Aalborg Universitet



## Impact of accelerometer calibration modelling on GRACE Follow-On precise orbit determination and intersatellite ranging

Papanikolaou, Thomas: Tsoulis, Dimitrios

Publication date: 2021

Document Version Publisher's PDF, also known as Version of record

Link to publication from Aalborg University

Citation for published version (APA):

Papanikolaou, T., & Tsoulis, D. (2021). Impact of accelerometer calibration modelling on GRACE Follow-On precise orbit determination and intersatellite ranging. Abstract from International Association of Geodesy (IAG) Scientific Assembly, Beijing, China.

https://www.researchgate.net/publication/352840826\_Impact\_of\_accelerometer\_calibration\_modelling\_on\_GRA CE\_Follow-On\_precise\_orbit\_determination\_and\_intersatellite\_ranging

### General rights

Copyright and moral rights for the publications made accessible in the public portal are retained by the authors and/or other copyright owners and it is a condition of accessing publications that users recognise and abide by the legal requirements associated with these rights.

- Users may download and print one copy of any publication from the public portal for the purpose of private study or research.
- You may not further distribute the material or use it for any profit-making activity or commercial gain
  You may freely distribute the URL identifying the publication in the public portal -

Take down policy If you believe that this document breaches copyright please contact us at vbn@aub.aau.dk providing details, and we will remove access to the work immediately and investigate your claim.

See discussions, stats, and author profiles for this publication at: https://www.researchgate.net/publication/352840826

Dynamic orbit analysis of satellite gravity missions View project

# Impact of accelerometer calibration modelling on GRACE Follow-On precise orbit determination and intersatellite ranging

Poster · June 2021

CITATIONS 0 69 0 2 authors, including: 2 authors, including: Thomas Papanikolaou Aalborg University 19 PUBLICATIONS 46 CITATIONS SEE PROFILE Some of the authors of this publication are also working on these related projects: Gravity and Geoid Modelling View project

All content following this page was uploaded by Thomas Papanikolaou on 30 June 2021.

## IAG 2002 Beijing June 28-July 2 2021

# **Geodesy for a sustainable Earth Scientific Assembly of the International Association of Geodesy**



## Impact of accelerometer calibration modelling on GRACE Follow-On precise orbit determination and intersatellite ranging

Thomas Papanikolaou<sup>1</sup>, Dimitrios Tsoulis <sup>2</sup>

<sup>1</sup>Department of Planning, Aalborg University, Copenhagen, Denmark (thomasp@plan.aau.dk) <sup>2</sup>Department of Geodesy and Surveying, Aristotle University of Thessaloniki, Greece

Precise orbit determination is a major objective in satellite geodesy and data analysis of several geoscientific satellite missions. Satellite gravity missions such as the Gravity Recovery And Climate Experiment (GRACE) missions (GRACE-FO and GRACE) are equipped with on-board accelerometers that form a key observation instrument for the measurement of nongravitational perturbations at orbital altitude. The accelerometers calibration through a data processing scheme is essential for GRACE applications such as precise orbit determination, gravity field mapping and non-gravitational forces modelling.

The present study focuses on the estimation of the accelerometer calibration parameters within an orbit determination approach. We apply an adapted dynamic orbit determination algorithm with extended variational equations. The orbit parameter estimation considers accelerometry calibration parameters such as bias, drift and scale factors in combination with empirical forces of cycle-per-revolution (CPR) terms. The applied approach leads to orbit residuals within 2 to 4 cm (RMS) while the LRI and KBR rangerate data residuals vary within a few  $\mu$ m/sec (RMS: GRACE-FO 1.7, GRACE 1.4  $\mu$ m/sec).

GRACE-FO/GRACE Orbit Determination and Accelerometer calibration modelling	
Orbit arc length / Date	1 day   18/7/2019 - 17/11/2009
Earth Rotation	IERS Conventions 2010
EOP	IERS 08 C04
Numerical Integrator	Gauss-Jackson 12 <sup>th</sup> order; RKN7(6)-8 start integrator
Integration step	2 sec / 5 sec
Pseudo-Observations	Kinematic Orbit XYZ (Suesser-Rechberger et al. 2020)
Gravity Model (d/o)	GOCO06s (Kvas et al. 2019)
Planetary Ephemeris	DE423
Solid Earth Tides	IERS Conventions 2010
Ocean Tides	FES2004
Relativistic effects	IERS Conventions 2010
GRACE-FO Accelerometers	ACC1B, Full Scale matrix (9 parameters), Bias (XYZ), Bias drift (XYZ)
GRACE Accelerometers	ACC1B, Diagonal Scale matrix (Sx,Sy,Sz), Bias (XYZ)
Empirical Forces (GRACE-FO)	1-CPR (along & cross-track), Bias-along
Intersatellite range-rate data	K-band ranging KBR1B & Laser Ranging Interferometry LRI1B





