# HYDRAULICS OF RIVER FLOW UNDER ARCH BRIDGES 001360 JUNE 1964 NO. 11 VOL. I unt ighway esearch

J.W.DELLEUR

PURDUE UNIVERSITY LAFAYETTE INDIANA

roject

#### HYDRAULICS OF RIVER FLOW UNDER ARCH BRIDGES

TO:	K. B. Woods, Director Joint Highway Research Project	<b>June 19, 196</b> 4
FROM:	H. L. Michael, Associate Director Joint Highway Research Project	Project: HPS-R-1 (36 File: 9-8-2

The Final Report on the project "Hydraulics of River Flow Under Arch Bridges" is attached. The title of the report is the title of the project and it has been prepared by Dr. J. W. Delleur, under whose direction most of the study has been performed.

This project was initiated in April 1958 and will be terminated upon acceptance of this final report by the cooperating agencies. During the six years of the project many monthly progress reports and seven larger progress reports were submitted. The attached report summarizes all the work performed during the entire project period and is in two volumes with Volume II containing only the figures.

This research has provided information on the backwater effects and energy losses of arch bridge constrictions with rigid boundaries. The information provided should be of great value to highway engineers in evaluating the problems associated with arch bridge crossings of streams.

The report is submitted for the record and will also be sent to the Indiana State Highway Commission and the Bureau of Public Roads for their acceptance of this Final Report on this project, subject of course to their review and comments and any subsequent action required.

Respectfully submitted,

Though I Musherd

Harold L. Michael, Secretary

HIM: bc

Attachment

Copy:

F. L. Ashbaucher J. R. Cooper W. L. Dolch W. H. Goetz F. F. Havey F. S. Hill G. A. Leonards J. F. McLaughlin R. D. Miles R. E. Mills M. B. Scott J. V. Smythe E. J. Yoder

# Funal Report

HYDRAULLISS OF RIVER FLOW UNDER ARCH BRIDGES

# Vol. I

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S. W. Dolleur Professor of Hydraulic Inginesing

Join's Mighuay Rescarch Project

Projest: HPS-R-1(36)

File. 9-8-8

Propaged as Firth of an Investigation

Conducted by

Joint Highway Research Project Engineering Experiment Station Pardue University

in cosperation with

Indiana State Maghway Countration

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Bureau of Public Roads U.S.Lepartment of Controls

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Indiana State Highway Commissio or the Eureau of Public Roads

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June 19, 1964

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# I. PREFACE AND ACKNOWLEDCEMENTS

This report completes a general program sponsored by the State Highway Department of Indiana and the U.S. Bureau of Public Roads at Purdue University on the Ardraulics of giver flow under Arch bridges.

The project has been active from April, 1958, to December, 1963, From the beginning to June, 1961, and from September, 1962 to December, 1963, the project was under the direction of Professor J. W. Delleur. From June, 1961, to September, 1963, it was under the direction of Professor G. H. Toebes. Several reports on different phases of the study have been submitted. In 1958, Mr. S. T. Husain<sup>1</sup>\* reported his work, "Preliminary Model Investigation of Hydraulic Characteristics of River Flow Under Arch Bridges". This was an investigation in a 12 foot long, 6 inches wide tilting flume with smooth boundaries which led, in 1959, to the building of a 64 foot by 5 foot by 2 foot, all steel tilting flume. The design of the 64 foot flume was undertaken by Mr. H. J. Cwen. This, along with the first tests in this flume was described in the report entitled: "Design and Construction of Hydraulic Flure and Backwater Effects of Semi-circular Constrictions in Smooth Rectangular Channel", by H. J. Owen and J. W. Delleur.<sup>2</sup> Mr. A. A. Sooky<sup>3</sup> derived an equation for the discharge through a sharp edged semi-circular constriction for the case of free surface flow through the opening. He also extended Mr. S. T. Husain's preliminary tests in the 6 inch wide flume to include the effect of rough boundaries. The work of Husain, Socky and Owen was summarized in a report entitled, "Hydraulics of River Flow Under Arch Bridges - A Progress Report";3

In 1960 and 1961, Mr. P. F. Biery undertook an investigation of artificial roughness in the 5 foot wide flume and studied single span arch bridge constrictions with free surface flow. The experiment did not include skew, eccentricity or entrance rounding, but tests were made with smooth and rough boundaries. A backwater equation

\*Superscripts refer to references in the bibliography.

was obtained together with a bridge design procedure for free surface flow through the opening, and a procedure for indirect discharge calculation from high water marks. The work was summarized in two reports entitled: "Hydraulics of SingleSpan Arch Bridge Constrictions"<sup>4</sup> and "Discussion on Roughness Spacing in Rigid Open Channels".<sup>5</sup> and both authored by P. F. Biery and J. V. Delleur. These reports were later published in the Journal of the Hydraulics Division of the American Society of Civil Engineers.<sup>6</sup>

In 1961 and 1962, Messrs. S. Lippai and T. P. Cheng investigated the free surface geometry due to dual bridges, wing walls, eccentricity, skaw and arch segments. No formal report was submitted on the works of Messrs. S. Lippai and T. P. Cheng. A summary of this work was presented in the monthly progress reports prepared by Professor G. H. Toebes.

In 1962 and 1963, O. Eikeri undertook the study of the effect of submergence of arch bridge constrictions on backwater superrelevation. His work was presented in a report entitled "Hydraulics of Submerged Arch Bridges"<sup>8</sup>. The experimental data of Mesers. P. F. Biery, S. Lippai and T. P. Cheng were reanalyzed in the fall of 1963. The necessary computer programs and plots were prepared by M. Mushtaq.

Also in 1962, a program of field verification was undertaken. Messrs. T. P. Cheng and J. T. Strange of the Indiana Flood Control and Water Resources Commission investigated over 100 arch bridge sites and selected 10 field verification sites<sup>9</sup>. Air photographs of 9 of these sites were made by the State Highway Department of Indiana. Topographic maps were prepared from these photographs in the Air-Photo Laboratory at Purdue University under the direction of Professor R. Miles.

The research reported herein was performed in the Hydraulics Laboratory, School of Civil Engineering at Purdue University, under the auspices of the Joint Highway Research Project, Professor K. B. Woods, Director and Professor H. Michael, Associate Director.

The present report summarizes the work done during the entire duration of the project. It includes the following parts: a literature abstract, a theoretical analysis, a description of the experimental procedure used in the testing program, and finally the presentation and analysis of the test data.

Regarding the motivation for studying the hydraulic characteristics of arch bridges, the following question could well be asked: How many arch bridges are there in the entire United States? Although this figure is probably unknown, Indiana alone has about 900 arch bridges, over 100 of these are in Mariom County and adjacent counties. This explains the particular interest that has existed im Indiana regarding the hydraulics of arch bridges.

By increasing the backwater upstream from the bridge site many of these bridge constrictions cause additional flooding during high flow; people and valuable property are endangered. In recent years, the problem of minimizing the backwater effect due to bridge constriction has become increasingly important. The highway engineers are faced with a multisided problem; not only do they have to build bridges to convey a specified traffic volume and are safe with respect to flooding, but they also must find the most economic design possibles. It is common knowledge to the highway engineer that a bridge crossing will interfere with the natural flow of the stream and will result in a rise of stage upstream from the bridge and in an increase in velocity through the bridge constriction. The engineers are also aware of alternatives of the inexpensive soil fill compared to a structural span. The optimum design is the shortest span that will not cause damage due to stage increase during serious flood conditions. In order to meet these requirements, exhaustive investigations were undertaken to study the hydraulic characteristics of the water flow through different bridge constrictions.

The other motivation for the study of hydraulics of arch bridges is the indirect determination of flood discharge from high water marks. The existing methods are

not applicable to the arch geometry. Due to the large number of arch bridges in Indiana, particularly on small streams, many of which are ungaged, the determination of peak discharge from backwater observation at bridge site would provide a valuable addition to the knowledge of small watershed hydrology.

The scope of the present research is limited to the study of the backwater effects and energy losses through arch bridge constrictions with rigid boundaries. In other words the effects of scour at the bridge site are not considered. This is not an excessive limitation as a correction curve for backwater with scour is available in the literature<sup>10</sup>.

- 1. Busain, J. T., "Predinanary del Intestigante for all of a state of River Flow Under trob Erddyes", Nathers the as
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- 3. Socky, A. C. Hulan, S. T., Orman H. S. Surger, M. C. Strand, M. S. Strand, M. S. Strand, S. S. Surger, No. 100, Force, Eng. Science, S. S. Surger, No. 100, Force, Eng. Science, S. S. Surger, No. 100, Force, Eng. Science, S. S. Surger, S. S. Surger, S. S. S. Surger, S. Surger, S. S. Surger, S. S. Surger, S. Surger, S. Surger, S. S. Surger, S.
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- Eikeri, C. and Dollaws and the cliss is the state of the
- 9. Cheng, T. " the Bornney T 1 WA meliate study of the Probotype Completion of Humaniaks of firsh of the Source In-
- 10. Bradley, S. M. Bornance of Brady, Steer of J. Discover of Pulla and U.S. Government Prinzing Office (Lynch 1);

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# II. REVIEW OF LITERATURE

To the author's knowledge there is no previous systematic investigation of the hydraulic characteristics of arch bridges in particular There is, however an extensive body of literature on the hydraulics of bridge constrictions, on open channel constrictions in general, on culverts and on the furdamental asystems of the mechanics of contracting and expanding streams.

An annotated bibliography covering some of the most recent literature is given below. This bibliography was compiled by the author as part of his accirity in the Task Force on Hydraulics of Bridges of the American Society of Civil Engineers.

#### I. CONSTRICTION AND BACK '/ TER

BRADLEY, J. N. Use of Backwater in Deaign of Bridge Waterways, Pub. Roads Vol. 30 No. 10 Oct. 1959. pg. 221-6.

"Investigations carried out by Division of Hydraulic Research Bur, Pub. Roads centered on determination of backwater produced by bridges. Scour at bridge abutment, scour around piers, and methods for alleviating scour; research results, design information derived and application of bridge backwater to waterway design; data presented are based on experimental backwater studies using hydraulic model- and field measurements", from Engr. Ind. 1960, pg. 160. For a more detailed discussion see Bradley, "Hydraulics of Bridge Vaterways".

ERADLEY, J. N., Hydraulics of Bridge Waterways. U.S. Dept. of Connerce, Bureau of Public Roads. Govt. Printing Office. 1960

This report gives the hydraulic design criteria for bridge waterway design and for the computation of backwater caused by bridges, and is written for highway departments engineers. Design procedures and illustrative problems are given for normal crossing, dual bridges, skew crossing, eccentric crossing, abnormal state discharge, and backwater with scour. The methods of computing the backwater are based almost entirely on model tests conducted at Colorado State University (see Liu, Eradley and Plate)

CHOM, VEN TE, Open Charnel Hydraulics. McGraw-Hill Book Company, 1959.

In addition to being the most recent and comprehensive text on hydraullos of open channels, Professor Chow's book contains a chapter on "Flow Through Nonprismatus Channel Sections", (Chapt. 17). A detailed summary (Art. 17-16) with complete set of figures of the work of Kindsvater Carter and Tracy, (reproduced from U.S.C.S. circular 284, 1959) gives all the necessary information to calculate the discharge through constrictions, or the backwater ratio due to bridge constrictions.

CRACE/ALL, JR. J. Indirect Methods of Discharge Measurement. Proceedings 6th Hydraulics Conference 1955 State University of Iowa. Studies in Engineering-Bulletin 36, 1956.

The 1951 flood in the Kansas river basin is taken as an example to discuss the necessity for indirect discharge measurements. The indirect methods are classified in four groups: 1) the slope area, 2) the contracted opening, 3) the flow through culvert, 4) the flow over dam. Each method is discussed briefly.

For a more detailed discussion of indirect discharge measurement at bridge constrictions see Kindsvater, Carter, and Tracy, "Computation of Peak Discharge at Contractions".

HENRY, H. R. Discussion on "Backwater Effects on Open Channel Constrictions". Trans. ASCE, 120, 1013-1017 (1955)

By simultaneous use of the momentum equation and the specific energy relation both in dimensionless form, Mr. Henry obtains graphically a theoretical solution for the backwater ratio, thus eliminating the trial and error calculation required by the method of Tracy and Carter (see also Ven Te Chow, Open Channel Hydraulics, example 17-3). The effect of the roughness on the decrease of momentum occasioned by boundary shear in the zone of expansion downstream of the constriction is compared to the roughness effect on the loss of energy on the contracting flow upstream of the constriction.

IZZARD, C. F. Discussion on "Tranquil Flow" by C. E. Kindsvater and R. W. Carter. Trans., ASCE, Vol. 120, pg. 985-89. 1955.

The experiments of Kindsvater and Tracy on "Tranquil Flow through Open-Channel Constrictions", were limited to the case of a horizontal bed. Izzard extends the analysis to the case of sloping channels, and making use of experimental data of Tracy and Carter ("Eackwater Effects of Open-Channel Constrictions") he shows that the backwater of the constriction may be expressed approximately as the product of a velocity head (velocity at normal depth in downstream section of constriction) times a coefficient which depends primarily on the contraction ratio of the constriction Whereas Kindswater and Tracy were concerned with the problem of estimating the discharge from measurement of water levels in the vicinity of a channel constriction, Izzard is concerned with the reverse problem of estimating the backwater caused by a channel carrying a known flow at normal depth.

IZZ RD, C. F. Discussion on "Backwater Effects of Open Channel Constriction" by H. J. Tracy and R. W. Carter. Trans. ASCE Vol. 120 pg. 1008-13. 1955.

The paper by Tracy and Carter is analyzed from the highway envineer viewpoint which is that of calculating expected backwater elevations due to floods of various frequencies. As the accuracy of the flood peak estimates is seldom better than a t 20%, simplifications may be introduced in the backwater calculation. The following simplification is proposed by Izzard. Neglecting minor effects (roughness, and length of constriction) the ratio of the maximum backwater depth upstream of the constriction to the normal depth in the unconstricted channel may be correlated to the contraction ratio and the velocity head in the constricted section. The velocity head in the constricted section is based on the area at normal depth.

IZZARD, C. F. and BRADLEY, J. N. Field verification of model tests on flow through highway bridges and calverts. Proc. 7th Hydr. Conf. Iowa Inst. of Hydr. Res. June 1958, Iowa City, Iowa, State University Iowa 1959, pg. 225-43. AMR 13-4641. Sept. 1960.

The paper reports on comparison of prototype measurements with computed values, derived from model tests for backwater caused by birdges, scour at bridge abutments, and head-discharge characteristics of culverts. Computed and measured values of the drop in water level across the bridge embankment is given for ten sites, two of which are for submerged deck girder. The smallest error is 0.5%, the largest is 13%. KINDSVATER, C. L.; CARTER, R. '.; TRACY, H. J. "Computation of Peak Discharge at Contractions. USGS - Circular 284, 1953.

This report gives a procedure for computing subcritical peak discharges at contractions based on laboratory studies. The discharge formula given includes a discharge coefficient which may be obtained from sets of curves which contain the essential geonetric and hydraulic factors governing the flow at a constriction. The factors are: the contraction ratio, the relative length of the abutment, the Froude number, the antrance rounding, the abutment chamfers, the angularity of the constriction, the side lepth at abutments, the side slops of the abutment, the eccentricity of the constriction, the bridge submergence, and bridge piles and piers. The primary variables are the contraction ratio and the relative length of the abutments. Standard values of the discharge coefficient as a function of the primary variables are given for four types of geometries: 1) vertical embankments and vertical abutments; 2) sloping mbankment and vertical abutments; 3) sloping embankments with sloping abutments; .) sloping embendments with vertical abutments with wing walls. The standard value of the discharge coefficient is then multiplied by adjustment coefficients, which take into account the effect of the remaining variables, to obtain the discharge coeffisient. Detailed field and office procedures are given.

See also: Windsvater and Carver, "Tranquil Flow through Open-Channel Constrictions" -, for a summary including working curves see Ven Te Chow, "Hydraulics of Open Channels".

INDSVAT R, C. E. and CARTER, R. H., "Tranquil Flow through Open-Channel Constrictions". Trans. ASCE 120, 955-992, 1955.

A practical mithod of solving the discharge equation for tranquil flow of water through open channel constrictions is described. The functional relationships between the coefficient of discharge and the principal independent variables (contraction ratio, length of contractior, Froude number, entrance rounding, eccentricity of opening, angularity and guide walls) is presented from experimental data. Boundary conditions considered include vertical constrictive elements, channel cross-sectional shapes and roughness patterr. Within the limits tested the proposed computation procedure yields satisfactory results. Tests were run in a horizontal flume. See also the discussion by C. F. Izzard. (adapted from author's conclusions).

ANS, E. M. "Experiments on the Flow of Mater through Contractions in an Open Channel. Trans. ASCE, 83, 1149 (1919 - 1920).

This is probably the first laboratory study on open channel constriction in the I.S. The Froude numbers used are higher than those usually found in bridge waterways. There is a limited number of boundary shapes. Experiments were made in four different "low contractions, namely rounded-edge plate, sharp edge plate, short flume, Venturi "lume. Coefficients of discharge were first computed using D'Aubuisson's and "eisbach's "cornulas. Based on the experimental data a general equation of contraction was developed "he results of flume tests were presented in detail and discussed.

IU, H. K; BR DLEY, J. N.; PLATE, E. J. "Backwater Effects of Piers and Abutments." Pept. CER 57 HKL 10; Colorado State Univ. 1957.

This is the final report of a project undertaken at Colorado State University in cooperation with the Eureau of Public Reads, U.S. Dept. of Commerce. Maximum backwater lue to bridge constriction is given for simple normal crossing, abnormal stage-discharge condition, dual bridge crossing, skew crossing, eccentric crossing piers, partially submerged bridge girders. The water surface profiles, coefficient of contraction, komation of maximum backwater, are presented. This report is perhaps the most compreuensive work on hydraulics of bridge constrictions in the American technical literature. NAGAI, S., On the two-dimensional analyses of suddenly contracted flows. Houille Blanche. Nov., 1950. Pg. 662-73. AMR 7-1849

Position of contraction in suddenly contracted flows is determined by use of the Schwarz-Christoffel theorem. Study is made for flows about a suddenly contracted pipe and about that having a round corner. The position of contraction is approximately given with determination of the stationary points of vortex center. Corner radius can be found by which vortex till vanish.

TRACY, H. J., C/RTER, R. W. Backwater Effects of Open Channel Constriction. Trans. ASCE 120, 993-1018. 1955.

A method of computing the backwater due to open channel constriction is given. It is based on ampirical discharge coefficients and on a laboratory investigation of channel shape, constriction geometry and influence of channel roughness. The solution involves the computation of water-surface drop through the constriction and the determination of a factor which is the ratio of backwater to water surface drop. This ratio is shown to be a function of channel roughness, per centage of channel contraction, and constriction geometry. See also the discussion by Insard (adapted from author's synopsis). ALBERTSON, M. L., JENSEN, R. A., DAI, Y. B., ROUSE, H. Diffusion of submerged jets. Trans. ASCE Vol. 115 (1950) pg. 639-697.

This paper deals with the turbulence generated at the edges of a free air jet issuing from orifices and slots. Results are given of measurements of the distribution of the longitudinal velocity and of turbulence characteristics. Of particular interest to bridge hydraulics is the discussion by N. R. Henry (1950) on flow under sluice gates.

ARCHER, W. H. Experimental Determination of Loss of Head dusto Sudden Unlargement in Pipes. ASCE Trans. Vol. 76 (1913) pg. 999.

Sudden expansion head loss:

$$h = 1.098 (\overline{U}_{m1} - \overline{U}_{m2})^{1.919}$$
2g

where Unl and Um2 are the mean velocities in the upstream and downstream conduits.

BLANCHET, C. Sur Le Probleme Des Renous Et Des Pertes De Charge Produits par Les Singularities Dans Les Canaux Et Rivers. Houille Blanche No. 1, Nov. 1945 -Jan. 1946. pg. 39-62.

"Problem of turbulence and loss of head caused by obstructions in channels and rivers; turbulence curves related to curves of specific energy; influence of sill at channel bed and of contraction of cross section area on kin tic energy; results of study can be applied only to channels or rivers of stable bottom not liable to form deposits". From Engr. Ind. 1946 pg. 428.

CHATURVEDI, M. C. Flow Characteristics of Axisymmetric Expansions. Jour. of the Hydraulics Division ASCE, Vol. 89-HY3; May 1963.

Characteristics of flow for four abrupt expansions which half angles of 15°, 30°, 45° and 90° have been determined by a combination of analytical and experimental means. The transformation of mean energy, the rate of production of turbulence, and the rate of dissipation of turb lence energy are determined. The evaluated head-loss results from the air-duct studies are shecked by independent measurements in a water-pipe assembly. The kinetic energy of the mean motion, kinetic energy of turbulence, pressure distribution, turbulence production, and turbulence shear are presented in the form of their spatial distribution, for all expansion angles. Head loss in abrupt expansions is given for different angles of separation; for 90° (half angle) the head loss coefficient is practically equal to that obtained in the assumption of constant pressure at the inlet section and one-dimensional analysis ( p. 80 and Fig. 13). The variation of the head loss with changing expansion ratio is considered (p. 89 and Fig. 20 DAVIDIAN, J., C/RRIGAN, P. H. JR., SHEN, J. (USGS) Flow through Opening in Midth Constrictions. Water Supply Paper 1369-D, 1962, p. 91-122, plate.

"Flow pattern at constrictions with 2 to 7 openings; laboratory experiments and analysis were diffected toward development of methods for computing discharge through multiple opening constrictions, apportioning given total discharge among several openings and predicting backwater caused by constrictions". Engr. Index 1962.

DELORME, A. Considerations Sur Les De Bouches Des Petits Duvrages Sous Routes. Annales Des Ponts Et Chaussees. Vol. 129. No. 2. March-April 1959. p. 141-67.

"Water passages under small road bridges; size of underpass is often overdimensioned, because formulas of calculation use too high safety factor; it is recommended that permissible factor of incidence of submersion should be defined by considering possible damage due to submersion; calculations should be based on flood statistics, also law of large values; practical applications." Engr. Ind. 1959, p. 171.

GIBBINGS, J. C., DIXON, J. R. Two-Dimensional Contracting Duct Nerw. Quart. J. Mech. Appl. Math. 10, 1, 24-41; Feb. 1957. AMR 10-2954. Sept. 1957.

This paper is primarily for wind tunnel design dualing with potential flow through two-dimensional contracting channels of finite length. Method of eliminating adverse velocity gradient along channel wall was presented in great detail. / numerical example is given.

HENRY, H. R. Discussion on Submerged Jets. Transactions, ASCE, Vol. 115 (1950) p. 6.

This is a discussion to a paper by M. L. Albertson, Y. B. Dai, R. A. Jensen and H. Rouse on diffusion of submerged jets. The effect of boundary conditions different from those used in the main paper are investigated. These, for the case of the flow under a sluic gate, are the effects of the free surface i.e. gravitational effect and the presence of a solid boundary, i.e. the flume floor, instead of a plane of symmetry of a two-dimensional jet.

Hydraulic design conclusion: Experimental discharge coefficients for the flow under a sluice gate are given in terms of a dimensionless gate opening, a dimensionless tailwater depth and the orifice Froude number (Fig. 35, p. 691). The expansion of the jet (which is limited by a fixed lower boundary and by a free surface upper boundary) is reported to have a slope of 1 on 6 approximately.

KINDSVATER, G. E. Energy Loss Associated with Abrupt Enlargements in Pipes. USGS. Water Supply Paper 1369-B-1961.

In connection with the USGS study of hydraulics of bridge-waterways, an experimental investigation was made of the flow of water through abrupt concentric enlargements in circular pipes. Particular attention is paid to the influence of pipe-wall roughness and to methods of computing the energy loss resulting from unlargements. The conventional method of computing the energy loss from the Borda-Garnot equation was found adequate for practical use. PETERS, H. Conversion of Energy in Cross Sectional Divergences Under Different Conditions of Igflow". NACA TM 737, 1934.

Hydraulic design conclusion: Super imposing a spiral motion on the flow entering an expansion increased the efficiency, as the rotational motion delays the separation.

ROUSE, H. Energy Transformation in zones of separation. Proceedings 9th IAHR Conf. Durbrounik, 1967.

The integrated equations of momentum and of mean energy are examined for the case of two-dimensional flow over a normal wall, and attention is given to the variablen of the sum of the terms of Bernoulli equation along a streamline in either the primary flow or the zone of separation.

This paper establishes some of the theoretical background on the mechanism of flow separation used in the experimental studies of Chaturvedi (1963)

ROUSE, H. Repartion de l'energis dans les Zones de Decollement. La Houille Blunche No. 3, May 1960 and No. 4, June, 1960.

Distribution of energy in regions of separation

Described in the present paper is the determination of the mean and secondary patterns of arisymmetric flow for two comparable boundary forms: the abrupt inlet and the blunt shaft.

Measurements available for analysis included the distributions of mean velocity, mean pressure, longitudinal and radial intensities of turbulence, turbulent shear, the longitudinal intensity gradient. Through use of the equations of mementum and of energy for the mean and the secondary motion, the measured distributions were adjusted to yield the required balance of the essential terms in the equations, thus yielding results in general accord with physical requirements. These are presented in the form of the flow patterns. (from outpor's statract)

TULTS, H., Flow expansion and pressure recovery in fluids. Transactions ASCE Vol. 121, 1956, Pg. 65. (AMR 8-2394. Aug. 155)

Investigating the possibility of improving the pressure recovery in flow expansions, author observed visually the separation phenomena and measured the development of velocity profiles and pressure distribution in a gradual unilaterally expanding two-dimensional rectangular test canal with varied divergence from zero degrees to 20 degrees in range of Reynolds number  $5 \times 10^4$  to  $3 \times 10^5$ .

A simple analytical dependency is established between the angle of divergence and the rate of separation at which the maximum pressure recovery in each section occurs. This permits us to predict the optimum divergence for any required rate of gradual expansion. VALLENTINE, H.R., Flow in rectangular channels with lateral constriction plates, Houille Blanche 13, 1, 75-84, Jan.-Feb. 1958. (AMR 12-834.)

Characteristics of flow in a rectangular channel with sharp-edged lateral constriction plates placed symmetrically, normal to the flow, are examined in a small tilting channel. The flow rate Q is related to the upstream depth yiby means of a discharge equation  $Q = Gby_1^{3/2}$ , where b is the width of opening and C is an experimental coefficient which depends upon the constriction ratio and the Froude number of the unconstricted flow. The values of C are established for Froude numbers up to 2.1 and constriction ratios up to 95%.

The conditions under which insertion of constriction plates produces an increase in upstream depth are investigated and the extent of the increase is evaluated. (From author's summary) BENJAMIN, T. B., On the flow in channels when rigid obstacles are placed in the stream, J. Fluid Mech. 1, 2, 227-213. July 56 (MR 11-503)

Author uses three physical quantities 0, R, S introduced in a previous paper (see Benjamin, T. B., and Lighthill, "On choid I waves and bores", <u>Proc. roy. Soc.</u> (A), 224, 448-440, 1954) with the meaning

Q = vh,  $R = \frac{1}{2}v^2h$ ,  $S = v^2h + \frac{1}{2}gh^2$ 

i.e., Q is the flow rate, R the energy for unit lass, and S the accentum flow rate for unit span (corrected for changes in horizontal pressure force due to changes in depth) and divided by density. Invariability if those quantities at different cross sections implies, respectively, conservation of flow rute, of energy, and of momentum.

This present paper is divided in five parks: 1-Introduction; 2-General theory of flow; 3-Flow under a sluice gate; 4-Then under a planning correace; 5-Experimental tests.

Parts 1 and 2 deal with one properties of flow and verse. In the last three parts, author studies the flow under a write of writing of rolling gets. This subject has also been treated in the Italian place of Gentlini <u>Energia elect</u>. April and **June 1941** and hareh <u>Ann. Mat. pure appl.</u> (1V) 54, 1953.

BINNIE, A.M., The flow of wat r un et sluids gate, "mari J. Mochn. appl. Math 5, Part 4, 395-407, Dec. 1352, (MR 6-2256)

By introducing the Fronde number of the first submor simplifies relationship between conditions upstream and deviations of first systes. Well-known expressions are derived for alternate depth of first data are and first in the gate. A simple explanation is presental for the absorpt of the or opth sides of the gate, Coefficient of contract. The absorpt of the first is given as a review of experimental and theoretical investigation. Actually, and are dependention for, but presents instead the gate opening requires to produce five reode number.

BIRKHOFF, G. Calculation of Potential slow with Free Streamlines. ASCE Jour. Hydr. Div. Vol. 87-HT6. Nov., 1961.

The most recent survey of the methods for computing these classes of steady potential flows with free stranlines: (1) plane flows having free boundaries and curved fixed boundaries without gravity; (2) plane flows having free boundaries and straight fixed boundaries with gravity and (3) exially symmetric flow having free boundaries without gravity.

Discussion by H. Rouse (Narch 1962, p. 187); L. Landweter (May, 1952 p. 203); S. P. Garg, T. S. Strelkoff and Y. S. Yu (July 1962, p. 293), and the closing discussion by G. Birkhoff, (March 1963, p. 147) point out the limitations of the mathematical methods; and comparisons between theoretical calculations and experimental results are given. The paper along with its discussions provides an excellent source of references on the subject. BLISDELL, F. ". Comparison of Sluice Cate Discharge in Model and Prototype, Trans. ASCE Vol. 102 p. 544. 1937.

"(1) Models of sluice gates can be depended upon to predict the discharge of their prototypes with reasonable accuracy, (2) Froude model law will apply to the discharge of sluice-gate, (3) Roughness of both model and prototype must be given consideration in the construction of the model; and (4) good results should not be expected for small gate-openings and low velocities". (From Authors Conclusion.)

FRANKE, P., Jet contraction in flow under cludet gates in rectangular channels. Bautechnik 32, 8, 257-259, Aug. 55. (ATR 9-2240)

Analytical study is these on energy of rederations of the flow under a sluice gate in a rectangular channel. Futhor computes the downshream depth in the case where energy losses are neglected and in the case there they are taken into consideration. Values of the coefficient of contraction are given for different ratioes of upstream depth to gate opening. All results are end on the dimensionless form.

FRANKE, P., Theoretical consideration of procession in flows under a sluide gate. Bautechnik 33, 3 73-77, Farch 56. (AMP 5-565)

Author analyses the contraction of the rational dramatissuing from the opening of a slulesgate. In its first that the analyses use is made of the results of von Mises who did but consider gravity in flow from a tax-dimensional opening. The computed westite that the pare will think a emperimental data given in the paper.

The only exact (in the sense that no composite in filter of invotation hivy is made) solution for the slutca-gate public is that of controll and Valeey Phil. Trans. (A) 240, 117-161 which public consists in tertor agreements with author's experimental curve that any of his sector obtained from calculation.

GENTILINI, B. Flow under incline of real and the getes (E.S. Translation %1-3; 11 pp., Nov. 195., (AMR 5-1242)

Theoretical and experimental determination of the mails of coefficient of discharge for inclined and radial sluice gates. Experimental charmel was 16 cm wide, and angles of inclination of 15°, 20°, 45°, 60°, 7°° and 0° word used. Opening ranged from 3 cm to 9 cm. Results closely checked theoretical coefficients that were derived from the specific energy relation.

ROUSE, H., and ABUL-FETOUR, A. H., Charact mistics of irrotational flow through axially symmetric orifices. J. Appl. North 17,4, 121-426, Dec. 1950 (AMR 4-3279.)

Although an exact analytical solution of the orifice problem has not yet proved feasible, use of the relaxation include has permitted a numerical determination of flow characteristics to be redered with sufficient precision for the problem to be considered solved. The coefficient of contraction as found to be practically identical with that evaluated by you block for two-dimensional flow from slots over the entire range of area ratio, and reasonable agreement is shown to exist between measurement and computation. Coordinates of the jet profiles are presented in tabular and graphical form, and are found to differ appreciably from those previously adapted from the two-dimensional case. A composite dimensionless chart is also provided showing the distribution of pressure along the boundary and center line and across the efflux section for the various area ratios. (From authors summary.)

#### CHAPTER III THEORETICAL ANALYSIS

## III-1 Flow Configuration at an Open Channel Constriction

The backwater due to a constriction in an open channel depends mainly on the boundary geometry, the discharge and the regime of the flow, i.e., the Froude Number. The phenomenon is usually so complicated that the resulting flow pattern is not entirely subject to analytical solution. A practical solution is only possible through experiments.

In the natural streams, the Froude Number of the flow usually ranged free 0.1 to 0.5. In this range of Froude Numbers the constriction will induce a pronounced backwater effect that extends a long distance upstream. A critical control section may exist at the constriction. When a critical control section exists at the constriction, a hydraulic jump occurs downstream of the constriction.

When a constriction is introduced into an otherwise uniform, friction-controlled flow in a prismatic channel of mild slope (see Fig. 3-1), a backwater of the M-1 type is developed upstream of the constriction. Upstream of the backwater curve the flow is undisturbed and the flow distribution is governed by the channel characteristics. Approaching the constriction, the flow decelerates as the depth increases. The deceleration process continues until a section (designated section (1)) is reached. Farther downstream the flow begins to accelerate owing to the convergence of the flow into the contracted opening (section 2). The flow pattern between sections 1 and 2 is essentially dominated by the constriction geometry.

As the fluid passes through the opening, the live stream contracts to a width somewhat less than that of the opening and the corresponding average longitudinal profile drops sharply. At the section of minimum (section 3) the expansion process begins and continues until normal flow again is established at some distance downstream (section 4).

The boundary roughness plays an important role in the downstream pattern of motion, especially when the flow just downstream of the constriction becomes supercritical. The roughness may force the hydraulic jumps to move upstream, and may eventually reach the downstream face of the constriction.

When the constriction is submerged, the centerline profile and plan view are as shown in figures 3-2 and 3-3 respectively. In figure 3-2 the approaching water surface is shown to have a profile of an  $M_1$  curve;  $Y_0$  or  $Y_n$  is the normal depth of the unobstructed channel flow,  $Y_1$  is the depth at the section of maximum backwater,  $Y_2$  is the downstream depth taken in the deadwater flow portion of the channel, and  $Y_3$  is the depth of the live stream at the vena contracta. The flow may be fully **Cubmerged as** shown by the solid line or the discharge may be free at the downstream end of the constriction. The plan view of the channel constriction flow in figure 3-3 shows the streamlines of the flow entering the opening. The velocity upstream of the constriction is low and the streamlines follow the walls very closely. The eddies sketched on the upstream side of the opening moves in a vertical plane in a screw motion towards the opening. A particle in the corner would not remain in this region but due to these eddies it moves towards the opening. The eddies downstream of the constriction, shown on both sides of the jet, are in the horizontal plane and are less active. Particles could often remain there for longer periods of time.

### III-2 Basic Equations

The three basic equations of fluid machanics: conservation of mass, conservation of momentum, and conservation of energy are used in the following study. These are applied to a definite volume called the control volume (c.v.) and the boundary of this volume is known as the control surface (c.s.). The shape of the control volume remains constant, but the amount and identity of the fluid in it may vary in time.

The flow configurations in the problem at hand are strictly three-dimensional. However, in order to simplify the analysis, a one-dimensional flow in which the velocity and the depth of flow vary only along the path of the channel is often used. In addition, the flow is that of a real fluid, that is to say that friction is present; toundary layers exist and flow separation may occur. The flow is thus rotational. The rotationality of the flow and its three-dimensional characteristics preclude the unit of potential flow theory. The free streamline theory, so ably reviewed by Birkhoff<sup>(1)</sup>(1961) is not applicable.

The errors introduced by using a one-dimensional analysis may be eliminated by using suitable correction coefficients. However, in practice these coefficients are not known exactly, and even in the laboratory, their evaluation requires timeconsuming measurements and calculations. It is, therefore, often customary to use the one-dimensional analysis without correction factors, realizing the limitations of such an analysis. The approximations so-obtained are nevertheless useful for engineering design; their accuracy is consistent with that of the data of the problem. It must be recognized that in the design of a bridge waterway opening, the design flow obtained from by bologic computation is selders known exactly and errors of  $\pm 20\%$  are common. Furthermore, all the irregularities of a natural water course cannot be taken into consideration, and average values or typical cross sections of the stream are considered. In the reverse problem of calculating a flood discharge from high water marks, the several inaccuracies inherent to the high water mark measurement and changes in roughness coefficient at high stages make it difficult to evaluate the discharge closer than within 10 to 15%.

The conservation of mass is expressed by the continuity equation which for a steady incompressible flow takes the form

in which the integral of the net efflux is taken around the whole control surface (C.S.). This equation simply states the inflow into the control volume (C.V.) is equal to the outflow from the control volume, and its one-dimensional form takes the usual form

$$V_1 A_2 = V_2 A_2$$
 (3...2-2)

The conservation of momentum for steady flow states that the sum of the external forces on the control surface and of the body forces (usually gravity) acting inside the control volume is equal to the net rate of efflux of momentum across the control surface:

$$\vec{F}_{S} + \iiint \vec{B} p d v = \iint \vec{\nabla} (p \vec{\nabla} \cdot \vec{d} \vec{\Lambda}) \qquad (3-2-3)$$

$$C.S.$$

where  $\overline{F}_{s}$  represents the external surface forces and B is the body force per unit of mass. The one dimensional counterpart of this equation is

$$F_{\mathbf{x}} = \beta Q \left( \beta_2 \nabla_{\mathbf{x}2} - \beta_1 \nabla_{\mathbf{x}1} \right)$$

$$(3-2-4)$$

where f is the momentum correction factor which takes into account the non-miformity of the velocity distribution

$$\beta = \frac{1}{A} \iint (u/V)^2 dA \qquad (3-2-5)$$

The momentum equation applied to a straight open channel reach is:

$$PQ(\beta_2 V_2 - \beta_1 V_1) = P_1 - P_2 + W \sin \Theta - F_f \qquad (3-2-6)$$

where P1 and P2 are the resultant forces acting on the end sections of the reach

W is the weight of water between the sections 6 is the angle of inclination of the channel

Ff is the external friction force exerted by the boundaries on the fluid

The friction force Ff may be evaluated from Manning's formula written for the boundary shear stress

$$T_{0} = \frac{14.5 \text{ n}^{2} \rho v^{2}}{\text{R}^{1/3}}$$
(3-2-7)

where  $\mathcal{V}_{o}$  is the average boundary shear and R is the hydraulic radius.

The specific force equation is obtained by applying the momentum equation to a short prismatic horizontal open channel reach, neglecting the friction force and assuming a hydrostatic pressure distribution:

$$\beta_{1} \frac{Q^{2}}{gA_{1}} + \tilde{I}_{1} A_{1} = \beta_{2} \frac{Q^{2}}{gA_{2}} + \tilde{I}_{2} A_{2} \qquad (3-2+8)$$

or

where  $F_g$  is the specific force, and where T is the head over the center of gravit, of the cross section.

The conservation of energy for steady flow may be written

$$\iint_{C.s.} (\nabla^2/2g + Z + p/\gamma) (\forall \nabla \cdot dA) = hydraulic losses (3-2-9)$$

and its one dimensional counterpart is

$$\alpha'_1 v_1^2/2g + Z_1 + p_1/\gamma = \alpha'_2 v_2^2/2g + Z_2 + p_2/\gamma + h_f$$
 (3-2-10)

where a is the energy correction factor which takes into account the non-uniformity of the velocity distribution

$$= 1/A \iint (u/V)^3 dA$$
 (3-2-11)

The head loss in steady uniform flow in open channel flow may be evaluated from Manning's formula

$$V = (1.486/n) R^{2/3} S^{\frac{1}{2}}$$
 (3-2-12)

where

$$S = h_f / L$$
 (3-2-13)

is the head loss per unit length.

In open channel flow, it is convenient to consider the energy with respect to the bottom as a datum. Thus is the specific energy first introduced by Bakhmeteff<sup>(2)</sup>:

$$\mathbf{E} = \nabla^2 / 2g + \mathbf{Y} \tag{3-2-14}$$

In this form it is assumed that the velocity is uniform and that the streamlines are essentially parallel, thus the pressure is hydrostatic. If the velocity is not uniform, and if the streamlines are curved, the velocity head is multiplied by the kinetic energy coefficient, and the depth y may be multiplied by a pressure coefficient  $(\checkmark)$ 

$$\alpha' = 1 + 1/Qy \iint c \nabla d A \qquad (3-2-15)$$

where

$$c = (d/g) v^2/z$$
 (3-2-26)

and is positive for concave flow and negative for convex flow, d being the depth, and r the radius of curvature of a streamline. The specific energy thus becomes in general

$$E = \sqrt{2/2g} + c' Y$$
 (3-2-2)

The opecific force and the specific energy equations are used to define a number of open channel flow characteristics. When the specific energy or the specific force is minimum, the flow is called critical. The following relationship for critical flow may be derived from the specific energy equation for a stream of an arbitrary cross section, with  $\sqrt{-1}$ 

$$Q^2/g = A_c^3/W_c$$
 (3-2-18)

$$Q/\sqrt{g/\kappa} = Z; Z = A/A/W = A/D$$
(3-2-19)

$$d V^2/2g = A_0/2W_0 = D_0/2$$
 (3-2-20)

$$\Psi = \sqrt{(g/s_c)} (A_c/W_c) = \sqrt{g D_c/s_c}$$
(3-2-21)

$$V / g D_c / z = F = 1$$
 (3-2-22)

where W is the width of the free surface, Z is the section factor for critical flow computation, D is the hydraulic depth, and F is the Froude number. For a stream of rectangular cross section, letting g to the discharge per unit width, the above relationships take the following simplified forms:

$$q = \left[ g r e^{3} / 4 \right]^{\frac{1}{2}}$$
 (3-2-25)

$$\sqrt{V^2/2g} = y_0/2$$
 (3-2-2).

$$V_{c} = \left[ gy_{c}/x \right]^{\frac{1}{2}} \tag{3=2=25}$$

$$\underline{\mathbf{Y}}_{e} = [\underline{\mathbf{x}}_{g}^{2}/\underline{g}_{g}]^{\frac{1}{3}}$$
 (3-2-26)

$$E = 3/2 \Sigma_{c}$$
 (3-2-27)

For flow in rectangular channels the specific force and the specific energy equations may be written in a dimensionless form, making use of the critical flow relationships. They are respectively

$$\frac{\overline{Y}_1}{\overline{Y}_1} \left[ \frac{\overline{Y}_1}{\overline{Y}_c} \right]^2 + \beta_1 \frac{\overline{Y}_c}{\overline{Y}_1} - \frac{\overline{Y}_2}{\overline{Y}_2} \left[ \frac{\overline{Y}_2}{\overline{Y}_c} \right]^2 + \beta_2 \frac{\overline{Y}_c}{\overline{Y}_2}$$
(3-2-28)

and

$$\frac{Y_1}{Y_c} + \alpha_1 \left( \frac{Y_c}{Y_1}^2 - \frac{Y_2}{Y_c} + \alpha_2 \frac{Y_c}{Y_2}^2 \right)$$
(3-2-29)

where the ratio  $\overline{Y}/Y$  is a geometric characteristic of the cross section and is equal to 0.5, for rectangular channels, y being the head over the center of gravity of the section.

#### III-3 Types of Flow at a Bridge Constriction

Six classes of flow may axist at a bridge constriction: Class 1 - The flow is subcritical throughout the transition Class 2 - The flow passes from subcritical to supercritical in the transition Class 3 - The flow is supercritical throughout the transition Class 4 - The flow passes from supercritical to subcritical near the transition Class 5 - The transition inlet is submerged but the outlet is free, resulting in a free orifice flow.

Class 6 - The transition inlet and outlet are submerged resulting in a submerged orific flow.

The following analysis illustrates a method of distinguishing between the first four types of free surface flow. The latter two types are not truly free surface flow. A method to discriminate between the free surface flow and the orifice flow is also given.

As the flow towards the constriction is an accelerating flow, and as the energy losses are relatively small in a converging flow, the energy equation is suitable to describe the flow between sections 1 and 3. The expanding flow between sections 3 and 4 is better described by a momentum equation.

The first four types of flow may be delineated by the following simplified analysis. Writing an energy equation between sections 1 and 2,

$$\alpha_1 \frac{v_1^2}{2g} + \alpha_1^2 \frac{v_1}{2g} = \alpha_3 \frac{v_3^2}{2g} + \alpha_3^2 \frac{v_3 + h_2}{2g}$$
 (3-3-1)

Assuming  $\ll_1 = \ll_3 = \ll_1' = \ll_3' = 1$  and  $h_f = 0$ , letting by the mean width of the stream at section 3, making use of the continuity equation

$$V_1 B X_1 = V_3 b_3 X_3$$

the equation of energy may be re-written, after dividing both sides by Ya.

$$\frac{V_{1}^{2}}{2g_{11}} + 1 = \frac{V_{1}^{2}}{2g_{11}} \left(\frac{B}{b_{3}} + \frac{V_{1}}{V_{3}}\right)^{2} + \frac{V_{3}}{V_{1}}$$
(3-3-2)  
$$\frac{1}{2}F_{1}^{2} \left[\left(\frac{B}{b_{3}} + \frac{V_{1}}{V_{3}}\right)^{2} - 1\right] = 1 - \frac{V_{3}}{V_{1}^{2}}$$
(3-3-3)

26

or

or

$$F_{1}^{2} = \frac{2 (1 - \frac{V_{3}}{2})}{\left(\frac{B}{b_{3}} + \frac{V_{1}}{2}\right)^{2} - 1}$$
(3-3-4)

$$= \frac{2 \left( \frac{X_{3}}{2} \right)^{2} \left( 1 - \frac{X_{3}}{2} \right)}{\left( \frac{B}{2} \right)^{2} - \left( \frac{X_{3}}{2} \right)^{2}}$$
(3-3-5)

F<sub>1</sub> is plotted as a function of  $\frac{T_3}{Y_1}$  in terms of  $\frac{B}{b_3}$  in Fig. 3-2. The curve for which the flow is critical at section 3 corresponds to F<sub>3</sub> = 1, and is obtained from the continuity equation:

$$V_1 B Y_1 = V_3 b_3 Y_3$$
 (3-3-6)

OI.

$$v_1^2 B^2 v_1^2 = v_3^2 b_3^2 v_3^2$$
 (3-3-7)

or

$$V_1^2$$
 B<sup>2</sup>  $V_1^3$   $V_3^2$   
 $gY_1$   $Y_3^3$   $F_3^3$   $GY_3^2$  (3-3-6)

01

$$\frac{B^2}{b_3^2} \frac{Y_{13}}{Y_{33}^3} = F_3^2$$
(3-3=9)

When the flow is critical at section 3,  $F_3 = 1$  and

$$P_1^2 = (\frac{1}{2})^3 \left(\frac{b_3}{B}\right)^2$$
(3-3-10)

Solving the two equations simultaneously:

F12

$$F_{1}^{2} = \frac{2 \left(\frac{r_{3}}{r_{1}}\right)^{2} \left(1 - \frac{r_{3}}{r_{1}}\right)}{F_{1}^{2} \left(\frac{r_{3}}{r_{1}}\right)^{-3} - \left(\frac{r_{3}}{r_{1}}\right)^{2}}, \qquad (3-3-11)$$

$$F_1^4 - F_1^2 \left(\frac{Y_3}{Y_1}\right)^5 - 2 \left(\frac{Y_3}{Y_1}\right)^5 \left(1 - \frac{Y_3}{Y_1}\right) = 0$$
 (3-3-12)

from which

or

$$2F_{1}^{2} = \left(\frac{X_{3}}{Y_{1}}\right)^{5} \pm \sqrt{\left(\frac{X_{3}}{Y_{1}}\right)^{10} + \left(\frac{X_{3}}{Y_{1}}\right)^{5} \left(1 - \frac{X_{3}}{Y_{1}}\right)}$$
(3-3-13)

Only the positive sign in front of the square root has a physical significance. This equation gives the values of the Froude number at section 1 corresponding to a Froude number of unity at the vena contracta.

The plane of Fig. 3-2 is thus divided in 4 regions:

Above the line  $F_1 = 1.0$  the flow is supercritical in the approach of the constriction, below the line  $F_1 = 1.0$ , the flow is subcritical at section 1. Above the curve  $F_3 = 1$ , the flow is supercritical, and below the curve  $F_3 = 1$  the flow is subcritical. The four regions of the figure are labelled according to the four classes of flow listed at the beginning of this section.
## III-4 Discrimination Between Free Surface Flow and Orifice Flow

Orifice flow will occur when the head over the crown of the arch exceeds the velocity head plus the head loss in the orifice.

Thus for the limiting case, calling r the semi-circular arch radius:

$$\frac{V_{2}^{2}}{2g} + \left(\frac{1}{C_{V}^{2}} - 1\right) \frac{V_{3}^{2}}{2g} = Y_{1} - T$$
(3a)

where  $(\frac{1}{Cv^2} - 1) \frac{V_3^2}{2g}$  is the orific head loss.

$$\frac{1}{C_{v}^{2}} = \frac{V_{1}^{2}}{2g} = \frac{Y_{1}}{2g} = \frac{Y_{1}}{2g}$$
(3-4-2)

By continuity V1 BV1 = V3 b3 V3 = V3 Cc Ao where Ao is the opening area.

Then

$$\frac{1}{C_{v}^{2}} \frac{V_{1}^{2}}{2g} \frac{B^{2} V_{1}^{2}}{C_{c}^{2A_{0}}^{2}} = Y_{1} = r \qquad (3.4...)$$

$$\frac{\mathbf{v}_{1}^{2}}{2g\mathbf{v}_{1}} = \left(1 - \frac{r}{\mathbf{v}_{1}}\right)^{2} \mathbf{c}_{d}^{2} \left(\frac{A_{0}}{A_{1}}\right)^{2} \qquad (3 \ \ell_{1} - \ell_{1})^{2}$$

$$F_1^2 = 2C_d^2 (1 - \frac{r}{T_1}) (\frac{A_0}{A_1})^2$$
 (3-4-5)

There will be submergence when

$$\mathbf{F}_{1}^{2} \ll \mathbf{C}_{d}^{2} \left(1 - \frac{\mathbf{r}}{\mathbf{Y}_{1}}\right) \left(\frac{\mathbf{A}_{2}}{\mathbf{A}_{1}}\right)^{2} \tag{3-4-6}$$

where Ca 20.55.

or when 
$$F_1^2 < 1/8 \ C_d^2 \ \pi^2 (1 - \frac{1}{2} \ \frac{b}{Y_1}) \ (\frac{b}{B})^2 \ (\frac{b}{Y_1})^2$$
 (3-4-7)

and since  $\frac{b}{B} = M$ 

when

$$\frac{r_1^2}{r_1^2} 4 \frac{1}{8} \frac{\pi^2}{\pi^2} c_d^2 \left(1 - \frac{1}{2} \frac{b}{Y_1}\right) \left(\frac{b}{Y_1}\right)^2$$
(3-4-8)

An empirical relationship to distinguish between free surface flow and orifice flow, can be obtained, making use of the notation developed in the article on dimensional analysis, and of the results of chapter 5.

From free surface flow experiments (see chapter 5)

$$\frac{Y_{1}}{Y_{n}} = 1 \div 0.47 \left(\frac{F_{n}}{M^{\circ}}\right)^{2.26} = 1 \div 0.47 \quad \frac{1}{Cm^{2.26}} \left(\frac{F_{n}}{M}\right)^{2.26} \quad (3 \to 4 \to 5)$$

where Cm " Mº/M. (See Figure 3-12) From orific flow experiments (see chapter 5)

$$\frac{\mathbf{X}_{1}}{\mathbf{Y}_{n}} = \mathbf{1} + \mathbf{1} \cdot \mathbf{16} \begin{bmatrix} \frac{\mathbf{F}_{n}}{\mathbf{A}_{0}} \\ \overline{\mathbf{B}_{T}_{n}} \end{bmatrix}^{\mathbf{1} \cdot \mathbf{S}} = \mathbf{1} + \mathbf{1} \cdot \mathbf{18} \left(\frac{\mathbf{S}}{\mathbf{H}}\right)^{\mathbf{1} \cdot \mathbf{S}} \left[\frac{\mathbf{Y}_{n}}{\mathbf{b}}\right]^{\mathbf{0} \cdot \mathbf{S}} \left[\frac{\mathbf{F}_{n}}{\mathbf{M}}\right]^{\mathbf{1} \cdot \mathbf{S}} (3-4-10)$$

Equating

 $\begin{bmatrix} F_n \\ M \end{bmatrix}^{0.46} = 13.46 \quad C_m^{2.26} \begin{pmatrix} Y_n \\ D \end{pmatrix}^{1.8}$ (3-4-11)

and since for d = 0

$$C_{M} = \frac{1}{2} \left\{ \sqrt{1 - \left(\frac{Y_{n}}{r}\right)^{2} + \frac{r}{Y_{n}}} \quad \sin^{-1} \quad \frac{Y_{n}}{r} \right\}$$
(3-4-12)

For the limiting case

$$\left(\frac{F_{n}}{M}\right)^{0.46} = 13.46 \left\{ \frac{1}{2}\sqrt{1 - \left(\frac{Y_{n}}{r}\right)^{2}} + \frac{r}{Y_{n}} \sin^{-1} \frac{Y_{n}}{r} \right\}^{2.26} \left(\frac{Y_{n}}{2r}\right)^{1.8}$$

A plot of this relationship is given in figure 3-5, from which it is possible to predict for a given Froude, normal depth and contraction geometry, whether the flow will be free or submerged.

#### a) General considerations

This is the most common case. The flow pattern divides itself in two parts: (see fig. 3-1).

1) a zone of contracting or accelerating flow, from the section of maximum backwater (section 1) to the section of minimum area or "wena contracta" - (section 3).

2) a zone of expanding or decelerating flow, from the vena contracta (section 3) to the section where the stream has regained its normal depth (section 4).

It is, therefore, expected that the analysis of the flow pattern would be divided in two parts. An energy equation may be used for the analysis of the contracting flow, and a momentum equation for that of the expanding flow. Section 4, at which the flow occurs at normal depth is the control section. The calculation of the flow profile for a given discharge could proceed from section 4 to section 3 by means of the momentum equation, and from section 3 to section 1 by means of the energy equation, the continuity equation being used to relate the areas at the several sections. The difficulty in this approach lies in the lack of proper definition of the geometry of section 3. The geometry of section 2, at the section of separation, is defined by the water depth, as the geometry of the constriction is assumed to be known — in our case a semi-circular arch.

At section 2 the flow separates from the solid boundaries, and the threedimensional live stream converges to the vena contracta. The geometry of the free boundary of the convergent live stream cannot be determined by means of the free bitmemaline theory as the latter is limited to two-dimensional flow. Consequently, it is customary to make some assumption on the flow geometry between sections 2 and

- 3. Typical assumptions are:
  - a) The geometry of section 3 is assumed.
  - b) The depths at sections 2 and 3 are assumed to be identical.
  - c) The depth at section 2 is assumed to be equal to the normal depth of the

#### uncontracted stream.

The three assumptions will be discussed.

# a) Assumed geometry at section 3.

Experiments by P. F. Biery indicate (Fig. 7-2-7) that the depth at section 3 is not uniform across the section, but that in the dead water region it is only 1% larger than across the live stream. The depth X<sub>3</sub> may thus be reasonably assumed constant across the section. The width of the live stream b<sub>3</sub> may be approximated using the two-dimensional coefficient of contraction and

$$b_3 = c_{0,2} b$$
 (3-1/1)

1

Thus the cross section of the livestream would be

$$A_{\rm S} = C_{\rm 0.2} \, \rm b \, Y_{\rm 3}$$
 (3-5-2)

The values of the coefficients of contraction for slots, after von Misses(j) is given in the following table:

**0 0.1 0.2 0.5 0.4** 0.5 0.6 0.7 0.8 0.9 1.0 **0.2 0.611 0.612 0.615 0.622 0.631 0.644 0.662 0.637 0 722 0.761 1.00** 

This assumption makes it possible to relate the depth  $Y_n = Y_i$  to  $Y_3$  by means of the momentum equation, and then  $Y_3$  to  $Y_1$  by means of the energy equation.

b) Assumed equality of depths at sections 2 and 3.

This assumption was used by Tracy and Carter (ASCE Trans. 1955 p. 997). The moments of P. F. Biery, (Jour. Hyd. Div. ASCE, Vol. 88 No. HT2 p. 96, Fig. 13) to not substantiate this assumption.

# c) Depth at section 2 assumed equal to normal depth of uncontracted stream.

This assumption has been implied by Izzard (ASCE Trans 1955, p. 1010). Although this assumption cannot be justified theoretically, the experiments of P. F. Biery show that this assumption is approximately satisfied. It is then possible to express the head loss in the constriction as a function of the velocity head at section  $2_9$ 

$$hf = K \frac{V_{n2}^2}{2g}$$
 (3-5-3)

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The depth of flow at section 1 may then be obtained by means of an energy equation between sections laand 2 or between sections 1 and 4. This assumption has also been used with reasonable success by I. P. Wu<sup>(4)</sup> in a preliminary calculation of indirect determination of flood discharge from contracted bridge openings and high water marks.

b) Momentum equation for diverging flow

The momentum equation is written between sections 3 and 4 with the assumption that the pressure distribution is hydrostaticate, These sections:

$$\mathbf{F}_3 + \mathbf{M}_3 = \mathbf{F}_L + \mathbf{M}_L + \mathbf{F}_D - \mathbf{F}_G \tag{3.56.4}$$

 $F_3$  and  $F_4$  represent the hydrostatic forces at section 3 and 4 respectively;  $M_3$  and  $M_4$  are the momenta,  $F_D$  is the arag force of the expanding stream on the bottom and  $F_C$  is the component of the gravity force in the direction of the flow.

To calculate the hydrostatic force at sections 3 and 4, let  $Y_3$  and  $Y_4$  be the heads over the respective enters of gravity of the cross sections. To calculate the momentum flux through section 3 it is necessary to evaluate the cross sectional area of the live stream at section 5. It is assumed that

The momentum equation may then by written

$$\delta \overline{\mathbf{x}}_3 = \mathbf{x}_4 + \mathbf{e} \mathbf{x}_3 + \mathbf{x}_4 + \mathbf{e} \mathbf{x}_4 + \mathbf{x}_D - \mathbf{x}_G$$
 (3-5-5)

Making use of the continuity equation, it follows that

$$\gamma T_3 B T_3 + \rho T_4 E T_4 + \rho Q^2 + F_D - F_G$$
 (3-5-7)

$$\tilde{X}_{3} X_{3} + \frac{a^{2}b}{gA_{3}} = \tilde{X}_{4} X_{4} + \frac{a^{2}b}{gA_{4}} - \frac{F_{D}}{\sqrt{B}} = F_{C}$$
 (3-5-6)

Introducing the critical depth  $Y_c^5 = q^2/g_r$ 

$$\mathbf{\tilde{I}}_{3} \mathbf{\tilde{I}}_{3} + \frac{\mathbf{\tilde{I}}_{2}}{\mathbf{\tilde{I}}_{3}} \mathbf{\tilde{b}}_{3} = \mathbf{\tilde{I}}_{1} \mathbf{\tilde{I}}_{1} + \frac{\mathbf{\tilde{I}}_{3}}{\mathbf{\tilde{I}}_{1}} + \frac{\mathbf{F}_{D} - \mathbf{F}_{G}}{\mathbf{\tilde{J}}^{B}}$$
(3-5.4)

The momentum equation may be written in a dimensionless form by dividing both sides of the above equation by  $\Sigma_c^2$ :

$$\frac{\overline{\mathbf{Y}}_{3}}{\overline{\mathbf{Y}}_{3}}\left(\frac{\overline{\mathbf{Y}}_{3}}{\overline{\mathbf{Y}}_{c}}\right)^{2} + \frac{\overline{\mathbf{Y}}_{c}}{\overline{\mathbf{Y}}_{3}} = \frac{\overline{\mathbf{Y}}_{L}}{\overline{\mathbf{Y}}_{L}}\left(\frac{\overline{\mathbf{Y}}_{L}}{\overline{\mathbf{Y}}_{c}}\right)^{2} + \frac{\overline{\mathbf{Y}}_{c}}{\overline{\mathbf{Y}}_{L}} + \frac{\overline{\mathbf{F}}_{D} - \overline{\mathbf{F}}_{C}}{\sqrt{\overline{\mathbf{E}}\mathbf{Y}_{c}}^{2}} \quad (3-5-.)$$

The quantity  $\frac{\overline{Y}}{\overline{Y}}$  is a geometric constant for many cross sections. For example for a rectangular section  $\frac{\overline{Y}}{\overline{Y}} = 0.5$ triangular section = 7.333semi-circle section = 1 - (L/3) Tparabola section = 3/5Let  $\frac{\overline{Y}}{\overline{Y}} = G$  a geometric constant of the section. Then the dimensionless momentum

equation becomes:

$$\mathbf{G}_{3}\begin{pmatrix} \mathbf{Y}_{2} \\ \mathbf{Y}_{c} \end{pmatrix} \overset{2}{\longrightarrow} \overset{B}{\overset{\mathbf{Y}_{2}}{\overset{\mathbf{Y}_{2}}{\overset{\mathbf{Y}_{3}}{\overset{1}}{\overset{\mathbf{Y}_{3}}}{\overset{1}}{\overset{1}}{\overset{1}}{\overset{1}}}$$

Neglecting for the time being the contribution of the term

$$(F_D - F_G)//BY_c^3$$

and calling the specific force Fs

$$\mathbf{F}_{\mathbf{B}} = \sqrt{\mathbf{X}} \mathbf{B} \mathbf{X} + \mathbf{C} \mathbf{Q} \mathbf{V} \tag{3-5-12}$$

then the generalized dimensionless momentum equation becomes:

$$\frac{Y_{x}}{Y_{c}}^{2} + \frac{B}{b} \frac{Y_{c}}{Y_{c}} = F_{B}/\gamma BY_{B}^{2}$$
(3-5-13)

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03

A plot of the relationship is given in the right hand side of figure 3-7, which may be used for an approximate solution of the depth  $Y_3$  if the depth  $Y_4$  is given. The width of the live stream at section 3 is usually unknown a priori, but may be taken as the width of the constriction times a contraction coefficient.

The drag force of the diverging stream on the channel bottom between the vena contracta and section 4 may be obtained by integrating the shear stress over the expansion area along the bottom. From the discussion by  $Henry^{(5)}$  it appears that the rate of expansion of a submerged jet is approximately 1 on 6. The model is assumed for the zone of expansion is shown Fig. 3-6.

From Manning's formula, the shear stress on the bottom is

$$\tau = \frac{14.5n^2 \, \ell \, v^2}{R^{1/3}} = \frac{14.5n^2 \, p \, Q^2}{Y^{1/3} \, b^{,2} \, Y^2} = \frac{14.5n^2 \, p \, q^2 \, B^2}{Y^{7/3} \, b^{,2}} \qquad (3-5-14)$$

and the elementary drag force is

$$dF_{\rm D} = \tau dA = \frac{14_0 5n^2 \rho_0^2 B^2}{\gamma^{7/3} b^0} dx \qquad (3-5-15)$$

or

$$\frac{dF_{D}}{fBY_{c}^{2}} = \frac{14.5n^{2} B Y_{c}}{Y^{7/3} b^{3}} \quad dx = 14.5 \left(\frac{n}{Y_{c}^{1}/6}\right)^{2} \left(\frac{Y_{c}}{y}\right)^{7/3} \frac{B}{Y_{c}} \frac{dx}{b^{3}} \quad (3-5-16)$$

The dimensionless drag force is

$$\frac{F_{\rm D}}{\gamma_{\rm BT_c}^2} = \frac{1}{4} \cdot 5 \left(\frac{n}{T_c} \frac{1}{6}\right)^2 = \frac{B}{T_c} = \int_0^1 \frac{L_{3-4}}{y} \left(\frac{T_c}{y}\right)^{7/3} \frac{dx}{b^3} \qquad (3-5-17)$$

Since b' =  $C_{db} + x/3$ , and taking an average value of the depth, an approximate value of the drag force is given by

$$\frac{F_{\rm D}}{\sqrt{BY_{\rm c}}^2} = \frac{14.5}{14.5} \left(\frac{n}{Y_{\rm c}} \frac{1}{1/6}\right)^2 \frac{B}{Y_{\rm c}} \left(\frac{2Y_{\rm c}}{Y_{\rm n}+Y_{\rm 3}}\right) \int_0^3 \frac{(B-bC_{\rm d})}{b} \frac{dx}{bC_{\rm d}+x/3} \quad (3-5-18)$$

$$= \underline{14}_{\circ} 5 \left(\frac{n}{\underline{Y_c}^{1/6}}\right)^2 = \frac{B}{\underline{Y_c}} \left(\frac{2\underline{Y_c}}{\underline{Y_n} + \underline{Y_3}}\right)^3 \ln \frac{B}{bC_d}$$
(3-5-19)

The gravity force is

$$F_{G} = B \frac{Y_{3} + Y_{4}}{2} L_{3-4} S_{0}$$
  
=  $B \frac{Y_{3} + Y_{4}}{2} 3 (B-bC_{d}) S_{0}$  (3-5-20)

and the dimensionless gravity force is

$$\frac{F_{G}}{BT_{c}}^{2} = \frac{Y_{3} + Y_{n}}{Y_{c}} = \frac{B - bC_{d}}{Y_{c}} = S_{0} \qquad (3-5-21)$$

In the actual computation, a first approximation of  $Y_3$  is obtained by neglecting  $F_D$  and  $F_G$ . With this approximate value of  $Y_3$ , the values of  $F_D$  and  $F_G$  may be calculated from the above equations.

c) Energy Equation for Converging Flow

The energy equation between sections 1 and 3 may be written

$$S_1 I_{1-3} + Y_1 + \alpha_1 \frac{V_1^2}{2g} = Y_3 + \alpha_3 \frac{V_3^2}{2g} - h_f$$
 (3-5-22)

where h<sub>f</sub> is the head loss between sections 1 and 3. Making use of the continuity equation

$$V_{1} \stackrel{A_{1}}{=} V_{3} \stackrel{A_{3}}{=} V_{1} \stackrel{A_{1}}{=} \frac{q^{2}}{2gY_{1}^{2}} \stackrel{A_{3}}{=} V_{3} \stackrel{A_{3}}{=} \left(\frac{A_{1}}{A_{3}}\right)^{2} \frac{q^{2}}{2gY_{1}^{2}} \stackrel{A_{1}}{=} h_{f} \stackrel{S_{0}}{=} L_{1-3} \quad (3-5-23)$$

Introducing the critical depth  $T_c^3 = q^2/g$ 

$$\frac{\mathbf{Y}_{1}}{\mathbf{Y}_{c}} + \alpha \mathbf{1} \left( \frac{\mathbf{Y}_{c}}{\mathbf{Y}_{1}} \right)^{2} = \frac{\mathbf{Y}_{3}}{\mathbf{Y}_{c}} + \alpha \mathbf{3} \left( \frac{\mathbf{B}}{\mathbf{b}_{3}} \right)^{2} \frac{\mathbf{Y}_{c}}{\mathbf{Y}_{3}}^{2} + \frac{\mathbf{h}_{f}}{\mathbf{Y}_{c}} - \mathbf{S}_{0} \frac{\mathbf{L}_{1-3}}{\mathbf{Y}_{c}}$$

$$(3-5-2l_{1})$$

which is the dimensionless form of the energy equation. Assuming  $\alpha_1 = \alpha_3 = 1$ , and neglecting the difference of the last two terms.

$$\frac{Y_1}{Y_c} + \left(\frac{Y_c}{Y_1}\right)^2 = \frac{Y_3}{Y_c} + \left(\frac{B}{b_3}\right)^2 \left(\frac{Y_c}{Y_3}\right)^2 = \frac{E_s}{Y_c}$$
(3-5-25)

The plot of this equation is given in the left-hand side of figure 3-7.

The friction loss between sections 1 and 3 may be calculated from the expression

$$h_{f} = L_{1-2} \frac{Q^{2}}{K_{1}K_{3}} + L_{2-3} \frac{Q^{2}}{K_{3}^{2}}$$
(3-5-26)

The value of the conveyance

$$K = \frac{1.486}{n} = \frac{1.2}{N} (3=5=27)$$

may be approximated as follows

$$K_1 = \frac{1.486}{n} = \frac{1.486}{n} = \frac{1.486}{1} = \frac{1.486}$$

$$x_3 = \frac{1.486}{n} \quad b_3 x_3^{5/3}$$
 (3-5-29)

then  $\frac{h_{f}}{Y_{c}} = \frac{q^{2}B^{2} n^{2}}{(1.486)^{2}Y_{c}} \left\{ \frac{L_{1-2}}{B_{0} Y_{1} 5/3 Y_{3} 5/3} + \frac{L_{2-3}}{b_{3}^{2} Y_{3} 10/3} \right\}$  (3-5-30)

$$= \frac{g Y_{c}^{2} n^{2} B}{2 e^{2}} \left\{ \frac{L_{1-2}}{b} + \frac{1}{Y_{1}^{5/3} Y_{3}^{5/3}} + \frac{L_{2-3} B}{b_{3}^{2}} + \frac{1}{Y_{3}^{10/3}} \right\}$$
(3-5-31)  
=  $\frac{E}{2 e^{2}} + \frac{B}{Y_{c}} \left( \frac{Y_{c}}{Y_{3}} \right)^{5/3} \left( \frac{n}{Y_{c}^{1/6}} \right)^{2} \left\{ \frac{L_{1-2}}{b} + \left( \frac{Y_{c}}{Y_{1}} \right)^{5/3} + \frac{L_{2-3}}{b^{2}_{3}} \left( \frac{Y_{c}}{Y_{3}} \right)^{5/3} \right\}$ (3-5-32)

The experiments of Biery show that  $L_{1-2} \approx L_{1-3} \approx b$ . Then

$$\frac{\mathbf{h}_{\mathbf{r}}}{\mathbf{Y}_{\mathbf{n}}} = \frac{B}{2 \cdot 2} \frac{B}{\mathbf{Y}_{\mathbf{c}}} \left(\frac{\mathbf{Y}_{\mathbf{c}}}{\mathbf{Y}_{\mathbf{3}}}\right)^{5/3} \left(\frac{\mathbf{n}}{\mathbf{Y}_{\mathbf{n}}^{1/6}}\right)^{2} \left\{ \left(\frac{\mathbf{Y}_{\mathbf{c}}}{\mathbf{Y}_{\mathbf{1}}}\right)^{5/3} + \frac{Bb}{b_{\mathbf{3}}^{2}} \left(\frac{\mathbf{Y}_{\mathbf{c}}}{\mathbf{Y}_{\mathbf{3}}}\right)^{5/3} \right\}$$
(3-5-33)

In an actual computation, a first approximation of  $Y_1$  is obtained by neglecting the difference between the head loss  $h_f$  and  $S_0 L_{1-3}$ . The term neglected is the loss in addition to the normal flow head loss due to the acceleration of the flow between section 1 and 3 and the separation of the flow from the boundaries at section 2. The quantity neglected is usually negligible.

#### d) Graphical Solution of Backwater due to a Constriction

The dimensionless specific force and specific energy curves may be used to obtain graphically the backwater upstream of a constriction, as was first suggested by H. R. Henry<sup>(6)</sup>. Part of the specific force and specific energy diagram has been enlarged in Fig. 3-8 for the illustration of the graphical method.

As an example, the conditions of the experiment 16-S, Geometry Ia (see table 7-1-2) are illustrated. The data of the problem are:

$$Q = 2.0$$
 cfs,  $Y_n = 20.30$  cm;  $M = B/b = 0.491$ .

Calculate

$$Y_{c} = \left(\frac{Q^{2}}{gB^{2}}\right)^{1/3} = 5.24 \text{ cm}$$

and

$$Yn/Tc = 3.88$$
.

Enter the diagram with the value of  $Y_{4_c}/Y_c = Y_n/Y_c = 3.88$  and proceed to the specific curve labeled B/b3 = 1.0, corresponding to no contraction. With a coefficient of contraction  $C_d = 0.64_s$  calculate

$$\frac{B}{b_3} = \frac{B}{bC_d} \approx \frac{1}{MC_d} = \frac{1}{O_0491 \times O_064} = 3.16$$

Proceed on the diagram along a vertical line, corresponding to a constant specific force, until an interpolated curve for  $B/b_3 = 3.18$  is reached. Proceed to the left and read  $Y_3/Y_c = 3.72$ . The experimental value of  $Y_3/Y_c$  was  $\frac{19.41}{3.88} = 5.0$ . With the obtained value of  $Y_3/Y_c$  of 3.72, proceed to an interpolated specific energy corresponding to  $B/b_3 = 3.18$ . Proceed upward following a line of constant specific energy until the curve  $B/b_3 = 1$  is reached, corresponding to no contraction at station 1. Read the value of  $Y_1/Y_c = 4.069$  and calculate  $Y_1 = 4.069 \times 5.24 = 21.30$  cms. The experimental value was 21.26 cms. A good agreement was obtained in spite of the neglect of the friction and head loss terms. However, it should be remembered that the example chosen corresponds to smooth boundaries, and consequently, the head losses are indeed small.

#### a) Geometric Properties of Semi-Circular and Circular Segment Channels

In this type of flow the control is in the constriction, where there exists a relationship between the depth of flow and the discharge. The depth at soction 1 may be obtained directly by writing an energy equation between the control section and section 1. As the calculation of the critical depth in a conduit having a semicircular section, or a circular segment section, is tied to the geometry of the section, this geometry is studied first. At critical stage

$$Z = \frac{Q}{\sqrt{g/a}}$$
(3-6-1)

where

 $Z = A\sqrt{\frac{\pi}{y}}$ (3-6-2)

and y is the hydraulic depth, defined by

where W is the free surface width.

The geometric properties of the semi-circular and circular segment arches are summarized in Fig. 3-9. The top right hand side quadrant gives the relationship between the section factor for critical flow computation and the critical depth in terms of the distance from the springline to the center of curvature of the arch. The curve labeled d/b = 0 corresponds to the semi-circular arch. In order to obtain the critical month in the constriction of a given diameter b for a given discharge  $Q_p$ one calculates the section factor for critical flow 7 by means of the relation

$$\sqrt{\frac{Q}{B/A}} = Z$$

and entering the diagram with the proper value of  $Z/b^{5/2}$  for the desired d/b one reads the ratio of the critical depth to the diameter  $Y_c/b$ . The value of the hydraulic depth  $\overline{y}$  and of the free surface width W can be obtained by means of the sets of curves in the top and bottom left hand side of the right respectively. With reference to Fig. 3-9 (bottom right hand side) the area of the section of depth y is:

$$A = \frac{b^2}{8} \sec^2 \frac{\theta_2}{2} (\theta_1 - \theta_2 + \sin \theta_1 - \sin \theta_2)$$
 (3-6-4)

the free surface is

$$W = b \frac{\cos \frac{\theta_2}{2}}{\cos \frac{\theta_2}{2}}$$
(3-6-5)

the hydraulic depth is

$$\frac{1}{2} = \frac{1}{8} - \frac{1}{8} = \frac{1}{2} + \frac{1}$$

the section factor for critical flow computation is:

$$Z = A \sqrt{\overline{y}} = \frac{b^{2/8}}{\cos^{2}} \frac{(\theta_{1} - \theta_{2} + \sin \theta_{1} - \sin \theta_{2})}{2} \sqrt{\frac{b}{8}} = \frac{\theta_{1} - \theta_{2} + \sin \theta_{1} - \sin \theta_{2}}{\cos \theta_{1} \cos \theta_{2}}$$

$$(3-6-7)$$

or in dimensionless form:

$$Z = \frac{0^{5/2}}{8^{3/2}} \frac{(\Theta_1 - \Theta_2 + \sin \Theta_1 - \sin \Theta_2)^{3/2}}{(\cos \Theta_2)^{5/2} (\cos \Theta_1)^{1/2}}$$
(3-6-8)

$$\frac{Z}{b^{5/2}} = \frac{1}{8^{3/2}} \frac{(\theta_1 - \theta_2 + \sin \theta_1 - \sin \theta_2)}{(\cos \frac{\theta_1}{2})^{5/2}} (\cos \frac{\theta_1}{2})^{1/2}$$
(3-6-9)

The above relationships were used to prepare the curves of Fig. 3-9.

b) Limiting Backwater

The purpose of the following analysis is to calculate the backwater at Section 1 when the flow goes through critical in the constriction. This type of flow represents the boundary between classes 1 and 2.

Assuming that the kinetic energy coefficient and the pressure coefficients are unity, the specific energy equation is

$$Y_1 + \frac{Y_1^2}{2g} = Y_2 + \frac{Y_2^2}{2g}$$
 (3-6-10)

OF

$$X_1 \left(1 + \frac{V_1^2}{2gX_1}\right) = \overline{y}_2 \left(\frac{X_2}{y_2} + \frac{V_2^2}{2g\overline{y}_2}\right)$$
 (3-6-11)

or 
$$\underline{Y}_1 \left(1 + \frac{F_1^2}{2}\right) = \overline{y}_2 \left(\frac{Y_2}{\overline{y}_2} + \frac{F_2^2}{2}\right)$$
 (3-6-12)

When the flow is critical at section  $2_{p} F_{2} = 1_{p}$  then

$$Y_1 (2 + F_1^2) = \overline{y}_2 (2 \frac{Y_2}{\overline{y}_2} + 1)$$
 (3-6-13)

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$$\frac{\mathbf{y}_2}{\mathbf{y}_1} \quad \frac{(2 - \frac{\mathbf{y}_2}{2} + 1)}{(2 + F_1^2)} = 1 \quad (3 - 6 - 2\lambda)$$

The equation of continuity

$$(\nabla_1 Y_1 B_1)^2 = (\nabla_2 \overline{y}_2 W)^2$$
 (3-6-15)

may be written

$$F_1^2 B^2 Y_1^3 = F_2^2 \overline{y}_2^3 W^2$$
 (3-6-16)

which for  $F_2 = 1$  becomes.

$$\frac{y_2}{y_1}^{5} = (\frac{B}{W})^2 F_{1}^{2}$$
 (3-6-17)

Introducing this result into the specific energy equation (3-6-14) it follows that

$$P_{1}^{2} \frac{B^{2}}{b^{2}} \frac{b^{2}}{W^{2}} \frac{(2 \frac{Y_{2}}{Y^{2}} + 1)^{3}}{(2 + P_{1}^{2})^{3}} = 1$$
(3-6-18)

or

$$\frac{b}{B}^{2} = M^{2} = F_{1}^{2} \frac{b^{2}}{W^{2}} \left\{ \frac{2 \frac{Y_{2}/b}{Y_{2}/b} + 1}{\frac{Y_{2}/b}{2} + F_{1}^{2}} \right\}^{3}$$
(3-6-19)

As the value of the critical depth is a function of the critical flow factor

$$\frac{Y_{c}}{b} = f(\frac{Z}{b^{5/2}})$$
 (3-6-20)

and as the values of the free surface width  $W_{\rho}$  and of the hydraulic depth  $\overline{y}$  depend on  $\mathbf{X}_{c,\rho}$  they are implicit functions of  $Z/b^{5/2}$ . Equation (3-6-20) may be then represented in a functional form as  $M = (\frac{b}{B}) = f(F_{1,\rho}, \frac{Z}{b^{5/2}})$  for a given value of d/b.

This relationship is presented in Fig. 3-10 several values of d/b ranging from 0 to 4. The values of  $\frac{W}{b}$ ,  $\frac{Y}{b}$ ,  $\frac{Y}{b}$  necessary for the preparation of these curves were obtained from Fig. 3-9 for values of  $7/b^{5/2}$  of 0.1; 0.2; 0.3; 0.4; and 0.5, and introduced in formula (3-6-19) for M<sup>2</sup> for several values of F<sub>3</sub>.

For a given discharge Q, and a trial opening ratio b/B one may calculate the dimensionless section factor for critical flow

$$\frac{Z}{b^{5/2}} = \frac{Q^2}{\sqrt{g} b^{5/2}}$$

This value of this parameter defines a curve in Fig. 3-10. For points located above the curve, the flow is subcritical in the constriction. Points located on the curve correspond to flow going through critical in the constriction. The limiting backwater may be obtained by reading the value of  $F_1$  on Fig. 3-10 corresponding to given values of  $Z/b^{5/2}$ , b/B and d/B, then  $Y_1$  is calculated from

$$\mathbf{x}_{1}^{3} = \mathbf{x}_{n}^{3} \left(\frac{F_{1}}{F_{n}}\right)^{2}$$
 (3-6-21)

For large discharge, the constrictions partially dam the flow, until it becomes an orifice flow.

#### III-7 Discharge Equations

a) The Equation of Discharge for Free Surface Flow

With reference to Fig. 3-1 and neglecting the velocity of approach, the discharge is found to be  $Q = \int V dA = \int_{0}^{X_{\rm L}} C \sqrt{2g(X_{\rm L} - h)} \times 2\sqrt{r^2 - h^2} dh$  (3-7-1)

An approximate value of the above integral may be obtained by expanding the integrand into a series, and integrating term by term. Making use of the fact that 2r = b

$$Q = C_{d} \sqrt{2g} \frac{17}{24} Y_{1}^{3/2} b \left\{ (1 - 0.1294 (\frac{Y_{1}}{r})^{2} - 0.0177 (\frac{Y_{1}}{r})^{4} \dots \right\}$$
(3-7-2)

This may be written as

$$Q = C_1 Y_1^{3/2} b T_1$$
 (3-7-3)

$$c_1 = c_d \frac{17}{2L} \sqrt{2g}$$
 (3-7-4)

where

and

 $T_1 = 1 - 0.1294 \left(\frac{Y_1}{r}\right)^2 - 0.0277 \left(\frac{Y_2}{r}\right)^4$  (3-7-5)

The above equation is valid as long as the constriction is unsubmerged and  $\mathbb{Y}_1 > \mathbb{P}_{+}$ 

The evaluation of the equation of discharge in the integral form (equation 3-7-1) can be accomplished in two ways. One by expanding into a series and integrating term by term or by evaluating the integral in terms of complete and incomplete elliptical integrals of the first and second kind. The approximate solution was given above and the exact solution follows.

The theoretical discharge may be obtained from equation 3-7-1 by making the coefficient of discharge  $C_d$  equal to unity:

$$Q_t = 2\sqrt{2g} \int_0^{Y_1} \sqrt{(Y_1 - h)(r^2 - h^2)} dh$$
 (3-7-6)

Let  $K^2 = \frac{Y_1 + r}{2r}$  or  $Y_1 = 2r (k^2 - 1), k \ge 1, Y_1 \le r$  (3-7-7)

and 
$$sn^2 u = \frac{h+2}{Y_1 + r}$$
 (3-7-8)

Since 
$$sn^2 u + cn^2 u = 1$$

then 
$$h = \mathbb{Z}_1 \sin^2 u - r \cos^2 u = 2rk^2 \sin^2 u - r$$
 (3-7-9)

and 
$$\frac{dh}{du} = 4 \circ rk^2 snu$$
  $\frac{d}{du} snu = 4rk^2 snu cnu dnu (3-7-10)$ 

From (3-7-8) and making use of (3-7-7)

$$I_{1} = h = Y_{1} (1 - sn^{2}u) + ren^{2}u$$
  
= Y\_{1} cn^{2}u + r^{2} cn^{2}u  
= 2rk^{2}cn^{2}u (3-7-11)

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Also from 3-7-8  

$$r^{2} - h^{2} = 4r^{2} k^{2} sn^{2} u (1 - k^{2} sn^{2} u)$$
  
 $= kr^{2} k^{2} sn^{2} u dn^{2} u$  (3-7-12)  
since  $dn^{2} u + k^{2} sn^{2} u = 1$ 

Substituting (3-7-10), (3-7-11) and (3-7-12) into (3-7-6) the expression for the theoretical discharge becomes

$$Q_{L} = 2\sqrt{2\pi} \int_{12}^{12} \left[ 2rk^{2}cn^{2}u \cdot 4r^{2}k^{2}sn^{2}u dn^{2}u \right]^{\frac{1}{2}} 4rk^{2}snu dnu cnu du = (3-7-1^{2})$$
  
=  $32\sqrt{g} r^{5/2} k^{4} \int_{12}^{12} cn^{2}u sn^{2}u dn^{2}u du$ 

The lower limit  $u_1$  is obtained from (3-7-8) as follows:

$$\operatorname{sn}^2 u_1 = \frac{0+r}{I_1+r}$$

$$\operatorname{snul} = \sin \phi = \sqrt{\frac{r}{Y_1 + r}}$$

(3-7-J.)

(3-7-35

where

or

and finally

F 
$$(\phi_s, k)$$
 is the incomplete elliptic integral of the first kind.  
the upper limit  $u_{2s}$  is obtained from (3-7-8) as follows:

 $\mathbf{u}_{1} = F(\phi, \mathbf{k}) = \int_{0}^{\phi} \frac{\mathrm{d}\phi}{\sqrt{1 - \mathbf{k}^{2} \sin \phi}} , \mathbf{k} < 1$ 

$$sn^2u_2 = \frac{Y_1 + y}{Y_1 + y} = 1$$

or

and

$$\sin u_2 = 1$$
  
 $e^{\pi/2}$  d d

 $u_2 = K = \int_0^{\infty} \sqrt{1 - k^2 \sin \theta}$ 

where K is the complete elliptic integral of the first kind The expression for the theoretical discharge (3-4-13) becomes

$$Q_t = 32 \sqrt{g} r^{5/2} k^4 \int_{F(\phi, k)}^{K} cn^2 u dn^2 u sn^2 u du$$
 (3-7-1.)

Upon performing the integration, and introducing the diameter b = 2r

$$\begin{aligned} \mathbf{Qt} &= \frac{4}{15} \sqrt{2g} \ b^{5/2} \left\{ 2(\mathbf{1} - \mathbf{k}^2 + \mathbf{k}^4) \left[ \mathbf{E} = \mathbf{E}(\phi, \mathbf{k}) \right] - (\mathbf{1} - \mathbf{k}^2)(\mathbf{2} - \mathbf{k}^2) \left[ \mathbf{K} = \mathbf{F}(\phi, \mathbf{k}) \right] \\ &- \mathbf{k}^2 \sin \phi \cos \phi \ \Delta \phi \ (3\mathbf{k}^2 \sin^2 \phi - \mathbf{1} - \mathbf{k}^2) \right\} \end{aligned} (3-7-17) \\ &= \int_{-\infty}^{\pi/2} \sqrt{1 - \mathbf{k}^2 \sin^2 \phi} \ d \ \phi \end{aligned}$$

where

 $E = \int_{0}^{\pi/2} \sqrt{1 - k^2 \sin^2 \phi} \, d\phi \qquad (3-7-18)$ 

which is the complete elliptic integral of the second kind, and

$$\mathbb{E}(\phi_{s}, \mathbf{k}) = \int_{0}^{\phi} \sqrt{1 - \mathbf{k}^{2} \sin^{2} \phi} \, \mathrm{d} \, 0 \qquad (3-7-19)$$

which is the incomplete elliptic integral of the second kind, and

 $k = \sqrt{\frac{Y_1 + Y}{b}}$ 

$$\phi = \sin^{-1} \sqrt{\frac{r}{Y_1 + r}}$$
(3-7-20)

and

$$4 \phi = \sqrt{1 - k^2 \sin^2 \phi} = \sqrt{0.5}$$
 (3-7-21)

and finally

Equation (3-7-17) yields the theoretical discharge for the flow through a semicircular constriction of diameter b =2r and where the maximum depth upstream of the constriction is y1. The quantities K, E,  $F(\phi, k)$ ,  $E(\phi, k)$  may be obtained from tables. Equation (3-7-17) is somewhat similar to that for the flow through circular weirs obtained by J. C. Stevens<sup>7</sup> which is

$$Q_{t} = \frac{4}{15} \sqrt{2g} D^{5/2} \left\{ 2(1 - k^{2} \div k^{4}) E - (2 - k^{2})(1 - k^{2}) K \right\}$$
(3-7-23)

where  $k^2 = H/D_9$  H being the head over the invert, and D is the diameter of the circular weir. Stevens also gives an infinite series approximation to equation (3-7-23) which is similar to equation (3-7-2):

$$Q_{t} = 2\sqrt{2g} D^{5/2} \left(\frac{1}{8}z^{2} - \frac{1}{32}z^{3} - \frac{5}{1024}z^{4} \dots \dots \right)$$
(3-7-24)

where z = H/D.

#### b) The Equation of Discharge for Orifice Flow

With reference to Fig. 3-11 and neglecting the velocity of approach, the discharge is found to be:

Q = 
$$\int V dA = 2 \text{ Gd} \sqrt{2g} \int_0^T (Y_1 - h)^{\frac{1}{2}} (r^2 - h^2)^{\frac{1}{2}} dh^2$$
 (3-7-25)

An approximate solution to the above integral is found by expanding each term of the integrand into a binomial series, multiplying the two series and integrating term by term. The result, using the 5 leading terms is

(3-7-22)

$$Q = 0.4019 C_{d} \sqrt{2g} Y^{\frac{1}{2}} b^{2} \left[ 1 - 0.2136 \left( \frac{r}{Y_{1}} \right) - 0.03216 \left( \frac{r}{Y_{1}} \right)^{2} - 0.0112 \left( \frac{r}{Y_{1}} \right)^{3} - 0.005344 \left( \frac{r}{Y_{1}} \right)^{4} \dots \right]$$
(3-7-26)

$$Q = C_1 \mathbb{Y}_1^{\frac{1}{2}} b^2 \mathbb{T}$$

where

n

$$C_1 = 0.4019 \sqrt{2g} \quad C_d = 3.22 \quad C_d \quad (3-7-23)$$

and

$$= 1 - 0.2136 \left( \frac{r}{Y_1} \right) - 0.03216 \left( \frac{r}{Y_1} \right)^2 - 0.0112 \left( \frac{r}{Y_1} \right)^3$$

$$-0.005344. \left(\frac{P}{Y_{1}}\right)^{4} \qquad (3-7-2\frac{1}{2})^{4}$$

The above equation is valid as long as the constriction is submerged, which means for Y1>r.

A simpler form of the orifice discharge equation when the approach velocity is very low relative to the velocity in the discharge jet, and when the orifice is considered as a whole is:

$$Q = C_d^2 \frac{T_d^2}{4} \sqrt{2g \, Y_1}$$
 (3-7-30)

where

Q = discharge through the opening

and

d = diameter of the opening

Y1 = the depth of the backwater

Cd = coefficient of discharge for type of inlet

Now rearranging equation (3-17-30):

T

$$\frac{q^2}{g d^4} = \frac{c_d^2}{32} \frac{\pi^2}{Y_1}$$
(3-7-31)

By dividing by  $d_s$  and taking the square root of both sides of  $(3-7-31)_s$  the following dimensionless equation is obtained

$$\frac{Q}{g^{\frac{1}{2}} d^{5/2}} = \frac{C_{d} \pi}{\sqrt{32}} \left( \frac{X_1}{d} \right)^{\frac{1}{2}}$$
(3-7-32)

where  $\frac{Cd \ F}{\sqrt{32}}$  is a constant.

The previous equation may be written in the general form

$$\frac{Q}{g\bar{z} d\bar{z}/2} = f(\frac{Y_1}{d})$$
 (3-7-33)

which is used in plotting experimental results.

#### III-8 Energy Loss in the Constriction

It has been shown that a complete theoretical analysis is not possible. However, some important conclusions regarding the classification of flow types and limiting cases have been obtained from momentum and energy considerations. The applicability of the momentum and energy equations is limited principally by the difficulty of formulating the drag force and the head losses. Iszard<sup>8</sup> and Bradley<sup>9</sup> have proposed an empirical relation for the head loss in a constriction. The head loss due to the contraction and expansion of the flow may be expressed as

$$h = K \frac{V_{2n}^2}{2g}$$
(3-2-1)

where V2n is a reference velocity, which is given by

$$V_{2n} = \frac{Q}{A_{2n}}$$
(3-8-2)

where  $A_{2n}$  is the area of section 2 corresponding to the normal depth in the unconstructed flow.

The energy equation, written between sections 1 and 4 is:

$$Y_{1} + \frac{\gamma_{1}^{2}}{2g} + S_{0} I_{1-4} = Y_{4} + \frac{\gamma_{4}}{2g} + E_{1-4}$$
 (3-8-1)

where  $E_{1-4}$  is the total energy loss between sections 1 and 4. This energy loss is made of two parts: the normal boundary resistance plus the additional loss due to the contraction and expansion of the flow:

$$E_{1-4} = S_0 L_{1-4} + K \frac{V_{12}^2}{2g}$$
(3-8-4)

Replacing in the energy equation

$$I_1 - I_4 = K \frac{V_D 2^2}{2g} + \alpha_b \frac{V_L^2}{2g} + \alpha_1 \frac{V_1^2}{2g}$$
 (3-8-5)

Since at section 4 the normal depth has been regained  $Y_4 = Y_n$ , and making use of the continuity equation

$$V_1 A_1 = V_4 A_4 = V_{n2} A_{n2}$$
 (3-8-6)

$$X_1 - Y_n = h_1^* = K \frac{V_{n2}}{2g} \int_{-\frac{1}{4}}^{\frac{1}{4}} (\frac{A_{n2}}{A_4})^2 - A_1 (\frac{A_{n2}}{A_1})^2 \int_{-\frac{1}{2g}}^{\frac{1}{4}} \frac{V_{n2}}{2g}^2 (3-8-7)^2$$

or approximately 
$$K = h_1 * / (I_{n2}^2/2g)$$
 (3-8-9)

 $K = \frac{h_1 + k_1}{V_2 - 2^2/2g} = \frac{k_1}{k_1} \left(\frac{An^2}{2}\right)^2 = \frac{a_1}{(An^2)^2}$ 

The same results are valid for submerged constriction, when the total cross sectional area of the constriction  $A_0$  is taken as reference area instead of  $A_{n2}$ . Then the head loss due to the contraction and expansion of the flow is written as

$$h_f = K \frac{V_0^2}{2g}$$
 (3-8-10)

where

$$V_0 = \frac{Q}{A_0}$$
(3-8-11)

and as before 
$$h_1^* = K \frac{V_0^2}{2g} \left[ \frac{V_4}{V_0} \left( \frac{V_4}{V_0} \right)^2 \frac{V_0^2}{2g} - \alpha_1 \left( \frac{V_1}{V_0} \right)^2 \right] \frac{V_0^2}{2g}$$
 (3-8-12)

and 
$$K = \frac{h_1^*}{V_0^2/2g} = \left[ \frac{A_0}{A_4} \left( \frac{A_0}{A_4} \right)^2 - \frac{A_1}{A_1} \left( \frac{A_0}{A_1} \right)^2 \right]$$
 (3-8-13)

In formulas 7-S-7, 7-S-8 and 7-S-12, 7-E-13, a simplification may be obtained by assuming  $\alpha_1 = \alpha_4 = \alpha_3$ 

An alternate simplified expression for the head loss in the contracting flow may be obtained by considering the contraction as an orifice, and by assuming that the depth at the orifice is the normal depth. With these assumptions, and neglecting the velocity head of approach  $V_{nl}^2/2g$ , the energy equation yields

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Introducing a cosfficient of valocity

$$\mathbf{C}_{\mathbf{y}} = \mathbf{C}_{\mathbf{y}} \left( \mathbf{M}_{\mathbf{y}}, \mathbf{F} \right)$$

where M' is the channel opening ratio, then

$$\nabla_{n2}^{2} = C_{y}^{2} 2g (\Sigma_{1} = \Sigma_{n})$$

Introducing an orblice Fronde number defined of

$$F_0^2 = \frac{V_{m2}^2}{gT_m^2} = 2C_y^2 + \frac{V_{m1}^2}{T_m^2} = 2$$

then

$$\frac{1}{n} \approx 1 \approx \frac{1}{2} \left( \frac{0}{2n} \right)^2$$

But as  $\left\langle \frac{\overline{r}_{0}}{\overline{r}_{1}} \right\rangle^{2} = \left\langle \frac{4\pi^{2}}{\overline{r}_{n1}} \right\rangle^{2} = \left\langle \frac{4\pi^{2}}{\overline{h}_{n1}} \right\rangle^{2} = \left\langle \frac{4\pi^{2}}{\overline{h}_{n1}} \right\rangle^{2}$ 

then  $\frac{V_{1}}{V_{1}} = \frac{1}{2C_{y}^{2}} - \frac{F_{1}}{W^{2}}^{2}$ 

The head loss through an orifice is given by

$$\frac{1}{\sqrt{2g}} = \frac{\sqrt{2g}}{2g}$$

so that the head loss in the contracting flow is, in dimensionless form

$$\frac{h_{f}}{y_{h}} = \left(\frac{1}{C_{V}^{2}} - 1\right) \frac{v_{h}^{2}}{2gIn} = \frac{1}{2} \left(\frac{1}{C_{V}^{2}} - 1\right) \left(\frac{F_{h}}{M_{1}}\right)^{2}$$
(3-8-22)

But, by means of (3-8-20)

$$\frac{1}{c_r^2} = \frac{2}{\frac{m_r^2}{(m_r^2)^2}} = \frac{2}{(m_r^2)^2}$$

and (7-8- 2) 1 scores:

$$\frac{h_{p}}{V_{n}} = \frac{Y_{1}}{Y_{n}} = \frac{1}{2} \left(\frac{F_{n}}{M}\right)^{2}$$

As experiments have shown that in general,

$$\frac{\mathbf{I}_{1}}{\mathbf{Y}_{n}^{1}} \sim \mathbf{f} \left(\frac{\mathbf{F}_{1}}{\mathbf{F}_{1}}\right)^{2}$$

then, in general

$$\frac{\mathbf{R}_{\mathbf{Y}}}{\mathbf{N}_{\mathbf{R}}} \approx \frac{\mathbf{P}_{\mathbf{X}}^{2}}{\mathbf{N}_{\mathbf{X}}} \frac{2}{\mathbf{P}_{\mathbf{X}}}$$

### III-9 Backwater Ratio Equations

The backwater ratio is defined as the ratio of the maximum anterline water depth to the normal depth of flow.

Expressions for the backwater ratio may be obtained from the energy equation or from the discharge equation. The use of the energy equation will be considered first.

The energy equation is written in dimensionless form by divising both sides by  $\mathbf{Y}_{n}$ :

$$\frac{h_{1}^{*}}{Y_{n}} = \frac{1}{2} \left( \frac{Y_{n}}{M^{*}} \right)^{2} \left[ \alpha_{I_{1}} M^{*2} - \alpha_{1} M^{*2} \left( 1 - 2 \frac{h^{*}}{Y_{n}} \right) - K \right]$$
(3-9-6)

Assuming of 1 = \$\alpha\_4 = \$\alpha\$ it follows that

$$\frac{h_1}{n} = \frac{1}{2} \left( \frac{Y_n}{N^{\gamma}} \right)^2 \left[ 2 \quad M^{\dagger} \quad \frac{h_1^{4}}{Y_n} + K \right]$$
(3-9-7)

or

$$\frac{\mathbf{h}_{1}^{*}}{\mathbf{Y}_{n}} \left[ 1 - \alpha \mathbf{F}_{n}^{2} \right] = \frac{1}{2} \left( \frac{\mathbf{F}_{n}}{\mathbf{M}^{*}} \right)^{2} \mathbf{K}$$
(3-9-8)

whence

$$\frac{\mathbf{h}_{1}^{++}}{\mathbf{I}_{n}} = \left(\frac{\mathbf{F}_{n}}{\mathbf{M}^{0}}\right)^{2} \quad \frac{\mathbf{K}}{\mathbf{I} - (\mathbf{F}_{n})\mathbf{Z}_{n}} \tag{3-9-9}$$

In general, the expression for the backwater ratio for free surgers flow is:

$$\frac{h_{\rm L}^{*}}{Y_{\rm L}} = D \left(\frac{F_{\rm H}}{M^{2}}\right)^{2} \tag{3-9-10}$$

where

$$D = \frac{1/2 K}{1 - (F_n)^2}$$
(3-9-11)

is a factor related to the head loss in the constriction.

This result is valid only when the flow is subcritical through the constriction and when  $h_1 * / Y_n < 1$ .

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Experimental results have yielded the empirical relation (for single span semi-circular arches, see Chapter V).

$$\frac{h_1}{Y_n} = 0.45 \left(\frac{F_n}{M}\right)^2$$
(3-9-12)

with this regult

K

$$D = \frac{1/2 K}{1 - F_n^2 \chi} = 0.45$$
(3-9-13)

or

$$(3-9-1)$$

or with a 1

$$K = 0.90 \ (1 - F_n^2) \tag{3-9.35}$$

The test results are presented in the form of plots of  $\frac{h_1^*}{T_1}$  vs  $(\frac{En}{M^*})^2$  and least square curves of the form

$$\frac{h_1^{\pi}}{Y_n} = C \left(\frac{F_n^2}{M^{1/2}}\right)^n$$
(3=9-16)

are obtained. A<sub>5</sub> from equation (3-9-10), D is the slope of the curve of  $h_1 * / T_n$ , vs  $(F_n/M^2)^2$ , it follows from equation (3-9-16) that

$$\frac{d(\frac{h_1}{Y_n})}{d(\frac{F_n}{M^0})^2} = nc \left(\frac{F_n^2}{M^0/2}\right)^{n-1} = 0$$
(3-9-17)

Alternate expressions for the backwater ratio may be obtained from the discharge equation for free surface and for orifice flow separately.

Consider the free surface flow case first. The equation for the discharge through the constriction is (from equ. 3-7-3)

$$Q = C_1 Y_1^{3/2} b T_1$$
 (3...9-13)

where

$$C_1 = Cd \frac{17}{24} \sqrt{2g}$$
 (3-9-19)

and

$$T_1 = 1 = 0.1294 \left(\frac{Y_1}{r}\right)^2 - 0.0177 \left(\frac{Y_2}{r}\right)^4$$
 (3=9-20)

The discharge in the approach channel is

$$Q = \nabla_n A_n = F_n \sqrt{g} = B Y_n^{3/2}$$
(3=9-21)

Equating the two expressions for the discharge one obtains

$$\frac{Y_{1}}{Y_{n}} = \left[\frac{12\sqrt{2} F_{n}}{170 d MT_{1}}\right]^{2/3}$$
(3-9-22)

or in general

$$\frac{x_1}{x_n} = c \left(\frac{F_n}{M^*}\right)^{2/3}$$
(3+9-2)

 $C = \frac{12}{17} \frac{2}{C_{\rm d}} \frac{2/3}{T}$ (3-9-2)

Since  $M = M^{\circ}/C_{m}$  (see equation 3-10-8)

Experimental results for free surface flow have yielded the following relation (see chapter VII)

$$\frac{Y_1}{Y_n} = 1 + 0.47 \left[ \left( \frac{Y_n}{M} \right)^{2/3} \right] 3.39 \qquad (3-9-2)$$

It has been observed that the equations derived by several different investigators for the backwater ratio produced by various constriction geometries seem to have a basic similarity. As an example, equation (23) in the present text for  $y_1/y_n$ appears to be a function of  $(F/M^2)^{2/3}$ .

$$y_1/y_n = g_1 (F_n/M^2)^{2/3}$$
 (3=9-20)

An equation for the backwater ratio given by Valentine<sup>8</sup> for lateral constriction plates is

$$y_1/y_n = (g F_n/C M)^{2/3} = g_2 (F_n/M^2)^{2/3}$$
 (3=9-27)

where C = a discharge coefficient

and  $M = b/B = M^{\circ}$  since  $C_M = 1$ 

Aluo Liu et al<sup>ll</sup> present an empirical formula for a two dimensional vertical board model

$$(h_1*/h_n)^3 = 4.483 F_n^2 (\frac{1}{M^2} - \frac{2}{3}(2.5 - M)) - 1$$
 (3-9-28)

where  $M = b/B = M^{\circ}$  since  $C_{M} = 1$ Considering only the leading term  $1/M^{2}$  of the quantity in brackets, equation (3=9-28) becomes

$$h_1^{*}/y_n = g_3 (F_n/M^{\circ})^{2/3}$$
 (3=9-29)

It appears that with the proper interpretation of the variables, namely  $M^{\circ}$  and  $F_n$ , the results of tests performed on different geometric shapes of bridge openings should produce the same results. For instance, a vertical abutment deck type bridge may physically appear completely different than a semicircular arch bridge. However, hydraulically speaking if they have the same opening ratio  $M^{\circ}$ , they should produce the same backwater ratio. The limitations of the assumption must necessarily lie in the fact that both bridges must have the same eccentricity, skewness and entrance rounding conditions. It is believed that this concept applies equally as well to multiple span bridges. An attempt has been made to compare the two dimensional semi-circular test results of the author, the segment data obtained by A. A. Sooky, and the Vertical Board (VE) data as given by Liu<sup>11</sup>. The results of this comparison will be shown and discussed in a later section.

A similar expression for the backwater ratio may be obtained from the theoretical discharge equation for orifice flow. The equation for the discharge through the constriction was found to be: (see equ. 3-7-27)

$$Q = C_1 Y_1^{\frac{1}{12}} b^2 T$$
 (3-9-30)

The discharge in the approach channel is:

$$Q = V_n A_n = F_n \sqrt{g} B Y_n^{3/2}$$
 (3-9-31)

Equating the two discharge equations (3-9-30 and 3-9-31):

$$c_1 Y_1^{\frac{1}{2}} b^2 T = F_n \sqrt{g} Y_n^{3/2} B$$
 (3-9-32)

$$C_{1} = \left(\frac{Y_{n}}{Y_{1}}\right)^{\frac{1}{2}} \frac{F_{n}}{b^{2}T} = \left(\frac{Y_{n}}{Y_{1}}\right)^{\frac{1}{2}} \left(\frac{B}{b} \cdot \frac{Y_{n}}{b}\right) F_{n} = \frac{1}{T}$$
(3-9-33)  
$$\frac{B}{b} \cdot \frac{Y_{n}}{b} = \frac{A_{n}}{k A_{0}} = \frac{1}{k M^{2}}$$

but

where

Now rearranging the equation (3-9-33):

$$\frac{Y_1}{Y_n} = \left(\frac{1}{k M^2} + F_n + \frac{1}{C_1} + \frac{1}{T}\right)^2$$
(3-9-34)

or in general a relation of the following form would be expected

$$\frac{Y_1}{Y_n} = C \left(\frac{F_n}{M^{\nu}}\right)^2.$$
(3-9-35)

It has been found empirically that, instead of equation (3-9-35), the experimental results can conveniently be represented by a function of the form:

$$\frac{h_1^*}{Y_n} = \frac{Y_1}{Y_n} = 1 + f \left(\frac{F_n}{M'}\right)^2$$
(3-9-36)

where h, " is the backwater superelevation.

Experimental results for the orifice flow have yielded the following relation (see Chapter VII)

$$\frac{\mathbf{Y}_{1}}{\mathbf{Y}_{n}} = 1 + 1.18 \left[ \left( \frac{\mathbf{F}_{n}}{\mathbf{M}^{0}} \right)^{2} \right] 0.90 \qquad (3-9-37)$$

00

$$\frac{Y_1}{Y_n} = 1 + 1.18 \left(\frac{F_n}{M^2}\right)^{1.80}$$
(3-9-38)

The experimental result for the backwater ratio equation for free surface was developed by P. F. Biery and found to yield the relation (see chapter V)

$$\frac{Y_1}{Y_n} = 1 + 0.47 \left[ \left(\frac{F_n}{M^2}\right)^{2/3} \right] 3.39$$
(3-9-39)

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$$\frac{Y_1}{Y_n} = 1 + 0.47 \left(\frac{y_n}{M}\right)^{2.26}$$
(3-9-40)

or

#### III-10 Dimensional Consideration

#### a) Dimensional Analysis

Figure 3-1 shows a definition sketch of the effects of a channel constriction on the water surface profile. Section view B illustrates the type of centerline profile obtained with a Class I flow. This is the most generally occurving situation that appears in actual practice. In the figure  $Y_0$  or  $Y_n$  is the normal depth of the unobstructed channel.  $Y_1$  is the depth at the point of maximum backwater elevation.  $Y_2$  is the depth at the section of minimum jet area or the vena contracta.  $Y_3$  is the minimum water depth of the regain curve, and  $Y_L$  is at a point sufficiently downstream from the contraction where the flow returns to the normal depth.

For any physical problem such as this, a dimensional analysis is convenient for the purpose of guidance and interpretation of a testing program. In this manner, the basic variables can be grouped into dimensionless quantities and their relationships investigated. In the problem at hand, it is desired to determine the maximum water depth upstream of the constriction. It is assumed that the variables which govern the backwater superelevation may be grouped into three categories as follows: the fluid properties, the kinematic and dynamic variables, and the dimensions defining the boundary geometry. Due to the two dimensional character of the constriction, the latter is expressed in terms of flow areas rather than the usual linear dimensions. (See figure 3-1 for an illustration of the terminology.) The variables are:

- a) Fluid Properties
  - 1)  $\gamma$  , the kinematic viscosity of the fluid
  - 2) e , density of the fluid
- b) Kinematic and Dynamic Flow Variables
  - 1) g, acceleration of gravity
  - 2) Y1, maximum water depth upstream of the constriction

- 3) Ing the normal depth of flow in the approach channel.
- 4)  $V_{\rm He}$  the velocity of flow at normal depth
- 5) n, Manning's roughness coefficient of the approach channel
- c) Properties of the Constriction Geanetry (see figures 2, 4, and 24 for definition of symbols).
  - 1) Anl, the total normal depth flow area at section 1.
  - 2)  $\Lambda_{n2_s}$  the normal depth flow area in the opening of the constriction (see Fig. 4).  $\Lambda_{n2} = \Lambda_0$  for submerged bridges.
  - L/b, thickness factor, where L is the length of bridge in direction of flow, b is the width of opening at the bottom.
  - bL<sub>d</sub>/A<sub>n2</sub>, distance factor where L<sub>d</sub> is the distance between two parallel identical bridges.
  - 5) \$1, wing wall angles
  - 6)  $\phi_{29}$  skow angle
  - 7)  $\beta_{2}$  segment factor defined by = d/r where d is the distance between the channel bottom and the center of the circular segment.
  - 8) N<sub>9</sub> number of spans
  - 9) w, eccentricity defined by  $e = 1 c/a_s$  where c and a are the width on either sides of the bridge openings.

From the above list of variables,

$$I_{1} = I_{1} (Y_{n}, V_{n}, n, v, h, V, e, g, a, A_{n1}, A_{n2}, L/b, \frac{bLd}{A_{n2}}, \beta_{1}, \beta_{2}, e, \beta_{n}, N)$$

Buckingham's theorem states that in a physical problem including n quantities in which there are m dimensions, the quantities may be arranged into (n-m) dimensionless parameters. With the mass, length and time system of units, the n-m or 14 dimensionless parameters are as follows:

$$\frac{1}{Y_{n}} = f_{2} \left( \frac{V_{n}^{2}}{Y_{ng}}, \frac{V_{n}Y_{n}}{Y_{n}}, \frac{n}{Y_{n}^{1/6}}, \frac{A_{n1}}{Y_{n}^{2}}, \frac{A_{n2}}{Y_{n}^{2}}, \frac{h}{Y_{n}}, \frac{L}{b}, \frac{bL_{d}}{A_{n2}}, \phi_{1}, \phi_{2}, e, \beta, N \right)$$

where  $\frac{V_n^2}{V_n g}$  is the square of the Froude number  $F_n$  and  $\frac{V_n Y_n}{v}$  is the Reynolds number R. It is known that:

1) Viscous forces play a negligible role in open channel flow. So the term R can be neglected. Furthermore, by combining the ratio  $A_{n2}/Y_{n2}$  and  $A_{n1}/Y_{n}^2$  into  $A_{n2}/A_{n1} = M^2$ , and letting n be constant, which means that the boundary roughness in the flume is kept the same, then the above equation simplifies to:

$$\frac{Y_1}{Y_n} = \langle (Fn, M', \frac{L}{b}, \frac{bLd}{A_{n2}}, p_{1}, p_{2}, e_{\beta}\beta', N \rangle$$
 (3-10-1)

Of the nine variables only two, the Froude number and the contraction ratio describe the flow field. The other seven variables in the dimensionless analysis describe the different model geometries that could be tested separately. For a definition of the separate geometries see section on Definition of Test Geometries.

# b) Definition of Channel Opening Ratio and Channel Midth Ratio

The channel opening ratio (M\*) is defined as that portion of the total normal depth flow that can pass through the bridge waterway without contraction. By definition it is equivalent to the ratio  $\Lambda_{n2}/\Lambda_{n1}$  obtained from the dimensional analysis. Along with the normal depth Froude number, the opening ratio is perhaps the most critical variable in the problem.

Referring to figure 3-12-a for the rectangular case, the total flow is that flow in area ADEH, and the flow that passes through the bridge opening without contraction is that represented by the area ECFG. Therefore the opening ratio M<sup>4</sup> is

$$M^{\circ} = q/Q$$
 (3-10-2)

If we assume that there is a constant uniform velocity  $V_n$  across the entire normal depth section, equation ( 3-10-2 ) becomes

$$1' = q/Q - A_{n2}V_n / A_{n1}V_n = A_{n2} / A_{n1} = by_n / By_n = b/B$$
(3-10-3)

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However, for an arch bridge, as shown also in figure (3-12-5), the surface with will be different for each and every normal depth  $y_n$ . Therefore in the same memory

$$M' = q/Q = A_{n2}V_{n}/A_{n1}V_{n} = A_{n2}/A_{n1} \qquad (3-10-4)$$

The ratio of the two areas is clearly not equivalent to b/B. (for simplicity, b/B is hereafter defined by the symbol M)

For the portion GIKF of a semicircular arch with radius r and depth  $y_{\rm R}$  the area becomes

$$A_{n2} = \int_{0}^{y_{n}} 2\sqrt{r^{2} - y^{2}} \, dy = 2 \left[ \frac{1}{2} \left\{ y_{n} \sqrt{r^{2} - y_{n}^{2}} + r^{2} \sin^{-1} y_{n} r^{2} \right\} \right] (1-1)$$

The segment of arch WECF shown in figure 3-12-b has a radius v and springline width b. The arch has been superimposed upon the flow area of depth  $y_{fl}$ . The center of curvature is at a distance d below the springline of the arch. The flow area (A<sub>nl</sub>) of the rectangular channel is  $By_{fl}$ , while the area GROF is given by

$$A_{n2} = \int_{0}^{D} 2\sqrt{r^{2} - y^{2}} \, dy - \int_{0}^{d} 2\sqrt{r^{2} - y^{2}} \, dy \qquad (3-20-6)$$

and the corresponding channel opening ratio is

$$M^{*} = A_{n2}/A_{n1} = \frac{D\sqrt{r^{2} - D^{2} + r^{2} \sin^{-1}D/r}}{By_{n}} = \frac{d\sqrt{r^{2} - d^{2} + r^{2} \sin^{-1}D/r}}{By_{n}}$$
(2.30 fb)

The channel opening ratio M<sup>4</sup> can be expressed in terms of three other dimensionles, ratios: the ratio of span to channel width  $M = b/B_s$  the ratio of depth of the arch center below the streambed to the arch radius  $\eta = d/r$  and the ratio of formal depth to arch radius  $\int = y_n/r$ . The channel opening ratio of equation (3-20-7) may thus be expressed as:

$$M^{\circ} = MC_{M}$$

in which M = b/B (3-10-8)

and the channel opening ratio coefficient is

$$C_{M} = \frac{1}{2} \left[ \left\{ \frac{\sqrt{1 - (n + 5)^{2}} + \frac{1}{n + 5} \sin^{-1}(n + 5)}{\frac{5}{n + 5} + \frac{1}{n + 5} \sin^{-1}(n + 5)} - \frac{\sqrt{1 - n^{2}} + \frac{1}{n + 5} \sin^{-1}(n + 5)}{\frac{5}{n + 5} + \frac{1}{n + 5} \sin^{-1}(n + 5)} \right]$$
(3-10-1)

with

and

$$3 = y_n/r$$

In the form of equation (3-10-8) the value of M = b/B is adjusted for the particular arch by an amount equivalent to  $C_M$  such that  $M^*$  is the same as the ratio of  $A_{n2}$  to  $A_{n1}$ . In the more general case, the values of 5 and n can take on numbers within certain limits, before the normal depth will submerge the crown of the arch. The limits are as follows:

For 
$$5 = y_n/r$$
  $0/r \le y_n/r \le (r-d)/r$  (3=10=10)  
or  $0 < 5 < (1 - n)$   
For  $\eta = d/r$   $0 < n < 1$  (3=10=11)

When  $\eta = 0$ , the case of a semicircular arch with the center of curvature at the springline exists. When  $\eta = 1$ , the contraction reduces to two parallel abutments.

The values of  $C_M$  have been calculated for several values of 3 and n and are summarized in the graph of figure 3-13. The submergence limit represents the upper limits of both 3 and  $\eta_0$ . The segment arch which is a constant radius arch with its center of curvature below the springline of the arch (i.e.  $\eta > 0$ ) can be used as an arch in its own right or as an approximation to an elliptical or a multiple radius arch. The value of M<sup>0</sup> for the elliptic and multiple radius arch could be determined directly from equation (3-10-4).

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Nine independent variables were considered in the dimensional analysis to determine the dependent variable  $T_1/T_n$ . Of these nine variables, seven describe different types of constriction geometries the other two, the Froude number and the contraction ratio describe the flow field and the amount of contraction. These seven geometric variables were used to define seven types of geometries that were tested separately. These types of geometries are defined laters.

### Geometry I-a

Two-Dimensional Semicircle Arch Bridge Constrictions:

(Figure 3-14) The characteristics of this type of geometry are the following:

- a) Since the model is two-dimensional.
- b) Single bridge case,  $L_{d} = 0$  and  $\frac{bLd}{An2} = 0$
- c) No wingwalls, according to definition of  $\beta_1$ ,  $\beta_2$  =  $00^{\circ}$
- d) None-skew case, according to definition of  $p_{2s} p_{2s} = 0^{\circ}$
- e) Semicircular case, d 0 and  $\beta = \frac{d}{d} = 0$
- f) One span case, N 🖷 📜
- g) No eccentricity, e = 0
- h) Froude number,  $F_n = \frac{V_n}{(gy_n)^2}$  designated as an independent variable.
- Constriction opening ratio M<sup>\*</sup> designated as an independent channel opening ratio M<sup>\*</sup>

Hence, seven independent descriptive variables,  $\frac{D}{r_0}$ ,  $\frac{bLd}{An2}$ , e,  $\beta_{1s}$ ,  $\beta_{2s}$ ,  $\beta_{1s}$  and N in the dimensionless equation: (3-10-1),  $\frac{Y_1}{Y_n} = f(F_{ns}, M^s)$ ,  $\frac{L}{bs}$ ,  $\frac{bLd}{An2s}$ ,  $\beta_{1s}$ ,  $\beta_{2s}$ ,  $\varepsilon_s$ ,  $\varepsilon_s$  are designated as constants and we get the relation:

$$\frac{\mathbf{Y}_1}{\mathbf{Y}_n} = \mathbf{f} \left( \mathbf{Z}_{ns} \ \mathbf{M}^{\mathsf{r}} \right)$$
### Geometry I-b

### Three-Dimensional Semicircular Arch Bridges:

This geometry differs from the previous one only in that L is varied, thus e parameter  $\frac{L}{b}$  describing the length of the model in the direction of the flow a variable and the dimensionless equation (3-10-1) for this case becomes

$$\frac{Y_1}{Y_n} = f \left(F_{n,p} M^0, \frac{L}{b}\right)$$

### Geometry II

### Dual Parallel Three-Dimensional Arch Bridge Constrictions:

This geometry consists of two identical bridges of geometry I-b, placed at iistance L<sub>d</sub> apart, measured center to center. One new variable, L<sub>d</sub>, is introduced, ich is characterized by the parameter  $\frac{BL_d}{A_{n2}}$  (Note: For submerged bridge constricons,  $A_{n2} = A_0$ , see Chapter III). Equation (3-10-1) simplifies for this case to:

$$\frac{Y_1}{Y_n} = f (F_{ns} M^{\dagger}_{s} \frac{bLd}{A_{n2}})$$

### Geometry III

Three-Dimensional Arch Bridge Constriction with Wingwalls:

The geometric characteristics were as follows:

- a)  $\frac{L}{b} = 0.25$  e)  $\beta = 0$ b)  $\frac{bLd}{An2} = 0$  f) N = 1
- c)  $g_1$  is a variable  $0_1 = 90^\circ$ ,  $60^\circ$ ,  $45^\circ$ , or  $30^\circ$  g) o = 0
- d)  $p_2 = 0^{\circ}$  h)  $F_n$  is a variable

### i) M' is a variable

Hence, equation (3-10-1) can be simplified to:

$$\frac{Y_1}{Y_n} = f(F_n, M, \phi_1)$$

### Geometry IV

Two-Dimensional Sanicircular Arch Bridge Constrictions with Second Arity:

The geometric characteristics were as follows:

a) 
$$\frac{1}{b} = 0$$
  
b)  $\frac{bL_d}{An2} = 0$   
c)  $\phi_1 = 90^0$   
d)  $\phi_2 = 0^0$   
e)  $\beta = 0$   
h)  $F_n$  is a variable  
h)  $F_n$  is a variable

And the equation (3-10-3) can be reduced to:

$$\frac{Y_1}{Y_n} = f (F_{n_2} M_{2}^s) e)$$

### Geometry V-a

Two-Dimensional Semicircular Arch Bridge Constrictions with Skew:

The geometric characteristics were as follows:

a) 
$$\frac{L}{h} = 0$$
  
b  
b)  $\frac{bL_d}{A_{n2}} = 0$   
c)  $\beta = 0$   
c)  $\beta = 0$   
d)  $O_2$  is a variable g)  $0 = 0$   
h)  $F_n$  is a variable

c) \$1 = 90° f) N = 1 i) M' is a variable

And the equation (3-10-1) can be reduced to:

$$\frac{Y_1}{Y_n} = f \left(F_{n,9} M_{s,2}^{s}\right)$$

### Geometry V-b

# Three-Dimensional Semiclocular Arch Bridge Constrictions with Skew:

This geometry is the same as the previous one, except that the length of the nstriction is allowed to vary, the parameter  $\frac{1}{2}$  is thus a variable and the dimenonless equation (3-10-1) can be simplified to:

$$\frac{L_1}{Y_{\rm II}} = f \left(F_{\rm II}, M^{\rm o}, \dot{p}_2, \frac{L}{\rm b}\right)$$

### Geometry 71

Three-Dimensional Two-Span Semicircular Arch Bridge Constrictions:

The geometric characteristics were as follows:

a)  $\frac{L}{m} = 0.50$ f) N = 2, Two-span case b)  $\frac{bL_d}{An2} = 0$ g) e = 0 c)  $p_1 = 90^{\circ}$ h) F<sub>n</sub> is a variable d)  $\phi_2 = 00$ 1) M' is a variable e) p = 0 j)  $p = \frac{b}{10}$  (for submerged tests)  $p = \frac{b}{3}$  (for free surface tests)

The equation (3-10-1) can be reduced to:

$$\frac{Y_1}{Y_n} = 1 \ (F_{ns} \ M^0)$$

#### Geometry VII

# Two-Dimensional Segment Arch Bridge Constrictions:

The geometric characteristics were as follows:

a)		е)	ß is a variable
b)	$\frac{bL_d}{A_{n2}} = 0$	f)	N = 1
c)	<b>9</b> <sub>1</sub> = 90°	g)	e = 0
d)	\$2 = 00	h)	F <sub>n</sub> is a variable

Hence, equation (3-10-1) reduces to:

$$\frac{Y_1}{Y_n} = f (F_n, M, \beta)$$

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A preliminary investigation was conducted in a small flume for the purpose of evaluating the design requirements for a larger testing flume and of establishing the testing procedure. The small flume also provided a facility where some preliminary tests could be run with scales and relative roughnesses different from those used in the large flume.

### IV-1 Small Flume, Models and Test Conditions

For the purpose of preliminary testing, a small variable slope flume 6" wide and 12° long was used. The channel sides and bottom were constructed of lucite and carefully aligned by means of adjusting screws. (Fig. 4-1) The slope of the flume was controlled by a hand operated scissor jack at the lower end of the flume. An aluminum I-beam mourted horizontally above the flume served as a track for the mechanical and electric point gages used in obtaining the water surface measurements. The electric point gage consisted of two metal points of slightly different length that were wired to a set of batteries and a galvanemeter. When the shorter metal point would make contact with the water surface, the longer one being already submerged, the circuit would close and the galvanemeter would deflect. The flow was metered by a 1 inch orifice plate in a 2 inch supply line. Two and three dimensional tests were run with both smooth and rough boundaries. For the rough tests, the bottom and the walls were lined with copper wire mesh of 16 meshes per inch.

The two dimensional semicircular models wars constructed with diameters of  $\beta_{\mu}$ 4 and 5 inches (see fig. 4-2). The material used was brass. The edges were machined to 1/32 of an inch and then beveled to a 45 degree angle. The two dimensional segment models were of the same type of construction as the semicircular models and had a value of  $\eta = d/r$  equal to 0.5. Three three-dimensional models were built for the purpose of this preliminary testing.

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The material used was "Lucite". The dimensions of the three-dimensional models are the following:

Lucite	Langth of Model	Arch Diameter	Rise
Model	Along Stream in	in Inches	Inches
	LINCNES	Constant of the second s	
1	9.7	7 ° 7	<b>1.4</b>
2	24	5	2.5
3	24	4	2

A photograph of the models is given in Figure 4-2. Model No. 1 is a reproduction to a scale of 1/60 of the Arch bridge in Clay County, Indiana, on State Road 246e over Branch Connley Ditch. For measuring the water surface under the bridge, vertical glass tubes were installed on the models as shown in Figure 4-4. Through these tubes the probe of an electrical point gauge could be introduced.

In all cases, an adjustable weir was used at the end of the flume. This permitted to increase the depth of flow, and to reduce the length of the M-2 curve at the free overfall so as to obtain a long test section with uniform flow. The weir height varied for the different tests. The weir heights used \$F\$ indicated for the several tests.

The three-dimensional model tests were conducted for the following variable conditions:

10	Wall roughness:	The channel sides and bottom were either smooth or rough
		as described in the previous paragraph,
2.	Slops:	Three different slopes were used: $\beta_0 = 0_s S_0 = 0.0003$
		S <sub>o</sub> = 0₀0005₀
3.	Discharges	The discharges used in cubic feet per second are:
	Q = 0,0114 0,0138 0,015	(with 0.413 in orifice meter)
	0.017 0.023 0.028	(with 1.034 in orifice meter)
	0.0379	11
	0.042	11
	0.0519	17
	0 0806	

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4. Models: Three models were used, designated as Nos. 1, 2, 3 and as discribed in previous paragraphs.

The steps of the test routine were the following:

- Adjustment of horizontality of beam on which the point gauges travel. This
  was done by maintaining a pool of water in the flume, and taking the point
  gauge readings of the water surface at several points. The beam position
  was adjusted so that the point gauge readings were the same all along the
  beam.
- 2. Control of the channel slope. This was accomplished by adjusting the scissor jack until the difference between the point gauge readings of the bottom of the flume at two points 10 feet apart would be equal to ten times the desired slope.
- Establishment of the flow. The flow was adjusted by means of a control
  valve so as to obtain the desired reading on the orifice meter manometer.
  The desired readings were obtained from the calibration curves.
- 4. Establishment of uniform flow. The adjustable weir at the downstream end of the flume was set so as to obtain uniform flow through most of the flume. The flow was considered uniform in a reach when constant depths were observed in that reach. Depth was measured at one foot intervals. The section of the flume with uniform flow was the test section.
- 5. Installation of model in flume. The models were installed in the test section so as to show the complete regain curve and as much as possible of the backwater curve. In the case of the tests with rough channel walls, no roughness was installed on the model. The bottom of the flume under the arch was covered with artificial roughness for models 2 and 3 only. For model 1 the roughness was not installed on the bottom under the bridge, but was installed along the rest of the flume.

6. Observation of the water surface profile. The water surface elevation was first measured with a mechanical point gauge along the center line of the flume upstream and downstream of the model. At a later time vertical glass tubes were installed on the models. The water surface elevation was then measured with an electrical point gauge along the centerline of the arch. The test data are tabulated in the Appendix. (Table 4-1)

### -2 Test Results

The observations of the water surface profiles observed are tabulated in Table . of the Appendix.

<u>Col. 1</u> is the station as read from the tape on the horizontal beam. The tape is graduated to oll foot from downstream (station 4) to upstream (station 14).

<u>Col. 2</u> is the distance along the fitme in the direction of the flow. Station 14 corresponds to zero and station 4 to 10.

Col. 3 is the flume bottom point gauge reading. At each 1 foct or 2 foot interval the bottom elevation was observed with the mechanical point gauge.

Gol. 4 is the water surface reading obtained with the mechanical point gauge before fixing the model, i.e., with uniform flow.

<u>Col. 5</u> is the water surface reading upstream and downstream of the model obtained with the mechanical point gauge and the water surface under the bridge obtained with the electrical point gauge.

Col. 6 is the difference between Col. 4 and Col.  $3_s$  namely the normal water depth.

<u>Col. 7</u> is the difference between Col. 5 and Col.  $3_f$  namely the depth of water with model in place.

<u>Col. 8</u> is used for general remarks such as: slope, diameter, rise, length of model, tail weir height that were used in the particular set of readings were noted in this column,

The data pertaining to the rough boundary, two dimensional, segment arch tests the small flume and the calculations necessary for the analysis of the segment are given in Table 4-2.

The results of the two-dimensional weir tests were put in graphical form by ting the coefficient of discharge <u>vs</u> the Froude Number with the channel width o as the parameter. The channel width ratio M is defined as the ratio of the weir diameter b to the flume width B. This graph is shown in the upper left corner of Figure 4=5. The lower graph shows the relation of the Froude Number and the ratio of depth upstream of the weir to the normal depth.

A typical water surface profile for the three-dimensional arch bridge models is shown in Figure 4-5. In that case, the flume walls were lined with copper wire mesh of 16 meshes per inch. This gave a Manning's roughness coefficient of approximately 0.025, which is typical of many canals and natural streams. Figure 4-7 shows the results of the three-dimensional tests with smooth boundaries using bridge models of width L = 24 inches. The coefficients of di charge Cd and the ratic of the backwater depth Y1 to the normal depth Y0 are plotted yz the Froude Number for several values of the channel width ratio M. The results of the two- and threedimensional tests with smooth boundaries of the ratio  $\frac{Y1}{Y_0}$ . are approximately the same for the two cases. For higher Froude Numbers, the threedimensional tests exhibit smaller values of Cd and larger values of  $\frac{Y1}{Y_0}$ .

As part of the preliminary testing, a series of 93 tests were run in the small flume on two dimensional segment weirs (Geometry VII) with a n = d/r value of 0.5 (See Table 4-2) The data obtained were reanalyzed in terms of the channel opening ratio M<sup>4</sup>. These tests were run in the small channel with rough boundaries which had a Manning's n of 0.0201. Results were plotted in the same manner as the large flume rough tests and are shown in Figure 4-Ea.

#### V EXPERIMENTAL EQUIPMENT

### -1 Design and Construction of the Testing Flume

Before the design of the flume itself could proceed, it was necessary to letermine whether the backwater and regain phenomena could be represented to a convenient and casily measurpole scale in the space available in the hydraulics aboratory.

Several sots of arch bridge plans provided by the Indiana State Highway Departeast were analyzed for the values of backwater. The theory of varied flow and the equations and tables presented by Bichmeteff<sup>(1)</sup> and by the U.S. Eureau of Public loads<sup>(2)</sup> were used. It was possible to make only an approximate calculation of the mackwater because of the unknown effect of an arch-type constriction.

One criterian for the selection of the flume size was that it had to be sufficiently ong to accompose the full length of the regain curve downstream of the constriction, the bridge width and part of the backwater curve upstream of the constriction within the test section of the flume. The test section of the flume includes only that pirt of the flume where uniform flow can be obtained. The upstream end of the flume where the flow is developing, and the downstream end which is affected by the survature of the free surface are not suitable for testing. The limitations on the flume size were imposed by the space available and by the capacity of the water supply.

A calculation was made for bridge S79 in Clay County in Indiana. This bridge has a span of 30 feet and the stream has a width of 46 feet. The backwater curve and the regain curve were calculated assuming a velocity of 6 ft/sec in the constriction. The prototype bridge had a width of 48 feet and the calculated length of the regain curve was 146 feet. At a scale of 1:10 this would represent a length of 19.4 eet. In a test section of 30 feet in length, it would be possible to reproduce a 40 foot section of the upstream backwater. Taking into account the boundary layer growth at the entrance of the flume and the drop-off curve at the downstream end, a total length of about 60 feet was required. A width of 5 feet and a depth of 2 feet was consistent with the scale of 1/10 for this bridge crossing. These flume dimensions required a pumping capacity which was close to the maximum supply available of 2100 GPM.

From this and similar computations and from the small flume tests<sub>p</sub>(1) it appeared that a flume length of 64 fest utilizing all of the available space in the laboratory would be satisfactory.

The width of the flume was fixed at 5 feet. This was based on a consideration of the scale ratios and the space available. The cross section of the flume was to be rectangular since this configuration lent itself wall to bot, case of construction and adaption.

In order to test the flows under varying slope conditions, the flume was to be illtable about one end. Screw jack were selected for the slope convol because of accuracy and ease of operation, is will as permanence and appearance.

In order to keep the deflection: due to the variable weight of water within the same order of magnitude of the small of readings of the point gay for detth measurement, 0.1 mm, the flume bottom was designed of 1/4 inch steel plat supported at a foot intervals on channels. The cartals in turn were to be supported by two main means riding on the jacks. The side plates were designed of 1/4 inch steel plate suppirted by vertical angles resting on the channel members. A long'tud'hal horicontal angle mounted on the vertical angles served at a support for the guide rails, the guide rails, which serve as a reference plane from which measurments the mased, area to be polished steinless steel to minimize correction and scale.

The design was based on a possible water depth of 2 feet. The distance between eam supports was set as 20 feet. Simply supported connection was assumed. The being were selected so that their deflections would be negligible. The beam first selected was an 18 I 54.7 which gave a calculated deflection of 0.00225 feet under the design loading. Contacts with the fabricator and erector were made at a later date and it was found that a 20 I 65.4 would be available at a cost less than that of the lighter beam. The use of the heavier beam was accepted and the design proceeded based on this beam. The deflection due to the variable water weight was approximately 0.002 feet for a depth of one foot. Adjustment bolts were provided for transverse leveling. The adjustment bolts were placed between the channels and the main beams. This arrangement gave the bottom plate full support acreas its width at 2 foot intervals.

The bottom plate was issigned slightly wider than the flume width. This permitted attaching an angle to hold the oottom edge of the vertical plate fixed. The upper edge of the vertical place hid nuts weld d on at the two foct points. Studs were attached between the suts and the virtical angles to support the plate and provide an adjustment for its longitudinal alignment. The inside of the flume was finished with an epoxy resin applied with a hand roller. The flume construction is shown in figure 5-1.

#### Tailgave

Control over the depth was exercise by a gate mounted at the end of the flume. The gate was manually operated from the lie of the flume. Figure 5-3 shows the gate. The gate was made in such a way that it could be used either as a sluice or as a weir. Throughout the tests, the same was used exclusively as a weir.

#### l'or abay

The forebay, 8 feet wide and 10 feet long, was constructed of plywood and lined with sheet metal; and is shown in figure 5.4. The 3 inch and 6 inch pipes entered the rear of the forebay at the top. The 6 inch line was centered and the 5 inth line was placed slightly off center. The diffusing mechanism for each supply line consisted of a tee and cross pipe of the same size as the line at the bottom of the forebay. The turbulence level of the entering water was controlled by a 4 inch gravel baffle and three wire mesh screens of 13 mech per inch. The transition section continuing

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into the bottom and side walls of the flume was made of quarter ellipses in the horizontal and vertical planes respectively with a ratio of major to minor axes of 1.5 to 1.0. The joint between the flume and the forebay was sealed with a fluxible rubber gasket mounted so as not to interfere with the flow.

#### Slope Control Mechanism

The flume rests upon six screw jacks and a hinge. The hinge is located at the joint of the flume and the forebay. The jacks are similar in all respects with the exception of the gear ratio. The jacks are divided into three pairs with rates of raise of one, two, and three inches for 96 turns of the shaft. Since the hinge was a fixed point and the opposite end of the flume was use point of maximum movement, the jacks were arranged such that the pair nearest the hinge noved the least and the pair at the opposite end of the flume had largest displacement per revolution. This maintained the bottom of the flume as a plane while it was being raised and lowered. The jacks on each side of the flume were driven by a common 1 inch shaft line connected to a single 60:1 ratio gear reducer. The power to operate all the jacks was supplied by a 1-1/2 iorseptime, 1750 revolutions per minute, reversible, electric motor connected directly to the gear raducer. This provides a rate of vertical displacement at the downstream jack station of approximately 1 inch per minute.

The jacks were arranged in such a way that the downstream and of the flume may move from 12 inches below horizontal to 3 inches above herizontal, resulting in a maximum positive slope of 1/60 and a maximum adverse slope of 1/240. The motor was controlled by a raise, lower, and stop control switch. Safety switches were located both near the motor and near the control switch. It was necessary to unlock these before the control circuit could be completed. In addition, automatic limit switches were provided to prevent running the jacks beyond their limits. The general arrangement of jacks and gears is shown in figure 5-5.

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In order to connect the ends of the jacks (which move in a vertical line) to the flume (which moves in an arc), it was necessary to use a pinned linkage. A photograph of the linkage is shown in figure 5-2. (Appendix C) Due to the ark of the linkage, the relation between the rise of the flume and the revolutions, turned by the jack shafts, was not linear. Therefore, it was necessary to make a calibration of the slope rather than computing it.

At the time of erection, the jacks were leveled at .001 foot before the flume was erected. During the alignment procedure, the jacks were raised or lowered individually as needed to obtain a level base. The shaft couplings were then installed and no further individual movements were made.

# V-2 Water Supply and Measurement System

The water available in the laboratory is recirculated through the system by two pumps rated at 300 Gpm and 2000 Gpm. This 3 inch line contained a  $3 \times 2.25$  inch venturi accurate to 0.5% over the range from 30 Gpm to 300 Gpm. A 67 inch differential manometer reading to 0.01 inch was connected to the venturi and filled with tetrabromoethane (specific gravity 2.95) which gave a monometer deflection of 51 inches with a flow of 336 Gpm.

The 2000 Gpm pump was connected to a 6 inch line which contained a 6 x 4.176 inch venturi accurate to 0.5% over the range 200 Gpm to 2000 Gpm. A 30 inch differential manometer reading to 0.01 inch was connected to the venturi and filled with mercury (specific gravity 13.6). This manometer gave a deflection of 14.9 inches for a maximum flow of 1790 Gpm. In both cases the venturis were fitted with air vents to insure proper measurements.

In order that the calibration of the venturi meters should have no error larger than that of the venturi meter, the scale to be used for the calibration was checked against standard weights by the Indiana State Board of Health, Division of Weights and Measures. The scale error was less than 0.2% or 2 pounds per 1000 pounds. For the purpose of calibrating of the venturi meters, branch lines led to a baffled concrete channel located above the mighing tank.

At the point immediately before the 3 inch and inch lines entered the forebay, values and value bypasses were installed. The 6 inch line had a 2 inch bypass and value and the 3 inch line was fitted with a 1 inch bypass and value. The manometers were mounted in a position easily visible to the person adjusting the values. The overall apparatus arrangement in the laboratory is shown in figure 5-4.

The calibration curves of the venturi meters are given in figures 5-6 and 5-7.

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### V-3 Measuring Equirment

### a) Instrument Carriage

An aluminum instrument carriage, mounted on two stainless steel guide rails running the length of the flume, was installed so that the bottom of the flume could serve as a reference plane. The four wheels of the carriage were grooved to give a precision fit on the rails. One of the wheels had a set screw to keep the carriage from moving when an experiment was conducted. A surveyors tape was mounted along the whole length of the flume and an indicating point on the carriage served as a reference point for all longitudinal measurements. Another surveyors tape was installed im the transverse direction of the flume on a grooved rail of the instrument, carriage. This rail served as a guide for the slide on which a point gage and Prandtl tube were mounted. This slide also had an indicating point serving as reference point for transverse measurements. By means of these two tapes any point in the flume could be located in the horizontal plane. The instrument carriage was also coulpped with a fluorescent lamp, several electrical outlets and a wide 3/4 inch plywood strip on one side to serve as a desk for notekeeping etc. The carriage could easily be moved to any location along the flume and locked in a specific position with an accuracy of 0,005 ft. (Figure 5-8 shows a top view of the carriage.)

### o) Point Gage

All water surface elevation measurements were obtained by means of an electric indicating point gage. It was mounted on the instrument carriage slide which could be positioned with an accuracy of  $\pm$  0.002 ft. in the transverse direction of the flume. The staff of the point gage was marked in millimeters and was equipped with a vernier which allowed a reading accuracy of  $\pm$  0.05 mm in the vertical direction. The accuracy of the water surface elevation measurements varied between 0.1 mm to 2.0 mm depending on the smoothness of the free surface.

#### c) Prandtl Tube

The flow velocity measurements were obtained by means of a  $1/8^{m}$  0.D. Prandtl tube of standard design. The Prendtl tube was mounted on a vertical staff which was marked in 0.2 cm and had a vernier allowing a vertical positioning within  $\stackrel{1}{=}$  0.01 cm of accuracy. A set screw prevented the position to change during test measurements. The support of the Prandtl tube permitted, in addition to vertical and traverse motions, a rotation so that the horizontal portion of the probe could be aligned according to the direction of the flow indicated by a freely swinging stiffened tuft attached to the nube. The amount of rotation was measured as an angle by means of an indicator sliding long a fixed protractor. For measurement of large velocities, the Prandtl tube as connected to an inverted U manometer which had a fluid of specific gravity 0.810.

# ) Variable-Reluctance Pressure Transducers

The small magnitude of the pressure differentials to be measured with the Prandtl ube in zones of small velocities made it necessary to employ a Variable-Reluctance ifferential Pressure Transducer (model P7D, Pace Eng. Comp.) instead of the convenional liquid manometer. In the low velocity regions a pressure transducer having a ange of  $\pm$  0.1 psid. was used. In the higher velocity regions - as in the discharge at of the models - range at  $\pm$  0.5 psid. was needed. Figure 5-10 shows sketch of elocity pressure transducer system.

# Calibration of Velocity Probe

By using the relation between the dynamic and static head  $\frac{V^2}{2g} = h$  (where V is alocity in fps., h is feet of water and g is the acceleration of gravity) a static ulibration could be performed.

Two reservoirs were connected to the stagnation and static openings of the Prendtl the respectively. The reservoirs were connected through a valve. A datum level was assured by an electric point gage when the connecting valve was open, and the roltmeter medie was adjusted to the zero mark. The pressure was then equal on both sides of the aphragm in the pressure transducer. The valve connecting the reservoirs was closed and distilled water was added to increase the level in the reservoir connected to the stagnation sponing. The water level was recorded for even increments in the volueter reading. Figure 5-11 shows the calibration setup. Calibration curves are shown in Figure 5-12.

### V-4 Boundary Roughness Analysis in the Large Flume

#### a) Smooth Boundaries

W

The actual testing program in the large flume was run under two different boundary roughness patterns. The first roughness which will be referred to as the smooth boundary consisted of the steel flume walls finished with an epoxy resin paint.

It was necessary to define the region of uniform flow prior to testing the models so that their effects on the flow pattern could be evaluated. The models and the affected flow reach must be contained within a uniform flow region. Special tests were run to determine the characteristics of uniform flow. The values of Manning's n<sub>g</sub> the Darcy-Weisbach friction coefficient and the length required for the full boundary layer development were calculated as discussed below.

The section used for model testing was the portion of the flume between 20.00 ft. and 50.00 ft. from the entrance. With the known uniform depth within the test equation, the discharge, channel geometry and slope, the Manning's equation could be used to determine a value of Manning's r. Throughout the testing program, it was felt necessar; to maintain good control of uniform depths. Based upon an average of all the calculations of Manning's n for the smooth boundaries, a value of 0.0110 was determined. The range of n was from 0.009 to 0.0130 for discharges and Froude numbers from 1 ofs to 4 ofs and 0.05 to 1.09 respectively.

### Darcy-Weisbach Coefficient

In addition, the Darcy-Weisbach friction factor was calculated according to the following equation.

	$f = \delta g R_n S / V_n$	(5-4-1)
hers	f - Darcy-Weisbach fiztion factor	
	$V_n \sim Average velocity$	
	$R_n \sim$ Hydraulic radius	
	S = slope	

The friction factor was plotted against the Reynolds number and compared to the Blasius and Prandtl Von-Karman curves for smooth surfaces. The resulting plot is shown in figure 5-13. The Reynolds number is that defined by  $Chow^{(3)}$  as VR//, where  $\gamma$  is the kinematic viscosity of the water. Some of the scatter of the smooth data points is due to the fact that the surfaces were so smooth that relatively flat slopes were necessary. As a result, the slopes and surface profiles were very difficult to measure and they introduce experimental errors in the calculations.

# Turbulent Boundary Layer Growth

To ensure that the model tests were being run in a regime of fully developed turbulent flow, the distance downstream from the entrance of the flume t: the point at which the boundary layer thickness 5 was equal to the normal depth  $y_{\pm}$  was computed. The equation used for the development of the turbulent boundary layer for smooth surfaces was that developed for flow over a flat plate. This equation is based upon the Blasius 1/7th power law. See for example Daugherty and Ingersoll<sup>4</sup>)

$$S/x = 0.377/ Rex^{1/5}$$
 (5-4-2)

where

S = the thickness of the turbulent boundary layer X = distance downstream from the point where S = 0

Rex = Vn X/w

Vn · average uniform velocity

Solving (27) for X

$$X = (\delta / 0.377)^{5/4} (y_n/\gamma)^{1/4}$$
(5-4-3)

Although this formula is restricted to flow over a flat plate, it will serve as a first approximation in the problem at hand. If S is made equal to the uniform depth In, then the value of X will estimate the distance to the point at which the turbulent boundary layer becomes fully developed.

Test results showed that the values of X ranged from about 8 feet for small discharges and steep slopes to about 40 feet at high discharges and steep slopes. For the majority of the runs, the values of X were well within the firs; 20 feet of the flume, indicating that the flow could be considered fully developed within the test section.

b) Rough Boundaries

It has been shown that Manning's n value for the smooth boundaries was 0.0110. This value was not representative of any natural stream condition. It was decided to run a second series of tests with a boundary roughness which would simulate a more natural condition, and would permit the use of steeper slopes in the flume.

For natural streams, Daugherty and Ingersoll<sup>4</sup> give the following values of Manning's n.

	Filmismum n	Maximum
Smooth Matural Streams	0.025	0.033
Roughest Natural Streams	0.045	0.060

Based upon a scale ratio of 1/10 between model and prototype, a roughness with n = 0.020 in the model would correspond approximately to an n = 0.030 in a natural stream. Also a Manning's n of 0.030 in the model would be approximately 0.045 in the field. Therefore, it would be desirable to select a roughness pattern for model testing purposes that would have a Manning's n between 0.020 and 0.030.

Referring to the work done at Colorado State University  $\frac{5}{2}$ , the following roughness pattern was selected. The roughness along the bottom consisted of two layers of 1/4inch aluminum rodd; a bottom layer of longitudinal bars placed 12 inches on center of a top layer of transverse bars 6 inches on center. Along the side walls, there was one layer of vertical bars 6 inches on center placed 1/4 inch from the valls, there bottom layers of bars were tied together with screen wire. The vertical bars were tied at the bottom to the transverse bars and clamped to the walls above the free surface. This roughness pattern can be seen in figure 5-14.

Figure 5-15 is a diagram of two velocity profiles. One profile was taken 0.03 feet downstream of a roughness element and the other at a point widway between two adjacent bars. It may be concluded that a significant amount of flow was beneath the elements.

# Manning's n and Darcy-Weisbach f

As in the case of the smooth boundaries, several uniform flow tests were made. The tests were conducted for discharges of 1, 2 and 3 ofs and for 24 different slopes ranging from 0.000010 to 0.00450. Results show that the Frouds numbers langed from 0.05 to 0.56. An average value of Manning's n was computed as 0.0238. The actual value of n ranged from 0.022 to 0.025.

Darcy-Weisbach friction flotors were calculated for the rough boundaries and plotted in figure 5-13. The data for rough boundaries was broken down according to constant discharge and constant hydraulic radius. The raw data for the 24 normal dep. tests in the rough channel are listed in table 5-1 of the Appendix.

# Equivalent Sand Roughness

Mikuradse's equivalent sand roughness was determined from an equation given by Chow<sup>3</sup> for uniform flow in rough open channels. (page 204)

$$n = V g (6.25 + 5.75 \log (E_n/k))$$

where

 $V_n = average uniform velocity$  $V_f = \sqrt{g R_n S} = shear velocity$ 

R<sub>n</sub> = hydrailic radius

k = Nikuralse's equivalent sand raughnees

Results showed the average k to be 0.095 ft, with a range from 0.05 ft, to 0.12 ft.

# Turbulent Boundary Layer Growth

As for the smooth boundary, the growth of the turbulent boundary layer was investigated. For the rough flow, an approximate method proposed by Bauer and presented in Chow<sup>4</sup> (page 193) was used. Although this method was primarily developed for steep slopes, it was found applicable to channels of small slope provided the flow was uniform. The equation proposed by Bauer was

where

S = thickness of the turbulent boundary layerX = distance from the point where S = 0

k = Nikuradse's equivalent sand roughness

Having already evaluated k for the rough boundaries, the distance to the point where the boundary layer thickness was equal to the normal depth  $T_n$  could be evaluated. For the maximum depth, the value of X determined from equation 5-2-5 was 80 to However, for the majority of tests, the value of X was equal to or less than 25 feet, in which cases the flow was fully developed in the test section.

Sayre and Albertson<sup>7</sup> have presented a comprehensive report on the effect of rough ness elements in rigid open channels. They state that a roughness parametery/(chi) which depends "on the size, shape and spacing of the roughness elements", should completely describe the boundary roughness. The true value of  $\chi$  depends on whether or no 1) the boundary is hydrodynamically rough - negligible viscous effects, and 2) the channel is sufficiently wide such that any appreciable side wall effects are essential eliminated.

According to Robinson, Keloseus and Sayre, the friction coefficient for fully rough turbulent flow may be expressed as a function of the relative roughness and a parameter representing the roughness spacing. In terms of Chezy C, as given by Sayre and Albertson<sup>7</sup>

 $C / \sqrt{g} = A \log \left( \mathbf{x}_{n} / a \right) + C_{2}$ 

C2 = a roughness parameter representing the roughness spacing and independent of roughness size.
a = parameter describing the actual roughness height
A = slope of the plotted line

The constant  $C_2$  may be found by graphical extrapolation in a  $C/\sqrt{g}$  vs  $Y_n/a$  plot as the value of  $C\sqrt{g}$  at which  $y_n/a = 1$ . The value of  $Y_n$  extrapolated to  $C/\sqrt{g} = 0$  defines the roughness parameter  $\gamma$ . The properties of a and  $C_2$  in equation (5-a-6) are com-

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(5-4-5)

bined in  $\chi$  according to the relationship

$$C_2 = -A \log (\chi / a)$$

Sayre decononstrated that by describing the boundary roughness by a single parameter the data would group about a single curve described by the equation

$$C/\sqrt{g} = 6/0E \log(\frac{\gamma_n}{\chi})$$
 (5-1-8)

Extensive comparisons to the work of Sayre and Albertson were made. Since the true value of  $\chi$  depends on having fully rough flow, the testing flume vas set to its maximum slope of 0.0125 and a series of six tests at various discharges were run. The data for these tests are given in Table 5-2 of the Appendix.

Figure 5-16 shows a plot of the resistance function  $C/\sqrt{g}$  against the value of og  $y_n/a$  in equation (5-4-6). This plot was made by assuming the value of a to be /2 inch for the particular pattern of 1/4 inch aluminum rods. The value of the slope as 6.06 as in equation (5-4-8) and the extrapolated value of  $C_2$  was 3.15. With the alues of  $C_2$ , A, and a the value of  $\chi$  was determined to be 0.0126 feet. Figure 5-17 ives a plot of equation (5-4-6). The determined  $\chi$  value was used to plot this curve. Valocity Distribution

# Sayre also demonstrated that equation (5-4-8) can be used to represent velocity rofiles. The resistance function $C/\sqrt{g}$ is equivalent to $\nabla_n/\sqrt{C_0/\rho}$ where the shear elocity $\sqrt{T_0/\rho}$ is equal to $\sqrt{g} \nabla_n b$ for open channel flow. Along with the formal opth tests to determine the $\chi$ parameter, a conterline velocity profile was taken at slope of 0.0125 and a discharge of 3.714 effs. The profile is shown in dimensionless rm in figure 5-18. It is compared to the similar profile presented by Sayre. The wation obtained was

e noticable difference in the constants (1.e. 4.6 and 2.6) was probably due to the Genent type of roughness used. The roughness baffles used by Sayre were claced such a way buat theore was to flow beneath the slements. The numerical value of a in equation (5-4-6) was assumed to be 1/2 inch. In actuality an effective value of a less than the assumed value could probably have been used. If a smaller value of a less than 1/2 inch had been used, then the value of 2 would decrease and the numerical values of y/2 in figure 5-18 would increase causing the velocity profile to shift to the left. The constants would then be approximately equal.

# General Resistance Diagram

Figure 5-19 shows a portion of the general resistance diagram for uniform flow in rigid open channels as presented by Sayre. The equations shown are those derived in his original paper. Added on this graph are the six special tests for rough boundaries and the value corresponding to the uniform flow depths used in both the smooth and rough boundary model testing program. In a later section of this report, the concepts presented in the paper of Sayre and Albertson have been extended to apply to field conditions. Although their work pertained only to rigid open channels, if is shown that the X perameter could possibly be applied to natural rivers and streeds.

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VI PROCEDURE AND DATA PROUESSING FOR FREE SURFACE FLOW ITSIS

### VI-1 Selection of Variables

For each geometry, several models were built to cover the practical range of the geometric and flow variables. The principal geometric variable is M = b/E. Nodels were built to cover 3 to 5 different values of M. A listing of the range of all the geometric variables tested for the seven basic geometries is given in the table, page 93

The principal flow variable that was varied was the Froude number. Netto were generally made for a range of values of the Froude number between 0.1 and 0.5. Each value of the Froude number could be obtained with different values of the discharge. The discharges used varied between 1 and L of 5.

Once the Froudo number and the discharge were selected, the normal diptit the tail gate elevation and the fluxe slope could be calculated. A homograph propered for this purpose is given in Figs. 6.1 and 6-2. Entering this diagram with the released value of the Froude number and using the chosen discharge, one may read directly the slope and the corresponding revolution counter setting, the seil gate height and the normal depth.

For definition of the Geometries see Fig. 3-14.

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# TABLE - SELECTION OF VARIABLES FOR FREE

SURFACE FLOW TESTS

Geogram		Anterior contraction of the second	1		Contemporer and Contemporer Statistics	and the second statement	CEDERAL SCREWING ANALOS	And the second se	
able	Ia	Ib	II	III	IV	Va	٧b	VI	VII
F	• <b>04-</b> 1•25	•04-•804	<b>.05-0</b> .4	0.1 0.15 0.25 0.3 0.4 0.5	0,1 0,15 0,2 0,25 0,3 0,4 0,5-0,25	0.1 0.14 0.2+2.3 0.4=0.5	0.09-0.5	0。1∞),475	10.2-0.5
М	<b>.3</b> 9	0.3-0.9	0.3 0.5 0.7 0.9	0.3 0.5 0.7	0.3 0.5 0.7	0.3 0.5 0.7 0.9	0.2 0.5	0.3 0.5 0.7	0.35 0.5 0.75
L b	0	0 0.5 1.0	0.25	0.25	0	0	0 <sub>°</sub> 5		a uses waterseasons
ld b In2	0	C	0 0-7.5 7.5-15 15-25 25-30	0	0	Otophetadeer Liverano, evid teagou	С	Ŭ	0
ġı	90	90	90	30 4.3 60 90	90	90	93	90	90
ø2	0	С	0	0	0	0 15 30	25 30	Ũ	)
β	0	0	0	0	0	0	0		0.3
N	1	1	1	1	3.	1		2	1
n	.011 .0238	.0238	°0238	.0238	.0238	.0238	.0238	o0238	.0238
9	0	0	0	0	0.0 0.8 0.85 0.9 0.95 1.0	0	Ò	0	0

# VI-2 Obtention of Uniform Flow

Once the Froude number and the discharge have been selected and the normal depth slope and tail gate setting have been calculated, the discharge is adjusted to the proper value, the slope and the tail gate are set to the calculated values.

The depth of flow is then measured at several stations along the flume, usually at 10 foot intervals from station 10 to station 50. If the depth at these stations varies by more than 1 mm the tail gate position is adjusted until uniform flow is obtained.

# VI-3 Free Surface Measurements

Usually the water surface measurements were taken along the centerline of the arch opening. Additional measurements of the free surface were taken for two-span arch bridge models, for two-dimensional segment models, for three-dimensional skewed models. The upstream face of the model was located at station 30. The maximum backwater depth occurredusually between stations 20 and 30. Water surface measurements were usually taken from station 15 to the end of the regain curve downstream of the bridge. The interval of the measurements varied with the slope and the curvature of the free surface, but in the vicinity of the point of maximum backwater depth, measurements were made at intervals of 0.5 to 1 foot.

The water surface measurements were taken by means of an electric point gage, which could be read to the nearest 0.1 mm.

### VI-4 Data Processing

Y

In the theoretical development it has been shown (see equ. 3-9-29 and 3-9-35) that the ratio of the backwater depth  $Y_1$  to the normal depth  $Y_n$  could be expressed as a function of the ratio of the Froude number  $F_n$  (which is the governing flow parameter) to the channel opening ratio M<sup>3</sup> (which is the governing geometric parameter) of the types:

$$\frac{\mathbf{Y}_1}{\mathbf{Y}_n} = \mathbf{f}_1 \left(\frac{\mathbf{F}_n}{\mathbf{M}^*}\right)^{2/3} \tag{6=1,-1}$$

or

$$I = I_2 \left(\frac{F_n}{M^0}\right)^2 \qquad (\delta = l_0 = 2)$$

It was found from the experimental data analysis that it was more convenient to present the backwater superelevation ratio  $h_1 * / Y_n$  as a function of  $F_n / H^0$  in the forms

$$\frac{Y_1}{Y_n} - 1 = \frac{h_1 *}{Y_n} - f_3 \left(\frac{F_n}{M^2}\right)^{2/3}$$
(6-4-3)

$$\frac{\mathbf{r}_1}{\mathbf{r}_n} - \mathbf{l} = \frac{\mathbf{h}_1^{\mathbf{h}}}{\mathbf{z}_n} = \mathbf{f}_4 - \frac{(\mathbf{F}_n)^2}{\mathbf{M}_1^2}$$
(6-4-4)

The general form of equ. 6-4-3 was used in the presentation of the results in the several progress reports. This presentation of the results is given in the collowing chapter.

In addition, a new analysis of the data was made according to the form of equ. -4-4. This presentation is given for the first time in this report. The value of constriction head loss coefficients have been calculated for all seven geometries by even of the formula (see equ. 3-8-13).

$$K = \frac{h_1^*}{v_{n2}^2/2g} = c \left[ \left( \frac{A_{n2}}{A_4} \right)^2 - \left( \frac{A_{n2}}{A_1} \right)^2 \right]$$
(6-4-5)

The value of a used in the computations was 1.20, which was obtained from velocity distribution measurements. (See chapter VII).

It has been shown (see equ. 3-9-10) that the backwatter ratio  $\frac{h_1}{V_n}$  is proportional to the square of the ratio of Froude number  $F_n$  and the contraction ratio:

$$\frac{h_1^*}{Y_n} = D \left(\frac{F_n}{K_1^*}\right)^2$$

The value of D was calculated from equ. 3-9-17, namely

$$D = n C \left[ \left(\frac{F_n}{M}\right)^2 \right] n - 1$$
(6-4-7)

and plotted vs.  $\frac{h_1^*}{Y_n}/(\frac{T_n}{N^*})^{<}$  for the purpose of evaluating the consistency of the results.

A geographical multiple correlation technique was used for the presentation of the data of geometries II to VII, This technique is described in the following paragraphs. The remaining calculations were performed on the digital computer. Figure 5-3 shows the program flow chart for data analysis using Fortran II on the IEM 7090 computer.

# VI-5 Four Variables Graphical Multiple Correlation

### (I) Principles of correlation

- A. The coaxial method of graphical correlation is based on the premise that if any important factor is emitted from a relation, the scatter of the points in a plotting of the observed values of the dependent variable vs. the values computed from the relation will be at least partly explained by the emitted factor. In other words, if the points on such a plotting are labeled with the corresponding values of the new factor, a family of curves can be drawn to modify the values computed from the original relation. For example:
- B. For 2-D segment arch bridge model test, the following relation resulted from dimension analysis for the unsubmerged case

$$\frac{\mathbf{x}_1}{\mathbf{x}_n} = f(\mathbf{F}_{r,s}, \mathbf{M}, \boldsymbol{\beta})$$

# (II) Procedures for Correlation

- (1) Tabulate the required data:  $F_{ns}\beta$  , M, Yg/In for each run
- (2) Consider  $\frac{Y_1}{Y_n}$  as the dependent variable, plotted along the final axis. (See fig. 6-4)

The F values vary in all tests and should be used as the first correlating variable. The values of  $\beta$  and M are fixed for a given model; it will be convenient to consider them in the second and third steps (quadrants A and E).

(3) Assume 
$$\frac{Y_1}{Y_n} = f_1(F,\beta)$$
 and neglect the variable M.

### (III) Remark:

The reading sequence in Fig. 6-10 should follow the sequence of correlation. For example, given  $F_n$ ,  $\beta$ ,  $M_p$  to find  $Y_1/Y_n$ . The sequence of reading must follow the sequence  $F_n \rightarrow \beta \rightarrow M \rightarrow Y_1/Y_n$ . The sequence  $F_n \rightarrow M \rightarrow \beta \rightarrow Y_1/Y_n$ does not yield the correct answer.  $F_n$  is used as the absciesa and  $\frac{\chi_1}{\chi_n}$  as the ordinate, as shown in Fig. 6-9a

A point  $P_1(F_n, I_1/I_n)$  can be located in the quadrant A and the  $\beta$  value of this point is shown by an assigned symbol. For example, in Fig. 6-5 symbol 0 shows that the value of  $\beta$  is 0.5.

In Fig. 6-5, the points are scattered because the essential factor M has not been considered. However, an average line for each  $\beta$  can be found through the average positions of the points.

- (4) Add the factor M (see Fig. 6-6),  $\frac{Y_1}{Y_n} = I_2 (P_n, \beta, M)$ With given values of  $F_n$ ,  $\beta$  and  $Y_1/T_n$  a point  $P_2$  may be located and the value of M is labeled or shown by an assigned s, ibcl. An average line for M is plotted through the average position of the p-ints.

(5) Check the accuracy of the slove correlation. See Fig 6-3. With given values of  $F_n$ ,  $\beta$  and  $I_p$  a computed  $Y_1/T_n$  can be found from Fig. 6-6. Then the computed  $Y_1/T_n$  and the measured  $Y_1/T_n$  define a point in quadrant 0 of Fig. :=7. Draw a line through the origin disecting the quadrant 0. Then the degree of statter of the points around the line will show the degree of scatter of the above correlation.

(\*) The  $\beta$ -lines and the M-lines in quadrants A and B are adjusted to improve the correlation, see Fig. 6-S.

a. With the given values of  $F_{n_0}$  M,  $Y_1/Y_n$  and using the M-lines obtained in the previous step, the points are adjusted in quadrant A. The scatter is reduced, and a better fit of the points is obtained.

b. In the same way with given values of  $F_{n,\rho}\beta$ ,  $Y_1/Y_n$  and the newly obtained  $\beta$ -lines, new M-lines are obtained and their accuracy is improved.

c. Then check the accuracy of the correlation in quadrant C. The points will be closer to the bisecting line than before. This means that the accuracy has been improved.
d. Try the procedures (a to c) a number of times until the points in quadrant no longer change their position, then the accuracy of the correlation has reached ts maximum possible value.

c. The  $\beta$ -lines and M-lines may also be presented as shown in Fig. 5-9.

(7) Change Fig. 6-9 to a more convenient form, Fig. 6-10, by combining quadrants and B together.

#### VII ANALYSIS OF TEST RESULTS

## I-1. Ceonetry I

#### a) Introduction

The results discussed in this chapter pertain to it has a line a line and the two dimensional model tests in and the and rouch boux which the element is and the discharge coefficient are given in terms of the charge. Low does not the normal depth Provide Number. For model cests is rough bound in the discharge coefficient are given in terms of the charge. Low does not the points of maximum and minimum depths are given. I not only the minimum depth is presented. The elects of different orige lengths on the backwater ratio and on the set is in the set of the correction of the set is in a compared. Finally, surface topography and velocity provides et is a set is a compared. Finally, surface topography and velocity provides et is a set is a set of the set of the correction of the set of

b) Smooth Boundaries - Tests by H. J. over

The consistency of the data is wall illustrated by the lack of scatter of the

1 1

experimental points as plotted in figures 7-1-1 and 7-1-2. These test r whis may be compared with the small scale tests done as part of the Preliminary Tests (see Chapter IV) for the contraction ratio of 0.5 which is sommon to both test series. Inspection of figure 4-8 and figure 7-1-1 and 7-1-2 shows that to take the discrete almost identical. For example, at a Fronds number of 0.2, the value of the discrete coefficient  $C_{\rm d}$  from the small scale tests (figure 4-3) is 0.38 and the superlevation ratio  $Y_{\rm d}/Y_{\rm n}$  is 1.1. Compared to this, the large stall tests, figure 4-1-1 and 7-1-2) give  $C_{\rm d}$  at 0.275 and  $Y_{\rm d}/Y_{\rm n}$  as 1.119 for a kineticity of 0.00 which corresponds to the Fronde number of 0.2 At 1 Fronde number of 0.4 the results of the mall scale tests indicated of was equal to 0.53 and  $Y_{\rm d}/Y_{\rm n}$  was 1.4. Similarly if corresponding kineticity of 0.08, the large scale tests shows 31 to be 0.48 and  $A_{\rm d}/Y_{\rm h}$  to be 1.432. The reliability of the date may be better discussed in terms of the values of the friction coefficient and of the backwater satio.

The Darcy-Weitbach friction factor and the Reynolds number for the orfers flow stablished before each mest wave calculated in vable 7-1-2 as follows:

e experimental friction factors there camp and its the theoreminal values cannot d adapting the Blasius and Frankfil formulas for flow in smooth pipes to the destance r open channel.

The formulas for smooth pipe flow are:  
Blasius 
$$f = 0.3164 \left(\frac{VD}{V}\right)^{-\frac{1}{4}}$$
  $R_{e} < 10^{5}$   
Prancifle-Von Karmin  $\frac{1}{\sqrt{T}} = 2.0 \log \left(\frac{VD}{V} \sqrt{T}\right) = 0.6$ 

placing D by 4R and simplifying, the equations become:

Blesta: ' = D.223 478

Prandt' Fon Kauran 
$$\frac{1}{\sqrt{2}} = 2.1 \log \left(\frac{R}{p} \sqrt{2}\right) = 0.0$$

Powell and Posty<sup>2</sup>, we while while a triangle of fluxe for the formal started by

for tranquil flow in a stooch on and flow and flow in a stooch on and flow in a stooch on an internation for and flow in a stooch of the Darch to solve and flow on a set of the Darch to solve and flow on a set of the Darch to solve and flow on a set of the Darch to solve and flow on a set of the Darch to solve and flow on a set of the Darch to solve and flow on a set of the Darch to solve and flow on a set of the Darch to solve and flow on a set of the Darch to solve and the solve and th

in figure 7-1- to filting andi cent 1 at 1 for Rep discourses in the normal depth, as mine two filters of the Althouse the as input sompletely true for the depth to visit and the rest in the some missible compare the deta wight the experiment of the rest of the solution of the solution

In general, as shown in figure ( ) is the date fail as a first of the first of the second se

The backwater ratio  $h_1^{*}/\lambda$ , h for such the lest was compared to the relation of a matrix and by Tracy and Carter<sup>4</sup> for rectangular constrictions. This comparison is shown a figure 7-1-5 which shows that the back ther ratios are of the size order of the size of th

## c) Smooth Boundaries - Tests by P. F. Hiery

The experimental results of the two-dimensional semi-circular arch model tests in the large flume with smooth boundaries were plotted as the backwater ratio  $Y_1/T_n$  vs. the channel opening ratio M<sup>0</sup> with the normal depth Froude number F<sub>n</sub> as the parameter. This plot is shown in Fig. 7-1-6. As would be expected, the ratio  $Y_1/T_n$ would decrease to unity for all Froude numbers as the value of M<sup>0</sup> approaches 1.00. Also, it goes to infinity as M<sup>0</sup> goes to zero. In a similar manner, the discharge coefficient C<sub>d</sub> was plotted vs. M<sup>0</sup> for F<sub>n</sub> and is shown in figure 7-2-7. The value of C<sub>d</sub> tends to 0.611 at M<sup>0</sup> = 0 and approaches the same value as the parameter F<sub>n</sub> when M<sup>0</sup> = 1.

The curves of figure 7-1-6 and 7-1-7 have actually been interpolated for constant Froude numbers. Although the results covered the entire range of variables, the Froude number of normal depth was never exactly equal to 0.5, 0.6 or any other value of the parameter on the figure. The amount of error produced during the interpolating process was found in most cases to be less than one percent. The wave data and calculations are given in tables 7-1-2 and 7-1-5 respectively.

The data of P. F. Biery for smooth boundaries have been processed and analyzed as described in section VI-4. The backwater superelevation ratio  $h_1 * / T_n$  is plotted as a function of  $(F_n/M^2)^2$  in Fig. 7-1-8; the head loss coefficient K is given in terms of M<sup>2</sup> in Fig. 7-1-9, and the backwater ratio coefficient D is given in Fig. 2-1-10. VII-2 <u>Geometries Ia and Ib - Rough Boundaries - Tests by P. F. Biery</u> (See Fig. 3-14 for definitions of geometries Ia and Ib). a) Selection of Rough Boundary Model Tests

From the nomographic chart, (Fig. 6,2) the conditions for the rough boundary model tests were selected. The X's in the table below indicate the selected normal depth conditions for each of which the following values of M and L/b were tested.

For M = b/B = 0.3, 0.5, 0.7, 0.9

and L/b = 0.0, 0.5, 1.0

F1 Ra	low ate	0.06	150.00.00	Froude Number			HORE CO
		Vov2 - Pri- Vo	1.7 0.20 0.27 0.	20 0.32 0.40	0.45 0.50 0.60	0,70 0,80 0,90	128131734
1	ciu	X	X	X	X	X	
2	cîs	X	X	X	X	x	
3	ofs	X	X		X	X	

#### b) Test Data

Table 7-2-1 presents the raw measurements of all of the large flume, rough boundary, semicircular tests. Included are both the two and three dimensional results. The designation of the run numbers is as follows. The first number represents the normal depth set-up. The second number represents the particular model being tested. The pertinent data and calculations are summarized in Table 7-2-2.

The particular measurements that were taken on each of the smooth and rough tests in the large flume were those required to calculate the following quantities: the hydraulic radius, the Reynolds number, the Froude number, the Darcy-Weisbach friction factor, the channel oponing ratio  $M^{\circ}$ , the discharge coefficient, the backwater ratio  $(T_1/T_n)$ , the backwater superelevation  $(h_1^{*})$ , the surface profile ratio  $(h_1^{*}/\Lambda h)$ , the length to the maximum backwater  $(L_{1-2})$ , the length to the point of minimum depth  $(L_{2-3})$ , the length  $L_{1-3}$ , and the Manning's n. (See figure 3-1 for the definition of terms.) In view of the large amount of data that was to be analyzed and the repetitive character of the calculations, a program was prepared for processing the data on the Royal McBee LGP-30 digital computer.

In Table 7-2-3, the surface topography measurements for run number 4-4 are listed. It was felt that this run was indicative of average natural stream conditions. For the same run, the velocity measurements at the uniform depth, maximum depth, vena contracta and minimum depth are given in Table 7-2-4. In addition to the uniform depth isovel diagram of run number 4-4 above, velocity measurements of unform flow were also taken for normal depth runs number 2 and 10. This data is also listed in Table 7-2-4.

With the isovel diagrams for uniform flow, the kinetic energy correction factor of and the momentum correction factor  $\beta$  were determined. This was accomplished by obtaining the areas between two adjacent constant velocity lines with an area planimeter. Once the areas were determined, the equations for  $\alpha$  and  $\beta$  were solved in the following form.

$$\alpha = \int \frac{1}{V^{2} da} = i \frac{\sum_{i=1}^{N} v_{i}^{3} A_{i}}{V^{3} a}$$
(7-2.1)

and

where n = number of sub-areas

$$V = \sum_{i=1}^{n} v_i \Delta a_i \qquad (7-2-3)$$

#### o) Teet Secults

#### <) Backator ratio and discharge coefficient

The backwater ratio is plotted vs. the channel opening ratio with the normal depth Fronds number as a parameter for the two-dimensional semicircular arch models as observed in the large flume with rough boundaries. In view of the importance of these curves, the scale was expanded and the results are shown in two parts. Figure

(7-2 2)

7-2-la gives the results of  $Y_1/Y_n$  vs. M<sup>o</sup> for the range of backwater ratios less than 1 1.50. For the ratios greater than 1.50, figure 7-2-lb should be used.

As for the smooth boundaries, the experimental values of the discharge coefficient for the same test conditions are presented in figure 7-2-2. This figure and figures 7-2-la and b have been interpolated for constant Froude numbers. The hump that appears in the Froude number lines of 0.25 to 0.60 was a phenomena which appeared in all of the curves of  $C_d$ . This was true for the smooth, rough and all L/b plots.

When compared to the preliminary test results of Fig. 4-8, the values of the backweter ratio for a given Fn and M<sup>o</sup> are almost identical.

(P) Location of the points of maximum backwater, and minimum depth

In orver to describe the centerline profile, it is desirable to have an estimate of the distants from the upstream face of the constriction to the point of maximum backwater elevation. This distance is referred to as L1-2. Because of the flatness of the surface profile in the vicinity of the maximum point, it was extremely difficult to get an exact measurement of 11-2. The actual measurements teren could have been in error by as much as 20.5 feet. However, with the large amount of data which was available, it was possible to study I1-2 on an average basis. Average values of  $L_{1-2}$  were calculated for several combinations of  $b/B_p$  L/b, L/B<sub>p</sub> etc. In this nanner, it appeared that the effect of the variable bridge length and the change in 1º were of the same order of magnitude as the experimental error. The most consistent elationship was found by plotting the dimensionless ratio 17-2 vs. the Froude number n with M = b/B as the parameter. This relationship is shown in figure 7-2-3a. The alues of L1-2 obtained from the smooth boundary tests also compared favorably with **'Igure 7-2-3a.** In a similar manner it was found that the length  $L_{1=3}$  (distance from he maximum depth point to the minimum depth point) varied only with the constriction cometry. The values of  $L_{1,\infty}3/b$  are plotted vs. M = b/B with  $L_1/b$  as a parameter in igure 7-2-3b. These curves are good for both two and three dimensional semicircular rch bridges.

#### X) Determination of the minimum depth

Several other investigators have used the Froude number at section 3 ( $F_3 = v_3/\sqrt{Er}$ ) see figure 3-1) as an estimator of the maximum backwater. Others have used  $F_3$  as a controlling parameter in making indirect measurements of flood discharges. Due to the extremely irregular flow pattern at the minimum point, it would seem that the use of  $F_3$  may be misleading. In the present research, the normal depth Froude number  $F_n$ was found to be a very reliable estimator of  $V_1/V_n$ . In order to test, the variability of  $F_3$  with  $F_n$ , a correlation curve of  $F_3/F_n$  vs.  $F_n$  was prepared. This curve is shown in figure 7.2-4. Below a Froude number of 0.5, the correlation was good. However, above  $7_n = 0.5$ , the depth  $y_3$  was often below the critical depth and the correlation of  $F_3/F_n$  to  $F_n$  was very plor. The scatter seemed to increase with increasing values of L/b. Therefore, only the test results of the L/b = 0 tests are shown. If used with caution, these curves can be used to estimate the minimum depth  $y_3$ . It appears from this curve that  $F_n$  is a much more reliable estimator than  $F_3$ .

# >) Relain Curve Length

The regain curve was measured for a few runs. This was done in order to establish the fact that the water surface profile returns to normal depth before the end of the testing flume was reached. The tests selected were those which would have the longest regain curve. The results are summarized below.

Run No.	М	L/b	$\mathbf{F_n}$	L2~1;
2-4	0.491	0.0	0.0489	20 ft.
2-7	0.693	0.0	0.1005	9 ft.
4-4	0.491	0.0	0.1984	- 9.5 2%.

In all cases the regain curve was within the test section.

## E) Comparisons of roughness effects

Comparisons between the model tests in the smooth and rough channel were made by plotting the backwater ratio and the discharge coefficient against the normal depth Froude number  $F_n$  for constant channel opening ratios. Similar comparisons were made between the several three-dimensional model tests in the rough flume.

The values of the backwater ratio  $Y_1/Y_n$  and the discharge coefficient  $C_1$  in the smooth and rough channel boundaries are compared in figures 7-2-5and 7-2-5b. It appears that the values are essentially the same for both smooth and rough conditions at Froudo numbers  $F_n$  less than 0.5. Since the practical range of Froude numbers for natural channels are those less than 0.5, these curves show that for all practical purposes the effect of roughness can be ignored.

3) Comparison of bridge length effects

Similarly, all of the L/b results were compared at constant values of the channel opening ratio M<sup>\*</sup>. Figures 7-2-62 and b illustrate these comparisons. Again it appears from the plots that for the practical range of field conditions (L/b  $\leq$ 1.0 and F<sub>n</sub>  $\leq$ 0.5) the effect of bridge length is negligible. The effect of length did seem to increase with a decrease in the channel opening ratio. However, as the value of M<sup>\*</sup> gets small, the physical proportions are closer to those of a culvert rather than a bridge opening.

 $\eta$ ) Surface topography

In order to complete the analysis of the maximum backwater, additional studies were made of the velocity distributions and the surface profiles. These studies were made for the condition of a sharp-crested semicircular constriction with M = b/2= 0.0and  $F_n$  approximately 0.20.

A detail of the surface topography both upstream and downstream of the model was observed. The results of this study are shown in figure 7-2-7. The numbers shown indicate the depths in centimeters. Only a detail of one half of the surface is given since the pattern is essentially symmetrical. The graph shows lines of equal surface elevation. The centerline profile is also given in the figure. It is interesting to note that the actual maximum water surface elevation is not along the centerline, but on the upstream face of the abutment. Although this may be expected, since there is a stagnation point at the abutment, the actual magnitude of the difference in elevation between the centerline maximum elevation and the actual maximum is the important question. The actual maximum shoreline elevation was found to exceed the maximum centerline elevation by as much as 5% of the centerline depth. This fact was verified in the surface topographies taken at other conditions. Liu<sup>6</sup>, as well as Herbich<sup>9</sup> gave similar surface topographies of other geometric constrictions, and the differences in water surface elevations was again found to be about 5% of the centerline depth.

#### i) . Velocity profiles

Traverses were run with the Prandtl Tube at four sections with the  $M=0.5_{\rm p}$ L/b = 0.0 and  $F_n$  approximately 0.20 model tests. The first section was in the norm mal depth flow without the model. The second was at the section of maximum backwater, the third at the upstream face of the bridge and the fourth at the section of minimum lepth. At the section of normal depth and maximum depth, vertical velocity traverses were taken at the centerline and 1 ft. and 2 ft. left and right. At the vena contracta, they were taken at the centerline, 0.5 ft. and 1 ft. left and right. Finally at the section of minimum depth, traverses were taken at the centerline, 1 ft., 1.5 ft., and 2.2 ft. left and right. In general a more detailed traverse was taken at the centerine of each section. From these measurements, plots of equal velocity lines were repared for each section. This was done first by plotting the velocity profiles at ach location within each section on arithmetic graph paper. Typical are the pro-11es shown in figure 7-2-8 at the maximum depth condition. From these curves, points f equal velocity were found and plotted in the appropriate location on a cross-section iew of the channel. A composite picture of the isovel diagrams is shown in figure -2-9. Only half of the diagram is given due to symmetry. All of the diagrams were ntegrated with an area planimeter and the discharge so obtained was checked against ne venturi meter discharge. In all cases they checked within 1%.

In addition to the isovel diagrams for uniform flow in figure 7-2-9, two other "form flow velocity profiles were plotted. The test conditions were similar to 1053 of a natural stream. All of the diagrams were integrated and the values of the discharge, the kinetic energy correction factor and the momentum correction factor were computed. Listed below are the determined values of  $\mathcal{A}$  and  $\beta$  for each profile.

Q	Slope	Уn	Apprex. Fn	$\sim$	β
2 cfs	0.000131	0.799 ft.	0.10	1.145	1.055
1 cfs	0.000584	0.319 ft.	0,20	1.216	1.084
1 cfs	0.004080	0.173	0.50	1250	1.090

k) The generalized backwater equation and head loss coefficients

With the introduction of the channel opening ratio M<sup>\*</sup>, the assumption was made that if properly interpreted, the backwater produced by constrictions of the same M<sup>\*</sup> would be equal, regardless of its geometry (i.e., a semi-circle, or a circular segment, or a rectangle etc.) as long as the constriction effects are not modified by kew, chamfere, eccentricity, wingwalls, etc.

To test this hypothesis, the results of this section (VII-2) are compared with the circular segment data obtained in the preliminary investigation (Fig. 4-8) and ith the vertical board data given by Liu<sup>7</sup>. The latter tests were run in a fluxe hich was widerthan the large fluxe used in this investigation. The roughness witern and by Liu was also different. It produced a Manning's a of 0.024. It is filt streamely interesting to note that the test data taken by three investigation geometries roduced almost identical results. This clearly verifies that, as defined, the hannel opening ratio M<sup>1</sup> is the governing geometric parameter. Of course, the date compared were those where the eccentricity was zero, the skew was zero, and the entrance as sharp. It is still necessary to apply correction terms for these conditions.

As mentioned previously in the analysis, a similarity was noticed between the everal different backwater equations. The term  $(F_n/M^1)^{2/3}$  appeared in several plutions of  $y_1/y_n$ . In general it appeared that

$$y_1/y_n = C \left[ (F_n/M^{\circ})^{2/3} \right]^{N}$$
 (7-2-5)

where C is a coefficient which would take into account the effects of the discharge coefficient, approach velocity, and non-uniform velocity distributions and other empirically determined factors. Equation 7-2-5 is actually the equation of a straight line on logarithmic paper with a slope of f. A total of 50 semicircular L/b = 0 test values, 44 vertical board values (Colorado) and 50 segment values were plotted in the form of  $X_1/T_n = 1$  vs  $(F_n/M^{\pm})^{2/3}$  and are shown in figure 7-2-10. The value of  $Y_1/T_n = 1$  was used instead of the backwater ratio. It is quite apparent that the data tends to collapse into one straight line relationship.

The method of least squares was applied to a random sample of the 144 test points to determine the empirical straight line relationship. After solving for  $\gamma$  and  $C_g$ equation 7-2-5 became

$$\mathbf{x}_{1}/\mathbf{x}_{n} = 1 + 0.47 \left[ (F/M^{0})^{2/3} \right] 3.39$$
 (7-2-6)

Equation 7-2-6 is a very simple and easy solution for the backwater produced by any type of constriction. In actual practice, this equation will give as goed an estimate of the maximum backwater y as any previously suggested method.

It has been suggested by C. F. Iszard that equation 7-2-6 could be approximated

$$Y_1/T_n = 1 + 0.45 (F/M^0)^2$$
 (7-2-7)

nd still fit the data very closely.

The data were also reanalyzed in terms of the simpler equation

$$\frac{\mathbf{h}_1^*}{\mathbf{Y}_n} = \frac{\mathbf{Y}_1}{\mathbf{Y}_n} = \mathbf{1} = \mathbf{f} \left(\frac{\mathbf{F}}{\mathbf{M}^*}\right)^2$$

igures 7-2-11 to 7-2-13 give plots of this relation for values of L/b = 0; 0.5; rd 1.0. Fig. 7-2-14 summarizes the backwater ratio information for geometry I, for mooth and rough boundaries. The curves shown in this figure are the same as those reviously obtained in figures 7-1-3 and 7-2-11 to 7-2-13. In addition, a least quare fitting of all the data (smooth and rough) taken by P. F. Biery is given. The head loss coefficient X is given as a function of M<sup>3</sup> for L/b = 0; 0.5; L.O in figures 7-2-15 to 7-2-18. A summary of the head loss coefficient curves is given in fig. 7-2-18. The backwater ratio coefficient D is given in Fig. 7-2-19 to 7-2-21. corresponding to values of L/b = 0; 0.5 and 1.0.

## VII-3 Geometry II Tests by S. Lippai and T. P. Chang

## Dual Parallel Three Dimensional Semi-Circular Arch Bridge Constrictions: (See Fig. 3-14 for definition of Geometry II)

For simplicity, only the case of two identical arch bridges placed parallel to each other and normal to the direction of flow, was considered. The effect of Froude Number  $(F_n)$ , the channel opening ratio  $(N^{\circ})$ , and the distance between the centerlines of dual parallel bridges  $(I_d)$ , on the backwater ratio  $(I_1/I_n)$  has been determined.

The raw and calculated data for the 117 tests are given in table 7-3. Typical surface profiles are shown in Fig. 7-3-0. Figures 7-3-1 to 7-3-5 were obtained by plotting backwater ratio  $T_1/T_n$  against the dimensionless bridge spacing  $bL_d/A_{n2}$  with the channel opening ratio M' as parameter for  $F_n = 0.10$ , 0.15, 0.20, 0.25, 0.30 and 0.40.

It was found that for low Froude Numbers ( $F_n \leq 0.25$ ) the backwater due to closely blaced dual parallel bridges ( $bL_d/A_{n,2} \leq 5$ ) was less than what it would have been for a single bridge. The lower part of the graphical correlation of Fig. 7-3-6 shows blearly this effect.

It was also found that for higher Froude Numbers  $(F_n \ge 0.25)$  and small channel opening ratios (N' = 0.3), the placing of models at  $bL_0/A_{n2} = 8$  to 9 resulted in strong wave action between the models and an abrupt increase in backwater ratio was observed as can be seen in Figures 7-3-3 to 7-3-5. For the latter conditions, the st expansion below the upper bridge was unstable. It oscillated from the left to right and back with a period of about 2 sec. The backwater upstream of the upper widge was minimum when the jet was symmetrical along the centerline. The backwater increased as the jet was deflected sideways. The values plotted were the maximum ackwater elevations observed.

Figure 7-3-6 shows the graphical multiple correlation for 4 variables ( $F_{n_s} M^s_{,s}$   $1/I_{n_s} bL_d/A_{n_s}$ ). Knowing any three of them, the fourth one may be determined from ig. 7-3-6.

A strength line relationship between  $I_1/I_n$  and  $(F_n/M)^{2/3}$  on log-log paper was obtained for dual parallel bridges placed at  $bL_d/A_{n,2} = 0$ , 10, 20, and 30. Figure 7.5-7 was arrived at through the following steps:

1. Fig. 7-3-6 was used to obtain values of  $Y_1/Y_n$  for  $bL_d/A_{n2} = 0$ , 10, 20, 30 M<sup>0</sup> = 0.3, 0.5, 0.7 and 0.9,  $F_n = 0.10$ , 0.20, 0.30 and 0.40.

2. This data was processed and an arithmetic plot of  $Y_1/Y_n$  vs.  $(F_n/M^0)^{2/3}$  for  $M_n/A_n = 0$ , 10, 20 and 30 was constructed.

3. To obtain a straight line relationship between  $Y_1/Y_n$  vs.  $(F_n/M^s)^{2/3}$  for  $bL_6/A_{n2} = 0$ , 10, 20 and 30 on log-log paper, appropriate values of a ware determined by using the method described in reference 8.

4. The curve of figure 8 doesn't apply to the unstable flow condition with heavy wave action between the dual bridges.

In addition the data were reanalyzed making use of the computer program. The backwater superelevation ratio  $h_1^{*}/T_n$  is plotted as a function of  $(F_n/M^*)^2$  in figures 7-3-8 to 7-3-13, for the following values of the parameter  $J_0 h_{n2}^2$ :

Fig.	7-3-8	Lab/Ano	 0
Fig.	7-3-9	W . 116	0 20 7.5
Fig.	7-3-10		7.5 to 15
Fig.	7-3-11		15 to 25
Fig.	7-3-12		25 to 30

The equation of a least square fluting is given for each range of the presenter  $L_{db}/A_{n2}$ . A summary of these curves is given in Fig. 7-,-13.

The head loss coefficient K is given in the charmal opening ratio W for the following values of the parameter  $L_{db}/\Lambda_{n2}$ :

r'1g.	7-3-14	Lab/Ano = 0
Fig.	7-3-15	0 20 7.5
Fig.	7-3-16	7.5 to 19
Fig.	7-3-17	16 +0.25
Fig.	7-3-18	
5-		20 UU )L

Whenever a good least square curve fitting could be obtained its equation is given on the graph. In the other cases, the average values of the several groups of test for a given value of M<sup>1</sup> was obtained, and a line was fitted by eye through these points. Figure 7-3-19 is a summary of the head loss coefficient curves. Figure 7-3-20 gives the values of the backwater ratio coefficients D. The raw and calculated data for Geomety II are given in Table 7-3.

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## II-4 Geometry III Tests by T. P. Chang

# Three Dimensional Semi-Circular Arch Bridge Construction with Wingwalls (See Fig. 3-14 for definition of Geometry XII)

The influence of wingwalls on the backwater ratio  $(X_1/X_n)$  has been tosted for dingwall angles  $p_1 = 30^\circ$ , 45°, 60° and 90, using a model with L/b ratio of 0.25. The raw and calculated data are listed in table 7-4. Figures 7-4-1 to 7-k-4 mere obtained by plotting the backwater ratio,  $(X_1/X_n)$  against the channel opening ratio, (M<sup>1</sup>) with the Froude Sumber (F<sub>n</sub>) as parameter for wingwell angles  $p_1 = 30^\circ$ , 5°, 60° and 90°.

Fig. 7-4-5 is obtained by graphical multiple correlation. Knowing any three of our variables,  $X_1/X_n$ ,  $F_n$ ,  $M^*$ , and  $\phi_1$ , the fourth one may be determined by this chart.

Fig. 7-4-6 shows the relation between the backwater ratio  $(Y_1/Y_n)$  and  $(F_n/M^2)^{2/3}$ ith  $p_1$  as parameter. It was based on the graphical correlation of Fig. 7-4-5. Due the indirect plotting method, the accuracy of Figure 7-4-6 is less than that of lowers 7-4-1 to 7-4-4.

The data were reanalyzed making use of the computer program. The results are resented in the form of plots of the backwater superclevation ratio  $h_1^{*}/I_n$  vs. the pare of the ratio of the Frence number to the channel opening ratio  $(F_n/M)^2$  for ingwall angle of 30°, 45°, 60° and 90° in <sup>P</sup>igures 7-4-7 to 7-4-10.

For each value of the parameter  $\beta_{1,r}$  the equation of the least square curve fitting given on the figure. A summery of these results is given in Fig. 7-4-11.

The head loss coefficient K has been calculated for each of the wingwall angles. He results are presented in the form of plots of K vs. the channel opening ratio. Herever possible a least square fitting of the data was obtained, and the equation if the line is given. In the other cases, average values of the coefficient K for grouping of points with the same value of M<sup>4</sup> was obtained, and a straight line was tted by eye through these average points. These results are presented in Fig. 7-4-12 7-4-15 for wingwall angles of  $30^{\circ}$ ,  $45^{\circ}$ ,  $60^{\circ}$  and  $90^{\circ}$ . A summary of the curves is esented in Fig. 7-4-16. Finally, the values of the backwater ratio coefficient D are presented in ig. 7-4-17 to 7-4-20 for the four wingwell argles.

The raw and calculated data for Geometry III are given in Table 7-4.

## VII-5 Geometry IV Two-Dimensional Semi-Circular Arch Bridge Constrictions with Eccentricity - Tests by T. P. Chang

(The eccentricity is defined as one minus the ratio of the small embadiment to the larger embankment. (See Fig. 3-14 for definition of Geometry IV)

The influence of eccentricity on the backwater ratio along the centerline  $(Y_1/Y_n)$ has been determined for eccentricities e = 0; 0.80; 0.85; 0.90; 0.95; and 1.00. Figures 7-5-1 to 7-5-6 give the backwater ratio  $(Y_1/Y_n)$  plotted vs. the channel opening ratio  $(M^{\circ})$  with the Froude Number as a parameter for each of the six eccentricities tested. Figure 7-5-7 shows the relation between the backwater ratio  $(Y_1/Y_n)$ and  $(F_n/M^{\circ})^{2/3}$  with the eccentricity as a parameter. Figure 7-5-7 was obtained by graphical multiple correlation.

The results of the re-analysis of the same data making use of the computer program is given in the following figures. The backwater superelevation ratio  $(h_1*/T_n)$  is plotted vs.  $(F_n/M^{\circ})^2$  for the six eccentricities tested in figures 7-5-8 to 7-5-13. For each eccentricity the equation of the least square curve fitting the data is given. A summary of the results is given in  $F_{ig.}$  7-5-14.

The values of the head loss coefficient K are plotted against the channel opening ratio M<sup>9</sup> for each of the six eccentricities tested in figures 7-5-15 to 7-5-20. The straight lines were obtained by obtaining average values for K for groups of points aving the same value of M<sup>9</sup>. A straight line was fitted by eye through these average points. A summary of the head loss coefficient K is given in Fig. 7-5-21.

The values of the backwater ratio coefficient are given in Fig. 7-5-22 to 7-5-27. The raw and calculated data for Geometry IV are given in Table 7-5.

#### VII-6 Geometry Va

o Dimensional Semi-Circular Arch Bridge Constrictions with Sky Tests of T.P. Chan. (Sse Fig. 3-14 for definition of Geometry Va)

The influence of skew on the backwater ratio along the conterline,  $(Y_1/T_1)$  has endetermined for skew angles of  $0_2 \rightarrow 0^{\circ}$ ,  $1^{-\circ}$ ,  $30^{\circ}$  and  $45^{\circ}$ . The max and calculated te are given in Table 7-6.

Fig. 7-5-1 to 7-6-4 were obtained by plottime the backwater ratice  $(X_1/T_1)$  against e channel opening ratio,  $(N^{\circ})$  with Froule No  $(F_1)$  as parameter for sket and es = 0°, 15°, 30° and 45°.

Fig. 7-6-5 was obtained through graphical on the correlation. Knowing any the four variables,  $Y_1/Y_n$ ,  $F_n$ ,  $F_r$ , and  $f_r$  the fouril tariable may a determined from the prelation chart.

Fig. 7-6-6 shows a relation between  $\mathbb{Z}_{1-n} := (\mathbb{F}_n/\mathbb{N})^{2/3}$  the parameter. was based on the graphical correlation of Neuron 6-5. During the inducer plotter shod the accuracy of Fig. 7-5-6 is less than the coff Figures ? I to 7-6-4

The same data were reanilyzed miking use of the computer optam. The beak the paralevation ratio  $(h_1*/T_n)$  is plotted on  $(T_n/n)$  for the four thermales the difference 7-6-7 to 7-6-10. The equation of the 1. the quare 1.1 and 1.1 and

The raw and calculated data for Geometry Va all given in Table 1-6.

# res-Dimensional Sevi-Gircular Arch Bridges Constrictions with Skew Tests by T.P. Chang (See Fig. 3-14 for definition of Geometry Vb)

The influence of the skew on the backwater superelevation ratio along the centerline  $(h_1*/I_n)$  has been investigated for skew angles of  $\phi_2 = 15^\circ$  and 30°. Figures 7-7-1 and 7-7-2 give the backwater superelevation  $h_1*/I_n$  as a function of  $(F_n/M)^2$ for the two skew angles tested. A summary of these results is given in Fig. 7-7-3.

The raw and calculated data for Geometry Vb are given in Table 7-7.

# /II-S Geometry VI Two-Span Seri-Circular Arch Enidge Construction of the berry (1.)

The pier width, P, used in the nodels two equil to one third of the spin or  $\pi/2$ . A different pler width was used for submerget tests.) The onfly use of this wy a of constriction on the backwater super-levation ratio has been in using the for 2 values of the channel width ratio M = 0.33 C.5 and C.7.

The backwater superelevation ratio  $h_2 \approx h_1$  are plotted we  $(e_1/1)^2$  in figure of an **in the equation of the least square fill line is given in the figure.** The values of the loss confrictent K are plotted with the charmed opening ratio N<sup>4</sup> in figure of the line obtained by the least square selled as given in the figure he values of the backwater ratio occificient 0 are given in fig. 7-8-5.

The raw and calculated data for lowester 'If and given in Table 7-J.

## JII-9 Geometry VII-Two-Dimensional Gircular Segment Arch Bridge Constrictions Tests by T. P. Chang (See Fig. 3-14 for definition of Geometry WII)

The influence of the ancunt of depression of the center of the arch with respect to the bottom of the stream, (as given by  $\beta = d/r$ , where d bh. distance between the channel bottom and the center of the circular segment of radius r) on the beauwater superelevation ratio has been investigated for 3 values of the depression parameter  $\beta = 0$ ; 0.3 and 0.5.

The values of the backwater superelevation ratio  $h_1^*/Y_n$  we been plotted vs.  $(F_n/M^3)^2$  in figures 7-9-1 to 7-9-3 for the three values of the depression parameter in each figure the equation of the lengt square fit has been given. A summary of the results is given in fig. 7-9-4.

Values of the head loss coefficient K are plotted vs. We thankel opening ratio ", for the three values of the depression parameter tested, and fight 7-9-5 to -9-7. In each figure the equation of the least square fit = raight line regiven. summary of the head loss coefficient values is given in fit. 7-9-8.

The values of the backwater rando coefficient as given in figures 7-0-9 to

#### TI-10 Bibliography of Chapter 7

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#### VIII-1 Test Selection

The selected models were to be tested in the 64 foot long flume, thest with smooth and then with rough boundaries. The large number of variables involved made it necessary to establish a systematic procedure for selecting the quantities allowed to very and those held constant in the several tests so that the possible range of the variables could be covered uniformly and completely. This was of connern because the range of discharges and slopes for which submargence could be attained was very limited. The procedure followed was based on the theoretical analysis and on the testing of simple geometries.

The channel width ratios, H' = b/B for which submergence was possible with the flow available in the laboratory was determined from prelimenty tests. The testing of geometry I-a with the selected channel width ratios gave the flow rates necess for submergence for each rodal. The dimensionless discharge  $\frac{2}{\pi^2 - b^5/2}$ , which

according to equ. 3-7-33 is related to the cackwater ratio Ng/b, was calculated for several discharges and channel width ratios to be tested as indicated in the following table.

ନ			<u>д</u> за ъ572	<ul> <li>And Managare (see )</li></ul>	and the second state of a state part ( ) and so
cîs	M = 0.15	M ≈ 0,20	N = 0.25	$M = C_0 275$	M 52 G.30
2.0	0.7095	0.353	0.202	0.159	0.328
2.5	0.885	0.441	0.2525	0.199	0.160
3.0	1.062	0.529	0.303	0.238	0.19:
3.5			0.354	0.278	0-121,
4.0	-	-22	0.405	0.318	0.256

By choosing flow rates in the test runs that give different values of  $Q/g^2 b^{5/2}$  grouping of data in the graphs could be prevented. The seven tables that follow this section show the test run. all for the different geometric. The condition relevant

1 Table I through VII were chosen in part from the table developed in this section.

1

ī

# Table I - Smooth Boundary Tests

#### n = 0.0110

## Geometry I-a

Flow	Fre	ucle Numb	er	Channel Width Ratio (b/B)						
cfs.	0.10	0.15	0.20	0 <b>.20</b>	0.27	0.275	0.30			
2 <sub>0</sub> 0	X			X	X	X	in Sur A			
2.0		X		s.		*				
2.0			X	X	*	36				
2.5	X			0	X	x				
2.5		X		X	x	X	ж			
2.5			X	X	x	Х	n an			
3.0	X			0	x	Х	X.			
3.0		х		0	х	X	A			
3.0			Х	0	x	х	X			
3.5										
3.5		X		0	X	x	X			
3.5			X	0	X	X	6			

X Regular test measurements obtained

0 The flow overtopped the model and channel walls

\* Submergence was not possible for these flow conditions

## Table II - Smooth Boundary Test

#### n = 0.011.0

#### Geometry 1-b

Flow	Froud	e Numb F <sub>n</sub>	OT?		Ohann O	el Vi . 20	dth R	atio 0.	b/3 25			ł	0.30		
Rate cls.	.10	.15	.20	,25	Thi.	a.cnes 。75	s Fac 1.0	tor L	/ъ 。50	.75	10	.25	. 50	75	1-7
2.0	х			X	X	X	X.	K	X	enderlaher i X	X	n de la composición d La composición de la composición de	K	a dalakan K	ete () fel d antender Start av 11 <sup>6</sup>
2.0		Ma		14	1. 1. 1. 1.	-15 <u>6</u>	÷.	46	44		÷.	44	ž	24	L.
2.0			Xs	*	25		<i>21</i> -	36	-94-	łc	22	22	÷	36	1 8 <i>8</i> 1 - 1 - 1
2.5	х			0	0	0		X	X	X	X				
2.5		X		X	X	X		Х	X	X	X	X		29 25	10
2.5			х					r			X	X	se A	X	1
3.0	<u>X</u>			0	0	-3	6								
3.0		X		0	0	Q	C	N A	X	X	à	Z	7 31	τ.	X
3.0			X	о	0	.)	0				X				
3.5	Xź			0	0	0		C	0	0	С				
3.5		Х		С	0	J	Q	X.	X	X	X	X	X	7P 63	X
3.5			X	0	0	0	0			99 22	A				
i+0	X3			0	0	0	G	C	0	С	0				
4.0		X		0	0	0	đ	C	0	0	6				-
400	and parts over 30	Widdle All Statistics	Xj	C	0	0	¢								

X Regular test measurements obtained

O The flow overtopped the model and channel walls

\* Submergence was not possible for these flow conditions

Xi Experimental condition investigated

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## Table III - Rough Soundary Tests

#### n = 0.0238

#### Geometry I

Flow	Pi	ude liu	awar	Channel dth lat.o - E						
	0.20	.015	0.20	0,20	0 25	9.0-5	0.30			
2.0	x			X	X	7 <u>.</u>	X			
2.0		Z-j			14		ΥĘ.			
2.0			X	X	*		75			
2.5										
2.5		X		Я	X	X	X.			
2,5			X	X	.0	X				
3.5										
3.5		X		0	- X0					
3.5 Catalogue and the Alastan states and the second	and Artific Children	Startin markets	Х. ж. м. м. м.	0						

X Rogular test in acurements obtained

O The flow overtopped the model and charnel wall :

\* Submergence was not possible for these flow an littions

Xi Experimental condition investigated

# Table IV - Rough Boundary Tests

#### n = 0.0238

Geometry I-b

Flow	enterore strength	la Numb Fn 015	.20		Chanal Fidth Estis (b/B) 0.25 0.30 Thickness Factor (1/b)										
CIS.	30-stanzationan-Rooten	aller Stratfort to Bridgebauers		.25	. 50	.75	1.0	.25	.50	. 75	I.J	25	5.3	.75	۳.,
2.0	X			X			X	X		4.0	X	dentar a sudana	**. *	- 2000 a second 	170 F. S 1605.
2.0		II							55		22		4.5	14 14 14	- 4-
2.0			Х	Х	X		Х		*		25	85	4-	D. <sup>6</sup>	
2.5	Xź			0	0	0	0								
2.5		X		X		X	X	Х			Х	X	λ	Х	
2.5			X	X		X	х				X			x	
3.5	X.			С	0	0	0	0	0	0	0				
3.5		X		С	0	0	0	X	X		Х				
3.5	THE ROOM PROPERTY	40.01-02-021-021-021-021-021-021-021-021-02	X,	C	0	0	0								

X Regular test measurements ch'ained

0 The flow overtopped the model and channel wells

\* Submergence was not possible for these flow conditions

Xj Experimental condicion investigated

# Table V - Rough Boundary Tests

#### n = 0.0238

#### Geometry V-b

Plow Rate cfs.	Fro.	Thickness Factor L/b = 0.50 Charnel Width Ratio (b/B) 0.20 0.30 /ngles of Skaw 0.							
	ning of Colors and an an an an an an an an	an a	Condentification and support	150	2220	300	Í50	22월0	30 <sup>0</sup>
2.0	X			X	X	x		X	X
2.0		X		X	X	X	踅	÷	*
2.,0			x	x	12	X	36	Se.	ĄĘ
3 <sub>0</sub> 0	X			x		x	x		Х
3.0		x		x		x	х		X
3.0			X	X		X	X		x

X Regular test measurements obtained

O The flow overtopped the model and channel walls

\* Submergence was not possible for these flow conditions

# Table VI - Rough Loundary Tests

## n = 0.0238

#### Geometry VI

Flow Rate	Froud	le Numb Fr	9 <b>7</b> °	Thickne. Factor L/b = 0.59					
cfs.	c 10	.15	.20	015	.20	25			
2.0	X			X	5	S .			
2,0		X		X	À	44 44			
2.0			x	X	x				
2.5	X			X	X	- ie			
2,5		X		Х	χ	х			
2.5			x	Х	2				
3.0	X			X		X			
3.0		х		Х	A				
3.0			X	X	X	X			

X Regular test measurments oftained

O The flow overtopped and model and change is

\* Submergence vas not possible for these flow one.tion

# Table VII - Rough Boundary Tests

## n = 0.0238

#### Geometry VII

Flow	Fro	ude Fr	Numl	er		Channel Wigth Ratio b/B					
cfs.	.1	。2	.3	.4	5	∞ () ∞	<b>*</b> .3	= <u>.</u> 5	en ()	<b>₽</b> .3	= o5
1.0	x					*		X	41-	*	
2,0	X					*	X		đē		X
2.0		X				*	X		*	4	x
2.0			X			#	Х		t.	*	Х
2 <b>.0</b>				x		₩.			46	-t-	X
2.0					X	<b>#</b>			÷.	16	X
2.5	X					X			46		
2.5				X		*	X			¥	
2.75					Х	갖	X		*	. 6	
3.0	X									X	
3.0		X				X				24	X
3.85	X							0	X		
3.85		X						0	×	X	
3.85			X			X		0	4÷	9.	X
3 <b>.85</b>				X		X		0	*	*	X
3.85			11.1.1.1.T. 1.1.1		X	X		0	*	Ť	X

X Regular test measurements obtained

O The flow overtopped the model and channel walls

\* Submergence was not possible for these flow conditions

## IL-2 Definitions: Fully Summerged and Partly Submerged Flow

The terms free outflow, fully submerged and partly submerged outflow are freuently used in this chapter and are defined as follows. Free outflow exists when he jet discharging from the downstream face of the bridge opening is surrounded, xcept for the channel bottom, only by air and a hydraulic jump occurs further downtream. Fully submerged outflow exists when the jet discharges into a stream that as a depth equal to or greater than the downstream height of the bridge opening, partly submerged outflow is the case when the jet discharges into a stream that as a depth less than the height of the bridge opening and no downstream (imp occurs,

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(See fig. 3-14 for definition of Geometry Ia)

 a) Coefficients of Contraction, of Discharge and of Velocity for Free Jet Outflow In order to investigate the coefficients of velocity, dontraction and discharge for submerged bridge inlets, 67 test runs were made (see Table I - VII). The coeffisients are defined as follows: Coefficient of contraction. C<sub>c</sub>, is the ratio of jet area at vena contracta A to the area of the orifice / o.

Coefficient of discharge, Co, is the ratio of the actual discharge Qa, obtained from the Venturimeter reading, to the theoretical Qb, as calculated by formula

$$C_d = \frac{Q_d}{Q_0}$$

coefficient of velocity is the ratio of actual velocity  $V_{\alpha}$  to the theoretical velocity t

$$G_V = \frac{V_2}{V_2}$$

t is custe any to combine the three coefficient as follows: From continuity quation the following and be writt n:

$$\frac{Q_{\rm fl}}{Q_{\rm fl}} = \frac{A_{\rm T}}{A_{\rm O}} = \frac{V_{\rm fl}}{T_{\rm fl}}$$

rom which it follows t at:

To find the coefficient of contraction the v na contracta had to be located and he cross-sectional are measured. This was done by means of the electronic point age.

A sufficient number of readings wer, taken at the vena contracta to plot the cross ectional area on graph paper. By use of a planimeter the area  $A_v$  was obtained and he coefficient of contraction could be calculated, as the area of the model opening >> known.
The coefficient of con rection are proved varue the value  $b/I_1$  of the modul opening width to the backwater depth using the channel width ratio  $b/B_p$  as parameter (see Figure 8-3-1).

To obtain values for the calculation of the coefficient of discharge  $C_{d,s}$  a theoretical discharge equation was developed. The development of this equation (3-7-26) was presented in the theoretical analysis in Chapter III. The actual rate of flow through the constriction was obtained from the manometer readings and the vonturi meter callor theoretical curves were plotted for the coefficient of discharge  $C_d$  versus the ratio  $b/T_1$  of the charged width ratio was  $c_i$  in used as parameter (see Figure 8-1-1). The coefficient of velocity was then for d from the relationship between the trace coefficients  $C_d = C_c C_c$  and plotted your us  $b/T_1$  in Figure 8-3-1.

The d silves see a to conterpoint the value 0.61 as the  $\frac{b}{l_1}$  ratio tends to zero. The curves ar closely proof straight lines and the C<sub>d</sub> values decrease with increasing channel width ratios. The C<sub>c</sub> curves also read towards the 0.61 value at the  $\frac{b}{l_1}$  at is decreases. Values of C<sub>c</sub> decrease for increasing channel width ratios. For interacting c eluce, within the test range, the C<sub>c</sub> curves tend to increase out i the value 0.77, an their stabilities at this value. The C<sub>v</sub> curves have the sine of the value 1.0 as a limitin value when the  $\frac{b}{l_1}$  ratio decreases toward zero.

The st data are scord d in test runs 1 - 67 shown in Table 8-1. The calculated values to find C<sub>c</sub>, C<sub>d</sub> and C<sub>v</sub> are sound in Table 8-4.

b) V.loc ty Distribu i n at Vena Contrac a for Free Jet Outflow

Extensive velocity measurements were taken in the jet downstream of constriction at the vena contracta in order to observe the velocity distribution. The Franklitube together with the pressure transducer were used. The readings were taken in a grid system to obtain full coverage over the cross section. The borderline of the area at the cross-section was also measured and plotted on graph paper together with the velocity grid values. From this the fleovelocity curves were drawn. Figure 8-3-2 shows one of the typical isovelocity curves drawn from such grid maxaments. Figure 8-3-3 shows the result obtained by following the individual isovelocity curves in the water cross-section itself. This was done by moving the Prandtl tube through the cross-section and mapping the path for the individual isovelocity curves obtained by means of the horizontal and vertical traversing devices. The data of figures 8-3-2 and 8-3-3 correspond to discharges 3.75 ofs and 2.62 ofs; normal depth Froude number 0.142 and 0.105 and opening ratios 0.30 and 0.30 respectively.

In Figure 8-3-4, the values of the velocities are plotted at different depths and a mean velocity curve is drive. The data for Figure 8-3-4, are the large as troufor Figure 8-3-3. Figure 8-3-4 shows the velocity variation over the opening width at different depths. A similar study was made for other flor conditions is recorded in Table 8-3. All the recorded values when plottid, show the case characterizate velocity distribution as shown in Figure 8-3-2 to 8-3-4.

c) The Generalized Backwater Equation - Outflow fully and Write Subranged

From the theoretical analysis ion in Chapter III if it expected that the general relation between the backwater and normal depth is of the form

$$\frac{\mathbf{x}_{1}}{\mathbf{x}_{n}} = \mathbf{C} \left[ \left( \frac{\mathbf{F}_{1}}{\mathbf{H}} \right)^{2} \right]^{-1}$$
(8-S-1)

where C is a coefficient that would take into account the eff ots of the expression velocity, discharge coefficient, non-hydrosistice pressure distribution, con-which in velocity distribution and other empirically determined factors. In a log-rithmic plot of  $Y_1/T_n$  verses  $(F_n/M^1)^2$ , the slope of the line is  $\gamma'$ . A total of 30 tests were made on geometry I-a and processed on the IEM 1620 computer to obtain plotting values of  $(Y_1/Y_n - 1)$  versus  $(F_n/M^1)^2$ . The graph is shown on figure 8-3-5. From this graph the values  $\gamma'$  and C were found to be 0.90 and 1.18 respectively, and equation 3-3-1 can be written for the submerged conditions of geometry I-a in the following way:

$$\frac{Y_1}{Y_n} = 1 + 1.18 \left[ \frac{y_1}{(1-1)^2} \right]^{0.90}$$
(8-3-2)

$$\frac{I_{1}}{Y_{n}} = 1 + 1.18 \left(\frac{F_{n}}{N^{2}}\right)^{1.80}$$
(8-3-3)

The test data taken to obtain this generalized backwater equation 8-3-3 are recorded in test runs 102-134 which are located in Table S-2. The calculated values from the data are presented in Table 8-3. A least square fitting of the backwater ratio point; for fully some sed flow is given in Fig. 8-10-1. The values of the head loss coefficient K are given in Fig. 8-9-1.

A general backwat: equation was not obtainable for the fires jet outflow This is becaus the Froude numbers for test runs -6 are all in the ringe of 0.550 = 0.587which reall in the first out of numbers, and a line could not be defined.

d) Discharge Coefficient for re- Suberged of Farty Subrerge Outflow

A we phote made if the discharge node, clout,  $G_{d1}$  we much sharped opening ratio, M° for the submarged noch bridge opening, Geor if y I-a (Figure 8-3-6/ The curve obtained as for the submarged di energy jet no pt for the two upper most curves which are for partly submarged jets. If all to wes are similar to those estimated by P. F. Elevy (a Art VII-1-c) for the unsubmarged case and for channel opening r the of 0.2 and up. The present a loss for the submarged case were obtained of the right of 10 between C.10 and 0.35. Figure 8-3-6 shows the  $C_{d1}$ versus M° curves for submarged and partly elever resc files as well as Elevy's discharge coefficient curves for submarged and partly elevers. An Figure 8-3-6 are performed in test run 102 to 134 which are located in Table 8-2. The calculated values are presented in Table 8-4.

The values of the discharge coefficient ... summarized in Figure 8-3-7. For the submerged jet, two sets of curves are given one uses the Froude number as a parameter, the other uses the ratio  $Y_n / \frac{b}{2}$  of the normal depth to the arch radius. The free jet discharge coefficient curves previously presented in Figure 8-3-1 have been reproduced here for completeness. The calculations are shown in Table 8-4.

# VIII-4 Geometry I-b - oo h Founce ce

As can be seen from the definition of term, sometries, (fig. -1/) secnetry I-e is a special case of geometry I-b. By setting the variable  $\frac{L}{b}$  equal to zero in the definition of geometry I-b, the secnet -1 is obtained. Geometry I-a therefore represents the limiting case of geometry I-b, and can be threated together with I-b.

in the theoretical analysis mide in Chapter All the dimensionle a relation

$$\frac{1}{b} = \frac{1}{b} \frac{1}{2} = \frac{1}{b} \frac{1}{b}$$
(8-4-1)

was developed for the mifile up for through an tribe and shell with the south chanal. total of 92 test runs are conducted in sub triber in the south chanal. Three sets of dimensionless cloves are ploted in the form of the it easier to compare the curves, ther was all plotted in the form of the  $\frac{1}{2}$  which is used as parameter, and the for the number is conducted in the form of the ball to show all three sets of enrors clearly of one of the form of curves are form in Figure 8-4-1. For higher Frede numbers for the curves is clover pared. This means that the L/b ratio becomes insignificant for off protein proves with the Fourie number reaches a value of abolt 0.20 for the mooth (a = 0.0110) boundary case of geometry 1-b. The data from the 92 test runs to list in run to ber 101-192 and recorded in Table 8-2. The calculated value of the plotted form's are shown in Table 6-4.

The backwater sup relevation with  $h_1^*/Y_r$  is plotted vo  $(F_n/M^2)^2$  for values of the length p rameter L/ = 0.25; 0 0; 0 75 . . .0 in f gures 8-10-2 to 8-10-5. The equations of the lines obtained by lengt square are given in the figure. Figure 8-10-6 summarizes the lackwater sup relevation d callor prometries Ia, and Ib with smooth boundaries. The values of the read loss coefficient K and the lines obtained by least square fitting are given in figures 3.9-2 to 8-9-5 for the same values of the length parameter L/b. A summary of the head loss coefficients for geometries Ia and Ib with smooth boundaries is given in Fig. 8-9-6.

# VIII-5 Geometries I-a and I-b - Rough Boundaries

# a) Test Data

A total of 38 test runs were made in order to plot the dimensionless curves shown in Figure 8-5-1. These three sets of curves were also based on the dimensionles equation 3-7-33 developed in Chapter IIX. The data are recorded in test runs 193-231 and can be found in table 2-2. The calculated values from the data are listed in Table 8-4

The values of the backwater superelevation ratio  $h_1^{*}/\Gamma_0$  were plotted vs.  $[F_0/\hbar^2]^2$ for the five values of the length parameter tested (1/L = 0; 0.25; 0.50; 0.75; 1.70)in figure 5.10-7. The lines were fitted by eye, and their equations are given on the figure, he values of the discharge coefficient were plotted vs. the channel orning rat; in fig. 3-9-7 for the same values of the length parameter. The equations of the lines fitted by eye are given in the figure.

The investigation, shows that the width of the constriction in the direction of flow, L<sub>p</sub> influences the backwater to a certain degree when the Fronde number he low. The wider the bridge in the direction of flow, the lower is the backwater depth obtained. For a Fronde number of 0.15 the backwater ratio for an L/b ratio of 1.0 was as much as 30% lower than for a L/b ratio of 0.000. For the more practical car as where the  $Y_{\rm L}/b$  ratio is in the order of about 0.50 to 1.00, a lowering of about 10-20% in the backwater ratio is usual then the L/b ratio is increased from 0.00 to 1.00. Flow through three dimensional arch bridges is very similar in behavior to the flow through short alverts. For short culverts, the flow characteristics are relatively independent of the slope and the factors that involve the length are comparatively unimportant. The control section for equare edge openings, such as investigated here is at the inlet.

Separation of flow around the opening, at the inlet section, was frequently observed. The flow contracted towards wena contracts section in the first part of the barrel expanded in a spiral motion as shown in Figure 8-5-2 and ended in a slug flow at the submerged outlet. Figure 8-5-3 shows the case where slug flow occurs at a exit of the barrel. Figure 8-5-4 presents the free jet case for geometry I-a the rough boundaries and the same flow.

Vortex action was also often observed. This air-vortex occurs from the surface ight above the crest of the inlet and introduces air through the opening. This acreases the water area of the opening and hence the flow through the constriction, and as a result the backwater increases.

An important effect upon the separation phenomena, that controls the change of low from a sluice type to full conduit flow, is the existence of a negative pressure in the barrel. This is caused by the relative magnitudes of air inflow from the inflet and from the vortex action over the inlet. The outflow if air due to entrainent at the air-water surface also has an effect upon the separation phenomena, the ore air entrained the less separation will occur and the barrel will tend towards all conduit flow.

) Comparison Batween Smooth and Rough Foundary Tests

A comparison between the limiting curves of Figures 8-4-1 and 3-5-1 for the mooth and rough boundary conditions respectively are made in Figure 6-5-5. This urves show that a rough boundary condition is of advantage for higher Frond in bers  $F_n > 0.15$ ) as could be expected from observing the flow behavior. For  $F_{ab} = 0$  offic unbers the approach velocities will increase and consequently give a light to refere evel due to the natural rouganess. This will improve the ability of the brind to roduce full conduit flow, because it increases the air entrainment in the barrol t the air-water surface and hence reduces the separation at the inlet.

# 

The discharge jet shouts to the side of the flume in a line tion of the entry perpendicular to the face of the bridge

The test results are per and both first ichtrap the blance of Figure 8-5-2, based on on theorem and have 1-7-35 on these in Company and Three sets of curves are platted blue the Taria notes of the up of the set 2, is parameter. The with any find here constant is the will be any The performance curves that the sale said he ught the second se and \$2 = 30° is selative only or equivation of parent to the first out of the the Proven numer. To be sound a state to Lor of the state of the constrictions for lever subtranting and D company of the second states and the second st of Geometry I-2. The application of the last of pro-termine to restrict the constriction grade as the second in the second and the dition closer to condition in the second state of the second state and both Geometrics. Inc. addition of the but he sate from the sate of the flow the difference in the puller of the output of the output of the second s having Y1/b ratio 1 1 2 m monore de l'arthur and 2000 bener 2 but are merely an app of the second second of the second second of the second second second second second second 24 test rurs the milde of the second of the test of the second of the se 8-4. The values of the same is read then rate have the outland the in Figure 2-10-3. The concourted of the Mission for the state of the values of the discharge coefficient K as plotted ve, the characteristic ratio in Fig. 8-9-8. The equation of a viscal fit time is given on the torre

# 7111-7 Geometry VI - Two Span Semicircular Arch Bridge Constriction Rough Boundaries (See Fig. 3-14 for definition of geometries)

This geometry represents about 11% of all existing arch bridges in the state of Indiana. The great majority of these bridge constrictions have a thickness factor, I/b, of about 0.50 which was chosen for the experimental models of Geometry VI.

The test results are presented in the dimensionless graphs shown in Figure 8-7-1. To be noted is that the opening width of the models, b, is the width of one of the openings. The performance of the modes in Geometry VI is approximately the same as for Geometry I-b of same thickness factor.

A total of 27 test runs were made and the data taken are recorded in test runs no. (256-282) as shown in Table 8-2. The calculated values are presented in Table 3-5. The values of the backwer superslevation ratio  $h_1^{*}/V_n$  were plotted vs.  $(F_n/A^{*})^2$ in fig. 8-10-9. The equation of a visual fit line is given on the figure. The values of the discharge coefficient K are plotted vs. the channel opening ratio h in Fig. 3-9-8. The equation of a visual fit line is given on the figure.

# 111-5 Geometry VII - Segment Arch Bridge Rough Boundaries (See Fig. 3-14 for definition of geometries)

With reference to Figure 3-12, it is readily seen that the dimensionless relationship of the discharge as a function of the backwater depth and the width of the constriction holds true also for the segment arch openings. The curves seen in figure 8-8-1 are based on this relationship as expressed in Equation 3-7-33, terived in Chapter III. A total of 22 test runs (No., 285 to 307) were done and recorded it vable 8-2. The calculated values are found in Table 8-5.

Figure 8-8-1 shows that the spacing between the curves becomes very small as he Froude number increases beyond 0.3. This means that the Froule number becomes nsignificant as a parameter when the degree of turbulence increases beyon a cost. In alue ( $F_n > 0.3$ ).

The values of the backwater superclevation ratio http://in were plotted vs. (1) // in fig. 8-10-10. The equation of a visual fit line is given in the figure. In alues of the discharge coefficient K are plotted vs. the channel opening ratio in in ig. 8-9-10. The equation of a visual fit line is given in the figure.

# VIII-9 Head Loss due to Submerged Arch Bridge Constriction-

In the theoretical analysis in Chapter III an expression for the additions let loss coefficient due to the bridge constriction was found, equation (3-8-13) Assuming  $\alpha'_1 = \alpha'_1 = \alpha'_2$  then

$$K = \frac{h_1^*}{V_0^2/2} - \alpha \left[ \left( \frac{A_0}{A_y} \right)^2 - \left( \frac{A_0}{A_1} \right)^2 \right]$$
(3.34)

This equation, which holds for subwerged bridge constrictions is sitular to the cas for free surface flow through a bridge constriction, and K various primarily its

- a) The channel opening ratio Mt
- b) Typ. and shape of bridge constriction with as inlet shape, win, wall tight is skewness angle \$2 and thickness factor, L/b
- c) Number of bridge spars, N
- d) Eccentricity of buildge opening, e
- e) Froude number Fn

The head loss coefficient K was computed or the IBM 70 0 from equation 2-4The corresponding value of the channel opening ratio, M<sup>0</sup>, was also computed, and by a least mean square routine the slope of the line of the plot K versus M<sup>0</sup> as found in arithmetic scale. A significant value such as  $1/b_0$ ,  $f_2$  and was used at parameters both for smooth and rough boundary tests. The plot of these curves are shown in Figures 8-9-1 to 8-9-10.

For the smooth boundaries a sufficient number of data was available to use the least square method on the computer. In the case of rough boundaries, however, the data available for each parameter were not always sufficient and some of the lines had to be fitted by eye.

The data used for the calculations are listed in runs 101-306 presented in Table 8-2. The calculated values from the computer are listed in tables 8-4 and 8-5.

# VIII-10 General Backwater Equations for Geopetries I-a, I-b, VI and VII - Smooth and Rough Boundaries

The theoretical development of these equations was done in Chapter III. The generalized backwater equation is of the form

$$\frac{\mathbf{x}_{1}}{\mathbf{x}_{n}} = C \left[ \left( \frac{\mathbf{F}_{n}}{\mathbf{M}} \right)^{2} \right]^{\frac{1}{2}}$$
(3-10-1)

However, it was found more convenient to plot the test results for the fully submerged discharge jet in the form

$$\frac{\mathbf{I}_{1}}{\mathbf{Y}_{n}} \rightarrow \mathbf{1} = \mathbf{C} \left[ \left( \frac{\mathbf{F}_{n}}{\mathbf{N}^{0}} \right)^{2} \right]^{\frac{1}{2}}$$
(3-10-2)

If  $h_1$  is substituted for  $Y_{-} - Y_{ns}$  quation 8-10-2 can be written in the following form:

$$\frac{h_1 *}{Y_n} = C \left[ \left( \frac{F_L}{M^1} \right)^2 \right]$$
(8-1)-3)

This is the presentation used in the graphs at the end of this section. Figure33-1)-. to 8-10-10 present the generalized backwater equations for the geometries considers, Significant values of the geometries are used as parameters.

Values for  $\frac{h_1}{n}^*$  and  $\frac{h_2}{n}^*$  were calculated in the same computer program as used for the head loss considerations in the previous section. These points were to be plotted on a log log scale, so the computer program was also written to determine the slope of the best fitting line through chese points. A least mean square rout the for log log plots was used. For the smooth boundaries a sufficient number of is that available to use the least square method in finding the slope of the line. In the case of rough boundaries, however, it was sometimes necessary to fit the lines by eye from the few data available.

The data used for the calculations are listed in runs 101-305, presented in table 8-2. The calculated values are listed in tables 8-4 and 8-5.

There are some advantages in doing all these calculation; in one computer program, Only one set of data has to be prepared and the computer stores the values in the memory once. The computer time used, therefore, becomes a minimum. For all the calculations pertaining to the figures 8-9-1 to 8-10-10, the time used by the IBM 7090 was less than 10 minutes.

.

# VIII-11 Conclusions - Submerged Tests

The results of this report are applicable to bridges with submerged inlet and geometries I-a, I-b, V-b, VI and VII (see definition of geometries in Chapter III and figure 3-14.) To eccentricity, entrance rounding or ving the are counted for.

Findings was the following.

1. Equation of discharge for original type flow through a mai-2 reliar sub-

2. Semi-empirical expression is the operation of the product of the boundary boundary with the boundary of the

3. Graphs presenting the data have a finite of the recent file "relater depth to the arch sorres

4. Graphs showing the ratio of the balance of the normal lepth Frederic in a contact and a contact of the normal lepth Frederic in a contact and a contact of the normal lepth frederic in a contact o

5. Empirical generalists between on thems for an bound of our files from the graphs in point 4.

6. Graph giving the value of the output the second of figure to a function of the channel op. Ung the optime if in the optime the second optime if it is a second optime is a second optime is a second optime if it is a second optime is a second optime is a second optime if it is a second optime is

## Seat 2 days in a

1. The back rater to the ergen of the disc of relations of the stream decreases as the longth of the story of the the discussion of the more areas

2. For equal brid angth the orthwater dutic carnet get and bridges is further decreased when the constriction is a wed to expect to the direction of flow, at least for skew angles up to and

3. The value of the head loss co fficing the to a short orginal arch bridge constriction is independent of the normal depth roude number as long as both inlet and outlet of the bridge are submerged 4. For hydraulically short submerged arch bridges (that is where the barrel is not sufficiently long to allow the expanding depth of flow below the contraction to raise and fill the barrel), the backwater depth increases for an increase in the normal depth Froude number.

5. To obtain a minimum backwater depth at a submerged arch bridge constriction the barrel should flow full (full conduit flow).

# IX FIELD VORK

# IX-1 A Preliminary Field Survey of Sites for Mc.el.-Provet pe Comparison of the Avdraulics of Arch Bridge Constrictions by T. P. Chang and J. T. Strange

INDI'NA FLOOD CONTROL AND WATER RESOURCES M MISSIC. INDIANAPOLIS, INDIANA

# Introduction

A series of model-testing studies of the horaulies of series also constricteds have been add during the pist Youn years at the Hydrablic line of Furche University, unler the middance of Fromsmor D.T. unler ad Fromesery 1 The Inclana Flood Control and Vater Resources Contission his black of some of the arch bridges in Indiana solution to be supported to the a field check of some of the arch bridges in Indiana solution is only provide to the some by taking field test which during a flood

The proliminary field reconcussions work to be units including the boot of this year. Hore the LCC such bridges, both one-special in the second secon

# Criter's last for the 20 - 1. on or Briles Sives

1 Locata n

For the transformed of future filles end only the set of the set o

# 2. Channel Condition.

(a) The channel reach close to this bridge site both upser and downstream, spuld be as clear as possible so that the flow during the flood will not be unduly disturbed.

(b) The cross sections of the channel reach, both upstream and downstream from the bridge, should be uniform in order to avoid the effect of head loss due to enlargement and contraction, and the water should be confined within the stream banks during the flood,

(c) The channel reach should be nearly straight. There should be no confluence close to the bridge.

(d) The shape of the channel cross sections should have as few irregularities as possible.

(e) There should be no structures, such as other bridges, dams, wiers, etc., near the bridge site.

(f) Bridge skew should not be excessive. Showed bridges, in which the center line of the openings are not parallel with the direction of the channel, should be avoided.

3. Arches of the Bridges:

(a) The shape of the arch should be close to sendcirbular or a part of a circular arch.

(b) The opening ratio of the arch should be small enough to produce an appreciable backwater effect.

(c) Excessive eccentricity should be avoided. The arth opening should be symmetrical with respect to the center line of the channel.

4. Flow Condition:

(a) Flow velocity during a flood should be great enough to create a relatively high Froude number, F vin .
(b) The drainage basin should be relatively large (about 15 square miles or more) and topographic relief within the basin relatively small in order to produce a flood of sufficient duration so that

desired data may be obtained.

# scription of the Field Situation

As far as the purpose of this report is concerned, all of the arch bridges ich have been inspected can be put into three general classifications. These e; Highway bridges, City bridges and County Road bridges. Each has its general atures and problems.

1. Highway Arch Bridges:

(a) Arch shape is excellent, but almost all of them have a very large opening. Flood waters will probably never reach the middle part of the arch. Hence, the arch effect is negligible.

(b) Usually the road side drainage ditches are near the faces of the bridge, both upstrian and downstream. Thise drainage ditches suddenly enlarge the channel cross section.

(c) Channels are not very clear or uniform. Channel and bank conditions are such that if flood levels were high enough to produce the desired arch effect, water would spill out into the fields; thereby, greatly changing the shape of the channel cross sections.

(d) Since the traffic is usually heavy, working conditions on the bridge are not good.

(e) Usually the length of the arch constrict on is relatively long2. City Arch Bridges:

(a) Channel conditions are fairly good, especially in the park ar ds.

(b) Flood water is confined within the banks. Flood water levels are high enough to produce an arch effect.

(c) The shaps of the arches are not good, either too flob or not semicircular.

(d) Jacques are in the channel reach close to the bridge or under the bridge. This is a common and serious problem with the city bridges.

# 3. County Road Arch Bridges:

(a) The openings of the arch are usually small.

(b) Channels are not uniform and are sometimes croked by brush and small trees.

(c) Farm fences across the channel near the bridge catch logs and debris which seriously affect the flood flow under the bridge.

(d) The channel and bridge size is usually smaller than the highway arch bridge or the city arch bridge.

# Conclusion

After three weeks of field inspection of the arch bridges in the eight counties near Indianacolis, the writers are of the opinion that it is extremely difficult to find an arch bridge for the purpose of model-prototype comparison without any defects. However, ten of the most suitable bridge sites have been selected for further investigation. Each one has some good features and some defects. The list of the bridges are as follows:

1. Oleny Street, Indianapolis, crossing Pogues Run (Bridge No. 2a)

2. Brookside Parkway, Indianapolis, crossing Pogues Run (Bridge No. 2b)

3. Jefferson Avenue, Indianapolis, crossing Poques Run (Bridge No. 2c)

4. South Belmont Avenue, Indianapolis, crossing Little Buck Creek (Bridge No. 8a)

5. State Road 100, crossing Williams Creek (Bridge No. 13)

6. Villa Avenue, Indianapolis, črossing Pleasant Run (Bridge No. 15a)

7. Linden Avenue, Indianapolis, crossing Pleasant Run (Bridge No. 15b)

8. State Road 44, crossing HurricansCreek (Bridge No. 51)

9. White Lick Creek near State Boys School, Plainfield (Bridge No. 59a)

10. Dean Road, Indianapolis, crcSsing Howland Ditch (Bridge No. 66a)

The complete information for each of the above bridges are compiled on a eparate form. A suggestion given by the writers has been made for each of those ridges.

# K-2 Summary of Information on Bridge Sites

For each of the ten bridge sites a resume sheet, one plan view and four cross ections were prepared by the Indiana Flood Control and 'ater Resources Conmission. ereal photographs were made by the State Highway Department of Indiana for nine if these sites. Topographic maps were prepared from these aer al photographs in the irphoto Laboratory, School of Covil Engineering, Purdue University, under the upervision of Professor R. Mile .

# I. OINEY STREET, INDIANAPOLIS, CROSSING POCUES RUN

- 1) Compiled number of bridge: No. 2a, 2) Location: Lat. 39°47'25", Long. 86°05'26", Indianapoli. east Indiana, quadrangle sec. 32, T. 16 N., R. 4 E., at bridge on Olney Street, Indianapolis, crossing Poguas Run, Eastern central part of Indianapolis.
- Drainage area: 8.7 square miles.
- 3) (4) (5) (6) (7) (8) (9) (9) (1) Tributary to: Tributary to West Fork Unite River Vabal River basin.
- Slope of channel: 3.1/1000.
- Bridge type: One-span concret ar h bridge.
- Skew toward upstream: 2º I.
- Skew toward downstream 25° L.
- Width of Bridge: 50 feet.
- Clearance of arch opening: 12 7 feet
- Midth between banks of charnel: 90 feet,
- 2) Width of channel botton: 24 feet
- Eccentricity: Low flow part of channel located at lero in the hannel. 3) Center line of arch is about 7 feet right or center line of low flow channel, but the .rch center line early concludes which the center line between the two banks.
- 4) High water mark: 3 sect below top of arch opening but dote is uncertain.
- .5) Channel condition:
  - (a) Bend about 175 fe t up bream from upstream face of bridge. Bend about 180 fe t downstream from downstream face of bridge
  - (b) Trees and brush distributed along right banks Lot. upstream and downstream.
  - (c) Two 1-foot diamet r sewers in upstream side.
  - (d) Channel bo t m is irre lar.
  - (e) A bridge of them on Drive loc ted 1,500 feet unstre from this site A bridge on Brooksid Parkway (Bridge 1. 21) ocaled 1 00 fet downstream from this site.
- 6) Arch condition: In upper part, arch is in good shape but lower part is straight wall.

ggestion: The opening ratio is small, For a very high Mood the channel irregularity and the straight part of the creating could be neglected.

gures 9-1-1 to 9-1-5 show the plan view and cross sections at the bridge site. gure 9-1-6 is a topographic map prepared from aereal photographs.

#### II. BROCKSINE PARKMAY, INDIANAPOLIS, CROWING POGUES RUN

#### RESUME

- (1) Compiled number of bridge: No. 2b.
- (2) Location: Lat. 39°47°26", Iong. 86°05′kl", Indianapolis east, Indiana, quadrangle, sec. 32. T. 16 N., R. 4 E., at bridge on Brock ide Parkway, Indianapolis, crossing Pogues Run, eastern central part of Indianapolis.
- (3) Drainage area: 9.2 square miles.
- (4) Tributary to: Tributary to Unite Fiver Sabash River basin.
- (5) Slope of channel: 2,9/3.000.
- (6) Bridge type: One-span concrete anch bridge,
- (7) Skew toward upstream: 15°L.
- (8) Skew toward downstream: 51.
- (9) Width of bridge: 20 feet.
- (10) Clearance of arch opening: 16 feets.
- (11) Lidth between banks of channel: 127 fast
- (12) Eccentricity: No.
- (13) high water mark: 4.8 feet below top of tich, date is unertained
- (14) Channel condition:
  - (a) Bend at 170 feet upstream from upstream face or bridge.
  - (b) Bushes distributed along barks both upstream and downstream,
  - (c) Channel bottom is regular but parabolic shape.
  - (d) 18-inch diameter sever in downstream tide.
  - (e) A bridge on North Rual Street located 2,200 rest downstream from the bridge Mo 2b.
- (15) Arch condition: It is a good whape arch
- Suggestion: The channel condition is very gap, only one server in Samatrees file Skew is small. It is recommended to be used for a common flood. However, the influence of 80° ten in upstream is uncertain.

Figures 9-2-1 to 9-2-5 show the plan view and arous sections on the brid site.

# III. JEFFERSON AVENUE, INDIANAPOLIS, CROSSING POGUES RUN

#### RESUME

- (1) Compiled number of bridge: No. 2c.
- (2) Location: Lat. 39947'11", Long. 86°07'28", Indianapolis east, Indiana quadrangle, sec. 31, T. 16 N., R. & E., at bridge on Jefferson Avenu-Indianapolis, crossing Pogues Run, eastern central part of Incianapolis.
- (3) Drainage area: 10.3 square miles.
- (4) Tributary to: Tributary to Vest Fork of White River, Wabash River basin.
- (5) Slope of channel: 2.4/1000.
- (6) Bridge type: One span concrete arch bridge.
- (7) Skew toward upstream: 5° R.
- (8) Skew toward downstream: 5° L.
- (9) Width of bridge: 45 feet.
- (10) Clearance of arch ovening: 11 feet.
- (11) Width between banks of channel: 70 feet,
- (12) Midth of channel bottom: 35 feet.
- (13) Eccentricity: None.
- (14) High water mark: Two high water marks, lover one is bout 6.7 feet below the top of arch opening, the higher one is about 1.4 feet below top of arch opening. Dates are uncertain.
- (15) Channel condition:
  - (a) No bend close to the bridge site both rostream and downstream side, Channel is straight.
  - (b) Trees and bushes distributed along both side banks of downstream channel.
  - (c) Two sewers upstream side, one 2-foot 6-inch diameter, another one 3-foot 6-inch diameter.
  - (d) Channel bottom is very good, regular and flat.
- (16) Arch condition: Upper part of opening is arch shape, but lower part is straight wall.

Suggestion: The two big sovers are major defect, but the influence is uncartain, otherwise, this is a good size.

Figures 9-3-1 to 9-3-5 show the plan view and crossections at the bridge site. Figure 9-3-6 is a topographic map of the bridge site prepared from acreal photographs.

# IV. SOUTH PELMONT AVENUL, INDIANAPOLIS, CROSSING LITTLE PUCK CREEK

# RESUME

(1)

- Compiled number of bridge: No. 82. Location: Lat. 39°40°00", Long. 86°11°47", Maywood, Indiana, quadrangle, (2)on east line sec. 9, T. 14 M., R. 3 E., at bridge on south Belmont Avenue, Indianapolis, crossing Little Buck Greek, southwestern corner of Indianapolis.
- (3) Drainage area: 15.7 square miles.
- Tributary to: Tributary to West Fork Thite River, Tabach Wiver this
- Slope of channel: 1.5/1000.
- Bridge type: One-span corcrete arch bridge.
- Skew toward upstream: 10°L
- Skew toward downstream: 150 R.
- Midth of bridge: 40 feet.
- (4) (5) (6) (7) (8) (10) (10) (11) (12) (13) Clearance of arch opening: 16.5 feet.
- Midth between banks of channel: 109 fest.
- Hidth of channel bottom: Not clear,
- Econtricity: The low flow pert is located in the lost the of the charten The center line of arch is about the same position is the center Mine between banks.
- High water mark: Flocd of 1955, 7.5 feet below tob of weh. (14)
- (15) Channel condition:
  - (a) The bridge site is just on a 25° curve.
  - (b) The channel is clear,
  - (c) Eanks are high enough to confine the flood.
  - (d) Channel both n is irregular.
  - (e) No sewers Leity.
- (16) Arch condition: It is in good shape but hoo flat-

Suggestion: The channel is clear and uniform, can's confine the flood flow we g well. It is recommended to be used during very ligh flood, & dut or more than the flood of 1955.

igures 9-4-1 to 9-4-5 show a plan view and cross sections at the bridge die. Igure 9-4-6 is a topographic map of the bridge site prepared from acreal photoraphs.

V. STATE ROAD 100, CROSSING WILLIAMS CREEK

# RESUME

- (1) Compiled number of bridge: No. 13.
- (2) Location: Lat. 39°54°44", Long. 86°10°28", Carmel, Indiana, quadrangle, on north line sec. 22, T. 17 N., R. 3 E., at bridge on State Road 100 crossing Williams Creek, 2 1/2 miles northwest of Augusta, Marion County, and 3.8 miles upstream from mouth.
- (3) Drainage area: 17.4 square miles.
- (4) Tributary to: Tributary to West Fork White River, Wabash River basin.
- (5) Slope of channel: about 2/1000.
- (6) Bridge type: One-span concrete arch bridge.
- (7) Skew toward upstream: 15° R.
- (8) Skew toward downstream: No skew.
- (9) Width of bridge: 42 feet.
- (10) Clearance of arch opening: 22 feet.
- (11) Width between banks of channel: About 160 feet.
- (12) Width of channel bottom: About 30 feet.
- (13) Eccentricity of arch opening: Center line of arch opening is 10 fest left of the center line of stream.
- (14) High water mark: Flood of April 25, 1961, 14.5 feet below top of arch opening. (15). Channel condition:
  - (a) No bend both upstream and downstream within 1,000 feet.
  - (b) Trees and bushes distributed on banks both upstream and downstream.
  - (c) Right banks both upstream and downstream are very low so that the flood water levels were limited to the lower part of arch opening.
  - (d) No sewer nearby.

Suggestion: The arch shape is very good. It is recommended to be used for very high flood.

Figures 9-5-1 to 9-5-5 show a plan view and cross sections at the bridge site. Figure 9-5-6 is a topographic map of the bridge site prepared from aereal photographs.

# VI. VILLA AVENUE, INDIANA POLIS, CROSSING PLEASANT RUN

# RESUME

- Compiled number of bridge: No. 15a. (1)
- (2) Location: Lat. 39°44' 53", Long. 86°07' 35", Maywood, Indiana, quadrangle. northeastern corner of sec. 18, T. 15 N., R. 4 E., at bridge on Villa Avenue, Indianapolis, crossing Pleasant Run, south part of Indianapolis.
- Drainage area: 12.9 square miles
- (3) (4) (5) (6) (7) (8) (9) Tributary to : Tributary to West Fork White River, Wabash River basin.
- Slope of Channel: 2.5/1000.
- Bridge type: One span concrete arch bridge.
- Skew toward upstream: 15° R.
- Skew toward downstream: 20° R.
- Width of bridge: 40 feet.
- 10) Clearance of arch opening: 14.5 feet.
- 11) Width between banks of channel: 116 feet.
- 12) Width of channel bottom: 65 feet.
- 13) Eccentricity of arch opening: Center line of arch opening is at 7 feet right of center line of low flow channel.
- 14) High water mark: Flood of April 1961, 7 feet below top of arch opening.
- 15) Channel condition:
  - (a) Bend at 150 feet downstream from downstream face. Bend at 150 feet upstream from upstream face.
  - (b) Channel is relatively clear and fairly uniform.
  - (c) The banks are high enough to confine the flood water.
  - (d) Bottom is very regular.
  - (e) A fairly large tree near upstream face may obstruct flood flow.
  - (f) Two 1-foot diameter sewers in upstream side.
- 16) Arch condition:
  - (a) Upper part is too flat.
  - Two ends of the arch are elliptic shape. **(b)**
- uggestion: The channel is clear and uniform but the arch seems too flat. It is recommended to be used for segment arch bridge model - prototype comparison. On the other hand, the degree of disturbances due to bends and sewers are uncertain.

'ignes 9-6-1 to 9-6-5 show a plan view and cross sections at the bridge site. 'igure 9-6-6 is a topographic map of the bridge site prepared from aereal photographs.

# VII. LINDEN AVENUE, INDIANAPOLIS, CROSSING PLEASANT RUN

# RESUME

- Compiled number of bridge: No. 15h. (1)
- Location: Lat. 39°44'45", Long. 86°08'14", Maywood, Indiana quadrangle, (2)west of sec. 18, T. 15 N., R. 4 E., at bridge on Linden Avenue, Indianapolis, crossing Pleasant Run, south part of Indianapolis.
- (3) Drainage area: 13.3 square miles.
- Tributary to: Tributary to West Fork White River, Wabash River basin. (4)
- (5) (6) Slope of channel: 1.5/1000.
- Bridge type: One span concrete arch bridge,
- (7) Skew toward upstream: 5º L.
- (8) Skew toward downstream: 15° Ĩ.
- (9) Width of bridge: 40 feet.
- (10)Clearance of arch opening: 11.5 feet.
- (11)Width between banks of channel: 107 feet.
- (12) Width of channel bottom: 55 feet.
- (13) Eccentricity: None.
- (14)High water mark:
  - (a) Flood of April 1961, 5.8 feet below top of arch.
  - (b) Highest water mark; uncertain date, 1 foot below top of arch.
- (15)Channel condition:
  - (a) Bend at about 200 feet upstream from upstream face.
  - (b) At arch bridge on Shelby Street crossing Pleasant Run at 1000 feet downstream from the bridge.
  - (c) The channel is uniform and clear.
  - (d) The banks are high enough to confine the flood.
  - (c) Channel bottom is very regular.
  - (f) Three 8 inch diameter sewers, one in downstream side and two in upstream side. One 12 inch diameter sewer in right bank of downstream side.
- (16) Arch condition: It seems too flat.
- Suggestion: The channel is uniform and clear, but the arch seems too flat. It is recommended to be used for segment arch bridge model - prototype comparison. On the other hand, the degree of disturbances due to upstream hend, downstream belidge and sewers are uncertain.

Picures 9-7-1 10 9-7-5 they a plan view and cross sections at the bridge site. Figure 9-7-6 is a topographic sap of Win bridge site prepared from acreal photographs。

### VIII. STATE ROAD 44 GROSSING HUPPICANE CRIEK

#### RESUME

- (1) Compiled number of bridge: No. 51
- (2) Location: Lat. 39º28' 51", Long. 86º02' 53", Franklin, Indiana, quadrangle, sec. 14, T. 12 N., R. 4 E., at bridge on State Road 44, crossing Hurrican Creek, eastern part of Franklin.
- (3)Drainage area: 27 square miles.
- (4)Tributary to: Tributary to Youngs Creek.
- (5) (6) Slope of channel: 1.3/1000.
- Bridge type: Two-span stony arch bridge.
- (7) Skew toward upstream: 35° R.
- Skew toward downstream: 35° R. (8)
- (9) Width of bridge: 45 feet.
- (10) Clearance of arch opening:
  - 7.5 feet for left side arch.
  - 9 feet for right side arch.
- (11)Width between banks of channel: 43 feet,
- (12)Width of channel bottom: 38 feet.
- (13) Eccentricity: The left part of channel is about 2 feet higher than right side, A part of the right side arch is obstructed by the right bank.
- (14) High water mark: 2.4 feet below the top of arches. Date is uncertain.
- (15) Channel condition:
  - (a) The channel is perfectly straight and uniform.
  - (b) Two sewers, one 3.5 feet diameter and one 2 feet diameter are under the right side arch opening.
  - (c) An obstacle, a 2 feet diameter sewer, is at 25 feet downstream from downstream face of bridge.
- (d) The bank slopes are vertical, both upstream and downstream.
- (16) Arch condition:
  - (a) The shape of the arch is very good.
  - (b) A part of the right side arch is obstructed by the right bank.
- Suggestion: The channel is uniform and straight. Since the skew is considerable, it is recommanded to be used for skew arch bridge model-prototype comparison. However, the influence of sewers are uncertain.

Figures 9-8-1 to 9-8-5 show a plan view and cross sections at the bridge sitz. Figure 9-8-6 is a topographic map of the bridge site prepared from acreal photographs.

#### RESUME

- (1) Compiled number of bridge: No. 59a.
- (2) Location: Lat. 39041'35", Long. 86023'53", Plainfield, Indiana, quadrangle, sec. 35, T. 15 N., R. 1 E., at bridge near State Boys School, Plainfield, crossing White Lick Creek, southern edge of Flainfield.
- (3) Drainage area: 101 square miles.
- (4) Tributary to: Tributary to Mest Fork White River, Mahash River basin.
- (5) Slope of channel: 1.4/1000.
- (6) Bridge type: Two-span concrete arch bridge.
- (7) Skew toward upstream: 5° R.
- (8) Skew toward downstream: 5º R.
- (9) Width of bridge: 18 feet.
- 10) Clearance of arch opening: 16 fest for right side arch, 12 fest for left side arch.
- 11) Width between banks of channel: 172 feet.
- 12) Width of channel bottom: Not very clear.
- 13) Eccentricity: Low flow part of channel is close to right bank.
- 14) High water mark: Flood of 1957, just about the same height as the top of arch opening.
- 15) Channel condition:
  - (a) Bend about 250 feet upstream from the bridge.
    - Bend about 300 feet downstream from the bridge.
  - (b) Sand heaps distributed along the left part of channel.
  - (c) Left part of channel is & feet, average, higher than right side.
  - (d) The banks are able to confine the flood & feet lower than the flood of 1957.
- 16) Arch condition: The arches are in good shape except the ends close to the pier.
- uggestion: This is the best two-span arch bridge under inspection. The defects are the non-symmetric channel cross section and the low bank on left side. So it is recommended to be used for a medium flood.

igures 9-9-1 to 9-9-5 show a plan view and cross sections at the bridge site. igure 9-9-6 is a topographic map of the bridge site prepared from asreal photographs.

#### RESUME

- (1) Compiled number of bridge: No. 66a.
- (2) Location: Lat. 39°53°33", Long. 86°05°54". Fishers, Indiana, quadrangle, sec. 29, T. 17 N., R. 4 E., at bridge on Dean Road, Indianapolis, crossing Howland Ditch, northern part of Indianapolis.
- (3) Drainage area: 4.5 square miles.
- (4) Tributary to: Tributary to West Fork Unite River.
- (5) Slope of channel: 3.5/1000.
- (6) Bridge type? Original bridge is a one-span concrete arch. Roadway was widened by addition of a one-span concrete flat decked structure on the downstream side of original bridge.
- (7) Skew toward upstream: 15º L.
- (8) Skew toward downstream: 15° L.
- (9) Width of bridge: Arch part, 18 feet.
- flat part, 13 fest.
- (10) Clearance of arch opening: 7 feet.
- (11) Width between banks of channel: 46 feet.
- (12) Width of channel bottom: 30 feet.
- (13) Eccentricity: None.
- (14) Highwater mark: None.
- (15) Channel condition:
  - (a) The cross section near the upstream face is larger than cross sections 10 feet upstream.
  - (b) Channel cross sections are regular.
  - (@) Small bend at 120 feet downstream from bridge.
  - (d) Channel is clear.
- (16) Arch condition:
  - (a) Arch opening is in good shape.
  - (b) Flat deck part of this bridge is high enough to be free from disturbing the flow.

Suggestion: Arch is good, channel is clear and regular. Perhaps the drainage area is not big enough to avoid the flash flood.

Figures 9-10-1 to 9-10-5 show a plan view and cross sections at the bridge site. Figure 9-10-6 is a topographic map of the bridge site prepared from aereal photographs.

# II-3 Preliminary Study of the Indirect Determination of Flood Discharge from Contracted Bridge Openings and High Water Marks, by I. P. Mu, Indiana Flood Control and Water Resources Commission

As soon as a bridge is constructed across a natural stream, it serves in at least some degree as a contracted opening to confine the stream flow, particularly in the higher ranges of discharge. Moder suitable conditions, such contracted openings provide opportunity for the indirect determination of the flood discharge passing through the bridge opening.

This office has for some years been engaged in a program for the field establishment of high water marks along a number of strems in the State following major floods, including the setting of such marks at and in the vicinity of bridges. The purpose of this study is to develop relationships whereby the high water mark data collected by this office at various bridges may be used to estimate the flood discharge at that point.

The experimental data used herein was obtained from a study by Purdue University on arch bridges and by the University of Colorado on simple normal crossings with a vertical-sided model.

# Basic Hydraulics and Assumptions for Deriving the Relationship Between the Flood Discharge and High Water Marks

When the flow is contracted by the bridge opening, the flow profile along the center line of the stream can be plotted roughly as follows:



where section 1 is the section of the highest heading up, the depth section o is the section right at the opening, the depth is yo  $y_n = normal depth$  $\Delta y = the maximum heading up = y_1 - y_n$  $v_n = normal velocity$  $v_1 = velocity$  at section 1  $v_0 = velocity$  at section 0  $S_0 = channel bottom slope$ L = longth from the bridge opening to the section of maximumheading up.

by conservation of energy, we have the relation

$$\frac{v_1^2}{2g} + y_1 + S_0 \Delta L = \frac{v_0^2}{2g} + y_0 + h_2^2$$

where hg is the friction loss from section 1 to section o. If So is small, so that the term Soll may be neglected, and assuming  $y_{ij} = y_{nj}$ Sq. 1 becomes:

$$\frac{\mathbf{v_1}^2}{2g} + \mathbf{y_1} = \frac{\mathbf{v_0}^2}{2g} + \mathbf{y_0} + \mathbf{h_f}$$
$$\frac{\mathbf{v_1}^2}{2g} + \mathbf{\Delta} \mathbf{y} + \mathbf{y_n} = \frac{\mathbf{v_0}^2}{2g} + \mathbf{y_n} + \mathbf{h_f}$$
$$\frac{\mathbf{v_1}^2}{2g} + \mathbf{\Delta} \mathbf{y} = \frac{\mathbf{v_0}^2}{2g} + \mathbf{h_f}$$

Since the high water marks are located on the banks of the stream, the elevation of the marks will be higher than the elevation of the water at the center line of the stream, due to differences in velocity. Assuming that the velocity of flow near the banks is very low or approaching zero, it might be said that  $\Delta y_{E}$ , which is defined as the difference in elevation between the high water marks on the banks and the iepth of normal stream flow  $y_{ns}$  is  $\Delta y$  plus its velocity head, that is  $\Delta y_{E} = y + \frac{v_{1}^{2}}{2\sigma}$ 

then 
$$\Delta y_{\rm E} = \frac{{\bf v}_0^2}{2{\rm g}} + {\rm h_f}$$

12

$$h_f = K \frac{v_0^2}{2e}$$

where K = friction loss coefficientthen  $\Delta y_E = (1 \div K) \frac{v_0^2}{2\sigma}$ 

or  $\frac{\Lambda yE}{y_n} = (1 + K) \frac{v_0^2}{2gy_n}$ 

and  $\frac{\Delta y_E}{y_n} = \frac{1+K}{2} F_0^2$ 

where  $F_0^2$  is the Froude number at the bridge opening.

This gives the relation that  $\Delta y_E$  is a function of the velocity head at the bridge opening and  $\frac{\Delta y_E}{y_R}$  is a function of the Froude number of that opening.

# Study of Experimental Data

Data from the experimental tests of flow through contracted openings made by the University of Colorado and Purdue University were collected. Since the solual field conditions investigated show that the opening ratio K<sup>0</sup> is large and the ratio of  $\Delta y$ and  $y_n$  is small, only those experimental data were selected which met the criteria that M<sup>0</sup> > 0.6 and  $\frac{\Delta y}{y_n} < 0.1$ . The relations of  $\frac{\Delta y_E}{y_n}$  and  $F_0^2$  can be seen in Figure 11-1, which shows that the data holds the relation:

$$\frac{\Delta y_E}{y_{ir}} = \frac{1}{2} F_0^2$$
  
r  $\Delta y_E = \frac{v_0^2}{2g}$ 

0

# Indirect Determination of Flood Discharge for Indiana Streams

Since the above study shows the relations of high water marks and the velocity at the opening, this will serve as a tool to estimate the peak flood discharge which passed the bridge opening during the particular high water event. Several bridges which are located on a relatively straight and uniform stream, and where the high water mark data is considered to be reliable were selected for this study. There are also U.S.G.S. gaging stations nearby at which the flood discharge was measured directly.

Information as to high water marks, computed discharge, and the actual discharge for the eight bridges selected for study is as follows:

Bridge Location	Date of Flood	Aye It.	Conputed Q cfs	Measured Q cfs
Eagle Creek, B&O RR Bridge at Speedway	June 1957	2.37	27,2850	25,200
Sugar Creek, at Edinburg	May 1956	0.90	2 <mark>3,000</mark>	27,600
Sugar Creek, Hwy. 231 near Crawfordsville	May 1956	0.81	23 <sub>8</sub> 700	18,200
Sugar Creek, Hwy. 231 near Crawfordsville	June 1957	1.012	31,400	26 <b>,</b> 300
Sugar Creek, Hwy. 41 Bear Byron	Ma <b>y 19</b> 56	0.34	19,600	20 <sub>9</sub> 600
Raccoon Creek, Hwy. 59 near Mansfield	1957	1.00	37,400	38,400
Flat Rock River, 孙汀。421 near St. Paul	Jan. 1959	0.274	13 <sub>9</sub> 300	14,,500
Flat Rock River, Hwy. 421 near St. Paul	Nov. 1955	0,265	9,600	10 <sub>9</sub> 800

There is shown in Figure 11-2 a plot of  $\Delta y_E$  against the velocity head. Data from the eight bridges shown in the above tabulation, and the experimental study from the University of Colorado and Purdue agree well with the theoretical line  $\Delta y_E = \frac{v_o^2}{2g}$ . It appears that this theoretical line can be used for a rough estimate of the peak flood discharge which passed through the bridge opening.

# Discussion

Theoretically, the flood discharge passed through the bridge opening is a function of the normal depth of stream flow, the cross section of the channel, the size of the bridge opening, the friction loss in the channel and the bridge opening, and the emergy coefficient. Since the normal depth of the natural channel is difficult to determine, the assumption was made that normal depth occurs at the bridge opening. This was based on the results of the experimental study made by Purdue University, which showed from the flow profiles that normal depth occured somewhere close to the opening under the condition that the opening ratio  $M^{\circ} > 0.6$  and that the Froude number is smaller than 0.5. The other major assumption was that the velocity of flow close to the bank is nervy small or approaching zero, such that the elevation of the high water mark on the pank is equal to the elevation of the flow profile along the center line plus its relocity head. Neither of these assumptions can be proved at this time and hence, constitute the weak points of this study.

As the velocities used in this study are all mean velocities, the velocity head actually is not  $\frac{v^2}{2g}$  but  $\alpha \frac{v^2}{2g}$ . The energy coefficient  $\alpha$  is a little larger than me for large uniform channels, but is far greater than one for the section close to a contracted opening. Since it is assumed to be unity in this study, a loss of accuracy undoubtedly results.

As shown in Figure H-1 data from the experimental tests by the University of colorado and Purdue agree well with the relation  $\frac{A_{YE}}{y_{B}} = \frac{1}{2F_{0}}^{2}$ , or  $A_{YE} = \frac{v_{0}^{2}}{2g}$ . This does not mean that the friction coefficient K is zero, but rather indicates that this factor may be cancelled by the energy coefficient  $\alpha'$ , or that there are other uncertained effects as a result of the two basic assumptions.

By studying the relation  $\Delta y_{\rm E} = \frac{v_0^2}{2g}$ , it will be noted that this is nothing more han the simple statement that the velocity head at the bridge opening is just equal o the maximum heading upstream from the contracted opening. Inasauch as assumptions ade in this study are not quite correct and since the friction loss coefficient K and energy coefficient  $\Delta$  are not precisely determined, the results of this study can ally be used as a rough estimate of the flood discharge or as a check on the flood ischarge as determined by other methods.

Further work is necessary to use the experimental data in conjunction with that rom actual natural streams to evaluate all the specific factors involved.

# APPENDIX A

# NOTATIONS

SYMBOL	DIMENSIONS	DEFINITION
A	L <sup>2</sup>	Area
Al	L <sup>2</sup>	Total depth water area at section 1, $A_1 = BY_1$
Ae	L <sup>2</sup>	Critical flow area
Anl	L <sup>2</sup>	Normal depth water area, Anl = BYn
An2	r <sub>5</sub>	Total depth flow area at section 2
Ao	L <sup>2</sup>	Area of bridge opening
В		Rectangular channel width; Body force per unit of mass
P.1	L	Diameter of circle of which the arch segment is a part
Ъ	L	Span width at the springline of the arch
C		A coefficient
C	L <sup>1</sup> /T	The Chezy roughness coefficient= $V_n/\sqrt{R_nS}$
C1		Coefficient, see equation 3-7-4
C ₀V ₀		Control volume
Ce		Coefficient of contraction
C.S.		Control surface
Cd		Coefficient of discharge
Cm		Correction factor = M <sup>0</sup> /M (see Fig. 3-13)
C.w		Coefficient of velocity
d	L	Diameter of a circle, distance between center of circular arch segment and channel bottom (Fig. 3-12), depth of flow measured perpen- dicularly to the bottom.
D	L	Hydraulic depth as defined by $\Lambda/W$ ; pipe diameter

SYMBOL	DIMENSIONS	DEFINITION	
d,	L	Distance from the springline to the center of curvature of the arch	
đ	L	Orifice diameter	
Eledi		Energy loss between section 1 and 4	
0		Econtricity	
F		Denotes a mathematical function, Froude number	
Ff		Friction force	
Fn		Normal depth Froude number = $\nabla_n / \sqrt{g X_n}$	
Fs	F	Surface force on control volume	
Î		Denotes a mathematical function, friction coefficient	
G		Denotes a mathematical function	
g		Denotes a mathematical function	
g	L/T <sup>2</sup>	Acceleration of gravity	
Н	L	Total energy head	
hl*	L	$\mathbf{X}_{1} = \mathbf{X}_{\mathbf{n}}$	
h		Static head	
hg	L	Headloss	
K		Weir coefficient as defined in Francis weir formula	
K		Head loss due to bridge constriction alone	
k		Nikuradse's equivalent sand roughness	
L	L	Length of the bridge in the direction of flow	
Ld	L	Distance between bridges	
L1-3	L	Distance between sections 1 and 3	
SYMBOL	DIMENSIONS	DEFINITION	
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Link	L	Distance between sections 1 and 4	
M		Channel width ratios b/B	
M0		Channel opening ratio, $\Lambda_0/\Lambda_n$ for submerged opening; $\Lambda_{n2}/\Lambda_{n1}$ for unsubmerged opening.	
Ml		Mild slope backwater curve in an open channel	
М		Number of spans	
n	L1/6	Manning's roughness coefficient	
n		Subscript which refers to the normal depth for uniform flow	
P	$F/L^2$	Pressure intensity; pier width (See Fig. 3-14)	
Р	F	Pressure force	
Qa	l <sup>3</sup> /T	Actual discharge through opening	
Qt		Theoretical calculated discharge through opening	
Q	$L^3/r$	Total flow	
q	L <sup>3</sup> /T	That portion of the total flow which could pass through the bridge without contraction	
R	L	Hydraulic radius	
R, Ro		Reynolds number V <sub>RRn</sub> /2	
r	L	Radius of curvature of the arch, radius of curvature of streamline	
S		Slope; head loss per unit length	
So		Slope for normal depth flow in unconstricted channel	

SYMBOL	DIMENSIONS	DEFINITION
T		A series defined in Art. 17-7a
t	T	Time
V	L/T	Average velocity
vo	L/T	Average velocity through constriction, $Q/A_{O}$
Vn	L/?	Average velocity when uniform, normal flow occurs, Q/BYn
V		Vclt
W	L	Free surface width; weight of water within a control volume
X		Distance along flume
3	L	Hydraulic depth y = A/N
Y	L	Depth of flow
Ä	L	Head over the center of gravity of a section
Z1	L	Depth of flow at section of maximum backwater
Yc	L	Critical depth
Yn or Yo	L	Depth of the normal unconstricted flow
¥ <b>t</b>	L	Tailgate height
Z		Section factor for critical flow compu- tation $2^2 = A^3/N$ ; elevation head
a		Kinetic energy coefficient
x'		Pressure coefficient
ß		Segment factor d/r, Momentum coefficient
8	F/L <sup>3</sup>	Specific weight of water
8		Boundary layer thickness

SYMBOL	DIMENSIONS	DEFINITION
٤		Roughness size
7		d/r (See fig. 3-12)
ν	L <sup>2</sup> /T	Kinematic viscosity of the fluid
P	FT <sup>4</sup> /L <sup>4</sup>	Fluid mass density
<i>с</i>	F/L <sup>2</sup>	Shear stress
Ø1	degrees	Mingwall angle (See Fig. 3-14)
\$2	degrees	Skew angle (See Fig. 3-14)
Θ	degrees	Angle of inclination of channel

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	Δ <u>H</u>	- 4-1			F	2.1	
STAT C	NG NG	ELLMI	MATLE	10 e	Not a mar		
1	LUNG		MITH T	MCC'	, Forth	i di	
	2	3	an derendenseren sola delanadaria suna Espi-	.5	F-	e de serie se presente en la companya de la compa	
4.0	10	.420	.515	.515	.095	.075	0
5.0 5.5 5.6 5.7 5.8 5.9	9 8.5 8.4 8.3 8.2 8.1	.420	.515	.515 .515 .514 .514 .514 .514 .513	.095	.095 .095 .094 .094 .094	2 4" 2 2" 2 24" 2 6/3"
6.0 6.1 6.2 6.3 6.4 6.5 6.6	8 7.9 7.3 7.7 7.6 7.5 7.4		.515	.513 .513 .513 .513 .513 .513 .513 .513	.095	.093 .093 .093 .093 .093	1 .002 · h · .002 · 2 · .005 · Postion to
6.8 7.0 7.065 7.72	7.2 7 6.935 6.28	.420	.515	.512 .512 .512 .513E	.095	.092 .092 .092 .093	.047
8 21	6		.515	51.5	.095	001	32
9	5		.515	• )1413	.095	•094	
9.065 9.1 9.3 9.5	4.035 4 4.7 4.5			.515 .515 .516		.095 .095 .096	
10.0	$l_{+}$	.420	.515	.517	.095	.097	
11.0	3		.515	.517	.095	.097	
12.0 13.0 14.0	2 1 0	.420	.515 .515 .515	.517 .517 .517	.095 .095 .095	.097 .097 .097	

4-1

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P2

N. FN

		- 1	÷ ē.			1 5 2	
	<i>i</i> .			1		2	
4.0	10	.1425	.538	.538	.113	.113	
5.0	9 8.5		.538	.538	.113	.113	- 4 <sup>n</sup>
6.C 6.5	8 7.5		.538	.538	.113	.113	2"
7.0 7.5	7 6.5	1	.538	.538	.113 .113	.113	- · 1"
7.75 8.00 8.1 8.2	6.25 6 5.9 5.8		. 538	.538 .537 .537 .537	.113	.113 .112 .112 .112	.001
8.4 8.5 9.0	5.6 5.5 5		.538	•535 •534 •535	.113	.110 .109 .110	1 ε 3.51 to M 5 511
9.21 9.5	4.5			.536E	1	.111	
9.725	4		.538	•537E	.113	.112	.035
10.25 10.5 10.6 10.7 10.8	3.5 3.4 3.3 3.2 3.1			•538E •538 •539 •539 •539		.113 .113 .114 .114 .114	
11.0 11.5	3		.538	•539 •539	.113	.114	
12.0 12.5	2 2.5		.538	•539	.113	.114	
13.0 13.5	1		.538	.539	.113	.114	1
14.0	0	.425	. 538	. 539	.113	.114	

		6-1			, i e	2.3	178
STALLA	N N S	F IF.	VATER	F LOF	N_=YA	N N N	
	-	3	-+	E	5		8
4.0 4.5 5.0 5.5	10 9.5 9 8.5	.420	.515 .515	.515 .515 .514	.095 .095	.095 .095 .094	Sc= 0 - 2.17.0114 Dis 5"
5.9 6.0 6.1 6.2 6.3 6.4 6.5 6.6 6.7	8.1 8 7.9 7.8 7.7 7.6 7.5 7.4 7 3	.420	.515	.514 .513 .513 .513 .513 .513 .513 .513 .513	.095	.094 .093 .093 .093 .093 .093 .093 .093 .093	L= 24" Z= 6/8" Weir Heig <sup>s</sup> *= h <sup>A</sup> = .001 △h= .003
6.8 6.9 7.0 7.24 8.0	7.2 7.1 7 6.76 6	.420	.515	.513 .513 .513 .514E	.095 .095	.093 .093 .093 .094	En the 51 Hosition to Along 71 Flume 71 Flume 71
8.24 9.0 9.1 9.2 9.3 9.4 9.5	5 4.9 4.8 4.7 4.6 4.5		.515	.515£ .516 .516 .516 .516 .516 .516	.095	.095 .096 .096 .096 .096 .096 .096	"ono <del>3</del> 1:
10.0	4	.420	.515	.516	.095	.096	
11.0	3		.515	.516	.095	.096	
12.0	2		.515	.516	.095	.096	Í
12.5	1		.515	.516	.095	.096	•
13.5	0	.420	.515	.516	.095	.096	

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P.4

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nelji 1 Angger so munitacijimu nima iti us anumi mu	• •			h.			
a deservativ a de a como como como como				tantona a carto ange			
4.0	10	.425	.538	.538	.113	.113	0
5.0	9		.537	.538	.113	.113	5"
6.0	8		. 538	.538	.113	.113	2.5" 2L"
7.0	7.0		.538	.538	.113	.113	1"
8.0 8.1 8.2 8.3 8.4 8.5	6.0 5.9 5.8 5.7 5.6		.538	.538 .537 .536 .535 .534 .534	.113	.113 .112 .111 .110 .109	.001 .005 3.51
8.695 9.0	5.315 5.0		.538	•533E	.113	.108	to 5,51
9.19 9.5	4.5			•534E	,	.109	0.0 4
9.693 10.0	4.0		.538	•535E	.113	.110	.035
10.183 10.5	3.5		. 538	.536E .537	.113	.111	
10.6 10.7 10.8 10.9	3.4 3.3 3.2 3.1			•537 •538 •539		.112 .113 .114	
11.0	3.0		.538	.539	.113	.114	
12.0	2.0		.538	.539	.113	.114	
13.0	1.0		.538	.539	.113	.114	
14.0	0	.425	.538	.539	.113	.114	

		4-1			F	2.5	180
IST4 IN	and the second second		WATER	INFACE	NEHTIL		
Support Supervise from the sector of	1 · ·		MOFEL	MUDEL IN PLACE		IN ACE	
		5. 	4	E	5	7	8
4.0	10 9.5	.420	.515	.515	.095	.095	Sof 0
5.0	9	.420	.515	.515	.095	.095	7.7"
6.0	°•2 8 7•5	.420	.515	.515	.095	.095	Hora 1.4" - 9.7"
7.0	7		.515	.515	.095	.095	2 6/8"
7.7 7.8 7.9 8.0 8.1	6.3 6.2 6.1 6 5.9	.420	.515 .515	.515 .514 .514 .514 .514	.095	.095 .094 .094 .094 .094	vier <u>:</u> Heyrt ∧ <sup>*</sup> = .001 ∧lic .003
8.2 8.5 18.75 9.0	5.8 5.5 5.25		51 5	.514 .514 .513	095	.094 .094 .093	Along 5.251
9.55	4.45		• / 4 /	.515	.075	.095	FR VIEw
9.50 9.7	4.4		.515	.516	.045	.096 .096	E O NO <del>30.</del>
9.8	4.2		.515	.516	.095	.096	5.
10.0	4.1	.420	.515	.516	.095	.096	
10.5	3		.515	.516	.095	.096	
12.0	2	.420	.515	.516	.095	.096	
12.5	1		.515	.516	.095	.096	
14.0	0	.420	.515	.516	.095	.096	

	· ·····		ICH SJAAPA	JE WILAD!	LIVE VILLE	. 6	
	DISTANCE	POIN	IT GAGE R	ENDING	NODAAN	DEPTH	
STATION	ALONS	FLUME	WATER S	SURFACE	NORMAL	MODEL	
	FLUME	BC. TOW	MUDEL	MODEL IN PLACE	DEPTH	PLACE	
	2	3	4	5	6	7	8
4.0 5.0	10 9 8-5	.420	.520 .520	.520 .520	.10 .10	.10 .10	S <sub>C</sub> = 0 Q <sub>mī</sub> .0138 Dia= 4"
6.0 7.0 7.5 7.6 7.7 7.8	8 7 6.5 6.4 6.3 6.2	.420	.520 .520	.520 .520 .520 .519 .518 .518	.10 .10 .10	.10 .10 .099 .098 .098	Rise=2" L= 24" Z- 6/8" Wer Haght=
7.9 8.0 8.1 8.2 8.3 8.4 8.5 8.6 3.7 8.6 3.7 8.8 8.8	6.1 5.9 5.8 5.7 5.6 5.5 5.4 5.3 5.2 5.2	.420	.520	.518 .518 .519 .518 .518 .518 .517 .517 .517 .517 .517	.10	.098 .098 .098 .098 .098 .098 .097 .097 .097 .097	h <sup>*</sup> = .001 ! Δh = .004 ! Bridge = 3 ! Position to Along 5 ! E PG-M.PG= .047 FIG. NO.
9.0	5		.520	.517 .517	.10	.097	
9.71	4.29	.420	.520	*218F	.10	.098	
10.22 11.0 11.1 11.3 11.5	3.78 3 2.9 2.7 2.5		.520	.519E .520 .521 .522 .522	.10	.099 .10 .11 .12 .12	
12.0 13.0 14.0	2 1 0	.420	.520 .520 .520	.522 .522 .522	.10 .10 .10	.12 .12 .12	

TABLE 4-IWATER SURFACE MEASUREMENTS P. 7

	DISTANCE	POIN	IT GAGE RI	EADING		DEPTH	
STATION	ALONG	FLUME	WATER S	URFACE	NORMAL	MODEL	
UNATION	FLUME	BOTTOM	WITHOUT	MODEL	DEPTH	IN PLACE	
	2	3	4	5	6	7	8
4.0	10	.420	.520	.520	.100	.10	$S_0 = 0$
4.5 5.0	9	.420	.520	.520	.10	.10	Dia= 5"
5.5	8	.420	. 520	.520	.10	.10	Rise=2.5"
6.5 7.0 7.3 7.4 7.5 7.6 7.7 7.8 8.0 8.2	7 6.7 6.5 6.5 6.4 6.3 6.2 6 5.8	.420	.520 .520	.520 .519 .519 .519 .519 .519 .519 .519	.10	.10 .099 .099 .099 .099 .099 .099 .099	$Z^{-} = 6/8^{11}$ Weir = Height = h <sup>*</sup> = .001! $\Delta h = .003!$ Bridge = 3! Position to
8.4 8.5 8.7 9.0 9.23	5.6 5.5 5.3 5 4.67			.519 .518 .518 .518 .519E		.099 .098 .098 .098 .098	Along Flume 5' E.R.GM.PG= •047 FIG. NO.
10.23 11.0 11.1 11.2 11.3 11.4	3.67 3 2.9 2.8 2.7 2.6	.420	.520	.519E .520 .520 .521 .521 .521	.10	.099 .10 .10 .101 .101 .101	
11.5 11.75 12.0 12.5 13.0	2.5 2.25 2	.420	.520	.521 .521 .521 .521 .521	.10	.101 .101 .101 .101	
13.5	0	.420	.520	.521	.10	.101 .101	

TABLE 4-1 WATER SHREACE MEAS REMENTS P.8

		POIN	IT GASE RI	FIDING		DEPTH	
	DISTANCE		WATER S	LARFACE	NORMAL	MODEL	
STATION	ALONG	FLUME	WITHOUT	MODE.	DEPTH	IN	
	FLUME	BOTTOM	MUDEL	IN PLACE		PL ACE	
I	2	3	4	Ę.	6	7	8
4.0 5.0 6.0 7.0 7.25	10 9 8 7.0 6.75	.420	.520 .520 .520 .520	.520	.10 .10 .10 .10	.10 .10 .10	So= 0 Qni .0138 Dia= 7.7" Rise= 1.4"
7.7 7.9 8.0 8.3 8.5 8.75 9.0	6.3 6.1 6.0 5.7 5.5 5.25 5.25	.420	.520 .520	.520 .519 .518 .518 .518	.10	.10 .099 .098 .098 .098	Z- 6/8" Weir Height <sup>±</sup> h <sup>*</sup> = .001!
8.75 9.0 9.50 9.55 9.60 9.70 9.8 10.0 10.25 10.50 11.0 11.5 11.75 12.0 13.0 14.0	5.25 5.0 4.45 4.4 4.3 4.2 4.0 3.75 3.5 3.0 2.5 2.25 2.0 1.0 0	.420	.520 .520 .520 .520 .520	.518 .519 .520 .520 .520 .521 .521 .521 .521 .521 .521 .521 .521	.10 .10 .10 .10 .10	.098 .099 .100 .100 .100 .101 .101 .101 .101	Ah = .0031 Fridae =4.451 Position to Along 5.251 E Pr5-M F G= FIG NO. <del>33</del> .

TABLE 4-1 WATER SURFACE MEASUREMENTS P. 9

	DISTANCE	POIN	T GAGE RI	EADING		DEPTH	
STATION	DISTANCE	FLUME	WATER S	URFACE	NORMAL	MODEL	
STATION	FLUME	BOTTOM	WITHOUT	MODEL	DEPTH	IN PLACE	
1	2	3	4	5	6	7	8
4.0 4.1 4.2 4.3 4.4 5.5 5.0 5.14 5.5 5.6 6.0 6.5 7.0 7.5 8.0 9.5 10.0 10.5 11.0 13.5 14.0	2 10 9.9 9.8 9.7 9.6 9.5 9.345 9 8.86 8.7 8.6 8.5 8.4 8.0 7.5 7 6.5 6 5.5 5 4.5 4 3.5 3 2.5 1 0	3 .425 .425 .425 .425 .425	4 .542 .542 .542 .542 .542 .542	5 .542 .541 .540 .539 .538 .537 .539E .540E .541 .541 .542 .543 .543 .543 .543 .543 .543 .543 .543	6 .117 .117 .117 .117 .117 .117 .117 .11	7 .117 .116 .115 .114 .113 .112 .114 .113 .112 .114 .115 .116 .116 .117 .117 .117 .118 .118 .118 .118 .118	8 So= 0 Qrrf.0138 Dia = 7.7" Rise = 1.4" L = 9.7" Z- 1" Weir = Height = h*= .001 Ah = .006 Bridge = 8.7' Position to Along 9.5' E PG-M PG: .042 FIG NO. <del>-29</del> -

2.6

-Δ	1	 4-1	1814	14.7	5.	de		5 D	·	5 TL		110	P	12	
	-	 a start reason	Arr 1,0000 - 1778	And Income the Party Name			see "tipe"	to a support statements	- the same ages		other and the	the opportunities	And in case of the local division of the loc	C	-

			THE SEA			16	
President and a second s		PD.	T GAGE I	E a Dirus		DEPTH	
1	L'ELANCI		WITTR	URELOF	LAMARCON	MODEL	
ISTATION	ALONG	FLUVE	MITHOUT	NY OF	DEDTU	1NI	
ee - 6	FLIME	BOTTOM	MODEL	IN PLACE		PLACE	
	2	3	4	5	6	7	8
	<u>}</u>						
4	10	.425	.542	.542	.117	.117	50= 0
4.5							0 <sub>TF</sub> .0139
5.0	9		.542	.542	.117	.117	Dia= 4"
2.2	0	1.25	510	510	117	117	Rise=2"
6.5	° 75	.427	• 742	• 742	• + + /	• 1 1 /	1 = 24"
2.0	7.0	1	51.2	51.2			
7.4	6.6	r t	.51.2	.51.2	.117	.117	Z- 1"
7.5	6.5		•/+~	.541		.116	Heat
7.6	6.4			.540		.115	L Congre
7.7	6.3			.539		.114	h <sup>*</sup> = .001
7.8	6.2			.539		.113	ah: .005
7.9	6.1		4	.537	6	.112	
8.0	6.0	.425	.542	.538	.117	.113	Postiun
8.70	5.30			•537E			Along 61
9.0	5.0		.542	.538E	.117		Flame 0.
9.72		1.05	510	.541E	220	110	EPG-MES.
10.0	4.0	.425	. 542	.542	•111	•11/	.035
10.1	2.7	1		• 545		118	FIG NO -20-
10.2	37	1		.51.3		.118	10.11
10.1	3.6			-543	- any more than the second secon	.118	
10.5	3.5	3		.543	-		
11.0	3.0		.542	.543	.117	.118	i
11.5				.543			
12.0	2.0	.425	.542	.543	.117	.118	1
12.5	1			.543		.118	
13.0	1.0		.542	.543	.117	.118	
13.5	0	1.05	51.0	.543		.118	-
14.0	0	.425	.542	. 543	.117	*118	
	1						1

TABLE 4-1 WATER SURFACE MEASUREMENTS P. 13

	DIGTANOS	POIN	IT GAGE RE	EADING		DEPTH		
STATION	DISTANCE	EL LINGE	WATER S	URFACE	NORMAL	MODEL		
STATION	FLUME	BOTTOM	WITHOUT	MODEL	DEPTH	IN PLACE		
	2	3	4	5	6	7	8	
4	10	.425	.542	. 542	.117	.117	$S_0 = 0$ $Q_{min} \cdot 0139$	
4.5 5.0	9	.425		.542	.117	.117	Dia= 5"	
5.5 6.0	8	.425	.542	.542	.117	.117	Rise=2 <mark>2</mark> " L= 24"	
7.0 7.5 7.6 7.7 7.8 7.9 8.0	7 6.5 6.4 6.3 6.2 6.1 6.0	.425	.542 .542	.542 .542 .540 .539 .539 .539 .539	.117 .117	.117 .117 .115 .114 .114 .114 .114	Z- 1" Weir = Height h <sup>*</sup> = .001 	
8.19 8.5	5.5		.542	.540E	.117		Position to Along 6	
8.69 9.0 9.185	5	.425		.541E			EPG-M.PJ=	
9.185 9.69 10 10.1 10.2 10.3 10.4 10.5 11.0 11.5	9.185 9.69 10 10.1 10.2 10.3 10.4 10.5 11.0 11.5 12.0 12.5	4 3.9 3.8 3.7 3.6 3.5 3 2	.425	.542 .542 .542 .542	.541E .542 .543 .543 .543 .543 .543 .543 .543 .543	.117 .117 .117 .117	.117 .118 .118 .118 .118 .118 .118 .118	FIG. NO. <del>19-</del>
12.5	1	1.25	51.2	.543	.117	.118		
13.5	1	•42)		.543	117	.118		
14.0	о	.425	.542	.543	.117	.118		

1 4-1		2.5	тç.	r.	:		TAIT	r	P. 14	F
-------	--	-----	-----	----	---	--	------	---	-------	---

	DISTANCE				N PWA		
STATIC	AL PNG FLUME	IFLIMI IBD T. M	W F NJE T	A LACE	JE T	A.L	í.
-	2	1	4	5	1	*	1
4.0 5.0 5.5 5.75	10.0 9.0 8.5 8.25	.423	.541	.541 .541 .541	.118	.118 .118 .118	0 .015 
6.0 6.5 6.8 7.C 7.72	8.0 7.3 7.2 7.0 6.28	.423	.541	.540 .539 .538 .538 .539E	.113	.117 .116 .115 .115 .116	24"
8.0 8.24	6.C 5.76	.423	.541	.540E	.118	.117	.0021
9.C 9.1 9.3 9.5	5 4.9 4.7 4.5	.423	.541	.541 .542 .543 .543	.118	.118 .119 .120 .120	hid 51 Field to
10.0 11.0 11.5	4 3 2.5	.423 .423	.541 .541	.543 .543 .543	.118 .118 .119	.120 .120 .120	.047
12.0 13.0 14.0	2 1 0	.423 .423 .423	.541 .541 .541	.543 .543 .543	.118 .118 .118	.120 .120 .120	

TABLE 4-I WATER E AS REME . . P. 15

p	y						
	DISTANCE	POIN	IT UNGE H	E'. DINC		DEPT r.	
STATION	ALONG	FLOME	WATER S	SURFACE	NURMAL	MODEL	
	FLUME	EOTTOM	WITHOUT	MODEL	DEPTH	IN DOL	former - no - no
			MCLET.	PH - LAUE		FUMUL	
1	-	3	4	5	0	?	9 #
4 5 5.5	10.0 9.0 8.5	.423 .423	.541 .541	.541 .541	.118 .118	.118 .118	So- 0 Qr-f .015 Dia= 5"
5.7 5.8 6.0 6.2 6.5	8.3 8.2 9.0 7.8 7.5	.423	.541	.541 .540 .540 .539	.118	.118 .117 .117 .116	Rise 2.5" 1 24" Z- 1"
6.8 7.0 7.24 7.75	7.2 7.0 6.76 6.25	.423	.541	•539 •539 •540E •540E	.113	.116 .116 .117 .117	hegr h <sup>*</sup> 001; ah003;
8.24 8.75 9.0 9.2 9.5	5.76 5.25 5 4.8 4.5	.423	.541	.541E .541E .542 .542 .542	.118	.118 .118 .119 .119 .119	Endon 51 Fosifiori to Along 71 Flume 71
9.75 10.C 10.5	4.25 4 3.5	.423	.541	.542 .542 .542	.118	.119 .119	.047 Figine
11.0 12.0 13.0	3 2 1	.423	.541	.542 .542 .542	.118	.119 .119 .119	
14	0	.423	.541	.542	.118	.119	

	a de la companya de l		<u> </u>		- 14	. 16	
ETAT N	AL UNC AL UNC AL UNC	FLOM!	WALR WALR	S FFAE	NORMAL	DE MOLEI N	
Pagapar utilizations as a contract		.3		-	÷ ÷		fann ann an maranna
4 6.0 6.5 6.75	10.0 8.0 7.5 7.25	.423 .423	.541 .541	.541 .541 .540 .540	.118 .118	.119 .118 .117 .117	$S_{1} = 0$ $Q_{1} = .015$ = 1.7.7"
7.0 7.25 7.5 7.8	7 6.75 6.5 6.2	.423	.541	.540 .540 .539 .539	.118	.117 .117 .116 .116	- 9.7"
0.8 8.8	6 5.2	.1423	.541	.539	.118	.116	lifen 1 Heigri
9.0 9.3 9.5	5 4.7 4.5	.123	.541	. 543 . 543 . 543	.118	.120 .120 .120	h <sup>*</sup> 0021 cit: .0041
10.0	43	.423	.541	.543 .543	.113	.120	Pustion 5.21
12.0	2	.423	.541	.543	.118	.120	Along 6.01
13.0 14.0	1 0	.423 .423	.541 .541	•543 •543	.11° .118	.120 .120	EF 5-124 .047

P. 17

		Fr. 1	5 	R. /	1 55 - - - - -		8
4.0 5.0 6.0 7.0 8.0 9.0 10.0 10.10 10.2 10.3 10.4 10.5 10.685	10 9 8 7 6 5 4 3.9 3.6 3.7 3.6 5.5 3.515	.421 .421 .421 .421 .421 .421 .421	.495 .495 .495 .495 .495 .495 .495 .495	.495 .495 .495 .495 .495 .495 .495 .494 .494	.C74 .O74 .O74 .O74 .O74 .O74 .O74	.074 .074 .074 .074 .074 .074 .074 .074	S C 2Cl7 7.7" 9.7" 3/8" .OOl1 .OOL1
11.0 11.485 11.57 11.6 11.65	3 2.515 2.43 2.4 2.35		.495	•494 •495 •495	.074	.073 .074 .074	F. 3.251
12.0 12.5 13.0 13.5	2 1.5 1	.421	• 495 • 495	.496 .496 .496	.074	.075 .075 .075 .075	· <del>34</del>
14.0	0	.421	.495	. 196	.074	.075	

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	7.4	4-1		1	P	18	19
STATE N	DISTANI A CNE FLUM	FLUME	MATER S	NURF-ADE MORT	NU- MAL		
alaa ja aa aha ti'aa ahartii aa ahaadhaadhaa.	2	2			E		1
4.C 5.0 6.0 7.0 8.C 9.0 10.0 10.1 10.2 10.3 10.6	10 9 8 7 6 5 4 3.9 3.8 3.7 3.4	.1,21	.512 .512 .512 .512 .512 .512 .512 .512	.512 .512 .512 .512 .512 .512 .512 .512	.091 .091 .091 .091 .091 .091	.091 .091 .091 .091 .091 .091 .091 .090 .039 .088 .087	$S_{r} = 0$ $G_{rr} \cdot 028$ $= 0  7.7"$ $R = 1.4"$ $= 9.7"$ $3/8"$ $= 3/8"$ $r$ $= 10005'$
10.77 11.0 11.55 11.7 11.8 11.9 12.0 12.5 13.0	3.23 3 2.45 2.3 2.2 2.1 2 1		.512 .512 .512	.514 .515 .516 .517 .517 .517	.091 .091 .091	.093 .094 .095 .096 .096 .096	-0091 Proving 10451 Aure 3.231 Fine 3.231
13.5 14.0	0	.421	.512	.517	.091	.096 .096	

TABLE 4-1 WATER SURFACE MEASUREMENTS P. 19

	DISTANCE	CE POINT GAGE READING				DEPTH	
CTATION	DISTANCE	EL LIBAT	WATER S	URFACE	NORMAL	MODEL	
STATION	ALONG	CONTENT TOWN	WITHOUT	MODEL	DEPTH	IN	
	FLUME	BOLLOW	MODEL	IN PLACE		PLACE	
!	2	3	4	5	6	7	8
4.0 5.0 6.0	10 9 8	.423	.571 .571 .571	.571 .571	.143 .148 .148	.148 .148	So= 0 Q <sub>17</sub> = .031 Dic= 4"
5.5 7.0 7.1 7.3 7.5 7.8 8.0	7.5 7 6.9 6.7 6.5 6.2 6		.571	.571 .570 .569 .568 .568 .567	.148 .148	.148 .147 .146 .145 .145 .144	Rise 2" L= 24" Z- 6/8" Weir Height =
8.2 8.3 8.6 8.7 8.9 9.0 10.0	5.8 5.7 5.4 5.3 5.1 5		. 57]	•565 •565 •565 •565 •565 •565	.148	.142 .142 .142 .142 .142 .142 .142	h <sup>2</sup> : .005 ah = .011 Sudde = 3 Position to Alung 5 Flume 5
11.0 11.1 11.2 11.3 11.4 12.0 12.5 13.0	3 2.9 2.8 2.7 2.6 2 1.5 1		• 7   2	.572 .573 .574 .576 .576 .576 .576 .576	.148	.149 .150 .151 .152 .153 .153 .153	EDG-M.F.S. FIG NO <del>37.</del>
13.0 13.5 14.0	1 0	.423	.571	.576 .576 .576	.143	.153 .153 .153	

TABLE 4 - I WATTER STREAT MELS REMEATS 20

	and the second s				4	_0	
	DISTANCE -	POIr	T SAUE :	FL.D.NG		DEPTH	
STATION	ALONG	E. L.ME	WATER S	SURFACE	NORMAL	MODEL	Contract of the second s
	FLUME	EDL TOM	WITHOUT	MODEL	DEPTH	IN	
			MULTEL	IN PLACE		FLACE	
	2	3	4	5	6	7	8
4.0 5.0	10 . 9	.421 .421	.611 .611	.611 .611	.190 .190	.190 .190	So= 0 Qn: .0519
7.0 8.0 8.8 9.0	7.0 6.0 5.2 5.0	.421	.611 .611	.611 .611 .611 .611	.190 .190 .190	.190 .190 .190 .190	Dig = 5" Rise = 2.5" = 24"
9.1 9.3 9.5 9.532 10.0 10.5	4.9 4.7 4.5 4.468 4.0 3.5	.421	.611	.610 .609 .607 .606	.190	.189 .188 .187 .186	2- 6/8" Weir Haight <sup>±</sup> h <sup>*</sup> 006: Ah= .011!
11.0 11.5 11.532 11.6 11.7 12.0 12.5 13.0	3.0 2.5 2.468 2.4 2.3 2.0 1.5 1.0	.421	.611	.612 .615 .616 .617 .617 .617	.190	.191 .194 .195 .196 .196 .196	Pridue 2.462 Positio to Along 4.468 Flume 4.468 F.PM.P.
13.5	0	.421	.611	.617 .617	.190	.196 .196	

Ρ.	2	I	

	ee française e The Berland					r •• •• •		
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		E SS		w li		Dett		
·	m	-			1 .		7	
4.0		10 9.5	.422	.531	.531	.109	.109	000003 00113
5.0		9	.422	.531	.531	.109	.109	10
2•2 6.0 6.5		8 7.5	.423	.532	.532	.109	.109	2" ∟ 24"
7.0 7.1 7.2 7.3 7.4 7.5 7.93 8.19 8.70 9.20		7 6.9 6.8 6.7 6.6 6.5 6.02 5.31 5.30 4.20	.123	.532	.532 .531 .531 .531 .530 .530E .530E .530E .531E .532E .532	.109	.109 .108 .108 .108 .108 .107	1" .002 .005 4.5' to 6.5'
9.5 9.5 9.7 9.3 9.9		4.5 4.4 4.3 4.2 4.1		• 534	•534 •534 •535 •535 •53		.112	.042 <del>22</del> -
10.0 10.5 11.0 11.5 12.0		4 3.5 3	.127	• 533	•535 •535 •535 •535 •535 •535	.109	.112	1
12.5 13.0 13.5		1	.1,25	.534	. 534	.109	.109	
14.0		0	.425	.534	.534	.109	.109	

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STATI - N	LISEANUR ALCNG FLUNS	FLUV.	MATE 9	MIFICE - MIDFL N.FLACE	DEPT.		
		1 3			ő	و ۱ ۱	8
4.0 4.1 4.2 4.3 4.4 4.5 4.68 4.93 5.0 5.3 5.4 5.5	10 9.9 9.8 9.7 9.5 9.5 9.32 9 8.7 8.6 8.5	.422	.538	.538 .537 .537 .536 .535 .535 .536 .537E .537E .539 .539 .539	.116 .116 .116	.116 .115 .115 .114 .113 .114 .113 .114 .115 .117 .117	S0003 .0138 7.7" 1.4" 9.7" 1" 5 <sup>*</sup> 001 .004
5.6 5.7 5.8 5.9	8.4 8.3 8.2 8.1			•539		.117	Briane - 8.71 P situx to P .72 P .51
6.0	8	.423		.540		.117	I and an
7.0	7 6.5	.423	.539	. 540	.116	.117	.042
8.0 8.5 9.0 9.5 10.5 11.0 12.0	6 5.5 5 4.5 3.5 3	.424	.540	.540 .541 .541 .541 .541 .541 .541	.116	.117 .117 .117 .117 .117 .117 .117	
13.0 14.0	1.5 1 0	.425 .425	.541	.542	.116	.117 .117	

	r = 9	P	23	
ALE S ATER FIL	NF-XCE Mr Chil IN FLALE	NUDIALL CELTH	NEPT MOLTL A FLACE	
4	-	Ö	7	
.531	.531 .531 .530 .530 .529 .529	.111 .111	.111 .111 .110 .110 .109 .109	Sc= .0003 Grf .C114 ^ - 4" H-6-2" - 24"
E 0 3				1

	2	3	4		0	7	
4 4.5 4.6 4.7 4.8	10 9.5 9.4 9.3 9.2	.420	.531	.531 .531 .530 .530 .530	.111 .111	.111 .111 .110 .110 .109	Sc= .0003 Grif .C114 ^ - 4" High= 2"
L.9 5.0 5.76 6.0	9.1 9 8.24 8	.420	.531 .531	.529 .529 .530E	.111	.109 .109 .110	241 21 1" Weir Height
6.28 6.78 7.0 7.1	7.78 7.22 7 6.9		.531	.530E .530E .531 .531	.111	.110 .110 .110 .111	ah .0011 Bridde 71
7.3 7.4 7.5 7.6 7.7	6.5 6.3	.421	•532	•531 •532 •533 •533 •533	.111	.111 .112 .112 .112 .112 .112	Along 91 Flume 91
7.8 7.9 8.0 9.0 10.0 11.0	6.2 6.1 6 5 4 3	.422	•532 •532 •532 •533	.533 .533 .533 .532 .532 .532 .533	.111 .111 .111 .111	.112 .112 .112 .112 .111 .111 .111	F , "¥1
12.0 13.0 14.0	2 1 C	.423	•533 •533 •534	•533 •533 •534	.111 .111 .111	.111 .111 .111	

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	r= = · · · · ·	- G-1 Varia		Est a	EVE JES	P. 24	
1	DISTANT	F bur	T VISE P	5 717 3		orp	
STATION	1 1' i Alla	IC N.	NATIP S	SURFACE	NORMUL	MODEL	
1	IN LUIND	1 C _ 200	NTHNIT	MUDEL	NOT.	IN.	1
1	FLOME	N. TUN	M DEVEL	IN FLACE		ALL	
heater and the mean of the second	· · · · · · · · · · · · · · · · · · ·	, 2					
) at charmen		l su'			0		6
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4	10	.420	.531	.531	.111	.)11	.0003
4.25	9.75			.531	.111	.)11	· 277 • 0114
4. C	9.4			.531	.111	.133	Dias 5#
4.1	7.2			.530 1	.113	.110	R.SP 2.5"
4.0	0 T • 4			.530	.11	.110	L= 24"
5.0	9.0	120	E 2 7	.530		.110	
5.22	3.70	• 460	لارز ه	• 220 L	•111	.110	- 1º
6.0	8.0 1		531	1 20(0.	111	.110	Hacht
6.22	7.79		• <i>) )</i> <b>⊥</b>	5300	• + + +	110	at the second se
7.0	7.0		. 53]	.529	111	100	F"= .0011
7.1 4	6.9		• > > =	.530	1	110	a' .0021
7.2	6.8			. 531	1	.111	
7.3	6.7	1		.532	1	.112	Signal A
7.4	5.6			.532			i na ot
7.5	6.5	.420	.531	.53?	.111	.112	Furne 7
7.6	6.4			.532			an anna i
7.7	6.3			.532		-	.045
7.5	5.2			.532			512 N
6.9	0.1	1.00	500	.532		1	
0.0	D F	.421	.532	.532	.111	.111	
10.0	2		.532	.532			
17.0	4	100	· 532	•532			
12.0	2	+ LL 4' 4'	+	• > > > >	•111	111.	
13.0	ĩ	1		• 222	122	• L [ ] _ [	
14.0	Ō	.1.23	.531	521.	.111	• ± ± ± 1	
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4.0 4.1 4.2	10 9.9 9.8 9.7	.422	.532	.532 .531 .531	.110	.110 .109 .109	.0C03 .0114 7.7 "
4.4	9.6 9.5			.531 .529	.110	.109 .107	1.4" 9.7"
4.68	9.32	100	532	•530E •532E	110	.108 .110	1"
5.3 5.4 5.5	8.7 8.6 8.5	• 4 <i>2</i> ~	• ))2	•532 •533 •534	.110	.110 .111 .111	.001 .005
5.3 5.9	8.2 8.1			. 794	.110		8.71 to
6.0	8.0 7.5	.423	500	.534 .534	110	.111	9.51
7.5 8.0 8.5	7.0 6.5 6.0 5.5	.423	•>>>	•534 •534 •534 •534	.110	.111 .111 .111	.042 <del>-27:</del> 1
9.0 9.5 10.5	5.0 4.5			•534 •534 •535	.110	.111	
11.0	3.0 2.0	.424	• 534	•535 •535	.110	.111	
14.0	0	.425	•535	•535	.110	.110	

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ABLE 4-IV.	TEL	S 1 5 1	1	"TEME"	5 P.26
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alternative terms and state of the	and the second second of the second sec	and a support of the second se	the state of the second s	And a second second second second			
	DISTANCE	E C III	T GAUL H	CHUNG		DEFTH	
STATION	DI CNG	ELUME	WATER S	URFACE	NORMAL	MODEL	
0151.0 1		COTTON	WITHOUT	MODEL	DEPTH	IN	
	TLUME	EOL OW	MCDEL	I'N PLACE		PLACE	1
1	2	3	ý.	F	6	Ĩ	Fi.
4 4.1 4.2 4.3 4.4 4.5	1C 9.9 9.3 9.7 9.6 9.5	.420	.531	.531 .531 .531 .530 .530 .530	.111 .111 .111 .111	.111 .111 .110 .110 .110	So0003 Q <sub>p</sub> : .0114 Dia: 7.7" Bise- 1.4" Lat 9.7"
5.0 5.3 5.4 5.5 5.6 5.7 5.8 6.0 6.5 7.0	9 8.7 8.6 8.5 8.4 8.2 8 8.2 8 7.5 7		.531 .531	.531 .532 .532 .532 .532 .532 .532 .532 .532	.111 .111 .111 .111 .111 .111 .111 .11	.111 .111 .112 .112 .112 .112 .112 .112	2 1" Weir Haght A <sup>X</sup> = .0011 Ats .0021 Scoon 2.71 Position to Along 9.51
7.5 8.0 9.0	6	.420 .421	.531 .532	.532 .532 .532	.111 .111 .111	.112 .111 .111	EF MARCO - 045 File 101
11.0 12.0 13.0	3 2 1	.422	• 533	•533 •533 •533	.111 .111	.111 .111	
14.0	0	.423	.534	.534	.111	.111	

		4-1	2		P P	2.27	21
r ۲ ف					с. Р.		
			-		ar tanan ka u	7	
4.0	10	.422	.534	•534	.112	.112	COO3 .0115
5.0	9		.534	•534	.112	.112	5"
6.7 6.8 6.9	8 7.2 7.1	.423	.534 .534 .534	• 534 • 534 • 534	.112	.112 .112	24" 1"
7.0 7.1 7.2 7.3	6.9 6.8 6.7	•42)	• >>>	• 534 • 533 • 533	.112	.111 .110 .110	· .001 · .006
7.5 7.12 8.13 8.50	6.5 6.12 5.87	.423	• 535	.535 .531 .531E .530E	.112	.108 .108 .107	4.5 to 6.5
8.63 9.0	)• ,			.532E .534E		.109	.042
9.5 9.6 9.7	4.5 4.4 7.3	.423	• 535	•536 •537	.112	.113 .113	-21-
9.8 9.9	4.2 4.1			•537 •537	.112	.113	
10.0 10.5	4 3.5	.424	.536	•537 •537	.112	.113	
11.0 11.5 12.0	3 2.5 2	. 425	•537	•537 •537 •537	.112	.112	
12.5 13.0	1.5 1		.537	•537 •537	.112	.112	
14.0	0	.425	.537	.537	.112	.112	

P.28

							( ~ 1
4.0 4.5 5.5 5.6 5.7 5.9 6.1 6.3	10 9.5 9 8.5 8.4 8.3 8.2 8.1 8 7.9 7.8 7.7	.422	.541 .541 .541 .541 .541	.541 .541 .541 .540 .540 .540 .540 .540 .540 .540 .540	.119 .119	.119 .119 .119 .119 .118 .118 .118 .118	.0003 .015 6" 2" 24" 1"
6.4 6.5 7.00 7.22 7.5 7.735 8.0 8.5 8.6 8.7 8.8 8.8	7.7 7.6 7.5 7 6.78 6.265 6 5.5 5.4 5.3 5.2	.423	. 542	.538 .537 .536 .536E .537E .539E .541 .542 .543 .543	.119	.116 .116 .116 .116 .118 .119 .120 .120	.008 5.5' to Flume 7.5' .042 .042
9.0 9.5 10.0 10.5 11.0 11.5 12.0 12.5 13.0	5.1 5 3.5 4 3.5 3 2.5 2 1.5 1	.424	•543	.543 .544 .544 .544 .544 .544 .544 .544	.119	.120 .120 .121 .121 .121 .120 .120 .120	
13.5	0.5	.425 .425	• 544	•545 •545	.119	.120 .120	

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A.

TADE 4-1 LITER HAR BERLAS P.29

	LUCZAN E	PIK	TLAS. I	127000		05P	
STATION	ALONG	FLUME	WATER S	SURFACE	14http://AL	MOUT	
	FLUME	BOTT M	MOREL	N'A HOE	0 E 1 1 2		
	2	3	-4	F	6	7	n - raatoo - waxaa ahaadaa waxaa
4.0 4.1 4.2 4.3 4.4 4.5 5.30	10 9.9 9.8 9.7 9.6 9.5 8.70	.420	• 534	.534 .533 .533 .532 .532 .532 .532 .532	.114	.114 .113 .113 .112 .112 .112 .112	.0003 .0130 .4" .2" .24" .24" 
5.81 6.0 6.32 6.50 5.6 6.8 6.9 7.0 7.1 7.2 7.3 7.4 7.5 7.6	3.19 8 7.68 7.5 7.4 7.2 7.1 7 6.9 6.2 6.7 6.5 6.4	.421	.534 .534 .534	.532E .532 .532 .533 .533 .533 .534 .535 .535 .535 .535	.114	.112 .112 .112 .112 .113 .113 .113 .114 .115 .115 .115 .115 .115 .115	.001 .003 .003 .003 .005 .045 .045
8.0 18.5 9.0 10 11.0 12.0 13.0 14	6 5.5 4 3 2 1 0	.422 .422 .422 .422	.535 .536 .536 .536 .526 .536 .537 .537	.536 .536 .536 .536 .536 .536 .536 .536	.114 .114 .114 .114 .114 .114 .114 .114	.115 .114 .114 .114 .114 .114 .114 .114	

P. 30

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carte					a contra		,		e di secon di disensi di secondo d
4.0 5.0 5.7 5.8 5.9 6.0 6.1 6.2 6.3	10 9 8.3 8.2 8.1 3.0 7.9 7.8 7.7		422	•533 •533	.539 .538 .538 .537 .537 .537 .537 .537 .537 .537	.116 .116 .116	.116 .116 .116 .115 .115 .115 .115 .115		.0003 .0138 4" 2" 24" 1"
6.4 6.5 7.0 7.22 7.73	7.6 7.5 7 6.78 6.37	•	422 423	•539	.536 .535 .534 .535E .536E	.116	.117 .114 .113 .112	1700	.002 .007 5.5'
8.25 8.5 8.6 8.7 8.8	5.75 5.5 5.4 5.3				•537£ •53₿ •539 •540		.115 .117 .117		to 7.5'
8.9 9.0 9.5 10.0	5.1 5 4.5 4	• L • 2	23		.541 .541 .541 .541 .541 .541		.118 .118 .118 .118 .118		-042 - <del>24</del> -
10.5 11.0 11.5 12.0	3.5 3 2.5 2	.1	24	.540	.541 .541 .541	.116	.117 .117 .117		
12,5 13.0 13.5	1	- 4	.25	•540 •540 •541	.541 .541 .541	.116	.117 .117 .116		
14.0	0	.4	25	. 541		116	116		

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4-1

TABLE 4-1 WATER SURFACE MEASUREMENTS P. 31

	DISTANCE	POIN	IT GAGE R	EADING		DEPTH	
IST VTION	ALONG	FLUME	WATER S	URFACE	NORMAL	MODEL	
	FLUME	BOTTOM	WITHOUT MODEL	MODEL	DEPTH	IN PLACE	
1	Ê.	3	4	5	6	7	8
4.0 5.0 6.0	10 9 8 7.9	.422	.538	• 538 • 538 • 538	.116 .116 .116	.116 .116 .116	S <sub>0</sub> = .0003 Q <sub>mi</sub> .0138 Dia= 5"
6.2 6.3 6.4 6.5 6.74 7.0 7.24 8.24 8.5 8.5 8.5 8.5 8.7 8.3 9.0 9.5 10.0	7.3 7.7 7.6 7.5 7.26 7 6.76 5.76 5.5 5.4 5.5 5.4 5.3 5.2 5	.423 .424	•539	.537 .536 .534 .534 .535E .537E .539E .540 .540 .540 .540 .540 .540 .540 .540	.116	.115 .114 .113 .113 .113 .117 .117 .117 .117 .117	Rise= 2.5" L= 24" Z= 1" Wur = h <sup>*</sup> = .001 sh1 .007 B
11.0	3	.424	.540	.541 .541	.116	.117	F16 40 #
12.0	2	125	.540	.541	114	.117	
13.0 13.5	1	• 440	• 541	• 541	.115	.110	
14.0	0	.425	.541		.116	.116	

TABLE 4 - I WATER SURFACE MEASUREMENTE P. 32

	DISTANCE	POIN	IT GAGE R	EADING		DEPTH	
STATION	ALONG	FLUME	WATER S	URFACE	NORMAL	MODEL	
	FLUME	BOTTOM	MODEL	MODEL IN PLACE	DEPTH	PLACE	
	2	3	4	5	6	7	8
4.0 4.10 4.20 4.30 4.40	10 9.9 9.3 9.7 9.6	.420	•535	• 535 • 534 • 534 • 534 • 534 • 534	.115	.115 .114 .114 .114 .114	So= .0003 Qrrf .0138 Dia = 5" Rise = 2.5"
4.50 4.72 5.23 5.5 6.22	9.5 9.28 8.77 8.50 7.78		• 535	.534 .534E .534E .534E	.115	.114 .114 .114 .114	Z- 1" Weir : Height
6.5 6.6 6.7 7.0 7.1 7.2 7.3	7.5 7.0 7.3 7.0 6.9 6.3 6.7	.420	•535	.535 .536 .537 .537 .537 .537 .537	.115	.115 .116 .117 .117 .117 .117 .117	n <sup>112</sup> .0021 <u>Ah = .0051</u> Bridge <u>-</u> 7.51 Position to Along Flume 9.51 E Pi3-M E J
7.5 7.6 7.75 8.0 9.0	6.5 6.40 6.25 6 5	.421 .421 .422	.536 .536 .536 .537 .537	•538 •538 •538 •539 •539	.115 .115 .115 .115	.117 .117 .117 .117 .117	.045 Fig Na
11.0 12.0 13.0 14.0	3 2 1 0	.422 .423 .423	•537 •538 •538	•539 •539 •539	.115 .115 .115	.117 .116 .116	

2	0	6
ے ا	U	0

TABLE 4 -1 WATER SURFACE ME IN FONT P. 33

	10-211		TEN OIL A				
	DIGTANCI	P0".	T CAGE RI	EADING		DEFTH	
arar 1.	TOTSTALVOL TOTSTALVOL	TT SHAT	WATER S	URFACE	NORMAL	MODEL	
OTAL ON	FALLEN J	FLURE	WITHOUT	MODEL	DEPTH	IN	
	- L. L. 1972	p. 211 OW	MODEL	IN FLACE			
ngis-uga aja gindinis tu Li even	2	3	4	5	6	7	B
4.0	10	.422	.541	.541	.119	.119	Sc: .0003
5.0	9	1.22	617	617	110	110	Qm .015
2.2	8.7	.422	.541	• 541	.119	.119	Dia= 5"
5.9	8.1	1 1		.540	i :	.118	Hire 23"
6.0	8			.540		.118	_ 24 <sup>¶</sup>
6.1	7.9			.540		.118	
6.2	7.8			• 539		.117	12- 1"
6.3	7.7			.538		.116	He 1.
6.5	7.0	1.22	512	• > > > ( 5 2 7	110	.115	122 000
6.7!	(•)	•462	• 74~	-536E	• • • • •	.113	.002
7.26				.537E		.116	1411007 1
7.5				1			0 1418 - 5.51
7.73				.538E	1	.115	turna to
8.24	5.76			- 539E	Í	.116	Turk 7.5'
8.5	2+2			. 540		.110	125 g 13.
8.7	5.3			.543		.110	.042
8.8	5.2			.542		.119	Ar
8.9	5.1			.542		.119	· · · · · · · ·
9.0	5.0			.543	.119	.120	
9.5				.543		.120	
10.0	4.0			• 244		.121	
11.0	3.0	.1.21	5/13	- 54.5	.119	.121	
11.5	240			.545	/	.121	
12.0	2.0			.545		.121	
12.5				.545		.121	8
13.0	1.0		511	• 545		.121	
11.0	0	1.25	• 244	• 242	110	121	
14.0	U	.42)	• 744	• 740	• 1 1 7	+141	

TABLE 4-1 WATER SUMFACE IN AS REVENTS P.34

DISTANCE POINT GAUL CONTACT DEPTH	
STATION ALONG FLUME WATER SURFACE NODEL	
FLUME BOTTOM MODEL IN PLACE PLACE	
1 2 3 4 5 6 7 8	
1         2         3         4         5         6         7         6           4.0         10	03 7 "" \$3" 01' 05' .45' .45' .45' .45'
ABLE 4-14 JER SURFACE MEAS TEME TO P. 35

	14.00		ELCODE EL				
	DISTANCE	PUIN	II GADE M	E - UNK	NORMAL	DEFTH	
STATION	ALONG	FLUME	WATER S	URFLICE	NORMAL_	MODEL	
	ELINAE	ROTTUM	WITHOUT	MODE'	DEPTH	IN DI ATT	
	FLUME	DOLLOW	MODEL	IN FLACE		PLASE	
ł	2	3	4	Ę	6	7	8
37	10.3	120	555	EEE	125	125	Se= 0003
3.8	10.2	•4×0	• ) ) )	• 222	•1))	.137	023
3.9	10.1			553	tr ab	122	
4.0	10	.420	.555	. 552	.135	.122	Dig= 4"
4.1	9.9		• / / /	• > > ~	**//	•1)2	Rise - 2"
4.2	9.8			.552		.132	L- 24"
4.3	9.7			.552	Concentration of the second seco	.132	7 ( / 4
5.0	9					-	12" 6/8"
6.0	8			t d			Height
6.3	7.7			.554		.134	L×- 000.
6.4	7.6			•555		.135	n0021
0.5	1.5			.550		.136	ah= .005!
6.75	1.4			• 557		.136	Bridge -
7.0	7	127	556	• > > /	125	.130	Position
7.5	1	044K.1	• • • • •	.557	•1))	.136	I Along Elume
8.0	6.0			.557		.136	EDG HAGY
8.5				.558		.137	E PUTAPG-
9.0	5.0			.558		.137	
10.0	4.0	.422	.557	. 559	.135	.137	FIG AR
11.0	3.0						
11.5	2.5			.559		.137	
12.0	2.0			• > > 9		.137	
13.7	1.0	1.22	550		125	127	
14.0	0	1.23	.558	560	135	137	
14.0	Ŭ		• ) ) 0	. )00	• • • • • •	1620	
					4		

TABLE 4-1 WATER SURFACE N. 45 REM VTS P. 36

	DISTANCE	PDR	T GAGE R	E .DING		DEPT -	
STATION	ALONG	FLUME	WATER S	USFACE	NORMAL	MODEL	
	FLUME	BUTTOM	MODEL	MCD9L H4 PLACE	DEPTH	F. ACE	
	2	3	4	5	6	7	8
4.0 4.1 4.3 4.5 4.6 4.7 5.0	10 9.9 9.8 9.5 9.4 9.3 9	.420	.561	.561 .550 .559 .559 .558 .558	.141	.141 .140 .139 .139 .138 .138	S <sub>0</sub> = .0003 G <sub>rtf</sub> .028 Dia= 5" Rise- 2.5" L= 24"
4.7 5.0 5.5 6.0 6.5 6.7 6.8 6.9 7.0 7.25 7.50 8.0 8.0 8.5 9.0 10.5 11.0 11.5 12.0 13.5 14.0	9.3 9 8 7.5 7.3 7.2 7.1 7 6 5 4 3 2 1 0	.421 .422 .422 .423 .423	.562 .563 .564 .564 .564	.558 .558 .559 .560 .561 .562 .563 .564 .564 .564 .564 .564 .564 .564 .564	.141 .141 .141 .141 .141	.138 .139 .140 .141 .141 .142 .143 .143 .143 .143 .143 .143 .143 .144 .141 .141	L= 24" Z= 6/8" Weir Harght R*= .002! Ah= .006! Pridre = 7.3! Position to Atong 9.3! E Pri-th Pur FIG NO

					the day of the second s		
	DISTANCE	POIN	IT GAGE R	- LING	NORMA	DEPTH	
STATION	ALONG	FLUME	WATER S	UPFACE	TROUMPAL.	MODEL	
	FLUME	BOTTOM	MODEL	MODEL IN PLACE	DEPTH	PLACE	
	2	3	4	E.	6	7	8
4.0 4.20 4.3 4.4 4.6 4.70 5.0 5.5	10 9.3 9.7 9.5 9.4 9.3 9	.420	.564	.564 .563 .563 .562 .561 .561	.144 .144	.144 .143 .143 .142 .141 .141	Sof .0003 Qrf .031 Lat 5" Risef 2.5" L= 24" Z 6/8"
5.5 6.0 6.5 6.70 6.3 6.9 7.0 7.1 7.2 7.3 7.5 8.0	8.7 8 7.5 7.3 7.2 7.1 7 6.9 6.3 6.7 6.6	.421	.565	.563 .564 .564 .566 .566 .566 .566 .566	.144	.143 .143 .143 .143 .145 .145 .145 .145 .145	HegH <sup>2</sup> M <sup>3</sup> = .001 <sup>1</sup> A <sup>2</sup> : .005 <sup>1</sup> Bridge = 7.3 <sup>1</sup> Position to Along 9.3 <sup>1</sup> FIDG-M.40-
8.5 9.0 9.5 10.0 10.5 11.0 11.5 12.0 12.5	5.5 5 4.5 4 3.5 3 2.5 2 1.5	.422	.566 .567 .567	.566 .567 .567 .567 .567 .567 .567 .567	.144	.145 .145 .145 .145 .145 .145 .145 .145	FIG NC
13.0 13.5	1	4	.557	.567		.145	
14.0	0	.423	.567	.567	.144	.144	

POINT GAGE READING DEPTH DISTANCE NORMAL WATER SURFACE MODEL STATION ALONG FLUME WITHOUT MODEL IN FLUME BOTTOM MUTEL IN PLACE 5 3 4 7 6 8 10 4.0 .420 .530 .530 .110 .110 So= .0005 9.9 .529 4.1 .109 Q<sub>rrf</sub> .0114 9.8 4.2 .529 .109 Dia- 7.7" 9.7 4.3 .529 .109 Rise = 1.4" 4.4 9.6 .529 .109 4.5 9.5 .421 .531 .531 .110 L = 9.7" .110 5.0 9 .421 .531 .110 8.7 Z-5.3 .531 .110 1 11 Neir Haight = 5.4 8.6 .531 .110 5.5 8.5 .532 .111 5.6 8.4 .532 .111 h<sup>×</sup>= .0011 5.7 8.3 .532 .111 Δh= .0031 5.8 8.2 5.9 8.1 .533 Bridge \_ 8.71 .111 Position to 6.0 8 .422 .532 .533 .110 .1)) 6.5 7.5 .532 .110 9.51 .532 .110 Flume 7.0 7 .422 .532 .532 .110 .110 8.0 6 .532 .532 5 9.0 .423 .533 .533 .110 .110 10.0 4 .423 .533 .533 .110 .110 3 11.0 .424 .534 .534 .110 .110 2 .534 12.0 .424 .534 .110 .110 1 .534 13.0 .424 .534 .110 .110 14.0 0 .425 .535 .535 .110 .110

TABLE 4-1 WATER SURFAUE MEASUREMENTS P. 38

10 - 10 - 10 - 10 - 10		- 4 - 1 V		L V S	NC	P. 39	21
		12.3	1241 = 11	-A7141		DEPTH	
	Die 194	101 Test	11123	N. 5	THE ACT	, MODEL	4
	5. 1. 7	1	V 1 1 1		DETH	N	
I grant the control of the second	·					C. L. F. O'Lea.	nan et Kall directe sageber (Bibliote, de 1999) au e part de
1 happenen um ummun ut	-	2. 	4.4 		) Jan mananan unun mitta man	·	
4.0	10	.420	.528	. 528	.108	.108	Se= .0005
4.1	9.9		.,	.527	1200	.107	.0115
4.2	9.8			.527		.107	Digital
4.3	9.7			.527	e ê	.107	3.58.21
4.4	9.6			.527		.107	- 211
4.5	9.5		.528	.527	.108	.107	- 24"
5.0	9.00		.528	1000	.108	2.05	
5.21	8.79			.527E		.107	
2.7	8.7			5070		104	nengi "
5.12	0.20	1.21	520	• ) ~ ( E	100	.100	1 A 002
6.22	C.V	•4KI	• 247	526F	.100	105	.002
6.5	7.5		529	. 528	1.08	.107	.005
6.6	7.1		• /~ /	.528	.108	.107	1 - e = 7.51
6.7	7.3			.528	1200	.107	to to
6.8	7.2			.527		.108	15 9.5'
6.9	7.1			.530		.109	G. 115.12
7.0	7.0	.421	.529	.531	.108	.110	010
7.1	6.9			.531		ŝ	.040
7.2	6.8			.532			0 X X0
7.3	6.7			.532	.108		1
7.4	6.6			.532	.108		
7.5	0.5	1.00	520	.532	301.	.110	
8.0	0.0	• 4 2 2	.530	.532	.108	.110	
0.7	2.2	122	530	530	108	1100	
9.5	1.5	+4KZ	- 531	532	.100		
10.0	4.0	1.23	. 531	.532	.108	109	
10.5	3.5	•+~;	• / / ±	.532	1100		
11.0	3.0	.423	.531	.531	.108	.108	
11.5	2.5						
12.0	2.0	.426	.532	.532	.108	.108	
12.5	1.5						
13.0	1.0	.424	.532	.532	.108	.108	
13.5	•5						
14.0	0	.425	•533	.532	.108	.1.08	

	(MW)	win YVM	TEN JUNI	The Is LANDUN	CTICIVIO .		a the second states where a state of the second states
	DISTANCE	POIN	JT GAGE R	EADING	ton 1	DEPTH	All and a start of the
STATION	ALONG	CI LIBAC	WATER S	SURFACE	NORMAL	MODEL	Other the A. H. The
JIA ION		DOTTON:	WITHOUT	MODEL	DEPTH.	· IN	and the man while the
	FLUME	BOI TON.	MODEL	IN PLACE.	-	PLACE	
	2	3	4	5	6	7.	martin B. 1 mman
formation and the second secon				<u>.</u>			1 16 27 8X
4	10	.420	.528	.528	.108	.108	So= .0005
4.2	9.8		1	.528		.108	Qm .0115
4.3	9.7			.527		.107	Dia= 5"
4.4	9.6		ander og	.527		.107	Rise= 2.5"
4.5	9.5		.528	.527		.107	L= 21.1
4.75	9.25			.527E		.107	a state of the state of the
5.0	9	.420	.528	.528E	.108		Z-44 1 1
5.25	0 50			.528E		.107	Weir Hant
2.2	8.26	1		. 728E		107 3	HIGIGIN ES
6.0	8	.1.21	. 529	• 275	.108	•10,7	h <sup>2</sup> = .002
6.465	7.76	*****	• /~ /	-528E		.107	ah=
6.5	7.5		.529	.529	.108	.108	Bridge 75
6.6	7.4			. 530	.108	.109	Position to
6.7	7.3			.531	.108	.110	Along 9.5
6.8	7.2			.531	.108	.110	riume
0.9	7.1	101	600	.531	.108	.110	EPG-M.PG
7.0	7.0	.421	• 749	• 231	.108	.110	.040
7.2				- 531			FIG. NO
7.3		1		.531			
7.4				.531			
7.5				.531			
8.0	6.0	.1,22	.530	.531	.108	.109	i
8.5			.530	.531			
9.0	5.0	.422	.530	-531	.108	.109	
10.0	1.0	122	521	523	100	100	
10.5	4.0	• 44.42.)	• • • • • •	• JJI	.100	.100	
11.0	3.0	.423	.531	.531	.108	.108	
11.5						1200	
12.0	2.0	.424	.532	.532	.108	.108	1.
12.5							· 7
13.0	1.0	.424	.532	.532	.108	.108	
13.5	0	1.05	600	500	2.04	100	
14.0	0	.425	. 523	.533	*T08	•108 -	
		1					

# TABLE 4-1 WATER SURFACE MEASUREMENTS P. 36

		· ۵.	-	6-1	-			r =	THENT P.	37 A	21	4
Istanos	1	AL ONLY FL 14	· · ·	7.)- 		10 7 1 10						
n narran na r			baller -	3		2			in and in the second	2	dealer an	
4.0 4.1 4.2 4.3 4.4 4.5		10 9.9 9.8 9.7 9.6 9.5		.420		.534		•534 •533 •533 •533 •532 •531	.116	.114 .113 .113 .113 .113 .112 .111	So <sup>5</sup> .0005 Graf .0137 Soc. 4" Soc. 4"	a second the state of a speculate a detraction of
5.0		9 8,79						.530E		.110	1"	1
5.5 5.72 6.0 6.22 6.5 6.6 6.7 6.8 6.9 7.0 7.2		7.5 8.28 8 7.5 7.4 7.3 7.2 7.1 7		.421		•535	1	.531E .533E .535 .536 .537 .537 .537 .537 .537 .537	.114	.110 .112 .114 .115 .117 .117 .117 .117 .117	.002 .006 .006 .006 .051 .04	the second s
7.5 8.0 8.5 9.0	10	6.5 6.C 5.5 5.0		.1,22 .423		.536		• 537 • 537 • 537 • 538	.114	.117 .115 .115 .115		-
9.5 10.0		4.5		.423		.537		•538 •538	.114	.115	1	
10.5		3.5 3.0		.424		. 538		•538 •539	.114	.115 .115		
12.0		2.0		.424		. 538		. 539	.114	.115		
12.5 13.0 13.5		1.5 1.0 .5		.425		.538		• 539	.114	.115	1	-
14.0		0		.425		.539		.539	.114	.114		-

	16.000	5 4 - L + m	LEN SUFE	+ N/4	FORVER	38 A	215
GY	MISTUMOR AL ANG FL IM-	POIN FLUME EDT TOM	NOTER S	UKF-UE MODEL MODEL	NURMAL DEPTH	DEPTH MODEL IN PLACE	
laiseanaine ann ann ann ann ann ann ann ann ann a	2	3	4			7	8
4.0 4.1 4.2 6.3 6.4 6.5 5.0 5.25 5.74 6.25 6.6 5.5 5.74 6.25 6.6 6.7 6.8 9 7.0 7.25	10 9.9 9.8 9.7 9.6 9.50 9.25 9 8.5 8 7.79 7.5 7.4 7.3 7.2 7.1 7 6.8	.420	. 534 . 534	.534 .533 .533 .533 .533 .533E .532E .533E .533E .534 .534 .534 .534 .534 .534 .535 .536 .536 .536 .536 .536	.114	.114 .114 .113 .113 .113 .113 .113 .113	Sup .0005 Gut 5" Cut 5" Cut 5" Cut 5" Cut 5" Cut 5" Cut 7" Cut
8.0	6	.422	.536	.537	.114	.115	
9.0	5	.423	• 537	.537	.114	.114	i i
9.5 10.0 10.5 11.0	4.2 4 3.5 3	.423	•537	•537 •537	.114	.114	
11.5 12.0 12.5	2.5 2 1.5	.424	.538	. 538	.114	.114	Not the second
13.5	•5	.425	•539	• 539	.114	.114	

	TABL	4-1 VIA	154 2-1-41	LE IVIER H	ENEVISI	. 39	angegel en Ferrenzensen er eine samme mehr besonen er eine
	DISTANCE	POIT	IT GAGE R	EADING	NODIAN	DEPTH	
STATION	ALONG	FLUME	WATER S	SURFACE	NURMAL	MODEL	
	FLUME	BOTTOM	WITHOUT MODEL	MODEL IN FLACE	DEPTH	IN FLACE	
I	2	3	4	5	6	7	8
4.0 4.1 4.2 4.3 4.4	10 9.9 9.3 9.7 9.6	.420	•535	.535 .534 .534 .534 .534 .534	.115	.115 .114 .114 .114 .114	S <sub>0</sub> = .0005 Q <sub>πτ</sub> .0138 Dia = 4" Rise = 2" L = 21/"
4.5 5.0 5.22 5.73 6.0	9.5 9 8.78 8.27 8	.421	•535 •536	.534 .534E .534E	.115	.114 .114 .114	Z 1" Weir Height <sup>a</sup>
6.23 6.5 6.6 6.7 6.8 6.9	7.77 7.5 7.4 7.3 7.2 7.1			•534E •534 •535 •536 •538		.113 .113 .114 .115 .117	Ah= .0031 Bridge = 7.51 Position to Along 9.51 Flume 9.51
7.0 7.1	7 6.9			.538	.115	.117	EPG-M PG= .053
7.2	6.8 5.7 6.6			-			FIG NO
7.5 8.0 8.5	6.5 6 5.5	.421	.536	.538 .538	.115	.117 .117	
9.0 9.5	5 4.5	.422	.537	•538 •538		.116	
10.0	4 3.5	.422	•537	.538 .538	110	.116	
12.0 13.0 14.0	2 1 0	.422 .423 .423 .425	•537 •538 •538 •540	•538 •538 •538 •540	.115 .115 .115 .115	.116 .115 .115 .115	

TABLE 4 - I WATER SURFACE MEAS HEMENTS P.40

		DOU				1.40	
	DISTANCE	POIN	UNDE RI	- LOING	NORMAL	DEPTH	
STATION	ALONG	FLUME	WATER	URFACE	- Continue	MODEL	
	FLUME	BOTTOM	WITHOUT	MODEL	DEPTH	DIACE	
	· come	1301101	MODEL	IN PLACE		PLAUE	
	2	3	4 -	5	6	7	З
						1	
1.0	סי	1.20	535	. 535	115	115	So= .0005
4.0	9.8	14.000	• / / /	535		.115	0 <sub>mf</sub> .0138
4.3	9.7			.534		.111	Dia= 5"
4.4	9.6			.534	Supreme to a	.114	Rise= 2.5"
4.50	9.5			.534		.114	L= 21.11
4.71	9.29			.534E	er - e della de	.114	~ ~ ~ ~ ~ ~ ~ ~ ~ ~ ~ ~ ~ ~ ~ ~ ~ ~ ~ ~
5.0	9		.535		.115		Z l"
5.21	8.79			.534E		.114	ver -
6.0	8	.421	.536		.115		Ineigin
6.21	7.79			-534E		.114	h*= .0011
0.5	1.7			• > 54		.115	Ah= .0021
6.0	73			• JJJ 536		115	
6.8	7 2			537		.116	Position = 7.51
6.9	7.1			.537		1120	Alona
7.0	7.0	.421		.537		.116	Fiume 7•21
7.1	6.9			.537			EP -WES
7.2	6.8			.537			.053
7.3	6.7	.421	.536	.537	.115	.116	E G NG
7.5	6.5		.536	.537			O . K
8.0	6.0		.536	•537	.115	.116	
8.5	5.5	.422	•537	• 537	.115	.115	
9.0	2	.422	• > > /	• > > 1 5 2 8	•115	.115	
10.0	4	.4<>	- 538	538	115	115	
12.0	2	1.21.		539	.115	•117	
13.0	ĩ	0 647-64	• / / /	539	.115	.115	
14.0	ō	.425	. 540	.540	.115	.115	
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4.0		10	Ι.	420	.536	.536	.116	.116		.0005
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5.6 5.7		8.4 8.3			.536	•536 •535	.116	.116		2" 24"
5.8 5.9 6.0		8.2 8.1 8	1	421	.537	•535 •534 •534	.116	.115 .114 .113	1	l"
6.5 6.75		7.5				.534E	r and	.114		.002
7.0 7.27		7 6.73				•534E	9 I	.115		.007
7.72 8.0 8.1 8.2 8.3 8.4 8.5		6.28 6 5.9 5.8 5.7 5.6 5.5		422	.538	.537E .538 .539 .540 .540 .540 .541	.116	.116 .116 .117 .118 .118 .118 .119		to 81
8.7	1	5.3		.423		.541		.118	1	
10.0 10.5 11.0		4.9 3.5 3		423	• 539	.541 .541 .541	.116	.118 .118 .118	Amount	
11.5 12.0 12.5		2.5 2 1.5		424	.540	.541 .541 .541	.116	.117		
13.0 13.5 14.0		1 .5 0		425	.541 .541	.541 .541	.116	.117 .116 .116		

		4-1			P-	42	
37 XTI			4 	F	- L	v	
4 4.5 5.0 5.5 5.6 5.7 5.8 5.9 6.0 6.21 6.5 6.715	10 9.5 9 8.5 8.4 8.3 8.2 8.1 8 7.79 7.5 7.285	.420	.536 .536 .536 .536 .536	.536 .536 .536 .536 .535 .535 .535 .534 .534 .530E .532E	.116 .116 .116 .116 .116	.116 .116 .116 .115 .115 .115 .114 .114 .109 .112	.0005 .015 5" 2.5" 24" 1" .001 .006
7.0 7.435 7.5 8.0 8.1 8.2 8.3 8.4	7 6.565 6.5 6 5.9 5.8 5.7 5.6	<i>.1</i> ,22		•533E •535 •535 •536 •538		.112 .113 .113 .114 .116	6; to 8; .C4
8.5 8.75 9.0	5.5 5.25 5	.423	•538	•539 •540 •540	.116	.117 .117 .117 .117	
10.0 10.5 11.0	4 3.5 3	.423	•539	.540 .541 .541	.116	.117	- contract of the
11.5 12.0 12.5	2.5 2 1.5	.424	.540	.541 .541 .541	.116	.117	
13.C 13.5	1	.425	.541	.541	.116	.116	
14.0	0	.425	.541	.541	.110	.110	

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P. 10	S. S. E. E.	FADNG		TOTA	
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+422	.549	.549	.127	1 121	ADEU LO
		.550		.128	Flume 9.251
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		.551		.129	1 4 14 2
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571	<sup>1</sup> M			2, 1	4	h.	1
1	- 17 <sub>2</sub> 0	TIN	54 -				
and a second sec	n	3			5		E
4.0	10	.421	.562	.562	.141	.141	.0005 .031
5.0 5.5 6.0 6.1 6.3	9 7.9 7.7	.422	.563	.563 .563 .562	.141	.141 .141 .140	5" 5 5 2.5" 1 24" 1 6/9"
6.9 7.0 7.2 8.0	7.2 7.8 6.8	. 423	. 564	.562 .561 .560	.141	.140 .140 .139 .138	140.1 h <sup>1</sup> : .0021 .0C61
9.2 9.3 9.4 9.5	2.8 4.7 4.6 4.5			.565 .566 .566 .567 .566	.141	.142 .143 .143 .144	Clong Fure 6.81
10.0 10.5 10.8	4 3.5 3.2	.424	.565	.566 .566 .566	.141	.142 .142	95 - N
11.0 12.0 13.0 14	3 2 1 0	.425 .425 .426	.566 .566 .567	.566 .566 .566 .567	.141 .141 .141	.142 .141 .141 .141	:
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TATION	ALL NUS	FOT N	W. H. W. H.	NO FROF MCIRI LINIACE	NORMAL LEPTH	MCS- IN FLASE	
	2	7	4	5	6	1	te la
4.0	10	.421	.571	.571 .571	.150	.150	So .0005 Grr .0379
5.5 6.0 6.1 6.2 6.3 6.4 6.5 6.8	8.5 8 7.8 7.7 7.6 7.5 7.2	.422 .422	.572	.572 .572 .572 .572 .571 .570 .569	.150 .150	.150 .150 .150 .149 .148 .147 .147	Che = 5" F set 2.5" L = 24" L = 6/8" Wen = Harr =
7.0 7.2 8 8.5 9.0	7 6.8 5.5 5	.423	•573	.568 .568		.146 .146	Ab0061 Bridge 4.81 Position to Along 6.81
9.2 9.3	4.7			.573 .574	.150	.150 .151	EPG-VPD
9.5 9.75 10.0 11.0 12.0 13.0 14.0	4.5 4.25 4 3 2 1 0	.424 .424 .424 .425 .426 .426	.575 .575 .576 .576	.575 .576 .576 .576 .576 .576	.150 .150 .150 .150	.152 .152 .152 .152 .151 .150 .150	F13 N.
6.8 7.0 7.2 8 9.0 9.2 9.3 9.4 9.5 9.75 10.0 11.0 12.0 13.0 14.0	7.2 7 6.8 6 5.5 5 4.7 4.6 4.7 4.6 4.5 4.25 4 3 2 1 0	.423 .424 .424 .424 .425 .426 .426	.573 .575 .575 .576 .576	.568 .568 .573 .574 .576 .576 .576 .576 .576	.150 .150 .150 .150 .150	.144 .146 .146 .146 .150 .151 .152 .152 .152 .151 .150 .150	Ab . 000 Bridge L Pas fan t Along Elume Elume Elume

TABLE 4-1 WALLIN SURFACE ASUREMENT P.46

		Charles and the second se	L	1. 1. 4.30.11	Line. I	. 46	
	DISTANCE	POIL	1 JAUS 23	L+ UNG	1.00140	DEPTH	
STATION	ALONG	FLUME	WATER S	URFACE	NORMAL	MODEL.	
	FLUME	BOTTOM	MUDEL	MODEL	DEPTH	IN PLACE	
ł	2	3	4	5	6	7	ε
4.0 5.0 6.0 6.6 6.8 7.0 7.2 7.5	10 9 8 7.4 7.2 7 6.8 6.5	.421 .422	.580 .580 .581	.580 .530 .581 .580 .579 .578 .578	.159 .159 .159	.159 .159 .159 .158 .157 .156 .156	S <sub>0</sub> = .0005 Q <sub>nf</sub> .042 Dia= 5" Rise= 2.5" L = 24" Z- 6/8"
8.0 8.5 9.0 9.2 9.225 9.3 9.4 9.5	5 4.8 4.775 4.7 4.6 4.5	.423	.582	.580 .581 .582 .583	.159	.157 .158 .159 .160	Weir Height h <sup>#</sup> = .0021 Ah = .0051 Bridge _ 4.81 Position to Along 6.31
10.0 10.5 11.0 11.5	4 3	. 424	.583	.584 .585 .585 .585	.159	.160 .161 .161 .161	EPG-MPG=
12.0 12.5 12.0	2 1	.425	. 584	.585 .585	.159	.160 .160	
14.0	0	.426	.585	.585	.159	.159	

TABLE 4-1 WATER REPENDENT OF TE P.47

	DOFANCE	POL	T GARE -	ADINE		FPTH	
STATION	DISTANCE	ET LINAC	WATER S	SURFACE	N FRIAL	MODEL	
	FLUME	E DTTOM	WITHOUT	MODEL	DENIN	IN NOT	
			MUDEL	TIN I ACE		T L AVUE	
1	£	3	4		6	7	5
4.0	10	.421	.590	.590	.169	.169	Sc0005
4.1	9.9			.539		.168	Port .0525
4.2	9.7			.589		.100	Dia = 5"
4.4	9.6			. 588		.167	IF.se= 2.5"
4.5	9.5			.587		.166	∟= 24"
4.7	9.2			.505		.164	Z- 6/811
5.0	9			.583		.162	Vieur - Height
6.0	8	.422	.591	580	.169	167	h <sup>×</sup> = 0031
7.1	6.9			.592	* T O /	.170	
7.2	6.8			.592		.170	.01
7.3	6.7	123		593		170	Position to
7.5	6.5	• 4~ 2		.593		1210	Alony 91
8.0	6	.423	.592	• 593	.169	.170	EPAMALE
8.5	5.5 5			• 594		.171	
9.5	4.5			. 595		.172	106 × 0
10.0	4	.424	.593	.596	.169	.172	s versions i deve
10.5	∴•> 3			.596			
12.0	2	.425	. 594	.597	.169	.172	
13.0	1	1.26	505	,598 598	169	.172	
14.0	U	.420	• 272	• 170	.107	• 1 / K	
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	and the second se				1		
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# Geometry VII

TABLE 4-2 - SMALL FLUME - SEGMENT TESTS - ROUGH BOUNDARIES

(All model tests are two dimensional)

Measured Data

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### Calculations

(1) Run .No.	(2) Q	(3) Slope	(4) yn	(5) yl	(6) <u>F</u> n	(7) M	(8) M*	(9) C <sub>d</sub>	(10) y <sub>l</sub> /y <sub>n</sub>	$(F_n/M^{11})^{2/3}$
1	0,010	0.00005	0.1116	0.1398	0.0957	.4364	-	.208	1.251	-
2	11	11	11	0.1190	11	.5820	~	.312	1.067	
3	11	tr	11	0.1135	Ħ	.7270	~	.080	1.018	-
4	11	0001	0.0848	0,1081	0.1441	.4365	-	.206	1.278	-
5	11	11	11	0.0899	11	•5820	-	.142	1,060	-
6	11	11	11	0.0861	11	•7270		.104	1.017	-
7.	11	0.0003	0.0672	0.0906	0.2035	•4364	-	• <b>2</b> 28	1.350	-
8	11	2.8	T?	0.0736		•25550	.1650	.171	1.098	0.576
9		t1		0.0697	11	•727	•5194	.129	1.039	0.476
10	11	H.	0.0792	0,1050	0.16.50	•4364		.210	1.325	-
11	11	11	11	0.0841	H	•5850	•4596	.151	1.063	0,518
12	11	11	11	0.0806	11	•7270	•5936	<b>-1</b> 08	1.019	0.415
13	11	0.0010	0.0580	0.0792	0.2544	•4364	• 3247	•254	1.367	0.850
14	11	51	11	0.0653	11	•2850	•7683	.192	1.125	0.643
15	11	11	11	.0605		.7270	•6354	.147	1.042	0.743
10	51	0.0020	0.0526	0.0786	0 <b>.</b> 2950	•4364	• 3421	.254	1.496	0,906
17	11	11	11	0.0023		•2850	•4988	.202	1.1°2	0.704
18				0.0556	и 	•7270	•6448	.162	1.059	0.594
19		0.0040	0.0467	0.0744	.0.3510	.4304	•35.72	•266	1.591	0.985
20	**	11	11	0.0583		•2850	•5087	.216	1.249	0.7%1
21		"		0.0510		•7270	.0543	.175	1.091	0.661
22	11	0.0070	0.0380	0.0700	0.4790	•4364	• 3766	•280	1.841	1.174
23	11			0.0540		•5820	•5232	•22×	1.441	0.943
24	11	0.0000	0 0 2 2 2	0.0462	0 5600	•7270	.5688	.194	1.217	0.801
25		0.0005	0.0343	0.0095	0.5000	.4364	• 3827	.282	2.025	1.289
20	11	18	11	0.0530		.5020	-5290	•232	1.505	1.039
-1	11	0 0100	0 0011	0.0445	0 41.20	• (210	.0754	.202	1.299	0.503
20	11	0.0100	0.000TT	0.0090	0.0470	-4304	• 3000 Tabe	• 200 01-0	2.6219	1.100
27	11	н		0.0523		•502U	•5347 6707	011	1.001	1.135
21	11	0 01 20	0.0000	0.0425	0.7100	1210	-0171	- CTT - 264	1,000 0,000	U,700
30		10.0120- 11	11	0.0521	U • ( 190 11	£820	• 3710 5378	01.0	1 700	1.500 1.51b
22	11	11	11	0.0122	11	7270	6831	211	1 157	1 034
31.	11	0.0150	0 0277	0.0671	0 7700	1,361	-2033	202	2 1.21	1 565
35	11	U-U-U-U-U-U-U-U-U-U-U-U-U-U-U-U-U-U-U-	11	0.0510	11	5820	5101	21.6	1 81.1	1 267
36	11	11	TE	0.011	tt	7270	6856	221	1 1.85	1 080
37	11	0 0180	0.0265	0.0666	0 8220	1361	3051	205	2.517	1.630
38	11	п 0.0100	1	0.0501	11	5820	5/18	2/18	1.900	1.321
29	11	!!	- 4	0. 1400	tt	7270	.6877	.223	1.510	1 125
40	11	0.0210	0.0251	0.0559	0.8910	.136h	3976	296	2.622	1.71)
41	11	1.0570	11	0.000	1	5820	51.1.2	0C3	1 071	1 380
12	11	rt -	R	0.0385	11	7270	5802		1 522	1187
43	11	0.0210	0.0225	0.0612	0.0780	•121U	1004	• 500		1 810
1.1.		11	11	0.01.70	0.9700	-4304	-4000 CLGE	• 522 569	C+272	1.013
1,4	17		11	0.0260		7070	•5405 4007	200	1 1 2 2 2	1.4/5
45				0.0302		•1210	•072T	• 241	7.222	1.259

(l) Run No.	(2) Q	(3) Slope	(4) yn	• (5) 	(6) F <sub>n</sub>	(7) M	(8) MI	(9) C <sub>d</sub>	(10) y <sub>l</sub> /y <sub>n</sub>	(11) $(F_n/M')$	2/3
46	0.020	-	0.1526	0.2782	0.1193	•4364	-	-	1.825	-	
47	11	-	11	0.1847	11	•5820	-	.207	1.250	-	
48	n	-	11	0.1654	11	.7270	-	.121	1.006	、 <del>-</del>	
49	11	0.00005	0.1463	0.2738	0.1269	•4364	-	-	1.870		
50	11	11	11	0.1847	н	•5820	-	.200	1.260	-	
51	11	11	17	0.1589	11	.7270	-	<b>.</b> 123	1.084	-	
52	TR.	0.0006	0.1185	0.2444	0.1741	•4364	-	-	2.062	-	
53	11	11	11	0.1529	п	<b>.</b> 5820	-	.197	1.290	-	
54	н	11	11	0.1284	11	.7270	-	.145	1.064	-	
55	п	0.0010	0.0974	0.2160	0.2340	•4364	-	-	2.218	-	
56	11	н	H	0.1298	11	•58 <b>2</b> 0		.212	1.331	-	
57	11	H	11	0.1045	11	•7270	•5365	•174	1.072	0.575	
58	п	0.0014	0.0811	0.1987	0.3080	<b>.</b> <u>1</u> 764	-	-	2.444	- 075	
59	11	11	n	0.1116	11	•5820	•111.90	•236	1.374	.0.815	
60	11	11	11	0.0899	tt	•7270	.5874	•195	1.108	0.650	
61	11	0.0030	0.0688	0.1876	0.3950	.4364	-	-	2.724	-	
62	11	11	н	0.1004	11	.5820	•4604	.256	1.460	0.903	
63	п	11	11	0.0803	11	.7270	<b>.61</b> 58	•233	1.108	0.744	
64	11	0.0050	0.0629	0.1860	0.4510	.4364	-	-	2,960	-	
65	11	11	11	0.0964	11	•2850	.4772	.266	1.532	0,963	
66	н	11	11	0.0770	11	.7270	.6274	•233	1.224	0.802	
67	11	0.0070	0.0567	0.1820	0.5290	•4354	• 3295	•972	3.208	1.371	
68	11	tt	:1	0.0937	11	<b>.</b> 5820	.4900	•275	1.651	1.052	
69	11	11	n	0.0734	11	.7270	•6383	•245	1.291	0.884	
70	н	0.0090	0.0500	0.1803	0.6380	•4364	• 3500	.856	3.606	1.405	
71	Ħ	11	11	0.0921	н	<b>-</b> 5820	•5034	.261	1.642	1.172	
72	11	11	17	0.0723	11	.7270	.6485	-248	1.447	0.989	
73	п	0.0110	0.0459	0.1792	0.7230	.4364	• 3609	•824	3.908	1.508	
74	11	**	11	0.0912	11	<b>.</b> 5820	•5104	.280	1.989	1.201	
75	11	11	11	0.0706	11	.7270	•6550	•252	1.540	1.060	
76	11	0.0140	0.0416	0.1796	0.8360	•4364	•3692	-850	4.310	1.725	
77	tt	11	11	0.0900	11	•2650	.5174	-254	2.152	1.316	
73	11	п	11	0.0697	11	.7270	.6623	•259	1.678	1.100	
79	и.	0.0180	0.0382	0.1796	0.9510	•4364	•3762	•850	4.700	1.915	
80	tt	it	11	0.0868	11	•5820	.5226	.286	2.323	1.490	
81	"~	11	!!	0.0670	11	•7270	-6681	-250	1.152	1.205	
82	11	0.0220	0.0362	0.1774	1.0320	•4364	•3797	•769	4.900	1.949	
83	71	11	11	0.0862	11	-5820	.5255	.296	2.302	T.200	
84	11	н	11	0.0625		•7270	.6717	-200 	1.(27	1.334	
85	11	0.0240	0.0360	0.1751	1.0410	.4364	.3801	.708	4.070	1.755	
86	11	11	11	0.0864	11	.5820	.5267	.294	2.390	1.5/5	
87	11	tt.	11	0.0614	11	•7270	.6717	•292	1.702	1.340	•
88	11	0.0260	0.0351	0.1762	1.0800	• 4364	. 3810	.(35	5.020	2.000	
89	11	11	rt	0.0863	11	.5820	.5219	.296	2.450	1.012	
90	11	11	11	0.0625	11	.7270	.6732	•2r6	T. (CT	1.310	

TABLE 5-1- DATA FOR NORMAL DEPTH TEST RUNS - ROUGH BOUNDARIES

Run No.	Q	Slope	Уn	${\mathcal T}_{{\mathbb T}}$	Temp.
	cfs	ft./ft.	cm.	cm.	F
1	1.J	0.000010	22.91	19.50	71
5	11	0.000075	16.49	12.70	70
3	13	0.000200	13.55	9.70	70
4	н	0.000400	10.90	7.00	70
5	п	0.001700	9.27	5.40	59
5	11	0.001200	7.95	4.20	59
7	11	0.002.00	6.96	3.60	69
8	8.8	0.003500	5.97	3.30	68
9	2.0	0.000025	30.80	25.00	64
10	11	0.000100	23.35	17.00	. 6li
11	11	0.000250	18.60	12.50	66
12	11	0.000500	15.26	9.00	55
13	77	0.000800	13.17	7.00	67
14	71	0.001/00	11.17	5.00	67
15	11	0.002500	2.45	3.80	67
16	11	0.004000	8.18	3.00	68
17	3.0	0.000050	39.16	31.00	71
18	н	0.000150	30.90	22,50	70
19	11	0.000300	21:02	15.50	70
20	11	0.000600	19.04	10.60	70
21	11	0.001000	15.99	7.70	72
22	17	0.001600	13.73	5.70	72
23	11	0.003000	11.41	4.00	72
24	12	0.004500	.10.07	3.10	72

TABLE 5-2- TESTS FOR THE ROUGHNESS PARAMETER X

Run No.	yn cm∙	Ç cfs	S	C/Vg	y <sub>n</sub> /a	y <sub>n</sub> /χ
1 2	8.66 8.44	3.714 3.574	0.0125	8.169 8.162	6.829 6.650	22.548 21.976
3	8.05	3.273	88	8.005	6.348	20.960
4	7.72	3.066	17	7.982	6.086	20.095
5	7.07	2.586	21	7.646	5.574	18.405
6	6.06	1.969	11	7.283	4.779	15.778

A) Normal Depth Tests

B) Velocity Profile Data ( y measured from the bottom )

Q = 3.714 cfs;  $y_n = 0.275 \text{ ft}$ ; S = 0.0125:

y ft.	y/2	v fps	v√ <b>t₀</b> /e
0.010	0.794 1.190	1.89 1.94	5.985 6.143
0.020	1.984	2.00	5.333 6.872
0.030	2.381	2.25	7.125
0.040	3.175	2.42	7.663
0.045	3.571	2.59	8.2.1
0.055	4.365	2.74	8.675
0.060	4.762	2.83	8.961
0.070	5.556	2.94	9.468
0.080	6.349	3.10	9.816
0.100	7.937	3.38	10.703
0.110	8.730	3.48	11.020
0.130	10.317	3.65	11.558
0.140	11.111	3.68 3.75	11.653 11.875
0.180	14.286	3.86	12.223
0.200	15.873 17.460	3.94 1.00	12.476 12.666
0.210	19.048	1.06	12.856

TABLE 7 - 1 - 1

OBSERVED AND CALCULATED DATA

GEOMETRY Ia

Contraction	Slope	3	y.	ľ	۲,	æ	#1 #1	3	•	-	п≜∜гн	V <sup>2</sup> /267 <sub>6</sub>
01200	ม/น น/น	cfs	5	X III Com	<b>1</b> 8	ಕ	긟>	Equation 2		OTx		
	0	~	23.78	24.53	<b>5.2</b>	1665.	25,147	.2067	I	۱	87.	.0053
'		+	22.88	87.62 53.78	x 8	. 5762		.1962	8010	1380	• 55	100
'	\$	+	32.15	33.33	31.17	.7400	34, 795	.2095	5710	1060	5.	00/18
	8	2	22.62	27.90 11.10	21.11	2017	25,606 35,5606	.2294 2280	.0231	.1276	67.	<b>69067</b>
		L	19.86	20.90	18.73	5160	26,188	2607	0347		19	1600
0.5	901	<b>~</b> √	2.12	23.66	20.19	.5531	36,533	3286	1610.	111	8	.0158
	185	3.5	20.45	2.42	17.70	.5280	47,008 50.555	.2892	.0288	.1255	ż S	6600.
		~	12.07	14.72	9.61	315	28,516	4329	02120	.1127	3	0397
	250	m -4	15.94	23.53	12.21	4319	40,959	.3807	.0208	1911.	84.	6110
1	350	5.	15.40	2.2	36.3	1614	56.385	.4803	7610		33	1060 ·
	55	~	9.25	13.57	3.50	.2704	29,459	5026	.0197	1046	: <del>.</del>	1233
	2	n 4	12.38	18.09 21 15	1.67	3490	12,015	.3888	0202	7011.	ŝ	.0256
		* ~	23.78	23.93	23.46	10.05	25,147	1506	.0138	-0920	46	88
	0	3	33.97	34.18	33.57	.7688	21,213	7351.			i zi	1,00
'			31.15	41.76	40.93	6978.	169. 21	.1385	1	1	8.	0700*
	25	N (**	20.15	10.02	9 7 7 7	5762	25,377	-1562	.0128	-0957	3:	0900
	2	1-4	11.05	17.07	39.63	0098	190	00ML.	64TO-	7711	÷۶	9100
1		~	20.02	22.22	21.7	-5594	25,606	.1668	1620	227	jej	6900
	8	<b>~</b> .	29.75	30.06	29.31	.7003	35,590	.1625	.0235	.1319	1	.0075
		+	37.50	37.82	36.97	.8222	161.44	.1579	.0246	. 1406	Ŗ	1500.
5	100	"	21.70	22.18	20.98	2 2 2 2 2 2 2 2	391 07	1474	1000		j.	1600.
'		4	30.69	51.16	29.95	1912.	47,025	2062	.0287	1811	5	6600
'	185	4	20.45	21.19	ET-61	.5280	50,555	.3765	.0227	.1281	Ŗ	20110-
•	250	<b>v</b> m	5.5	8.71	10-01	1115	28,516	-3737	.0212	1127	5.	7660.
'			18.97	20.06	16.80	1975	52,860	5167.	0191	100	÷.	1060
1	350	2	15.40	16.84	13.20	1614.	56, 385	8584.	7610.	a701.	07.	.084.8
	200	~	9.25	10.68	0.94 0.75	102.2	29,459	Lan.	.0197	1046	8	88.
	0	-	33.92	33.81	33.61	2688	34,213	BLOIT		1011.	<u>;</u> =	300
	25	2	22.86	22.86	22.79	.5756	25, 385	CE21.	.0128	.0955	17	990
'			1.01	10.17	39.97	8600	43,180	.1095	8410.	1111.	.15	1100.
	Ş		3.2	80.72	8.3	-5554	25,606		.0231	1276	R.	-0067
	ł	-	37.50	37.67	17. 27	200	240,00	lear.	Server and a	101	ગેમ	58
P	8	-	2.12	21.78	21.57	.5531	38, 533	.1984	1610	111	) <b>R</b>	130
6.0			50.73	30.78	30.51	1971.	47,008	.1597	0288	C8977.	.19	6600
'	181	1	20.45	23.43	22.22	82.5	50,555	SELE.	1220.	552.	ia	2010
	250		15.91	16.04	14.41	Ì	10.948	0100		j.	ş,	1450.
1			16.97	18.99	18.52	\$167.	52,860	422	1610.	100	şą	100
'	370	-	15.40	3.5	8.3	611.	56, 385	0611	*010*	5587-	E.	3
	005		10. ST		19.01	10/2	29,459		610-	1016	Ŗ	Tar.
			13.20	13:57	1.21	198	562.95	3006	50	1001	i ș	100

GEOMETRY I

TABLE 7-1-2 RAW DATA - LARGE FLUME - SHOOTH BOUNDARLES P. 1

(Semicircular Model Teste)

ы еге са. - 0.300 677 33.95	. Depth (cm.)		85 fre cm. 0.889 66F 34.90	. Depth (cm.)	<sup>8</sup> 8 8 8 8 8 8 8 8 8 8 8 8 8 8 8 8 8 8
Run No. 5 Q = 1 0 M = 2/03 M = b/B - L/b = 0.0 L/b = 0.0 L/b = 0.0 Slope = 0 Temp. = 1	Cent. Sta (ft.)	52222222222222222222222222222222222222	Run ko. 1 Q = 2 2 Fn = 20,3 M = 0/8 L/b = 0.0 L/b = 0.0 L/b = 0.0 Jamp. = Model Sta	Cent. Sta (r)	84888888888888888888888888888888888888
• •.300 00229 17 - 33.91	Depth (cm.)		с га очб93 6F - Э4.35	Depth (cm.)	50.52 50.55 50 50 50 50 50 50 50 50 50 50 50 50 5
Run No. 85 Q = 1 cf Yn - 3.60 c M - b/B - ( L/b - 0.0 L/b - 0.0 Slope - 0.( Temp b Model 3ta.	Cent. Sta. (ft.)	82888888888889999999999999999999999999	Run No. 17 Q = 2 c Yn = 20.3 M = 5/8 L/b = 0.0 L/b = 0.0 Temp. = 6 Hodel Sta	Cent. Sta. (ft.)	8 000000000000000000000000000000000000
.300 	Depth (cm.)	11 22 22 22 23 24 24 24 25 24 25 24 25 24 25 25 25 25 25 25 25 25 25 25 25 25 25	55 "" 0.491 " 32.66	Depth (cm.)	44444444444 88882844444444
Run No. 75 4 - 1 cfi 7a - 4.06 cu M - b/8 - 0 M - b/8 - 0 L/b - 0.0 Slope - 0.0 Slope - 0.0 Flemp 66	Cent. Sta. (ft.)	2800 2800 2800 2900 2000 2000 2000 2000	Run No. 1 Q = 2 cf yn - 20.3 y - b/8 - L/b - 0.0 Slope - 0.0 Tamp. 65 Hodel Sta.	Cent. Sta. (ft.)	88.88 88.89 89.90 80 80 80 80 80 80 80 80 80 80 80 80 80
". 300 01260 .F - 31.87	Depth (cm.)	12.29 12.27 12.29	s [s 0,889 0,889 6F - 33,08	Depth (cm.)	~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~
Run No. 65 Q = 1 cf: Yn - 4.31 cf: M - 5/8 - 0 M - 5/8 - 0.0 L/6 - 0.0 Slope - 0.0 Slope - 0.0 Slope - 66	Cent. Sta. (rt.)	200 200 200 200 200 200 200 200 200 200	Run No. 15 4 - 2 c: 7 - 25 32 M - b/B - L/b - b/B - L/b - 0. 510pe - 0. 7 emp 6 Hodel Sta.	Cent. Sta. (ft.)	88.88 99.99 90.99 90.99 90.99 90.99 90.99 90.99 90.90 90 90.90 90 90.90 90 90 90.90 90 90 90 90 90 90 90 90 90 90 90 90 9
.300 0.0653 - 32.24	Depth (cs.)	4.22 2.22 2.22 2.22 2.22 2.22 2.22 2.22	s ta 0.693 66P = 33.46	Depth (cm.)	88888888888888888888888888888888888888
Run Mo. 55 4 = 1 cf ya - 5.11 cf M - b/B - 0 1/b - 0.0 510pe - 0.0 7emp 64 Model Sta.	Cent. Sta. (ft.)	20 23 23 23 23 25 25 25 25 25 25 25 25 25 25 25 25 25	Run No. 145 9 - 2 cfi 9 - 25.32: M - b/B - 6 1/b - 0.0 510p - 0.0 510p - 0.0 7 cmp 6 Model Sta.		20 20 20 20 20 20 20 20 20 20 20 20 20 2
	Depth (cm.)	22.68 22.69 22.69 22.69 22.69 22.69 22.69 22.69 22.69 26.11 26.69	s .a 0.491 6F - 33.41	. Depth (cm.)	25.95 25.95
Run Ko. 45 4 - 1 cfs yn - 6.10 cm H - b/B - 0 L/b - 0.0 L/b - 0.0 Slope - 0.0 Slope - 64 Hodel Ste.	Cent. Sta. (ft.)	20 20 20 20 20 20 20 20 20 20 20 20 20 2	Run No.13 2 cf 2 cf 2 n - 2 cf 2 n - 2 cf 2 n - 2 cf 1 - 1 - 1 - 1 2 cf 2 cf 2 cf 2 cf 2 cf 2 cf 2 cf 2 cf	<b>Gent.</b> Sta (ft.)	4 000000000000000000000000000000000000
.300 .0286 - 35.85	Depth (cm.)	20011111111111111111111111111111111111	5 5.889 0.889 0.889 0.889 0.889 0.889 - 34.10	Depth (cm.)	88.88.88.88.88.88.88.88.88.88.88.88.88.
Run No. 35 Q = 1 cfe Y <sub>n</sub> - 7.50 cm M = b/B = 0. L/b = 0.0 Slope = 0.0 Slope = 0.0 Model Sta.	Cent. Sta. (ft.)	22,22 22,25 23,25 24,25 24,25 24,25 24,25 25,25 24,25 25,25 26,25	Run No. 12: Q = 2 cf Vn -38.65 c H = b/8 - 1 L/b = 0.0 Slope = 0.0 Slope = 0.0 Model Sta.	Cent. Sta. (ft.)	88 88 88 88 88 88 88 88 88 88 88 88 88
	Depth (cm.)	13.78 13.78 13.77 13.78 13.76 13.76 13.76 13.77 7.87 7.87 7.87	15 .fa cm. 0.693 00000712 67F 34.43	. Depth (cm.)	ĸĸĸĸĸĸĸĸĸĸ ĸĸĸĸĸĸĸĸĸ
Run No. 25 Q. 1 cfe Yn - 9.82 cm M b/8 - 0.0 L/b - 0.0 Slope - 0.0 Slope - 0.0 Model Sta.	Cent. Sta. (ft.)	20.0 28.0 29.0 29.5 29.5 29.5 29.5 29.5 29.5 29.5 29.5	Run No. 1 4 - 2 - 2 7 - 36.65 7 - 9/8 36.65 1 - 9/8 - 0.0 51 ope - 0.0 51 ope - 0.0 7 - 1000 - 1000	Cent. Sta (r.)	88 88.0 88.0 88.0 88.0 88.0 8 8.0 8.0
, 300 00281 - 34,39	Depth (cm.)	17.58 17.58 17.58 17.55 17.55 17.55 17.55 17.55 17.55 17.55 17.55 17.55 17.55 17.55 17.55 17.55 17.55 17.55 17.55 17.55 17.55 17.58		Depth (cm.)	911122222222 88.62.22222222 8.6222222222
Run No. 13 $V_{11} - 15.64cm$ , $V_{12} - 15.64cm$ , M - 5/8 - 0 M - 5/8 - 0.00 1.0 = 0.00 31079 - 0.00 7emp 627 Model Sta	Cent. Ste. (ft.)	8 9 9 9 9 9 9 9 9 9 9 9 9 9 9 9 9 9 9 9	Run No. 103 Q = 1 cfr Yn = 2.89cr H = b/B = ( L/b = 0.0 Slope = 0.0 Temp = 0.0 Modal Sta.	Cent. Sta. (ft.)	848844888659 999999999

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ď	
(continued)	
BOUNDARIES	
SHOOTH	
- 1	
FLUNE	
LARGE	
- 1	
DATA	
2 BAW	
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2	
TABLE	

# (Semicircular Model Testa)

693 0628 33₊34°	Depth (cm.)	10.46 10.50 10.46 6.04 6.04 6.76
$\begin{array}{llllllllllllllllllllllllllllllllllll$	Cent. Sta. (ft.)	284.0 2000 2000 2700 2700 2700
491 0628 - 33.65	Depth (cm.)	15.44 25.21 25.21 25.22 25.25 21.25 21.85 21.85
Run No. 265 Q 2 cfa Yn - 8.86cm N - b/8 - 0. L/b - 0.0 Slope - 0.00 Slope - 0.00 Slope - 671 Tamp - 671	Cent. Sta. (ft.)	24.0 28.0 31.0 31.0 31.5 31.5 31.5 31.5 31.5 31.5 31.5 31.5
389 34,86 33.87	Depth (cm.)	9.85 9.87 9.87 9.87 9.87 9.87 9.87 9.85 9.85 9.53 9.53
Run No. 255 4 - 2 cfa y <sub>n</sub> - 9,71cm. M - b/8 - 0.6 1/b - 0.0 Slope - 0.000 Tamp 66F Tamp 66F	Cent. Sta. (ft.)	2200 2800 2810 2810 2810 2800 2800 2800
. 693 00486 - 33.86	Depth (cm.)	10.95 10.95 10.95 8.23 8.23 8.23 8.23 8.23 8.23 8.23 8.23
Run No. 245 Q = 2 cfs Yn -9.71 cm H + b/B - 0 Ly <sup>A</sup> b - 0.0 Ly <sup>A</sup> b - 0.0 Slope - 0.0 Slope - 66 Temp 66	Cent. Sta. (ft.)	20.0 28.0 30.0 33.5 33.5 33.5 33.5 33.5 33.5 33
5 • 491 • 34.40	Lepth (cr.)	5.52 5.52 5.52 5.52 5.52 5.52 5.52 5.52
Run No. 23 4 - 2 cfa 7 - 9 71cm N - 5/8 - 0 1/b - 0.0 510pe - 0.0 70mp - 660 Model Stu.	Cent. Sta. (ft.)	20.0 28.0 33.7 33.5 33.5 33.5 33.5 33.5 33.5 33.5
300 34,86 33.87	Depth (cm.)	20.68 20.73 20.77 20.77 20.68 2.25
Run No. 228 4 - 2 cfa Yn - 9.71cm M - b/8 - 0. 1/b - 0.00 510P - 0.00 510P - 0.00 510P - 0.00 76mp - 66F Hodal 16a - 66F	Cent. Sta. (ft.)	24.0 28.0 31.5 31.5 37.12
869 010 03.10	Depth (cm.)	16.37 16.47 16.47 16.47 16.47 16.37 16.37 16.33 16.33 16.33
Run No. 215 4 - 2 cfs y <sub>n</sub> -16.46cm. M - b/8 - 0. 1/b - 0.00 Slope - 0.00 Temp 66F Hoodi Sta	Cent. Sta. (ft.)	22.00 23.00 23.00 23.00 23.00 23.00 23.00 23.00 23.00 23.00 23.00 23.00 23.00 23.00 23.00 23.00 23.00 23.00 23.00 23.00 20 20 20 20 20 20 20 20 20 20 20 20 2
693 010	Depth (cm.)	16.02 16.72 16.72 16.72 16.72 15.93 15.93 15.93 15.93 15.93
Run No. 205 q 2 cfe yn - 16.46cm. H 5/8 - 0. Lyb - 0.0 Slape - 0.00 Slape - 0.00 Fremp 671	Cent. Sta.	35:00 35:000
191	Depth	17.82 17.82 17.79 17.79 17.77 15.11
Run No. 195 4 - 2 cfe 7n - 16.46m. H - 5/8 - 0.4 L/b - 0.0 Sibpe - 0.000 Temp 577	Cent. Sta.	32.00 33.00 34.5 34.5

65 fe 0.889 0.889 0.0015 7F - 34.02	. Depth (cm.)	7.73 7.61 7.61 7.77 7.77 6.83 6.83 6.83 6.83 7.05
Run 3 4 - 2 c. 7 yn -7.55 8 - b/5 - 1/b - 0.0 1/b - 0.0 1 emp 6 1 emp 6 1 emp 6	Cent. Sta. (ft.)	20.0 331.0 3
s .693 D115 - 34.23	Depth (cm.)	9.75 9.95 10.02 10.03 9.99 9.99 3.04
Run No. 35: Q = 2 cf3 Yn -7,35 cr H - b/8 -0,0 L/b -0,0 Slope -0,00 Tiemp. 7 671 Tiemp. 7 671	Cent. Sta. (ft.)	28.0 28.0 37.7 37.7
с 	Depth (cm.)	2.32 2.51 2.51 2.51 2.51 2.51 2.51 2.51 2.5
Run No. 34 4 - 2 cf: 7 - 7,35 cc 7 - 1,8 - 0,0 8 - 0,0 510pe - 0,0 510pe - 0,0 7 Temp 67 Model Sta.	Cent. Sta. (ft.)	28.0 28.0 28.0 29.0 29.0 29.0 29.0 29.0 29.0 20.0 20
а 0.300 0115 - 34,35	Depth (cm.)	20, 55 20, 75 20, 85 20, 88 20, 88 20, 88 20, 84 20, 84 20, 18
Run No.335 4 - 2 cf: 4 - 2,55 c: N - b/8 - 1 N - b/8 - 0 1,b -0,0 510pe -0,0 510pe - 60	Cent. Sta. (ft.)	28.0 28.0 33.5 33.5 33.5 5 42.63 33.5
s • • 889 • • 200779 - 34. 85	Depth (cm.)	8.53 8.53 8.53 8.53 8.53 8.54 8.57 7.48 7.48 7.85 7.85 7.85
Run No. 32 4 - 2 cf yn - 8,29 c N - b/8 - 0 N - b/8 - 0 N - 0,0 Slope - 0, Temp, - 60 Model Sta.	Cent. Sta. (ft.)	788600 78860 78860 79960 7900 7000 70
693 6770 53.65	Depth (cm.)	10.10 10.21 10.21 5.55 5.53 5.53 5.33
Run No. 315 2 - 2 cfa 3n - 8,29 cm M - b/8 - 0 L/b - 0.0 Slope - 0.00 Slope - 6.01 Actal Sta.	Cont. Sta. (ft.)	88.00 37.000 37.0000000000
191. 0779 - 34.17	Depth (cm.)	30.34 30.51 30.55 30.55 30.55 30.55 30.55 20.51 20.51
Run No. 30S Q - 2 cf3 Yn -8.29 cm N - b/8 - 0 L/b -0.0 Slope -0.00 Slope -0.00 Slope -0.00	Cent. Sta.	22.00 23.00 23.00 23.05 23.05 23.05 23.05 23.05 23.05 23.05 23.05 23.05 23.05 23.05 23.05 23.05 23.05 23.05 23.05 23.05 24.05 25.05
5° •300 •300	Depth	20.65 20.77 20.77 20.83 20.83 20.83 20.73 20.73
Run No. 2% 4 - 2 cft 14 - 82 cft 14 - 9/8 - 0 11/5 - 0.0 51006 - 0.00	Cent. Sta.	22:00 22:00 22:5 22:5 22:5
889	Depth	9.02 9.05 9.05 9.05 9.05 9.05 9.05 9.05 8.45 8.51
ыл No. 285 1 - 2 сfe 1 - 8.86сm. 1 - b/B - 0. 1 - b/B - 0.0 Slope - 0.00 Slope - 0.00	Model Ste	88.00 89.00 89.00 89.00 89.00 89.00 89.00 89.00 89.00 80 80 80 80 80 80 80 80 80 80 80 80 8

TABLE 7-1-2 RAY OFTA - LARGE FURGE - SHOOTH BOUNDHULES (continued) P. 3 (Semicircular Model Teste)

34•3E	Dept (cm	19.78 20.43 20.76 20.76 20.77 20.77 20.77 1.19
Run No. 458 Q = 2 cfo Ya - 5.70 cm. Ya - 5/8 - 0.5 L/b - 0.0 Slope - 0.00 Slope - 66F Model Sta 66F	Cent. Sta. (ft.)	20.0 33.0 33.5 33.5 47.40
889 889 216 - 34.25	Depth (cm.)	6.55 6.75 6.72 6.92 7.07 7.07 7.05 7.05 7.00
Run No. 445 Q = 2 cf3 Y <sub>n</sub> - 6.15 cm M = b/3 - 0.0 Slope - 0.0 Slope - 0.0 Temp 665 Femp 665	Cent. Sta. (ft.)	20.0 20.0 20.5 20.5 20.5 20.5 20.5 20.5
693 693 00216 - 33.82	Dupth (cm.)	9.18 9.71 9.83 9.85 9.85 9.79 2.15
Run No. 4,35 4 - 2 cie 7 - 6,15 cm M - 6,15 cm 1,16 - 0,0 Slope - 0,0 Slope - 0,0 Slope - 0,0 Slope - 0,0	Cent. Sta. (ft.)	20.0 30.5 31.0 31.5 39.34
, 491 216 - 33.82	Depth (cm.)	12.65 13.26 13.28 13.29 13.29 1.58
Run Mc. 425 Q = 2 cfs Yn -6.15 cm M = 9/9 -0. Lyb - 0.0 Slope -0.00 Slope -0.00 Temp 65F Moulel Sta.	Cont. Sta. (ft.)	20.0 30.0 31.0 31.5 40.66
	Lepth (cr.)	20.07 20.54 20.54 20.68 20.73 20.75 20.75 20.75 20.68
Run Mo. 415 4 = 2 cfs 70 - 6.15 cm H = b/8 - 0.0 L/b = 0.0 Slope = 0.0 Slope = 0.0 Model Sta.	Cent. Sta. (ft.)	28.0 30.0 32.5 28.5 28.5 28.5 28.5 28.5 28.5 28.5 2
889 889 0161	Depth (cm.)	7.00 7.20 7.10 5.15 5.15 5.10 6.30
Run No. 405 Q - 2 cf3 Yn -6.77 cm M - b/B -0. L/b - 0.0 Slope - 0.0 Slope - 0.0 Rodel Ste.	Cent. Sta. (ft.)	29.0 29.5 33.5 35.5 36.5 36.5
15 1,693 1,693 1,693 1,693 1,693 1,147	Depth (cm.)	9.81 9.90 9.94 9.84 2.18*
Run Mo. 3: 4 - 2 cfr 7 cfr 7 -6.77 cfr 7 c	Ceat. Sta. (ft.)	28.0 31.0 32.0 32.5 39.71
s 491 161 - 34.54	Depth (cm.)	12.85 13.32 1.55 1.55 1.55
Run Mo. 385 Q - 2 cfe Yn - 6.77 cm M - b/8 - 0.0 L/b - 0.0 Slope - 0.00 Slope - 0.00 Temp 671 Temp 671	Cent. Sta. (ft.)	20.0 28.0 32.5 23.5 41.2 41.2
.300 [ [ - 34.27	Depth (cm.)	20,32 20,67 20,80 20,81 20,81 20,81 20,83 20,83 20,75 20,75
Run Mo. 375 4 - 2 cfe 7 - 6.77cm, 7 - 6.77cm, H - 5/8 - 0. 1/b - 0.0 1/b - 0.0 1/b - 0.0 1 femp, - 661 1 femp, - 661	Cent. Ste. (ft.)	28.0 28.0 29.0 29.0 29.0 20.0 20.0 20.0 20.0 20

ы в 1,693 8Р - 34,10	Depth (cu.)	8.5.5.5.5.5.5.5.5.5.5.5.5.5.5.5.5.5.5.5
Run 5 4 - 3 cf 7 - 3 cf 7 - 3,01c 73,01c 73,01c 73,01c 70,0 510pe - 0,0 1 eeyu - 0,0 1 e	Cent. Sta. (ft.)	22.00 23.25.00 23.25.00 23.25.00 23.25.00 23.25.00 23.25.00 23.25.00 25.00 26.00 27.000 27.000 27.000 27.0000000000
491 20025 34 <b>.</b> 08	Depth (cm.)	24.10 24.10 23.46 23.46 23.46 23.46 23.46 23.46 23.46 23.46 24.100
Run No. 535 4 - 3 cfs 7 - 3 cfs 7 60 1,0 0,0 510p - 0,00 16mp - 67P Nodel Sta	Cant. Sta. (ft.)	20.0 224.0 235.0 235.0 25.0 25.0 25.0 25.0 25.0 25.0 25.0 2
s 8 0.889 003495 6F - 33.41	Depth (cm.)	6.42 6.45 6.45 7.25 7.25 7.25 7.25 7.25 7.25 7.25 7.2
Run No. 52 4 - 2 cfr 7 - 5.24 cr N - b/B - 6 1/b - 0.0 510pº - 0.0 Tamp - 0.0 Model Sta.	Cent. St (ft.)	28.00 29.00 29.00 29.00 29.00 29.00 29.00 20.000 20.000 20.000 20.000 20.000 20.000 20.0000 20.0000 20.0000 20.00000000
5 1.693 23495 - 33.33	Depth (cm.)	8.35 9.55 9.55 9.65 9.45 2.06
Run No. 51 4 - 2 cti 7n -5.24 cu M - b/B - ( M - b/B - 0.0 Sloye -0.0 Sloye -0.0 Temp 661 Kudel Sta.	Cent. Sta. (ft.)	20.0 28.0 31.0 31.5 32.5 38.5 38.5 38.5
ks 1491 03495 - 33.15	Depth (cm.)	22.22.23.25.25.25.25.25.25.25.25.25.25.25.25.25.
Run No. 50 Yn - 2 cfa Yn - 5.24 cm M - b/8 - 0 L/b - 0.0 Slope - 0.0 Slope - 0.0 Model Sta.	Cent. Sta. (ft.)	20.0 28.0 31.0 31.5 31.5 32.5 32.0
800 33,88	Depth (cm.)	10.35 20.18 20.55 20.55 20.55 1.12*
Hun No. 495 4 - 2 cfa 3h - 5,24 cm. h - 5,24 cm. 1/b - 0,0 Slope - 0.0 Slope - 0.0 Slope - 66F Houel Sta.	<b>C</b> ent. Sta. (ft.)	2000 2000 2000 2000 2000 2000 2000 200
55 1,889 1,889 1,889 1,81 - 33,81	Depth (cm.)	5.05 7.07 7.15 7.15 7.15 5.29 5.29 5.29 5.29
Run No. 44 Q = 2 cfr Yn -5.70 cu M = 5.8 - 0 M = 64 Slope -0.0 Slope -0.0 C Temp 64 Model Sta.	Cent. Sta. (ft.)	20.0 20.0 32.0 33.5 33.5 33.5 33.5 33.5 33.5 33.5 33
5 •693 •24•16 - 34•16	Depth (cm.)	8.85 9.57 9.78 9.98 2.10
Run No. 47 4 - 2 cf <b>7</b> n - 5.70 cm M - b/8 - 0.0 L/b - 0.0 Slape - 0.0 Temp 651 Hodel Sta.	Cent. Sts. (ft.)	20.0 23.5 29.5 20.5 20.5 20.5 20.5 20.5 20.5 20.5 20
65 16 10 1491 00.491 66 7 33.30	. Depth (cm.)	4.4.4.4.4.4.4.4.4.4.4.4.4.4.4.4.4.4.4.
Run No. 41 Q - 2 ct Yn -5.70 c M - b/B - 0.0 Slape - 0. Temp 0.	Cent. Ste. (ft.)	888888888 00000000000

TABLE 7-1-2 RAW DATA - LARGE FLUME - SMOUTH BOUNDARLES (continued) P.4

(Semicircular Model Tests)

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с •,889 00185 - 33.04	Dupt (cm	19.58 19.55 19.55 19.55 19.55 19.25
Run No. 61 4 - 3.5 cf Yn - 19.445 M - b/B - C L/b - 0.0 Slope - 0.0 Temp 70 Model Sta.	Cent. Sta. (ft.)	20.0 28.0 37.0 35.0 35.0 35.0 35.0 35.0
ы ч • 693 60185 - Эч.56	Depth (cm.)	20,27 20,27 20,27 20,24 18,35 18,15 18,15 18,13 18,28
Run No. 66 Q - 3.5 cf Yn - 19.44 H - b/B - 0 L b - 0.0 Slope - 0.0 Temp 65 Temp 65	Cent. Sta. (ft.)	20.0 28.0 33.0 37.5 37.5 37.5
s a 0.491 6.491 - 34.17	Depth (cn.)	22.78 22.75 22.73 22.73 22.68 22.68 117.33 10.53 10.53
Run No. 59 Q = 3.5 cf. M = 0.9.444 M = b/B = L/b = 0.0 Slope = 0.0 Temp. = 70 Model Sta.	Cent. Sta. (ft.)	20.0 28.0 33.0 33.0 33.5 33.5 33.5 35.6
s • 889 • 00050 F = 33.98	Depth (cm.)	23.65 23.66 23.66 23.66 23.55 23.66 23.55 23.55 23.55 23.55 23.55 23.55 23.55 23.55 23.55 25.55
Run No. 58 Q = 3 cf. Q = 2.65 M = b/B = 0 L/b = 0.0 Slope = 0.0 Slope = 0.0 Slope = 68 Model Sta.	Cent. Sta. (ft.)	20.0 233.5 233.5 233.5 233.5 233.5 233.5 233.5 233.5 233.5 233.5 233.5 235.5 25.5 2
s 	Depth (cm.)	29.93 29.91 29.91 29.91 29.23 29.23 29.23 29.23 29.23
Run No. 57 4 - 3 cf. 4 - 29.65m M - b/B - 0.0 1/b - 0.0 51096 - 0.0 51090 - 68 Madel Sta.	Cent. Sta. (ft.)	20.0 28.0 33.0 37.0 37.0 37.0
s 	Depth (cm.)	30.87 30.87 30.89 30.89 30.89 30.85 30.85 28.55 29.55 29.55 20.87
Run No. 56 Q - 3 cfs Yn - 29.65 M - b/B - 0 M - b/B - 0.0 L/b - 0.0 Slope - 0.0 Temp 68 Model Sta.	Cent. Sta. (ft.)	20.0 28.0 33.5 33.5 33.5 33.5 33.5 33.5 33.5 33
, 889 00025 - 34,20	Depth (cm.)	ĸĸĸĸĸĸĸĸĸĸĸ 9,9,9,9,9,9,8,8,8,8,9,9,9,9,9,9,9,9,9,9
дил No. 555 1 - 3 cfs 1 - 33.01cm 4 - b/B - 0 /b - 0.0 Slope - 0.0 Bemp 67 bodel Sta.	Cent. Sta. (ft.)	22.00 22.00 23.00 23.00 23.00 23.00 23.00 26.00 27.000 27.000 27.000 27.000 27.000 27.0000 27.0000 27.0000000000

\*Only the minimum depth was measured on the downstream side of the model.

GEOMETRY I. <u>TABLE7-1-3</u> CALOULATIONS FOR THE LARDE FLARE SMOTH BUNDARY MORT, TREFE

Measured Data

Calculated Data

ສ ຕິ:	888888888888888888888888888888888888888
<sup>ع</sup> د "	4%, 1%, 1%, 1%, 1%, 1%, 1%, 1%, 1%, 1%, 1
8 1 1 1	ੑੑਗ਼
$y_1 - y_3$ (ft.)	0.000000000000000000000000000000000000
بوريد کې	0.000 0.0000 0.00000 0.00000 0.00000 0.00000 0.00000 0.000000 0.000000 0.00000000
20 n/H13/3	
19 1/ <sub>n</sub> (1	19999999999999999999999999999999999999
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15 Re	
r "	0.0.0977 0.0.2077 0.0.2071 0.0.2071 0.0.2072 0.0.20
13 8 8 8 (n.)	11,11,11,11,11,11,11,11,11,11,11,11,11,
12 1. Sta. (n.)	%E%34%E%2%13614386%4%2%2%2%2%2%2%2%2%2%2%2%2%2%2%2%2%2%2%
ц Sta Мо	00000000000000000000000000000000000000
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y <sub>1</sub> st (ft.	
× 10° × °	
8 y3 (cn.)	Ҷݘݭӊѻѻҫѻѻ。ݡݵҶݵӼჇჇჇჇჽႨႿႨჇႹႵჾჿჄჿ๛ႹႷႦჄႽႹႹႱჿႥႷჃႦჁႹႹႷჃႹႷႦჄႹႵႽჿႾႳႦႦჅჾჇჾ ჅႽჇჇჇჇჇჇჇჇჇჇჇႦႺႦႺႦႵႨჂႱჄႦჇჂႦႳჂჂჂႱႦႦႦჂႦႦႹႹჅჂႱႨჇჇჂႵႽჄჂႽჇႽႷႽჅႦႦႦ Ⴥ
7 y1 (cm.)	ឣឣឣឣឣឣឣឣឣឣឣឣ៓៳៳៱៶៸៹៶៹៹៹៹៹៹៹៹៹៹៹៹៹៹៹៹៹៹៹៹៹៹៹៹៹៹៹៹៹៹៹៹៹
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5 6 Cfa)	
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L/b	ç
н г. Ко.	ਸ਼ਲ਼ਖ਼ਸ਼ਲ਼ਫ਼ਫ਼ਫ਼ਲ਼ਫ਼ <u>ਸ਼ਗ਼ਲ਼ਸ਼</u> ਲ਼ਫ਼
4	

GEOMETRY I<sub>d</sub> AND I<sub>b</sub> t<u>abe 7-2-1 em orth - leiger fung</u> - rough boundarees

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(Two and Three Dimensional Semicircular Model Teste)

bun Mo. 1-1 1 - 1 cfe 1 - 24.32mm. 1 - 5/8 - 0.3 1 - 0.0 1 - 0.0000 1 - 0.00000 1 - 0.0000 1 - 0.0000 1 - 0.0000 1 - 0.00000 1 - 0.00	8 <del>8</del> 8	Run No. 1-2 Q - 1 cfe Yn -24.32cm. M - b/8 - 0.5 Slope - 0.00 Temp 64F Model Sta	311 30,002 30,00	Run No. 1-3 < - 1 cfs Yn - 24.32 cm. N - 10/B - 0. N - 10/B - 0. L/b - 1.0 Slope - 64,F Temp 64,F	і 115 200302 20.00	Run No. 1- 4 - 1 cf 7n -24.32 cf 7 - b/B - C 8 - b/B - C 1/b - 0.0 510pe - 0.0 7 - 64 Temp - 64	-4*** 's 0.491 0.491 - 30.00	Run Mo. 1- 4 1. cf V <sub>II</sub> -24.52 cf N - b/B - 0 L/b - 0.5 Slope - 0.6 Temp 66 Hodel Stu.	5** "'' 1,500 000302 8F - 30,00	Run No. 1- Q - 1 cf: Yn -24.52 cc M - b/B - 0 H - b/B - 0.0 Slope - 0.0 Slope - 0.0 Temp 66 Model 3ta.	6** 1,500 1,500 1,500 1,000 1,000	Run No. 1- 4 - 1 cfe Yn -24,52cm M - b/B - 0 M - b/B - 0.0 Slope - 0.0 Slope - 0.0 Temp 68 Hodal Sta.	7** • 693 • 30.002 - 30.00	Run No. 1- Q - 1 cfs Yn - 24.52m N - b/8 - 0 N - b/8 - 0 L/b - 0.5 Slope - 0.0 Tamp 68 Nodel Sta.	8** 700 700 30.002	Run No. 1-9 Q = 1 cfs Yn -24.52cs. Yn - b/B = 0. L/b = 1.0 Slope - 0.00 Slope - 0.00 Slope - 0.00 Slope - 0.00 Slope - 0.00	700 30.00
Cent. Ste. (ft.)	Depth (cm.)	Cent. Sta. (ft.)	Depth (cu.)	Cent. Sta. (ft.)	Depth (cm.)	Cent. Sta. (ft.)	Depth (cm.)	Cent. Ste. (ft.)	Depth (cm.)	Cent. Sta. (ft.)	Depth (cm.)	Cent. Sta. (ft.)	Dupth (cm.)	Cent. Sta. (ft.)	Depth (cm.)	Cent. Sts. (R.)	Depti (cm.
Subburg	pes	Submo	argod	Suba	e rgod	20.0 25.0 25.0 28.0 28.0 29.5 29.5 29.5 29.5	11111222233 11111222233	20.0 25.0 25.0 27.5 28.5 29.0 29.0 29.0 29.0	22.00 22.00 22.00 22.00 22.00 22.00 22.00 22.00 22.00 25.000	20.0 25.0 25.0 28.0 28.0 28.0 29.5 29.5 29.5 29.5	555558283555 33333333333333	20.0 21,00 25.0 26.5 28.5 28.5 28.5 28.5 28.5 29.5 29.5 29.5 29.5	55.28 55.28 55.28 55.28 56.28 57 58 58 58 58 58 58 58 58 58 58 58 58 58	20.0 25.0 25.5 26.0 26.5 26.5 26.5 26.5 27.5 29.5 29.5 29.5 29.5	525,522,522,528 55,525,525,525,58 55,525,525,525,58 55,525,525,525,525,58 55,525,525,525,525,58 55,525,525,525,525,58 55,525,525,525,525,58 55,525,525,525,525,58 55,525,525,525,525,58 55,525,525,525,525,58 55,525,525,525,525,58 55,525,525,525,525,58 55,525,525,525,525,58 55,525,525,525,525,58 55,525,525,525,525,58 55,525,525,525,525,585,585,585,585,585,5	20.0 24.0 25.5 25.5 25.5 25.5 28.5 28.5 28.5 28.5	25,25 27,25

.

Ann No. 1-10 1 - 1 cfa fn -24.52ma 4 - b/B - 0. 14/b - 0.0 51ape - 0.000 1amp 68F 60al Sta	(n.)	20.0 25.0 25.0 26.0 26.0 26.0 26.0 26.0 26.0 26.0 26
**************************************	Depth (cm.)	5.4.5.5.5.5.5.5.5.5.5.5.5.5.5.5.5.5.5.5
Run No. 1-1 4 - 1 cfr Jn - 24.50cm Jn - 0.8 - ( 1.0 - 0.5 Slope - 0.6 Tomp - 681 Model Ste.	Cent. Sta. (ft.)	20.0 25.0 25.5 26.5 27.5 28.5 28.5 28.5 28.5 28.5 28.5 28.5 28
0,900 - 30,000	Depth (cm.)	<u> </u>
Run No. 1- q - 1 cf yn -24,52c H - b/B - L/b - 1.0 Slope - 0 Temp, - 0 Hodel Sta.	Cent. Ste. (ft.)	2000 2000 2000 2000 2000 2000 2000 200
.12** 13 0.900 0.900 - 30.00	Depth (cm.)	353883385885 <b>8</b> 333333333333
Run No. 2-1 2 - 2 cfo 3 - 24,09 cm 1 - 24,09 cm 1 - 10 1 - 0.0 1 - 0.0 1 - 0.0 1 - 681 1 - 681 1 - 681 1 - 681 1 - 681	Cent. Sta. (ft.)	đuč
.300 .300	Depth (cm.)	pegree
Run No. 2-: U - 2 cfi Yn - 24,0922 M - b/B - ( L/b - 0.5 Slope - 0.6 Temp 06 Nouel Sta.	Cent. Sta. (ft.)	5 3 3
2  0.311 0.311 3F - 30.00	Depth (cm.)	t so t
Run No. 2. Q = 2 cfs Yn = 24,092 M = 1/8 = C L/b = 1.0 Slope = 0.0 Slope = 0.0 Temp. = 68	Cent. Sta. (ft.)	<b>8</b> 3 3
-3 	Depth (cm.)	ar god
Run No. 2- U - 2 cf Yn -24.092 M - b/8 - L/b - 0.0 Slope - 0. Temp 6 Model Sta.	Cent. Sta. (ft.)	20.0 25.0 25.0 26.5 28.0 28.0 28.5 28.5 28.5
4 9 0.491 8F - 30.00	Depth (cm.)	5 <b>8888888</b> 84 57555888884 5755555555555555555555555
Run No. 2- Q - 2 cfs Yn - 24.09m M - b/B - 4 L/b - 0.5 Slope - 0.6 Temp 68 Model 3ta.	Cent. Ste. (ft.)	22222222222222222222222222222222222222
5 0,500 3.F - 30,00	Depth (cm.)	8858885898 <b>6</b> 5 333333333388865
Run 1:0. 7 4 = 2 9 n - 24.0 N = b/B 1./b = 1.0 Slope = 0 Temp. =	Cent. Sta (ft.)	24.0 25.0 25.0 26.5 28.0 28.0 28.0 28.0 23.0 28.0 28.0 23.0 23.0 23.0 24.0
2-6 11a 72a 0.500 0.667 0.000131 0.667	. Depth (cm.)	4444444666

311 0289 30 <b>.</b> 00	Cupth (cm.)	rge d	00 289 30.00	p <b>th</b> cn.)	
Hunn No. 3-3 Q = 3 cfe Yn -24,05 cm, M = b/8 = 0. L/b = 1.0 Slope = 0.00 Slope - 76F Model Sta	<pre>Cont. Sta. (ft.)</pre>	9 <b>00 10</b> 0	Huno. 3-12 4 - 3 cfa 7n -24.05cm. M - b/B - 0.9 Slope - 0.000 Slope - 0.000 Hadal Sta	Cent. Sta. De (ft.) (	25.0 24 26.0 24 26.5 24.5 27.5 24.5 24.54 24
311 3289 30.00	Depth (cm.)	-Sed	900 30,000	Depth (cm.)	24.13 24.13 24.04 24.04 24.06 24.06 24.06 24.06 24.06
Run No. 3-2 Q - 3 cfs Yn -24.05cm. M - b/8 - 0. L/b - 0.5 Slope - 0.65 Tlemp 76R Nodel Sta	Cent. Sta. (ft.)	Sutace	Run No. 3-11 Yn -2, 05 cm, Yn -24,05 cm, N - 54,05 cm, N - 0,5 L/b - 0,5 L/b - 0,5 Fomp - 737 Fomp 54	Cent. Sta. (ft.)	20.0 22.5 22.5 22.5 22.5 22.5 23.5 23.5 23.5
1 .300 .00289 - 30.00	bepth (cm.)	a r Sed	10 	Depth (cm.)	24.12 24.09 24.09 24.10 24.08 23.99 24.08 22.33.98 24.08 24.08 24.08
Run No. 3- 4 - 3 cfa 7 - 24.05 cm 8 - 64 - 0.0 1/b - 0.0 1/b - 0.0 Slope - 0.0 Femp 76! Houdel Sta.	Cent. Sta. (ft.)	<b>19</b> 19	Run No 3- 4 - 3 cfs yn -24.05 cm yn - b/8 - 0 1, b - 0,0 51.0p - 0,0 51.0p - 76 Tomp 76	Cent. Sta. (ft.)	80.0 28.0 28.5 38.5 31.0 31.0 5 28.5 5 31.0 5 28.5 5 28.5 5 28.5 5 28.5 5 28.5 5 28.5 5 28.5 5 28.5 5 5 28.5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5
12 .900 .0131 - 30.00	Depth (cm.)	24.10 224.09 224.09 224.06 224.05 224.05 224.05	9 	Dopth (cm.)	24.33 24.35 24.38 24.34 24.31 24.33 24.33 22.42 20.33 23.80
Run. No. 2. Q 2 cfs Y <sub>11</sub> - 24.07 cm H - b/B - 0 14 <sup>4</sup> b - 1.0 Slope - 0.06 Temp 701 Temp 701	Cent. Sta. (ft.)	24.0 25.0 26.5 27.5 27.5 27.5 27.5 27.5 27.5 25.6 35.6 35.6 35.6	Run No. 3- 4 - 3 cfs Yn -24,05cm Yn - b/8 - C L/8 - 1,0 Slofe - 1,0 Slofe - 0,0 Slofe - 73 Yomp - 73	Cent. Sta. (ft.)	24.0 25.0 22.5.0 22.5.5 22.5.5 23.80 33.80 33.80
11 • 900 • 30.00	lepth (cu.)	24,09 22,06 22,06 22,06 22,06 22,06 22,06 22,06 22,06 22,07	8 • 700 • 30.00	Depth (cm.)	24, 32 24, 32 24, 32 24, 32 24, 27 24, 27 23, 75*
Run No. 2. 4 = 2 cfs Y <sub>0</sub> -24.07 cm H = b/8 = 0 L/b = 0.5 Slope = 0.0 Temp. = 70 Model Stu.	Cent. Sta. (ft.)	20.0 25.0 25.5 25.5 25.5 25.5 25.5 25.5	Run No. 3- 4 - 3 cfr Yn -24.05 cn 1/6 - 0.5 5100 - 0.5 5100 - 1.73 Temp 73	Cent, Sta. (ft.)	20.0 24.0 25.5 27.5 28.5 28.5 28.5 28.5 28.5 28.5
0 889 00131	Depth (cm.)	24,12 24,11 22,11 22,10 22,000 22,000 22,000 22,000 22,000 22,000 22,000 22,000 22,000 22,000 20,00000000	.693 00289 	Depth (cm.)	24,44 24,44 24,44 24,44 24,44 23,59 24,59 25,59
Run No. 2-1 4 - 2 cfa Yn -24.99 cm. M - b/8 - 0. L/b - 0.0 Slope - 0.0 Teap 73F Model Sta	Cent. Sta. (ft.)	20.0 28.0 28.0 28.5 28.5 31.0 31.0 32.5 32.5 32.5 32.5 32.5 32.5 32.5	Run No 3-7 4 - 3 cfa 3 cfa - 05 cm 1/b - 0/B - 0. 1/b - 0.00 Slopa - 0.00 Temp 70	Cent. Sta. (ft.)	2010 2010 2010 2010 2010 2010 2010 2010
700 00131 . 30,00	Depth (cm.)	24,12 24,15 24,15 24,15 23,91 23,91 23,91 23,91 23,91 23,91 23,91 23,91 23,91	.500 00289 - 30.00	Depth (cm.)	25.12 25.14 25.14 25.12 25.12 25.12 25.12 25.20 21.20
Run No. 2-9 ~ - 2 cfs yn -24.07 cm M - b/8 - 0. L/b - 1.0 Slope - 0.0 Tomp 70F Mudel Sta	Cent. Sta. (ft.)	24.0 24.0 25.0 24.0 24.0 25.0 25.0 25.0 25.0 25.0 25.0 25.0 25	Run No. 3-6 4 - 3 cfs 2n -24.05 cm H - b/b - 0 L/b - 1.0 Slup - 1.0 Slup - 7 Temp 7 Vodel Stu.	Cent. Sta. (ft.)	24.0 25.0 28.0 28.0 29.0 29.0 29.0
700 0131 30,00	Depth (cm.)	24.11 224.113 224.113 22.93 23.93 23.93 23.93 23.93 23.93 23.93 23.93 23.93 23.93 23.93 23.93 23.93 23.93 24.11	5 	Depth (cm.)	2, 23, 25, 25, 25, 25, 25, 25, 25, 25, 25, 25
Run No. 2-6 9 - 2 cfe Yn -24,07cm. M - b/8 - 0. L/b - 0.5 Slope - 0.00 Temp 70F Model Sta	Cent. Sta. (ft.)	20.0 28.0 28.0 28.7 5 38.0 38.0 38.0 38.0 5 38.0 5 38.0 5 38.0 5 38.0 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5	Run No. 3- 4 - 3 cf 7 -24.05c 8 - b/8 - 1/6 - 0.5 Slope - 0.5 Femp 7 Femp 7	Cent. Sta. (ft.)	20.0 24.0 28.0 28.5 28.5 28.5 28.5 28.5 28.5 28.5 28.5
 .693 .30.00	Depth (cm.)	22,22 23,99 23,99 23,99 23,99 23,99 23,99 23,99 23,99 23,99 23,99 23,99 23,99 23,99 23,99 23,99 23,99 23,99 24,22 25,22,22 25,22 25,22 25,22,22 25,22,22,22 25,22,22,22 25,22,22,22,22,22,22,22,22,22,22,22,	.491 00289 - 30,00	Depth (cm.)	25.87 25.867 25.87 25.87 25.69 22.97
un No. 2-7 - 2 cfe n - 24.09m - 5/8 - 0 /5 - 0.0 10p6 - 0.0 10p6 - 0.0 241 Sta	ent. Sta. (ft.)	20.0 24.0 25.0 25.0 25.0 25.0 25.0 25.0 25.0 25	aun No. 3-4 i - 3 cfe i - 24.05 cfa i - 24.05 - 0.0 iope - 0.0 bep: - 78: bep: 5 ta.,	ent. Sta. (fr.)	20.0 24.0 25.5 26.5 28.5 32.07 32.07

TABLE 7-2-1 RAM DATA - LARGE FLUME - ROUCH BUUNDARIES (continued) P.2

(Two and Three Dimensional Semicircular Model Teste)

TABLE 7-2-1 RAN DATA - LARDE FLUME - ROUCH BOUNDALLES (continued) P. 3

(Two and Three Ofmonsional Semicircular Model Tests)

....

9*** 0.700 1.30.584 1.30.005	Depth (cm.)	9.93 9.92 9.91 9.91 9.91 9.87 9.87	6 8 0,500 0,500 - 30,00 - 30,00 (su.)	15.02 15.02 14.87 14.87 14.87 11.96
Raun No. 4- 4 - 1 cfe 7 - 9.64 cm M - b/B - L/b - 1.0 Slope - 0.0 Temp 68 Model Sta.	(ent. Sta. (ft.)	20.0 25.5 25.5 29.5 29.5 29.5 29.5 29.5 29.5	Run -0. 5-4 4 - 2 cf. 2 - 1.28:28: 2 - 1.28:28 2 - 1.28:28:28:28:28:28:28:28:28:28:28:28:28:2	26.0 276.5 28.5 28.5 33.11 33.11
-8** -700 -30.00	Depth (cm.)	9,90 9,88 9,88 9,88 9,88 9,88 9,88 9,87 9,87	5 3 0,500 0,500 74F - 30.vu (cm.)	14.89 14.90 14.90 14.85 11.85 11.85 11.87
Run No. 4. Q 1 cfs Yn - 9.64 cm H - b/8 - 0. 1/b - 0.5 Slope - 0. Tamp 68 Model Sta.	Cent. Sta. (ft.)	20.0 22.5 22.5 22.5 22.5 22.5 22.5 22.5	Run No. 5- 4 - 2 c: 3 - 13,28 M - b/8 - L/b - b/8 - L/b - b/8 - J.p - 0, fomp, - 1 Model 54a, (ft.)	24.0 25.0 266.0 288.0 32.95
-7*** 	Depth (cm.)	9.94 9.93 9.94 9.99 9.89 9.89 9.89 9.89	-4 f3 cm. 0.4910.491 0.4	15,58 15,62 15,62 15,63 15,63 15,50 11,78
Raun No. 4 4 = 1 cfi Yn - 9,64 cc M - b/8 - 0 M - b/8 - 0 L/b - 0,0 Slope - 0,0 Temp 66 Hodel Sta.	Cent. Sta. (ft.)	20.0 25.0 25.5 25.5 29.5 29.5 29.5 29.5 29.5 29.5	Run No. 5 7 - 2 - 2 7 n -13-28 7 - b/8 0 1/0 - 0 51op - 0 51op - 0 51op - 5 7 7 Model Sta	20.0 25.0 25.0 21.5 21.5 21.9 21.9 21.9 21.9 21.9 21.9 21.9 21.9
ъ 1 2,500 8F - 30,00	Depth (cm.)	0.59 1	3 E.fa Ben. - 0.311 - 0.311 - 0.311 - 0.311 30.00 50.00 50.00 50.00 50.00 50.00 50.00 	19-32 19-35 19-35 19-32 19-27 11-22
Run No. 4 Q = 1 cf Yn = 9.64 cf M = b/8 = ( M = b/8 = 1.0 Slupe = 1.0 Teurp = 50 Mouel Sta.	Cent. Sta. (ft.)	255.5 255.5	Run No. 5 4 - 2 ( 7 - 12,52 7 - 13,52 7 - 12,0 1,0 1,0 1,0 1,0 1,0 1,0 1,0 1,0 1,0 1	25.00 26.0 28.0 28.0 28.0 28.0 28.0 28.0 28.0 28
5 • 500 • 500 • 30,000	bepth (cn.)	0100 88.001 100.001 100.001 100.000 100.00 100.00 100.00000000	-2 cm. 0.311 73F 30.00 0.5th (cm.)	18,86 18,90 18,95 18,95 18,95 18,95 18,85 18,85 8,55*
Run Mo. 4- 4 - 1 cf ya - 9.64 cn ya - 9.64 cn M - b/8 - 0 L/b - 0.5 Slope - 0.5 Slope - 0.5 Hodel Sta.	Cent. Sta. (ft.)	88888999999999999999999999999999999999	Run No. 5 4 - 2 c 2 7 - 13.28 M - b/B - 0.5 10/b - 0.5 510/b - 0.5 510/b - 0.5 10/b - 0.	20,0 25,0 28,5 28,5 28,5 28,5 28,5 28,5 28,5 28,5
ены 191 30584	Depth (cm.)	e VIII pography"	L 	21,89 21,95 21,95 21,95 21,95 22,02 22,02 21,84 8,87 8,87
Run No. 4-4 4 = 1 cfs Y <sub>n</sub> = 9.64 cm. M = b/8 = 0.4 L/b = 0.00 Slope = 0.00 Slope = 0.00 Fromp. = 62F	Cent. Sta. (ft.)	See Tabl	Run No 5-, 2 cf M-13.28 c M - b/B - L/b - 0.0 L/b - 0.0 Slope - 0, Slope - 0, Moiel - 24 Moiel	31.54 31.54
3** 1,311 200584	Depth Depth (cm.)	2.22 2.23 2.23 2.24 2.25 2.25 2.25 2.25 2.25 2.25 2.25	12 +** fs nn. 0.900 68 F 68 F 0.900 1 betth 1 betth	66.69 66.69
Run No. 4- 4 - 1 cfs y <sub>n</sub> - 9.64 cm H - b/8 - 0 L/b - 1.0 Slope - 0.6 Temp 68	Cont. Sta.	25,50 25,50 25,50 25,55 26,55 28,55 28,55 29,55 29,55 29,55 29,55 29,55 29,55 29,55 29,55 29,55 29,55 29,55 20,55	Run No. 4: 4 - 2 c 4 - 2 c 4 - 2 c 4 - 2 c 1 - 1 - 1 1 - 1 -	29.00 29.00 29.00 29.00 29.00 29.00 29.00 29.00 29.00 29.00 29.00 29.00 29.00 29.00 20 29.00 20 20 20 20 20 20 20 20 20 20 20 20 2
** 311 00584	. 30.00 Lepth	12.55 12.55	-11 ** 15 ** 0.900 0.900 1.00054 *	9.69 9.69 9.69 9.69 9.69 9.69 9.69 9.69
Run No. 4-2 4 - 1 cfe Yn - 9.64 cm. H - b/8 - 0. L/b - 0.5 Slope - 0.0 Temp 68F	Model Sta Cent. Sta.	22.0 22.0 23.0 23.0 23.0 23.0 23.0 23.0	Run No. 4 V = 1 C V = 2 C V =	20.0 254.0 254.0 265.0 265.0 265.0 265.0 288.0 288.0 299.0 294.0 294.0
+* -300 20584	- 30.00 Depth		0.889 0.889 0.889 - 30.00 - 50.00	9.65 9.66 9.66 9.66 9.66 9.65 9.65 9.65
Run No. 4-1* 4 - 1 cfo y <sub>n</sub> -9.64 cm. y <sub>n</sub> -5.64 cm. L/c - 0.0 510pe - 0.00 68F	Model Sta Cent. Sta.	(1) 2000 2000 2000 2000 2000 2000 2000	Run No.4-1 r - 1 cft yn - 9,64ct M - b/B - 0.0 510p - 0.0 310p - 0.0 7 Temp - 0 Model 3te.	23.5 23.5 23.5 23.5 23.5 23.5 23.5 23.5

.

311 0123 - 30.00	(cm.)	rge d	L2 - 9 <b>00</b> - 30,00	(	22222222222222222222222222222222222222
Run No. 6-3 Yn - 15.20m M - b/8 - 0. L/b - 1.C Slope - 0.C Slope - 7.3 Houdel Sta.	(ent. Sta. (ft.)	Sutence	Run io. 6- 4 - 3 cfs yn -15.28cm M - b/8 - 0. L/b - 1.0 Slope - 0.0 Hodel Sta. 1 Cart. Sta. 1	('u')	25.0 28.0 33.0 35.0 35.0 35.0 35.0 35.0 35.0 35
311 123 30,00	Depth (cm.)	ged	1 30,00	(cm.)	, 22, 22, 22, 22, 22, 22, 22, 22, 22, 2
$\begin{array}{llllllllllllllllllllllllllllllllllll$	Cent. Sta. (ft.)	Submor <sub>1</sub>	Ram No. 6-1 Q - 3 cf3 yn -15,28 cm. H - b/8 - 0. Kub - 0.5 Slope - 0.60 Temp 764 Hodel Sta.	(U.)	8 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2
300 1223 30 <b>.</b> 00	Depth (cm.)	0 81	. 889 - 30.00 - 30.00	Depth (cm.)	22222222222 26222222222 26222222222222
Run No. 6-1 4 - 3 cfe 4 - 3 cfe 7 - 15,28m. N - 1/8 - 0. 1/b - 0.0 51ope - 0.0 73F Temp 73F Model Sta	Cent. Sta. (ft.)	and do so that a	Run No. 6-1 4 - 3 cfa yn -15,28 cm M - b/8 - 0 1/b - 0.0 510pe - 0.0 1 famp 7 Tamp 7 Hodel Sta.	Cent. Sta. (ft.)	20.0 26.0 26.0 26.5 26.5 31.5 31.5 31.5 31.5 31.5
900 0670 30,00	Lepth (cm.)		9 • 700 • 30.00	Depth (cm.)	16.15 16.17 16.17 16.17 16.17 16.23 16.23
Run No. 5-12 Q - 2 cfs Y <sub>n</sub> -13.28m H - b/9 - 0. Ifb - 1.0 Slop - 0.00 Temp 74F	(ft.) (ft.)	25.0 28.5 28.5 38.4 38.4 5 38.4 5 38.4 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5	Run No. 6- 4 - 3 cist Yn - 15,28cs H - b/8 - 0. 1/b - 1.0 1/b - 1.0 Slope - 0.C Temp 73 Heddl Ste.	Cent. Sta. (ft.)	20.0 24.0 26.0 33.87 33.87
1 900 30,000	bepth (co.)	22.21 2.21 2.21 2.21 2.21 2.21 2.21 2.2	8 0.700 0.23	Depth (cm.)	16.05 16.05 16.06 16.05 16.03 15.98 15.83 15.83
Run Mo. 5-1 4 - 2 cf3 Y <sub>n</sub> -13.28tm, M - b/8 - 0. 1/b - 0.5 Slope - 0.0 Temp 744 Hodel Stu	Cert. Sta. (ft.)	24 25 25 25 25 25 25 25 25 25 25 25 25 25	Run No. 6- u - 3 cfu yn - 15.28= M - b/8 - L/b - 0.5 Slope - 0. Tamp 75	Cent. Sta. (ft.)	20.0 25.0 25.0 25.0 25.0 25.0 25.0 23.0 33.05
39 0870 30 <b>.</b> 00	Depth (cm.)	22222222222222222222222222222222222222	693 - 30.00	Depth (cm.)	16.33 16.37 16.37 16.37 16.33 16.33 16.33 16.33 16.33
Run No. 5-10 4 - 2 cfs Yn -13,28cm. M - b/8 -0.8 L/b - 0.0 Slope - 0.0 Tomp 72F Model Sta	Cent. Sta. (ft.)	20.0 25.0 23.5 23.5 23.5 23.5 23.5 23.5 23.5 23.5	Run No 6 4 - 3 cfa 2 - 3 cfa 2 - 2 cfa 2 - 2 - 3 2 - 2 - 2 5 - 2 - 0 5 2 - 0 1 - 1 7 3 - 7 3 Houlal Sta.	Cent. Sta. (ft.)	20.0 24.0 25.0 25.0 26.5 33.03 33.03
700 0870 30.00	Depth (cm.)	11.77 11.77 11.77 11.77 11.88 12.88 12.88 12.88 12.88	6 • 500 • 30.00	Depth (cm.)	18.35 18.37 18.38 18.38 18.34 17.98 17.98 12.15
Ram No. 5-9 4 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2	Cent. Sta. (ft.)	24,0 28,0 28,0 28,0 34,4 34,5 28,0 34,4 34,5 28,0 28,0 28,0 28,0 28,0 28,0 28,0 28,0	Run No. 6- Q 3 cfr yn -15.28 cr H b/8 - 0 1./b - 1.0 Slope - 0.0 Slope - 0.0 Temp 74 Hedel Sta.	Cent. Sta. (ft.)	22,00 22,00 23,00 23,00 23,00 23,00 23,00 23,00 23,00 23,00 23,00 23,00 23,00
30-00 30-00	Depth (cm.)	113.72 113.72 113.66 113.66 113.66 112.86 122.87 122.86	۲-5 0,500 1-30,000 1-30,000	Depth (rm)	18.15 18.19 18.19 18.19 18.19 18.19 18.19 18.19 18.19
t Run No. 5–8 t Run No. 5–8 yn −13.285m H − b/8 −0.7 1/b −0.5 1/b −0.5 1000 −0.000 1000 − 74.5 Hoddai Xta. −	Cent. Sta. (ft.)	20.0 28.0 28.0 28.5 28.5 33.5 33.5 33.5 33.5 5 33.5 5 33.5 5 28.5 5 28.5 5 33.5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5	Run No. 6 4 - 3 c: 3 -15.28 M - 5/8 - 10 - 6 S100 - 0.5 Temp - 74 Hodel Sta.	Cent, Sta.	22.00 22.00 22.50 22.55 28.0 28.0 28.0
593 3870 30.00	Depth (cm.)	12122222222222222222222222222222222222		Depth (cm.)	19.38 19.45 19.45 19.46 19.42 19.42 19.42
Run No. 5-7 yr - 2 cfa yr -13.28cm. H - 5/B - 0.6 L/b - 0.0 Sippe - 0.00 Cart. 72F Model 540	Cent. Sta. (ft.)	8.4%% % % % % % % % % % % % % % % % % % %	Run No. 6-1 Q - 3 cfa Yn -15,28cm N - b/B - 0 Li/b - 0,0 Slope - 0,1 Hodel Sta.	Cent. Sta.	2000 2000 2000 2005 2005 2005 2005 2005

TABLE 7-2-1 RAW DATA - LARGE FLUME - ROUCH BOUNDALIES (continued) P.4 (Two and Three Dimensional Semicircular Model Teets)

ь. Э TABLE 7-2-1 RAN ONTA - LANGE FILLINE - ROUCH BUNNARLES (continued) ("Do and Three Mismontonal Somicircular Model Teste)

.

- 30.00	Depth (cm.)	7.40 7.42 7.43 7.43 7.43 6.11*
Run No. 7- 2 - 1 cf. yn - 6,69 ct. M - b/B - 1. 1/b - 1.0 Slope - 0. Temp 70 Temp 70	Cent. Sta. (ft.)	24.0 25.0 26.5 28.5 29.0 34.42 34.42
.700 .9191 .30.00	Depth (cm.)	7.28 7.31 7.31 7.32 6.10 *
Run No. 7-8 Q = 1 cfa Yn -6.69 cm M - b/B - C L/b - 0.6 Slope - 0.6 Tomp 701 Nodel Sta.	Cent. Sta. (ft.)	20.0 24.0 25.5 26.5 28.5 28.5 28.5 28.5 28.5 28.5 28.5 28
-7 0.693 - 30.00	Depth (cm.)	7.41 7.45 7.45 7.45 7.45 7.45 7.45 5.18
Run No. 7. Q = 1 cfe y <sub>n</sub> - 6.79 cm M = b/8 - 0 L/b = 0.0 Slope - 0. Temp. = 700 Nodel Sta.	Cent. Sta. (ft.)	20.0 25.5 26.5 31.55 31.52 31.52 31.52
• • • • • • • • • • • • • • • • • • •	Depth (cm.)	8,63 8,64 8,70 8,8,68 5,8 5,94 8,8 5,8 5,94
Run Mo. 7-6 q - 1 cis yn - 6.69 cm M - b/8 - 0 Ly <sup>4</sup> b - 1.0 Slope - 0.0 Temp 70 Kodell Sta.	Cent. Sta. (ft.)	24.0 26.5 28.5 29.5 29.5 28.5 29.5 28.5 29.5 28.5 28.5 29.5 28.5 28.5 28.5 28.5 28.5 28.5 28.5 28
• 500 • 500 • 30•00	Depth (cr.)	88888888888 4,5558888888 5,55888 5,528 5,5
Run Mo. 7-5 4 - 1 cia 7 - 5,79 cm 7 - 5,79 cm M - b/8 - 0 1/b - 0.5 Slope - 0.0 Temp 70 Temp 70	Cent. Sta. (ft.)	20.0 28.0 28.0 28.5 28.5 28.5 28.5 28.5 28.5 28.5 28.5
1611 1910 1910 1910	Depth (cm.)	8.91 8.92 8.93 9.00 5.54 5.57 5.57 5.57 5.57 5.57 5.57
Run No. 7-4 u - 1 cfs y <sub>n</sub> - 6,79 cm N - b/8 - 0 N - b/8 - 0 L/b - 0.00 Slope - 0,00 Temp 69 Hodel Sta.	Cent. Sta. (ft.)	20.0 22.0 22.5 22.5 22.5 22.5 22.5 22.5
, 311 , 311 , 30,00	Depth (cm.)	11.17 11.28 11.28 11.28 10.21 5.37 10.21 5.37
Run No. 7-3 4 - 1 efe Yn -6.69 cm M - b/8 - 0 M - b/8 - 0 L/b - 1.0 Slope - 0.0 Temp 70 Temp 70	Cent. Sta. (ft.)	25.0 26.0 28.0 28.0 29.5 29.5 29.5 29.5
311 0191	Depth (cm.)	10.25 10.43 10.45 10.55 10.57 5.16* 5.16*
Run No. 7-2 q - 1 cfe Yn - 6.69 cm M - b/8 - 0. L/b - 0.5 Slope - 0.5 Slope - 70F Tearp 70F	Cent. Sta. (ft.)	20.0 24.0 28.0 28.5 28.5 28.5 28.5 28.5 28.5 28.5 28.5
- 30,00	Depth (cm.)	22.23 22.48 22.55 22.55 22.55 5.46 5.46 5.46
Run Mo. 7-1 4 - 1 cfe 7 - 6.79 cm 1 - 5/8 - 0. 1 - 5/8 - 0.0 1 - 0.0 1 - 0.0 1 - 0.0 1 - 0.0 1 - 70	Cent. Sta. (ft.)	20.0 28.0 28.5 38.5 38.5 38.5 39.5 39.5 39.5 39.5 39.5 5 39.5 5 39.5 5 39.5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5

8-6 cf3 6 cm. - 0,500 0,00231 76F a 30,00	e. Dupth (cm.)	2.51 5.51 5.51 5.51 5.51 5.51 5.51 5.51
Run No. 4 - 2 Yn - 9.64 M - b/B L/b - 1. Slope - Temp Model Ste	Cent. St. (ft.)	24.0 25.0 26.5 26.5 28.0 28.0 35.42 35.42
500 500 - 30.00	Depth (cm.)	0.2222222 0.222222 0.2222222
Ràm No. 3-5 Q - 2 cfa Q - 2 cfa M - b/8 - 0. M - b/8 - 0. L/b - 0.5 Slope - 0. Tempe - 76 Tempe - 76	Cent. Sta. (ft.)	24.0 25.0 26.0 28.0 28.5 28.6 33.00
.491 5F - 30.00	Depth (cm.)	13.69 13.69 13.69 13.69 13.73 13.73 13.73 14.61 15.73 14.63 15.73 15.73 15.73 15.73 15.73 15.73 15.73 15.75 15.65 15.75
Run No. 8-4 9 - 2 cfs 7 - 9,66 cm M - b/B - 0 1,b - 0,0 51 op - 0,0 51 op - 0,0 1,7 amp - 7,7 Tamp - 7,7 Model Sta.	Cent. Sta. (R.)	20.0 28.0 28.0 32.65 32.
	Depth (cm.)	18,95 19,00 19,00 19,00 18,96 18,85 18,82 5,77*
Run No. 8- 4 - 2 cfs yn - 9.66 cm y - b/R - 0, L/b - 1,0 Slope - 0,0 Slope - 777 Temp, - 777	Cent. Sta. (ft.)	25.0 26.0 276.5 28.5 28.5 33.51 33.51
2 • 311 77 - 30.00	Depth (cm.)	17.89 18.19 18.36 18.47 18.40 18.42 18.42 5.66 5.66
Run No. 8. 4 - 2 cft Yn - 9.66 cm Yn - b/B - 0 L/b - 0.5 Slope - 7 Tempe - 7 Model Ste.	Cent. Sta. (ft.)	20.0 255.0 255.0 258.0 258.0 258.0 258.0 258.0 258.0 258.0 258.0 258.0 258.0 258.0 258.0 258.0 258.0 258.0
300 0231 F _30,00	Depth (cm.)	20.31 20.55 20.57 20.72 20.72 4.29
Run No. 8-1 2 - 2 - 2 - 2 3 - 4 - 5 - 6 3 - 10 - 0, 1/b - 0, 510pe - 0, 10 - 0, 10 - 0, 10 - 10, 10 - 10,	Cent. Sta. (ft.)	339.0 300.0 300.00
2 1,900 0191 - 30,00	Depth (cm.)	6,85 6,85 6,88 6,88 6,88 6,89 88 88 88 88 88 88 6 6 6 88 88 6 88 6
Rhun No. 7-] Q ] cfa Yn - 6.69 cm Yn - b/8 - ( 1,b - 1.0 Slope - 0.0 Tampe - 70 Model Sta.	Cent. Sta. (ft.)	22,000 20,000 20,00000 20,0000 20,0000 20,0000 20,0000 20,0000 20,0000 20,0000 20,00000000
1 0,900 191 - 30,00	Depth (cm.)	
Run No. 7-1 4 - 1 cfr 3 - 6.65 cm 1 - b/B - 1 - b/B - 1 - b/B - 1 - 0.5 51 op - 0.5 7	Cent. Sta. (ft.)	4 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2
0 .889 .1910 30.00	Depth (сш.)	6.55 6.53 7,73 7,73 7,73 7,73 7,73 7,73 7,73 7,
hn No. 7-1 1 - 1 cfe n - 6.79 cm n - 6.79 cm n - 5/8 - 0. /b - 0.0 inpe - 0.0 inpe - 0.0 inpe - 0.0	ent. Sta. (ft.)	20.0 22.0 22.0 22.0 22.0 22.0 22.0 22.0

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(continued)
BOUNDARJES
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(Two and Three Dimensional Semicircular Model Taste)

Ran Mo, 8–7 Ran I 1. 2 Ceta 9. 7 2 Ra - 9,66 cat, 9	Cent. Ste. Depth Cent. (ft.) (cm.) (ft	2.2.0 2.	Мал №, 9-4 Run 4 - 3 сге ч 71 - 11,59а. л. 11,59а. л. 11,6 - 0,0 1,91 М
Mo. 8-8 2 cfs 9,66cm. 7/8 - 0,700 7/8 - 0,700 1 - 0,00231 1 - 7/7 5ts 30,00	Sta Depth t.) (cm.)	0.10,72 0.00 0.0	i No. 9-5 - 3 cfa -11.59cm. - 0.5
Run No. 8-9 Q. 2 cfe Yn -9.66 cm. Yn - b/B - 0.' X - b/B - 0.' X - b/B - 0.' Slope - 1.0 Slope - 777 Temp. 777	Cent. Sta. (ft.)	25.0 284.5 284.5 35.1 35.1	Run No. 9-4 Q - 3 cfs Yn 11.59 cm M - b/8 - 0 L/b - 1.0
700 231 8-00	Depth (cm.)	11,00 111,04 11,02 10,98 10,95 8.45*	2.200
Run No. 8-10 2 - 2 cfa 2n - 9.66 cm. M - b/B - 0.6 1/b - 0.0 Slope - 0.002 1 Pamp 77F Famp 77R	Cent. Sta. (ft.)	20.0 24.0 28.5 28.5 28.5 28.5 28.5 28.5 28.5 28.5	Ran No 9-7 4 - 3 cfe M - b/8 - 0.0 L/b - 0.0
0 889 30,00	Depth (cm.)	9.73 9.73 9.75 9.75 9.55 9.55 9.55 9.55 9.55 9.55	
Run Mo. 8-1 9. 2 cf3 7. 9.66 cm. M - b/8 - 0. 1./b - 0.5 Slope - 0.5 Temp 77F Model Stu	Cent. Sta. (ft.)	24.0 2550 28.5 28.5 28.5 28.5 28.5 28.5 28.5 28.5	Run No. 9-6 4 3 cfo 7 - 11,52m M - b73 - 0 L/b - 0,5
. 00 30,00 30,00	Lepth (cm.)	9.81 9.76 9.67 9.65 9.65 9.45*	3.700
Run No. 8-J Yn - 2 cfs Yn - 9.66 cm M - b/8 - 0. I/b - 1.0 Slope - 0.0 Temp 77 Wodel Sta.	Cent. Sta. (ft.)	24.0 25.0 28.5 28.5 28.5 28.5 28.5 28.5 28.5 28.5	Run No. 9- ч - 3 cfs ул -1.59 cf M - b/B - 1.0
	Depth (cm.)	9.72 9.75 9.79 9.79 9.78 9.78 9.78 9.78	9 0.700
Run No. 9-1 Q = 3 cfe Yn -11.59cfe Yn -11.59cfe Yn -0,8 L/b -0,0 Slope -0,00 Slope -0,00 Slope -0,00	Cent. Ste. (ft.)	n n N	Run No, 9- 4 - 3 cfs 91 - 15 cfs 11-59 cu 11/b - 0,0
200 100 12 12 12 12 12 12 12 12 12 12 12 12 12	Dupth (cm.)	5 8 8	10 1889
Run Ho. 9–2 Q. = 3 cfa Yn -11.59cn. H. = b/8 - 0.5 L/b - 0.5 Slope - 0.0 Temp 72P	Cent. Sta. (ft.)	Subac	Run No. 9-1. J 3 cfs M b/8 - 0.5 L/b - 0.5
118 1920,00 1920,00	Depth (cm.)	erged	п. <sup>06</sup>
Run No. 9- q - 3 cf Yn -11.59 M - b/8 - 0. Slope - 0. Tamp 0. Tamp 0.	Cent. Sta. (ft.)	ал 9	Run ioo, 5/н: ч - 3 сf3 Ул -11,59 ст8 Н - b/8 - 0, L/b - 1,0
-3 0,311 72F - 30,00	Dept (cm	tear gad	900

Run No. 9-4 9 - 3 cfe 9 - 11.52a. 9 - 12.52a. 1/b - 0.0 210pe - 0.00 7amp 721 Model Sta	(n.) (f)	20.0 25.0 26.0 26.5 28.5 28.5 28.5 28.5 28.5
491 294 30.00	Depth (cm.)	16.86 17.17 17.28 17.28 17.28 17.28 17.14 17.14
Run No. 9- U = 3 cf Fn = 11.59 cn H = b/8 = 0 L/b = 0.5 Slope = 0.6 Toup. = 73 Model Sta.	Cent, Sta. (r.)	20.0 25.0 25.0 28.0 28.0 28.0 28.0 28.0 28.0 28.0 28
-5 a 2,500 20294 18 - 30,00	Depth (cm.)	17.28 17.28 17.41 17.45 17.45 17.45 17.45
Run No. 9. 2 - 3 ci yn 11.59 c M - b/8 - 1/b - 1.0 Slope - 0. Slope - 73 Hedel Sta.	Cent. Sta. (ft.)	25.0 28.5 328.5 33.68 33.68
13 13 0.500 37 37 37 30.00	Depth (cm.)	17.59 17.68 17.70 17.68 17.58 17.39 7.47*
Run No 9- 4 - 3 cf. Xh -11,89 ci Xh - 11,8 - 0 K - 11,8 - 0 L/b - 0,0 Slope - 0,0 Slope - 0,0 Houel Sta.	Cent. Sta. (Ct.)	22.50 22.50 22.50 22.55 25.55
7 •693 37 - 30,00	Depth (cm.)	2222222222 8888288 8888288
Run No. 9 K - 3 ci Yn - 11.59 M - b/8 - L/b - 0.5 Slope - 0.5 Tomp 1 Model Sta.	Cent. Sta. (ft.)	25,0 26,5 27,5 285,0 286,0 286,5 286,5 286,5 33,12
не Ге ст. 0.700 13F 73F - 30.00	Depth (cm.)	12.52 12.52 12.52 12.52 15.555
Run No. 9. 4 - 3 ci yn -11.59 c yn -11.59 c Model 244. Model 344.	Cent. Sta. (ft.)	20.0 24.0 25.0 25.5 26.5 28.5 28.5 28.5 28.5 28.5 28.5 28.5 28
-9 "a" 0.700 .00294 3F - 30.00	Depth (cm.)	49.51 49.51 49.51 59.555 59.5555555555
Run No, 9 2 - 3 cf yn -11,59 c N - b/B - ( L/b - 0.0 Slope - 0.0 Temp 7 Model Sta	Cent. Std. (ft.)	20.0 26.0 26.5 26.5 276.5 276.5 31.30
-10 :8 0,889 0,889 - 30,00	Depth (cm.)	11.72 11.79 11.79 11.81 11.78 11.79 11.76
Run No. 9- 2 - 3 cfi yn -11.9 ca N - b/8 - 0 L/b - 0.5 Slope - 0.0 Slope - 73 Kodel Sta.	Cent. Sta. (ft.)	20.0 25.0 26.5 28.5 28.5 28.5 28.5 28.5 28.5 32.5 32.5 32.5 32.5 32.5 32.5 32.5 32
11 •900 •900 - 30,00	Depth (cm.)	111111111111 22288888888888888888888888
Run 104 9 4 - 3 c 7 - 11.59 8 - 1.0 1./b - 1.0 510ps - C. 10mp	Cent. Sla. (ft.)	25.0 265.0 265.0 265.5 27.5 28.5 28.5 28.5 29.0 33.73
-12 fs 0.900 0.900 745 745 0	Lepth . cn. )	11.78 11.80 11.80 11.80 11.77 11.77 11.75 11.75

TABLE 7-2-1 RAW DATA - LARDE FLURE - ROUGH BOUNDARLES (continued) P.7

(Two and Three Dimensional Semicircular Model Tests)

$ \begin{array}{cccccccccccccccccccccccccccccccccccc$	Depth (ent. Sta. Dep (cm.) (ft.) (ca	614 614 614 614 614 614 614 614
Run No. 10-8 4 - 1 cfs Yn -5.35 cm. M - b/8 - 0.700 L/b - 0.5 Slope - 0.0040 Temp 70F Model Sta	Cent. Sta. (ft.)	5 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2
⊷7 0.693 00408 F 30.00	Dupth (cm.)	6.20 6.44 6.44 6.447 6.47 6.43 6.61 6.61 4.70*
Ann No. 10 4 - 1 cf( 7 - 5-35 cf 7 - 5-35 7	Cent. Sta. (ft.)	20.0 24.0 26.5 26.5 27.5 28.6 29.6 33.00
н 0.500 0.500 0.60 - 30.00	Depth (cm.)	7.47 7.98 7.98 8.07 8.23 8.23 8.23 8.23 4.19*
Run No. 10 4 - 1 cf 7 - 5.35 ci 7 - 5.35 ci 8 - 5.45 8 - 1.0 5 - 1.0 5 - 1.0 5 - 1.0 1 - 1.0 5	(ft.)	20.0 254.0 254.0 26.5 275.5 28.5 28.5 28.5 28.5 28.5 28.5 28.5 2
	bepth (cu.)	7.52 7.536 7.75 7.75 7.75 7.75 7.75 7.75 7.75 7.7
Run Ho. 10. 4 - 1 cfs Yn -5.35 cm M - b/B - ( 1/b - 0.5 Slope - 0.6 Tamp 711 Model Stu.	Cert. Sta. (ft.)	20.0 22.0 22.5 22.5 22.5 22.5 22.5 22.5
-4 1,491 5,408 - 30.00	Depth (cm.)	7.71 8.08 8.108 8.41 8.42 8.42 9.79
Fun No. 10 - 1 cff Yn - 5.35 cff M - b/B - 0 L/b - 0.0 Slope - 0.0 Temp 700 Stat.	Cent. Sta. (ft.)	20.0 24.0 26.0 26.5 26.5 28.5 29.5 28.5 29.5 32.70 32.70
	Lepth (cm.)	11.11.11.11.11.11.11.11.11.11.11.11.11.
Run No. 1 < 1 cf y <sub>n</sub> +5.35 cl N - b/B - L/b - 1.0 Slope - 0. Temp 70 Model Sta.	Cent. Sta. (ft.)	26.5 27.5 28.6 28.6 29.5 33.78 33.78
 - 	Depth (cm.)	9.63 10.07 10.031 10.34 10.35 10.25 10.25 10.25 10.25 10.25
Run No. 10 4 - 1 cfs Yn - 5.35cm M - b/8 - ( L/b - 0.5 Slope - 0.6 Temp 70 Model Sta.	Cent. Sta. (ft.)	20.0 26.0 26.5 26.5 28.5 29.5 28.5 29.5 29.5 22.70
1 .300 54,08 - 30,00	Depth (cm)	11.22 11.64 11.65
Run No. 10- 4 - 1 cfs 7n - 5.3 Sam. M - b/8 - 0. 1/b - 0.0 Slppe - 0.0 Slppe - 0.0 Xodel Sta	Cent. Sta. (ft.)	88888888888888888888888888888888888888

3		
11-6 efs 9.cm. - 0.500 6.00534 4 30.	4. Depth (cn.	12.66 12.85 12.33 12.47 4.47
Run loo. $y_n = 7.4$ $y_n = 7.4$ $y_n = 1.6$ $y_n = 1.6$ $y_n = 1.6$ $y_n = 1.6$ $y_n = 1.6$ $y_n = 1.6$	Cent. Ste (ft.)	28.5 28.0 28.0 29.0 34.9]
-5 0,500 534 - 30,cu	Certh (cr.)	22.22 22.22 22.23 23.23
Run No. 11. 4 - 2 cfs Yn - 7449 cm H - b/B - ( L/b - 0.5 510pe - 0.5 510pe - 0.0 femp 801 femp 801	Cont. Sta. (ft.)	24.0 25.0 25.5 26.5 28.5 33.6 5 33.6 5
	Depth (cm.)	11.89 12.4.88 12.4.88 12.02 13.02 13.02 13.02 13.02 4,33*
Run No. 11 - 2 cfs yn - 7.49 cm N - b/B - L/b - 0. Slope - 0. Temp. 78F Yodel Stá.	Cent Sta. (ft.)	20.0 24.0 26.0 26.0 26.0 26.0 27.5 27.5 27.5 27.5 27.5 27.5 27.5 27.5
+3 6,311 0534 - 30,00	Deptil (cm.)	19,28 19,55 19,58 19,58 3,48 3,48
Run No. 11. $u = 2 c_{13}$ $u = -2 c_{13}$ u = -1.49 cm M = -1.0 Slore = 0.00 Slore = 0.00 Temp = 80. Notel Sta.	Cent, Sta. (ft.)	27.0 28.5 28.5 29.5 34.67 34.67
-2 0,311 0534 0534	Depth (cm.)	18.48 18.65 18.65 19.04 19.00 19.09 19.03 3.77*
Run No. 11 4 - 2 cfs yn - 7.49 un M - b/8 - L/b - 0.5 L/b - 0.5 Temt 8 Temt 8 Moulel Sta,	(ent. Sta. (ft.)	24.0 26.0 28.5 28.5 28.5 28.5 28.5
1 • 300 • 30•00	Depth (cm.)	18.97 19.65 19.55 20.05 20.18 20.19 3.43*
Run No 11- N - 2 cfs N - b/8 cm. N - b/8 - 0. L/b - 0.0 Slope - 0.0 Temp 78F Temp 78F	Cent. Sta. (ft.)	20.0 254.0 26.6 26.6 26.6 28.6 23.6 33.25 33.25
+12 0.900 04.08 05.00	Depth (cm.)	5, 71 5, 78 5, 79 5, 78 5, 78 5, 88 5, 88 4, 97 *
Run No. 10 4 - 1 cfs Yn - 5.35 cm Yn - 5.35 cm Yn - 1.0 1/b - 1.0 Slope - 0.0 Iwbell Sta.	Cent. StA. (ft.)	25.0 26.5 28.5 28.5 28.5 28.5 29.5 29.5 35.0 35.0
-11 -900 -900 - 30,00	lepth (cm.)	**************************************
Run No. 10 U = 1 cf: Fn = 5.35 cm N = b/B = 0 I/b = 0.5 Slope = 0. Tomp. = 7 Hodel Sta.	Cent. Sta. (ft.)	20.0 22.0 26.5 28.5 28.5 28.5 29.5 29.5 29.5 29.5 29.5 29.5 29.5 20.5 20.5 20.5 20.5 20.5 20.5 20.5 20
⊷10 a. 0,889 20,08 99F	Der th (cm.)	**************************************
Rur. No. 10 4 - 1 cf: 2n - 5,35cm M - b/B - -/b - 0.0 -/b - 0.0 1 emp - 0.0 1 emp + 6 Model 3ta.	" "". Sta. (11.)	20° 20° 20° 20° 20° 20° 20° 20° 20° 20°

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(continued)
BOUNDARIES
ROUGH
- 34
BOB PU
1
A DATA
wa   - 2
Z-7 Z
TAB

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(Two and Three Dissuational Samicircular Modol Tests)

hun No. 12-3 2 - 3 cfo 7 - 8.70 ca. N - V8 - 0.311 L/0 - 1.0 3109 - 0.007µ8 3109 - 0.007µ8 3640 - 320 Model Sta 20 00	Cent. Sta. Depth (ft.) (cm.)	Jum ged
22 8. 0.301 - 30.00	Depth (cm.)	0 ° 0
Run No. 1 q - 3 cf yn -8.70 c M - b/8 - L/b - 0.5 Slope - 0. Temp 72. Modell Sta.2	Cent. Sta. (ft.)	8
12-1 fs ca. 0.300 0.300 2748 27.00	Depth (cm.)	n Fged
Run No. 9 - 8.70. 9 - 8.70. 1.76 - 0.0 Slope - 0.0 Slope - 7. Model Sta.	Cent. Sta. (ft.)	<b>9</b> 7 8
11-12 15a cm. 0.900 0.900 78F 78F	. Depth (cm.)	8.05 7.97 7.97 8.20 8.17 8.17 7.25 8.15 7.25
Run No. Q = 2 c Y <sub>n</sub> -7.49 H = 0.8 - Ly <sup>4</sup> = 1.0 Slope = 0 Slope = 0 Temp. = -	Cent. Sta. (ft.)	24.0 255.5 276.5 28.0 28.0 28.0 28.0 28.0 28.0 28.0
1-11 15 0.900 0.900 0.900 78F 78F - 30.00	Lepth (cm.)	7.93 8.00 8.02 7.99 7.99 7.95 7.95 7.95
Run Mo. 11 Q 2 C. Ya7.49 H. + b/8 - L/b - 0.5 Slope - 0.5 Temp 76.	Cent. Sta. (ft.)	24.0 25.0 26.5 28.5 29.0 28.5 29.0 28.5
L1-10 f3 0.889 0.889 0.889 0.889 0.889 57 - 30.00	Depth (cm.)	7.80 7.91 7.93 7.93 7.95 6.33 6.33 6.33
Run No. 4 - 2 c Yn -7.49 - H - b/8 - L/b - 0.0 Slope - 0 Slope - 0 Slope - 71 Hodel Sta	Cent. Sta. (ft.)	20.0 24.0 24.0 27.5 27.5 28.5 28.5 28.5 28.5 28.5 28.5 28.5 28
ts ts 0.700 0.700 187 - 30.00	Depth (cm.)	10.01 10.20 10.20 10.20 5.45 5.45
Run Mo. Q - 2 c Yn -7.49 N - b/8 - L/b - 1.0 Slope - 0 Teap 7 Hodel Sta.	Cent. Sta. (ft.)	26.0 276.5 28.6 35.26 35.26
11-8 8. 0.700 0.700 - 30.00	Depth (cm.)	9.45 9.45 9.45 9.48 9.48 88.8 9.48 88.8 9.48 88.8 9.48 88.6 2.48
Run H0. 1 4 - 2 c1 7 - 7.49 c H - b/8 - L/b - 0.5 Slope - 0. Slope - 0. Flemp E	Cant. Sta. (ft.)	20.0 22,5 33,5 33,5 33,5 33,5 33,5 33,5 33,5
1-7 1-7 1534 87 - 30,00	Depth (cm.)	9.01 9.99 9.99 9.72 9.72 9.72 7.99 9.72 7.99 9.72 7.99 9.72 7.73 7.73 7.73 7.73 7.73 7.73 7.73 7
Run Ho. 1 2 - 2 ci 7 - 7.49 c H - b/B - L/b - 0.0 315 p 0.00 315 p 0.00 1 32 s.	Cent. Ste. (ft.)	22.00 22.00 22.00 23.00 23.00 23.00 23.00

. Eepth (cm.)	9.61 9.63 9.63 9.60 9.72 7.75
Cent. Sla (ft.)	26.0 28.5 28.5 28.5 28.5 28.5 28.5 28.5 28.5
Lepth (cm.)	999999999999999999 88999999999995 86682882899995
Cont. Sta. (ft.)	200 2500 2500 2500 2505 2505 2505 2505
Depth (cm.)	9.27 9.20 9.40 9.41 9.41 8.041 8.041 8.041 8.041
Cent Sta. (ft.)	240.0 24.0 25.0 25.0 26.0 26.5 276.5 28.5 28.5 28.5 31.93
Depth (cm.)	12.77 12.85 12.83 12.73 6.00*
Cent. Sta. (ft.)	26.5 27.5 28.0 29.0 35.48 35.48
Depth (cm.)	11.11 11.11 11.12 11.13
Cent. Sta. (ft.)	20.0 27.0 28.0 28.0 28.0 29.0 29.0 29.0 29.0 29.0 29.0 29.0 29
Depth (cm.)	11122222222 111222222222 111222222222
<b>Cent</b> , Sta. (ft.)	20.0 25.0 25.0 26.5 28.0 28.0 23.33 33.33
Depth (cm.)	17.26 17.26 17.25 17.25 17.14 4.55*
Cent. Sta. (ft.)	26.5 27.5 28.0 28.0 29.0 36.48 36.48
Depth (cm.)	15.% 16.33 16.52 16.52 16.52 16.52 16.52 16.52 16.53 16.55
Cent. Sta. (ft.)	20.0 22.0 22.0 22.0 22.0 22.0 22.0 22.0
Derth (cm.)	14.75 15.66 16.33 17.34 16.33 17.34
ent. Ste. (ft.)	20.0 24.0 26.5 28.5 28.5 28.5 23.59
	ent. Sta. Depth Cent.

TABLE 7-2-1 RAW DATA - LARGE FLUDE - ROUCH BOUNDARLES (continued) P. 9

(Two and Three Dimensional Semicircular Model Tests)

-9 -700 - 30,00	Depth (cm.)	3,55*
Run No. 13- Q = 1 cfa Ya - 1,12 cm M = b/B - 0, 1/b - 1,0 Slope - 0,0 Tamp 801 Nodel Sta	Cent. Sta. (ft.)	28,00 35,08
	Depth (cm.)	5.95**** 3.38*
Run No. 13 Q = 1 cfs Yn - 4.12cm M - b/8 - 0 L/b - 0.5 Slope - 0.0 Temp 80 Nodel Sta.	Cent. Sta. (ft.)	27.90 33.51
-7 	Depth (cm.)	4-22 4-63 5-88 3-24 3-29
Run No. 13 G - 1 cfc Yn - 4.12 cm H - b/8 - 0 L/b - 0.0 Slope - 0.0 Slope - 78 Hodel Sta.	Cent. Sta. (ft.)	20.0 25.0 25.0 26.5 28.0 33.02 33.02
, 500 5112 91.20	Uepth (cm.)	7.50**** 3.09*
Run No. 13-6 4 - 1 cfs Yn - 4.12 cm H - b/8 - 0 14 <sup>6</sup> - 1.0 Slope - 0.0 Temp 8 Houel Sts.	Cent. Sta. (ft.)	29,00
-5 -500 - 30,00 - 30,00	Depth (cm.)	7.28**** 3.15*
Run No. 13- 4 - 1 cfs yq - 4,12 cm H - b/8 - 0, L/b - 0,5 Slope - 0,0 Temp 801 Model Stu.	Cent. Sta. (ft.)	28,50 33.65
4 491 112 30,00	Depth (cm.)	7,90**** 2.83*
Run No. 13- 4 - 1 cfs yn - 4.12 cm. M - b/8 - 0. 1/b - 0.0 Slope - 0.0 Teny 805 Model Sta: -	Cent. Sta. (ft.)	28. <b>81</b> 33.50
-3 -311 -30,00 -30,00	Depth (cm.)	11,08**** 2,66*
Run No. 13. 4 - 1 cfe yn -4.12 cm N - b/8 - 0. L/b - 1.0 Slope - 0. Slope - 0. Model Sta.	Ceat. Sta. (ft.)	28, 80 34, 64
2	bepth (cm.)	10, 77 <del>****</del> 2,52*
Run No. 13 Q - 1 cfo Yn - 4.12 cm H - b/B - 0.5 L/b - 0.5 Slope - 0.0 Temp 801 Hodel Sta	Cent. Sta. (ft.)	28.62 33.28
1 0,300 1112 - 30,00	Depth (cm.)	2,23,84,74,74,74,74,74,74,74,74,74,74,74,74,74
Run No. 13. 4 - 1 cfa Yn - 12 cfa M - 5/8 - 0.0 1/b - 0.0 51ppe - 0.0 Tamp 1	Cant. Sta. (ft.)	25.00 25.00 25.00 25.00 23.05 23.05 23.05 23.05 23.05 23.05 23.05 23.05 23.05 23.05 23.05 25 25 25 25 25 25 25 25 25 25 25 25 25

				0.00	10	Part 41 11		- N 11-		Thus No. 71		Dave No. 12.	7-	Barn Mar 115		Due to	
Nun no. 1 of	2	U - 1 cfs	1	0 - 1 cfa	4	4 - 2 cfa		4 - 2 cls		v - 2 cfs		4 - 2 cfs	,	Q - 2 cfs		4 - 2 c	
Yn -4.12 cm.		Zn -4.12 cm.		Yn - 4.12cm.		yh- 6.02 cm.		yn - 6.02cm.		yn - 6.02cm.		yn - 6.02cm		yn - 6.02cm.		yn - 6.02	ш.
H - b/B - 0.	889	N - b/8 - 0.	900	H - b/B - 0.	.900	N - b/B - 0	300	M - b/8 - 0.	.311	M - b/B - C	311	M = b/B = 0	.491	M - b/B - 0.	200	H - b/B -	0.500
L/b - 0.0		L/b - 0.5		1/b - 1.0		L/b - 0.0		L/b = 0.5		L/b - 1.0		1/p - 0.0		L/b = 0.5		L/b - 1.0	
Slope - 0.(	211	Slope - 0.01	21	Slope - 0.(	2110	Slope - 0.01	23	Slope - 0.0	123	Slope - 0.01	-23	Slope - 0.0	123	Slope - 0.01	23	Slope - 0.	0123
Model Sta.	30.00	Nodel Sta	30*00	Model Sta	- 30,00	Model Sta	30,00	Model Sta, -	- 30.00	Model Sta	30,00	Model Sta.	00.00	Nodel Sta	30,00	Model Sta.	- 30,00
Cent. Sto. (ft.)	Depth (cm.)	Cent. Sta. (ft.)	Depth (cm.)	Cent. Sta. (ft.)	Depth (cm.)	Cent. Sta. (ft.)	Depth (cm.)	Cent. Sta. (ft.)	Depth (cm.)	Cent. Sta. (ft.)	Depth (cm.)	Cent. Sta. (ft.)	Depth (cm.)	Cent. Sta. (ft.)	Depth (cm.)	Cent. Sta. (ft.)	Durth (cm.)
27.83 32.86	4. 55**** 3.84*	22.77	4. 58**** 3.32*	34.60 34.60	4. 55**** 3.97*	22,20 20,20 20,2	16.65 18.17 19.55 19.55 19.55 19.58 19.58 19.88 19.88 19.88 19.88	28.90 34.55	16.81**** 3.31*	28,92 35,16	17,60**** 3.25*	24.0 25.5 28.5 28.5 28.5 28.5 28.5 28.5 28.5	10,98 11,39 11,69 11,69 12,45 12,45 12,44 12,48 12,48 12,48 12,48 12,48	28,88 34 <b>.</b> 04	11.75**** 3.55*	35,30	J. 90*
TABLE 7-2-1 RAW DATA - LARGE FLUME - ROUGH BOUNDARLES (continued) P.10

(Two and Three Dimensional Semicircular Model Tests)

30°00 30°00	Depth (cm.)	7.03*** <sup>+</sup>
Run No. 14- Q = 2 cf3 Yn - 6.02 cm. M - b/B - 0.5 14b - 1.0 Slope - 0.012 Temp 74F Wodel Sta	Cent. Sta. (ft.)	29.38 36.50
11 • 900 123 30 <b>.</b> 00	Depth (cn)	6.95**** 5.13*
Run No. $14-$ q - 2 cfs $y_n - 6.02$ cm M - b/B - 0 L/b - 0.5 Slope - 0.0 Temp $74F$ Model Sta	Cent. Sta. (ft.)	29 <b>.</b> 34 32 <b>.</b> 86
10 • 889 122 30,00	Depth (cm.)	6.69****
Run No. $14-1$ $1 - 2 cf_3$ $y_1 - 2 cf_3 cf_3$ $y_1 - 6.02 cm$ . M - b/B - 0.0 1/b - 0.0 Slope - 0.0 Slope - 744 Hodel Sta	Cent. Sta. (ft.)	27.31 32.57
700 30 <b>.</b> 00	Depth (cm.)	9.40**** 4.42*
Run No. 14-9 Yu - 6.02 cm. Yn - 6.02 cm. M - b/B - 0. 1/b - 1.0 Slope - 0.01 Temp 76F Mouel Sta	Cent. Sta. (ft.)	28.73 36.10
30°0 30°0	Depth (cm.)	9 <b>.1</b> 3**** 4.23*
Run No. 14-8 q - 2 cfs Yn - 6.02 cm. M - b/B - 0.7 L/b - 0.5 Slope - 0.01 Temp 76F Model Sta	Cent. Sta. (ft.)	28.20 34.5 <b>9</b>
\$93 23 30,00	Depth (cm.)	9.01**** 4.56*
un No. 14-7 l - 2 cfs r - 6.02 cm. (1 - 1/B - 0.t (2 - 0.0 3100 - 0.01 3100 - 0.01 4 - 74F	Cent. Sta. (ft.)	28.82 33.10

Notes:

Only the minimum depth was measured on the downstream side of the bridge model. \*

Runs for which no measurements were taken on the downstream side of the bridge model. Ŧ

\*\*\* Runs for which a detail of the regain curve is given in Table VIII.

Runs for which only the maximum and minimum depths were read and recorded. No other measurements were taken. \*\*\*\*

SEOMETRY IGAND Ib

VABLE 7-2-2 CALCULATIONS FOR THE LARGE FLIME BOUGH BOUNDARY MODEL TESTS

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Measured Data

Calculated Data

17. (F1.) 8 J. 22 (ft.) ÷۲4 20 (F<sub>R</sub>/H13/3 18.1 216.0 1891 1891 1891  $\frac{19}{n^{1/y_{n}}}$ 81 O 0.4.5.4.0 0.4.5.4.5.0 0.6.6754 0.6.6754 0.6.6754 0.6.6754 0.6.6754 0.6.6754 0.6.6754 0.6.6754 0.0.2022 0. ч. 11100 f P 57 æ 0.11000 1 ° 0.046 0.0000 ц в (г.,1) 12 Mod. Sta. (ft.) 11 y<sub>3</sub> Sta. 10 y<sub>1</sub> Sta.) (ft.) \* روم ۲ ه в У3 (св.) 7 21 (cm.) 6 Уп (ст.) 24.52 5 G (cfs) **;**• 0.00030 22000-10 22000-10 \* °° Ψ× ~ \$ H کوبہ

TABLE 7-2-2 CALCULATIONS FOR THE LARGE FLUGE ROUGH BOUNDARY MODEL TESTS (continued) P. 2

Measured Data

Calculated Data

24 1-3 (ft:	
5 1-5 (11.)	10000000000000000000000000000000000000
22 ¥1 - ¥3 (It.)	
21 1, 1, 2, 1 (ft)	20000000000000000000000000000000000000
го /н13/3	2805285536556566666555555555555555555555555
P P	
1 <sup>1</sup>	
C.d.	
н,	
f p	
15 Re	
7 ° (	
5 e (;	
12 Kod. št (ft.,	4 1 1 8
11 V3 Sta.)	
10 y <sub>1</sub> Sta. (ft.)	
° <b>∧</b> <sup>°</sup>	00000000000000000000000000000000000000
8 ¥3 (cm.)	11542522525252525255255255255255255555555
7 y1 (cm.)	
6 yn (cm.)	2
5 Cf6)	%·····
2° °	.000.91 200.000 200.000 200.000 200.00000000
σ×	
2 L/b	0.000.000.000.000.000.000.000.000.000.
I No.	ŢŢŢĴſŢŢŢŢŢŢŢŢŢŢŢŢŢŢŢŢŢŢŢŢŢŢŢŢŢŢŢŢŢŢŢŢŢ

۵.	
(continued)	
-2-2 CALCULATIONS FOR THE LANDE FLUKE ROUCH BOUNDARY MODEL TESTS	
TABLE	

'n.

Messured Data

Calculated Data

24 1-3 (fft.)		
23 175 (117)	44444646446466 44444646666666666666666	
22 (1 - <sup>y</sup> 3	0,0,9,9,9 0,0,0,9,9,9 0,0,0,1,1,1,1,1,2,1,1,1,1,1,1,1,1,1,1,1,	
17 ° ° ° ° ° ° ° ° ° ° ° ° ° ° ° ° ° ° °	2.2.2.2.2.2.2.2.2.2.2.2.2.2.2.2.2.2.2.	
20 n/H13/3	11,570 11,271 11,272	
19 y <sub>1</sub> /y <sub>n</sub> (F	2, 996 2, 996 2, 996 2, 996 2, 996 2, 996 2, 998 2,	
0.d	9984) 1000000000000000000000000000000000000	
17	2,245,2 2,455,2 2,4	
r Ib	11114 	
15 R.		
Li. Fn		
13 R - 13	0,13,20	
12 . Sta.		
11 Sta. Moc (rt.)	24455444555844 2685856544455544 268585654445544 268585655544454 2685855555444454 26854444544454 268544445444 268544444544 268544444444 2685444444444 26854444444444444 2685444444444444444444444444444444444444	
10 Sta. Y <sub>3</sub> ft.)	「 、 、 、 、 、 、 、 、 、 、 、 、 、	
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(c 10 10 10		
	011200 - 1 - 1 - 1 - 1 - 1 - 1 - 1 - 1 - 1 -	
	Lt a second seco	
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1 bun No. L/		

Geometry Ia

TABLE 7-2-3\_ SURFACE TOPOGRAPHY & REGAIN CURVE DATA

A) Surface Topography

Run No. 4-4; Q = 1 cfs; S = 0.000584;  $y_n = 9.72$  cm.; M = 0.491;

**L/b = 0;** Model Station = 29.90 ft.:

Centerline Station	Station Left	Depths Centerline	(cm.) Ieft
	2002 0		
15.00		10,58	
	1.00		10.57
	2.00		10.54
16.00		10.56	
17.00		10.58	
18.00		10,61	/-
	1.00		10.61
20.00	2.00		10.05
19.00		10.63	
20,00	00.05	. TO 03	10 61
	1:00		10°01
21 00	2.00	10 62	10.57
22.00		10.63	
22.00	1.00	10,00	10.61
	2.00		10,58
23.00	2000	10.63	
24.00		10.63	
	1.00		10.63
	2.00		10.61
25.00		10.64	
26.00		10.65	
	1.00		10.66
	2.00		10.63
26.50		10.65	
27.50	7	10.64	70 (1
	1.00		10.04
00 00	2.00	10 61	10.05
28.00	1 00	TOOT	10.65
	1.00		10.66
28 EO	2.00	10 56	10.00
20.50	1 00		10.60
	2 00		10.66
29.00	L. 00	10.47	
27.00	0.50		10.51
	1.00		10.56
	- 1.50		10.64
	2.00		10.72

### 7-2-3 ( CONTD.)

Centerline Station	Station Left	Depths ( Centerline	cm。) Left
29.25	0.50 1.00 1.50 2.00	10.39	10.43 10.50 10.65
29 <b>.</b> 50	0.50 1.00 1.50 2.00 2.32	10.21	10.25 10.34 10.75 10.72 10.70
29.79	0.50 0.75 1.00 1.25 1.50 1.75 2.00 2.32	10.09	-10.09 10.07 10.17 10.53 10.72 10.76 10.76 10.73
30.20	0.50 0.75 1.00 1.25 1.50 2.00	9•95	9.86 9.74 9.55 9.09 9.07 9.05
30.50	0.50 1.00 1.50 2.00	9.57	9.45 9.17 9.07 9.06
31.00	1.00 1.50 2.00	9.27	9 <b>.1</b> 3 9 <b>.</b> 08 9 <b>.</b> 07
31.50	1.00 2.00	9.06	9 <b>.07</b> 9 <b>.0</b> 8
32.0	1.00 2.00	9.02	9 <b>.1</b> 0 9 <b>.</b> 07
32•50 33•00	1.00	9.05 9.04	9.07 9.13
34.00	2.00	9.15	9,14
35.00		9.12	

#### 7-2-3 (CONTD.)

Centerline Station	Station Left	Depths Centerline	(cm.) Left
36.00	0.00	9.22	
37.00	2.00	9.35	9.18
38.00		9.38	
39.00		9.40	
40.00		9.43	
41.00		9.45	
42.00		9.50	
43.00		9.53	
44.00		9.55	
45.00		9.57	
46.00		9,58	

### B) Regain Curves

Run No. 1-4; Q = 1 cfs;  $y_n = 24.50 \text{ cm.}$ ; Run No. 2-7; Q = 2 cfs;  $y_n = 24.30 \text{ cm.}$ ;S = 0.0000302; M = 0.491; L/b = 0;Nodel Station = 30.00 ft.:

Centerline	Depth (cm.)	Centerline	Depth (cm.)
31.00	24.24	31.00	24,20
31.50	24.15	31,50	24.15
32.00	24.13	32.00	21.12
32.50	24.11	32,50	24.13
33.00	24.10	33.00	21.71
33.50	24.07	33,50	21.11
34.00	24.14	34.00	21.15
34.50	24.17	35.00	211.19
35.00	24.20	36.00	21.21
36.00	24.22	37.00	2/1.2/1
37.00	24.26	38.00	21.27
. 38.00	24.30	39.00	24.30
39.00	24.34	40.00	24.29
42.00	24.35	41.00	24.30
44.00	24.40		
46.00	24.14		
48.00	24.46		
50.00	24,50		
52.00	24.51		
54.00	24.50		

Geometry 1

### TABLE 7-2-4 VELOCITY MEASUREMENTS

.

# A) Velocity Profiles for Run No. 4-4

0.837 0.837

0.850

0.200 0.240

0.280

# Q = 1 cfs; S = 0.000584; $y_n = 9.72$ cm.; M = 0.491; L/b = 0; Model Station = 30.00 ft.:

#### a. Normal Depth

Center	rline	l ft	. Left	2 ft.	Left
Depth	Velocity	Depth	Velocity	Depth	Velocity
ft.	fps	ft.	fps	ft.	fps
0.010 0.015 0.020 0.025 0.030 0.035 0.010 0.015 0.050 0.055 0.025	0.147 0.490 0.490 0.565 0.565 0.565 0.618 0.635 0.635 0.695 0.695 0.695 0.695 0.695 0.695 0.748 0.748 0.788	0.010 0.020 0.030 0.040 0.050 0.060 0.070 0.080 0.090 0.100 0.120 0.140 0.160 0.200 0.240 0.280	0.147 0.125 0.490 0.470 0.528 0.535 0.600 0.650 0.650 0.650 0.665 0.722 0.708 0.735 0.718 0.775	0.010 0.020 0.030 0.040 0.060 0.100 0.100 0.140 0.160 0.200 0.240 0.280	0.347 0.347 0.347 0.347 0.490 0.528 0.565 0.600 0.635 0.650 0.695 0.748 0.762

b. Maximum Depth

Center	rline	l ft.	Left	2 ft.	Left
Depth	∀elocity	Depth	Velocity	Depth	Velocity
ft.	fps	ft.	fps	ft.	fps
0.0145 0.0145 0.050 0.055 0.060 0.070 0.080 0.090 0.100 0.110 0.120	0.528 0.547 0.547 0.547 0.547 0.547 0.546 0.566 0.634 0.618 0.650 0.695	0.040 0.050 0.060 0.070 0.080 0.090 0.100 0.120 0.140 0.160 0.180	0.566 0.602 0.618 0.634 0.650 0.695 0.695 0.708 0.722 0.749 0.749	0.050 0.060 0.070 0.080 0.090 0.100 0.120 0.140 0.140 0.160 0.180 0.200	0.1117 0.1417 0.1468 0.168 0.509 0.1489 0.566 0.566 0.566 0.566 0.6314 0.650

/-2-+ (					
0.130 0.140 0.160 0.200 0.200 0.220 0.240 0.260 0.280 0.280 0.300	0.695 0.695 0.749 0.749 0.749 0.749 0.749 0.749 0.788 0.800 0.837	0,200 0,220 0,240 0,260 0,280 0,280 0,300	0.775 0.788 0.813 0.837 0.837 0.813	0.220 0.240 0.260 0.280 0.300	0.650 0.680 0.695 0.695 0.695

2 1 LCON

#### c. Vena Contracta

Cente	rline	0 <b>.5</b> ft	. Left	1.0 ft.	Left
Depth	Velocity	Depth	Velocity	Depth	Velocity
ft.	fps	ft.	fps	ft.	fps
0.010	1.145	0.010	1,206	0.010	1.090
0.015	1.154	0.020	1.215	0.020	1.180
0.020	1.163	0.030	1.230	0.030	1.300
0.025	1,170	0.040	1.240	0.040	1.390
0.030	1.180	0.050	1.280	0.050	1.490
0.035	1.206	0.060	1.313	0.060	1.581
0.040	1.230	0.070	1.367	0.070	1.594
0.045	1.265	0.080	1.432	0.080	1.642
0.050	1,300	0.090	1.470	0.090	1.654
0.060	1.350	0.100	1.490	0.100	1.690
0.070	1.367	0.120	1.524	0.120	1.690
0.080	1.410	0.140	1.534	0.140	1.724
0.090	1.420	0.160	1.543	0.160	1.770
0.100	1.455	0.180	1.555	0.200	1.758
0.120	1.490	0.200	1.570	0.240	1.834
0.140	1.510	0.220	1.588		
0.160	1.517	0.240	1.618		
0.180	1.517	0.260	1.623		
0.200	1.524	0.280	1.642		
0.220	1.555				
0.240	1.550				
0.260	1.555				
0.280	1.570				

### d. Minimum Depth

Cente	rline	1.0 ft	. Left	1.5 ft	. left	2.21	ft. Left
Depth ft.	Velocity fps	Depth ft.	Velocity fps	Depth ft.	Velocity fps	Depth ft.	Velocity fps
0.010 0.015 0.020 0.025 0.030	0.695 0.708 0.722 0.749 0.775	0.010 0.020 0.030 0.040 0.050	0.915 1.036 1.162 1.265 1.280	0.010 0.020 0.030 0.040 0.050	0.447 0.425 0.425 0.400 0.400	0.010 0.020 0.030 0.040 0.060	0.200 0.200 0.200 0.200 0.200 0.200

-2-4 (6	UNID.)						
0.035 0.040 0.045 0.050 0.060 0.070 0.080 0.090 0.100 0.120 0.140 0.160 0.200	0.850 0.936 0.980 1.170 1.265 1.420 1.505 1.570 1.700 1.700 1.824 1.940 2.030	0.060 0.070 0.080 0.090 0.100 0.120 0.140 0.160 0.160 0.200 0.220 0.240	1.360 1.5142 1.618 1.735 1.770 1.835 1.813 1.758 1.758 1.758 1.770 1.687 1.735	0.060 0.070 0.080 0.090 0.100 0.120 0.140 0.160 0.200 0.240	0.347 0.347 0.347 0.347 0.347 0.347 0.347 0.347 0.347 0.347 0.347 0.347 0.347	0.080 0.100 0.120 0.160 0.200 0.240	0,200 0,200 0,200 0,200 0,200 0,200

B) Velocity Profile for Normal Depth Run No. 2

٠

Q = 2 cfs; S = 0.000131;  $y_n = 24.30$  cm.:

Center	rline	l ft.	Left	2 It	Leit
Depth ft.	Velocity fps	Depth ft.	Velocity fps	Depth ft.	fps
0.010 0.020 0.030 0.040 0.050 0.060 0.080 0.090 0.100 0.120 0.200 0.	0.200 0.245 0.315 0.317 0.315 0.315 0.315 0.375 0.375 0.401 0.425 0.401 0.425 0.401 0.421 0.401 0.491 0.510 0.550 0.550	0.010 0.050 0.100 0.150 0.250 0.300 0.350 0.400 0.500 0.600 0.700	0.347 0.401 0.401 0.547 0.566 0.602 0.650 0.665 0.665 0.665 0.694 0.722	0.010 0.100 0.200 0.400 0.500 0.600 0.700	0.101 0.547 0.650 0.665 0.665 0.665 0.665

# 7-2-4 (CONTD.)

# . C) Velocity Profile for Normal Depth Run No. 10

Q = 1 cfs; S = 0.00408;  $y_n = 5.35$  cm.:

Cente	rline	1 ft.	Left	2 ft.	Left	2.23	ft. Left
Depth ft.	Velocity fps	Depth ft.	Velocity fps	Depth ft.	Velocity fps	Depth ft.	Velocity fps
0.010 0.015 0.020 0.025 0.035 0.035 0.035 0.050 0.050 0.050 0.060 0.070 0.080 0.090 0.100 0.120 0.110	0.665 0.695 0.755 0.825 0.875 0.915 1.180 1.215 1.265 1.340 1.475 1.540 1.630	0.010 0.020 0.030 0.040 0.050 0.060 0.070 0.080 0.090 0.100 0.120 0.140	0.750 0.775 0.825 0.875 1.015 1.015 1.145 1.250 1.330 1.390 1.505 1.580	0.010 0.020 0.030 0.040 0.050 0.060 0.060 0.060 0.100 0.120 0.140	0.775 0.775 0.775 0.875 0.915 1.035 1.035 1.095 1.180 1.340 1.555 1.550	0.010 0.020 0.030 0.040 0.060 0.080 0.100 0.120 0.140	0.775 0.775 0.825 0.980 1.145 1.230 1.340 1.475

TABLE 7-3 GEOMETRY II RAW and CALCULATED DATA.

	3	e.	4	5	6	7	80	6	10	ц	ส	13
0	М	<sup>^</sup> <b>B</b>	cfs	с Ч С	8 <sup>0</sup>	"W	P P	۲ <sup>۳</sup>	יב <b>י</b> ן גי	$\left(\frac{F_{n}^{2}}{M}\right)$	ĸ	۵
.~	5	C.050	:	24.09	C. CCCO3	C.464	0	24.32	0.00955	0.01161	1.85832	0.50180
2	C.5	0.050	1.C	24.09	C.CCC03	0.464	4.1	24.29	0.00830	0.01161	1.64962	0.36721
~ ·	5 U	0.050		24.09	C. COCC3	0.464	8.2	24.34	0.01038	0.01161	1.57545	U. 366668
<b>t</b> u	יי יי יי ב		ے ر 	24.02		1010	13.7	24.28	0.00956	0-01161	1.85299	0.56668
- <b>-</b>		10.0		24.05	0.0000	0.675	0	24.11	0.00249	0.00927	1.42707	0.50452
, r-	C • 7	C.065	1.C	24.05	C. C.C.C.C.C.C.C.C.C.C.C.C.C.C.C.C.C.C.	C.675	3.)	24.13	0.00333	0.00927	1.72052	0.15816
æ	C . J	0.065	1.0	24.C5	C. COCO3	0.675	7.9	24.25	0.00832	0.00927	3.48124	0.56350
5	د ، ا	C.C.C.E.S	1•C	24.15	C.CCC03	C.675	10.5	24.23	0.00748	0.00927	3.18778	0.56350
J	C.7	C. G65	1.C	24.C5	C. CCCU3	0.675	13.1	24.24	0.00790	0.00927	3.33451	0.56350
_	5•3	ເ€).J	1.C	24.15	C. COCO3	0.882	.0	24.23	0.06748	0.00543	5.44274	0.55560
2	5•)	C.LE5	1 • C	24.C5	C • C O C O 3	C.882	3.9	24.25	0.00832	0.00543	5.94379	0.33751
<b>(</b> ")	5°)	0.065	1•C	24 • C 5	C. CCCO3	0.882	7.8	24.26	0.00873	0.00543	6.19431	0.55601
4	5°2	430.0	1.C	24.C5	C. CCCO3	0.882	10.3	24.27	0.00915	0.00543	6.44484	0.55601
ŝ	5°)	G.CE5	1•C	24.05	C. COCO3	0.842	12.9	24.28	0.00956	0.00543	6.69536	0.55601
Ļ	C•5	C.356	2 • C	9 ° FC	C.CC23C	0.896	•	9.88	0.00816	0.19533	1.03396	0.46894
~	5°)	C.356	2•C	0.9°6	C • C C Z 3 C	0.896	<b>9.4</b>	9.98	0.01837	55541.0	1.12211	0.60811
<u>م</u> ،	ა. ე	9320		ວຸ ເມື	C. CO23C	0.846	18.1	9.46 0.00	0.01033	55561 0 55561 0	16401•1	0.60000
ۍ د	י ג י נ	0.150			1.00231	478 O	0.02	0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0		0 10533	1.221.1	
	, , , ,				C. CC2333	0.030	0.10 2.15	11 26.0	0.15918	0.22465	1.44173	
					C.00233	0.645	26.1	11.35	0.15816	0.32465	1 43601	0.81028
		90° C • O	2 ° C		C. CC233	0.645	I H B	11.26	0.14898	0.32465	1.38466	0.81028
1.5		956.0		08.0	C. CC233	0.695	4.6	11.05	0.12755	0.32465	1.2655C	0.61588
	C - 7	0.156	2.0	9. 60	C.C0233	0.695	•0	10.77	0.09898	0.32465	1.10816	0.46325
	C.5	0.356	2°C	9 ° C	C. CC233	0.495	°.0	13.30	0.35714	0.64000	1.30955	0.45577
~	c.5	0.356	2.C	9.60	C.CC233	0.495	31.4	14.55	0.48469	0.64000	1.69398	•0
8	ç•)	L.356	2°C	9 <b>.</b> 8C	C.CC233	0.495	25.1	14.55	0.48469	0.64000	1.69358	1.19302
5	C.5	0.356	2 • C	9 • FC	C.CC233	0.495	18.8	14.34	0.46327	0.64000	1.62892	1.19302
J	(°3	0.353	2 • C	9 ° F C	C.CC233	0.291	• 0	19.53	0.99286	1.82389	1.13038	0.44446
-		0.353	2.C	9 ° °C	. C.CC233	0.291	9.6	23.12	1.35918	1.82389	1.53068	0.64304
~	<b>.</b> .)	C.353	2°C	9.80	C.CC233	0.291	19.2	23.09	1.35612	1.82389	1.52732	2.16718
	3	C.353	2°C	9°80	C.CC232	0.291	25.6	22.65	1.31122	1.62300	1.47808	<b>.</b>
<b>5</b> 1		252.0			C. CC233	167.0	1.25	CQ•27	77116.1	1.00250	1.47808	0.45447
<b>^</b> 4		5,55,0 2,75,0 2,75	י ג גיר	2 ° C	C. CCC13	- 403	<b>.</b>	00.02	1 63265	0 72048	4.02014	14404-0
			, C.		0.0001			25.75	1.62755	0.72048	4.62191	0.62828
- a				0.40	C. COO12	0.463	20.2	25.88	1.64082	0.72048	4.65891	1.27635
ათ-		0.353	2.0	9.80	c.coc13	0.463	26.9	25.91	1.64388	0.72048	4.66745	•
ں .	5.0	0.353	2°C	9. 80	C.COC13	0.463	33.6	25.96	1.64898	0.72048	4.68168	••
-	C•3	0.155	1.C	9.88	C.COC71	0.291	•0	13.33	0.34919	0.44904	1.64844	0.45966
2	с <b>.</b> Э	0.155	1.C	9.88	C.00071	0.291	9.5	14.47	0.46457	0.44904	2.16624	0.62090
~	C•3	0.155	1.C	9.68	C.COC71	0.291	19.1	14.15	0.43219	0.44904	2.02069	0.97483
•	3	C.155	1•C	9.68	C.COC71	u.291	25.4	14.62	0.47976	0.44904	2.23452	0.97483
<u>د</u>		C.155	1•C	9.88	C.CC071	0.291	31.8	14.38	0.45547	0.44904	2.12529	0.
υr		251.0	ہ ر 	8. S 5. C	0.00071	264°0		16.01	0.13457	0.1251.0		0.52200
~ a		C 51 - D		2. 2. 2. 2. 2. 2. 2. 2. 2. 2. 2. 2. 2. 2	C.00071	0.4430	191	11 26	26021.0	0 15510	2 04276	
50	יא ר - ב כ			.0			21.2 2	11 28	0.141.0	0.15519	2.09558	
- u	33	0.750	) C	13.40	C. COC 85			20.64	0.54030	0.78593	1.48359	0.45353
,	33	0.250		13.40	C. COCRS	0.282		23.60	0.16119	0.78593	2-06425	0.58627
• •	10	C.25C	2.0	13.40	C. COC85	0.282	14.5	23.33	0.74104	0.78593	2.01113	0.62965

มิจ		1 34120	0.51884	0.61255	0.71669	0.71609	0.48043	0.40103	0.48064	0.47356	0.47730	0.46291	0.59702	0.59702	0.59702	0.55570	0, 50570	0.4254B	0.48608	0.46152	0.54076	· 0.61829	0.88565	C0C88.0	66666°0	14704-0	0.50051	0.46930	0.46930	0.45289	1.28657	1.38657	•0	•••	0.94049	0-61992	0.62158	0.58627	0.58524	41469-0	0.61797	0.64672	51543	0.82096	0.50358	0.92675	0.59435	0.44425	0.59893	0.58760	0.58903	0.39688	0.51251	0.37874	0.48521	0.36180
27 א		0 53376	1.58778	1.70819	1.74837	1.77134	1. 37853	1.26705	1.23131	1.23131	1.57646	1.36834	1.67635	1.79302	1.89301	0000001	1 447765	102-1	1.21915	1.46843	1.55479	1.70139	1.7/244	1 4.0020	47604°I	1.30547	1.17054	1.13839	1.13839	1.30688 1.50348	1.49445	1.57860	1.57286	1.37707	1.35980	1.30816	2.17263	1.56053	15051-2	1.84615	1-68302	1.18769	1.52493	1.38322	1.11666	1 78134	1-37374	1.09089	1.93592	1.238570	1.93374	1.22242	1.85821	1.49325	1.37868	0.99312
$\left(\frac{F_{n}}{M^{1}}\right)^{2}$		0.78593	0.26137	0.26137	0.26137	0.26137	11671.0	0.12977	0.12977	0.12977	0.09355	0.09355	0.09355	0.09355	0.09355	92670 0	91CE70 0	91640-0	0.04376	0.37948	0.37948	0.37948	0.37040	0.114004	0.18904	0.18904	0.18904	0.18904	0.18904	0.83317	11669-0	0.83317	0.83317	0.42166	0.42166	0.42166	0.46913	0.78593	0-04216	1.08507	0.37180	05206 1	0.66639	0.33220	0.19974	0.41041	0.18849	0.11311	0.10633	0.024953 0.02912	0.05459	0.02338	0.1/654	04490-0	0.13204	0.07890
"un		AAF71-0	0.17164	0.18731	0.19254	0.19552	66750 0	0.05224	0.04328	0.04328	0.06265	0.05280	0.06738	0.07289	0.07702		0.02040	0.01970	0.01497	0.23625	0.25307	0.28195	0.28130 7 2020E	00000	0.08155	0.07443	0.06084	0.05761	0.05761	0.49871	0.58398	0.62188	0.61929	0.21102	0.20612	0.19380	0.44600	0.56706	0-03641	1.04007	0.29530	105206	0.46087	0.15731	0.01897	0.30214	0.08187	0.00845	0.0792B	0.00392	0.04290	19200.0	0.14414	0.19909	0.05389	0.00170
9 Lr		15.73	15.70	15.91	15.98	16.02	14.11	14-10	13 98	13.98	26.97	26.72	27.09	27.23	26.12	04.90	00.00	25.88	25.76	19.10	19.36	19.80	10 00	16.76	16.71	16.60	16.39	16.34	16.34	19 46	18.39	18.83	18.80	14.06	13.99	13.86	14 • 46	21.03	15.37	23.42	14.87	21.12	18.48	14.64	12.89	20-04	16.65	15.52	27.50	25.58	34.28	33.12	20.05	21.14	18.58	17.66
An2 An2		19.4	7.0	14.0	18.6	6.62	1 4 4	8.61	6 9	••	•0	3.9	7.8	10.4	13.0				•0	•0	6•1	7.21	7.02	0.00	16.0	12.0	6.0	•	•	о - н	16.0	21.3	26.6	26.5	15.9	8.0	9 <b>°</b> 6	4°8	12.7	8 3	6.7	0 - 0	7 4	24.3	7.3	20.4	20.1	20.0	13.0	3.7	10.8	6.0 16.0	15.5	17.9	17.6	17.5
w* ~		0.282	0.489	0.489	0.489	684.0	1001	0.694	0.694	0.694	6.461	0.461	0.461	0.461	19401	0.674	0.674	0.674	0.674	0.487	0.487	0.481	10400	0.640	C.690	0.690	0.690	0.690	0.690	0.493	0.493	0.493	0.493	0.693	669-0	0.693	0.292	0.282	0.487	0.288	0.492	0.240	0.490	Ù.694	C.895	0.486	0.691	0.892	C.460	0.879	0.428	C-654	0.484	0.482	0.688	0.890
ઝ જ <sup>4</sup>		C.CCCB5	C.COC85	C.COC85	C. CCC85	C 00005	C-00085	C. COC85	C. COCH5	C. COOB5	C. COC25	c.c0c29	C.C0C29	C. COUZY	C.CUC25	0.00024	C. CCC24	C. COC2 9	C.COC25	C.C0123	C.00123	C-CU123	C. CO123	C.00123	C.C0123	C.C0123	C.C0123	C.00123	0.00294	C. CC234	C. C0294	C. CC294	c.cc294	C.C0294	C. C0294	C.CC294	C.COC13	C. 00018	C. COCL4	C.C013C	C.0013C	0.00014	C.C023C	C. CC23C	C. CO23C	C.CO021	C.00021	C.00021	C. COC2E	C.00025	C.C0012	C.00C12	r.cutut r.consc	C.CCC85	C. CO085	C.00CB5
S n n − 5		13.40	13.40	13.4C	13.4C	12.40	13-40	13.40	13.40	13.40	25.38	25.38	25.38	86.62	25, 38	25. 28	25.38	25.38	25.38	15.45	15.45 15.45	15 45	107.01	15.45	15.45	15.45	15.45	15.45	12.40	11.61	11.61	11.61	11.61	11.61	11.61	11.61	10.00	14.23	14.83	11.48	11.48	4.77	12.65	12.65	15.65	15.39	15.39	15.39	25.48 25.48	25.48	32.67	32.67	20.12	17.63	17.63	17.63
cfa 6		2°C	2°C	2 • C	5°C	2 ° C		2.0	2.C	2°C	3.0	3°C	0.0		эс • т	0	0.0	3.0	3°C	ບ ຕ				0	0	Э•С	3°C	0	).		0	3°C	3°C		5 U n m	Э•С	1.C	) · · ·		2.C	0.0	, . , .	Э•С	3°C			3°C	0. M			3°C	0 L M M	. c	3.0	3.0	з•с
e n		C.250	C.250	C-250	C.25U	0.250	0.2.0	C.250	0.250	U.25U	C.141	C.141	C.141	141.0	1.1.	C 141	0.141	0.141	U.141	c.3C0	0.200			0.300	C.3CO	C.3CO	0.300	0.300	0.250	0.440	0.450	0.450	0.4.0	0.4°0	0.450	0.450	0.200		C.1CO	C.3C0	0.300	0.400	0.400	0.400	0.14-0	000000	0.300	0.300	0.1.0	0.150	0.100	0.100	0.200	0.250	C.250	0.220
າ່ສ		<b>.</b>	c.5	5	5	55	5		C.7	c.7	6.5		5.		32	C • 7	6.7	C.7	C.7	5	5	32		C • 2	C . 7	C.7	c. 2			55	C.•5	C.5				C.7	3	50	5.5	č• 3	5.		C.5	3	د د		C.7	5.0	30	5.0	c.5	C • 3	C.7	C.5	C • 7	۲•۶
1 Run No		53	54	55	2	- 4	56	90	61	62	63	49	0	0 4	- 4	59	70	71	72	13	5 U	12		7.8	51	8C	81	7.8		6 6	96	87	au u ag a	50	16	92	66	56	96	16	a 5 6 6	100	101	102	104	105	106	107	106	110		112	114	115	116	111

Contd.

	TABL	E 7-4 GE(	METRY	III RAW	and CALCULAT	ED DATA.							
1	2	3	4	5	6	7	80	6	10	п	21	13	77
Run No	W	e.a	ر در fs	у <sup>п</sup> сш	su u	'n	ቅ	y1 cm	*u	$\left(\frac{F_{n}}{M}\right)^{2}$	K	D	$\left(\frac{h_{1}^{*}}{r_{n}}\right)\left(\frac{M_{1}}{F_{n}}\right)^{2}$
I-I-M	0.0	0.096 0.096 0.096	0. 0. 0. 0.	32.70 32.70 32.70	0.00012 0.00012 0.00012	C.427 C.427 C.427	30.00 45.00 60.00	<u>33.56</u> 33.63 33.71	0 <u>.02630</u> 0.02844 0.03089	0.05049 0.05049 0.05049	1.03070 1.11463 1.21055	0.23441 0.27311 0.35825	0.521 0.563 0.612
V-1-2	0.5	0.096 0.096 6.096 0.096	ດ	32.70 32.70 32.70 32.70	0.00012 0.00012 0.00012	C.427 C.652 C.652 C.052	90.00 90.00 45.00 90	<b>33.7</b> 6 <b>33.</b> 90 32.94	0.03242 0.03670 0.00734 0.00765	0.05049 0.02166 0.02166 0.02166	1.27051 3.35284 0.67026 0.69819	0.37485 0.22092 0.26626 0.35389	0.642 1.694 0.339 0.353
W-1-3	0.7 C.7	0.147 0.147 0.147 0.147	0000	24.65 24.65	с. 00028 u. 00028 с. 00028	C.674 D.674 C.674	3C.00 45.00	24.88 24.98 24.98	0.00933 0.01339 0.01339	0.04741 0.04741 0.04741	0.38360 0.55047 0.55047	0.23338 0.27259 0.37325	0.197 0.282 0.282
W-1-4		C.147 C.147 C.147 C.147		24.65 24.65 24.65 24.65	u.00028 u.00028 C.00028	U.462 C.462 C.462	30.00 45.00 40.00	25.87 26.12 26.21	0.04949 0.05963 C.06329 0.06572	0.10092 0.10092 0.10092 0.10092	0.95725 1.15381 1.22460 1.27180	0.24605 0.27884 0.38129 0.39293	0.490 0.591 0.627 0.651
W-1-5	000000	C.201 C.201 O.201	0000 1 m n n	20.00 20.00	U. 00048 U. 00048 U. 00048	0.476 0.476 0.476 0.476	45.00 60.00	22.23 21.00 20.12	0.011150	0.17818 0.17818 0.17818	1.19983 0.53600 0.06412	0.25604 0.28364 0.40130	0.626 0.281 0.034
W-1-6	0.1	6.201 C.201 0.201 0.201		00000000000000000000000000000000000000	C.00048 U.00048 U.00048 C.00048 C.00048 C.00048	C. 4 0.693 0.693 0.693 0.693 0.693 0.693 0.693 0.0000 0.0000 0.0000 0.0000 0.0000 0.0000 0.0000 0.0000 0.0000 0.0000 0.0000 0.0000 0.0000 0.0000 0.000000	90.00 90.00 90.00 90.00 90.00 90.00 90.00	20.24 20.24 20.42 20.48 20.48	0.01200 0.01750 0.02100 0.02400 0.02400	0.17818 0.08639 0.08639 0.08639 0.08639	0.12830 0.38606 0.46339 0.52971 0.52971	0.40841 0.24339 0.27754 0.37599 0.38879	0.203 0.243 0.278 0.278
W-1-7	000000	C.201 0.201 0.194 C.194 0.194		20.00 20.00 9.82 9.82 9.82 9.82	u.00048 u.00048 U.00050 u.00050 u.00050	0.887 C.887 C.290 C.290 O.290	40.00 40.00 45.00 40.00 40.00 40.00	20.12 20.11 12.26 12.69 12.89 13.15	U.00600 U.00550 O.25051 O.29226 O.31263 O.33910	0.05123 0.05123 0.44869 0.44869 0.44869 0.44869 0.44869	0.22302 0.20443 1.08015 1.26214 1.35106 1.46677	0.23465 0.37522 0.27314 0.29160 0.43608 0.43488	0.117 0.107 0.558 0.651 0.697 0.756
<b>W-2-1</b> W-2-2	000000	0.250 0.250 0.250 0.250		17.27 17.27 17.27 17.27 17.27	C. 00084 C. 00084 C. 00084 C. 00084 C. 00084	C + 482 C + 482 C + 482 C + 482 C + 482 C + 82 C + 82	4 5 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0	19.01 19.49 19.92 20.28	0.10075 0.15855 0.17429 0.17429	0.26938 0.26938 0.26938 0.26938 0.13245	0.69935 0.89451 1.07002 1.21742 0.08092	0.26356 0.28717 0.41651 0.42005 0.25078	0.374 0.477 0.570 0.647 0.044
	1.00 1.00 6.0	C. 250 C. 250 O. 250 O. 250	00000	17.27 17.27 17.27 17.27 17.27	C. 01084 L. 00084 C. 00084 U. 00084 C. 00084	C.687 0.687 0.687 C.890 C.890	4 4 4 4 4 4 4 4 4 4 4 4 4 4 4 4 4 4 4	17.85 17.88 17.94 17.37 17.40	0.00579 0.03880 0.03880 0.00579 0.00753	0.13245 0.13245 0.13245 0.07896 0.07896	0.4/08/ 0.49532 0.54425 0.13575 0.17651	0.28112 0.39073 0.40026 0.24186 0.38642	0.254 0.267 0.073 0.095
¥-2-3	0.7	0.299 0.299 0.299	0000 • • • • •	15.32 15.32 15.32 15.32	U.00123 C.00123 C.00123 G.00123	0,690 0,690 0,699 0,699 0,690	30.00 45.00 60.00 90.00	15.81 15.92 16.09 16.16	0.03198 0.03916 0.05026 0.05483	0.18824 0.18824 0.18824 0.18824 0.18824	0.30495 0.37385 0.48062 0.52469	0.25703 0.28410 0.40329 0.40994	0.170 0.208 0.267 0.291
¥-2-5	0000 00000 0000	C.299 0.299 0.299	0000 	15.32 15.32 15.32 15.32	0.00123 0.00123 0.00123 0.00123 C.00123	0.486 0.486 0.486 0.486	30.00 45.00 60.00 90.00	17.66 17.93 18.48 18.85	0.15274 0.17037 0.20627 0.23042	0.37964 0.37964 0.37964 0.37964	0.73454 0.82101 0.99801 1.11768	0. 0.29015 0.42957 0.42997	0.402 0.449 0.543 0.607
W-2-7	0.9 0.9	0.299	3.0 3.0	15.32 15.32	0.00123 0.00123	0.892 0.892	30.00 90.00	15.38 15.39	0.00392 0.00457	0.11258 0.11258	0.06214	0.24794 0.39586	0.035 0.041

3	io. M Fn	8 C.5 0.344	0.5 0.394	0.394			0.7 0.394	0.9 0.394	0.9 C.394	L 0.5 0.507	105°0, 6°0			0.7 0.507	0.7 0.507	0.7 0.507	0.9 0.507	0.9 0.507				405-0 5-0	405-0 5-0	0.3 0.394	0.3 0.394	0.301	0.3 0.301	0.3 0.301		0.3 0.232	0:3 0.232	0.3 0.232	0.3 0.131	0.3 0.131	0.3 0.131	0.3 0.131	0.3 0.103		0.3 0.103	
4	دfs	0.0	3.0	3.0				3.0	9.0	3.0	3.0	0 0 n n	0 C		3.0	3.0	3.0	3.0	2.0	2•0 5	0 0 7 7	0.2		2.0	2.0	2.0	2.0	2.0	2•0	0 • 1		0-1	1.0	1.0	1.u	1.0	1.0	1•0		2
5	ч <sup>п</sup> Ш	12.76	12.76	12.76	12.76	12.10	12.76	12.76	12.76	10.78	10.78	10.78	10 70	10.78	10.78	10.78	10.78	10.78	9.54	9.54	4°24	40°4	5 · · 0	72.0	9.74	11.66	11.66	11.66	11.66	8. / 3 2 - 7 3	0.19 272	8.73	12.80	12.80	12.80	12.80	14.96	14.96	14.96	11.70
6	о <sub>п</sub>	Ú.00220	U.00220	0.00220	C.00220	0.00220	0.200.00	0-00220	0.09220	0.00360	<b>U.00360</b>	u.00360	C.00360		003600	0.00360	0.00360	0.00360	0.00360	0.00360	0.00360	0.00360	07700.0	0.000200	0.00220	0.00135	0.00135	0.00135	0.00135	0.00088	0.00000	0-00088	0.00031	0.00031	0.00031	0.00031	0.00014	0.00014	0.00014	4T000.0
7	W	C • 490	0.490	C.490	C.093	0.693	0.693 7.503	0.845	1995	664.0	C.493	0.493	0.443	0 4045	0,000	C. 695	C. 896	0.896	0.291	0.291	0.291	0.291	0.290	067.0		0.220	0.289	0.289	0.289	0.292	0.292	262.0	0.636	0.283	0.283	0.283	0.276	0.276	0.276	0.216
60	¢	30.00	60.00	90.00	30.00	45.00	60.00		00.06	30.00	45.00	6C.00	90.06	30.00		90 ° 00	30.00	90.00	30.00	45.00	60.00	90.00	30.00	45.00	00.00		45.00	60.00	90.00	30.00	45.00	60 <b>.</b> 00	00.06		00-00	00-06	30.00	45.00	60.00	90.00
6	ч св <sub>1</sub>	16.82	17.12	17.43	13.74	13.82	13.89	10.41	10.21	12.76	12.89	13.00	17.18	12.76	12.89	13.09	10.96	11.06	18.26	18.81	18.97	19.24	18.55	18.90	19.14	17 05	18.29	18.90	19.65	11.85	12.04	12.25	12.58	13.68	14.13	14.40	16.23	16.52	16.63	16.73
10	<sup>r</sup> u k	0.31818	0.33114	0.36594	0.07680	0.08307	0.08856	0.09196	26600°0	0.18367	0.19573	0.20594	0.59369	0.18367	0.19573	46602.0	0.01670	0.02597	0. 11405	0.97170	0.98847	1.01677	0.90452	0.94045	0.96509	0.93152	18050.0	10000-0	0.68525	0.35739	0.38488	0.40321	0.44101	0.06875	14160°0	0 12500	0.08480	0.10428	0.11163	0.11832
ц	$\left(\frac{F_n}{M^{1}}\right)$	0.64533	0.64533 0.64533	0.64545	0.32291	0.32291	0.32291	0.32291	0.19382	U.14382	1.05808	1.05808	1.05808	0.53249	0.53249	0.53249	6476C*0	55025-0 55055-0	1.94996	1.94946	1.94996	1.94996	1.83732	1.83732	1.63732	1.83732	1.07956	1 07954	1.07956	0.62957	0.62957	0.62957	0.62957	0.21297	0.21297	16712.0	0.12402	0.13993	0.13993	0.13993
12	K	0.86364	0.91953	4/066.0	1.00038	0.42947	0.45850	0.50845	0.03295	0.05276	C0C03.0	0.29811	0.94530	0.52389	0.56088	0.59237	0.61826	68710°0	0.86360	00,000,00	0.9379A	0.96629	0.91127	0.94936	0.97551	0.99296	0.92591	0.443/8	1 • 20441	1.08843	1.17358	1.23041	I.34780	0.63365	0.84295	0.95852	1.15368	106661.1	1.57803	1.67269
13	Q	0.28018	0.29480	0.45058	0.2449/0	0.28874	0.42336	0.42526	0.25755	0.41075	+0062.0	0.47109	0.46101	0.27643	0.29311	0.44286	0.43998	0.256677		C120C • 0	52207 O	0.48058	0.30147	0.30420	0.49508	0.47863	0.29045	0.24939	0.46164	696220	0.29458	0.44958	0.44501	0.25926	0.28516	0.40780	0.41340	67 167 O	0.20267	0.40176
۴ī	$(\frac{h_1}{y_n})(\frac{M_1}{F_n})$	п. п. 0.493	0.523	0.529	0.567	0.2570	0.274	0.303	0.020	0.032	4, T•0	0.195	0.561	0.345	0.368	0.387	0.402	0.052	0.081	0.409	0.448	00	0.492	0.512	0.525	0.534	0.492	0.527	2/5-0 35-0	0.568	0.611	0.640	0.700	0.323	0.429	0.488	0.587	0.607	0. 745 0 700	0.846

7- 4 COntd.

		TABLE 7-	-5 - 6	DEOMETRY IN	V RAW and	CALCULATEL	DATA.						
ч	N	m	4	5	9	7	80	6	10	11	12	EL	7 <b>7</b> ,
Run No	W	Ł	ď	$\mathbf{y}_{\mathbf{n}}$	ง <sup>น</sup> ี้	, M	Ð	y1	* <sup>I</sup>	$\left(\frac{F_n}{M!}\right)$	К	Q	$\left(\frac{h_{1}}{y_{n}}\right) \left(\frac{H_{1}}{F_{n}}\right)$
			cfs	CB				CE	1				:
E-2-1	0.0	0.108	<u>°</u>	14.57	C.00014	0.278	0	16.70	0.14619	0.15004	1.92652	0.55879	0.974
	n n 0 0	0.108	1.0	14.57	0.00014 0.00014	C.278	0.80	16.66 14 48	0.14345	0.15004	1.89027	0.62400	0.956
	0.3	0.108		14.57	0.00014	U. 278	06.0	16.69	0.14550	0.15004	1 - 41746	0.57252	0.965
	0.3	0.108	0.1	14.57	C.00014	0.278	0.95	16.69	0.14550	0.15004	1.91746	0.58354	0.970
		0.108	0 : -	14.57	0.00014	0.278	1.00	16.85	0.15649	0.15004	2.06250	0.62834	1.043
E-2-3	n.⊷ ⊃ 0	0.151	- 0 - 1	11.63	U. 00038	C. 286 C. 286	0.80	14.52	0.24850	0.27786	1.75339	0.55687	0.894
	0.3	0.151	1.0	11.63	U. 0003H	C.286	0.85	14.59	0.25451	0.27786	1.19611	0.56107	0.916 0.916
		0.151		11.63	U.00038	0.286	06.0	14.48	0.24506	0.27786	1.72898	0.56376	0.882
		0.151		11.63	U.00038	G. 286	0.95	14.53	0.24936	0.27786	1.75949	0.57760	0.897
5-3-]	0.3	0.196	• T	9.78	u. 00060	0.290		13.51	14502.0	U.21186	1.86326	0.61759	0.950
+ 1	0.3	0.196	<b>1.</b> J	9.74	000000	C. 2 30	0.80	13.41	0.37117	0.45397	1.58783	0.1905	0.840
	0.9	0.196	<b>1</b> •0	9.78	0.00060	0.230	0.85	13.52	0.38241	0.45397	1.63651	0.55917	0.842
	• • •	0.196		9.18	L. 00060	0.290 0.290	C.30	13.42	0.37219	0.45397	1.59226	0.55689	0.820
		0.196		41.0 42.0	0.00060	0.62.0	0°0°	13.47	0.37730	0.45397	1.61438	0.57292	0.831
E-3-3	0.3	0.243	) ) • []	8.46	0.00000	C. 293		12.89	0.52366	U.45397 D.48054	1.82257	0.60916	0.937
	0.3	0.243	1.0	8.46	U-00110	0.293	n.80	12.89	0.52364	0.68956	1.46019	0.61719	0.759
		0.243	]•∩	8.45	0.100.0	C.293	0.85	12.91	0.52600	0.68956	1.46691	0.55756	0.763
		0.243		0 4 9 0 4	0.100.0	C. 293	0.90	12.94	0.52955	0.68956	1.47699	01154.0	0.768
		0.243			01100-0	6 4 4 0 6 6 7 - 0 6 6 7 - 0		12.51	0.52600	0.68956	1.46691	0.56895	0.763
F-3-5	0.3	0.316	2.0	11.27	0.00120	0.287	• • •	20.34	0.80479	000000 1.21396	1.24123	0.60207	0.821
	e.0	0.316	2.0	11.27	C.09120	C.287	0.80	20.27	U.79858	1.21396	1.24732	0.61468	0.658
		0.316	5°C	11.27	C.0C120	0.287	0.35	20.38	0.80834	1.21396	1.26307	0.55539	0.666
		0.316		12.11	0.00120	C.287	0.9C	20.38	0.80834	1.21396	1.26307	0.54336	0.666
	0.3	0.316	2.0	11.27	C.00120	C. 287	1.00	20.94	U.808034	1.21346	1.26307	0.56364 0 50361	0.666
E-4-1		C.407	2.0	9.52	U.00210	C.291	•	19.88	1.08824	1.96174	1.03119	0.55080	0.555
		104-0	0 ° C	20.50 2012	01200	C•291	0.80	19.99	1.09979	1.96174	1.04271	0.61256	0.561
	0.3	0.407	2.0	9.52	01200-0	1.50	00000	20.02	1.10294	1.96174	1.04585	0.55356	0.562
	C•3	0.407	2.0	9.52	0.00210	167.0	0.95	20.21	1.12290	1 - 96174	20020-1	U.53688 D.55017	0.564
	m r 0 c	0.407	<b>7</b> •0	9.52	u.00210	C.291	1.00	20.74	1.17857	1.96174	1.12139	0.58470	0.601
E-4-3		16410		3° - 5	G.00360	0.293	•	19.56	1.34532	2.87497	0.85155	0.54963	0.468
		0.497	0 · · · ·		0.00360	C. 293	0.80	19.56	1.34532	2.87497	0.85155	0.61087	0.468
	0.3	C.497	2. Ú	8.34	0.00360	0.243	06.0	14.72	1 36651	2.8/49/	0.88009	0.55210	0.483
	0.3	0.497	2.0	8.34	0.00360	0.293	0.95	19.77	1.37050	70478.C	0.85439	0.55542	0.415
1	~ ~ ~	0.497	<b>2</b> •0	8.34	L.00360	0.293	1.00	20.53	1.46163	2.87497	0.93073	0.57848	0.508
1		0.094		53°C3	0.00012	C.651	0.0	33.38	0.00876	0.02098	0.82654	0.56498	0.418
	0.7	0.094	00	33.09	0.00012	0.651	0.40	33.36 12 27	0.00816	0.02098	0.76953	0.63290	0.389
	0.7	0.094	3.0	33.09	U. 00012	0.651	06.0	5.4.EE	0.01027	0.02098	0.19803	0.57116	0.403
		0.094	3.0	33.09	U.00012	C.651	0.95	33.40	0.00937	0.02098	0.88355	05100-0	0.440
	•••	1.044	<b>3.</b> U	33.09	0.00012	0.651	1.00	33.46	0.01118	0.02098	1.05459	0.66392	0.533

14	$\mathbf{y}_{n}^{\mathrm{F}}$		0.743	0.725	0.731	C.737	0.761	0.798	0.691	0.712	0.729	0.729	0.758	0.179	0.217	0.217	0.217	0.217	0.204	112.0	086.0	- V + • O	0.456	0.468	0.503	0.711	0.727	0.717	0.714	0.740	0.785	0.406	0-684	0.699	0.708	0.758	C.338	0.386	0.404	0.404	0.341	0.452		0-0-0	0.695	0.668	
13	Q		0.56229	0.62903	0.56781	0.58869	0.59444	0.64826	0.56021	0.62604	0.56523	4061 4.0	0.s8795	0.63637	0.56021	0.62604	0.56523	0.21904	66186.0 50,00	16060.0	0 63656	0.5558	0.58070	0.58907	0.63841	0.55830	0.62325	0.56285	G.57028	0.58203	0.62559	121220	0.56122	0.56429	0.57796	0.61824	0.55920	0.62459	0.56397	0.5/440	0.50482	U.03U65 0 56570		0.55974	0.55893	0.57431	
75	К		1.47087	1.43439	1.44655	1.45871	1.50736	1.58034	1.35211	1.39293	l.42558	1.42558	1.48273	1.52356	0.42242	0.42292	0.42242	26224-0	06966.0	0. 775 .1	0.81536		0.87136	0.89376	0.96098	1.36572	1.39480	1.37667	1.37120	1.42051	1.50823		1.28905	1.31792	1.33442	1.42940	0.62899	0.71943	0./5221	12261.0		U.8421U	102721	1 - 2 - 1 1 - 2 - 1 - 1 - 1 - 2 - 2 - 2	1.28213	1.23120	
ц	$\left(\frac{F_n}{M}\right)$		0.04922	0.04922	0.04922	0.04922	C.O4922	C.04922	0.09533	0.09533	C.09533	0.04533	0.04533	C.09533	0.09533	0.09533	0.04533	66480.0	000000 V		0.0850.	C.00503	0.08503	0.08503	0.08503	0.17548	0.1771.0	0.17548	0.17548	0.17548	0.17548 0.26761	0.26761	0.26761	0.26761	0.26761	0.26761	0.13156	0.13156	04151.0	0.13196 0.13156		0.10100		50265.0	0.39205	0.39205	
10	<sup>*</sup> u <sup>*</sup>		C.03657	C.03566	91.550.0	0.03626	0.03747	C.03929	0.04540	U.06788	0.06947	0.06947	0.07225	0.07424	0.02064	C.02064	0.02054	1.01040	0 000047	0.0000	0.03630	16460-0	0.03879	0.03978	L.04276	0.12481	U.12874	0.12581	0.12531	0.1221 U	0.151/4	0.18602	C.18313	0.18718	0.18949	0.20277	0.04448	0.05084	0.05315 0 05315	05142		00460.0	0 26535	0.26865	0.27261	0.26205	2,000 0
6	У1 сш		34.30	34.27	<b>34.2</b> 2	34.29	34.33	34.39	26.85	26.90	26.94	26.94	10.72	27.06	11.62	25.71	1.02	11.02	50.00 12	10.76	20.84	20.82	20.89	20.91	20.97	22.62	22.62	22.64	22.63	21.22	20.00 20.00	20.53	20.48	20.55	20.53	20.82	18.08	18.19	10.23	18.20		19.21	19.17	19.22	19.28	19.12	10 60
8	Ð	ļ	•	ر <b>8</b> ر	0.85	- ° ° °	0.95	1.00	.0	04.0	C.85	00.0	5. °C	- 1 - 1		0.80		30.0			0.80	(- 85	0.90	0.95	1.00	•	0°°°	0.85 20	0.30			0.80	C • 35	06-0	0.95	1.00			00.00	0 • 4 0 C • 4 0			0.80	0.85	0.90	0.35	00
7	'n		C.425	C.425	0.425	C • 4 2 5	C.425	C.425	0.460	C.++0	C•400	C.460	C • 4 6 0	C•460	0.440	0 0	0.400			0.083	C. 083	C.083	C.683	C. 693	C. 683	C.475	C.475	C.475	0.475	C• + - 0		C.482	C . + H 2	C.482	C.482	C.482	C. C 3 -	1000	C.647	C.687	6.687	0.486	C - 4 H 6	C.485	C.486	C.486	0 494
6	$s_n$		0C012	(.00012	C.00012	u.00012	L.00012	L.0501∠	UC021	L.9CJ27	U. 00027	0n027	12000-1	12000	(.uuuz/	L-00027	12000-0	00027	L. 00027	00046	0.00046	L.00046	0 <b>.0</b> 0046	<b></b> 00045	00046	L.C0346	v.00346	00046	00044	010000	U. 00085	J.03085	C.00085	<ul> <li>00085</li> </ul>	U.00085	0.00085	C. UCURS		0.00085	u. 00085	000.85	0.00011	11000.0	C.00011	0.00011	u.00011	0.00011
5	у <sub>п</sub> сш		33.09	\$3 <b>.</b> C9	33.09	53.09	13.04	<b>33.</b> C9	25.19	<1.14	25.1V	20.19 21.20	61.62	61.62	20•14	61.02	50.10 25.10	25.19	25.14	20.11	11.02	20.11	11.02	20.11	20.11	20.11	20°C4	11.02	11.07		17.31	17.31	17.31	17.31	17.31	11.31	10.11	12.11	17.31	17.31	17.31	15.15	15.15	15.15	15.15	15.15	15.15
4	ي cfs	I	J.L	3.0		ۍ . د	ः • १	<b>ر.</b> د	<b>ع</b> • د	ر. د	ر. ۲۰	- - -	<b>.</b>		)				0.0		<b>3.</b> )	3.0	3.U	3.0	<b>3.</b> U	<u>ع</u> • ر		) : • •	- - -		າ ຕ	5.0	3•C	ر <b>، د</b>	بە ت	n : • •				3.0	3.0	a. ن	3.0	3.∪	J. U	0.0	3.0
e	F		0.CJ4	C.94	C.C94	0-0-0-0	C.C.14	C • C 44	C•142	C.142	0.142	C•142	C•142	2+1-0	0 • T + V	C•145	0 142	0.142	C.142	C.199	0.199	0.149	0.144	0.199	661.0	C.194	0.700	0 100	0.139	0.199	C.249	0.249	0.249	C.249	C.249	0.244	0.440	C - 249	0.249	C.249	0.249	C.304	0.304	0.304	C . 304	0.304	0.304
2	¥	I	0.5	5. 0	0 0 0	0 °	•••	5 i 1 i	0.2 0			0, u	0 u	• •	- u - c	n u 5 c	. 5	0.5	c.5	C.1	0.7	0.7	C.7	0.7	0.7	0. 0			2.5		0.5	0.5	0.5	5 C				0.7	0.7	0.7	0.7	c.5	0.5	0.5	5 C	0.0 1	c.0
ı	Run No.		E-5-3					1	E-6-1					c 7 4	(-0-j					E-6-5						E-6-7					E-7-1					5-7-3						E-7-6					

7-5 Contd.

	7	)(  = 7 )(	ч -	370	380	101	104	+28	+72	392	420	+22	+22	448	489	539	542	529	550	563	707	590	597	595	594	508	505	471	465	493	505	531	568
	*	₽́	<b>1</b>	0	•	0	•	•	0	•	0	0	0	•	•	0	•	•	0	5	0	0	•	••	•	•	•	•	0	0	•	••	• 0
	ព	Q		0.55798	0.62283	0.56245	0.56881	0.58103	0.62379	0.55621	0.62029	0.56025	0.56079	0.57558	0.61395	0.55406	0.61721	0.55759	0.55118	0.56901	0.60217	0.55249	0.61497	0.55564	0.54425	0.56425	0.59033	0.55462	0.61801	0.55828	0.55366	0.57071	0.60521
	ศ	м		0.66554	0.68407	0.73352	0.72733	0.77065	0.85125	0.65457	0.70302	0.70707	0.70707	0.75166	0.82495	1.12790	1.13444	1.11048	1.14970	1.17372	1.25690	0.99247	1.00540	1.00229	1.00073	1.02563	1.00367	0.72253	0.71141	0.75880	0.77840	0.82343	0.88580
	я	1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1		n.19452	0.19452	0.19452	0.19452	0.19452	0.19452	0.34321	0.34321	0.34321	0.34321	0.34321	0.34321	0.68533	0.68533	0.68533	G.68533	0.68533	0.68533	1.13695	1.13695	1.13695	1.13695	1.13695	1.39302	0.57257	0.57257	0.57257	0.57257	0.57257	0.57257
	10 ,	* <i>E</i>	yn.	0.07195	6.01393	0.07921	0.07855	0.08317	0.09175	0.13440	0.14400	0.14480	0.14480	0.15360	0.16800	0.43760	0.44000	0.43120	0.44560	0.45440	0.48480	0.67110	0.67871	0.67641	0.67586	0.69106	0.84318	0.26996	0.26616	0.28232	0.28897	0.30418	0.32510
	6	y1	E	16.24	16.27	16.35	16.34	16.41	16.54	14.18	14.30	14.31	14.31	14.42	14.60	17.97	18.00	17.89	18.07	18.18	18.56	17.58	17.66	17.64	17.63	17.79	18.10	13.36	13.32	13.49	13.56	13.72	13.94
	80	0			0.80	0.85	0°-90	0.95	1.00	•0	0.80	C.85	ს • 90	0.95	1.00	•0	C.80	0.85	C.90	0°35	1.00	с	C.8C	0.85	0.90	0.95	1.00	• •	0.80	0.85	06.0	0.95	1.00
	7	-x		0.690	C.690	C.640	C. 090	0.690	0.690	0.693	C.693	0.693	C. 693	C.693	C. 693	0.491	C.491	C.491	0.491	0.491	0-491	C.493	0.493	0.493	0.493	C.493	C.494	0.695	0.095	0.695	0.095	0.695	0.095
	6	sn		0.00011	1100011	11000.0	0.00011	L.00011	000011	C.00229	U.00220	U.00220	U.00220	0.00220	00220	U.00220	U.07220	0.00220	0.00220	U.00220	0.00220	0.00360	0.00360	U.00360	0.01360	U.00360	0.00360	C.00360	U.O0360	0.00360	U.00360	0.00360	0.00360
	5	$\mathbf{y}_{\mathbf{n}}$	EI C	15.15	15.15	15.15	15.15	15.15	15.15	12.50	12.50	12.5C	12.50	12.50	12.50	12.50	12.50	12.50	12.50	12.50	12.50	10.52	10.52	10.52	10.52	10.52	9.82	10.52	10.52	10.52	10.52	10.52	10.52
	4	ď	cls	0.0	<b>J</b> •€	J.U	3.0	3.0	3.0	3.)	3•0	3.0	3.0	3.0	3.0	<b>3</b> •0	3 <b>.</b> Ū	3.0	3.0	3.0	3.0	3.0	9°0	3.0	3.0	3.0	3.0	3.0	3.0	Э°О	3.0	3.0	3•0
5 Contd.	e	Fn		C.304	0.304	0.304	0.304	C.304	0.304	0.406	0.406	0.406	0.406	0.406	0.406	0.406	0.406	0.406	0.406	0.406	0.406	0.526	0.526	0.526	0.526	0.526	0.583	0.526	0.526	0.526	0.526	0.526	0.526
4	8	M		12.0	0.7	0.7	0.7	0.7	0.7	0.7	0.7	C.7	0.7	0.7	0.7	0.5	0 <b>.</b> 5	0.5	0.5	0.5	C.5	0.5	0.5	0.5	0.5	0.5	0.5	0.7	0.7	0.7	0.7	0.7	0.7
	Ч	Run No		8-7-8	-					5-8-1	1					<b>E-8-</b> 3						5-8-5						E-8-7					

	ы	CABLE 7-6 . G	ROMETRY	Va RAV	I and OALCUI	LATED DATA	.1						
٦	7	ę	4	5	6	7	80	6	o <b>r</b> *	11	12	ព	14 1
Run No.	W	F n	ູ cfs	y cmsn	sn	- ¥	€ <sup>7</sup>	y_1 cms	ur k	$\left(\frac{F_n^2}{M}\right)$	К	Q	$\begin{pmatrix} h_1 \\ Y_n \end{pmatrix} \begin{pmatrix} M^t \\ F_n \end{pmatrix}^2$
S-1-1	0.5	0.502	3.0	10.85	0.00360	0.493	.0.	17.95	0.65438	1.03811	1.26071	0.58523	0.630
	0.5	0.502	0.0	10.85	U.00360	0.493	30.0	16.54	0.52442	1.03811	1.01034	0.53265	0.505
	<b>9.</b> 0	0.502	0°. 6	10.85	0,00360	0.493	45.0	15.32	0.41198	1.03811	1/661.0	0.43046	0.397
S-1-2	0 C	0.502		10.85	U.00360	0.493	15.0	13.60	0.25346	1.03811	0.48830	0.58523	0.244
	0.5	0.502	3.0	10.85	u.00360	0.493	0.01	13.13	0.21014	1.03811	0.40485	0.53265	0.202
	0.5	0.502	3.0	10.85	u.00360	0.493	45.0	12.68	0.16866	1.03811	0.32494	0.43046	0.162
S-1-3	6.0	0.504	ວ. ຕິ	10.83	0.00210	0.896	.0.	11.17	0.03139	0.31594	0.19873	0.55936	0.099
-	0.5	0.393	0.0	12.78	0.00210	0.490	0.41	11.13	0.43036	0.31594	26271.0 78985.1	0.57208 74657	0.088 (1.670
*-T-0	0.5	0.393	3.0	12.78	0.00210	0.490	15.0	17.89	0.39484	0.64239	1.24486	0.59817	0.622
	0.5	0.393	3.0	12.78	U.00210	0.490	30.0	17.21	0.34664	0.64739	1.07921	0.51683	0.540
	• 0 0	0.393	0 : M	12.78	0.00210	0.490	45.C	16.10	0.25978	0.64239	0.808/9	0.41247	0.404
	0	676.0 696.0	0 0 0	12.78	0.00210	0.693	0. 1 5.0	14.45	0.13061	0.32142	0.81310 0 76928	0.55315	0.401
	0.7	0.393	0.6	12.78	U.00210	0.693	30.0	14.19	0.11033	0.32142	0.68651	0.49484	0.343
	0.7	0.393	3.0	12.78	0.00210	0.693	45.0	13.81	0.08059	0.32142	0.50149	0.38785	0.251
S-1-5	1.0	0.289	3.0	15.70	0.00120	0.690	•0	16.72	0.06497	0.17516	0.74182	0.54696	0.371
	0	0.289		15.70	02100.0	0.690	1 0 • 0 F	16.58 16.55	0.05605 0.05414	0.17516	0.64001 0.61819	0.01648	075.0
	0.7	0.289	3.0	15.70	0.00120	0.640	45.0	16.36	0.04204	0.17516	0.48000	0.36747	0.240
	0.5	0.289	3.0	15.70	0.00120	0.485	•0	19.69	0.25414	0.35380	1.43664	0.56171	0.718
	0°2	0.289	0 ° C	15.70	U.00120	0.485	15.0	19.41	0.23631	0.35380	1.33582	0.55918	0.668
	0 C	0.289	00	15.70	0.100120	0.485	30.0	18.87 18.05	0.20191	0.35380	1.14139 0.86616	0.49783	0.571
<u>م</u> اردي	0.5	0.193	0.0	20.55	U.00048	0.474		22.98	0.11825	0.16525	1.43113	0.54575	0.716
ļ	<b>6.</b> 0	0.193	3.0	20.55	0.00048	0.474	15.0	22.79	0.10900	0.16525	1.31923	0.51309	0.660
		0.193	0.1	20.55	0.00048	0.474	30.0	22.39	0.08954	C.16525	1.08366	0.47459	0.542
		661 • O	- - - -	20.55	U. 00048	0.414	0°0	21.15	0.03930	67691.0 7 07984	1.38402	0.36758 0.43099	0.692
	0.7	0.193	9°0	20.55	0.00048	0.682	15.0	21.10	0.02676	0.07986	0.67024	0.41262	0.335
	0.7	0.193	3.0	20.55	U.00048	0.692	30.0	21.00	0.02190	0.07986	0.54838	0.45341	0.274
	0	0.193	0.0	20.55	0.00048	0.682	45.0	20.91	0.01752	0.07986	0.43870	0.34269	0.219
S-2-1		460.0		33.11	0.00012	0.651	15.0	95.55 85.55	0.00815	66020°0	0.77852	U.5U456 0 35955	0.389
	0.7	0°044	3.0	33.11	0.00012	0.651	30.0	13.33	0.00664	0.02095	0.63435	0.41686	0.317
	0.7	0.094	3.0	33.11	0.00012	0.651	45.0	33.27	0.00483	0.02095	0.46135	0.30425	0.231
S-2-3	0°2	0.094	0.0	33.11	0.00012	0.425	•	34.33	0.03685	0.04915	1.49921	0.52118	0.750
	n 1 2 C	160.0		11.66	0.00012	624°0	90°0	16.66	0.02597	0.04915	1.05682	0.43980	0.528
C-C-2		0.105	1.0	14.78	0.00014	0.777	5°.C	17.07	0.15494	0.04915	0./0045 2.14560	0.52822	066.0
	0.3	0.105	1.0	14.78	0.00014	0.277	15.0	17.03	0.1523	0.14447	2.10753	0.50536	1.054
	6.0	0.105	1.0	14.78	0.00014	0.277	30.0	16.52	0.11773	0.14447	1.62982	0.47060	0.815
	0.0	0.105	1.0	14.78	0.00014	0.277	45.0	16.21	0.09675	0.14447	1.33945	0.36123	0.670
S-2-3	10	0.189	1.0	10.00	0.00013	0.290	0. 15.0	13.95 13.84	0.39500	0.42598 0.42598	1.85453 1.80289	0.56575	0.927
							ļ						

	0.716 0.716 0.6578 0.6578 0.65978 0.65978 0.6932 0.5978 0.5978 0.3719 0.3719 0.3719 0.3719 0.3719 0.3719 0.3719 0.3719 0.5590 0.55000 0.55000 0.55000 0.55000 0.55000 0.55000 0.55000 0.5500000000	0.377 0.213
D I3	0.550367 0.58691 0.58691 0.53518 0.59713 0.55713 0.55713 0.55713 0.55713 0.557122 0.55728 0.55728 0.55728 0.55721 0.55722 0.57722 0.57	0.48206 0.37374
I2 K	1.42729 1.42729 1.185728 1.185728 1.25353 1.25353 1.25353 1.25353 1.25353 1.25353 1.25353 1.25353 1.25358 1.25328 1.25328 1.25328 1.25328 1.25328 1.25328 1.25328 1.25328 1.25338 1.25	0.75319 0.42681
$\frac{11}{(\frac{F_n}{M})}$	0.42598 1.11951 1.11951 1.11951 1.11951 1.11951 1.76376 1.76376 1.76376 1.76376 1.76376 2.69036 2.69036 2.69036 0.09205 0.09205 0.09205 0.09205 0.09205 0.04293 0.022118 0.022118 0.022118 0.022118 0.022118 0.022118 0.022118 0.022118 0.022118 0.022118 0.022118 0.022118 0.022118 0.022118 0.022118 0.022118 0.02218 0.02218 0.02218 0.02218 0.02218 0.02218 0.02218 0.022	0.21186 0.21186
<b>v</b> n 10	0.30400 0.80172 0.65262 0.65262 0.65262 0.55862 1.11437 1.015364 0.88158 0.88158 0.88158 0.88158 0.98641 0.05797 0.05797 0.05797 0.05797 0.057881 0.057881 0.057878 0.01371 0.01371 0.01371 0.01371 0.01371 0.055881 0.012378 0.01237 0.105788 0.11544 0.11544 0.11544 0.11544 0.105788 0.115440 0.1154400 0.11544000000000000000000000000000000000	0.07979 0.04521
9 v1 cms	13.04 20.90 20.90 19.28 20.89 20.89 20.89 19.59 20.88 19.59 25.41 19.89 25.88 19.29 25.88 19.20 25.88 19.20 25.88 19.20 25.88 19.20 25.88 19.20 25.88 19.20 25.88 19.20 25.88 25.88 25.88 25.88 20.20 25.88 20.20 25.88 20.20 25.88 20.20 25.88 25.88 25.88 25.95 25.88 20.20 25.88 25.95 25.95 25.95 25.95 25.95 25.95 25.95 25.95 25.00 20.00 25.00 25.00 20.00 25.00 20.00 25.00 20.00 25.00 20.000 20.000 20.000 20.000 20.000 20.000 20.000 20.0000 20.00000000	20.30 19.65
œ	30.00 30.00 45.00 30.00 45.000	30.00 45.00
7 W	0.290 0.286 0.286 0.286 0.290 0.290 0.293 0.293 0.293 0.285 0.285 0.285 0.285 0.285 0.285 0.285 0.285 0.285 0.285 0.285 0.292 0.293 0.292	0.478 0.478
ං ග් <sup>ස</sup>	0.00013 0.00013 0.00013 0.00013 0.00013 0.000220 0.000280 0.000280 0.000280 0.0000000000	0.00089 0.00089
ر vn cms	111.60 111.60 9.88 9.88 9.88 9.88 9.88 8.53 8.53 8.53 8.53 8.53 12.12 12	18.80 18.80
4 Q Cfs		3.0
e e	0.200 0.3185 0.303 0.303 0.303 0.385 0.385 0.385 0.385 0.385 0.385 0.385 0.385 0.385 0.385 0.385 0.189 0.189 0.139 0.139 0.139 0.139 0.139 0.139 0.139 0.231 0.231 0.231 0.231 0.231 0.231	0.220
W N	00000000000000000000000000000000000000	0.5
1 Run No.	5-3-1 5-3-2 5-4-2 8-5-1 8-5-1	

7-6 Contd.

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ч	3	e	4	5	ý	7	Ø	6	10	ц	12
tun No	Σ	г ц	ل د fs	vn cm	$s_n$	- W	or r	у1 св	$\frac{\lambda_n}{n}$	$\left(\frac{\mathbf{F_n}}{\mathbf{M^1}}\right)$	К
-1	0.2	0.462	2.0	9. I b	0.00364	0.241	15.00	23.90	I.12831	3.6/1/9	0.81961
2	0.2	0.388	1.0	6.2C	i.00254	0.245	15.00	13.44	1.16774	2.49464	0.87930
e	0.2	0.310	1.0	7.20	C.00138	C.244	15.00	14.22	0.97500	1.61427	1.15494
- 4	0.2	0.194	1.0	9.84	U. 00058	0.238	15.00	14.92	0.51626	0.66266	1.51967
. 10	0.2	0.046	1.0	15.70	0.00013	0.217	15.00	18.40	0.17197	0.19649	1.73509
. 9	C•2	0.462	2.0	8.76	U.00364	0.241	30.00	22.54	1.57306	3.67779	0.79641
2	0.2	0.750	2.0	6.34	L.00254	0.245	30.00	13.67	1.15615	9.34820	0.19012
• 60	0.2	0.310	1.0	7.19	0.00138	0.244	30.00	13.80	0.91933	1.62078	1.08246
6	0.2	0.194	1.0	<b>9.</b> 84	ú.00058	0.238	30,00	15.79	0.60467	0.66266	1.78335
2	0.2	0.096	1.0	15.70	6.00013	C.217	30.00	18.33	0.16752	0.19649	1.69002
Ħ	0.5	0.467	2.0	8.69	C.00364	0.496	15.00	13.27	0.52704	0.88892	1.01749
7	0.5	0.387	2.0	9.85	L.00234	0.494	15.00	13.59	0.37970	0.61357	1.09851
ភ	0.5	0.292	2.0	11.89	0.00128	0.492	15.00	14.63	0.23045	0.35263	1.20857

TABLE 7-7 GEOMETRY V<sub>h</sub> RAW and CALCULATED DATA.

		A-/ THAN	121025	H TA XH	tAW and CALC	ULATED DA	TA.					
г	2	Э	4	ŝ	6	7	80	6	10	ц	12	ព
Run No.	М	Ł	് Cfs	yn cu	s <sup>n</sup>	-w	у <sub>1</sub>	* [ ] %	$(\mathbf{F_n}^2)$	ĸ	Q	Å, <sup>H</sup>
-	0.5	C.486	2.0	8.46	0.1750	0.546	12 42	0.46.800	01002 3			:
2	0.5	C.389	2.0	9.82	C.00231U	0.545	12.98	0.32179	0.50961	26888.0	0.73540	0.590
m	<b>6.</b> 0	0.303	2.0	11.50	0.001350	C.543	14.04	0.21139	0.31233	1.24103	04042-0	0.677
-4 '		0.197	2.0	15.44	0.000610	C.537	17.09	0.10687	0.13493	1.52036	0.80601	0.797
<u>``</u>	2 c	0,452	.) : 	15.40	0.000175	C. J 3 /	15.87	0.03052	0.03399	1.77568	0.88646	0.848
0 0	0 C	764.0		09.6 2	C. 604080 2. 882.52	C. 347	11.37	1.03036	1.64008	1.10963	0.67701	0.610
<u>~ o</u>		0.004	•	0.4	0.002450	C. 346	11.58	0.78980	1.101.1	1.33494	0.69731	0.717
ρα		0.189	- - -	1	0.001380	C. 345	11.99	0.55512	0.65662	1.60710	0.72265	0.845
~ ~		001.0	) · ·	10.00	0.00084	0.341	13.24	0.31610	0.30188	2.03513	0.76245	1.047
3;		0.000	) ·	15.68	0.000132	C. 328	18.60	0.18622	0.08623	4.28165	0.83130	2.160
1;		0.448	⊃: ∽:	10.52	0.003680	0.145	13.17	0.20604	0.44557	0.71653	0.74224	0.462
Ľ,		0, 392	<b>3.</b> 0	12.80	0.002230	0.144	14.40	0.12500	0.27749	0.76010	0.76680	0.450
7;		162.0	: : :	15.60	0.001220	C.140	16.65	0.06731	0.15487	0.78886	0.79839	0.435
4 k		0.1.0		20.13	0.000523	C. / 3.3	21.36	0.03039	0.06736	0.86489	0.84559	0.451
4		0.100		21.42	0.000131	C. 125	25.10	0.0112H	0.01781	1.25267	0.42688	0.633
		0.100 0.00E		12.10	0.000135	0.331	13.01	0.0/521	0.08778	1.69501	0.83028	0.857
2		1001		14.60	0.000140	0.342	10.19	0.08289	0.07621	2.15473	0.83842	1.088
1			⊃ ⊂ • •	4. · · · ·	062600.0	0.526	23.84	0.14726	0.12978	2.18955	0. HOB1B	1.135
5		- 4 A - 0	- - - -	00 • H V	191000.0	C. 316	26.14	0.06002	0.03592	3.30676	0.88309	1.671
4 C		104	•	00.0		5 - C - C	22.435	1.65648	1.90630	1.61674	0.67141	0.869
		0.296		04 11	0.001.040	24 <b>6.</b> 0	+c • 7 7	1.51411	1.32619	1.86779	0.68843	166.0
800		0.142	0.0	1 5 7 0	0.0001780	0.338	23.12	0.96265	0.76689	2.40901	0.71495	1.255
		0 096		01.01	096000.0	C. 3/H	14.02	0.68217	0.34374	3.88548	0.15565	1.985
		0.040			666000.0	0.328	18.60	0.18622	0.08623	4.28165	0.83130	2.160
140	7.0			+ C • 7 4	0.00012F	C. 135	14.54	0.12365	0.08262	2.96521	0.83376	1.497
BOT		0 147	5 a 6 a	0.000	0.000201	50/ •D	33.77	0.02240	0.01800	2.46396	0.42622	1.245
	0	0,080	 	10.02	0.000001		29.68	0, 03523	0.04206	1.63389	0.87352	0.838
	O	0.201	• •	10 00		C. 1 10	24.05	0.01008	0.01251	1.59942	0.94976	0.806
) t	•		2	1 7 . 0 7	011600.0	C• 7 / 8	22+84	0.14832	0.14517	1.9624B	0.80196	1.022

AW and CALCULATED DATA.

		TABLE 7-9	GEOMET	TRY VII	RAW and	CALVULATED	DATA.						
-	3	e	4	5	ó	7	80	6	10	ц	12	13	<b>۲</b> ,
Run No.	И	г ц	Э	y <sub>n</sub>	sn	ч,	β	ጙ	'E' :	$(\frac{F_n}{m})^2$	K	Q	h <sup>*</sup> ( <u>+</u> ) ( <u>+</u> )
			cf9	СĦ				튑	vn v	- 14			yn r <sub>n</sub>
-	0.35	C.447	0.1	5.64	C.0041C0	C.329	0.50	12.22	1.16667	1.84420	6.41940	0.39976	0.633
101	0.35	0.355	1.0	6.58	0,002540	C.318	0.50	12.39	0.88298	1.23920	5.36265	0.40714	0.713
e	0.35	0.278	1.0	7.74	C.CO1380	C.310	0.50	12.67	0.63695	0.80499	4.43386	0.41530	0.791
-4 1	0.35	C.187	- 1 - 0	10.08	0.000584	C.294	0.20 .0	13.83	0.37202	0.40454	3.55381	0.42865	0.920
Ś	0.00	0.632	· · ·	12.13 a 50	C.000124	L.211		19 49	0.12U19	19611.0	64140 6.65206	0,4540	1.040 0.505
0 6	0.35	0.391	2.0	67.0	0.002340	C.318	0.30	19.70	1.01226	1.50498	5.81817	0.42427	0.673
- 00	0.35	0.278	1.0	7.73	C.001380	C. 328	0.30	12.24	0.58344	0.72092	4.25285	0.42930	0.809
. 6	0.35	C.191	1.0	9.94	C.00C560	C. 320	06.0	13.28	0.33602	0.35711	3.45520	0.43415	0.941
5	0.35	0.096	<b>1.</b> )	15.71	0.000132	C.286	0.30	17.50	0.11394	0.11296	2.52687	0.44222	1.009
7	0.35	0.478	2.0	8.56	C.003640	C.326	•	17.68	1.06542	2.15100	4.64210	0.65525	0.495
ន	0.35	C. 397	0 0 7 ~	9.68	C.002340	C. 125	ن ن	16-11	0.85640 0.43073	1.49063	4.2/092	0.63444	67 G • 0
ង	0.00	0.103		15.67	0.000560	C. 534	• ·	07.02	0,000.0	1.37354	2.90076	0.56170	0.823
<b>1</b> 2	0.35	0.096	1.0	15.71	0.000132	C. 327	••• ت ت	17.04	0.03466	0.08643	2.32732	0.49382	0.979
272	0.50	C.478	2.7	10.59	0.003410	C.445	C.50	18.59	0.75543	1.15158	4.53756	0.40851	0.656
35	0.50	C.388	2.0	9.83	0.002340	C.452	0.50	15.02	0.52798	0.73655	3.67511	0.41700	0.717
18	0.50	0.241	2.Ŭ	11.91	C.0012R0	C.436	0 <b>5°</b> J	15.92	0.33669	0.44504	2.88341	0.42678	0.757
19	0.50	0.191	1.0	9.94	0.000584	0.428	0.50	11.46	0.15292	0.19860	2.11948	0.44291	0.770
20	0.50	c.101	1.0	15.22	0.000132	C.412	0°20	16.03	0.05322	0.05970	2.00016	0.46809	0.892
น	0.50	0.490	. B	13.02	0.003280	C.46C	0.00	21.16	0.62519	1.13661	3.32239	0.42618	0.550
ន	04.0	0.392	9 : 9 :	10.21	0.002090	C. 444	0.30	22.62	1,656.0	61191.0	3.64584	0.42892	0.103
ខ្លះ	00	C - 100	5 C	15.33	061100-0	C • 4 9 9		17.67	0.15264	0.19636	3.UBUOU 2.14544	0.43630	181.0
4 K	0.50	C. 1CO	00	15.27	0.000132	C • 4 4 9	0.30	15.93	0.04322	0.04978	1.91126	0.44895	0.868
2,43	0.50	0.490	3.8	13.02	C.003280	0.470	• ت	22.56	0.73272	1.08644	4.58154	0.61703	0.674
27	0.50	0.392	3. й	15.12	0.002040	C.472	•0	21.96	0.45238	0.68786	3.07178	0.59270	0.658
28	0.50	0.295	3°8	18.26	0.001500	C.468	•	23.33	0.27766	0.39723	2.44863	0.56475	0.699
59	0.50	195 0	ກ ຕ	24.08	C.000501	C • 4 5 3	• •	21.24	0.13123	0.18486	1.88590	0.52798	0./10
9 F	0.15	0.496		10.94	0.003680	0.697	0.50	13.51	0.23492	0.50599	1.72257	0.42426	0.464
í,	0.75	0.400	3.0	12.62	C.002230	C. 690	0.50	14.39	0.14025	0.33683	1.25428	0.43228	0.416
(8	0.75	0.295	2.0	11.80	0.001280	C.694	0.50	12.64	0.07119	0.18115	0.98696	0.44479	0.393
3	0.75	0.195	2.0	15.56	0.000131	0.670	0.50	16.06	0.03213	0.08477	0.84279	0.46060	0.379
35	0.15	0.101	0 0 7 0	24.13	0.000131	C.615	0.50	24.44	0.01285	0.02696	0.98953	0.48553	0.477
ጽ	0.1	0.440		10.74	0.003680	0 210	0.30	12.88	0.11/133	0.21501	1.16261	0.43220	0.5.0
37		0.400		12.21	0.002200.0	0.715	0.30	12.51	0.12490	10616.0	1. 10/22	0.43931	0.353
n n	0.75	0.195	2.0	15.56	0.000560	0.106	00	15.96	0.02571	0.07618	0.74123	0.44502	0.337
20	0.75	0.101	2.0	24.06	0.000131	C.668	0.30	24.32	0.01081	0.02303	0.97040	0.45362	0.469
43	0.75	0.496	3.0	10.94	0.003680	0.729	•0	12.88	0.17733	0.46321	1.30753	0.57243	0.383
13	0.75	0.402	3.0	12.58	0.002300	0.742	•0	13.92	0.10652	0.29366	1.03667	0.54993	0.363
£3	0.15	0.443	0 ° n	11.80	0.001280	C.744	••	12.41	0.05169	0.35440	0.39313	0.55910	0.146
4	0.75	0,101	0.2	15.54	0.000560	C. 144	• •	15.87	0.02124	0.06896	0.61082	0.48410	0.308
45		101.0	<b>v•</b> 2	61.62	101000.0	071.0	•	63.42	N0CON*0	00210.0	41610.0	1 1 7 5 4 * 0	0000

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	pth t.	081 277 277 277 278 5587 5587 5587 5587 558			٠
	ssection a 30.23 fi			tion Depth .175 ft.	0,000 0,186 0,478 0,478 0,559 0000000000
cfs. cm. ft. sq. ft.	Cr Sta. Traverse ft.	0.814 1.0 1.1 1.1 1.1 1.5 1.5 2.1 2.162 2.162 2.162	ې ۲	Crossect Sta. Traverse 30. ft.	0,805 0,850 0,850 0,850 0,850 11,000 11,000 11,600 11,600 11,600 2,050 2,150 2,150 2,150
5.37 3.3.58 3.3.58 3.3.58 3.58 30.00 30.00 1.30 1.40 1.40 1.40 1.40 1.40 1.40 1.40 1.4	ile of Jet Depth ft.		10 24,43 cm 24,43 cm 24,11 cm 22,11 cm 30,00 ft 0,864 sq 0,0	of Jet Depth ft.	0.629 0.572 0.572 0.459 0.419 0.419 0.205 0.206
Run No. 91 Y2- Model Sta. L9b	Sta. Frof		Run No. = Y Y2 = = Madel Sta. = L/b	Profile Sta. ft.	a. 88 8.98 8.98 8.98 8.98 8.98 8.98 8.98
	section Depth 30.21 ft.	0,000 0,000000		stion Depth 0.19 ft.	0.00 0.175 0.175 0.175 0.176 0.1760 0.160 0.160 0.160 0.160 0.160 0.160 0.160 0.160 0.160 0.160 0.160 0.160 0.160 0.160 0.160 0.160 0.160 0.160 0.1750
cís. cm. cm. ft. sq. ft.	Cros Sta. Traverse ft.	0.8175 0.86 0.86 0.86 0.11.00 1.20 1.20 1.20 2.15 2.15 2.15 2.15 2.15 2.15	រុំ រូ	Crossec Sta. Traversa 30	0.08 0.09 0.09 0.09 0.09 0.09 0.09 0.09
	lle of Jet Depth ft.	0.619 0.58 0.554 0.573 0.237 0.237 0.237 0.237	8 27.83 cm 27.83 cm 27.83 cm 27.83 cm 27.00 ff 30.00 ff 30.00 ff 30.00 ff 30.00 ff	s of Jet Depth ft.	0,612 0.564 0.487 0.443 0.1443 0.177 0.230
Run No. X1 X2 b/B Model Sta. L/b	Sta. ft.	81. 05 90. 05 90	Run No. '≤ Y1 Ma = b/8 Model Sta.	Profile Sta. ft.	8. % 8. % 8. % 8. % 8. % 8. % 8. % 8. %
	ertion Deptn .215 ft.	0.00 0.10 0.117 0.1187		ssection Depth 30.215 ft.	0.000 0.000 0.182 0.287 0.287 0.287 0.287 0.287 0.287 0.287 0.287 0.287 0.282
t t	Cross Sta. fraverse 30 ft.	0.815 0.985 0.985 0.985 0.980 11.50 11.88 11.88 11.88 11.88 2.15 2.15 2.15 2.15	cfs. cm. cm. ft. sq. ft.	Cro Sta. Traverse ft.	0.85 0.98 0.98 0.98 0.98 0.98 0.98 0.98 0.98
3 41-04 41-04 41-00 61 0-30 61 61 0-88 6 0-88 6 0-88 6 1 1 0-88 6 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1	le of Jet Depth ft.	0,621 0,592 0,592 0,512 0,512 0,125 0,225 0,225	7 30.45 30.45 14.60 14.60 30.00 30.00 30.084 0.884 0.084	.le of Jet Depth ft.	
Run No. 9 9 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1	Sta. Profi ft.	24.18 30.55 31.55 31.55 31.55 31.55 31.55	Run No. Q Yl M2= b/8 Model Sta. L/b	Profi Sta. ft.	
	ion Depth ft.	0.0 0.133 0.		ssection Depth 30.208 ft.	00.0 255.0 264.0 2
ړ. ۲	Crossect Sta. Traverse 30,235 ft.	00008888888888888888888888888888888888	s. . ft.	Cro Sta. Traverse ft.	0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0
<pre>* 2 * 74 cfs. = 55.52 cm. = 55.52 cm. = 0.30 cm. = 0.984 sq. = 0.984 sq.</pre>	ils of Jat Depth ft.		<pre>6 3.340 cf 3.340 cf 3.340 cf 3.375 cm 3.375 cm 1.377 cm 2.030 ft 3.040 ft 0.0884 sq 0.0</pre>	le of Jet Depth ft.	0.612 576 0.543 0.543 0.798 0.798 0.798 0.295 0.295
Run No. 2 7 7 7 7 8 8 8 8 8 8 8 8 8 8 1 9 8 1 1 0 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1	Sta. Prof Sta. ft.		Run No. 37, 32, 32, 18 46 b 1 Sta. 10 b 1 Sta.	Sta. Profi	80.30 20.50

...

TABLE VIII - RAN DATA - AREA OF VENA CUNTRACTA - BACKWATER AND PROFILE

Crossection Star Traversa 30.18 ft. 0.00 0.138 0.138 0.138 0.431 0.436 0.431 0.472 0.472 0.472 0.472 0.472 0.472 0.472 0.472 0.472 0.554 0.472 0.554 0.472 0.554 0.554 0.554 0.555 0.472 0.555 0.472 0.555 0.472 0.555 0.472 0.555 0.472 0.555 0.472 0.555 0.472 0.555 0.472 0.555 0.472 0.555 0.472 0.5555 0.555 0.555 0.555 0.555 0.5555 0.555 0.555 0.555 0.555 0.555 0.00 0.155 0.155 0.362 0.392 0.529 0.529 0.588 0.588 0.588 0.588 0.588 0.588 0.588 0.588 0.588 0.588 0.588 0.588 0.588 0.588 0.588 Crossection Depth Urossection Sta. Depth Traverse 30.20 ft. ft. 2.095 2.005 г. ед. ft. cfa. cm. ft. 89. ft. Frontie of Jet. 3.a., bepth ft., ft., 1... 3.... 1... 1... 3... 3... 1.... 1... 1... 1... 1... 1... 1... 14 3.33 cfe. 3.28 cm. 0.98 cm. 30.00 ft. 0.884 sq. 0.0 18 3.635 1.53 0.275 0.741 0.741 
 Frontile of Jet

 Sta.
 Depth

 At.
 ft.

 ft.
 ft.

 ft.
 ft.

 st.
 ft.
 Run No. = Q = 7 Y1 = b/B Mcdel Sta. = Ag Run No. 2 Y1 M= b/B Model Sta. 0.00 0.136 0.255 0.513 0.510 0.513 0.5150 0.5150 0.5150 0.5150 0.5150 0.5150 0.5150 0.5150 0.510 Crossection Depth η. Croesection Sta. Depth Traverse 30.23 ft. ft. Sta. De Traverse 30.20 1 ft. 1 17 3.705 cfe 6.0.73 cm 6.0.73 cm 6.0.73 cm 0.771 94 ft. 0.0 ft. eq. ft. 
 Frofile of Jet
 Stat.

 Stat.
 Dapth

 Stat.
 T.

 ft.
 T.

 ft.
 T.

 Stat.
 0.566

 Stat.
 0.472

 Stat.
 0.472

 Stat.
 0.472

 Stat.
 0.431

 Stat.
 0.432

 Stat.
 0.432

 Stat.
 0.432

 Stat.
 0.432

 Stat.
 0.432
 13 3.735 cfs. 36.214 cm. 1.23 cm. 30.00 ft. 0.884 eq. f 
 Frontia
 of Jet

 Sta.
 Depth

 Sta.
 Depth

 ft.
 ft.

 ft.
 ft.

 p0.18
 0.572

 p0.18
 0.572

 p0.40
 0.572

 p0.40
 0.572

 p0.40
 0.572

 p0.50
 0.528

 p1.00
 0.528

 p1.00
 0.528

 p1.00
 0.528

 p1.00
 0.528
 ...... Run No. 4 Yl Y2 Madal Sta. L/b ft. 59. ft. 16 4.180 cfs. 52.24 cm. 0.00 cm. 30.00 ft. 30.00 ft. cfs. cm. ft. sq. ft. Frofile of Jet Sta., Bubth ft. ft. 30.00 0.573 30.20 0.521 30.20 0.521 30.50 0.514 30.50 0.514 31.50 0.514 31.50 0.253 31.50 0.253 Run No. = 12 Run No. = 12.02 Y. = 11.20 Y. = 10.78 Rodel Sta. = 30.00 Nodel Sta. = 30.00 Y. = 10.00 Nodel Sta. = 0.00 N 1.11.11.11 Run No. 2 Y\_ Model Sta. Profile Sta. r. 330,50 330,50 31,50 31,50 31,50 31,50 31,50 31,50 31,50 31,50 31,50 31,50 31,50 31,50 31,50 31,50 30,50 3 
 Gamma Ling
 Channel Ling

 Gamma Ling
 Day 10

 Fr.
 <t ът. гд. ft. 4.1. - 0.00 to - 1.0. to сfв. сп. сп. ele. rt. 0.604 0.556 0.435 0.435 0.435 0.315 0.315 Frontie of Jet Ban, Boph ft. ft. 20,20,20 20,20,20 20,20 20,20 20,20 20,20,20 20,20 20,20, rt. 0.624 0.598 0.571 0.571 0.534 0.534 0.360 0.360 Profile of Jet Depth ц 4.75 49.75 Run No. Q Y1 X2 Y2 Y2 Hodel Sta. Run No. 91 92 Model Șta. 87.888.898.898 87.888.898 87.888.898 87.888.898 87.888 87.897 87.897 87. 1

TABLE 8-1- RAW DATA - AREA OF VENA CONTRACTA - BACKWATER AND FROFILE

Quar No. = 2. Quar No. = 2. Y = 2.7/32 cf.a. Y = 1.660 cm. Model Stat. = 30.00 Model Stat. = 20.00 LD = 0.01 stat. P.	Frofile of Jet Crossection Sta. Depth 25a. Depth ft. ft. ft. ft.	Jule         0.55k         0.66k         0.00k           Jule         0.55k         0.6k         0.00k           Jule         0.55k         0.4k         0.05k           Jule         0.00k         0.4k         0.05k           Jule         0.00k         0.4k         0.05k           Jule         0.00k         0.4k         0.05k           Jule         0.0118         1.10         0.04k           Jule         0.0128         1.10         0.04k           Jule         1.10         0.054         0.04k           Jule         1.10         0.054         0.04k           Jule         1.10         0.04k         0.04k           Jule         1.10         0.054         0.04k           Jule         1.10         0.04k         0.04k           Jule         1.10         0.04k         0.04k           Jule         1.10         0.04k	$ \begin{array}{rclcrc} \text{Fan No.} & = & 26 \\ \text{Q} & = & 2.140 & \text{cra.} \\ \text{Y} & \text{P} & \text{P} & 0.35 & \text{cm}. \\ \text{Y} & \text{P} & \text{P} & 0.373 & \text{cm}. \\ \text{Model Size, } & = & 0.273 & \text{cm}. \\ \text{Model Size, } & = & 0.701 & \text{sd}. & \text{fr.} \\ \text{L} & 0.0 & \text{fr.} & \text{sd}. & \text{fr.} \end{array} $	Profile of Jat Crossection Sta. Depth Sta. Depth ft. ft. ft. ft.	2.09 0.01 2.09 2.09 2.09 2.09 2.09 2.09 0.02 2.09 0.02 2.09 0.02 2.09 0.02 0.02
Quan No. = 2. 2. 2. = 2.300 cfr. 2. = 2.300 cfr. 1.380 cm. Madal 354. = 0.201 2.001 394. fr.	Profile of Jet Crossection Sta. Depth Sta. Depth ft. ft. ft. ft.	yy, ii         0.500         0.601           yy, ii         0.516         0.691           yy, ii         0.516         0.691           yy, ii         0.451         0.491           yy, ii         0.446         0.446           yy, ii         0.446         0.446 <td><math display="block"> \begin{array}{llllllllllllllllllllllllllllllllllll</math></td> <td>Profile of Jat Crossection Sta. Dapth Sta. Depth Sta. ft. ft. ft.</td> <td>30.18         0.5%         0.48         0.00           30.20         0.5%         0.48         0.00           30.20         0.05%         0.48         0.00           30.20         0.05%         0.49%         0.179           30.20         0.05%         0.19%         0.179           31.00         0.23%         1.10%         0.179           31.10         0.23%         1.11%         0.128           31.10         0.137         1.12%         0.128           31.20         0.137         1.12%         0.128           31.20         0.137         1.12%         0.128           31.20         0.137         1.12%         0.128           31.4%         0.138         1.12%         0.128           31.5%         0.138         1.13%         0.552           2.00         0.148         0.158         0.158           2.10%         0.056         0.128         0.158           2.10%         0.128         0.128         0.128           2.10%         0.128         0.128         0.128</td>	$ \begin{array}{llllllllllllllllllllllllllllllllllll$	Profile of Jat Crossection Sta. Dapth Sta. Depth Sta. ft. ft. ft.	30.18         0.5%         0.48         0.00           30.20         0.5%         0.48         0.00           30.20         0.05%         0.48         0.00           30.20         0.05%         0.49%         0.179           30.20         0.05%         0.19%         0.179           31.00         0.23%         1.10%         0.179           31.10         0.23%         1.11%         0.128           31.10         0.137         1.12%         0.128           31.20         0.137         1.12%         0.128           31.20         0.137         1.12%         0.128           31.20         0.137         1.12%         0.128           31.4%         0.138         1.12%         0.128           31.5%         0.138         1.13%         0.552           2.00         0.148         0.158         0.158           2.10%         0.056         0.128         0.158           2.10%         0.128         0.128         0.128           2.10%         0.128         0.128         0.128
Run No. 2 20 2 3 200 cfs. 3 3 2.130 cfs. 3 3 2.130 cfs. 2 3 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2	Profile of Jet Crossection Sta. Depth Sta. Depth ft. ft. ft.	0.09 0.09 0.09 0.0282 0.0282 0.0282 0.0282 0.0282 0.0282 0.0282 0.0282 0.0282 0.0282 0.0282 0.0282 0.0282 0.0282 0.0282 0.0288 0.000 0.0288 0.000 0.0288 0.000 0.0288 0.000 0.0288 0.000 0.0288 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.0000 0.0000 0.0000 0.0000 0.0000 0.0000 0.00000 0.00000 0.00000 0.000000	$ \begin{array}{llllllllllllllllllllllllllllllllllll$	Profile of Jet Crossection Sta. Depth Sta. Depth ft. ft. ft. ft.	X0.18         0.353         0.607         0.009         0.139           X0.20         0.047         0.047         0.049         0.139           X0.20         0.0470         0.049         0.139           X0.20         0.0470         0.049         0.139           X0.20         0.0470         0.049         0.139           X0.20         0.0461         1.120         0.149           X0.20         0.2461         1.120         0.1493           X0.20         0.1461         1.120         0.1493           X0.20         0.1461         1.149         0.1493           X0.20         0.1461         1.149         0.1493           X0.20         0.1461         1.149         0.1493           X0.20         0.1461         1.149         0.1493           X0.20         1.149         0.1493         0.1493           X0.20         1.149         0.1493         0.1493           X0.20         1.149         0.1493         0.1493           X0.20         2.100         0.1493         0.1493           X0.20         2.149         0.1493         0.1493           X0.20         2.149         0.1493
Run No. = 19 3 = 1.5.4 go ers. 25 = 1.6.5.4 ers. 25 = 1.6.5.4 ers. 25 = 1.6.5.4 ers. 2.2.7.4 so. 7.2. 2000 - 0.0.1 so. 7.4 so	Frofile of jet Crossection Sta. Depth Sta. Depth Traverse 30.18 ft. ft. ft.	X0.18         0.558         0.487         0.000           X0.20         0.558         0.487         0.000           X0.20         0.487         0.487         0.487           X0.60         0.486         1.100         0.478           X0.60         0.486         1.100         0.476           X1.00         0.486         1.100         0.476           X1.00         0.524         1.100         0.555           X1.00         0.524         1.100         0.555           X1.00         0.524         1.100         0.555           X2.00         0.524         1.100         0.555           X2.00         0.524         1.100         0.555           X2.00         0.577         2.09         0.577           X2.00         0.578         2.09         0.578	Pan No. = 23 Pan No. = 2.720 cfs. 21 = 2.720 cfs. 22 = 2.13 cs. 22 = 2.13 cs. Packed Sts. = 0.027 Packed Sts. = 0.01 sq. ft.	Profile of Jet Crosssction Star Depth Traverse 30.15 ft. ft. ft.	0.08 0.09 0.09 0.09 0.09 0.09 0.09 0.09

TABLE **8-1**- RAW DATA - ABEA OF VENA CONTRACTA - BACKWATER AND PROFILE

TABLE 8-4 RAW DATA - AREA OF VENA CUNTRACTA - BACKWATER AND PROFILE

Crossection Star. Star. Theorem Jo. 285 Array 2000 0.942 0.012 1.00 0.012 1.00 0.012 1.00 0.012 1.00 0.042 1.00 0.042 1.00 0.042 1.00 0.042 1.00 0.042 1.00 0.042 1.00 0.042 1.00 0.012 2.00 0.012 2.00 0.012 Crossection Sta. Depth Traverse 30.21 ft. ft. 34 34,10 cfs. 1,700 cm. 0,00 cm. 0,25 ft. 0,614 sq. ft. 0,0 
 Profile of Jet

 Sta.
 Depth

 Sta.
 Depth

 ft.
 ft.

 ft.
 ft.

 pt.30
 0.491

 30.30
 0.491

 31.00
 0.447

 32.40
 0.447

 31.00
 0.447

 31.00
 0.447
 0,872 0,90 1,00 1,20 1,30 1,30 1,40 1,40 1,40 1,40 1,40 1,50 1,50 1,50 2,00 2,00 2,00 2,00 2,00 ft. 89. ft. cfs. cm. cm. Profile of Jet Sta. Dopth ft. ft. ft. ft. 30,30 0.552 30,40 0.4298 30,40 0.4298 31,40 0.4253 31,40 0.4354 31,40 0.4354 31,40 0.4354 30.225 c 3.225 c 3.225 c 3.003 c 0.0 c 1.275 c 0.741 c 0.741 c Run No. V1 V2 M2 b/8 Model Sta. Run No. 2 Y1 Y2 Y2 Model Sta. Crossection Dept. Traverse 30.22 Ppt Ft. 90.22 Ppt ft. 90.22 Ppt 1.20 0.00 0.422 1.20 0.429 1.20 0.429 1.20 0.429 1.20 0.429 1.20 0.429 1.20 0.428 1.20 0.428 1.20 0.428 1.20 0.208 2.201 0.018 0.00 0.160 0.222 0.212 0.484 0.484 0.484 0.484 0.484 0.484 0.484 0.484 0.289 0.289 0.216 0.108 Creasertian Statementian Freeworks 20,185 Freeworks 20,185 Freeworks 20,185 Freeworks 20,185 Freework 20,00 0,070 1,200 1,200 0,275 1,200 0,275 1,200 0,275 2,073 0,000 2,073 2,073 0,000 2,073 0,000 2,073 0,000 0,073 1,200 1,200 = 33.485 cfs. = 50.485 cfs. = 0.000 cm. = 0.014 sq. ft. = 0.01 cfs. cm. ft. sq. ft. 
 Profile of Jat

 Sca.
 Depth

 St.
 ft.

 ft.
 ft.

 ft.
 ft.

 30.38
 0.498

 30.30
 0.443

 30.40
 0.443

 30.40
 0.443

 31.50
 0.443

 32.60
 0.443

 32.60
 0.443

 32.60
 0.443

 32.60
 0.443
 Run Ho. = 29Y = 2.963 ct Y = 2.9203 ct 2.200 cs 9.200 cs M = b/B = 0.200 ft Model bta. = 0.200 ft L/b = 0.741 sc Profile of Jet S.a. Depth ft. ft. 30.18 0.555 30.40 0.455 30.400 0.487 30.400 0.487 31.00 0.487 31.00 0.487 31.00 0.487 31.00 0.487 31.00 0.487 ft. 0.555 0.486 0.437 0.430 0.329 0.329 Run No. Y Y Madel 3ta. 
 Channell and Channell and Frawres 30, 192 th Frawres 30, 192 th Frawres 30, 192 th Channell and 100 th 32 3.690 cfs. 3.690 cfs. 0.0 cm. 0.25 cm. 0.25 ft. 0.614 sq. ft. ft. 3q. ft. 
 Profile of Jet

 Star
 berth

 Star
 berth

 ft
 ft

 Dubbe
 0.250

 Dubb
 0.250

 Dubb
 0.461

 Dubb
 0.463

 Dubb
 0.463
 cfa. cm. cm. 28 29,28 29,28 0,88 30,00 20,741 0,00 
 Frafile of Jet

 Sta.
 Depth

 ft.
 ft.

 ft.
 ft.

 90.18
 0.552

 90.18
 0.552

 90.10
 0.471

 90.40
 0.471

 91.50
 0.471

 91.50
 0.471

 91.50
 0.473

 91.50
 0.473

 91.50
 0.473

 91.50
 0.473

 91.50
 0.473

 91.50
 0.473
 Run No. V Y2 Mc b/8 Nodel Sta. 19b 0.00 0.227 0.227 0.398 0.538 0.524 0.525 0.525 0.525 0.525 0.442 0.442 0.250 0.250 0.250 Crossection Sta. Depth Traverse 30.27 ft. ft. ft. 39. ft. 2.535 efe. 2.535 efe. 2.1.02 cm. 1.08 cm. 0.275 cm. 30.00 ft. сfз. ст. ст. 31 3.56 e 41.73 e 1.35 e 0.275 30.00 f 0.741 s Profila of Jet Sta. Depth ft. ft. ft. 20.30 0.522 20.30 0.528 20.40 0.528 20.40 0.4378 20.40 0.4378 31.50 0.4373 31.50 0.2381 21.50 0.2381 20.50 0.2381 
 Frofile of Jet

 Star
 Bepth

 Star
 ft

 ft
 ft

 ft
 ft

 ft
 ft

 g0.30
 0.500

 90.500
 0.500

 90.500
 0.500

 91.000
 0.231

 91.000
 0.232

 91.000
 0.233

 91.000
 0.233

 91.000
 0.233
 Run No. 27 37 M2 = b/B M2 = b/B Ag Run No. 2 Y1 K = b/B Model Sta. 1/b

0.00 0.127 0.127 0.127 0.127 0.127 0.127 0.127 0.127 0.127 0.123 0.123 0.123 0.123 0.123 0.121 0.120 0.

	sction Depth 30.22 ft.	0.00 0.116 0.1250 0.2550 0.2550 0.2550 0.250 0.2510 0.2510 0.2510 0.2510 0.2510 0.2510 0.2510 0.2510 0.00 0.00 0.00 0.00 0.00 0.00 0.00 0		ection Depth 30.188 ft.	0.00 1457 0.2445 0.2445 0.2445 0.2495 0.2495 0.2495 0.2495 0.2495 0.2495 0.2495 0.2400
ու ա. ա. ե.	Cross Sta. Traverse ft.	0.925 0.05 1.00 1.10 1.10 1.10 1.10 1.10 2.032 2.032 2.032	임 바바 나 나	Crosse Sta. Traverse ft.	0.922 1.00 1.10 1.10 1.10 1.10 1.80 1.80 1.80
35.22 5 35.22 5 35.22 5 0.93 5 30.00 1 0.614 3	e of Jet Depth ft.	0.506 0.454 0.454 0.436 0.436 0.229 0.229	42 42 42 42 42 42 42 42 42 42 42 42 42 4	ie of Jet Depth ft.	0.504 0.458 0.458 0.335 0.337 0.148 0.128 0.128
Run No. 71 Y2 Mc = b/8 Model Sta. L/b	Profil Sta. ft.	87.98 87.99 87.90 87.00 87.90	Run Ko. 2 7 M = b/B Model Sta. L/S	Profi Sta. ft.	81.05 80.05
	sction Depth 30.23 ft.	0.00 0.106 0.306 0.306 0.400 0.400 0.400 0.400 0.400 0.400 0.400 0.400 0.234 0.400 0.1234 0.00 0.00		section Depth 30.181 ft.	0.167 0.1670
Бз. с. 1. Г.	Crosse Sta. Traverse j ft.	0.935 1.00 1.10 1.150 1.150 1.150 2.00 2.00 2.00 2.00 2.00 2.00 2.00 2.		Cros Sta. Traverse ft.	0.95 0.95 1.00 1.50 1.50 1.50 2.03 2.03 2.03
37 2.920 c 0.00 c 0.25 30.00 f 0.614 c	f Jet Depth ft.	0.510 0.484 0.405 0.425 0.425 0.425 0.425 0.425 0.425 0.425 0.482 0.482	41 2,100 cf 2,5,83 cn 2,59 cn 2,59 cn 30,00 ft 1,614 sc	of Jet Depth ft.	0.506 0.462 0.403 0.368 0.368 0.265 0.265 0.2136
Run No. X Y Ma = b/B Model Sta. L/b	Profile o Sta. ft.	81.25 51.25 51.55 51.55 51.55 52.55	Run ho. * * * * * * * * * * * * * * * * * * *	Profile Sta. ft.	80.18 20.45
	ection Depth 30.275 ft.	0,100 0,110 0,1110 0,112 0,112 0,112 0,122 0,122 0,122 0,122 0,122 0,122 0,122 0,122 0,122 0,100 0,100 0,100 0,100 0,100 0,100 0,100 0,100 0,100 0,100 0,100 0,110 0,0112 0,012 0,010 0,012 0,010000000000		sction Jepth 30.20 ft.	00.0 765.0 865.0 864.0 864.0 864.0 864.0 864.0 864.0 864.0 864.0 864.0 864.0 864.0 866.0 866.0 866.0 860.0 800.0 8
fa. t. q. ft.	Cross Sta. Traverse ft.	0.938 0.938 1.50 1.50 2.00 2.00 2.00 2.00	į.	Cross Sta. Fraverse ft.	0.926 0.95 1.00 1.20 1.20 1.40 1.40 1.40 1.40 1.40 2.02 2.028
36 36.020 0.02 0.02 0.01 30.00 1 30 1 3	s of Jet Depth ft,	0.512 0.485 0.485 0.443 0.4432 0.4532 0.258 0.258 0.194	40 2,385 eff 29,94 cm 1.95 cm 1.95 cm 30.05 30.05 0,614 sq 0,00	of Jet Depth ft.	0.604 0.469 0.412 0.313 0.313 0.291 0.291 0.291
Run No. 91 Y1 M = b/B Model Sta.	Profil Sta. ft.	30, 10 30, 50 30, 50 31, 50 31	Run No. × y1 y2 Nodel Sta.	Profile Sta. ft.	30, 28 30, 58 31, 58 31, 58 21, 58 21
	ection Deptn 30.24 ft.	0.00 0.1331 0.1331 0.1331 0.1430 0.1430 0.1430 0.1430 0.1430 0.1332 0.1332 0.000		section Jo.21 ft.	0.00 0.223 0.223 0.254 0.254 0.496 0.4496 0.4496 0.4496 0.4496 0.249 0.249 0.249 0.249 0.249 0.249 0.249 0.249 0.249
cfs. cm. cm. ft. sq. ft.	Cross ita. Traverse ft.	0.93 0.95 0.95 0.95 0.95 0.95 0.95 0.95 0.95	د: 15. 15.	Cros Sta. Traverse	0.935 0.935 0.935 0.935 0.935 0.935 0.935 0.935 0.9550 0.9550 0.9550 0.9550 0.9550 0.9550 0.9550 0.9550 0.95
35 3.098 0.00 0.25 0.614 0.01	of Jet Depth ft.	0,513 0,469 0,461 0,451 0,437 0,437 0,437 0,437 0,203	39 22.55 22.55 20.05 30.05 30.05 50 50 50 50 50 50 50 50 50 50 50 50 5	of Jet Depth ft.	0.505 0.472 0.441 0.404 0.303 0.303 0.303 0.161
Run No.	Profile Sta. ft.	30.18 30.16 30.16 30.16 31.00 20.05	°un No. ∑ 1 = b/B ⇒ ab/B sta. [/b	Profile Sta. ft.	80.18 30.50 30.50 31.60 31.55 21.55

TABLE 8-1- RAW DATA - AREA UF VENA CUNTRACTA - BACKWATER AND FRUFILE

Crossmett, on Crossmett, on Treverse 20, 1991 1.067 0.00 1.109 0.234 1.100 0.234 1.100 0.234 1.100 0.234 1.000 0.234 1.000 0.234 1.000 0.234 1.000 0.000 Crossection Sta. Depth Traverse 30.19 ft. ft. 0.00 0.2267 0.3256 0.3556 0.433 0.4730 0.4730 0.4730 0.4730 0.4730 0.4730 0.4730 0.4 Run No. = 51 Y = 52.52 efs. Y = 59.95 cm. K= b/8 = 0.20 cm. Model Sta. = 0.00 ft. L/b = 0.33 ss. ft. 0.915 0.950 0.950 0.950 0.911.05 0.11.95 0.11.95 0.11.95 0.11.95 0.11.95 0.11.95 0.11.95 0.05 0.915 0. cfs. cm. ft. ft. ft. Profile of Jet Sta. Depth ft. 0.421 0.407 0.377 0.377 0.377 0.377 0.326 0.326 0.326 10,000 Profile of Jet Sta. Depth ft. ft. 23.25.06.05.18 23.25.06.05.18 23.25.06.05.18 Run No. y y2 M = b/8 Model Sta. A96 Crassetion Sc. Sec. 2010 Transverse Jula Burth Transverse Jula 0.993 0.109 1.009 0.239 1.009 0.239 1.009 0.2499 1.100 0.2499 1.100 0.2499 1.100 0.2499 1.190 0.2499 1.990 0.2499 1.990 0.2499 2.049 0.018 Crossection Sta. Depth Treverse 30.23 ft. ft. Run No. = 49 Y = 21.53 cfa. Y = 21.53 cfa. Y = 0.55 cfa. Nodel Sta. = 20.00 ft. 10 = 0.014 eq. ft. 45 3.70 cfs. 55.50 cm. 0.25 cm. 0.25 cm. 30.00 ft. 0.214 sq. ft. 0.014 2.095 Profile of Jet Sta. Deith ť Profile of Jet Sta. Depth ft. ft. Run No. Q Y1 M<sup>2</sup> b/3 M<sup>2</sup> b/3 M<sup>2</sup> b/3 L/b Ŀ. Crossection Sta. Depth Traverse 30.185 ft. ft. Grossection Surveys 20.175 Fr. 20.175 Fr. 20.175 Fr. 20.175 Fr. 20.175 Fr. 20.253 Fr. 20 - 48 2.112 cfa. 2.5140 cm. 2.33 cm. 0.250 ft. 90.00 ft. 0.014 sq. ft. 2.00 44 1.742 cfs. 1.742 cfs. 2.048 cm. 2.040 cm. 30.02 ft. 30.04 sq. ft. Profile of Jet Sta. Depth ft. ft. Profile of Jet Sta. Depth ft. 0.510 0.398 0.314 0.314 0.213 0.110 Run No. VI Nodel 34e. Run No. Crossection Sta. Depth Traverse 30.205 ft. ft. 2,005 Profile of Jet Sta. Depth ż 0.507 0.454 0.378 0.341 0.341 0.236 0.162 Profile of Jet Sta. Depth Ŀ. 822282828 825282828 85588888 ť

CABLE B - I - RAW DATA - AREA OF VERA CONTRACTA - BACKWATER AND PROFILE

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	ssection De 30.25 ft	22000000000000000000000000000000000000	. u	Grossect sta. averse 30 it.	1.059 1.20 1.20 1.20 1.70 1.70 1.70 1.70 1.70 1.90
cfs. cm. cm. sq. ft.	Crc Sta. Traver ft.		9 1.75 efs. 2.55 em. 0.20 ft. 0.393 sq. 0.0	f Jet Depth S ft. Tr	0.398 0.375 0.354 0.354 0.319 0.158 0.120
55 2.12 30.00 30.00 0.333 0.333	of Jet Depth ft.	0.159 0.382 0.388 0.388 0.359 0.2500 0.2500 0.159	, еко е л пи питен ~ с с с	rofile of	18 50 50 50 50 50
Run No. 2 7 M2 = b/B Model Sta. L9b	Profils Sta. ft.	8.888.88 8.998.98 8.988.98 8.988.98 8.988.98	Run N VL MAZ L∕b L⁄b L/b	Str. 3	ୡଢ଼ଢ଼ଢ଼ୡୖ୷୷୴
	th	0 152 152 152 152 152 152 152 152 152 152		stion Depth 30.228 ft.	0.00 0.164 0.378 0.378 0.376 0.376 0.376 0.108 0.00 0.108
	isection Dep is 30.26 ft.	000000000000000000000000000000000000000	ມ ຊູ	Crossec Sta. Traverse 2	1.063 1.120 1.20 1.60 1.60 1.92 1.927
cfs. cm. ft. sq. ft.	Cros Sta. Travers ft.	1.065 1.665 1.660 1.980 1.980 1.980	1,800 cfs 34,75 cm 34,75 cm 0,23 cm 0,293 sq. 0,0 ft	r Jet spth	102 364 364 326 326 326 127 127
54 54 54 54 54 54 54 54 54 54	e of Jet Depth ft.	0.114 0.396 0.378 0.366 0.233 0.233 0.221 0.160	۰۰. ۱3ta, ۱۱۳	Profile o ta. D	0,18 0,18 0,100 0,100 0,100 0,100 0,11 1,000 0,000 0,100 0,100 0,000000
Run No. Q Y1 Y2 Model Sta. L/b	Profi Sta. ft.	30,18 30,50 30,60 31,50 31,50 31,50	Hode Hode Hode Hode Hode	ю ч	
	pth.	00 354 354 354 255 00 255 00 255 00 255 00 255 00 255 00 255 00 255 00 255 00 255 00 255 00 255 00 00 00 00 00 00 00 00 00 00 00 00 0		ion Depth 0.235 ft.	0.00 0.161 0.352 0.378 0.378 0.336 0.256 0.103
	section De se 30.282 ft		ſt.	Grossect Sta. Traverse 3 ft.	1,900 1,900 1,900 1,900 1,900 1,900 1,922
cfs. cm. ft. sq. ft.	Cros Sta. Traver ft.	2007 1 - 2007 1 - 2007 1 - 2007 1 - 2007 1 - 2007 1 - 2007 2 - 200	937 cfa. 45 cm. 33 cm. 20 ft. 393 sq.	н 	
= 53 2,290 50,68 = 30,00 = 335 0,03 = 0,00	e of Jet Depth ft.	0, 377 0, 389 0, 360 0, 272 0, 272		le of Jet Depti ft	00000000000000000000000000000000000000
Run No. 2 Y1 Ma = b/8 Model Sta. L9b	Profil Sta. ft.	30,18 30,200 31,500 31,500 31,500 31,500 32,000	Run. No. V1 Nd b/B Nodel Sta.	Profi Sta. ft.	30,18 30,40 30,50 31,50 31,50 31,50 31,50 31,50 31,50 31,50 31,50 31,50 31,50 31,50 31,50 31,50 31,50 31,50 31,50 31,50 31,50 31,50 30,500
	. 2 th	000 358 358 337 337 337 337 234 8772 234 8772 00 772		n Depth 221 ft.	0,00 0,164 0,352 0,3396 0,3396 0,3396 0,3396 0,339 0,339 0,00 0,00
	ossection . De erse 30.3	800000000		rossectio a. verse 30.	20000000000000000000000000000000000000
cfa. cm. ft. sq. ft.	Cr Sta Itav		cris. cm. ft. sq. ft.	1 S C C	
52 442 56.03 56.03 30.00 30.00 30.00	le of Jet Depth ft.	0.420 0.376 0.376 0.376 0.355 0.255 0.185	56 10.44 10.44 30.00 0.393	e of Jet Depth ft.	0.107 0.356 0.356 0.356 0.357 0.347 0.271 0.108
Ran No. 32 Md= b/8 Md= 1 Sta. L/b	Frofil Sta. ft.	30,50 31,00 31,00 32,00 33,00 33,00 32,00 33,00 33,00 33,00 33,00 33,00 33,00 33,00 33,00 33,00 33,00 33,00 34,00 35,00 30,000 30,0000 30,0000 30,00000000	Hun No. V1 M <sup>2</sup> = b/B Mdel Sta.	Profil Sta. ft.	30.18 30.400 30.560 31.000 32.000 32.000

		n Depth 165 ft.	0.00 0.178 0.353 0.385 0.385 0.381 0.381 0.381 0.381 0.381 0.381		n Depth ft.	0.00 0.184 0.348 0.338 0.338 0.338 0.338 0.338 0.270 0.145 0.00
	ې پړ •••••	Crossectic Sta. Traverse M	1.03 1.10 1.20 1.30 1.50 1.50 1.90 1.92 1.92	. <i>i</i>	Crossectic Sta. Traverse 30.1	1.051 1.10 1.10 1.10 1.10 1.10 88 1.10 88 1.10 88 1.10 88 1.10 88 1.10 1.00 1.10 1.00 1.0
	63 15,145 15,145 15,146 15,146 15,146 15,145 15,155	le of Jet Depth ft.	0,391 0,299 0,248 0,125 0,125 0,092	+ 67 = 1.08 cfa = 1.93 cm, = 1.98 cm, = 1.98 cm, = 0.20 cm, = 0.393 m,	s of Jet Depth ft.	0.392 0.392 0.294 0.258 0.258 0.258 0.258 0.097 0.000
	Run No. 2 Notol Sto L/b	Profi Sta. ft.	8.838.88 8.848.88 8.848.84	Run No. 2 M = M = L9 L9	Profil Sta. ft.	8.08 8.09 8.09 8.09 8.09 8.09 8.09 8.09
		n Depth ft.	0.00 0.176 0.2355 0.3355 0.3355 0.3355 0.3362 0.3362 0.268 0.268		cion Depth 0.155 ft.	0,000 0,185 0,324 0,324 0,328 0,328 0,368 0,368 0,368 0,141 0,141
	ž.	Crossactic Sta. Traverse 30.1	1.05 1.10 1.40 1.40 1.40 1.40 1.40 1.40 1.40		Crossec Sta. Traverso 30	1.046 1.100 1.100 1.100 1.100 1.940
	62 1.530 cfs 2.1.530 cm 2.1.1.3 cm 1.1.1.3 cm 30.00 ft. 0.393 sq.	t of Jet Depth ft.	0,390 0,362 0,317 0,235 0,235 0,235 0,135 0,103	66 11,20 11,20 11,73 11,	.le of Jet Depth ft.	0.392 0.307 0.274 0.213 0.213 0.106 0.106
	Run No. 71 M2= b/8 Model 3ta.	Profilie Sta. ft.	2.18 20.58 21.66 2	Run Ko. 7 7 Mc b/8 Model Sta.	Profi Sta. ft.	8.98.98.99 9.99 9.99 9.99 9.99 9.99 9.9
		pth	00 2173 331 331 339 339 339 234 224 224 224 200 00		n Depth ft.	0,00 181 0,181 0,288 0,288 0,389 0,389 0,389 0,389 0,389 0,389 0,00 0,00 0,00 0,00 0,00 0,00 0,00 0,
	,ĩ	Crossection Sta. De averse 30.185 ft. ft	00000000000000000000000000000000000000	ft.	Crossectic Sta. Traverse 30.1	1,002 1,0020
IIE	61 1.570 cfs. 28.28 cm. 1.38 cm. 30.00 ft. 0.393 sq. f	of Jet Depth ft. Tr	0.389 0.352 0.222 0.200 0.210 0.210 0.102 0.102	65 1.27 cfe 2.1.71 cm 1.77 cm 30.00 ft. 0.03 sq.	of Jet Depth ft.	0,390 0,345 0,2814 0,2814 0,285 0,285 0,255 0,0172 0,0172 0,013
KWATER AND PROF	Run No. 71 Y2 Model Sta. L/b	Profile Sta. Sta. ft.	32.65 32.65 33.65 33.66 33.1.66 33.1.66 33.1.66 33.1.66 33.1.66 33.1.66 33.1.66 33.1.66 33.1.66 33.1.66 33.1.66 33.1.66 34.1.66 35.1.6	Run No. 71 Ma b/8 Model Sta. 196	Pmfile Sta. ft.	80.80 80.80 80.80 80.80 80.80 81.80
CONTRACTA - BAC		pth	00 237 352 352 337 337 337 2266 2366 2366 2366 2466 2466 246 246 246 246 246 246 246		pth.	000 3,821 3,821 3,924 3,956 3,956 1,3355 1,335 1,335 1,335 1,335 1,3355 1,3355 1,3355 1,3355 1,3355 1,
EA OF VENA C		Crossection a. De ersa 30.190	502003000000000000000000000000000000000		Crossection ta. De verse 30.165 t. ft	9.9.9.9.9.9.9.9.9.9.9.9.9.9.9.9.9.9.9.
RAW DATA - AR	0 1.62 cfs. 3.65 cm. 1.35 cm. 1.20 ft. 1.393 sq. ft.	l Jet Depth St ft. ft	0.334 0.334 0.334 0.3217 0.112 0.112 0.112 0.112 0.112 0.112 0.112 0.112 0.112 0.112 0.112 0.112 0.112 0.112 0.120		Jet bepth S ft. Tra	0.388 0.352 0.2320 0.22200 0.22200 0.22200 0.22200 0.22200 0.22200000000
TABLE 8-1-	Run Ko. Run Ko. Ky Mean by R. Model Sta.	Profile of Sta. ft.	30.30 30.50 31.60 31.60 31.50 31.50 31.50 31.50	Run No. = 64 Y1 = 22 M=b/B = 1 M=b/B = 2 Model Sta. = 36 L/9	pmfile of Sta. I ft.	8.8.8 8.9.8 8.9.8 8.9.8 8.9 8 8.9 8 8 8 8

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ARLE 8-2 RAW DATA - SUBARAGED TESTS

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 8-2- RAW DATA - SUBVERCEO TABLE

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TABLE 8-2- RAW DATA - SUBMERCIED TESTS

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TABLE G-2 RAW DATA - SUBJERGED TESTS

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- 275	14 -	- 0.023	- 2.0 c.	- 0.50	- 0.20	- 0,000	- 0,20	- 0.50	- 68° F	- 30.0	- 16.60	- 12.60	- 2.0	- 0.0	Denth	Cent	ġ.	16.39	10.40	16.44	16.45	16.50	10.60		13.52
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- 274	IV -	- 0.0	- 2.0	- 0.5	- 0.2	- 0.0	- 0.1	- 0.5	1 680		- 24	- 12	2	•	2	18	5	24			24	22	07**	0	11
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73		.0238	.0 cfs.	.80 ft.	.15	.00028	.25	.50	100 E	0.0	5.14 cn	2.0 ct	2.0	0*0	t to a	ent. R	п.	5.14	=	=	=	=	=	22	2.9
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Run No.	Geometr	g	Y	Υ'n		s.	<b>M</b> =b/B	L/b	Temp.	Nodel 2	Y,	<u>ب</u>	2	62	a+5		ż	10	15	20	25	27	29.3	30.9	11
		20	.f3.	Ŀ.		0285					o cm.	сп.				-	E.	0						21.8	
- 272	IA -	- 0.02	- 3.0	- 0.80	- 0.15	- 0,00	- 0.20	- 0.50	- 700	- 30.0	- 29.5	- 21.0	- 2.0	- 0.0	Pert	Cent	e U	29.5	=	±.	=	=	" 20	`	22.7
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1	y - V	1	ĩ	•	•	•	•	•		ita 3	- 1	1	•	•	6		сш°	4					42.27	20.0	4
Run No.	Geometr	5	3	Y,			8/0-1	L/b	Temp.	Model 2	۲ı	5	N	92	St.a.		£t.	10	15	20	25	27	29.55	30.3	11
																•								<i>•</i>	
0		2238	5 cfs.	706 ft.	15	00295	25	2	<b>6</b> .	0	-50 CB.	-2 Cm.	0	0	th.	с. В	đ.	80	1	9	5	_	_	20.	ę.
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209	17	· 0.0238	- 2.5 cf	· 0.706	. 0.15	- 0.0002	- 0.20	- 0*20	- 70° F	0.0	. 23.75	. 18.95	. 2.0	0.0	Denth	Cont.	св.	23.75	=	=	=	=	=	-	20.0
40.	stry -					·				L Sta.				•			ca.						23.75	18.95	
Run	Geom	¢	σ	Y,		s	/q=H	L P	Temp.	Mode	τ,	Y.	z	б Ф	Star		z	10	15	8	25	27	29.5	ŝ	1
		238	cfs.	06 ft.	5	00295	5	0	٥.,	0	8	0	0	0	oth	nt. R	10	-50	=				*	17.5	
- 268	IN -	- 0.0	- 2.5	- 0.7	- 0.1	- 0.0	- 0.1	- 0.5	2.	30.	- 34.	P.	- 2	•	đ	e S	e e	75					8	•	2
un No.	sometry						=b/8	٩	emp.	odel Sta		104		2			t.	0	\$		\$	-	9.55 24	0.3 16	4 15
- Æ	ර	5	o	×	-	S	Σ	нĩ	e	2	÷	34	25	a	0		54	Ä	÷,	2	N	N	Ń	б.	3

o 286 Run No stry - VII Geometry - - 2.0 cfs - - 2.0 cfs -	0.44 0.05 0.005 M=b/ 0.05 1/b/ 0.05 1/b/b	- 1.0, 22 - 1.0, 22 - 0.0 02 - 0.5 02 - 0.5 03 - 0.5 15 -	26,06 10 26,16 15 26,17 28 26,19 25 26,19 25 26,19 25 27,2 27,08 10,10 20,10
285 Run N .ry - VII Geome - 0.0238 n - 2.0 cfo. 0	- 0.5 - 0.3006 - 0.3006 - 0.35 - 0.35 - 0.0 - 0.0 - 0.0 - 10 - 10 - 0.0 - 10 - 0.0 - 10 - 0.0 - 10 - 0.0	$ \begin{array}{cccccccccccccccccccccccccccccccccccc$	25,10 10 25,16 15 26,58 20 26,58 20 26,58 27 26,55 27 26,55 27 7,73 30.3
7 - VI - 0.0238 Run No - 0.0238 n - 3.0 cfs. 9 - 3.0 cfs. 9	-0.20 -0.200525 F -0.200525 S -0.25 L/B -0.50 F -0.50 -20.60 cm -21.50 cm -21.50 cm -21.50 cm	10:0 N 0:0 N - 0:0 A -	21.0 10 21.1 10 21.1 20 21.4 20 21.5 21.5 20.5 8.0 18.41 30.3
- 281 Run No. - 11 Geometry - 0.0238 n - 3.0 cfe. 9	- 0.20 - 0.200525 F - 0.200 - 0.20 - 0.50 - 1.4b - 2.574 cm. X - 25.74 cm. X	- 2.0 can M2 - 2.0 can M2 - 0.0 can M2 Depth Sta. Cont. R th. can. cm. ft.	25.74 10 15 15 15 15 15 14 17.8 20.3 2 24.3 2 2.4 17.8 20.3 1
7 - VI Geometry - VI Geometry - 3.0 cfs. 9	- 0.00 22 - 0.000525 8 - 0.15 8 - 0.15 8 - 0.15 8 - 0.15 8 - 0.12 1 - 0.10 m. 1 - 30.10 m. 1	- 15.70 cm. 12 - 12.0 km - 0.0 km L Depth Sta. L Cent. R Sta. cm. cm. ft.	38.10 10 15 25 38.10 15.08 30.3 1 16.70 30.3 1
tko 279 Rún ko. astry - VI Geometri - 2.5 cfe. 9	() () () () () () () () () ()	- 2.0 M2 - 0.0 02 - 0.0 02 - Depth Sta. can. ca. ft	NGT 10 SUBSERGED 25 INLET 25 30.3
No 278 Run setry - VI Geo - 2.5 cfs. 9	= 0.200 ft	- 15.30 cm. 17 - 2.0 M. - 2.0 22 - 2.0	20.60 20.60 15.80 15.83
- 277 Run - VI Geo⊞ - 0,0238 n	- 0.205 II. In - 0.205 F - 0.205 S - 0.205 Hub/ - 0.15 Hub/ - 0.15 Wub/ - 0.20 Mode a 30.56 on II	- 15.6 cm. 17 - 2.0 K - 0.0 92 - 0.0 92 - Depth 8ta. L Cont. R cm. cm. ft.	30,56 10 15 12 14 14 14 14 14 14 14 14 14 14 14 14 14

TABLE 8-2- RAW DATA - SUBHERGED TESTS

																	24	e B							10.48	
96	Ħ	0,0238	8.85 cfi	.49 ft	4.0	67000*0	.35	•	990 E	0.0	<b>33.43</b> cl	Lu. 76 ci	0.1	0.0	0.0	Apth	Cent.	ġ	32.20	18.2	33.09	33-41	33.49	33.43		
•		'	1	1	'	•	-	•	ï	ta	1	7	•	1	•	-	Ц	e e							4.76	
on No.	sometr			. 8		:.	=p/B	ę	emp.	odel S		10		a	2	ta.		ي.	0	ŝ	0	2	7.5	6.6	1.0	
æ	9	ء ~	cfs. 9	r t		2	x	-	μ	24	cia.	đ	~	0	•		<i>c</i> 4	U							9*65	
- 295	IIA -	- 0.023	- 3.85	1.0.0	- 0.50	.00,000	- 0.35	0.0 -	- 0% F	- 30.0	- 32.71	- 9.BL	- 1.0	0.0	0.0	napth	Cent.	сн.	31.24	31.69	ы. Ж	3	32.76	32.71		
:	Ę	Ì		·		•				Sta.							ы	đ							9.84	
Run N	Geome	5	ð	ř		·.,	H-b/E	٩ ۲	Temp.	Model	ř	Y <sup>2</sup>	¥	õ	¢	Sta.		ż	10	15	20	25	27.5	29.9	30.3	
		_		ډ.		9					CB.	<b>.</b>					æ	đ	_			_			26.34	
294	11A	0.0238	2.0 cf	0.80 f	1.0	0,0000	0.35	0.0	4\o99	- 30.0	34.84	27.10	0.1	0.0	°.	Depth	Cent.	ë	34.60	34.86	34.8	34.90	34.85	34.81		
•	- E	'	ľ	•	'	1	'	'	'	Sta.	'	'	'	'	'	,	ń	ca.							27.10	
Bun N	Geome	6	ð	Ϋ́	4	s	B/d=M	۹ \1	Temp.	Hodel	۲	ŝ	2	õ	C	Sta.		Ŀ.	10	15	20	25	27.5	29.9	30.3	
				تې		6					elle,	en e					~								17.29	
293	ΝĪΙ	0.0238	2.0 cf	0.50 1	0.20	0,0001	0.35	0.0	66° F	30"0	25.67	16.72	1.0	0.0	0.3	Depth	Cent.	ca.	25.70	25.92	26.03	26.03	26.05	25.67		
•	-	.'	'	'	'	'	'	'	'	Sta	'	'	'	'	'			an C							16.60	
Bun No	Geomet	c	ð	Yn		۰°	H=b/B	<u>م</u>	Temp.	Model	Y,	£۲	z	õ,	Ľ	Sta.		ż	10	15	20	25	27.5	29.9	30.3	
		38	cfs.	ft.		1600			ρ.,	~	70 CH.	LO Cm.	_	~	~	5	н.	en.	0	a a	5	52	75	2		
- 292	TIA -	- 0.0	- 2.0	- 0.3	- 0.3	0.0	- 0.3	0.0 -	- 660	- 30.0	- 22	- 12.	-		0	Dept	Cent	ġ	23.	23.1	23.	23.6	23.	ສ	o,	
No.	stry						8			al Sta.							ы	E.					5	~	1.21 6	
Bun	Geo	5	đ	Y,		S	(q.	< T	Ten	Mode	۲	r,	S	õ	8	Sta.		ż	q	15	20	25	27.	29.	8	
_	_	26.36	5 cfe.	37 R.		2005	35	~	p.,	0	.03 cm.	.97 cm.	0	0		th th	с. В	e.	- 82	ħ	2	5	ອ	6	<b>9</b> •6	
- 291	(IA -	- 0.	- 2	0.	- 0	- 0.	- 0	-0	38	- 3	- 28	1	-	•	•	Dep	Cen	ca.	27.	27.	27.	28.(	28.	28.	2	
No.	Dotry.	•					8/9	<u>م</u> .	.0	iel Sta				~			1	5					ŝ	•	· 3 9.5	
TT A	ž	4		Y.	<b>.</b>	- Solution	Ξ.	2	Ę.	Ma	۰. ۲,	£۲ •	'z	Ô	2	St		2	10	15	20	25	27	29	8	
6		02.38	75 cfe	.34 ft.	5	000	35	0	4 o 1	0.0	0.67 cm	9.70 cm	0-1	0	°.3	pth	nt. R	5	10	53	æ.	34	8	•67		
4	-	1	1	-	1	•	-	•	-	a 3	ī	1	1	1	1	å	8	5	29	67	29	8	8	8	2	
un No.	eometry					-	= p/8	٩		odel St	_	••		2	6	ta.	ч	ۍ د	0	\$	0	\$	7.5	6.6	e.o	
8	3		0		-	- S	×	1	· 64	R	۳. ۲	4	20	U		ó		с. а	~	-	â	N	~	ŝ	8.54 J	
69		02.38	0 cfa	50 11	9	00025	.35	0	4 o9	0.0	1.56 ci	8.25 cr	0.1	0.0	5.0	epth	ent. R	8 1	77.1	1.78	1.78	1.79	8	1.56	Ä	
•		•	1	;	1	1	1	ì	•	ta 3		1	'	•	•	Ц	с _		~	~	~	7	J	~	8.25	
hun No.	e cane t.m.				1.1	٩.,	<b>L</b> b/B	4	ED.	lodel S	5			n A	ح	ta.		j.	0	ŝ	2	5	7.5	6.6	0°,3 L	
10.				تو		75 5								Ĩ			æ	ch.							16.27	
288	IIA	0.0238	2.0 cf	0.59 1	0.20	0.000	0.35	0.0	4 o99	30.0	29.81	15.27	0.1	0.0	0.5	Depth	Cent.	ŧ	27.78	29.76	29.91	29.92	29.98	29.81		
• •	-	, '	•	'	'	'	'	'	'	Sta	'	•	'	'	•		1	8							15.27	
Burn No	Geomet		. 0		1		H-b/B	2	Temp.	Model	Υ.	4	ž	8	2	Sta.		ż	2	5	20	25	27.5	29.9	30.3	

\* Table **8~ 2-** ray data - subreked tests

а. 66.2 Сп.	в. св. 9
- 304 - VII - VII - 0.0238 - 2.0 cf - 0.0 - 0.0 - 0.0 - 29.95 - 20.95 - 26.95 - 26.95 - 26.95 - 26.95 - 26.95 - 0.5 - 0.	Depth Cent. 29,95 30,05 30,05 30,05 30,05 30,05
sta.	L cm,
Run N Geome P R R R R R R R R R R R R R R R R R R	sta. 11 15 25 25 25 30.3
Can.	а ст. 21.95
7303 10023 10023 10023 10020 10020 10023 1002000 10020 100000000	Depth Cent. cm. 29,09 29,18 29,12 29,17 29,17 29,17 29,17
Sta.	L cm. 21.12
Run N Geo.mei S S M=b/8 N M=b/8 N M=b/8 N M=b/6 1 M=b/6 A 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2	Sta. ft. 10 20.3 27.5 27.5 27.5 27.5
cm.	R ст.
902 1000 1000 1000 1000 1000 1000 1000 1	Depth Cent. cm. 31.15 31.48 31.48 31.48 31.48
Sta 3	L cm.
Run No Geomet Fr Nodel X2 X1 Nodel X2 X2 X2 X2 X2 X2 X2 X2 X2 X2 X2 X2 X2	ft. ft. 10 20 29,9 20,3
. É É	ст.
01 10238 85 cf 85 cf 100048 9.81 c 9.81 c 9.81 c	apth ant. 3.73 3.95 3.95 4.30 4.30 4.30 4.30 4.30 4.54 1.30 4.54 1.30 3.55 1.30 3.55 1.30 3.55 1.30 3.55 1.30 3.55 1.30 3.55 1.30 3.55 1.30 3.55 3.73 3.73 3.73 3.73 3.73 3.73 3.73
N>000000000000000000000000000000000000	400 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0
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50 - 27 - 2 - 2 - 2 - 2 - 2 - 2 - 2 - 2 -	3 4 9598746 v
38 cfs. 7 cm.	с с л с л с л с с л с с л с с л с с л
00100000000000000000000000000000000000	Lepth Cent. cm. 27.7 28.2 28.9 28.9 28.9 28.9 28.9
Sta.	ст. ст.
Run N Geome Fn Mibb/B Nodel	Sta. ft. 10 15 20 25 29,9 30,3
2 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1	в ст. 30.71
- 299 - 219 - 2000 - 20	Depth Cent. 35.97 35.99 35.99 35.99 35.97 35.79
Sta.	ст. 30.04
Run N Geome Fun Tomp. Z Z Z Z Z Z Z Z Z Z Z Z Z Z Z Z Z Z Z	Sta. ft. 10 20 22 20.3 30.3
•	ет. 22.20
298 10,000 10,000 20,000 20,00 20,00 20,00 20,00 1,00 1	Depth Cont. 31.15 31.28 31.28 31.28 31.28 30.33
.e	сm.
Run No Geomet I F R M M A 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2	Sta. ft. 110 115 225 225 225 225 225 225 225 225 225
÷.	6.62
70 135 135 135 135 135 135 135 135	pth nt. 1 600 722 722 722 722
00000000000000000000000000000000000000	సినినినిని కొండి కె
metry metry el Sta	
สี่ง อาระเขารี่รี่ต่องราวราช์ ๙	84 303 303 303 303 303 303 303 303 303 30

		~	cfs.	2		57					св.	CB.						æ	сш.							40.22
307	IIA .	0.023	3.85	1.45	0.10	0.000	0.50	0.0	68° F	30.0	43.19	40.57	0.1	0.0	0.0		Depth	Cent.	cm.	43.49	13.47	43.44	13.12	13.61	43.19	
•	try -	'		'	'		'	'	'	Sta.	'	'	'	'	'			ы	cm.							40.57
Run N	Geome	£	0	Y.	'n.		M=b/8	L/b	Temp.	Model	Y,	12	'z	0 0	~	-	Sta.		ſt.	q	15	20	25	27.5	29.9	30.3
						52					.8	н.						8	св.							34.48
306	VII	0.0238	3.0 cf	1.04 f	0.1	00000*0	0.5	0.0	680 F	30*0	38.85 0	34.84 0	1.0	0.0	o.3		Depth	Cent.	cm.	38,83	36.96	39.02	39.02	38.98	38.85	
•	2	'	'	'	•	1	'	'	'	Sta	1	'	'	1	1			ы	сш.							34.84
Run No	Geomet	u	ð	Yn	5	S	B/d=M	۹/1	Temp.	Model.	Ľ,	L.	z	õ	¢		Sta.		Ŀt.	10	15	20	25	27.5	29.9	30.3
			fs.			•					. ш.	ев.						æ	сш.							25.61
305	IIV	0.0238	3.85 c	0.78 f	0.2	0.0001	0.5	0.0	68° F	30.0	31.43	24.67	1.0	0.0	0 <b>•</b> 3		Depth	Cont.	сш.	32.18	32.22	32,25	32.26	32.18	31.43	
•	try -	•	'	'	'	,	1	'	'	Sta	'	'	•	•	'		,	_	cm.	Е						24.67
Run N.	Сеоте	5	ð	Yn.	Fn.	s	Marb/B	٩ ۲	Temp.	Model	۲,	¥2	Ň	ñ	e e		Ste.		ż	0	52	ຸ	52	27.5	56.9	<b>%</b> .3

Run No 70 Geometry - Ia n - 0.0110 Q / 3.75 cfs.																			
Machine      - </th <th></th>																			
Depth from Bottom Pt.			*J4	0.7	9*0	0.5	7*0	0.3 HOF	11ZUNTAL 0.2	. UISTANG 0.1 AGITY R	CE FROM 0.0 EADINGS	CENTER 0.1 (fps.)	0.2 (F		0 <b>†</b> •0	••5	0		74
0.664 0.464 0.498 0.1045 0.236 0.10062 0.10062			6.24	6.24 5.24	6.02 5.93 5.93	5,550 5,500 5,5000 5,5000 5,5000 5,5000 5,5000 5,5000 5,5000 5,5000 5,5000 5,5000 5,	5.17 5.17 5.16 5.15	4.68 4.88 4.85 4.85 4.85	1, 168 1, 160 1, 160 1, 168 1, 168 1, 168 1, 168 1, 168 1, 168	4-03 4-15 4-15 4-15 4-15 4-15 4-15 4-15 4-15	4-03 4-49 4-55 4-55 55 4-55 55 55 55 55 55 55	4-66 4-66	4 - 54 4 - 54 4 - 70 4 - 72 4 - 72	4-88 4-88 4-91 4-93	5.15 5.15 5.16 5.16 5.10 5.10	5) 5.55 5.68 5.35 5.35	5.85 5	24	5.24
							1	ŝ	0	RUSS 39	CTLUN R	CADIAGS	00	06	07	05		2	74
Distance from Centerline (ft.)			71.	.70	8	20	0.4.	₹.	R	.10		24.		2					
Water Depth (ft.)			0.0	.352	.480	.562	÷05	.630	•700	727	.727	.727	869	.629	\$04 <b>*</b>	.561	. 480	353	0*00
$\begin{array}{llllllllllllllllllllllllllllllllllll$																			
Depth from Bottom (ft.)				.70	09.	• 50	07*	• 30 HO	61204TA •20 V	L DISTAN .10 ELUCITY	VCE FHUN 0.0 READING	CEMPER 0.10 S (fps.	LINE (	(**) 0,30	0**0	0.50	0.60	70	
4200 1000 1000 1000 1000	s			s.	0 6.10 6.10 6.10	(	5.15 5.10 5.15 5.15	4.75 4.75 4.75 4.75	4.50 4.55 4.55	4.20 4.35 4.40 4.40	4, 05 4, 20 4, 30 4, 30 4, 35	4.10 4.30 4.30 4.30	4-40 4-40 4-40 4-40	4.60 4.60 4.60 4.65	5.00	5.25 ( 5.50 5.30	5.90 - 52) 5.95 - 52	6,10	
										CHOSS S	sômuk l	EADINGS							
Distance from Centerline (ft.)	<b>,</b> 68	9	.55	.50	:45	07.	.35	• 30	•20	01,0	0.0	<b>.</b> 05	.15	.25	.35	•45	÷5•	.65	.70
Water Depth (ft.)	0.0	.239	.321	.381	664.	484.	.512	.535	•574	.621	6633	.633	.627	.574	.524	.453	.353	.190	00.

									.675 725	0.62 0.63 8.82 0.63 8.82 0.63 6.89 6.81 6.76 6.72 715 775 .713 .167 0.0
									ŝ	6.33 6.33 6.25 6.25 6.12 6.12 6.12 6.25
									.575	6,03 6,08 6,08 5,97 5,68 5,68 5,68 5,68
									.525	52.52 5.47 5.47 5.4785 5.47855 5.47855555555555555555555555555555555555
									.475	5.60 5.59 5.57 5.57 5.60 5.29 5.29
									.425	5.37 5.43 5.42 5.42 5.42 5.42 5.42 5.28 5.28 5.28 5.28 5.28 5.28
									375.	5.25 5.26 5.24 5.19 5.18 375 584
									• 325	5.10 5.17 4.96 5.14 5.14 5.10 5.10
									.275	4.96 5.02 4.99 4.99 4.99 4.99
									.225	4.99 4.97 5.02 5.02 5.02 5.02 5.02 5.02 5.02 5.02
	(U.)	0		880 882 91 91 92					£71.	4.80 4.88 4.98 4.98 4.98 4.98 4.98 4.98 4.98
	NTERLINE	10	(tps.)	1011 4- 1012 4					\$21*	. 72 4. 87 4. 88 4. 88 6. 88 6
	FROM CE	• • • •	SADINGS (	429 445 451 451 451 477 484 487 487 487 487 1.08 1.08 1.08 1.00 1.00 1.00 1.00 1.00	0.0	;u.		ft.)	*075	26.7 27.7 27.7 27.7 27.7 27.7 27.7 27.7
	DISTANCE	.10	UCITY HE	437 4.61 4.61 4.65 4.87 4.87 4.87 4.87 4.87 4.91	so.	.703		CRLINE (1	+025	.025 .736
	TATNOLL	• 20	VEI	4.92 4.92 4.92 4.92 5.03 5.03	.15	.081		ON CEVIE	0.0	(1) 1, 7 1, 7 1
	HOF	.30		( <u>21.)</u> 5.03 51.25 51.25 51.25 51.25 51.25	-25	140.		TAUCE FI	\$0°	ITT HEAL 4.088 4.77 4.77 4.77 4.77 4.77 4.77 4.77 4.77 4.77 4.77 4.77 4.77 4.77 4.77 4.77 4.77 4.77 4.77 7 4.77 7 4.77 7 4.77 7 4.77 7 7 1.77 7 1.77 7 1.72 1.72 1
		07		5.42	•35	£65*		NTAL DIS	.10	VELOC VE
		3.		5.75 5.79 5.75 5.75	\$45	• 5teh		HORIZO	57° 0	1
		8		6,17 6,17 6,17 6,21	÷55,	097.			52 °5	
		φ.		(125) 6119 6121	÷65	.x5			۰ ۶	4.98 4 5.04 4 5.05 4 5.05 4 5.05 4 4.99 4 5.01 4 5.
		÷7.		<u>رده،</u> او د د او	2.	£13			.35	5.17 5.16 5.19 5.19 5.24 5.24 5.24
				77	.74	.089			07.	5.29 5.29 5.29 5.21 5.21 5.11
					~				54.	45 • • 530
21		()			a from ine (ft.	ft.)			8.	5.65 5.65 5.65 5.68 5.65 5.65 5.42 5.42 5.42
RGEO TES		Depth 1 Bottom		.649 .616 .984 .198 .236 .0062	Distand Centerl	Depth (			• 55	8.8.8.8.6.2.2.5 5.5 5. 5 5. 5 5. 5 5. 5 5. 5 5. 5
= SUBHE		ċ							. 60	6.11 6.11 6.12 6.12 6.12 6.12 6.12 6.12
RAW DATA	22	0.0110 3.575 cf	325	64.0 F 17.1 1.69 cm 1.0 0.0			73 10 10 10 10 10 10 10 17 17 17 17 17 10 11 10 10 10 10 10 10 10	ę	70 .6	
6-3-	•	5		S. C			Sta.		. 27.	6 6 6 6 6 6 6 6 6 6 6 6 6 6 6 6 6 6 6
TABLE	Run N	80 2 7 4	Helphane and a second				Run N Run N Reby Reby Ry Ry Ry Ry Ry Ry Ry Ry Ry Ry Ry Ry Ry		.80 Death Eane	Settin (11) 3456 4466 4466 333 3555 333 3555 333 3555 333 3555 333 3555 3333 3555 3333 3555 3333 3555 3333 3555 3333 3555 3333 35555 3555 355555 35555 35555 35555 355555 35555 35555 355555 355555 355555 355555 355555 3555555

BLE <b>8-4</b>	CALC	ULATE	D VA	LUES	- RUN	C	2-10	¥0 ¥n	Fn	
	Run No.	Aar	чt	°c	ď	-0	Y 1	. 25	600	
	2 3 4 5 6 7 8 9 10 11 22 11 11 15 16 71 81 9 20 12 22 3 25 26 6 72 82 9 30 11 22 33 35 36 77 88 9 40 12 24 4 4 4 4 4 4 4 4 4 51 52 53 24 55 56 57 58 5. 6 6 6 26 6 6 6 6 6 6 6 6 6 6 6 6 6 6 6	$\begin{array}{c} .635\\ .632\\ .634\\ .629\\ .624\\ .644\\ .629\\ .544\\ .644\\ .644\\ .594\\ .594\\ .644\\ .491\\ .594\\ .594\\ .491\\ .500\\ .505\\ .501\\ .506\\ .522\\ .498\\ .441\\ .425\\ .422\\ .423\\ .422\\ .423\\ .432\\ .432\\ .432\\ .262\\ .263\\ .275\\$		718 715 715 6088 705 6098 705 702 6094 711 6094 702 6094 702 6094 702 6094 702 6097 700 6097 700 600 6097 700 600 600 600 600 600 600 600 600 60	570 5754 5754 5754 5754 5754 5754 5754 5754 5754 5754 5754 5754 5754 5754 5754 5750 5750 5750 5750 5750 5750 5750 5750 57700 57700 57700 57700 57700 57700 57700 57700 57700 57700 577	. e.c. e.e.	.901 1.220 1.227 1.223 1.420 1.227 1.420 1.429 1.420 1.428 1.420 1.410 1.418 1.4288 1.4288 1.4288 1.4288 1.4288 1.4288 1.4488 1.4488 1.	-435 3933 3979 3970 3970 3970 3970 3970 3970	.587 .588 .589 .5879 .5770 .5770 .5770 .5770 .5770 .5770 .5770 .5770 .5770 .5770 .5770 .5770 .5770 .5770 .5771 .5770 .5771 .5770 .5771 .5773 .5771 .5771 .5771 .5773 .5771 .5771 .5771 .5773 .5771 .5771 .5771 .5773 .5771 .5771 .5773 .5771 .5773 .5771 .5773 .5771 .5773 .5771 .5770 .5770 .5773 .5771 .5770 .5580 .55500 .555000 .55500 .55500 .555000 .555000 .555000 .55500000000	
	Run	К	π,	h1*	$\left(\frac{F_n}{M}\right)^2$	5205	/2 <u>Y1</u>	$\frac{Y_1}{b/2}$	Yn b/2	Ψt
	-16c, 102 103 103 104 105 107 107 107 107 107 107 107 107	2.132 1.886 1.955 2.232 2.210 2.200 2.200 2.200 2.200 2.1960 2.020 2.1960 2.020 2.200 2.188 2.020 2.2000 2.2000 2.2000 2.2000 2.2000 2.2000 2.200000000	141 171 203 104 162 196 234 1171 203 120 234 1271 1592 2271 1592 2271 1592 2277 120 2277 120 2277 120 227 120 20 20 20 20 20 20 20 20 20	In 637 637 637 637 637 637 637 637	8" -003 -004 -286 -286 -286 -290 -200 -2	e	- 	2,290 1,702 1,480 1,480 1,487 1,497	1.375 1.272 1.106 1.521 1.107 1.107 1.272 1.107 1.272 1.007 1.407 1.	$\begin{array}{c} 5,404\\ 5,769\\ 7,417\\ 3,874\\ 4,647\\ 5,124\\ 4,647\\ 5,124\\ 4,647\\ 5,124\\ 4,656\\ 5,175\\ 5,124\\ 4,956\\ 5,175\\ 5,128\\ 5,175\\ 5,128\\ 5,175\\ 5,128\\ 5,175\\ 5,128\\ 5,175\\ 5,128\\ 5,128\\ 5,175\\ 5,128\\ 5,$

Cd

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TABLE 8-5- CALCULATED VALUES - RUN NE. 141 - 306

r

Run No.	к	M1	$\frac{h_1^*}{Y_n}$	$(\frac{F_{n}}{H^{1}})^{2}$	g <sup>205/2</sup>	- <u>Y</u> 1 6		Run No.	К	×	$\frac{h_1}{Y_n}$	$\left(\frac{F_{n}}{k_{1}}\right)^{2}$	g2 b5,	$\frac{Y_2}{b}$
$\begin{array}{c} 141\\ 142\\ 144\\ 144\\ 144\\ 144\\ 144\\ 144\\$	1.527 1.436 1.431 1.227 1.431 1.223 1.237 1.237 1.237 1.237 1.237 1.203 1.207 1.203 1.207 1.203 1.207 1.203 1.207 1.203 1.207 1.207 1.207 1.207 1.207 1.207 1.207 1.207 1.207 1.207 1.203 1.207 1.203 1.207 1.203 1.207 1.203 1.207 1.203 1.207 1.203 1.207 1.203 1.207 1.203 1.207 1.202 1.207 1.207 1.207 1.207 1.207 1.207 1.207 1.207 1.207 1.207 1.207 1.207 1.207 1.207 1.207 1.207 1.207	.154, .154, .223, .223, .234, .134,	.390 .306 .124 .124 .123 .090 .088 .477 .398 .427 .598 .237 .598 .123 .577 .728 .123 .577 .728 .123 .577 .728 .125 .125 .125 .125 .220 .221 .225 .225 .225 .225 .221 .225 .221 .225 .225		2022 2022 2022 2022 2023 2023 2023 2023	.843 .835 .606 .528 1.100 1.030 1.030 1.030 1.030 1.030 1.030 1.030 1.030 1.030 1.030 1.030 1.030 1.031 1.030 1.030 1.031 1.031 1.031 1.031 1.031 1.031 1.031 1.031 1.031 1.031 1.032 1.031 1.031 1.031 1.031 1.0300 1.0300 1.0300 1.0300 1.0300 1.0300 1.0300 1.0300 1.0300 1.030000000000	· ·	222 223 224 225 226 227 228 231 231 231 233 234 235 235 235 235 235 235 235 235 235 241 22 245 235 235 241 24 245 255 255 259 259 259 259 259 259 259 25	1.580 1.085 1.398 1.240 1.073 1.219 1.211 2.251 1.259 2.220 2.200 2.220 2.2000 2.2000 2.2000 2.2000 2.2000 2.2000 2.2000 2.2000 2.2000 2.2000 2.2000 2.2000 2.2000 2.2000 2.20000 2.200000000	252 252 252 252 252 252 252 252 252 252	.285 196 1.129 1.021 1.021 1.177 1.137 .234 3.325 .353 .353 .022 .004 .258 .355 .355 .353 .022 .044 .258 .024 .258 .024 .258 .024 .258 .024 .258 .024 .258 .024 .258 .024 .258 .024 .258 .024 .258 .024 .258 .024 .258 .024 .258 .024 .258 .024 .258 .024 .258 .024 .258 .024 .258 .024 .024 .024 .024 .258 .024 .257 .024 .257 .024 .257 .024 .257 .024 .257 .024 .257 .024 .257 .024 .257 .027 .257 .027 .257 .027 .257 .027 .257 .027 .257 .027 .257 .027 .257 .027 .257 .027 .257 .027 .257 .027 .257 .027 .257 .027 .257 .027 .257 .027 .257 .027 .257 .027 .257 .027 .257 .027 .027 .027 .257 .027 .257 .027 .257 .027 .027 .027 .027 .027 .027 .027 .02	$\begin{array}{c} .353\\353\\353\\1017\\ .$	253 160 2353 3553 2513 2513 2513 2513 2513 2513 2523 2523 2523 2523 2523 2529 1922 2529 1922 2529 1922 2533 2533 2533 2529 1922 2533 2533 2533 2529 1922 2533 2533 2533 2529 1922 2533 2533 2533 2529 1922 2529 1922 2533 2533 2533 2533 2533 2533 2529 1922 2529 1922 2533 2533 2533 2529 1922 2529 1922 2529 1922 2529 1922 2533 2533 2529 1922 2533 2533 2533 2529 1922 2529 1922 2529 1922 2533 2533 2529 1922 2529 1922 2529 1922 2529 1922 2529 1922 2529 1922 2533 2533 2529 1922 2533 2529 1922 2529 1922 2529 1922 2533 2529 1922 2529 2033 2529 25	b .005 .562 1.075 1.270 .200 .200
197 198 199 200 201 202	1.500 1.949 2.082 1.650 1.599 1.370	.112 .175 .212 .252 .158 .135	1.352 .724 .529 .297 1.312 1.499	1.789 .735 .501 .353 1.017 2.159	.441 .253 .199 .160 .353 .441	1.600 .974 .784 .010 1.155 1.463		278 279	.196 .110	.120 .187	.280 .069	2.755 1.131	.529 .303	•843 •565
203 204 205 206 207 208 209 210 211 212 213 214 215 216 218 219 220 221	1.515 1.603 1.448 1.640 1.398 2.589 1.471 1.199 1.398 1.199 1.396 1.119 1.319 1.419 .723 1.674	.211 .256 .140 .100 .150 .154 .154 .223 .154 .223 .112 .175 .211 .175 .211 .140 .252	.086 .497 .246 .836 .710 .540 .307 .121 1.150 .121 1.150 .121 .21 .417 .643 .551 .424 .302	. 580 . 604 1.154 1.011 . 415 . 415 . 200 1.017 . 200 1.017 . 200 1.017 . 734 . 734 . 734 . 734 . 154 1.154 1.154	. 253 .199 .354 .353 .202 .202 .202 .202 .202 .202 .202 .20			285 286 287 288 299 290 291 292 293 294 300 301 302 303 304 305 306	1.855 1.786 1.568 1.366 2.941 3.045 3.082 3.148 3.093 3.151 1.919 1.814 1.650 1.810 2.586 2.523 3.972	.288 .257 .207 .161 .190 .450 .413 .392 .305 .191 .451 .323 .289 .242 .400 .300	1.647 1.401 .978 .658 1.960 1.486 .684 .429 1.263 .996 .721 .421 .421 .322 .225	1.697 1.513 1.217 .949 .280 1.202 .913 .554 .432 .270 1.189 1.019 .832 .453 .168 .169 .247 .113	.045 .045 .045 .045 .012 .120 .169 .087 .087 .087 .087 .087 .087 .087 .087	.500 .508 .509 .404 .559 .526 .425 .425 .425 .425 .4380 .392 .413 .381 .391 .392 .510