Taxonomy of fibre mat misalignments in pultruded GFRP bridge decks

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| 8 | Abstract: This paper presents a taxonomy of the fibre mat misalignments found in a multi- | | | | |
| 9 | celled pultruded GFRP deck, based on high-resolution images of the polished cross sections. A dual | | | | |
| 10 | approach to misalignment taxonomy is presented, one based on misalignment morphology, the other | | | | |
| 11 | grounded in the manufacturing provenance of the misalignments. Each misalignment is characterised | | | | |
| 12 | using, alternately, a Gaussian function, the angle to its steepest tangent, its length and its height. Some | | | | |
| 13 | scatter in the type and severity of the misalignments was found within the web-flange junctions. The | | | | |
| 14 | deck's top flange was found to be thicker in regions containing a double-backed or flipped mat layer, | | | | |
| 15 | which was probably due to displacement of the internal mandrels during pultrusion. A previously verified | | | | |
| 16 | analytical approach based on classical laminate theory predicts that, owing to stress concentrations, the | | | | |
| 17 | present Gaussian misalignments lead to a 57% knockdown in tensile load capacity. | | | | |
| 18 | Keywords: Glass fibres, fabrics/textiles, defects, pultrusion | | | | |
| 10 | 1 Introduction | | | | |
| 19 | | | | | |
| 20 | Glass fibre reinforced polymer (GFRP) cellular bridge decks offer numerous advantages | | | | |
| 21 | compared to conventional steel or reinforced concrete alternatives. The cellular design and high stiffness- | | | | |
| 22 | and strength-to-weight ratio of GFRP enables prefabricated decks to be quickly lifted onto the abutments | | | | |
| 23 | using low capacity cranes, which minimises traffic disruption [1]. The modular design is eminently suited | | | | |
| 24 | to increasing the load capacity of existing bridges through dead weight reduction. Also, the superior | | | | |
| 25 | corrosion resistance (to de-icing salts in particular) can reduce the maintenance and inspection costs over | | | | |
| 26 | a prolonged design life. | | | | |
| 27 | Of the range of manufacturing techniques available, pultrusion has proved popular for GFRP | | | | |
| 28 | decks as it facilitates the continuous production of standardized, thin walled sections of complex | | | | |

29 geometries and fibre architectures. The standardization of the geometry and material properties of

30 pultruded GFRP sections has been crucial to their adoption in bridge decks and in civil engineering

31 structures more widely. Thus, any discrepancies that exist between the assumed material properties and

To that end, this paper investigates a key source of material variability in pultruded decks;

32 those of the as-built deck must be properly understood.

33

- 34 namely, misalignment of the fibre mats. As shown in Figure 1(a), the fibre mats in a GFRP deck are often 35 randomly misaligned in the transverse (normal-to pultrusion) direction. The term "fibre waviness" is commonly used to describe this type of defect [2]. However, for pultruded decks the trajectory of the mats 36 37 can take various forms. Hence, throughout this paper the term *misalignment* is used to refer to any 38 deviation of the mats from their optimal location; that is, the location that maximises the mechanical 39 properties of the deck when subject to realistic in-service loads. 40 Fibre mat misalignments are only problematic if their adverse effects are not accounted for at the 41 design stage. There are three reasons why this is often the case for GFRP decks: 42 1. The mat misalignments arise due to variabilities in the pultrusion process, and so their location 43 and severity are highly random. 44 2. The mechanical properties of the deck provided by the manufacturer are established from 45 standard tests on flat coupons taken from the webs and flanges. However, the properties of the
- flats are not representative of the web-flange junctions (WFJs) due in part to the presence of mat
 misalignments and other defects (including voids and resin-rich zones) that are *unique* to the
 junctions.

49 It should be noted that, given the definition of misalignment provided above, even in the absence of

50 the *random* misalignments that arise during the pultrusion process, the fibre mats may still be considered

51 misaligned if they do not produce the *optimal* deck performance. This becomes an important

52 consideration at the junctions, which are subject to complex, multi-directional stress states due to the local

53 contact pressure distribution exerted by heavy lorry tyres [3]. The positioning of fibre mats that produces

- 54 the optimal resistance to these loads is difficult to determine and may not be consistent with the
- 55 positioning specified by the manufacturer. Furthermore, manufacturers of GFRP decks do not typically

56 disclose detailed information regarding the 'intended' fibre architecture and mat positioning.

57 The above concerns have precluded quality standards from adequately addressing the issue of mat misalignments in pultruded decks. For example, the US standard ASTM D4385-19 [4] states that 58 59 misalignments are permissible if the section achieves the material properties stated by the manufacturer, 60 which is inadequate for the reasons outlined in point 2 above. The European standard EN 13706-2 [5] 61 outlines a defect state in pultruded sections as "An unintentional or unspecified misalignment of mat or 62 fabric reinforcing material in relation to the contour of a pultruded section", which is limited to an out-63 of-plane deviation of 1.5 mm (or 20% of the part thickness). However, as outlined in the paragraph above, 64 in areas of complex geometry the 'contours' of the GFRP deck are ill-defined and may not describe the 65 optimal trajectory for the fibre mats. Thus, the specified limits on mat misalignment should be relative to 66 the performance of the deck.

67 A lack of clear guidance regarding the acceptable positioning of the fibre mats has resulted in 68 many previous studies reporting on misaligned fibre mats in pultruded GFRP decks. For example, 69 Coogler et al. [6] described "misoriented" and "wavy" fibre mats in the WFJs of a section of Duraspan 70 deck. For the same decking system, Yanes-Armas et al. [7] described "wrinkling" of the inner mats, 71 "variable prolongation" of the mats from the WFJ into the adjoining flange and "asymmetric distribution" 72 of the mats through-the-thickness of the web. These misalignments were inconsistent between two 73 nominally identical WFJs. Xin et al. [8] commented on mat "folding" in the WFJs of a cellular GFRP 74 deck, which was described as more pronounced for the WFJs with thicker webs, flanges and fillet radii. In 75 a recent study by Sa et al. [9], Figure 3 of their paper clearly shows misalignment of the mats passing through the WFJ for a 'snap-fit' pultruded GFRP deck manufactured in Korea. Sebastian et al. [2] 76 77 presented a first attempt at measuring the fibre mat "waviness" in a unit of Fiberline ASSET deck, by 78 manually plotting points along each fibre mat layer on a transparent plastic sheet using a sharp-pointed 79 marker.

Aside from bridge decks, several other studies have demonstrated similar mat misalignments in standard pultruded sections. For example, misalignments in both the flats [10], free ends [11] and WFJs [12] [13] of I-sections are reported to produce significant variability in the local buckling loads, likewise for box beams [14]. This demonstrates the prominence of mat misalignments as a generic issue in pultruded sections. However, beyond these initial observations, no data that relate the severity of mat misalignment to the properties of the pultruded section have been produced. Indeed, a search through

Mottram's constantly renewed database of over 2700 publications on research and development with pultruded profiles [15] shows that the present paper is the first to address this important issue.

88 Despite this paucity of data, the influence of variable mat misalignments on the failure of 89 pultruded sections is evidenced by the fracture patterns that develop during failure. The effect is most 90 pronounced at key junctions within the deck that exhibit significant misalignments. As shown in Figure 91 1(b), failure of the WFJs manifests as delaminations along the misaligned mat layers and kinking at the 92 steepest tangent angle, which is accompanied by transverse cracks that occasionally propagate to the 93 surface. These surface cracks create paths for water ingress, which can further deteriorate the fibre-to-94 resin bond strengths. A comprehensive summary of the figures in previous studies that show similar 95 fractures within the WFJs of both pultruded GFRP decks and I-sections is provided in Table 1 of 96 Sebastian et al. [2].

A complete understanding of fibre mat misalignments starts with the pultrusion process. To that end, Figure 2 shows an illustration of the pultrusion process, where the key stages that influence the positioning of the mats are emphasized (in bold). The specific process details for a given pultruded section vary and are typically undisclosed by the manufacturer, so the illustration is based on generic information provided in Shaw-Stewart and Sumerack [16] and Meyer [17]. A description of the key stages is given below.

103 During pultrusion, layers of fibre mats are pulled and progressively formed from their initial 104 planar shape into that of the final section using a series of metal or high-density polyethylene forming 105 guides. Rovings are pulled in a similar manner through a series of guiding holes and inserted as discrete 106 layers between the mats. This process (referred to as reinforcement *collimation*) is more complex for 107 cellular sections as the fibre mats must be completely wrapped around an internal mandrel in order to 108 form the hollow shape. This wrapping method is preferable to inserting separate flat mats into each of the 109 webs and flanges, which does not achieve continuity of the mats through the web-flange junctions 110 (WFJs).

Thick, multi-celled pultruded sections with multiple layers of fibre mats and rovings – as used in GFRP bridge decks – require many forming guides and mandrels that afford only approximate control over the positioning of the fibres. Lateral shifting and crowding of the rovings prior to die entry can misalign the surrounding mats, and wrinkles in the mats can also arise during the wrapping process.

Within the die, compression of the fibre rovings and impregnation of the resin using die-injection can result in further movement of the fibres, producing wrinkles and resin-rich zones near the surface of the section. For cellular sections, transverse gaps or overlaps in the mats are inevitable due to the wrapping process, but their location and extent are subject to variability during manufacture.

As a result, the cross section of the cured profile can exhibit multiple *types* of misalignment that occur randomly and simultaneously throughout the section, as illustrated in the insert to the bottom right of Figure 2. It is important to distinguish between these different types, as they may have a differing impact on the structural performance of the GFRP deck. Unlike the previous studies listed above that use terms such as "misoriented", "wavy", "wrinkled" and "folded" interchangeably, the study reported on in the rest of this paper defines multiple categories of mat misalignment based on their morphology and origin in manufacture.

126 The issue of fibre misalignments has been studied extensively in the context of aerospace- and 127 automotive-grade FRPs [18]. In this case, experimental investigations typically involve inducing a 128 localised misalignment in a small flat coupon under lab conditions, enabling the severity to be precisely 129 controlled and quantified. For example, the effect of wrinkles in CFRP prepregs has been investigated by 130 inserting either steel rods [19], aluminium wire [20], or prepreg strips [21] transversely across the 131 laminate. Potter et al. [22] built on these studies by developing a taxonomy of defect states in prepreg 132 composite components, highlighting that a significant proportion of misalignments are influenced 133 predominantly by design decisions, including the choice of corner radii and part curvature, rather than by 134 errors in the manufacturing process itself.

135 The studies mentioned in the preceding paragraph produce misalignments in flat FRP coupons 136 that are easily characterised using simple sinusoidal functions, which enables theoretical prediction of the 137 laminate stiffness and strength reduction as a function of the wave height and amplitude. Bogetti et al. 138 [23] applied 2D piecewise classical laminate theory to a cross-ply laminate with a half sine-wave middle 139 layer. This method was later shown by Hsiao and Daniel to correlate well with experimentally obtained 140 stiffness [24] and strength [25] values for carbon- and glass-FRP coupons under uniaxial compression. El-141 Hajjar and Peterson [19] used the method to accurately predict the stiffness of unidirectional carbon-FRP 142 laminates with wrinkles described using a graded Gaussian function. It is now prudent to transfer the 143 above concepts to the mat misalignments found in as-manufactured pultruded GFRP bridge decks.

2. Objectives of the present study

| 145 | From the issues outlined above, the objectives of this paper are as follows, namely to: | | | |
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| 146 | • Develop a suitable technique for imaging and measuring the transverse fibre mat misalignments | | | |
| 147 | within a pultruded GFRP bridge deck. | | | |
| 148 | • Present a taxonomy of fibre mat misalignment based on morphology and origin in manufacture. | | | |
| 149 | • Quantify the inconsistency of misalignments observed in nominally identical WFJs of the deck. | | | |
| 150 | • Assess whether the fibre mat misalignments can be characterised using simple mathematical | | | |
| 151 | functions that can be implemented into an analytical model to predict the reduction in | | | |
| 152 | mechanical properties of the deck. | | | |
| | | | | |
| 153 | A specific pultruded bridge deck has been used throughout this paper to illustrate the above | | | |
| 154 | objectives. The mat misalignments observed in the bridge deck used in this study are broadly | | | |
| 155 | representative of those found in other commercial pultruded, cellular bridge decks to date. Similar | | | |
| 156 | misalignment morphologies to those presented in this paper can be seen in previously published cross- | | | |
| 157 | sectional images of a variety of bridge decks from different manufacturers (for example [7], [8] and [9]) | | | |
| 158 | and from a larger decking system from the same manufacturer [2]. Thus, the imaging and characterisation | | | |
| 159 | technique proposed in this paper can be applied to a wide range of pultruded, cellular decks. | | | |
| 160 | The scope of this paper is limited to mat misalignments in the transverse-to-pultrusion direction, | | | |
| 161 | as can be seen in the cut cross-sections of the deck. It is noted that the tensile force applied to the fibre | | | |
| 162 | mats during pultrusion should minimise misalignments in the longitudinal direction. | | | |
| 163 | The following section outlines the imaging technique used and gives details of the different fibre | | | |
| 164 | mats observed in the chosen deck specimens. The taxonomy of fibre mat misalignments is presented in | | | |
| 165 | Section 4, which is then used in Section 5 to compare the WFJs from each of the five specimens. Section | | | |
| 166 | 6 presents the analytical analysis, which is followed by conclusions and recommendations for future | | | |
| 167 | work. | | | |

3. Methodology

169 **3.1 GFRP deck preparation and imaging**

170 Figure 3(a) shows the geometry of the GFRP decking system used in this study, which is a 171 rectangular unit that encloses two diagonal stiffeners to form three internal cells. The short sides of the 172 deck have complementary tongue-and-groove connectors to facilitate adhesive bonding. The GFRP 173 material consists of E-glass fibres embedded in an isophthalic polyester resin. Specific details of the fibre 174 mats and rovings used in the deck are provided in Section 3.2. The deck geometry was optimised to 175 satisfy the traffic loading in the Dutch standard NEN 6788: 1995 [26] and was manufactured by Fiberline 176 using resin injection within the die. The decking system has been used in several bridges throughout 177 Europe, including a 99 m cyclist and pedestrian bridge in Reinbek, Germany in 2009. In this and other 178 bridges, the deck has spanned transversely between longitudinal girders with the thicker flange on top to 179 provide adequate resistance to local tyre loads. 180 Five panels of bridge deck were selected for analysis based on differences in the type of mat 181 misalignments observed within the cross section of the deck, indicating that the panels were produced in 182 different batches. These were labelled specimens A to E, where specimen A exhibited the fewest 183 misalignments and E the most. A comparison of the misalignments within the WFJs of each specimen is 184 provided in Section 5 of this paper. 185 A single 80 mm wide section of deck was cut from each panel, and the cut surface was then 186 deburred and polished using progressively finer sandpaper up to 600-grit. The application of an oil-based 187 substance to the cut surface of the deck increased the reflectivity of the 0° fibres, thus enhancing the 188 contrast between the rovings and mats. Petroleum jelly was used for this purpose as it did not fade 189 significantly with time. 190 The cross sections from each specimen were imaged using a Canon 7D DSLR camera with a 191 telephoto zoom lens that had a minimum focal length of 200 mm at maximum magnification, as shown in 192 Figure 3(b). At this distance, the image window was approximately 80 mm by 53 mm (2/3 aspect ratio) 193 with a resolution of 3456 x 5184 pixels. Through trial and error, it was found that placing a light source at 194 a 45° angle above the cut surface produced maximum contrast between the roving and mat layers. 195 For each specimen, images were taken of the four WFJs at the ends of the diagonal stiffeners, 196 named top left (TL), top right (TR), bottom left (BR) and bottom right (BR) as can be seen in Figure 3(a).

197 Centring the image at each of these junctions provided sufficient coverage of the adjoining 'flats' of the198 deck and any additional misalignments that occurred in the webs or flanges.

The images were post-processed in ImageJ, a freely available image editing software that enables scaling, transformation and measurements of images. First, each image was scaled using the estimated pixels/mm obtained from scaling the horizontal ruler aligned with the top surface of the deck. Then, the images were rotated and translated to ensure the top deck surface was perfectly horizontal and at a fixed distance from the edge of the image.

The distortion effects from the lens were checked by taking an image of a 4 mm square grid and measuring the displacement of the vertices relative to a uniform grid overlaid onto the image in postprocessing. The image showed a minimal degree of pincushion distortion, with a maximum corner displacement of 0.2 mm. In the centre of the image (where the web-flange junction was located) this distortion was negligible.

Once scaled, the trajectory of each mat could be traced by placing markers every 0.25 mm along the top and bottom edges of the fibre mat layers. The misalignment of the layer was evaluated from the midline of the top and bottom markers. The uncertainty with regards to the placement of the markers was influenced by the geometry of the fibre mats, as outlined in the following section.

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3.2 Geometry of fibre mats

214 The cross-sectional images of the WFJs revealed the geometry and layup of the fibre mats and 215 rovings used within the bridge deck panels. As shown in Figure 4(a), the GFRP deck comprised a 216 combination of biaxial, triaxial and complex stitched mats either side of a large core of textured rovings. 217 The biaxial mats appear as a thick white layer (90° bundles) with a row of blue ellipses (0° bundles) that 218 have areas of pure white resin in the gaps between them. The triaxial mats have a similar cross-sectional 219 appearance to the biaxial, as the $+45^{\circ}$ and -45° bundles appear as a single white layer, with an adjacent row of blue ellipses. Note that these mats were inserted either with the 0° bundles above or below the off-220 221 axis bundles. The third type of fibre mat present in the deck was a thin complex mat, shown to the right of 222 Figure 4(a), which had a combination of small $0^{\circ}/90^{\circ}$ woven bundles and randomly orientated fibres. A 223 surface veil also lined the outer perimeter of the deck.

The fibre rovings within the core of the deck exhibit a marble-like texture due to patches of nonstraight and non-uniformly distributed fibres. Detailed micro-CT images of the fibres within a pultruded roving core showing a similar macro-structure can be found in a recent study by Baran et al. [27].

Given this fibre architecture, the trajectory of the mats was measured by manually placing markers along the top and bottom interfaces of the off-axis bundles (white bands). The fibre mat-toroving interface was very clear, and the markers could be placed with an error of less than 0.08 mm (the height of 5 pixels). However, marking the fibre mat-to-fibre mat interface required occasional interpolation through the small areas of pure resin that were present due to the gaps between the 0° bundles.

233 At this point it is useful to define a bridge deck section that has the 'idealised' mat positioning 234 that can be used as a reference for the fibre mat misalignments reported on in the rest of this paper. This 235 idealisation is shown in Figure 4(b), which also gives a magnified view of the WFJ in the insert at the top 236 of the figure. As the manufacturer's intended fibre architecture is not disclosed, the idealised section 237 assumes the mats are parallel with the contours of the profile, as is consistent with EN 13706-2 [5]. 238 From Figure 4(b), it is seen that just below the external surfaces of the deck is a surface veil 239 (black) and a biaxial mat (red). The internal cells are lined with a veil, biaxial and triaxial mat (blue), 240 which wrap completely around the cells. In the top flange, an additional horizontal biaxial mat runs 241 continuously from the left to the right side, passing through the WFJs. The underlying complex mat layer 242 (green) is terminated immediately before the junction. The rovings (grey) form a thick layer within the core of the webs and flanges between the mat layers. The schematic diagrams, such as the one shown in 243 244 Figure 4(b), are used throughout the following section in conjunction with the images of the deck cross-245 sections to clearly illustrate the trajectory of the fibre mat layers for each misalignment category.

246 **4. Taxonomy**

Figure 5 presents a summary of the different categories of mat misalignments observed within the GFRP bridge deck cross-sections. As shown, six categories are defined based on the morphology of the misalignment, where the name given to each category refers to a distinct shape as follows:

250 1. Double-Back (DB): a mat layer returns upon itself.

251 2. Flip (F): at a web-flange junction, the end of a mat layer is flipped to the other side of the
252 junction.

253 3. Corner Wrinkle (CW): a localised loop in the corner of a mat layer.

- 4. Near-Surface Wrinkle (NSW): a localised dip in the mat layers near the surface of the
 specimen.
- Waviness (W): a pronounced undulation of a mat layer, generally as it passes between layers of
 roving.

258 6. **Drop (D)**: a mat layer that drops down due to the termination of the layer below.

A detailed description of each category is provided in the following sub-sections. An example image is provided for each category. The top of each figure containing the image indicates where on the deck cross section and from which specimen (A to E) the image was taken. Within each figure a schematic of the fibre mat trajectories is provided for clarity. Note, that when discussing a single WFJ, the terms 'closed side' and 'open side' refer to the adjoining flanges that form acute and obtuse angles with the web, respectively.

265 4.1 Double-back

266 An extreme example of mat misalignment is a double-back, which is caused by errors during the mat wrapping process of cellular profiles. Figure 6 shows an example of this feature; note that the 267 268 schematic diagram shows a larger portion of the top flange than the image below. It is seen that the two 269 internal fibre mats did not follow their intended path along the top flange of the deck. Rather, near the 270 local midpoint of the flange the two layers simultaneously double-backed, and again for a second time at 271 the centre of the WFJ before terminating on the closed side of the junction. It is only this second double back that is seen in the example image given in the figure. The biaxial mat layer that ran continuously 272 273 along the top flange was forced up and over these double-backed layers. Consequently, the WFJ was 274 highly asymmetric; the closed side of the junction had more than double the intended number of fibre mat 275 layers, which then also significantly increased the thickness of that side of the junction from 9.0 mm to 276 10.5 mm (due to movement of the mandrel, as explained in Section 5.3).

277 **4.2** Flip

278 A mat flip is a class of misalignment that occurred only within the WFJs where the end of a mat 279 layer was flipped over from one side of the junction to another, as shown in Figure 7. In this example, the 280 end of the triaxial mat layer on the closed side of the junction was flipped over and terminated on the open side, rather than traverse around the fillet radius as intended. The flipped layer pushed up the biaxial 281 282 mat layer on the open side of the junction, which then dropped down a steep gradient within the junction 283 before rising again as it entered the closed side. Another side-effect was a triangular-shaped area of pure 284 resin that formed in the gap between the flipped triaxial mat and the biaxial mat above. Shrinkage of the 285 pure-resin area during curing created a void at the centre of the WFJ, as seen in the example image. The 286 flipping of the mat is also a consequence of errors in the collimation process.

287 4.3 Corner wrinkle

A corner wrinkle is a localised loop in a mat layer, typically manifesting at the corner of one of the internal mats that was wrapped around the mandrel during pultrusion. The example shown in Figure 8 occurred in the triaxial mat layer at the bottom right WFJ of specimen B. Corner wrinkles form during collimation of the mat, as incorrect wrapping of one layer can result in excess width of mat being taken up in the form of a wrinkle, which then gets pushed to the corners of the mat when the fibres are compressed into the die. A pure resin zone also formed in the gap behind the wrinkle.

294 4.4 Near-surface wrinkle

295 A second type of wrinkling occurred in the mat layers near the outer surfaces of the specimen 296 due to a different stage of the pultrusion process. During resin injection within the die, the flow of the 297 resin through the fibre mats can cause depressions in the outer layers, typically adjacent to the injection 298 ports within the die. The example in Figure 9 shows a surface wrinkle in the top biaxial mat and surface 299 veil, which occurred at the centreline of the WFJ. Note how the height of the wrinkle is greater in the 300 biaxial mat than in the surface veil. This is likely due to the lower porosity of the biaxial mat leading to 301 greater drag forces as the resin passed through the fibres. A pure-resin area filled the 'well' formed by the 302 wrinkle, which could be seen from the outer surface of the specimen.

The morphology of surface wrinkles roughly approximates a bell-curve, so the severity of misalignment is characterised by fitting a Gaussian function to the coordinates of the layer mid-thickness. The resulting curve-fits are used to compare the WFJs in Section 5, and as the basis for the analytical analysis in Section 6. The Gaussian function is given by the equation:

$$f(x) = A_{\rm w} e^{-\frac{x^2}{2c^2}}$$

(1)

307 Where A_w is the amplitude (height) of the wrinkle and *c* is the horizontal distance between the 308 centre of the curve and the inflection point (point of maximum misalignment angle).

309 4.5 Waviness

Waviness is defined as the gradual undulation of one or more mat layers within the section, in contrast to wrinkles that occur over a finite length of mat. Waviness is a consequence of the non-uniform distribution of rovings either side of the mat, which arises due to lateral shifting or crowding of the rovings during reinforcement collimation.

314 For the GFRP deck used in this study, there is only one layer of rovings within the core of the 315 section. Hence, the only location where a fibre mat passes through the middle of the roving core is at the 316 WFJs. As seen in the example in Figure 10, the rovings within the web were shifted up and to the closed 317 side of the junction, whilst in the flange they were shifted down, creating a misalignment in the biaxial mat layer that was 'intended' to pass horizontally through the junction. As shown later in Section 5, the 318 319 randomness of roving shifting led to a range of wavy shapes that were inconsistent between the different 320 WFJs. Unlike wrinkles, the morphology of the wavy mat layer within the junction did not lend itself to 321 characterisation using a mathematical function. Instead the severity of waviness was evaluated using the 322 length (L_w), height (H_w) and maximum misalignment angle ($\theta_{max,w}$), as defined in Figure 10.

323 **4.6 Drops**

A drop refers to a mat layer that is terminated within the section. For cellular sections, the wrapping of the internal mats inherently creates a discontinuity at the point where the ends meet. The guidance in EN 13706-2 [5] specifies a minimum of 5 mm overlap of the mats for closed sections. However, in practice the size and location of the overlap are highly random, due in part to the wrinkling

328 and waviness of the mat layer which shortens the path length. For the bridge deck used in this study, large 329 gaps between the ends of the two internal mats were observed along the top flange of the deck. These gaps caused pronounced drops in the mat layers above, as shown in Figure 11. The thin complex mat 330 331 above these layers also terminated immediately before the WFJ. The distance from the junction at which 332 the drops occurred were inconsistent between different specimens. This inconsistency was evaluated by 333 calculating the distance from the WFJ centreline of termination of the biaxial (D_b) , triaxial (D_t) and 334 complex (D_c) mat layers, as shown in Figure 11. A comparison of these values for each WFJ is presented 335 in Section 5.

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5. Inconsistency of misalignments

337 5.1 WFJ comparison

338 Using the taxonomy of misalignments defined in the previous section, it is now possible to 339 compare the misalignments observed in the five different bridge deck specimens used for this study. This 340 comparison focuses on the area surrounding the internal WFJs on the top side of the deck, as these 341 junctions are directly beneath the applied tyre load. Figure 12(a) to (e) show the top-left and top-right 342 WFJs for each of the five specimens. The junctions are labelled with two letters, the first denoting the 343 specimen designation (A to E) and the second whether the junction was on the left or the right side, for 344 example 'CL' for specimen C, left side. On each image, the misaligned mat layers have been identified 345 and the class of misalignment labelled using the coding system defined at the bottom of the figure. Note 346 that in Figure 12(e) the image of junction ER has a discontinuity on the open side of the junction due to a 347 cut made in the specimen prior to imaging for other purposes.

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• **Double back (DB):** the double-back occurred only in EL.

Flip (F): the flip occurred only in DL, DR and CR. However, the flips in DL and DR were
 inconsistent. In DL, the flipped mat layer kinks towards to closed side of the junction before
 flipping over to the open side. Conversely, the flipped mat in DR had almost no kink and flipped
 immediately to the open side. The kink in DL decreased the area of the triangular zone of pure
 resin to approximately 4.1 mm², compared to 10.3 mm² in DR and 6.7 mm² in CR. The smaller

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From this figure, the inconsistencies pertaining to each class of misalignment are as follows:

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area of pure resin in DL precluded the development of a shrinkage hole that can be seen clearly in the other two flipped junctions.

- **Corner wrinkle (CW)**: as noted in Section 4.3, the corner wrinkle appeared only in the *bottom* right WFJ of specimen B, and thus is not included in Figure 12.
- 359 Near-surface wrinkles (NSW): specimens A and B had negligible wrinkling of the top mat 360 layers, whereas specimens C and D had pronounced wrinkles at the centre of the junction. For 361 comparison, Figure 13 shows the Gaussian function fits of the surface wrinkles at the centre of 362 junctions CL, CR, DL, DR. The Gaussian constants A_w and c, along with the R^2 values for each 363 curve are given in the top right of the figure. It is seen that this function provides a good fit to the 364 wrinkle geometry, with R^2 values of greater than 0.97 except for CL, which had the smallest 365 wrinkle height and was asymmetric about its peak. The c values (horizontal distance to inflection 366 point of curve) ranged between 4.7 mm and 5.9 mm and were largely independent of the wrinkle 367 height. The right WFJs of specimens C and D showed a 135% and 34% increase in wrinkle 368 height compared to the left side, respectively.
- 369 Waviness (W): the waviness of the horizontal biaxial mat was highly variable both *between* the 370 different specimens and within the same specimen in the left and right junctions. The measured 371 length (L_w), height (H_w) and steepest tangent angle ($\theta_{max,w}$) of the wavy profiles are given in 372 Figure 14(a), (b) and (c) respectively, as defined in Section 4.5. The maximum waviness length was 30.52 mm (for BL) due to the severe shifting of the rovings, which were pushed up and into 373 374 the closed side of the junction. Compared to BR, this shifting produced a 66%, 30% and 16% 375 increase in the waviness length, height and angle respectively. The maximum height and tangent 376 angle were 2.9 mm and 26°, respectively, both of which occurred in CR. However, in this case the increased waviness is primarily a consequence of the flipped layer itself rather than shifting 377 378 of the rovings.
- Drops (D): the dotted lines labelled 'D' in Figure 12 show the drops in the biaxial mat layer that
 runs across the full width of the top flange due to termination of the three underlying layers
 (refer back to Figure 11 for a clear illustration of these drops). The distance from the WFJ
 centreline at which the complex, biaxial and triaxial mats terminated are shown in Figure 15(a),
 (b) and (c) respectively, but only for the junctions that did *not* contain a flip or double back.

384 Negative distances are on the closed side of the junction and positive on the open side. This
385 figure demonstrates yet again the asymmetry of the junctions, this time both between the left and
386 right WFJs (AL vs AR for example) and on either side of the same junction.

For context, the values of misalignment severity reported above can be compared to analogous

misalignment morphologies observed in industrial FRP components manufactured using techniques other 388 389 than pultrusion. For example, for a CFRP prepreg component, Wang et al. [21] reported a maximum 390 waviness length, height and tangent angle of 28.17 mm, 0.50 mm and 5.8°, respectively. While the 391 waviness length of the CFRP prepreg is comparable to that of the GFRP deck, the maximum height and 392 tangent angle are 480% and 350% greater in the GFRP deck, respectively. 393 In conclusion, the six classes of mat misalignment defined in Section 4 were highly inconsistent 394 between the top WFJs of the deck. This inconsistency was often greater between WFJs from the same 395 deck specimens than between the WFJs from *different* specimens. As outlined earlier, this is likely due to

396 lack of control over the positioning of the fibre mats during pultrusion, which introduced significant

397 randomness into the observed misalignments.

398 5.2 Flange wrinkles

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399 In addition to the near-surface wrinkles at the centre of the WFJs, significant wrinkling was 400 observed in the top flange of specimen D (labelled DT) and the bottom flange of specimen B (labelled 401 BB), shown in Figure 16(a) and (b) respectively. From the Gaussian curve fits in Figure 16(c), it is seen 402 that these winkles exhibited the maximum and minimum values of amplitude A_w (2.0 mm) and width c 403 (2.8 mm), respectively, when compared to the WFJ wrinkles. That these wrinkles occurred near to the 404 local flange midspan is problematic, as the top mat layer is subject to high stresses that alternate between 405 tensile and compressive as a tyre rolls over the surface of the deck. 406 The influence of these near-surface wrinkles on the mechanical properties of the top flange are

investigated in Section 6 using an analytical model based on classical lamination theory.

408 5.3 Mandrel displacement

409 The WFJs also varied with respect to the thicknesses of their adjoining flanges. Figure 17 shows 410 the measured flange thicknesses either side of each WFJ, along with a horizontal line indicating the 411 manufacturers stated flange thickness of 9.0 mm. It is seen that the measured values differ significantly from the intended thickness, are inconsistent between the different WFJs and show a high degree of 412 413 asymmetry between the open and closed side of the same WFJ. The largest asymmetry occurred in EL, 414 where the closed-side thickness was 16% larger than that on the open side. That the increased thickness 415 occurred on the side containing the double-backed mat layers suggests that the variable flange thicknesses 416 were a result of displacement of the mandrel during pultrusion.

417 As illustrated in Figure 18(a) during pultrusion of the decking system used in this study, there 418 are three mandrels that are cantilevered into the die. Within the die, the free ends of the mandrels are 419 subject to compressive forces from the fibre mats and rovings, and from the resin that is injected at high 420 pressure. If these forces are unbalanced around the perimeter of the mandrels, for example due to a 421 double-back or flipped mat layer, then the mandrels may move out of concentric alignment with the die. 422 The severity of mandrel misalignment can be estimated by measuring the wall thicknesses, which was 423 carried out for the left-hand cell of specimen E. The results are shown in Figure 18(b), where the centroid 424 of the intended mandrel position (in black) is compared to that of the estimated mandrel position (in red) 425 based on the wall thicknesses. It is seen that the double backed layers appear to have rotated the mandrel 426 approximately 1° towards and shifted 1.02 mm away from the junction, causing a step-change in 427 thickness in both the top and bottom junctions. This effect likely explains the variable flange thicknesses 428 in all specimens. Indeed, the WFJs containing the flipped mat layer (CR, DL and DR), which transferred 429 a layer from the closed to the open side, also had a pronounced step change in thickness. Conversely, the 430 flange thickness of specimen A was relatively constant, owing to minimal misalignment of the mats.

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6. Influence of near-surface wrinkles on mechanical properties

432 6.1 Methodology

433 An analytical approach based on the work of Bogetti et al. [23] is used in this section to evaluate 434 the influence of near-surface wrinkles on the local stiffness and failure of the top flange of the GFRP

435 deck. An idealised model of top the flange with a near-surface wrinkle is shown in Figure 19(a). The

436 model comprises a core of rovings sandwiched between two mat layers with 90° fibres. The top and

437 bottom mat layers are of equal thickness $t_{\rm m}$. The top mat undergoes a single out-of-plane undulation as

438 described by the Gaussian parameters A_w and c using Eq.(1), with the centroid located at x = 0.

439 As shown in Figure 19(b), the local orientation of the wrinkled mat layer (in 1-3 coordinates) 440 rotates an angle θ with respect to the global (*x-z*) coordinates of the laminate. The variation of θ as a 441 function of *x* is given by the following relation:

$$\theta(x) = \tan^{-1}\left(-\frac{A_{w}x}{c^{2}}e^{\frac{-x^{2}}{2c^{2}}}\right)$$
(2)

442 where A_w is the wrinkle amplitude and c the width. Table 1 gives the geometric values used in 443 the analysis, as well as the maximum and minimum values of the Gaussian parameters A_w and c based on 444 the curve fits in Figure 16(c). The length L of the laminate was 30 mm, calculated from the height of the 445 wrinkle at $\pm L/2$ being less than 1% of A_w for the maximum wrinkle width c_{max} . This ensured that the 446 stiffnesses were calculated only for the portion of the flange containing the wrinkle. The mat thickness $t_{\rm m}$ 447 was measured from the cross-sectional images and the total flange thickness $t_{\rm T}$ was taken to be the 448 manufacture's specified thickness. For the sensitivity analysis, c was varied between values based on 449 those measured on the real specimens. The wrinkle amplitude $A_{\rm w}$ was varied from 0 mm to the maximum 450 practical amplitude of 6.5 mm, where the top fibre mat dips down and touches the bottom fibre mat. Note 451 that this amplitude is over three times greater than the maximum measured amplitude of 2.0 mm shown in 452 Figure 16(c).

453 Both the elastic properties of the top flange and the failure load were determined using the 454 approach outlined in Bogetti et al. [23], which is based on dividing the model into discrete segments dx 455 and systematically applying 2D classical laminate theory to each segment (Figure 19(b)). The elastic and 456 strength properties of each lamina were taken from those reported in a previous study [28] for a similar 457 pultruded GFRP material, and are given in Table 2. The average axial and bending stiffnesses were 458 obtained by integrating the laminate stiffness matrices of each differential segment over the total length L. 459 A quadratic failure criterion was used to determine intralaminar failure for each dx segment 460 within the laminate. Additionally, interlaminar failure between the wrinkled mat-to-roving interface was 461 considered by applying a weakening factor of 0.3 to the normal and shear strengths of the mat layer. This

weakening factor was derived in [28] for a FE model of a similar pultruded GFRP material using a sensitivity analysis. A uniaxial tensile load was considered in this analysis as this partially reflects the stress state at the near surface wrinkles present in the pultruded deck under tyre loading. Under a tensile load the wrinkled mat layer will develop local out-of-plane normal σ_3 and interlaminar shear τ_{13} stresses due to the misalignment angle θ . These stresses are calculated by tensorial transformation of the *x*-

467 direction stress σ_x .

468 6.2 Results and discussion

Figure 20(a) and (b) show the reduction in axial and bending stiffnesses as a function of wrinkle amplitude A_w , respectively. The stiffnesses are normalised relative to the values for $A_w = 0$. In each figure, two curves are shown for different wrinkle widths of c = 2.0 mm and c = 5.0 mm, and the area less than $A_w = 2.0$ is shaded in grey to reflect the maximum range of measured wrinkle amplitude in the bridge deck specimens. From these figures the following conclusions can be drawn:

- The maximum *practical* wrinkle amplitude of 6.3 mm results in a reduction in axial and bending stiffnesses of 13% and 20%, respectively. As expected, the influence on bending stiffness is greater than that on axial stiffness due to the reduction in second moment of area as a consequence of the wrinkled fibre mat being closer to the flange mid-surface.
 At the maximum *measured* amplitude of 2.0 mm, the axial and bending stiffnesses reduced by
- 479 4.8% and 7.5% respectively. However, at this amplitude, a *smaller c* value of 2.0 mm produced a 480 greater reduction in stiffness. The cross-over point of the two curves for c = 2.0 mm and c = 5.0481 mm occurs because a narrower wrinkle produces a greater misalignment angle but over a smaller 482 length of flange compared to a wide wrinkle.

In Figure 21, the normalised knockdown tensile load – defined as the tensile failure load divided by the failure load at $A_w = 0$ – is plotted as a function of wrinkle amplitude for two different wrinkle widths of c = 2.0 mm and c = 5.0 mm, for an applied uniaxial tensile load. For $A_w = 0$, failure manifests as tensile failure in the mat layers. However, as the wrinkle amplitude is increased, the interlaminar normal and shear stress demand increases, resulting in interlaminar failure at the mat-roving interface. This transition point occurs at $A_w = 0.3$ mm and $A_w = 0.8$ mm for values of c = 2.0 mm and c = 5.0 mm, respectively.

| 489 | The predicted tensile load capacity reduction for the two measured flange wrinkles shown |
|-----|---|
| 490 | previously in Figure 16, which had $A_w = 2.0$ mm, $c = 4.7$ mm for specimen DT and $A_w = 1.5$ mm, $c = 2.8$ |
| 491 | mm for specimen BB, are 53% and 57%, respectively. The greater load capacity reduction in BB is due to |
| 492 | the fact that the wrinkle has a higher maximum tangent angle and thus a higher interlaminar stress |
| 493 | demand, despite the wrinkle having a smaller amplitude. |
| 494 | An important feature of Figure 21 is the sensitivity of the failure load to small changes in |
| 495 | wrinkle amplitude once interlaminar failure becomes the limiting criterion. For example, for a wrinkle |
| 496 | width of $c = 2.0$ mm, starting at $A_w = 0.3$ mm a 100% increase in A_w leads to a 40% drop in failure load. |
| 497 | This sensitivity suggests that the manufacturing tolerance for wrinkle amplitude should be well below that |
| 498 | which leads to interlaminar failure, which for small wrinkle widths is approximately $A_{\rm w} = 0.3$ mm. |

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7. Conclusions and future work

- The following conclusions can be drawn from this study:
- The fibre mats observed in a specific pultruded, cellular GFRP bridge deck exhibited different
 classes of misalignment in the transverse-to-pultrusion direction, each of which had a distinct
 morphology. Six different classes of mat misalignment were distinguished: double-backs, flips,
 near-surface wrinkles, corner wrinkles, waviness and drops.
- Double-backs, flips and corner wrinkles arose due to errors during the wrapping of the mats
 around the internal mandrel, near-surface wrinkles were caused by resin injection, waviness
 from shifting of the longitudinal rovings adjacent to the mats, and drops from the gaps between
 the internal mats.
- The mat misalignments were inconsistent between nominally identical WFJs taken from the different specimens of the same decking system. Out of the ten WFJs analysed, double backs occurred only in one, flips in three and wrinkles in four of the WFJs. The length, height and steepest tangent angle of the wavy mat layer passing through each junction were highly variable, as was the distance from the centre of the junctions at which the different mat layers were terminated.

515 Extreme forms of mat misalignment, for example double backs and flips, can result in 516 displacement of the mandrels during pultrusion due to unbalanced compression forces from the 517 fibres. Mandrel displacement resulted in variable flange thicknesses either side of the WFJs. 518 Mat misalignments are not solely a feature of the WFJs. Near-surface wrinkles near the local 519 flange midspan were more severe than those in the WFJs. Predictions based on classical laminate 520 theory showed that the near surface wrinkles observed in the flanges led to a laminate tensile 521 load capacity reduction of up to 57% due to the increased interlaminar stress demand and the 522 point of steepest tangent angle at the fibre mat-to-roving interface.

523 The taxonomy of fibre mat misalignments provided in this paper can be readily adopted by 524 pultruders as a framework for improving the nature and consistency of mat misalignments within cellular 525 profiles. In particular, the different misalignment categories outlined in this paper have been intentionally 526 linked to a specific aspect of the pultrusion process so that remediation measures can be targeted 527 appropriately. For example, the near-surface wrinkles observed at the junctions and local flange midspans 528 may be minimised or relocated through better control of the resin injection pressure within the die. 529 Currently, this defect occurs in areas of maximum local bending and shear when the deck is subjected to 530 the local contact pressure distribution of heavy lorry tyres, and hence may be compromising the deck's 531 long-term performance. Furthermore, the optical technique for measuring and characterising the severity 532 of mat misalignment can, with further research into the underlying mechanics, be implemented into 533 existing quality standards for pultruded profiles.

534 To further substantiate and extend the practical implications of the findings presented in this 535 paper, future work should use experimental observations of fracture patterns developed during both quasi-536 static and fatigue loading of the deck to establish which categories of misalignment are most significant to structural performance. This can be supplemented with data from FE simulations that independently vary 537 538 the severity of each type of mat misalignment identified in this paper. Furthermore, the variability of mat 539 misalignments should be assessed in the direction of pultrusion by making multiple section cuts along a 540 representative length of deck, although such longitudinal misalignments are expected to be much less 541 owing to the longitudinal tension exerted on the associated fibres during the pultrusion manufacturing 542 process.

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