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Integral High Order Sliding Mode Control of a Brake System

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Abstract—The aim of this paper is to present the design of a robust sliding mode control scheme for a vehicle system which consists of active brake systems. The proposed control strategy is based on the combination of high order sliding mode control methods and integral sliding mode control, taking advantage of the block control principle. The brake controller induces the antilock brake system feature by means of tracking the slip rate of the car, improving the stability in the braking process and preventing the vehicle from skidding.

Index Terms: Antilock braking systems, Automotive control, Sliding mode control, Integral control

I. INTRODUCTION

One of the most important control objectives in automotive systems is to provide adequate safety and to keep the vehicle occupants comfortable. An example is the controller design for the brake system. The wheel skidding is an undesirable phenomena which often results when the wheels get locked, leading to lose control of the car. Therefore, it is necessary to design brake controllers that induce the anti-lock brake system (ABS), preventing wheels from locking up during hard braking (Petrov *et al.*, 1977; Rittmannsberger, 1988; Emig *et al.*, 1990).

For many drivers the ABS usually offers improved vehicle control and decreases stopping distances on dry and slippery surfaces. In contrast, ABS can significantly increase braking distance on loose surfaces like gravel or snow-covered pavement, although still improving vehicle control (Burton *et al.*, 2004). Generally, the ABS design is based on the idea of reducing the pressure on the brake cylinder when the wheel is close to lock. This is done until the wheel velocity exceeds some threshold value. This breaking process may be improved by regulating the wheel slip (Tan and Chin, 1991). Thus, modern ABS systems not only try to prevent wheels from locking but also try to maximize the breaking forces generated by the tires to prevent that the longitudinal slip ratio exceeds an optimal value (Rajamani, 2011).

The main difficulties arising in the controller design for the brake system are due to uncertainties and the high nonlinearity presented in the system model. Therefore, the ABS has become an attractive examples for research in area of robust control. As proposals, several works have been

applied the sliding mode (SM) technique to design a slip-ratio controller for the vehicle stability; some examples are (Tan and Chin, 1991; Unsal and Pushkin, 1999; Ming-Chin and Ming-Chang, 2003; Tanelli *et al.*, 2009; Delprat and Ferreira de Loza, 2011), including the extremum seeking controllers in order to maximize the friction between the tire and the road (Chin *et al.*, 1992; Drakunov *et al.*, 1995; Tunay, 2001).

In this paper, the main purpose is to propose an ABS controller in presence of external disturbances and parametric variations, which are supposed to be unknown but bounded. The solution for the exposed problem is based on SM algorithms. These algorithms are proposed with the idea to drive the dynamics of a system to a sliding manifold, that is an integral manifold with finite reaching time (Drakunov and Utkin, 1992); exhibiting features such as SM motion finite time convergence, robustness to uncertainties including plant parameter variations and external bounded disturbances (Utkin *et al.*, 2009). To design the brake controller, a combination of block control feedback linearization (Loukianov, 2002), integral SM control (Utkin *et al.*, 2009) and high order quasi-continuous SM algorithms (Levant, 2005) is used in order to obtain an exact rejection of the system disturbances, similarly to the method shown by (Estrada and Fridman, 2010). The integral SM control methods guarantee the robustness of the closed-loop system throughout the entire response starting from the initial time instant, and reduces the sliding functions gains in comparison with the standard SM control. This approach is based on the integral nested SM control technique (Huerta-Avila *et al.*, 2007), which combines the integral SM control (Utkin *et al.*, 2009) with the nested SM control (Adhami-Mirhosseini and Yazdanpanah, 2005) and, has been applied for the ABS design as shown in (Sánchez-Torres *et al.*, 2013), bounding the effect of disturbances to an arbitrary small vicinity of zero, however without exact rejection.

The work is organized as follows: The mathematical model for the brake system is presented in Section II. Section III exposes the design of the SM controllers for the brake, with special emphasis in the sliding manifold design. The simulation results are presented in Section IV,

verifying the performance of the proposed control strategy. Finally, the conclusions are presented in Section V.

II. MATHEMATICAL MODEL

In this section, the dynamic equations which represent brake system, are presented. This system consider a quarter of the vehicle model and includes the dynamics of pneumatic brake system, wheel and vehicle.

A. Pneumatic Brake System Equation

The specific configuration of this system model considers the brake disk system. The brake holds the wheel, as a result of the air pressure increment in the brake cylinder. The air entrance trough the pipes from the central reservoir and the air expulsion from the brake cylinder to the atmosphere, are regulated by a common valve.

Considering Figure 1, the brake torque T_b is supposed to be proportional to the pressure P_b in the brake cylinder

$$T_b = k_b P_b \quad (1)$$

with $k_b > 0$. For the brake system, an approximated first order model of pressure changes in the brake cylinder due to the opening of the valve, is applied (Clover and Bernard, 1998):

$$\tau \dot{P}_b + P_b = P_c \quad (2)$$

where P_c is the pressure inside the central reservoir, and τ is the time constant of the pipelines. The atmospheric pressure, P_a , is considered as equal to zero.

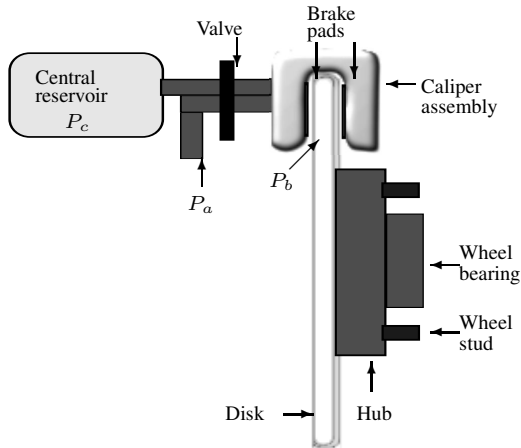


Fig. 1. Pneumatic brake scheme

B. Wheel Motion Equations

To describe the wheel motion, the mathematical model derived in (Kruchinin *et al.*, 2001) is used.

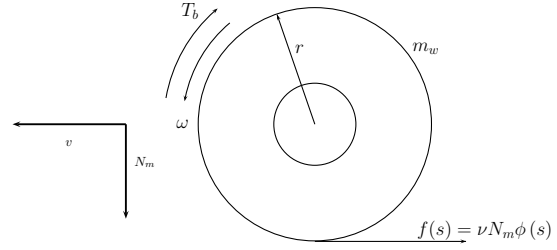


Fig. 2. Active suspension and wheel forces and torques

Considering Figure 2, the dynamics of the angular momentum variation relative to the rotation axis, are represented by

$$J\dot{\omega} = r f(s) - b_b \omega - T_b \quad (3)$$

where ω is the wheel angular velocity, J is the wheel inertia moment, r is the wheel radius, b_b is a viscous friction coefficient due to wheel bearings, $f(s)$ is the contact force of the wheel and s is the slip rate.

The expression for the longitudinal component of the contact force in the motion plane is

$$f(s) = \nu N_m \phi(s) \quad (4)$$

where ν is the nominal friction coefficient between the wheel and the road, N_m is the normal reaction force in the wheel and it is defined by $N_m = mg$, with g the gravity acceleration and m the mass supported by the wheel.

The function $\phi(s)$ represents the longitudinal friction/slip characteristic relation between the tire and road surface. In general, its calculation is based on experimental tests (Bakker *et al.*, 1987). A plot of this function for different surfaces is shown in Figure 3.

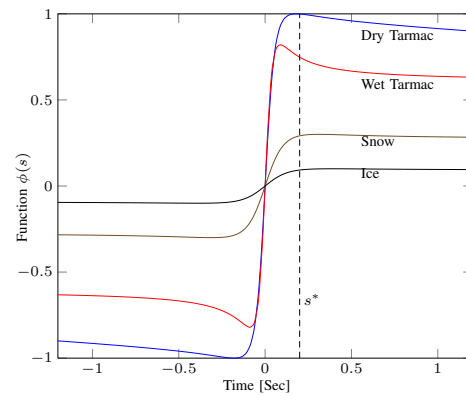


Fig. 3. Characteristic function $\phi(s)$

To approximate the function $\phi(s)$, the Pacejka formula (Bakker *et al.*, 1987) is used. This model is defined as follows:

$$\phi(s) = D \sin(C \arctan(Bs - E(Bs - \arctan(Bs))))$$

where B , C , D and E are the model parameters which are adjusted by using experimental data.

Based on the mentioned experimental tests, the Pacejka formula produces a good approximation of the tire/road friction interface, fitting accurately the data obtained via measurement of the road conditions (Pacejka, 1981; Bakker *et al.*, 1989; Pacejka, 2012). Table I shows the estimated parameters for different road conditions.

Surface	B	C	D	E
Dry Tarmac	10	1.9	1	0.97
Wet Tarmac	12	2.3	0.82	1
Snow	5	2	0.30	1
Ice	4	2	0.10	1

TABLE I

PACEJKA MODEL PARAMETERS FOR DIFFERENT ROAD SURFACES.

The slip rate s is defined as

$$s = \frac{v - r\omega}{v} \quad (5)$$

where v is the longitudinal velocity of the wheel mass center. For $s > 0$, the function $\phi(s)$ represents braking conditions. In Figure 3, s^* represents the desired value for the slip rate s . A suitable choice of s^* is value in the small interval $0.15 \leq s^* \leq 0.25$, this selection ensures a high longitudinal friction for the different road conditions.

The equations (3)-(5) characterize the wheel motion.

C. The Vehicle Motion Equation

The vehicle longitudinal dynamics considered without lateral motion, are described by

$$M\dot{v} = -F(s) - F_a(v, t) \quad (6)$$

where M is the total vehicle mass, F_a is the aerodynamic drag force which is proportional to the vehicle velocity. This force is defined as

$$F_a(v, t) = \frac{1}{2}\rho C_d A_f (v + v_w(t))^2$$

where ρ is the air density, C_d is the aerodynamic coefficient, A_f is the frontal area of vehicle, v_w is the wind velocity, and the contact force of the vehicle $F(s)$ is modeled of the form

$$F(s) = \nu N_M \phi(s) \quad (7)$$

where N_M is the normal reaction force of the vehicle, defined as $N_M = Mg$.

D. State Space Equations

The complete system (2)-(6), using the state variables $\mathbf{x} = [x_1 \ x_2 \ x_3]^T = [\omega \ P_b \ v]^T$, is presented in the following form:

$$\begin{aligned} \dot{x}_1 &= -a_1 x_1 + a_2 f(s) - a_3 x_2 \\ \dot{x}_2 &= -a_4 x_2 + bu \\ \dot{x}_3 &= -a_5 F(s) - a_5 F_a(x_3, t) \end{aligned} \quad (8)$$

with the output

$$y = x_1 \quad (9)$$

where $a_1 = b_b/J$, $a_2 = r/J$, $a_3 = k_b/J$, $a_4 = 1/\tau$, $a_5 = 1/M$, $b = 1/\tau$, $u = P_c$.

III. CONTROLLER DESIGN

The task of controlling the wheels rotation is studied in this section. The controller is designed such that the longitudinal force due to the contact of the wheel with the road, is near to the maximum value in the time period valid for the model. This maximum value is reached as a result of the brake valve effort (see Figure 1).

Considering the direct action of the pressure P_b in the brake cylinder over the wheel, the output tracking error

$$e_1 = x_1 - \frac{1 - s^*}{r} x_3. \quad (10)$$

is defined.

From (8) and (10), with the variables x_1 and x_3 written in terms of e_1 , it follows that

$$\dot{e}_1 = f_1(e_1) + b_1 x_2 + \Delta_1(\mathbf{x}, t) \quad (11)$$

where $f_1(e_1) = \left[a_2 N_m + \frac{1-s^*}{r} a_5 N_M \right] \nu \phi(s) - a_1 x_1$ and $b_1 = -a_3$. The disturbance term $\Delta_1(\mathbf{x}, t)$ in (11) contains the unknown aerodynamic drag force $-\frac{1-s^*}{r} a_5 F_a(x_3, t)$, the reference derivative \dot{s}^* and, the variations of the friction parameter ν . In the control design, the term $\Delta_1(\mathbf{x}, t)$ will be considered as a bounded disturbance.

To stabilize the dynamics for e_1 in (11), x_2 can be selected as a stabilizing term in form of a virtual controller, the desired value for x_2 , x_{2des} is determined as

$$x_{2des} = x_{2des}^0 + x_{2des}^1 \quad (12)$$

where x_{2des}^1 will be designed to reject the disturbance $\Delta_1(\mathbf{x}, t)$ in finite time by using the integral sliding mode technique (Utkin *et al.*, 2009) in combination quasi-continuous SM control (Levant, 2005). The term x_{2des}^0 is such that e_1 converges exponentially to zero.

In order to establish the control x_{2des} in (11), the error variable e_2 is defined as

$$e_2 = x_2 - x_{2des} \quad (13)$$

and (11) is rewritten as

$$\dot{e}_1 = f_1(e_1) + b_1 e_2 + b_1 x_{2des} + \Delta_1(\mathbf{x}, t). \quad (14)$$

To calculate the term x_{2des}^1 , the variable σ_1 is proposed as

$$\sigma_1 = e_1 + z_1 \quad (15)$$

where z_1 is a SM integral variable to be defined below.

From (11) and (15), the dynamics for σ_1 are given by

$$\begin{aligned} \dot{\sigma}_1 &= f_1(e_1) + b_1 x_{2des}^0 + b_1 x_{2des}^1 + b_1 e_2 \\ &\quad + \Delta_1(\mathbf{x}, t) + \dot{z}_1 \end{aligned} \quad (16)$$

where the derivative \dot{z}_1 is assigned of the form

$$\dot{z}_1 = -f_1(e_1) - b_1 x_{2des}^0 \quad (17)$$

with $z_1(0) = -e_1(0)$ in order to fulfill the requirement $\sigma_1(0) = 0$. With this selection of \dot{z}_1 , the system (16) reduces to

$$\dot{\sigma}_1 = b_1 x_{2des}^1 + b_1 e_2 + \Delta_1(\mathbf{x}, t). \quad (18)$$

To enforce sliding motion on the manifold $\sigma_1 = 0$ despite of the disturbance $\Delta_1(\mathbf{x}, t)$, the term x_{2des}^1 in (18) is chosen as

$$x_{2des}^1 = b_1^{-1} \xi \quad (19)$$

with ξ the solution of

$$\dot{\xi} = -\alpha \frac{\dot{\sigma}_1 + \beta |\sigma_1|^{\frac{1}{2}} \text{sign}(\sigma_1)}{|\dot{\sigma}_1 + \beta |\sigma_1|^{\frac{1}{2}}|} \quad (20)$$

where $\alpha > 0$ and $\beta > 0$.

When the motion on the manifold $\sigma_1 = 0$ is reached, the solution of $\dot{\sigma}_1 = 0$ in (16)

$$b_1 \{x_{2des}^1\}_{eq} = \Delta_1(\mathbf{x}, t). \quad (21)$$

shows that the disturbance $\Delta_1(\mathbf{x}, t)$ is rejected by the equivalent control $b_1 \{x_{2des}^1\}_{eq}$ (Utkin, 1992). Therefore, the dynamics on $\sigma_1 = 0$ is given by

$$\dot{e}_1 = f_1(e_1) + b_1 x_{2des}^0. \quad (22)$$

Thus, the desired dynamics $-k_1 e_1$ for \dot{e}_1 in (22) are introduced by means of

$$x_{2des}^0 = b_1^{-1} [-f_1(e_1) - k_1 e_1] \quad (23)$$

where $k_1 > 0$. Hence, with (23) in (17), \dot{z}_1 reduces to

$$\dot{z}_1 = k_1 e_1. \quad (24)$$

From (13), it follows that

$$\dot{e}_2 = bu - \Delta_2(\mathbf{x}, t) \quad (25)$$

where the term $\Delta_2(\mathbf{x}, t) = \dot{x}_{2des}$ is a bounded disturbance.

Considering the types of valve that can vary its position in a continuous range, the integral SM control method is applied to (25) in order to induce a sliding mode on the manifold $e_2 = 0$ in spite of the disturbance $\Delta_2(\mathbf{x}, t)$. With this aim, the control u is decomposed as

$$u = u_0 + u_1 \quad (26)$$

where u_1 will reject the disturbance $\Delta_2(\mathbf{x}, t)$ in finite time and u_0 is designed to stabilize e_2 in finite time.

To calculate the term u_1 , the variable σ_2 is proposed as

$$\sigma_2 = e_2 + z_2 \quad (27)$$

where z_2 is a SM integral and it will be defined below.

Considering the equations (25) and (27), the dynamics for σ_2 are given by

$$\dot{\sigma}_2 = bu_0 + bu_1 - \Delta_2(\mathbf{x}, t) + \dot{z}_2. \quad (28)$$

Selecting \dot{z}_2 as

$$\dot{z}_2 = -bu_0 \quad (29)$$

the system (28) reduces to

$$\dot{\sigma}_2 = bu_1 - \Delta_2(\mathbf{x}, t). \quad (30)$$

To enforce sliding motion on the manifold $\sigma_2 = 0$ despite of the disturbance $\Delta_2(\mathbf{x}, t)$, the term u_1 in (30) is chosen as the *super-twisting* algorithm (Levant, 1993)

$$u_1 = b^{-1}(u_{11} + u_{12}) \quad (31)$$

where $u_{11} = -k_{11} |\sigma_2|^{\frac{1}{2}} \text{sign}(\sigma_2)$ and $\dot{u}_{12} = -k_{12} \text{sign}(\sigma_2)$, with $k_{11} > 0$ and $k_{12} > 0$.

On the manifold $\sigma_2 = 0$, the solution of $\dot{\sigma}_2 = 0$ in (28) leads to

$$b \{u_1\}_{eq} = \Delta_2(\mathbf{x}, t). \quad (32)$$

showing that the disturbance $\Delta_2(\mathbf{x}, t)$ is rejected by the equivalent control $b \{u_1\}_{eq}$ (Utkin, 1992). Thus, the dynamics for $e_2 = 0$ is given by

$$\dot{e}_2 = bu_0. \quad (33)$$

Finally, the desired dynamics for \dot{e}_2 in (33) are introduced with

$$u_0 = -k_2 b^{-1} |e_2|^{\frac{1}{2}} \text{sign}(e_2) \quad (34)$$

where $k_2 > 0$. Thus, with (34) in (29), \dot{z}_2 yields to

$$\dot{z}_2 = k_2 |e_2|^{\frac{1}{2}} \text{sign}(e_2). \quad (35)$$

Using the nonsingular transformation (10), (13), (19) and (23)

$$\begin{aligned} e_1 &= x_1 - \frac{1-s^*}{r} x_3 \\ e_2 &= x_2 + b_1^{-1} [-f_1(e_1) - k_1 e_1] + b_1^{-1} \xi \\ \sigma_1 &= e_1 + z_1 \\ \sigma_2 &= e_2 + z_2 \end{aligned}$$

with the integral variables z_1 , z_2 and ξ defined by (24), (35) and (20), respectively; the extended closed loop system (11) and (25) is presented as

$$\begin{cases} \dot{e}_1 &= -k_1 e_1 + \xi + b_1 e_2 + \Delta_1(\mathbf{x}, t) \end{cases} \quad (36)$$

$$\begin{cases} \dot{\sigma}_1 &= \xi + b_1 e_2 + \Delta_1(\mathbf{x}, t) \\ \dot{\xi} &= -\alpha \frac{\dot{\sigma}_1 + \beta |\sigma_1|^{\frac{1}{2}} \text{sign}(\sigma_1)}{|\dot{\sigma}_1 + \beta |\sigma_1|^{\frac{1}{2}}|} \end{cases} \quad (37)$$

$$\begin{cases} \dot{e}_2 &= -k_2 |e_2|^{\frac{1}{2}} \text{sign}(e_2) - k_{11} |\sigma_2|^{\frac{1}{2}} \text{sign}(\sigma_2) \\ &+ \dot{u}_{12} + \Delta_2(\mathbf{x}, t) \end{cases} \quad (38)$$

$$\begin{cases} \dot{\sigma}_2 &= -k_{11} |\sigma_2|^{\frac{1}{2}} \text{sign}(\sigma_2) + u_{12} + \Delta_2(\mathbf{x}, t) \\ \dot{u}_{12} &= -k_{12} \text{sign}(\sigma_2). \end{cases} \quad (39)$$

Considering the disturbances in the closed-loop system (36) - (39) as fulfilling the following conditions:

$$\left| \dot{\Delta}_1(\mathbf{x}, t) \right| \leq L_1 \quad (40)$$

$$\left| \dot{\Delta}_2(\mathbf{x}, t) \right| \leq L_2 \quad (41)$$

in some admissible region Ω_0 with $L_1 > 0$ and $L_2 > 0$, the stability of the closed-loop system (36) - (39) is outlined in the stepwise procedure

Step A) The SM stability of the projection motion (39).

Step B) Reaching phase of the projection motion (38).

Step C) The SM stability of the projection motion (37).

Step D) The SM motion stability of (36) on the manifold $e_2 = 0$.

Step A) Using the transformation $q = u_{12} + \Delta_2(\mathbf{x}, t)$, the system (39) yields to

$$\begin{aligned}\dot{\sigma}_2 &= -k_{11}|\sigma_2|^{\frac{1}{2}}\text{sign}(\sigma_2) + q \\ \dot{q} &= -k_{12}\text{sign}(\sigma_2) + \dot{\Delta}_2(\mathbf{x}, t).\end{aligned}\quad (42)$$

Under the assumption (41), it follows that if $k_{11} > 0$ and $k_{22} > 3L_2 + 4\left(\frac{L_2}{k_{11}}\right)^2$, then $(\sigma_2, q) = (0, 0)$ in finite time despite of the perturbation $\Delta_2(\mathbf{x}, t)$ (Moreno and Osorio, 2008).

Step B) In (38), the motion on the manifold $\sigma_2 = 0$ is given by

$$\dot{e}_2 = -k_2|e_2|^{\frac{1}{2}}\text{sign}(e_2).\quad (43)$$

Let the Lyapunov candidate $V = \frac{1}{2}e_2^2$. From (43), it follows

$$\dot{V} = -k_2 2^{\frac{3}{4}} V^{\frac{3}{4}}.\quad (44)$$

Therefore, e_2 converges to zero in finite time; establishing a SM in the manifold $e_2 = 0$.

Step C) For (37), the motion on the manifold $e_2 = 0$ is given by

$$\begin{aligned}\dot{\sigma}_1 &= \xi + \Delta_1(\mathbf{x}, t) \\ \dot{\xi} &= -\alpha \frac{\dot{\sigma}_1 + \beta |\sigma_1|^{\frac{1}{2}}\text{sign}(\sigma_1)}{|\dot{\sigma}_1| + \beta |\sigma_1|^{\frac{1}{2}}}\end{aligned}\quad (45)$$

With the transformation $\varphi_1 = \sigma_1$ and $\varphi_2 = \dot{\sigma}_1$, the system (45) is written as

$$\begin{aligned}\dot{\varphi}_1 &= \varphi_2 \\ \dot{\varphi}_2 &= -\alpha \frac{\varphi_2 + \beta |\varphi_1|^{\frac{1}{2}}\text{sign}(\varphi_1)}{|\varphi_2| + \beta |\varphi_1|^{\frac{1}{2}}} + \dot{\Delta}_1(\mathbf{x}, t).\end{aligned}\quad (46)$$

Under the condition (40), there exist $\alpha > 0$ and $\beta > 0$ such that the system (46) is finite time stable (Levant, 2005), i.e., its solution converges in finite time to the origin $(\varphi_1, \varphi_2) = (\sigma_1, \dot{\sigma}_1) = (0, 0)$, inducing a SM on $\sigma_1 = 0$.

Step D) The motion for (36) on the set $(e_2, \sigma_1) = (0, 0)$ is given by the system

$$\dot{e}_1 = -k_1 e_1\quad (47)$$

which is exponentially stable.

Note that during the braking process $x_3 > 0$ and with $s \approx s^*$, due to the control action, the dynamics for x_3 becomes

$$\dot{x}_3 = -a_5 F(s^*) - a_5 F_a(x_3, t).\quad (48)$$

From the vehicle mechanics, the term $a_5 F_a(x_3, t)$ can be considered to be bounded $|a_5 F_a(x_3, t)| \leq \delta_3$, with $\delta_3 > 0$. In addition, it can be considered that $a_{11} F(s^*) \gg \delta_7$.

Therefore, let $\gamma = a_{11} F(s^*) - \delta_7$, and $W = \frac{1}{2}x_3^2$, then $\dot{W} = -x_3 [a_5 F(s^*) + a_5 F_a(x_3, t)] < -\gamma x_3$. Hence, $\dot{W} < -\gamma_0 \sqrt{W}$, where $\gamma_0 = \gamma \sqrt{2}$. Thus, $x_3 = 0$ in finite time. Also, from (10) it follows that $x_1 = 0$ in finite time.

IV. SIMULATION RESULTS

To show the effectiveness of the proposed control law, simulations are carried out on the vehicle model, the system parameters used are listed in Table II.

Parameter	Value	Parameter	Value	Parameter	Value
m_c	1800	τ	0.0043	C_d	0.65
m_w	50	J	18.9	ρ	1.225
A_f	6.6	k_b	100	v_w	-6
b_b	0.08	g	9.81	r	0.35
ν	0.95	-	-	-	-

TABLE II
VALUES OF THE SYSTEM PARAMETERS (MKS UNITS).

To expose the performance, the proposed controllers are simulated in iced road conditions. The numerical integration was carried by applying the Euler method with a step time of 10^{-3} s. The parameters for this case are given in Table I. For the proposed SM controller, the parameters used in the control law for the brake are $\alpha = 30$, $\beta = 0.001$, $k_1 = 70$, $k_{11} = 10$, $k_{12} = 50$ and $k_2 = 100$.

In order to increase the longitudinal friction force, it is supposed that slip tracks a constant signal $s^* = 0.2$. In Figure 3 it can be seen that the selected reference s^* produces a value close to the maximum of the function $\phi(s)$ in every road condition.

On the other hand, to show robustness property of the control algorithm in presence of parametric variations we introduce a change of the friction coefficient ν which produces different contact forces, namely $F(s)$ and $f(s)$. Then, $\nu = 0.8$ for $t < 10$ s, $\nu = 0.95$ for $t \in [10, 25)$ s, and $\nu = 0.9$ for $t \geq 25$ s. It is worth mentioning that just the nominal values were considered in the control design.

The figure 4 shows the wheel speed x_1 . The figure 5 presents the longitudinal speed x_3 . The brake controller should be turned off when the longitudinal speed is close to zero.

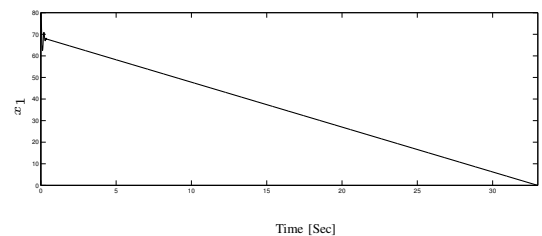


Fig. 4. Wheel velocity x_1

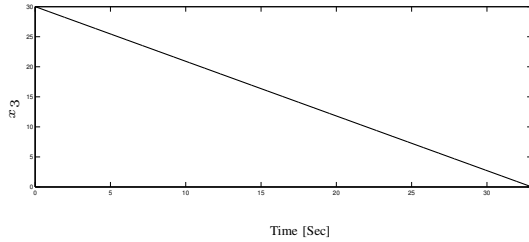


Fig. 5. Vehicle velocity x_3

The figures 6 and 7 shows the tracking error e_1 for the slip and the error e_2 respectively.

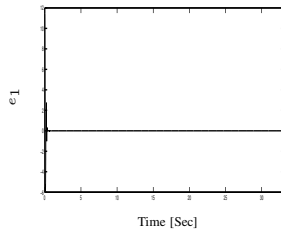


Fig. 6. Tracking error e_1

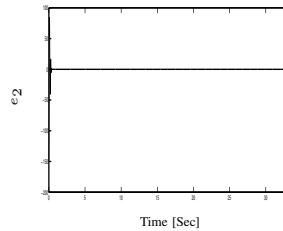


Fig. 7. Error e_2

The control signal u response for the brake is presented in Figure 8.

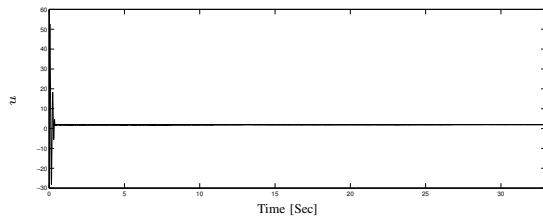


Fig. 8. Control signal u

V. CONCLUSIONS

In this work a robust control scheme was designed for the brake systems using integral SM control technique combined with quasi-continuous SM algorithms. The controller uses the SM control for achieve the error tracking asymptotically to a zero and the integral SM guarantees the disturbance rejection. The simulation results show good performance.

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