

Twin Ion Engine Demonstration for Small Spacecraft Applications

SSC22-X-04

Michael Tsay, Riley Terhaar, Kyle Emmi and Carl Barcroft

Busek Co. Inc., Natick, MA, 01760 USA

36th Small Satellite Conference
Utah State University, Logan, UT, USA
August 6-11, 2022



APPROVED FOR DISTRIBUTION

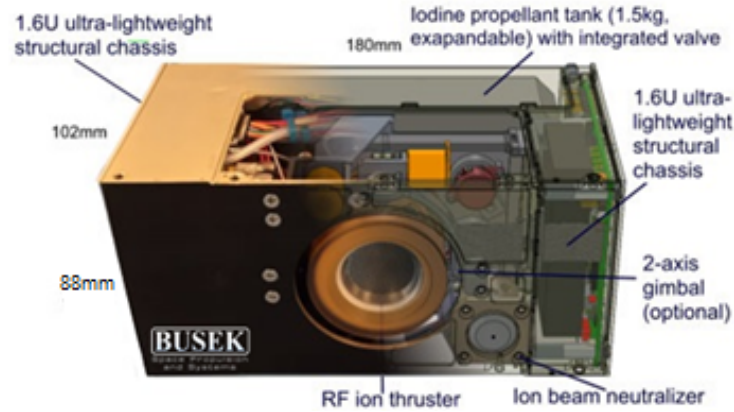
Copyright © by Busek Co. Inc.



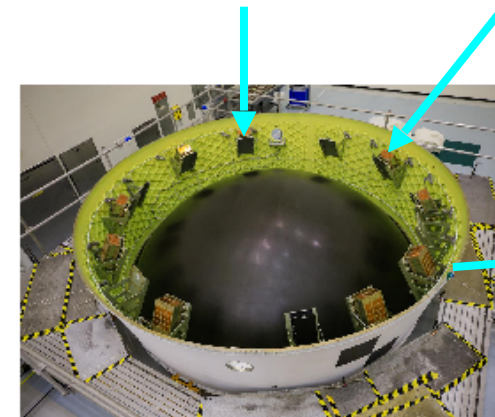
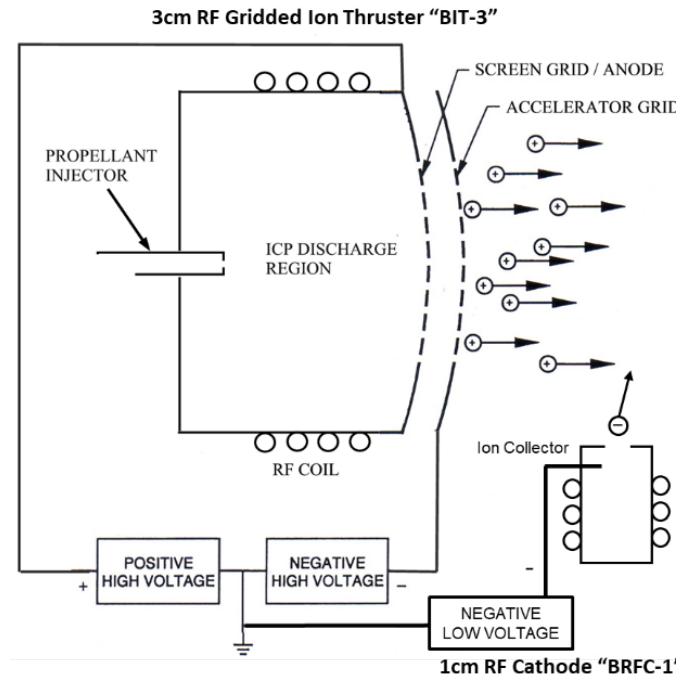
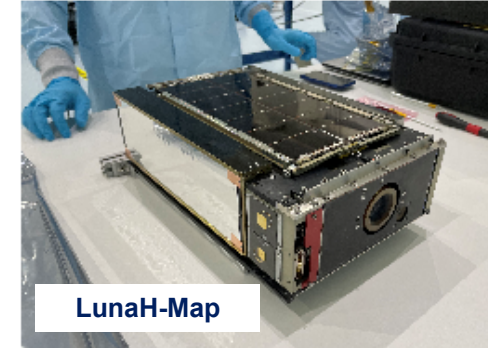


BIT-3 Overview and Gen-1 Upcoming Flights

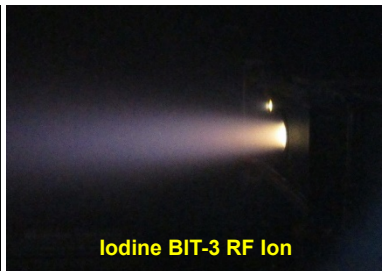
- BIT-3 is an RF gridded ion thruster with an RF cathode neutralizer, both fueled by iodine
- Iodine has many advantages over xenon while equal in performance
 - Solid storable; simpler tank requirement
 - 4.9 g/cc I₂ storage density vs. 1.8-2.0 g/cc Xe
 - <\$100/kg I₂ cost vs. >\$27,000/kg Xe (at least 300X difference)
- Busek one of pioneers on iodine EP
 - I₂ HET
 - I₂ RF gridded ion
 - I₂ RF cathode



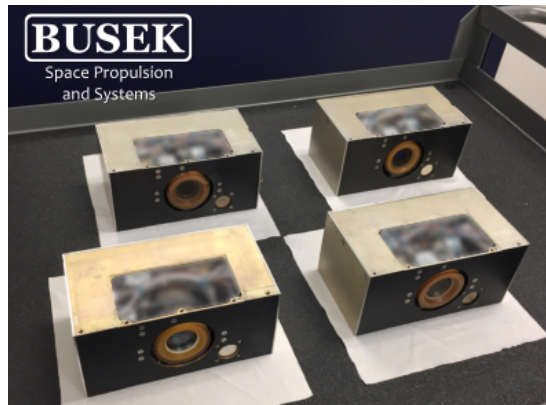
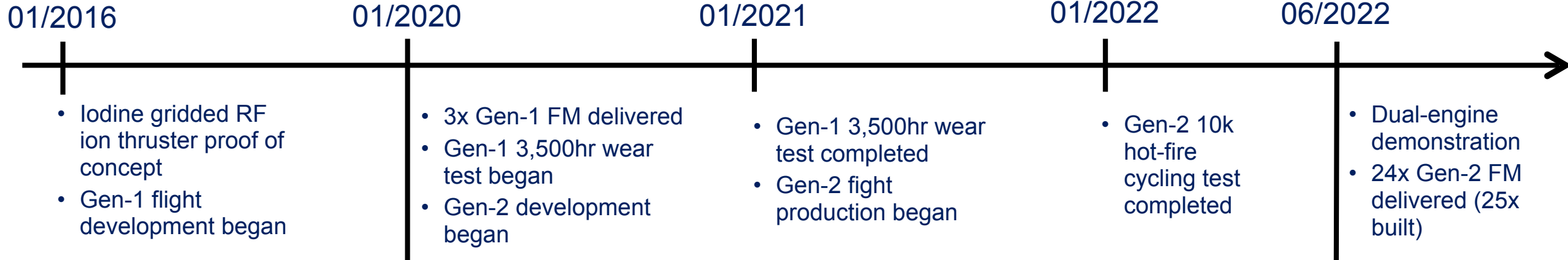
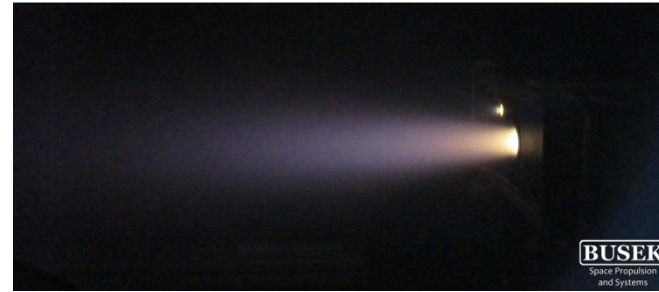
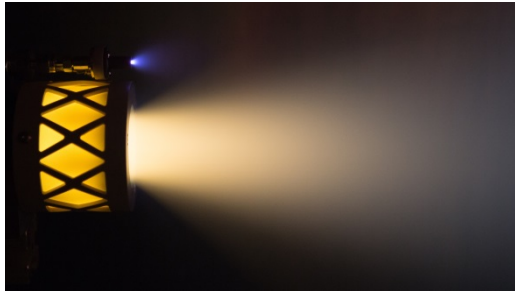
Two Gen-1 BIT-3 on SLS Artemis 1



SLS Artemis 1's Ten 6U CubeSat Secondary Payloads Installed in the Orion Stage Adapter (OSA)



BIT-3 Development Timeline



Gen-1 Units



Gen-2 Fleet

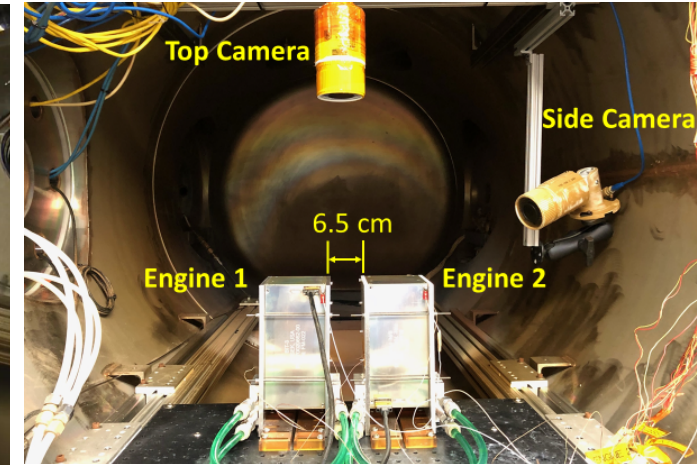
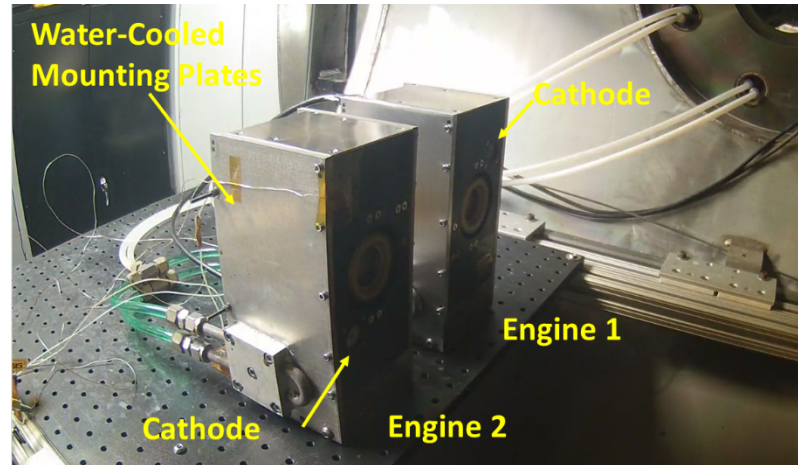
Dual Engine Test Objectives and Setup

Objectives

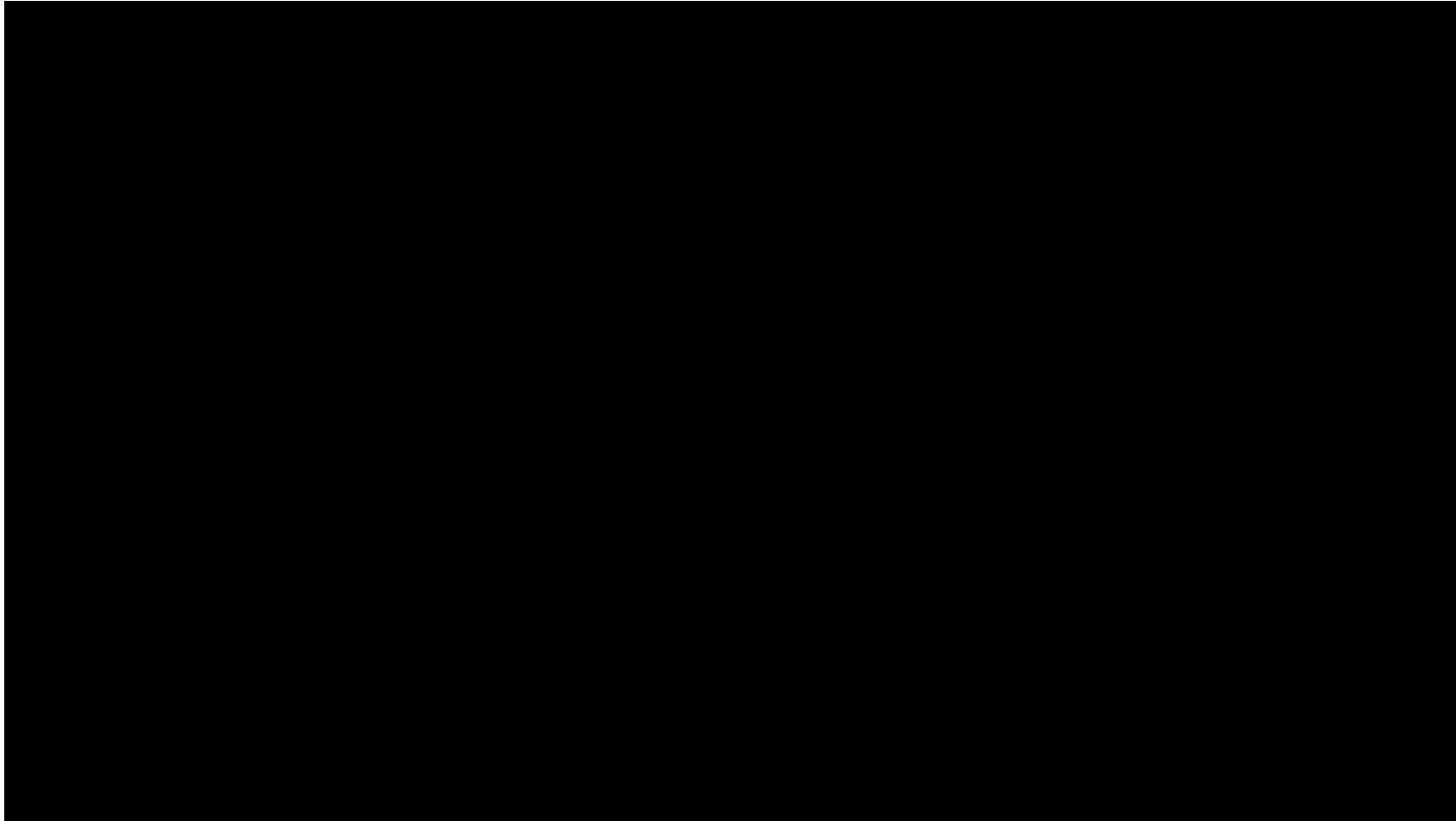
- Demonstrate two BIT-3s can fire in close proximity (<10cm separation), simulating micro-sat volume constraints
 - Operating gridded ion clusters is not new (e.g. JAXA's Hayabusa mission), but never so close
- Demonstrate there is no electrical or plasma interferences during thruster startup and throttling (3 scenarios tested)
- Demonstrate ion beam neutralization
 - Total neutralizer electron emission current equal or higher than ion beam current

Setup

- Water-cooled mounting plates, simulating spacecraft mounting interface
- Two sets of bench power supplies and LabView/RS-485 controls
- Busek T-4 vacuum chamber, dedicated to BIT-3 iodine hot fire tests



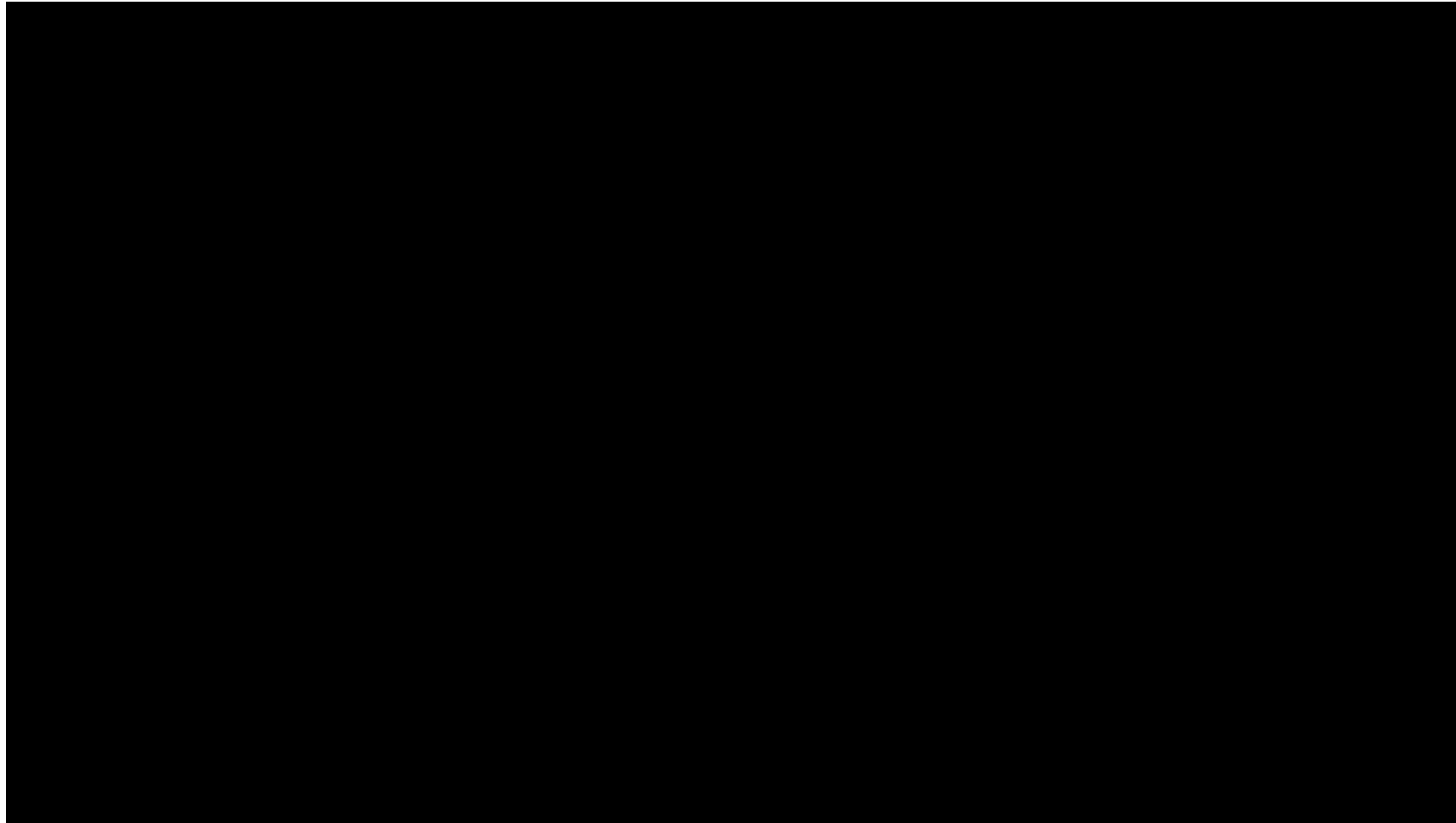
Result – Simultaneous Startup



Video Shown at 4x Speed



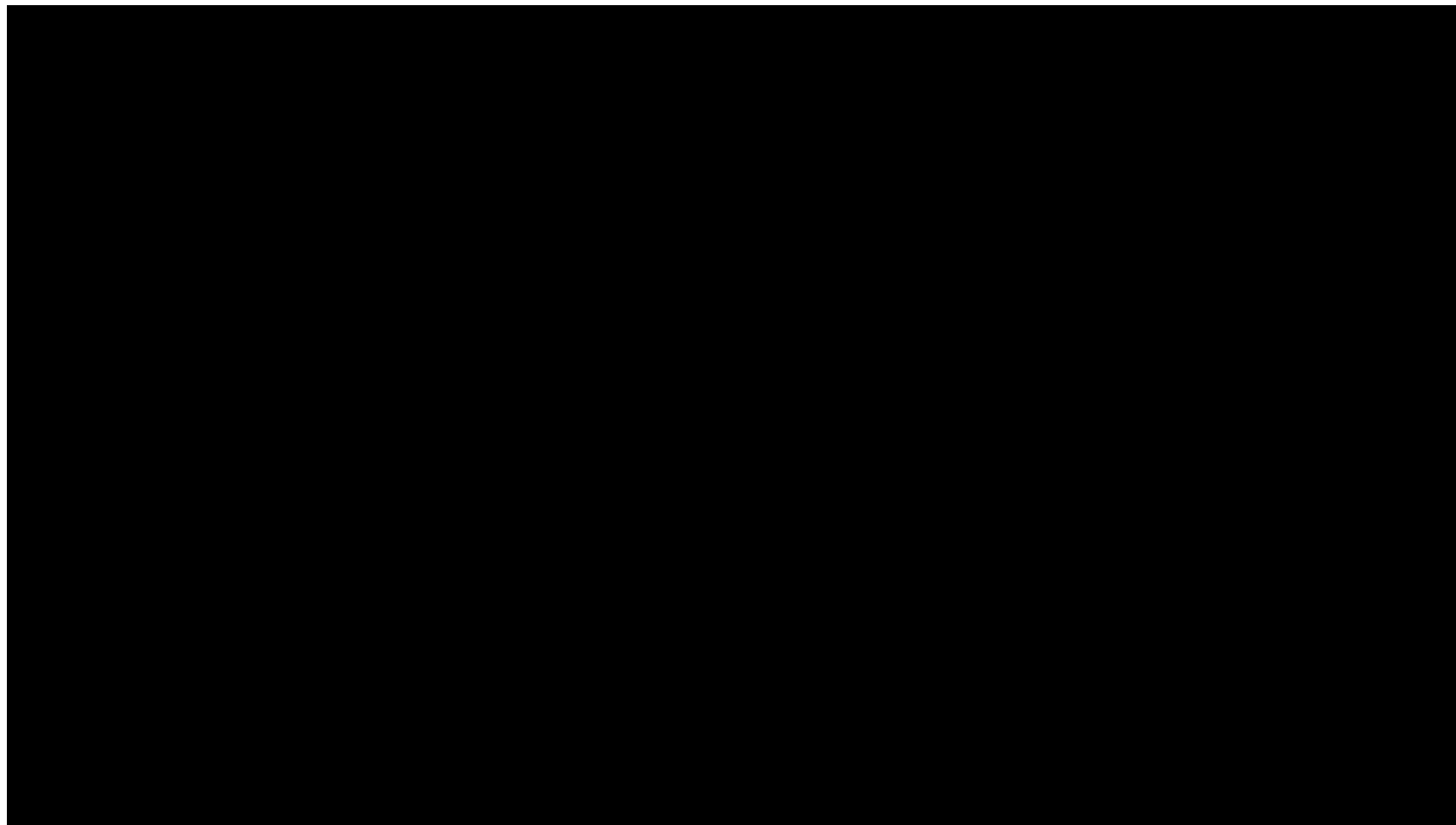
Result – Sequential Startup



Video Shown at 4x Speed



Result – Throttling

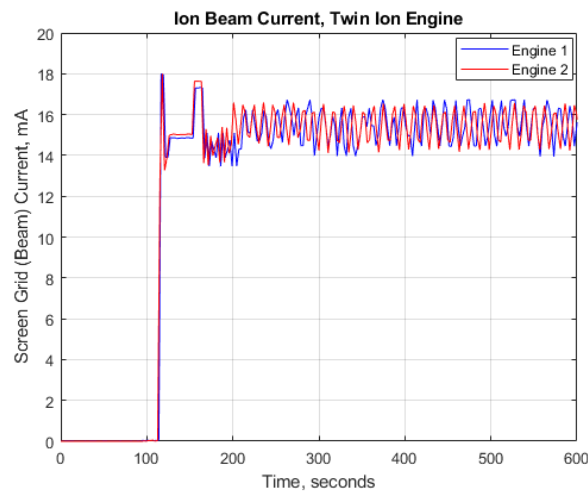
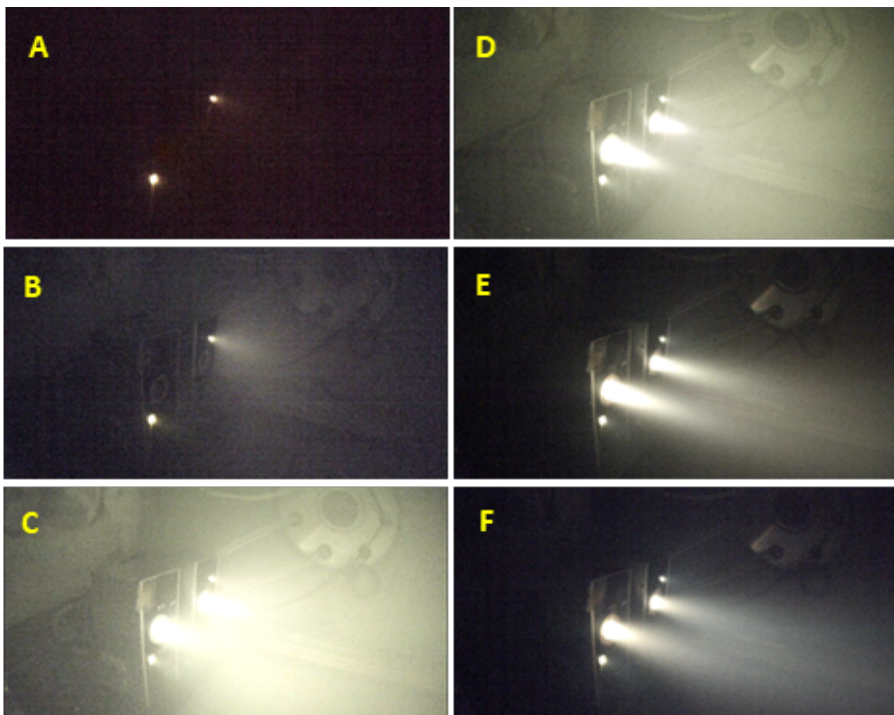


Video Shown at 2x Speed



Result – Simultaneous Startup

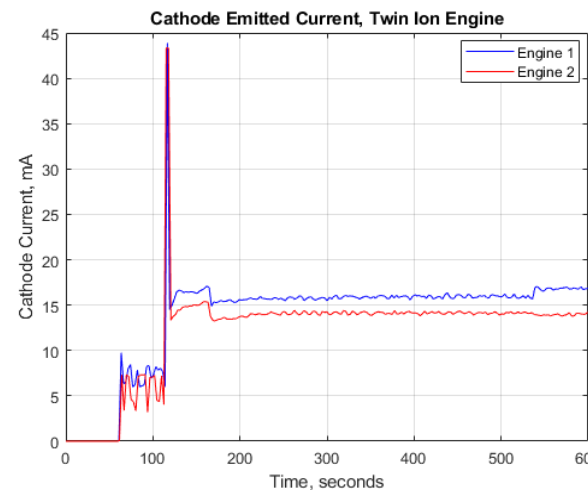
- Both thruster started up near simultaneously (<1sec offset due to command timing)
- No startup plasma interference issue
- Achieved overall neutralization (total cathode current > or = total ion beam current)



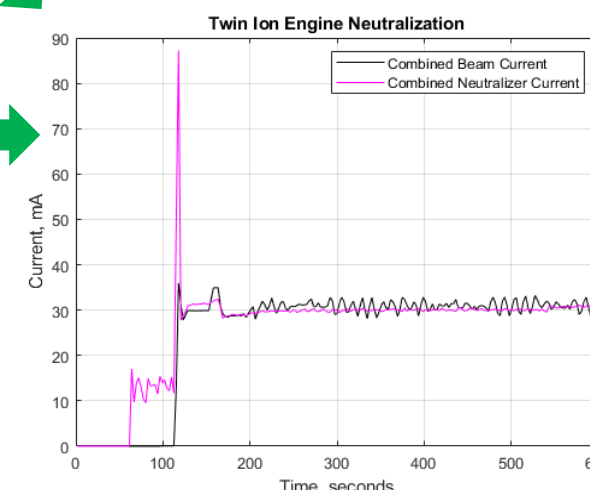
Thruster Ion Beam Current



System Input Power



Cathode Electron Current

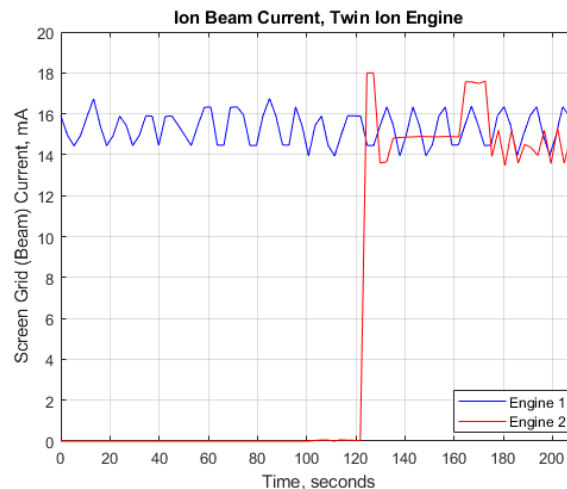
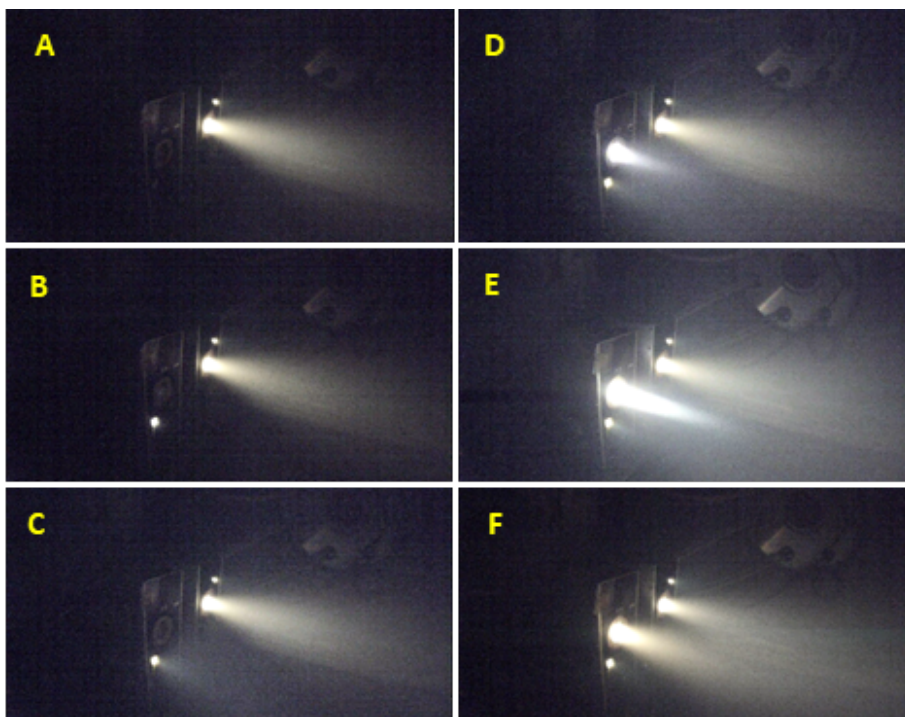


Combined Beam vs. Cathode Current

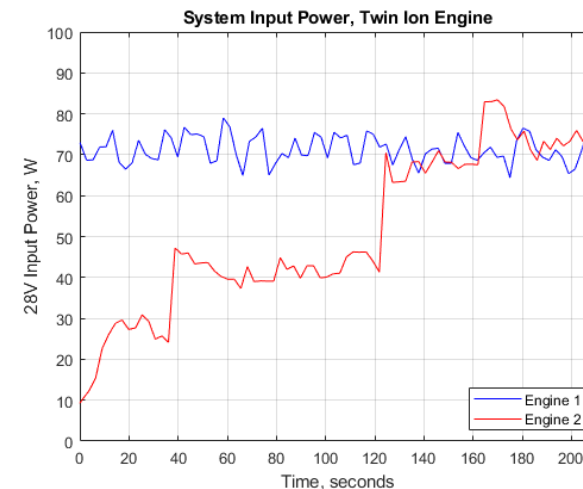


Result – Sequential Startup

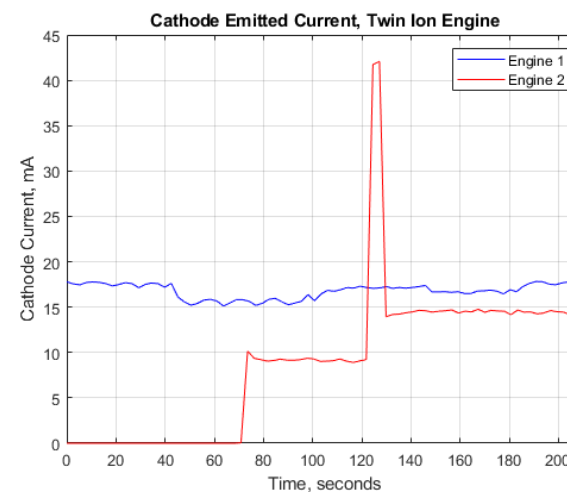
- Thrusters fired sequentially to simulate unsynchronized startups or single engine flame-out recovery
- No startup plasma interference issue
- Achieved overall neutralization (total cathode current > or = total ion beam current)



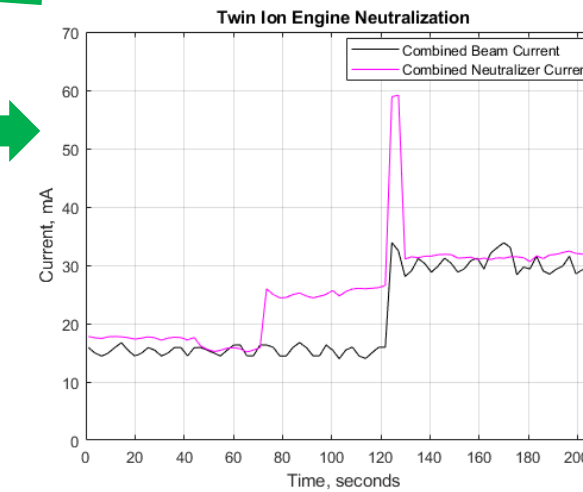
Thruster Ion Beam Current



System Input Power



Cathode Electron Current

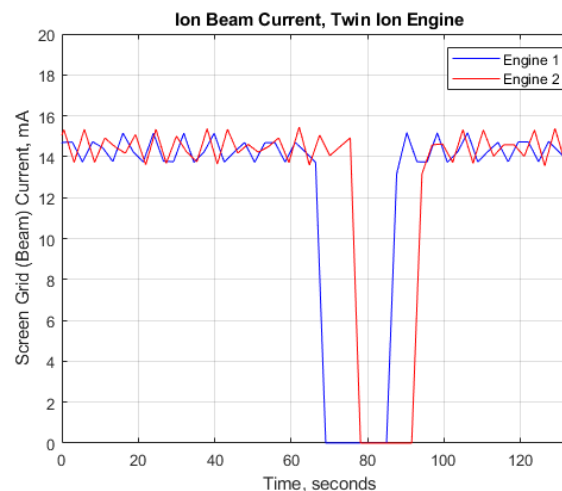
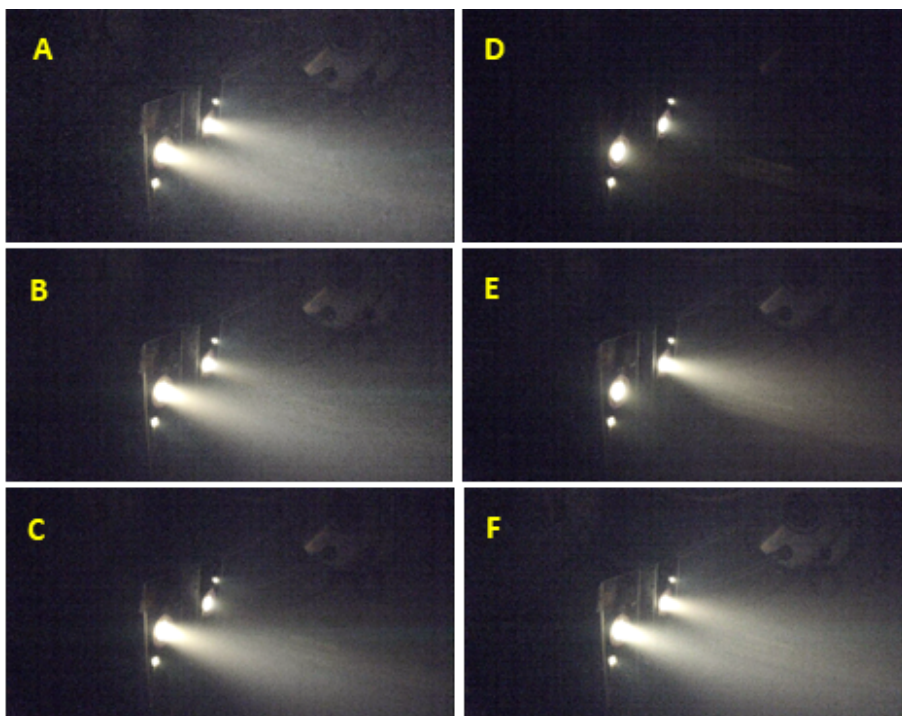


Combined Beam vs. Cathode Current

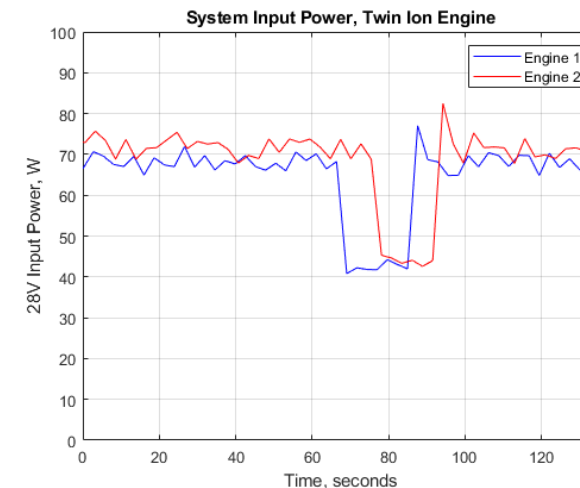


Result – Throttling

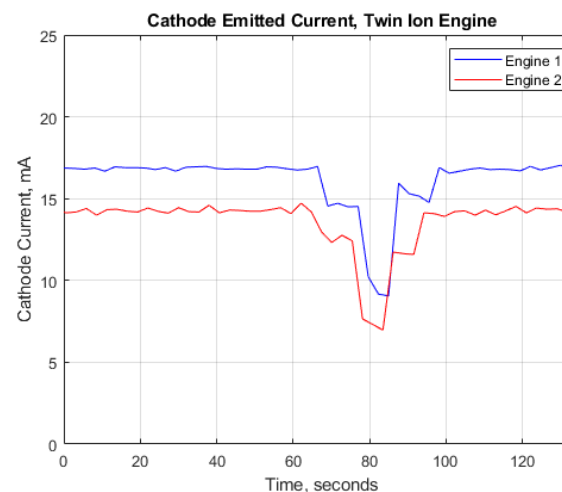
- Thrusters throttled sequentially from Lv4 (1.0mN) to Lv0 (no thrust) and back to Lv4
- No plasma interference issue / no flameouts
- Achieved overall neutralization (total cathode current > or = total ion beam current)



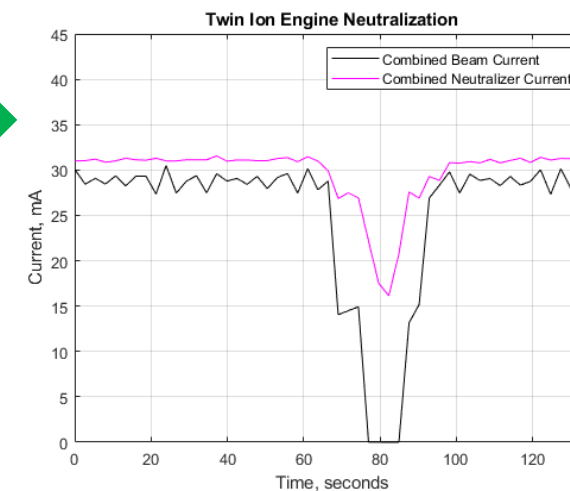
Thruster Ion Beam Current



System Input Power



Cathode Electron Current



Combined Beam vs. Cathode Current



Conclusion

- Two Gen-2 BIT-3 systems successfully demonstrated iodine hot firing side by side
 - Close proximity (6.5cm separation) had no observable electrical or plasma-interaction issues
 - Plasma plumes not entangled
 - Greatly reduces risk of clustering BIT-3 in confined s/c volume
- Three scenarios examined; thruster and cathode stable in all cases
 - Simultaneous ignition
 - Sequential startup
 - Independent throttling
- Iodine ion plumes were fully neutralized
 - The two cathode neutralizers were able to couple to the plumes via plasma bridge, and emit greater than or equal to the ion beam current