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- 12 Keywords: Lunar transparent glass, regolith simulant magnetic beneficiation, ISRU at lunar base,
- 13 window glass made on the Moon, iron-bearing mineral removal, XRF and optical analysis
- 14 Abstract

15 International space agencies are aiming to establish permanent outposts on the lunar surface. For that 16 purpose, new technologies and equipment are being developed which will enable and augment these 17 mission goals. To increase the duration of a long-term planetary mission and to expand mission 18 capabilities, the ability to manufacture transparent glass in-situ could be an important enabler on the 19 lunar surface. Results presented in this work show that it is feasible to use different lunar regolith 20 simulants to manufacture optically transparent glass by magnetically beneficiating regolith prior to 21 processing. Beneficiated regolith simulant was melted, cast into glass nuggets which were then 22 ground, lapped and polished into glass slides of 1mm thickness. The glass slides' surface roughness 23 and geometry were measured, prior to optical analysis, which showed an average transmission of 24 about 80 % of light in the wavelength range from 250 to 1250 nm. A comparable reference glass 25 sample performed only about 9 % (absolute) better on average. From these results it seems viable to 26 manufacture transparent glass from actual lunar regolith on the lunar surface as well, however, 27 differences in regolith simulant and actual regolith still need to be fully explored – Regolith may be available on the lunar surface in unlimited quantities and therefore open up new strategic possibilities. 28

30 Introduction

Missions like the Lunar Orbital Platform – Gateway (LOP-G), seek to enable establishing a first permanent human presence on the Moon and potentially Mars [1]. One of the goals for these missions is to allow for the assembly and operation of surface missions by improving the "payload-to-lunarsurface" metric and cost per kilogram compared to the Apollo missions.

35 During the Apollo missions the actual single launch payload-to-lunar-surface mass was about 6,000 kg 36 [2], which was the mass of the lunar lander after touch down (fuel of decent stage fully burnt). With 37 SpaceX's new Starship, a fully reusable transport system, which shall be capable of servicing Earth's 38 orbit as well as the Moon's and Mars's, this mass may now increase to 100+ metric tons when 39 refuelling in orbit around Earth [3]. Further, technology developed and tested for the international 40 space station (about 400 metric tons [4]) potentially enable smaller space station designs for a future 41 cis-lunar station in the range of about 50 metric tons [5]. Space-X's Falcon 9 showed a 10-to-1 42 reduction in costs for development [6] and a 20-to-1 reduction in payload launch cost to Low Earth 43 Orbit (LEO) [7].

Despite these improvements, future human missions to the lunar surface will still be constrained by
mass transportation logistics [8] [9]. This is likely to be especially true if mining equipment has to be
flown [10] [11] to the lunar surface for, for example, mining water, aluminium, titanium, iron or oxygen
[12] .

48 To allow for sustainable long-term space exploration and exploitation, means to lessen the mass 49 constraints have been devised. "In-Situ Resource Utilization" (ISRU) aims at utilising and harnessing 50 space resources for the purpose of creating items and products which enable space missions by 51 significantly reducing the mass, cost, and risk lunar surface exploration [13] [14]. Multiple concepts 52 and ideas for lunar missions using lunar resources [15] have been investigated up to the present day 53 [16], but only a small number of these studies focused on using local resources to manufacture 54 synthetic glass (fibres) [17] [18] [19] or even glass parts [20] [21] [22]. Utilising local lunar regolith to 55 manufacture transparent glass has only been investigated by considering pure anorthite, a lunar 56 regolith simulant, as an input material rather than bulk regolith simulant and actual regolith [23].

Artificially fabricated synthetic glass containers and glazing are amongst the oldest glass applications on Earth. Ever since first using glass as building material for the first time a multitude of other applications such as fibres, displays or electronic components have been developed. Glass can be transparent, recycled, strong, chemically inert and is readily castable amongst other characteristics of which most can be tailored to the specific needs today [24]. Making glass available as raw material on the lunar surface may open possibilities for using glass in construction on the lunar surface. Possible applications are windows, mirrors, solar cells, fibres or insulation foams, [23] which all require glass material of different qualities and properties. For example, using synthetically manufactured glass as backplate for front coated mirrors, glass composition is of minor importance compared to glass surface quality. In comparison window glass or cover glass for, for example, solar cells, will require optically transparent glass.

68 The goal of this work was to determine how transparent glass can be manufactured from bulk lunar 69 regolith by using artificial lunar regolith (regolith simulant) as starting material, and then to determine 70 the optical quality of the glass produced.

71 Terrestrial basaltic glass, lunar glass and synthetic glass

72 For this work, glasses found in nature (on the Moon and on Earth), are considered natural glasses and 73 glasses manufactured from sand, rocks and minerals are considered synthetic glass. Compared to 74 research conducted on terrestrial synthetic glasses, which are used in an increasing number of 75 applications on Earth, research conducted on natural (basaltic) glasses is comparably limited. On the 76 Moon glass has been found in quantities from 1-17 % in the mare regions and 5-25 % in the highland 77 regions [25]. There have been numerous geochemical studies of natural lunar glasses [26] [27], 78 principally aimed at determining how glasses and related volcanic products were formed, and what 79 they may indicate about the nature of the lunar interior [28] [29]. Studies on natural terrestrial glasses 80 also typically focus on geochemical characteristics, which can indicate conditions and geological 81 settings under which parental melts formed [30] [31] [32]. However, there has also been considerable 82 research on crystallisation behaviour of terrestrial basaltic glasses [33] [34] as well as their 83 physical/chemical [35] and magnetic [36] properties. Practical applications of synthetic basalt glass 84 are, for example, basalt fibre reinforced concrete [37] or immobilisation of transuranic wastes [38]. 85 With respect to potential lunar applications, synthetic lunar glass can be produced from bulk lunar 86 material [19] and has already been used to manufactured synthetic glass substrate from artificial lunar 87 regolith (regolith simulant) for mirrors [20]. To the knowledge of the authors, only one study has been 88 conducted on manufacturing transparent synthetic lunar glass from anorthite [23] but not using a 89 regolith simulant as a starting point for manufacturing.

90 Glass sheet manufacturing from regolith simulant

91 The overall goal of this study was to manufacture a transparent glass sheet and to analyse its optical
92 properties, using available analogue lunar material. Previous work on basaltic glasses as a start did not
93 provide guidelines on the actual manufacturing process. Thus, terrestrial soda-lime glass

94 manufacturing was targeted, as this has been studied for centuries [39] [40]. Glass manufacturing on 95 Earth has changed much from early glass production [41], to the rise of the float glass process for 96 glazing [42], and many different types of glass have been developed and utilised. Other than the glass 97 composition and selected process, key parameters are processing temperature, material 98 combinations and correct cooling and annealing. This is in order to avoid, for example, bubble 99 formation or stress building up in the glass [43]. Due to the lack of practical experiments conducted 100 with basaltic glasses, it was required to develop a manufacturing process first before samples could 101 be produced and measurements could be conducted. The developed method will be described after 102 glass colour and the regolith beneficiation process have been discussed briefly.

103 Basaltic glass colour

104 Terrestrial basaltic glasses are typically black-brown-green and low transparency and lunar glasses are 105 variable in colour. Glass beads recovered during Apollo missions can be characterised based on Ti 106 content and colour, varying from green to orange-black [26]. Although multiple elements influence 107 glass colour, one of the elements having a major impact is iron [44] [45]. Typically, natural terrestrial 108 basaltic glasses have a green-brown/black taint which results primarily from the presence of Fe²⁺ and 109 Fe³⁺. The presence of smaller amounts of other metal ions such as Cr, Mn, V, and Co may also 110 contribute to the colour. Only considering iron, synthetic glasses with very low, approx. 0.01 weight 111 percent (wt%) ferric oxide content do not show the blue-green coloration of typical window glass with 112 0.1 wt% ferric oxide content, especially for thicknesses >> 1cm [46]. Other applications require a 113 certain amount of iron oxide to manufacture highly coloured glasses such as car windows or beer 114 bottles with Fe₂O₃ contents of 1.4-4 wt% [47]. However, it should be noted that terrestrial glasses are 115 not typically synthesised from basaltic material, which is chemically complex and typically contains appreciable amounts of iron. Other elements potentially impacting synthetic glass colour, even in 116 117 quantities as small as 0.1 wt%, are Ti, Cr and S [45] [44]. On Earth, a geologically complex environment 118 means that silica-rich, relatively Fe-poor material for making transparent glass is readily available. This 119 is not the case for the Moon, whose surface represents products formed by cooling on an extensive 120 magma ocean, overprinted by later, dominantly basaltic volcanism [26].

For this study, the prime focus was on removing iron oxide, and a secondary focus on removing titanium, in order to synthesise transparent glass from lunar regolith simulant. The regolith simulants used, which represent a range of six available regolith simulant materials (BP-1, EAC-1, FJS-1, JSC-1A, JSC-2A, LHT-3M; more details in methods section), contained between 5.56 to 13.18 wt% Fe_2O_3 and 0.11 to 2.15 wt% TiO₂ in the raw bulk. Since these two were considered the prime contributor to the

- 126 colour of any glasses produced, the first step in trying to manufacture optically transparent glass from
- 127 these simulants was to remove as much Fe and Ti oxide from the input materials as possible.

128 Beneficiation of regolith

129 The goal of this work was to extract iron from a variety of six regolith simulants composed of igneous 130 rocks and minerals. Extraction of Fe, Si, Ti, H₂O, Al, Mg and O from extra-terrestrial sources has been 131 studied in the past and an overview is presented in [16]. Works reviewed in [16] include methods to 132 extract, for example, oxygen, water, and metals, and use principles such as vapor phase pyrolysis, 133 molten regolith electrolysis or carbothermal reduction. Some of these techniques are based on 134 terrestrial exploitation techniques, and although modifications to these conventional techniques may 135 work on the lunar surface, they are often energy intense and/or heavy on consumables and/or 136 complex. Further, the reviewed techniques in [16] sometimes considered pure minerals as inputs 137 rather than raw regolith, which seems unrealistic. However, it is more realistic to consider unaltered 138 regolith to be available as a resource. Unaltered regolith contains all minerals found at the landing site 139 which will probably show a range in both mineral and chemical composition. The yield of an extraction 140 method may suffer if it has to treat mineral fractions it was not designed for. However, only 141 considering certain minerals as input materials seems unrealistic in the authors' view as it is unlikely 142 that only a specific mineral, such as ilmenite, is found near the landing site in required purities and in 143 the right grain size spectrum.

144 Therefore, the authors suggest beneficiation of regolith material prior to extraction which will increase 145 the yield of the most often standalone solutions. Beneficiation is regarded as prior separation of a raw 146 material's minerals by means of mechanical processing, while using no consumables and as little 147 power as possible. Further, in searching for a suitable beneficiation process the goal was to find a low-148 tech, reliable approach which would be capable of handling a wide variety of input materials. 149 Although, the mining industry has a wide range of potential beneficiation methods available, the 150 ultimate process was derived from a series of techniques geologist use to separate heavy, often Fe-151 rich minerals from a sand/crushed or powdered sample.

152 Geologists approach to separating high density minerals

One approach to beneficiation is to process material to remove any denser minerals particularly enriched in iron. Geologists often use three techniques to separate heavy minerals such as apatite and/or zircon from a sand sample: shaker tables, electromagnetic separation, and heavy liquids. Since a shaker table requires large amounts of water and comparably much space to work, it was deemed unrealistic to be used on the lunar surface. Further, the impact of the reduced gravity environment on this technique is not clear. Heavy liquids in general could work on the lunar surface and are an attractive approach as they use zero power, little space and are simple. However, each liquid can only separate a sample into two fractions, one fraction heavier and one lighter than the liquid. Thus, they are rather a crude method of modifying a concentrate composition. This led to using electromagnetic separation to split regolith (ore) into multiple different fractions with the goal to remove as much iron, as possible.

164 Materials and Methods

Preparing transparent glass samples from lunar regolith simulants required a multistep workflow asdepicted in Figure 1. The following sections provide details with respect to every step on the work

167 flow in Figure 1.

168 Selection of Simulants

169 Six different lunar regolith simulants were selected with the aim to provide a variety of relevant 170 compositions. This helped to test whether the developed process is sensitive to input material 171 variations. The selected simulants were Black Point -1 (BP-1), European Astronaut Centre -1 (EAC-172 1), Fuji Japanese Simulant – 1 (FJS-1), Johnson Space Center - 1A (JSC-1A), Johnson Space Center - 2A 173 JSC-2A, NASA/USGS - Lunar Highland Type - 3 Medium (LHT-3M). All but LHT-3M were mare simulants 174 comprised primarily of igneous rock rich in iron oxide. LHT-3M is a highland simulant and is 175 constructed from plutonic rock, rather than igneous rock. It was correspondingly iron sparse compared to the other five simulants. All simulants were designed to match Apollo sample grain size 176 177 distribution but have been sourced from different suppliers and geolocations and are listed in Table

178 1.

Simulant	Description
BP-1	for Black Point – 1, was sourced from the Black Point basalt flow (San Francisco
	Volcanic Field) in northern Arizona. The sample was kindly provided by NASA Swamp
	Works (KSC) and is a mare simulant [48].
EAC-1	for European Astronaut Centre – 1, was sourced from the so called "Huehnerberg"
	quarry located in the Eifel region in Germany, south of Cologne. The sample was kindly
	provided by the European Astronaut Centre and is a mare simulant [49] [50].
FJS-1	for Fuji Japanese Simulant – 1, was procured from the Shimizu Corporation, which
	sourced it near Mount Fuji and is a mare simulant [51].
JSC-1A	for Johnson Space Center - 1A, was sourced from the volcanic ash field (San Francisco
	Volcanic Field) in northern Arizona. The sample was kindly provided by NASA Swamp
	Works (KSC) and is a mare simulant [52].
JSC-2A	for Johnson Space Center - 2A procured from Zybek Advanced Products in Westminster
	(CO) [53], USA and it was manufactured to be like JSC-1A and is a mare simulant.

LHT-3M for NASA/USGS - Lunar Highland Type - 3 Medium, procured from and manufactured by Zybek Advanced Products in Westminster (CO) [53], USA, thus not directly linked to NASA or USGS any longer. The precursor simulants to NU-LHT-3M were the "medium 1 and 2", NU-LHT-1/2M [54] which were manufactured and developed by USGS and NASA. All LHT family simulants are supposed to be roughly the same and all represent highland regolith simulants.

The oxide composition of each simulant, as provided by the manufacturer, is listed in Table 5 in supplemental material. For most simulants only a range for each oxide was provided by the manufacturer, rather than absolute values. Nevertheless, the manufacturer values provided enough information for the selection of the simulants. After arrival of the simulants, all raw materials have been sampled and XRF measurements were taken as well as XRD measurements on EAC-1, FJS-1, JSC-2A and LHT-3M. No XRD measurements have been taken on BP-1 and JSC-1A, however, JSC-1A is likely very similar to JSC-2A. Details on the measurement procedures are listed in the end of the methods

187 section under analysis.



188

Figure 1 Workflow undergone by each of the selected six regolith simulants. After initial measurements, beneficiation via sieving and magnetic separation followed. Next, all beneficiated samples were analysed via means of XRF, prior to hot processing. Glass formed from the regolith simulants was cold processed to produce final glass slides which were then analysed by means of profilometry to determine roughness and flatness and in a UV-vis-spectrometer to determine reflectivity and transmission of the gasses.

- 194 Beneficiation Method
- 195 200 grams of each regolith simulant sample have been dried, sieved and magnetically processed.
- 196 After sieving three grain size groups for each selected simulant were available. Each of these grain
- 197 size groups was processed separately via magnetic beneficiation.

198 Drying and sieving

Prior to magnetic separation, 200 grams (each) of all six simulants were dried at 220°C for two hoursand sieved afterwards.

By sieving, the simulants were divided into 6 groups, according to grain sizes: > 1000 μ m, 1000-500 μ m, 500-250 μ m, 250-125 μ m, 125-63 μ m and < 63 μ m. Afterwards, the remainders from three grain size groups were used: 500-250, 250-125 and 125-63 μ m and the other groups were discarded as waste. The combined weight of these three grain size groups was different for all six simulants and within the range 130 to 95 grams (details in Table 6 in supplemental material). Next, the remaining amount for these three grain size groups was processed via (electro)magnetic separation.

207 Magnetic separation

208 As shown in Figure 1 the sieved and dried samples were first processed with a hand magnet (left in 209 Figure 3) and then using the electromagnetic setup depicted in Figure 2. Before using this 210 electromagnetic setup, the standard hand magnet (hm) depicted on the left in Figure 3 was used to remove highly magnetic material from the bulk (samples marked "hm"). A total of approximately 100 211 212 grams of each regolith simulants split in three grain size groups was processed. For each of these, 213 three different electromagnetic field intensities were used during separation. Ultimately this led to 214 five samples types (hm, high, fair, low, non) for each grain size group, each with a different magnetic 215 susceptibility. After each run, the entered regolith simulant group was split into a magnetic and non-



Figure 2 Electromagnetic separator, laboratory setup (left), schematically (right). Entry funnel (1) for regolith, copper slide (2), leading trough dedicated space in-between an electromagnet (3), at the end of the slide splitting regolith in two buckets containing non-magnetic (4) and magnetic material (5). Pitch and roll angel can be adjusted using the wheel (6).

- 216 magnetic fraction. In this case, "non-magnetic" simply refers to the fraction of material not 217 susceptible to exposure of the specific magnetic field used in that run. Hence, during another run, 218 processing regolith with a stronger magnetic field, parts of the regolith may be susceptible.
- The three different grain size groups (500-250, 250-125 and 125-63 μ m) were processed separately for each simulant, to avoid contamination and clogging of the machine. This ultimately led to 6 regolith simulants, each processed at 3 different grain sizes, and split into 5 categories of magnetic

- susceptibility. Thus, 6*3*5 = 90 samples were obtained. After processing, the 3 different grain sizes
- for each simulant magnetic susceptibility subgroup were recombined, which led to 6*5 = 30 samples.



Figure 3 Two steps of magnetic separation, left separation utilising a hand magnet, right using an electromagnetic setup
both splitting the regolith into a magnetic (1) fraction and a non-magnetic fraction (2)

The physical size of the electromagnet and slide led to a processing time of 4-5 hours for each grain size group and led to the sample amounts listed in Table 6 in supplemental material. Each sample was separated into five groups: hand magnet (HM), high, fair and low magnetic susceptibility and nonmagnetic (non) in the last run. Further, losses have been calculated based on the input amount of sample.

232 During each separation step, samples were split into two groups, tailings and concentrate. Here, 233 tailings refer to the fraction of the sample which was magnetically susceptible during a processing 234 step. These tailings were removed and not further processed but an XRF measurement was taken. The 235 concentrate refers to the second group which was not magnetically susceptible during a processing 236 step. This group was processed further until the final step, which led to the "low" (tailings) and "non" 237 (concentrate) group. Hence, all measurements have been taken from the tailings but the last sample 238 "non" which was the concentrate. Note that prior to manufacturing a 1 g sample was taken from each 239 magnetically beneficiated group (30 in total) for XRF analysis.

240 Hot Processing Method

224

After the simulants had been separated into different groups, the "non" group was used to attempt to manufacture a transparent glass sheet. Since LHT-3M-non was the only group which delivered enough sample (>10 g) to manufacture a glass sheet of typical size, it was the only one leading to 3 "good" samples. However, despite the lack of a large amount of "non" sample, manufacture of transparent glass from BP-1-non, FJS-1-non and JSC-2A-non was also attempted, as described below.

246 Heating and casting

247 Figure 4 shows parts of the heating and casting process for manufacturing a glass sheet from 248 magnetically beneficiated regolith simulants. On the left side in Figure 4 LHT-3M-non (1) and JSC-2A-249 non (4) samples are depicted prior to processing and combining of gran size distributions. In the top 250 middle of Figure 4 a platinum crucible is shown next to a vial of LHT-3M-non prior to heating. Each 251 sample was processed following the same procedure. Samples of all grain sizes processed via magnetic 252 beneficiation were combined, placed in a platinum crucible, and heated at 1550 °C for 15 minutes in 253 a resistive heated furnace (temperature empirically determined as best working point temperature). 254 Then the crucible was removed from the furnace and the molten regolith sample was cast from the 255 platinum crucible into a graphite mould. Typically, the sample was then allowed to cool to room 256 temperature before being removed from the mould and further processed. Due to the small amount 257 of sample available for this experiment, two more samples of LHT-3M glass were manufactured by 258 entering a smaller amount of molten regolith into the graphite mould and pressing down on the 259 sample with a graphite plunger. These two resulting samples are shown in (3) and (6) in Figure 4 and 260 appear transparent. Also shown in subfigure 5 is a platinum crucible from the experiments with JSC-261 2A with the small amount of JSC-2A-non, which also turned into a transparent glass.

262 No simulant other than LHT-3M, delivered an amount of more than 10 g of non-magnetic material. 263 However, BP-1, FJS-1 and JSC-2A delivered enough to attempt limited glass manufacturing as well. 264 Unfortunately, the thermal mass of these three samples was not sufficient to allow casting from a 265 platinum crucible, since the sample solidified in the crucible before it could be cast. Use of higher temperatures (1700 °C) did not overcome this issue. Thus, samples were put into a graphite crucible 266 267 (known not to bond with regolith materials) and entered in the furnace at 1550 °C for only 3 minutes. After removing the crucible from the furnace, the sample was kept in the crucible and a graphite 268 269 plunger was used to press down on the hot liquid glass to obtain a thin, elongated piece of glass similar 270 to (6) in Figure 4.



Figure 4 Iron sparse regolith "non" (after magnetic beneficiation) on the left (1 - LHT-3M and 4 - JSC-2A), iron sparse LHT-3M regolith next to crucible prior to melting next to platinum crucible (2), melted, cooled and compressed LHT-3M-non in the mould after processing (3) and after removing the samples from the crucible (6). Melted and cooled iron sparse JSC-2A in platinum crucible without contamination depicted in 5.

271 Annealing

After heating, melting, casting, and cooling to ambient temperature, a total of six samples (3 LHT-3M, 1 BP-1, 1 FJS-1, and 1 JSC-1A) were obtained. For annealing these samples were all heated to 700 °C, starting at room temperature, at a rate of 170 °C/h. Next, samples were cooled from 700 to 550 °C at a rate of 150 °C/h until in a final step all samples were cooled back to room temperature at a rate of 60 °C/h. Utilising this annealing process with samples of the given size or smaller avoided any breakage or cracking of the samples during processing or while handling them.

278 Cold Processing Method

After heating and annealing, six shapeless samples were obtained. As shown in the workflow diagram 279 280 in Figure 1 the next steps were cold processing the samples by means of grinding, lapping and polishing. Figure 5 shows the largest obtained raw sample (approx. 40*20*12mm) after casting (left) 281 282 and after all cold processing steps (right). To achieve a flat, parallel glass slide it was necessary to grind 283 the sample first with a 74 μ m diamond disc into the rough shape of a glass slide. Next, the samples were mounted on a glass slide using Crystal-bond[™]. Then, the samples were processed on a lapping 284 machine (to obtain parallel surfaces) with silicon carbide slurry to a thickness of 1.5 to 0.5 mm. In a 285 286 last step samples were machine polished using 0.3 µm aluminium slurry. After polishing the first surface, each sample was heated, the Crystal-bond[™] was removed, the sample separated from the 287 glass slide, flipped and remounted on the sample holder. This ensured that both sides of the sample 288 289 were as flat and smooth as possible, as required for optical measurements. All six (3 LHT-3M, 1 BP-1,

- 290 1 FJS-1, and 1 JSC-1A) samples could be processed into slides. However, latter three samples delivered
- 291 limited usable samples due to contamination shown in Figure 6 and discussed below.



292

- **293** Figure 5 Transparent glass made from regolith simulant LHT-3M. Sample after casting and annealing (left) and after
- 294 processing (right).



- 295
- 296 Figure 6 Glass panes made of regolith simulants BP-1 (left), FJS-1 (middle) and JSC-2A (right). Partial transparency could be
- 297 achieved with visible black streaks of carbon (indicated by red arrowheads) which penetrated into the samples during
- 298 manufacturing from the graphite crucible.

299 Analysis Methods

This section describes methods employed to measure mineralogical (XRD) and oxide (XRF) content of the six regolith simulants utilised. Further, it describes methods used to measure the surface properties (roughness and flatness) and the optical properties (reflectivity and transmission) of the
 transparent glasses manufactured from some of these simulants.

304 Mineralogic analysis of regolith samples via XRD and SEM

305 X-ray powder diffraction (XRD) was used to identify phases in four unprocessed (EAC-1, FJS-1, JSC-2A 306 and LHT-3M) regolith simulant samples. All four samples were micro-ground and then spray dried 307 prior to being analysed on a Bruker D8 Advance diffractometer using Cu K-alpha radiation produced 308 by a 40kV accelerating voltage and a tube current of 40mA. The detector used was a sol-x energy 309 dispersive detector tuned for Cu K-alpha radiation, capable of quantitative and qualitative 310 identification of crystalline materials. Bruker Diffrac.EVA software in combination with latest 311 International Centre for Diffraction Data (ICDD) database was used for phase identification, and TOPAS 312 3.0 software for full profile Rietveld analysis and determination of phase proportions.

313 The proportion of glass/amorphous material present was not determined directly. Rather, its present 314 was first identified by visual identification in thin sections, then by its effect on peak intensities in the 315 XRD traces. The amount of amorphous material present in each sample was determined by a 'spiking' 316 method. For this, a known amount of an exotic mineral (in this case Calcite, CaCO₃) was added to the 317 sample. This composite sample was then scanned by XRD and the proportion of each mineral present (including the CaCO₃ spike) determined by Rietveld analysis. As the spike mineral concentration is 318 319 known, the absolute amount of each mineral present can then be calculated by determining its 320 absolute concentration with respect to the known concentration of the spike mineral. Thus, when the 321 amounts of all the minerals present are summed, any deficit from 100% must either be accounted for 322 by a mineral phase not included in the Rietveld analysis or by the presence of a 'diluting' amorphous 323 phase. As all the diffraction peaks in the XRD traces have been assigned to mineral phases we discount 324 the possibility of a significant amount of an unknown phase(s) being present in the sample. Thus, any 325 deficit from 100% in the sum of the mineral assemblage must indicate the presence of an amorphous 326 phase.

In addition to the XRD measurements, thin sections of the samples were prepared for compositional
 analyses using a scanning electron microscope with quantitative energy dispersive spectrometer
 (SEM/EDX). In-situ compositional data was obtained using a Carl Zeiss SIGMA HD VP FEG SEM fitted
 with Oxford AZtec EDX system.

331 Analysis of regoliths' geochemical composition using XRF

All six simulants used for this experiment were analysed using x-ray fluorescence spectroscopy (XRF)
 to determine elemental composition. Two separate sets of measurements have been taken, one at

the beginning taken from each regolith simulant in its raw unaltered state and one after magnetic
beneficiation. Latter one led to a total of 30 samples as described earlier in the beneficiation methods.
This led to a total of 36 measurements.

Bulk composition of the sample was carried out by X-ray fluorescence using a Philips PW2404 wavelength dispersive sequential X-ray spectrometer at the School of GeoSciences at the University of Edinburgh. The system is fitted with a rhodium anode end window X-ray tube operating at an accelerating voltage of 50kV and a tube current of 50mA. Fused glass discs were prepared as described in [55]. Samples were dried at 100°C and fused into glass discs using a borate flux (Johnson and Mathey Spectroflux 105¹) in a ratio of 1:5, sample:flux.

For each of the 36 XRF measurements the loss on ignition (LOI) was determined, which is the weight loss shown by a sample after heating, in air, to 1100°C for 20 minutes. This operation was carried out immediately prior to fusion of the sample into glass discs as described above. LOIs are listed in Table 7 in supplemental material.

347 Surface and parallelism analysis

348 Surface roughness and flatness of two LHT-3M-non glass samples versus one 'off the shelf' microscopy 349 slide was determined using a profilometer. For this measurement a Taylor Hobson Talysurf-5 modular 350 system was utilised which records the results on electro-sensitive chart paper, which was then 351 digitised. The systems sensitivity tolerance is ± 2.0% and is driven by a synchronous motor. The 352 instrument's stylus made one trace across the surface of about 1mm in the centre of the sample and 353 along the longest dimension of the sample.

The parallelism of all the samples (microscopy slide, 3 LHT-3M samples & BP-1, FJS-1 and JSC-2A samples) was determined by measuring thickness of each sample at each corner using a micrometre screw and calculating the difference in thickness between these points.

357 UV-vis-IR spectroscopy

Reflectivity and transmission of the six regolith glasses and one reference microscopy slide were determined over a wavelength range of 350 -1250 nm by using a PerkinElmer Lambda 950 UV-vis-NIR spectrometer.

¹ Spectroflux 105 consists of a mixture of 47 % Lithium tetraborate, Ll2b4O7, 37 % Lithium carbonate (Li2O) and 16 % of La2O3, Lanthanum oxide as an X-ray heavy absorber.

361 Results

During the process of preparing a transparent glass slide from regolith simulant, three main measurements have been conducted. First, the analysis of the regolith samples composition prior and after the magnetic beneficiation. Second, the samples surface roughness after grinding, lapping and polishing. Third, the optical properties of each sample with a focus on transmission measurements. All percent values in this chapter are weight percent (wt %) if not indicated otherwise.

367 XRD

368 Results obtained from XRD analysis of four (EAC-1, FJS-1, JSC-2A and LHT-3M) of the initial, raw and

unaltered regolith samples is shown in Table 2. Additionally, for reference supplier values for BP-1 arealso included.

Group	BP1 ^M	EAC1	FJS1	JSC2A	LHT3
Plagioclase	57.7	13.8	55.9	45.7	66.6
K Feldspar	7.3	13.4	5.0	5.0	2.9
Pyroxene	13.8	35.5	26.3	6.8	24.4
Olivine	12.9	13.3	5.0	11.9	1.3
Oxide Minerals	8.3	2.0	1.5	0.6	0.3
Glass	na	14.4	0.7	23.9	0.0
Mica	na	3.3	3.2	1.8	0.9
Alteration	na	4.3	2.4	4.4	3.8
Sum	100.0	100.0	100.0	100.0	100.0

371 Table 2 XRD results overview, values for mineral groups are displayed in wt %

^M Manufacturer Data [56]

372 Mineral phases detected by XRD with Rietveld refinement include numerous members of solid

solution series, and can be categorised into minerals groups, as listed in Table 3.

374 Table 3 Detected minerals groups and individual minerals by XRD with Rietveld refinement

Group	Minerals contained
Plagioclase	Albite, Andesine An50, Anorthite, Bytownite An85, Labradorite An65, Oligoclase
	An16
K Feldspar	Microcline maximum, Orthoclase, Sanidine Na0.07, Nepheline
Pyroxene	Aegirine, Augite, Diopside, Enstatite, Pigeonite
Olivine	Forsterite (iron)
Oxide Minerals	Ilmenite, Titanomagnetite
Glass	Amorphous material
Mica	Annite Mica, Muscovite 2M1
Alteration	Chlorite, Illite, Kaolinite (BISH), Phlogopite

375 XRF

The results of the XRF analysis of the initial 6 raw, unaltered regolith simulants, as well as the 30 magnetically beneficiated samples are listed in Table 5. Again, the categories for the 30 magnetically altered samples are hm, low, fair, high and non, and for the raw unaltered regolith its unaltered name is used. The table shows the unaltered composition of each simulant in the first row (full simulant name) and the according changes for each level of magnetic treatment. Further, the according loss on ignition is listed with every sample.

382 Table 4 XRF analysis results of magnetically altered regolith simulant samples, shown in wt%, not corrected for LOI.

383 Unaltered samples are shown in the first line named after the simulant, results of the tailings of each step are shown as

384 *"hm", "high", "fair" and "low" as well as the final concentrate is shown under "non".*

Sample	SiO ₂	Al ₂ O ₃	Fe ₂ O ₃	MgO	CaO	Na ₂ O	K ₂ O	TiO ₂	MnO	P ₂ O ₅	LOI	Total
BP-1	46.13	15.91	11.96	6.31	10.28	3.08	1.01	2.04	0.17	0.39	2.04	99.32
B-hm	46.14	15.86	13.86	6.52	9.25	3.28	1.04	2.56	0.19	0.42	0.88	100.01
B-high	46.82	15.56	11.58	7.27	10.08	2.98	0.99	1.68	0.17	0.37	2.22	99.73
B-fair	46.44	14.74	9.98	8.29	11.11	2.71	0.93	1.22	0.15	0.31	3.76	99.63
B-low	45.85	15.20	5.03	3.96	15.93	2.47	0.95	0.76	0.09	0.23	9.11	99.56
B-non	45.78	19.29	1.39	1.12	19.08	2.50	0.44	0.18	0.03	0.06	9.80	99.66
EAC-1	43.58	11.45	12.66	14.08	10.18	2.62	1.18	2.15	0.21	0.59	1.40	100.11
E-hm	44.20	12.96	12.26	11.58	10.79	2.67	1.35	2.37	0.19	0.67	1.19	100.24
E-high	43.63	9.21	13.27	18.66	8.58	1.51	0.99	1.45	0.24	0.42	1.74	99.69
E-fair	41.93	2.53	13.99	36.40	3.08	0.03	0.27	0.28	0.24	0.08	1.08	99.90
E-low	46.80	5.53	7.45	23.75	8.44	0.67	0.96	0.38	0.19	0.11	6.05	100.34
E-non	40.78	5.82	1.44	5.27	23.39	0.84	1.37	0.13	0.14	0.08	20.31	99.57
FJS-1	49.82	16.56	12.90	5.91	9.71	2.42	0.66	1.46	0.20	0.28	-0.25	99.67
F-hm	49.40	14.50	14.85	6.73	9.15	2.15	0.76	1.75	0.22	0.32	-0.19	99.66
F-high	50.32	18.35	9.81	7.04	10.06	2.29	0.53	0.69	0.16	0.21	-0.16	99.31
F-fair	49.75	17.33	10.03	10.15	10.03	1.92	0.32	0.44	0.17	0.11	-0.37	99.88
F-low	51.09	25.25	4.42	3.40	12.25	2.85	0.28	0.24	0.07	0.06	-0.04	99.88
F-non	51.81	29.39	1.42	0.01	13.30	3.36	0.25	0.10	0.02	0.02	0.22	99.89
JSC1-A	46.76	16.24	12.62	8.57	9.90	2.91	0.82	1.80	0.19	0.70	-0.46	100.04
J1-hm	46.66	16.32	12.60	7.28	10.16	3.07	0.87	1.92	0.19	0.76	-0.46	99.39
J1-high	46.26	15.50	13.14	9.87	9.34	2.86	0.80	1.76	0.20	0.68	-0.57	99.83
J1-fair	46.36	15.58	12.71	10.22	9.44	2.81	0.76	1.67	0.19	0.64	-0.67	99.71

J1-low	47.89	21.05	8.67	4.76	11.88	3.20	0.64	1.31	0.13	0.50	-0.20	99.82
J1-non	39.53	21.46	1.29	1.03	22.87	1.94	0.16	0.18	0.02	0.19	11.11	99.77
JSC-2A	46.28	16.63	13.18	7.98	9.65	3.11	0.82	1.83	0.20	0.71	-0.79	99.59
J2-hm	46.32	15.94	13.62	8.15	9.59	3.05	0.88	1.96	0.20	0.77	-0.65	99.82
J2-high	45.75	15.13	13.54	10.29	8.94	3.35	0.84	1.76	0.20	0.68	-0.64	99.86
J2-fair	46.21	15.87	12.76	10.72	9.18	2.72	0.72	1.57	0.18	0.61	-0.70	99.82
J2-low	46.55	17.26	11.60	9.81	9.78	2.72	0.65	1.42	0.17	0.54	-0.46	100.05
J2-non	37.89	18.93	0.93	0.90	24.29	1.97	0.29	0.13	0.02	0.21	14.05	99.60
LHT-3M	49.34	21.59	5.56	9.49	12.54	1.04	0.08	0.11	0.09	0.02	0.12	99.97
L-hm	49.30	16.64	9.17	13.00	10.93	0.52	0.05	0.14	0.14	0.02	-0.36	99.56
L-high	50.27	12.38	8.36	17.33	8.24	0.34	0.03	0.14	0.15	0.00	2.22	99.47
L-fair	50.82	14.82	7.50	15.64	9.83	0.45	0.04	0.13	0.14	0.00	0.21	99.59
L-low	48.23	27.46	2.76	4.52	15.22	1.07	0.06	0.06	0.05	0.01	0.26	99.70
L-non	47.32	33.13	0.72	n.d.	17.22	1.32	0.07	0.03	0.01	0.01	0.25	100.06

n.d. = not detected

385 Surface Properties

386 Surface roughness of two LHT-3M samples were measured versus one 'off the shelf' microscopy

387 slide. Results are depicted in Figure 8 and show the three samples over the distance of up to 0.9 mm

and a deviation from the mean line within 15 to -20 nm. LHT-3M samples are depicted in green and

the microscopy slide in violet.

390 Thicknesses measurements obtained with a micrometre screw reach from 1.4 mm to about 0.8 mm

and differences between corners of regolith samples are 0.040 to 0.010 mm. Detailed values can be

392 found in Table 9 in the supplementary material.



Figure 7 Surface roughness of two polished transparent substrate made from LHT-3M (shades of green) which has been magnetically treated to remove iron oxide. Reference; 'off the shelf' microscopy slide (violet). The abscissa shows length on the samples surface in mm and the ordinate deviation from the mean line in nm.

393 UV-vis-IR results

- 394 Results of reflectivity measurements are depicted in Figure 10 and results of the transmission in Figure
- 395 9. Also included on Figure 9 is an AMO spectrum, superimposed on the figure, showing the spectrum
- which would be seen by a solar cell or a sample in space, in the vicinity of Earth.
- 397 From the results in Figure 9 average transmissions over the entire wavelength range have been
- 398 calculated and are listed in Table 8 in supplementary material.



Figure 8 Transmission of transparent regolith simulant glass substrates over a wavelength range of 350 to 1250 nm, made from four different regolith simulants and compared to a microscopy slide ("Micro"), superimposed by the solar spectrum AMO which would be seen by a solar cell or a sample in space, in the vicinity of Earth. Lines of samples LHT-3M-1 and LHT-3M-2 coincide.

399

Figure 9 Reflectivity of transparent regolith simulant glass samples over a wavelength range of 350 to 1250 nm, made from four different regolith simulants and compared to a microscopy slide ("Micro").



401 Discussion

400

402 Beneficiation

403 Drying and sieving went as expected with the only exception that the large amount of fine grained

404 regolith (<63 μ m) led to longer processing times in the sieving tower. This was to allow for all the fine

grain material to migrate to the bottom of the sieve tower and not be incorrectly incorporated inanother grain size group.

407 For magnetic separation, differences were observed between the different grain sizes processed. The 408 smallest grains sizes tended to clog the system more often than the larger ones, and different grain 409 sizes sometimes led to different yields. However, this latter observation requires confirmation from a 410 larger number of data points as well as repetitions for quantitative results. The overall yields for each 411 simulant and all grain sizes recombined (500-63 µm) are shown in Table 6 in supplemental material. 412 Due to cleaning and unblocking the system for some sample groups it was necessary to rerun these 413 groups up to three times. This added to the long processing times (4-5 hours each run) which were 414 inherent to the utilised system. Further, the long processing time and small amounts of about 100 415 grams of regolith which could only be processed in a reasonable time led to, in some cases, less than 416 1 gram of sample in a specific group. Also apparent from Table 6, all mare simulants have comparably 417 low yields in the non-magnetic ("non") area. It is further apparent that, the sample with the highest 418 yield in the non-magnetic group is the highland simulant LHT-3M. Looking at the manufacturer's data 419 in Table 5 (supplemental material) and the obtained XRF measurements in Table 4, LHT-3M also has 420 the lowest iron oxide content of all simulants. Although the yields in Table 6 may suggest that more 421 iron oxide sparse input material seems to produce more non-magnetic material, which can be used 422 for glass manufacturing, it is important to consider that these oxides are contained in minerals rather 423 than in individual oxide form. Hence, the mineral composition of an input material will be the driver 424 for how well a regolith (simulant) separates. With respect to manufacturing glass from iron free/iron 425 sparse material, the method used works but for most simulants processed, a starting amount of 100 426 grams was not enough to separate a sufficient quantity of material to manufacture a sheet of glass. 427 Typically, about 10 grams of processed material was enough to manufacture a glass sheet of 30*20*1 428 mm, and only LHT-3M provided this amount. This was after recombination of the three different grain 429 size groups back to 30 samples. This was also necessary to obtain enough material to be able to obtain XRF data for all 30 samples. 430

Although, beneficiation was successful, as shown in the discussion of the XRF results later, it remains to be tested whether the process would be feasible in a lunar environment with reduced gravity, which will impact transport of the grains during processing, and at temperatures as low as < -100 °C [57] [58] [59], where magnetic susceptibility of minerals may change [60] [61]. Further, it seems viable to automate this process and to reduce power consumption by possibly utilising only permanent magnets instead of electromagnets, both to increase mission capabilities further.

437 Hot processing

While melting and casting LHT-3M-non and JSC-2A-non it was noticed that JSC-2A-non had a slight blue tint compared to the greenish taint of the LHT-3M-non sample. Glass colouration is complex, so it is hard to tell what exactly causes a certain colouration without measuring the trace element amounts as well. Unfortunately, such measurements require about 3-4 grams of sample which were not available, and therefore, not conducted.

Annealing parameters were determined empirically, to work for all regolith simulants at the same time. Since annealing temperature is related to the glass transition temperature of a regolith/glass, energy saving potential exist for future experiments and applications. Via optimisation and tailoring of the process temperature to the actual regolith glass transition temperature (e.g. 620 °C for JSC-2A [62]) and composition this potential can be explored in the future. For this experiment run the process efficiency was of secondary importance and sample quality had priority.

449 Carbon contamination

450 For manufacturing of the contaminated samples shown in Figure 6 the heating time was reduced to 3 451 minutes (compared to 15 for LHT-3M-non) since their mass was smaller (< 6 g) compared to the 40 452 grams of LHT-3M non-magnetic material. The further intention was to keep the reaction time between 453 graphite and the sample material to a minimum. However, despite these measures, the process 454 resulted in variable carbon contamination of the glasses. Carbon contamination is clearly visible in 455 Figure 6 as black streaks or lines and blocks most of the light from being transmitted. However, 456 transparent spots are visible in-between the graphite streaks. This suggests that it should be possible 457 to also manufacture transparent glass from other regolith compositions than LHT-3M, a highland 458 simulant. The fact that carbon contamination can still be seen after polishing of the samples suggests 459 that it is not simply surface contamination. Hence, carbon likely penetrated the glass entirely during 460 heating in the furnace. Presence of a free carbon phase likely means that the samples are fairly 461 reduced, as carbon will readily react with oxygen in the air to produce CO₂ and CO. Furthermore, this 462 also implies that graphite crucibles cannot readily be used for heating/processing samples at high 463 temperatures since the crucible will interact with the sample in the form of a carbothermal reduction. 464 An exception can be made for using graphite as moulds for casting, since temperature drops rapidly 465 during casting which does not allow for any reduction to take place.

Glass colour is different for all of these samples. This is likely a result of trace elements being contained
in the samples which give the glasses their green, brown, and blue colours. To be able to tell which
elements are responsible for the colouration, enough sample for a trace element analysis would need
to be collected. However, it is likely that green-brown colouration arises due to the presence of iron,

as commonly noted in terrestrial basaltic glasses. Green colouration due to the presence of Ti³⁺ is,
however, noted in reduced clinopyroxene from the Allende meteorite [63]. A blue colouration may
be due to the presence of Ti, especially in its reduced form Ti³⁺. Blue colouration due to Ti-Ti
intervalence charge transfer, or due to Ti³⁺ colour centres, is commonly noted in other material [64].

474 Cold processing

Although cold processing led to suitable glass samples for analysis, it is noted that the described cold processing process utilised a number of consumables not readily available on the Moon. Thus, alternative hot manufacturing processes may be utilised in the future to obtain an ideal surface in one shot. Terrestrially this can be observed in the float glass process for example. Alternatively, or additionally locally available materials (regolith) may be used as grinding, lapping and polishing agent instead of the described materials.

481 Analysis of effects of magnetic beneficiation

482 This section discusses the results of XRD/SEM, XRF, surface and optical analysis.

483 Discussion of magnetic beneficiation using XRF, XRD and SEM results

484 Prior to the discussion it was pointed out that the LOIs during XRF sample preparation, listed in Table 485 7 in supplemental material, show especially high (more than 10 % and in one case (EAC-1, non) in excess of 20 %) losses for samples labelled with "-non", the least magnetic samples. High LOI values 486 487 in a sample could be derived from water in the sample (either adsorbed or structural water in minerals 488 such micas and amphiboles), or contaminants, such as polymers from packaging and/or bottling, as 489 well as contained organic materials. Water is however, not very likely since samples have been dried 490 prior to processing. The other materials are typically not magnetic and will collect in the non-magnetic 491 group and lead to the weight loss during sample preparation. Although, on the lunar surface organic 492 contamination of any kind are unknown so far, foreign materials entering the regolith concentrate 493 during processing, from the process itself may need to be accounted for when utilising magnetic 494 separation.

495 XRD and SEM data confirmed the occurrence of pyroxene, feldspar, olivine, alteration phases and 496 oxide minerals in all simulants and the likely occurrence of larger amounts (>10wt%) glass in EAC-1 497 and JSC-2A. Respectively, it is very likely that JSC-1A also contains a similar amount of glass and other 498 minerals since these two simulants are very similar by design. From this list of minerals, looking at the 499 oxide content obtained by means of XRF conclusion can be drawn from the oxide content changes 500 with respect to the changes in the minerals content of a regolith simulant. 501 Prior to discussing changes caused by the beneficiation process, the manufacturer values are 502 compared with respect to the obtained raw, unaltered regolith XRF measurement results. 503 Manufacturer values of simulants supposed oxide composition are provided in Table 5 in 504 supplemental material. In there, the simulant with the initially highest amount of iron oxide is FJS-1 505 (13.1 wt%) and the one with the lowest LHT-3M (4.2 wt%). However, the actual measurements of the 506 raw regolith, shown in the according first line for ach simulant in Table 4, show that JSC-2A had the 507 highest iron oxide content with 13.18 wt% and FJS-1 only contained 12.9 wt%. Further, LHT-3M still 508 showed the lowest initial iron oxide content but with 5.56 wt% rather than 4.2 wt%. The example of 509 iron oxide content is only one where oxide values provided by the manufactures differ significantly 510 from the actual values of the shipped material. Since standards and quality control are not yet 511 established in the field of regolith simulants, it will therefore always be necessary to determine the 512 geochemical composition of a newly shipped batch, prior to utilisation, to allow to draw conclusion 513 and to establish comparability between scientific works.

514 To gain a better overview of the effect the magnetic beneficiation had on the regolith simulants, 515 values from Table 4 were taken and translated in the form of stacked bar charts depicted in Figure 10. 516 The values depicted have been corrected for the LOI and rescaled to 100%. Since the SiO_2 content of 517 all samples was always higher than 40%, the first 40% of the samples' compositions are not depicted. 518 Looking at both, Figure 10 and Table 4, differences in magnetic separation of different simulants can 519 be seen. Precise percentage values are listed in Table 4 and the following will discuss rounded values 520 from Figure 10 with the aim of identifying trends focusing on samples that experiences the most 521 drastic changes.

522 Although it was not possible to conduct a detailed mineralogical analysis of the altered samples, due 523 to insufficient amounts of samples, the XRF results presented in Table 4 and in Figure 10 provide 524 insight into which minerals may have been removed in the process. Seen from XRD results in Table 2, 525 in general, all mare simulants utilised consist of mostly pyroxene minerals (both ortho- and clino-), 526 plagioclase feldspar and olivine. Thus, for example, a reduction in iron (titanium) content likely 527 coincides with the removal of the more iron (and possibly titanium) bearing minerals like pyroxenes 528 and/or olivine, in addition to minor/accessory minerals such as spinels (e.g. titanomagnetite and 529 magnetite) and hematite.

In general, after magnetic beneficiation the iron oxide content of the simulants at the "non" level (the
least magnetic samples) could be reduced to less than 2 wt%, with LHT-3M reaching a value as low as
0.72 wt%. Thus, it can generally be assumed that magnetic beneficiation was working. Whether this
can be true for actual lunar regolith will be discussed after detailed discussion of the effects of

- 534 magnetic beneficiation onto the six selected simulant samples. As a reminder, measurements
- 535 displayed in Table 4 and in Figure 10 were always taken from the tailings, the iron enriched samples
- of each step and only in the end ("non" samples) taken from the concentrate.
- 537 Looking at each regolith simulant and the changes over the course of magnetic beneficiation, taking
- 538 into account XRD results in Table 2 and XRF results from Table 4, changes of each individual simulant
- are described next:



Figure 10 Oxides contained in samples (from top left to bottom right BP-1, EAC-1, FJS-1, JSC-1A, JSC-2A and LHT-3M.
Comparison of different stages of magnetic separation: first bar in each chart (unaltered regolith), "-hm", "-high", "-fair", "low", "-non"; after unaltered regolith, in decreasing order of magnetic susceptibility. Values below 40 wt% are all SiO2 content.

545 BP-1: From the hand magnet (hm) sample to the "non" sample, iron content gradually decreases from 546 about 14 % to less than 2 %, a relative reduction of more than 80 % compared to the raw regolith with 547 about 12%. Furthermore, MgO and TiO₂ reduced to almost 1 % from >6 % (MgO) and >2% (TiO₂), and 548 in a similar fashion the iron oxide content from 2.1 % to 0.2 %. On the contrary Al_2O_3 and CaO relative fractions increased in "B-non" to 19 % (from about 16 %) for Al₂O₃ and to 19 % (from about 10 %) for 549 550 CaO. These changes may indicate that B-non mostly consists of anorthite-rich plagioclase feldspar. 551 Coupling of changes in Fe_2O_3 and TiO_2 likely indicate that both are enriched in certain phases. This 552 could be in pyroxene or in smaller amounts of accessory Fe-Ti oxides.

553 EAC-1: Similar BP-1, the E-non sample iron oxide content was reduced to almost 2 % compared to the 554 raw regolith of approx. 12%. The two groups showing the most drastic changes are E-fair and E-non. 555 The sample E-fair showed significant relative reductions (shown in brackets) compared to the 556 unaltered bulk for Al₂O₃ (78%), CaO (70%), Na₂O (99%) and K₂O (77%). This likely corresponds to almost 557 complete absence of any feldspar minerals in this sample. On the other hand, iron oxide content as 558 well as MgO of this sample is the highest for all EAC-1 samples analysed, which suggests a large 559 amount of olivine and smaller amounts of pyroxene to be present. For the E-non sample the TiO₂ 560 content was 0.13 % (94%). This in combination with changes in other elements suggest low olivine 561 content, high feldspar content and medium pyroxene content.

FJS-1: For FJS-1, the magnetic separation also worked and gradually lead to a reduction of iron oxide
 content to less than 2 % in sample F-non, compared to almost 13 % in raw FJS-1. This suggests low to
 no Fe-bearing pyroxenes and olivine to be present after processing. Further the 29 % of Al₂O₃ suggest
 that F-non is mostly comprised of feldspar minerals.

566 JSC-1A & JSC-2A: Due to JSC-2A intentionally mimicking JSC-1A the composition of these two simulants 567 is similar and are discussed in parallel. The behaviour of those two simulants with respect to magnetic 568 treatment further is also very similar. Similar to the regoliths already described, the Fe-containing 569 phases drop from an initial proportion of approx. 13% to almost 2% in J1-non and J2-non. For both 570 simulants the overall oxide content of the sample did not change drastically from J1/J2-hm to J1/J2-571 fair and for JSC-2A even until J2-low. Only significant changes to be noticed were in J1-low and J1/J2-572 non. Compared to the unaltered regolith, for the "-non" samples SiO_2 (~-15/-18%), Fe_2O_3 (~-90/-93%), 573 MgO(~-88/-89%), Na₂O(~-33/-37%), K₂O(~-80/-64%) and TiO₂(~-90/-93%) decreased relatively (J1/J2) 574 and Al_2O_3 (~+32/+14%) and CaO(~+131/+152%) increased. This suggest that J1/J2-non may be very 575 plagioclase (anorthite) rich since MgO and Fe_2O_3 are at almost zero percent for these samples.

576 **LHT-3M:** The only highland sample, based on plutonic rather than igneous rock started with a lower 577 proportion of iron oxides than the other regoliths. However, this was reduced to less than 1% in 578 sample L-non. Other differences include an increased amount of iron and magnesium oxide in the L-579 high sample, while at the same time reduced calcium and aluminium oxide content. This suggests an 580 elevated pyroxene content. Further, for L-non almost no Fe₂O₃ or TiO₂, and no MgO was detected at 581 all (n.d.). This suggests that L-non is comprised of almost exclusively plagioclase feldspars with a high 582 amount of anorthosite.

583 From all observations, it seems apparent that it was possible to remove iron-bearing minerals, as 584 targeted for glass production. However, since the change of all simulants' oxide compositions' is 585 different from simulant to simulant, the way magnetically processing might alter regolith on the Moon 586 is expected to be dependent on material composition. For future experiments it will be crucial to 587 measure and observe mineralogy of a simulant in addition to the oxide content as done for this 588 experiment. Since elements are typically contained in certain minerals, magnetic separation separates 589 minerals from each other rather than individual oxides. If the mineral composition of a regolith is 590 known, it should be possible to target and mostly remove certain minerals. Importantly, processing 591 indicates that for many simulants there are coupled changes in Fe_2O_3 and TiO_2 . This might indicate 592 that a proportion of the iron in most simulants is present in mixed Fe-Ti-rich phases, facilitating 593 beneficiation. However, although it is possible in theory to preferentially remove certain phases, 594 practically this is complicated by that fact that regoliths are typically composed of igneous rock 595 fragments. Hence, individual regolith grains are most often composed of a multitude of different 596 minerals "glued together", which makes it unlikely that they can be fully separated in practice. 597 Furthermore, the presence of inclusions of Fe-rich oxides in some phases, or the preferential 598 association of oxides with certain phases, can result in variations in the extent to which different 599 phases, and overall composition, is modified during magnetic separation. As such, not just the mineral 600 proportions, but also texture and mineral associations are also important to determine, along with 601 the particulate size distribution. Adaption of any magnetic beneficiating process on the lunar surface 602 would require, therefore, a much fuller understanding of the composition, size distributions, phase 603 relations and textural relationships within lunar regolith.

Although, it was not within the scope of this project to increase or concentrate iron oxide content, the method of magnetic separation can also be used to enrich a samples iron oxide content. An additional single test conducted on LHT-3M, aiming at increasing the amount of iron oxide by repeatedly using a hand magnet on it, delivered a sample with an iron oxide content of almost 60 % (59.03 %) after correcting for the LOI. LHT-3M was likely a good simulant for both, iron oxide reduction and increase, due to most of the iron oxide being contained in individual minerals, rather than in igneous rocks. This made the iron bearing minerals accessible to the magnet and led to a good removal rate. 611 How this process may be applicable to actual regolith remains to be tested. Actual lunar regolith will 612 contain iron oxide as well but it may be part of different minerals and/or present as nano phase iron 613 coatings on the regolith's grains surfaces. Furthermore, environmental conditions, especially low 614 temperatures, may be challenging when relying on magnetic susceptibility. Although it cannot be 615 excluded after this study, it seems likely that (other) iron bearing minerals are also magnetically 616 attracted on the lunar surface. Due to the low gravity environment (1/6th of Earth's gravity) susceptibility on the surface may even been better. However, only testing with actual regolith will 617 618 provide more reliable results whether or not magnetic separation works on actual lunar regolith 619 minerals. Further, with respect to the nano phase iron coating of the grains, magnetic beneficiation 620 as conducted for this study does not seem to be an efficient technique. Instead, attrition grinding may 621 be used to remove these fine dusts and thereby decreasing the overall amount of iron contained in 622 the samples even further. However, this will first need to be tested on suitable regolith simulants and 623 then on actual lunar regolith to provide more reliable data. Last but not least, environmental testing 624 of the described magnetic beneficiation process will need to be conducted to better understand the 625 impact of low temperature low gravitation environment on the process.

626 Analysis of Optical Properties of Regoliths Simulant Glass

627 Mechanical properties

628 After mineral separation and determining changes in oxide/mineral content, LHT-3M samples with 629 the lowest iron oxide contents were used to manufacture glass. As described in the methods sections, 630 flat, parallel, glass sheets were manufactured to, ultimately, assess transparency. Prior to optical 631 measurements of the samples, sample mechanical properties were measured to determine the effect 632 of sample thickness and surface roughness on transparency. To achieve highest transparency, an 'as 633 smooth as possible' surface was desired to avoid light scattering. Although both LHT-3M samples 634 analysed (Figure 8) show higher deviations than the reference microscopy slide, they were both 635 deemed acceptable at a roughness of about 35 nm, compared to about 20 nm of the microscopy slide. 636 Differences between the sample's parallelism are listed in Table 9 and show that the microscopy slide 637 used showed differences in thicknesses of maximum 0.004 mm between its corners. The LHT-3M-1 and LHT-3M-2 sample showed maximum differences of 0.034 and 0.021 mm between their corners. 638 639 Considering that the LHT-3M samples were about half as long as the microscopy slide this leads to an 640 about one order of magnitude difference between the parallelism of the microscopy slide and the LHT-3M-non glass samples. At this level of parallelism, all transparent glass samples were deemed 641 642 good enough for optical analysis.

643 Optical

The microscopy slide (marked "Micro" in Figure 9 and Figure 10) showed the overall highest reflectivity measurements, which seems consistent with the surface roughness measurements, showing that the microscopy slide had the smoothest surface compared to LHT-3M-1/2. Compared to the microscopy slide, the two LHT-3M samples (-1 and -2) show a lower reflectivity by about 2-4 % absolute difference, less in the infrared range and more in the UV range (details Figure 10). Further, the contaminated samples (BP-1, FJS-1 and JSC-2A) show similar reflectivity as the other samples, all in the range from about 4% to 8% reflectivity.

651 Transmission results (shown in figure 9) reveal much larger differences between samples than 652 reflectivity. As expected, the microscopy slide shows the highest transmission of all samples. Although, 653 the LHT-3M-3 sample is within only a few percent less reflectivity and next, offset by about 20 % 654 (absolute), the LHT-3M-1/2 samples achieve transmissions of between 60-80 % for most of the 655 measured spectrum. Important to notice, since not clearly visible in Figure 9, the lines of samples LHT-656 3M-1 and LHT-3M-2 coincide. This is expected due to their very similar geometry/thickness and same 657 geochemical composition. The contaminated samples showed, despite their carbon cloudiness, 658 transmissions between 0 and 15 %. In comparison, other non-beneficiated samples have shown 0% 659 transmission over the entire wavelength range in tests conducted outside this work.

660 Considering the spectral irradiance in space, in the form of the AMO spectrum depicted in orange in 661 Figure 9, the glasses manufactured from lunar regolith simulant allow most of the energy from the 662 sun to pass the glass in the area of high spectral irradiance. Around 450 nm, the LHT-3M samples show 663 transmission of about 80 % (LHT-3M-3) or 60 % (LHT-3M-1/2). Based on these results, it is fair to say 664 that these glasses are optically transparent and with further improvements to the beneficiation 665 process it may even be possible to achieve similar transparency to terrestrial window or cover glass.

Listed in Table 8 in the supplementary material are average transmission versus average thicknesses of the samples. The three samples made of beneficiated LHT-3M show average transmissions of above 60% with the most transparent sample, LHT-3M-3, achieving results of >80% average transmission. This brings the LHT-3M-3 sample within a range of only 9% of the reference microscopy slide. Since the LHT-3M-3 sample is only almost half as thick (Table 8 and Table 9) as the LHT-3M-1/2 samples, it seems logical, that it would also allow for the transmission of more light. The increased transmission for thinner samples is thus likely connected to less iron blocking light during transit through the glass.

673 Considering that all LHT-3M samples' transmissions start dropping at about 550 nm compared to the 674 microscopy slide, the average transmission from 550 to 1250 nm is even closer to the microscopy slide 675 in that range than on average. The drop of the samples in this range may be explained by their vestigial 676 iron content, which results in blocking of shorter wavelength light. This suggests that if the iron content can be manipulated accordingly, glass made from regolith is not only transparent but may be
engineered to act as a filter over a certain wavelength range. This could be beneficial for applications
which require little to no light in the wavelength range <550 nm.

From these discoveries, it seems viable to engineer glass from lunar regolith with just the right amount of iron, to be utilised, for example, as cover glass for a solar cell on the lunar surface. Other potential applications include optical filters, windows, building materials, (optical) glass fibres, backplates for mirrors or solar cells and more.

684 Conclusion

The results presented in this work successfully show that it is feasible to use lunar in-situ resources to manufacture transparent glass, and open up new possibilities for supporting planetary surface missions to the Moon. Manufactured transparent glass from lunar regolith could be the first raw material which can viably be manufactured on the lunar surface. Having a supply of a raw material as versatile as glass will significantly boost the goal of permanently becoming a multiplanetary species.

- Optical analysis of transparent glasses manufactured from lunar regolith simulant show, that they are suitable as cover glass for solar cells or general-purpose windows, for example. With these first samples, average transmissions of 80 % were achieved, which was close to the 89% of the reference glass sample. With further optimisation to the beneficiation and manufacturing processes it seems possible to even increase the amount of transmitted light further.
- 695 Next steps on the way to increasing the technology readiness level (TRL) of transparent glass 696 manufacturing on the lunar surface should include, increasing the amount of regolith processed, 697 improving/tailoring magnetic beneficiation and utilising actual lunar regolith for testing. Further, 698 automating processes as well as environmental testing, such as thermal vacuum tests, will be required. 699 The latter is especially important to test the magnetic susceptibility of minerals at low temperatures, 700 as well as the effect of more reducing atmospheric conditions on glass transparency. Additionally, low 701 gravity testing in the form of a parabola flight or an experiment on the international space station will 702 help to understand process parameters in a low g-environment. Improvements to the magnetic 703 beneficiation process should also, ultimately, make it possible to target specific minerals only and/or 704 reduce elements such as iron or other glass colouration elements further. This would enable further 705 engineering of glass' colour/optical properties, and may allow for glass to be tailored to specific 706 applications.

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715 Compliance with ethical standards

- 716 Conflict of interest: Juergen Schleppi is an employee at Maana Electric. The other authors declare
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Supplementary material

Table 5 Oxide composition of the utilised simulant, manufacturer information in weight % (wt%). All iron contained iscollectively listed under Fe_2O_3 . No further information on Loss of Ignition (LOI) or more detailed information were provided.

Simulant	SiO ₂	Al ₂ O ₃	Fe ₂ O ₃	MgO	CaO	Na₂O	K ₂ O	TiO ₂	MnO	P ₂ O ₅
	43.0-	16.4-	9-	5.6-	9.2-					
BP-1	47.2	18.0	11.7	10.0	14.0	3.45	0-1.1	-	-	-
EAC-1	43.7	12.6	12.0	11.9	10.8	2.9	1.3	2.4	0.2	0.6
FJS-1	49.1	16.2	13.1	3.8	9.1	2.8	1.0	1.9	0.2	0.4
		14.5-	10-	8.5-			0.75-		0.15-	0.6-
JSC-1A	46-49	15.5	11.5	9.5	10-11	2.5-3	0.85	1-2	0.20	0.7
		14.5-	10-	8.5-			0.75-		0.15-	0.6-
JSC-2A	46-49	15.5	11.5	9.5	10-11	2.5-3	0.85	1-2	0.20	0.7
LHT-3M	46.7	24.4	4.2	7.9	13.6	1.3	0.1	0.4	0.1	0.2

Table 6 Yields for magnetic sample separation, shown in grams. Out of initially 200 grams unaltered regolith simulant only the grainsize fraction of 500 to 63 µm was processed and the amounts left after sieving are listed under "start". Each simulant type was split into 5 groups, "HM" (hand magnet), "high", "fair", "low" and "non" with descending order of magnetic susceptibility. The measured samples were taken from the tailings (the material with higher magnetic susceptibility after each run. Under "loss" the amount of material lost during processing is listed.

500-63	start	НМ	high	fair	low	non	loss
BP-1	123.4	62.0	42.0	7.6	2.7	3.1	6.0
EAC-1	132.8	96.9	22.9	9.7	1.3	0.4	1.6
FJS-1	96.2	69.0	6.2	3.9	8.5	5.6	3.0
JSC-1A	114.3	25.1	49.8	33.9	2.6	0.5	2.4
JSC-2A	113.6	24.9	63.7	22.9	0.3	0.3	1.5
LHT3M	129.7	2.7	17.1	50.3	14.2	44.4	1.0

Table 7 Loss on ignition of samples magnetically altered, in wt%

Sample	BP-1	B-hm	B-high	B-fair	B-low	B-non
LOI	2.04	0.88	2.22	3.76	9.11	9.80
Sample	EAC-1	E-hm	E-high	E-fair	E-low	E-non
LOI	1.40	1.19	1.74	1.08	6.05	20.31
Sample	FJS-1	F-hm	F-high	F-fair	F-low	F-non
LOI	-0.25	-0.19	-0.16	-0.37	-0.04	0.22

Sample	JSC1A	J1-hm	J1-high	J1-fair	J1-low	J1-non
LOI	-0.46	-0.46	-0.57	-0.67	-0.20	11.11
Sample	JSC-2A	J2-hm	J2-high	J2-fair	J2-low	J2-non
LOI	-0.79	-0.65	-0.64	-0.70	-0.46	14.05
Sample	LHT-3M	L-hm	L-high	L-fair	L-low	L-non
LOI	0.12	-0.36	2.22	0.21	0.26	0.25

Table 8 Average transmission and thickness of transparent glasses. Thickness is shown in mm.

Sample	BP-1	FJS-1	JSC-2A	LHT-3M-1	LHT-3M-2	LHT-3M-3	Micro
Avg. trans.	10.72%	4.07%	7.17%	60.86%	60.57%	80.58%	89.01%
Avg. thick.	1.336	1.321	1.341	1.400	1.385	0.838	1.075

Table 9 Dimensions of transparent glasses all samples 3*LHT-3M, BP-1, FJS-1, JSC-1 and microscopy slide ("Micro") are listed in mm. Show are length, width, thickness at all four corners and average thickness (t_{avg}) of the samples.

Sample	Length	Width	t _{LF}	\mathbf{t}_{LB}	t _{RF}	t _{RB}	t _{avg}
BP-1	24.2	18.1	1.353	1.346	1.323	1.320	1.336
FJS-1	33.7	21.1	1.314	1.343	1.303	1.322	1.321
JSC-2A	26.7	15.0	1.345	1.353	1.336	1.330	1.341
LHT-3M-1	34.5	19.0	1.391	1.391	1.412	1.404	1.400
LHT-3M-2	29.8	19.8	1.396	1.370	1.403	1.369	1.385
LHT-3M-3	18.9	19.6	0.833	0.838	0.839	0.843	0.838
Micro	75.8	21.9	1.077	1.073	1.074	1.075	1.075

L: left, R: right, B: back, F: front, avg: average