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DTU Wind Energy Department of Wind Energy

Track: Wind Power

Project:

OffWindChina - Research and development of optimal

wind turbine rotors under offshore wind conditions in China

Integrated Individual Flap and Pitch Control for active load alleviation

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Abstract: This work present the development and application of an high-fidelity linear model of the aero-servo-elastic response of a smart rotor wind turbine with Individual Flap Control (IFC). The linear model allows for a rapid evaluation of the smart rotor response, and is here applied to design and tune the flap control system. The performance of the Individual Flap Control is then evaluated with non-linear time marching simulations; the non-linear simulations confirm the load alleviation potential of the IFC smart rotor configuration

Keywords: Smart rotor; active load control; flaps; linear aero-servo-elastic modelling

Introduction and motivation:

Both on-shore and off-shore wind turbines experience largely fluctuating forces, caused for instance by: atmospheric turbulence, wind shear, tower shadow effect, interaction with wakes from others turbines.

Such forces often results in fatigue and ultimate loads that drive the design of the wind turbine components, and hence its



Application and results:

DTU 10 MW Reference wind turbine [6]: pitch control for power limitation, flaps on 30 % of the blade span, Individual Flap Control above rated wind speed, with two configurations: LP freq. of 1.4 Hz and 8.0 Hz.



cost.

Smart Rotors, with a combination of sensors, control units, and actuators, are able to actively alleviate the loads the turbine experiences, thus lowering the design loads, and possibly leading to a reduction of the Cost of Energy [1,2,3].

Method:

1) HAWCStab2 with Flaps: high-fidelity linear modelling tool [4,5].

2) Linear aero-servo-elastic model of a smart rotor: Modelling: Wind turbine + Pitch and Generator power regulation + Flap Actuator.



3) Individual Flap Control (IFC): PI control on Low Pass filtered measurements of root bending moments.

4) Closed-Loop aero-servo-elastic system: System characteristics (poles) depend on IFC tuning.

The smart rotor performance is evaluated with non-linear HAWC2 aeroelastic simulations, following the IEC standard [7] prescriptions for a class A turbine.



A reduction of lifetime fatigue Damage Equivalent Loads on blade root flapwise moment (ctrl. objective) is reported, together with a slight increase in blade and tower torsion DEL, and a slight decrease of the ultimate loads on the tower, and main bearing (extreme loads in DLC 1.x [7]).





Conclusion and future work:

An high-fidelity linear modelling tool for smart rotor with flaps was developed. The tool can be used for rapid evaluation of the smart rotor design, as well as

5) Ziegler-Nichols tuning of IFC control parameters:



for design and tuning of its control system.

- An individual flap control configuration was proposed and tuned with the linear model. Non-linear time marching simulations confirmed the load alleviation potential of the smart rotor with IFC; better performances are reported for the IFC with higher corner frequency of the LP filter (8 Hz)
- To achieve an higher alleviation potential, future application of the linear tool might include: integration of individual flap and pitch control, design of model based control algorithm, integration of smart rotor configurations in concurrent design and optimization methods.

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