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Hinge and overturning moments due to unsteady heliostat pressure distributions in a turbulent atmospheric boundary layer

Matthew J. Emes*, Azadeh Jafari, Farzin Ghanadi, Maziar Arjomandi

Centre for Energy Technology, School of Mechanical Engineering, The University of Adelaide, SA 5005, Australia

Abstract

Non-uniform pressure distributions on the heliostat surface due to turbulence in the atmospheric boundary layer (ABL) have a significant impact on the maximum bending moments about the hinge of and pedestal base of a conventional pedestal-mounted heliostat. This paper correlates the movement of the centre of pressure due to the mean and peak pressure distributions with the hinge and overturning moment coefficients using high-frequency pressure and force measurements on a scale-model heliostat within two simulated ABLs generated in a wind tunnel. The positions of the centre of pressure were calculated for a range of heliostat elevation-azimuth configurations using a similar analogy to those in ASCE 7-02 for monoslope-roof buildings, ASCE 7-16 for rooftop solar panels, and in the literature on flat plates. It was found that the maximum hinge moment is strongly correlated to the centre of pressure movement from the heliostat central elevation axis. Application of stow and operating load coefficients to a full-scale 36 m² heliostat showed that the maximum hinge moment remains below the stow hinge moment at maximum operating design gust wind speeds of 29 m/s in a suburban terrain and 33 m/s in a desert terrain. The operating hinge moments at elevation angles above 45° are less than 60% of the stow loads with a constant 40 m/s design wind speed. The results in the current study can be used to determine heliostat configurations and appropriate design wind speeds in different terrains leading to the maximum design wind loads on the elevation drive and foundation.

Keywords: heliostat; hinge moment; overturning moment; pressure distribution; centre of pressure; turbulence

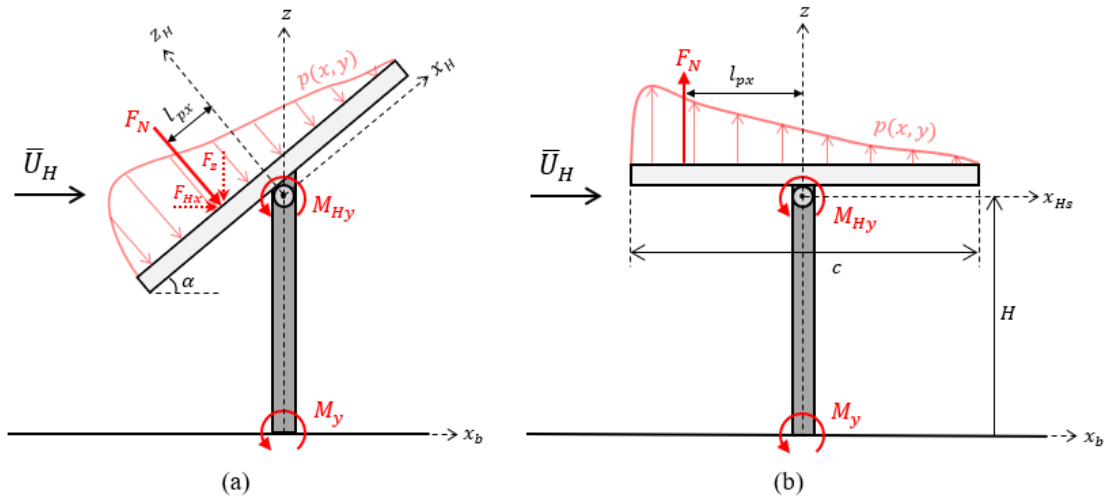
26 Nomenclature

27	A	Helio­stat mirror area (m ²)
28	$\alpha_{\bar{u}}$	Power law velocity profile exponent
29	α	Elevation angle of helio­stat mirror plane with respect to the horizontal (°)
30	β	Azimuth angle of wind with respect to frontal projected helio­stat mirror plane (°)
31	c	Helio­stat mirror chord length (m)
32	c_{Fx}	Drag force coefficient
33	c_{Fz}	Lift force coefficient
34	$c_{M_{Hx}}$	Hinge moment coefficient about the axis x_H
35	$c_{M_{Hy}}$	Hinge moment coefficient about the elevation axis y_H
36	c_{M_y}	Overtur­ning moment coefficient about the axis y_b at the base of the helio­stat pylon
37	C_p	Pressure coefficient
38	F_N	Net force normal to the helio­stat surface (N)
39	F_{Hx}	Drag force on the helio­stat surface (N)
40	F_z	Lift force on the helio­stat surface (N)
41	G_u	Velocity gust factor
42	H	Elevation axis height of helio­stat hinge above the ground (m)
43	I_u	Longitudinal turbulence intensity (%)
44	I_w	Vertical turbulence intensity (%)
45	l_p	Distance to the centre of pressure from the centre of the helio­stat mirror plane (m)
46	l_{px}	Longitudinal distance to the centre of pressure from the y -axis of the helio­stat (m)
47	l_{py}	Lateral distance to the centre of pressure from the x -axis of the helio­stat (m)
48	L_u^x	Longitudinal integral length scale (m)
49	L_w^x	Vertical integral length scale (m)
50	M_{Hx}	Hinge moment on helio­stat about the axis x_H (N·m)
51	M_{Hy}	Hinge moment on helio­stat about the elevation axis y_H (N·m)
52	M_y	Overtur­ning moment about the axis y_b at the base of the helio­stat pylon (N·m)
53	ρ	Air density (kg/m ³)
54	p	Differential pressure between the upper and lower helio­stat surfaces (Pa)
55	P_i^f	Pressure fluctuations on the upper helio­stat surface (Pa)
56	P_i^b	Pressure fluctuations on the lower helio­stat surface (Pa)
57	$\bar{U}(z)$	Mean velocity profile (m/s)
58	\bar{U}_H	Mean velocity at helio­stat elevation axis height (m/s)
59	\hat{U}_{op}	Operating gust wind speed at a 10-m height for helio­stat design (m/s)
60	\hat{U}_{st}	Stow gust wind speed at a 10-m height for helio­stat design (m/s)
61	x	Dimension parallel to the elevation axis on the helio­stat mirror plane (m)
62	x_H	Longitudinal axis at the helio­stat hinge height (m)
63	x_b	Longitudinal axis at the base of the helio­stat pylon (m)
64	y	Dimension perpendicular to the elevation axis on the helio­stat mirror plane (m)

65	y_H	Lateral axis at the heliostat hinge height (m)
66	y_b	Lateral axis at the base of the heliostat pylon (m)
67	z	Vertical azimuth axis of heliostat (m)
68	z_0	Surface roughness height of logarithmic velocity profile (m)

69 **1. Introduction**

70 The development of concentrating solar thermal (CST) as an emerging renewable technology in
71 recent decades has been accompanied by the increased deployment of large-scale power tower (PT)
72 plants. A large number of heliostats are required to achieve the high temperatures and power cycle
73 efficiencies in a central receiver PT plant (IEA-ETSAP and IRENA 2013), such that the heliostat field
74 represents the largest contribution of almost half of the plant's total cost (Kolb *et al.* 2011; Pfahl *et al.*
75 2017b). The economic viability of PT systems relies on the reduction of the overall heliostat cost per
76 unit area. For instance, lowering the strength and stiffness requirements, following a three-halves power
77 law with the heliostat area (Kolb *et al.* 2011), can be achieved through the manufacturing of lighter
78 wind-sensitive components of heliostats (Télliez *et al.* 2014; Emes *et al.* 2015). The elevation and
79 azimuth drives, pedestal, foundation and mirror support structure of a conventional elevation-azimuth
80 heliostat account for up to 80% of the heliostat capital cost (Kolb *et al.* 2011). These costs can be most
81 effectively reduced with an accurate estimation of the wind loading on a heliostat to maintain the
82 structural integrity during high wind periods while achieving good optical performance during operation
83 of the field (Pfahl *et al.* 2017a). Heliostats are designed to maintain structural stiffness during operation
84 (Figure 1a) at different elevation angles ($\alpha > 0^\circ$) for maximum optical accuracy. Furthermore, they
85 require the structural strength to withstand the maximum loads during high-wind conditions when
86 aligned parallel to the ground ($\alpha = 0^\circ$) in the stow position (Figure 1b). The design wind loads on
87 heliostats are commonly defined using a combination of non-dimensional peak load coefficients that
88 account for the turbulence in the wind and the mean wind speed \bar{U}_H at the elevation axis height (H in
89 Figure 1) above the ground.



90

91 Figure 1. Wind loads on a heliostat due to a non-uniform pressure distribution $p(x, y)$ on the heliostat mirror
 92 plane caused by atmospheric turbulence in: (a) operating positions $\alpha > 0^\circ$; (b) stow position $\alpha = 0^\circ$.
 93 Reproduced from Emes *et al.* (2019). Positive values of the hinge M_{Hy} and overturning M_y moments are defined
 94 by anti-clockwise rotations about the elevation axis y_H and the y -axis of the pylon base y_b , respectively.
 95

96 The design method for heliostat wind loads (Peterka and Derickson 1992) outlines the critical
 97 configurations of the elevation angle α of a square heliostat mirror (chord length c in Figure 1) relative
 98 to the horizontal, and the azimuth angle β of the frontal projected heliostat mirror plane relative to the
 99 wind direction. Peterka and Derickson (1992) reported the azimuth-elevation configurations (α, β) for
 100 the most unfavourable working conditions represented by the maximum values of the peak coefficient
 101 of the drag force F_x in the horizontal x direction, lift force F_z in the vertical z direction, hinge moment
 102 M_{Hy} about the central elevation axis and overturning moment M_y about the foundation at the base of
 103 the steel pedestal of a conventional heliostat. The forces and moments were calculated from high-
 104 frequency measurements using strain gauges mounted on a square-facet heliostat model ($c = 0.27$ m,
 105 $H = 0.13$ m) and in the base of a force balance in a wind tunnel with test section of 2.13 m height, 1.8
 106 m width and 18.29 m length (Peterka *et al.* 1988; Peterka *et al.* 1989). Aerodynamic force and moment
 107 coefficients used in heliostat design are therefore commonly calculated using the mean wind speed at
 108 the elevation axis height. This follows a quasi-steady approximation that the ratio of the peak and mean
 109 forces are proportional to the square of the velocity gust factor defined by the ratio of the 3-second gust
 110 wind speed to the mean wind speed (Peterka and Derickson 1992; Mendis *et al.* 2007). The gust factor
 111 method can provide reliable estimations of the wind loads on physical structures with standard

112 geometries, such as low-rise buildings with monoslope roofs (ASCE 7-02 2002; AS/NZS 1170.2 2011).
113 However, the aerodynamic mechanisms such as corner vortices generated by buildings have been
114 shown to cause significant differences between the loading on roof-mounted and ground-mounted solar
115 panels (Kopp *et al.* 2012). Furthermore the quasi-steady assumption can under-estimate the load
116 predictions on small physical structures with non-standard geometries, such as the hinge moments on
117 stowed heliostats (Ghanadi *et al.* 2017) due to the large amplitude fluctuations during high-wind events
118 caused by gusts over short time intervals (Durst 1960; Mendis *et al.* 2007).

119 The design of the load-bearing heliostat components, such as the drive units, pedestal and
120 foundation, requires the distribution of the loads over the mirror to be accurately estimated in operating
121 and stow positions. Eddies embedded in the turbulence lead to fluctuations in the wind velocity and
122 direction, resulting in a non-uniform pressure distribution $p(x, y)$ on the heliostat mirror that varies
123 both temporally and spatially. The temporal distribution of the surface pressure is represented by the
124 mean, root-mean-square (RMS) and peak pressure coefficients, whereas the position of the net force on
125 the heliostat surface due to a non-uniform pressure distribution is given by the centre of pressure. Wind
126 codes and standards for open buildings provide recommendations for the maximum movement of the
127 centre of pressure from the leading edge of the roof surface as a function of the aspect ratio of the roof
128 dimensions and the inclination angle. For example, the centre of pressure distance from the windward
129 edge of a square-cross-section monoslope roof is given in ASCE 7-02 (2002) as 30% of the roof
130 dimension parallel to the wind direction over the 10-20° range of inclination angles, and 40% for a roof
131 inclination angle of 30°. In contrast, EN 1991-1.4 (2010) recommends the maximum movement of the
132 centre of pressure from the centre of a flexible, centrally-supported plate-like structure, such as a
133 signboard, to be less than 25% of the plate chord length (c) normal to the wind direction. Although a
134 square cross-section signboard separated from the ground by a height greater than $c/4$ closely represents
135 the surface geometry of a heliostat, a sign board is a stationary structure with constant inclination angle
136 and therefore has a more limited range of centre of pressure movement compared to a heliostat tracking
137 over a large range of elevation and azimuth angles. There is a linear increase of the centre of pressure
138 toward the leading edge of a thin, square flat plate aligned parallel to the ground ($\alpha = 0^\circ$) with

139 decreasing β from 90° to 0° (Holmes *et al.* 2006). Gong *et al.* (2013) investigated the effect of α and β
140 on the mean, RMS and peak pressure distributions on the heliostat surface. However, the variation of
141 the position of the centre of pressure where the net force acts on the heliostat surface for different
142 azimuth-elevation configurations is not well understood. Hence, the first objective of this paper is to
143 determine the positions of the centre of pressure corresponding to the mean and peak pressure
144 distributions on a heliostat at a range of elevation and azimuth angles.

145 Peterka and Derickson (1992) derived non-dimensional peak load coefficients to account for the
146 turbulence in the wind from the measured mean wind speed $\bar{U}_H \approx 12.6$ m/s and the turbulence intensity
147 $I_u = \sigma_u / \bar{U}_H = 18\%$, defined as the ratio of the root-mean-square of the fluctuating velocity to the mean
148 wind velocity at the elevation axis height $H = 0.155$ m. The maximum design aerodynamic load
149 coefficients have been reported in scale-model heliostat wind tunnel experiments (Peterka *et al.* 1988;
150 Peterka *et al.* 1989; Peterka and Derickson 1992) over a range of elevation-azimuth configurations in
151 an open country terrain ($z_0 = 0.03$ m) with $I_u = 18\%$ and $G_u = 1.6$ at the heliostat elevation axis height.
152 Turbulence intensities $I_u \geq 10\%$ have been found to significantly influence the peak drag, lift and
153 overturning moment coefficients in operating positions ($\alpha = 15-90^\circ$ and $\beta = 0-180^\circ$) by Peterka *et al.*
154 (1988) and Yu *et al.* (2019), and the peak lift and hinge moment coefficients in stow position ($\alpha = 0^\circ$)
155 by Pfahl *et al.* (2015) and Emes *et al.* (2017). Furthermore, the peak drag and lift coefficients on normal
156 and stowed heliostats have been shown to depend on both the turbulence intensity and the longitudinal
157 integral length scales of the energy-containing eddies in the longitudinal and vertical directions,
158 respectively (Jafari *et al.* 2018; Jafari *et al.* 2019a). The aerodynamic coefficients reported by Peterka
159 *et al.* (1989) only specified one worst-case scenario for the peak hinge moment at $\alpha = 30^\circ$ for a range
160 of α between 0° and 180° and $\beta = 0^\circ$. Increasing lift force and pitching moment coefficients have been
161 observed on wings (Holloran and O'Meara 1999) and flat plates (Ortiz *et al.* 2015) as they are positioned
162 closer to the ground. Further knowledge of the effect of changes in β between 0° and 180° on the peak
163 hinge moments, due to the maximum movement of the centre of pressure from the central elevation
164 axis, is critical for the design of the elevation drive to maintain the structural rigidity of the heliostat
165 during operation. Hence, the second objective of this paper is to determine the effect of turbulence

166 intensity on the position of the centre of pressure and the resulting hinge and overturning moment
167 coefficients on heliostats at a range of azimuth and elevation angles.

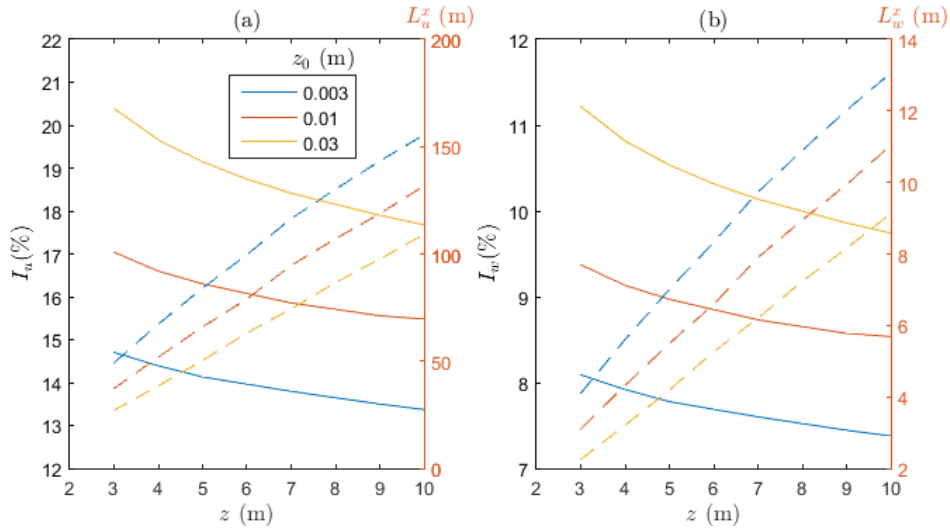
168 The influence of the temporal and spatial variations of turbulence in the lowest 10 m of the ABL
169 on the peak load coefficients with changes in aerodynamic surface roughness and heliostat size is an
170 important consideration for the design loads on the drive units, pedestal and foundation. Wind loads on
171 heliostats are highly dependent on the surrounding terrain of a heliostat field, which can be characterised
172 by location, the height above sea level, total area of land, maximum height variations across the terrain
173 and a description of the ground roughness, including any natural topography or structures larger than 2
174 m in height (AS/NZS 1170.2 2011). Table 1 shows three terrain categories defined by Xu (2013) with
175 the estimated surface roughness parameters z_0 (m) and $\alpha_{\bar{v}}$ for the logarithmic law and power law
176 velocity profiles, respectively. The vertical profiles of turbulence intensities and length scales of the
177 longitudinal u and vertical w velocity components in Figure 2 are taken from ESDU 85020 (2001) using
178 similarity theory formulations of full-scale ABL data as a function of z_0 . A flat “open country” terrain
179 is commonly assumed as the surroundings of a heliostat field (Peterka and Derickson 1992; Pfahl *et al.*
180 2015), where the wind characteristics are derived from the 10-m reference height defined in design
181 wind codes and standards (Holmes 2007). However, the expected loads for the single turbulence
182 condition ($I_u = 18\%$, $G_u = 1.6$) can only be applied to a height of 10 m in an open country terrain with
183 $z_0 = 0.03$ m from ESDU 85020 (2001) full-scale atmospheric boundary layer data in Figure 2(a). The
184 largest heliostats ($A \geq 120$ m²) currently deployed by Abengoa Solar and Sener are typically designed
185 with $H \leq 6$ m, however smaller heliostats ($A \leq 20$ m²) developed by eSolar and Brightsource Energy
186 are closer to the ground with $H \leq 3$ m (Télez *et al.* 2014). Furthermore, heliostat fields are commonly
187 positioned in low-roughness terrains, such as flat deserts and grassy plains (Table 1). The turbulence
188 intensities in a flat desert are approximately 25% smaller at all heights below 10 m compared to an open
189 country terrain, whereas the turbulence length scales increase by 43% at $z = 10$ m and by as much as
190 82% at $z = 3$ m. Hence, the third objective of this paper is to identify the critical elevation-azimuth
191 configurations of the heliostat corresponding to the hinge and overturning moments in operating

192 positions and determine the maximum operational design wind speeds that allow the operating loads to
 193 remain below the ultimate stow design loads.

194 Table 1. Terrain categories in the ABL (Xu 2013)

Terrain description	z_0 (m)	$\alpha_{\bar{u}}$
Open country with isolated trees and buildings	0.03	0.17
Grass and very few trees	0.01	0.15
Flat desert	0.003	0.12

196



197

198 Figure 2. (a) Longitudinal $i = u$, and (b) vertical $i = w$ turbulence intensity I_i and length scale L_i^x profiles from
 199 ESDU 85020 (2001) as a function of aerodynamic roughness height z_0 at heights below 10 m where heliostats
 200 are positioned in the lower atmospheric surface layer. Solid lines indicate the turbulence intensity profiles and
 201 dashed lines indicate the integral length scale profiles.

202 2. Method

203 The aerodynamic force and moment coefficients on heliostats are calculated from experimental
 204 measurements in a wind tunnel at the University of Adelaide. Surface pressures on an instrumented
 205 heliostat (Figure 3) with square cross-section chord length $c = 0.8$ m and forces at the base of the
 206 heliostat model with elevation axis height $H = 0.5$ m were sampled at 1 kHz using four three-axis load
 207 cells mounted on a force balance. The heliostat facet is attached to a circular hollow section pylon by a
 208 hinge pin joint to adjust the elevation angle α in increments of 15° between 0° and 90° and an electronic
 209 turntable to adjust the azimuth angle β in increments of 30° between 0° and 180° . The instrumented
 210 heliostat is positioned within three simulated part-depth atmospheric boundary layers (ABLs) at
 211 longitudinal turbulence intensities of 8%, 13% and 26% at the heliostat elevation axis height $H = 0.5$

212 m. Further details of the experimental setup of spires and roughness elements for the generation of the
 213 ABLs are provided in Yu *et al.* (2019) and Jafari *et al.* (2019a). The experimental devices used for the
 214 heliostat surface pressure and force measurements are described in the previously published papers by
 215 the authors (Emes *et al.* 2017; Emes *et al.* 2019; Yu *et al.* 2019). The pressure coefficients at each of
 216 the 24 tap locations i on the heliostat mirror surface were calculated as:

$$217 \quad C_{p_i}(t) = \frac{p'(t)}{1/2\rho\bar{U}_H^2}, \quad (1)$$

218 where ρ (kg/m³) is the air density, \bar{U}_H (m/s) is the mean wind speed at the heliostat elevation axis height
 219 H , and $p(t) = P_i^f(t) - P_i^b(t)$ (Pa) is the instantaneous differential pressure between the upper and
 220 lower surfaces of the heliostat mirror. The net force acting perpendicular to the mirror surface F_N ,
 221 defined in Figure 1 for operating positions ($\alpha > 0^\circ$) and stow position ($\alpha = 0^\circ$), was calculated in the
 222 current study from the area-averaged pressure coefficient in equation 1,

$$223 \quad F_N = 1/2\rho\bar{U}_H^2 \oint -C_{p_i} dA. \quad (2)$$

224 Here $A = c \times c$ (m²) is the area of the heliostat mirror projected onto the x_H - y_H plane in operating
 225 positions (Figure 1a) and x_{HS} - y_{HS} plane in stow position (Figure 1b). For the derivation of mean and
 226 peak wind loads on heliostats (Peterka and Derickson 1992), the net force F_N acting perpendicular to
 227 the heliostat mirror at elevation angle α is decomposed into the drag force F_x in the horizontal wind
 228 direction and the lift force F_z in the vertical direction as follows:

$$229 \quad F_x = F_N \sin \alpha, \quad (3)$$

$$230 \quad F_z = F_N \cos \alpha. \quad (4)$$

231 The normal force F_N acts on the heliostat mirror surface at the centre of pressure $l_p = \sqrt{l_{px}^2 + l_{py}^2}$.

232 Turbulence in the approaching ABL causes a non-uniform differential pressure distribution $p(x, y)$ on
 233 the heliostat. The position of the centre of pressure is calculated in the current study as the distance in
 234 the longitudinal and lateral directions from the centre of the heliostat surface in Figure 3 by

$$235 \quad l_{px} = \frac{c}{2} - \frac{\int_0^c xp(x,y) dx}{\int_0^c p(x,y) dx}, \quad (5)$$

$$236 \quad l_{py} = \frac{c}{2} - \frac{\int_0^c yp(x,y) dy}{\int_0^c p(x,y) dy}. \quad (6)$$

237 The hinge moment about the elevation y_H axis of the heliostat mirror is

$$238 \quad M_{Hy} = F_N l_{px} , \quad (7)$$

239 as the product of the area-averaged net force in equation 2 and the centre of pressure distance l_{px} in
 240 equation 5 from the elevation y_H axis in Figure 3b. Similarly, the hinge moment about the x_H axis in
 241 the heliostat mirror plane in Figure 3 can be calculated as $M_{Hx} = F_N l_{py}$ using the same coordinate
 242 system as in Figure 1 of Peterka and Derickson (1992). The peak forces ($F_{peak} = \bar{F} + 3\sigma_F$) and
 243 moments ($M_{peak} = \bar{M} + 3\sigma_M$) are calculated as the sum of the mean values and three times the standard
 244 deviation of the fluctuating surface pressure measurements. The three-sigma approach provides peak
 245 values with a 99.7% probability of not being exceeded based on extreme value analysis (Simiu and
 246 Scanlan 1996). The peak forces derived from the surface pressure measurements are within $\pm 5\%$ of
 247 those calculated from the load cell measurements using the same three-sigma approach. Hence, the
 248 mean and peak aerodynamic coefficients of the forces and moments are calculated in the current study
 249 using the positions of the centre of pressure within the non-uniform pressure distributions, such as those
 250 in Figure 5 and Figure 6. Peak coefficients were calculated using the peak force/moment and the mean
 251 velocity \bar{U}_H at the heliostat elevation axis (hinge) height H in Figure 1, following the method outlined
 252 in Peterka and Derickson (1992) and the equations:

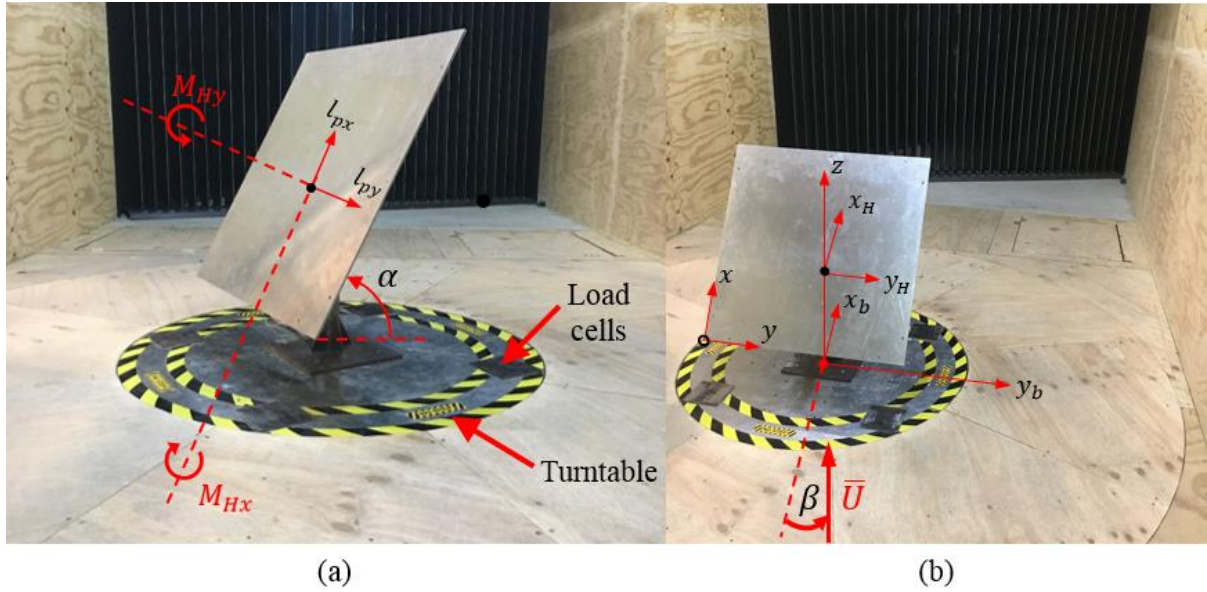
$$253 \quad c_{Fx} = \frac{F_x}{1/2\rho\bar{U}_H^2 A} , \quad (8)$$

$$254 \quad c_{Fz} = \frac{F_z}{1/2\rho\bar{U}_H^2 A} , \quad (9)$$

$$255 \quad c_{MHx} = \frac{F_N l_{py}}{1/2\rho\bar{U}_H^2 A c} = c_{FN} (l_{py}/c) , \quad (10)$$

$$256 \quad c_{MHy} = \frac{F_N l_{px}}{1/2\rho\bar{U}_H^2 A c} = c_{FN} (l_{px}/c) , \quad (11)$$

$$257 \quad c_{My} = \frac{M_y}{1/2\rho\bar{U}_H^2 A H} = c_{MHy} \left(\frac{c}{H}\right) + c_{Fx} . \quad (12)$$



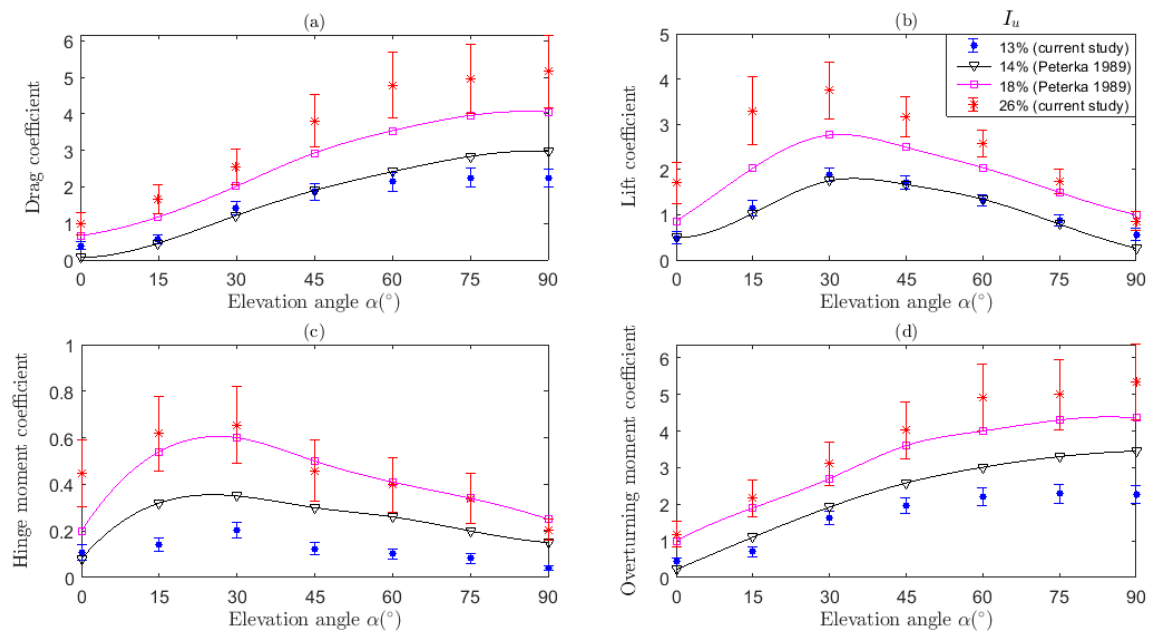
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259 Figure 3. Experimental setup for measurement of the pressures and forces on a model heliostat at a range of
 260 elevation α and azimuth β angles: (a) the centre of pressure $l_p(x, y)$ is defined by the streamwise l_{px} and spanwise
 261 l_{py} distances from the centre of the heliostat mirror plane; (b) definition of the coordinate axes (x, y) for the
 262 pressure measurements on the heliostat mirror plane, elevation axes (x_H, y_H) at the heliostat hinge height, azimuth
 263 z -axis and axes (x_b, y_b) at the base of the heliostat pylon.

264 3. Results and Discussion

265 Figure 4 compares the calculated peak aerodynamic force and moment coefficients at $\beta = 0^\circ$ for
 266 the two simulated ABLs ($I_u = 13\%$ and 26%) in the current study with those reported by Peterka *et al.*
 267 (1989) at $I_u = 14\%$ and 18% . The peak drag (Figure 4a) and lift (Figure 4b) coefficients for operating
 268 elevation angles at $I_u = 13\%$ in the current study follow a similar trend to those of Peterka *et al.* (1989)
 269 at $I_u = 14\%$. It can be observed that the lift and hinge moment coefficients increase more significantly
 270 at smaller elevation angles $\alpha \leq 30^\circ$ and in stow position ($\alpha = 0^\circ$) compared to the study by Peterka *et*
 271 *al.* (1989). Furthermore, the drag and overturning moments at $\alpha \geq 60^\circ$ show a smaller increase with
 272 increasing α to 90° in the current study. The largest differences in the peak load coefficients between
 273 the two studies at $\alpha = 0^\circ$ and 90° are likely to be caused by variations in the ratio of the turbulence
 274 length scales relative to the heliostat chord length (Jafari *et al.* 2018; Jafari *et al.* 2019a). The distribution
 275 of turbulence length scales varies significantly with height compared to turbulence intensity in the two
 276 part-depth ABL simulations with scale factors of 1:151 and 1:90 in the current study (Jafari *et al.*
 277 2019a). When comparing the peak load coefficients reported by Peterka *et al.* (1989) at $I_u = 18\%$ with
 278 the current study at $I_u = 26\%$, the peak drag coefficient at $\alpha = 90^\circ$ increases by 29% from 4.0 to 5.16

279 and the peak lift coefficient at $\alpha = 30^\circ$ increases by 34% from 2.8 to 3.75. This confirms the
 280 approximately linear increases of the peak drag and lift coefficients on heliostats in operating (Peterka
 281 *et al.* 1988) and stow (Pfahl *et al.* 2015; Emes *et al.* 2017) positions observed with increasing turbulence
 282 at $I_u \geq 10\%$. This is caused by increases in the standard deviation of the fluctuating drag and lift forces,
 283 shown by the error bars in Figure 4, with increasing I_u from 13% to 26%. The standard deviation of the
 284 drag coefficients increases by a factor of 3 at $\alpha = 0^\circ$ and by up to a factor of 4 at $\alpha = 90^\circ$, whereas
 285 those for the lift coefficients increase by factors of between 1.5 at $\alpha = 90^\circ$ and 4.4 at $\alpha = 30^\circ$. It is
 286 noted that the standard deviations of the force coefficients at $I_u = 26\%$ are significantly larger than
 287 those at $I_u = 13\%$, such that they are similar in magnitude to the mean coefficients. Hence, the peak
 288 force and moment coefficients derived for this high-turbulence case are likely to have a larger error
 289 margin for estimating the full-scale heliostat loads within the expected range of turbulence conditions
 290 (Figure 2) in the lowest 10 m of the ABL.



291
 292 Figure 4. Peak aerodynamic coefficients at azimuth angle $\beta = 0^\circ$ as a function of elevation angle α and
 293 turbulence intensity I_u (%) compared with Peterka *et al.* (1989): (a) drag force coefficient c_{FHx} ; (b) lift force
 294 coefficient c_{Fz} ; (c) hinge moment c_{MHy} ; (d) overturning moment c_{My} . Error bars indicate one standard deviation
 295 of the coefficients from the mean values.

296 3.1. Heliostat Pressure Distributions

297 Figure 5 and Figure 6 show the mean and standard deviation pressure coefficient distributions,
 298 respectively, on the heliostat surface at different elevation α and azimuth β angles within the simulated

299 ABL at a turbulence intensity $I_u = 13\%$. The region of high suction on the heliostat surface at $\beta = 0^\circ$
300 with mean $C_p > 2$ and standard deviation $C_p > 0.4$ moves towards the leading edge ($x = 0$ m) with
301 decreasing α from 60° to 15° . The maximum values of the peak pressure coefficients calculated
302 following the three-sigma approach at $\alpha = 30^\circ$ in the current study ($C_p = 2.92$) are consistent with peak
303 C_p values of 3.4 near the leading edge and 2.62 near the central elevation axis in the distribution of
304 Pfahl *et al.* (2011) at $\alpha = 30^\circ$ and $\beta = 0^\circ$. A similar trend to Gong *et al.* (2013) is found that the high-
305 magnitude region of the mean pressure coefficient distribution in Figure 5 is concentrated near the
306 windward edge of the heliostat with the mean C_p at $\alpha = 60^\circ$ increasing to 2.1 at $\beta = 0^\circ$, 2.7 at $\beta = 60^\circ$,
307 -1.8 at $\beta = 150^\circ$ and -2.02 at $\beta = 180^\circ$. The maximum values of mean C_p at $\alpha = 60^\circ$ in the current
308 study are larger than the mean $C_p = \pm 1.2$ at $\beta = 0^\circ$ and 180° and $C_p = \pm 1.5$ at $\beta = 60^\circ$ and 150° in
309 Gong *et al.* (2013). The reason for the variability in the pressure coefficients of the current study, Pfahl
310 *et al.* (2011) and Gong *et al.* (2013) is likely to be the differences in the heliostat model size, turbulence
311 intensities and length scales in the simulated ABLs. Although the longitudinal turbulence intensities are
312 similar in these studies, differences in the ratio of the turbulence length scales and the heliostat model
313 dimensions and the consequent mismatch of the turbulence spectra can lead to variations in the unsteady
314 loads measured on the models (Jafari *et al.* 2019b). The standard deviation pressure coefficients in
315 Figure 6 show that the large magnitude pressure fluctuations ($C_{p,std} > 0.4$) extend further from the
316 windward edge towards the centre of the heliostat than the mean pressure coefficients. For example, the
317 position of the high-magnitude fluctuating pressures extends to 0.2 m downstream of the windward
318 edge in the x -direction at $\alpha = 15^\circ$, compared to a maximum lateral distance of 0.13 m from the side
319 edge of the heliostat ($y = 0.8$ m) at $\alpha = 30^\circ$. In contrast for stow position ($\alpha = 0^\circ$), the mean pressure
320 coefficients are smaller in magnitude but the high-pressure region spans a greater portion of the
321 windward edge of the heliostat. This corresponds to the maximum movement of the centre of pressure
322 towards the windward edge at $\beta = 0^\circ$ and 180° with highly correlated pressures across the width of the
323 stowed heliostat surface. Further, the magnitudes of the mean and standard deviation pressure
324 coefficients in stow and operating positions indicate that the maximum forces on the heliostat structure
325 occur when the wind is perpendicular to the heliostat surface at $\beta = 0^\circ$ and 180° . There is a change of

326 sign of the mean pressure coefficients on operating heliostats as β increases beyond 90° , such that the
327 wind approaches the inclined heliostat from behind and the differential pressure between the upper and
328 lower surfaces becomes negative. The high-pressure regions from separation of the approaching
329 turbulent flow over the maximum width of the windward edge causes the largest variation in pressure
330 along the mirror in the x -direction. The maximum hinge moment is thus likely to be characterised by
331 strong suction near the leading edge at $\alpha = 15^\circ$ and 30° , as the standard deviation pressure coefficients
332 become more highly correlated across the span of the heliostat at $\beta = 0^\circ$ and 180° . This suggests that
333 the distribution of the turbulent pressure fluctuations indicated by the movement of the centre of
334 pressure is critical for the peak hinge moments on heliostats at smaller elevation angles $\alpha \leq 30^\circ$. The
335 peak hinge moment is decisive for the design wind loads on the elevation drive of a conventional
336 pedestal-mounted heliostat, particularly at smaller elevation angles and in stow position. Hence, the
337 spanwise-averaged distribution of pressure in the along-wind x -direction of the heliostat surface and
338 the position of the centre of pressure where the net force acts corresponding to the peak hinge moment
339 on the heliostat should be considered for design loads.

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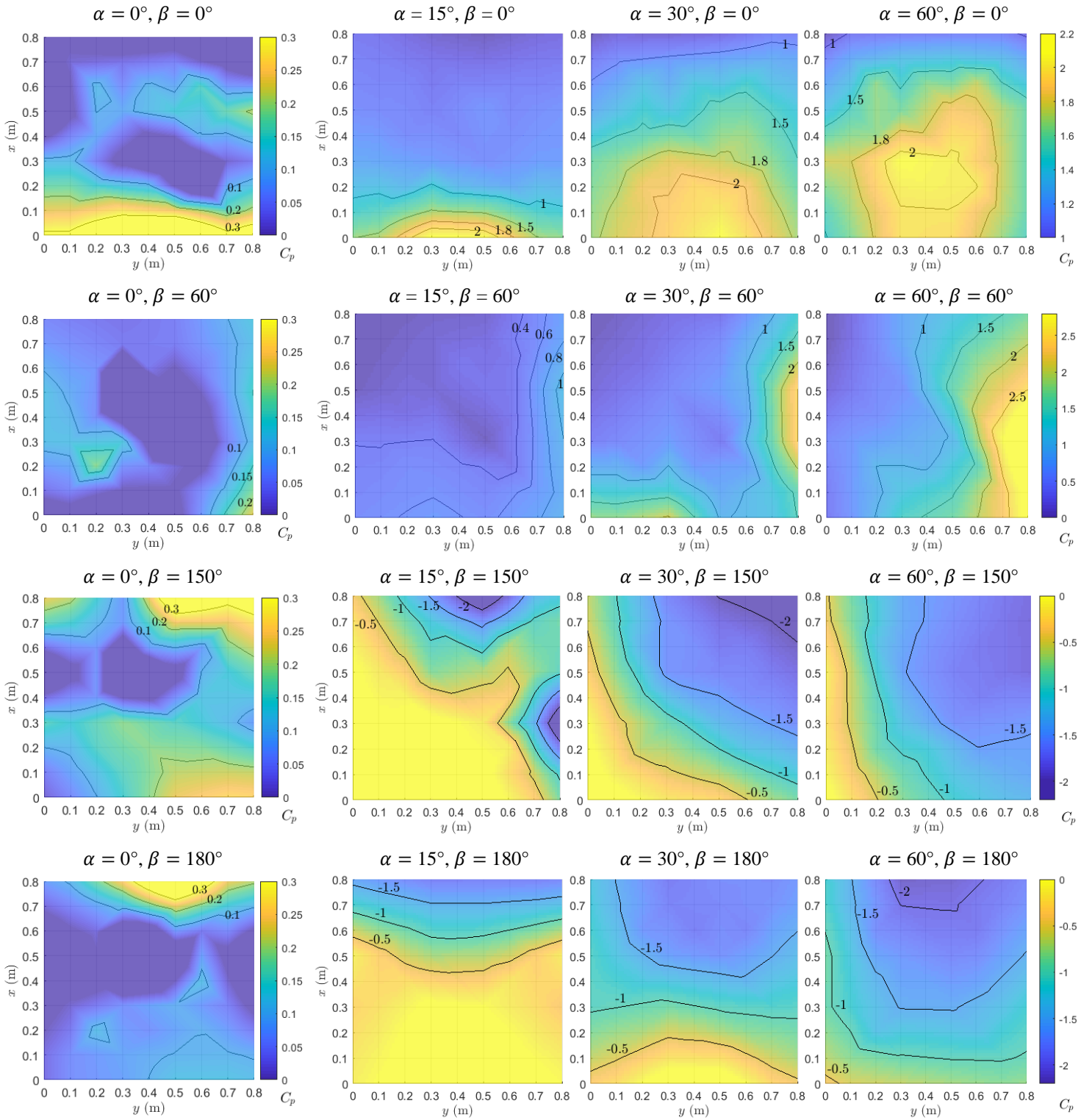
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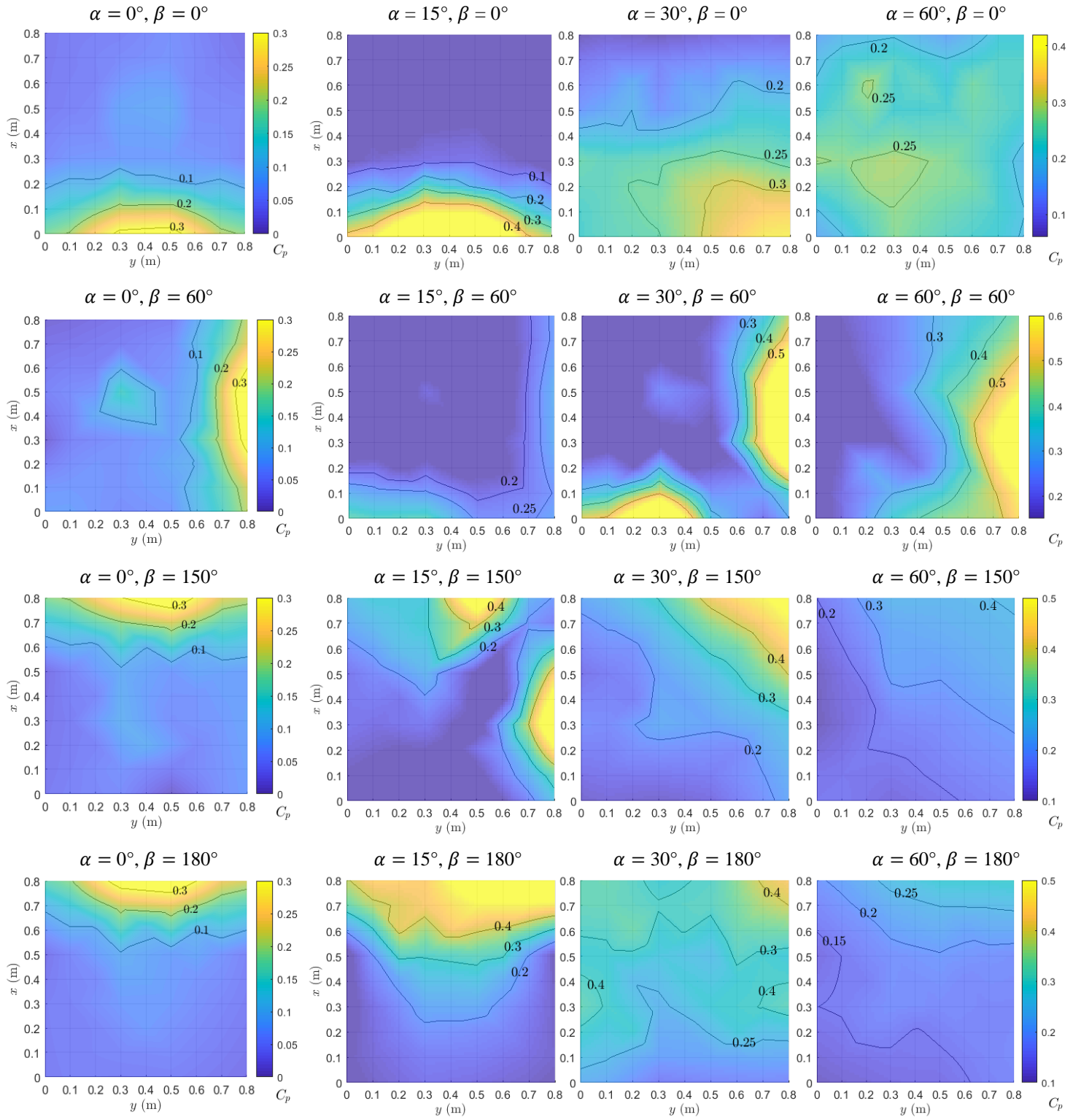
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353 Figure 5. Surface contour distributions of the mean pressure coefficients on the heliostat surface at different
 354 elevation α and azimuth β angles within a simulated ABL at a turbulence intensity $I_u = 13\%$ at the heliostat
 355 elevation axis height. The leading edge at $\beta = 0^\circ$ corresponds to $x = 0$ m. The black lines indicate constant C_p
 356 values and the coloured bar on the right side of each row (at constant β) shows the range of mean pressure
 357 coefficients.
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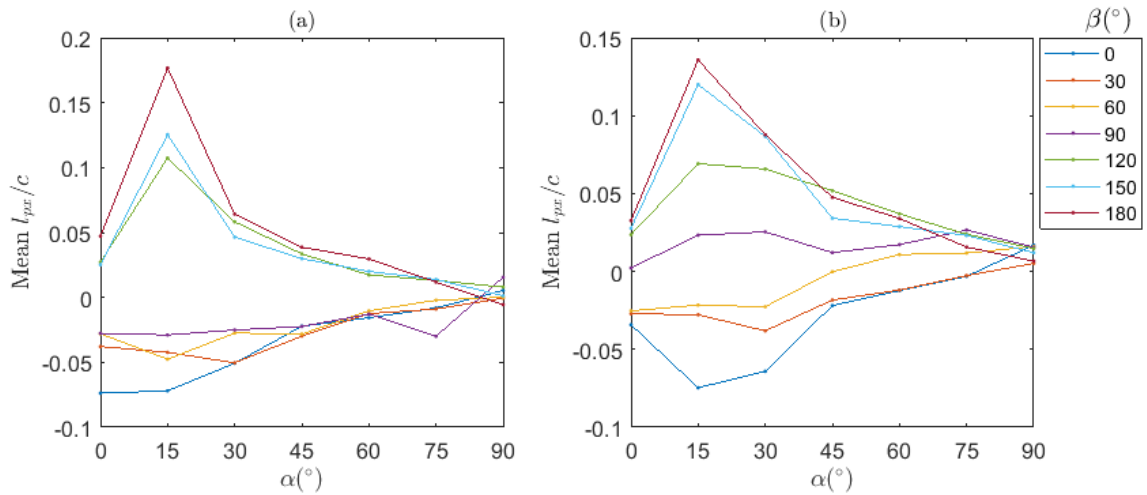
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360 Figure 6. Surface contour distributions of the standard deviation pressure coefficients on the heliostat surface at
 361 different elevation α and azimuth β angles within a simulated ABL at a turbulence intensity $I_u = 13\%$ at the
 362 heliostat elevation axis height. The black lines indicate constant C_p values and the coloured bar on the right side
 363 of each row (at constant β) shows the range of the standard deviation pressure coefficients.
 364

365 Figure 7 shows the average position of the centre of pressure l_{px} from the central elevation axis in
 366 the longitudinal direction x on the heliostat surface in Figure 1(a), as a non-dimensional ratio of the
 367 chord length of the heliostat at different elevation and azimuth angles. The average distance where the
 368 net force acts on the heliostat surface increases with decreasing elevation angle α from 90° to 15° for
 369 all of the azimuth angles β . The largest positive values of l_{px}/c from the movement of the normal force
 370 towards the trailing edge of the heliostat occur at $\beta = 180^\circ$, whereas the largest negative values of l_{px}/c
 371 at $\beta = 0^\circ$ correspond to the movement of the normal force towards the heliostat's leading edge in Figure
 372 3. The smallest values of $|l_{px}/c| \leq 0.03$ occur at all elevation angles when the wind approaches the
 373 heliostat at $\beta = 90^\circ$ and all azimuth angles in stow position at $\alpha = 0^\circ$ for $|l_{px}/c| \leq 0.11$. The profiles
 374 of l_{px}/c are similar at $I_u = 13\%$ (Figure 7a) and $I_u = 26\%$ (Figure 7b), which indicates that the effect
 375 of turbulence intensity is not significant on the mean pressure distribution on the heliostat. The average
 376 position of the centre of pressure $|l_{px}/c| \geq 0.1$ is most significant at the smaller operating elevation
 377 angles $\alpha = 15^\circ$ and 30° and for wind approaching the windward ($\beta = 0^\circ$) or leeward ($\beta = 180^\circ$) edges
 378 of the heliostat surface. The maximum l_{px}/c occurs at $\alpha = 15^\circ$ for these two critical azimuth angles,
 379 where the absolute magnitude of l_{px}/c at $\beta = 180^\circ$ is approximately double that at $\beta = 0^\circ$ for both
 380 turbulence intensities in Figure 7(a) and Figure 7(b). The largest movement of the time-averaged centre
 381 of pressure l_{px}/c toward the leading edge at $\alpha = 15^\circ$ is due to the increased pressure on the lower
 382 surface of the heliostat caused by the ground not allowing the flow to expand as it would without the
 383 presence of a lower boundary. This is analogous to the "ground effect" observed on wings (Holloran
 384 and O'Meara 1999) and flat plates (Ortiz *et al.* 2015) at small angles of attack near the ground at $H/c \leq$
 385 0.5. The small difference between the mean l_{px}/c in stow position ($\alpha = 0^\circ$) at $\beta = 0^\circ$ and 180° is likely
 386 to be due to the small average differential pressures measured for this case that are close to the maximum
 387 error of the pressure sensors. The heliostat surface is supported by a hinge pin joint and telescopic pylon
 388 in the absence of a torque tube in the heliostat model in the current study, as shown in Figure 8. The
 389 decreased magnitudes of the mean pressure coefficients (Figure 5) at $\beta = 180^\circ$ compared to $\beta = 0^\circ$ on
 390 the heliostat in operating positions ($15^\circ \leq \alpha \leq 60^\circ$) confirms the finding by Gong *et al.* (2013).
 391 Furthermore, the differences in the mean pressure distributions at these two azimuth angles leads to an

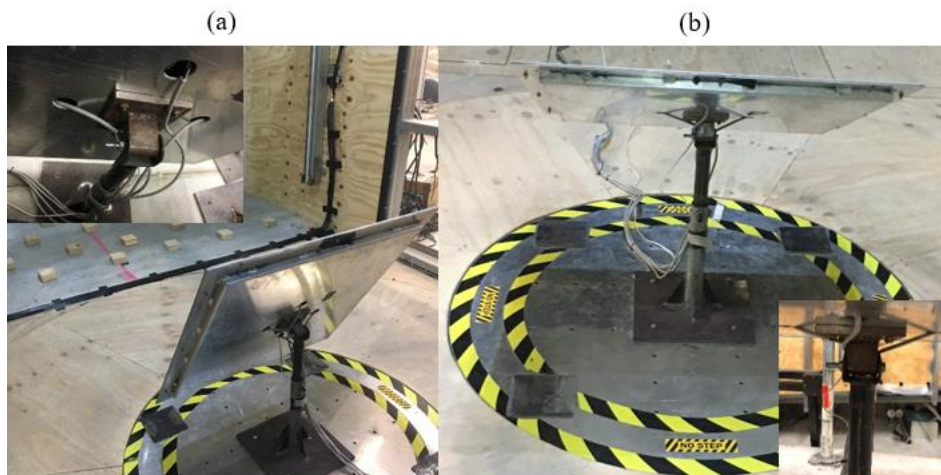
392 increased movement of the centre of pressure (Figure 7) from the central elevation axis at $\beta = 180^\circ$
 393 caused by flow separation generated by the rectangular prism shape of the hinge joint protruding from
 394 the back of the heliostat surface in Figure 8(a). Hence, wind approaching the back of the heliostat ($\beta =$
 395 180°) should be considered in addition to the front of the heliostat ($\beta = 0^\circ$) for the critical mean pressure
 396 distributions on operating heliostats, such as at $\alpha = 15^\circ$, that lead to the largest movement of the time-
 397 averaged net force from the central elevation axis.



398

399 Figure 7. Average position of the centre of pressure l_{px}/c in the streamwise x direction from the central elevation
 400 y_H axis of a heliostat at different elevation and azimuth angles: (a) $I_u = 13\%$; (b) $I_u = 26\%$.

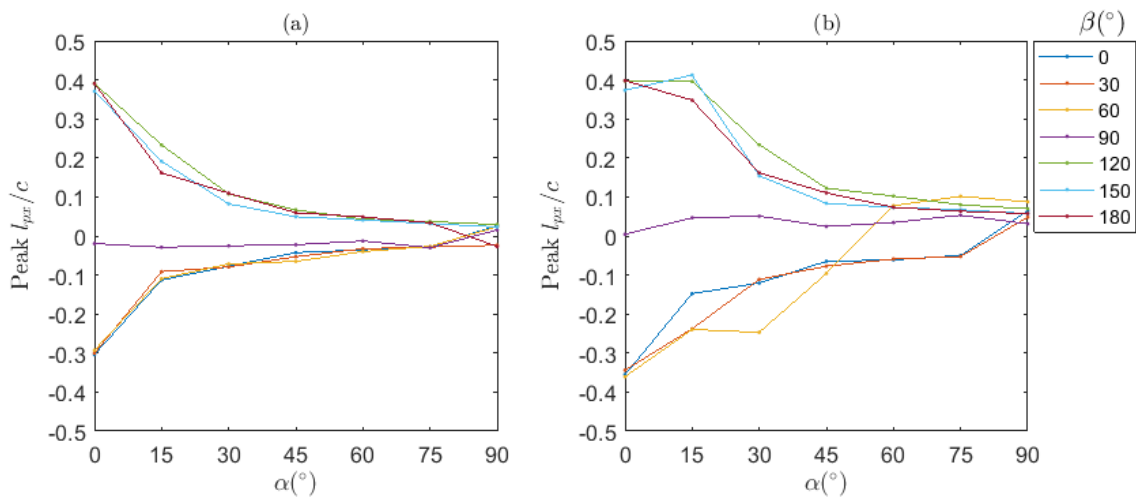
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402 Figure 8. Heliostat model and inset photos of the hinge pin joint and telescopic pylon design: (a) back view of the
 403 model in a high-elevation ($\alpha = 75^\circ, \beta = 0^\circ$) operating position; (b) front view of the model in stow ($\alpha = 0^\circ, \beta =$
 404 180°) position.

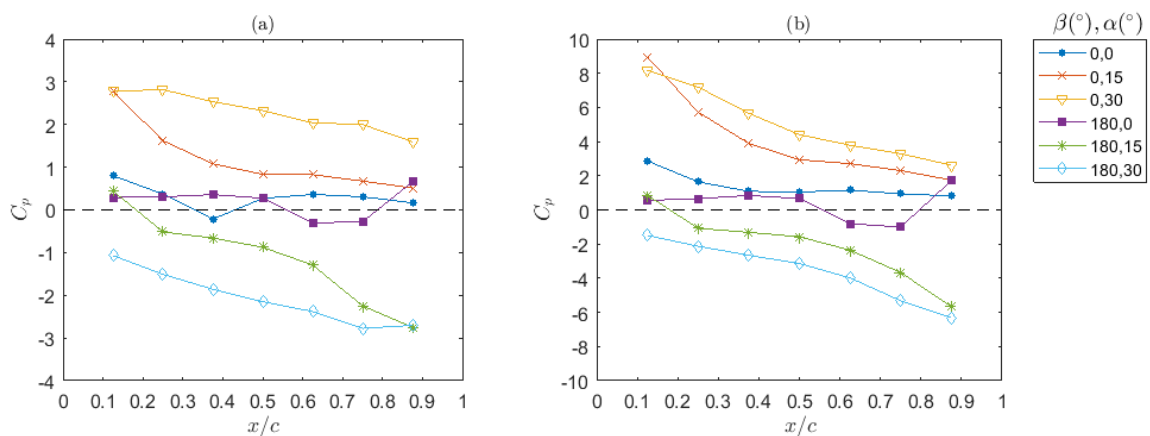
405 Figure 9 shows the peak movement of the centre of pressure l_{px} in the longitudinal direction from
 406 the central elevation y_H axis of the heliostat, calculated following the three-sigma approximation as a

407 non-dimensional ratio of the chord length of the heliostat at different elevation and azimuth angles.
408 Similar to the mean l_{px}/c in Figure 7, the peak l_{px}/c decreases in magnitude with increasing α to
409 approximately zero at $\alpha = 90^\circ$. The maximum absolute values of the peak l_{px}/c of 0.3-0.4 occur in
410 stow position ($\alpha = 0^\circ$) at all azimuth angles β investigated. The magnitudes of $l_{px}/c \approx 0.3$ and their
411 variation with β at $\alpha = 0^\circ$ in stow position are consistent with experimental data reported by Holmes
412 *et al.* (2006), however at $\alpha \leq 30^\circ$ the maximum movement of l_{px}/c is larger at $\beta = 180^\circ$ due to the
413 increased build-up of pressure near the trailing edge ($x = 0.8$ m in Figure 5 and Figure 6) of the lower
414 surface of the heliostat. The peak l_{px}/c is also less sensitive than the mean l_{px}/c (Figure 7) to changes
415 in β at all of the elevation angles tested. The turbulence intensity of the simulated ABL has a more
416 significant impact on the peak l_{px}/c when comparing Figure 9(a) and Figure 9(b). There is a more
417 pronounced increase in l_{px}/c with decreasing α at the higher turbulence intensity $I_u = 26\%$ in Figure
418 9(b). For example, $l_{px}/c = 0.07$ and 0.16 at $\alpha = 60^\circ$ and $\alpha = 30^\circ$ for $\beta = 180^\circ$ at $I_u = 26\%$, compared
419 to $l_{px}/c = 0.05$ and 0.11 for the same elevation-azimuth configurations at the lower turbulence intensity
420 $I_u = 13\%$. Hence, the maximum movement of the centre of pressure from the central elevation axis is
421 highly sensitive to the turbulence of the approaching flow and less affected by changes in the azimuth
422 angle from the maximum cases at $\beta = 0^\circ$ and $\beta = 180^\circ$.



423
424 Figure 9. Peak position of the centre of pressure from the central elevation y_H axis of a heliostat at different
425 elevation and azimuth angles: (a) l_{px}/c at $I_u = 13\%$; (b) l_{px}/c at $I_u = 26\%$.

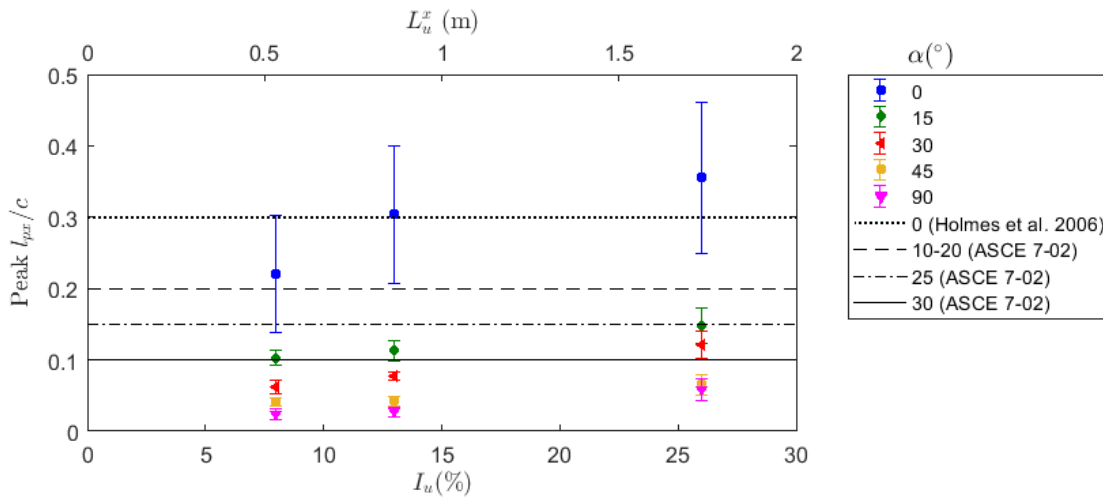
426 Figure 10 presents the peak pressure coefficients, calculated from the sum of the mean and three
427 times the standard deviation of the spanwise-averaged pressure coefficients in Figure 5 and Figure 6,
428 as a function of the longitudinal distance x/c on the heliostat surface at elevation and azimuth angles
429 corresponding to the maximum l_{px}/c in Figure 9. Comparison of the peak C_p profiles at $\beta = 0^\circ$
430 indicates that the peak normal force acting toward the upward-facing heliostat surface, calculated from
431 the integral of the C_p profile as a function of x in equation 2, increases to larger positive values with
432 increasing α from 0° to 30° . Similarly, the peak normal force acting toward the downward-facing
433 heliostat surface at $\beta = 180^\circ$ becomes increasingly negative. The peak normal force in Figure 10(a) at
434 $\beta = 180^\circ$ is 48%, 1% and 10% smaller than the corresponding force at $\beta = 0^\circ$ for $\alpha = 0^\circ$, 15° and 30° ,
435 respectively. With increasing turbulence in Figure 10(b) the differences in the peak normal force
436 between $\beta = 0^\circ$ and 180° at these elevation angles are 80%, 45% and 29%, respectively. However, the
437 smaller absolute magnitude of the normal force at $\beta = 180^\circ$ is accompanied by a larger movement of
438 the centre of pressure l_{px}/c than for $\beta = 0^\circ$ in Figure 9. This can be observed at $\beta = 0^\circ$ and 180° in
439 Figure 10(a) and Figure 10(b) by the larger increase in the absolute magnitude of C_p near the leading
440 edge at $\alpha = 15^\circ$ compared to $\alpha = 30^\circ$. The larger increase of l_{px}/c near the leading edge at $\alpha = 15^\circ$
441 contributes to a larger moment arm for the hinge moment M_{Hy} about the central elevation y_H axis of
442 the heliostat.



443
444 Figure 10. Spanwise-averaged profiles of the peak pressure coefficients in the longitudinal x -direction of the
445 heliostat surface at elevation-azimuth configurations corresponding to the maximum movement of the centre of
446 pressure at: (a) $I_u = 13\%$; (b) $I_u = 26\%$. The leading edge of the heliostat for azimuth angles of $\beta = 0^\circ$ and 180°
447 is at $x/c = 0$ and 1 , respectively.

448 Figure 11 shows the peak movement of the centre of pressure l_{px}/c from the central elevation y_H
 449 axis of the heliostat ($c = 0.8$ m) at $\beta = 0^\circ$ as a function of longitudinal turbulence intensity I_u and
 450 integral length scale L_u^x measured at the heliostat elevation axis height ($H = 0.5$ m). The effect of
 451 increasing longitudinal turbulence in the approaching boundary layer flow has the largest influence on
 452 the fluctuating component of l_{px}/c in stow position ($\alpha = 0^\circ$), as indicated by the error bars representing
 453 one standard deviation from the mean l_{px}/c . The peak l_{px}/c increases by 38% from 0.22 to 0.3 for an
 454 increase in I_u from 8% to 13%, and by 20% from 0.3 to 0.36 with an increase in I_u from 13% to 26%.
 455 The large error bars of the peak l_{px}/c at $\alpha = 0^\circ$ are caused by the unsteady pressure fluctuations with
 456 near-zero mean values in stow that are highly concentrated near the leading edge of the heliostat surface.
 457 Hence, the uncertainty of the calculated peak l_{px}/c in stow is relatively large compared to the other
 458 elevation angles tested in the current study. Nevertheless, the maximum movement of l_{px}/c toward the
 459 leading edge at $\alpha = 0^\circ$ in the current study is consistent with $l_{px}/c = 0.3$ reported by Holmes *et al.*
 460 (2006) in experimental measurements on a thin flat plate aligned parallel to a longitudinal flow with an
 461 unknown turbulence intensity. As the elevation angle of the heliostat increases in operating positions,
 462 the peak l_{px}/c decreases significantly in magnitude and varies linearly with I_u and L_u^x . For instance,
 463 the peak l_{px}/c increases to a maximum of 0.15 at $\alpha = 15^\circ$, 0.12 at $\alpha = 30^\circ$ and 0.06 at $\alpha = 90^\circ$. At
 464 constant I_u and L_u^x in Figure 11, the values of peak l_{px}/c on the heliostat at $\alpha = 15^\circ$ and 30° increase
 465 due to the build-up of uniform pressure on the lower surface from the flow acceleration in the gap
 466 between the windward heliostat edge and the ground. The “ground effect” causes an increased normal
 467 force and centre of pressure movement on the heliostat with increasing turbulence compared to that on
 468 a square cross-section (length $c \times$ depth c) monoslope roof with mid-roof height $H > 18$ m in ASCE 7-
 469 02 (2002). The aerodynamic effects of corner vortices generated by flow separation at the building
 470 edges also cause differences in the position of the centre of pressure in the load distributions on ground-
 471 mounted and roof-mounted solar panels (Kopp *et al.* 2012). At the elevation angles $\alpha > 0^\circ$ of operating
 472 heliostats, the position of the centre of pressure shows an approximately linear increase with the spatial
 473 and temporal variations of turbulence at heights below 10 m is the maximum hinge moments on
 474 heliostats. However, it is suggested that the effect of changes in L_u^x/c on l_p/c should be investigated in

475 future studies, considering the large range of heliostat sizes that are currently deployed in operational
 476 PT plants and under construction in active projects.

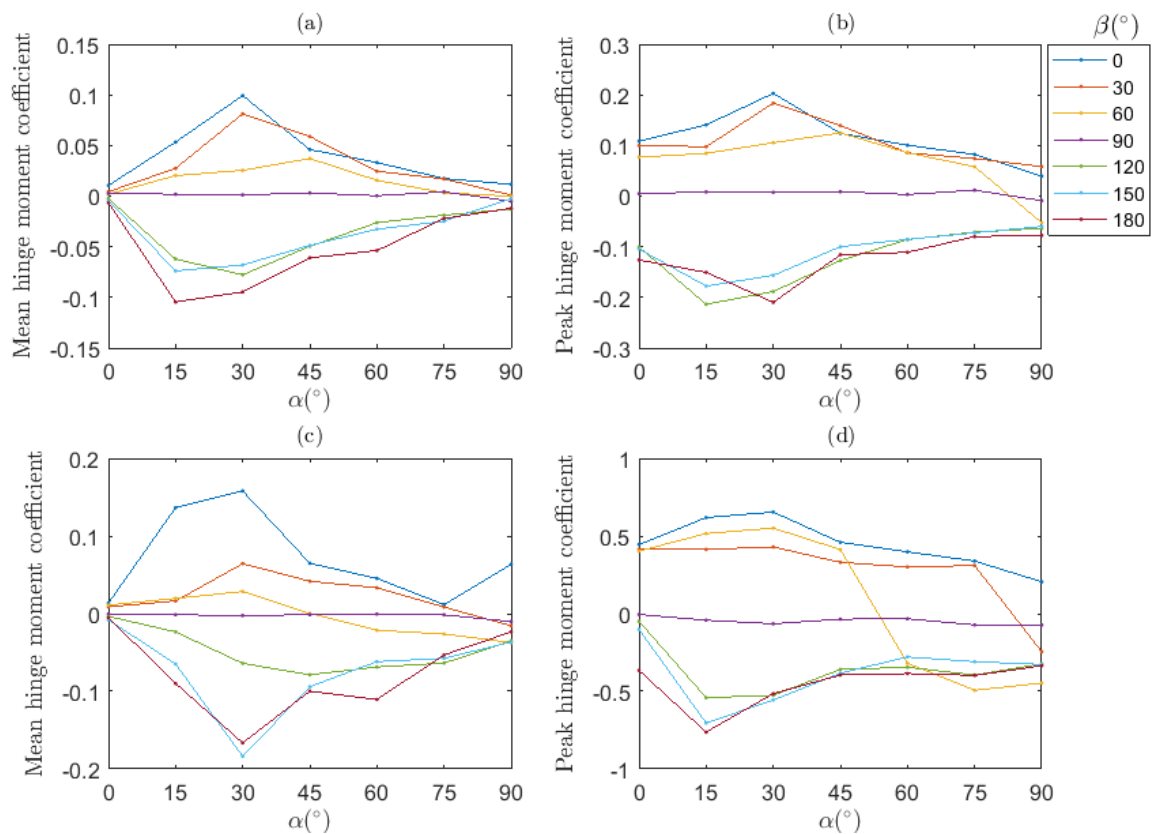


477
 478 Figure 11. Effect of longitudinal turbulence intensity I_u and length scale L_u^x on the peak movement of the non-
 479 dimensional centre of pressure l_{px}/c from the central elevation y_H axis of a heliostat with chord length c at
 480 different elevation angles for $\beta = 0^\circ$. Values of l_{px}/c are compared with a square flat plate (Holmes *et al.* 2006)
 481 and a square cross-section monoslope roof with mid-roof height $H > 18$ m (ASCE 7-02 2002). The error bars
 482 represent one standard deviation of the non-dimensional centre of pressure distance from the mean values.

483 3.2. Heliostat Hinge and Overturning Moment Coefficients

484 Figure 12 shows the mean and peak hinge moment coefficients c_{MHY} , calculated using equation 11
 485 as the product of the normal force coefficient and the non-dimensional distance to the centre of pressure,
 486 as a function of the elevation and azimuth angles for the two turbulence intensities investigated in the
 487 current study. For the lower turbulence case at $I_u = 13\%$, the peak positive (anti-clockwise direction in
 488 Figure 1a) c_{MHY} of 0.20 occurs at $\alpha = 30^\circ$ and $\beta = 30^\circ$, whereas the peak negative (clockwise direction
 489 in Figure 1a) c_{MHY} of -0.22 corresponds to the heliostat configuration of $\alpha = 30^\circ$ and $\beta = 180^\circ$. In
 490 comparison at $I_u = 26\%$, the peak positive c_{MHY} is 0.65 at $\alpha = 30^\circ$ and $\beta = 0^\circ$ and the peak negative
 491 c_{MHY} is -0.76 at $\alpha = 15^\circ$ and $\beta = 180^\circ$. Hence, the mean c_{MHY} are most sensitive to the elevation-
 492 azimuth configuration of the heliostat and the area-averaged normal force on the heliostat surface. In
 493 contrast, the peak c_{MHY} about the central elevation y_H -axis are highly dependent on the maximum
 494 distance to the centre of pressure l_{px}/c (Figure 9) within the non-uniform pressure distribution resulting
 495 from the turbulent velocity fluctuations in the ABL. The maximum values of the c_{MHY} within a turbulent
 496 ABL typically correspond to small elevation angles $\alpha = 15-30^\circ$ with wind approaching the front ($\beta =$

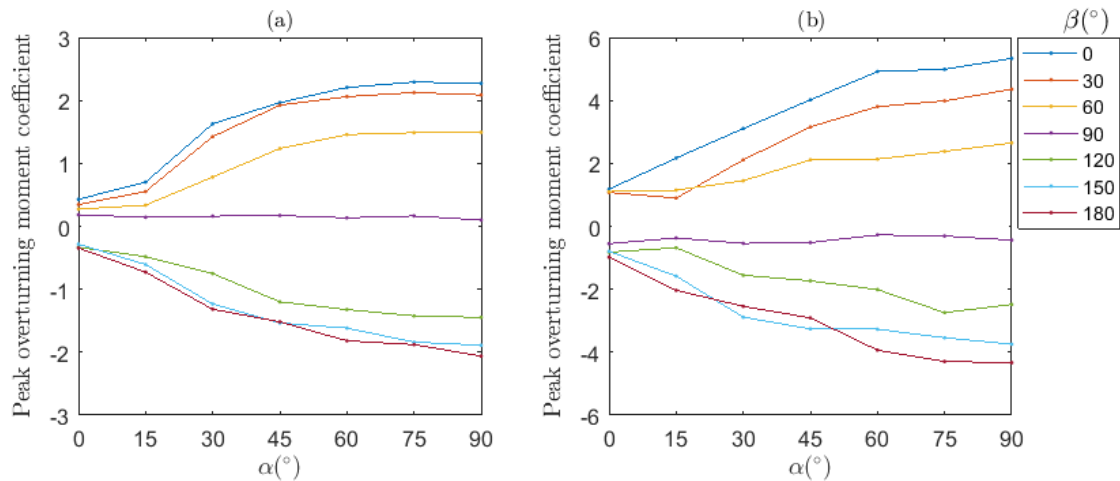
497 0°) or back ($\beta = 180^\circ$) of the heliostat, whereas Peterka *et al.* (1989) only reported the absolute values
 498 of the maximum hinge moment and lift force coefficients at $\alpha = 30^\circ$ and $\beta = 0^\circ$. Despite a smaller
 499 normal (and lift) force on the heliostat at $\beta = 180^\circ$, the maximum positive movement of the l_{px}/c at
 500 $\alpha = 15^\circ$ in the highly turbulent flow at $I_u = 26\%$ is larger than at $\beta = 0^\circ$ due to the increasing pressure
 501 near the windward edge of the lower heliostat surface. The maximum absolute value of the hinge
 502 moment coefficient therefore may not occur at the same azimuth-elevation heliostat configuration as
 503 the maximum lift force coefficient, as suggested by Peterka and Derickson (1992). Hence, the effect of
 504 turbulence intensity on the movement of the centre of pressure, particularly for cases of wind
 505 approaching the front ($\beta = 0^\circ$) and leeward ($\beta = 180^\circ$) sides of the heliostat at $\alpha \leq 30^\circ$, should be
 506 considered for the design operating loads on the elevation drive of a conventional pedestal-mounted
 507 heliostat.



508
 509 Figure 12. Hinge moment coefficients c_{MHy} about the central elevation axis of the heliostat at different elevation
 510 and azimuth angles: (a) Mean at $I_u = 13\%$; (b) Peak at $I_u = 13\%$; (c) Mean at $I_u = 26\%$; (d) Peak at $I_u = 26\%$.

511 Figure 13 shows the peak overturning moment coefficient c_{My} about the base of the heliostat as a
 512 function of elevation and azimuth angles for the two simulated ABLs in the current study. The peak

513 c_{My} increases with increasing elevation angle from stow ($\alpha = 0^\circ$) to maximum values at $\alpha = 90^\circ$ for
514 wind approaching the front ($\beta = 0^\circ$) or back ($\beta = 180^\circ$) of the heliostat. Similarly to the peak c_{MHY} in
515 Figure 12, the peak c_{My} is relatively independent of β and lowest in magnitude at $\beta = 90^\circ$ due to the
516 small drag force acting on the minimum projected area of the thin heliostat facet. The effect of β
517 increasing from 0° to 60° corresponds to a maximum reduction in peak c_{My} on a normal heliostat ($\alpha =$
518 90°) of 34% and 50% at $I_u = 13\%$ and 26% , respectively. The increase of c_{My} with increasing α and
519 decreasing β is largely caused by the strong correlation of the overturning moment to the drag force on
520 operating heliostats. However, the effect of β is attenuated in stow position at $\alpha = 0^\circ$ due to the reduced
521 influence of drag and the increasing impact of the hinge moment. Hence, the overturning moment
522 coefficients on a heliostat are largely dependent on azimuth-elevation configuration of the heliostat due
523 to their dependence on the drag force increasing with an increase in the projected frontal area of the
524 heliostat to the wind. In contrast in stow position, the overturning moment coefficient is independent of
525 azimuth angle and is more closely correlated to the hinge moment resulting from the movement of the
526 centre of pressure within the unsteady pressure distribution.



527
528 Figure 13. Peak overturning moment coefficients c_{My} as a function of elevation angle α and azimuth angle β at:
529 (a) $I_u = 13\%$; (b) $I_u = 26\%$.

530 Table 2 compares the critical heliostat α - β configurations as a function of turbulence intensity
531 corresponding to the maximum and minimum moment coefficients with the peak coefficients reported
532 by Peterka *et al.* (1989). The minimum (negative) hinge moment coefficients c_{MHY} are 5% and 17%
533 larger in magnitude than the maximum (positive) c_{MHY} at $I_u = 13\%$ and 26% , respectively. Further, the

534 absolute maximum c_{MHY} values occur at the heliostat configurations of $\beta = 180^\circ$ and $\alpha = 30^\circ$ at $I_u =$
535 13%, and for $\beta = 180^\circ$ and $\alpha = 15^\circ$ at $I_u = 26\%$. This result suggests that the unfavourable working
536 condition of $\alpha = 30^\circ$ and $\beta = 0^\circ$ found by Peterka *et al.* (1989) may not correspond to the maximum
537 operating hinge moment for all turbulence conditions. Hence, wind approaching the leeward side of the
538 downward-facing heliostat surface at $\beta = 180^\circ$ in the operating range $\alpha = 15\text{-}30^\circ$ should be considered
539 for the critical hinge moment cases. The critical heliostat configuration of $\alpha = 90^\circ$ and $\beta = 0^\circ$ for the
540 peak overturning moment coefficient c_{MY} is consistent with the finding of Peterka *et al.* (1989). This
541 confirms that critical load cases for c_{MY} are not strongly correlated to the hinge moment resulting from
542 the non-uniform pressure distribution, but are largely dependent on the maximum drag force with the
543 maximum frontal projected area of the heliostat surface to the wind.

544 Table 2. Critical heliostat configurations for the peak positive (anti-clockwise direction of rotation in Figure 1a)
545 and negative (clockwise direction of rotation in Figure 1a) moment coefficients as a function of turbulence
546 intensity, compared with the absolute maximum coefficients reported by Peterka *et al.* (1989).

Load coefficient	Heliostat configuration		Current study		Peterka <i>et al.</i> (1989)	
	α ($^\circ$)	β ($^\circ$)	13%	26%	14%	18%
Hinge moment c_{MHY}	30	0	0.20	0.65	0.35	0.60
	30,15	180	-0.21	-0.76		
Overturning moment c_{MY}	90	0	2.29	5.33	3.45	4.35
	90	180	-2.01	-4.21		

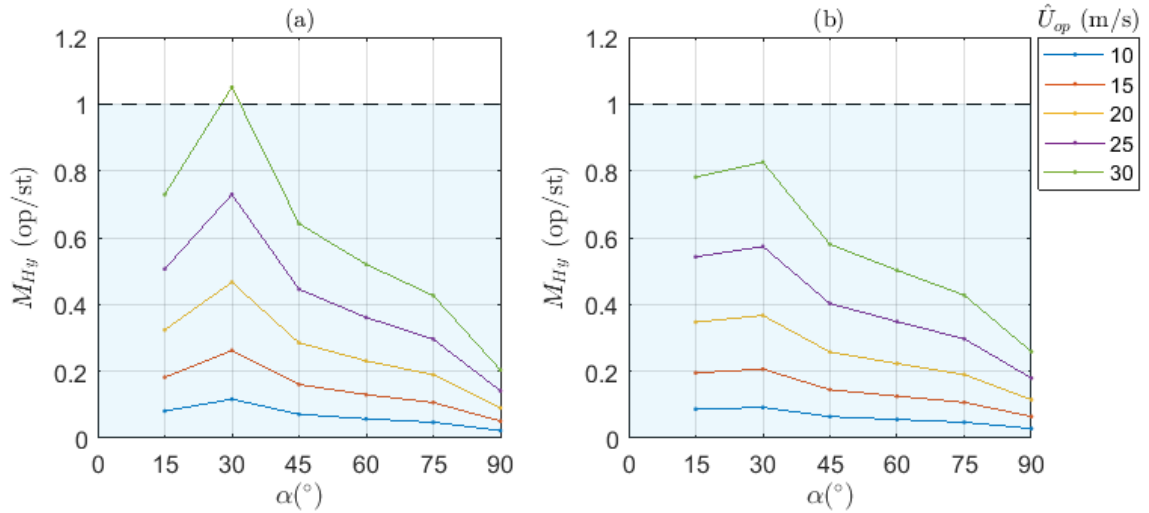
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548 3.3. Effect of Design Wind Speed on Operating Heliostat Loads

549 This section presents the ratios of the maximum operating and stow hinge and overturning moments
550 as a case study on a 6 m \times 6 m heliostat with $H/c = 0.5$ as a function of elevation angle and the operating
551 design gust wind speed (\hat{U}_{op}). The maximum operating loads, calculated at selected gust wind speeds
552 between 10 m/s and 30 m/s for different elevation angles at $\beta = 0^\circ$, are normalised with respect to a
553 constant maximum stow load for an assumed ultimate design condition $\hat{U}_{st} = 40$ m/s gust wind speed
554 at a 10 m height. The survival design wind speed for heliostats in stow position is 40 m/s at a 10-m
555 height based on a 100-year mean recurrence interval (Murphy 1980). This is equivalent to the 3-second
556 gust wind speed specified at the 10 m height in design wind codes and standards for buildings and other

557 physical structures, such as in exposed open terrains within Region A of AS/NZS 1170.2 (2011). The
558 peak operating and stow moments in this section are calculated using the peak aerodynamic coefficients
559 at $\beta = 0^\circ$ in Section 3.2 and equations 11-12. The mean wind speed at the elevation axis height $H = 3$
560 m is calculated from the 10-m height gust wind speed using the gust factor $G_u = U_{des}/\bar{U}_H$ and velocity
561 profile exponent calculated in the wind tunnel for the two turbulence cases: a moderate turbulence
562 intensity $I_u = 13\%$ corresponding to a flat desert terrain in Table 1 and Figure 2 with power law velocity
563 profile exponent $\alpha_{\bar{U}} = 0.12$ and gust factor $G_u = 1.44$, and a high turbulence intensity $I_u = 26\%$
564 corresponding to a suburban terrain with $\alpha_{\bar{U}} = 0.2$ and $G_u = 1.70$.

565 Figure 14 shows that the maximum operational hinge moment on the 36 m^2 heliostat at $\alpha = 30^\circ$
566 exceeds the stow hinge moment by 5% in the desert terrain with an operating design gust wind speed
567 $\hat{U}_{op} = 30 \text{ m/s}$ at a 10 m height, whereas the maximum operating hinge moment with $\hat{U}_{op} = 30 \text{ m/s}$ is
568 22% smaller than the stow hinge moment with $\hat{U}_{st} = 40 \text{ m/s}$ in the suburban terrain. Lowering the
569 operating design gust wind speed to 29 m/s in the desert terrain and increasing the operating design gust
570 wind speed to 33 m/s in the suburban terrain ensures that the operating load is maximum while
571 remaining below the stow load for all operating conditions of the heliostat. It is notable that the
572 maximum operating loads at larger elevation angles, such as $\alpha \geq 45^\circ$, are less than 60% and 70% of
573 the stow load for the optimal operating design wind speeds of 29 m/s and 33 m/s in the desert and
574 suburban terrains, respectively. This presents an opportunity to increase the design wind speed and thus
575 the operating hours of those regions of the heliostat field with favourable configurations (e.g. $\alpha \geq 45^\circ$)
576 for reducing the maximum hinge moments without compromising the strength and mass of material in
577 a heliostat design with drives that are able to resist the maximum loads in stow and operating positions.

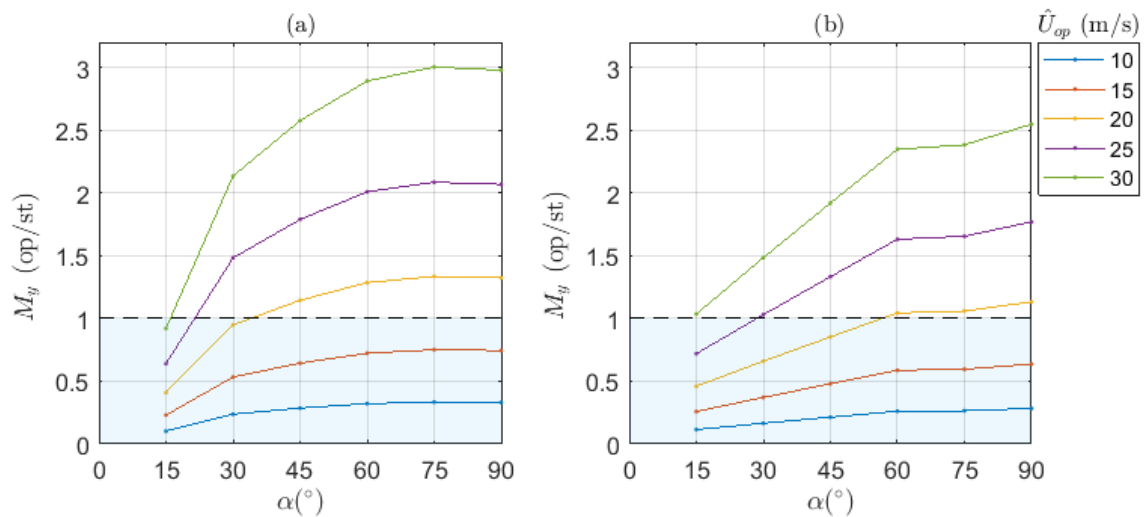


578

579 Figure 14. Ratios of the peak operating hinge moment to the stow ultimate design ($\hat{U}_{st} = 40$ m/s at $z = 10$ m)
 580 hinge moment M_{Hy} as a function of operating elevation angle α and maximum operational design wind speed
 581 (\hat{U}_{op} at $z = 10$ m) on a $6 \text{ m} \times 6 \text{ m}$ heliostat with $H/c = 0.5$ positioned in: (a) flat desert with $\alpha_{\bar{j}} = 0.12$, $I_u =$
 582 13% and $G_u = 1.44$; (b) suburban terrain with $\alpha_{\bar{j}} = 0.2$, $I_u = 26\%$ and $G_u = 1.70$.

583 Figure 15 presents the maximum operating overturning moments normalised with respect to the
 584 ultimate design stow overturning moments on a $6 \text{ m} \times 6 \text{ m}$ heliostat with $H/c = 0.5$ as a function of
 585 elevation angle and operating design gust wind speed (\hat{U}_{op}). It can be seen that the critical operating
 586 configuration of the heliostat is $\alpha = 90^\circ$, such that $\hat{U}_{op} = 17$ m/s in a flat desert (Figure 15a) and $\hat{U}_{op} =$
 587 18 m/s in a suburban terrain (Figure 15b). It is relevant to consider that the heliostats are likely to only
 588 be operating at α close to 90° for small periods of the day near sunrise and sunset, which may lead to
 589 over-designed loads for the structural rigidity of the pedestal and the concrete depth of the foundation.
 590 For example, the operating design wind speed that allows the maximum operating load to remain below
 591 the stow load can be increased to 18 m/s in a flat desert and 21 m/s in a suburban terrain for the operating
 592 range of $\alpha \leq 45^\circ$. As an example, the elevation angles of two heliostats positioned 100 m and 362 m to
 593 the north of a 100 m tower can differ by up to 15° when tracking throughout a day (Zeghoudi and
 594 Chermitti 2014). For a maximum gust wind speed of 20 m/s during a summer day (21 May to 22 July
 595 in northern hemisphere) from 8 am to 4 pm that would nominally stow the entire field in a suburban
 596 terrain (Figure 15b), the majority of in-field heliostats (100 m) close to the tower with $\alpha = 40\text{-}55^\circ$
 597 would need to be stowed. However, the heliostats in the outer region of the field (362 m) with $\alpha = 25\text{-}$
 598 40° could continue to operate throughout this period. It should be noted that such a partial stowing

599 strategy of the field can only be realised when the operating range of elevation angles is below the
600 elevation angle corresponding to the maximum operating load ($\alpha = 90^\circ$ for overturning moment).
601 Despite the favourable working conditions for the overturning moment at smaller elevation angles, such
602 a stowing strategy would need to be avoided for the operating hinge moment in Figure 14. This is
603 because a transition to stow (of the order of minutes) in the event of increasing wind speed would expose
604 the heliostat to the maximum operating hinge moment at $\alpha = 30^\circ$ and potential structural failure.
605 Characterisation of the dynamic effects, including torsional motions and displacements of the heliostat
606 resulting from the critical wind load cases investigated in the current study, warrants further
607 investigation to optimise the mass and cost of the heliostat support structure, pylon thickness and
608 foundation depth.



609

610 Figure 15. Ratios of the peak operating overturning moment M_y to the stow ultimate design ($\hat{U}_{st} = 40$ m/s at
611 $z = 10$ m) overturning moment as a function of operating elevation angle α and maximum operational design
612 wind speed (\hat{U}_{op} at $z = 10$ m) on a $6 \text{ m} \times 6 \text{ m}$ heliostat with $H/c = 0.5$ positioned in: (a) flat desert terrain with
613 $\alpha_{\bar{u}} = 0.12$, $I_u = 13\%$ and $G_u = 1.44$; (b) suburban terrain with $\alpha_{\bar{u}} = 0.2$, $I_u = 26\%$ and $G_u = 1.70$.

614 4. Conclusions

615 The maximum loads caused by turbulent wind conditions during the operation of a heliostat field
616 are an important design consideration to maximise the solar output of a power tower plant, while
617 maintaining structural integrity and performance of the high-cost, wind-sensitive heliostat components,
618 such as the drive units. The current study investigated the effect of the non-uniform pressure distribution
619 at turbulence intensities and gust factors representing a flat desert and a suburban terrain according to

620 the lowest 10 m of full-scale atmospheric boundary layer (ABL) data from the Engineering Sciences
621 Data Unit (ESDU) 85020 model. It was found that the maximum hinge moment coefficient was strongly
622 correlated to the position of the centre of pressure from the heliostat central elevation axis. The
623 maximum lift coefficient was at $\alpha = 30^\circ$ and $\beta = 0^\circ$ in agreement with Peterka and Derickson (1992),
624 whereas the maximum hinge moment coefficient was found at $\alpha = 15^\circ$ and $\beta = 180^\circ$ in a highly
625 turbulent flow with 26% turbulence intensity due to a larger movement of the centre of pressure toward
626 the leading edge with decreasing α . Furthermore, the movement of the centre of pressure on heliostats
627 with $\alpha \leq 30^\circ$ is larger at $\beta = 180^\circ$ than at $\beta = 0^\circ$. The increase in pressure on the lower surface of the
628 heliostat is caused by the “ground effect” observed on wings and flat plates with $H/c \leq 0.5$ at small
629 angles of attack. Hence, wind approaching the upward-facing ($\beta = 0^\circ$) and downward-facing ($\beta =$
630 180°) surfaces of a heliostat in the operating range of $\alpha = 15\text{-}30^\circ$ should be considered to determine
631 the maximum hinge moment on the elevation drive of a conventional pedestal-mounted heliostat.

632 The maximum hinge and overturning moment coefficients on the heliostat in the current study were
633 shown to follow a linear increase with the longitudinal turbulence of the approaching flow in the ABL.
634 The peak hinge moment coefficient was highly correlated to the position of the centre of pressure and
635 the elevation angle of the heliostat, whereas the overturning moment coefficient was largely dependent
636 on the maximum drag force with the maximum frontal projected area of the heliostat surface to the
637 wind. This is highlighted in the difference of the maximum movement of the centre of pressure
638 increasing with decreasing elevation angle from approximately 5% of the heliostat chord length at $\alpha =$
639 90° to more than 30% of the heliostat chord length in stow position ($\alpha = 0^\circ$). With a doubling of
640 turbulence intensity from 13% to 26%, the position of the centre of pressure due to the peak pressure
641 distribution at $\alpha = 30^\circ$ increases above the 10% threshold specified for square cross-section monoslope
642 roofs with $\alpha = 30^\circ$ in ASCE 7-02 (2002). The increased movement of the centre of pressure contributes
643 to the maximum hinge moment coefficients of 0.65 at $\alpha = 30^\circ$ and 0.76 at $\alpha = 15^\circ$ for wind at $I_u =$
644 26% approaching the windward ($\beta = 0^\circ$) and leeward ($\beta = 180^\circ$) edges of the heliostat, respectively.
645 The maximum hinge moments due to the centre of pressure movement investigated in the current study

646 were based on a single heliostat size, however the effect of the turbulence length scales with respect to
647 the heliostat dimensions warrants further investigation in future studies.

648 Application of the peak hinge moment coefficients on a 36 m² heliostat design, with drives that are
649 able to resist the maximum loads in stow and operating positions, to full-scale ABL data showed that
650 the maximum operational design wind speed can be increased to 29 m/s in a desert terrain and 33 m/s
651 in a suburban terrain for an ultimate design stow wind speed of 40 m/s. However, the maximum
652 operating loads at larger elevation angles $\alpha \geq 45^\circ$ are less than 60% and 70% of the stow load for the
653 same operating design wind speed specification in desert and suburban terrains, respectively. In
654 contrast, the maximum operating design wind speed that allows the maximum operating overturning
655 moment for $\alpha \leq 45^\circ$ to remain below the stow load is 18 m/s in a flat desert and 21 m/s in a suburban
656 terrain for the operating range of $\alpha \leq 45^\circ$. This presents an opportunity to increase the operating hours
657 of those regions of the heliostat field with favourable elevation-azimuth configurations that have smaller
658 operating loads than the maximum stow loads and would not expose the heliostat to the maximum
659 operating load during the transition to the stow position in the event of increasing wind speed. The
660 methodology and results for deriving the hinge and overturning moments in the current study can form
661 part of structural design considerations for determining the appropriate design wind speeds in different
662 terrains and the critical heliostat configurations that lead to the maximum design wind loads on the
663 elevation drive and foundation. The effects of turbulence intensity and the position of the centre of
664 pressure on the design hinge and overturning moments are critical for the strength and stiffness of the
665 elevation drive during operation and the structural rigidity of the pedestal and foundation in stow
666 position.

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