

Experimental Assessment of the Effects of a Porous Double Skin Façade System on Cladding Loads

G. Pomaranzi^{a,*}, N. Daniotti^b, P. Schito^a, L. Rosa^a, A. Zasso^a

^aPolitecnico di Milano, Department of Mechanical Engineering
via G. La Masa 1, 20156 Milan, Italy

^bUniversity of Stavanger, Department of Mechanical and Structural Engineering and Material Science
N-4036 Stavanger, Norway

Abstract

The high-efficiency façades, such as porous double skins, have become increasingly popular due to the recent technological progress in architecture. The so-called porous double skin façade (DSF) systems, which are constituted by a permeable layer over a closed inner façade, are often adopted to reduce the system energy demand. However, as expected, the porous skin alters the wind-induced pressures acting on the inner façade. Therefore, the cladding loads for such a façade system has to be accurately estimated performing wind tunnel tests. Using the low-rise buildings of the New Bocconi Campus as a case study, we present the experimental wind tunnel methodologies utilized to assess the wind-induced peak pressures acting on the inner glazed skin of the porous double skin façade system designed for the case at hands. In particular, the reduction of both the positive and negative peak pressures estimated for the inner façade is addressed when comparing the standard façade to the porous DSF case. In addition, the valuable data set of the pressure signals acquired for the porous DSF system studied, allows one to investigate the dependence of the computed peak pressures on the averaging time utilized for the extreme value estimates.

Keywords: Wind Tunnel Tests, Pressure Measurements, Double Skin façade, Porous Double Skin façade, Porous façade

1. Introduction

The Double Skin Façade (DSF) is a multilayer system characterized by two façades (or skins) which place the intermediate cavity airflow between the indoor and the exterior environment (Figure 1(a)). This cladding system is assuming ever-greater importance in modern building practice (Xiaotong and Chen, 2010) (Moon, 2011) (Poizaris, 2004) (da Silva and Gomes, 2008) for several reasons, such as interior and exterior aesthetics, sun-blocking, wind-blocking, and energy savings. When the outer skin is characterized by a porous metallic mesh, the system is defined as permeable double skin façade. The latter is primarily utilized to control the building solar radiation and, at the same time, allows one to have a nearly unobstructed view of the outside. The porous outer skin is constituted by perforated screens which are typically

*Corresponding author. Politecnico di Milano, Department of Mechanical Engineering, via La Masa 1, 20156 Milan, Italy.
e-mail: giulia.pomaranzi@polimi.it

Email addresses: giulia.pomaranzi@polimi.it (G. Pomaranzi), nicolo.daniotti@uis.no (N. Daniotti), paolo.schito@polimi.it (P. Schito), lorenzo.rosa@polimi.it (L. Rosa), alberto.zasso@polimi.it (A. Zasso)

20 constructed from aluminum, but they can also be made from galvanized steel, cold-rolled steel or stainless
21 steel.

22 The presence of the outer skin may alter the wind-induced pressure distribution on the inner façade and,
23 thus, the design cladding loads. The expected potential reduction of the design peak pressures compared
24 to the standard glazed façade case, if correctly estimated, has a relevant impact on the economics of the
25 building since the cladding system can achieve up to 25% of the total building costs (Overend and Zammit,
26 2006). However, Eurocode and the other National Codes do not currently address the calculation of wind
27 loads when these façade systems are adopted (Kilpatrick et al., 2009). In fact, as pointed by da Silva
28 and Gomes (2008), the impact of the DSF system on the inner skin's wind-induced pressure distribution
29 relevant to the cladding design has been sparsely ascertained, especially when the porosity of the external
30 skin is relatively high, e.g. $10 \div 50$ %. The porous DSF systems characterized by a porosity lower than
31 1% such as rain screens and porous roofs are found to be effective in reducing the wind-induced pressures
32 on the inner façade compared to the standard glazed façade case (Gerhardt and Janser, 1994)(Gerhardt and
33 Kruger, 1997). Belloli et al. (2014) have evaluated the effects of the porosity of perforated steel plates as
34 the cladding of a tall slender tower. The results highlight the peak pressure reduction due to the installation
35 of the permeable elements.

36 Nowadays the porous structural elements are utilized for civil engineering application also for their
37 aerodynamic behavior. For example, perforated shroudings have been investigated in the wind tunnel as
38 they may represent a suitable device able to control vortex shedding induced vibrations (Belloli et al., 2012).
39 Belloli et al. (2016) have analyzed the aerodynamic behaviour of a cylinder with perforated shrouding,
40 varying the porosity level and the gap between the external mesh and inner surface. Test performed in
41 smooth flow condition and sub-critical Reynolds Number range have highlighted a reduction of vortex
42 shedding induced vibration with the application of the mesh, especially for porosity higher than 40%. Hu
43 et al. (2017) have studied the effects of a double-skin façade with vertical openings installed in front of the
44 windward face on the wind-induced response of the bare CAARC model. The cross-wind response of the
45 structure, associated with vortex shedding phenomena, has been proved to be significantly reduced due to
46 gap flow between the windward face and the inner façade which results in less severe pressure fluctuations
47 acting on the side faces. Kemper and Feldmann (2019) have proposed different wind load recommendations
48 for rectangular shaped buildings with single porous cladding and porous double skin façade. For the latter,
49 the porous cladding elements are made by stainless steel wire meshes with 20% porosity. Both full-scale
50 measurements and wind tunnel tests in a scaled model have been performed to address the aerodynamic
51 behavior of the porous skin and the corresponding wind loads.

52 The main scope of this research is to assess the external wind-induced pressure distribution on the inner
53 glazed façade, when an outer porous mesh is installed. The case at hands is the New Bocconi Campus
54 low-rise buildings, the cladding system of which is constituted by a porous DSF, having a diamond-shaped
55 expanded metallic mesh as the outer skin. The multi-scale experimental methods utilized during the wind
56 tunnel tests aim to provide a reliable estimate of the design wind-induced peak pressures acting on the inner
57 façade. To this purpose, the buildings of the case study are first modeled adopting a standard façade system.
58 Thereafter, the metallic mesh is scaled to guarantee the same aerodynamic behaviour of the prototype one,
59 and the wind-induced pressures acquired on the inner skin of the porous DSF system are assessed. Thus,
60 the expected reductions of the peak pressures compared to the standard glazed façade are quantified. The
61 valuable pressure data set is further utilized to study the characteristics of the pressure signals, focusing on
62 the beneficial effects of the adopted outer skin and the impact of the averaging time used in the extreme
63 value analysis.

64 The paper is organized as follows: Section 2 describes the multi-scale experimental methodologies

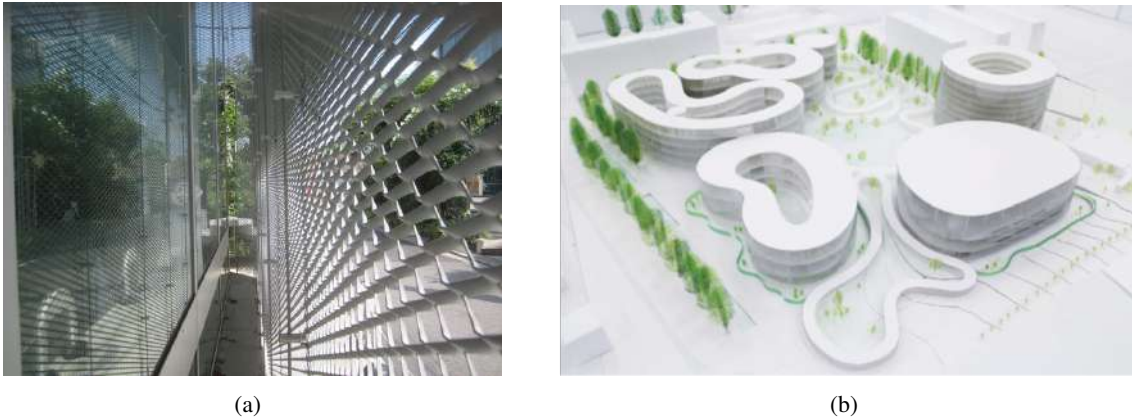


Figure 1: (a) Example of Double Skin Façade porous system; (b) the *New Bocconi Campus* rendering.

65 adopted in the wind tunnel to study a porous DSF system using the New Bocconi Campus low-rise buildings
 66 as a case study. Section 3 illustrates the output the pressure measurements performed whereas Section 4
 67 investigates the reduction of the peak pressures on the inner façade.

68 2. Wind tunnel tests set-up

69 2.1. Case Study: the New Bocconi Campus buildings

70 The New Bocconi Campus is a building complex located within the urban area of 36000 m^2 in southern
 71 part of Milan (Italy) and comprises six different low-rise buildings, the designated use of which will be
 72 offices, dormitories as well as a recreation centre. Figure 1(b) shows a rendering of the structures: they
 73 are characterized by cylindrical shape with high curvature radius for the vertical walls. The height of the
 74 structures varies from 21 m to 35 m . The buildings are designed utilizing a double skin porous façade
 75 system. The inner glazed façades are covered by a metallic mesh, which thus allows the so-called gap-flow
 76 to develop (see Figure 1(a)) in between the inner and the outer skin. Figure 2(a) depicts the porous double
 77 skin façade designed for the case at hand, whereas Figure 2(b) illustrates the close-up of the prototype outer
 78 skin, namely a diamond-shaped expanded metallic mesh. According to the peculiar design of the structures,
 79 the perimeter defined by the outer skin does not comply with the one corresponding to the inner glazed
 80 façade. In fact, the metallic mesh is bent in the horizontal plane in such a way that the wavy profile shown
 81 in Figure 3 is obtained. The distance between the outer and inner skins varies in the range $750\text{ mm} \div 950\text{ mm}$
 82 for all the buildings. The basic geometrical characteristics of the mesh are summarized in Table 1. The listed
 83 quantities refer to Figure 4 whereas k is the pressure loss coefficient, which is introduced in section 2.4. The
 84 expected 50-years return period mean wind speed at the reference height $H_{ref} = 35\text{ m}$ (roof height of the
 85 tallest building) is computed according to the Eurocode (AFNOR, 2008) assuming terrain category IV, and
 86 is equal to 22.5 m/s .

87 2.2. POLIMI Wind Tunnel and Instrumentation

88 The experimental activities are conducted in the wind tunnel of Politecnico di Milano, which is a 1.5
 89 MW closed-circuit wind tunnel, the boundary layer test section of which is 4 m high, 14 m wide and 36 m
 90 long. The latter distance is adequate to develop a stable boundary layer whereas the large dimensions of the
 91 test section allow one to adopt concurrently a large geometric scale λ_L ($\lambda_L = \text{model}/\text{real}$), and a low value

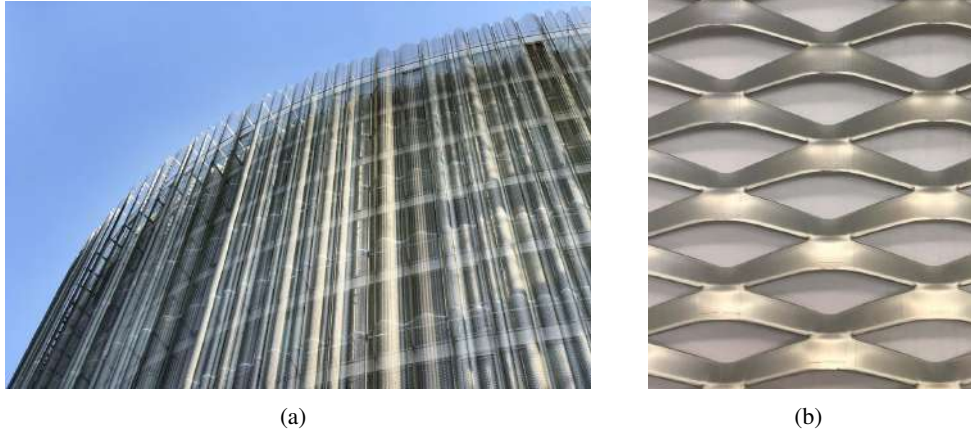


Figure 2: (a) The double skin with closed inner façade and the outer diamond shaped expanded metallic mesh of the New Bocconi Campus; (b) close-up of the prototype outer mesh.

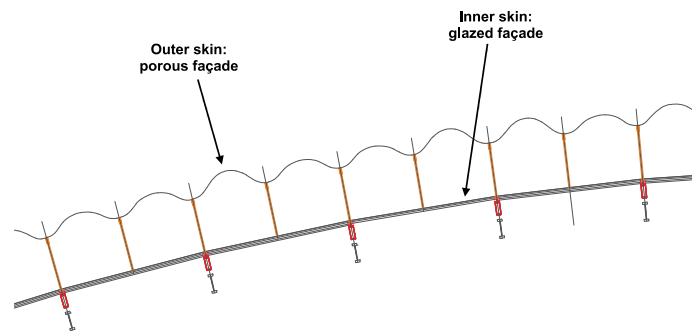


Figure 3: Top view of the Double Skin Façade system: detail of the wavy profile of the outer mesh adopted for the New Bocconi Campus.

Table 1: Geometrical characteristics of the full-scale and model-scale meshes.

		Full scale	Model scale
DS	shorter diagonal	35 mm	3 mm
DL	longer diagonal	85 mm	6 mm
Av	width	11 mm	0.8 mm
Sp	thickness	2 mm	0.5 mm
k	pressure loss coefficient	2.24	2.32

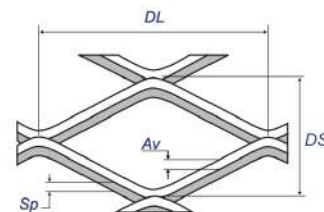


Figure 4: Geometry of the diamond shaped expanded metallic mesh.

Table 2: The two stages of the wind tunnel tests: name, description, geometric scale λ_L and velocity scale λ_V .

Stage	Name	Façade System	λ_L	λ_V	Sampling time
FC1	Façade Configuration 1	Standard Glazed Façade	1:100	1:2.53	120 s
FC2	Façade Configuration 2	Double Skin Porous Façade	1:50	1:2.14	120 s

92 of the blockage ratio, which is equal to 1.25% for the worst case tested. These aspects are fundamental
 93 to reduce the effects related to the Reynolds number dependency and avoid flow distortions during the
 94 experimental tests, respectively.

95 To understand the aerodynamic behaviour of the metallic mesh concerning the pressure distribution on
 96 the inner façade, this research project is divided into two stages, during which rigid models are tested in
 97 turbulent flow conditions. In the first stage, which is referred to as *FC1* (Façade Configuration 1), cladding
 98 loads on rigid models ($\lambda_L = 1 : 100$) with no outer skins are assessed. As we model a standard glazed
 99 façade system, the main scope of the aforementioned test session is to analyze the peak pressure envelopes
 100 of all the buildings and, consequently, identify which building is affected by the most severe cladding loads
 101 scenario. The aerodynamic behavior of the outer metallic mesh is adequately modeled before entering the
 102 second stage, which is referred to as *FC2* (Façade Configuration 2) in the following. In fact, the adequately
 103 modelled porous element and the increased geometrical scale ($\lambda_L = 1 : 50$) allow one to measure effectively
 104 the pressure distributions on both the inner façade and the metallic mesh. Hence, the effects of the porous
 105 element on the inner façade's pressures can be investigated.

106 The models of the building analyzed during both the test sessions are instrumented with 8 PSI ESP-32
 107 HD high-speed pressure scanners, each of them having 32 pressure channels. The scanners are connected to
 108 the Initium data acquisition with a sampling frequency of 500 Hz. The sampling time in both stages is 120 s
 109 model scale, which corresponds to 79 minutes and 47 minutes full scale for *FC1* and *FC2* respectively. The
 110 maximum pressure tubing length adopted in the pneumatic chain is kept lower than 90 cm to minimize the
 111 induced distortion in the pressure measurement system. The pressure signals are corrected in the frequency
 112 domain, based on the frequency response function estimated according to Bergh and Tjrdeman (1965).

113 2.3. *FC1: set-up and model description*

114 In the first stage, all five buildings of the New Bocconi Campus are tested. Rigid models, which re-
 115 produce the full-scale geometry of the structures, are instrumented with pressure taps (Figure 5(a)). The
 116 geometric scale is $\lambda_L = 1 : 100$ whereas the velocity scale is $\lambda_V = 1 : 2.53$, as reported in Table 2. Being
 117 the buildings characterized by curved shapes, the Reynolds number dependency is adequately addressed.
 118 The external surfaces of the tested buildings are roughened using suitable sandpapers to improve local flow



Figure 5: (a) Set-up of *FC1*; (b) set-up of *FC2*.

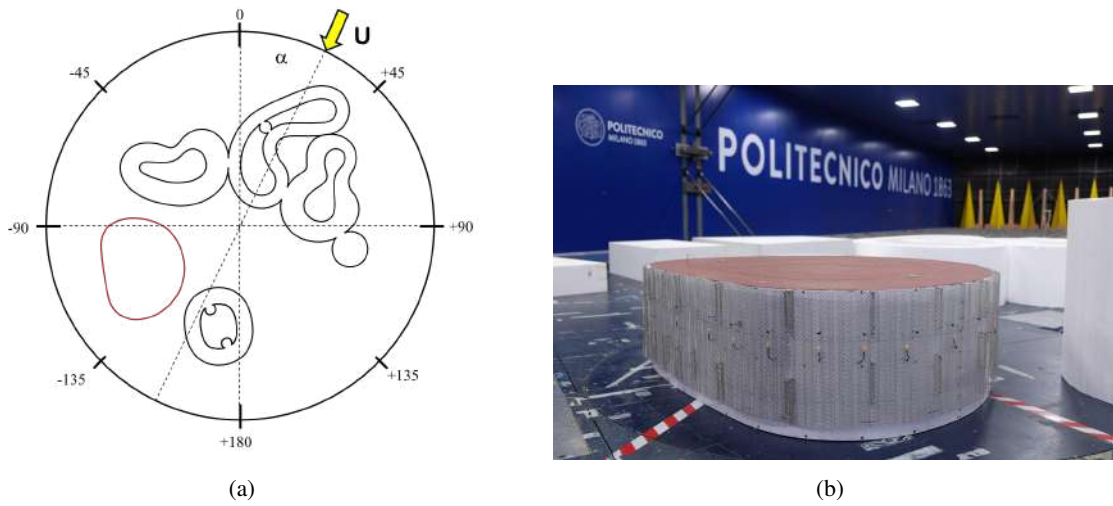


Figure 6: (a) Wind direction convention and reference system; (b) *FC2* building during the second stage of the tests.

119 similarity and reduce the dependence on the Reynolds number effects. In fact, the surface roughness in-
 120 creases the turbulence locally, allowing one to reproduce the fluid-dynamic effects corresponding to higher
 121 Reynolds number. The aforementioned practice is commonly adopted in boundary layer wind tunnel tests
 122 when dealing with curved surfaces, see for example Diana et al. (2013). As the structures are low-rise build-
 123 ings, the pressure taps are distributed along one single level, at $2/3$ of the building height with an average
 124 spacing of 4 cm model-scale for each building. The total amount of the measurement points is 256.

125 2.4. Scaling procedure for the permeable façade

126 The identification of the porous medium to be used during the *FC2* tests is essential to reproduce cor-
 127 rectly the aerodynamic behaviour of the double skin façade system. As the air flows through a porous
 128 medium, an interaction between the permeable element and the fluid occurs. The geometry of the mesh, its
 129 porosity, and the fluid properties are the main parameters the interaction depends on. Specifically, when the
 130 flow approaches the porous medium, it passes through narrow openings followed by short section enlarge-
 131 ment. This mechanism leads to a pressure drop between the two sides of the permeable element.

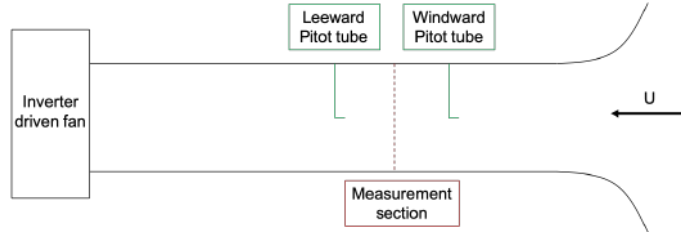


Figure 7: Schematic representation of the open circuit wind tunnel with pressure sensors and location of the porous screen.

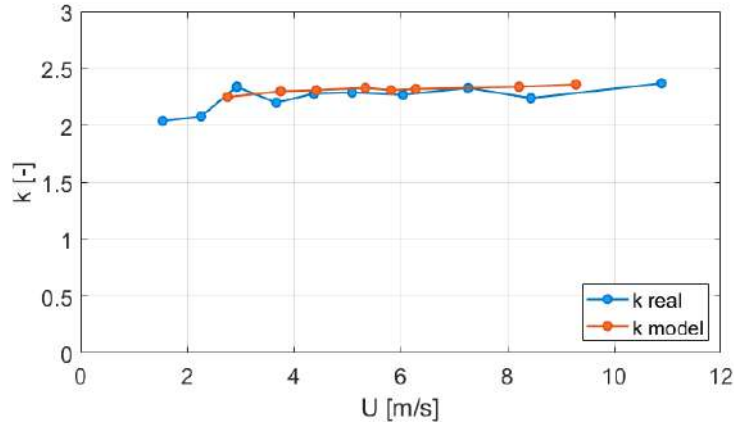


Figure 8: Experimental pressure loss coefficient with varying velocity: real mesh and model mesh.

132 The geometrical scaling applied to the global structure is not straightforward for those details which are
 133 already small at full scale. To ensure the kinematic similitude, e.g. the same flow conditions between real
 134 and model scale, the pressure loss coefficient k must be maintained (Belloli et al., 2014) (Letchford et al.,
 135 2000). It is defined as:

$$k = \frac{\Delta p}{\frac{1}{2}\rho U^2} \quad (1)$$

136 where Δp is the pressure difference between the windward and the leeward side of the porous screen; ρ
 137 is the air density; U is the mean wind speed. The k parameter is an index of the aerodynamic resistance to
 138 the flow through a porous element; it accounts for the effects of porosity and the shape of the holes.

139 The pressure loss coefficient is experimentally evaluated, performing dedicated tests in a mostly lami-
 140 nar open circuit wind tunnel, with circular cross-section (diameter 400 mm). Figure 7 shows a schematic
 141 representation of the facility. Two Pitot tubes are positioned at the windward and leeward side of the porous
 142 screen, which is placed in the measurement section. Based on the measured pressure values, the pressure
 143 loss coefficient is computed through equation 1.

144 The k parameter is firstly determined for the real scale mesh. Several samples of mesh models are
 145 then tested and the one characterized by the same k value is finally chosen. The pressure loss coefficient
 146 comparison of the prototype and model meshes is shown in figure 8: an overall agreement is found between
 147 the k parameter curves within the range of wind speed of interest.

148 2.5. *FC2: set-up and model description*

149 The second stage of the research project, e.g. *FC2*, assesses the pressure distribution on the permeable
150 DSF system, which is adequately reproduced for the building which experiences the highest wind loads
151 during the *FC1* wind tunnel tests. The building model equipped with the scaled DSF system is depicted in
152 Figure 5(b). The geometrical scale $\lambda_L = 1:50$ is twice as big as *FC1* geometrical scale. This enables one to
153 model properly the porous skin, guaranteeing a gap between the two skins large enough to let the gap flow
154 develop. It is worth noting that the velocity scale is $\lambda_V = 1:2.14$, as reported in table 2.

155 The model of the building is rigid and the outer porous skin, the characteristics of which are reported
156 in table 1, is installed onto the external surface of the model using rectangular aluminum profiles. The
157 latter guarantee a geometrically scaled distance between the two skins and, at the same time, minimize the
158 potential flow distortion of the gap flow (Figure 9(a)). Before installing it on building model, the mesh is
159 bent to reproduce the wavy profile of the prototype metallic mesh, which can be appreciated from the top
160 view depicted in Figure 3. To measure the pressure distribution on the inner façade, the pressure taps are
161 positioned along the inner perimeter of the building model in such a way that a correspondence between
162 the measurement points of *FC1* and *FC2* can be established. Additionally, pressure sensors are also placed
163 in columns (Figure 10) to investigate the vertical variation of the wind-induced pressures on the inner skin.
164 The location of such vertically distributed acquisition points is selected based on the most severe wind
165 directions in terms of peak pressures, which are evaluated during *FC1* test. Further pressure taps are located
166 along the aforementioned inner perimeter to increase the spatial resolution of the pressure measurements.
167 As previously mentioned, this stage aims also to assess the wind pressure distributions on both the inner
168 façade and the metallic mesh. As the local wind loads on a permeable skin are estimated based on net
169 pressures, the outer skin of the model is instrumented to measure both the external pressure and the inner
170 gap pressure. The former is acquired using pressure taps positioned at the center of a single opening of the
171 mesh previously filled with resin material, trying to minimize the interference between the measurement
172 points distributed on the inner skin. Additionally, pressure taps fixed by small supports (button-like pressure
173 taps shown in Figure 9(a)) are mounted onto the external surface of the model mesh to measure the inner
174 gap pressure. The pressure tubing system utilized for the mesh is assumed to have a negligible influence on
175 the pressure signals acquired on the inner façade, given the relatively small outer diameter (2.4 mm) of the
176 adopted tubes. The total amount of the measurement points during this test is 204.

177 Multi-holes cobra probes are utilized to retrieve the 3-components of the velocity of the gap-flow during
178 the *FC2* stage. The sensors are installed in the gap between outer and inner skin as well as in front of the
179 external surface of the mesh. Hence, insights into the aerodynamic behaviour of the DSF system comple-
180 mentary to the pressure measurements can be addressed. As the probes are mounted to point towards the
181 expected direction of the oncoming gap-flow, they provide a reliable estimate of the flow fields within a
182 range of $\pm 45^\circ$. The adopted sampling frequency is 2000 Hz.

183 2.6. *Flow Configuration*

184 As the comparison between the cladding's pressure distribution obtained during *FC1* and *FC2* stages are
185 based on different geometrical scales, an accurate scaling of the atmospheric boundary layer is essential.
186 The latter is adequately modeled based on the characteristic of the site and the surrounding territory. To
187 ensure that the flow conditions agree with the predicted ones, the profiles of the mean wind speed, turbulence
188 intensity and turbulence integral length scale are scaled according to the geometric scale, as shown in Figure
189 11. It is worth noting that the reference heights $H_{ref,j}$, $j = FC1, FC2$ correspond to 35 m full-scale and
190 the plotted model heights are normalized accordingly. The position of the pressure measurement points are
191 located at 0.4 of $H_{ref,j}$, $j = FC1, FC2$ on tested building models. An overall agreement between the profiles

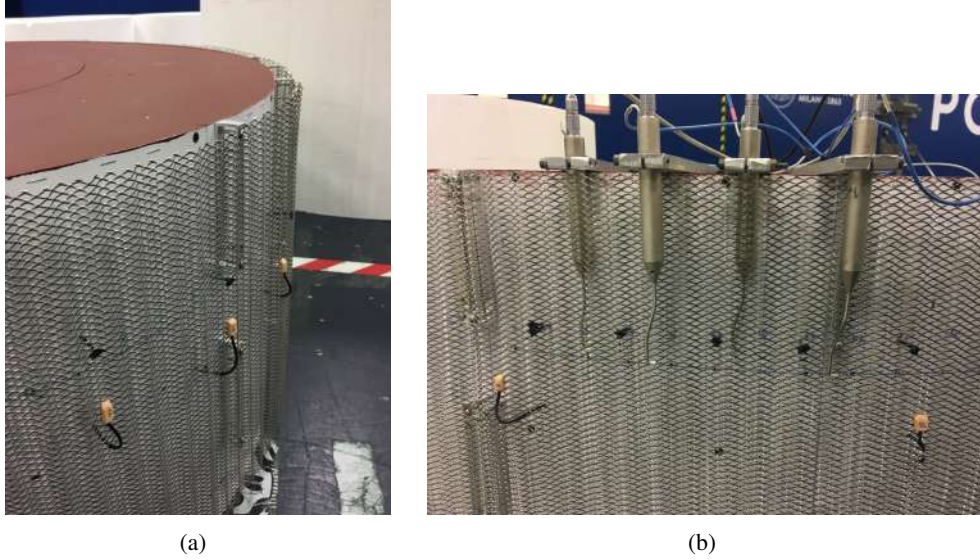


Figure 9: (a) Close-up insight into the rectangular aluminum profiles used to anchor the outer mesh as well as the measurement system of the net pressure acting on the metallic mesh and (b) Cobra Probes set-up during the *FC2* stage.

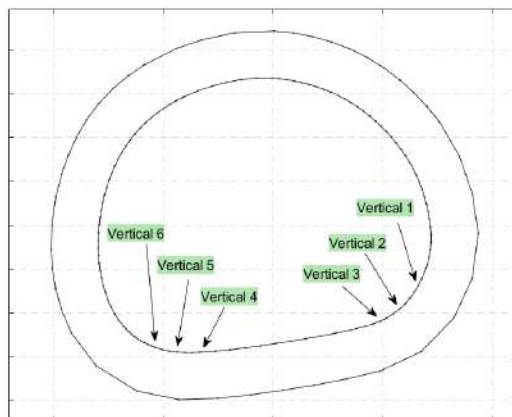


Figure 10: Position of the pressure taps columns on the inner skin of the *FC2* model, top view.

192 of the mean wind speed normalized by reference mean wind speed at $H_{ref,j}$, $j = FC1, FC2$ is found. By
 193 fitting the logarithmic profile of the mean wind speed to the measured ones, we obtain a roughness length
 194 approximately equal to 1 m.

195 The profiles' shapes of the along-wind, across-wind, and vertical turbulence intensities are adequately
 196 reproduced within the investigated heights of the atmospheric boundary layer. However, the turbulence
 197 intensity magnitudes estimated during *FC2* are 5% to 7% lower than the *FC1* ones at H_{ref} . Discrepancies
 198 in the turbulence intensity level may affect the extreme value analysis of the pressure signals and, thus,
 199 the estimated peak pressures. However, when it comes to pressures distribution on low-rise buildings, the
 200 pressure fluctuations on the façade are also determined by the fluid-structure interactions due to the specific
 201 geometry of the surrounding and the shapes of the buildings, which are accurately modeled during the tests.
 202 Within the height range of interest, e.g. $z/H_{ref} < 1$, some discrepancies between the turbulent length scale
 203 profiles estimated during the tests are found, especially for the along-wind turbulence length scale. However,
 204 this is expected since the computation of the auto-correlation function depends on the degree of stationarity
 205 of the time histories and, therefore, the resulting turbulence length scale profile is more scattered than other
 206 turbulence characteristics. However, as pointed out by Aly and Bitsuamlak (2013), discrepancies on the
 207 integral length scale scaling in a wind tunnel test should not affect the prediction of the wind loads for low-
 208 rise buildings. Such differences may have an impact on the spatial correlation of the pressure fluctuations,
 209 which is out of the scope of the present study.

210 The agreement between the one-point spectrum of the turbulence components is also addressed, as
 211 shown in Figure 12. The one-point spectral density functions normalized by the variance of the velocity are
 212 estimated at the reference heights $H_{ref,j}$, $j = FC1, FC2$ and compared to Von-Karman velocity spectrum
 213 model. The latter spectral model superimposes well to the experimentally estimated velocity spectra.

214 3. Experimental Results

215 In this section, the wind tunnel tests results are presented. The main aim of this study is the assessment
 216 of the impact of the outer permeable skin on the inner façade pressure distribution through a comparison
 217 of the wind tunnel tests outcomes. Before proceeding in such comparison, some preliminary ascertainment
 218 have to be addressed. Firstly, given the circular shape of the buildings, the Reynolds dependency has to be
 219 investigated and discussed properly. Then, it has to be ensured that the different geometric scales are not
 220 affecting the results.

221 3.1. Pressure Coefficients Definition

222 The acquired pressure data are treated in non-dimensional form as local pressure coefficients defined,
 223 at the k -th pressure tap, as follows:

$$C_p^k(t) = \frac{p^k(t) - p_s}{q_H} \quad (2)$$

224 where $p^k(t)$ is the local pressure, p_s is the reference static pressure in the test section and $q_H = 1/2\rho U_{ref}^2$
 225 is the wind reference dynamic pressure at $H_{ref,j}$, $j = FC1, FC2$. From the definition itself, positive pres-
 226 sure loads represent thrusts while negative pressure loads represent suctions (inward and outward arrows
 227 respectively in the pressure distributions figures). The pressure coefficients are hereafter presented using
 228 their mean value (\bar{C}_p), positive extreme (\hat{C}_p) and negative extreme (\check{C}_p).

229 When dealing with the metallic mesh, the total wind load acting on the porous skin is computed as the
 230 net pressure coefficient, namely the difference in the time domain between the external (*ext*) and internal
 231 (*int*) pressure coefficients acquired at the k -th tap:

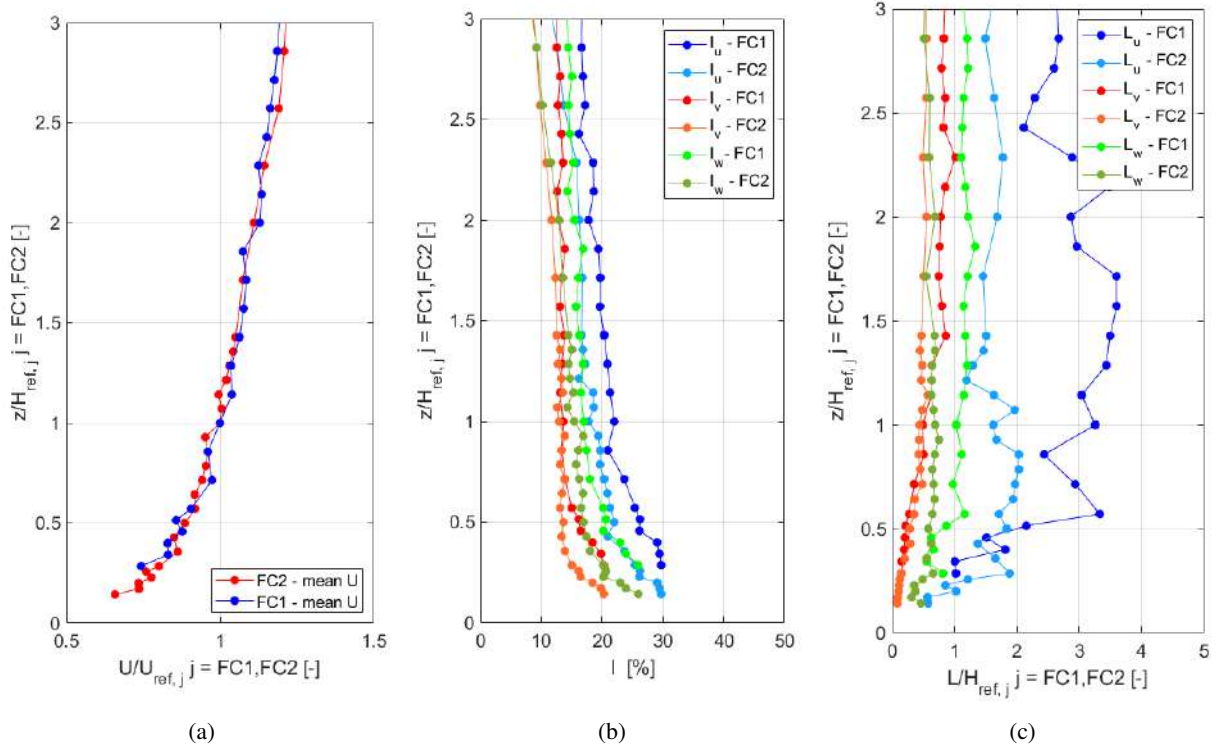


Figure 11: (a) mean velocity profile, (b) turbulence intensity profile and (c) integral length scales for the *FC1* and *FC2* tests.

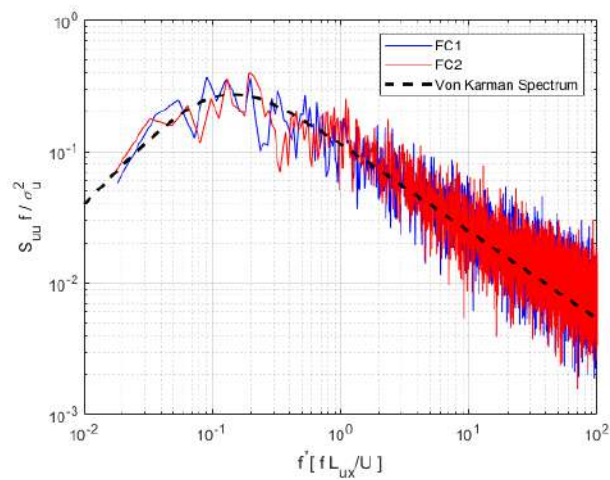


Figure 12: One-point velocity spectral density estimates of the along-wind turbulence at $H_{ref,j}$, $j = FC1, FC2$.

Table 3: Reference wind velocities to test Reynolds dependency. Values are in m/s .

	<i>FC1</i>	<i>FC2</i>
$U_1(H_{ref})$	4.61	7.68
$U_2(H_{ref})$	6.76	8.11
$U_3(H_{ref})$	8.57	10.01
$U_4(H_{ref})$	9.34	11.42

$$C_{p,net}^k(t) = C_{p,ext}^k(t) - C_{p,int}^k(t) \quad (3)$$

232 Cladding pressure time-histories are recorded and statistically analyzed to estimate the extreme pressure
 233 coefficients (positive and negative) by applying the method introduced by J. Cook and J.R. Mayne (Cook,
 234 1986). Each $C_p(t)$ time history is divided into 10 windows, the extreme positive/negative pressure coef-
 235 ficient is extracted from each window and, finally, the Gumbel distribution is fitted to the extreme values
 236 (Cook, 1986). Providing that the sampling time is longer than the shortest practicable period of 10 *minutes*
 237 full-scale (Cook and Mayne, 1979) (Mayne and Cook, 1980), the method allows on to correctly estimate the
 238 peak pressure coefficient. For the present case, *FC1* and *FC2* acquisition period corresponds to 79 *minutes*
 239 and 47 *minutes* full-scale respectively. Hence, the method can be confidently applied and the differences in
 240 time history duration do not affect the extreme value estimation.

241 According the current practice in wind engineering design façade, before proceeding with the extreme
 242 value analysis, a low-pass moving average filter with an averaging time equal to 3 s real scale (the so-called
 243 3 s *gust*) is applied to the signals to account for the lack of correlation of the wind gusts over large areas in
 244 such a way that the effects of the eddies smaller than the loaded area are removed.

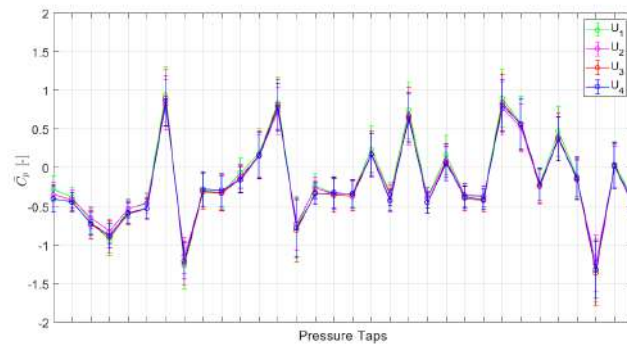
245 3.2. Reynolds number dependency

246 As the Reynolds number similitude requirement cannot be achieved in the boundary layer wind tunnel
 247 tests and the models are characterized by curved surfaces, the corresponding scaling effects are addressed.
 248 In particular, the dependency of the mean pressure coefficients upon the tested mean wind speed is analyzed
 249 to assess the sensitivity of the experiments to the Reynolds number. For both the test sessions, four different
 250 wind speeds measured at the corresponding $H_{ref,j}$, $j = FC1, FC2$ are considered, as reported in table 3.
 251 Based on the maximum mean wind speed considered, the tested Reynolds number are $Re_{FC1} = 3.2 \times 10^5$
 252 and $Re_{FC2} = 7.9 \times 10^5$. Figure 13 reports the mean pressure coefficients \bar{C}_p of 32 pressure taps randomly
 253 selected during *FC1* (Figure 13(a)) and *FC2* tests (Figure 13(b)).

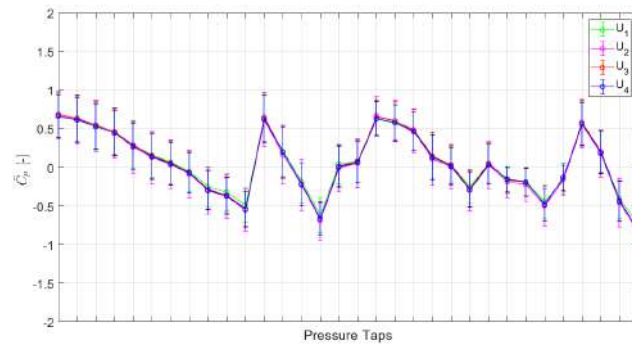
254 Within the considered wind speed range, for both the test sessions the mean pressure coefficients exhibit
 255 a negligible discrepancy when varying the oncoming mean wind speed. Such a result underpins the inde-
 256 pendence of the acquired pressure signals on the Reynolds number for the tested mean wind speed range
 257 and thus make it possible to proceed confidently with the wind tunnel test.

258 3.3. Vertical pressure distribution

259 The main scope of this research is to deeper into the wind pressure distribution of the inner glazed façade
 260 when an outer permeable skin is installed. The research structure, which consists of two different test stages,
 261 allows one to compare the pressure distribution from *FC1* with the one acquired on the inner cladding in the
 262 *FC2*. Before proceeding in that, it has to be ensured that differences in wind-induced pressure are not due to
 263 different geometric scales of the two stages that are responsible for some discrepancies in the boundary layer
 264 characteristics, as already pointed out in section 2.6: the turbulence intensities profiles show differences up



(a)



(b)

Figure 13: \bar{C}_p dependency on test wind speed for (a) *FC1* and (b) *FC2* tests.

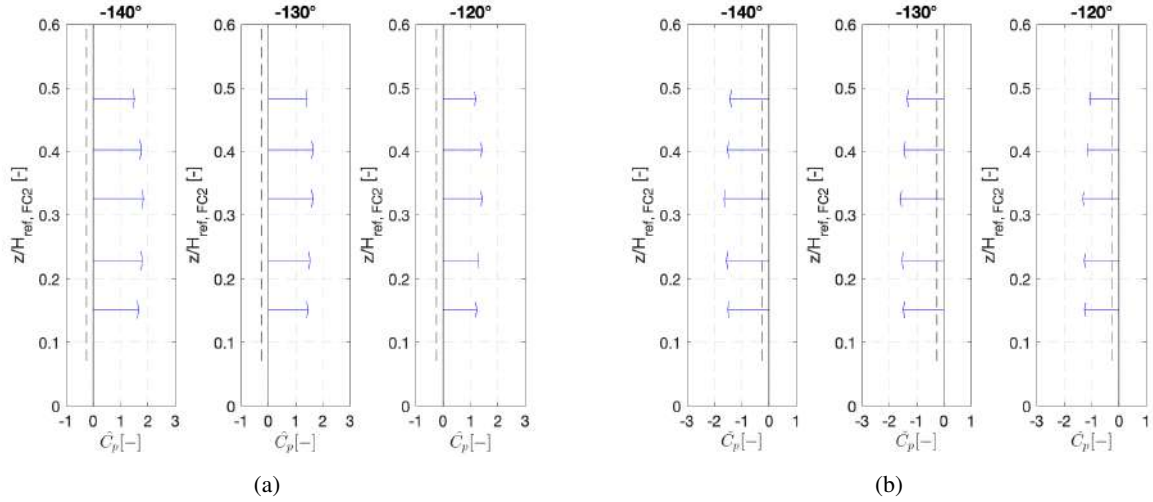


Figure 14: (a) \hat{C}_p distribution for the vertical 2 and (b) \check{C}_p distribution for the vertical 5, -140° , -130° and -120° wind directions.

265 to 5% between the FC1 and FC2 stages at the same z coordinate while running along the building height,
 266 turbulence intensities vary more than 8%.

267 The pressure sensors placed in columns during the *FC2* session (as depicted in Figure 10) allows the
 268 possibility to investigate the vertical distribution of the peak pressure coefficients in the inner façade and to
 269 discuss their eventual dependency on the turbulence intensity components.

270 For example, Figure 14 shows the \hat{C}_p from Vertical 2 and \check{C}_p from Vertical 5 for the wind directions
 271 -140° , -130° and -120° ; a little dependency on the z coordinate of the peak pressure coefficient is found.
 272 In particular, the positive and negative peak pressures show the largest values at half height for all the
 273 considered exposures. These findings demonstrate that the presence of the outer skin makes the dependency
 274 of the extreme values on the height (and so on the turbulence intensities) less severe. This is the main reason
 275 why discrepancies in turbulence intensities between the two stages are assumed to not significantly affect
 276 the pressure distribution comparison, which can be thus confidently performed.

277 3.4. *FC1 and FC2 results comparison*

278 Figure 15(a) shows the mean pressure distribution for wind direction -140° (Figure 6(a)) from the
 279 *FC2* test: the inner ring represents the pressure acting on the glazed façade, the outer one refers to the net
 280 pressures on the metallic mesh. It emerges that the net pressure on the mesh is highly correlated with the
 281 external pressure distribution on the inner façade, even if the former denotes lower values of C_p (note that
 282 the scale factor for the mean pressure on the mesh is doubled with respect to the one for the inner façade).

283 The high correlation and the sign of the net pressure on the porous external layer is worth of comments
 284 and additional studies. At this aim, by linking the velocity measurements (that will be presented in section
 285 3.5) and pressure distributions, the following mechanism can be highlighted:

- 286 • intake with a positive mean net pressure load on the outer skin, coherently with a drop in the velocity
 287 component passing from outside to the inner gap shown by the Cobra measurements;
- 288 • flow inside the skins gap: the air flows along an ideal line parallel to the inner façade. The flow
 289 recorded along the outward normal is found to be negligible compared to the inner façade and there-

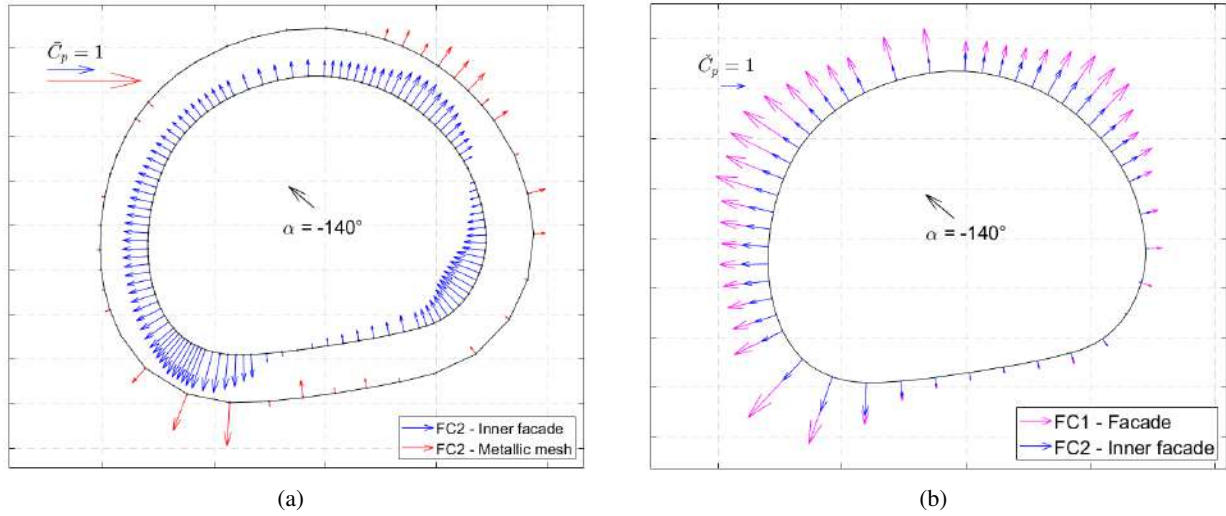


Figure 15: (a) \bar{C}_p distribution for wind direction -140° , *FC2* test. (b) \check{C}_p distribution comparison between *FC1* vs *FC2*, -140° wind direction.

290 fore, as confirmed by the pressure measurements, a negligible pressure gradient arises between the
 291 inward part of the mesh and the façade;

- 292 • exit at the large external suction (both on the inner and the outer skin).

293 Figure 15(b) compares the peak pressure distributions for the *FC1* and *FC2* tests on the glazed façade:
 294 focusing on the negative peak pressures, strong suction recorded during the *FC1* test do not occur in the
 295 *FC2* system. The outer porous skin beneficial effect is well demonstrated by the envelopes diagrams of the
 296 peak pressure coefficients as well, shown in figure 16: the filtering effect holds true for both positive and
 297 negative peak pressures.

298 3.5. The Effect of the Mesh

299 3.5.1. Cobra Probe measurements

300 The main issue involved in the Cobra Probe measurements is associated with the dependency of the
 301 accuracy on turbulence levels: the measurements remain relatively accurate for turbulence intensities lower
 302 than 30%. At the height of the boundary layer the Cobra are operating, $\sim 0.4H_{ref,FC2}$, turbulence intensities
 303 are relatively high but lower than such limit. In addition, the proximity of the inner anemometers to the
 304 façades (the distance is approximately equal to 1 cm in model scale) might affect the accuracy of the
 305 measurements. However, such flow field study aims essentially to characterize qualitatively the flow within
 306 the gap.

307 Table 4 summarizes the most significant results, presented as the ratio between the mean of the velocity
 308 components and the mean reference velocity measured at $H_{ref,FC2}$; firstly, the percentage of good data ac-
 309 quired by the anemometer is shown to ensure that the anemometers are correctly oriented. Figure 17 shows
 310 the corresponding configurations. The velocity components *U* and *V* are defined on an ideal horizontal
 311 plane and the first one is associated with the direction along which the sensor is aligned to and *V* to the
 312 perpendicular one. Finally, the vertical velocity component *W* is also measured.

313 Let us first focus on the -120° configuration (Figure 17(a)): the *U* component is always higher than
 314 the others, since the sensors have been oriented according to the expected flow direction. By comparing the

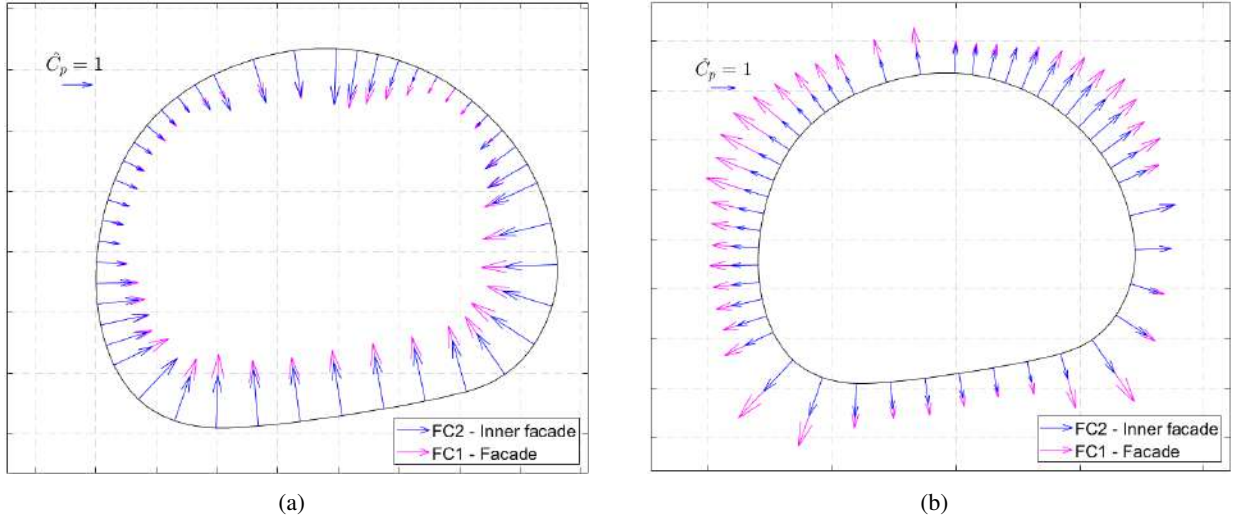


Figure 16: (a) \hat{C}_p and (b) \check{C}_p envelope distributions for the glazed façade: *FC1* test and *FC2* test comparison.

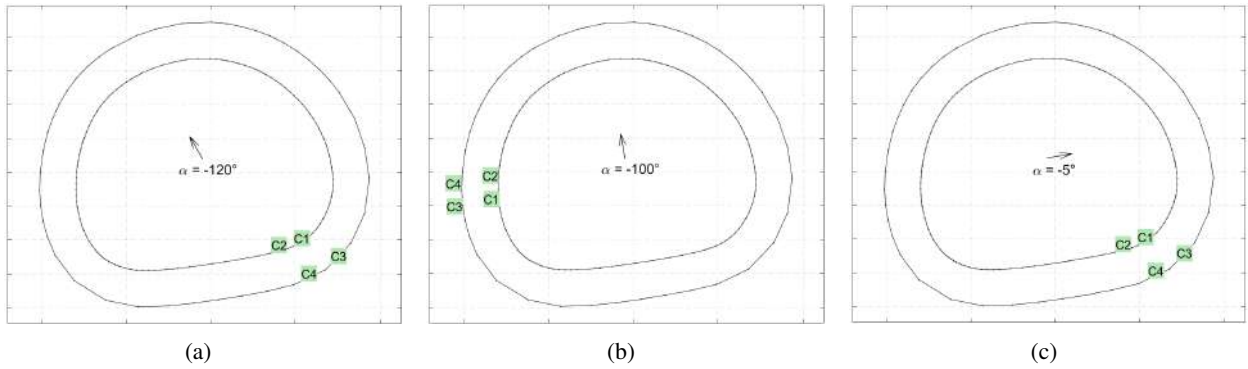


Figure 17: Cobra probes position for incoming wind directions (a) -120° , (b) -100° and (c) -5° .

315 outer anemometers to the ones placed in the gap between the façades, it can be noted that there is an intake
 316 with a drop in the U velocity component passing from outside to the inner gap.

317 Switching to the set up depicted in Figure 17(b), the anemometers have been placed such that they
 318 are parallel to the façade perimeter. We still have that the U component is dominant compared to V and
 319 W values, which are marginally contributing to the total flow velocity. Results confirm that we have flow
 320 running along an ideal line parallel to the inner façade while a negligible flow occurs along the outward
 321 normal with respect to the glazed skin.

322 Finally, referring to the -5° configuration (Figure 17(c)), where the sensors have been oriented again
 323 according to the building perimeter, the U component acquired by the inner anemometers is halved with
 324 respect to the values outside the mesh.

325 3.5.2. Time and frequency domain comparison

326 The effect of the mesh on the acquired pressure signals is further addressed considering the wind direc-
 327 tion -140° , which is significant for both thrusts and suctions. To this purpose, pressure signals recorded on

α		C1	C2	C3	C4
-120°	% Good Data	98	91	100	98
	$U/U_{H_{ref},FC2}$	0.66	0.49	1.18	0.97
	$V/U_{H_{ref},FC2}$	0.04	0.02	0.37	-0.04
	$W/U_{H_{ref},FC2}$	-0.10	-0.08	0.04	0.01
-100°	% Good Data	95	85	100	99
	$U/U_{H_{ref},FC2}$	0.57	0.44	1.10	0.78
	$V/U_{H_{ref},FC2}$	-0.03	-0.04	-0.01	-0.01
	$W/U_{H_{ref},FC2}$	0.02	0.04	0.04	0.05
-5°	% Good Data	93	95	98	100
	$U/U_{H_{ref},FC2}$	0.35	0.34	0.58	0.73
	$V/U_{H_{ref},FC2}$	-0.08	-0.06	-0.03	-0.10
	$W/U_{H_{ref},FC2}$	-0.09	-0.13	-0.05	-0.09

Table 4: Cobra measurement results

328 the façade during the *FC2* stage are compared in both time and frequency domain to the *FC1* ones acquired
329 in the corresponding points.

330 As an example in the thrust region, pressure tap in position T1, as depicted in Figure 18, is selected
331 for *FC1* and *FC2*. When considering the flow-detachment region, T2 sensor position for *FC1* and *FC2* is
332 chosen. The time histories of the analyzed signals $C_p(t)$ are depicted in Figures 19(a) and 20(a) for positive
333 and negative pressure cases, respectively. For the case at hands, the application of the mesh results in lower
334 absolute mean value and standard deviation of the *FC2* pressure signals, e.g. the one acting on the inner
335 façade. In fact, the sharp pressure spikes which are typically recorded on the standard façade, e.g. *FC1*
336 test, are filtered by the metallic mesh and, thus, disappear on the inner façade. This is particularly relevant
337 for the suction case, which is usually the most severe in terms of peak pressures. As reported in Table 5,
338 the standard deviation of the *FC2* pressure signals halves. Being the probability density function (PDF)
339 tails associated with the extreme value analysis (Lou and Peterka, 1981), the influence of the porous media
340 is also studied in that regard. Figures 21(a) and 21(b) show the PDF of the same pressure signals for the
341 thrust and suction cases, respectively. When comparing the *FC1* and *FC2*, the difference in the tails of the
342 distribution and thus in the degree of Gaussianity of the pressure signals is evident. As shown in Table 5, the
343 skewness of *FC2* pressure signals is $\sim 40\%$ lower than the ones estimated for the *FC1* case. This behaviour
344 is due to the application of the porous skin, which effectively filters the oncoming turbulence and therefore
345 determines the different dynamic content of pressure signals recorded on the inner façade during *FC2*.

346 The power spectral density functions (PSD) of the pressure fluctuations analyzed above are depicted in
347 Figures 19(b) and 20(b). As expected, the energy content of the *FC1* signal is higher than the corresponding
348 one recorded on the inner façade within the entire frequency axes. In fact, spectra vary due to the nature
349 of the physical processes that occur near the surfaces in the boundary layer flow and lead to different
350 aerodynamic behavior in the two configurations due to the presence of the mesh.

351 4. Reduction of the Peak Pressure Values

352 4.1. Reduction Factors

353 As the peak pressure distributions are computed for both *FC1* and *FC2* systems, the reduction of the
354 design wind loads for the inner façade of the DSF system can be assessed. Therefore, a reduction factor

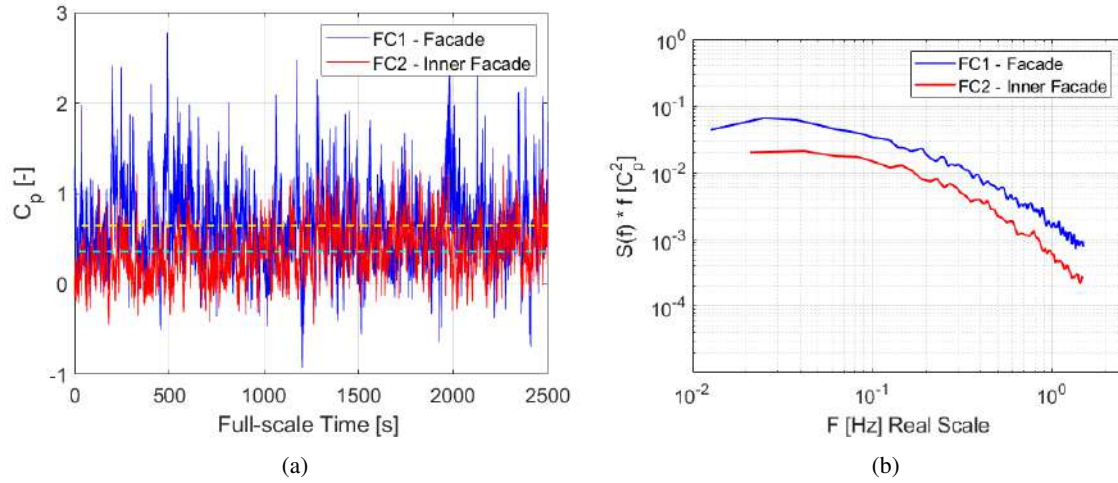


Figure 19: Example of C_p (a) time history and (b) PSD for positive pressure signal, -140° wind direction.

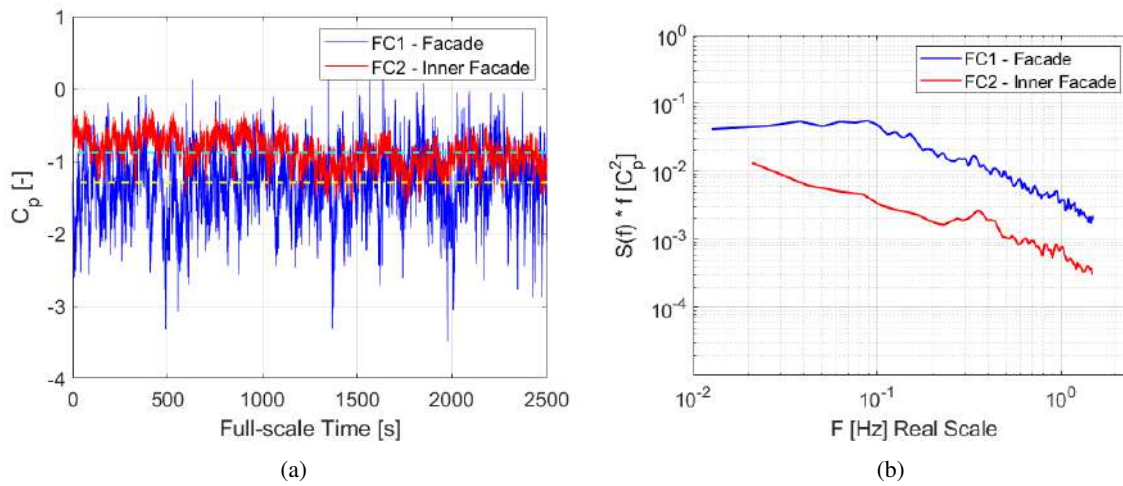


Figure 20: Example of C_p (a) time history and (b) PSD for negative pressure signal, -140° wind direction.

Table 5: Statistics of Pressure Time Series Data: *FC1* and *FC2* comparison.

Tap	Setup	Direction	Mean	Std	Skewness
T1	<i>FC1</i>	-140°	0.643	0.426	0.564
	<i>FC2</i>	-140°	0.360	0.277	0.298
T2	<i>FC1</i>	-140°	-1.293	0.443	-0.346
	<i>FC2</i>	-140°	-0.876	0.211	-0.203

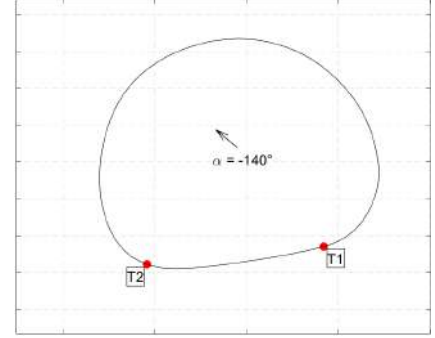


Figure 18: T1 and T2 pressure taps position.

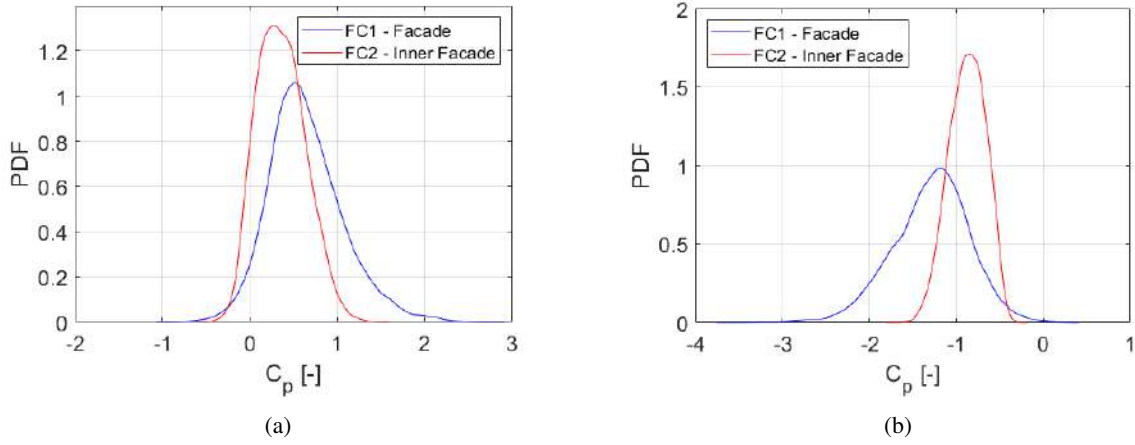


Figure 21: PDFs of signals which exhibit strong thrusts (a) and suction (b): comparison between *FC1* test and *FC2* test.

355 is calculated based on the peak pressures obtained with the standard façade system, e.g. *FC1* stage, by
356 considering the correspondence of the pressure taps locations during the two experimental tests. Making
357 reference to both the positive (\hat{C}_p) and negative (\check{C}_p) peak pressures, the reduction factors are defined,
358 respectively, as:

$$\hat{\eta} = \frac{\hat{C}_{p,FC1} - \hat{C}_{p,FC2}}{\hat{C}_{p,FC1}} [\%] \quad \check{\eta} = \frac{\check{C}_{p,FC1} - \check{C}_{p,FC2}}{\check{C}_{p,FC1}} [\%] \quad (4)$$

359 Since the reduction factors are essential for the design peak pressure, we focus only on non-dimensional
360 peak pressure coefficients $\hat{C}_{p,FC1} > 1.5$ and $\check{C}_{p,FC1} < -1.5$. The reduction factors corresponding to peak
361 pressure coefficients outside such a range are set equal to zero. The $\hat{\eta}$ and $\check{\eta}$ factors are computed based on
362 the positive and negative peak pressure envelopes (the most critical pressure value among all the incoming
363 wind directions for each pressure tap) of the standard façade system. Figure 22 shows the calculated reduc-
364 tion factors. The decrease of peak pressures acting on the inner façade of the DSF system is remarkable for
365 both positive and negative peak pressure envelopes. In addition, the highest values of the reduction factors
366 are found in the regions which correspond to the most severe peak pressure estimated for the *FC1* case.
367 In particular, the strongest suction of the *FC1* test is reduced by 63% if the DSF system is utilized: \check{C}_p
368 increases from -3.21 to -1.18 (Table 6). These findings, which further highlights the filtering features of the

Table 6: Extreme values from envelopes of *FC1* test and corresponding reduction factors.

	<i>FC1</i>	<i>FC2</i>	η [%]
\check{C}_p [-]	-3.214	-1.189	63
\hat{C}_p [-]	2.509	1.581	37

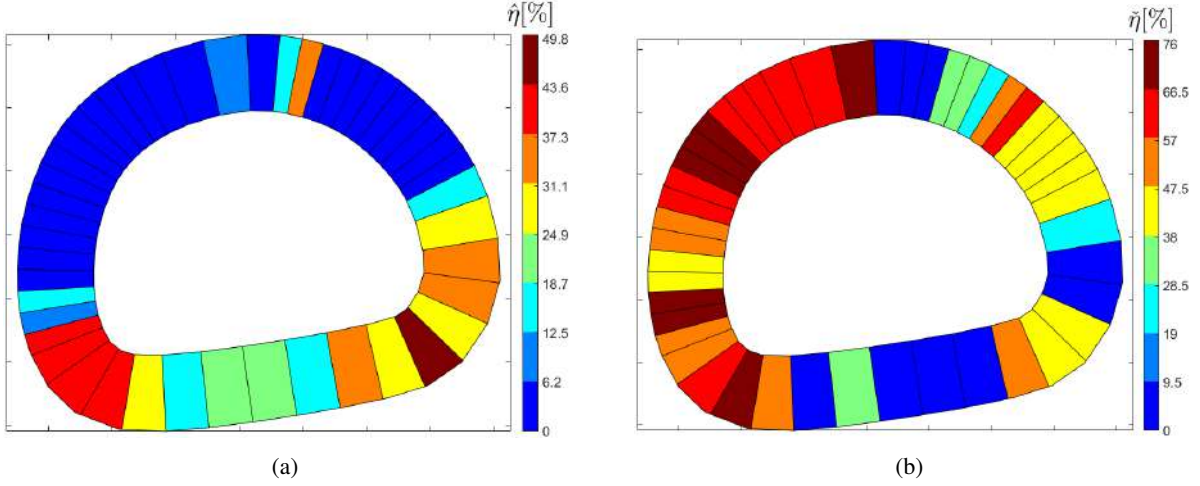


Figure 22: Reduction factors for the positive (a) and negative (b) pressure envelopes for the tested building.

369 mesh, are essential since the peak pressure envelopes are the design tool for the cladding wind loads.

370 4.2. Generalization

371 The computation of the reduction factors is extended to obtain a statistically significant population. To
 372 this purpose, for each pressure measurement point, the peak pressures computed for every wind direction
 373 are considered. The same lower bound of non-dimensional pressure coefficient mentioned above is utilized
 374 here. Figure 23 shows the computed reduction factor based on both positive and negative peak pressures
 375 estimated during the *FC1* test. The effect of the metallic mesh on the peak pressures computed for the
 376 inner façade is relevant. In particular, the higher the absolute value of the *FC1* peak pressure, the higher the
 377 reduction factors are. Hence, a lower bound can be arbitrarily set for the reduction factors of both positive
 378 and negative peak pressures, as depicted in Figure 23. The lower bound is chosen to be described by a
 379 straight line using the following expressions:

$$\hat{\eta} = \begin{cases} 0 & \text{if } \hat{C}_p \leq 1.90 \\ 45(\hat{C}_p - 1.90) & \text{if } \hat{C}_p > 1.90 \end{cases} \quad \check{\eta} = \begin{cases} 0 & \text{if } \check{C}_p \geq -1.80 \\ -35(\check{C}_p + 1.80) & \text{if } \check{C}_p < -1.80 \end{cases} \quad (5)$$

380

381

382 The definition of a lower bound for the reduction factors aims to obtain a qualitative estimate on the safe
 383 side of the peak pressures expected on the inner façade of the DSF system. Figure 23 provides a relevant
 384 design tool for cladding loads, underpinning the prime importance of the wind tunnel tests for the DSF
 385 systems.

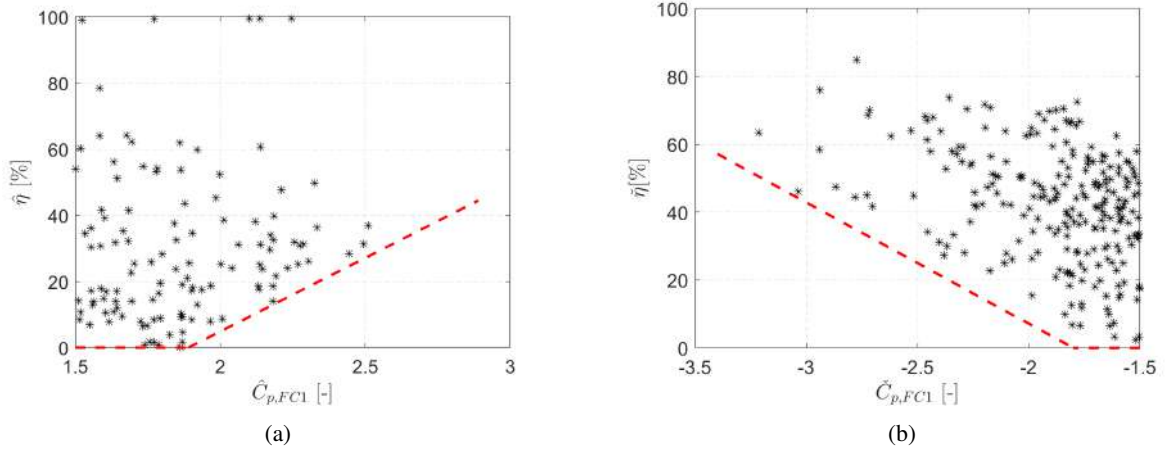


Figure 23: $FC1$ test \hat{C}_p (a) and \check{C}_p (b) and the corresponding reduction factors.

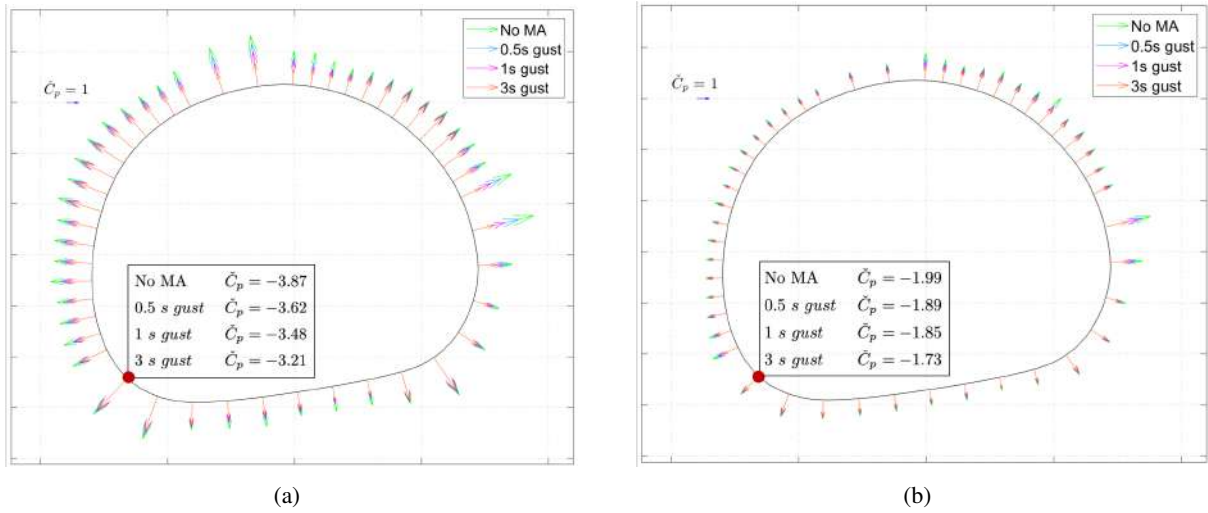


Figure 24: \check{C}_p envelopes for the $FC1$ (a) and the $FC2$ (inner skin) (b) stages, for different values of the averaging time (no moving average filter, 0.5 s, 1 s and 3 s).

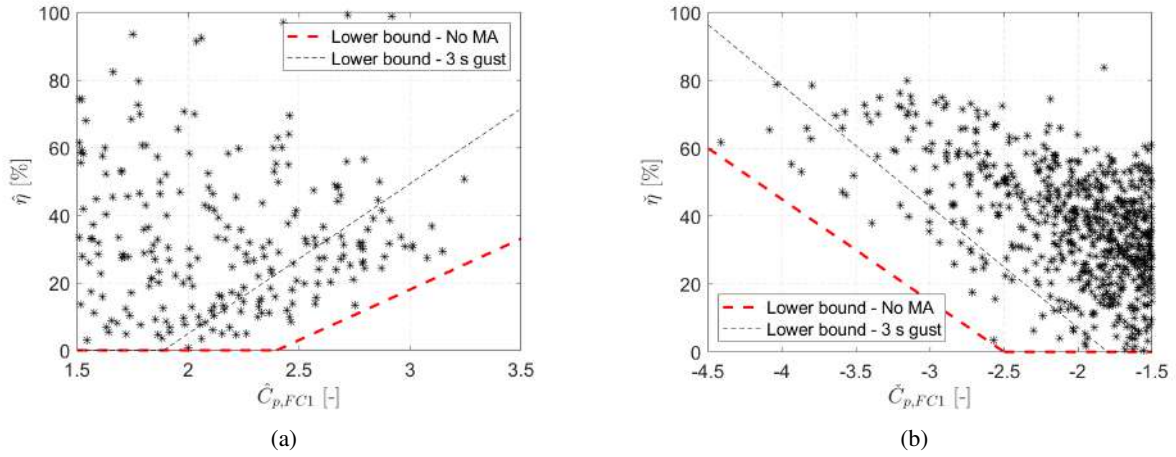


Figure 25: $FC1$ test \hat{C}_p (a) and \check{C}_p (b) with no moving average filter and the corresponding reduction factors. The red dotted lines highlight the lower bound of the point cloud and the black dotted lines correspond to the same limit for the 3 s gust case.

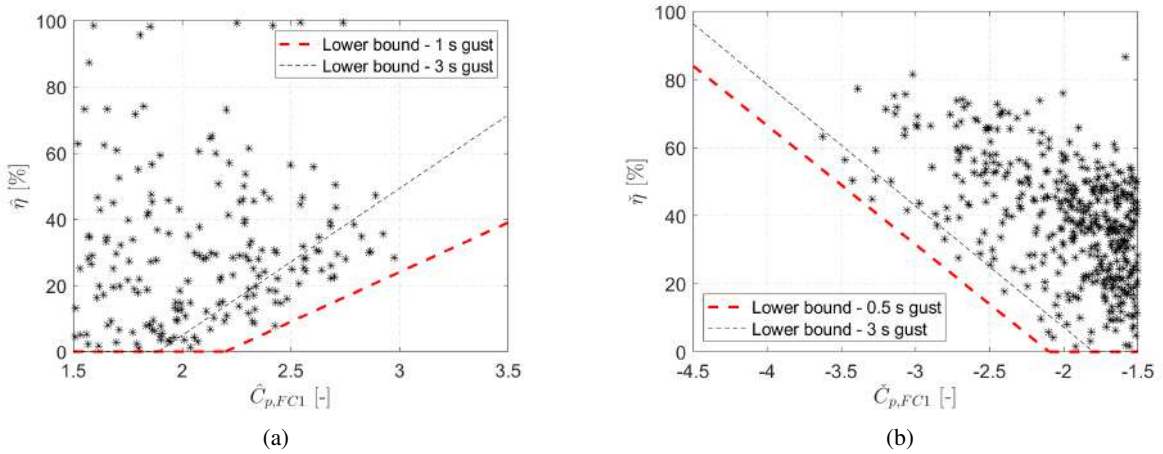


Figure 26: $FC1$ test \hat{C}_p (a) and \check{C}_p (b) and the corresponding reduction factors for the 0.5 s gust case. The red dotted lines highlight the lower bound of the point cloud and the black dotted lines correspond to the same limit for the 3 s gust case.

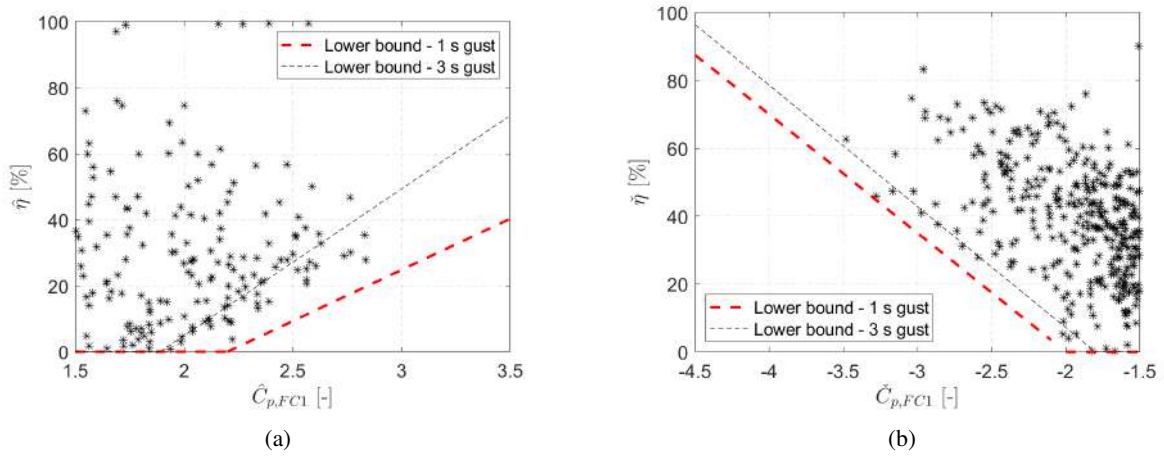


Figure 27: $FC1$ test \hat{C}_p (a) and \check{C}_p (b) and the corresponding reduction factors for the 1 s gust case. The red dotted lines highlight the lower bound of the point cloud and the black dotted lines correspond to the same limit for the 3 s gust case.

386 4.3. Remarks on the 3 s gust

387 All the extreme values analysis results presented up to this point come from the post-processing of the
 388 acquired pressure time series filtered by a moving average operator with an averaging time of 3 s real scale.
 389 This filter is commonly used in the wind tunnel current practice (Amerio, 2018) to account for the lack of
 390 simultaneous correlation in space of pressure signals.

391 Further investigations about the effects of the choice of the averaging time on the peak pressure distri-
 392 butions and the envelope diagrams have been performed, since the existing literature declares a lack of a
 393 fully justified procedure in the determination of the most suitable averaging time (Holmes, 1997) (Amerio,
 394 2018). At this aim, in addition to the 3 s gust , different averaging times have been selected, i. e. 1 s , 0.5 s
 395 and no averaging at all.

396 The analyses previously shown have highlighted how the presence of the outer skin modifies the peak
 397 pressure distributions and the flow conditions on the inner glazed skin with respect to the standard façade
 398 case. It is shown that signals from the $FC1$ have a higher dynamic content, with spikes greater in amplitude
 399 than the ones from the $FC2$ signals. This different "nature" reflects on a different effect of the moving mean
 400 operator: pressure signals from the $FC1$ test are much more affected by the operator than the ones from the
 401 inner façade of the $FC2$ stage. This impacts on the estimation of the extreme values: by applying Cook and
 402 Mayne method to the same signal, previously filtered by the moving mean operator with different averaging
 403 times, the resulting C_p peaks are quite different. As shown by the envelope diagrams depicted in figure 24(a)
 404 changing the averaging time (no filter, 0.5 s , 1 s , 3 s), the difference in length is appreciable and the longest
 405 arrows are more than halved if the 3 sec gust is applied with respect to the extreme values corresponding to
 406 the no averaged signals. The text in Figure 24(a) reports the \check{C}_p values for the tap that experiences one of the
 407 greatest suctions: the difference switching from the first and the last filter (i.e. No MA and 3 s gust) is about
 408 $0.6 C_p$. Figure 24(b) reports the same plot for the inner facade C_p negative peak values: here, switching
 409 from the green to the red arrows, the length reduction is less evident. Focusing on the tap marked in red, the
 410 negative peak pressure coefficient is increased by $0.26 C_p$ applying the 3 s averaging time with respect to
 411 the raw acquired signal. The smaller effect of the averaging filter on the $FC2$ extreme values is mainly due
 412 to the fact that those signals have been affected by the mesh effect and already "smoothed" by it.

413 The different effect of the choice of the averaging time on the peak values has a direct impact also

414 on the *reduction factors*, that have been re-calculated for all the signals treated with the different filters.
 415 Figures from 25 to 27 show the points clouds for positive and negative peak pressures, computed varying
 416 the averaging time. The dashed red line highlights the lower bound of the cloud and the black dashed line
 417 the limit associated with the *3 s gust* filter. For sake of completeness, we list the expressions of the lower
 418 bounds, varying the averaging time:

- 419 • if no moving average filter is applied:

$$\hat{\eta} = \begin{cases} 0 & \text{if } \hat{C}_p \leq 2.40 \\ 30(\hat{C}_p - 2.40) & \text{if } \hat{C}_p > 2.40 \end{cases} \quad \check{\eta} = \begin{cases} 0 & \text{if } \check{C}_p \geq -2.50 \\ -30(\check{C}_p + 2.50) & \text{if } \check{C}_p < -2.50 \end{cases} \quad (6)$$

- 420 • for the *0.5 s gust* case:

$$\hat{\eta} = \begin{cases} 0 & \text{if } \hat{C}_p \leq 2.20 \\ 30(\hat{C}_p - 2.20) & \text{if } \hat{C}_p > 2.20 \end{cases} \quad \check{\eta} = \begin{cases} 0 & \text{if } \check{C}_p \geq -2.10 \\ -35(\check{C}_p + 2.10) & \text{if } \check{C}_p < -2.10 \end{cases} \quad (7)$$

- 421 • for the *1 s gust* case:

$$\hat{\eta} = \begin{cases} 0 & \text{if } \hat{C}_p \leq 2.20 \\ 31(\hat{C}_p - 2.20) & \text{if } \hat{C}_p > 2.20 \end{cases} \quad \check{\eta} = \begin{cases} 0 & \text{if } \check{C}_p \geq -2.00 \\ -35(\check{C}_p + 2.00) & \text{if } \check{C}_p < -2.00 \end{cases} \quad (8)$$

422 Decreasing the averaging time, it appears that the lower bound is horizontally shifted toward higher (in
 423 module) peak pressures, which means that for the same $\check{C}_{p,FC1}$ or $\hat{C}_{p,FC1}$, a lower value of η than the one
 424 corresponding to the *3 s gust* filter should be considered for the design of the inner façade of a DSF system.

425 5. Conclusions

426 The paper presents the experimental methodologies utilized during the wind tunnel tests to assess the
 427 cladding loads for a porous double skin façade (DSF). Using the low-rise buildings of New Bocconi Campus
 428 as a case study, a multi-scale approach is adopted to investigate the aerodynamic effects of the porosity on
 429 the pressure distribution on the inner façade of the DSF system. In particular, the reduction of the peak
 430 pressures, which are relevant for the cladding design, is addressed, leading to the following findings for the
 431 case at hands:

- 432 • Both the positive and negative peak pressures estimated on the inner façade are found to be up to 40%
 433 lower than the corresponding one for standard façade system. In addition, the largest decreases of the
 434 peak pressures acting on the inner façade of the DSF system studied correspond to the most severe
 435 peak pressures estimated when testing the standard façade system. For example, the strongest suction
 436 which dictates the cladding design for the latter case is reduced by 63% (\check{C}_p increases from -3.21 to
 437 -1.18) with the adoption of the porous DSF system designed for the low-rise buildings analyzed. This
 438 finding is essential since the cladding design is generally dictated by a unique value.
- 439 • For the case at hands, additional investigations have been addressed, i.e. velocity measurements and
 440 pressure signals comparison. The former have highlighted the presence of a flow within the gap
 441 between the two façades. The latter has allowed to remark the aerodynamic effects of the porous skin
 442 on the pressure signals acquired on the inner façade are remarkable. In fact, both the absolute mean

443 values and standard deviations of the pressure signals are found to be generally lower when testing
444 the porous DSF system. Such a filtering effect of the mesh is particularly evident for the suction
445 considering the most severe wind direction, where the standard deviation of the recorded pressures
446 is halved. The lower skewness estimated for the inner façade pressures, which is relevant for the
447 extreme value distribution, is also highlighted. Spectra comparison shows a lower energy content for
448 the inner skin pressure signals with respect to the one acquired in the standard façade case.

- 449 • The higher the peak pressures estimated in each pressure measurement point for the standard façade
450 case considering every wind direction, the higher the corresponding reductions are.
- 451 • The dependency of the extreme values of pressures on the averaging time is more evident for the
452 standard façade case for the case at hands. This reflects the filtering effect of the porous skin on inner
453 façade pressure signals, which, thus, exhibit a higher correlation in space.

454 References

- 455 AFNOR, 2008. Eurocode 1: actions on structures - part 1:4: General actions - wind actions.
- 456 Aly, A. M., Bitsuamlak, G., 2013. Aerodynamics of ground-mounted solar panels: test model scale effects. *Journal of Wind
457 Engineering and Industrial Aerodynamics* 123, 250–260.
- 458 Amerio, L., 2018. Experimental high resolution analysis of the pressure peaks on a building scale model facades. Ph.D. thesis,
459 Italy.
- 460 Belloli, M., Muggiasca, S., Rosa, L., Zasso, A., 2016. Experimental study on the aerodynamic behavior of a circular cylinder with
461 perforated shrouding. In: 8th International Symposium on Bluff Body Wakes and Vortex-Induced Vibrations. pp. 1–9.
- 462 Belloli, M., Rocchi, D., Rosa, L., Zasso, A., 2012. Wind tunnel studies on the effects of porous elements on the aerodynamic
463 behavior of civil structures. In: The Seventh International Colloquium on Bluff Body Aerodynamics and Applications (BBAA7).
464 pp. 1132–1141.
- 465 Belloli, M., Rosa, L., Zasso, A., 2014. Wind loads and vortex shedding analysis on the effects of the porosity on a high slender
466 tower. *Journal of Wind Engineering and Industrial Aerodynamics* 126, 75–86.
- 467 Bergh, H., Tjrdeman, H., 1965. Theoretical and experimental results for the dynamic response of pressure measuring systems.
- 468 Cook, N., Mayne, J., 1979. A novel working approach to the assessment of wind loads for equivalent static design. *Journal of Wind
469 Engineering and Industrial Aerodynamics* 4 (2), 149–164.
- 470 Cook, N. J., 1986. Designers guide to wind loading of building structures. part 1.
- 471 da Silva, F. M., Gomes, M. G., 2008. Gap inner pressures in multi-storey double skin facades. *Energy and Buildings* 40 (8),
472 1553–1559.
- 473 Diana, G., Fiammenghi, G., Belloli, M., Rocchi, D., 2013. Wind tunnel tests and numerical approach for long span bridges: the
474 messina bridge. *Journal of Wind Engineering and Industrial Aerodynamics* 122, 38–49.
- 475 Gerhardt, H., Janser, F., 1994. Wind loads on wind permeable facades. *Journal of Wind Engineering and Industrial Aerodynamics*
476 53 (1-2), 37–48.
- 477 Gerhardt, H., Kruger, O., 1997. Double skin glass facades-investigations into the load sharing possibilities. *Proc. of ICBEST 97*,
478 335–339.
- 479 Holmes, J. D., 1997. Equivalent time averaging in wind engineering. *Journal of Wind Engineering and Industrial Aerodynamics*
480 72, 411–419.
- 481 Hu, G., Hassanli, S., Kwok, K. C., Tse, K.-T., 2017. Wind-induced responses of a tall building with a double-skin façade system.
482 *Journal of Wind Engineering and Industrial Aerodynamics* 168, 91–100.
- 483 Kemper, F., Feldmann, M., 2019. Wind load assumptions for permeable cladding elements considering the installation context.
484 *Journal of Wind Engineering and Industrial Aerodynamics* 184, 277–288.
- 485 Kilpatrick, J., Crosbie, E., Jain, A., Kumar, S., 2009. Load sharing on double skin façades.
- 486 Letchford, C., Row, A., Vitale, A., Wolbers, J., 2000. Mean wind loads on porous canopy roofs. *Journal of Wind Engineering and
487 Industrial Aerodynamics* 84 (2), 197–213.
- 488 Lou, J.-J., Peterka, J. A., 1981. Extreme value analysis of peak wind pressures on buildings. *CER*; 81/82-32.
- 489 Mayne, J., Cook, N., 1980. Acquisition, analysis and application of wind loading data. In: *Wind Engineering*. Elsevier, pp. 1339–
490 1355.
- 491 Moon, K. S., 2011. Structural design of double skin facades as damping devices for tall buildings. *Procedia Engineering* 14,
492 1351–1358.

- 493 Overend, M., Zammit, K., 2006. Wind loading on cladding and glazed façades. In: Proceedings.
- 494 Poizaris, H., 2004. Double skin facades for office buildings: literature review. Tech. rep., Report EBD.
- 495 Xiaotong, P., Chen, L., 2010. Cfd modeling for thermal behavior of open-type double-skin facades. In: Intelligent Computation
496 Technology and Automation (ICICTA), 2010 International Conference on. Vol. 1. IEEE, pp. 918–921.