  $\eta_K \ll \delta_L \ll L$ .

By referring to the papers by Klimov [32] and Williams [36,37], Bray [8] considered a constraint of  $\tau_F u'/\lambda = 0(1)$  to limit the existence of laminar flames in a turbulent flow. Moreover, Bray [8] drew a line of  $\delta_L = \eta_K$  as a boundary of the flamelet combustion regime on a 2D plane { $Re_T, u'/S_L$ } but did not specify other combustion regimes on this plane, contrary to subsequent multi-regime diagrams [9–22].

Kovasznay [29], Klimov [32], Williams [12–14,36,37], and Bray [8] did not apply the term "Karlovitz number" to turbulent combustion, while Bray [8] introduced a criterion of  $K_2 = \tau_F u'/\lambda$  and used a similar symbol  $K_1$  to designate the Karlovitz stretch factor when referring to combustion quenching in laminar flows and to the pioneering study by Karlovitz et al. [30]. To the turbulent combustion literature, the term "Karlovitz flame stretch factor  $\tau_F u'/\lambda$ " was likely introduced by Abdel-Gayed et al. [38], who discussed that premixed combustion could be quenched by turbulence when  $\tau_F u'/\lambda = O(1)$ .

Thus, the authors of Refs. [8,29,32,36–38] stressed the importance of the number  $\tau_F u'/\lambda$  for characterizing the influence of turbulence on the local flame structure. All these authors agreed that, at  $\tau_F u'/\lambda \ll 1$ , (i) combustion should be localized to thin inherently laminar flames and (ii) an increase in burning rate resulted from an increase in the wrinkled flame surface area, in line with the first Damköhler hypothesis [5]. As far as combustion at a large  $\tau_F u'/\lambda$  is concerned, scenarios discussed in the cited papers are different. Kovasznay [29] and Klimov [32] hypothesized a transition to distributed burning due to local combustion quenching. Williams [36] emphasized the annihilation of hot spots of burned gas. Bray [8] highlighted local combustion quenching by velocity gradients and noted that the quenching could play an important role even if  $\tau_F u'/\lambda$  was smaller than unity. Abdel-Gayed et al. [38] also highlighted flame quenching by turbulence based mainly on their experimental data obtained from statistically spherical flames expanding in turbulence generated by fans in a closed vessel.

There is another important difference between phenomenological scenarios presented in Refs. [8,13,29,32,36–38]. While a constraint of  $\tau_F u'/\lambda = O(1)$  was written in Refs. [8,29,36–38], Klimov [32] clearly stated that thin reaction zones could dominate even at significantly larger  $\tau_F u'/\lambda$ . Accordingly, Klimov (private communications, 1981–2010)

never accepted the term "Klimov-Williams criterion", which was widely applied to the constraint  $\tau_F u'/\lambda = 0(1)$ . Recent experimental and numerical data reviewed elsewhere [22,39,40], as well as experimental [41] and numerical [42–47] papers published over the past two years, indicate that inherently laminar flamelets can survive and control statistical characteristics of premixed turbulent flames at large  $\tau_F u'/\lambda \gg 1$ , in line with Klimov's standpoint. Williams did not highlight the constraint  $\tau_F u'/\lambda = 0(1)$  in his book [13] either.

While the studies cited above placed the focus of consideration on the straining of local laminar flames by velocity gradients created by small-scale turbulent eddies, such eddies can also perturb the local flame structure by entering the local flames and intensifying mixing inside them. This mechanism was emphasized in the premixed turbulent combustion regime diagrams by Borghi [10,11] and Peters [15]. Under the simplifications invoked to arrive at Equation (1), the "quenching criterion", i.e.,  $\tau_F = \lambda/u'$ , occasionally coincides with the "mixing criterion", i.e.,  $\delta_L = \eta_K$ . Therefore, the two criteria simply read Ka = 1. Accordingly, in the majority of premixed turbulent combustion regime diagrams [8,10,11,15–22], the same line Ka = 1 plays the most important role and is considered to be (i) a boundary of a regime associated with a substantial probability of local flame quenching and (ii) a boundary of a regime associated with local flames broadened by small-scale turbulent eddies. However, in a general case, the two boundaries should be different, as will be discussed in the next section.

In addition, in many recent diagrams, there is another line of Ka = 100. This line was first drawn by Peters [18,48] to highlight the so-called thin reaction zone regime of premixed turbulent combustion. In this regime, the smallest turbulent eddies are sufficiently small to enter the preheat zones of the local flames, i.e.,  $\eta_K < \delta_L$ , but are too large to enter significantly thinner reaction zones of these flames, i.e.,  $\delta_r < \eta_K$ . Here,  $\delta_r$ is the reaction zone thickness, and Peters [18,48] assumed that  $\delta_r = 0.1 \delta_L$ . It is worth remembering that, within the framework of the classical Activation Energy Asymptotics (AEA) theory of laminar premixed flames [49],  $\delta_r \ll \delta_L$  and the flame speed is controlled by a reaction time scale and molecular heat diffusivity in the reaction zone. Accordingly, penetration of small-scale eddies into thicker preheat zones and intensification of mixing inside these zones can change the local flame structure but weakly affect the local burning rate in a turbulent flow, provided that  $\delta_r < \eta_K$  [50]. Thus, the influence of turbulent eddies on local flames can be different in the flamelet and thin reaction zone regimes. In the former regime, the entire local flames retain their laminar structure, whereas preheat zones are broadened in the latter regime, but the local burning rates retain the laminar values. However, while the division of a laminar premixed flame into a thick preheat zone and a thin reaction zone is fully justified within the framework of the classical AEA theory of single-step-chemistry laminar premixed flames [49], such a division could be disputed for certain complex-chemistry flames. Moreover, for the latter flames, an estimate of  $\delta_r =$  $0.1\delta_L$  appears to be too strong. These two points, which are of importance for accurately specifying the upper boundary of the thin reaction zone regime, will be discussed in Section 4.

#### 3. Differently Defined Karlovitz Numbers

Let us, following common practice [2–4], define Kolmogorov length, time, and velocity scales as follows

$$\eta_K = (\nu^3/\varepsilon)^{1/4}, \qquad \tau_K = (\nu/\varepsilon)^{1/2}, \qquad u_K = (\nu\varepsilon)^{1/4}.$$
 (2)

Henceforth, all turbulence characteristics are evaluated in unburned gas upstream of a flame if the opposite is not specified.

In homogeneous isotropic turbulence [51],

$$\varepsilon = C_{\varepsilon} \frac{{u'}^3}{L} = 15\nu \left(\frac{u'}{\lambda}\right)^2. \tag{3}$$

Here,  $C_{\varepsilon}$  is a constant, which is often assumed to be of unity order, but its exact value is not known a priori. In the simplest case of homogeneous isotropic turbulence, an increase in  $C_{\varepsilon}$  with decreasing Reynolds number  $Re_{\lambda} \equiv u'\lambda/\nu$  is well documented in Direct Numerical Simulation (DNS) studies [52,53], with the effect being strongly pronounced at  $Re_{\lambda} < 150$ . These DNS data also indicate that  $C_{\varepsilon}$  tends to a finite constant value  $C_{\varepsilon,\infty}$  (about 0.5) as  $Re_T \to \infty$ , but this asymptotic value depends on "details of forcing at low wavenumbers" [52]. Burattini et al. [54] summarized experimental data on  $C_{\varepsilon}$ , obtained by themselves and by other research groups from different flows under different conditions. Figure 1 in the cited paper shows that  $C_{\varepsilon}$  measured at approximately the same  $Re_{\lambda}$  (in a range of  $100 < Re_{\lambda} < 300$ ) in different flows varies from 0.5 to 2.6, with a dependence of  $C_{\varepsilon}$  on  $Re_{\lambda}$  being either weakly pronounced or even increasing in some of the experiments. Accordingly, Burattini et al. [54] have concluded that "a universal value for " $C_{\varepsilon}$  "is not tenable" and "the flow type and initial conditions (for any given flow type) seem to have a persistent influence even in the fully developed region of the flow". Recently, Vassilicos [55] has argued that in spatially decaying turbulent flows (e.g., flows behind various grids or wakes), there exists a significant nonequilibrium region, where  $C_{\varepsilon}$  is roughly proportional to a ratio of an inlet Reynolds number, which is constant for each specific flow, to the local turbulent Reynolds number, which could vary as the turbulence decays. Accordingly, data measured at different distances from a grid or wake could show an increase in  $C_{\varepsilon}$  with a decreasing turbulent Reynolds number.

In the combustion literature, the values of  $C_{\varepsilon}$  are even more scattered because different methods are adopted to evaluate the length scale *L*, with the chosen method being poorly described in many papers. For instance, if the length scale *L* is evaluated as follows  $L = u'^3/\varepsilon$ , the constant  $C_{\varepsilon} = 1$ . This value of  $C_{\varepsilon}$  was adopted in some experimental [56–59] and DNS [60–65] studies. If  $L = k^{3/2}/\varepsilon$  [51,66,67], where k = $\mathbf{u' \cdot u'}/2$  designates the mean value of turbulent kinetic energy, then  $C_{\varepsilon} = (3/2)^{3/2} =$ 1.83. If *L* designates longitudinal or transverse integral length scale, the "constant"  $C_{\varepsilon}$  is poorly known and is not constant [51–55]. By referring to the results of their measurements, Abdel-Gayed et al. [38] have set  $(\lambda/L)^2 = 40.4Re_T^{-1}$ , which is equivalent to  $C_{\varepsilon} = 0.37$  in Eq. (3). In commercial CFD codes, the default value of  $C_{\varepsilon}$  is commonly equal to  $(3/2)^{3/2}C_{\mu}^{3/4}$ , i.e.,  $C_{\varepsilon} = 0.30$  if  $C_{\mu} = 0.09$ . Steinberg et al. [27] recommended  $C_K = 0.5$  in  $L/\eta_K = C_K Re_T^{3/4}$ . Accordingly,  $C_{\varepsilon} = C_K^4 = 0.0625$ , because  $L/\eta_K =$  $Lv^{-3/4}\varepsilon^{1/4} = C_{\varepsilon}^{1/4}Re_T^{3/4}$ . These examples show that the values of  $C_{\varepsilon}$  can differ by a factor as large as 30. In an anisotropic turbulent flow associated with a typical burner, evaluation of  $C_{\varepsilon}$  is even more difficult.

Using Equations (2) and (3), the following relations could be obtained

$$\eta_{K} = LC_{\varepsilon}^{-1/4} Re_{T}^{-3/4}, \qquad \tau_{K} = \tau_{T} C_{\varepsilon}^{-1/2} Re_{T}^{-1/2} = \frac{1}{\sqrt{15}} \frac{\lambda}{u'}, \qquad u_{K} = u' C_{\varepsilon}^{1/4} Re_{T}^{-1/4}. \tag{4}$$

Subsequently, substituting Equations (3) and (4) into Equation (1) and introducing a flame-thickness factor (or a flame counterpart of Reynolds number)

$$\Gamma_F \equiv \frac{\delta_L S_L}{\nu},\tag{5}$$

we arrive at

$$Ka_{0} = \frac{\tau_{F}}{\tau_{K}} = \sqrt{15} \frac{\tau_{F}u'}{\lambda} = \frac{1}{\Gamma_{F}} \left(\frac{\delta_{L}}{\eta_{K}}\right)^{2} = \Gamma_{F} \left(\frac{u_{K}}{S_{L}}\right)^{2} = \sqrt{C_{\varepsilon}} \Gamma_{F} \left(\frac{u'}{S_{L}}\right)^{3/2} \left(\frac{L}{\delta_{L}}\right)^{-1/2} = \sqrt{\Gamma_{F}} \left(\frac{\varepsilon\delta_{L}}{S_{L}^{3}}\right)^{1/2}$$
$$= \Gamma_{F} \sqrt{C_{\varepsilon}} \left(\frac{u'}{S_{L}}\right)^{2} Re_{T}^{-1/2} = \sqrt{C_{\varepsilon}} \frac{Re_{T}^{1/2}}{Da}.$$
(6)

Various parts of Equation (6) were earlier adopted to calculate a Karlovitz number. For instance, Abdel-Gayed et al. [38] have introduced a Karlovitz number as follows

$$Ka_{1} = \frac{\tau_{F}u'}{\lambda} = 0.157 \left(\frac{u'}{S_{L}}\right)^{2} Re_{T}^{-1/2}$$
(7)

by setting  $C_{\varepsilon} = 0.37$  in Equation (4) and assuming that  $\Gamma_F = 1$ . Since that seminal work, Equation (7) was adopted in many papers. In certain papers, the second equality in Equation (7) was not invoked, and the Karlovitz number was evaluated using measured values of rms turbulent velocity and Taylor microscale [56,68].

In other experimental papers, e.g., see Ref. [57], these measured values were adopted to calculate a differently defined Karlovitz number

$$Ka_2 = \sqrt{15} \frac{\tau_F u'}{\lambda}.$$
(8)

The same Karlovitz number written in the form of  $Ka_2 = \tau_F/\tau_K = \tau_F(\nu/\varepsilon)^{-1/2}$  is sometimes used in DNS papers, e.g., see Ref. [66], where the dissipation rate  $\varepsilon$  is directly evaluated.

Conditions of experiments are often characterized by setting both  $C_{\varepsilon} = 1$  and  $\Gamma_F = 1$  [69–71], i.e., by invoking the following number

$$Ka_3 = \left(\frac{u'}{S_L}\right)^{3/2} \left(\frac{L}{\delta_L}\right)^{-1/2}.$$
(9)

The same number is widely used in DNS [63,72–74] and review [22,39] articles also. This number may be interpreted to characterize a ratio of the dissipation rate  $\varepsilon$  to its flame counterpart  $\varepsilon_F = C_{\varepsilon} S_L^3 / \delta_L$ , i.e.,  $Ka_3 = \sqrt{\varepsilon/\varepsilon_F}$ .

In other DNS papers [61,65,75,76], the factor  $\sqrt{\Gamma_F}$  is retained, i.e., the DNS conditions are characterized with

$$Ka_4 = \sqrt{\Gamma_F} \left(\frac{u'}{S_L}\right)^{3/2} \left(\frac{L}{\delta_L}\right)^{-1/2}.$$
 (10)

Finally, many authors evaluate one more Karlovitz number using a ratio of a laminar flame thickness and Kolmogorov length scale [58,77–82], i.e.,

$$Ka_5 = \left(\frac{\delta_L}{\eta_K}\right)^2. \tag{11}$$

The above brief review shows that several differently defined Karlovitz numbers are adopted in the literature. It is worth stressing that this ambiguity is even greater because differently defined laminar flame thicknesses could be substituted in each of Equations (7)–(11), e.g., (i)  $\delta_L = v_u/S_L$ , (ii)  $\delta_L = \kappa_u/S_L$  and  $\Gamma_F$  is inversely proportional to Prandtl number  $Pr = v/\kappa$  in this case, (iii)  $\delta_L = D_u/S_L$ , where *D* designates molecular diffusivity of deficient reactant, and  $\Gamma_F$  is inversely proportional to Schmidt number  $Sc = D/\kappa$  in this case, (iv) each of the aforementioned molecular transport coefficients could be taken at an intermediate temperature, etc.

Based on the historical overview provided in Section 2, the use of the ratio  $\tau_F u'/\lambda$  as a Karlovitz number appears to be the most appropriate choice. While application of Equations (7) and (9) or (10) seems to be easier, the Taylor microscale can be evaluated in the state-of-the-art measurements and DNSs either directly or indirectly (e.g.,  $\lambda = u'\sqrt{15v/\varepsilon}$ , where the dissipation rate is either measured or sampled from DNS data). In the following,

$$Ka \equiv \frac{\tau_F u'}{\lambda} \tag{12}$$

and Equations (6) and (8)–(11) read

$$Ka = \frac{1}{\sqrt{15}} Ka_2 = \sqrt{\frac{C_{\varepsilon}\Gamma_F}{15}} Ka_3 = \sqrt{\frac{C_{\varepsilon}}{15}} Ka_4 = \frac{1}{\Gamma_F\sqrt{15}} Ka_5.$$
(13)

It is also worth noting that (i) DNS data by Girimaji et al. [83,84] show that the mean strain rate in a turbulent flow is about  $0.28\tau_K^{-1} \approx 1/\sqrt{15}\tau_K^{-1}$ , in line with the above relation between *Ka* and *Ka*<sub>2</sub>, and (ii) difference in *Ka* and *Ka*<sub>4</sub> can reach an order of magnitude depending on the value of  $C_{\varepsilon}$ .

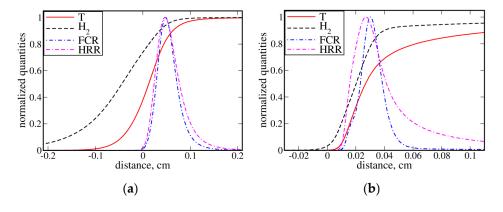
The use of the term "Karlovitz number", i.e.,  $Ka_5$ , defined by Equation (11), as a criterion of penetration of the smallest-scale turbulent eddies into flame preheat zones does not seem to be historically justified because Karlovitz did not highlight this physical mechanism. Moreover, the criteria of Ka = 1 and  $Ka_5 = 1$  are different in a general case (see Equation (13)), where  $\Gamma_F$  can be large, as discussed in the next section. Therefore, another name for a ratio of  $\delta_L/\eta_K$  as a boundary of broadened preheat zone regime appears to be more appropriate both from the historical and fundamental perspectives.

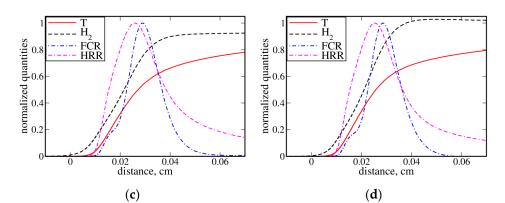
The same comments hold for a criterion of  $\delta_r = \eta_K$ , introduced by Peters [18,48]. Relation between the number  $\delta_r/\eta_K$  or  $(\delta_r/\eta_K)^2$  and *Ka* or *Ka*<sub>5</sub> depends on a ratio of  $\delta_r/\delta_L$ , which, in its turn, depends on mixture composition, pressure, and temperature, as discussed in the next section.

#### 4. Preheat and Reaction Zone Thicknesses of Complex-Chemistry Flames

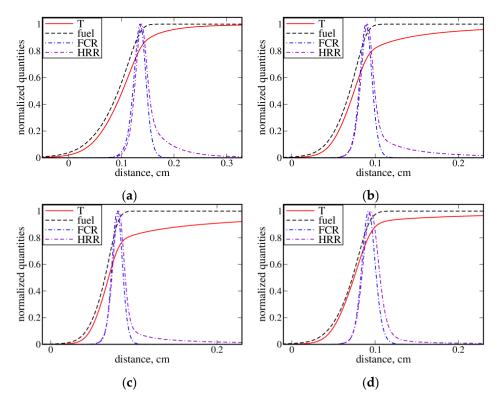
To evaluate  $\Gamma_F$  and  $\delta_r/\delta_L$ , numerical simulations of unperturbed, complexchemistry, adiabatic, laminar premixed flames were performed by running PREMIX code [85] of CHEMKIN-II software package [86] to numerically integrate stationary, onedimensional transport equations for species mass fractions and energy, supplemented with the ideal gas state equation and the continuity equation. Thermo-diffusion and multi-species diffusion options were activated for H<sub>2</sub>, CH<sub>4</sub>, and C<sub>3</sub>H<sub>8</sub>. For methane, the GRI mechanism [87] (53 species and 325 reversible reactions) was adopted. Hydrogen, propane, and *n*-heptane–air flames were simulated, invoking chemical mechanisms by Konnov [88] (15 species and 75 reversible reactions), Chaos et al. [89] (117 species and 755 reversible reactions), and Huang et al. [90] (114 species and 632 reversible reactions), respectively.

Figures 1–4 show profiles of the temperature-based combustion progress variable  $c_T = (T - T_u)/(T_b - T_u)$  (solid red lines), fuel-based combustion progress variable  $c_F = (X_F - X_{F,u})/(X_{F,b} - X_{F,u})$  (black dashed lines), fuel consumption rate (FCR) (blue dotted-dashed lines), and heat release rate (HRR) (magenta or violet double-dashed-dotted lines) obtained from complex-chemistry lean, stoichiometric, and rich hydrogen–air (Figure 1), methane–air (Figure 2), propane–air (Figure 3), and *n*-heptane–air (Figure 4) unperturbed laminar flames under room conditions. Here,  $X_F$  designates fuel mole fraction, subscripts *u* and *b* refer to unburned reactants and burned products, respectively, and the rates are normalized using their peak values in the flame. Differently defined flame thicknesses obtained using these profiles are reported in Figure 5.

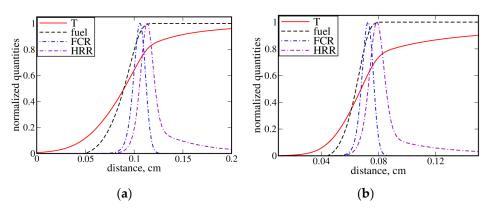


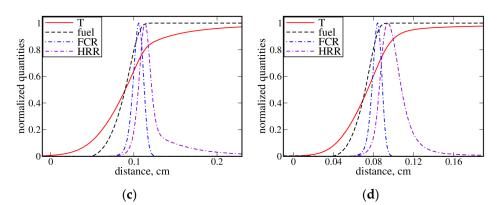


**Figure 1.** Profiles of  $c_T$ ,  $c_F$ , fuel consumption rate, and heat release rate, normalized using their peak values. The profiles have been obtained from unperturbed hydrogen–air premixed flames characterized by different equivalence ratios: (**a**)  $\phi = 0.35$ , (**b**)  $\phi = 0.8$ , (**c**)  $\phi = 1.0$ , and (**d**)  $\phi = 1.3$ .

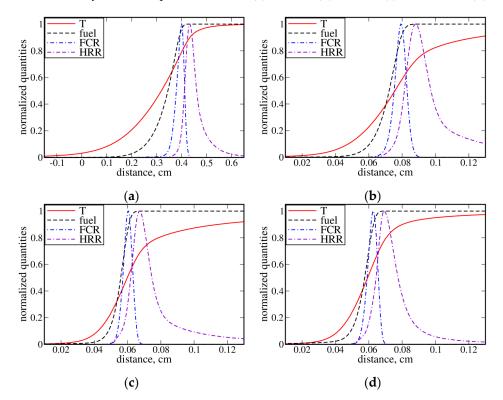


**Figure 2.** Profiles of  $c_T$ ,  $c_F$ , fuel consumption rate, and heat release rate, normalized using their peak values. The profiles have been obtained from unperturbed methane–air premixed flames characterized by different equivalence ratios: (**a**)  $\phi = 0.6$ , (**b**)  $\phi = 0.8$ , (**c**)  $\phi = 1.0$ , and (**d**)  $\phi = 1.25$ .

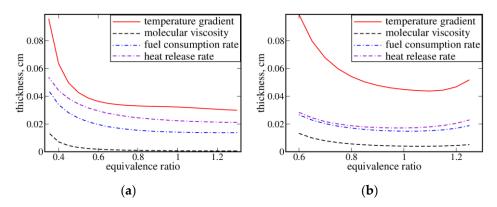


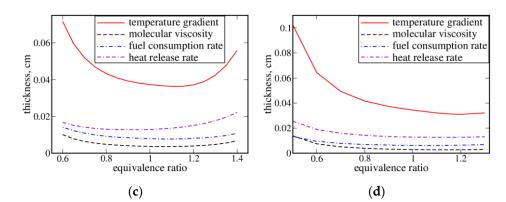


**Figure 3.** Profiles of  $c_T$ ,  $c_F$ , fuel consumption rate, and heat release rate, normalized using their peak values. The profiles have been obtained from unperturbed propane–air premixed flames characterized by different equivalence ratios: (a)  $\phi = 0.6$ , (b)  $\phi = 0.8$ , (c)  $\phi = 1.0$ , and (d)  $\phi = 1.4$ .



**Figure 4.** Profiles of  $c_T$ ,  $c_F$ , fuel consumption rate, and heat release rate, normalized using their peak values. The profiles have been obtained from unperturbed *n*-heptane–air premixed flames characterized by different equivalence ratios: (**a**)  $\phi = 0.4$ , (**b**)  $\phi = 0.7$ , (**c**)  $\phi = 1.0$ , and (**d**)  $\phi = 1.3$ .





**Figure 5.** Dependencies of differently defined laminar flame thicknesses on the equivalence ratio, computed for (a) H<sub>2</sub>-air, (b) CH<sub>4</sub>-air, (c) C<sub>3</sub>H<sub>8</sub>-air, and (d) n-C<sub>7</sub>H<sub>16</sub>-air mixtures under room conditions.

In all studied flames (maybe, with the exception of the leanest *n*-heptane–air flame, see Figure 4a), the laminar flame thickness is controlled by the maximal temperature gradient and evaluated as follows

$$\delta_L = \frac{T_b - T_u}{\max|dT/dx|} \tag{14}$$

seems to be quite comparable with the reaction zone thickness  $\delta_r^F$  or  $\delta_r^T$  equal to the halfwidth of the fuel consumption rate profile or the heat release rate profile, respectively. In the H<sub>2</sub>–air mixtures, with the exception of the leanest one, see Figure 1a, the thickness  $\delta_L$  seems to be close to  $\delta_r^T$ , cf. curves plotted in red solid and magenta double-dashed-dotted lines in Figure 1b–d. Moreover, in these flames, there are wide radical recombination zones, where the temperature grows gradually due to heat release in slow three-molecular reactions between radicals [91]. Thus, in these three moderately lean, stoichiometric, and moderately rich H<sub>2</sub>–air flames, reaction zones are not significantly thinner than other (preheat and radical recombination) zones, and the separation of broadened preheat zone and broadened reaction zone regimes of premixed turbulent combustion does not seem to be fundamentally justified. Therefore, a criterion of  $\eta_K = 0.1\delta_L$ , introduced by Peters [18,48] as a boundary of thin reaction zone regime, appears to be irrelevant to moderately lean, stoichiometric, and moderately rich H<sub>2</sub>–air flames.

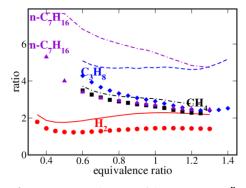
In the leanest H<sub>2</sub>-air flame, see Figure 1a, fuel consumption and heat release zones are a little thinner than preheat zone. While difference in  $\delta_r^F$  or  $\delta_r^T$  and a laminar flame thickness is substantially increased if the latter thickness is evaluated as follows

$$\delta_L^F = \frac{Y_{F,u} - Y_{F,b}}{\max\left| dY_{F,u}/dx \right|'} \tag{15}$$

this measure of laminar flame thickness is seldom used in the literature, contrary to Equation (14). Here,  $Y_F$  designates fuel mass fraction and  $\delta_L^F$  is significantly larger than the thickness  $\delta_L$  defined by Equation (14) because the molecular diffusivity of hydrogen is much larger than the molecular heat diffusivity in a lean H<sub>2</sub>–air mixture.

In hydrocarbon–air flames, see Figures 2–4, the thickness  $\delta_L$  is distinctly larger than  $\delta_r^F$  or  $\delta_r^T$ , but the difference seems to be really large in the leanest *n*-heptane–air flame only, see Figure 4a. Note that the difference is significantly reduced if the thickness of propane–air or *n*-heptane–air flame is quantified with  $\delta_L^F$  defined by Equation (15).

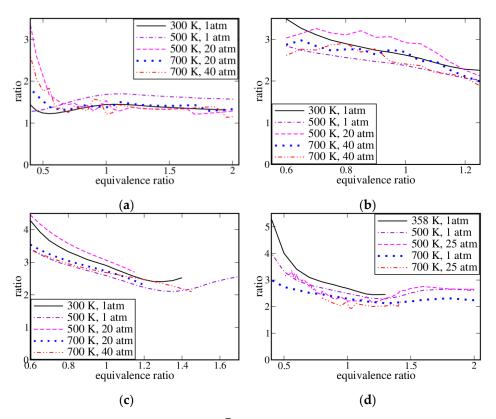
Figure 5 further emphasizes that in hydrogen–air or paraffin–air complex-chemistry laminar premixed flames under room conditions, differences between thicknesses of preheating zones, see solid red lines, and reaction zones, see blue and violet dotteddashed lines, are sufficiently small. As shown in Figure 6, such differences are substantially less than an order of magnitude. More specifically,  $\delta_r^T$  is close to  $\delta_L$  in the studied H<sub>2</sub>–air mixtures, see red circles, with a ratio of  $\delta_L/\delta_r^T$  (i) being slowly increased with decreasing the equivalence ratio and (ii) reaching two at  $\phi = 0.35$ . In the studied paraffin–air flames, the ratio  $\delta_L/\delta_r^T$  is slightly above two in rich mixtures, increases moderately with decreasing the equivalence ratio, is close to four in the richest methane–air and propane-flames and is about 5.5 in the richest *n*-heptane–air flame.



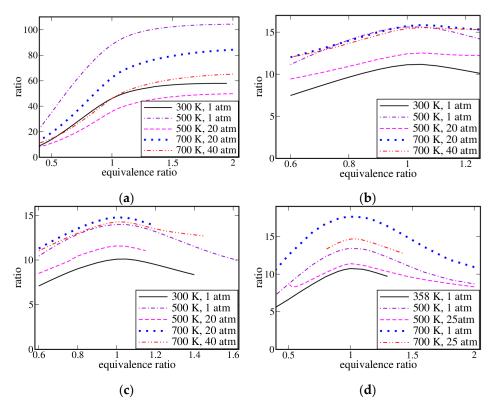
**Figure 6.** Dependencies of the ratios  $\delta_L/\delta_r^F$  (lines) and  $\delta_L/\delta_r^T$  (symbols) calculated for various fuels specified in legends.

All in all, Figures 1–6 do show that a factor of 0.1 in the criterion  $\eta_K = 0.1\delta_L$ , considered to demarcate thin reaction zone regime [18,48] on a premixed turbulent combustion regime diagram, is too small for complex-chemistry flames. Moreover, for the vast majority of the studied mixtures,  $\delta_L/\delta_r^T < 4$ , thus, putting into question the foundation of the concept of a thin reaction zone (especially for hydrogen-air flames, where  $\delta_L/\delta_r^T < 2$ ). Under room conditions, such a concept appears to deserve consideration for very lean mixtures of heavy paraffin and air only. One may note that  $\delta_L/\delta_r^T < \delta_L/\delta_r^F$ , cf. symbols and lines in Figure 6, because a fuel consumption rate zone is thinner than a heat release zone, cf. curves plotted in blue and violet dotted-dashed lines, respectively, in Figure 5. However, penetration of small-scale turbulent eddies into a thicker heat release zone is sufficient to claim that the boundary of the discussed regime is crossed. Therefore, the boundaries of regimes of broadened preheat zones and broadened reaction zones are so close that the separation of the two regimes does not seem to be fundamentally justified under room conditions. Figure 7 and similar results obtained for  $\delta_L/\delta_r^F$  (not shown for brevity) indicate that this separation is not fundamentally justified under elevated pressures and temperatures either.

Dependencies of the factor  $\Gamma_F$  on the equivalence ratio, calculated for various fuels under different pressures and unburned gas temperatures are reported in Figure 8. These numerical results indicate that the factor  $\Gamma_F$  (i) is significantly larger than unity, (ii) is very large for near stoichiometric and moderately rich hydrogen-air mixtures, (iii) depends on the equivalence ratio, (iv) is increased by the unburned gas temperature, and (v) is decreased with increasing pressure. Therefore, the number  $Ka_5$ , defined by Equation (11) and widely used as a criterion of penetration of the smallest-scale turbulent eddies into flame preheat zones, is not equal to the Karlovitz number Ka, defined by Equation (12) and adopted as a criterion of local combustion quenching. Consequently, if the influence of complex combustion chemistry on the laminar flame thickness  $\delta_L$  is taken into account, the use of a single line of Ka = 1 as a boundary of (a) broadened preheat zone regime, and (b) regime associated with local combustion quenching is not justified. In other words, the single line Ka = 1, drawn in many combustion regime diagrams, should be split into two different lines. As  $\Gamma_F$  is significantly larger than unity, the boundary of the former (broadened preheat zone regime) is associated with less intense turbulence (a lower Karlovitz number) when compared to the boundary of the latter regime, see Equation (13). Moreover, distance (in a 2D combustion regime diagram) between the boundaries of the two regimes should depend on fuel formula, equivalence ratio, unburned gas temperature, and pressure, which affect the factor  $\Gamma_{F}$ .



**Figure 7.** Dependencies of the ratio  $\delta_L/\delta_r^T$  on the equivalence ratio, calculated for (**a**) H<sub>2</sub>, (**b**) CH<sub>4</sub>, (**c**) C<sub>3</sub>H<sub>8</sub>, and (**d**) *n*-C7H16, calculated for various pressures *P* and unburned gas temperatures  $T_u$ , specified in legends.



**Figure 8.** Dependencies of the factor  $\Gamma_F$  on the equivalence ratio, calculated for (**a**) H<sub>2</sub>, (**b**) CH<sub>4</sub>, (**c**) C<sub>3</sub>H<sub>8</sub>, and (**d**) *n*-C7H16, calculated for various pressures *P* and unburned gas temperatures  $T_u$ , specified in legends.

#### 5. Concluding Remarks

First, while the numbers (i)  $\tau_F u'/\lambda$  or (ii)  $(\delta_L/\eta)^2$ , which are widely adopted to demarcate (i) a combustion regime associated with the importance of local flame quenching by small-scale turbulent eddies or (ii) a combustion regime associated with broadening of preheating zones by small-scale turbulent eddies, respectively, are both proportional to the same *Ka*, the critical values of the two numbers, used to demarcate the combustion regimes, are significantly different in complex-chemistry flames. More specifically, a constraint of  $(\delta_L/\eta)^2 = 1$  should be reached at a substantially lower *Ka* when compared to a constraint of  $\tau_F u'/\lambda = 1$ . Moreover, a ratio of the two numbers,  $(\delta_L/\eta)^2$  and  $\tau_F u'/\lambda$ , depends on fuel formula, equivalence ratio, unburned gas temperature, and pressure. In particular, the ratio of these numbers can be very large in moderately lean, stoichiometric, or rich hydrogen–air mixtures, especially at elevated unburned gas temperatures.

The use of Kolmogorov time and length scales in the discussed constraints can be disputed by noting that Kolmogorov eddies rapidly disappear and, consequently, cannot substantially affect a premixed flame during the eddy lifetime. For instance, following Klimov [32], the criterion of  $\tau_F u'/\lambda = 1$  could be changed to  $\tau_F u'/\lambda = Ka_{cr} \gg 1$ . Moreover, based on results of numerical [92,93] and experimental [94] studies of vortex filaments (worms or tubes) in incompressible turbulence, the smallest eddy length scale was argued to be larger than  $\eta$  by a factor of about 8. We may also note that experimental data analyzed by Monin and Yaglom ([2], Figure 77) show that the highest rate of dissipation of turbulent energy is also localized at a length scale of about  $8\eta$ . If this smallest length scale, which was already used in combustion research [27,95], is compared with  $\delta_L$ , then, the classical criterion of  $(\delta_L/\eta)^2 = 1$ , should be changed to  $(\delta_L/\eta)^2 = \Delta_{cr}$ , where  $\Delta_{cr}$  can be as large as 64. These simple reasoning could explain the utility of the flamelet paradigm even at high Karlovitz numbers, which (utility) was emphasized in recent review articles [22,39,40], as well as in subsequent experimental [41] and numerical [42–47] papers. In any case, criteria of local flame broadening, i.e.,  $(\delta_L/\eta)^2 = \Delta_{cr}$ , and local flame quenching, i.e.,  $\tau_F u'/\lambda = Ka_{cr}$ , should be different in a general case.

Second, numerical simulations of complex chemistry laminar premixed flames do not warrant separation of a thick preheat zone and a much thinner reaction zone, especially in moderately lean, stoichiometric, and rich hydrogen-air mixtures. Such separation could be acceptable for lean mixtures of heavy paraffin with air, but the ratio of the two thicknesses is significantly less than 10, even in this case. Nevertheless, recent experimental and numerical studies reviewed elsewhere [22,39,40] do show that the reaction zone can retain its (laminar flame) thickness even if turbulence is sufficiently intense to significantly broaden preheating zones. This apparent inconsistency (comparable thicknesses of preheating and reaction zones in laminar flames and wellpronounced broadening of the former zone in certain turbulent flames) could be attributed to the rapid disappearance of the smallest eddies in thick preheat zones due to thermal expansion and a significant increase in the mixture viscosity with the temperature, e.g., see Ref. [96]. DNS data by Bobbitt et al. [97] and by Apsden [98] do indicate that an increase in  $\nu$  with the temperature significantly affects the evolution of enstrophy  $(\nabla \times \mathbf{u})^2$  in premixed flames. While recent experiments by Wabel et al. [99] did not show substantial variations in the local turbulent kinetic energy within broadened preheat zones, an increase in the length scale of turbulent eddies conditioned to such zones, was reported in the cited paper, thus, implying a decrease in the local turbulent strain rates. While the discussed hypothesis definitely requires further assessment, it is worth stressing already now that, if confirmed, the hypothesis challenges the utility of combustion regime diagrams that do not allow for the influence of combustion on turbulence. Such an influence has yet been addressed in a few diagrams either by considering [19,20,100] hydrodynamic instability of laminar premixed flames [49,101] or by parameterizing results [17] of 2D DNS of the interaction of premixed flames with a vortex pair [102].

Third, the above discussion was restricted to combustion regime boundaries given by a constant Karlovitz number. Other criteria have also been proposed to demarcate regimes of premixed turbulent combustion. For instance, by analyzing recent experimental data, Driscoll et al. [22,103] argued that preheat zones are broadened by turbulent eddies if  $Re_T > 2\,800$ , but we are not aware of any support for this criterion from the fundamental perspective. Governing physical mechanisms and regimes of highly turbulent combustion characterized by large Karlovitz numbers and small Damköhler numbers are still poorly understood, as reviewed elsewhere [21,22,27], and further research in this direction is definitely required.

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