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Effect of carbon on cryogenic tensile behavior of CoCrFeMnNi-type high entropy alloys

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ABSTRACT

High entropy alloys (HEAs) with a face-centered cubic (fcc) structure are considered as promising structural materials, in particular due to their impressive ductility and toughness at cryogenic temperature; at the same time strength of these HEAs is often quite low. An addition of interstitial elements like carbon substantially increases the strength of the fcc HEAs at room temperature, however the effect of C on cryogenic properties has not been properly studied. Therefore in this work we examined cryogenic tensile behavior of the fcc high entropy alloys with different carbon content (0-2 at.%). The alloys had non-equiatomic proportions of principal elements, i.e. Co1Cr0.25Fe1Mn1Ni1. The lower Cr concentration in comparison with the equiatomic alloy led to the higher solubility of carbon confirmed by both ThermoCalc calculations and experimental results; only in the alloy with 2 at.% C a small (<1%) fraction of Crrich M₇C₃ carbides was found in the as-cast condition. The microstructure of the alloys was not significantly affected by the carbon content and generally consisted of coarse (250-300 µm) fcc phase grains with dendritic segregations. In turn, the carbon content influenced on mechanical behavior substantially: the strength of the alloys progressively increased with the carbon content along with some reduction in ductility. Solid solution strengthening by carbon at 77 K was much stronger than that at room temperature: 67 MPa/at% and 178 MPa/at%, respectively. The increase in solid solution strengthening agreed well with an anticipated increase in lattice friction at lower temperatures. Plastic deformation was associated with dislocations slip both at 293 K and 77 K; a decrease in temperature and an increase in the carbon concentration increased the inclination to planar slip. The obtained results offer new approaches to increase the cryogenic properties of fcc HEAs.

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1. Introduction

The so-called high entropy alloys (HEAs) have been attracting significant attention from materials scientists worldwide over the last two decades [1,2]. Initial works strongly emphasized maximizing configurational entropy via the usage of multiple (\geq 5) principal components taken in close to equiatomic concentrations (5–35 at.%) [1]. High configurational entropy was supposed to stabilize disordered solid solution phase in favor of ordered or intermetallic ones. However, later it was found that high configurational entropy is neither a necessary nor sufficient condition for solid solutions formation; therefore alloys composed of multiple

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principle elements can have a complex multiphase structure [2–5]. Nevertheless these alloys were revealed to have structures and properties not readily available in a conventional alloy based on single principle elements [2,6–10]. The current alloy design strategy focuses more on alloys which can fill the "gaps" in the performance of single-principle-element alloys; as such, no strict requirements on the chemical composition of these complex, concentrated alloys (CCAs) is claimed [2].

Meanwhile some of the HEAs possess a surprisingly stable solid solution single-phase structure. One of the examples is the equiatomic CoCrFeMnNi alloy, widely known as the Cantor alloy [11]. It has a face-centered cubic (fcc) single-phase structure stable at $T \le 900$ °C [12–15] and possesses quite intriguing mechanical properties, namely a combination of high strength and ductility along with exceptional fracture toughness at cryogenic temperatures [16–19]. Promising characteristics of the alloy were attributed







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to the synergy of different deformation mechanisms, in particular, dislocation slip and mechanical twinning [16,19–21]. However, quite low room temperature strength of the alloys caused a number of studies focused on an increase in the strength [15,20,22–29]. At the same time not much attention has been given to the further improvement of the cryogenic properties of the alloy [30–33].

Alloving with interstitial elements like carbon was found to be one of the most effective ways for improving the strength of CoCrFeMnNi and similar alloys at room temperature [34–48]. An increase in strength was associated with (i) solid solution strengthening [34,38,41,43] and/or (ii) precipitation strengthening due to carbide particles [39,42,45,47,49,50]. Although solid solution strengthening by carbon was found to be particularly effective, the equilibrium solubility of carbon in the fcc matrix of the equiatomic CoCrFeMnNi alloy was found to be rather low (~0.1 at.% at 1000 °C [42]) due to the strong interaction of carbon with chromium. Carbon when dissolved in the fcc matrix can also strongly affect deformation mechanisms of the alloys by varying the stacking fault energy (SFE) [36,44,51] and/or friction stress [52]. Meanwhile controversial information on the effect of carbon on active deformation mechanisms in CoCrFeMnNi-type alloys is available [36,37,43,46]. Also, to the best of the authors' knowledge, very limited information on the effect of carbon on cryogenic mechanical properties of such alloys can be found in the literature [46,53].

Therefore, in the present paper, we report the effect of carbon content (0–2 at.%) on room and cryogenic tensile properties of a CoCrFeMnNi-type HEA. To increase the solubility of carbon in solid solution, the molar fraction of Cr in the "base" alloy was reduced to 0.25 whereas the molar fractions of all other elements were ~1. Post-deformation microstructures were examined to estimate the effect of carbon on deformation mechanisms.

2. Materials and methods

Alloys used in the current study had non-equiatomic concentrations of the principal elements: Co1Cr0.25Fe1Mn1Ni1 (the subscripts indicate the molar fraction of the respective components; hereafter the CoCr_{0.25}FeNiMn designation will be used instead of Co₁Cr_{0.25}Fe₁Mn₁Ni₁ for the sake of simplicity). Different amounts of carbon (the nominal concentrations were 0, 0.5, 1.0, and 2.0 at.%) were added to the experimental alloys. The ingots were produced from mixtures of pure (\geq 99.9%) constitutive elements by vacuum induction melting. An extra amount of Mn was added to the initial mixtures to compensate its loss due to evaporation in accordance with [16]. The produced ingots of ~400 g had a rectangular shape measured $65 \times 40 \times 17 \text{ mm}^3$. Results of chemical analysis (Table 1) indicated that the measured composition was close to the nominal one. The concentrations of the principal elements, listed in Table 1, were determined by energy dispersive spectrometry (EDS) analysis; the concentration of carbon was determined by LECO thermalcombustion analysis.

Specimens for microstructure analysis and mechanical testing were cut from the as-cast ingots using an electric discharge machine. X-ray diffraction (XRD), scanning (SEM) and transmission

Table 1

Chemical composition of the program alloys depending on the nominal carbon concentration, in at.%.

Nominal carbon content, at.%	Concentration of the elements, at.%							
	Со	Cr	Fe	Mn	Ni	С		
0	23.29	6.22	23.90	23.09	23.46	0.03		
0.5	23.17	6.42	23.97	23.67	22.24	0.53		
1.0	23.14	6.31	23.13	23.14	23.34	0.95		
2.0	23.42	6.23	22.41	22.02	23.82	2.11		

(TEM) electron microscopy were used for structural analysis. The XRD analysis was performed using RIGAKU diffractometer and Cu K α radiation. The samples for SEM observations were prepared by careful mechanical polishing. SEM back-scattered electron (BSE) images were taken using an FEI Quanta 3D microscope equipped with an EDS detector. TEM investigations were performed using a JEOL JEM-2100 microscope operated at 200 kV and equipped with an EDS detector. Samples for TEM analysis were prepared by conventional twin-jet electro-polishing of mechanically pre-thinned to 100 mm foils, in a mixture of 90% CH₃COOH and 10% HClO₄ at 30 V potential at room temperature.

Tensile testing was used to evaluate the mechanical properties of the alloys. Tensile tests at room (293 K) temperature (RT), or cryogenic (77 K) temperature of dog-bone flat specimens (gage measured $6 \times 3 \times 1 \text{ mm}^3$) were conducted using an Instron 5882 universal testing machine at a constant crosshead velocity corresponded to a nominal strain rate of 10^{-3} s⁻¹. For cryogenic testing, the test specimen and both grips were immersed in an open-top vessel filled with liquid nitrogen whose level was continuously monitored and adjusted to ensure adequate temperature control. The specimen held for a couple of minutes before the test onset to equilibrate the temperature. Elongation to fracture was determined by measurements of the spacing between marks designating the gauge length before and after the test. At least 3 specimens for each composition and temperature were tested. Extra specimens were tensioned to 20% strain; these specimens were used further for TEM examination of the deformed microstructure.

3. Results and discussion

3.1. Microstructure

Fig. 1a shows XRD patterns of the CoCr_{0.25}FeNiMn alloy with the different carbon contents. Diffraction maximums from the fcc lattice dominated in the observed patterns; no other peaks were found even in the alloy with 2 at.% of carbon (some variations in the peaks intensity were most likely associated with coarse grains in the as-cast structure). However, the lattice parameter of the fcc phase gradually increased from 0.3590 nm to 0.3613 nm due to an increase in the carbon content from 0 to 2.0 at.% (Fig. 1b). Note that a linear relationship between the carbon content and the fcc lattice parameter was observed (dashed line in Fig. 1b). The slope of the line was ≈ 0.001 nm/at%.

SEM-BSE images of microstructure of the CoCr_{0.25}FeMnNi alloys with the different carbon contents are shown in Fig. 2. Structure of the alloys in all conditions consisted of coarse grains $(\sim 150-200 \,\mu m)$ with dendritic segregations; in accordance with earlier obtained results [54] the interdendritic area (marked as #1 in Fig. 2a) were found to be enriched with Mn and Ni, while dendrites (#2 in Fig. 2a) were enriched with Co, Cr, and Fe (the results of EDS chemical analysis are not shown). Some pores, visible as black dots, were also found in the alloys. An increase in the carbon content from 0 at.% (Fig. 2a) to 0.5 at.% (Fig. 2b) or 1 at.% (Fig. 2c) did not result in any noticeable differences in the microstructure. Meanwhile in the alloy with the highest concentration of carbon (2 at.% (Fig. 2d)) fine second phase(s) particles were detected at grain boundaries (high magnification insert in Fig. 2d). Unfortunately, the size and volume fraction of the particles were too small to identify them reliably using XRD/SEM techniques.

Further investigation of the CoCr_{0.25}FeNiMn-C alloys by TEM (Fig. 3) showed that both the alloy without carbon (Fig. 3a) and the alloy with the low carbon content (0.5 or 1 at.%; Fig. 3b and c, respectively) had a single fcc phase structure; no signs of a second phase(s) were detected. Low-angle boundaries and dislocation walls in which individual dislocations can be recognized were



Fig. 1. XRD analysis of the $CoCr_{0.25}$ FeMnNi high entropy alloys with different carbon content (x, at.%): (a) – XRD patterns; (b) – dependence of fcc lattice parameters on carbon content.





Fig. 2. SEM-BSE images of the microstructure of the $CoCr_{0.25}$ FeNiMn high entropy alloys with different carbon content (x): (a) - x = 0; (b) - x = 0.5; (c) - x = 1.0; d - x = 2.0.

observed in the microstructure; some of the boundaries separated the dendritic and interdendritic areas. With an increase in the carbon content to 2 at.% elongated second phase particles formed along (sub)boundaries (Fig. 3d). The length and width of the

particles were ~300 nm and ~80 nm, respectively. The particles, identified as M_7C_3 type carbides in accordance with the selected area electron diffraction patterns interpretation (insert in Fig. 3d), were composed mostly of Cr in accordance with EDS analysis (note



(a)

(b)



Fig. 3. TEM bright-field images of the CoCr_{0.25}FeMnNi high entropy alloys with different carbon percentage (x): (a) - x = 0; (b) - x = 0.5; (c) - x = 1.0; d - x = 2.0. Selected area electron diffraction pattern from the second phase particle is inserted in Fig. 3d.

that the energy resolution of the EDS system used in the current study did not allow quantitative analysis of light elements like C). The volume fraction of the carbide particles was below 1%.

3.2. Tensile properties

Fig. 4 shows tensile stress-strain curves obtained at room (Fig. 4a) or cryogenic temperatures (Fig. 4b). Resulting mechanical properties, namely yield strength (YS), ultimate tensile strength (UTS), uniform elongation (UE), and elongation to fracture (EF), are summarized in Table 2. Qualitatively, all tested alloys demonstrated rather similar mechanical behavior at both temperatures: an extensive strain hardening stage after yielding resulted in high ultimate tensile strength and considerable uniform elongation. Note that some serrations were observed at the stress-strain curves of the carbon-containing alloys at RT (insert in Fig. 4a); meanwhile no serrations were found in the same alloys at cryogenic temperature.

Quantitatively, two main trends can be drawn from the mechanical behavior analysis: (i) an increase in the carbon concentration resulted in a pronounced increase in strength of the alloys and some reduction in ductility; (ii) a decrease in the testing temperature resulted in a pronounced increase in strength and some increment in ductility (Fig. 4, Table 2). Note that ductility of the alloy with the maximum carbon percentage (2 at.%) did not increase with decreasing the testing temperature. Yield strength of the alloys was especially sensitive to both the carbon content (at room temperature YS almost doubled from 185 MPa to 320 MPa with an increase of carbon from 0 to 2 at.%) and deformation temperature (YS at 77 K was almost two times greater than that at RT - 315 MPa and 605 MPa at 2 at.% C, respectively).

The dependence of the yield strength of the carbon-doped CoCr_{0.25}FeMnNi alloys on the carbon content is plotted in Fig. 5. At both temperatures (293 and 77 K) a linear relationship between strength and the carbon concentration is observed, however their slope significantly depended on the testing temperature: 67 MPa/at % at 293 K and 147 MPa/at% at 77 K.

Fig. 6 shows strain hardening curves of the CoCr_{0.25}FeMnNi alloy with different carbon contents (0, 0.5 and 2 at.%) obtained during tensile testing at room or cryogenic temperatures. At 293 K the alloys showed qualitatively similar behavior irrespective of the carbon content (Fig. 6a). Strain hardening rate was ~1100–1500 MPa at the initial stages of deformation with a noticeable (30–40%) increase in the interval of strain $\varepsilon \approx 0.05-0.3$ following by a gradual decrease. The alloys with the low carbon content (0 or 0.5 at.%) showed a similar maximum value of $d\sigma/d\varepsilon \approx 1400$ MPa, while the alloy with 2 at.% of carbon reached $d\sigma/d\varepsilon \approx 2000$ MPa.



Fig. 4. Tensile stress-strain curves of the CoCr_{0.25}FeMnNi high entropy alloys with different carbon contents obtained at different temperatures: (a) – 293 K; (b) – 77 K.

Table 2

Mechanical properties of the CoCr_{0.25}FeMnNi high entropy alloys with different carbon contents at 293 K or 77 K: yield strength (YS), ultimate tensile strength (UTS), uniform elongation (UE), and elongation to fracture (EF).

Carbon concentration, at.%	Testing temperature, K	YS, MPa	UTS, MPa	UE, %	EF, %
0	77	315	750	59	70
	293	185	460	52	64
0.5	77	370	795	58	68
	293	215	535	47	58
1	77	445	905	52	61
	293	255	610	48	55
2	77	605	1060	47	53
	293	320	685	48	53



Fig. 5. Dependence of the $CoCr_{0.25}$ FeMnNi alloys yield strength on the carbon content at room (293 K) or cryogenic (77 K) temperatures.

At the lower temperature the values of $d\sigma/d\epsilon$ at the initial stage of deformation ($\epsilon \approx 0.05$) were found to be ~1750–1950 MPa for the alloys with 0 or 0.5 at.% and ~2300 MPa for the alloy with 2 at.% of carbon (Fig. 6b). The shapes of the curves for the alloys contained 0 or 2 at.% of carbon were similar to those at room temperature; however the maximum values of $d\sigma/d\epsilon$ (~2000 MPa in case of 0.5 at.% of carbon and ~2600 MPa for the alloys with 2 at.% of carbon) were reached at slightly lower strain $\epsilon \approx 0.16-0.25$. Then both curves gradually decreased until a rapid drop at the final stages of deformation. The alloy with 0 at.% of carbon showed a very slow decrease of $d\sigma/d\epsilon$ to ~1900 MPa during straining at cryogenic temperature.

Serration was observed at both temperatures in all the alloys;

some increase in the serration amplitude can be noticed with an increase in the carbon content at room temperature.

3.3. Deformation microstructure

Fig. 7 shows TEM bright-field images of the CoCr_{0.25}FeMnNi alloy structure with 0, 0.5 or 2.0 at.% of carbon tensioned to 20% at 293 or 77 K. Tensile deformation of the alloy at RT resulted in the formation of elongated dislocation arrays heterogeneously distributed in the microstructure (Fig. 7a). Addition of 0.5 at.% of carbon resulted in activation of intensive slip along a preferable direction and the formation of dislocation bands (Fig. 7c). However, boundaries of the bands were quite diffuse and sometimes masked with dislocation pile-ups located in between the bands most probably due to the operation of secondary slip systems (marked by arrows in Fig. 7c). Further increase in the carbon content to 2 at.% resulted in more obvious planar dislocation slip with the formation of denser dislocation bands (Fig. 7e).

Deformation of the CoCr_{0.25}FeMnNi alloy at cryogenic temperature resulted in a higher dislocation density and homogeneous dislocations distribution for all carbon concentrations (Fig. 7b, d, f), most probably due to more active operation of secondary slip systems in comparison with RT deformation. For example operating of multiple slip systems in the alloy with 0.5 at.% of carbon during deformation at cryogenic temperature led to the formation of a cell-like structure (Fig. 7d). In the alloy with 2 at.% of carbon multiple planar slip was also observed (Fig. 7f), however, slip was more localized in comparison with the alloy with 0.5 at.% of carbon. In some places individual crystallographically aligned slip bands with a high dislocation density were observed (insert in Fig. 7f).



Fig. 6. Strain hardening curves of the CoCr_{0.25}FeMnNi alloys with 0, 0.5, and 2 at.% of carbon at: (a) - 293 K; (b) - 77 K.

4. Discussion

The present study has revealed a significant effect of carbon concentration on the mechanical properties of $CoCr_{0.25}$ FeMnNi alloys; this effect became even more pronounced with a decrease in deformation temperature from 293 to 77 K. One of the important features of the program alloys was associated with a nearly single fcc phase condition, i.e. carbon was almost completely dissolved in the matrix; insignificant amount of carbides was found only in the alloy with 2 at.% of carbon (Figs. 2d and 3d). The linear dependence of the fcc lattice parameter on nominal carbon concentration (Fig. 1b) also confirmed dissolution of carbon in the alloys [42].

The single-phase structure of the program alloys is in contrast with carbon-doped equiatomic CoCrFeNiMn alloys [37,39,42] in which a lot of carbides appeared when the content of carbon was ≥ 1 at.%. This effect was obviously associated with different percentages of Cr, which is the strongest carbide-forming element among all the constitutive elements of the Cantor alloy.

The effect of Cr on the solubility of carbon in the fcc solid solution is shown qualitatively in Fig. 8. One can clearly see that a decrease in the Cr molar fraction from 1 to 0.25 greatly extended the single fcc phase field toward higher carbon concentrations. For example, the equiatomic CoCrFeMnNi alloy can dissolve only 0.36 at.% of C in the fcc solid solution at 1000 °C, whereas the corresponding amount for the $CoCr_{0.25}FeMnNi$ alloy is 1.61 at.%. Given a strong solid solution strengthening effect of carbon (see below), a reduction in the Cr content with an addition of carbon can be considered as a promising alloying strategy. It should be noted however that due to non-equilibrium processes during solidification some amount of carbides can be formed in the alloy with 2 at.% of carbon (Fig. 3d) although complete dissolution of this amount in the fcc matrix at T \geq 1040 °C was predicted (Fig. 8).

Since except for the insignificant fraction of carbides in the alloy with 2 at.% of C, the microstructures of the examined alloys were almost identical, it is reasonable to suggest that the mechanical properties were mostly affected by the carbon concentration in the fcc phase, i.e. by carbon-induced solid solution strengthening. Indeed, the linear relationship between the yield strength of the alloys and the (nominal) carbon concentration is observed (Fig. 5) with the slope of 67 MPa/at% at 293 K. This value correlates reasonably well with the carbon-induced solid solution strengthening in stainless (76.6 MPa/at% [55]) and TWIP steels (46 MPa/at% [56]) and in a carbon-doped equiatomic CoCrFeMnNi alloy (65 MPa/at% [37]). It should be noted, however, that a much higher carbon-induced solid solution strengthening effect was reported for some other fcc HEAs, for example 178 MPa/at% for a $Fe_{40.4}Ni_{11.3}Mn_{34.8}Al_{7.5}Cr_6$ alloy [34]. This discrepancy can most likely be ascribed to a significant difference in the chemical compositions of these two alloys (i.e. $Fe_{40.4}Ni_{11.3}Mn_{34.8}Al_{7.5}Cr_6$ and $CoCr_{0.25}FeMnNi$).

The solid solution strengthening effect of carbon became at cryogenic temperature ~2.2 times stronger (147 MPa/at%) in comparison with that at RT. Solid solution strengthening is generally ascribed to the lattice friction which becomes stronger with a decrease in temperature. According to Ref. [57], the temperature (*T*) dependence of the friction stress (σ_{fr}) can be expressed as:

$$\sigma_{fr} = \sigma_{fr}(0)^* \exp \frac{-2\pi\omega_0}{(3bt_m T)}$$
(1)

where $\sigma_{fr}(o)$ is the friction stress at T = 0 K, ω_0 denotes the dislocation width at T = 0 K, b is the Burgers vector, and T_m is the melting temperature. Using equation (1), we have calculated the ratio between σ_{fr} at 77 K and σ_{fr} at RT. $\omega_0 = b$ was used for the calculations in accordance with the results of [57] for a number of equiatomic fcc solid solutions of the CoCrFeMnNi system. T_m of the CoCr_{0.25}. FeMnNi alloy with 1 at.% of carbon was estimated by the Thermo-Calc software as 1493 K. The results of calculations showed that the expected $\sigma_{fr}(77 \text{ K})/\sigma_{fr}(\text{RT})$ ratio is 1.94, which agrees well with the observed difference in solid solution strengthening by carbon at RT and 77 K.

High frictional stress can be one of the reasons for intensive strengthening of the alloys during deformation (Fig. 4). In general such type of the mechanical behavior is usually ascribed to the operation of at least one of the deformation mechanisms: transformation-induced plasticity (TRIP), twinning-induced plasticity (TWIP) or microband-induced plasticity (MBIP). Each of these mechanisms gives rise to relatively high strain hardening rate which even can increase during deformation [58,59] and intensive strengthening thereby attaining high elongation to fracture of the alloy. The present study did not reveal any signs of deformation twinning (or deformation-induced phase transformation) even at cryogenic temperature in contrast to the equiatomic CoCrFeMnNi alloy where intensive twinning was observed at a cryogenic temperature [16,22,60]. For activation of both these mechanisms (TRIP or TWIP) low stacking fault energy (SFE) is a necessary condition. Meanwhile reduction in the Cr content (which is particularly effective for decreasing SFE in Co-Cr-Fe-Mn-Ni alloys [61,62]) in the investigated CoCr_{0.25}FeMnNi alloy in comparison with the equiatomic CoCrFeMnNi most probably resulted in an increase in SFE and therefore in suppression of both TRIP and TWIP mechanisms.

Strain hardening due to the formation crystallographic slip





(f)

Fig. 7. TEM bright-field images of the CoCr_{0.25}FeMnNi alloys microstructure with different carbon percentage (x) after tension to $\varepsilon = 20\%$ at (a, c, e) - 293 K and (b, d, f) - 77 K: (a, b) - x = 0; (c, d) - x = 0.5; (e, f) - x = 2.0. The applied load direction is indicated with the arrow in Fig. 7a.

bands is usually associated with low SFE (which is not low enough to induce the TWIP effect, however), short-range ordering (SRO) or high friction stress [63,64]. The possibility of SRO is quite discussible since there are no direct experimental proofs in our case. The observed serrations at the stress-strain curves of the carbon-doped alloys (Fig. 4a) can be the indication of the SRO [65,66]. Note however that for carbon-doped Fe_{40.4}Ni_{11.3}Mn_{34.8}Al_{7.5}Cr₆ HEAs (up to 1.1 at. %) atom probe tomography results showed randomly distributed atoms without formation any clusters [41]. The experience from the TWIP steels [66] suggests that there can be a few



Fig. 8. Boundaries of the single fcc phase field in the carbon-doped $Co_1Cr_xFe_1Mn_1Ni_1$ alloys (subscripts indicate the molar concentrations of the respective elements) depending on the Cr and C concentrations. The equilibrium phase diagrams were constructed using the Thermo-Calc software.

possible alternative explanations; in broad sense all suggest different kinds of interaction between the defects (dislocations) and the carbon atoms. Further studies are needed to establish the nature of the serrations in the program alloys.

Meanwhile the observed increase in lattice friction stress along with a decrease in SFE [41] both due to the addition of carbon and a decrease in temperature from 293 to 77 K can result in the intensification of planar slip [64]. That is why the most obvious planar slip was observed at 77 K in the alloy with 2 at. % of carbon (Fig. 7f) and an increase in the carbon content resulted in tending to planar slip even at room temperature (Fig. 7 a,c,e). Apparently, planar slip increased the strain hardening capacity (Fig. 6) of the alloy. Formation of dislocation pile-ups (Fig. 7c) also increased strain hardening rate at the initial stages of deformation, similarly to the carbon-doped Fe_{40.4}Ni_{11.3}Mn_{34.8}Al_{7.5}Cr₆ HEAs [35]. Yet the ductility of the alloys became lower with increasing the carbon content (Fig. 4, Table 2). Evidentially, this is the consequence of increased strength together with the lack of new effective strengthening mechanisms (e.g. TWIP/TRIP) in the alloys with higher carbon percentage.

To sum up, the present study has revealed that the strength of the single fcc solid solution phase HEAs at 77 K can be substantially increased by the addition of carbon, without noticeable loss in ductility. This finding may open new pathways for improving impressive fracture toughness, already demonstrated by such alloys [18,30]. The carbon-doped alloys with high ductility, strength, and toughness can be considered as promising materials for cryogenic structural applications, for example, for natural gas transportation [67]. However, further studies on the effect of carbon on fcc HEAs toughness and other properties at different temperatures are required to verify these suggestions.

5. Conclusion

In the present work structure, mechanical properties and deformation mechanisms of the as-cast $CoCr_{0.25}FeMnNi$ alloys doped with 0-2 at.% of carbon were examined during uniaxial tension at 293 and 77 K. Following conclusions were drawn:

 In the as-cast condition the alloys had the single fcc phase coarse-grained structure when carbon content was in the range of 0–1 at.%. An insignificant fraction of the M₇C₃ carbides was found in the alloy with 2 at.% of carbon. Higher solubility of carbon in the CoCr_{0.25}FeMnNi alloy in comparison with the equiatomic CoCrFeMnNi one was proven by ThermoCalc calculations to be the result of a reduction in the Cr concentration.

- 2) Tensile mechanical properties of the alloys showed a strong dependence on the carbon content and testing temperature. Strength of the alloys increased with an increase in the carbon concentration and with a decrease in the testing temperature. For example, yield strength at room temperature increased from 185 MPa to 320 MPa with an increase in the carbon concentration from 0 to 2 at.%, while corresponding values at 77 K were 315 MPa and 605 MPa, respectively. The increase in strength was attributed to the carbon-induced solid solution strengthening effect, which was estimated as of 67 MPa/at% at 293 K and 178 MPa/at% at 77 K. Increased solid solution strengthening at the lower temperature agreed with an anticipated increase in lattice friction. Ductility of the alloys slightly decreased with an increase in the carbon percentage and generally was higher at 77 K in comparison to ductility at 293 K.
- 3) Plastic deformation in the program alloys was associated with dislocations slip both at 293 K and 77 K. A decrease in temperature and an increase in the carbon concentration increased the inclination to planar slip during deformation.

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