Stamford, Cal. TECHNICAL REPORT NO. 3

Institute, June 1951, Stamford Ulan

RSC

MAY 1.6 1953

HAYDEN

LIBRARY

Published in the Journal of Theat Transfer

QC320 M41 H43 NO.3

HEAT TRANSFER AND PRESSURE DROP DATA FOR HIGH HEAT FLUX DENSITIES TO WATER AT HIGH SUBCRITICAL PRESSURES

BY

WARREN M. ROHSENOW AND JOHN A. CLARK

FOR

THE OFFICE OF NAVAL RESEARCH CONTRACT N5ori-07827 NR-035-267 D.I.C. PROJECT NUMBER 6627

APRIL 1, 1951



MASSACHUSETTS INSTITUTE OF TECHNOLOGY DIVISION OF INDUSTRIAL COOPERATION CAMBRIDGE 39, MASSACHUSETTS



Room 14-0551 77 Massachusetts Avenue Cambridge, MA 02139 Ph: 617.253.5668 Fax: 617.253.1690 Email: docs@mit.edu http://libraries.mit.edu/docs

DISCLAIMER OF QUALITY

Due to the condition of the original material, there are unavoidable flaws in this reproduction. We have made every effort possible to provide you with the best copy available. If you are dissatisfied with this product and find it unusable, please contact Document Services as soon as possible.

Thank you.

Some pages in the original document contain pictures, graphics, or text that is illegible.

TECHNICAL REPORT NO. 3

HEAT TRANSFER AND PRESSURE DROP DATA FOR HIGH HEAT FLUX DENSITIES TO WATER AT HIGH SUBCRITICAL PRESSURES

by

Varran M. Rohsenow and John A. Clark

For the Office of Naval Research

N5 ori-07827

NR 035-267

DIC PROJECT NO. 6627

April 1, 1951

DIVISION OF INDUSTRIAL COOPERATION

Massachusetts Institute of Technology

Cambridge, Massachusetts

• •

HEAT TRANSFER AND PRESSURE DROP DATA FOR HIGH HEAT FLUX DENSITIES

TO VATER AT XIGH SUBCRITICAL PRESSURES

by

Warren M. Rohsenow# and John A. Clark##

SUMMARY

Local surface coefficients of heat transfer, overall pressure drop data and mean friction factor are presented for heat fluxes up to 3.5×10^6 Btu/hr ft² for water flowing in a nickel tube under the following conditions: mass rates of flow up to 5.6×10^6 lb_m/hr ft² (or inlet velocities up to 30 ft/sec), absolute pressures up to 2000 psia, and liquid subcooling between 50 F and 250 F. The test section dimensions were 0.180 inch I.D. and 9.4 inches long.

INTRODUCTION

Very little information has been available to predict heat transfer and pressure drop performance for liquids with severe temperature gradients adjacent to the surface, particularly when the surface temperature exceeds the normal boiling temperatures. Recent works published by Knowles (3), Kreith and Summerfield (1) (2), and McAdams et al (4) present data for this type heat transfer process in the lower pressure ranges. The data presented here for water was obtained in August, 1950, and represents some of the initial data available for water at pressures up to 2000 psia. Simultaneously, similar data has been gathered at U.C.L.A. This particular type of heat transfer process is encountered in liquid-cooled rocket motors and in more recent steam power plant equipment.

* Asst. Prof. Mech. Eng., M. I. T., Cambridge, Mass. ** Instructor, Mech. Eng., M. I. T., Cambridge, Mass.

APPARATUS

-2-

The test apparatus is a closed system consisting of a vertical test section of pure nickel, a Hayward-Tyler centrifugal pump, a calibrated orifice, a heat exchanger, a pressure vessel, and an ion exchanger. A layout is given in Fig. 1. Power is supplied from two 36 KW 12 volt DC Generators driven by 440 volt, 3 phase, AC 600 rpm synchronous motors. The generator outputs are connected in series and provide a range of 0-24 volts and 0-3000 amperes. Thermocouples are located at the test water inlet and outlet and along the outer wall of the test section, readings being taken by means of a Rubicon potentiometer, Model 2703 and a Rubicon spotlight galvanometer Model 3401-H. A bourdon type Heise pressure gage 0-2000 psi is located at the inlet to the test section. A Barton differential pressure gage 0-300 inches of water is connected across inlet and outlet of the test section to record the pressure drop.

The test section (figure 2) consists of a pure nickel tube (International Nickel Co. "L" Nickel) of .1805 in. inside diameter, .2101 in. outside diameter, and a length of 9.4 inches. Threaded bushings of "L" nickel are gold soldered to each end of the test section, and make contact with bronze end mounts, which support the test section assembly and carry current from the bussbars.

The test water was circulated by a four-stage centrifugal pump manufactured by Hayward-Tyler Company, Ltd., of England. The pump impeller and the motor assembly are both enclosed in a single stainless steel casing capable of being operated at a pressure of 4000 psia with a head of 300 ft. of water at 50 gpm. The heat exchange system which removes heat equal to the electrical energy supplied employes an intermediate fluid, silicone DC-701, between the test and city vater cooling water, in order to assure an absence of local boiling of the cooling, in this part of the circuit. Flow measurements were made by a calibrated orifice with a Barton differential pressure gage having a range of 0-100 inches of water. The system pressure is maintained by the vapor pressure in an insulated pressure vessel heated by chromalox electric heaters.

EXPERIMENTAL TECHNIQUES

- 3 -

<u>Determination of Temperatures at Liquid-Wall Interface</u>. Because of mechanical problems it was impractical to measure the tube inner wall temperature directly. Instead, this temperature was calculated from measurements of the outer wall temperature, the electric current, and the geometrical and physical properties of the nickel tube.

The tube outer wall temperature was measured at seven points above the tube as shown in fig. 3 from twelve chromel-constantan thermocouples electrically insulated from the tube wall by a small sheet of 0.0015 in. thick mica. Around the tube was a copper shield electrically heated in three sections, each controlled by a variac. The space between the tube and the shield was filled with Kaolin wool insulation.

The shield temperature was adjusted to the same temperature as the test section thermocouple. Under such conditions the reading of the outer wall thermocouple can be considered to be the temperature of the tube outer wall.

In the surface boiling region the tube wall temperatures are fairly uniform; hence a uniform setting of shield temperatures for any one operating condition resulted in accurately determined wall temperatures. In the non-boiling region, the test section had an axial temperature gradient requiring adjustments of the three variacs to cause the shield to follow closely the tube temperatures.

A mock-up of the test section assembly (fig. 2) was tested with condensing steam in the test section to determine the effect of shield temperature variation on tube thermocouple reading. It was found that when the difference between steam temperature and shield temperature was 10°F, the difference between steam temperature and tube thermocouple temperature was between 1/2 and 1°F as recorded by the various thermocouples along the length. These differences varied approximately linearly with each other. The inner wall temperature was calculated from a Taylor Series solution of the temperature distribution for heat conduction in an electrically heated tube with an adiabatic outer wall. This equation which is similar to that proposed by Kreith and Summerfield (1) is:

$$t_{o}-t_{w} = \frac{m}{k_{o} f_{o}} \left\{ \Delta \overline{x}^{2} + \frac{\Delta \overline{x}^{2}}{r_{o}} + \Delta \overline{x}^{4} \left[\frac{m(3\alpha + 4\alpha \beta t_{o} + \beta)}{6(1 + \beta t_{o})(1 + \alpha t_{o})} + \frac{1}{4\gamma_{o}^{2}} + \cdots \right] \right\} (1)$$
mere:
$$m \equiv \frac{3.412 \ I^{2} \ f_{m}^{2}}{2 \pi^{2} (t_{o}^{2} - \eta^{2})^{2}}$$

For accuracy within 0.5% the third term may be neglected, the working equation then becoming:

$$t_{o}-t_{w} = \frac{m}{k_{o}S_{o}} \left\{ \Delta \overline{x}^{2} + \frac{\Delta \overline{x}^{3}}{r_{o}} + \cdots \right\}$$
(1a)

This relation includes allowance for variation of electrical resistivity and thermal conductivity with temperature and for electrical resistivity with radius.

<u>Measurement of Heat-Flux Density</u>. The power dissipated in any portion of the test section is dependent upon the resistance, which, being a function of temperature, varies from point to point along the tube. The heat flux density then varies along the test section length and must be calculated to obtain accurate results. For a small element of length dx,

$$\frac{q}{A} = \frac{3.412 \ I^2 \ f_m \ dx}{T \ D_i \ dx \ T (r_i^2 - r_i^2)} = 1.148 \times 10^6 \ I^2 \ f_m \tag{2}$$

for the tube of 0.1805 in. I.D. and 0.2101 in. 0.D.

The electric current was determined from the measured voltage drop across a 0.00001667 ohm G. E. manganin shunt calibrated by the National Bureau of Standards. The shunt was in series with the test section and its voltage drop Was measured by a Rubicon potenticmeter. The voltage drop across the test section was measured by a potentiometer and a voltage divider network as illustrated in fig. 4. The resistors were General Radio Company resistors calibrated at the M.I.T. Electrical Instruments Laboratory. The potential taps were made of nickel held in place by stainless steel spring clips. The voltage drop across the test section is then

$$E_{ts} = E_1 \frac{R_1 + R_2 + R_3}{R_1}$$
(3)

<u>Test Water Bulk Temperature</u>. The method of determining the fluid temperature at any station along the tube consisted of integrating q/A from the inlet to the station in question and calculating the fluid temperature rise from the inlet. Neglecting axial heat flow and neglecting the effect of vapor bubbles on the effective specific heat of the water, an energy balance results in

$$\int_{T_{in}} dr = \frac{\pi P_i}{\varsigma_{iw}} \int_{0}^{n} \left(\frac{q}{A}\right) dx \qquad (4)$$

For pusposes of graphical integration, a plot as shown in fig. 5 was made. In general it was sufficiently accurate to make the actual calculation by a stepwise integration between any two stations n and (n-1) in the form

$$T_n - T_{n-1} = \frac{\pi D_i}{w \varphi} X_{n-(n-1)} (N_A)_{n-(n-1)}$$
 (5)

Data from this apparatus affords several checks on itself. The average (q/A) calculated from the power measurement should be equal to the average (q/A) as calculated from equation (2). Also the sum of the water temperature rises between stations should equal $T_{out} - T_{in}$.

Flow Measurement. The pressure drop across the orifice was measured with Barton Differential Pressure Gages with 0-100" H 0 scale. Its accuracy calibrated at atmospheric pressure against a water column was within 1% of full scale reading. The orifice coefficient calibrated by direct weighing at eight Reynolds numbers for pressure drop readings from 10 to 90" H 0 had a root-mean-square

- 5 -

deviation of 0.5% for the high flow range orifice and 0.9% for the low range orifice. The accuracy of the flow measurement should be within 1 per cent.

- 6 -

<u>Test-Section Pressure Drop Measurement</u>. Pressure drop across the test section is measured by means of a Barton Differential Pressure Gage, 0-300 inches of water, connected at the pressure taps as shown in Fig. 2. The pressure drop, Δp_i , across the heated section was determined from corrected gage reading by subtracting the pressure drops in the unheated sections at either end of the tube as determined from isothermal friction factors. The temperature of the water in the vertical test section is considerably higher than the temperature of the ambient water in the gage lead-in tubing, necessitating a correction factor to determine the actual pressure drop due to friction and momentum change. The equation relating these pressure drop values is as follows:

$$\Delta p_{f,m} = \Delta p_i + L_p \left(\gamma_0 - \overline{\gamma} \right) \tag{6}$$

Because of the addition of heat to the fluid its density changed in passing through the test section. This resulting momentum change must be subtracted from $\Delta \beta_{\rm int}$ to obtain the pressure drop associated with friction alone. Then

$$\Delta P_{f} = \Delta P_{f,m} - \frac{G^{2}}{g_{o}} \left(\frac{1}{f_{2}} - \frac{1}{f_{1}} \right)$$
(7)

A friction factor based on the friction pressure drop may be defined by the equation $G^2 \int dx$

$$\Delta p_{f} = 4 f_{f} \frac{G}{2 g_{o} D_{i}} \int_{o} \frac{dx}{f}$$
(8)

and when based on the pressure drop including both frictional and momentum effects

$$\Delta p_{f,m} = 4 f_{f,m} \frac{G^2}{2g_0 D_i} \int_0^{L} \frac{dx}{g} \qquad (9)$$

In each case the actual variation of density along the tube length was used in evaluating the integral.

Accuracy of Results. The tube thickness varied - 0.0003 in. Equation (2)

with the value of m introduced shows the first and major term of the series to be approximately inversely proportional to the square of the tube diameter. For the high heat flux tests the value of $T_o = T_w$ was about 60°F; then the maximum error of T_w attributable to tube thickness variation was 0.4°F and decreases as the heat flux is lowered.

The thermal conductivity data was obtained from the International Nickel Company and the electrical resistivity was determined experimentally to an estimated accuracy of $\pm 0.5\%$.

It is expected that the inner-wall temperature has been determined within a \pm 3 degree F error.

ION EXCHANGER

During early test runs black iron oxide was found to be depositing in the test section causing as much as a 50°F increase in tube wall temperature in two hours of operation at a fixed set of conditions. The source of iron ions was probably the Hayward-Tyler pump which, contrary to expectations, had a considerable amount of ordinary iron in contact with the test water. It was suggested that the deposit was formed because of the electrical potential gradient along the test water in the tube and the ions (hence electrical conductivity) of the test water. A mono-bed ion exchanger consisting of Rohn and Haas Resin MB-1 "Amberlite" in a stainless steel jacket equipped with suitable filters was added to the circuit as shown in figure 1. It is not as yet known whether this ionexchanger actually removed the objectionalbe ions or simply acted as a high grade filter, but subsequent to its installation all difficulty with deposition of scale on the heat transfer surface ceased.

EXPERIMENTAL PROCEDURE

The system was filled with freshly distilled water (0.70 ppm as NaCl and approximately 15 ml air/1) by means of an aspirator located at the top of the Hayward-Tyler pump. As soon as it was certain that the upper thrust bearing on

-7-

the Hayward-Tyler pump was immersed in water, the pump was started and water was circulated through the test loop. Degassing was accomplished by circulating the test circuit water through the heated pressure vessel vented to the atmosphere for a period of 1/2 to 3/4 of an hour. This period was found to be of sufficient length to reduce the oxygen content to approximately 1.5 ml air/1, as determined by the Winkler Technique. Subsequent to degassing, the system was sealed by closing the degassing vent valve and test water was circulated through the test section at high velocity. The bulk temperature of the water was increased by applying power to the test section and the system pressure was increased to the desired level by heating the water in the closed pressure vessel. The system pressure could be controlled to ± 2 psi by the chronilox strip heaters regulated by a variac. The inlet bulk temperature was controlled and maintained at the desired level by regulating the flow of silicone fluid through the intermediate heat exchanger. Cooling water flow through the ion-exchanger heat exchanger and the city water heat exchanger was fixed at its maximum rate at the beginning of the run and not thereafter adjusted. The water at the discharge of the ionexchanger was consistently at approximately 0.1 ppm NaCl.

EXPERIMENTAL RESULTS

Heat Transfer in Forced Convection Without Surface Boiling.

Local values of the heat transfer coefficient at stations 2 through 6 (Fig. 3) were evaluated by assuming a linear variation of fluid temperature with distance along the tube. The tube wall temperature varied along the tube in the non-boiling runs; so q/A was not uniform because the wall electrical resistivity varies with temperature. Nevertheless the assumption of linear fluid temperature resulted in at most a 3% error in the resulting local heat transfer coefficient for runs involving high rates of heat transfer at lower velocities.

The results of these runs are shown in Table I and figure 6. The local value of j is shown plotted against the local N_{Ref} . It is noted that there exists a separate curve for the points along the tube for each run and the points near the

- 8 -

end of the tube are correlated by the equation

$$\frac{h}{c_p G} \left(\frac{c_p \mu_f}{k}\right)^{4/3} = 0.019 \left(\frac{G D_i}{\mu_f}\right)^{-0.2}$$
(10)

» 9 m

which is below the Colburn correlation line by about 17%. The values of the heat transfer coefficient at the points toward the inlet end of the tube are higher than the correlation line which is drawn for the points near the outlet end of the tube. This is to be expected because of the build up of the thermal boundary layer. The tube has an L/d of 52. For most cases an L/d of approximately 50 is found to be necessary to form a fully developed thermal boundary layer; hence, the trend of the data seems to be reasonable.

There is, of course, the possibility that a film of contamination on the heat transfer surface would result in too high a temperature difference and thus reduce the j value. However, this effect is discounted as being negligible since it would have to account for an interface temperature error of from 20°F to 40° F to bring the correlation in line with the Colburn correlation. Doubtless the film exists since examination of used tubes showed a slight discoloring of the heat transfer surface. It was extremely thin, however, reflecting incident light as blue suggesting its thickness as the order of the wave length of that color light. Also, inspection of Fig. 8 shows that if the true Δ t is as much as 10° F below that reported then boiling would occur at temperatures less than saturation, which is improbable.

Energy balances comparing enthalpy change of liquid with the electrical energy were all within $\pm 2\%$, most of them being within $\pm 0.5\%$. Isothermal runs with the liquid temperatures above 400°F showed the inlet and outlet liquid thermocouple to agree within $\pm 1°F$ of the tube wall temperatures.

The values of the fluid properties were obtained from data taken from Wellman (8).

Heat Transfer in Forced Convection with Surface Boiling

Again local values of q/A as a function of temperature difference are studied. In these boiling runs the tube temperature, and hence q/A, is very nearly uniform along the tube. The temperature of the fluid is assumed to vary linearly with distance along the tube. This assumption is not strictly valid for conditions of high heat flux or heat transfer to a liquid with low subcooling where the percentage of volume occupied by vapor becomes significant. Since the effect of these vapor bubbles on bulk temperature changes could not be determined from the measurements taken it was assumed the fluid behaved as a liquid and as a check on the assumption the longitudinal variation in bulk enthalpy and thus bulk temperature was determined by a numerical integration of q/A with length. The results agreed within 2°F of the assumed linear variation.

Figure 7 illustrates the type of information obtained from a set of runs. Five data points are obtained for a run, each at nearly the same q/A value but a different value of liquid subcooling. Similar curves for other conditions are shown in figures 13 through 17 from the data in Tables I and II.

The curves of figure 7 for various values of subcooling can be brought together to fall on a single line by plotting q/A vs. $(t_v - t_{gat.})$, the vall superheat, as shown in figure 8. Here data in the region of surface boiling (Table II) are plotted for various values of fluid velocities and pressures. The points plotted here are the average values of the five points along the tube for each run. The value of q/A along the tube did not vary significantly but the value of $(t_v - t_{gat})$ varied within ± 2 °F from the mean value plotted. It is observed that at these high pressures the amount of wall superheat is very shall, generally less than 10°F at 2000 psia and less than 15°F at 1500 psia. At lower values of pressure near atmospheric Kreith and Summerfield (1) found values of wall superheat to be around 60°F. Errors in the smaller values of wall superheat are magnified on the log-log type of plot in figure 8. The uncertainties of ± 2 or 3°F in wall temperature values have greater emphasis here. These discrepancies

- 10 -

when referred to the value of $(t_w - t_b)$ are very small, however.

The general trend of the curves of figure 8 shows the same effect reported by previous investigators (1), (2), (3), and (4) in the lower pressure range, e.g., at the higher heat transfer rates the effect of fluid velocity decreases and the agitation of the fluid by the bubbles governs the rate of heat transfer and at the higher pressures less wall superheat exists at a given rate of heat transfer. <u>Pressure Drop in Forced Convection Without Surface Boiling</u>

Local values of pressure drop could not be obtained because such measurements would interfere with the heat transfer measurements; hence, overall values of pressure drop were obtained and are tabulated in Table III, and friction factors are shown in figure 9 as a function of bulk Reynolds number at the arithmetic mean value of inlet and outlet fluid temperatures. The f_f and $f_{f,m}$ values are shown compared with the isothermal values found in figure 51 of McAdams (5).

In figure 10 the ratio of isothermal friction factor to the friction factor with heat transfer is plotted as a function of $(\mathcal{M}_b, \mathcal{M}_b)$ at the arithmetic mean value of inlet and outlet fluid temperatures. The ratio involving $f_{f,m}$ is seen to correlate with the generally accepted line having a slope of 0.14, but the ratio involving f_f is correlated with a line having a slope of 0.60 for these data. <u>Pressure Drop in Forced Convection With Surface Boiling</u>

Data of Table IV is plotted in figures 11 and 12 and show the effect of heat flux on the pressure drop quantities for friction alone and for friction plus momentum change. When surface boiling begins the pressure drop begins to rise slightly as q/A is increased. A velocity effect and an effect of liquid subcooling or absolute pressure is observed. It appears that the effect of liquid subcooling is more pronounced than the effect of absolute pressure.

Since five local values of heat transfer coefficients were obtained for each value of overall pressure drop there is much less pressure drop data available. Generally one particular range of liquid subcooling values were associated with particular values of absolute pressure and velocity. At 2000 psia and 20 ft/sec

- 11 -

most of the data were taken with a mean subcooling of around 160°F; however, two runs were made at a liquid subcooling of around 237°F. The curves drawn through points indicate that the offect on pressure drop of liquid subcooling is probably more important than the effect of absolute pressure in ranges of these tests. More test data is needed to explore more fully this effect.

Summary of Results

1. For fully developed flow the non-boiling heat transfer data at 1500 psia and 2000 psia is correlated by the equation

$$\frac{h}{c_{p}G}\left(\frac{c_{p}\mu_{f}}{k}\right)^{2/3} = 0.019\left(\frac{GD_{i}}{\mu_{f}}\right)^{-0.2}$$

2. With surface boiling the heat transfer data plotted against $(t_w - t_{sat})$, wall superheat, shows a principal effect due to absolute pressure and a secondary effect due to fluid velocity. At high q/A values the wall superheat becomes nearly independent of fluid velocity and decreases as pressure increases. 3. For non-boiling heat transfer the friction factors may be correlated by

$$\frac{f_{\text{(isothermal)}}}{f_{f_{gm}}} = (\mathcal{M}_{b}) \quad 0.14$$

$$\frac{f(\text{isothermal})}{f_{f}} = (H_{W})^{0.60}$$

4. With surface boiling the pressure drop increases with increasing heat flux and decreasing subcooling and decreasing pressure.

Ac'mowle gements

The work reported here was sponsored by the Office of Naval Research, Contract N5-ori-07827, NR-035-267. The authors gratefully acknowledge this aid and also the help of the following persons. The many problems associated with the design of the high pressure apparatus were for the most part solved by Mr. Milton W. Raymond, project design engineer, who also originally conceived and successfully built the ion-exchanger with very little design data. The maintenance and operation of the equipment and the gathering of the test data was accomplished by Messrs. William W. Barton, Edward H. Somma, Paul V. Osborn, Francis J. Ziumerman and Frank Mullin, graduate students in the Department of Mechanical Engineering, Massachusetts Institute of Technology.

Bibliography

- Kreith, F. and Summerfield, M., "Heat Transfer to Water at High Flux Densities With and Without Surface Boiling," <u>A.S.M.E. Trans</u>., Vol. 71, No. 7, p. 805, Oct., 1949.
- Kreith, F. and Summerfield, M., "Pressure Drop and Convective Heat Transfer with Surface Boiling at High Heat Flux; Data for Aniline and n-Butyl Alcohol," <u>A.S.M.E. Trans.</u>, Vol. 72, No. 6, p. 869, Aug., 1950.
- 3. Knowles, J. W., "Heat Transfer With Surface Boiling," <u>Canadian Journal of</u> <u>Research</u>, 26A, 268-278, 1948.
- 4. McAdams, W. H., Kennel, W. E., Minden, C. S., Rudolf, C., Picornell, P. M., Dew, J. E., "Heat Transfer at High Rates to Water With Surface Boiling," <u>Ind. and Eng. Chem.</u>, Vol. 41, p. 1945, Sept., 1949.
- 5. McAdams, W. H., Heat Transmission, McGraw-Hill, 1942.
- 6. Rohsenow, W. M., Somma, E. H., Osborn, P. V., "Construction and Operation of Apparatus for Study of Heat Transfer with Surface Boiling," Tech. Report No. 2, D.I.C. Project No. 6627, Mass. Inst. of Tech., July, 1950.
- 7. Clark, J. A. and Rohsenow, W. M., "Statement of Progress for Period 23 July 1950 to 1 September 1950," D.I.C. Project No. 6627, Mass. Inst. of Tech.
- 8. Wellman, E. J., "A Survey of the Thermodynamic and Physical Properties of Water," M. S. Thesis, Purdue University, 1950.
- 9. Rohsenow, W. M., and Clark, J. A., "Heat Transfer and Pressure Drop Data for High Heat Flux Densities to Water at High Subcritical Pressures," Heat Transfer and Fluid Mechanics Institute, Stanford, California, June, 1951.
- 10. Buchberg, H., Romie, F., "Heat Transfer, Pressure Drop, and Burnout Studies With and Without Surface Boiling For De-Aerated and Gassed Water at Elevated Pressures in a Forced Flow System," Heat Transfer and Fluid Mechanics Institute, Stanford, California, June, 1951.

Nomenclature

A	Heat transfer area, sq feet
°p	Specific heat, Btu/1b or
Di	Inner diameter of test section, feet
E	Test section voltage, volts
G	Mass velocity, 1b/hr ft ²
h	Surface coefficient of heat transfer, Btu/hr ft2op
I	Test section current, amperes
k	Thermal conductivity of fluid, Btu/hr ft of
ko	Thermal conductivity of tube wall at t, Btu/hr ft of
Lp	Test section heated length, ft
N _{Re,f}	$=\frac{D_{10}}{M_{e}}$
P	Pressure, psia
AP1	Pressure drop reading across test section, in. H O
APf.m	Pressure drop due to friction and momentum change, in. H O
AP	Pressure drop due to friction alone
q/A	Heat flux density, Btu/hr ft ²
(q/A)	$(n-1) = \frac{1}{2} \left[(q/A)_n + (q/A)_{(n-1)} \right]$
ſ	Defined by eq. (8)
f _{f,m}	Defined by eq. (9)
1	$= \frac{h}{C_{p}G} \left(\frac{\gamma p/r}{k}\right)^{2/3}$
ro	Outer radius of test section, feet
ri	Inner radius of test section, feet
R	Resistance, ohms
T _b	Test water bulk temperature, or
Υ _{in}	Test water inlet bulk temperature, or
Cout	Test water outlet bulk temperature, or
to	Test section outer wall temperature, oF

- 14 -

tw	Test section inner wall temperature, oF
tsat	Saturation temperature, or
T _x	Wall temperature minus saturation temperature, tt_, or
V	Flow velocity, feet/sec
W	Flow rate, 1b/hr
Δx	$= (r_o - r_i)$
X .	Distance between thermocouple stations, 1.4 inches
ß	Temperature coefficient of electrical resistivity, oF-1
ØK.	Temperature coefficient of thermal conductivity, or-1
8	Mean density of fluid in test section, 1b,/cu ft
V.	Density of liquid in lead lines to pressure gage, 1b,/cu ft
8	Electrical resistivity at t, ohm feet
Sm	Electrical resistivity at temperature $t_{m} = \frac{1}{A_{T}} \int (t-t_{m}) dr$
9	Mass density of fluid at fluid temperature, 1b/cu ft
Mg	Viscosity at film temperature, $\frac{T_{3}+t_{w}}{2}$, lb/hr ft

Figure Captions

- 1. Layout of Test Apparatus
- 2. Test Section Assembly
- 3. Location of Thermocouples and Potential Taps
- 4. Test Section Voltage Measurement
- 5. Heat Flux Density and Bulk Temperature Along the Tube, Non-Boiling Run
- 6. j-Factor vs. Reynolds Number for Forced Convection Without Surface Boiling
- 7. Heat Flux vs. Temperature Difference for Forced Convection with Surface Boiling of Water at 2000 psia and 20 ft/sec Inlet Velocity
- 8. Heat Flux vs. Wall Superheat for Forced Convection with Surface Boiling
- 9. Friction Factor vs. Reynolds Number for Forced Convection without Surface Boiling
- 10. Correction Factor for Friction Factor with Heat Transfer Without Surface Boiling
- 11. Effect of Heat Flux on APf.m
- 12. Effect of Heat Flux on A P.
- 13. Heat Flux vs. Temperature Difference for Forced Convection with Surface Boiling of Water at 2000 psia and 30 ft/sec Inlet Velocity
- 14. Heat Flux vs. Temperature Difference for Forced Convection with Surface Boiling of Water at 2000 psis and 10 ft/sec Inlet Velocity
- 15. Heat Flux vs. Temperature Difference for Forced Convection with Surface Boiling of Water at 1500 psia and 30 ft/sec inlet Velocity
- 16. Heat Flux vs. Temperature Difference for Forced Convection with Surface Boiling of Water at 1500 psia and 20 ft/sec Inlet Velocity
- 17. Heat Flux vs. Temperature Difference for Forced Convection with Surface Boiling of Water at 1500 psia and 10 ft/sec Inlet Velocity

FIGURE I







LOCATION OF THERMOCOUPLES AND POTENTIAL TAPS

FIGURE 3

TEST SECTION VOLTAGE MEASUREMENT













FIGURE 9



FIGURE 10









∆t = tw - tb



 $\Delta t = t_w - t_b$





At = tw - to

Table I Heat Transfer Data Without Surface Boiling

Run No.	Station	Pin	Vin	G	tin	tout	Δp	Nub	Ref	5 Mg	1
		psia	f.p.s.	Bec/ft ²	oF	op	aH O		x10 ⁻³	R	× 103
19-1	2 3 4 5 6	2000	29	15770	410	437	152	503 477 482 479 471	292 300 302 303 307	.82 .77 .82 .82 .80	1.84 1.7. 1.7 1.6 1.6
20-1	2 3 4 5 6	2000	29.8	1585	400	436		468 475 458 461 462	299 305 307 309 311	•85 •785 •78 •79 •795	1.6 1.6 1.6 1.6
20-2	2 3 4 5 6	2000	29.9	1585	406	454		448 442 444 444 445	314 323 329 332 339	•765 •76 •755 •745 •73	1.5 1.5 1.4 1.4 1.4
21-1	23456	2000	29.8	1561	427	482	140	443 443 442 447 446	329 341 348 353 359	•75 •73 •73 •715 •715	1.4 1.4 1.4 1.4 1.3
22-1	2 3 4 5 6	2000	30.0	1558	446	498	150	481 478 472 483 478	343 350 353 359 373	•73 •715 •715 •725 •72	1.5 1.5 1.5 1.5 1.4
22=2	2 3 4 5	2000	30 .0	1548	453	520	155	495	360	.695	1.5
23-1	6 N metado	2000	29.9	1585	383	388	193	441 434 402 397	254 254 355 256 256	•945 •94 •94 •935	1.7 1.7 1.6

Table I Heat Transfer Data Without Surface Boiling (cont.)

						2.					
Run No.	Station	Pin	Vin	G lb	tin	tout	Vb	Nub	Ref	Colut	J_3
		psia	f.p.s.	sec/ft ²	oŗ	or	"H 0		x 10"3		XIO
23-2	2345	2000	30.0	1562	442	502	130	498 502 506	346 356 362	.735 .71 .71	1.60
1	6								327 ,		11
23-3	2	2000	30.0	1970	444	504	130	503	349	.725	1.61
	4 5			¥.				493 477	356 368	.72 .71	1.55 1.45
	6								19 AL		
33-11	2 3 4	2000	30.0	1618	386	470		453 446	345 361	.705 .685	1.48
	6										
23-4	2 3 4 5	2000	19.9	1051	407	462	64.5	338 335 330 327	210 217 221 228	.785 .75 .74	1.74 1.70 1.65
	6							324	233	.73	1.55
23-5	2 3 4 5 6	2000	20.0	1049	426	491	63.5	347 341 332 349	226 233 238 243	.735 .715 .715 .705	1.70 1.64 1.56 1.61
		2000	21		5.)					200	
23=0	2 3 4 5	2000	20.0	1043	438	514	67	350	241	.705	1.63
	6								and the state		
30-1	2345	2000	20.1	1099	373	414	53.6	313 326 315	197 198 202	.835 .825 .815	1.69 1.75 1.67
	6					1		313	210	.790	1.64

î		Tat	ole I Heat Tran	asfer Data	Without	Surface Ba	dling .				
Run No.	Station	Pin	Vin	3 G 1b_	tin	tout	∆ P	Nub	Ref	Colle	4
		psia	f.p.s.	sec/ft ²	oF	oF	"H O	۲	x 10 ⁻³	F	Vino
30=2	2 3 4 5 6	2000	20.0	1083	409	467	52.9	315 332 321 321 317	225 226 236 241	•740 •745 •72 •715	1.55 1.55 1.52 1.49
31-1	2	2000	20.1	1132	227	200	22.9	211	244	.71	1.46
31_0	3 4 5 6	2005	10 595		521	308	57 •2	311 322 308 308 308	170 173 178 180 181	•96 •935 •916 •90 •89	1.86 1.91 1.78 1.77
2206	2	2000	20.0	1111	350	391	55	311	100	,	- 014
21_2	456	árana.	10.0 545					325 308 315 309	189 195 199 202	.855 .865 .835 .825 .820	1.75 1.30 1.63 1.69
<i>C-2C</i>	2 3 4 5 6	2000	20.1	1100	382	429	54.3	324 334 321 323 319	203 209 212 217	.805 .785 .785 .765	1.72 1.73 1.64 1.63
32-4	2	2000	20.4	1083	108	105	FO O	327	222	.750	1,53
31_5	3 4 5 6	1900	30.31 1723	50		400	72.2	327 338 325 328 328	221 226 233 239 244	.745 .745 .725 .710 .720	1.63 1.65 1.55 1.54 1.50
	34	2000	20.3	1063	431	504	53	354 347	239 246	.710	1.66
	56								234	0120	1631
	1. 1. 1. 1. 1. 1. 1. 1. 1. 1. 1. 1. 1. 1			- 10							
, ,	2 3 4	2000	20.2	1130	340	454	62	331	236	.70	- 1.58
	56			-							

				Table I Hea	t Transfer	Lata With	out Surface	Boiling (ccht.)	6	-
Run No.	Station	Pin	Vin	G	tin	tout	ÞP	Nu_o	Fej.	but	· il-
		psia	f.p.s.	m sec/ft ²	oF	oF	"H O		x 10 ⁻³	k	XIO .
34-2	2 3 4 5 6	2000	10	590	223	252	25	163 164 157 160 161	59.9 61.3 63.8 64.7 66.1	1.38 1.35 1.31 1.28 1.25	2.22 2.22 2.25 2.25 2.25 2.25
34-3	2 3 4 5 6	2000	10	544	371	470	22.9	168 183 198	113 118 122	.715 .690 .685	1.60 1.81 1.84
34-4	2 3	2000	10	560	389	512	22.9				
	4 5 6							275	124	.695	2.60
35-1	2 3 4 5 6	2000	10.0	548	355	370		173 169 159 155 157	84.1 85.3 86.5 86.8 87.1	•945 •940 •940 •930 •935	2.10 2.01 1.88 1.83 1.85
35-2	2 3 4 5 6	2000	10.0	552	357	405		172 177 166 169 172	94.8 97.9 100 102 104	.855 .830 .830 .820 .795	1.92 1.92 1.77 1.77 1.79
32-1	2 3 4 5 6	1500	30.0	1727	305	318	117	404 422 403 406 397	227 228 230 232 234	1.06 1.06 1.04 1.04 1.04	1.72 1.78 1.78 1.77
32-2	2 3 4 5 6	⁻ 1500	30.0	1711	314	333	114	439 432 413 409 412	238 242 244 245 280	1.01 .995 .990 .990 .990	1.04 1.76 1.70 1.67 1.68

			Table I Heat	t Transfer Da	te Withou	ut Suific	e Dulatin	e Verment			
Run No.	Station	Pin	Vin	G 1b	tin	• t _{out}	0			Cpuc	i
		psia	foposo	Bec/ft2	QŢ	oF	™H_C		× 10 ⁻³	Fr	XIEB
32-3	23	1500	30.0	1665	348	373	110	. 442 433	262 • 264	.910	• 1.47.4+
	4 5 6		2 * *5					414 413 405	* 268 272 276	910 .895 .875	
32-4	2	1500	30.0	1641	379	412	106	455	593	1.850	ala.
	4 5 6							446 426 428 421	295 299 303 312	.825 .820 .810 .795	4-1 1455 1455 1455 + 845
32-5	2 3 4	1500	30.2	1570	423	467	202	439 461 442	321 327 335	.765 .745	1 44
	6							298 298	341 349	.730 .715	
- 332	2	1500	20.6	1161	323	357	56.2	323	174	.960	2.5
	4 5 6	,						907 305 302	182 185 188	940 910 895	1.77
33-3	2	1500	20.2	1090	376	427	52.2	33/	201	dan	4.50
	3 4 5					007 - 22		330 319 320	206 211	.795 .770	1.073 1.65
	6							316	222	.760	1,62
35-3	2345	1500	10.0	545	355	404	16.4	177 178 169	94 96.5 52.8	.845 .835 .825	1.97
	6							103	103	.790	1.82
35-4	2 3	1500	10.0	549	357	439	16.4	162	105	.765	1.39
	4 5 6							175	114	.730	- 2 - 62 - 2 - 10

TABLE II

HEAT TRANSFER DATA WITH SURFACE BOILING

Run No.	Station	G lb lb/sec.ft ²	p psia	Vin ft/sec.	tw 9F	T °F	Tx °F	At subcool	q [/] A Btu [/] hr.ft ²	Tx avg.
21-2	2345	1585	2000	30.1	641.5 642.5 643.5 641.5	445 457 469 481	5.7 6.7 7.7 5.7	191 179 167 155	2.4×10^{6} 2.4×10^{6} 2.41×10^{6} 2.40×10^{6} 2.40×10^{6}	6.5
21-3	2	1561	2000	29.9	641	493	5.2	143 .	2.40 x 10 2.57 x 106	6.6
	3456		2000		642 643.5 642 643.5	461 473.5 486 499	6.2 7.7 6.2 7.7	175 152.5 150 137	$2.58 \times 10^{\circ}$ $2.59 \times 10^{\circ}$ $2.58 \times 10^{\circ}$ $2.59 \times 10^{\circ}$ $2.59 \times 10^{\circ}$	
22-2	3 4 5 6	1548	2000	30.0	640 642 641 642	476.5 486.5 496.5 506.5	4.2 6.2 5.2 6.2	159.5 149.5 139.5 129.5	2.04×10^{6} 2.055×10^{6} 2.054×10^{6} 2.055×10^{6}	5.45
22-3	2 3 4 5	1553	2000	30.0	642 643 645 644	468.5 480 492 504	6.2 7.2 9.2 8.2	167.5 156 144 132	$2.44 \times 10_6^6$ $2.44 \times 10_6^2$ $2.44 \times 10_6^2$ $2.44 \times 10_6^2$	7.8
22-4	6	1553	2000	30.0	644	516	8.2	120	$2.44 \times 10^{\circ}$ $2.89 \times 10^{\circ}$	9.8
23-5 30-5	3 4 5 6	1049	2000 2000	20.0	645 647 646 646	487 501 515 529	9.2 11.2 10.2 10.2	149 135 121 107	2.90×106 2.90×106 2.90×106 2.90×106 2.90×106	
22-5	23456	1553	2000	30.0	646 646 648 646 647	471.5 485.5 499.5 513.5 527.5	10.2 10.2 12.2 10.2 11.2	164.5 150.5 136.5 122.5 108.5	2.895×10^{6} 2.895×10^{6} 2.91×10^{6} 2.895×10^{6} 2.895×10^{6} 2.91×10^{6}	10.8
22-6	23456	1548	2000	30.0	647 647 649 647 646	473 489 505 520 534	11.2 11.2 13.2 11.2 10.2	163 147 - 131 116 102	$3.15 \times 10_{6}^{6}$ $3.15 \times 10_{6}^{6}$ $3.16 \times 10_{6}^{6}$ $3.15 \times 10_{6}^{6}$ $3.15 \times 10_{6}^{6}$	11.4
22-7	23456	1548	2000	30.0	646.5 646.5 645.5 645.5 645.5	471 486.5 501.5 516.5 531.5	10.7 10.7 10.7 9.7 8.7	165 149.5 134.5 119.5 104.5	$3.12 \times 10_{6}^{6}$ 3.11×10^{6}	10.1

10.00	and the	-	-	10-
80.	ыь.	11		18.
6.62	242.3.1	And .	100	45

14

HEAT TRANSFER DATA WITH SURFACE BOILING

Run No.	Station	G lb lb/sec.ft ²	p psia	Vin ft/sec.	tæ of	T	Tx °F	At subcool	g/A Btu'hr.ft ²	Tx avg.
23-2	56	1562	2000	30.0	640.5 640.5	481 489.5	4.7 4.7	155 146.5	1.86 x 10 ⁶ 1.86 x 10 ⁶	4.7
23-3	56	1570	2000	30.0	641 639	483 491.5	5.2 3.2	153 144.5	1.86 x 10 ⁶ 1.86 x 10 ⁶	4.2
33-11	4 56	1618	2000	30.0	643.5 643.5 642.5	428 440.5 453	7.7 7.7 6.7	208 195.5 183	2.53×10^{6} 2.53 x 10^{6} 2.53 x 10^{6}	7.3
33-12	23456	1610	2000	30.0	642 643.5 644 643 641.5	416.5 430 443.5 457 470.5	6.2 7.7 8.2 7.2 5.7	219.5 206 192.5 179 165.5	2.75×10^{6} 2.76 x 10^{6} 2.76 x 10^{6} 2.76 x 10^{6} 2.76 x 10^{6} 2.75 x 10^{6}	7.0
33-13	23456	1602	2000	29.7	644.5 644.5 644.5 643.5 642.5	423 438 453 468 483	8.7 8.7 8.7 7.7 6.7	213 - 198 183 168 153	3.00×10^{6} 3.00×10^{6} 3.00×10^{6} 3.00×10^{6} 2.99×10^{6}	8.1
23-6	3456.	1043	2000	20.0	639 642 643 642	465 476 487 499	3.2 6.2 7.2 6.2	171 160 149 137	1.57×10^{6} 1.58×10^{6} 1.58×10^{6} 1.58×10^{6} 1.58×10^{6}	5.7 6.5
23-5	6	1049	2000	20.0	639.5	478	3.7	158	1.37 x 10 ⁶	3.7
30=3	23456	1051	2000	20.1	642.5 643.5 644.5 644.5 643.5	454 467 479 492 505	6.7 7.7 8.7 8.7 7.7	182 169 157 144 131	1.78×10^{6} 1.78×10^{6} 1.79×10^{6} 1.79×10^{6} 1.79×10^{6} 1.78×10^{6}	7.9
30-4	2 3 4 5	1052	2000	20.1	643 643.5 645 645 644	460 475 489 504 519	7.2 7.7 9.2 9.2 8.2	176 161 147 132 117	$\begin{array}{c} 2.03 \times 10^6 \\ 2.03 \times 10^6 \end{array}$	8.3
30=5	23456	1059	2000	20.1	644.5 645 646.5 646 645	450 466 482 498 514	8.7 9.2 10.7 10.2 9.2	186 170 154 138 122	$\begin{array}{c} 2.21 \times 10^6 \\ 2.20 \times 10^6 \\ 2.23 \times 10^6 \\ 2.23 \times 10^6 \\ 2.23 \times 10^6 \\ 2.23 \times 10^6 \end{array}$	9.6

-3-

HEAT TRANSFER DATA WITH SURFACE BOILING

Run No.	Station	G 1b 1b/sec.ft ²	p psia	Vin ft/sec.	tw op	Top	Tx ØF	At subcool	q/A Btu/hr.ft ²	Tx avg.
30-6	23456	1059	2000	20.1	643 644 644 644 643	445 459 474 488 503	7.2 8.2 8.2 8.2 7.2	191 177 162 148 133	2.02×106 2.02×106 2.03×106 2.03×106 2.03×106 2.03×106	7.8
31-5	4 5 6	1063	2000	20.3	643 643 643	468 479 489	7.2 7.2 7.2	168 157 147	1.54×106 1.54×106 1.54×106 1.54×10^{6}	7.2
31-6	2 3 4 5	1060	2000	20.3	642.5 643.5 644.5 644.5	452 464 477 489	6.7 7.7 8.7 8.7	184 172 159 147	1.74×10^{6} 1.74×10^{6} 1.74×10^{6} 1.74×10^{6} 1.74×10^{6}	7.9
31-7	6 2 3 4	1057	2000	20.1	643.5 644.5 644.5	502 440 454 468	7.7 7.7 8.7 8.7	134 196 182 168	1.74×10^{-1} 1.91×10^{-1} 1.92×10^{-1} 1.92×10^{-1}	8.3
	56				644.5	482 495	8°7 7°7	154 141	$1.92 \times 10^{\circ}$ $1.91 \times 10^{\circ}$	
31-8	2 3 4 5 6	1057	2000	20.1	644.5 645.5 645.5 644.5 643.5	204.5 190.5 175.5 159.5 143.5	8.7 9.7 9.7 8.7 7.7	196 181 166 151 136	2.09×10^{6} 2.11×10^{6} 2.11×10^{6} 2.09×10^{6} 2.07×10^{6}	8.9
31-9	3 4 5 6	1130	2000	20.2	645 647 646 645	380 397 413 430	9.2 11.2 10.2 9.2	256 239 213 206	$\begin{array}{r} 2.34 \times 106 \\ 2.34 \times 106 \\ 2.34 \times 106 \\ 2.34 \times 106 \\ 2.34 \times 106 \end{array}$	9.95
31-10	2 3 4 5 6	1118	2000	20.1	643.5 645.5 646.5 644.5 643.5	365 384 403 421 439	7.7 9.7 10.7 8.7 7.7	271 252 234 216 197	$2.53 \times 10^{6} \\ 2.54 \times 10^{6} \\ 2.54 \times 10^{6} \\ 2.54 \times 10^{6} \\ 2.54 \times 10^{6} \\ 2.53 \times 10^{6} \\ $	8.9
34-3	56	544	2000	10.0	639.5 641.5	434 450	3.7	200 185	1.08×10^{6} 1.08×10^{6}	4.7
34-4	2356	560	2000	10.0	642 638 641 641.5	413.5 431.5 467.5 485.5	6.2 2.2 5.2 5.7	222.5 204.5 168.5 150.5	1.29×10^{6} 1.28×10^{6} 1.29×10^{6} 1.29×10^{6} 1.29×10^{6}	(4.8) (5.4)

TABLE II

HEAT TRANSFER DATA WITH SUPFACE BOILING

-

Run No.	Station	C lb lb/sec.ft ²	p psia	Vin ft/sec.	tu op	T	Tx of	At subcool	q'A Btu'hroft ²	Tx avg.
34-5	2 3 4 5 6	541	2000	10.0	641.5 643 644.5 643.5 642	415 436 457 478 499	5.7 7.2 8.7 7.7 6.2	221 200 179 158 137	1.46×10^{6} 1.47 x 10^{6} 1.47 x 10^{6} 1.47 x 10^{6} 1.46 x 10^{6}	7.1
34-6	23456	545	2000	10.0	642.5 643.5 645 643.5 642	407 430 453 476 499	6.7 7.7 9.2 7.7 6.2	229 206 183 160 137	1.61×10^{6} 1.62×10^{6} 1.62×10^{6} 1.62×10^{6} 1.61×10^{6}	7.5
34-7	23456	550	2000	9.8	643 644 645 643.5 642.5	346 374 402 428 454	7.2 8.2 9.2 7.7 6.7	290 262 234 212 182	1.82×10^{6} 1.82×10^{6} 1.82×10^{6} 1.82×10^{6} 1.82×10^{6} 1.81×10^{6}	7.8
32-6	23456	1552	1500	29.9	604 607 610 609 608.5	451.5 460.5 469.5 478.5 487.5	7.8 10.8 13.8 12.8 12.3	144.5 135.5 127.5 118.5 108.5	1.82×10^{6} 1.84×10^{6} 1.85×10^{6} 1.84×10^{6} 1.84×10^{6} 1.84×10^{6}	11.5 (12.4
32-7	2 3 4 5 6	1552	1500	29.9	608.5 610 611 609.5 609	445.5 456 466.5 477 487.5	12.8 13.8 14.8 13.3 12.8	150.5 140 129.5 119 108.5	2.09×10^{6} 2.10 x 10^{6} 2.11 x 10^{6} 2.10 x 10^{6} 2.10 x 10^{6} 2.10 x 10^{6}	13.5
32-8	2 3 4 5 6	1618	1500	30.0	600.5 607 609.5 609 608.5	406 417.5 429 440.5 452	4.3 10.8 13.3 12.8 12.3	190 178.5 167 155.5 144	2.25×10^{6} 2.28×10^{6} 2.29×10^{6} 2.29×10^{6} 2.29×10^{6} 2.29×10^{6}	10.7 (12.3)
32-9	2 3 4 5 6	1610	1500	29.8	607 609.5 610 609 608.5	413.5 425.5 437.5 449.5 461.5	10.8 13.3 13.8 12.8 12.3	182.5 170.5 159.5 146.5 134.5	2.40×10^{6} 2.42×10^{6} 2.42×10^{6} 2.42×10^{6} 2.42×10^{6} 2.41×10^{6}	12.6
32-10	23456	1610	1500	29.8	610 610.5 610.5 610.5 610	414.5 427.5 440.5 453.5 466.5	13.8 14.3 14.3 14.3 14.3 13.8	181.5 169.5 155.5 142.5 129.5	2.58×10^{6} 2.59×10^{6} 2.59×10^{6} 2.59×10^{6} 2.59×10^{6} 2.58×10^{6}	14.1

TABLE II

HEAT TRANSFER DATA WITH SURFACE BOILING

-5-

Run No.	Station	G 1b 1b/sec.ft ²	p psia	Vin ft/sec.	tw op	T	Tx of	At subcool	q/A Btu hr.ft ²	Tx avg,
32-11	23456	1593	1500	29.7	611 611 610.5 609.5	414.5 428 442 456 469.5	14.8 14.8 14.8 14.3 13.3	181.5 168 154 140 126.5	2.75×10^{6} 2.75 x 10^{6} 2.75 x 10^{6} 2.75 x 10^{6} 2.75 x 10^{6} 2.74 x 10^{6}	14.5
33-4	2 3 4 5 6	1051	1500	20.1	606 608.5 610 609.5 609	444 454 464 474 484	9.8 12.3 13.8 13.3 12.8	152 142 132 122 112	1.40×10^{6} 1.41×10^{6} 1.42×10^{6} 1.42×10^{6} 1.42×10^{6} 1.42×10^{6}	12.4
33-5	2 3 4 5 6	1059	1500	20.1	609.5 610.5 610.5 610 609.5	441 452.5 464 475.5 487	13.3 14.3 14.3 13.8 13.3	155 143.5 132 120.5 109	1.62×10^{6} 1.62×10^{6} 1.62×10^{6} 1.62×10^{6} 1.62×10^{6} 1.62×10^{6}	13.8
33-6	2 3 4 5 6	1051	1500	20.0	610.5 611.5 611 610.5 609.5	443.5 457 470 483.5 496.5	14.3 15.3 14.8 14.3 13.3	152.5 139 126 112.5 99.5	1.81×10^{6} 1.82×10^{6} 1.81×10^{6} 1.81×10^{6} 1.81×10^{6} 1.81×10^{6}	14.4
33=7	23456	1051	1500	20.0	611.5 612 611.5 610.5 609.5	441 455 469 483 497	15.3 15.8 15.3 14.3 13.3	155 141 127 113 99	1.91×106 1.91×106 1.91×106 1.90×106 1.90×106 1.90×10^{6}	15.2
33-8	2 3 4 5 6	1050	1500	20.0	613 612.5 611.5 610.5 609.5	445 461 476 491 506	16.8 16.3 15.3 14.3 13.3	151 135 120 105 90	2.06×106 2.06×106 2.05×106 2.05×106 2.05×106 2.04×10^{6}	15.2
33-9	23456	1051	1500	20.1	612 612 611 610.5 609.5	449 465.5 482 498.5 515	15.8 15.8 14.8 14.3 13.3 .	147 130.5 114 97.5 81	2.32×10^{6} 2.31×10^{6} 2.31×10^{6} 2.31×10^{6} 2.31×10^{6} 2.30×10^{6}	14.8
33-10	23456	1057	1.500	20.1	611.5 611.5 610.5 610.0 609.5	437 455 473 491 509	15.3 15.3 14.3 13.8 13.3	159 141 123 105 97	2.49×10^{6} 2.49×10^{6} 2.48×10^{6} 2.48×10^{6} 2.48×10^{6} 2.48×10^{6}	24.04

	1400	100.44	1000	1000	
eg a	- A -	1.17	C 227	- 100	æ
- 80	м.	511	6 C	- 10.1	ъ

Run No.	Station	G lb lb/sec.ft ²	p psia	Vin ft/sec.	tw op	T	Tx of	At subcool	q/A Btu/hr.ft ²	Tx avg
35-4	56	549	1500	10.02	603.5 606.5	410 422	7.3 10.3	186 174	.896 x 10 ⁶ .896 x 10 ⁶	8.
35-5	23456	543	1500	10.02	605 606 608 607.5 607.5	398.5 415 431.5 448 464.5	8.8 9.8 11.8 11.3 11.3	197.5 181 164.5 148 131.5	1.12×10^{6} 1.13×10^{6} 1.14×10^{6} 1.14×10^{6} 1.14×10^{6} 1.14×10^{6}	10.
35-6	2 3 4 5 6	541	1500	10.03	604 604 605 604.5 604	422.5 442 461.5 481 500.5	7.8 7.8 8.8 8.3 7.8	173.5 154 134.5 115 95.5	1.39×10^{6} 1.39×10^{6} 1.39×10^{6} 1.39×10^{6} 1.39×10^{6} 1.39×10^{6}	8
35-7	23456	541	1500	10.01	604.5 604.5 604.5 604.5 603	415 437 459 481 503	8.3 8.3 8.3 8.3 6.8	181 159 137 115 93	1.56×10^{6} 1.56×10^{6} 1.56×10^{6} 1.56×10^{6} 1.55×10^{6}	8.(
35-8	2 3 4 5 6	559	1500	9.9	606.5 606.5 605.5 605.5 604.5	344 370 396 422 448	10.3 10.3 10.3 9.3 8.3	252 226 200 174 148	1.81×10^{6} 1.81×10^{6} 1.81×10^{6} 1.80×10^{6} 1.80×10^{6}	9.,
35-9	23456	556	1500	9.9	607 606.5 606.5 604.5 603.5	355 382 409 436 463	10.8 10.3 10.3 8.3 7.3	241 234 187 160 133	1.85×10^{6} 1.84×10^{6} 1.84×10^{6} 1.84×10^{6} 1.84×10^{6} 1.83×10^{6}	9.1

HEAT TRANSFER DATA WITH SURFACE BOILING

~6~

Table III PRESSURE DROP WITHOUT SURFACE BOILING

.

Run No.	G	Vin	Tw, avg	Tin	Tout	Ma / Mu	& Pf,m	f	ſ,m	fisotherm	N _{Ra} , t
	1b.m sec.?t ³	ft. sec.	op	of	oF	at Tavg	"H 0 2	x10 ⁵	x 10 ⁵	x 10 ⁵	attor
					Pressure	2000 psia					C. V (
30-1	1098	20.1	492	373	414	1.275	54.8	360	388	405	172 500
2	1081	20.0	583	409	467	1.34	54.4	345	392	205	100,000
31-1	1132	20.1	420	327	358	1.25	58.1	383	401	115	170,000
-2	1114	20.0	473	350	391	1.31	56.0	365	302	44.7	157,000
-3	1098	20.1	522	382	429	1.305	55.6	350	282	410	109,400
-la	1081	20.4	581	408	466	1.35	53.7	328	272	204	185,500
32-1	1728	30.0	349	305	<u>Pressure 1</u> 33.8	1.13	117.3	355	360	200	197,000
-2	1712	30.0	378	314	333	1.17	115.0	3/3	251	200	217,200
-3	1665	30.0	42.8	348	373	1,185	111 0	220	274	385	226,000
=4	1640	30.0	486	379	412	1.26	107 0	330	340	378	248,500
-5	1570	30.2	566	122	2682	1.20	107.02	320	342	372	268,500
27-2	2260	00 (423	201	1.31	103.1	295	333	365	293,000
3.1-a	110%	20.0	426	323	357	1.29	57.4	355	375	412	161,200
-3	1090	20.2	527	376	427	1.33	53.5	338	374	400	184,000
					5 F F F						

Table IV PRESSURE DROP WITH SURFACE BOILING

ion No.	p	Vin	G	q/Ax10 ⁻⁰	tw,avg	T_{in}	Tout	t sat avg	APfm	. APr
	psia.	ft/sec	lb sec.ft ²	Btu/hr.ft ^a	F	F	F	F	**H 0	"H O
33-11 12 13	2000	30.0 30.0 29.7	1618 1610 1602	2.53 2.76 3.00	643 643 644	386 398 403	470 491 503	208 191 183	111.1 115.8 123.1	90.1 90.3 94.8
30- 3 4 56 31- 5 6 7 8 9 10	2000	20.1 20.1 20.1 20.1 20.3 20.3 20.3 20.1 20.1 20.2 20.1	1051 1052 1059 1059 1063 1060 1057 1057 1130 1118	1.78 2.03 2.23 2.20 1.54 1.74 1.92 2.10 2.34 2.54	644 645 644 643 644 644 644 645 645	436 440 428 424 431 435 421 420 340 340	523 539 537 523 504 519 514 520 454 464	156 146 153 162 168 159 168 166 239 234	57.0 63.8 65.6 65.7 53.5 57.7 62.6 62.7 59.9	44.7 49 5 50.3 51.9 43.7 46.1 50.3 49.5 47.8
3%= 5 6 7	2000	10.0 10.0 9.8	541 545 550	1.47 1.62 1.82	643 643 644	387 376 309	527 531 487	179 182 242	26,8 29.6	21.9
32- 6 7 8 9 10 11	1500	29.9 29.9 30.0 29.8 29.8 29.8 29.8	1552 1552 1618 1610 1610 1593	1.84 2.10 2.29 2.42 2.59 2.75	608 609 608 609 610 611	439 431 390 397 397 396	500 501 468 478 484 489	107 130 167 158 156 154	107.5 118.1 115.4 117.3 123.0 126.0	25.3 89.5 97.5 95.1 96.1 99.5
5 6 7 8 9 10	1500	20.1 20.1 20.0 20.0 20.0 20.1 20.1	1051 1059 1051 1051 1050 1051 1057	1.42 1.62 1.81 1.91 2.05 2.31 2.59	609 610 611 611 611 611	430 425 426 422 426 427 413	499 504 515 517 527 539 534	132 131 125 126 119 113 122	55.6 61.2 67.7 72.0 74.3 81.8 84.6	46.4 52.6 55.2 59.2 60.3 54.0
35-56789.	1500	10.0 10.0 10.0 9.9 9.9	543 541 559 556	1.14 1.39 1.56 1.81 1.84	607 604 604 606 606	376 396 385 309 319	487 528 534 485 499	164 134 136 199 187	22.6 26.7 27.7 27.9 29.0	19.3 22.2 22.5 21.9 24.0

ONP:438:3MC:ba NR-035-267

		10-7403 Mar 124, 16	E
1.	1 2 3 3 3 3 4 3 4 3	2-1-2 / 20.	1 2 6 3 52
1.5 4 5 5 4	1.1.2.1.2.2	2-2-2-2-2-22	1 1 1 1 1 1 1 1
ALC: NO. 341-14	n all contracts of	a see the state	 And the Dark the

	DISTRIE
Chief of Naval Research	
Department of the Nevy	
Washington 25. D. C.	
Atta: Code 438	(2)
Director, Naval Research Labora	tory
Attas Tash Infa Officer	101
Manhadaal Idhaama	127
Machanian Distador	
Hechanica Dialaton	(2)
Commanding Officer	
Office of Neval Research	
Branch Office	
495 Summer Street	
Boston 10, Mass.	(2)
The Englands Passers Courission	
Commanding Officer	
Office of Naval Research	. Criv
Brench Office	
346 Broadway	
New York 13, N. Y.	(1)
Contraction and a second	
Countending Ullicer	· [25]
Ollice of Navel Hesearch	
branch Office	
344 N. Rush Street	
Chicego 11, Illinois	(1)
Commanding Officer	
Office of Nexal Descent	
Brouch Office	
201 Davahua Câmaat	
Son Francisco 21 de246	1.2.2
Den Francisco 249 Gallio	(1)
Commandana OPPLaca	
Office of Nem 1 Decemb	
Guilde of Nevel Research	
1020 Company Changed	
1000 Green Street	1
resadena, Callo	(1)
Contract Martin and and	
Office of New Pressors SE Aree	
Ullice of Naval Research	
Department of the Nevy	
wasnington 25, D. C.	1.1.1
Acta: Mr. R. F. Lynch	(1)
Assistant Neval Attache for Resea	rch
London	
U. S. Nevy	
FPO 100	
New York, N. Y.	(5)
	S. 4 19

www.mapan	100
ON LIST April 27, 195	0
Library of Congress	
washington 25, D. C.	12-12-2
Attn: Nevy Research Section	(2)
Research and Development Board	
Department of Defense	
Pentegon Building	
Washington 25, D. C.	
Attn: Library (Code 3D-1075)	(1)
Chief of Burery of Lamoncution	
Nevy Depertment	
Washington 25. D. C	
Attn: TD-11 Tashniac 7 Tihmour	123
PP_22	11
4.5 mm	123
Chief of Buresu of Ordnance	
Navy Department	
Washington 25. D. C.	
Attn: Ad-3, Technical Library	(2)
ktomber Marker Distriction	
Chief of Bureau of Ships	
Navy Department	
Washington 25, D. C.	
Atta: Director of Research	(2)
Code 390	(3)
Code 646	(2)
Code 432	(1)
Superintendent	
Post Graduate School	
U. S. Naval Academy	
Annapolis, Maryland	(1)
Nam 7 Ordnaman Y. Louis	
Nevel Uranence Laboratory	
white Uak, Maryland	
AFD 1, Sliver Spring, Maryland	1-1
Atus Mechanics Division	(1)
Naval Ordnance Test Station	
Inyokern, California	
Atta: Scientific Officer	(1)
New Haven, Cent.	
Director,	
Nevel Engineering Experiment Station	
An abolis, Maryland	1-1
	(2)
Child C Charles	
Uniel of Stall	
nebeleneor of the stady	
Varbinaten OK D A	
Atta Dimates of Description	
RUMAN DATECNOT OI NESSERTON	

and Development

(1)

Distribution List (cont.)

Office of Chief of Ordnance Research and Development Service Department of the Army The Pentagon Washington 25, D. C. Attn: ORDTB

(1)

(1)

Commanding General U. S. Air Forces The Pentagon Washington 25, D. C. Attn: Research & Development Div. (1)

Office of Air Research Wright-Patterson Air Force Base Dayton, Ohio Attn: Chief, Applied Mechanics Group (1)

U. S. Atomic Energy Commission Division of Research Washington, D. C. (1)

Argonne National Laboratory P. O. Box 5207 Chicago 80, Illinois Attn: W. H. Jens (25)

U. S. Coast Guard 1300 E Street, N. W. Washington, D. C. Attn: Chief, Testing and Development Division (1)

National Advisory Committee for Aeronautics Cleveland Municipal Airport Cleveland, Ohio Attn: J. H. Collins, Jr. (2)

Dr. William Sibbitt Purdue Research Foundation Purdue University Lafayette, Indiana (1)

Professor E. D. Kane University of California Berkeley, California (1)

Dr. J. V. Foa Cornell Aeronautical Laboratory Euffalo, New York

Professor John W. Hazen Engineering Research Division University of California Los Angeles 24, California Professor W. H. McAdams Department of Chemical Engineering Massachusetts Institute of Technology Cambridge 39, Massachusetts (1)

Professor A. L. London Department of Mechanical Engineering Stanford University Stanford, California (1)

Professor Warren Rohsenow Mechanical Engineering Department Massachusetts Institute of Technology Cambridge 39, Massachusetts (15)

Lt. Cdr. E. P. Wilkinson c/o U. S. Atomic Energy Commission P. O. Box 1105 Pittsburgh 30, Pa.

(2)

(1)

(1)

(1)

(1)

111

Mr. R. A. Bowman Westinghouse Electric Co. Atomic Power Division P. O. Box 1468 Pittsburgh, Pa.

Deen L. M. K. Boelter Dept. of Engineering University of California Los Angeles, California

Professor J. H. Keenan Dept. of Mechanical Engr. Mass. Inst. of Technology Cambridge 39, Massachusetts

Professor J. N. Addoma Dept. of Chemical Engr. Mass. Inst. of Technology Cambridge 39, Massachusetts

Dean W. J. Wohlenberg School of Engineering Yale University New Haven, Conn.

Mr. E. E. Ross Room 20-E-226 Mass. Inst. of Technology Cambridge 39, Mass. Distribution Also to Include the Following

Mr. Robert Bromberg Engineering Research Department of Engineering	3. 1	Mr. Gilbert L. Campbe Library Los Alamos Scientific
University of California		P. O. Box 1663
Los Angeles 24, California	(1)	Los Alamos, New Mexic
Commandor E. B. Roth		J. C. Morris
Office of Chicago Directed Operations		Chief Librarian
U. S. Atomic Energy Commission		Oak Ridge National La
P. O. Box 4160 A		P. O. Box P
Chicago 80, Illinois	(1)	Oak Ridge, Tennessee
Mr. H. Etherington, Director		Miss Hazel Hutcheson,
Naval Reactor Division		Westinghouse Electric
Argonne National Laboratory		Atomic Power Division
P. O. Box 5207		P. O. Box 1468
Chicago 80, Illinois	(4)	Pittsburgh 30, Pennsy
Mr. Ben Pinkel		Mr. Joel V. Levy
National Advisory Committee		repartment of the Nat
for Aeronautics		Bureau of Ships
Levis Flight Propulsion Laboratory		Washington 25, D. C.
Municipal Airport		
Cleveland, Ohio	(2)	Mr. Spencer C. Stanfo Library
Reference Services Branch		Brookhaven National I
Technical Information Service		Upton, Long Island, M
Only Didra Courses		Ma W E Shallon
Ver ninge, fernessee	(1)	Nivestor of Reservab
Attals 1. A. Mainelt, chief .	(4)	Westinchouse Electric
Ma F D Kmonsfelder		Atomic Pover Division
Reheart & Wildow Company		P. O. Box 1/68
Respond & Davelanment Dent.		Pittsburgh 30. Pennst
Alliance Ohio	(2)	s's contraction to a second
urrance oneo	141	Library
Do Coorre & Haurina		Fairchild Engine & A
Disdia Induander		NEPA Division
Lefowette Indiana	(1)	P. O. Roy (15
rarayerve, indiana	121	Oak Ridge, Tennessee
Mr. Fred Gunther		
Jet Propulsion Laboratory		Dr. R. H. Lyon
California Institute of Technology		Technical Division
Pasadena, California	(1)	Oax Ridge National Le
Wa U C Table		Oak Ridge, Tennessee
Honfami Wanks		Mr. W. K. Words
D A Lay 550		7203 Building
Righland Haustanian	(1)	Janaral Flastina Con
reserved agains rom	8-2-8	Richiani, Saabinutar

Campbell ientific Laboratory ew Mexico (3) an ional Laboratory (3) nnessee tcheson, Librarian Electric Corporation Division , Pennsylvania (3) evy the Navy PB , D. C. (1) . Stanford tional Laboratory (3) sland, New York upp esearch Electric Corporation Division , Pennsylvania (1) ine & Airplane Corp. (3) onessee ision ional Laboratory nnessee 18 de Company

(1)

(1)

Mr. N. F. Lensing Cak Ridge National Laboratory P. O. Box P Oak Ridge, Tennessee Mr. Frederick T. Hobbs Technical Cooperation Branch Division of Research Washington, D. C. Mr. H. V. Lichtenberger Argonne National Laboratory P, 0. Box 5207 Chicago 80, Illinois (1) H. D. Young Information Division Argonae National Laboratory P. O. Box 5207 (2)Chicago 80, Illinois Mr. John R. Huffman Reacto: Engineering Division Argonne National Laboratory P. O. Box 5207 Chicago 80. Illinois (1)Mr. Frank Kreith Project Squid Princeton University Princeton, New Jersey (1)Dr. Martin Summerfield Project Squid Princeton University (1) Princeton, New Jersey Dr. Alfred G. Lundquist Code 429 Power Branch House 2714, Bldg, '-3 Office of Nevel Research Department of the Navy Washington 25, D. C. (1)