Determination of Martian Surface Reflectivity From 0.4 to 1.1 Micron Using a Vidicon Spectrometer

by

Douglas John Mink

Submitted in Partial Fulfillment of the Requirements for the Degree of Master of Science

at the Massachusetts Institute of Technology

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Certified by:____

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 Submitted to the Department of Earth and Planetary Science in partial fulfillment of the requirements for the degree of Master of Science on May 23, 1974

ABSTRACT

A new astronomical instrument, the vidicon spectrometer, is being developed at the M.I.T. Planetary Astronomy Laboratory. Based on the silicon diode vidicon system currently in use there. a low dispersion prism is added between the vidicon image tube and the telescope, allowing digital vidicon photographs to be taken of spectra. **These** spectra are stored on magnetic tape and computer processed to create intensity vs. wavelength curves for stars and The high spatial resolution of the vidicon image tube, planets. combined with a higher spectral resolution than photometer filters currently in use at M.I.T. give this instrument potential in the study of planetary surface composition from spectral reflectivity. Procedures for reducing the vidicon images to spectra have been tested on a set of spectra of tuo stars and the planet Mars. It is concluded that the vidicon response is not linear enough with variations in exposure time at low levels of incoming light for consistent star spectra, although it works well with Mars due to the planet's larger intensity where the vidicon tube has its The spectrometer slit is so narrow (one second poorest response. of arc for this data) that uavelength-dependent variations in refraction of light from a point source by the atmosphere cause star spectra of variable quality. Because of the low quality of the star spectra, direct spectral reflectivity measurements (which are obtained using Mars to star ratios) proved to be impossible. Although further tests of the spectral and intensity response of the silicon dinde vidicon should be carried out in the laboratory before good results can be guaranteed, the present Mars spectra may probably be used in conjunction with photometer-derived reflectivity data to expand coverage of the surface of Mars.

Thesis Advisor: Thomas B. McCord

Title: Associate Professor of Planetary Physics

Acknowledgements

Numerous people were involved in the design and construction of the vidicon spectrometer, although I have only met a few. Professor Thomas B. McCord, Mr. Jeffrey Bosel, and Ms. Carle Pieters have been informative about the design of the hardware, as well as the conditions under which they obtained the Mauna Kea data. Dr. Robert Huguenin lent his insight into the problems of Martian surface composition, as well as criticizing an early draft of this thesis. Steve Kent, who is also working with the spectrometer, provided criticism and astronomical knowledge. My spouse, Catherine, has greatly aided me financially and morally, while David McDonald greatly expedited the production of this thesis, with the aid of several computers.

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I. Introduction

Although Mariner 9 has returned a vast quantity of information about the planet Mars, little was learned about surface composition. From such experiments as the infrared spectrometer, particle size and silica composition were estimated, but these determinations had error bars so great as to be nearly useless in reaching conclusions about the composition of the surface materials of Mars. Until the Viking Lander in 1976, there is no way to physically look at a Martian rock with instruments.

Probably the most useful technique for remotely sensing surface composition is reflectance spectroscopy. Dollfus (1961), studying the polarization of light reflected by Mars, concluded that limonite, a hydrated iron oxide, was probably a major Hovis (1965) observed absorption bands in the nearconstituent. infrared reflectivity of limonite and suggested that they would be a diagnostic test for limonite on Mars. Sagan et al (1965) compared absorption pands they observed in laboratory specimens of limonite to Dollfus' Martian albedo curves and concluded that a surface with at least some limonite was not inconsistent with the Adams (1968) observed absorption bands between 0.5 and 2.5 data. microns in many iron-bearing minerals, the positions of which varied significantly from mineral to mineral. These bands are caused by electron transitions in iron ions and by vibrational bands in hydroxyl ions and water molecules. Adams suggests that

anometric albedo

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the absorption feature observed in Tull's (1966) geometric albedo curve is not inconsistent with a hydrated basalt composition. The feature observed at one micron in their spectra is not due to iron in iron oxides, but to iron ions in silicates. Adams and McCond 1967 (1969), using geometric albedoes obtained during the opposition discovered that curves for the bright areas had different shapes than those from the dark areas of the Martian surface. They concluded that the surface was composed of a combination of oxidized basalt and hydrated iron oxides. The bright and dark areas were modelled as being composed of the same material in different degrees of oxidation.

McCord and Westphal (1971, see also McCord, Elias, and Westphal, 1971) observed Mars during the 1969 opposition and noted that the iron ion absorptions were in different places, indicating compositional differences. Seven areas were observed, four dark and three bright, each being about five Martian longitudinal degrees in diameter. From this data, much compositional analysis has been done (see Figure 1 for examples of mineral reflectivities compared to Mars); however, from such small sample. а generalizations about the rest of the surface cannot be made. Despite over twenty additional spots obtained during the 1973 opposition, such interesting features as the Coprates canyon' and the Hellas basin remain uncovered; what is needed is whole disk at high spectral and spatial resolution. пен coverage technique, vidicon spectroscopy, has been developed to obtain the



Figure 1.

Comparison of Mars dark area reflectivity to reflectivity of sheet silicates. Note resemblance to the clay minerals, kaolinite and montmorillonite.

(Courtesy Dr. Robert Huguenin)

desired high resolution full-disk coverage. This thesis describes that technique.

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II. The Vidicon Spectrometer

The silicon diode array vidicon was originally developed for television and picturephone use, but because of its large dynamic range, high quantum efficiency, and linear response, it is being used by a growing number of astronomers as a digital replacement for photographic plates. The only advantage a photographic plate has over a vidicon is spatial resolution; however, that is not a limiting factor as atmospheric conditions are the resolutionlimiting factor in astronomy. McCord and Westphal (1972). Kunin (1972), and McCord and Bosel (1973) have reported on the development of a vidicon system for single-frame astronomical Planetary Astronomy photography at the Laboratory of the Massachusetts Institute of Technology (MITPAL). This system is based on an RCA silicon vidicon tube with a peak quantum efficiency of 85% at 0.5 microns, sloping off to about 6% at 1.1 microns (see Figure 2). Using filters this system has been developed as a two-dimensional imaging photometer, using filter sets similar to those used with photometers for spectral reflectivity work at MIT. As reported by McCord and Bosel, a vidicon spectrometer which would give the spatial resolution of the vidicon combined with a greater spectral resolution than such a vidicon photometer is under development.

The vidicon spectrometer is basically an optical system which is attached to the front end of the vidicon system on the telescope. Schematically it consists of a low-dispersion prism



Figure 2. Quantum efficiency of the RCA vidicon. This is the percentage of incoming photons which the diode array and affect it as opposed to being reflected or passing through without being absorbed. This graph was made by averaging the published curve over 250 angstrom segments.

through which light from a slit situated at the focus of the telescope is passed. The dispersed image of the slit is then refocused onto the surface of the vidicon diode array. In practice this is done through a system of mirrors (see Figure 3 for details) to avoid the infrared absorption of lenses.

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The vidicon tube consists of a 1024 by 1024 array of reverse biased diodes. A photon impinging on the vidicon target results in a decrease in charge in the diode it reaches. The image is read out by scanning the diode array with an electron beam which recharges the diodes as it hits them, producing a current proportional to the amount of charge lost. By knowing where the beam is at any given time, the intensity at each location in the diode array can be known. These intensity elements are then passed on to be recorded and displayed (for further details on the electronics of a silicon vidicon see Crowell and Labuda (1963)). The vidicon is read out as 250 rows of 256 image elements, each of which corresponds physically to four diodes. In such a lover resolution scan, less accurate positioning is required of the electron beam. No data is lost, and the vidicon's resolution is still better than the atmosphere allows. The intensity image is amplified, recorded on magnetic tape, and displayed on a slow scan TV monitor. This image is then available for further computer processing. The spectrometer system is diagrammed in Figure 4.

A portion of a vidicon spectrometer image is presented in Figure 5. The elements along the column correspond to spatial



Figure 3. Optics of the MITPAL vidicon spectrometer. The telescope is to the right.



Figure 4. The MITPAL vidicon system with the spectrometer attached.

elements along the slit. Wavelength is along the abscissa. The magnitude of each element is proportional to the current from the vidicon diode array at the time a corresponding diode was read by the scanning electron beam. The image is now ready to be turned into a spectrum.

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Figure 5. A portion of one vidicon spectrometer image of Mars. It runs from about 0.6 μ in the leftmost column to 0.8 μ in the rightmost.

PAPSE 1

III. Image Processing

The first processing that must be done to the image is to convert the column coordinate into wavelength. This is done through the use of a calibration function:

$$S = -S_0 + \frac{C}{(\lambda - \lambda_0)} \qquad \qquad \lambda = \lambda_0 + \frac{C}{(S + S_0)}$$

 S_0, λ_0 , and C being three constants determined from three column number-wavelength correspondences as follows:

$$C = \frac{(\lambda_1 - \lambda_2) (S_1 + S_0) (S_2 + S_0)}{(S_2 - S_1)}$$
$$S_0 = -S_1 + \frac{(\lambda_2 - \lambda_3) (S_2 - S_1) (S_3 - S_1)}{(\lambda_1 - \lambda_2) (S_3 - S_2) - (\lambda_2 - \lambda_3) (S_2 - S_1)}$$

$$\lambda_0 = \lambda_1 - \frac{C}{(S_1 + S_0)}$$

These correspondences are obtained by observing the spectrometer image of a calibration lamp with known sharp emission lines (as shown in Figure 6). From this calibration, which is redone periodically as data is taken, the wavelength-column relationship is known (see Figure 7 for an example). The resolution also varies as a function of wavelength, as would be expected (see Figure 8 for a sample dispersion function plotted from the first derivative of the calibration function).

Now enough is known to process a spectral image. A program



VIDCo	2 F.125								·
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10=	0.1541 C=-4	1.7388	50=-26	.2374		•			
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	2-0.3139	52	0.3517	- 102-	0.4130	152	0.5294	202	0.8357
	3 0.3145	53	0.3527	103	0.4146	153	0.5328	203	C.8471
	4 0.31-1	54	0.3536	104	0.4162	154	0.5362	204	0.8588
	5 0.3158	55	0.3546	_105 _	0.4179	155	_0.5398_	205	0.8708
	6 0.3164 ·	55	0.3555	106	0.4196	156	0.5433	206	C.8834
	7 C.3170	57	C.3565	107	. 0.4713	157	0.5470	207	0.8963
	8 0.3177	58	0.3575	108	0.4230	158	0.5507	208	0.9098
	9 C. 31 83	59	0.1585	109	0.4741	159	0.5546	209	0.9237
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	4 0.3216		0.3636		0.4338	164	-0.5747-	-214-	-1.0019-
1	5 0.32/3	65	0.3047	115	0.4357	165	0.5790	215	1.0194
	C 0.3229	66	0.3657	-116-	0.4376	-166-	0.5834	216	1.0378
î	7 0.3236	67	(+. 3668	117	0.4395	167	0.5878	217	1.0569
1	8 0.3243	68	0.3679	118	0.4415	168	0.5924	218	1.0768
1	9 0.3250	69	0.3690	119	0.4435	169	0.5970	219	1.0977
2	U U.3257	70	0.3701	120	0.4455	170	0.6018	220	1.1195
2	1 0.3264	71	0.3712	121	0.4476	171	0.6066	221	1.1424
4	2 0.3271	72 .	0.3124	122	0.4497	172	0.6116	222	1.1663
2	<u>3 C.3279</u>	73	_0.3735 _	. 123 .	0.4518	173	0.6167	223	1.1915
2	4 Q.3286	14	0.3747	124	0.4537	174	0.6219	224	1.2179
2	5 0.3293		0.3759	125	_0.4561_	175	0.6272	725	1.2458
2	6 0.3301	76	0.3770	126	0.4583	176	0.6326	220	1.2751
	7 0.3308	11	0.3782	-1/1-	0.4005	$\frac{1}{1}$	0.0381		1.3000
2	0 0.1310	·/8 70	0.2607	120	0.4620	170	0.6496	220	1.2722
	0 0 3321		0.3.10	120	0.4674	-180	0.6556	230	1.4100
2	1 0.3349	• 81	0.3832	131	0.4628	181.	0.6617	231	1.4489
	2 0.3346	82	0.3844	132	C.4722	182	0.6679	232	1.4904
ر . و	3 C.3354	83	0.3857	133	0.4746	183	0.6743	233	1.5346
3	4 0.3362	84	0.3870	134	0.4/71	184	0.6809	234	1.5818
·	5 J.3370	85	0.3483	135	0.4796	_ 185	0.6876	235	1.6324
3	6 0.3378	86	0.3396	136	0.4822	186	0.6945	236	1.6866
و	7 0.3346	87	0.3910	137	0.4848	_187_	_0.7016_	237	1.7451
3	8 0.3374	88	0.3723	138	0.4874	188	0.7089	238	1.8081
	9 0.3403	89	0.393/	139	0.4901	189	0.7164	239	1.8764
4	0 0.3411	90	0.3951	140	0.4928	190	0.7241	240	1.9505
	1 0.3419		0.3965	141			-0.7319	241	2.0313
4	2 0.3428	92	0.3001	142	0.4984	103	0.7494	242	2.2168
	6 (), 1 446	94	0.4008	144	0.5042	-194	0.7570	244	2.3240
2	5 0,3454	95	0.4022	145	0.5071	195	0.7658	245	2.4430
	6 0.3463	95	0.4(137	146	0.5101	196	0.7749	246	2.5758
4	7 0.3471	91	0.4052	147	0.5132	197	0.7843	247	2.7250
. 4	8 6.3480	98	0.4(57	148	0.5163	198	0.7939	248	2.8937
4	9 0.3490	99	0.4083	149	_0.5195	149	0.8039	249	3.0862
5	0.3499	.100	0.4098	156	0.5227	200	0.8142	250	3.3077

1

Figure 7. Wavelength as a function of vidicon column for a typical calibration function. The three column (Sn)-wavelength (Ln) pairs used to determine the function are given at the top. Column number is at the left, wavelength at right.



has been written which runs as a subroutine under the Planetary Astronomy Laboratory's image processing system (DIPSYS) which has been set up to provide a metastructure under which vidicon images A simplified diagram of this program may be easily processed. appears in Figure 9. The spectral image is read off the run tape by DIPSYS and stored on a disk where it is available to the spectral processing routine, which has three basic tasks. The first and easiest is to punch out the intensities along one row of the image onto computer cards for input into a plotting routine (this was how Figure 6 was produced). Second, it can subtract the average background from the image, column by column, where the rows over which the background is to be averaged are read from the input instruction cards. Last, and most important, the program can produce a new image in which all of the elements have the same For spectral reflectivity work, where the spectral resolution. range of interest is 0.4 to 1.2 microns, a resolution of 250 angstroms, the best resolution at 1.2 microns, was chosen. Figures 10 and 11 show the effects of this processing on an image of the standard star Xi 2 Ceti. Portions of these images are then Due to atmospheric and integrated spatially along the slit. telescope optical effects, a star image is not a point; it is smeared out spatialy into to a Gaussian distribution of intensity which is at its maximum where the point source would be. To use the full energy output of the star at a given wavlength, the image must be integrated across all rows where the image intensity is



20111	5																								
	151	152	153	154	155	155	157	158	159	160	101	162	163	104	155	165	167	16.4	151	113	1/1	112	173	174	17 .
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1-	440	437	4 36.	437	441	442	443	435	446	45)	420	448	454	447	443	449	440	421)	144	\$ \$ 2	4 1	4.41	- 14	7
2-	204	204	112	213	1.19	175	195	191	196	193	191	175	194	1.30	192	1.14	129	271	111	175	171	173	1.42	777	1 12
3-	155	160	150	155	16.1	157	157	147	167	167	163	163	160	153	162	166	161	1.50	159	150	157	104	1	110	1-2
	110	144	1.54	147	149	1.2	1 1 1 2	149	151	152	144	144	- 145		145	145	165			165	- 12:00	141	145	· • • • • • • • • • • • • • • • • • • •	
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6-	132	135	141	140	157	141	1.44	133	1 34	141	150	137	141	131	143	144	144	- D 2	141	144	1.1	1.24	141	1 1 2	14.
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9-	135	1 4 1	_1 \$1	135	134	131	132	129	137	141	143	137	139	135	141	137	131	133	139	144	141	125	1:7	13/	ا فق ا
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12-	133	132	121	137	141	142	135	132	143	136	135	129	135	130	127	136	125	127	136	135	172	124	132	135	135
13-	135	134	127	131	139	135	142	137	141	146	144	147	137	1 3 9	134	143	137	135	140	141	130	145	1 47	137	130
14-	132	133	141	137	135	134	127	127	127	129	1 17	137	133	132	124	124	129	124	125	134	141	124	1 54	1.25	125 3
15-	127	125	1.11	138	1.17	145	1 3 2	1.12	127	127	1 44	134	1.0	129	141	131	1.30-	1	1.2.4	131	1 4 7	1.5	1	1.57	1.57
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13-	429	131	130	123	171	133	121	1.29	1 20	130	1.57	110	170	121	1.1.7	132	120	1.75	121	132	1 1 4	124	120	4 71	44' 7 1 1 1 1
17-	139	140	1.57	135		125	135	1.55	_131	137	139	<u> </u>	133		12/		_14 2 .	- <u>144</u>	131	137	_ 4 %	125-	1,24	. 132 -	. 1.21
20-	133	137	138	139	127	153	122	136	134	137	137	130	175	127	135	131	130	135	127	155	143	14.3	123	147	127
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27-	139	135	138	138	131	140	135	139	129	138	146	129	134	132	129	129	140	135	111	132	1 + 1	1 33	134	14.	131
23-	134	135	1.51	131	141	1 37	124	131	133	132	142	139	128	130	132	142	133	1+7	130	141	13.	123	179	136	135 1
64-	131	129	1 34	143	131	127	1 34	136	-134	125	130	125	174	133	130	111	- 1 411	11	117	1,0	1 11	111	110	1.1	144
25-	131	135	130	137	133	135	128	141	127	129	137	1 3 1	133	1 34	126	126	1 14	125	111	1.1	11.	1.17	144	140	134
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20-	162	151	1 44	143	144	167	1 42	144	134	126	151	144	176	141	141	127	1.11	121	1.70	1/4	1	1 1	1.		1 2 2
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30-	205	761	7.7	112	220	705	407			401	-217	201		1214	200	1.7.5	200	193	192	1 - 1	1 - 3	1.2.1	1 4	11+	1.55
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32- 4	920	1 76 1	1919	1.141	1121	1933	1135	14.20	1717	1923	1430	1904	1844	1362	1072	1 31 2	1017	1731	1747	1122	1717	177	17.1	1	1635
11- 2		2454	2848	2.51	2333	2483	2436	2912	7712	2485	2912	2391	2.42.9	2418	2116	2735	2695	2002	2452	2-15	, ² ', •, •	1 44		Z529 :	1444
34- 1	845	1853	1549	1.57	1972	1375	1871	1465	1372	1264	1647	1315	1795	1 790	1112	1749	1103	128.4	1041	1,14	10.13	1	1574	1941 .	1522.
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56-	413	411	416	470	473	4 5 3	4 16	411	444	443	450	442	440	455	450	401	40.5	401	4:2'.	463	473	403	434	482	469
>7-	241	242	244	259	21.4	260	748	255	250	252	208	_206	253	260	269	257	203	271	259	157	271	257	254	253	270 1
38-	178	183	111	183	179	163	177	178	172	179	178	180	175	186	179	173	177	172	105	177	172	173	175	177	1:0 1
37-	147	154	143	143	1 11	135	135	141	135	142	141	145	177	136	139	133	136	139	131	131	1 + >	131	1 . 1	14 -	144
40-	143	131	135	137	137	1 54	1 35	127	127	140	1 45	132	135	1 37	135	125	1 30	131	121	131	111	142	1	121	133
41-	136	135	127	129	129	127	120	125	127	1 15	1 14	127	127	144	1.13	127	134	125	127	1.1	1.1	1.7	1 . :	1	111
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43-	127	15	116	13-	1.45	1 4 1	1 1 2	136	143	114	144	126	1 3 4	1 3 8	1 3 .	127	127	122	110	133	174	123	167	121	127
	111	134	1.25	100	1.31	126	134	129	121	130	1.47	129	-126		127	127	1.20	- 121	114		125	129	143	135	147
44-	122	121	127	121	1 204	120	1 3 4	147	1.14	121	131	110	120	1.71	161	121	127	1 31	127	151	1.4.5	123	1 10	1.57	139
	124	1 2 1	121	171	1.1	1.30	128	143	120	170	1.37	. 1.12	12/	1.35	1.54	119-	121	127	175	124	13.5	124	1	1-1-3	13.2
-0-	132	4.54	1.54	132	1.55	125	131	136	129	136	131	123	122	1.32	127	155	134	135	140	133	14+	133	141	100	135
	757	144.	163	137	131	12/	110	_137	131	125	129	133	122	131	13/	131	121	1 34	125	137	144	113	130	131	147
48-	134	132	125	124	137	12)	131	131	127	133	145	127	132	136	135	174	138	133	130	131	144	133	1.15	1 15	141.
49-	134	132	127	135	ا د ا	134	1 15	1 37	133	137	1 30	1 \$ 1	126	.141	132	117	ì 31	125	123	133	140	157	1.7	134	137
>0-	133	137	143	131	135	127	124	1 36	137	127	124	1 54	126	136	137	126	134	130	1 15	132	145	144	1.45	149	114
.>1	148	141	1 1 1	. 127	. 137	125	_121	_140	. 122	139	136	127	130	1 34	129	124	129	121	129	144	127	135	1 *	144	137 1
-20	121	125	127	135	129	135	132	137	132	125	127	137	127	1.57	134	135	127	1.15	129	127	1 4 3	144	144	1.2	145 1
53-	143	134	1.15	131	133	137	1 32	141	133	131	140	174	125	136	145	136	134	143	134	147	140	1	1 - 1	35.	15.1
24-	144	134	1.52	144	1.57	127	137	141	128	137	1 34	120	132	141	140	129	141	1 17	149	141	14	1 4	140	1	المتر المراجع
55-	122	135	136	119	121	12/	134	1 36	135	125	1 1 2 2	1 14	121	12-5-	130	13.5	1 1 0	14.1	14	147	141	1635	140	1	121
54-	137	135	118	127	133	126	132	137	135	145	1 4 4	117	134	140	144	1 3 6	1.1	110	- 1 6 2	1	1.1	14.5	1 4 4	1.71	1
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Figure 10. A portion of the vidicon image of the spectrum of ξ^2 Ceti from about 0.5 μ to over 0.6 μ . Note spatial spreading of image (vertically). The star is centered in row 32,

5X267116

_	1	· /	1	4	5	6	. 7	8	9	14	11	12	13	24	
	175 5		4 x1 7 2m	526.7-4	13 67*	6(6 11)	L/7 666								•
2-	65.676	1. 471		172.448	35.341	0 44	917+944	134 32	2 44 (40	21 . 36	494.331	491.379	487.122	467.32	
3-	17.2.3	28.044	4 7 /4	43.077	22.421	25.562	28.07	27 482	41.760	37.149	43.433	F9.52C	73.4.	\$1.54	
	7.6	1.125	14.848	17.724	16.121	12.199	11.460	9.547	17.813	16.197	4 + 237	33,103	17.599	11.47	
- 4 -	7.749	9.861	10 . 345	6.11	7.6.6	6.007	4.117	5.582	11.411		10.004	2.092		5.05	
**-	14.097	8	:1.1 3	2.4%7	2.617	3.534	3.976	2.441	4.399	6.737	4.971	. 0.000	A 604	2.C2 4.c2	
·I-	0.202	1.212	3.277	•8.1	1.463	3.595	4.255	1.320	11.514	1 .324	4.04.	6.246	A 040	·	
<u>Pe</u> .	- 4.413.	4.384	3.924	5.6 \8	. 2.576	2.363	1.295	4.269	0.372	C. 961	1.899	0.148	6.000	0.00	
- 9-	3.117	8.647	3.873	1.2.4	1.558	2.232	1.478	2.112	1.363	2.114	1.844	0.078	(. 733	0.00	
1 -	6.197	2.323	0.25	1.45	2.627	1.040	1.2.2	4.967		7.242	C . (00	0.000		0.00	
11-	2.943	3,834	1.646	1.2.8	3.669	C.579	0.560	0.000	6.196	0.274	C.C.C	9.000	0.010	0.00	:
			1.694	ú.215	0.712	0.634	2.023	1.166	4.399		C.240	0.000	0.000	0.00	1
14-	3.457	4.583	0.257	1.2 /	0.000	0.000	2.10/	2.451	5.277	6.147	1.349	0.0.0	2.011	C	1
15-	3./3P		C+117	V. 967	4.204	6.337	3.425	2.332	1.446	Ç.153	C.Cuu	6.000	0.000	6.00	1
14-	3,891	6.375	A 000	0.347	0.373	0.496	0.009	0.000	0.020	0.260	0.000	. 0.000	C.000	0.00	Ţ
17-	1.393	3.201	310-0	0.212	1.347	+040	1.644	1.597	1.373	1.180	5.100	•	0.030	3-05	
10-	1.206	6.266	6.404	0.015	0.006	0.000	1.747	0.000	0+134	0.024	C+C>>	6.000	0.030	30	
17-	5.103	2.435	150.0	0.812	0.928	1. 336	0.514	9.167	0.707		C •000	0.000	C.100		-
27-	0.767	fote"	C.000	1.728	1.088	1.756	1.240	0.734		1.734	7.175	0.0.0	0.002	0.03	
21-	1.936	2.615	C.0.0	1.3.4	9.673	C-165	0.735	G.184	1.035	0 177	0.513	0.000	0.003	0.02	1
22-	4.512	5.172	C. 100	0.355	0.019	0.103	0.618	0.734	3.515	2.112	0 920	0.000	0.033		1
23-	1.276	1.764	6.363	1.115	2.309	2.591	1.809	1.432	1,952	2.903	6.000	0.000	0.033	U	1
74-	1.4 7	:. 71	· • • • • •	n.4 it	6.579	0.165	0.009	0.418	1.033	0.236	0.000	0.000	0.000	0.00	1
25-	5.144	5. 341	Q•013	6.340	0.229	0-136	1.005	0.184	0.320	0.000	0.000	0.000	1.000	0.00	4
25-	1.223	•611	0.011	2.255	0.083	C.675	0.000	1.799	0.413	9.024	2.000	3.000	0.000	0.1.2	
29-	1.707	1.906	0.023	2.277	G.725	(.324	6.569	1 74	1.776	7.524	2.202	0.000	0.0.0	0.00	
23-	1.028	2.132	0.057	P. 111	(.782	0.3(3	C.009	0.991	3.179	1.333	0.5/3	0.000	0.050	3.30	1
2,5-	0.112	0.335	C.CCC	5.380	4.517	0.169	15.500	13.119	9.283	4.876	1.509	2.000	0.000	0.06	1
41-	4.9/1	13 483	11.114	47.48h	78+571	95.958	108.160	126.612	120.445	111./97	03.749	62.049	40.105	32.224	
17-	5.844	11.605	4 14 4 19 1 4 14 4 19 1	11,4647	351.710	726.357	815.275	935.781	929.018	651.12r	154.748	667.237	555.141	47: . 251	1
-14	5.991	.3.675		2.28 2.2	1 15 470	7393.904	263 .129	2925.239	2464.294	2841.002	5921.5	2446.450	27 797	1975. 23	
14-	5.834	-1-007	41.4 . 4.51	1125.450	1111.617	3060+341	4170+177	4400.492	4557.047	4464.301	4364.4"	3004.543	3418.51:	3-49.478.	
51-	4.814	72. 151	146.0634	404.613	631.461	787.394	9 2 716	2010.000	2674.633	2717.977	2406.230	53.0.044	2- 30-046	1844.412	4
:6-	2.429	12.817	110.321	271.254	268.051	357.481	343.263	442. 41	4 4 4 7 7	614 346	1115.98	1 54.440	1113.769	975. 53	1
37-	1. /08	5.151	41.445	#4.105	118.50%	141.133	153.197	175.017	147.233	201 406	335.829	5, 1.643	5.12.441	-77 92	1
33-	2.925	3.444	7.101	19.315	37.361	5 . 772	56.531	56-452	64.004	67.668	5 047	204.000	100.004	11.581-	
31-	1.256	1.953	0.+95	5.634	16.757	11.751	16.162	15.570	2.809	2.289	6.010	3.0.1	0.055	0.000	ł
4r)-	1.764	4.49.	1.006	0.397	1.107	1.972	0.007	141	C.960	6.201	B. Grid	Date f	0.066	0.000	1
41-	1.897	1.001	0.037	0.257	0.426	U.214	0-147	0.312	0.000	0.614	0.000	0.000	0.0)7	0.00	1
			. 0.4 4	7.536	1.340	0.00J	0.018	0.000	C.00D	0.000	0.0.0	0.01	000	0.350	
44-	2 1 2 2	2 24	1.525	0.022	0.630	0.344	0.156	0.000	2.013	1.133	C.000	0.303	0.363	2.311	-
41-	0.550	د درج د حد		1.50%	0.0.4	0.765	0.007	0-129	0.000	0.389	0.000	0.000	0.000	0. 3.76	
46-	Leph!	1.945	1.057	0.419	8 •445	1.019	1.721	0.514	C. 104	<.001	0.000	5.228	110.5	0.066	
47-	2.993	1.33	0.558	0.000	1.117	0.124	P•376.	0.035	0.000	C.CO0	1.349	0.028	0.033	1-106	
4	1.808	0.77 1	1.6.29	9.447	1 024	U•089 ∆≝•	0.107 A	2.451	0.000	0.024	0.825	1. 32	2.317	1.122	
43-	1.365	4.115	4.551		0.012	A. 466	0 000		1.027	<u>1.424</u>	715			0.1.2	
50-	3.575	3. 374	1.172	6.610	0.967	0.661	1 074	1 462	1.237	0.000	1.927	0.000	0.396	4.093	i
51-	1.695	0.515	1.962	2.084	1.694	9.300	0.541	2.104	0+000	0.024	1.101	0.579	3.401	14.413	Ĭ
57-	2.512	1.33"	7.752	4.305	6.223	1.474	0.020	a . 304	C 6 391	0.000	~ 000		0.063	1. 20	1
53-	4.267	5.975	3.601	1.648	0.006	U.75×	0-020	0.707	1 020	1 404	1.073	1.514	7.693	7.132	1
54-		1.205	6.235	_ 3.517	4.496	1.688	0.716	2.671	1.477	1.144	3.174	13.072	12.046	[•13]	1
55-	2.538	2.162	1.399	5.018	1.787	-124	0.000	746.0		1.581					1
56-	1.797	1.672	3.212	1.845	0.343	0.716	0.018	0.679	0.000	2.936	5,306	13.022	9.014	1+15. 771	1
	10	Norma 1	4 4								20200	1 11027			1

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Figure 11. A portion of the processed image of ξ^2 Ceti from 0.35 μ to 0.70 μ . Each image element represents intensity per ten angstroms averaged over a 250 angstrom resolution element. The background was first subtracted out of the vidicon image. 13.027

above the background. After this integration, the spectrum vector is punched out onto cards for plotting and further processing. A more advanced version of this processor will incorporate the plotting, ratioing, and other functions into one DIPSYS subsystem, where only disk files will be used.

The final procedure needed for good spectral reflectivity data of the surface of a planet is to know from what part of the surface the spectrum originates. A photograph is taken through the eyepiece, looking at the slit in a mirror tilted 45 degrees to the optical axis of the telescope (the first surface in Figure 3). A similar logging arrangement is used for photometer data. A plotting program has been written to create Calcomp plots of the coordinate grid of Mars (or any other planet) projected onto a disk using the physical ephemeris of the planet from The American Ephemeris and Nautical Almanac and the time of observation in Universal Time. Figure 12 is a block diagram of the program. while Figure 13 is a typical, although smaller than normal, output. To position the spectrometer slit on the disk of the planet, the negative of the photograph of the telescope image is projected onto the grid, and the slit marked by hand. At this point the original vidicon images have been reduced to constant resolution spectra of stars and known positions on Mars; and reduction to spectral reflectivity data, as well as testing, can begin.

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Figure 13. A typical grid plot produced by the program in Figure 12, the third produced for vidicon spectrometer Mars run C.

IV. Analysis of Data

The first major attempt to use the vidicon spectrometer to take spectra for reflectivity work occurred during the opposition of Mars during October, 1973. On two consecutive nights the Mauna Kea eighty-inch reflector was trained on the planet Mars, and about 75 spectra were taken, as well as an equal number of spectra of the standard stars Alpha Lyra and Xi 2 Ceti. Xi Z Ceti µas chosen because it was near Mars in the sky, while Alpha Lyra has a spectrum which is well known and is used to calculate planet/sun ratios to get reflectivity. Figure 14 demonstrates the reduction methods used to get spectral reflectivities from raw intensity To avoid airmass reductions, spectra of Alpha Lura and spectra. Xi 2 Ceti were taken when the two stars were at the same airmass. Since star/star ratios exhibit little variation with low 1.38. airmass changes, the ratio of the two stars obtained from these spectra can also be used to reduce reflectivities at other Before any data was reduced to reflectivities, airmasses. extensive testing was done to see whether the data would be usable. This portion of the thesis will describe that work, using the best results obtained to date.

Figure 15 shows a high resolution spectrum of Alpha Lyra which has been averaged over 250 angstrom segments to simulate the spectrometer output. Figure 16 is an Alpha Lyra spectrum from the vidicon spectrometer from which the vidicon response has been



Figure 14. Production of spectral reflectivity from raw spectra. Air mass correction not not needed if objects to be ratioed are at the same air mass.

350.25 280.20 *10 ² 210.15 TO.05 INTENSI 8 0.40 0.CO 0.80 WAVELENGTH IN MICRONS 1.00 1.20

Figure 15. Spectrum of αLyra, averaged over 250 angstrom resolution elements, from a 50 angstrom resolution spectrum provided by Steve Kent.



Figure 16. aLyra spectrum from vidicon spectrometer with vidicon response (Figure 2) divided out.

Note that the peak is shifted to a slightly longer removed. wavelength and that the shape is generally broader to about 0.7 To test the repeatability of the data, pairs of spectra microns. of the same star were ratioed to each other. Results of one such pair are shown in Figure 17 (all ratios plotted are normalized to 1.0 at 0.575 microns). Figure 17a is the ratio of two Alpha Lyra spectra with similar airmasses (1.40/1.38), but different exposure times (5sec/1sec). If the response of the system were perfectly linear, that is, if intensity recorded from a given source is a linear function of the integration (exposure) time, the curve would be flat. It is obvious that it is not; however, the relatively flat region corresponds with the peak intensities of the spectra, so it may be that low level signals are nonlinear representations of the intensity received from the star. To test this idea, a 'pedestal' was set up under the spectrum. AII intensities below a certain value would be ignored, and possibly. the nonlinear features of the curve would go away. Figures 17b and 17c show the results of installing pedestals of 300 and 400. respectively (the maximum intensity registerable is 4095). а pedestal of 300 seems to help from 0.5 to 0.8 microns, but a larger pedestal doesn't help at all. Figure 18 shows a similar ratio for two Xi 2 Ceti spectra with slightly different airmasses (1.67/1.32) and different exposure times (20sec/15sec). Once again the curve is relatively flat over the peak in incoming energy, this time from almost 0.4 to 0.8 microns. (Figure 19 is a



Figure 17a. Ratio of two α Lyra spectra, all elements above background included.







Figure 17c. Ratio of same two α Lyra spectra, this time including no elements less than 400.



Figure 18. Ratio of two ξ^2 Ceti spectra, including all image elements above background.

typical Xi 2 Ceti spectrum). this time, however, there is a smooth upturn which has some undetermined significance. Thus, star ratios seem to be usable, at best, from 0.4 to 0.8 microns.

Now that there is some idea as to the reliability range of the spectrometer, indefinite though it may be, the Mars spectra can be observed. Figure 20 is a typical Mars spectrum, summed over five vidicon elements down the slit. Note that the peak is in the red, rather than the blue like the two stars' spectra. This is because the stars are both of spectral type A0, while the sun, which is providing the light which is reflected from Mars is a cooler, redder type G. Figure 21 shows a saturated spectrum of Mars. The peak intensity of 4095 is surpassed from 0.5 to 1.0 microns, although around 1.1 microns, the signal is unsaturated. Originally it was thought that the unsaturated portions of a saturated spectrum could be used to extend the range of an unsaturated spectrum which had a very low signal beyond 1.1 The data show, unluckily, that there is little or no microns. overlap between the good signal from one and the good signal from the other type of spectrum. Once again, an attempt was made to do away with low, nonlinear signals with a pedestal. Figures 22a, b, and c show the progressive changes as pedestals of 300 and 400 are subtracted from the original spectrum. Ratios of Mars images seem to be more consistent than those of star images. Figures 23a, b, and c and 24a, b, and c are the results of ratioing different 'images of Mars to each other. The three images used



Figure 19. A typical ξ^2 Ceti spectrum. Note that the peak is at a longer wavelength and the shape is broader than α Lyra.









Figure 21. An overexposed spectrum of Mars. Arrows indicate intensities reading greater than 4095 in at least one element of the image which went into the resolution element.



Figure 22a. Mars spectrum







Figure 22c. The same Mars spectrum with a pedestal of 400.

were taken within 15 minutes of each other. The same portion of the image was used in each case. Each is a one minute exposure. Note the flat curve from 0.5 to 1.1 micron, indicating better repeatability than for the stars, possibly due to more signal above a nonlinear level. As the pedestal is increased, some of the apparently good data is lost, but the noise is gone by the time a pedestal of 400 is used (c). The Mars spectra are probably recoverable.

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Figure 23b. Ratio of two Mars spectra, each of which has a pedestal of 300.









Figure 24a. Ratio of two Mars spectra, without pedestals.



Figure 24b. Ratio of two Mars spectra, each of which has a pedestal of 300.



Figure 24c. Ratio of same two Mars spectra, each of which now has a pedestal of 400.

V. Recommendations for Future Use of the Vidicon Spectrometer

Although it appears that it will be impossible to do spectral reflectivity work using the vidicon spectrometer due to an inability to meaningfully ratio stars and planets over a useful range, the instrument has advantages which will make it worthwhile to develop it. The combination of good spectral resolution (250 angstroms or better, compared to 300 angstroms for a filter photometer), with complete spectral coverage and high spatial resolution indicate much promise. It appears that the limiting factor will be the response function of the vidicon tube, with its nonlinearities in wavelength and intensity. Once more lab work is done to quantify knowledge about this problem, the instrument will be ready to gather more data. Another problem which may affect the star spectra is the problem of differential diffraction of the star's light by the earth's atmosphere. Different wavelengths, diffracted at slightly different angles would show up at different positions in the smeared out star spectrum, and if the slit is smaller than the apparent diameter of the star, part of the star's spectrum would be lost, in a wavelength-preferential manner. The solution is to widen the slit; although the spectral resolution at the vidicon would be reduced, the spectrum would be much more reliable. But what about the Mars data from Mauna Kea? With the high spatial resolution and apparent good response of the vidicon. something should be recoverable. The planet in the slit occupies

up to 35 elements in a vidicon column when it is about 23 arc seconds in diameter, and the slit is two elements wide, so, with good seeing of 1.5 seconds or less, there are fifteen spectra per spectrometer image. Luckily, the slit passes over some photometer spots that were taken within days of the vidicon spectrometer run. allowing relative reflectivities to be obtained, basically extending the photometer data for more complete surface coverage. For example, Figure 25 shows the position of the slit on the planet's disk during one run. This one slit passes through the Coprates canyon as well as a large dust storm to the southwest of Using a photometer spot as a standard and modifying Coprates. resolution to match the photometer, some interesting data should be forthcoming.

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Figure 25. Position of one set of spectra across the disk of Mars. Latitude and longitude of the sub-earth point, S, given.

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