

Dynamic behavior of non-linear planetary gear model in non-stationary conditions

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Abstract:

The nonlinear effects in gearboxes are a key concern to describe accurately their dynamic behavior. This task is difficult for complex gear systems such as planetary gearboxes. The main aim of this work is provide responses to overcome this difficulty especially in non-stationary operating regimes by investigating a back-to-back planetary gearbox in steady conditions and in run up regime.

The nonlinear Hertzian contact of teeth pair is modeled in stationary and non-stationary run-up regime. Then it is incorporated to a torsional model of the planetary gearbox through the different mesh stiffness functions.

In addition, motor torque and external load variation are taken into account. The nonlinear equations of motion of the back-to-back planetary gearbox are computed through the Newmark- β algorithm combined with the method of Newton-Raphson. An experimental validation of the proposed numerical model is done through a test bench for both stationary and run-up regimes. The vibration characteristics are extracted and correlated to speed and torque. Time frequency analysis is implemented to characterize the transient regime during run-up.

Keys words: planetary gear, non-linearity, run up regime, stationary condition

Nomenclature

α_m	The operating pressure angle
F	The external applied force
G	Shear modulus
s_h	Shear factor
I_i	The inertia
A_i	Area moment
S_i	The cross section of tooth
ν	The Poisson ratio
E	Young's modulus
W	Tooth width
S_b	The slip at breakdown
T_b	the torque at breakdown,
a_1 and b_1	Motor constant properties
f_s	Sun rotational frequency
Z_s	Sun teeth number
Z_r	Ring teeth number
s	The proportional drop in the speed

Notation

HS	Hertzian Stiffness
BS	Bending Stiffness
FFS	Fillet Foundation Stiffness
TE	Transmission Error
FEM	Finite Element Method

1-Introduction:

Planetary gears are operated in several fields such as wind turbine and hybrid automotive where the performance of the whole system particularly depends on the split unit [1-3]. Many researchers proposed numerical models of planetary gears in order to ameliorate its dynamic behavior [4-13] and to have a better understanding of the vibratory signature, allowing the development of more efficient fault detection and diagnostic techniques [14-18].

The variable speed and load are the main non-stationary conditions in which planetary gears are usually running. These conditions externally excite the planetary gearbox [19]. For this, machinery dynamic behaviour for varying speed conditions is a very attractive research topic. Chaari et al., [20] studied planetary gearbox dynamic behaviour running under variable loading which induce variability in the motor speed. Obtained results are presented by using the STFT (Short Time Fourier Transform). Zimroz et al. [21] proposed an automatic time-frequency segmentation algorithm to highlight effects of load and speed varying conditions on planetary gearbox dynamics. Vicuña and Chaari [22] carried out running experiments of planetary gearbox under sinusoidal variable load to which the input speed is sensitive. They correlated these measurements with numerical results obtained from a dynamic model in which the gear mesh stiffness depends on the input speed. Lopatinskaia et al., [23, 24] proposed to analyze the vibration signal in the angle domain. Meltzer and Ivanov [25, 26] processed signals of a three-stage planetary gearbox test bench through time quefrequency techniques to detect tooth defects during start up and run down. Zimroz et al., [27] investigated the variable input wind power effect on gearbox vibrations of a wind turbine. They presented the STFT of the gearbox to show the frequency modulations. Viadero et al. [28] numerically studied the behavior of an offshore wind turbine gearbox which is subject to fast run-up and emergency run-down. During start up, the gear mesh stiffness between gears is modeled with sinusoidal functions where meshing period decreases in time. Hammami et al., [29-30] studied the planetary gearbox dynamic behavior during start up and stop regimes [29] and under variable speed [30]. The gear mesh stiffness sun-planets and planets-ring are modeled with rectangular waves respecting the meshing phase between planets. In these works [29, 30], the meshing period is decreasing when speed is increasing.

These cited works on time varying speed gearbox used linear gear mesh stiffness to model the teeth gear contacts which are the main internal excitations.

In general, the teeth contact are modeled by the gear mesh stiffness functions which are computed by considering only the teeth HS or considering both HS, BS of tooth and FFS.

In many researchers' works, the nonlinear contact between teeth is taken into account. In fact, Zhou et al., [31] investigated the nonlinear dynamic effects of backlash, friction and load of gear-rotor-bearing system through a lateral-torsional model and they highlighted the quasi-periodic behavior due to these effects. The sixteen d-o-f responses are computed by using Runge-Kutta algorithm. In addition, Chen et al., [32] computed the non-linear vibration response of a face gearbox which is excited by the mesh stiffness, backlash and stiffness of supports. They also used Runge-Kutta algorithm to compute the non-linear equation of motion. Fernandez et al., [33] used the hertzian contact theory and the FEM method to compute the load TE and the mesh stiffness function of spur gear.

The modeling of the non-linear planetary gear system is extensively studied nowadays. Guo et al. [34] examined the stability and the nonlinear behavior of wind turbine planetary gearbox by using the FEM and lumped parameters methods in which the corresponding dynamic response is computed with the help of the extended harmonic balance method. They correlated the analytical and the numerical results to those obtained experimentally. Zhao and Ji [35] have concluded that the external excitation and the mesh stiffness are the main significant factors which have influence on the nonlinear dynamic behavior of wind turbine gearbox which is computed by the numerical integration method. Guo and Parker [36] showed that the non-linearity caused by bearing clearance can decrease the resonances in helicopter gear sets. Also, the dynamic behavior of spur planetary gear in different backlashes is investigated by Liu et al., [37] through a lumped-parameter model which considered the nonlinearity introduced by the gravity effect and bearing oil film. They used Newmark integration to calculate the dynamic responses. Nevertheless, dynamic models of planetary gears should include a non-linear mesh stiffness function and gear contact loss nonlinearity [38-41]. The influence of nonlinear jumps and the chaotic motions on dynamic behavior of planetary gearbox is studied by Ambarisha and Parker [42] who correlated their analytical results with FEM results.

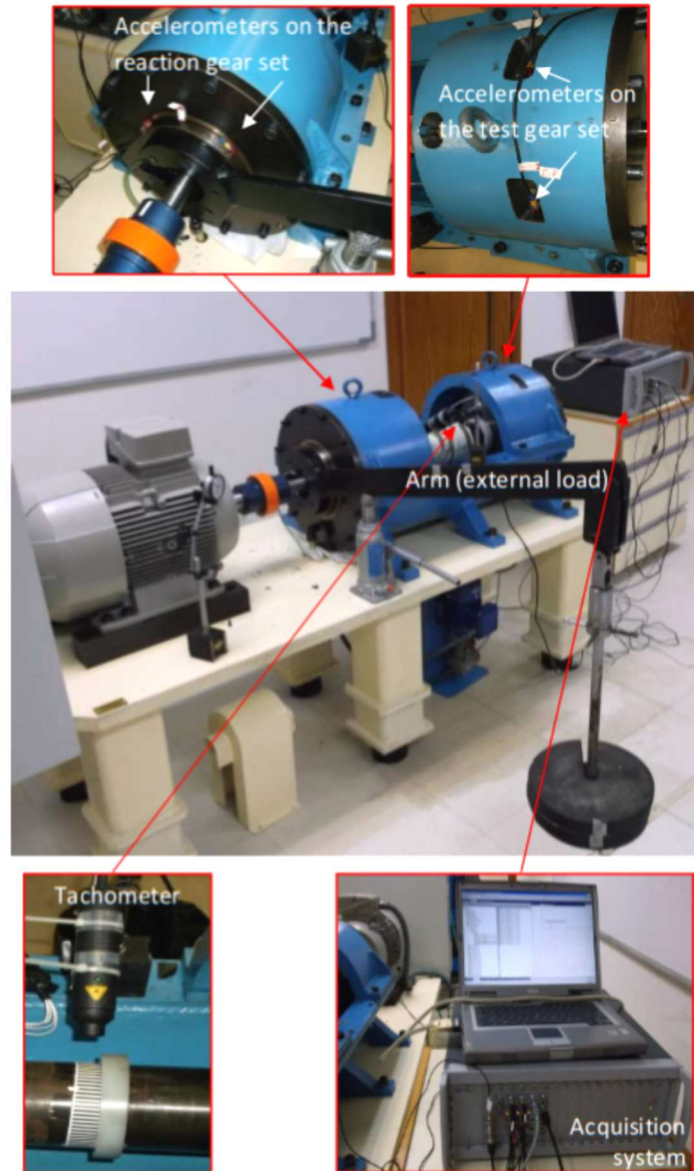
Most nonlinear cited models are exploited under stationary operation where the speed is constant and used Runge-Kutta method or Newmark integration to compute the nonlinear dynamic response. Although, Bouchaala et al., [43] investigated the effect of nonlinear Hertzian gear contact on the vibration behavior of one-stage spur gearbox in acyclism regime, their work presents only a theoretical contribution without any experimental validation.

In the concrete conditions, the contact between teeth in planetary gearbox needs to be considered in the dynamic models especially when the gearbox is running in the non-stationary conditions. Thus, the non-linear hertzian contact between teeth is a basic condition to refine the gear dynamic analysis and to allow a better diagnosis when the driven speed or the applied

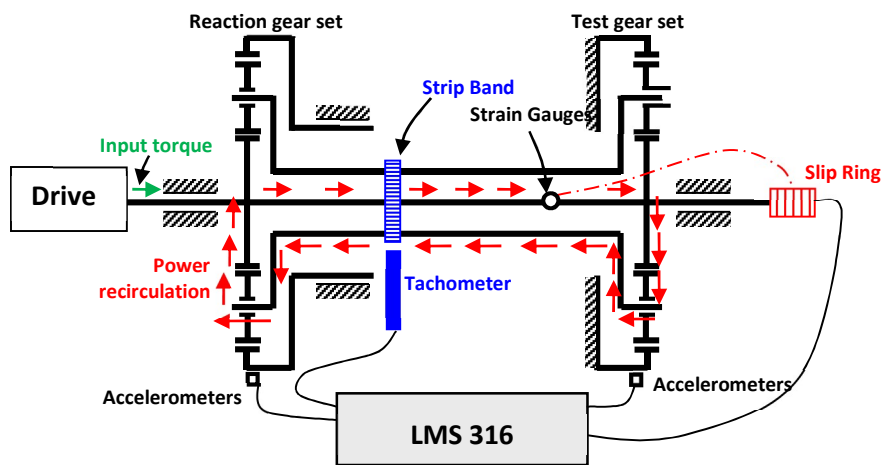
loads are variable. The aim of this work is to check the influence of considering non-linear (HS) between mating teeth on the vibration behavior of planetary gearbox. A corresponding nonlinear model is developed to study this effect. Using Newmark-Newton Raphson integration method, the dynamic response is computed under two regimes: stationary conditions and variable speed conditions and the example of run-up regime will be studied in this case. Numerical results are compared to those obtained from experience through a planetary gearbox test bench.

2- Test bench description:

The test bench is configured in a two-spur identical planetary gear with 3 planets (Fig. 1). The two gear sets are linked in a back-to-back configuration through a rigid shaft and a hollow shaft respectively called sun's shaft and carrier shafts. This special configuration allows minimizing costs and improving energy efficiency. The test gear set is the main planetary gear set in which output power from test carrier is reintroduced to the input test sun with the help of the reaction gear set [47]. Moreover, an external torque can mechanically be applied to the reaction ring gear by adding masses on the arm.



(a)



(b)

Fig. 1 (a) Planetary gearbox test bench (b) instrumentation layout

The driving electric motor is connected to the reaction gear with a rigid shaft and it is controlled by a variable frequency converter which is configured with “STARTER” software. Four accelerometers are fixed on each ring. In addition, an optic tachometer recorded the instantaneous angular velocity of the hollow shaft. The signals coming from accelerometers and tachometer are acquired by an acquisition LMS SCADAS 316 system. Additional key parameters of this test bench are illustrated in table 1. Further details about the test bench can be found in [44-46].

Table 1 Basic dimensions of planetary gear

	Ring	Planet (3)	Sun	Carrier
Number of teeth	65	24	16	-
Mass (Kg)	28.1	1.225	0.485	3.643
Moment of inertia (Kgm²)	$697767 \cdot 10^{-6}$	$2045 \cdot 10^{-6}$	$356 \cdot 10^{-6}$	$21502 \cdot 10^{-6}$
Base diameters (m)	0.2494	0.0921	0.0614	0.1728
Tip diameters (m)	0.2579	0.1006	0.0699	0.1813
Module (mm)		4.23		
Pressure angle (rad)		0.4621		

3-Modelling of gear mesh-stiffness:

The gear mesh stiffness of both sun-planets and ring-planets are modeled by putting in series the different stiffness (figure 2): BS “ K_b ”, the FFS “ K_f ” and the HS “ K_h ”.

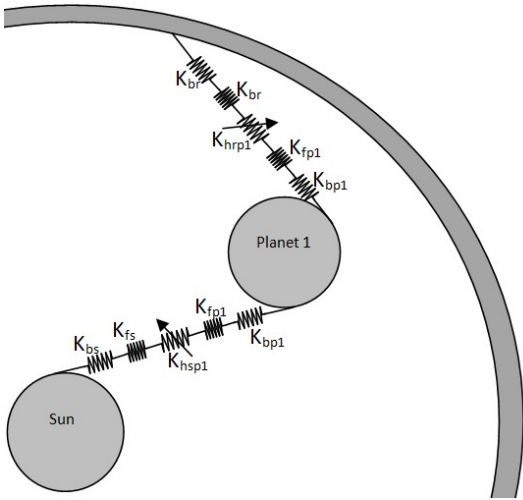


Fig. 2 Nonlinear mesh stiffness sun-planet 1 and ring-planet 1

$$K_{jik} = \frac{1}{\frac{1}{K_{bipk}} + \frac{1}{K_{fipk}} + \frac{1}{K_{bi}} + \frac{1}{K_{fi}} + \frac{1}{K_{hi}}} \quad (1)$$

Where $i=r,t$ (r for the reaction gear set and t for the test gear set), $i=s, r$ (s for the sun gear and r for the ring gear) and $k=1..3$ (planet order). The variable mesh stiffness $K_i(t)$ can be quantified based on the procedure given by Velex and Flamand [40]. The mesh period is defined by:

$$T_s = \frac{Z_s Z_r N}{(Z_s + Z_r) 60} \quad (2)$$

1-3-Bending deflection

Based on the results obtained by Cornell [48], the bending deflection δ_b is expressed as:

$$\delta_b = F \cos^2 \alpha_m \sum_{i=1}^n e_i \left\{ \frac{d_i - e_i d_i + \frac{e_i^2}{3}}{E' I_i} + \frac{1}{s_h G A_i} + \frac{\tan^2 \alpha_m}{E' A_i} \right\} \quad (3)$$

e_i is the considered segment width of the tooth which is supposed segmented uniform cantilever beam. d_i is the distance between the considered segment and the cross point between the tooth symmetric line and the line of action.

E' , I_i and A_i are defined as following :

$$E' = \frac{E(1-\nu)}{(1+\nu)(1-2\nu)} \quad (4)$$

$$\frac{1}{I_i} = \frac{\frac{1}{I_i} + \frac{1}{I_{i+1}}}{2} \quad (5)$$

$$\frac{1}{A_i} = \frac{\frac{1}{A_i} + \frac{1}{A_{i+1}}}{2} \quad (6)$$

The corresponding (BS) can be obtained by:

$$k_b = \frac{F}{\delta_B} \quad (7)$$

2-3-Fillet foundation deflection

It is modeled according to Muskhelishvili theory [49]:

$$\delta_f = \frac{F \cos^2 \alpha_m}{WE} \left\{ L^* \left(\frac{u_f}{S_f} \right)^2 + M^* \left(\frac{u_f}{S_f} \right) + P^* (1 + Q^* \tan^2 \alpha_m) \right\} \quad (8)$$

Where S_f is the dedendum surface of the tooth and u_f is the distance between the dedendum circle and the crossing point between the tooth symmetric line and the line of action.

L^* , M^* , P^* and Q^* are polynomial functions defined by Sainsot et al. [50].

$$X_i^*(h_{fi}, \theta_f) = \frac{A_i}{\theta_f^2} + \frac{B_i}{h_{fi}^2} + \frac{C_i h_{fi}}{\theta_f} + \frac{D_i}{\theta_f} + E_i h_{fi} + F_i \quad (9)$$

$h_{fi} = r_f/r_{int}$, r_f and θ_f are respectively the radius of the dedendum circle and the dedendum angular pitch of the tooth.

The (FFS) can be deduced by:

$$K_f = \frac{F}{\delta_f} \quad (10)$$

3-3-Contact deflection

The following equation presents the Hertzian elastic deformation in the line contact derived by Harsha [51]:

$$\delta_h = \frac{4.05F_0^{0.925}}{10^5 l_{eff}^{0.85}} \quad (11)$$

Where l_{eff} is the length of contact between teeth. The deflection contact force is indicated by:

$$F_h = 56065.703 l_{eff}^{0.92} \delta_h^{1.08} \quad (12)$$

Thus, the non-linear deflection stiffness is given by:

$$K_h = \frac{F_h}{\delta_h} = 56065.703 l_{eff}^{0.92} \delta_h^{0.08} \quad (13)$$

4-Numerical model:

A torsional model corresponding to the test bench is developed [47] and it is shown in Fig.3.

This paper assumes that all planetary gears component has a rotational rigid motion.

The model consisted of two similar planetary gear sets called reaction gear and test gear respectively. The reaction ring is located near the motor, it is characterized as a free ring.

Two rigid shafts were used to link the inner parts (suns and carriers) and a rigid housing was used to link the external parts (ring). The sun reaction gear is connected to input shaft and the free ring is connected to a rigid arm and extorted by the external torque.

The mass and inertia of each components were denoted m_c , I_c , m_s , I_s , m_r , I_r , m_{p1} , I_{p1} , m_{p2} , I_{p2} , m_{p3} , I_{p3} for the carrier, the sun, the ring and the three planets respectively.

The planets were linked to the reaction ring by the ring planets mesh stiffness K_{rt1} , K_{rt2} and K_{rt3} and to the reaction sun by the sun planets mesh stiffness K_{sr1} , K_{sr2} and K_{sr3} respectively. The same functions were used in the test gear. They are K_{rt1} , K_{rt2} and K_{rt3} ring planets mesh stiffness and K_{st1} , K_{st2} and K_{st3} sun planets mesh stiffness.

The sun's shaft and the carrier's shaft are respectively modeled by a torsional stiffness k_{st} and k_{ct} .

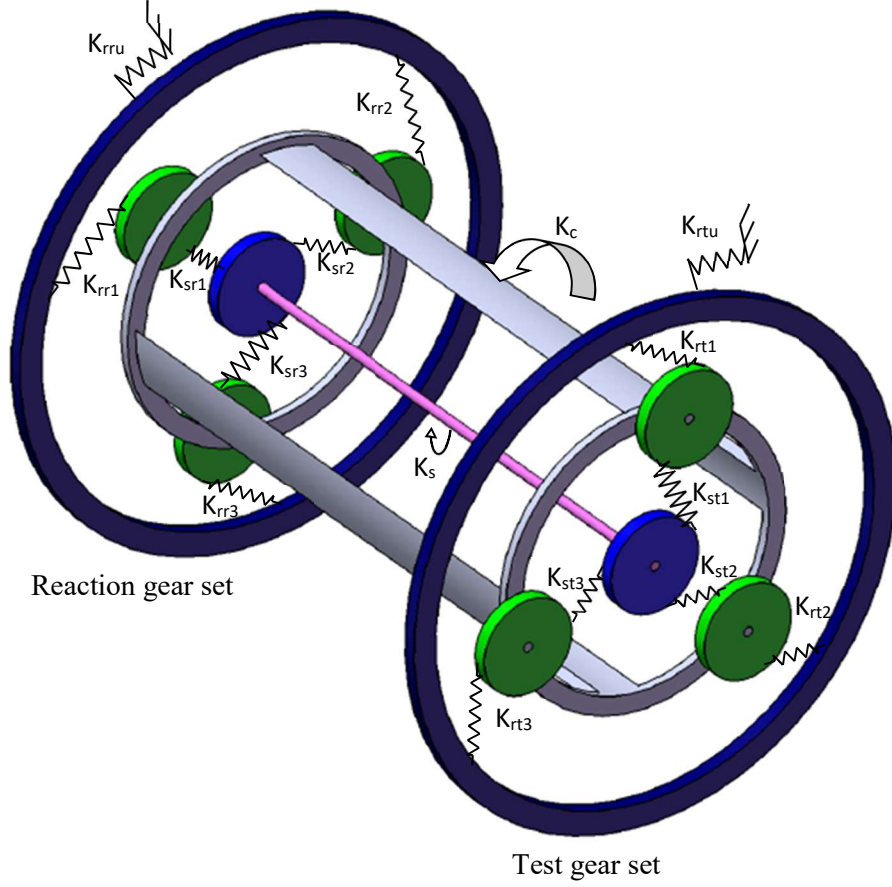


Fig. 3 Planetary gear torsional model

The corresponding equation of motion is:

$$M\ddot{q} + C\dot{q} + K_c q + F_{nl} = F_{ext}(t) \quad (14)$$

Only the rotational movements were considered.

$$q = \{\theta_{cr} \ \theta_{rr} \ \theta_{sr} \ \theta_{p1r} \ \theta_{p2r} \ \theta_{p3r} \ \theta_{ct} \ \theta_{rt} \ \theta_{st} \ \theta_{p1t} \ \theta_{p2t} \ \theta_{p3t}\} \quad (15)$$

q is the d-o-f vector of the system, M denotes the mass matrix.

$$M = \begin{bmatrix} M_r & 0 \\ 0 & M_t \end{bmatrix} \quad (16)$$

$$M_i = \text{diag} \left(\frac{I_{ci}}{r_{ci}} + n \cdot m_p \frac{I_{ri}}{r_{ri}} \ \frac{I_{si}}{r_{si}} \ \frac{I_{p1i}}{r_{p1i}} \ \frac{I_{p2i}}{r_{p2i}} \ \frac{I_{p3i}}{r_{p3i}} \right) \quad i=r,t \quad (17)$$

$$K(t) = \begin{bmatrix} K_{mr} & 0 \\ 0 & K_{mt} \end{bmatrix} + K_c \quad (18)$$

$$K_{mr} = \begin{bmatrix} \sum_{i=1}^3 (K_{sri}(t) + K_{rri}(t)) & -\sum_{i=1}^3 K_{rri}(t) & -\sum_{i=1}^3 K_{rri}(t) & K_{rr1}(t) - K_{sr1}(t) & K_{rr2}(t) - K_{sr2}(t) & K_{rr3}(t) - K_{sr3}(t) \\ -\sum_{i=1}^3 K_{rri}(t) & \sum_{i=1}^3 K_{rri}(t) & 0 & -K_{rr1}(t) & -K_{rr2}(t) & -K_{rr3}(t) \\ -\sum_{i=1}^3 K_{sri}(t) & 0 & \sum_{i=1}^3 K_{sri}(t) & K_{sr1}(t) & K_{sr2}(t) & K_{sr3}(t) \\ K_{rr1}(t) - K_{sr1}(t) & -K_{rr1}(t) & K_{sr1}(t) & K_{sr1}(t) + K_{rr1}(t) & 0 & 0 \\ K_{rr2}(t) - K_{sr2}(t) & -K_{rr2}(t) & K_{sr2}(t) & 0 & K_{sr2}(t) + K_{rr2}(t) & 0 \\ K_{rr3}(t) - K_{sr3}(t) & -K_{rr3}(t) & K_{sr3}(t) & 0 & 0 & K_{sr3}(t) + K_{rr3}(t) \end{bmatrix} \quad (19)$$

K_{mt}

=

$$\begin{bmatrix} \sum_{i=1}^3 (K_{sri}(t) + K_{rri}(t)) & -\sum_{i=1}^3 K_{rri}(t) & -\sum_{i=1}^3 K_{rri}(t) & K_{rr1}(t) - K_{sr1}(t) & K_{rr2}(t) - K_{sr2}(t) & K_{rr3}(t) - K_{sr3}(t) \\ -\sum_{i=1}^3 K_{rri}(t) & \sum_{i=1}^3 K_{rri}(t) & 0 & -K_{rr1}(t) & -K_{rr2}(t) & -K_{rr3}(t) \\ -\sum_{i=1}^3 K_{sri}(t) & 0 & \sum_{i=1}^3 K_{sri}(t) & K_{sr1}(t) & K_{sr2}(t) & K_{sr3}(t) \\ K_{rr1}(t) - K_{sr1}(t) & -K_{rr1}(t) & K_{sr1}(t) & K_{sr1}(t) + K_{rr1}(t) & 0 & 0 \\ K_{rr2}(t) - K_{sr2}(t) & -K_{rr2}(t) & K_{sr2}(t) & 0 & K_{sr2}(t) + K_{rr2}(t) & 0 \\ K_{rr3}(t) - K_{sr3}(t) & -K_{rr3}(t) & K_{sr3}(t) & 0 & 0 & K_{sr3}(t) + K_{rr3}(t) \end{bmatrix} \quad (20)$$

The damping matrix "C" is defined as:

$$C = \alpha M + \beta K \quad (21)$$

α and β are two constants.

The external torque vector F_{ext} is expressed as:

$$F_{ext} = \{0 \quad 0 \quad C_m(t) \quad 0 \quad 0 \quad 0 \quad 0 \quad 0 \quad C_r(t) \quad 0 \quad 0 \quad 0\} \quad (22)$$

The nonlinear force induced by the HS and expressed as following:

$$F_{nl} = \{X\} K_i(t) \{X\}^T \{q\} = \{X\} K_i(t) \delta \quad (23)$$

This force is defined as:

$$F_{nl} = \{F_{nl,r} \quad F_{nl,t}\}^T \quad (24)$$

$$F_{nl,j}(t) = \{0 \quad F_{rj}(t) \quad F_{sj}(t) \quad F_{p1j}(t) \quad F_{p2j}(t) \quad F_{p3j}(t)\}^T \quad (25)$$

$$F_{rj}(t) = -\sum_{i=1}^n K_{rij}(t) \delta_{ri}^j(t) \{r_{1i} \quad r_{2i} \quad r_{3i}\}^T \quad (26)$$

$$F_{sj}(t) = -\sum_{i=1}^n K_{sij}(t) \delta_{si}^j(t) \{s_{1i} \quad s_{2i} \quad s_{3i}\}^T \quad (27)$$

$$F_{ij}(t) = -K_{rij}(t)\delta_{ri}^j(t)\{r_{4i} \ r_{5i} \ r_{6i}\}^T - K_{sij}(t)\delta_{si}^j(t)\{s_{4i} \ s_{5i} \ s_{6i}\}^T, \text{ avec } i=1..3 \quad (28)$$

The tooth deflection components are highlighted in table 2.

Table 2 Tooth deflection components

Coefficients s_i	Coefficients r_i
$s_{1i}=s_{2i}=s_{3i}=-\Gamma_{br}$	$r_{1i}=r_{2i}=r_{3i}=-\Gamma_{bp}$
$s_{4i}=s_{5i}=s_{6i}=\Gamma_{bs}$	$r_{4i}=r_{5i}=r_{6i}=\Gamma_{bp}$

With:

$$\{X\} = \{r_{bcr} \ r_{brr} \ r_{bsr} \ r_{bp1r} \ r_{bp2r} \ r_{bp3r} \ r_{bct} \ r_{brt} \ r_{bst} \ r_{bp1t} \ r_{bp2t} \ r_{bp3t}\}^T \quad (29)$$

The TE “ δ ” is defined by Velex and Flamand [26]:

$$\delta_{ri}^r = r_{brr}\theta_{rr} + r_{bir}\theta_{ir} \quad (30)$$

$$\delta_{ri}^t = r_{brt}\theta_{rt} + r_{bit}\theta_{it} \quad (31)$$

$$\delta_{si}^r = r_{bsr}\theta_{sr} + r_{bir}\theta_{ir} \quad (32)$$

$$\delta_{si}^t = r_{bst}\theta_{st} + r_{bit}\theta_{it} \quad (33)$$

Characteristics of the mechanical transmissions are shown on table 3.

Table 3 Model parameters.

Torsional Shaft Stiffness (Nm/rd)	
Sun	$3.73 \cdot 10^4$
Carrier	$8.38 \cdot 10^5$
Gear Mesh stiffness[N/m]	
Sun-planets	$3.5 \cdot 10^8$
Planets-ring	$4.5 \cdot 10^8$

The equation of motion is computed using the classical Newton Raphson method coupled with the implicit Newmark algorithm.

Before applying the Newmark method, we need firstly to compute the initial value correspondent to $X_0, \dot{X}_0, \ddot{X}_0$. Then, the equation of motion (14) is transformed to an approximated version ($t = t_{n+1}$):

$$M\{\ddot{q}\}_{n+1} + C\{\dot{q}\}_{n+1} + K_c\{q\}_{n+1} + F_{nl}\{X\}_{n+1} = \{F_{ext}(t)\}_{n+1} \quad (34)$$

Firstly, the initial displacement, velocity and acceleration are introduced:

$$d_0 = q(0), v_0 = \dot{q}(0), a_0 = \ddot{q}(0)$$

$$\{X\}_{i+1} = \{X\}_i + \Delta t \{\dot{X}\}_i + b \Delta t^2 \left[\left(\frac{1}{2b} - 1 \right) \{\ddot{X}\}_i + \{\ddot{X}\}_{i+1} \right] \quad (35)$$

$$\{\dot{X}\}_{i+1} = \{\dot{X}\}_i + \Delta t(1-a)\{\ddot{X}\}_i + a \Delta t \{\ddot{X}\}_{i+1} \quad (36)$$

Δt denotes the time step which is chosen as 10^{-4} second.

Extracting $\{\ddot{X}\}_{i+1}$ from equation (35):

$$\{\ddot{X}\}_{i+1} = \frac{1}{b \Delta t^2} \left[\{X\}_{i+1} - \{X\}_i - \Delta t \{\dot{X}\}_i \right] - \left(\frac{1}{2a} \{X\}_i - 1 \right) \{\ddot{X}\}_i \quad (37)$$

(37) and (38) gives:

$$[\bar{K}]\{X\}_{i+1} = \{\bar{F}\}_{i+1} \quad (38)$$

Where

$$\begin{aligned} \{\bar{F}\}_{i+1} = & \{F\}_{i+1} + [M] \left(\frac{1}{b \Delta t^2} \{X\}_i + \frac{1}{b \Delta t} \{\dot{X}\}_i + \left(\frac{1}{2b} - 1 \right) \{\ddot{X}\}_i \right) + [C] \left(\frac{1a}{b \Delta t} \{X\}_i + \right. \\ & \left. \left(\frac{a}{b} - 1 \right) \{\dot{X}\}_i + \left(\frac{\Delta t}{2} \right) \left(\frac{b}{a} - 1 \right) \{\ddot{X}\}_i \right) \end{aligned} \quad (39)$$

So, the residue is computed as follows:

$$R = [\bar{K}]\{X\}_{i+1} + F_{nl} - \{\bar{F}\}_{i+1} \quad (40)$$

To ensure the convergence of the Newton Raphson method coupled with the implicit Newmark algorithm, R should be higher than ε ($R > \varepsilon$). ε is a small predefined value.

If this criterion is not verified, the residue R is defined at the k+1 iteration by a Taylor expansion:

$$R_{i+1}^{k+1} = R_{i+1}^k + \left. \frac{\partial R}{\partial q} \right|_{i+1}^k \Delta q \quad (41)$$

R_{i+1}^{k+1} must be zero, so Δq is defined as:

$$\Delta q = \left(\left. \frac{\partial R}{\partial q} \right|_{i+1}^k \right)^{-1} (-R_{i+1}^k) \quad (42)$$

$\{X\}_{i+1}$, $\{\dot{X}\}_{i+1}$ and $\{\ddot{X}\}_{i+1}$ are identified at the k+1 iteration when Δq was computed.

5-Results:

The presented results are obtained in two operating conditions: steady condition then the run up condition regime.

Using the Euler method, fig. 4 shows the measured speed evolution of the electrical motor. Two distinguished distinct regimes A and B are shown. (A) is a stationary regime while (B) is a non-

stationary one, specifically a run-up regime which is characterized by the input shaft speed increase as shown in Figure 4.

Fig. 5 displays the measured driving torque during the run-up regime.

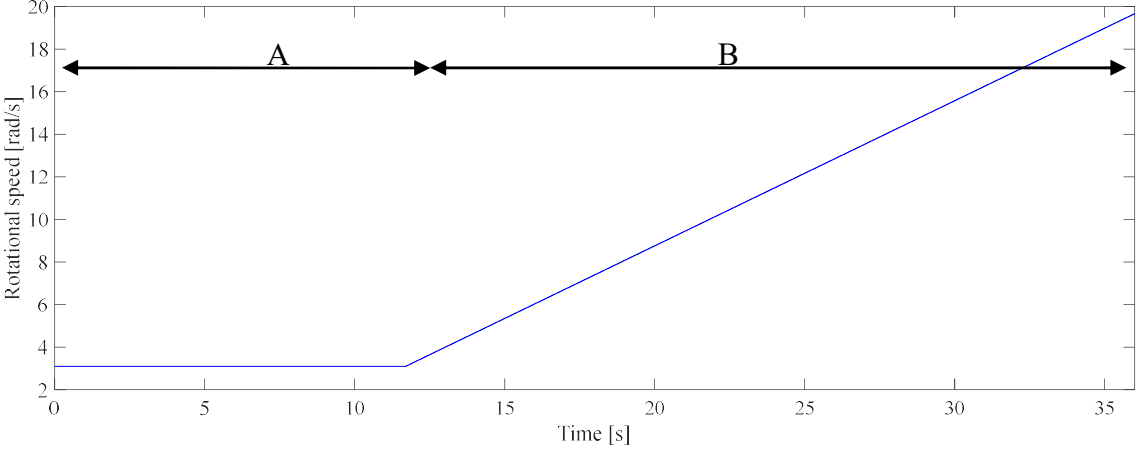


Fig. 4 Speed evolution of electrical motor

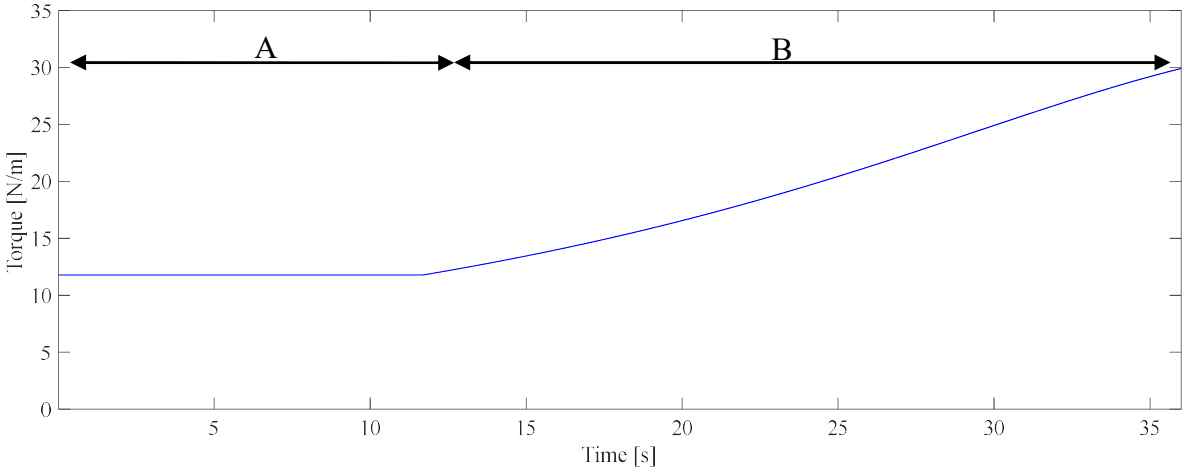
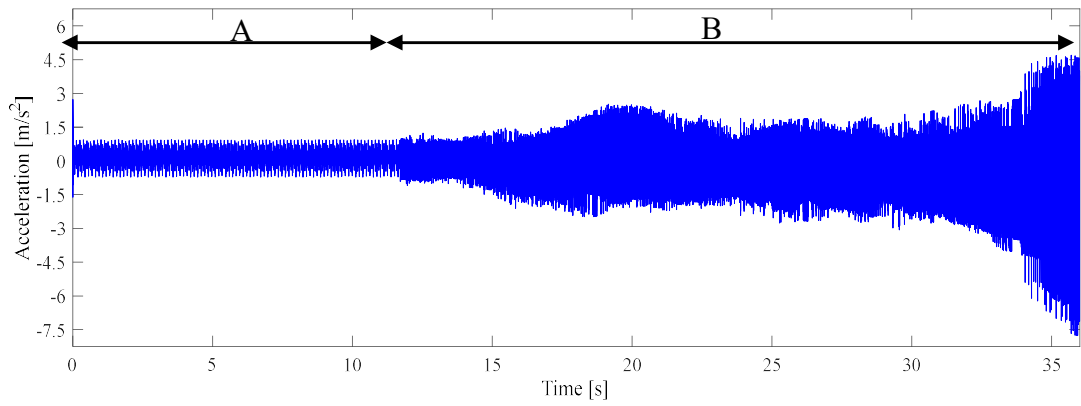


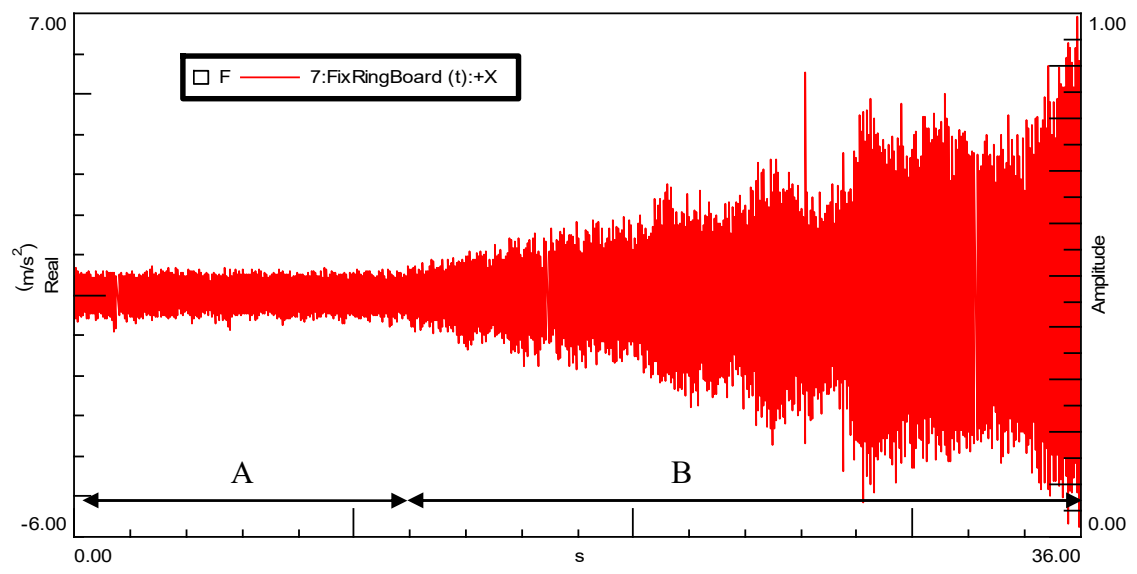
Fig. 5 Driven system mechanical characteristic

The input torque and the speed evolution presented in the two previous figures are induced in the numerical model in order to get the same running conditions.

Fig. 6 displays the acceleration on the fixed ring. The signal can be divided into two parts similarly to the evolution of the speed signal: part (B) presents run up régime, during this regime we can notice that the vibration is increasing. This phenomenon is due to the increasing of the accelerating torque.



(a)

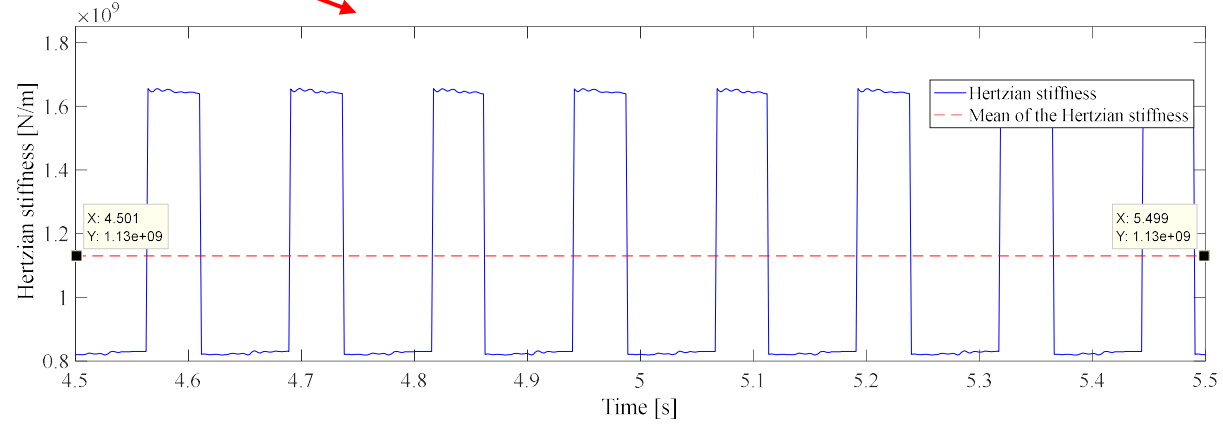
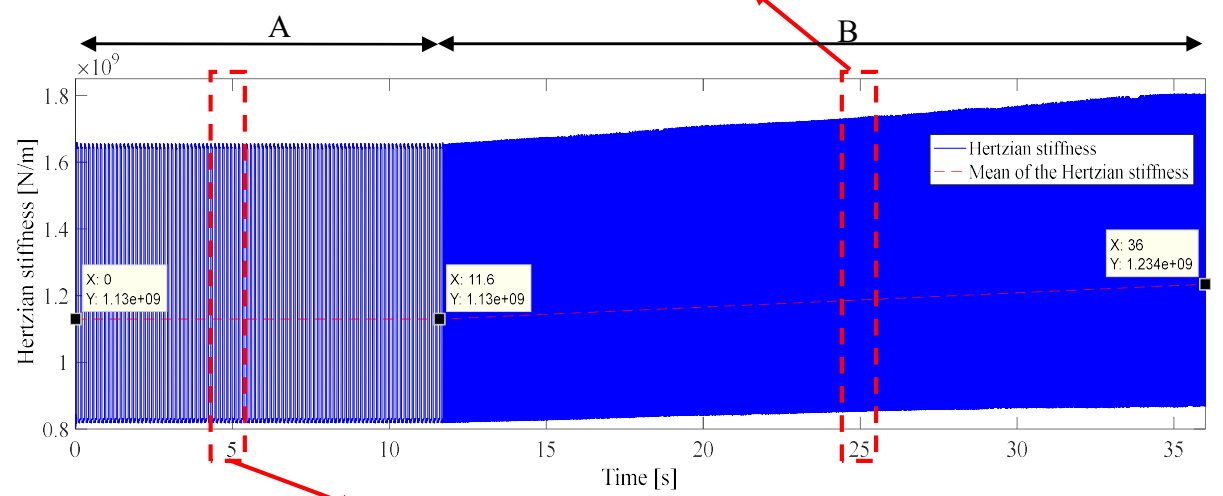
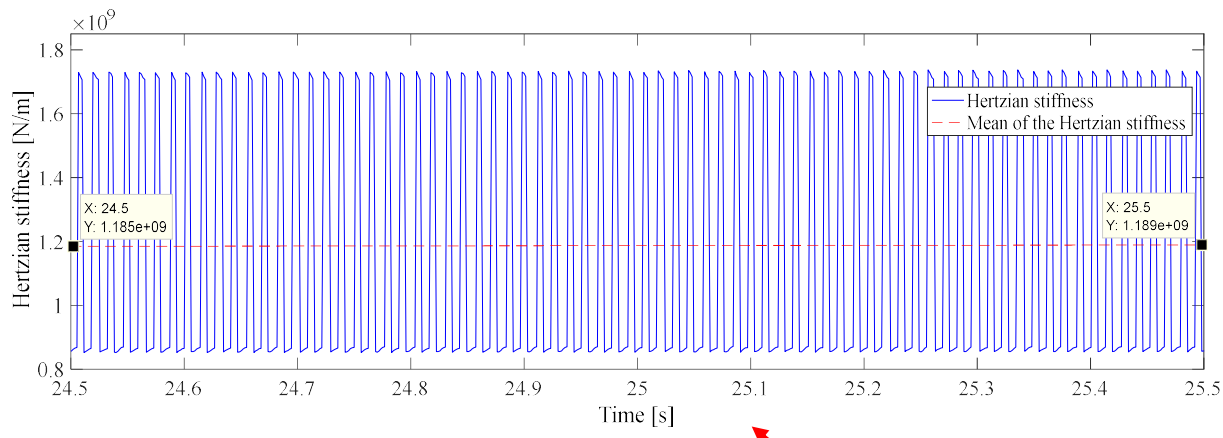


(b)

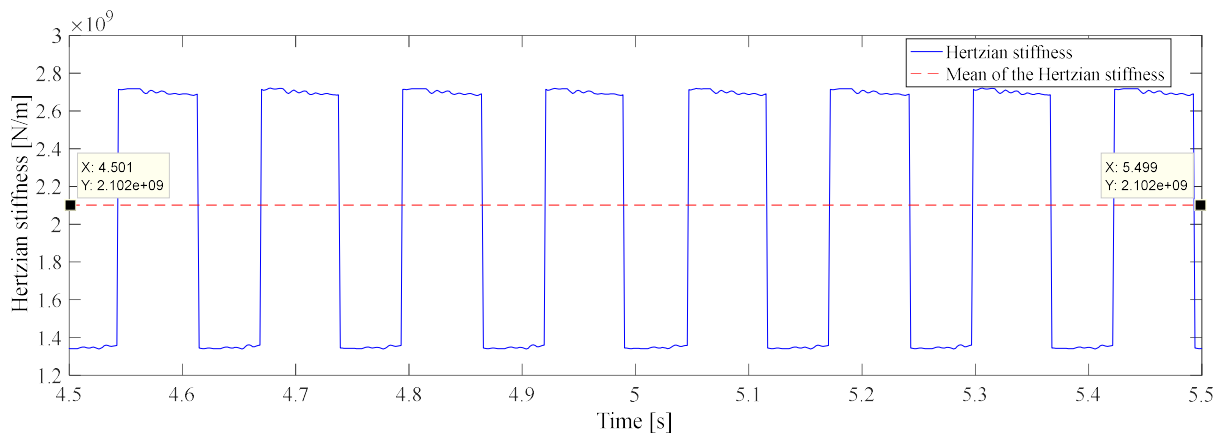
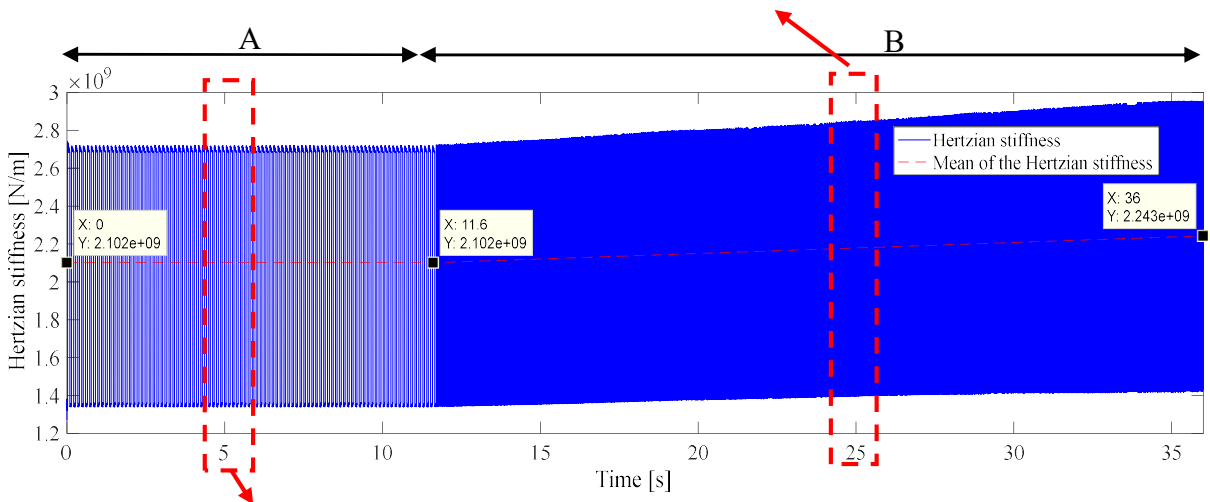
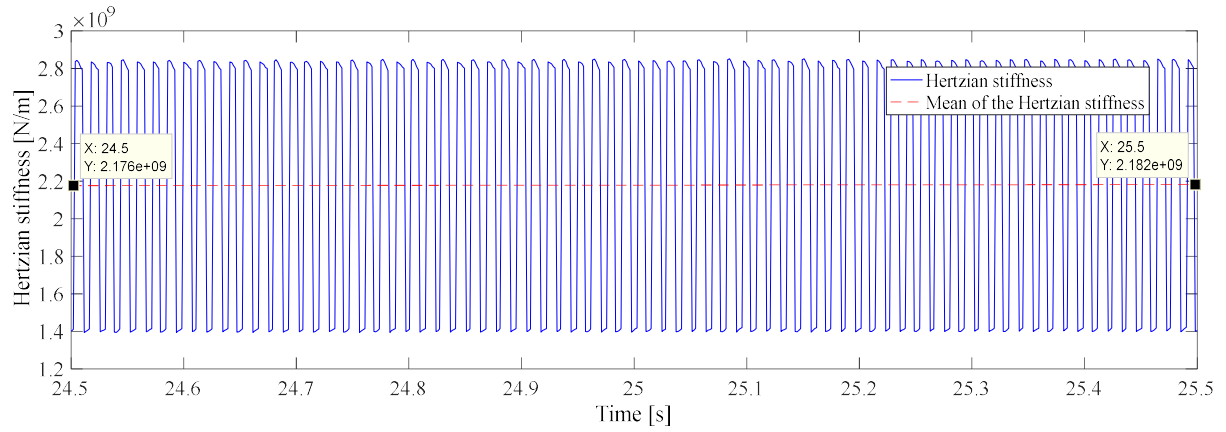
Fig. 6 Accelerations of the fixed ring (a) numerical result (b) experimental result

In part B while the speed is increasing (fig. 6), the acceleration signal time series have some amplitude modulations which are explained by crossing of the rotation frequency of motor with one of natural frequencies.

Fig. 7 shows the computed (HS) functions of the ring-planet1 and sun-planet1 respectively. The amplitude of HS is variable. The (HS) evolution is sensitive to the input torque applied to the planetary gear transmission. Thus, the mean value of this stiffness starts constant before acquiring its maximum in the run-up regime.



(a)

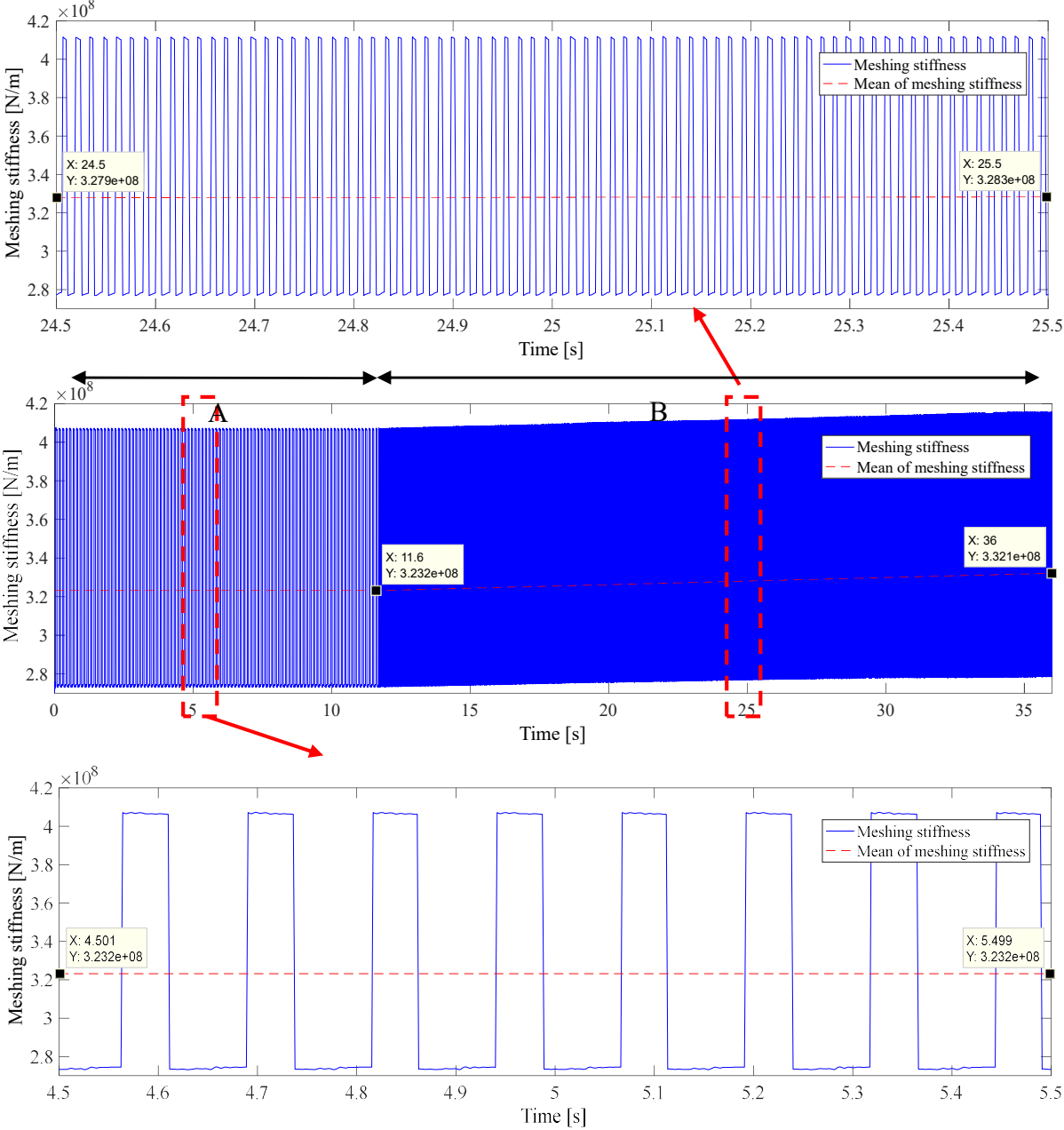


(b)

Fig. 7 (HS) evolution (a) sun-planet 1 (b) fixed ring-planet 1

According to equation (1), the (HS) of both sun-planets and planets-ring contacts is coupled to the (BS) and (FFS). Two (HS) behaviors of both sun-planets and planets-ring contacts are observed: in zone A, the mean and the period of fluctuation of (HS) is constant whereas the mean is increasing during run up (zone B) and the period of fluctuation is decreasing. In fact, the mean of (HS) of sun-planet 1 contact and ring-planet 1 contact **are** respectively increasing from 1.13×10^9 N.m to 1.234×10^9 N.m and from 2.102×10^9 N.m to 2.243×10^9 N.m.

This coupling allows computing of the gear mesh stiffness. Fig. 8 displays the gear mesh stiffness evolutions of sun-planet 1 and ring-planet 1 in which the mean evolutions are not the same in both regimes.



(a)

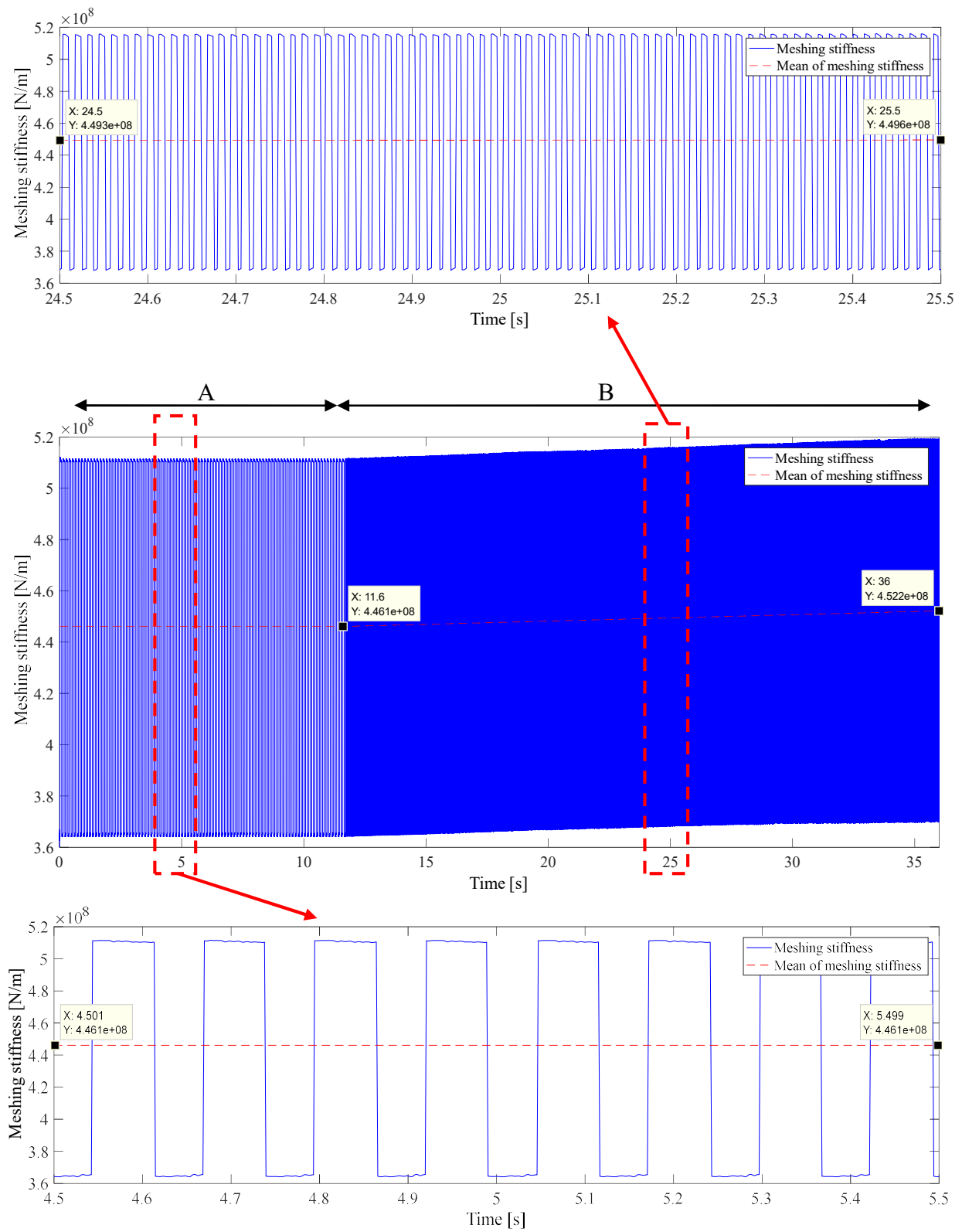


Fig. 8 Mesh stiffness evolution, (a) sun-planet 1 (b) fixed ring-planet 1

The gear mesh stiffness is steady in zone A in both sun-planets and planets-ring contacts whereas its mean values are increasing during run up.

To prove the influence of run-up regime on gear teeth, the (TE) and the inter mesh forces were carried out for all gear functions. The (TE) time evolution function matched to the ring-planet1 is presented in fig. 9. The (TE) attains a maximum value in the period (B). Nevertheless, they are constant during the first period (A).

The dynamic forces on teeth are computed according to the following equation:

$$F_d(t) = K_e(t)\delta(t) \tag{38}$$

Fig. 10 shows the meshing force between the ring and planet1. An overload behavior on teeth is observed during the run-up regime and it can cause defects [52].

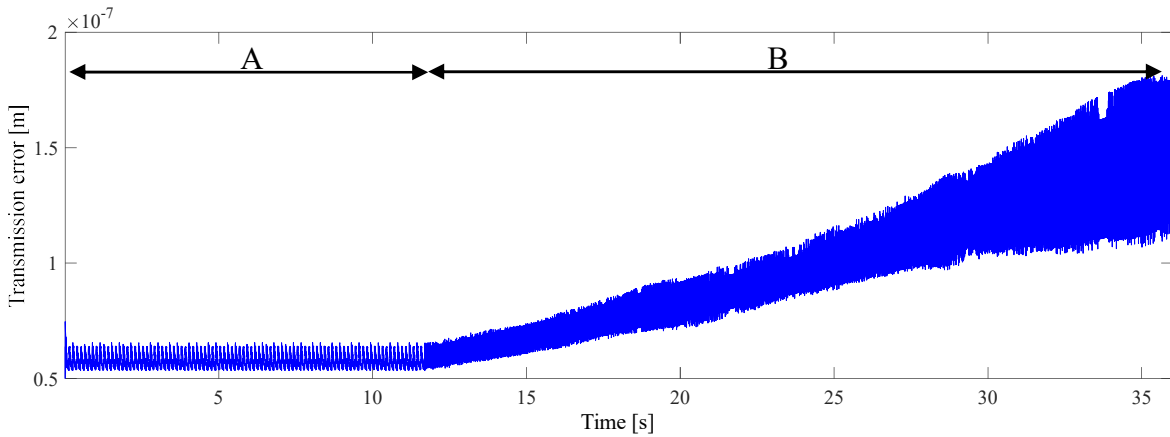


Fig. 9 (TE) function between the fixed ring and planets 1

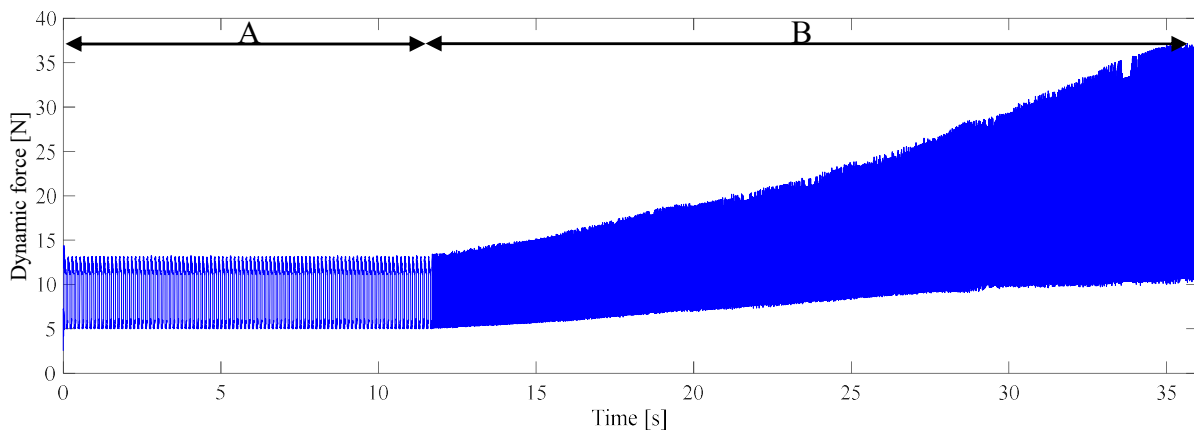


Fig. 10 Fixed ring-planet 1 meshing force

Fig. 11 displays a time-frequency map of the test ring acceleration with numerical simulation and experimental test. The two obtained behavior in time responses are also presented in both numerical and experimental results. From this figure, two different behaviors corresponding to the two parts A and B are shown. In part A, vertical lines appear, these lines show the stationary

regime, during which the speed is constant. In part B, the sloping lines show the mesh frequency with its harmonics increasing.

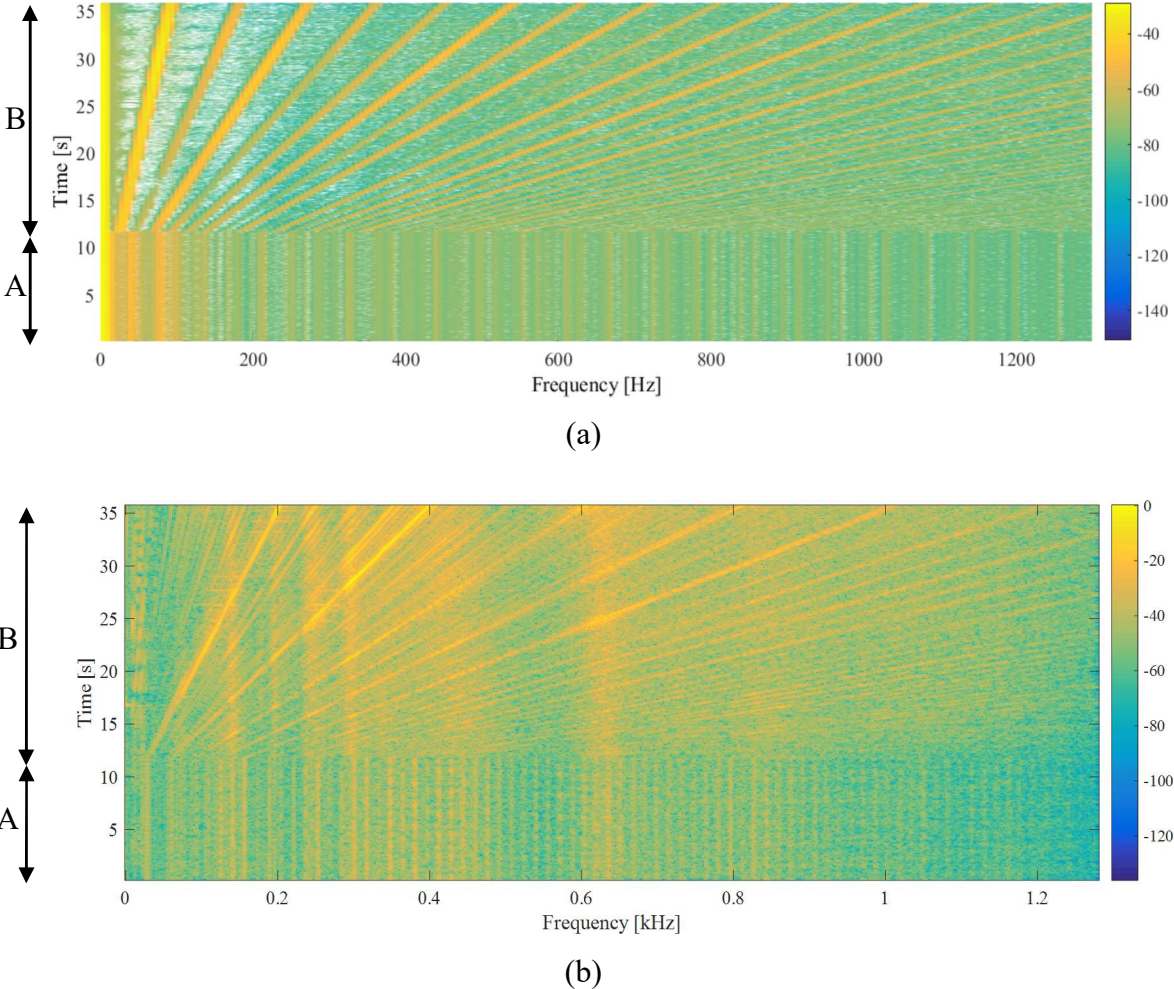


Fig. 11 STFT of acceleration of the fixed ring (a) numerical result (b) experimental result

6-Conclusion:

This paper examined the non-linear dynamic behavior of a back-to-back planetary gearbox transmission during stationary condition and then the run up regime. The objective was to characterize the dynamic behavior for such regimes when Hertzian contact between mating teeth is considered. To compute the mesh stiffness function which is the main excitation source of the transmission, bending, fillet foundation and Hertzian stiffness are put in parallel. A torsional model combining the nonlinear mesh stiffness functions modeled in the different mesh zones and the non-stationary regime has been developed. The system’s equation of motion is computed by using Newmark- β algorithm combined with Newton-Raphson technique. An experimental setup was used to validate numerical results. The main obtained results are listed as follows:

In the stationary condition, the mean of the Hertzian stiffness and the gear mesh stiffness are constant and the vibration level is steady. For the run-up regime, the mean value of the Hertzian stiffness and the gear mesh stiffness is increasing and the period of fluctuation is decreasing. The increasing acceleration torque and the variable gear mesh stiffness cause higher vibration levels, especially when the rotation frequency of motor cross with one of natural frequencies. The sensitivity inter-teeth dynamic force and transmission error to the run-up transient regime were also investigated showing an increase of the amplitude.

Seen the time varying frequency components, time frequency analysis was used to highlight the variation of gear mesh frequency and its harmonics during run-up regime. Experimental results confirmed the numerical ones. Future works will be dedicated to gear defect modeling and its influence on the nonlinear dynamic response of the planetary transmission.

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